

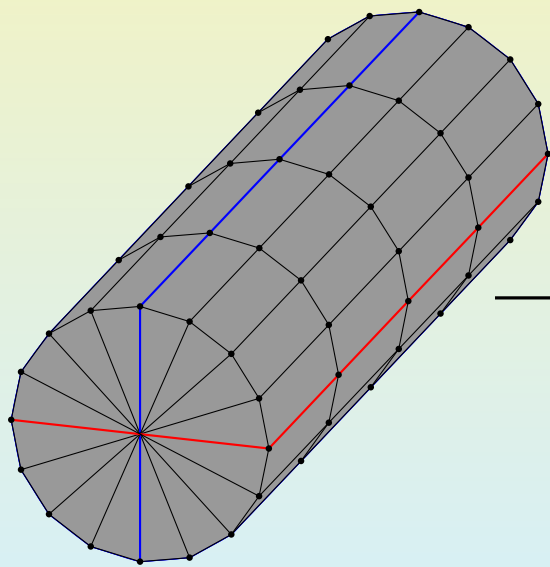
# Multi-Fidelity, Multi-Physics Analysis Degenerate Geometry

Rob McDonald – Cal Poly  
Joel Belben

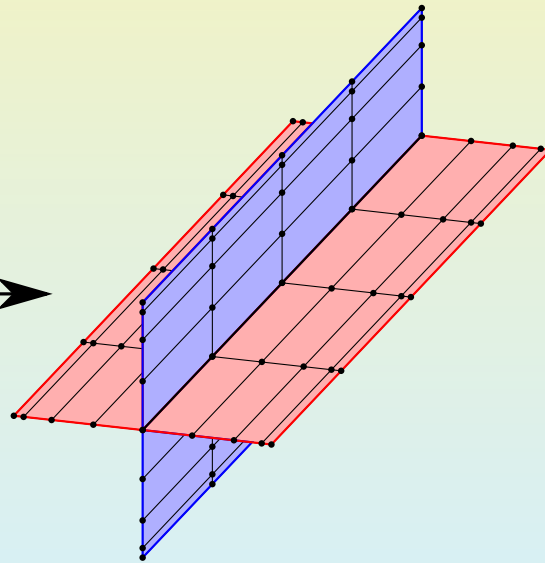
VSP Workshop  
August 8, 2013



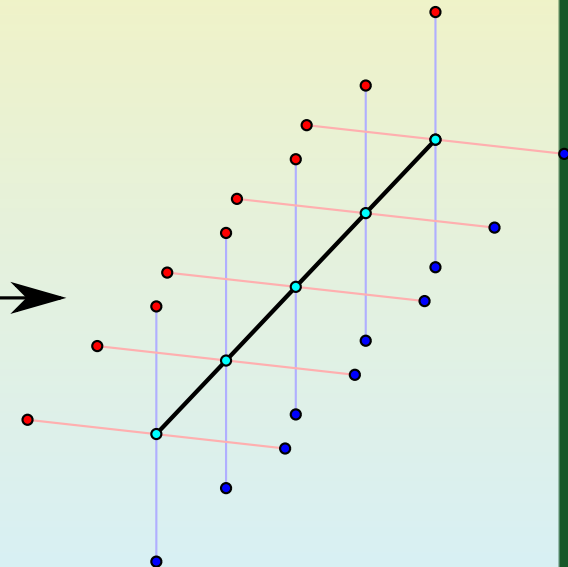
# Degenerate Geometry



Surface



Plate



Stick



# Degenerate Surface

## Surface

- Surface Node Locations
- Parametric  $u$  &  $w$
- Surface normal vectors

## Plate

- Plate Node locations
- Parametric  $u$  &  $w_{\text{top}}, w_{\text{bot}}$
- Plate normal vectors
- Camber surface height
- Camber normal vectors
- Thicknesses

## Stick

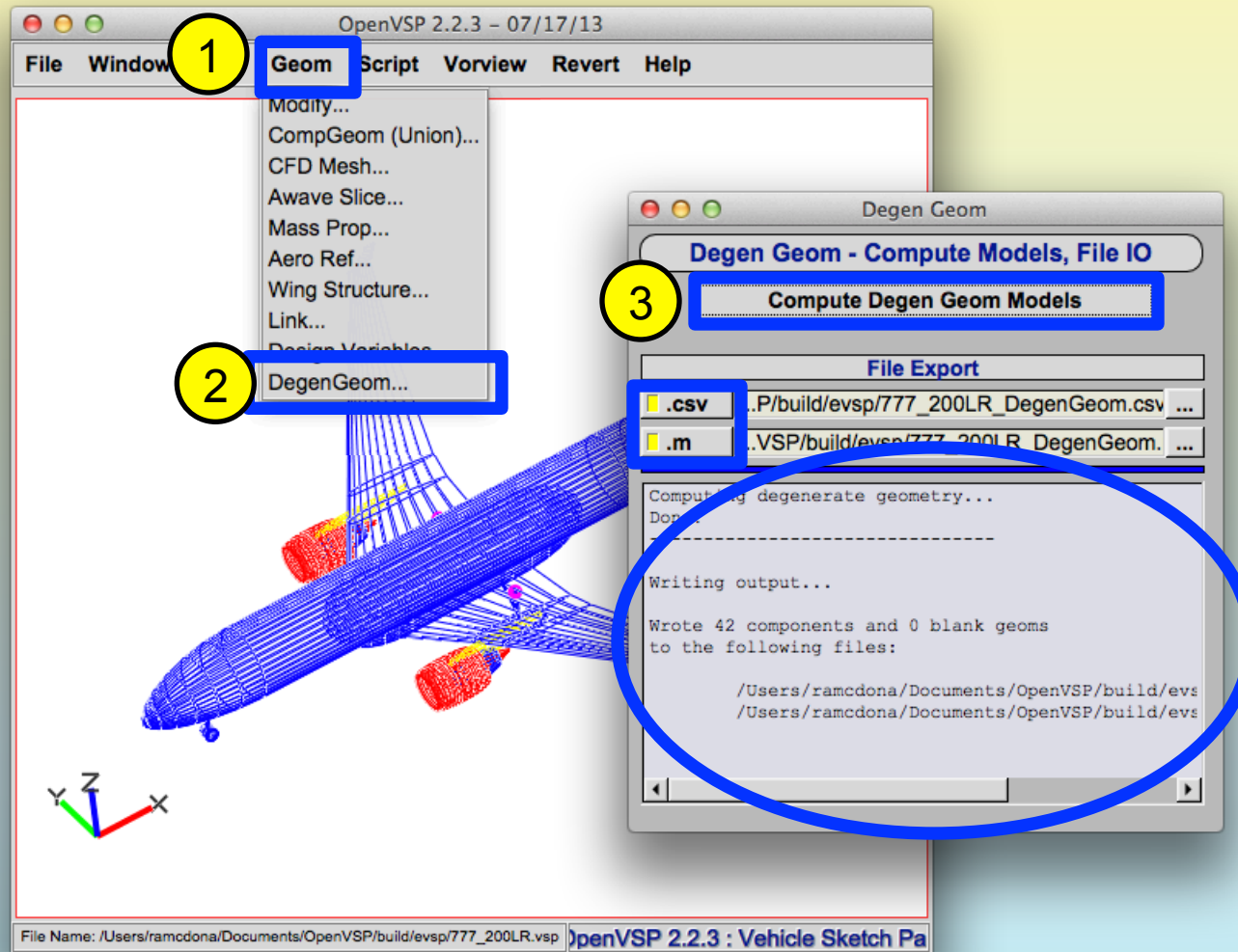
- LE & TE node locations
- Parametric  $u$
- Max  $t/c$  & location
- Chord
- Sweep
- Section line & area inertias & cg
- Area, perimeters

## Point

- Surface area & volume
- Wetted area & volume
- Shell & solid inertias & cg



# Creating Degenerate Geometry in VSP



# Output Formats

## CSV

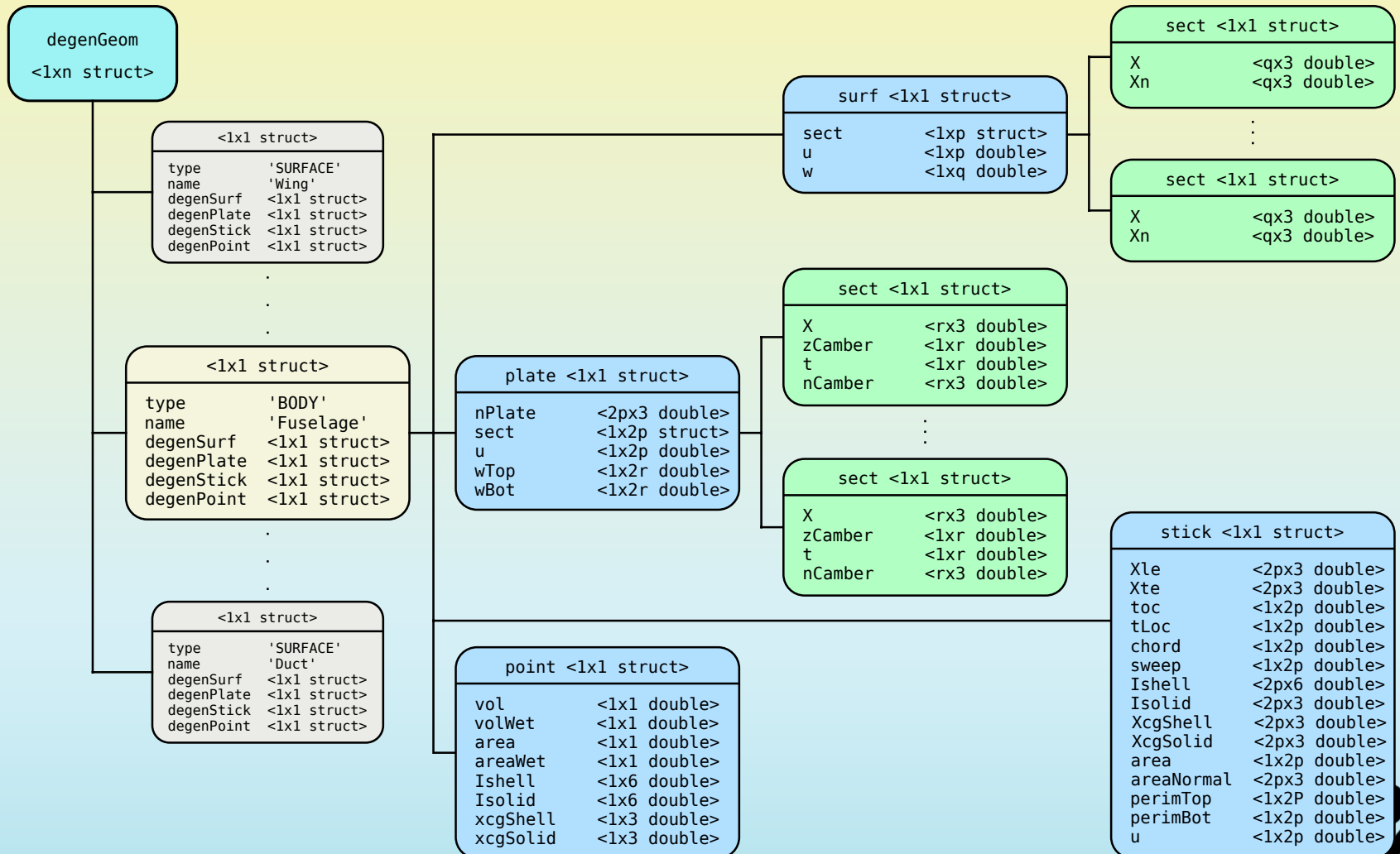
- Ready-made MS Excel file
- Easily human-readable (Comments)
- Easily parsed (C, Fortran, Java)

## Matlab

- Information direct to data structure
- Remove parsing barrier
- Design students in mind



# M-File Structure

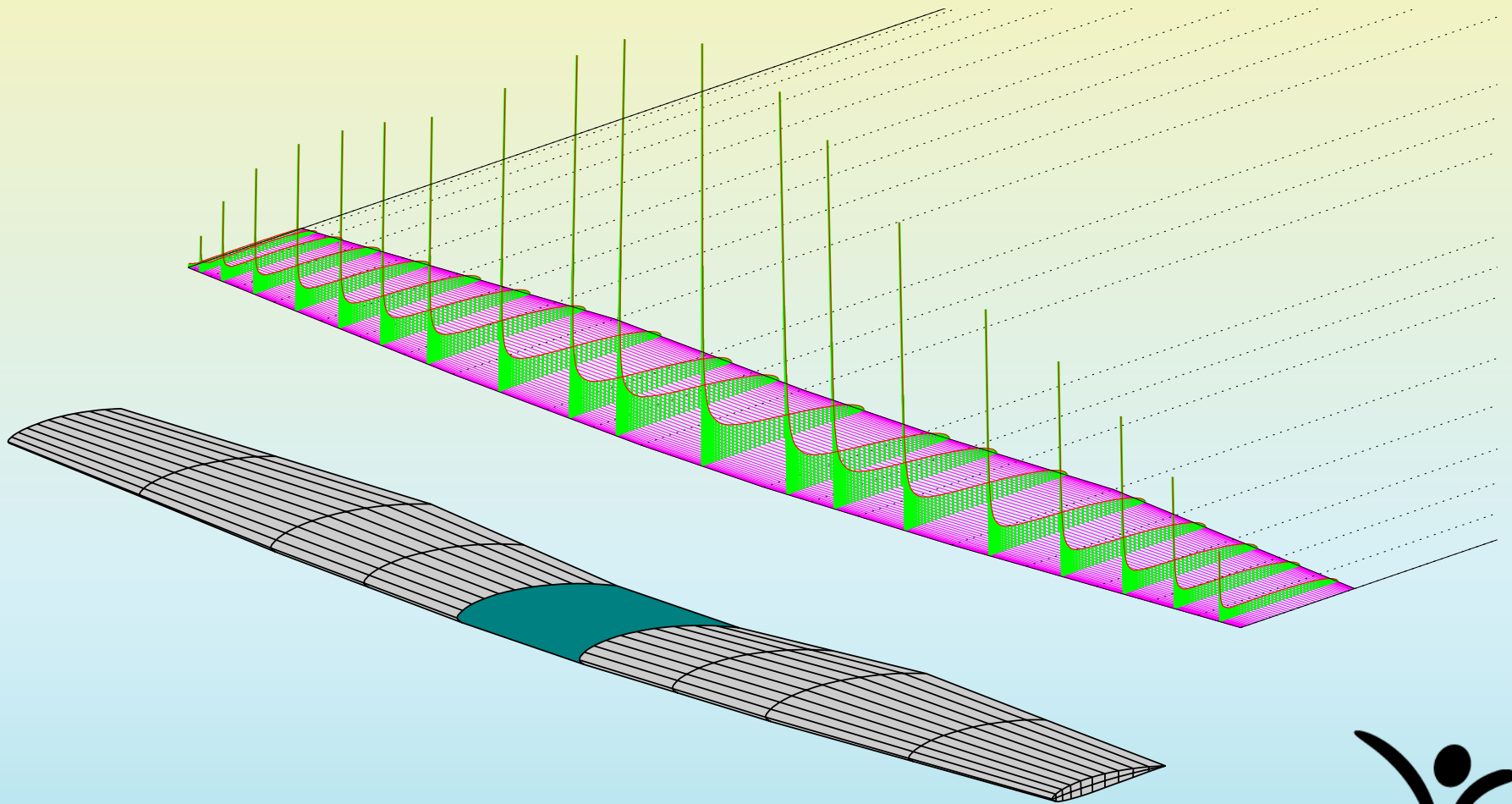


# Test Cases

- Aerodynamic
  - Lifting line theory
  - AVL (vortex lattice theory)
- Structural
  - Equivalent beam theory (sort of...)
  - ELAPS (equivalent plate theory)



# Test Cases: Vortex Lattice

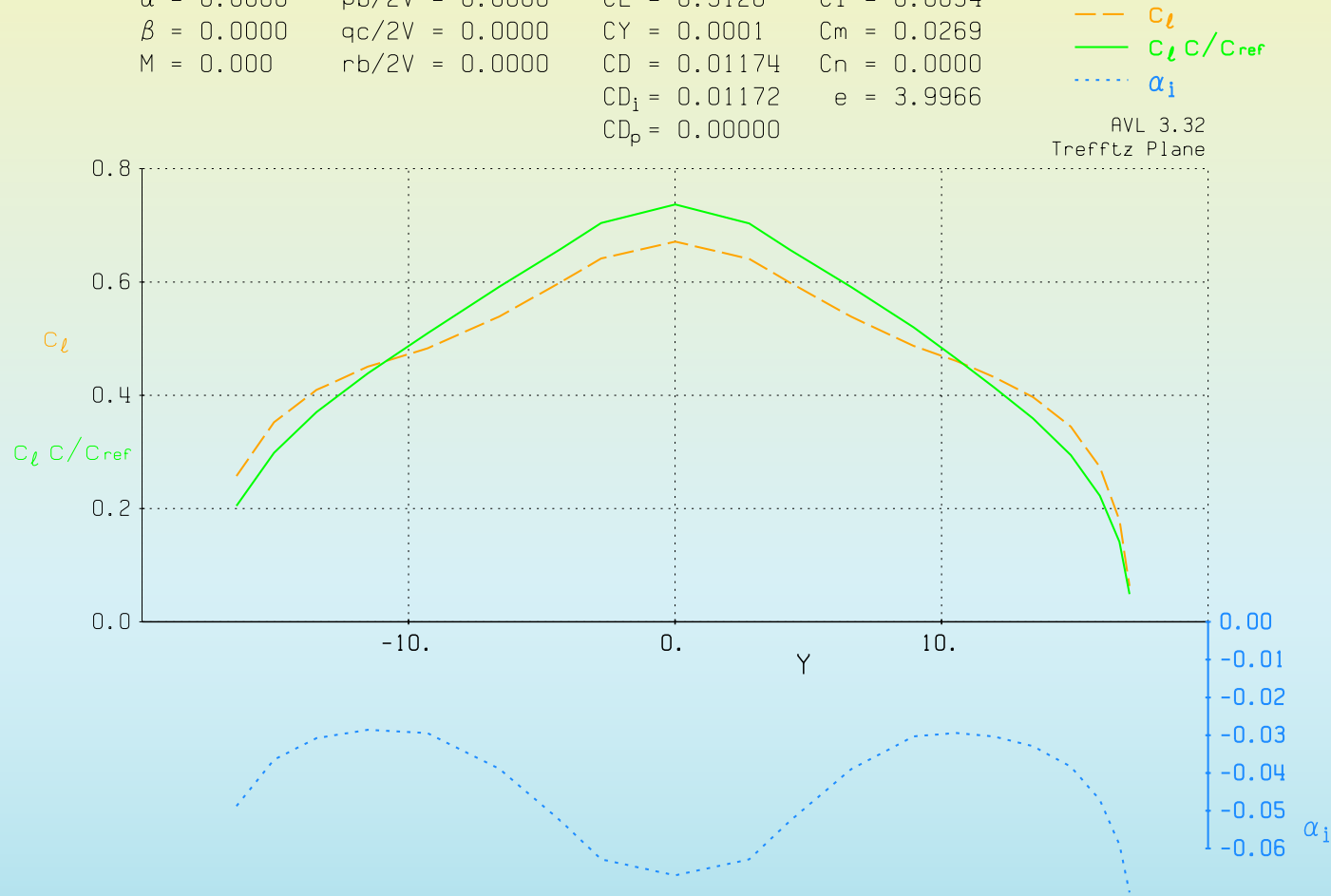




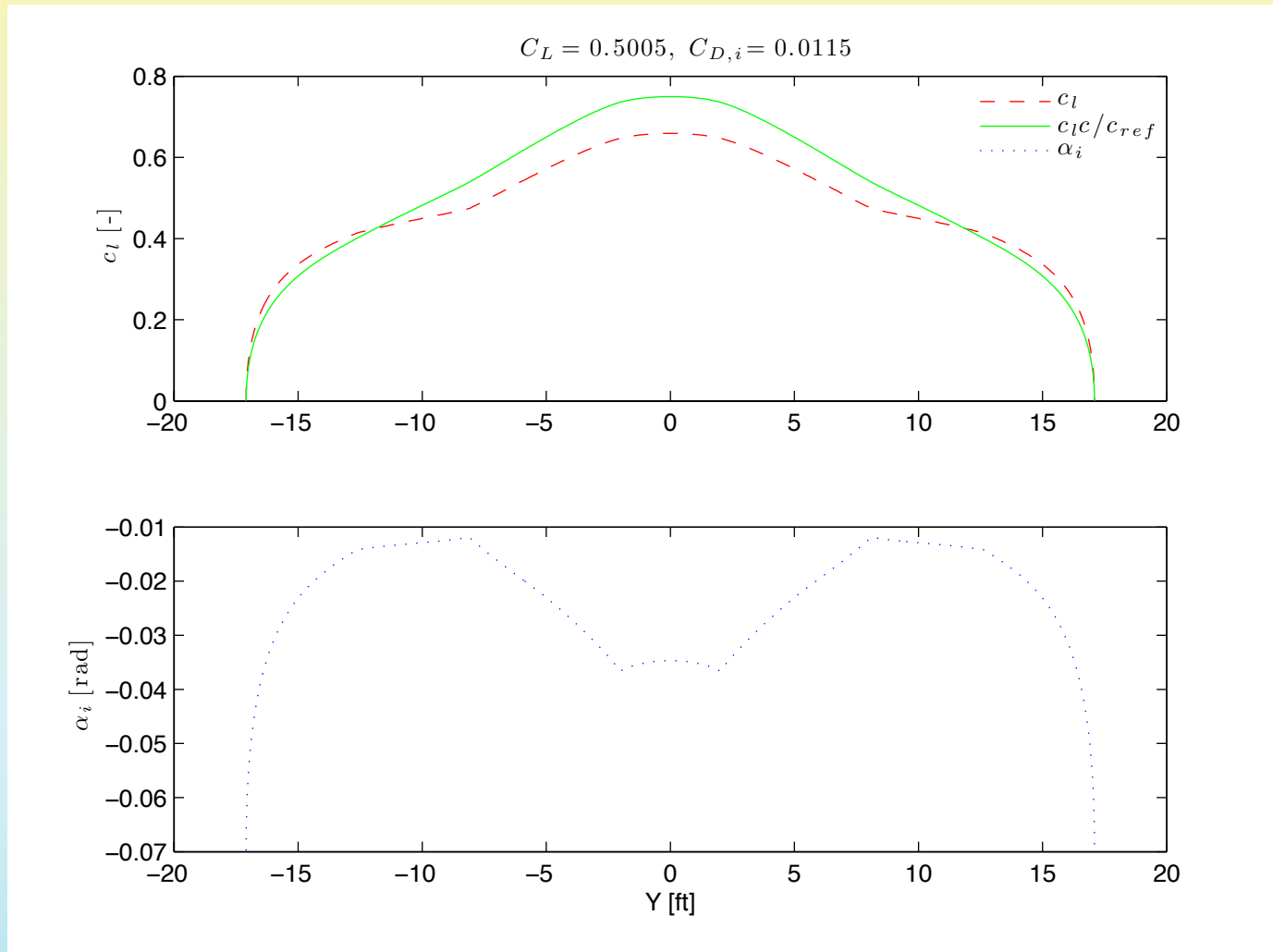
# Test Cases: Vortex Lattice (AVL)

Cessna Wing Test

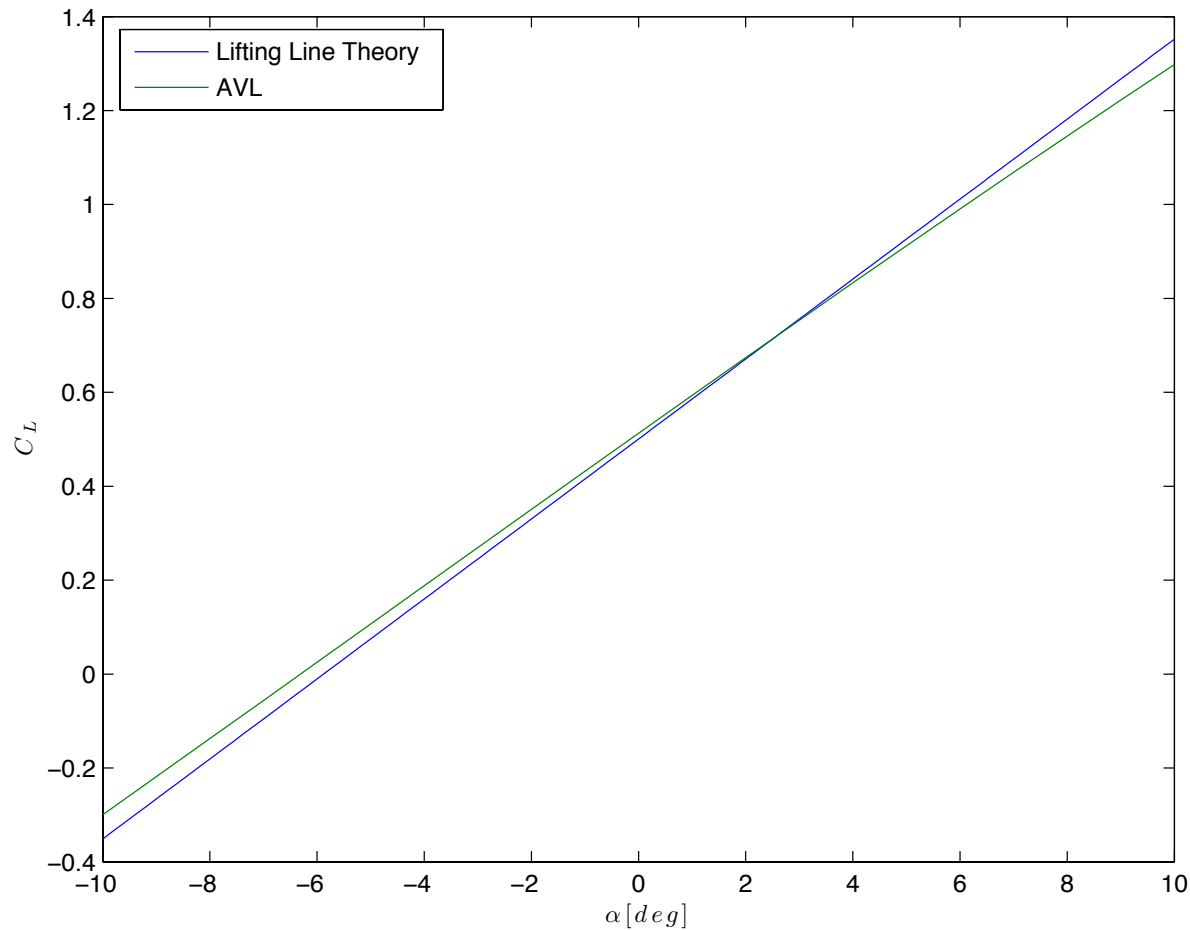
$\alpha = 0.0000$	$pb/2V = 0.0000$	$CL = 0.5128$	$Cl = 0.0034$
$\beta = 0.0000$	$qc/2V = 0.0000$	$CY = 0.0001$	$Cm = 0.0269$
$M = 0.000$	$rb/2V = 0.0000$	$CD = 0.01174$	$Cn = 0.0000$
		$CD_i = 0.01172$	$e = 3.9966$
		$CD_p = 0.00000$	



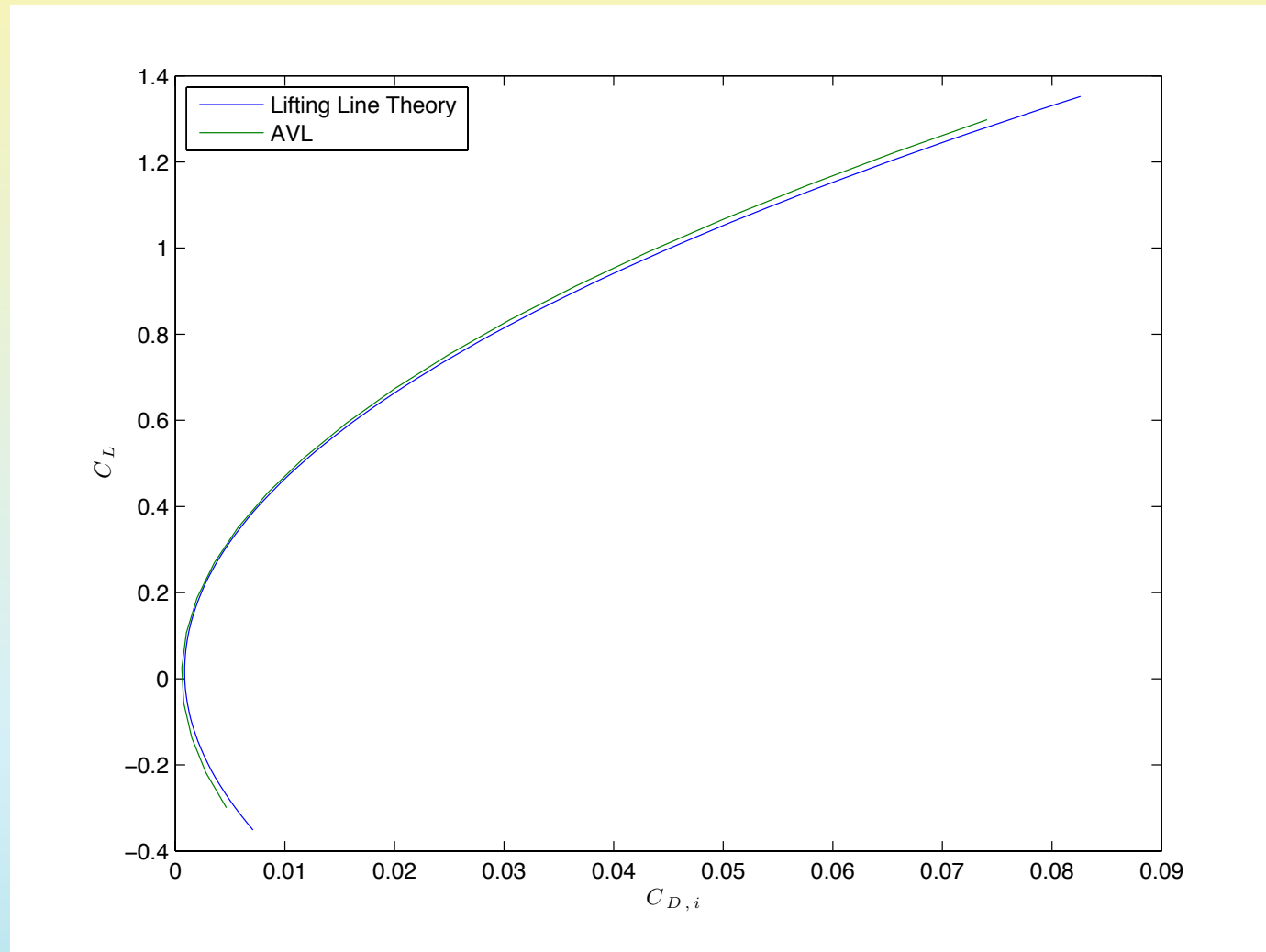
# Test Cases: Lifting Line Theory



# Test Cases: Aero Comparison

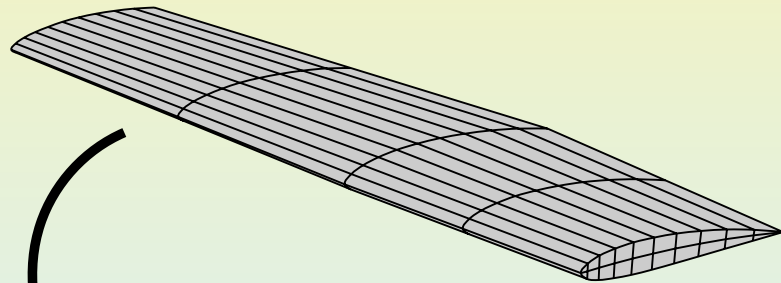


# Test Cases: Aero Comparison

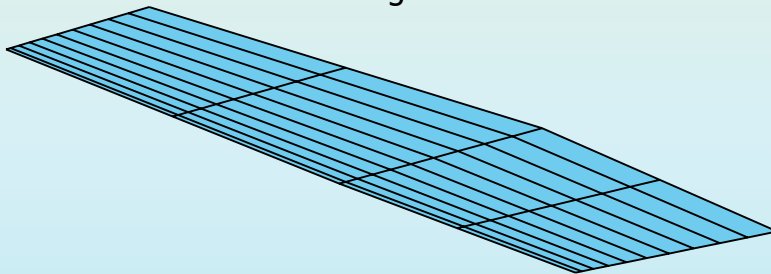


# Test Cases: Equivalent Plate (ELAPS)

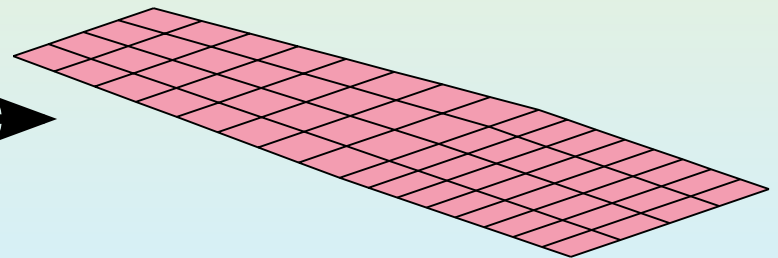
VSP Internal Definition



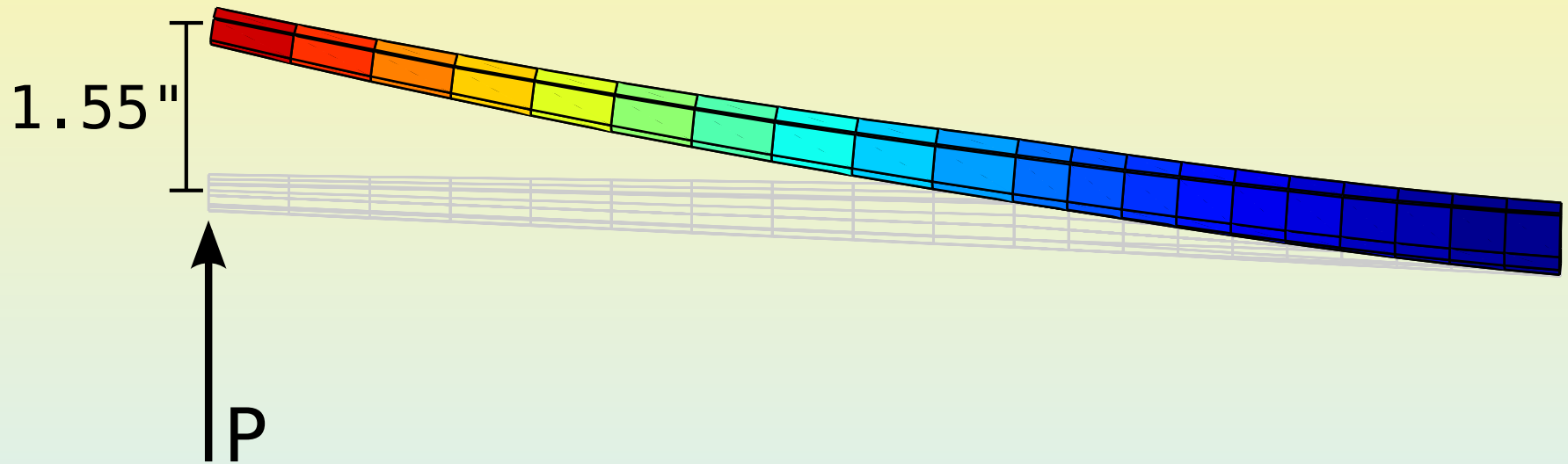
VSP Degenerate Plate



ELAPS Nodes on Reference Plane ( $z = 0$ )



# Test Cases: Equivalent Plate (ELAPS)



$E_{\text{chord}}$	3169 ksi
$E_{\text{span}}$	3169 ksi
$\nu$	0.32
G	1200 ksi
$\rho$	0.288 lb <sub>m</sub> /in <sup>3</sup>

Force of  $P = 1550$  lb applied at wing tip, mid chord



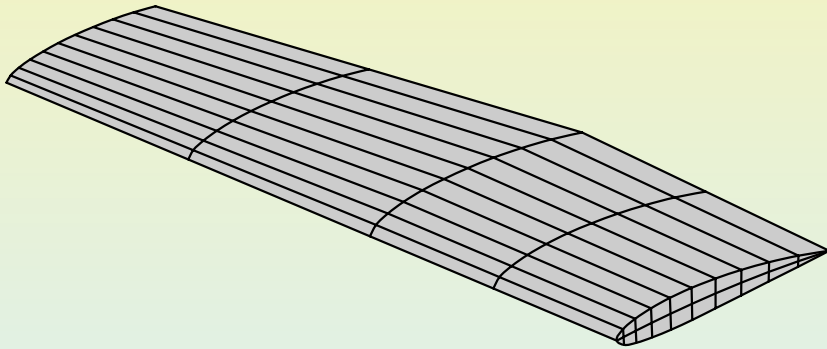
# Test Cases: Equivalent Plate (ELAPS)

	ELAPS	DegenPoint	% Difference
Volume	27.841	27.370	1.723
$x_{cg}$	4.167	4.153	0.334
$y_{cg}$	8.529	8.532	0.032
$z_{cg}$	1.956	2.106	7.109

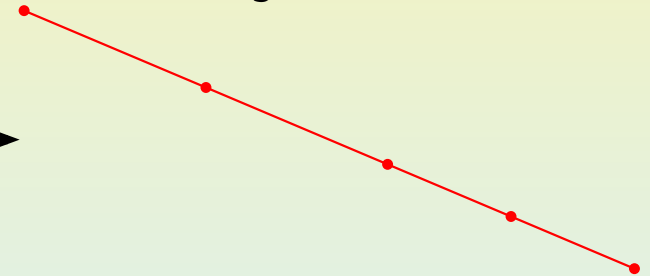


# Test Cases: Equivalent Beam

VSP Internal Definition



Beam Compsed of Leading  
Edge Nodes



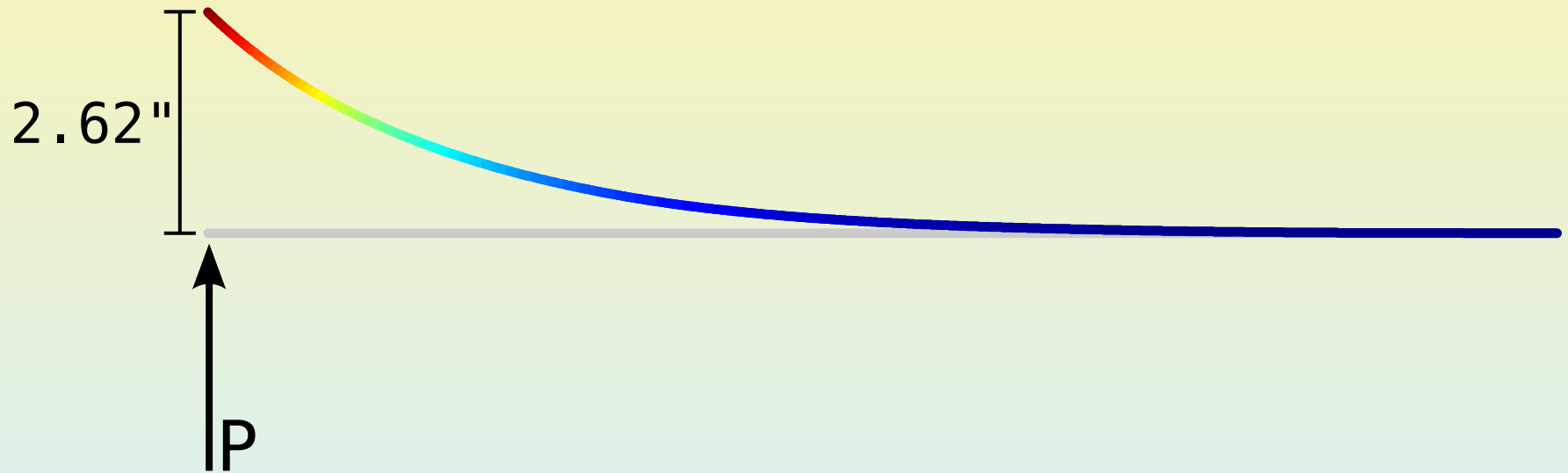
Euler-Bernoulli Beam Theory

$$\frac{d^2}{dy^2} \left( EI \frac{d^2 w}{dy^2} \right) = 0$$





# Test Cases: Equivalent Beam



E	3169 ksi
I	From DegenStick
P	1550 lb



Questions?

