



Georgia Tech  
College of  
Engineering

# COE 3001

# MECHANICS OF DEFORMABLE BODIES

*Lecture 1* – Introduction

Yazhuo Liu

Georgia Institute of Technology

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# General Information

## Course Materials

- All lecture notes, assignments, and solutions will be available at **Canvas**

## Office Hours

- In-person: MW 12:30 – 1:30 pm, MRDC 4114A (next to the meeting room)
- Online: Zoom meeting based on appointment

## Prerequisite

- COE 2001 or CEE 2020 or ME 2211 or AE 2120 (**Statics**)
- MATH 2403 or MATH 2552 or MATH 2413 (**Differential equations**)

# General Information

## Textbook

- ✓ Mechanics of Materials, by James M. Gere & Barry J. Goodno, 9th Ed, Cengage Learning.

Note:

- ✧ Homework may be taken from the textbook but will provide full context.
- ✧ Reading recommendations are based on the textbook.
- ✧ Different editions are **ACCEPTABLE**.

# General Information

## Grading

- (Bonus) Attendance: 6 pts.
- Homework:  $5 \times 6$  pts. + 4 pts. bonus
- Midterm Exam: 30 pts.
- Final Exam: 40 pts.

Points	Letter grade
90 ~ 100	A
80 ~ 89	B
70 ~ 79	C
60 ~ 69	D
0 ~ 59	F

## ➤ 3 attendance check (bonus points)

- ✓ 1 present – 1 pt.
- ✓ 2 presents – 3 pts.
- ✓ 3 presents – 6 pts.



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## 6 assignments: 6 pts each

- Lowest grade assignment will be dropped.
- Typed assignments have 1 bonus pt each time (maximum total: 4 pts)

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## Exam:

- Exam info will be announced separately.
- In class and close book
- 1 page (2 sides) equation sheet allowed
- Copy homework solutions on equation sheet prohibited



# General Information

## Using GenAI Responsibly – Key Guidelines

- ◆ Permitted Use:
  - Clarify concepts, review material, explore alternative explanations
  - Support learning—not replace understanding
- ◆ Critical Evaluation Required:
  - Verify all AI-generated content with authoritative sources (lectures, readings, course materials)
- ◆ Strictly Prohibited:
  - Using GenAI to complete assignments, homework, projects, or exams
  - Submitting AI-generated work as your own = academic misconduct
- ◆ Important Reminders:
  - Over-reliance without comprehension may result in penalties
  - Always adhere to Georgia Tech’s Academic Integrity Policy
  - Learn more: <https://oit.gatech.edu/ai/guidance>



# General Information

## Course Principles

1. Discussions on homework and lecture notes are encouraged. However, exchange of written information in completing assignments is **NOT** permitted.
2. Unless approval is granted, late assignments will receive a 30% late penalty. Assignments submitted after the solutions have been posted will not be accepted, except for delays due to serious illness or other documented difficulties.
3. Unless approval is granted, **NO** make-up exams will be given, except for serious illness or other documented difficulties.



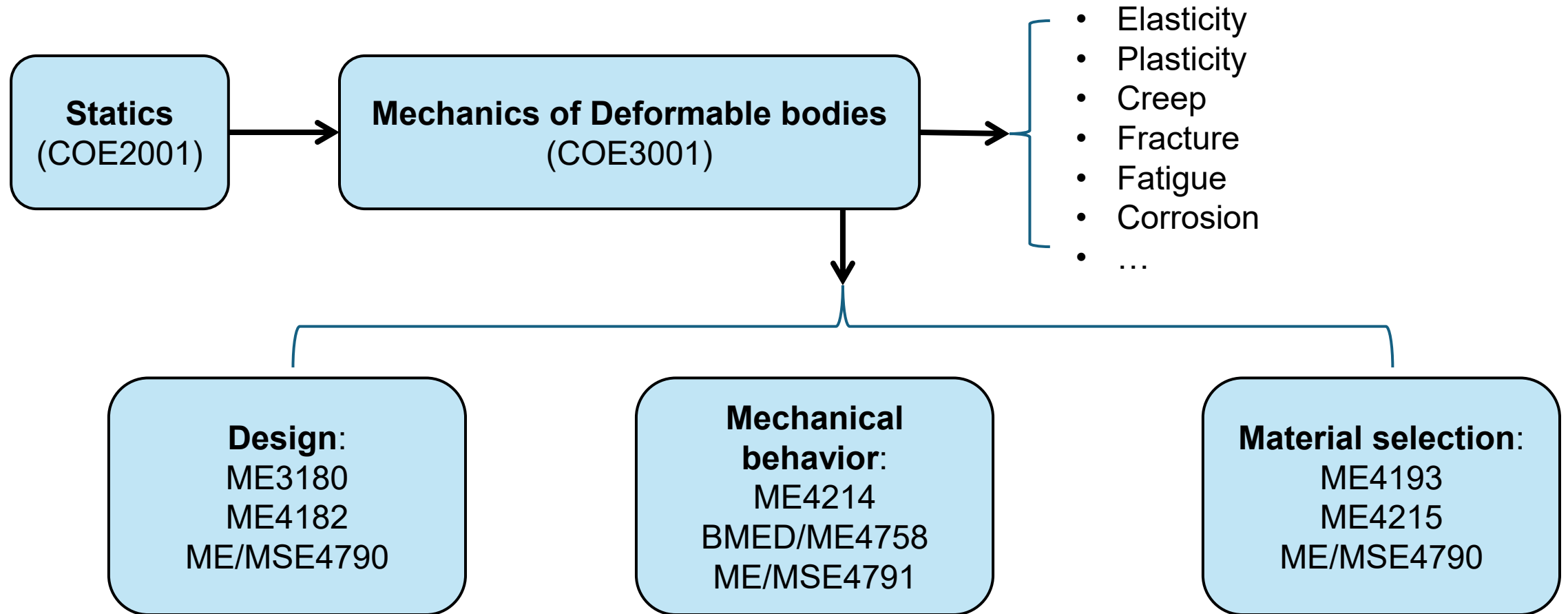
# Learning Objectives

By the end of this course, students will be able to:

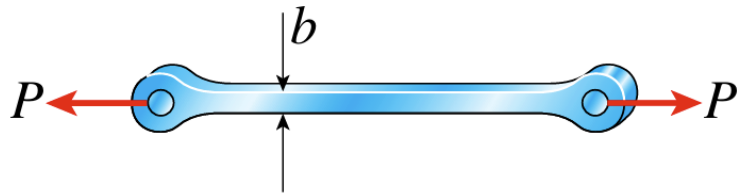
- Analyze the strength, stiffness, and stability of structural components.
- Solve real-world engineering problems — such as designing lightweight bridges
- Understand the mechanical behavior of materials in practical applications.
- Build a solid foundation for advanced courses in *machine design* , *structural mechanics* , and *mechanical systems*.

This course bridges everyday experience with engineering principles, helping students develop the analytical tools needed to design and evaluate safe, efficient, and reliable structures and systems.

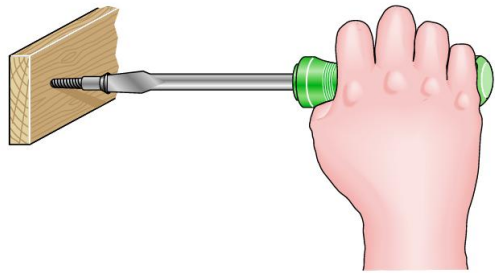
# Relation to other courses



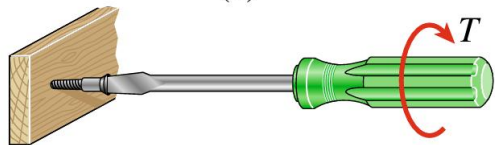
# Topics



Bars: tension/compression

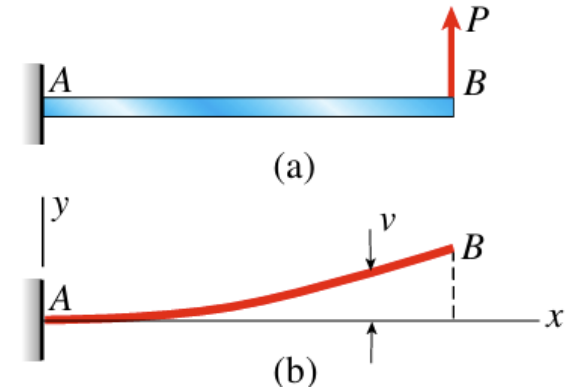


(a)

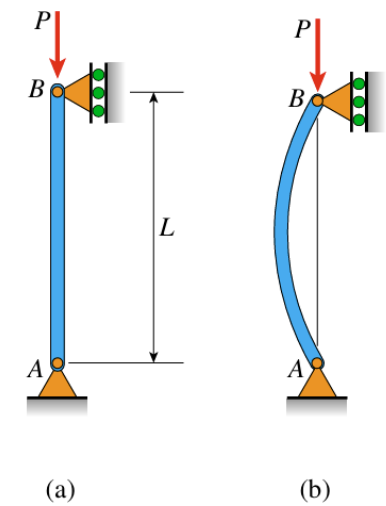


(b)

Shafts: torsion



Beams: bending

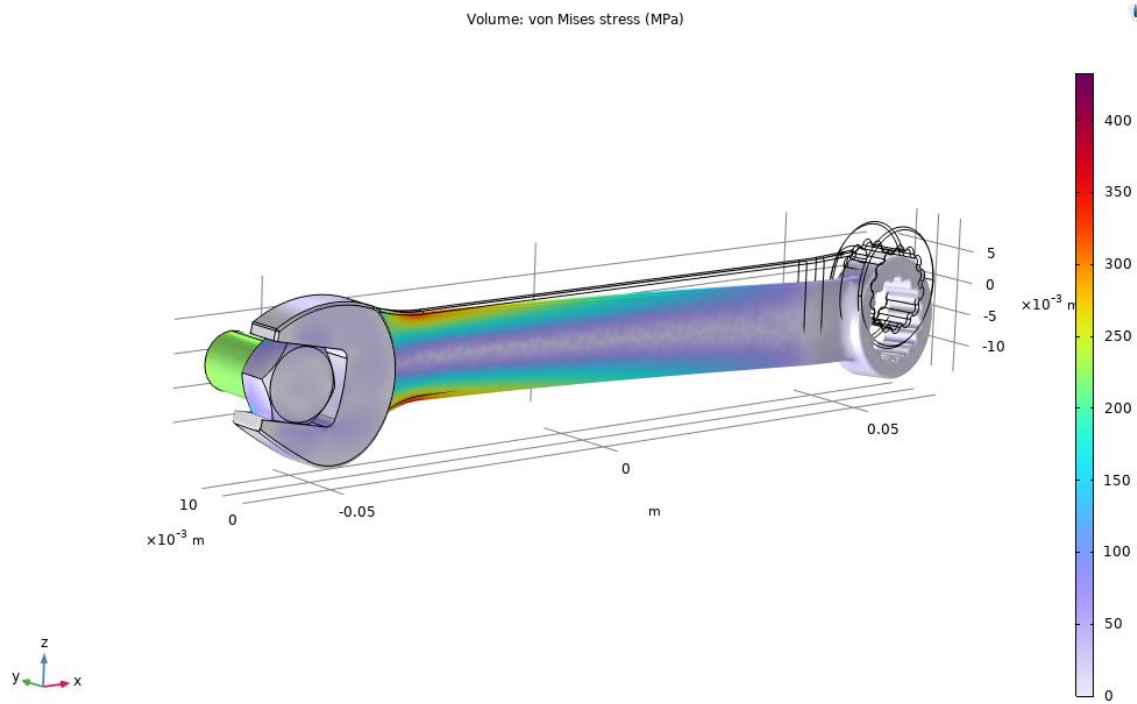


Columns: buckling

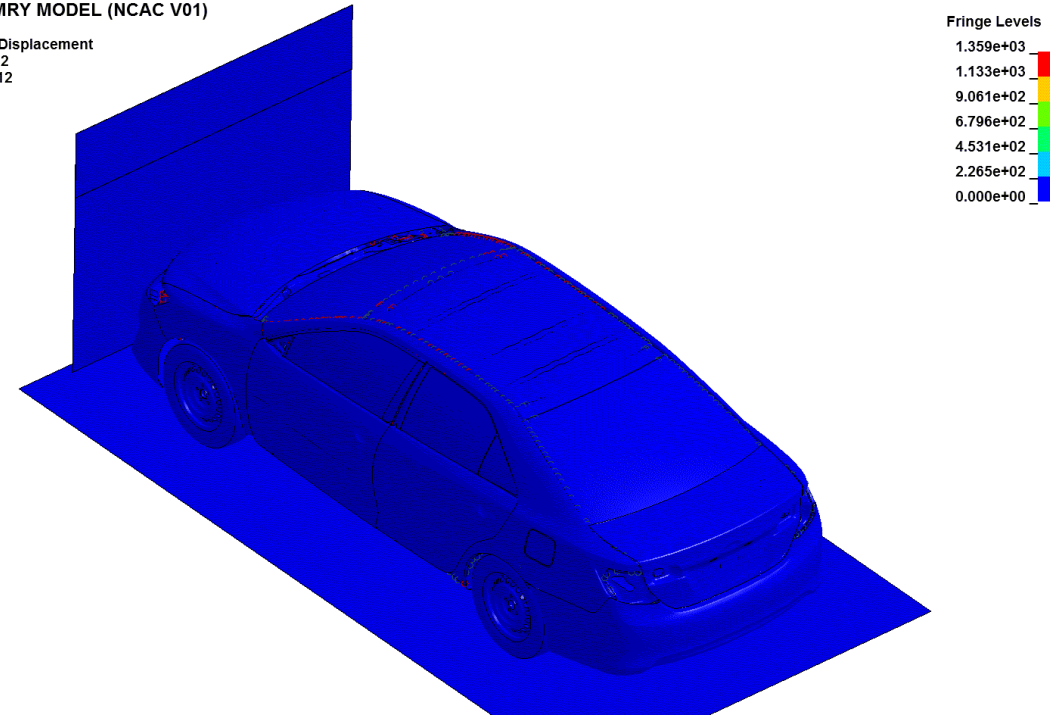
Don't worry if you don't understand it now. You'll gain a much better understanding when you revisit it later in the course.



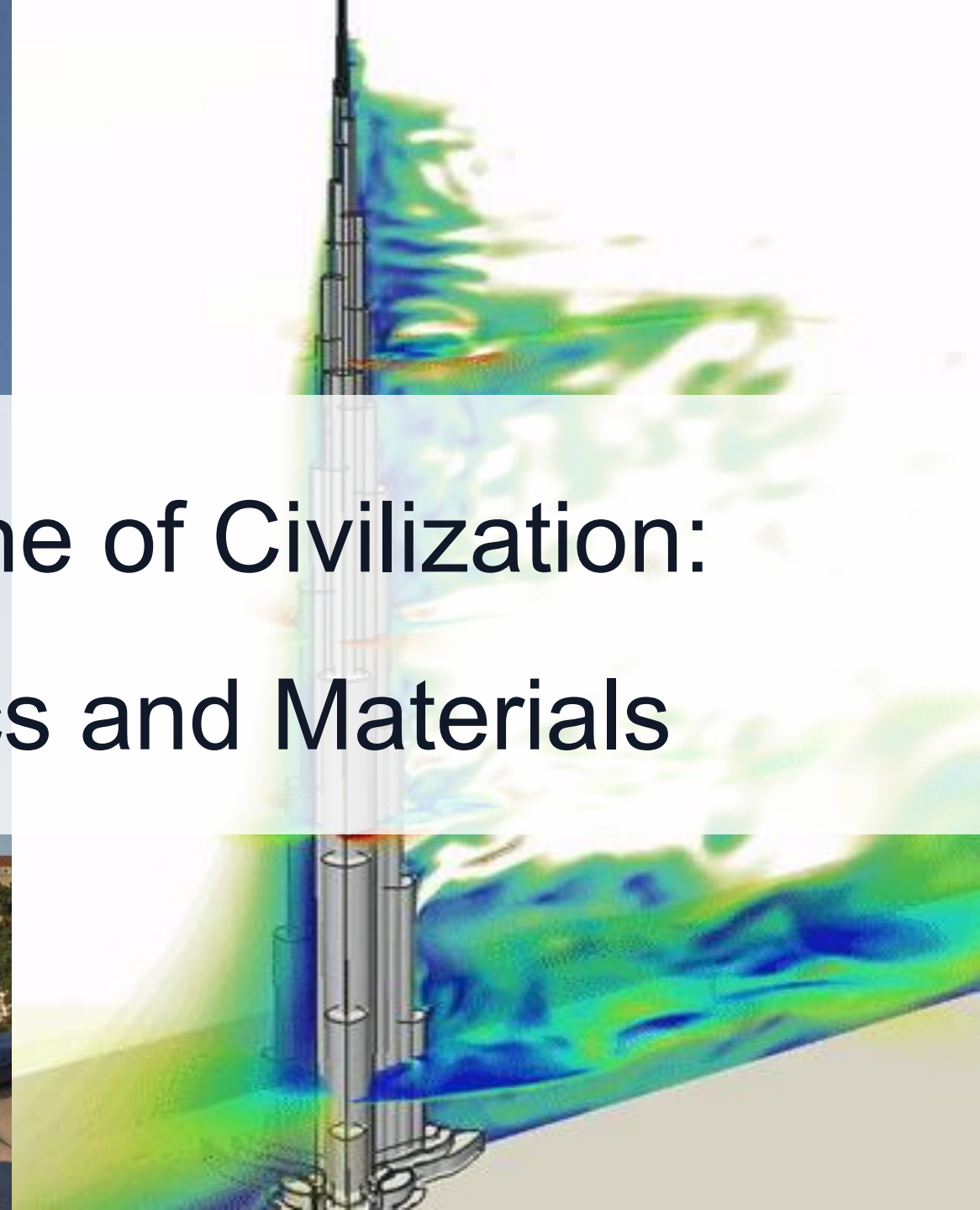
# Mechanics of life



2012 TOYOTA CAMRY MODEL (NCAC V01)  
Time = 0  
Contours of Resultant Displacement  
min=0, at node# 246412  
max=0, at node# 246412







# The Hidden Backbone of Civilization: Stories of Mechanics and Materials

# Saturn V – Reaching for the Moon

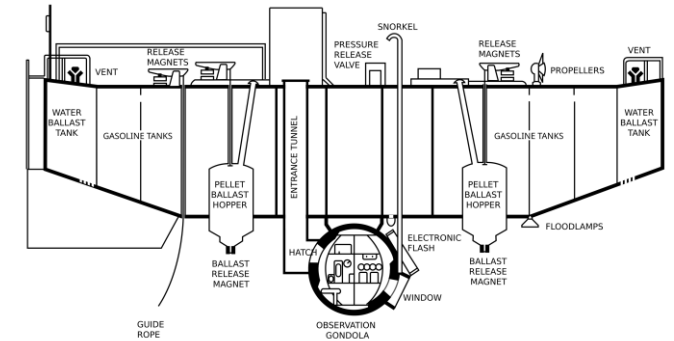
- Materials: Aluminum alloys, titanium alloys, etc.
  - *Honeycomb sandwich structures — lightweight yet strong enough to defy Earth's gravity.*
- Engine thrust and its immense weight:
  - *Compression/Tension*
- Aerodynamic forces:
  - *Bending*
- Roll control thrusters:
  - *Torsion*



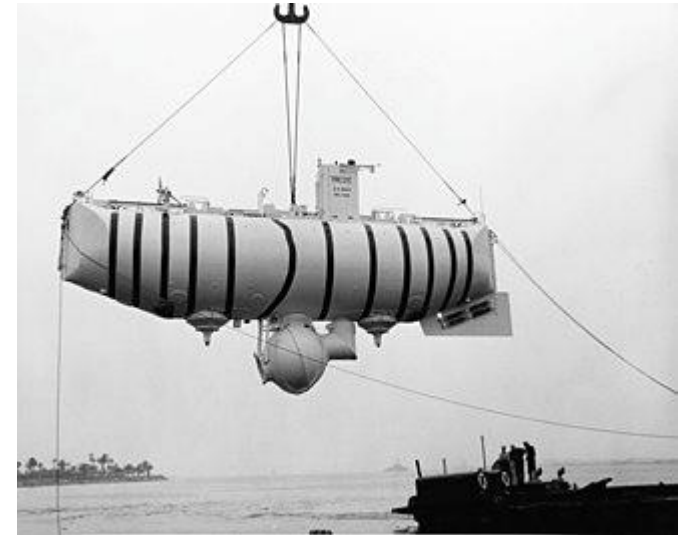
The launch of the historic Apollo 11 mission on Saturn V SA-506, July 16, 1969

# Bathyscaphe Trieste – Into the Abyss

- Materials: Steel spherical pressure hull
  - *Designed to withstand ~110 MPa external pressure.*
- External water pressure:
  - *Compression (uniform on hull)*
- Supporting cables & mounts:
  - *Tension*
- Deployment forces at ports or during mooring:
  - *Bending / Torsion*



GENERAL ARRANGEMENT DRAWING OF TRIESTE, CA. 1959

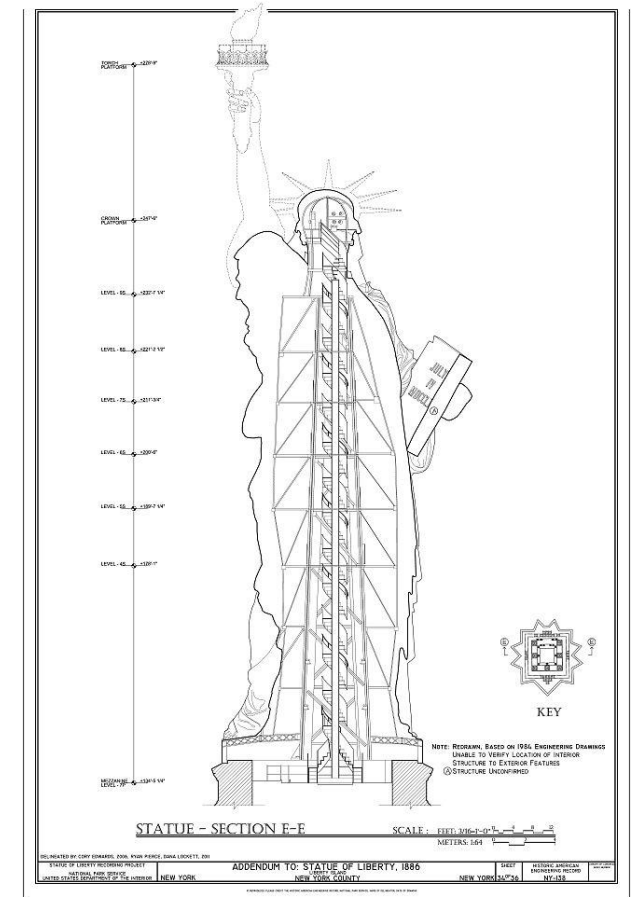


Bathyscaphe Trieste — In 1960, it became the first crewed vessel to reach the bottom of Challenger Deep in the Mariana Trench.



# Statue of Liberty – Engineering Aesthetics

- Materials: Wrought iron frame, copper skin
  - *Copper shell expands/contracts with temperature; iron framework resists deformation.*
- Central pylon:
  - *Compression (supports vertical load)*
- Armature bars:
  - *Tension (prevents collapse from wind or weight)*
- Wind loads (especially on torch arm):
  - *Bending / Torsion*



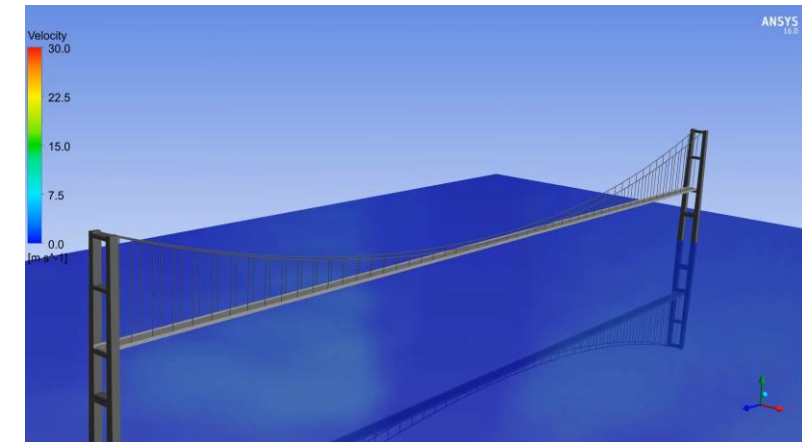
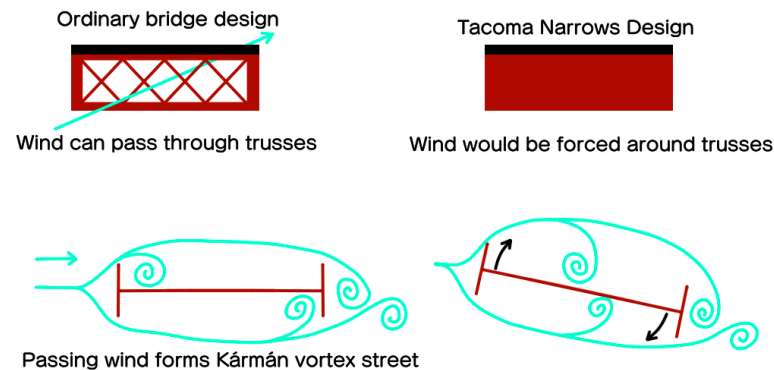
A massive iron pylon and secondary skeletal framework allows the Statue's copper skin to move independently yet stand upright.

# Mechanics Design is Important ...



Tacoma Narrows Bridge,  
Nov. 7, 1940, Washington

- The longest suspension bridge of its time.
- It opened to traffic on **Jul. 1, 1940**, and dramatically collapsed into Puget Sound on **Nov. 7** of the same year.
- Due to **design faults**.
- The violent swaying and eventual collapse resulted in the death of a cocker spaniel named “*Tubby*”.



# Understanding Mechanics is Important ...



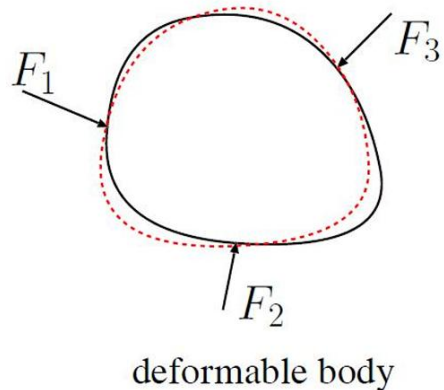
Aloha Airlines Flight 243, Boeing 737-200,  
Apr. 28, 1988

- The flight suffered extensive damage after an explosive decompression in flight.
- A large section of the roof had torn off, consisting of the entire top half of the aircraft skin extending from just behind the cockpit to the fore-wing area.
- Due to **fatigue cracks and poor maintenance**.
- The one fatality, flight attendant *Clarabelle "C.B." Lansing*, was ejected from the airplane. 65 passengers and crew were injured.

# COE 3001 is important !!!

**Mechanics of Deformable bodies** is foundational in **biomedical, mechanical, civil, and aerospace engineering**. It explores how materials respond to external forces — addressing the central question:

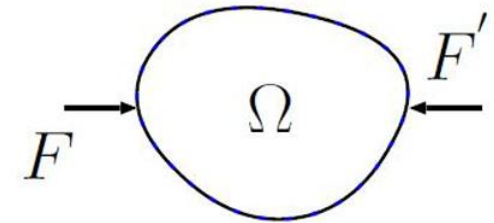
*How do materials resist or fail under external load?*



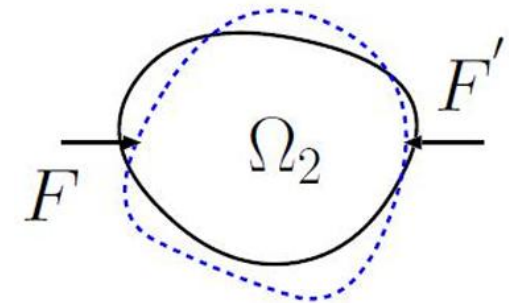
- **Strength:**  
capacity of a structure to resist loads (max load)
- **Stiffness:**  
ability to resist changes in shape (allowable deformation)
- **Stability:**  
ability to resist buckling under compressive stresses (stable or not while in service)

# Rigid/deformable bodies

Rigid bodies	Deformable bodies
can sustain <b>unbounded</b> load	<b>upper bound</b> exists
<b>NO</b> shape and size changes	with shape and size changes
<b>idealized model</b>	more <b>realistic</b>
cares only the <b>overall load carrying, kinematics, dynamics</b> , etc.	cares more about the <b>shape and size changes</b> due to loads
usually used in COE 2001	usually used in COE 3001



A rigid body



A deformable body



# Idealizations for deformable bodies

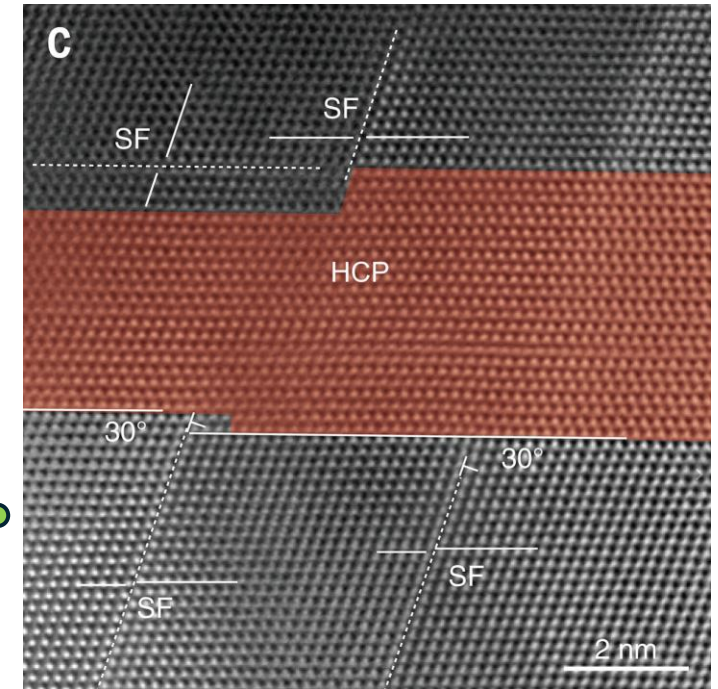
Assumptions that work well on macro-scales:

1. continuous (no voids, cracks, defects, etc.)
2. homogeneous (uniform in material properties)
3. isotropic (behaving identically in all directions)



Steel Highway Guardrail-Roller Barrier

Think about the  
size scale you  
are interested



Magnified HAADF-STEM  
image for Stainless Steel  
*Pan, et al., Science 2025*

# Isotropic

- In the study of mechanical properties of materials, isotropic means having identical values of a property in all directions.
- Isotropic materials: metals, glass, etc.
- Anisotropic materials: composites, etc.



isotropic



anisotropic



# Units

SI units are used in this course.

Quantity	SI Units	USC Units
Length	meter (m)	Foot (ft)
Mass	kilogram (kg)	slug
Force	Newton (N)	pound (lb)
Moment	Newton-meter (N*m)	foot-pound (lb*ft)
Energy	Joule (J)	foot-pound (lb*ft)
etc.		

Quantity	Metric (SI) Units of Measure	US Customary (USC) Units of Measure	Conversion Factors (SI) to (USC)
Time	second (s) 1 minute (min) = 60 seconds 1 hour (hr) = 60 minutes 1 day = 24 hours	second (s) 1 minute (min) = 60 seconds 1 hour (hr) = 60 minutes 1 day = 24 hours	--
Length	meter (m) 1000 millimeters (mm) = 1 m 1 kilometer (km) = 1000 m	foot (ft) 12 inches (in) = 1 ft 1 mile (mi) = 5280 ft	1 m = 3.281 ft
Area	square meter (m <sup>2</sup> )	square foot (ft <sup>2</sup> )	1 m <sup>2</sup> = 10.764 ft <sup>2</sup>
Volume	cubic meter (m <sup>3</sup> ) Liter (L) = .001 m <sup>3</sup>	cubic foot (ft <sup>3</sup> ) Gallon (gal) = .1337 ft <sup>3</sup>	1 m <sup>3</sup> = 31.315 ft <sup>3</sup> 3.785 L = 1 gal
Mass	kilogram (kg) metric ton / tonne (t) = 1000 kg	slug = 1 ( $\frac{lb \cdot s^2}{ft}$ )	14.59 kg = 1 slug
Force	newton (N) = 1 ( $\frac{kg \cdot m}{s^2}$ ) kilonewton (kN) = 1000 N	pound (lb) 16 ounces (oz) = 1 lb 1 kilo pound (kip) = 1000 lbs 1 ton = 2000 lbs	4.448 N = 1 lb
Velocity (Linear)	meter per second (m/s) 3.6 kilometers per hour (kph) = 1 m/s	feet per second (ft/s) mile per hour (mph) = 1.467 $\frac{ft}{s}$	1 $\frac{m}{s}$ = 3.281 $\frac{ft}{s}$
Acceleration (Linear)	meter per second squared ( $\frac{m}{s^2}$ )	foot per second squared ( $\frac{ft}{s^2}$ )	1 $\frac{m}{s^2}$ = 3.281 $\frac{ft}{s^2}$
Velocity (Angular)	radian per second ( $\frac{rad}{s}$ ) 1 rotation per minute (rpm) = $\frac{2\pi \text{ rad}}{60 \text{ s}}$	radian per second ( $\frac{rad}{s}$ ) 1 rotation per minute (rpm) = $\frac{2\pi \text{ rad}}{60 \text{ s}}$	--
Acceleration (Angular)	radian per second squared ( $\frac{rad}{s^2}$ )	radian per second squared ( $\frac{rad}{s^2}$ )	--
Moment of a Force; Torque	Newton-meter (N*m)	Foot-pound (lb*ft) Inch pound (lb*in)	1.356 N*m = 1 lb*ft
Area Moment of Inertia	meter <sup>4</sup> (m <sup>4</sup> ) 1 millimeter <sup>4</sup> (mm <sup>4</sup> ) = 10 <sup>-12</sup> * m <sup>4</sup>	foot <sup>4</sup> (ft <sup>4</sup> ) 20736 inches <sup>4</sup> (in <sup>4</sup> ) = 1 ft <sup>4</sup>	8.6x10 <sup>-3</sup> m <sup>4</sup> = 1ft <sup>4</sup>
Mass Moment of Inertia	kilogram meter <sup>2</sup> (kg * m <sup>2</sup> )	slug foot <sup>2</sup> (slug * ft <sup>2</sup> )	1 kg*m <sup>2</sup> = 1.356 lb*ft <sup>2</sup>
Work and Energy	joule (J) = 1 (N * m)	foot-pound (ft lb)	1.356 J = 1 ft lb
Power	watt (W) = 1 (J/s) = 1 ( $\frac{kg \cdot m}{s^2}$ ) kilowatt (kW) = 1000 W	foot pound per second ( $\frac{ft \cdot lb}{s}$ ) horsepower (hp) = 550 ( $\frac{ft \cdot lb}{s}$ )	1.356 W = 1 $\frac{ft \cdot lb}{s}$ 1 kW = 1.341 hp
Pressure or Stress	pascal (Pa) = 1 N/m <sup>2</sup> kilopascal (kPa) = 1000 Pa megapascal (MPa) = 10 <sup>6</sup> Pa gigapascal (GPa) = 10 <sup>9</sup> Pa	pounds per square inch (psi) = 1 ( $\frac{lb}{in^2}$ ) kilo-pounds per square inch (ksi) = 1000 psi	6894.76 Pa = 1 psi



# Prefixes

In the SI, designations of multiples and subdivision of any unit may be arrived at by combining with the name of the unit the prefixes

Name	Symbol	Meaning
giga	G	$10^9$
mega	M	$10^6$
kilo	k	$10^3$
milli	m	$10^{-3}$
micro	$\mu$	$10^{-6}$
nano	n	$10^{-9}$
etc.		

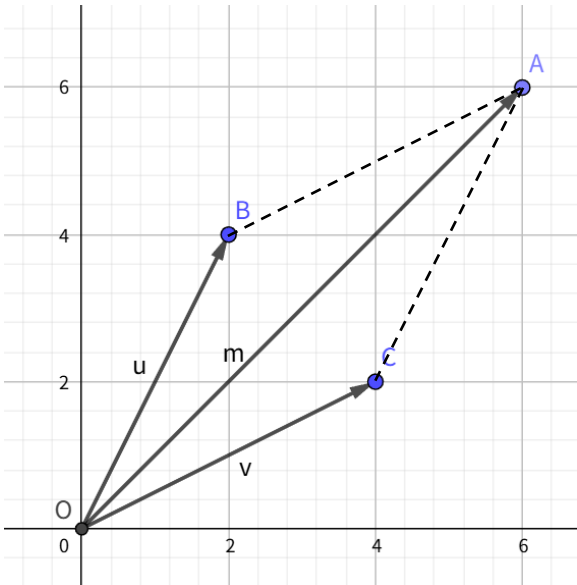
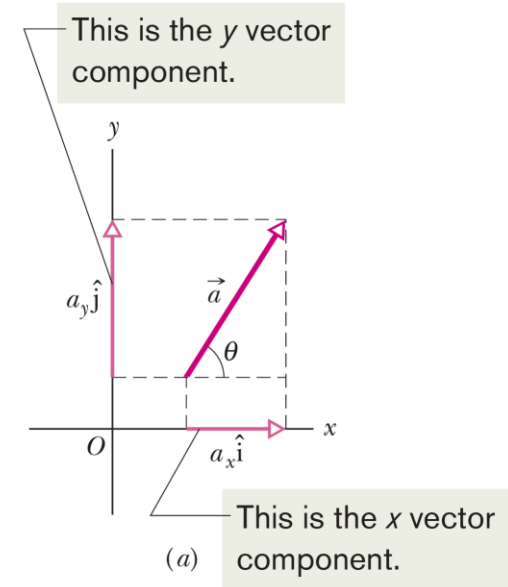
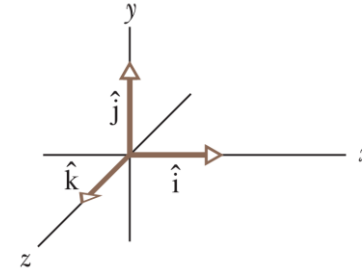
e.g.,

$$200 \text{ GPa} = 200 * 10^9 \text{ Pa}$$

$$100 \text{ MPa} = 100 * 10^6 \text{ Pa}$$

# Scalars vs. Vectors

- **Scalar:** A quantity that has **magnitude only**
  - Examples: mass, time, temperature, energy
- **Vector:** A quantity that has both **magnitude and direction**
  - Examples: force, velocity, displacement, acceleration

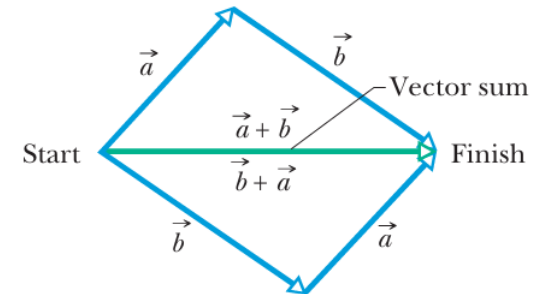


$$\mathbf{u} = 2\mathbf{i} + 4\mathbf{j} = \begin{pmatrix} 2 \\ 4 \end{pmatrix}$$

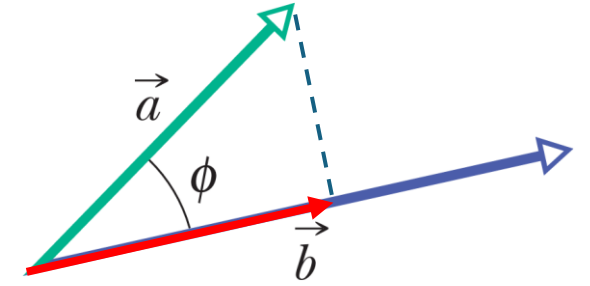
$$\mathbf{v} = 4\mathbf{i} + 2\mathbf{j} = \begin{pmatrix} 4 \\ 2 \end{pmatrix}$$

$$\mathbf{m} = \mathbf{u} + \mathbf{v} = (2\mathbf{i} + 4\mathbf{j}) + (4\mathbf{i} + 2\mathbf{j}) = \begin{pmatrix} 2 \\ 4 \end{pmatrix} + \begin{pmatrix} 4 \\ 2 \end{pmatrix}$$

$$\mathbf{m} = 6\mathbf{i} + 6\mathbf{j} = \begin{pmatrix} 6 \\ 6 \end{pmatrix}$$



# Multiplying Vectors



Component of  $\mathbf{a}$  along direction of  $\mathbf{b}$  is  $a \cos \phi$

- Dot product:

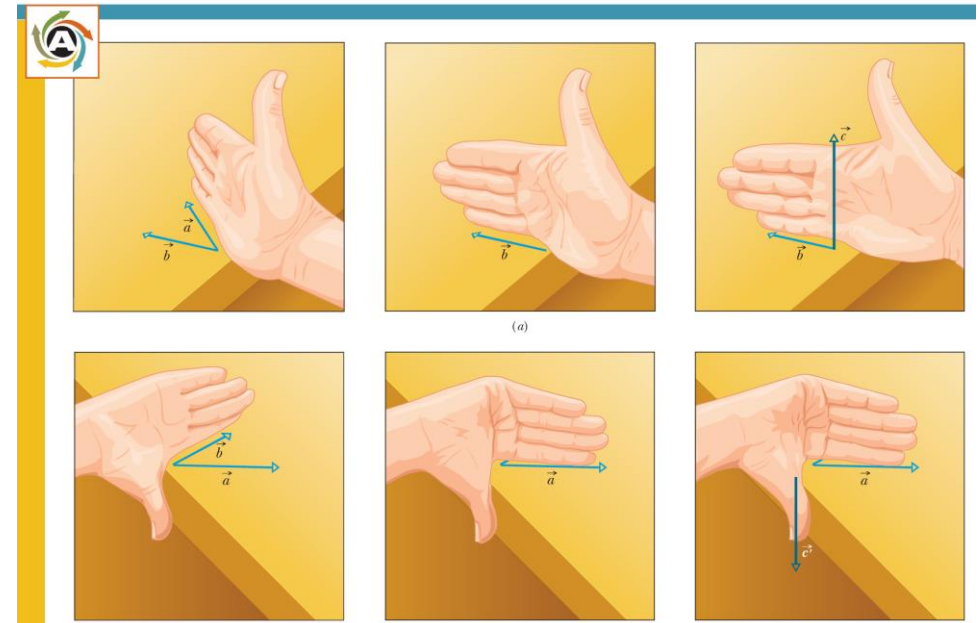
$$\mathbf{a} \cdot \mathbf{b} = ab \cos \phi = (a \cos \phi) b$$

$$\mathbf{a} \cdot \mathbf{b} = (a_x \mathbf{i} + a_y \mathbf{j} + a_z \mathbf{k}) \cdot (b_x \mathbf{i} + b_y \mathbf{j} + b_z \mathbf{k}) = a_x b_x + a_y b_y + a_z b_z$$

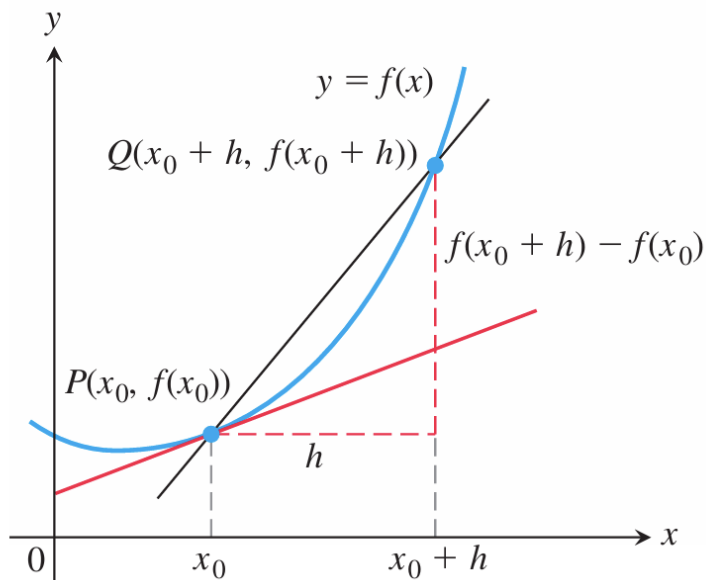
- Cross product:

$$\mathbf{a} \times \mathbf{b} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ a_x & a_y & a_z \\ b_x & b_y & b_z \end{vmatrix}$$

- Magnitude:  $|\mathbf{a} \times \mathbf{b}| = ab \sin \phi$
- Direction: Right-hand rule



# Derivatives and Differential Equations



**DEFINITION** The **derivative of a function  $f$  at a point  $x_0$** , denoted  $f'(x_0)$ , is

$$f'(x_0) = \lim_{h \rightarrow 0} \frac{f(x_0 + h) - f(x_0)}{h}$$

provided this limit exists.

**DEFINITION** The **derivative** of the function  $f(x)$  with respect to the variable  $x$  is the function  $f'$  whose value at  $x$  is

$$f'(x) = \lim_{h \rightarrow 0} \frac{f(x + h) - f(x)}{h},$$

provided the limit exists.

## DIFFERENTIATION RULES

### General Formulas

Assume  $u$  and  $v$  are differentiable functions of  $x$ .

Constant:  $\frac{d}{dx}(c) = 0$

Sum:  $\frac{d}{dx}(u + v) = \frac{du}{dx} + \frac{dv}{dx}$

Difference:  $\frac{d}{dx}(u - v) = \frac{du}{dx} - \frac{dv}{dx}$

Constant Multiple:  $\frac{d}{dx}(cu) = c \frac{du}{dx}$

Product:  $\frac{d}{dx}(uv) = u \frac{dv}{dx} + v \frac{du}{dx}$

Quotient:  $\frac{d}{dx}\left(\frac{u}{v}\right) = \frac{v \frac{du}{dx} - u \frac{dv}{dx}}{v^2}$

Power:  $\frac{d}{dx}x^n = nx^{n-1}$

Chain Rule:  $\frac{d}{dx}(f(g(x))) = f'(g(x)) \cdot g'(x)$

### Trigonometric Functions

$$\frac{d}{dx}(\sin x) = \cos x \quad \frac{d}{dx}(\cos x) = -\sin x$$

$$\frac{d}{dx}(\tan x) = \sec^2 x \quad \frac{d}{dx}(\sec x) = \sec x \tan x$$

$$\frac{d}{dx}(\cot x) = -\csc^2 x \quad \frac{d}{dx}(\csc x) = -\csc x \cot x$$

### Exponential and Logarithmic Functions

$$\frac{d}{dx}e^x = e^x \quad \frac{d}{dx}\ln x = \frac{1}{x}$$

$$\frac{d}{dx}a^x = a^x \ln a \quad \frac{d}{dx}(\log_a x) = \frac{1}{x \ln a}$$

### Inverse Trigonometric Functions

$$\frac{d}{dx}(\sin^{-1} x) = \frac{1}{\sqrt{1-x^2}} \quad \frac{d}{dx}(\cos^{-1} x) = -\frac{1}{\sqrt{1-x^2}}$$

$$\frac{d}{dx}(\tan^{-1} x) = \frac{1}{1+x^2} \quad \frac{d}{dx}(\sec^{-1} x) = \frac{1}{|x|\sqrt{x^2-1}}$$

$$\frac{d}{dx}(\cot^{-1} x) = -\frac{1}{1+x^2} \quad \frac{d}{dx}(\csc^{-1} x) = -\frac{1}{|x|\sqrt{x^2-1}}$$

### Hyperbolic Functions

$$\frac{d}{dx}(\sinh x) = \cosh x \quad \frac{d}{dx}(\cosh x) = \sinh x$$

$$\frac{d}{dx}(\tanh x) = \text{sech}^2 x \quad \frac{d}{dx}(\text{sech } x) = -\text{sech } x \tanh x$$

$$\frac{d}{dx}(\coth x) = -\text{csch}^2 x \quad \frac{d}{dx}(\text{csch } x) = -\text{csch } x \coth x$$

### Inverse Hyperbolic Functions

$$\frac{d}{dx}(\sinh^{-1} x) = \frac{1}{\sqrt{1+x^2}} \quad \frac{d}{dx}(\cosh^{-1} x) = \frac{1}{\sqrt{x^2-1}}$$

$$\frac{d}{dx}(\tanh^{-1} x) = \frac{1}{1-x^2} \quad \frac{d}{dx}(\text{sech}^{-1} x) = -\frac{1}{x\sqrt{1-x^2}}$$

$$\frac{d}{dx}(\coth^{-1} x) = \frac{1}{1-x^2} \quad \frac{d}{dx}(\text{csch}^{-1} x) = -\frac{1}{|x|\sqrt{1+x^2}}$$

### Parametric Equations

If  $x = f(t)$  and  $y = g(t)$  are differentiable, then

$$y' = \frac{dy}{dx} = \frac{dy/dt}{dx/dt} \quad \text{and} \quad \frac{d^2y}{dx^2} = \frac{dy'/dt}{dx/dt}.$$

# Derivatives and Differential Equations

A **first-order differential equation** is an equation

$$\frac{dy}{dx} = f(x, y) \tag{1}$$

in which  $f(x, y)$  is a function of two variables defined on a region in the  $xy$ -plane. The equation is of *first order* because it involves only the first derivative  $dy/dx$  (and not higher-order derivatives). We point out that the equations

$$y' = f(x, y) \quad \text{and} \quad \frac{d}{dx}y = f(x, y)$$

are equivalent to Equation (1) and all three forms will be used interchangeably in the text.

In this course, most likely:

$$\begin{aligned} \frac{dy}{dx} &= f(x) \\ \Rightarrow y &= \int f(x) \, dx \end{aligned}$$