



Xi'an Jiaotong-Liverpool University  
西交利物浦大學

# EEE220 Instrumentation and Control System

*2018-19 Semester 2*

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# Lecture 22

# Outline

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## **Frequency Response Methods**

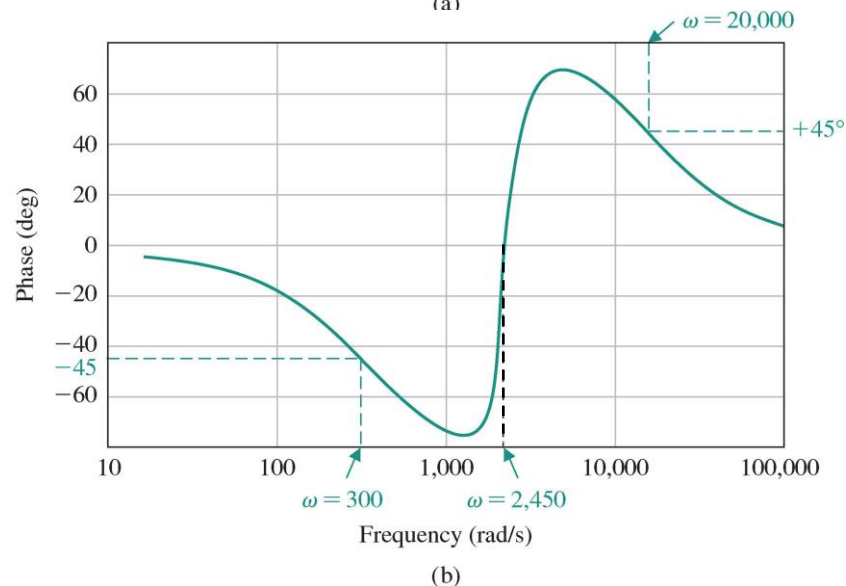
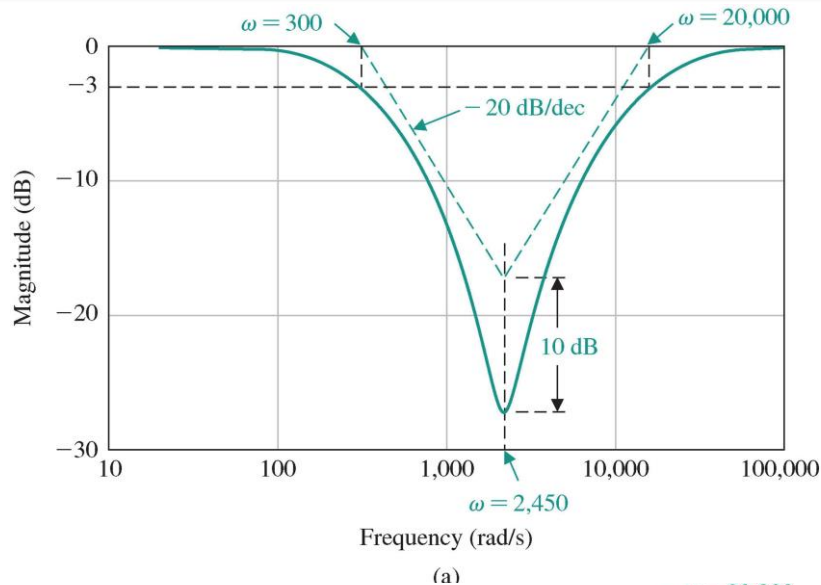
- ☐ Introduction
- ☐ Frequency Response Plots – Polar Plot & Bode Plot
- ☐ **Frequency Response Measurements**
- ☐ **Performance Specifications in the Frequency Domain**
- ☐ **Gain Margin and Phase Margin**
- ☐ **Compensators**
- ☐ **Frequency Response Methods Using Matlab**

# Frequency Response Measurement

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- A sine wave can be used to measure the open-loop frequency response of a system. In practice, a plot of amplitude versus frequency and phase versus frequency will be obtained. From these two plots, the transfer function (including both the loop TF and the closed-loop TF) of the system can be deduced.
- A device called a **wave analyzer** can be used to measure the amplitude and phase variations as the frequency of the input sine wave is altered. Also, a device called a **transfer function analyzer** can be used to measure the loop transfer function and closed-loop transfer function.
- A typical signal analyzer instrument can perform frequency response measurement from DC to 100 kHz. Built-in analysis and modeling capabilities can derive poles and zeros from measured frequency responses or construct phase and magnitude responses from user-supplied models. This device can also synthesize the frequency response of a model of a system, allowing a comparison with an actual response.

# Example 22.1

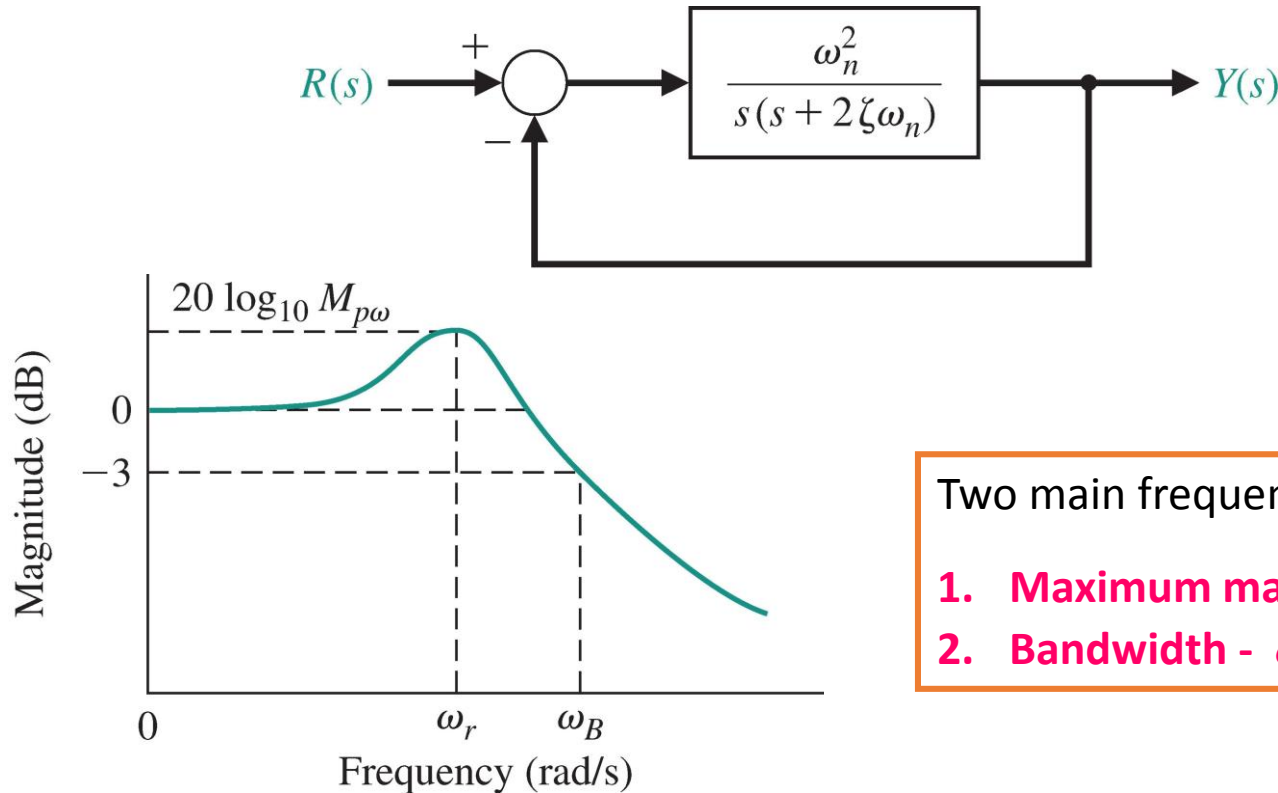


- The magnitude declines at about -20 dB/decade as  $\omega$  increase between 100 and 1000; and because the phase is  $-45^\circ$  and the magnitude is -3 dB at 300 rad/s  $\rightarrow$  there must be a pole at  $p_1 = 300$ ;
- Phase abruptly increases by nearly  $180^\circ$  and passing  $0^\circ$  at 2450 rad/s; also, the slope of the magnitude changes from -20 dB/decade to 20 dB/decade at 2450 rad/s  $\rightarrow$  there must be a pair of quadratic zeros existing at  $\omega_n = 2450$ ;
- Difference in the corner frequency of the asymptotes to the minimal response is 10 dB  $\rightarrow \zeta = 0.16$ .
- Slope of magnitude returns to 0 dB/decade as  $\omega$  exceeds 50000; specifically, magnitude is -3 dB and phase is  $45^\circ$  at 20000 rad/s  $\rightarrow$  there must be a second pole at  $p_2 = 20000$ .

$$T(s) = \frac{(s/2450)^2 + (0.32/2450)s + 1}{(s/300 + 1)(s/20000 + 1)}$$

# Performance Specifications in the Frequency Domain

- ❑ For a second-order system with a pair of complex poles:



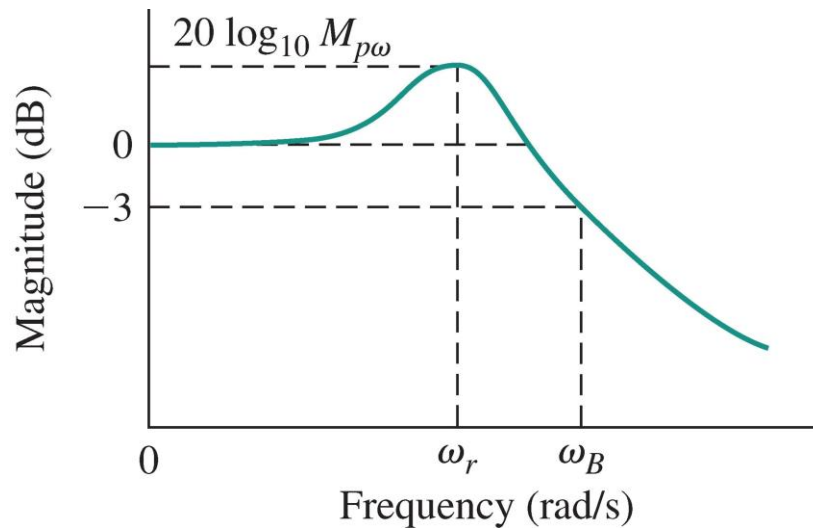
Two main frequency specifications:

1. **Maximum magnitude -  $M_{p\omega}$**
2. **Bandwidth -  $\omega_B$**

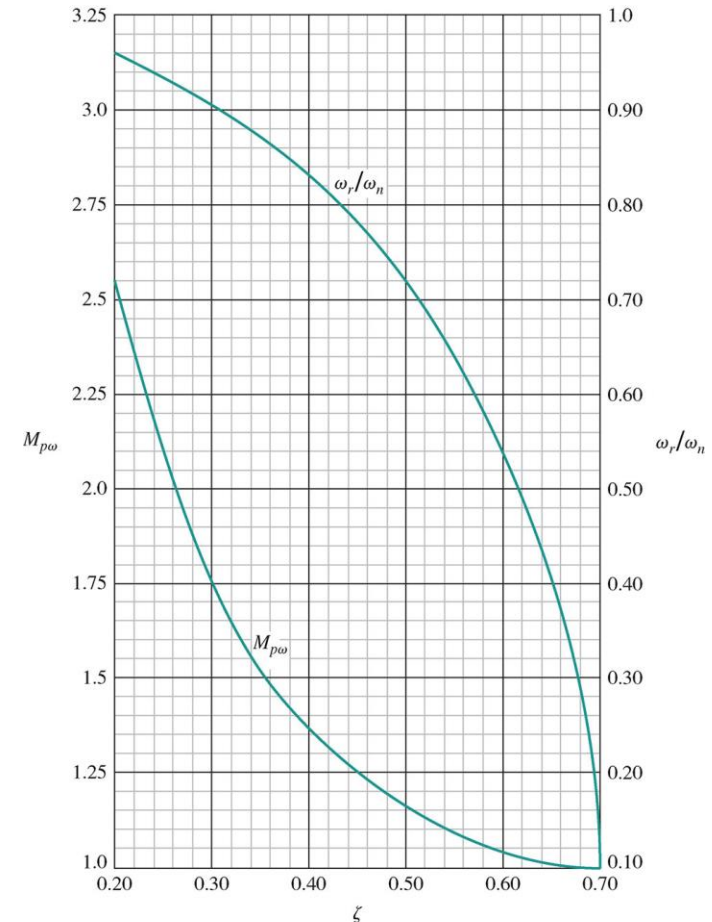
- ❑ For a higher-order system, if the frequency response is **dominated** by a pair of complex poles, the relationship between frequency response and time response will be valid.

# Maximum Magnitude

At the resonant frequency  $\omega_r$ , a maximum value  $M_{p\omega}$  of the frequency response is attained.



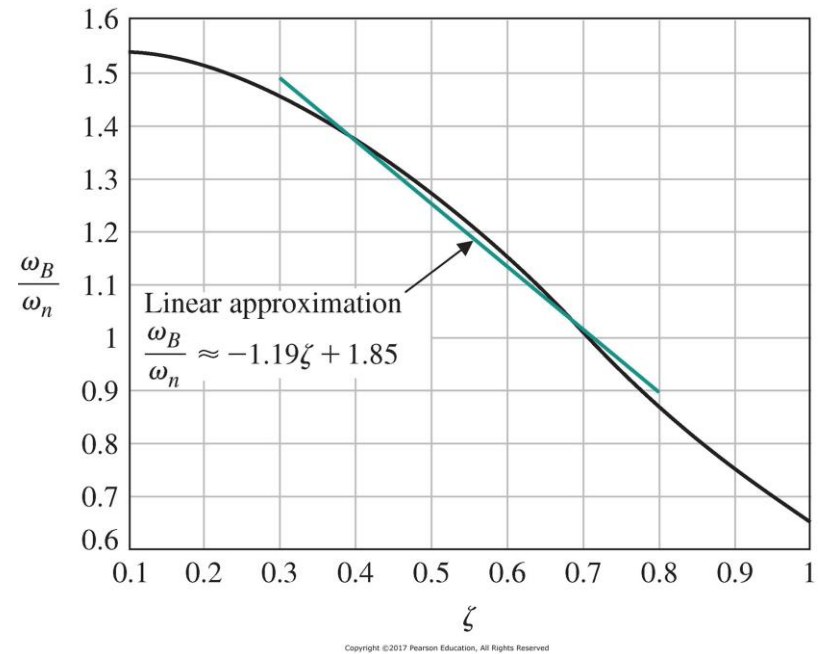
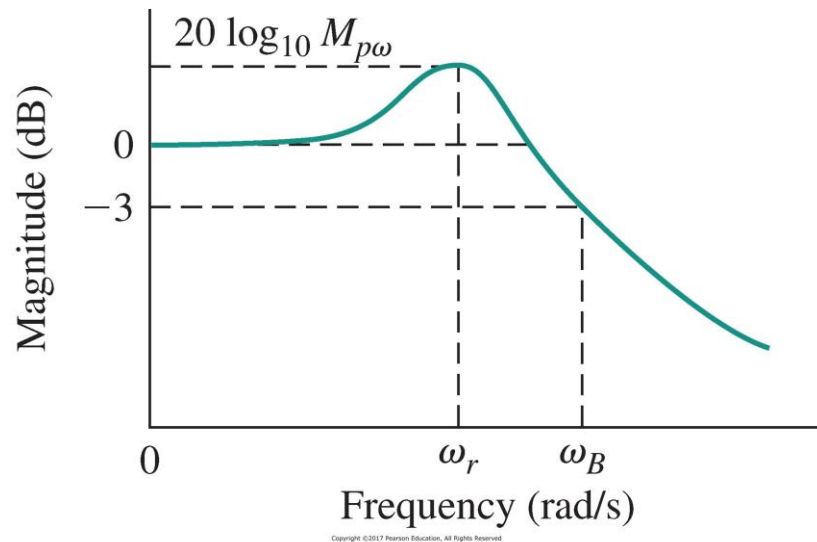
$$M_{p\omega} = |G(j\omega_r)| = (2\zeta\sqrt{1-\zeta^2})^{-1}, \quad \zeta < 0.707.$$



Maximum magnitude  $M_{p\omega} \uparrow \rightarrow$  Damping ratio  $\zeta \downarrow \rightarrow$  Percent Overshoot P.O.  $\uparrow$

# Bandwidth

The bandwidth is the frequency at which the frequency response has declined 3 dB from its low-frequency value.



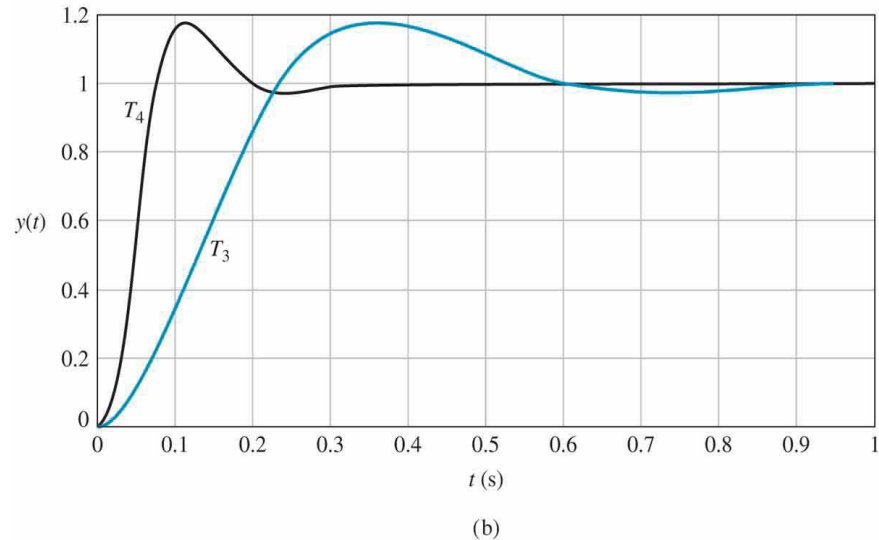
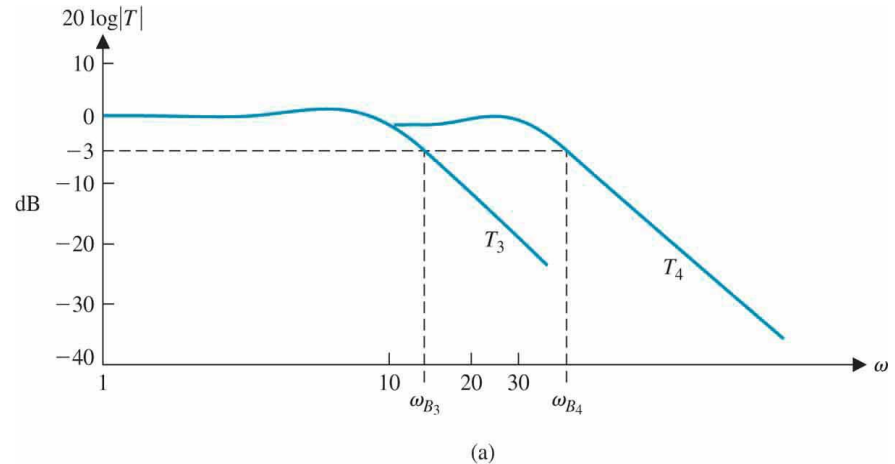
**Bandwidth  $\omega_B \uparrow$  (with a constant  $\zeta$ )  $\rightarrow$  Natural frequency  $\omega_n \uparrow \rightarrow$  Settling time  $T_s \downarrow$**



# Bandwidth vs. Settling Time

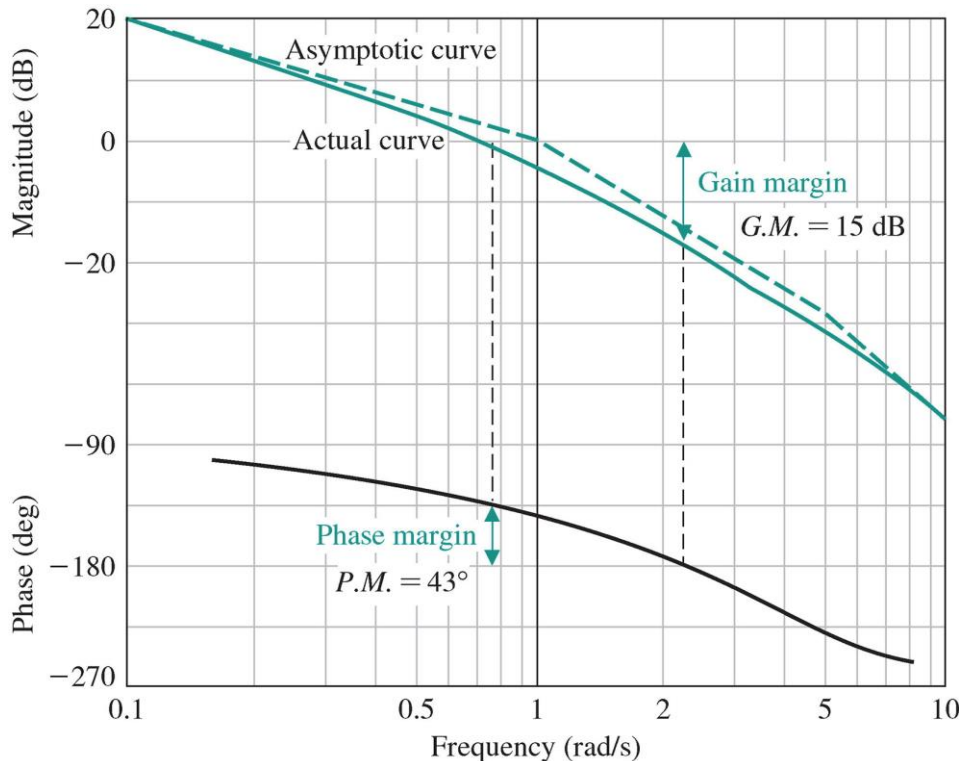
$$T_3(s) = \frac{100}{s^2 + 10s + 100}$$

$$T_4(s) = \frac{900}{s^2 + 30s + 900}$$



# Gain Margin and Phase Margin

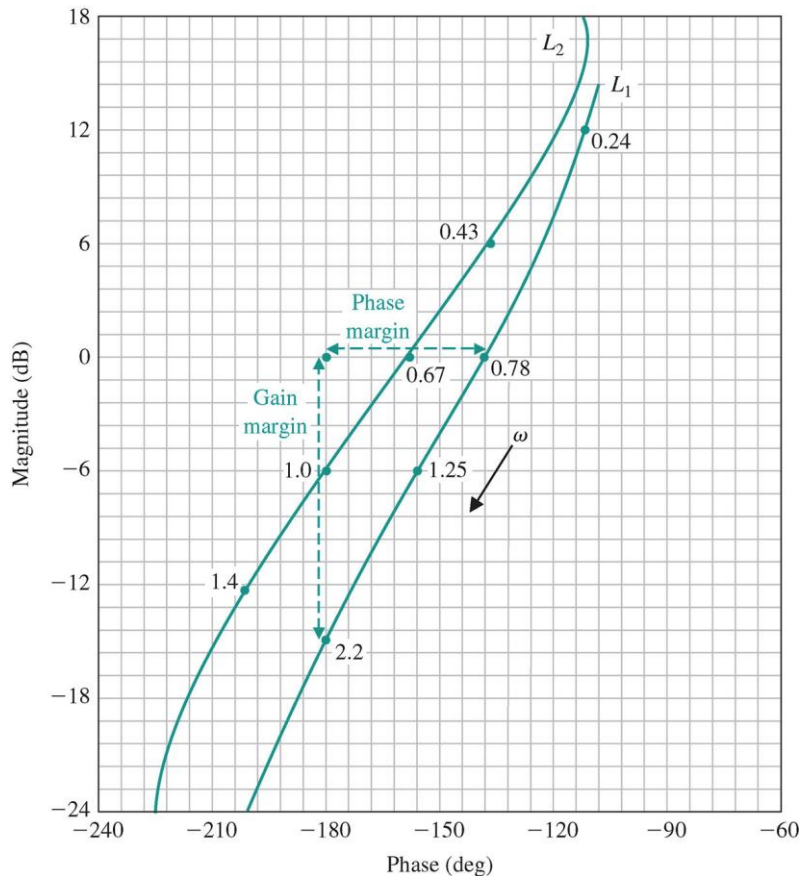
- For a closed-loop control system, with **loop transfer function**  $L(j\omega)$ , the characteristic equation is  $1 + L(j\omega) = 0 \rightarrow L(j\omega) = -1 + j0$ .
- It indicates that when  $|L(j\omega)| = 1$  (or  $20 \log(|L(j\omega)|) = 0$  dB) and  $\angle L(j\omega) = \pm 180^\circ$ , the closed-loop system is marginally stable.



- The gain margin GM is the increase in the gain of the  $L(j\omega)$  when phase =  $\pm 180^\circ$  that will result in a marginally stable system;
- The phase margin  $\phi_{pm}$  is the amount of phase shift of the  $L(j\omega)$  at unity magnitude (0 dB) that will result in a marginally stable system.

# Relative Stability

- $L_1(j\omega) = \frac{1}{j\omega(j\omega+1)(0.2j\omega+1)}$ , phase margin is  $43^\circ$ , gain margin is 15 dB
- $L_2(j\omega) = \frac{1}{j\omega(j\omega+1)^2}$ , phase margin is  $20^\circ$ , gain margin is 5.7 dB

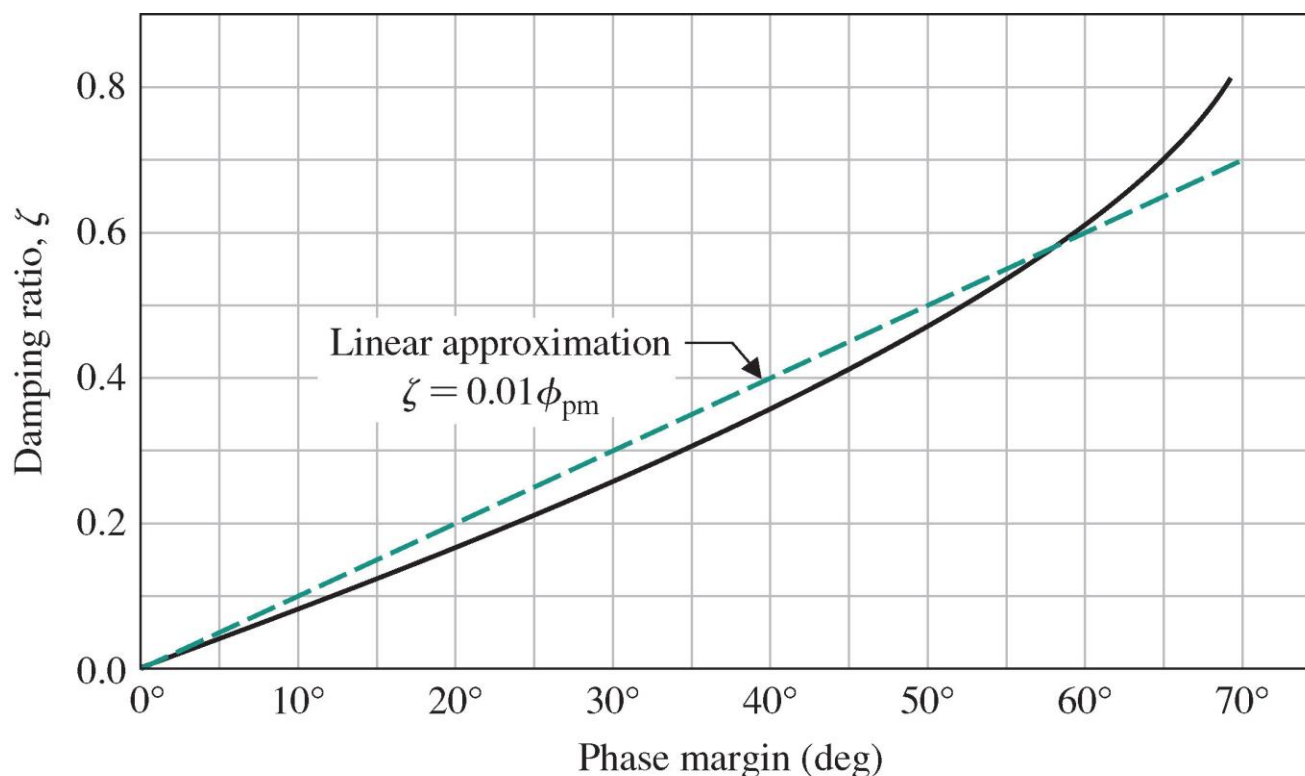


System 1 is MORE stable than system 2!

-- Gain margin or phase margin is actually a measure of **relative stability**.

# Phase Margin and Damping Ratio

$$\zeta = 0.01\phi_{pm}$$



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# Compensator

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- The performance of a feedback control system is of primary importance. A suitable control system is stable and result in an acceptable response to input commands, is less sensitive to system parameter changes, results in a minimum steady-state error for input commands, and is able to reduce the effect of undesirable disturbances.
- A feedback control system that provides an optimum performance without any necessary adjustments is rare.
- Thus, the design of a control system is concerned with the arrangement, or the plan, of the system structure and the selection of suitable components and parameters.
- The alteration or adjustment of a control system in order to provide a suitable performance is called compensation.
- A compensator is an additional component that is inserted into a control system to compensate for deficient performance.

# Compensator Structure

$$G_c(s) = \frac{K \prod_{i=1}^M (s + z_i)}{\prod_{j=1}^n (s + p_j)}$$

Compensator  $G_c(s)$  can be chosen to alter either the shape of the root locus or the frequency response.

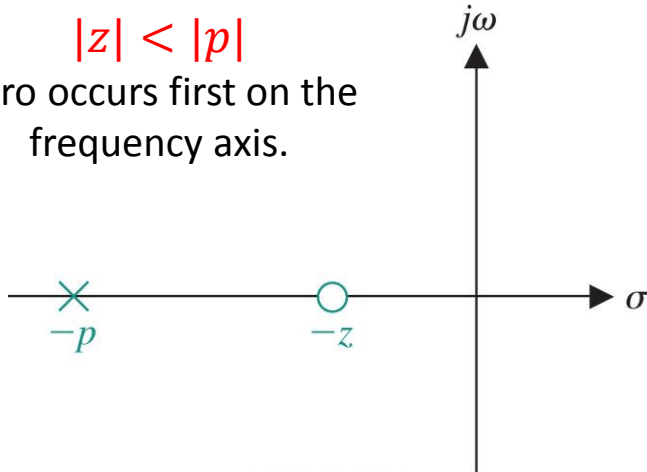
Compensator  $G_c(s)$  is used with a process  $G(s)$  so that the overall loop gain can be set to satisfy the steady state error requirement, and then  $G_c(s)$  is used to adjust the system dynamics favorably without affecting the steady state error.

First order compensator with the transfer function

$$G_c(s) = \frac{K(s+z)}{(s+p)}$$

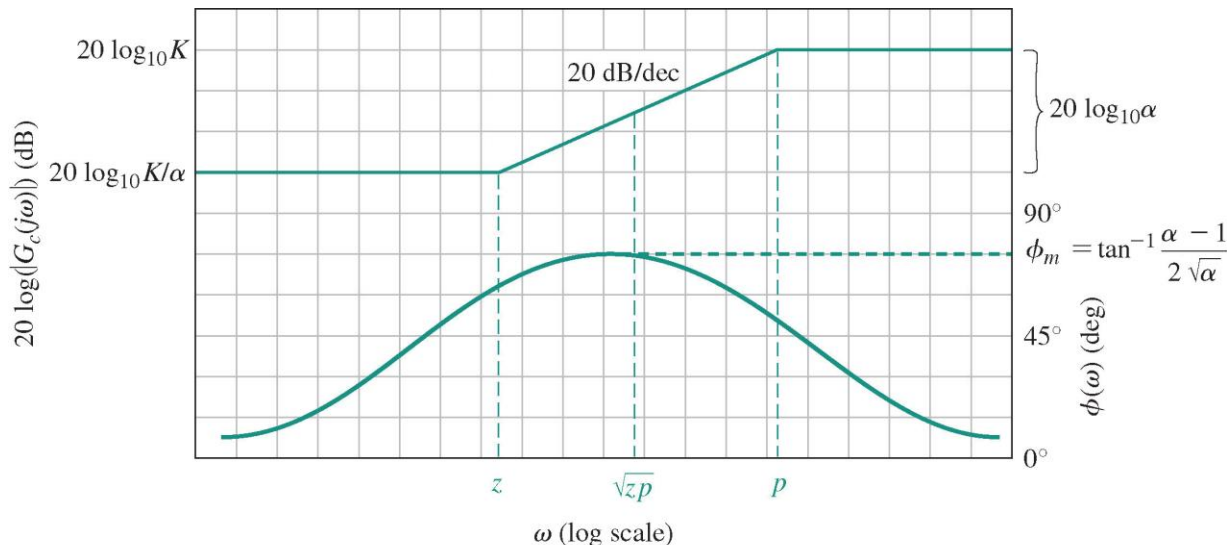
# Phase – Lead Compensator

$|z| < |p|$   
Zero occurs first on the frequency axis.



$$G_c(j\omega) = \frac{K(j\omega + z)}{(j\omega + p)} = \frac{K(1 + j\omega\alpha\tau)}{\alpha(1 + j\omega\tau)}$$

$$\tau = \frac{1}{p} \text{ and } \alpha = p/z > 1.$$



The maximum phase lead  $\phi_m$  occurs at  $\omega_m$ .

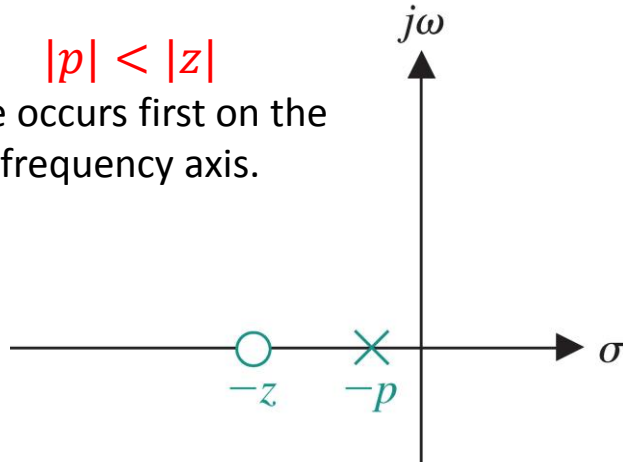
$$\omega_m = \sqrt{zp} = \frac{1}{\tau\sqrt{\alpha}}$$

$$\sin \phi_m = \frac{\alpha - 1}{\alpha + 1}$$

# Phase - Lag Compensator

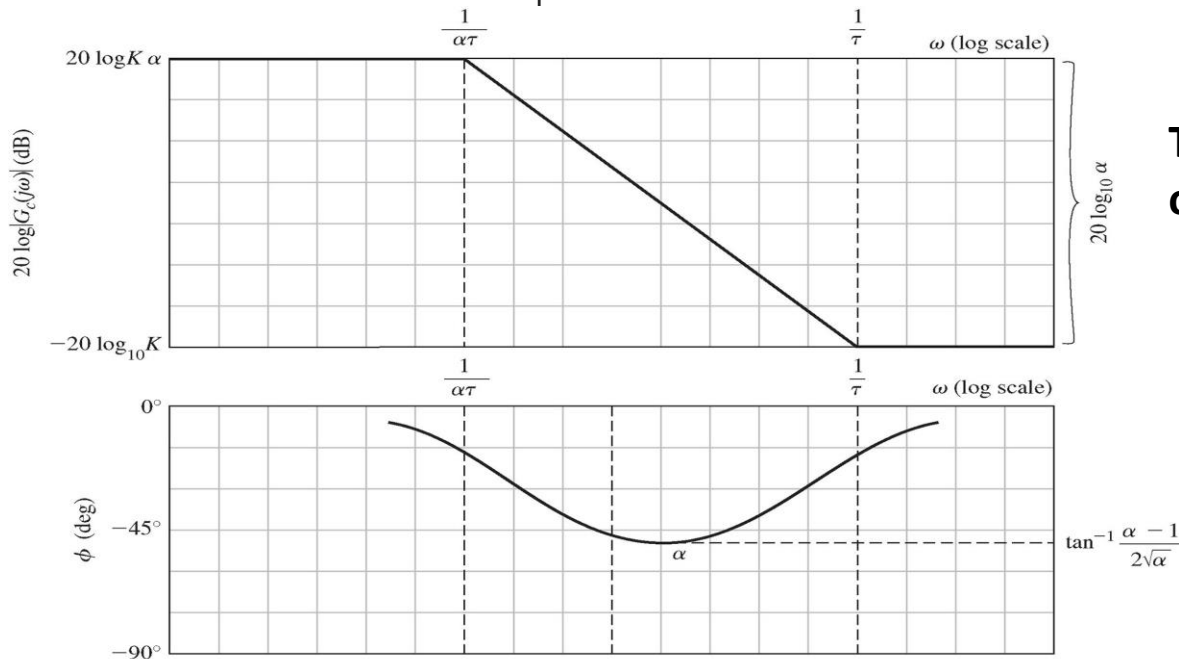
$$|p| < |z|$$

Pole occurs first on the frequency axis.



$$G_c(j\omega) = \frac{K(j\omega + z)}{(j\omega + p)} = \frac{K\alpha(1 + j\omega\tau)}{1 + j\omega\alpha\tau}$$

$$\tau = \frac{1}{z} \text{ and } \alpha = z/p > 1.$$



The maximum phase lag  $\phi_m$  occurs at  $\omega_m$ .

$$\omega_m = \sqrt{zp} = \frac{1}{\tau\sqrt{\alpha}}$$

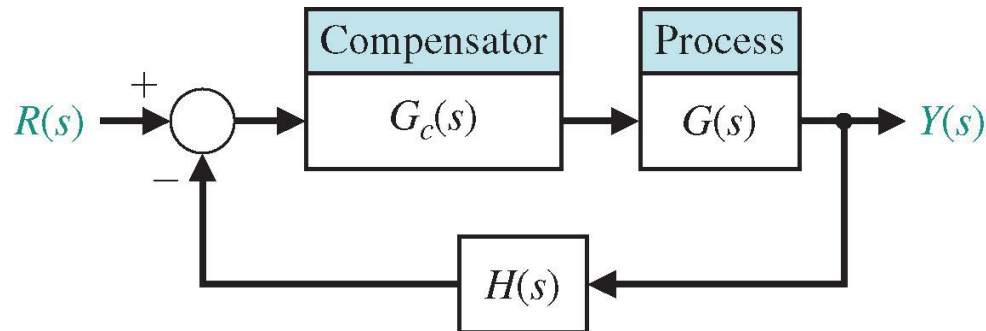
$$\sin \phi_m = \frac{\alpha - 1}{\alpha + 1}$$



# Example 22.2

Let's consider the below system, we assume  $G(s) = \frac{10}{s^2}$ , and  $H(s) = 1$ . It is clear that without compensator, the closed-loop system is marginally stable. Design the compensator to satisfy the following specifications using frequency response methods:

1. Settling time (with 2% criterion)  $T_s \leq 4s$ ;
2. System damping ratio  $\zeta \geq 0.45$ .



## Solutions:

### Step 1.

$$T_s = \frac{4}{\zeta \omega_n} = 4;$$
$$\omega_n = \frac{1}{\zeta} = \frac{1}{0.45} = 2.22.$$



$$\omega_B = (-1.19\zeta + 1.85)\omega_n = 3.00$$

→ The bandwidth of the closed-loop system should be larger than 3 rad/s.

## Step 2.

The phase margin of the system is required to be approximately

$$\phi_{\text{pm}} = \frac{\zeta}{0.01} = \frac{0.45}{0.01} = 45^\circ.$$

The phase margin of the uncompensated system ( $G(s)$ ) is  $0^\circ$  → The compensator needs to provide a phase-lead angle of at least  $45^\circ$  to the loop transfer function.

$$\frac{\alpha - 1}{\alpha + 1} = \sin \phi_m = \sin 45^\circ \quad \longrightarrow \quad \alpha = 5.8$$

We use  $\alpha = 6$ . The gain added to the system at the frequency  $\omega_m$  is

$$10 \log \alpha = 7.78 \text{ dB}$$

We want to have  $\omega_m$  equal to the compensated slope crossing the 0 dB axis, thus, the compensated crossover frequency is located by evaluating the frequency where the uncompensated magnitude curve is equal to -7.78 dB. So,

$$\omega_m = 4.95$$

### Step 3.

We can now determine the pole and zero of the compensator.

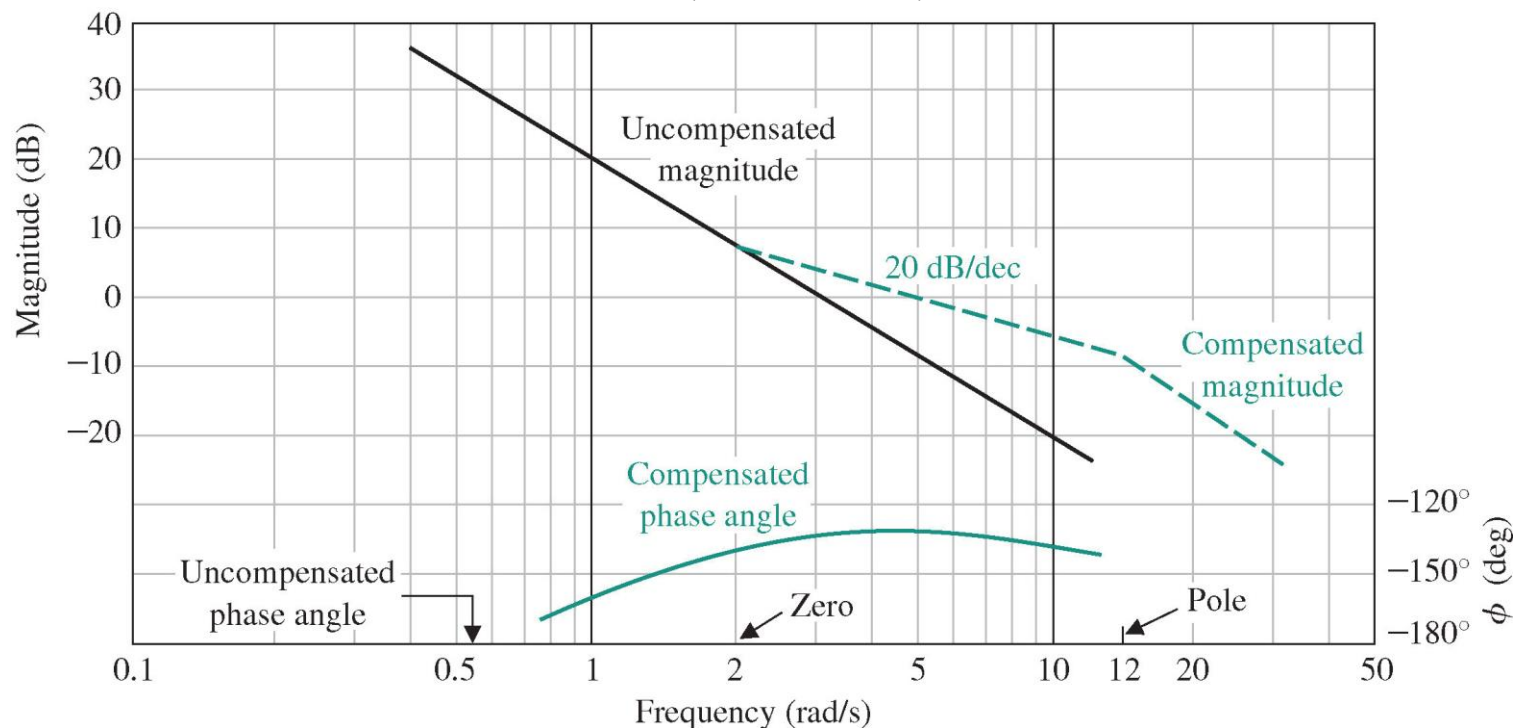
$$p = \omega_m \sqrt{\alpha} = 12.0$$

$$z = p/\alpha = 2.0$$

The compensator is

$$G_c(s) = \frac{K}{6} \frac{(1 + s/2.0)}{(1 + s/12.0)}$$

Choose  $K=6$  to keep the loop gain being still 10.



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#### Step 4.

Validate, and re-design if the specifications are not satisfied.

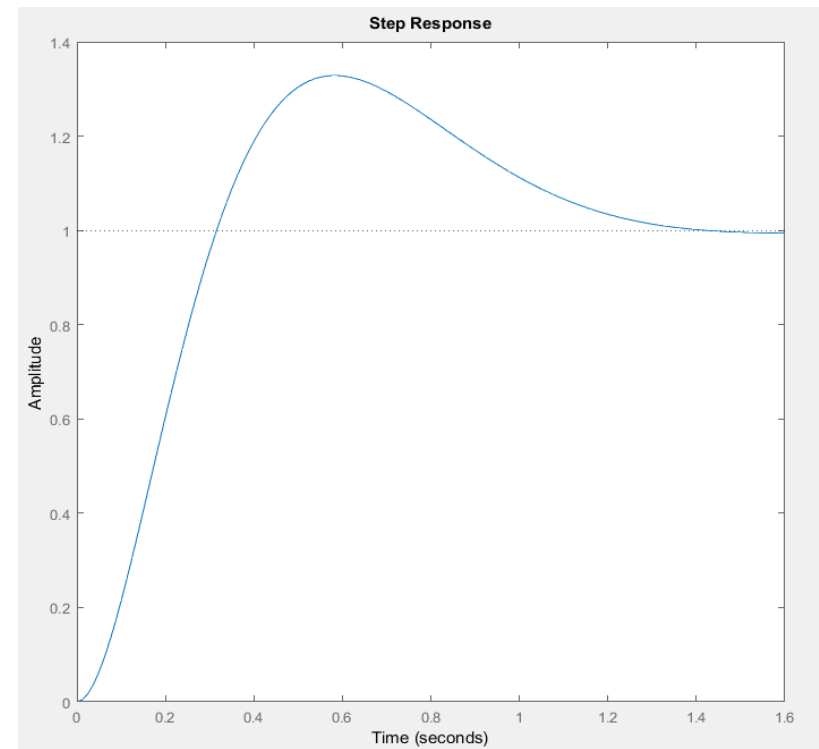
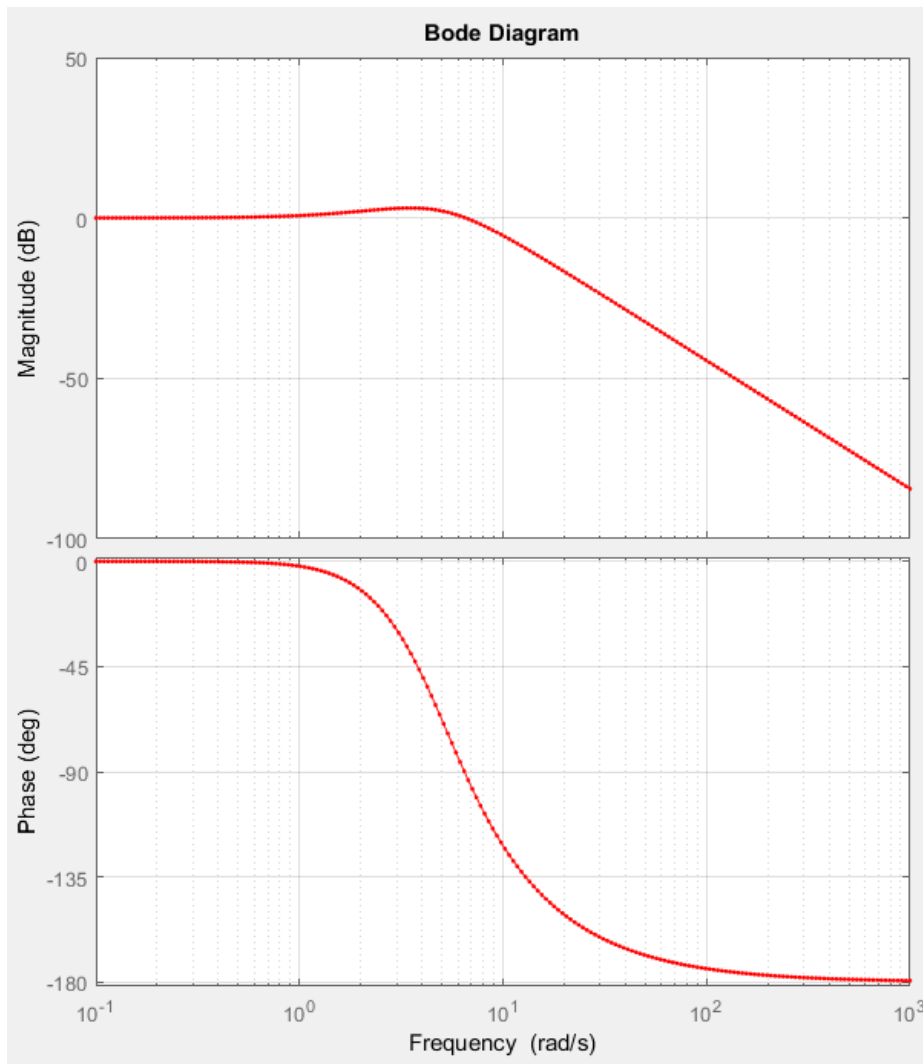
The total loop transfer function is

$$L(s) = \frac{10(1 + s/2)}{s^2(1 + s/12)} = \frac{60(s + 2)}{s^2(s + 12)}.$$

The closed-loop transfer function is

$$T(s) = \frac{60(s + 2)}{s^3 + 12s^2 + 60s + 120}$$

Obtain the bode plots for  $T(s)$  or step response of the system.

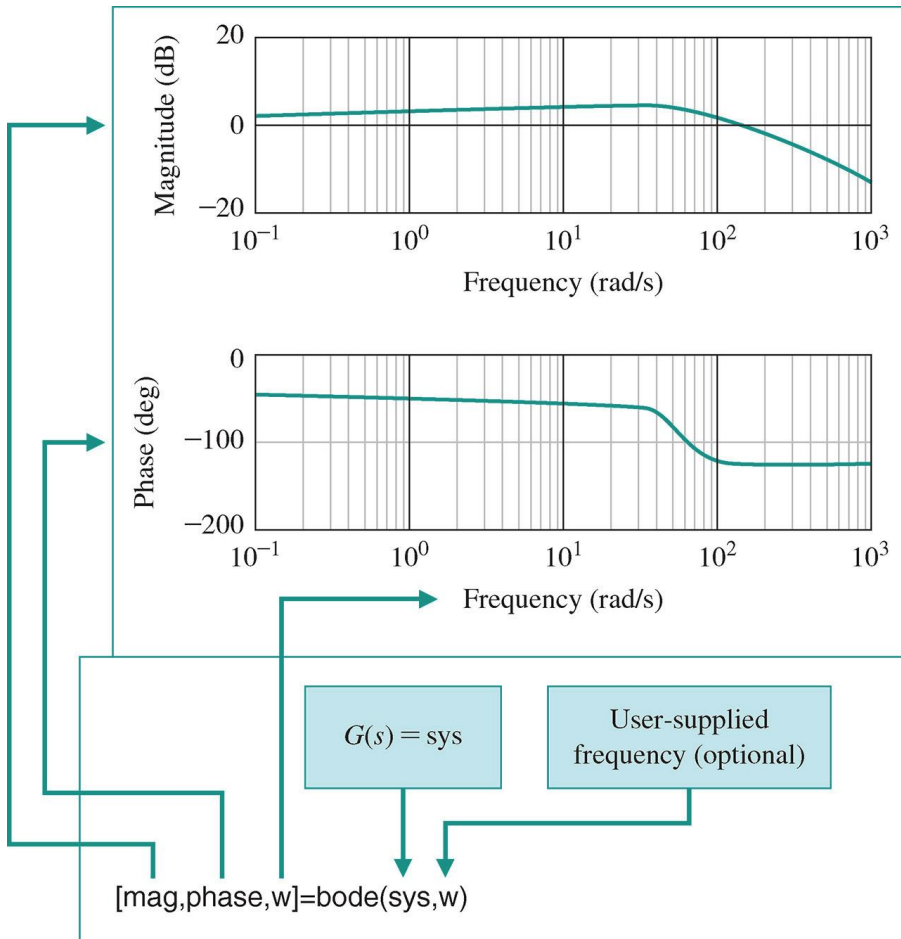


$$T_S = 1.3 \text{ s}; P.O. = 34\%;$$

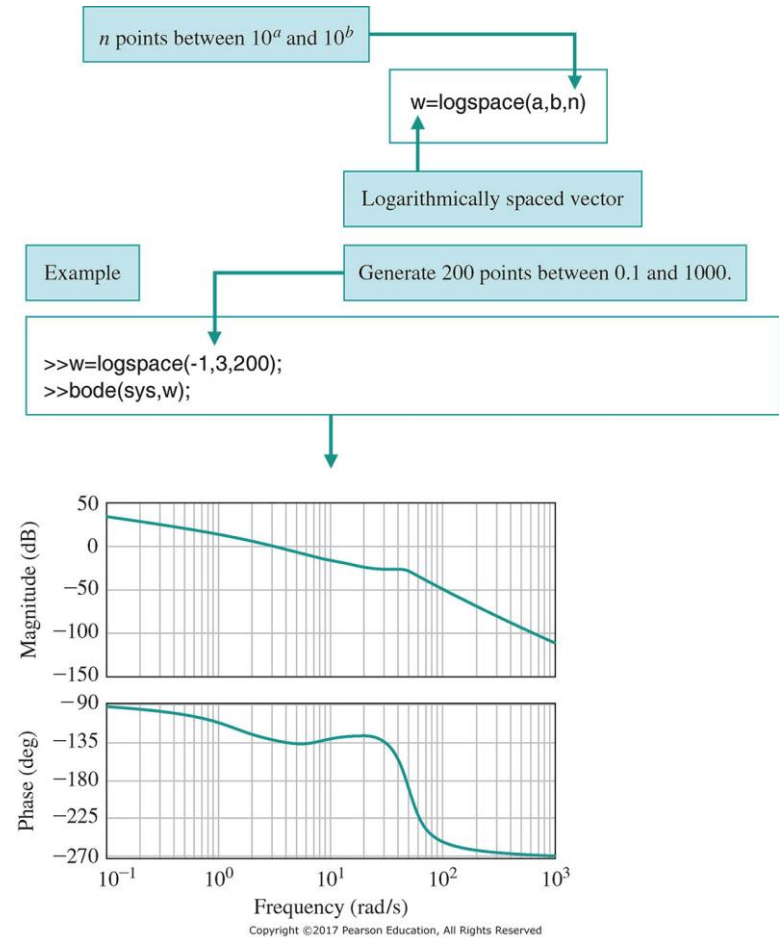
$$\omega_B = 8.4 \text{ rad/s}$$

→ The design specifications have been satisfied.

# Frequency Response using Matlab



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% Bode plot script for Figure 8.39

%

num=5\*[0.1 1];

f1=[1 0]; f2=[0.5 1]; f3=[1/2500 .6/50 1];

den=conv(f1,conv(f2,f3));

%

sys=tf(num,den);

bode(sys)

Compute

$$s(1 + 0.5s) \left( 1 + \frac{0.6}{50}s + \frac{1}{50^2}s^2 \right)$$

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bode(sys)

Transfer function model  
 $\text{sys} = \text{tf}(\text{num}, \text{den})$

State-space model  
 $\text{sys} = \text{ss}(\text{A}, \text{B}, \text{C}, \text{D})$

bode(sys)

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# Thank You !