MSc in Industry 4.0
IND5003 Data Analysis for Sense Making
Lecturer: Prof. Vik Gopal

Group 5 Project :
Airplane Engine Vibration

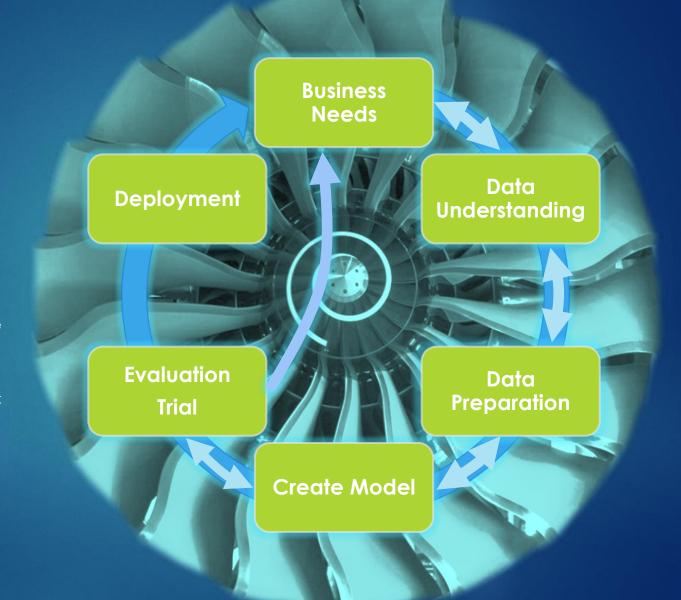
#### **Group member:**

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# **Methodology Overview**

- Business Needs create a common business understanding of current situation and project objectives.
- 2. Data Understanding took the data for further exploration, and by collaborating with the subject matter experts and understanding the build process.
- 3. Data Preparation Data was cleaned and formatted, several hundreds of features were extracted and final data sets for the analysis were selected.
- **4. Create Models** after initial models, the team iterated between the third and fourth step to further refine the models that most robust for the process.
- **5. Evaluation and Trial** against the most robust model for the build environment.
- 6. Deployment Proposal visualizations and layouts that help the stakeholders understand how the results can be operationalized in practice



#### **Business Needs**

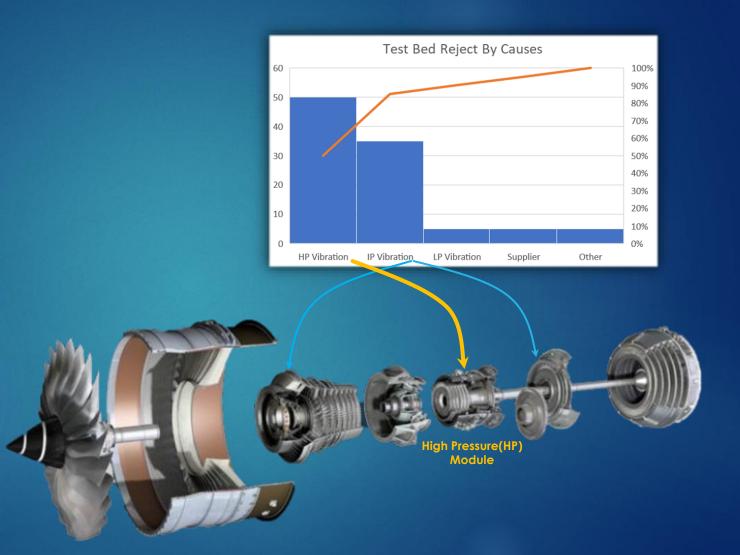
**Problems: Poor Production Yield** 

#### 1. Rejects at Test Bed:

- Test bed pass rate is about 80%.
- On each engine been rejected, the rework lead time is 20 days and direct cost more than 100k.
- The engine is build to order, any reject will delay the delivery to airplane factory and cause the airplane delivery delay to airline.
- The top failure modes are HP and IP vibration.
- By improving the quality of the HP and IP modules build by 50%, the overall test bed pass rate can be improved from 80 to 91%.
- Estimated 200 engines produced a year in a plant, that is 22 reworks can be avoided.

#### 2. Rejects at Build:

 High Pressure(HP) Module having highest rework for build, pass rate is 70%. Due to the flatness and concentricity measurement out of drawing specification.

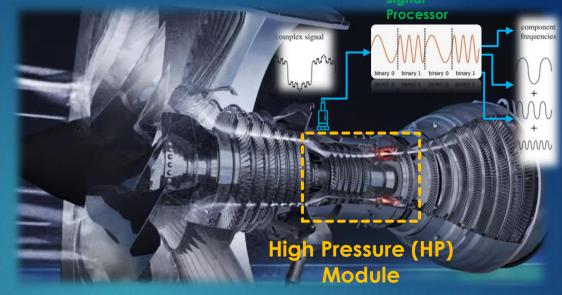


#### **Business Needs**

#### **Objective for This Project**

**Using Machine Learning to:** 

- Improve the build quality of HP module to increase the pass rate at Test.
- Explore opportunities to improve the yield for the HP module build.



Compressor



https://www.youtube.com/watch?v=JxkJ-FwFeVI (How jet engine works)

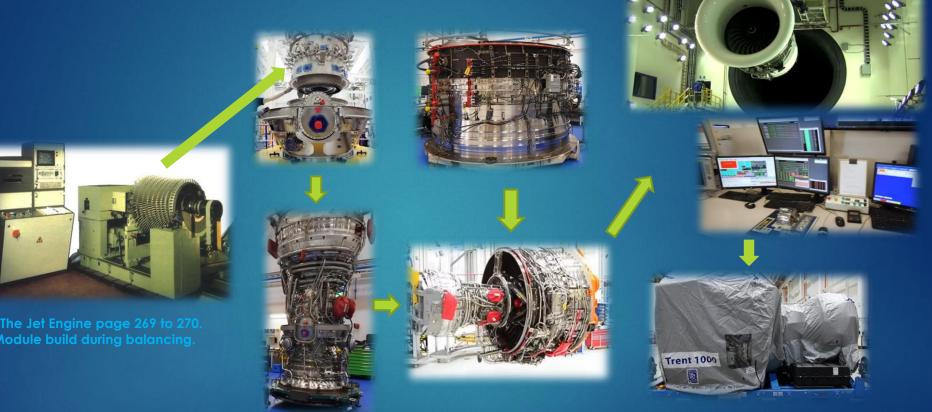
https://www.youtube.com/watch?v=m0X0o8aBrxk (3D modelling)

**Turbine** 

# **Data Understanding**

Engine build process:

Engine Test Where The Vibration signals to be recorded.



https://www.youtube.com/watch?v=K2R6NTgvEV4 (How jet engine been assembled and test)

# **Data Understanding**

#### Deep dive to the HPS build:

- The HPS is the power generation source of the engine. The HPS rotate at 12,000 rpm at the power of 60 F1 cars for a large commercial wide body airplane engine.
- Vibration will damage the component and caused early fatigue.
- Vibration caused by the off axis and unbalance of the rotating components.
- Measurements are carried out along the staking of the components to ensure the off-axis within a specification.
- The measurements data can be pulled out from the MES.
- The vibration data is recorded at the test bed for the complex performance test curve of the engine. For the evaluation of this project, the maximum vibration is used.
- 70% of the total rejected at HP module due to C1 measurement out of specification.

	Flatness Measurement						Balancing Measurement											
	Label	Label	A1	A2	B1	B2	C1	C3	D1	D3	Ax	Ay	Bx	Ву	Ex	Еу	Fx	Fy
Index	Pass/Fail	Vibration	um	um	um	um	um	Deg	um	Deg	um	Deg	υm	Deg	um	Deg	um	Deg
1	Pass	0.23	5	25	12	20	42	154	10	100	39	227	85	63	38	312	60	221
2	Reject	0.11	25	21	16	6	76	156	8	121	70	98	109	73	51	131	66	262

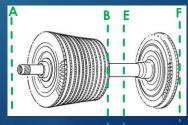
Components will never be manufactured to perfect flat and balanced. Multiple of components will be stacking together during the assembly. Hence, the component will be measure at stages to ensure best matching to reduce the axis offset.

NTERMEDIATI

L.P. COMPRESSOR(Fan) CASE

H.P. SYSTEM





See The Jet Engine page 269 for balancing couple planes.

I.P. AND L.P. TURBINE

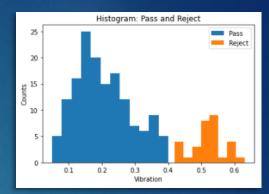
# **Data Understanding**

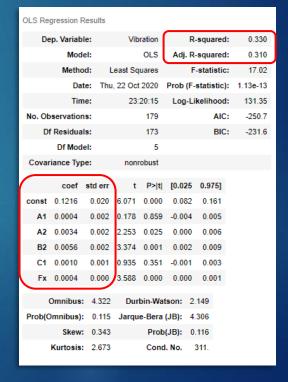
#### EDA:

- There are total 180 engine data set, 149 Pass, 31 Reject.
- The Pass and Reject kind of 2 distinctly near to normal distribution groups.
- As from the subject expert observation, the measurements data not having strong relationship to the Vibration reading. Which as demonstrated in the scatter matrix plot.
- Performed regression analysis and drop influencer point not helping.
   The residual plot too

	Vibration											
	count	mean	std	min	25%	50%	75%	max				
PassFail												
Pass	149.0	0.203020	0.084787	0.05	0.140	0.19	0.260	0.40				
Reject	31.0	0.519032	0.052240	0.42	0.495	0.52	0.545	0.63				

				Cor	relati	on M	atrix l	Plot	TAN-		77.75			1.0
Vibration	1.00	0.39	0.43	0.06	0.47	0.23	-0.04	-0.01	0.17	0.01	0.42	]		
A1	0.39	1.00	0.64	0.18	0.42	0.27	-0.07	0.00	0.26	-0.05	0.28			-0.8
A2	0.43	0.64	1.00	0.24	0.48	0.17	-0.05	-0.19	0.43	-0.00	0.30			
B1	0.06	0.18	0.24	1.00	0.33	0.06	0.09	-0.15	0.19	0.04	0.10		-	0.6
B2	0.47	0.42	0.48	0.33	1.00	0.24	-0.03	-0.00	0.39	0.01	0.41			
C1	0.23	0.27	0.17	0.06	0.24	1.00	-0.08	-0.11	0.20	0.01	0.24			0.4
D1	-0.04	-0.07	-0.05	0.09	-0.03	-0.08	1.00	0.05	-0.00	0.03	-0.11			
Ax	-0.01	0.00	-0.19	-0.15	-0.00	-0.11	0.05	1.00	-0.04	0.19	-0.04			0.2
Вх	0.17	0.26	0.43	0.19	0.39	0.20	-0.00	-0.04	1.00	0.09	0.39			
Ex	0.01	-0.05	-0.00	0.04	0.01	0.01	0.03	0.19	0.09	1.00	0.01			0.0
Fx	0.42	0.28	0.30	0.10	0.41	0.24	-0.11	-0.04	0.39	0.01	1.00			
\	/ibratio	n A1	A2	B1	B2	C1	D1	Ax	Bx	Ex	Fx			



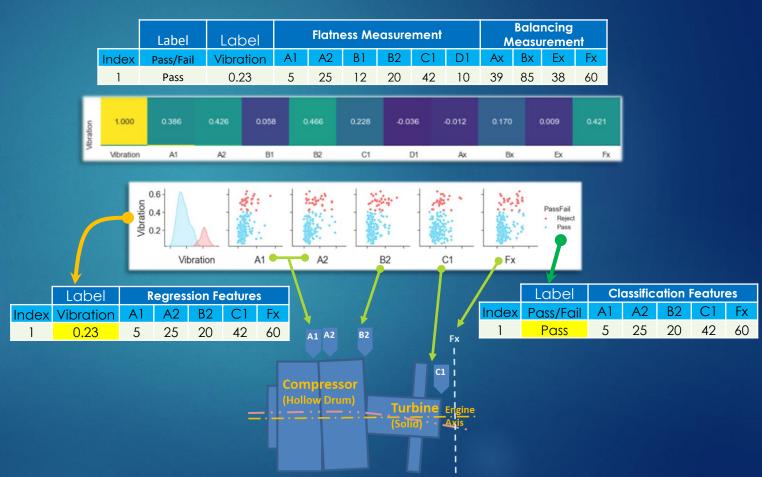


## **Data Preparation**

- Raw Data Collection
- Features in "Deg" to be dropped as the turbine is rotating based on the understanding of the operation of HP rotor.
- EDA
- Scatter Matrix: Low R-squared
- Regression Analysis
- Select Features
- Divided in to 2 sets:
   Regression and Categorical

- Select Features
- The features selected are good physical representation of the full HP module build.

		Flatness Measurement						Balancing Measurement										
	Label	Label	A1	A2	В1	B2	C1	C3	D1	D3	Ax	Ау	Bx	Ву	Ex	Еу	Fx	Fy
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1	Pass	0.23	5	25	12	20	42	154	10	100	39	227	85	63	38	312	60	221
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#### **Create Models**

## **Regression Model**

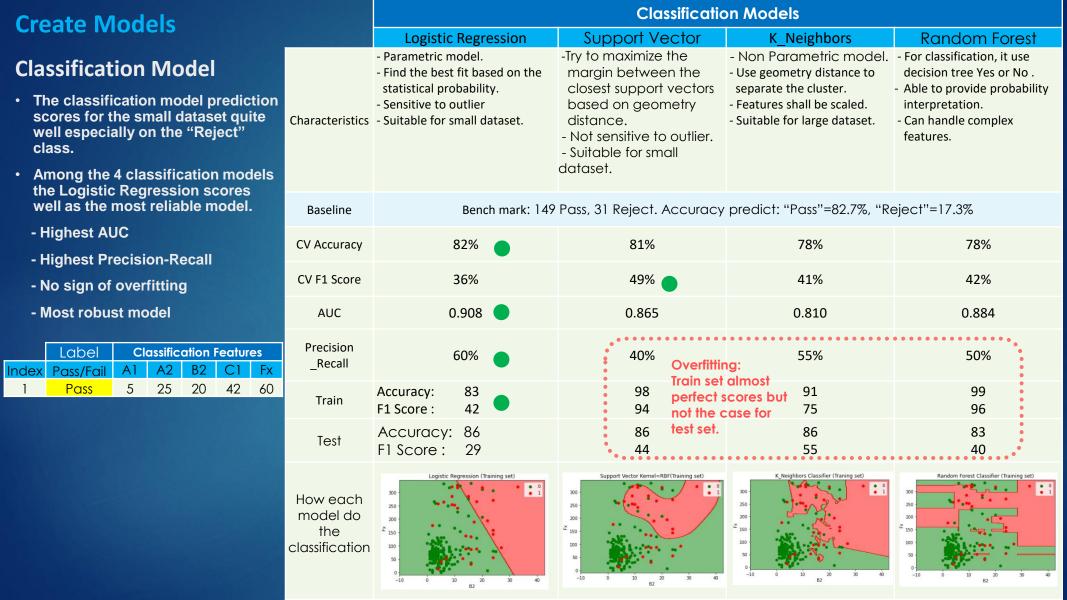
- Although from the EDA and regression analysis it is expected not gain result for the ML regression model.
- Nevertheless, the regression models are created to prepare for the future works. When new features are to be evaluated in regression models.
- Overall based on the current set of data, the regression cross validation score at about 0.2. Which unable to predict the Vibration confidently.

531	Label	Regression Features								
Index	Vibration	A1	A2	B2	C1	Fx				
1	0.23	5	25	20	42	60				

1]:				
	Parameters	Train_Accuracy	Test_Accuracy	4-Fold_CV_Score
Multiple Linear Regression	fit_intercept=False	0.235514	-0.007822	0.019539
Random Forest Regression	n_estimators=50, max_depth=4	0.697416	0.120070	0.197553
AdaBoost Regression	n_estimators=50	0.704885	0.182087	0.177160
Support Vector Regression	Kernel='rbf'	0.450199	0.121575	0.227106

# Test our best regression on some new measurement data In [51]: # Measurement data input from a new build MPS: A1-25 #<25 A2-25 #<25 B2-25 #<25 C1-25 #<25 Fx-250 #<250 measurement = pd.DataFrame({'A1': [A1],'A2': [A2],'B2': [B2],'C1': [C1],'Fx': [Fx]}) measurement = sc.transform(measurement) #perdict the Vibration of the measurements prediction-SVR reg.predict(measurement) print(f'The Vibration predicted by Support Vector is: {prediction[0]:0.3f}') The Vibration predicted by Support Vector is: 0.205

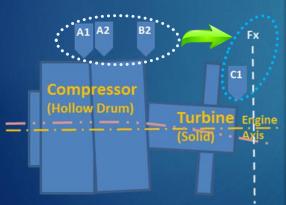


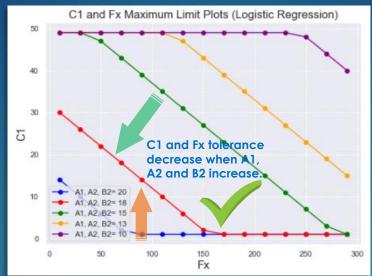


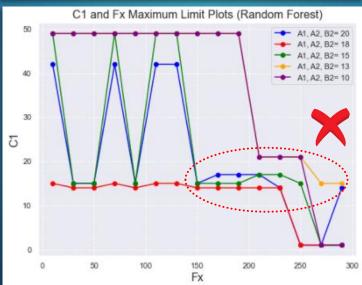
## **Evaluation**

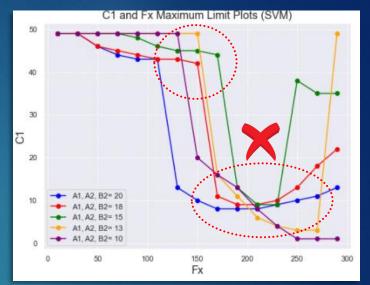
#### Test the model

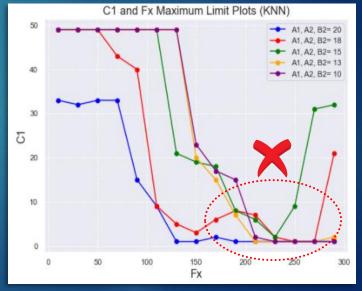
- A simulation of the models been carried out by using Compressor features A1, A2 and B2 to predict the Turbine feature C1.
- Rule: If the Compressor features values A1, A2 and B2 increase, the available tolerance of the Turbine C1 will need to decrease in order to prevent vibration reject at engine test.
- In this case only the Logistic Regression able show consistency to the Rule.











## **Deployment Proposal**

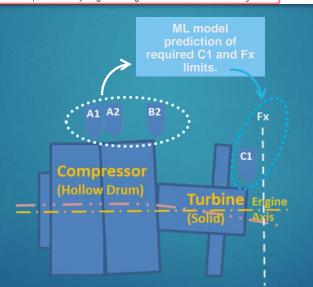
#### How the model help:

- 1. Predict if the HP module will cause high vibration during Test.
- 2. At build in-progress of HP module, with the compressor measurement, the model able to predict the passing limits for Turbine C1 and Fx.

# Logistic Regression Model Coefficient:

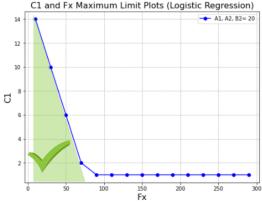
	Feature name	Coefficient	Odds_ratio
3	B2	0.770891	2.161691
5	Fx	0.453616	1.573994
1	A1	0.408341	1.504320
4	C1	0.237067	1.267526
2	A2	0.199571	1.220878
0	Intercept	-2.104775	0.121873

# 1. Sentencing the Pass or Reject with the HPS Measurements



# 2. Predicting the required Turbine Parameters C1 and Fx with the Compressor Measurements

```
# Measurement data input from Compressor
A1=20
      #<25
A2=20 #<25
B2=20 #<25
# Find the limit of the C1 and Fx on the matching Turbine
C1 list=[]
Fx list=[]
C1_range=np.arange(1,50,1)
Fx_range=np.arange(10,300,20)
for i in Fx range:
    for j in C1 range:
        measurement = pd.DataFrame({'A1': [A1], 'A2': [A2], 'B2': [B2], 'C1': [j], 'Fx': [i]})
        measurement = sc.transform(measurement)
        prediction=LR classifier.predict(measurement)
        if prediction>0:
   C1 list.append(j)
   Fx list.append(i)
# C1 and Fx predict tramline
plt.figure(figsize=(8,6))
plt.plot(Fx_list,C1_list,'o-',label='A1, A2, B2= 20', color='blue')
plt.grid(linestyle='--')
plt.xlabel('Fx',fontsize=16)
plt.vlabel('C1',fontsize=16)
plt.legend()
plt.title('C1 and Fx Maximum Limit Plots (Logistic Regression)', fontsize=16);
       C1 and Fx Maximum Limit Plots (Logistic Regression)
                                                 A1, A2, B2= 20
```



# **Deployment Proposal**

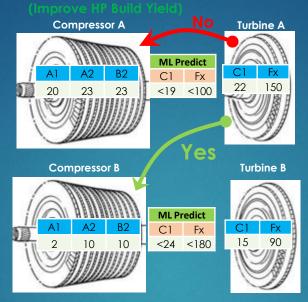
### Propose Strategy:

	Drawing Specification									
A1	A2	B2	C1	Fx						
<25	<25	<25	<25	<200						

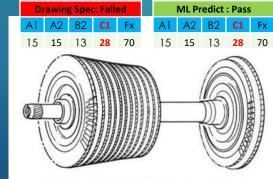
The Compressor and Turbine are build separately before assemble become full HP module.

- 1. Reduce the reject by Swap Turbines
- 2. Reduce the reject due to drawing spec one individual measurement.
- 3. For the chief engine to evaluate if the out of specification can be accepted. Increase speed of decision making.
- 4. Batch Matching to reduce the engine reject at Test Best. For example get the best combination among 5 HP compressors and turbines sets in order to achieve best margin to vibration reject.

1. Compressor and Turbine Swap



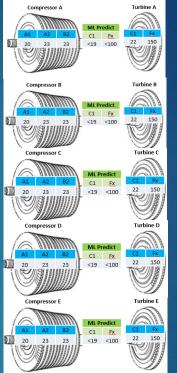
2. Drawing Spec VS ML Model

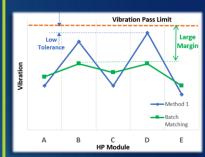


70% of the total rejected at HP build due to C1 measurement out of specification.

#### 4. Batch Matching

(Improve Engine Test Yield)





**Best Combination** with the largest margin to vibration reject at Test.





#### Conclusion

The machine learning is a effective tool to improve the yield of the production which can be easily proposed to the business because little capital investment is required.

#### **Challenges:**

- Small training dataset prone to overfitting.
- Yet to create a useful regression model which will be better illustrate the insight of vibration.

#### **Future Works:**

- Create a module of the model which can be used in standard application like Tableau to enable technician to use at production line.
- Create a report on how the model can help the Design Engineer to revise the drawing specification from individual measurement specification to matrix type specification.

#### **Lesson Learned:**

 Understanding the Business Needs, Processes and Operations from subject matter experts of relevant departments will be the keys to the success of a machine learning project.

# Thank You.