## ME 467 – ECE 564: Robotics and Automated Systems Spring 2025 — Project II: Localization

In this project, we will implement a couple (orientation) localization algorithms and compare them to a third algorithm, the so-called Mahony filter [1]. Mahony filter, which was first introduced in the seminal paper [4], is a type of nonlinear complementary filter. In our implementation, the orientation of our cell phone  $\{C\}$  with respect to the inertial frame  $\{0\}$  will be represented by the rotation matrix  $\mathbf{R}$  or the associated quaternion  $\mathbf{q}$ .

The purpose of these algorithms is to obtain an estimate of the orientation with an inertial measurement unit (IMU) strapped on a rigid body. Today, every cell phone has an IMU incorporated to its circuitry. For Android and iOS users, we will make use of HyperIMU [3] and SensorLog [5], respectively, in order to collect data from the on-board IMU.

We achieve this by numerical integration. The orientation kinematics is given by the differential equation

$$\dot{q} = \frac{1}{2} q \circ p(\omega) = \frac{1}{2} \Omega(\omega) q,$$
 (1)

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where the map p takes a three-vector  $\boldsymbol{\omega}$  and outputs a pure quaternion:  $\boldsymbol{\omega} \stackrel{p}{\mapsto} 0\langle \boldsymbol{\omega} \rangle$ . The operation  $\circ$  represents quaternion multiplication. The last equality expresses the quaternion multiplication as a matrix multiplication of the quaternion q, viewed as a 4-vector, with the 4-by-4 matrix  $\Omega(\boldsymbol{\omega})$ , defined by

$$oldsymbol{\Omega}(oldsymbol{\omega}) riangleq egin{bmatrix} 0 & -oldsymbol{\omega}^{ op} \ oldsymbol{\omega} & -oldsymbol{\omega}_{ imes} \end{bmatrix},$$

with  $\boldsymbol{\omega}_{\times}$  representing the skew-symmetric form of  $\boldsymbol{\omega}$ . Note that  $\boldsymbol{\Omega}(\boldsymbol{\omega})^2 = -\|\boldsymbol{\omega}\|^2 \boldsymbol{I}$ .

How does one integrate the orientation kinematics (1)? First, we assume that the integration period is very short:  $\delta t \ll 1/f$ , where f is the frequency of the fastest motion the rigid-body can undergo. This allows us to assume that  $\omega$  is constant during an interval of operation  $[t_0, t_0 + \delta t)$ . We then solve the differential equation (1) in closed-form under this assumption to obtain

$$\mathbf{q}(t) = e^{1/2\mathbf{\Omega}(\boldsymbol{\omega})t}\mathbf{q}(t_0), \quad t \in [t_0, t_0 + \delta t).$$

If we express this equation in terms of the integration step k so that  $t = k\delta t$ , we obtain the following update equation

$$q_{k+1} = e^{1/2\Omega(\boldsymbol{\omega})\delta t}q_k.$$

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Finally, note that the exponential in this formula may be evaluated in closed-form using the formula [6]

$$e^{\mathbf{\Omega}(\boldsymbol{\omega})\theta} = \cos(\|\boldsymbol{\omega}\|\theta)\mathbf{I}_4 + \sin(\|\boldsymbol{\omega}\|\theta)\frac{\mathbf{\Omega}(\boldsymbol{\omega})}{\|\boldsymbol{\omega}\|}.$$

Suppose at the initial time,  $t_0 = 0$ ,  $q(t_0) = 1\langle 0 \rangle$ . Integrate equation (1) with the following sequence of inputs:

- 1. Rotate about the body x-axis for 1/4 s at  $\pi$  rad s<sup>-1</sup>.
- 2. Rotate about the body z-axis for 1/4 s at  $\pi$  rad s<sup>-1</sup>.
- 3. Rotate about the body x-axis for  $^{1}/_{4}$  s at  $-\pi$  rad s<sup>-1</sup>.
- 4. Rotate about the body z-axis for 1/4 s at  $-\pi$  rad s<sup>-1</sup>.

Report the rotation q(1) at the end of this motion, i.e., after 1s has elapsed. Does it hold that q(1) = q(0)? Discuss why this is true or false.

$$t, m_x, m_y, m_z, \omega_{y,x}, \omega_{y,y}, \omega_{y,z}, a_x, a_y, a_z,$$

where t is the time,  $a_{\bullet}$  and  $m_{\bullet}$  are respectively the components of the linear acceleration and magnetic field measurements  $\boldsymbol{a}$  and  $\boldsymbol{m}$ , expressed in the body-frame. Finally,  $\omega_{y,\bullet}$  are the components of  $\boldsymbol{\omega}_{y}$ , which is the measured angular velocity through the gyroscope.

We will implement the Mahony filter so that we obtain a good estimate of the orientation of my cell phone for the duration of the time I collected the data in input. The Mahony filter is characterized by the following equations

$$\dot{\hat{\boldsymbol{q}}} = \frac{1}{2}\hat{\boldsymbol{q}} \circ \boldsymbol{p} \left( \boldsymbol{\omega}_{y} - \hat{\boldsymbol{b}} + k_{p} \boldsymbol{\omega}_{\text{mes}} \right) =: \frac{1}{2}\hat{\boldsymbol{q}} \circ \boldsymbol{p}(\boldsymbol{u}),$$

$$\dot{\hat{\boldsymbol{b}}} = -k_{I} \boldsymbol{\omega}_{\text{mes}},$$

$$\boldsymbol{\omega}_{\text{mes}} = k_{a} \boldsymbol{v}_{a} \times \hat{\boldsymbol{v}}_{a} + k_{m} \boldsymbol{v}_{m} \times \hat{\boldsymbol{v}}_{m},$$
(2)

Any estimated quantity has a *hat* symbol over it. For example,  $\hat{\boldsymbol{b}}$  denotes the estimate of the gyroscope bias vector. The normalized acceleration and the normalized magnetic north vector are respectively given by  $\boldsymbol{v}_a$  and  $\boldsymbol{v}_m$  so that  $\|\boldsymbol{v}_a\| = \|\boldsymbol{v}_m\| = 1$ . Their estimated counterparts are denoted  $\hat{\boldsymbol{v}}_a$  and  $\hat{\boldsymbol{v}}_m$ . The nonnegative gains  $k_p$ ,  $k_I$ ,  $k_a$ , and  $k_m$  are to be determined. Note that the first equation in equations (2) may be expressed in terms of the rotation matrix  $\hat{\boldsymbol{R}}$  associated with the quaternion  $\hat{\boldsymbol{q}}$  as follows

$$\dot{\hat{R}} = \hat{R} \left( \omega_y - \hat{b} + k_p \omega_{\text{mes}} \right)_{\times} =: \hat{R} u_{\times}.$$
(3)

The question is, how does one compute the estimates  $\hat{\boldsymbol{v}}_a$  and  $\hat{\boldsymbol{v}}_m$ ? We go back to the definition of  $\boldsymbol{v}_a$  in order to determine this. Note that  $\boldsymbol{a}$  is the linear acceleration of the rigid-body as expressed in the body frame. If we denote the linear acceleration of the rigid-body, expressed in the inertial frame by  $\boldsymbol{a}_0$ , then we have the relationship

$$oldsymbol{v}_a riangleq rac{oldsymbol{a}}{\|oldsymbol{a}\|} = oldsymbol{R}^ op rac{oldsymbol{a}_0}{\|oldsymbol{a}_0\|}.$$

The estimated counterpart of this vector is defined analogously, using the estimated orientation, that is,

$$\hat{oldsymbol{v}}_a = \hat{oldsymbol{R}}^ op rac{oldsymbol{a}_0}{\|oldsymbol{a}_0\|}.$$

Let's take the time-derivative of this, substitute from equation (3), and live with the consequences:

$$\dot{\hat{\boldsymbol{v}}}_a = \dot{\hat{\boldsymbol{R}}}^{\top} \frac{\boldsymbol{a}_0}{\|\boldsymbol{a}_0\|} = -\boldsymbol{u}_{\times} \hat{\boldsymbol{R}}^{\top} \frac{\boldsymbol{a}_0}{\|\boldsymbol{a}_0\|} = -\boldsymbol{u}_{\times} \hat{\boldsymbol{v}}_a. \tag{4}$$

This equation must be integrated in order to maintain the estimate  $\hat{v}_a$  of the normalized acceleration vector,  $v_a$ , expressed in body frame. An analogous construction shows that we also must integrate the equation

$$\dot{\hat{\boldsymbol{v}}}_m = -\boldsymbol{u}_{\times} \hat{\boldsymbol{v}}_m \tag{5}$$

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so as to maintain an estimate of the normalized magnetic north vector  $v_m$ , expressed in the body frame.

Remark 1. Typically, the magnetometer reading m has a component that points towards world's core, i.e., along the gravitational acceleration vector. Make sure to subtract off this component when constructing the vector  $\mathbf{v}_m$ . Otherwise, your estimation will suffer!

Now, from problem 1, we know how to integrate the first of equations (2). The second equation is straightforward to integrate

$$\hat{\boldsymbol{b}}_{k+1} = \hat{\boldsymbol{b}}_k - k_I \boldsymbol{\omega}_{\text{mes}} \delta t.$$

Finally, we have to integrate equations (4, 5) so that we can compute the estimator input  $\omega_{\text{mes}}$ . Again, assuming that  $\boldsymbol{u}$  is constant in the selected short integration step, we can integrate these equations in closed-form to obtain

$$\hat{\boldsymbol{v}}_{a,k+1} = e^{-\boldsymbol{u}_{\times}\delta t}\hat{\boldsymbol{v}}_{a,k}, \qquad \hat{\boldsymbol{v}}_{m,k+1} = e^{-\boldsymbol{u}_{\times}\delta t}\hat{\boldsymbol{v}}_{m,k}.$$

where the exponential map  $\mathfrak{so}(3) \xrightarrow{\exp} \mathbf{SO}(3)$ , is given by the Rodrigues formula:

$$e^{\boldsymbol{u}_{\times}\theta} = \boldsymbol{I}_{3} + \sin(\|\boldsymbol{u}\|\theta) \frac{\boldsymbol{u}_{\times}}{\|\boldsymbol{u}\|} + (1 - \cos(\|\boldsymbol{u}\|\theta)) \frac{\boldsymbol{u}_{\times}^{2}}{\|\boldsymbol{u}\|^{2}}.$$

Process the input file input using your Mahony filter. Compute the amount of rotation  $\theta(t)$  as a function of time, taking  $\theta(0) = 0$ . Plot this with respect to time and then find the total amount my cell phone has rotated during the time I collected this data.

**Remark 2.** For some nominal values that you can set for the gains  $k_p$  and  $k_I$ , please see Section VI of [4]. The remaining two gains  $k_a$  and  $k_m$  are supposed to be set relative to each other. If there is a lot of magnetic interference you ought to have  $k_m \ll k_a$ . Conversely, if the motion undergoes a lot of acceleration, you ought to have  $k_m \gg k_a$ .

- (a) [7 points] Find the roll-pitch-yaw angles from your estimated orientation,  $\hat{q}$ , and plot them with respect to time.
- (b) [7 points] Record all the necessary quantities to compute the estimate of the error the Mahony filter is making, given in eqn. (6), and plot it with respect to time.

$$E_{\text{mes}} = 1 - \boldsymbol{v}_a \cdot \hat{\boldsymbol{v}}_a + 1 - \boldsymbol{v}_m \cdot \hat{\boldsymbol{v}}_m. \tag{6}$$

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(c) [6 points] Record the gyroscope bias estimate and plot it with respect to time.

- (a) [5 points] Implement a naïve orientation estimator that integrates raw gyroscope measurements. The naïve estimator takes the gyroscope measurement  $\omega_y$  and estimates the orientation  $\hat{q}$  by integrating the differential equation in (1), with  $\omega$  replaced by  $\omega_y$ .
- (b) [5 points] Implement the triad (Tri-axial Attitude Determination) method, first described in [2] to algebraically estimate an attitude represented as a direction cosine matrix from two orthogonal vector observations.

Given two non-parallel reference unit vectors  $\mathbf{v}_1$  and  $\mathbf{v}_2$  and their corresponding unit vectors  $\mathbf{w}_1$  and  $\mathbf{w}_2$ , it is required to find an orthogonal matrix  $\mathbf{R}$  satisfying  $\mathbf{R}\mathbf{w}_i = \mathbf{v}_i$  for i = 1, 2.

Two vectors  $\mathbf{v}_1$  and  $\mathbf{v}_2$  define an orthogonal coordinate system with the **normalized** basis vectors  $\mathbf{q}$ ,  $\mathbf{r}$ , and  $\mathbf{s}$  as the following triad:

$$oldsymbol{q}_r = oldsymbol{v}_1, \qquad oldsymbol{r}_r = rac{oldsymbol{v}_1 imes oldsymbol{v}_2}{\|oldsymbol{v}_1 imes oldsymbol{v}_2\|}, \qquad oldsymbol{s}_r = oldsymbol{r}_r imes oldsymbol{q}_r.$$

These are represented as vectors to build an appropriate reference frame  $M_r$ :

$$oldsymbol{M}_r = egin{bmatrix} oldsymbol{s}_r & oldsymbol{r}_r & oldsymbol{q}_r \end{bmatrix}$$
 .

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Similarly, at any given time, two measured vectors in the body frame  $\mathbf{w}_1$  and  $\mathbf{w}_2$  determine the 3-by-3 body matrix  $\mathbf{M}_b$ :

$$oldsymbol{M}_b = egin{bmatrix} oldsymbol{s}_b & oldsymbol{r}_b & oldsymbol{q}_b \end{bmatrix},$$

where, like the first triad, the second triad is built as

$$oldsymbol{q}_b = oldsymbol{w}_1, \qquad oldsymbol{r}_b = rac{oldsymbol{w}_1 imes oldsymbol{w}_2}{\|oldsymbol{w}_1 imes oldsymbol{w}_2\|}, \qquad oldsymbol{s}_b = oldsymbol{r}_b imes oldsymbol{q}_b.$$

The attitude (rotation) matrix  $\mathbf{R} \in \mathbf{SO}(3)$  defines the coordinate transformation:  $\mathbf{R}\mathbf{M}_b = \mathbf{M}_r$  and solving for  $\mathbf{R}$  we obtain

$$oldsymbol{R} = oldsymbol{M}_r oldsymbol{M}_b^{ op}.$$

For estimations using an IMU, we identify two main reference vectors: gravity  $\mathbf{g}^{\star}$  and magnetic field  $\mathbf{m}^{\star} = \begin{bmatrix} m_x & m_y & m_z \end{bmatrix}^{\top}$ . A common convention sets the gravity vector equal to 0 along the  $\mathbf{z}-$  and  $\mathbf{y}-$ axes, and equal to  $\sim 9.81$  along the  $\mathbf{z}-$ axis. This assumes the direction of gravity is parallel to the vertical axis. Because triad uses normalized vectors, the  $\mathbf{z}-$ axis turn out to be equal to 1:  $\mathbf{g}^{\star} = \begin{bmatrix} 0 & 0 & 1 \end{bmatrix}^{\top}$ .

The magnetic field is defined from the geographical position of the measurement. Using the World Magnetic Model, we can estimate the magnetic field elements of our location on a given date.

Boise State University is located at  $43.603600^{\circ}$  N and  $116.208710^{\circ}$  W with an altitude of 835 m. On February 27, 2023 its magnetic elements were obtained as

$$m^* = \begin{bmatrix} 4525.28449 & 19699.18982 & -47850.850686 \end{bmatrix}^\top \text{ nT}.$$

Again, triad works with normalized vectors, so the reference magnetic vector becomes

$$m^* = \begin{bmatrix} 0.087117 & 0.37923 & -0.92119 \end{bmatrix}^\top$$
.

Both normalized vectors  $\boldsymbol{v}_1 = \boldsymbol{g}^{\star}$  and  $\boldsymbol{v}_2 = \boldsymbol{m}^{\star}$  build the reference triad.

Then, we have to measure their equivalent vectors, for which we use the accelerometer to obtain  $\boldsymbol{a} = \begin{bmatrix} a_x & a_y & a_z \end{bmatrix}^{\top}$  and the magnetometer for  $\boldsymbol{m} = \begin{bmatrix} m_x & m_y & m_z \end{bmatrix}^{\top}$ . Both measurement vectors,  $\boldsymbol{w}_1 = \boldsymbol{a}$  and  $\boldsymbol{w}_2 = \boldsymbol{m}$  are also normalized, meaning  $\|\boldsymbol{w}_1\| = \|\boldsymbol{w}_2\| = 1$ , so they can build the *body's measurement triad*  $\boldsymbol{M}_b$ .

- (c) [10 points] Process the input file input from problem 2 again with the naïve estimator and the triad method and plot the amount of rotation,  $\theta(t)$ , with respect to time. Also compute the total amount my cell phone has rotated during the time I collected this data using the naïve estimator and the triad method.
- (d) [5 points] Repeat problem 3 using the naïve estimator and the triad method. Since none of these methods keeps an estimate of the gyroscope bias, you can ignore part (c) of problem 3 for this question.

(e) [5 points (bonus)] Provide a discussion about which of the estimation methods works better for the input file that I have provided. Can you imagine a scenario in which the performances of the various estimators would be very different from what you observed for this dataset? What would the characteristics of such a dataset be? Can you create a such a motion with your cell phone and show such behavior?

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## References

- [1] Mahony Orientation Filter. URL https://ahrs.readthedocs.io/en/latest/filters/mahony.html.
- [2] H. D. Black. A passive system for determining the attitude of a satellite. AIAA journal, 2(7):1350–1351, 1964.
- [3] S. Campisi. HyperIMU. URL https://github.com/JohannesFriedrich/HypeRIMU.
- [4] R. Mahony, T. Hamel, and J.-M. Pflimlin. Nonlinear complementary filters on the special orthogonal group. *IEEE Transactions on automatic control*, 53(5):1203–1218, 2008.
- [5] B. Thomas. SensorLog. URL http://sensorlog.berndthomas.net/.
- [6] J. R. Wertz. Spacecraft attitude determination and control, volume 73. Springer Science & Business Media, 2012.