ELEC 3035: Review of Part I Ivan Markovsky

- Representations
- · Autonomous systems and stability
- · Controllability and observability
- Pole placement

Input/output representation $\mathcal{B}_{i/o}(P, Q)$

The difference equation in DT

$$P_0y(t) + P_1y(t+1) + \dots + P_ny(t+n) = Q_0u(t) + Q_1u(t+1) + \dots + Q_mu(t+m)$$

or differential equation in CT

$$P_0y(t) + P_1\frac{d}{dt}y(t) + \dots + P_n\frac{d^n}{dt^n}y(t+n)$$

$$= Q_0u(t) + \frac{d}{dt}Q_1u(t+1) + \dots + Q_m\frac{d^m}{dt^m}u(t+m)$$

where $m \le n$ defines linear time-invariant (LTI) system

The class of system that admit such a repr. is called finite dimensional.

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Transfer function

Consider a system $\mathscr{B}_{i/o}(P,Q)$ and let \mathscr{L} be the Laplace transform.

$$P(\sigma)y = Q(\sigma)u \implies P(s)Y(s) = Q(s)U(s)$$

where $Y := \mathcal{L}(y)$ and $U := \mathcal{L}(u)$.

The rational function

$$Y(s)U^{-1}(s) = P^{-1}(s)Q(s) =: H(s)$$

is called transfer function.

In the SISO case

$$\frac{\mathsf{Y}(\mathsf{s})}{\mathsf{U}(\mathsf{s})} = \frac{\mathsf{Q}(\mathsf{s})}{\mathsf{P}(\mathsf{s})} =: \mathsf{h}(\mathsf{s}).$$

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Input/state/output (I/S/O) representation

A finite dimensional LTI system admits a representation via

$$\sigma x = Ax + Bu, \quad y = Cx + Du$$

- x an auxiliary variable called state
- n := dim(x) state dimension, \mathbb{R}^n state space
- $A \in \mathbb{R}^{n \times n}$, $B \in \mathbb{R}^{n \times m}$, $C \in \mathbb{R}^{p \times n}$, $D \in \mathbb{R}^{p \times m}$ parameters of \mathscr{B}
- m := dim(u) input dimension, p := dim(y) output dimension

single input single output (SISO) systems — $\dim(u) = \dim(y) = 1$ multi input multi output (MIMO) systems — $\dim(u) \ge 1$, $\dim(y) \ge 1$

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Nonuniqueness of an I/S/O representation

There are two sources of nonuniqueness of an I/S/O representation:

- 1. redundant states n := dim(x) bigger than "necessary"
- 2. nonuniqueness of A, B, C, D choice of state space basis

minimal I/S/O representations — dim(x) is as small as possible

For any nonsingular matrix $T \in \mathbb{R}^{n \times n}$ and

$$\widetilde{A} = T^{-1}AT$$
, $\widetilde{B} = T^{-1}B$, $\widetilde{C} = CT$, $\widetilde{D} = D$

we have that

$$\mathscr{B}_{i/s/o}(A,B,C,D) = \mathscr{B}_{i/s/o}(\widetilde{A},\widetilde{B},\widetilde{C},\widetilde{D}).$$

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Nonuniqueness of an I/O representation

There are two sources of nonuniqueness of an I/O representation:

- 1. redundant equations g := row dim(P) bigger than "necessary"
- 2. nonuniquencess of P, Q equivalence of equations

minimal I/O representations — row dim(P) is as small as possible

In the single output case, P, Q are unique up do a scaling factor, i.e.,

$$\widetilde{P} = \alpha P, \qquad \widetilde{Q} = \alpha Q, \qquad \text{for } \alpha \in \mathbb{R}$$

we have that

$$\mathscr{B}_{i/o}(P,Q) = \mathscr{B}_{i/o}(\widetilde{P},\widetilde{Q}).$$

For multi output systems the nonuniqueness of *P*, Q is more essential.

I/S/O → transfer function

The transfer function corresponding to a system $\mathcal{B}_{i/s/o}(A,B,C,D)$ is

$$H(s) = C(sI - A)^{-1}B + D.$$

For the opposite direction "transfer function $\mapsto I/S/O$ ", see page 23.

State transition matrix

The dynamics of the sate vector *x* of the system

$$\sigma x = Ax, \quad y = Cx$$

is given by the equation

$$x(t_2) = \Phi(t_2 - t_1)x(t_1)$$

where $\Phi(t) = A^t$ in DT and $\Phi(t) = e^{At}$ in CT.

The matrix $\Phi(t)$ is called state transition matrix.

 $\Phi(t)$ shows how the initial state $x(t_1)$ is propagated in $t_1 + t$ time steps

Note: if t < 0, $\Phi(t)$ propagates backwards in time.

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State construction

Consider a scalar autonomous system defined by the equation

$$P_0y + P_1\sigma y + \cdots + P_{n-1}\sigma^{n-1}y + I\sigma^n y = 0.$$

How can we represent this system in a state space form

$$\sigma x = Ax, y = Cx$$

Choose, for example, $x(t) = \text{col}(y(t-1), \dots, y(t-n))$. Then

$$A = \begin{bmatrix} -P_{n-1} & -P_{n-2} & \cdots & -P_1 & -P_0 \\ I & 0 & \cdots & 0 & 0 \\ 0 & I & \ddots & \vdots & \vdots \\ \vdots & \ddots & \ddots & 0 & 0 \\ 0 & \cdots & 0 & I & 0 \end{bmatrix}$$
 companion matrix of P

$$C = \begin{bmatrix} -P_{n-1} & -P_{n-2} & \cdots & -P_1 & -P_0 \end{bmatrix}$$

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Characteristic polynomial of a matrix

The polynomial equation

$$\det(\lambda I_n - A) = c_0 \lambda^0 + c_1 \lambda^1 + \dots + c_n \lambda^n = 0$$

is called the characteristic equation of the matrix $A \in \mathbb{R}^{n \times n}$.

The roots of the characteristic polynomial

$$c(z) = c_0 z^0 + c_1 z^1 + \cdots + c_n z^n$$

are equal to the eigenvalues of A.

Cayley-Hamilton thm: Every matrix satisfies its own char. polynomial

$$c_0 A^0 + c_1 A^1 + \cdots + c_n A^n = 0.$$

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Stability

An autonomous system

$$\mathscr{B} = \{ x \mid \sigma x = f(x) \}$$

is stable if $x \in \mathcal{B}$ implies $x(t) \to 0$ as $t \to \infty$.

For a linear time-invariant system,

$$\mathscr{B} = \{ x \mid \sigma x = Ax \}$$

the eigenvalues of A determine the stability property of the system.

CT LTI system is stable iff all eigenvalues have negative real parts.

DT LTI system is stable iff all eigenvalues have absolute value < 1.

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State trajectories

The trajectories of the system

$$\mathscr{B}_{ss}(A,B) = \{ (u,x) \mid \sigma x = Ax + Bu \}$$

are in the DT case

$$x(t) = A^{t}x(0) + \sum_{\tau=0}^{t-1} A^{t-1-\tau}Bu(\tau)$$
 (1)

and in the CT case

$$x(t) = e^{At}x(0) + \int_0^t e^{A(t-\tau)}Bu(\tau)d\tau$$
 (2)

DT-CT analogy:
$$A^t \leftrightarrow e^{At}$$
 and $\sum_{t=0}^{t-1} (\cdot) \leftrightarrow \int_0^t (\cdot) d\tau$

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Controllability and state transfer

Controllability — a property of the system ensuring that

the system can be transferred from any given state $x_{\rm ini}$ to any desired state $x_{\rm des}$ over a period of time by proper choice of the input u.

 $\mathscr{B}(A,B)$ is controllable if and only if $\mathscr{R}_t = \mathbb{R}^n$

Minimum energy state transfer:

$$U_{\mathsf{ln},t} = \mathscr{C}_t^{\top} \left(\mathscr{C}_t \mathscr{C}_t^{\top} \right)^{-1} \left(x_{\mathsf{des}} - A^t x_{\mathsf{ini}} \right)$$

The minimum "energy" needed for $x_{\text{ini}} \mapsto x_{\text{des}}$ in t seconds is

$$\mathscr{E}_{\mathsf{min}} := \|U_{\mathsf{ln},t}\|_2^2 = \left(x_{\mathsf{des}} - A^t x_{\mathsf{ini}}\right)^\top \left(\sum_{\tau=0}^{t-1} A^\tau B B^\top (A^\tau)^\top\right)^{-1} \left(x_{\mathsf{des}} - A^t x_{\mathsf{ini}}\right)$$

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Controllability Gramian

 \mathcal{E}_{min} shows how "hard" is to transfer the state and depends on t.

Assuming that the system is stable

$$G_{\mathtt{c}} := \lim_{t o \infty} \left(\sum_{ au=0}^{t-1} A^{ au} B B^{ op} (A^{ au})^{ op}
ight)$$

exists and gives the minimum energy

$$\mathscr{E}_{\mathsf{min}} = (x_{\mathsf{des}} - A^t x_{\mathsf{ini}})^{\top} G_{\mathsf{c}}^{-1} (x_{\mathsf{des}} - A^t x_{\mathsf{ini}})$$

for state transfer without time limit.

 G_c is called the controllability Gramian of the system $\mathcal{B}_{ss}(A, B)$. It satisfies the matrix equation

$$AG_{c}A^{T} - G_{c} = -BB^{T}$$
 DT Lyapunov equation

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Controllability test for $\mathcal{B}_{i/o}(P, Q)$

GCD = greatest common divisor

Theorem The degree of the GCD d of p and q is equal to the rank deficiency of the Sylvester matrix $\begin{bmatrix} S_{\ell_p}(q) & S_{\ell_q}(p) \end{bmatrix}$, *i.e.*,

$$\deg(\textit{d}) = \ell_{\textit{p}} + \ell_{\textit{q}} - \text{rank} \left(\begin{bmatrix} S_{\ell_{\textit{p}}}(\textit{q}) & S_{\ell_{\textit{q}}}(\textit{p}) \end{bmatrix} \right).$$

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Output trajectories

The trajectories of the system

$$\mathscr{B}_{i/s/o}(A, B, C, D) = \{ (u, x) \mid \sigma x = Ax + Bu, \ y = Cx + Du \}$$

are in the DT case

$$y(t) = CA^{t}x(0) + C\sum_{\tau=0}^{t-1}A^{t-1-\tau}Bu(\tau) + Du(t)$$

and in the CT case

$$y(t) = Ce^{At}x(0) + C\int_0^t e^{A(t-\tau)}Bu(\tau)d\tau + Du(t)$$

DT–CT analogy:
$$A^t \leftrightarrow e^{At}$$
 and $\sum_{\tau=0}^{t-1} (\cdot) \leftrightarrow \int_0^t (\cdot) d\tau$

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Observability of DT systems

Suppose we have observed u and y over the period [0, t-1].

The system of equations

$$y(\tau) = CA^{\tau}x(0) + C\sum_{s=0}^{\tau-1}A^{\tau-1-s}Bu(s) + Du(\tau), \qquad \text{for } \tau = 0, 1, \dots, t-1$$

written in a matrix form is

$$\underbrace{\begin{bmatrix} y(0) \\ y(1) \\ \vdots \\ y(t-1) \end{bmatrix}}_{\mathbf{Y}_t} = \underbrace{\begin{bmatrix} C \\ CA \\ \vdots \\ CA^{t-1} \end{bmatrix}}_{\mathcal{O}_t} \mathbf{x}(0) + \underbrace{\begin{bmatrix} D & 0 & \cdots & 0 \\ CB & D & \ddots & \vdots \\ \vdots & \ddots & \ddots & 0 \\ CA^{t-1}B & \cdots & CB & D \end{bmatrix}}_{\mathcal{T}_t} \underbrace{\begin{bmatrix} u(0) \\ u(1) \\ \vdots \\ u(t-1) \end{bmatrix}}_{\mathbf{U}_t}$$

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$$Y_t = \mathscr{O}_t x(0) + \mathscr{T}_t U_t$$

- \mathcal{O}_t maps the initial state to the output over [0, t-1]
- \mathcal{T}_t maps the input to the output over [0, t-1]

Estimating the initial state requires to solve for x_0

$$\mathscr{O}_t x(0) = Y_t - \mathscr{T}_t U_t$$

Therefore, x_0 can be reconstructed uniquely if and only if $\ker(\mathscr{O}_t) = \{0\}$

 $\mathcal{B}_{i/s/o}(A, B, C, D)$ is observable if and only if $\ker(\mathcal{O}) = \{0\}$, where

$$\mathscr{O} := \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}$$

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Least squares observer

Assume that the output is observed with measurement noise *v*, *i.e.*,

$$y = Cx + Du + v$$

Then the system of equations for x_0

$$\mathcal{O}_t \mathbf{x}(0) = \mathbf{Y}_t - \mathcal{T}_t \mathbf{U}_t$$

(generically) has no exact solution. The least-squares observer is

$$\widehat{\mathbf{x}}_{ls}(0) = \underbrace{(\mathscr{O}_t^{\top} \mathscr{O}_t)^{-1} \mathscr{O}_t}_{F_{lo}} (\mathbf{Y}_t - \mathscr{T}_t \mathbf{U}_t)$$

It minimizes the output estimation error $\|\mathbf{Y} - \hat{\mathbf{Y}}\|_2$, where

$$\hat{Y} := \mathscr{O}_t \widehat{x}_{ls}(0) + \mathscr{T}_t U_t$$

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Duality between observability and controllability

The system $\mathscr{B}(A^{\top}, C^{\top}, B^{\top}, D^{\top})$ is called the dual of $\mathscr{B}_{i/s/o}(A, B, C, D)$.

The observability matrix of $\mathcal{B}_{i/s/o}(A, B, C, D)$ is

$$\mathscr{O}(A, \mathbf{C}) = \begin{bmatrix} \mathbf{C} \\ \mathbf{C}A \\ \vdots \\ \mathbf{C}A^{n-1} \end{bmatrix} = \mathscr{C}^{\top}(A^{\top}, \mathbf{C}^{\top})$$

equal to the transposed of the controllability matrix of

$$\mathscr{B}(A^{\top}, C^{\top}, B^{\top}, D^{\top})$$

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Motivation for pole placement

The desired dynamics is specified by the pole locations

$$\{z_{\mathsf{des},1},\ldots,z_{\mathsf{des},n_{\mathsf{cl}}}\}$$

of the closed-loop system

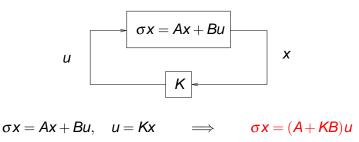
or equivalently by the characteristic polynomial

$$p_{\text{des}}(z) = \prod_{i=1}^{n_{\text{cl}}} (z - z_{\text{des},i}) = p_{\text{des},0} + p_{\text{des},1}z + \dots + p_{\text{des},n_{\text{cl}}}z^{n_{\text{cl}}}$$

of the closed-loop system.

- Example: in deadbeat control $z_{\text{des},i} = 0$, for all i, i.e., $p_{\text{des}}(z) = z^{n_{\text{cl}}}$
- The aim of pole placement control is to choose the feedback so that the closed-loop system achieve the desired char. polynomial

State feedback



The closed-loop system is autonomous with state matrix

$$A_{c} = A + KB$$
.

Pole placement by state-feedback aims to choose K, so that

$$\det(zI - A_c) = \rho_{des}(z)$$

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Controller canonical form (SISO case)

Fact: Any controllable system $\mathscr{B}_{i/o}(p,q)$ can be represented in a state space form $\mathscr{B}_{i/s/o}(A,b,c,d)$ with parameters

$$A = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \ddots & \vdots \\ \vdots & & \ddots & \ddots & 0 \\ 0 & \cdots & \cdots & 0 & 1 \\ -p_0 & -p_1 & \cdots & \cdots & -p_{n-1} \end{bmatrix}, \quad b = \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 1 \end{bmatrix}$$

$$c = \begin{bmatrix} c_0 & c_1 & \cdots & c_{n-1} \end{bmatrix}, \quad d = q_n$$

where c_0, c_1, \dots, c_{n-1} are the coefficients of $q(z) - q_n p(z)$

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Similarity transformation for controller canonical form

A more general result:

Lemma:

- Let A, b and A', b' be two controllable pairs and
- assume that A and A' have the same char. polynomials.

Then there is a unique similarity transformation given by the matrix

$$T := \mathscr{C}(A, b) \big(\mathscr{C}(A', b') \big)^{-1}$$

such that

$$T^{-1}AT = A'$$
 and $T^{-1}b = b'$.

Any controllable representation of the system can be transformed to the controller canonical form.

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State-feedback pole placement in controller form

Let the plant be given by $\mathcal{B}_{i/s/o}(A, b, c, d)$, with A, b in controller form.

Then

$$A_{cl} := A + bk = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \ddots & \vdots \\ \vdots & & \ddots & \ddots & 0 \\ 0 & \cdots & \cdots & 0 & 1 \\ k_1 - p_0 & k_2 - p_1 & \cdots & \cdots & k_n - p_{n-1} \end{bmatrix}$$

and the closed-loop characteristic polynomial is

$$p_{cl}(z) = (p_0 - k_1) + (p_1 - k_2)z + \dots + (p_{n-1} - k_n)z^{n-1} + z^n$$

The equation $p_{cl} = p_{des}$ has the unique solution

$$k_1 = p_0 - p_{\text{des},0}, \cdots, k_n = p_{n-1} - p_{\text{des},n-1}$$

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Observer design

The observer design is based on the following principles:

- 1. Internal model: the model run by u, gives an estimate \hat{x} for x
- 2. Feedback: correct the estimate \hat{x} , so that the error

$$x(t) - \widehat{x}(t) =: e(t) \to 0$$
 as $t \to \infty$

Let the feedback be a linear function of the output error

feedback correction =
$$L(y - \hat{y})$$

Then the observer for the model $\mathcal{B}_{i/s/o}(A, B, C, D)$ is

$$\widehat{\sigma}\widehat{x} = A\widehat{x} + Bu - \underline{L}(y - \widehat{y})$$
$$\widehat{y} = C\widehat{x} + Du$$

Observer design by pole placement

The condition $e(t) \rightarrow 0$ as $t \rightarrow \infty$ is a minimum requirement.

In fact we want $e(t) \rightarrow 0$ fast

(possibly in a finite (small) number of steps → deadbeat observer)

The error dynamics is governed by the poles of the matrix

$$A_0 := A + LC$$

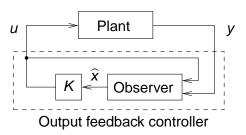
so for desired error dynamics we can

select desired pole locations of A_0 and choose L to achieve them.

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Closed-loop system with output feedback controller

Consider the closed loop system



where

Plant: $\sigma x = Ax + Bu$, y = Cx + Du

Observer: $\sigma \hat{x} = A\hat{x} + Bu - L(y - C\hat{x} - Du)$

State feedback controller: $u = K\hat{x}$

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Feedback controller:

$$\sigma \widehat{\mathbf{x}} = (\mathbf{A} + \mathbf{LC})\widehat{\mathbf{x}} + (\mathbf{B} + \mathbf{LD})\mathbf{u} - \mathbf{L}\mathbf{y}, \qquad \mathbf{u} = \mathbf{K}\widehat{\mathbf{x}}$$
$$= (\mathbf{A} + \mathbf{LC} + \mathbf{B}\mathbf{K} + \mathbf{LD}\mathbf{K})\widehat{\mathbf{x}} - \mathbf{L}\mathbf{y}$$

Note: the feedback controller is a dynamical system

Closed-loop system:
$$\sigma \begin{bmatrix} x \\ \widehat{x} \end{bmatrix} = \begin{bmatrix} A & BK \\ -LC & A+LC+BK \end{bmatrix} \begin{bmatrix} x \\ \widehat{x} \end{bmatrix}$$

Note: closed-loop system order = plant order + controller order

Error equation:
$$\sigma \begin{bmatrix} x \\ e \end{bmatrix} = \begin{bmatrix} A + BK & -BK \\ 0 & A + LC \end{bmatrix} \begin{bmatrix} x \\ e \end{bmatrix}$$

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Polynomial approach to pole placement

$$u = P(\sigma)y = Q(\sigma)u$$

$$P_{c}(\sigma)u = Q_{c}(\sigma)y$$

Plant: $P(\sigma)y = Q(\sigma)u$

Controller: $P_c(\sigma)u = Q_c(\sigma)y$

The closed-loop system is autonomous. In the SISO case

Closed-loop system:
$$(p_c(\sigma)p(\sigma) - q_c(\sigma)q(\sigma))y = 0$$

and the closed-loop characteristic polynomial is

$$ho_{
m cl}(z) :=
ho_{
m c}(z)
ho(z) - q_{
m c}(z) q(z)$$

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Diophantine equation

For SISO pole placement we need to solve the polynomial equation

$$p_{c}(z)p(z) - q_{c}(z)q(z) = p_{des}(z)$$
 (D)

in p_c, q_c with $degree(p_c) \ge degree(q_c)$ (for causality of the controller).

Notes:

- p_{des} is the desired char. polynomial of the closed-loop system
- $\underline{\text{degree}(p_{\text{des}})} = \underline{\text{degree}(p)} + \underline{\text{degree}(p_{\text{c}})}$ CL sys's order n_{cl} plant order n_{c} controller order n_{c}
- In state space, p_{des} includes plant and observer's desired poles.

The equation (D) is called Diophantine equation (also Bezout eqn).

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Diophantine equation

With $n_c := degree(p_c)$ and $m_c := degree(q_c)$ given,

$$ho_{\mathrm{c}}(z)
ho(z)-q_{\mathrm{c}}(z)q(z)=
ho_{\mathrm{des}}(z)$$

can be written as

$$\begin{bmatrix} S_{n_c}(p) & S_{m_c}(q) \end{bmatrix} \begin{bmatrix} p_c \\ q_c \end{bmatrix} = p_{des}$$
 (D')

where

$$\begin{split} p_{\text{c}} &= \text{col}(p_{\text{c},0}, p_{\text{c},1}, \dots, p_{\text{c},n_{\text{c}}}) \quad, \quad q_{\text{c}} = \text{col}(q_{\text{c},0}, q_{\text{c},1}, \dots, q_{\text{c},m_{\text{c}}}), \\ p_{\text{des}} &= \text{col}(p_{\text{des},0}, p_{\text{des},1}, \dots, p_{\text{des},n_{\text{cl}}}) \end{split}$$

 \implies solving (D) (with n_c , m_c given) is a standard linear algebra problem

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Part I: Linear system design

- 1. Review of linear algebra
- 2. Introduction to state space and polynomial methods
- 3. Autonomous systems and stability
- 4. Controllability and observability
- 5. Design by pole placement and observer design
- 6. Linear quadratic control and Kalman filter
- System identification

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Part II: Nonlinear system design

- Mathematical modelling of nonlinear systems
- · Lyapunov stability analysis
- Describing functions
- Feedback linearization
- Adaptive control