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# **Advanced Helicopter Dynamics**

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1. INTRODUCTION LIST OF TABLES

#### 1 Introduction

Somewhat surprisingly over the period of 2010 to 2017 total electricity generation in the United Kingdom fell by 18 % [1] as consumption also fell 10% across the period [2]. This is in spite of a rising population and number of households. This has been attributed to warmer winter lowering electric heating demands as well as improvements in refrigerator technology and increased use of LED lights [5]. Using data made available by the UK Department for Business, Energy & Industrial Strategy [1] it can be seen that wind power has defied the slowdown of electricity generation in the UK, rising an average 29% year on year over the same 2010 to 2017 period.

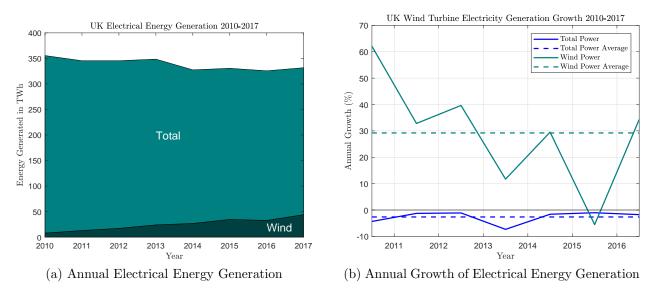


Figure 1: Statistics on UK Electrical Energy Generation

Traditionally the wind sectors growth was completely reliant on large government subsidies but in recent years governments such as the UK and Germany have adopted a new approach to subsidisation which has lead to significant shift in the sector's economics. The government policy change was to make energy companies bid for projects and associated subsidies meaning the actual subsidy received is the difference between the subsidy and the bid. This competitive approach (contract for difference) has driven innovation with off-shore wind cost more than halving in the UK in just three years [6]. In some cases large wind projects have become viable with no subsidy or tax-break at all. The significance is the rate of development within the wind energy sector is now not reliant on government renewable energy policy but is shifting to large scale private sector investment. Due to the sheer size of the energy industry this could lead to significant wind turbine technology development. With the age of the electric car expected to dramatically reverse the trend of falling electricity consumption seen in Figure 1b, research into wind turbine design has never been more relevant.

The optimal design of a wind turbine blade is one which produces the most power for the lowest unit cost [4]. This is would require a vast amount of cost data which is not readily available and so this paper shall instead focus on the design of a technically optimal design. Specifically the design which has the highest annual energy production (AEP) within the parameters given in Table 1.

2. METHODOLOGY LIST OF TABLES

Quantity **Symbol** Value Units Hub Height  $H_{\text{hub}}$ 35 m 3 **Hub-Tower Separation**  $d_{hub}$ m **DIMENSIONS** Root Radius  $R_{min}$ 1 m Maximum Radius 20  $R_{max}$ m Mean Chord  $c_{mean}$ m Maximum Root Bending Moment 0.5 MNm  $M_{root,max}$ **FORCES** Maximum Vertical Hub Force 70,000 N  $F_{Y,max}$ 2,000 Blade Density kg/m<sup>3</sup> Pblade **Blade Stiffness** (EI)<sub>blade</sub> GPa Cut-In Speed (assumed)  $V_{min}$ 5 m/s **QUANTITIES** 25 Cut-Out Speed (assumed)  $V_{max}$ m/s 7 Weibull Coefficient A Weibull Coefficient 1.8 k Rotational Speed 30 ω rpm

Table 1: Design Parameters for Wind Turbine Design [4]

### 2 Methodology

To find the optimal design blade element momentum theory was used to calculate power output for a given velocity and parameter set. Using Weibull's wind speed distribution and power output for a given velocity, AEP was calculated. Part A of the flow diagram of Figure 2 shows the flow of information for the computer program used to calculate the AEP.

The design parameter's which could be changed for blade design were inclination at root:  $\theta_0$ , rate of twist:  $\theta_{tw}$  and chord gradient:  $c_g$ . The theoretical maximum power output for a given wind speed,  $V_0$ , and reference area, A, is the called the Betz limit and is given by equation 1. The maximum theoretical AEP can be found using the Weibull distribution and the Betz limit. Using a minimum finding function from Matlab's library the combination of the three parameters which gives the least difference to the Betz limit AEP can be found thus finding the optimal design for the given parameter set.

$$P_{Betz} = \frac{16}{27} \left( \frac{1}{2} \rho V_0^3 A \right) \tag{1}$$

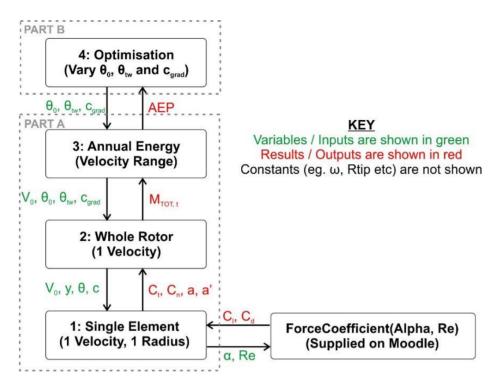


Figure 2: Flow Diagram of Method Used to Find Annual Energy Production for the Turbine Design [4]

#### 3 Results and Discussion

- 3.1 A: Optimal Design
- 3.2 B: Limitations of Model

#### 4 Conclusions

#### References

- [1] Department for Business, Energy & Industrial Stratergy, 2018. Historical Electricity Data:1920 to 2017.
  - Available: www.gov.uk/government/statistical-data-sets/historical-electricity-data
- [2] Digest of United Kingdom Energy Statistics (DUKES), 2018. Digest of UK Energy Statistics (DUKES): electricity. Available:https://www.gov.uk/government/collections/electricity-statistics
- [3] Cleaver, D., 2018. *Slides: WT Coursework*. University of Bath.
- [4] Cleaver, D., 2018. ME40343 Helicopter Dynamics: Coursework. University of Bath.

REFERENCES REFERENCES

[5] Vaughan, A., 2017. Electric cars will fuel huge demand for power, says National Grid. The Guardian.

[6] Arie, S., 2018. Renewables are primed to enter the global energy race. The Financial Times.

# Appendices

#### A Induced Calculations Function

```
1 function [a_out, adash_out, phi, Cn, Ct] = ...
      WTInducedCalcs(a_in, adash_in, VO, omega, y, theta, chord, B)
2 %1: SINGLE ELEMENT: use an iterative solution to find the ...
      values of a,
3 %adash, phi, Cn and Ct at a particular radius.
5 %% Set constants
6 \text{ rho} = 1.225;
                        %density of air in kg/m<sup>3</sup>
7 \text{ mu} = 18.81e-6;
                        %kinematic viscosity in Pa.s
9 error_tol = .0001; % setting allowable error
10 error = error_tol + 1; {	t \%}set first error to initiate while loop
12 %% loop over until error goes below tolerance
13 i = 0;
               %Initialize counter, will need to break loop if i ...
      exceeds 100
14 while error \geq error_tol && i <101
15
       i = i + 1;
16
17
       %% Calculate
                      angles
18
       phi = atan(((1-a_in)*V0) / ((1+adash_in)*omega*y));
19
20
       alpha = phi - theta;
21
22
       %% Calculate Reynolds number
23
       V_{rel} = ((V0*(1-a_{in}))^2 + ((omega*y)*(1 + adash_{in}))^2)^5;
24
25
       Re = (rho*V_rel*chord) / mu;
26
27
       %% Get non-dimensional lift and drag coefficients and convert
28
       [Cl, Cd] = ForceCoefficient(alpha, Re);
29
       Cn = Cl*cos(phi) + Cd*sin(phi);
                                                  %Normal Force ...
          Coefficient
       Ct = Cl*sin(phi) - Cd*cos(phi);
                                                  %Tangential Force ...
          Coefficient
31
32
       %% Calculate new values for a and a'
       sigma = (B*chord) / (2*pi()*y);
34
       a_{out} = 1 / (((4*sin(phi)^2)/(sigma*Cn)) + 1);
```

```
adash_out = 1 / (((4*sin(phi)*cos(phi))/(sigma*Ct)) - 1);
37
       %% Calculate error
38
       error = abs(a_out - a_in) + abs(adash_out - adash_in);
39
40
       %% If error greater than allowable error apply relaxation ...
          factor,
41
       %else the solution has been found and while loop can be ...
42
       if error > error_tol
            a_{in} = 0.1*(a_{out} - a_{in}) + a_{in};
43
44
            adash_in = 0.1*(adash_out - adash_in) + adash_in;
45
       else
46
            break
47
       end
48
49
  end
  %% If counter exceed 100 set a' to zero and solve for a
51
53 if i > 100
54
       a_in = 0;
56
       adash_out = 0;
57
       while error > error_tol
58
59
            %% If counter reaches 500 the error tolerance
            % can be decreased to get a result.
61
62
                         %Increase counter
            i = i+1;
63
64
            \textbf{if} \quad \textbf{i} \quad \geq \quad 500
                                      %Decrease error tol after 500 ...
               interations
                error_tol = 0.01;
66
            end
67
            %% Calculate angles
68
            phi = atan(((1-a_in)*V0) / ((1)*omega*y));
69
            alpha = phi - theta;
71
72
73
            %% Calculate Reynolds number
74
            V_{rel} = ((V0*(1-a_{in}))^2 + ((omega*y)*(1))^2)^.5;
75
76
            Re = (rho*V_rel*chord) / mu;
```

```
77
            %% Get non-dimensional lift and drag coefficients and ...
79
            [Cl, Cd] = ForceCoefficient(alpha, Re);
80
81
            Cn = Cl*cos(phi) + Cd*sin(phi);
                                                        %Normal Force ...
               Coefficient
82
            Ct = Cl*sin(phi) - Cd*cos(phi);
                                                        %Tangential ...
               Force Coefficient
83
84
            %% Calculate new values for a and a'
85
            sigma = (B*chord) / (2*pi*y);
86
            a_{out} = 1 / (((4*sin(phi)^2)/(sigma*Cn)) + 1);
87
88
            %% Calculate error
89
            error = abs(a_out - a_in);
90
91
            \ensuremath{\text{\%}}\xspace If error greater than allowable error apply \dots
               relaxation factor,
            \% else the solution has been found and while loop can ...
               be broken
            if error > error_tol
94
                 a_{in} = 0.1*(a_{out} - a_{in}) + a_{in}; %Relaxation ...
                    facto k=.1 added
            else
                 break
96
97
            end
98
99
        end
100
101 end
102 end
```

## B Wind Turbine Single Velocity Function

```
4 %root.
6 %% Set constants
7 \text{ rho} = 1.225;
                         %density of air in kg/m<sup>3</sup>
9 %% Set up geometry
10 y = 1.5:1:19.5;
                      %Vector of local radii (elements) (in m)
11 y_{\Delta} = y(2) - y(1); %\Delta is difference between adjacent elements
12
13 R = TipRadius;
14
15 %% Set up empty outputs arrays to improve efficiency
16 MN_local = zeros(length(y),1);
17 MT_local = zeros(length(y),1);
18\, %% 2: Calculate individual element moment both in plane and ...
      normal
19
20 for i = 1:length(y)
21
22
       %Calculate local chord and angle of attack
23
       chord_local = chord_mean + ((y(i)) - (R/2))*chord_grad;
24
       assert(chord_local \geq 0 && chord_local \leq 2, strcat('Error ...
           in chord length! Chord length = ', num2str(chord_local)))
25
       theta_local = theta0 + (y(i))*theta_twist;
26
27
       %Calculate Cn and Ct for the local element
28
       [a_out, adash_out, ¬, Cn, Ct] = WTInducedCalcs(0, 0, V0, ...
           omega, y(i), theta_local, chord_local, B);
29
       % Calculate Relative Velocity
31
       V_{rel} = ((V0*(1-a_out))^2 + ((omega*y(i))*(1 + ...
           adash_out))^2)^.5;
       %Calculate the local moment in normal and tangential ...
           (torque) directions
34
       MN_local(i) = (.5*rho*(V_rel^2)*chord_local*Cn)*y_{\triangle}*y(i);
       MT_local(i) = (.5*rho*(V_rel^2)*chord_local*Ct)*y_{\Delta}*y(i);
36
37 end
38
39 %% Integrate the moments to find totals
40
41 \text{ MN} = \text{sum}(\text{MN_local});
42 \text{ MT} = \text{sum}(\text{MT_local});
43
```

44 end

#### C Wind Turbine Velocity Range Function

```
1 function [Diff] = WTVelocityRange(Parameters)
2 %3: ANNUAL ENERGY - loop WTSingleVelocity to find the moments ...
      across the
3 %entire velocity range. Combine this with the frequency ...
      information to get
4 %the AEP. Parameters = [theta0, theta_twist, chord_grad]
6 %% Setting as constants%%%%
7 A = 7;
8 k = 1.8;
9 \text{ omega} = 3.14;
10 \text{ chord_mean} = 1;
11 TipRadius = 20;
12 RootRadius = 1;
13 B = 3;
14 \text{ MinVO} = 5;
15 \text{ MaxVO} = 25;
17 %% Constants
18 \text{ rho} = 1.225;
                        %density of air in kg/m<sup>3</sup>
19 Area = (TipRadius^2)*(pi);
                                           %Area in m^2
20
21 %% Calculate total tangential moment for each speed
22 \quad \triangle VO = 1;
23 V0 = MinV0:\Delta_V0:MaxV0;
                                 % Set up wing velocity range in m/s
24 MT = zeros(length(VO), 1);
                                      % Initialise tangential moment ...
      vector for each wind speed
25 MN = zeros(length(VO), 1);
                                      %Initialise normal moment ...
      vector for each wind speed
26 Power = zeros(length(VO), 1); %Initialise power vector for ...
      each wind speed
P_{\text{speed}} = zeros(length(V0)-1, 1);
                                                   %Initialise vector ...
      of speed probablilities
29 AEP_speed = zeros(length(V0)-1, 1);
                                                  %Initialise vector ...
      of annual power at each speed
30 Power_midpoint = zeros(length(V0)-1, 1); %Initialise vector ...
      of power inbetween speeds
```

```
31 AEP_speed_ideal = zeros(length(V0)-1, 1);
32 Power_ideal = zeros(length(V0)-1, 1);
34 %% Calculate power for each speed
36
  [MT(1), MN(1)] = WTSingleVelocity(VO(1), Parameters(1), ...
      Parameters (2), chord_mean, Parameters (3), TipRadius, ...
      RootRadius, omega, B);
38 \text{ Power}(1) = MT(1)*B*omega;
39
40
  for i = 2:length(V0)
41
42
       [MT(i), MN(i), \neg, \neg] = WTSingleVelocity(VO(i), ...
          Parameters (1), Parameters (2), chord_mean, ...
          Parameters (3), TipRadius, RootRadius, omega, B);
44
       Power(i) = MT(i)*B*omega;
45
46
       %%%Calculating power and probablilities in the speed ...
          intervals
47
       P_{speed(i-1)} = \exp(-(V0(i-1)/A)^k) - \exp(-(V0(i)/A)^k); ...
          "Calculate probability of speed between Vi and Vi+1
48
49
       Power_midpoint(i-1) = 0.5*(Power(i-1) + Power(i)); \dots
                    %Power at mid point is average of lower and ...
          upper bounds
       Power_ideal(i-1) = (16/27) * ...
          0.5*rho*(((V0(i-1)+V0(i))/2)^3)*Area;
                                                    %Theoretical ...
          maximum power defined by Betz limit
52
       AEP\_speed(i-1) = Power\_midpoint(i-1) * P\_speed(i-1) * ...
          8760;
                         %Calculate predictied anual power ...
          generated at each windspeed range
       AEP\_speed\_ideal(i-1) = Power\_ideal(i-1) * P\_speed(i-1) * ...
                   %Calculate the ideal anual power generated at ...
          each windspeed range
55
  end
56
57 \text{ AEP} = sum(AEP\_speed);
58
  AEP_ideal = sum(AEP_speed_ideal);
59
60 tip_deflection = WTBendingDeflection(Parameters(1), ...
      Parameters (2), Parameters (3));
```

#### D Wind Turbine Velocity Range Function

```
1 function [x] = TurbineOptimisation()
2 %%%%Script to find the optimal solution for theta0, ...
     theta_twist and
3 %chrod gradient
5 %OPTIMISE FIT - 3 variables: Theta0, Theta_grad and Chord_grad
6 opts = optimset('fminsearch');
7 opts.Display = 'iter'; %What to display in command window
8 opts. TolX = 0.001; %Tolerance on the variation in the parameters
9 opts.TolFun = 1e7; %Tolerance on the error
10 opts.MaxFunEvals = 150; %Max number of iterations
11
12 %fminsearch inputs: (function, initial guess, lower limits on \dots
     variables,
13 % upper limits on variables, options)
14
15 %Limit for ThetaO is set to +/- 20 degrees
16\, %Limit for rate of twist is set so 45\, degrees twis over blade ...
     is the
17 %maximum
18\, % Limit for chord grad set so there is at least 20cm chord at \dots
     root for
19 % attatchment and so chord is positive at R = 20
21 [x] = fminsearchbnd(@WTVelocityRange, [5*pi/180 -.3*pi/180 ...
      .0], [-20*pi/180 -2.36*pi/180 -0.105], [20*pi/180 ...
     2.36*pi/180 (8/95)], opts);
23 %disp(strcat('OPTIMAL VALUES: Theta0 = ', ...
     num2str(x(1)*180/pi), 'Twist Rate = ', ...
     num2str(x(2)*180/pi), 'Taper Rate = ', num2str(x(3)))
```

```
24
25 %% Check bending moment does not Exceed 0.5MNm
26 %WTSingleVelocity inputs => VO, thetaO, theta_twist, ...
      chord_mean, chord_grad, TipRadius, RootRadius, omega, B
27
28 %[MT, \neg] = WTSingleVelocity(25, x(1), x(2), 1, x(3), 20, 1, ...
      30, 3);
30 %disp(MT)
31
32 %% Compute and display how
33 BetzLimit = 1.99684E+09;
34 diff = WTVelocityRange(x);
35 eff = (BetzLimit - diff) / BetzLimit * 100;
37 disp(strcat('Percentage of Betz Limit = ', num2str(eff), ' \%'))
38
39
40
41 end
```

#### E Flow Chart of FEM Solver