# Captain: Online Multi-Goal Adaptation for Autonomous Unmanned Systems via Optimal Control

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#### A APPENDIX

### A.1 UUV Oceanic Surveillance

This example originates from [3, 4]. Rather than considering single-goal optimization for the UUV scenario in [3, 4], we extended it to achieve multiple dynamic goals under uncertainties and disturbance through elastic goal satisfaction, while other configurations like sensors are kept the same as [3]. The goals include:

- $G_l$ : A segment of surface over a distance of  $L_t = 100$  km is expected to be examined by the UUV within  $\Delta = 10$  hours, while the distance threshold is L = 90 km.
- G<sub>e</sub>: A total amount of energy E<sub>t</sub> = 5.4 MJ is expected to be consumed, while the maximum amount of energy is E = 6 MJ.
- $G_{\varphi}$ : The accuracy of sensor measurements is targeted at  $A_t = 90\%$ , while the accuracy threshold is set as A = 80%.

The UUV is equiped with 5 sensors for ocean surveillance. The scanning time 10 hours is 360 time instance,  $x_i, i \in [1,5]$  is the portion of time the sensor i should be used during system operation in each instance.  $Acc_i$  is the accuracy of sensor i;  $E_i$  is the energy consumed by sensor;  $V_i$  is the scanning speed of sensor.  $q_i$  is portion of accuracy of sensor and  $p_i$  is for scanning speed respectively in decimals. The energy consumed is related with working accuracy and speed of sensor. The corresponding measures are listed as follows:  $X_l = \sum_{k=0}^T \sum_{i=0}^N x_i q_i V_i \tau$ ,  $X_e = \sum_{k=0}^T \sum_{i=0}^N x_i E_i \cdot \frac{e^{p_i + q_i} - 1}{e^2 - 1} \tau$ , and  $X_{\varphi} = \sum_{k=0}^T \sum_{i=0}^N x_i p_i Acc_i$ , where T = 360, i.e., adaptations is performed every 100 surface measurements of the UUV state, and the time instance k incremented by  $1 \sim 100$ . The goal satisfactions are listed as follows:

$$DS^{2}(X_{l}) = \begin{cases} 1, & X_{l} \geq L_{t} \\ \frac{X_{l} - L}{L_{t} - L}, & L \leq X_{l} < L_{t} \\ 0, & X_{l} < L \end{cases}$$

$$DS^{1}(X_{e}) = \begin{cases} 1, & X_{e} \leq E_{t} \\ \frac{E - X_{e}}{E - E_{t}}, & E_{t} < X_{e} \leq E \\ 0, & X_{e} > E \end{cases}$$

$$DS^{2}(X_{\varphi}) = \begin{cases} 1, & X_{\varphi} \geq A_{t} \\ \frac{X_{\varphi} - A}{A_{t} - A}, & A \leq X_{\varphi_{k}} < A_{t} \\ 0, & X_{\varphi} < A \end{cases}$$

The Figure 1 shows the goal adaptation process of the UUV during operation with the case in [3]. As the capability of sensor reconfiguration increase, UUV could tune the sensor usage deliberately. At k=100 we change the available energy change from 5.4 to 5.0 MJ, , at k=160 we change the distance to be scanned from 100 to 105 km. The plots show that these changes in requirements lead to corresponding changes in the arrangement of sensor usage, as the

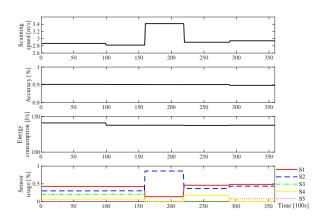


Figure 1: UUV adaptation optimal planning results.

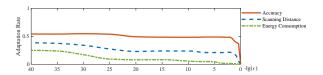


Figure 2: UUV case (# of incidents = 9)

time portion for S2 increases. Figure 1 also shows how Captain reacts to changes in sensor parameters and sensor failures. At k=220, the measurement accuracy of sensor S3 drastically decreases from 83% to 43%, at k=290, S4 stops working, and leads to switching the sensors to the optimal solution that S1 is more exploited. At last, the mission end with average measurement accuracy at 90.1%, scanning distance at 106.7 km, energy consumption at 4.98 MJ.

In this scenario, the three methods are compared with their performance by adding random failures to parameters of sensors, i.e., sensor accuracy, scanning speed and energy consumption. For each method, we simulated the UUV with different number of random disturbances added to certain time instants. For each configuration, we ran the simulation for 500 rounds and computed the average accuracy, scanning distance and energy consumption. The results are shown in Table 1. Captain outperforms the other two strategies at the goal satisfaction overall and is more robustness when sensor failure increases. The variance of scanning distance and energy consumption of three strategies are compared in Figure 3(b) and Figure 3(c). The choice of different violation tolerance  $\epsilon$  and the performance scanning accuracy achievement are illustrated in Figure 3(a). As the number of incidents increases, the impact of  $\epsilon$  on goal satisfaction is more obvious.

As discussed in the paper, the choice violation tolerance  $\epsilon$  determine which goal needs relaxation, while the larger the value is, the less number of goals in the *Violated Goal Set*, along with the underlying higher risk of no feasible solution. Thus the value of  $\epsilon$  should be determined based on experimental data. Figure 2 illustrates adaptation rate of each soft goal in UUV case, as the increase of violation tolerance. In the UUV case, we choose  $\epsilon_{\varphi,e,l} = \{10^{-3},0,0\}$ , when the adaptation rate of each goal tends to be gentle.

## A.2 UAV Delivery Scenario

Then, we will illustrate how to apply goal satisfaction modeling into the case of UAV delivery system based on goals illustrated in the paper.

• Safety: The indicator to evaluate the safety requirement is the collision risk of UAV during the flight. Supposing that the obstacles detected by the UAV at time instant k is  $O_k$ , while the current state of UAV is  $s_k$ . Thus, the QM of safety is  $X_{S_{o,k}} = \frac{\|\mathbf{x}_k - \mathbf{x}_o\|_2 - r_a - r_o}{D_o}$ ,  $\forall o \in O_k$ . Such that the average distance between UAV and the center of obstacle reflects the safety risk.

$$DS^2(\mathcal{X}_{S_{o,k}}) = \begin{cases} 1, & \mathcal{X}_{S_{o,k}} \geq 1 \\ 0, & \mathcal{X}_{S_{o,k}} < 0 \\ \mathcal{X}_{S_{o,k}}, & otherwise \end{cases}$$

• Timelines: The total traveling time from time instant i to j is denoted as  $\xi_{ij} = \sum_{k=i}^{j-1} \frac{\|\mathbf{x}_{k+1} - \mathbf{x}_k\|_2}{v_k}$ . The indicator of timeliness is  $\mathcal{X}_{\xi} = \xi_{0T}$ , the degree of satisfaction of the timeliness requirement of the whole trajectory  $DS_{\xi}$  is:

$$DS^{1}(\mathcal{X}_{\xi}) = \begin{cases} 1, & \mathcal{X}_{\xi} \leq \Delta_{t} \\ \frac{\Delta - \mathcal{X}_{\xi}}{\Delta - \Delta_{t}}, & \Delta_{t} < \mathcal{X}_{\xi} \leq \Delta \\ 0, & \mathcal{X}_{\xi} > \Delta \end{cases}$$

• Accuracy: The average quality of the information collected during the mission is denoted as  $X_{\varphi} = \frac{1}{\xi_{0T}} \sum_{k=0}^{T-1} \|\omega\| \tau$ , the degree of satisfaction is  $DS_{\varphi}$  is:

$$DS^{2}(X_{\varphi}) = \begin{cases} 1, & X_{\varphi} \ge A_{t} \\ \frac{X_{\varphi} - A}{A_{t} - A}, & A \le X_{\varphi} < A_{t} \\ 0, & X_{\varphi} < A \end{cases}$$

• Energy-saving: The total energy consumption from time instant i to j is denoted as  $e_{ij} = \sum_{k=i}^{j-1} \|x_{k+1} - x_k\|_2 + \eta_1 \cdot \|v_{k+1} - v_k\|_2 + \eta_2 \cdot \|\omega_k\| \tau$ . The indicator of energy consumption is  $X_e = e_{0T}$ , the degree of satisfaction of energy requirement  $DS_e$  is:

$$DS^{1}(\mathcal{X}_{e}) = \begin{cases} 1, & \mathcal{X}_{e} \leq E_{t} \\ \frac{E - \mathcal{X}_{e}}{E - E_{t}}, & E_{t} < \mathcal{X}_{e} \leq E \\ 0, & \mathcal{X}_{e} > E \end{cases}$$

A.2.1 Implementation in UAV delivery system. The whole process of applying goal adaptation method into UAV trajectory planning at each time instant k is shown in the algorithm 1.

**Algorithm 1:** Goal Adaptation in Trajectory Planning for UAV at time instant k

Input: Current state  $s_k$ , prediction horizon N, detected obstacles  $O_k$ , detected privacy regions  $C_k$ , and criteria  $\phi$  for iteration stopping. Requirement specifications: preferred travel time  $\Delta_t$ , travel time budget  $\Delta$ , preferred energy consumption  $E_t$ , energy budget  $E_t$ , preferred information quality  $A_t$ , information quality budget  $E_t$ , preferred non-conflict distance  $D_0$ ,  $D_c$ .

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Output: Action plan < v_k, \omega_k...v_{k+N}, \omega_{k+N} >
 1 while \phi = false do
         Initialize v_k...v_{k+N}, \omega_k...\omega_{k+N}, slash variables \varepsilon, \delta;
         CurConstr \leftarrow \emptyset;
         for i \in \{S, P, \xi, e, \varphi\} do
               X_i \leftarrow Obs_k(s, G_i);
               Constr_i \leftarrow SlashConstr(q_i, ub_i, lb_i, \delta_i, \varepsilon_i)
                 CurConstr = CurConstr \cup Constr_i;
         Construct Goal Satisfaction Checking problem P<sub>1</sub> with
           CurConstr;
          (\boldsymbol{\varepsilon}^*, \boldsymbol{\delta}^*, \boldsymbol{v}^*, \boldsymbol{\omega}^*) \leftarrow \text{Solve}(P_1);
         /*Solve the problem with Sequential Convex
           Programming*/;
         if (v^*, \omega^*) \neq \emptyset then
10
               for i \in \{S, P, \xi, e, \varphi\} do
11
                VD_i \leftarrow (\delta_i^*, \varepsilon_i^*);
12
               G_v, G_{nv} \leftarrow \text{GoalAnalyze}(\{VD_i, \epsilon_i\});
13
               for G_i \in G_{nv} do
14
                     GoalConstr \leftarrow ConstrConstruct(s, g_i);
15
                    CurConstr = CurConstr \cup GoalConstr;
16
               for G_i \in G_v do
17
                    HardConstr \leftarrow ConstrConstruct(s, lb_i, ub_i);
                    HardConstr = HardConstr \cup GoalConstr:
19
               Construct Goal Satisfaction Optimization problem
                 P_2 with current G_v, CurConstr, HardConstr;
               (\boldsymbol{v}_2^*, \boldsymbol{\omega}_2^*) \leftarrow \text{Solve}(P_2);
               if v_2^*, \omega_2^* \neq \emptyset then (v^*, \omega^*) \leftarrow (v_2^*, \omega_2^*);
22
23
         if \phi = true \wedge v^*, \omega^* = \emptyset then
24
               Report false and act some emergent actions, such as
                 emergency landing;
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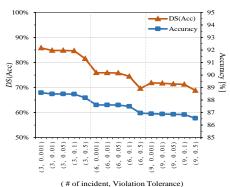
*A.2.2 Robustness of* Captain. In this section, we show the details of the experimental results in the UAV delivery scenarios discussed in the paper.

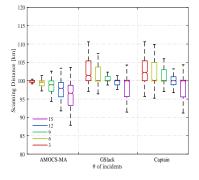
To test the robustness of Captain against the changes in the requirements and system behaviors, we randomly added several disturbances (up to 10% of target values) to the soft goals and system parameters at different time instants during the motion of the UAV. The simulation results are shown in Table 3, where the quantity of incidents denotes the adding of disturbances, and the values

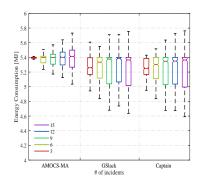
Table 1: Goal satisfaction results of UUV case

Goals Strategies		Accuracy [%]			Scanning Distance [km]			Energy Consumption [MJ]			Adaptation
		AMOCS-MA	GSlack	Captain	AMOCS-MA	GSlack	Captain	AMOCS-MA	GSlack	Captain	Rate
# of incidents	3	88.26%	95.30%	95.84%	96.64%	99.72%	99.79%	99.03%	99.65%	99.65%	34.33%
		88.85	89.53	89.58	99.93	103.91	104.23	5.34	5.24	5.24	34.33%
	6	74.84%	82.32%	83.40%	89.38%	97.76%	97.80%	97.89%	99.51%	100%	55.33%
		87.50	88.24	88.35	99.16	103.38	103.97	5.33	5.19	5.18	33.33%
	9	57.78%	67.68%	69.53%	79.03%	93.66%	93.37%	96.14%	99.42%	100%	71.67%
		85.78	86.77	86.96	98.2	103.04	103.49	5.31	5.16	5.14	/1.0/%
	12	43.77%	49.57%	54.12%	64.86%	85.14%	86.26%	92.29%	99.10%	98.73%	90 920
		84.38	84.96	85.41	96.64	100.50	100.95	5.34	5.17	5.16	80.83%
	15	24.05%	25.78%	31.39%	44.15%	70.84%	69.51%	90.28%	100%	99.99%	80.87%
		82.41	82.58	83.19	94.49	98.81	98.86	5.34	5.18	5.16	00.07%

<sup>&</sup>lt;sup>a</sup> The violation tolerance are  $\epsilon_{\varphi}=10^{-3}, \epsilon_{e}=0, \epsilon_{l}=0$ 







(a) Scanning accuracy with various incidents and  $\epsilon$ . (Captain)

(b) Variance of scanning distance.

(c) Variance of energy consumption.

Figure 3: Scanning accuracy, distance and energy consumption with various probability of incidents and violation tolerance.

represent the degree of related goal's satisfaction, i.e., the values of  $DS^j(\mathcal{X}_i)$ . From Table 3, in the case of both requirement changes and system parameter changes, we can find that the increase of the frequency of introduced disturbances in the flight task can lead to the decrease of goal satisfaction degrees. However, Captain shows higher degrees of goal satisfaction and lower variance against AMOCS-MA and GSlack.

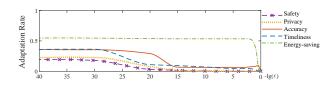


Figure 4: Goald adaptation rate in UAV case ( $\rho_o$  = 2%,  $\rho_p$  = 2%)

A.2.3 Scalability of Captain. To demonstrate the scalability of Captain with respect to different environments, we further simulated the UAV case on two selected real urban environments from the open building dataset of Portland in USA [1]. We used ArcGIS map to set up a 3D model based on the method in [2]. Their original

Table 2: Statistics on Overhead Data.

Case Study	Strategies	Average [s]	Standard Deviation [s]		
	AMOCS-MA	0.2039	0.0639		
UAV	GSlack	0.0991	0.1147		
	Captain	0.1301	0.1439		
	AMOCS-MA	0.0423	0.0092		
UUV	GSlack	0.0180	0.0078		
	Captain	0.0344	0.0217		

spaces are  $500 \times 500 \times 100 m^3$  and  $10^3 \times 10^3 \times 100 m^3$ , and compressed into  $50 \times 50 \times 10 m^3$  and  $100 \times 100 \times 10 m^3$ , respectively. In the dataset, we can also obtain the longitude and latitude of the center of each building, as well as its average height and building types (i.e., industrial and commercial buildings, houses and apartments for living). The buildings for industrial and commercial use are viewed as obstacles, while houses and apartments are viewed as private regions. We used ArcGIS map to set up a 3D model according to the method introduced in [2]. As shown in Figure 5, the process of real urban scenarios modeling is explained. We set the range of longitude and latitude of working space on the ArcGIS map, and all the building are marked as blue (Figure 5(a)). Then the ArcGIS map

Table 3: Goal Satisfaction Results of the UAV with Dynamic Disturbances in Requirements and System Parameters

Goals			Accuracy (%)			Traveling Time (s)			Energy Consumption (unit)			Adaptation
Strategies		AMOCS-MA	GSlack	k   Captain	AMOCS-MA	GSlack	Captain	AMOCS-MA	GSlack	Captain	Rate	
# of incidents	Requirement Changes	2	92.18% 89.39	97.18% 90.08	97.36% 90.27	95.10% 13.96	98.20% 11.96	98.85% 11.44	75.36% 23.70	86.21% 21.47	86.82% 21.35	16.84%
		4	91.37% 89.20	95.95% 89.82	96.35% 90.11	93.74% 13.51	97.94% 11.59	98.37% 11.15	73.97% 23.44	84.47% 21.28	85.04% 21.17	17.67%
		6	89.25% 89.14	94.33% 89.77	95.09% 89.95	92.42% 13.57	97.00% 11.78	98.15% 11.22	71.67% 23.64	81.89% 21.38	82.95% 21.23	19.12%
	System Parameter Changes	2	87.49% 88.75	92.83% 89.53	93.75% 89.75	98.55% 14.71	99.71% 12.83	99.89% 11.74	75.21% 24.96	84.94% 22.98	89.34% 22.08	19.07%
		4	84.24% 88.42	91.83% 89.32	92.79% 89.49	97.08% 14.84	99.91% 12.80	99.76% 11.52	73.03% 25.39	83.67% 23.26	88.41% 22.30	21.40%
		6	78.15% 87.82	86.32% 88.79	87.62% 88.97	96.12% 14.87	99.00% 12.70	99.89% 11.56	72.30% 25.54	83.47% 23.30	87.97% 22.40	26.28%

<sup>&</sup>lt;sup>a</sup> The violation tolerance is  $\epsilon_{SR,PR,\phi,\xi,e} = \{10^{-20},10^{-20},10^{-10},10^{-20},0.005\}$  with environment setting of  $\rho_o = 2\%, \rho_p = 2\%$ 

Table 4: Goal adaptation results for different scale of environment

Scale		Accuracy [%]	Traveling Time [s]	Energy Consumption [unit]	Safety Risk	Privacy Risk	Real-time Performance	
	Soft Goals	90	60	100	0	0	Adaptation Rate	4
50	Hard Constraints	80	90	150	1 0	1 0	6/120 Overhead	43
50	Results	90	60.00	105.57				4
	Satisfaction	100%	100%	88.85%	100%	100%	Avg. 0.066s Std. 0.139s	4
100	Soft Goals	90	90	200	0	0	Adaptation Rate	-
	Hard Constraints	80	150	300	1	1	3/203	4
	Results	90	101.50	210.94	0	0.1534	Overhead	4
	Satisfaction	100%	80.83%	89.06%	100%	99.98%	Avg. 0.132s Std. 0.180s	43

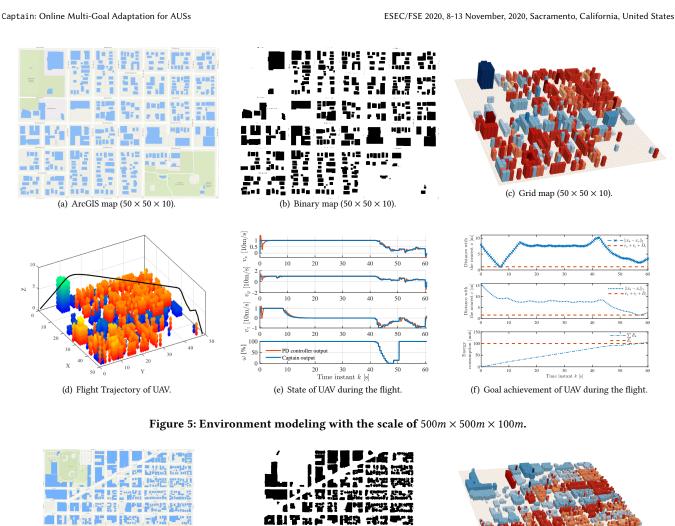
is turned into a binary map. Thus a 3D model of the urban scene is set up with the scale of  $500m \times 500m \times 100m$  in Figure 5(c).

In each case, the flight task of UAV is to travel from the position [0, 0, 0] to the destination [49, 49, 0] and [99, 99, 0] respectively, within the budget of accuracy, time and energy, as well as minor safety and privacy risk. The trajectory, state transition and goal achievement of the UAV at each time instant are shown in Figure 5(d), Figure 5(e) and Figure 5(f). So as to the scale of  $1000m \times 1000m \times 100m$  in Figure 6. Table 4 summarizes the goal adaptation results in these two settings. we find that the states generated from Captain and the PD controller are almost the same, indicating that the planning results of Captain can be translated by the PD controller and executed by the UAV effectively.

From our simulation results, in Setting 2, the flight task is completed in 60.07 s, consuming 106.27 units of energy without safety and privacy risk. In Setting 3, the task is finished in 101.5 s, consuming 210.94 units of energy. It does not have safety risk, but gets 4 points along the trajectory where the soft goal of privacy-preserving is violated. This results in an average 99.98% goal satisfaction along the path. Captain has high scalability as the two important steps Goal Satisfaction Checking and Goal Satisfaction Optimization are solved by SQP, which can handle large-scale optimization problems. In contrast, AMOCS-MA fails to compute a motion plan at real-time when the size of workspace increases.

A.2.4 Real-time performance of Captain. Finally, we analyze the cost to compute the optimal motion and sensor configuration plans for the given case studies. From Table 2, the gap of average computation time between AMOCS-MA and Captain is about 7 ms (0.0423 s versus 0.0344 s) in the UUV case, and 70 ms (0.2039 s versus 0.1301 s) in the UAV case. This is because AMOCS-MA considers the optimization of all goals while Captain only considers the determined violated goals. So AMOCS-MA requires more computations especially when there are more goals to be considered. Captain has larger variance since the number of goals to be optimized may vary at runtime.

Fig. 4 illustrates adaptation rate of each soft goal in UAV case, as the increase of violation tolerance. In the experiments before, we choose  $\epsilon_{S,P,\varphi,\xi,e} = \{10^{-20}, 10^{-20}, 10^{-10}, 10^{-20}, 0.005\}$  in the UAV case, when the adaptation rate of each goal tends to be gentle.

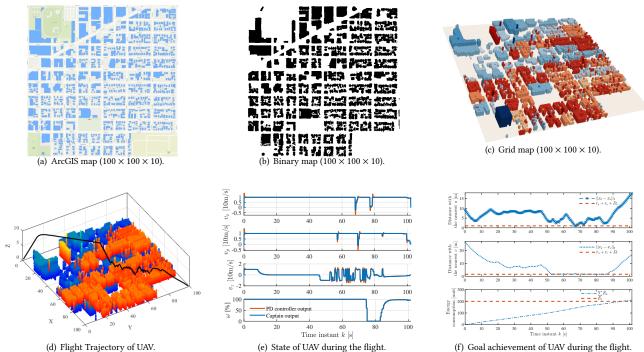


Figure 6: Environment modeling with the scale of  $1000m \times 1000m \times 1000m$ .

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