

Recursive Noise Adaptive Kalman Filtering by Variational Bayesian Approximations

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Abstract—This article considers the application of variational Bayesian methods to joint recursive estimation of the dynamic state and the time-varying measurement noise parameters in linear state space models. The proposed adaptive Kalman filtering method is based on forming a separable variational approximation to the joint posterior distribution of states and noise parameters on each time step separately. The result is a recursive algorithm, where on each step the state is estimated with Kalman filter and the sufficient statistics of the noise variances are estimated with a fixed-point iteration. The performance of the algorithm is demonstrated with simulated data.

Index Terms—Adaptive filtering, Kalman filtering, noise adaptive filtering, variational Bayesian methods.

I. INTRODUCTION

THE *Kalman filter* (KF) [1] considers estimation of the dynamic state from noisy measurements in the class of estimation problems where the dynamic and measurement processes can be approximated by linear Gaussian state space models. The *extended Kalman filter* (EKF) and the *unscented Kalman filter* (UKF) extend this method to non-linear dynamic and measurement models by forming a Gaussian approximation to the posterior state distribution [2]–[5]. A serious limitation in these filters is that they assume complete *a priori* knowledge of the measurement and dynamic model parameters, including the noise statistics. The exact knowledge of the parameters, and especially, the noise statistics is not in many practical situations a plausible assumption. Examples of such applications are low cost integrated GPS/INS positioning systems and fault tolerant systems (see, e.g., [6], [7] and references therein).

The classical way (see, e.g., [8], [9]) of solving the problem of uncertain parameters is to use *adaptive filters* where the model parameters or the noise statistics are estimated together with the dynamic state. The classical noise adaptive filtering approaches can be divided into Bayesian, maximum likelihood, correlation and covariance matching methods [8]. The *Bayesian approach* is the most general of these and the other approaches can often be interpreted as approximations to the Bayesian approach. Examples of Bayesian approaches to noise adaptive filtering are state augmentation based methods [10], [11], multiple model methods such as the *interacting multiple models* (IMM) algorithm [3], [9] and particle methods [12]–[15].

Variational Bayesian (VB) methods have been developed for a wide range of models to perform approximate posterior inference at low computational cost in comparison with the sampling methods (for a review of VB methods, see, e.g., [16]–[18]). These methods assume a simpler, analytically tractable form for the posterior. Two main approaches are either to derive a factored free form distribution (for models in the conjugate-exponential class), or to assume a fixed-form posterior distribution (e.g., a multivariate Gaussian, with possibly a suitable parametrization of the model).

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Smidl and Quinn [19] present a theoretical framework for VB-approximations, with special emphasis on recursive Bayesian inference and signal processing applications. Related to the present work, VB-approximations of AR-models with unknown variances are also considered, but general linear state space models with unknown time-varying variances are not. A general variational version of the Kalman smoother has been developed in [20] utilizing the conjugate-exponential framework, but with stationary noise. Time varying variance models have been addressed in [21] by using the fixed form approach.

Our proposed method approximates the joint posterior distribution of the state and the noise variances by a factorized free form distribution. This approximation is formed on each time step separately, and the result is a recursive algorithm, where on each step the sufficient statistics of the state and the noise variances are estimated with a fixed-point iteration of a Kalman filter. We also propose a heuristic dynamical model for the variances, which can be used for modeling the time behavior of the noise, still retaining the functional form of the approximate posterior distribution.

II. MAIN RESULTS

A. Problem Formulation

The discrete-time linear state space model considered here is

$$\begin{aligned} x_k &= A_k x_{k-1} + q_k \\ y_k &= H_k x_k + r_k \end{aligned} \quad (1)$$

where $q_k \sim \mathcal{N}(0, Q_k)$ is the Gaussian process noise, $r_k \sim \mathcal{N}(0, \Sigma_k)$ is the measurement noise with diagonal covariance Σ_k , and the initial state has a Gaussian prior distribution $x_0 \sim \mathcal{N}(m_0, P_0)$. The measurement y_k is a d -dimensional vector and the state x_k is an n -dimensional vector. Time is indexed by k and the matrices A_k , H_k , Q_k , as well as the parameters of the initial state (prior) distribution m_0 , P_0 are assumed to be known.

Now, departing from the case of standard Kalman filter, we assume that the observation noise variance parameters $\sigma_{k,i}^2$, $i = 1, \dots, d$, are stochastic with independent dynamic models. We denote the diagonal covariance matrix comprising of these variances by $\Sigma_k = \text{diag}(\sigma_{k,1}^2, \dots, \sigma_{k,d}^2)$. The construction of a suitable dynamical model of the observation noise variances will be discussed in Section II-C, and at this stage we denote it generically by $p(\Sigma_k | \Sigma_{k-1})$. We assume that the dynamic models of the states and the variance parameters are independent:

$$p(x_k, \Sigma_k | x_{k-1}, \Sigma_{k-1}) = p(x_k | x_{k-1}) p(\Sigma_k | \Sigma_{k-1}). \quad (2)$$

The goal of Bayesian optimal filtering of the above model is to compute the posterior distribution $p(x_k, \Sigma_k | y_{1:k})$. Formally, the well-known recursive solution to this filtering problem consists of the following steps [2]:

- **Initialization:** The recursion starts from the prior distribution $p(x_0, \Sigma_0)$.
- **Prediction:** The predictive distribution of the state x_k and measurement noise covariance Σ_k is given by the Chapman-Kolmogorov equation:

$$\begin{aligned} p(x_k, \Sigma_k | y_{1:k-1}) &= \int p(x_k | x_{k-1}) p(\Sigma_k | \Sigma_{k-1}) \\ &\quad \times p(x_{k-1}, \Sigma_{k-1} | y_{1:k-1}) dx_{k-1} d\Sigma_{k-1}. \end{aligned} \quad (3)$$

- *Update*: Given the next measurement y_k , the predictive distribution above is updated to a posterior distribution by the Bayes' rule:

$$p(x_k, \Sigma_k | y_{1:k}) \propto p(y_k | x_k, \Sigma_k) p(x_k, \Sigma_k | y_{1:k-1}). \quad (4)$$

The integrations involved in the general solution are usually not analytically tractable. In the following, we solve the above recursion effectively by using a VB approximation for the posterior update, accompanied by suitable (heuristic) dynamics for the observation noise variances.

B. Variational Approximation

Assume that we can approximate the conditional distribution for x_{k-1} and Σ_{k-1} given the measurements y_1, \dots, y_{k-1} as a product of Gaussian and independent Inverse-Gamma distributions¹ as follows:

$$p(x_{k-1}, \Sigma_{k-1} | y_{1:k-1}) = N(x_{k-1} | m_{k-1}, P_{k-1}) \times \prod_{i=1}^d \text{Inv-Gamma}(\sigma_{k-1,i}^2 | \alpha_{k-1,i}, \beta_{k-1,i}).$$

We have chosen this approximation, because the Inverse-Gamma distribution is the conjugate prior distribution for the variance of a Gaussian distribution. For this reason, using an Inverse-Gamma model for variances of Gaussian models is common in Bayesian analysis [22]. And for the same reason, the Inverse-Gamma distribution turns out to be the natural approximating distribution in the VB-approach considered here.

Because the dynamics of the state and observation noise variances are assumed to be independent, the factored form will remain in the prediction step, as can be easily seen from the (3). The linear dynamics will result in a Gaussian distribution for predicted x_k , with parameters m_k^-, P_k^- given by the standard KF prediction equation.

Let us assume that the dynamical models for the variance parameters $p(\sigma_{k,i}^2 | \sigma_{k-1,i}^2)$ are of such form that the prediction step will always result in an Inverse-Gamma predicted distribution for each $\sigma_{k,i}^2$, with parameters $\alpha_{k,i}^-, \beta_{k,i}^-$ (we will discuss possibilities for such dynamics later on). The joint prediction distribution is thus

$$p(x_k, \Sigma_k | y_{1:k-1}) = p(x_k | y_{1:k-1}) p(\Sigma_k | y_{1:k-1}) \\ = N(x_k | m_k^-, P_k^-) \times \prod_{i=1}^d \text{Inv-Gamma}(\sigma_{k,i}^2 | \alpha_{k,i}^-, \beta_{k,i}^-).$$

Now, in the posterior update step, the state and observation noise variance parameters will be coupled through the likelihood $p(y_k | x_k, \Sigma_k)$, and the exact posterior will not be of tractable form.

In order to make the computations tractable we will now form a variational approximation to the posterior distribution. We follow the standard VB-approach (see, e.g., [16]–[19]) and search for a free form factored approximate distribution for $p(x_k, \Sigma_k | y_{1:k})$ as follows:

$$p(x_k, \Sigma_k | y_{1:k}) \approx Q_x(x_k) Q_\Sigma(\Sigma_k).$$

The VB-approximation can now be formed by minimizing the the Kullback-Leibler (KL) divergence [23] between the separable approximation and the true posterior:

$$\text{KL}[Q_x(x_k) Q_\Sigma(\Sigma_k) || p(x_k, \Sigma_k | y_{1:k})] = \int Q_x(x_k) Q_\Sigma(\Sigma_k) \\ \times \log \left(\frac{Q_x(x_k) Q_\Sigma(\Sigma_k)}{p(x_k, \Sigma_k | y_{1:k})} \right) dx_k d\Sigma_k.$$

¹In this article, we use the same parametrization of Gamma and Inverse-Gamma distributions as in [22].

Using the methods from calculus of variations (see, e.g., [24]) for minimizing the KL-divergence with respect to the probability densities $Q_x(x_k)$ and $Q_\Sigma(\Sigma_k)$ in turn, while keeping the other fixed yields:

$$Q_x(x_k) \propto \exp \left(\int \log p(y_k, x_k, \Sigma_k | y_{1:k-1}) Q_\Sigma(\Sigma_k) d\Sigma_k \right) \\ Q_\Sigma(\Sigma_k) \propto \exp \left(\int \log p(y_k, x_k, \Sigma_k | y_{1:k-1}) Q_x(x_k) dx_k \right).$$

This cannot be solved directly, as the above equations are coupled. However, computing the expectation in the first equation gives the following:

$$\int \log p(y_k, x_k, \Sigma_k | y_{1:k-1}) Q_\Sigma(\Sigma_k) d\Sigma_k \\ = -\frac{1}{2} (y_k - H_k x_k)^T \langle \Sigma_k^{-1} \rangle_\Sigma (y_k - H_k x_k) \\ - \frac{1}{2} (x_k - m_k^-)^T (P_k^-)^{-1} (x_k - m_k^-) + C_1 \quad (5)$$

where $\langle \cdot \rangle_\Sigma = \int (\cdot) Q_\Sigma(\Sigma_k) d\Sigma_k$ denotes the expected value with respect to the approximating distribution $Q_\Sigma(\Sigma_k)$ and C_1 denotes terms independent of x_k . As a function of x_k , this is a quadratic implying that $Q_x(x_k)$ is a Gaussian, mean and covariance of which can be found with standard matrix manipulations.

Similarly, the second expectation can be computed as follows:

$$\int \log p(y_k, x_k, \Sigma_k | y_{1:k-1}) Q_x(x_k) dx_k \\ = -\sum_{i=1}^d \left(\frac{3}{2} + \alpha_{k,i} \right) \ln(\sigma_{k,i}^2) - \sum_{i=1}^d \frac{\beta_{k,i}}{\sigma_{k,i}^2} \\ - \frac{1}{2} \sum_{i=1}^d \frac{1}{\sigma_{k,i}^2} \langle (y_k - H_k x_k)_i^2 \rangle_x + C_2 \quad (6)$$

where $\langle \cdot \rangle_x = \int (\cdot) Q_x(x_k) dx_k$. From the equation it can be seen that $Q_\Sigma(\Sigma_k)$ is a product of Inverse-Gamma distributions. By evaluating the expectations in (5) and (6), and matching the parameters of the distributions on left and right hand sides gives the following densities:

$$Q_x(x_k) = N(x_k | m_k, P_k) \\ Q_\Sigma(\Sigma_k) = \prod_{i=1}^d \text{Inv-Gamma}(\sigma_{k,i}^2 | \alpha_{k,i}, \beta_{k,i}),$$

where the parameters $m_k, P_k, \alpha_{k,i}, \beta_{k,i}$ are the solution to the following coupled set of equations:

$$m_k = m_k^- + P_k^- H_k^T (H_k P_k^- H_k^T + \hat{\Sigma}_k)^{-1} (y_k - H_k m_k^-) \\ P_k = P_k^- - P_k^- H_k^T (H_k P_k^- H_k^T + \hat{\Sigma}_k)^{-1} H_k P_k^- \\ \alpha_{k,i} = 1/2 + \alpha_{k-1,i} \\ \beta_{k,i} = \beta_{k-1,i} + \frac{1}{2} \left[(y_k - H_k m_k)_i^2 + (H_k P_k H_k^T)_{ii} \right] \quad (7)$$

where $i = 1, \dots, d$ and

$$\hat{\Sigma}_k = \text{diag}(\beta_{k,1}/\alpha_{k,1}, \dots, \beta_{k,d}/\alpha_{k,d}).$$

In this manner, we have completed one prediction-update cycle ending up with an approximate posterior distribution of the same functional form as the one we began with. This same procedure can now be applied to the next measurement and the measurement after that, which results in a recursive type of filtering algorithm. However, in order to complete the algorithm description we will construct a suitable dynamic model for the noise variances in the next section.

C. Dynamics of Observation Noise Variances

It is not straightforward to choose a dynamical model $p(\Sigma_k | \Sigma_{k-1})$ such that the Chapman-Kolmogorov (3) for the observation noise vari-

ances would yield Inverse-Gamma distributions as their predicted distributions. Usually, the dynamical model of the noise variances is not known in detail, and hence we propose a heuristic dynamics for the variances, which simply “spreads” their previous approximate posteriors. We choose to keep the expected measurement noise *precisions* (inverse variances) constant, and increase their variances by a factor of ρ^{-1} , $\rho \in (0, 1]$. This is obtained by the following, for $i = 1, \dots, d$:

$$\begin{aligned}\alpha_{k,i}^- &= \rho_i \alpha_{k-1,i} \\ \beta_{k,i}^- &= \rho_i \beta_{k-1,i}.\end{aligned}\quad (8)$$

Note that the value $\rho = 1$ corresponds to stationary variances and lower values increase their assumed time-fluctuations. A strong practical motivation for this specific heuristic is the stability of the VB scheme: it ensures that the algorithm starts from the same posterior (parameter values) where it was terminated at the previous time point and thus if the parameters already have the correct values, they remain unaffected (see the fixed point iteration in Algorithm 1).

D. The Adaptive Filtering Algorithm

Algorithm 1: The Variational Bayesian Adaptive Kalman Filter (VB-AKF) algorithm

- *Predict*: Compute the parameters of the predicted distribution as follows:

$$\begin{aligned}m_k^- &= A_k m_{k-1} \\ P_k^- &= A_k P_{k-1} A_k^T + Q_k \\ \alpha_{k,i}^- &= \rho_i \alpha_{k-1,i}, \quad i = 1, \dots, d \\ \beta_{k,i}^- &= \rho_i \beta_{k-1,i}, \quad i = 1, \dots, d\end{aligned}$$

- *Update*: First set $m_k^{(0)} = m_k^-$, $P_k^{(0)} = P_k^-$, $\alpha_{k,i} = 1/2 + \alpha_{k,i}^-$, and $\beta_{k,i}^{(0)} = \beta_{k,i}^-$, for $i = 1, \dots, d$. Then iterate the following a few, say N , steps:

$$\begin{aligned}\hat{\Sigma}_k^{(n)} &= \text{diag} \left(\beta_{k,1}^{(n)} / \alpha_{k,1}^{(n)}, \dots, \beta_{k,d}^{(n)} / \alpha_{k,d}^{(n)} \right) \\ m_k^{(n+1)} &= m_k^- + P_k^- H_k^T \left(H_k P_k^- H_k^T + \hat{\Sigma}_k^{(n)} \right)^{-1} \\ &\quad \times (y_k - H_k m_k^-) \\ P_k^{(n+1)} &= P_k^- - P_k^- H_k^T \left(H_k P_k^- H_k^T + \hat{\Sigma}_k^{(n)} \right)^{-1} H_k P_k^- \\ \beta_{k,i}^{(n+1)} &= \beta_{k,i}^- + \frac{1}{2} \left(y_k - H_k m_k^{(n+1)} \right)_i^2 \\ &\quad + \frac{1}{2} \left(H_k P_k^{(n+1)} H_k^T \right)_{ii}, \quad i = 1, \dots, d,\end{aligned}$$

and set $\beta_{k,i} = \beta_{k,i}^{(N)}$, $m_k = m_k^{(N)}$, $P_k = P_k^{(N)}$.

The adaptive filtering algorithm is presented as Algorithm 1. The algorithm should be started from a prior of the form

$$p(x_0, \Sigma_0) = N(x_0 | m_0, P_0) \prod_{i=1}^d \text{Inv-Gamma}(\sigma_{0,i}^2 | \alpha_{0,i}, \beta_{0,i})$$

and the approximation formed by the algorithm on step k is then

$$\begin{aligned}p(x_k, \Sigma_k | y_{1:k}) &\approx N(x_k | m_k, P_k) \\ &\quad \times \prod_{i=1}^d \text{Inv-Gamma}(\sigma_{k,i}^2 | \alpha_{k,i}, \beta_{k,i}).\end{aligned}$$

The prediction step of the algorithm consists of the standard KF prediction step and the heuristic prediction presented in (8). The update step is simply a fixed point iteration algorithm for the (7). Note that the degrees of freedom parameters of the Inverse-Gamma distributions do not change in the fixed point iteration. Also, the update of $Q_x(x_k)$ is the standard KF update step with the known noise covariance $\hat{\Sigma}_k$. This is used to find a distribution for the noise variances, which would give the best separable match to the (intractable) joint posterior of state and observation noise variances. The KF update is then performed again with the obtained new expected noise covariance. The fixed-point algorithm is in fact a type of natural gradient method, which can be shown to have asymptotic optimal convergence properties [25].

In experiments we have noticed that the fixed point iteration tends to converge very fast and most of the time only few, say 2 or 3, iterations are enough in practice.

III. ILLUSTRATIVE EXAMPLE

A. Stochastic Resonator Model

In this simulation the phenomenon is a randomly drifting stochastic resonator, which is measured at time intervals of $\Delta t = 1/10$ seconds. The dynamic model for the resonator with angular velocity ω can be written in form

$$\begin{pmatrix} x_{1,k+1} \\ x_{2,k+1} \\ x_{3,k+1} \end{pmatrix} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos(\omega \Delta t) & \frac{\sin(\omega \Delta t)}{\omega} \\ 0 & -\omega \sin(\omega \Delta t) & \cos(\omega \Delta t) \end{pmatrix} \begin{pmatrix} x_{1,k} \\ x_{2,k} \\ x_{3,k} \end{pmatrix} + \begin{pmatrix} w_{1,k} \\ w_{2,k} \\ w_{3,k} \end{pmatrix}, \quad (9)$$

where $w_k = (w_{k,1} w_{k,2} w_{k,3})^T$ is a white Gaussian noise sequence with a known covariance Q . The measurement model at time step k is

$$y_k = x_{k,1} + x_{k,2} + v_k$$

where v_k is a Gaussian random variable with unknown time varying variance σ_k^2 . In the simulation, angular velocity $\omega = 0.05$ was used. The initial state was assumed to be known with variance 1 in each component, that is, $x_{0,i} \sim N(0, 1)$ for $k = 1, 2, 3$. Because no prior information was available on the measurement noise variance, a rather diffuse prior $\sigma_0^2 \sim \text{Inv-Gamma}(1, 1)$ was used.

The simulated data is shown in the Fig. 1. In the simulation the measurement noise first had the variance 0.2 (see Fig. 2). Around time $t = 100$ the variance quickly increased to 1 and then around $t = 200$ it again quickly decreased back to the value 0.2.

The following methods were tested:

- *Variational Bayesian adaptive Kalman filter (VB-AKF)* with the variance decreasing factor $\rho = 1 - \exp(-4)$. The number of fixed point iterations per time step was $N = 2$.
- *Interacting multiple models (IMM)* [3], [9], where a separate mode was used for each variance level 0.1 ... 1.2 in steps of 0.01 (total of 111 models). The mode transition probabilities were chosen such that the probability of switching from mode i to j was proportional to $p_{ij} = \exp(-\lambda \Delta t |i - j|)$ with $\lambda = 5$.
- *Kalman filter (KF)* [1] for each noise level 0.1 ... 1.2 separately, in steps of 0.01.

Note that none of the dynamic models is exactly matched to the true simulated variance dynamics. Instead, the models are selected to represent sufficiently smooth changes, which approximate the true dynamics of the variance.

The result of variance estimation, when VB-AKF and IMM methods were applied to the simulated data is shown in the Fig. 2. In the figure, the actual variance and the estimated variance (posterior means) of

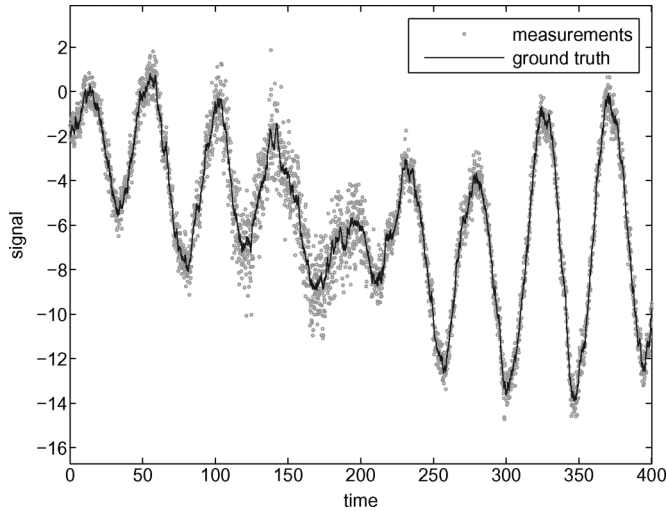


Fig. 1. Simulated noisy resonator model. The *ground truth* is the actual simulated position and the *measurements* are the noisy measurements obtained from it once per 1/10 seconds.

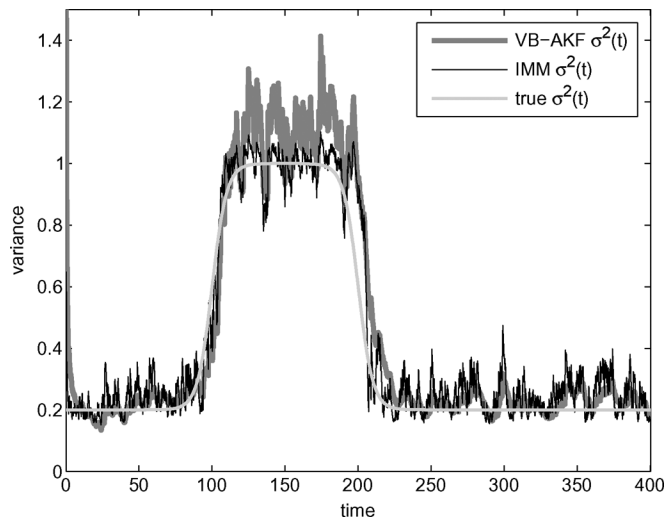


Fig. 2. Results of tracking the variance parameter of noisy resonator by VB-AKF and IMM.

VB-AKF and IMM estimates are shown. It can be seen that both the estimates follow the true variance trajectory quite well. IMM does seem to follow the trajectory a bit better than VB-AKF. But the performance of IMM comes with penalty of CPU time, because the CPU time needed for IMM was at least 100 times the CPU time of VB-AKF.

The root mean squared errors (RMSE) of VB-AKF and IMM are both lower than of any standard Kalman filters without the noise adaptivity. This can be seen from the Fig. 3 where the RMSE results of Kalman filters with noise variance 0.1, ..., 1.2 are shown together with the errors of VB-AKF and IMM. None of the Kalman filters has as low RMSE error as the VB-AKF or IMM. The CPU time of single KF was approximately half of the CPU time of VB-AKF.

As can be seen from the RMSE in Fig. 3 and variance trajectory in Fig. 2, the IMM algorithm performs a bit better than VB-AKF. This is natural, because IMM forms a grid approximation to the variance distribution and forms Gaussian state distributions conditionally to each of these grid values. In the limit of infinite number of grid points the

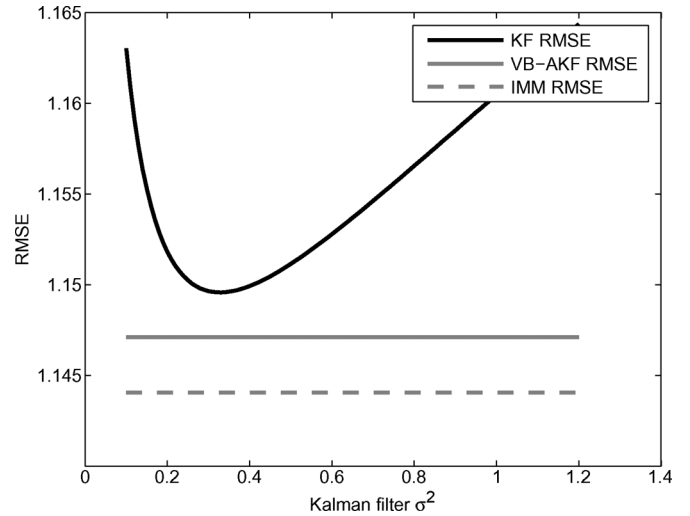


Fig. 3. RMSE errors of Kalman filters with different noise variances, and the RMSE errors of VB-AKF and IMM. It can be seen that both VB-AKF and IMM give lower RMSE errors than any of the standard Kalman filters.

estimate becomes exact and thus it is likely to be good approximation with a sufficiently large number of grid points—but with cost of high CPU time. The CPU time needed by IMM also grows exponentially in the number of unknown variances and thus IMM can practically be applied only to models with very low dimensional measurements.

IV. DISCUSSION

Although, the VB-AKF method presented in this article is a Bayesian method for noise identification, the algorithm much resembles the covariance matching methods presented in, for example, [8], [10]. This is because the algorithm essentially estimates noise parameters such that the parameters become consistent with the observed residual, which is also the basis of covariance matching methods.

Because the algorithm presented in this article contains explicit Kalman filter prediction and update steps as parts of the algorithm, the extension to non-linear filtering problems can be done by replacing them with, for example, the corresponding extended Kalman filter (EKF) or unscented Kalman filter (UKF) equations [2]–[5]. The update equations of the parameters $\beta_{k,i}$ can also be easily approximated with EKF and UKF type of approximations.

V. CONCLUSION

In this article, we have presented a new adaptive Kalman filtering algorithm, which is based on recursively forming separable approximations to the joint distribution of state and noise parameters by variational Bayesian methods. The performance of the algorithm has been demonstrated in a simulated application.

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Attitude Tracking and Disturbance Rejection of Rigid Spacecraft by Adaptive Control

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Abstract—In this technical note, we deal with the attitude tracking and disturbance rejection problem of spacecraft for a class of persistent disturbances with unbounded energy which include the sinusoidal disturbance as a special case. The approach involves the integration of techniques from robust control, adaptive control, and robust output regulation theory.

Index Terms—Adaptive control, nonlinear systems, space vehicle control.

I. INTRODUCTION

ATTITUDE control of spacecraft systems has been a benchmark control problem and has been extensively studied under various assumptions and scenarios [1], [6], [12], [17], [20], [21]. An important control method for disturbance attenuation of spacecraft systems is the \mathcal{H}_∞ control method. Most \mathcal{H}_∞ based methods can only attenuate the effect of disturbance on the output, quantified by \mathcal{L}_2 gain, to some degree. An exception was given recently in [17] where the authors employed the so-called inverse optimal adaptive control method to achieve attitude tracking. A distinct feature of their approach as opposed to others is that their control law allows the \mathcal{L}_2 gain of the closed-loop system to be chosen arbitrarily small so as to achieve any level of \mathcal{L}_2 disturbance attenuation. Moreover, for the special class of disturbances with *bounded energy*, the approach given in [17] can achieve exact asymptotic disturbance rejection. It should be noted that the inverse optimal adaptive control method was first developed in [14] without considering the external disturbances and plant uncertainties.

For disturbances with unbounded energy, the \mathcal{H}_∞ control method can only achieve *disturbance attenuation* (see, e.g., [12], [17], [20]) rather than *disturbance rejection*. Therefore, in this note, we will further consider a more interesting attitude control problem where a spacecraft involves mass parameter uncertainties and is subject to a class of external disturbances which is a sum of finitely many sinusoidal functions with unknown amplitudes and phase angles. As this class of disturbances has *unbounded energy*, the existing approaches, e.g., [12], [17], [20], cannot achieve exact asymptotic disturbance rejection. Thus, we need to develop a new methodology to deal with the disturbance rejection problem. The rest of the note is organized as follows. The attitude control and disturbance rejection problem of the spacecraft is formulated in Section II. The complete motion equation is called System A ((1)–(4)). In order to convert the problem to a more tractable problem, we need to perform two major transformations. First, we make use of the results in [17] and [22] to simplify our problem into a regulation

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