

Advanced Classical Mechanics (PHYS 615)
Extra Credit Project 1 (due on date of Exam I)

Fall 2024

The points obtained in this project will be added towards Exam I. To obtain full credit, your computer source code + calculations done on paper **MUST** be provided along with the plots.

During our second lecture, we discussed projectile motion with drag force of the form $F = -kmv$ from which the projectile flight time and range were derived

$$T = \frac{kV + g}{gk}(1 - e^{-kT}), \quad R = \frac{U}{k}(1 - e^{-kT})$$

where k is a drag coefficient and $V \equiv v_0 \sin \theta$ and $U \equiv v_0 \cos \theta$ are the initial y- and x-components of the projectile speed. The time of flight is a transcendental equation and can't be solved analytically. Two methods for solving this problem are (i) *Numerical* and (ii) *Perturbative*

Numerical Approach: **(20 pts)** To solve a transcendental equation, we use *Newtons's Method* numerical approach which is a root-finding algorithm, which has the recurring form:

$$x_{n+1} = x_n - \frac{f(x_n)}{f'(x_n)}$$

where x_{n+1} is the root after n -recursive steps. To use this method, the function must be of the form $f(x) = 0$ and it requires an initial guess of the root x that satisfies the equation. In our case, for a given value of k , we find the root T , and used the (k, T) pair as input into the range equation.

Solve the transcendental equation using Newton's Method.

- step 1: need to put the equation in the form: $f(k, T) = 0$

- **step 2:** for an arbitrary value of $k \leq 0.08$, find the root T , as follows

$$T_{n+1} = T_n - \frac{f(k, T_n)}{df(k, T_n)/dT}$$

Use the initial conditions to get the iteration process going, and try using $n=100$ iterations

initial conditions ($n=0$):

$$v_0 = 600 \text{ m/s}$$

$$\theta = 60^\circ$$

$$T_0 = 150 \text{ sec (initial guess will not necessarily satisfy } f(k, T_0) = 0)$$

after n -iterations, the algorithm should have found the root that satisfies the equation $f(k, T) = 0$,

-**step 3:** now that you have a (k_i, T_i) pair for an i -th arbitrary value of k , use it as input into the range equation, $R_i(k_i, T_i)$

-**step 4:** at this point, you should have a procedure established for determining the values (k_i, T_i, R_i) ; now repeat the procedure for a range of k -values from 0.0000001 to 0.08 with step-size of $dk = 0.001$; ** note: since k is in the denominator, $k \neq 0$

-**step 5:** make a plot of R vs. k to examine how the projectile range varies with the drag coefficient ** note: negative ranges are unphysical, therefore, when making the plots, set the lower limit on the y-axis to zero.
Add proper labels and a legend to each of the plots.

Perturbative Approach: **(10 pts)** Recall, during lecture we applied a Taylor expansion to the time-of-flight (T) and range (R) using k as the *expansion parameter* since it is usually small and can be considered a perturbation. The series was truncated for powers of k^n , $n \geq 2$ only keeping the linear term:

$$R' \approx R \left(1 - \frac{4kV}{3g} \right)$$

where $R \equiv 2UV/g$ (projectile range without drag)

Let's now attempt to go one step further and include a *quadratic term* (k^2) to check how well the perturbative approach using both a *linear* and *quadratic* term agrees with the numerical approach.

To keep the quadratic term k^2 in the range expansion, let's begin with:

$$R' \approx \frac{U}{k} \left(kT - \frac{1}{2}k^2T^2 + \frac{1}{6}k^3T^3 - \dots \right)$$

** notice that the k in the denominator cancels with a common factor k

Now substitute the approximate expression for the flight time (which only kept k to a linear term) to the expression above

$$T \approx \frac{2V}{g} \left(1 - \frac{kV}{3g} \right)$$

keeping only up to k^2 terms, and re-arrange the equation to show that

$$R' \approx R \left[1 - 4 \left(\frac{kV}{3g} \right) + 9 \left(\frac{kV}{3g} \right)^2 \right]$$

Now, to check the perturbative method versus the numerical method, make a plot overlay of: i) R' vs. k (*perturbative **linear** approximation*),

ii) R' vs. k (*perturbative **quadratic** approximation*),

iii) R vs. k (***numerical** approximation*)

Add proper labels and a legend to each of the plots.

Write a brief paragraph commenting on your results. Is it what you expected to happen? Elaborate.