# **Learning To Fly**

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## **Abstract**

## 1 Introduction

Autonomous flying machines are the future of aerial vehicles and they present an area of great interest to the control engineering fraternity. Conventional autopilot systems are commonly implemented through complex nested proportional-integral-derivative (PID) controllers, which require tedious parameter tuning and are vulnerable to control instability in perturbed flight conditions. Previous works applying reinforcement learning to aircraft control have been limited to methods simplifying the problem by discretizing the action space and reducing the number of dimensions in the state space Rennie [2018]. There is hence a need to explore more Artificial Intelligence based methods for autonomous flying which achieves full six degrees-of-freedom aircraft control with continuous actions and move towards a new generation of self-flying machines that operate safely and robustly.

Thus arises the problem we tackle in this work. The problem is part of the Airbus Learning to Fly Challenge(link). It deals with autonomous control of aircraft. This problem domain is well suited for reinforcement learning which enables agents to learn from interactions with an environment. Also, reinforcement learning has been successfully used in the past for many control system problems including autonomous flying of helicopters Ng et al. [2003] and recent advances in the application of deep neural networks to RL have allowed agents to perform well in increasingly complex tasks, including continuous control tasks Rennie [2018].

## 2 Problem Setting

The task at hand is known as the Heading Control Task as provided in the gym-jsbsim environment. In this environment, we pilot an Airbus A320 trying to maintain a fixed altitude while achieving various target setpoints.

#### The task at hand

- Achieve stable, steady flight at fixed altitude 10000ft
- New target headings are provided every 150 seconds with increasing complexity. The complexity is increased by increasing the difference between the current and new target headings. i.e. Delta between current and next headings increases linearly  $\pm 10 \deg$ ,  $\pm 20 \deg$ ,  $\pm 30 \deg$
- Termination of the task occurs when current heading is not within  $10\deg$  of the target heading at the end of 150 seconds, or if the aircraft is more than 100ft away from the target altitude
- Reward depends on deviations in heading and altitude throughout the duration of the flight

**State Space** The state space is a Nine-dimensional vector representing the altitude, heading, roll, pitch, yaw and velocity of the aircraft. Specifically, it is [Delta Altitude, Delta Heading, Pitch, Roll, Downward Velocity, Total air speed, Roll rate, Pitch rate, Yaw rate]

**Action Space** We are given control over four actions, namely the Aileron (which controls roll), Elevator (which controls pitch), Rudder (which controls yaw) and the Throttle (which controls the aircraft's power).

## 3 Theoretical Background

We use various approaches to solve this problem, starting with a simple PID-based baseline, a PPO-based approach and ending with policy search and hill climbing.

**PID Controller** The PID controller (Fig. 1), or the Proportional-Integral-Differential controller, is a control loop mechanism employing feedback that is widely used in applications requiring continuously modulated control. The controller constantly calculates the error in the target property (i.e. the difference between the desired set-point  $r\left(t\right)$  and the measured process variable  $y\left(t\right)$ ) and uses three terms, namely a term proportional to the error, one which is the integral of the error, and the third being the derivative of the error, to produce the required correction  $\left(u\left(t\right)\right)$ .

$$u(t) = K_p e(t) + K_i \int_0^t e(\tau) d\tau + K_d \frac{de(t)}{dt}$$

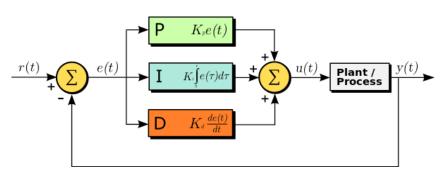


Figure 1: A block diagram of a PID controller in a feedback loop.  $r\left(t\right)$  is the desired process value or setpoint (SP), and  $y\left(t\right)$  is the measured process value (PV). Image taken from Wikipedia contributors [2022]

**Proximal Policy Optimization** PPO is a policy gradient method that improves on Trust Region Policy Optimisation using a clipped surrogate objective.

$$\begin{split} L^{PG}(\theta) &= \hat{\mathbb{E}}_t[\log \pi_\theta \left(a_t \mid s_t\right) \hat{A}_t] \\ r_t(\theta) &= \frac{\pi_\theta \left(a_t \mid s_t\right)}{\pi_{\theta_{old}} \left(a_t \mid s_t\right)} \\ L^{CLIP}(\theta) &= \hat{\mathbb{E}}_t \left[\min \left(r_t(\theta) \hat{A}_t, clip \left(r_t(\theta), 1 - \epsilon, 1 + \epsilon\right) \hat{A}_t\right)\right] \end{split}$$

**Policy Search** Policy Search is an application of Black Box Optimisation to Reinforcement Learning. The goal for black box optimisation is to use numerical derivative-free methods to optimize an objective function  $f:W\to\mathbb{R}$  (where  $W\subset\mathbb{R}^n$ )

Since black box optimisation does not involve gradient computation, the advantage with using such numerical methods is that we need little to no assumptions about the function f i.e. it can be discontinuous, non-linear or erratic. The only requirement to running such methods is that it should be relatively efficient to compute f(w) (generally using computer simulations). In terms of the success of these methods, they are known to work well with problems that are in general more difficult to solve compared to convex optimization. Audet and Kokkolaras [2016] However, most of these

methods are heuristic based and donot have optimality guarantees but the ambition here is to find good parameter values efficiently rather than searching for optimal solutions.

In the context of reinforcement learning, the set W is the set of parameters defining our agent policy  $\pi:S\to\mathbb{A}$  and f(w) can be the expected long term reward that we wish to optimize. Random or Grid search over the set W are two natural approaches to solve this problem but they quickly fail with increase in the number of dimensions n. Also, no numerical method is expected to work well for a large number of dimensions (1000's or higher). Local search is known to work well for intermediate n (10's,100's). One common approach to local search is hill climbing which involves starting with an initial guess  $w_0$  and then sampling points on a hypersphere in n dimensions with  $w_0$  as the center. The idea is to compare  $f(w_0)$  to f(w) for all the sampled points and move to the point which gives the highest increase in the function value. Thereafter, it becomes the new center and the process repeats. The algorithm stops when the center gives the highest function value.

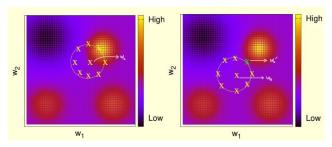


Figure 2: Illustration of hill climbing. The image is taken from: https://www.cse.iitb.ac.in/ shivaram/teaching/old/cs747-a2021/index.html

## 4 Experiments

**PID Controller** In order to tune the PID controller, we broke the control problem into controlling altitude first and then the heading. We aimed to maintain roll and pitch while performing the above mentioned control tasks. We use the tuning strategy proposed by Ziegler and Nichols [1993] to tune the same.

While tuning for altitude control, no setting of the  $K_p, K_i, K_d$  values seemed to provide stable altitude flight over long time periods. We thus tried using an offset in the elevator and throttle actions, and were able to provide near-stable altitudes over long time periods. We then tuned the PID controller over these offset action values. We finally obtained the following PID controller which was able to satisfactorily control roll, pitch, heading and altitude but was not able to control acceleration. These results corroborate findings that controls more complicated than PID are required to manoeuvre aircrafts.

**Proximal Policy Optimization (PPO)** In order to train agents using PPO, we first tried to learn a policy using the default reward function provided by the gym-jsbsim environment. The default reward function was a geometric mean of scaled gaussian rewards for each relevant variable i.e. heading, altitude, roll, speed and acceleration where the reward for each variable was

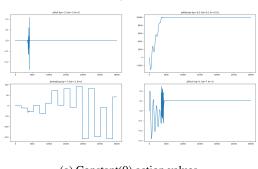
$$Reward = exp(-(VariableValue/ErrorScale)^2)$$

The error scales for the different variables are as in Table 1:

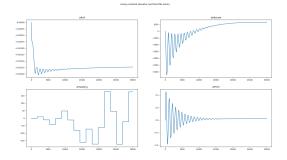
We trained PPO for 1M steps with default hyperparameters as given in StableBaselines3. Raffin et al. [2021]

Next we tried to modify the reward function by linearizing it to see if it improved performance. Thus instead of a geometric mean, if summed up rewards for each relevant variable, where the reward for each variable was -1 if it was beyond the error scale and a linear function between 0 to 1 if it was within the error scale. Finally each individual reward was given a weight and the weighted sum was the total reward. The weights for different variables are as follows:

We again trained PPO for 1M with the modified reward function and default hyperparameters. The results of the two PPO models are provided in Table 4



(a) Constant(0) action values



(b) Constant (non-zero) elevator and throttle action

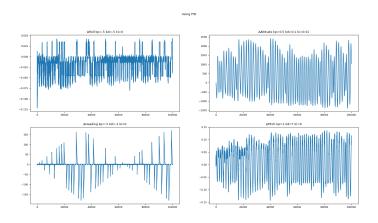


Figure 3: The baseline PID controller

**Policy Search** For applying policy search to this problem, we modelled our policy  $\pi:S\to\mathbb{A}$  with a single hidden layer of 16 neurons. The input is the 9-dimensional MDP state and the output is the 4-dimensional action. We use tanh activation function and appropriately scale the network output to match the bounds of the action space. In total we have 208 weights which form our search space.

Initially we kept our objective function as the default reward we want to maximise. Our initialisation, we searched over 100,000 seeds to check which gives the best initial policy. Then we tried to hill climb on that using reward function as our objective but were unable to do so. Then we changed our objective to minimise delta heading and alitude near the checkpoints and maximise the reward and run time elsewhere. We sampled 10,000 points on the hypersphere with a radius of 1 which was annealed with increasig steps. This led to some promising results.

Table 1: Error Scales

| Variable           | Error Scale  |
|--------------------|--------------|
| Heading            | 5 degrees    |
| Altitude           | 50 feet      |
| Roll               | 0.35 radians |
| Speed              | 16 fps       |
| Acceleration(x, y) | 0.1g         |
| Acceleration(z)    | 0.5g         |

Table 2: Reward Weights

| Variable     | Weight |
|--------------|--------|
| Heading      | 0.3    |
| Altitude     | 0.3    |
| Roll         | 0.1    |
| Speed        | 0.1    |
| Acceleration | 0.2    |

Table 3: PPO Models

| Model           | Run Time(in sec) | Reward  |
|-----------------|------------------|---------|
| Default Reward  | 134.083          | 234.899 |
| Modified Reward | 104.417          | 89.331  |

Table 4: Policy Search

| Sr No | Run Time(in sec) | Reward  |
|-------|------------------|---------|
| 1     | 468.67           | 1729.61 |
| 2     | 150.00           | 648.32  |
| 3     | 619.83           | 415.79  |

#### 5 Conclusion and Future Work

We have implemented various algorithms in this work in the hope of solving the autonomous flight challenge. We have seen how PID controllers are not quite capabale of achieving sustained flights with varying target headings. We tried the PPO algorithm as it showed some promise, but realised in the end that the search space is so large that tuning PPO is a near-impossible task. Policy search yet again shines in this task, providing stellar results and flight-times with minimal training times. While our PID controller could barely produce 50 seconds of safe (as defined by the gym environment) and sustained flight, the policy search model was easily able to control the flight for over 600 seconds, providing over a 12x improvement.

In order to extend this work, we would like to perform more extensive hyperparameter search and tuning on policy search and hill climbing. We would like to explore more deeply the Physics and Mathematics behind aircraft flight, and use this knowledge to handcraft even better state spaces. Optimizing over these state spaces using policy search may yield surprising results.

## Acknowledgments and Disclosure of Funding

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## Checklist

- 1. For all authors...
  - (a) Do the main claims made in the abstract and introduction accurately reflect the paper's contributions and scope? [Yes]
  - (b) Did you describe the limitations of your work? [Yes]
  - (c) Did you discuss any potential negative societal impacts of your work? [N/A]
  - (d) Have you read the ethics review guidelines and ensured that your paper conforms to them? [Yes]
- 2. If you are including theoretical results...
  - (a) Did you state the full set of assumptions of all theoretical results? [N/A]
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- 3. If you ran experiments...
  - (a) Did you include the code, data, and instructions needed to reproduce the main experimental results (either in the supplemental material or as a URL)? [Yes]
  - (b) Did you specify all the training details (e.g., data splits, hyperparameters, how they were chosen)? [Yes]
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