- 一、实验目的
- 1、熟悉 GPS 星历
- 2、利用 GPS 星历计算卫星位置
- 二、实验准备
- 1、台式电脑每人一台。
- 2、Visual studio2010 软件提前安装。
- 三、实验步骤
- 1、设置卫星位置计算的界面:

🔛 卫星在轨位置计算	
读取星历文件:	浏览
aO(时间偏差):	
al(频率偏差系数):	卫星地固坐标系下的坐标:
a2(漂移系数):	. [
Te(星历参考历元):	x:
IODE(星历数据龄期):	у:
a的开方(长半轴平方根):	
e(偏心率):	z:
i(轨道倾角):	
ω(近地点角距):	计算
Ω0(参考历元的升交点赤经):	
M(平近点角):	
△n(平均角速度误差):	
Ω dot ( 升交点赤经变化率 ):	
i dot(轨道倾角变化率):	
Cus(纬度幅角的正弦振幅):	
Cuc(纬度幅角的余弦振幅):	
Cis(轨道倾角的正弦振幅):	
Cic(轨道倾角的余弦振幅):	
Crs(轨道半径的正弦振幅):	
Crc(轨道半径的余弦振幅):	
GPD(GPS星期数):	
Tgd(电离层延迟误差):	
IODC(星钟数据龄期):	
卫星精度:	
卫星健康:	

2、运用课本里面 48-50 页的公式,完成卫星位置的计算。

参考代码如下:

using System;

using System.Collections.Generic;

 ${\color{red} using \ System.} Component Model;$ 

using System.Data;

using System.Drawing;

using System.Text;

using System.Windows.Forms;

```
using System.IO;
namespace SatellitePositionComputation
    public partial class Form1 : Form
         float u = float.Parse("0.3986005e+15");
         public Form1()
         {
              InitializeComponent();
         }
         private void button1_Click(object sender, EventArgs e)
              openFileDialog1.DefaultExt = "txt";
              openFileDialog1.Filter = "文本文件 (*.txt)|*.txt";
              openFileDialog1.FileName = "";
              //openFileDialog1.InitialDirectory = "c:\\";
              openFileDialog1.Title = "打开星历文件";
              if (openFileDialog1.ShowDialog()==System.Windows.Forms.DialogResult.OK )
                  try
                  {
                       textBox1.Text = openFileDialog1.FileName;
                       if (File.Exists(openFileDialog1.FileName))
                       {
                            string[] gData = File.ReadAllLines(openFileDialog1.FileName,
Encoding.Default);
                            a0.Text = gData[1];
                            a1.Text = gData[2];
                            a2.Text = gData[3];
                            te.Text = gData[4];
                            iode.Text = gData[5];
                            aRoot.Text = gData[6];
                            e0.Text = gData[7];
                            i0.Text = gData[8];
                            w0.Text = gData[9];
                            ascendingNode.Text = gData[10];
                            M.Text = gData[11];
                            meanMotionD.Text = gData[12];
                            RateascendingN.Text = gData[13];
                            Ratei.Text = gData[14];
                            cus.Text = gData[15];
```

```
cuc.Text = gData[16];
                    cis.Text = gData[17];
                    cic.Text = gData[18];
                    crs.Text = gData[19];
                    crc.Text = gData[20];
                    gpd.Text = gData[21];
                    tgd.Text = gData[22];
                    iodc.Text = gData[23];
                    psatellite.Text = gData[24];
                    hsatellite .Text = gData[25];
               }
          }
          catch
          {
               MessageBox.Show("读取文本文件出现错误!");
               Console.WriteLine("The process failed: {0}", e.ToString());
          }
     }
}
private void comp_Click(object sender, EventArgs e)
{
     try
     {
          //mean angular speed n
          \label{eq:double_n} \begin{array}{ll} \textbf{double} & n = Math.Pow(u,0.5)/Math.Pow(float.Parse(aRoot.Text~),3); \end{array}
          n = n + double.Parse(meanMotionD.Text);
          //time tk
          double tk = 0;
          //mean anomaly Mk
          double Mk = double.Parse(M.Text )+n*tk;
          //Eccentric anomaly Ek
          double Ek=Mk;
          double Ekt =0;
          while(Math.Abs( Ek -Ekt) >0.000000000001)
          {
               Ekt = Ek;
               Ek = Mk + double.Parse(e0.Text)*Math.Sin(Ekt);
          }
          //true anomaly Vk
          double e00 = double.Parse(e0.Text );
```

```
double Vk =
Math.Atan(Math.Pow(1-Math.Pow(e00,2),0.5)*Math.Sin(Ek)/(Math.Cos(Ek)-e00));
                   //argument of satellite and ascension node
                   double Qk = Vk + double.Parse(w0.Text );
                   //diturbed satellite motion correction term
                   double Cu = double.Parse(cuc.Text) *
Math.Cos(2*Qk)+double.Parse(cus.Text)*Math.Sin(2*Qk);
                   double Cr = double.Parse(crc.Text) *
Math.Cos(2*Qk)+double.Parse(crs.Text)*Math.Sin(2*Qk);
                   double Ci = double.Parse(cic.Text) *
Math.Cos(2*Qk)+double.Parse(cis.Text)*Math.Sin(2*Qk);
                   //argument of latitude
                   double Uk = Qk + Cu;
                   double Rk = Math.Pow(double.Parse(aRoot.Text),2 )*(1-e00*Math.Cos(Ek ))+Cr;
                   double Ik = double.Parse(i0.Text )+double.Parse (Ratei.Text )*tk +Ci ;
                   //orbit plane coordinate
                   double OrbitX = Rk * Math.Cos(Uk);
                   double OrbitY = Rk * Math.Sin(Uk);
                   //longitude of ascension node
                   double we = double.Parse("7.29211567E-5");
                   \label{eq:double_double} \begin{aligned} & \textit{double}. Parse (ascendingNode.Text) + (\textit{double}.Parse (RateascendingN.Text) - \end{aligned}
we) * tk - we * double.Parse(te.Text);
                   //Earth coordinate
                   double EarthX= OrbitX*Math.Cos(W)-OrbitY*Math.Cos(Ik)*Math.Sin(W);
                   double EarthY= OrbitX*Math.Sin(W)+OrbitY*Math.Cos(Ik)*Math.Cos(W);
                   double EarthZ = OrbitY * Math.Sin(Ik);
                   xcoord.Text = EarthX.ToString();
                   ycoord.Text = EarthY.ToString();
                   zcoord.Text = EarthZ.ToString();
                   //
              }
              catch
                   Console.WriteLine("此操作失败:{0}",e.ToString());
              }
          }
```