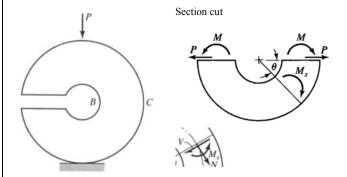
1) 9.7

9.7. The curved beam in Figure P9.7 has a circular cross section 50 mm in diameter. The inside diameter of the curved beam is 40 mm. Determine the stress at B for P = 20 kN.



sigma[theta, theta] :=
$$\frac{N}{A} + \frac{M(A - r \cdot A[m])}{A \cdot r(R \cdot A[m] - A)}$$

$$\sigma_{\theta, \theta} := \frac{N}{A} + \frac{M(-rA_m + A)}{Ar(RA_m - A)}$$
(1)

 $P := 20 \cdot 1000 \, \#N$

$$P := 20000$$
 (2)

$$A[m] := evalf(2 \cdot Pi \cdot (45 - sqrt(45^2 - 25^2)))$$

$$A_m := 47.64807164$$
(3)

$$A := evalf\left(\text{Pi}\left(\frac{10}{2}\right)^{2}\right) \# mm^{2}$$

$$A := 78.53981635$$
(4)

$$Rm := 40 + \frac{(50 - 40)}{2}$$

$$Rm := 45 \tag{5}$$

$$r := 20$$

$$r := 20 \tag{6}$$

#Sum force

$$F[x] := 0$$

$$F_{\mathbf{r}} := 0 \tag{7}$$

N := -P #Sum F

$$N := -20000 \tag{8}$$

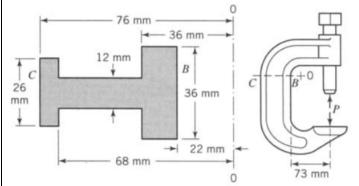
$$M := -P \cdot (24)$$

$$M := -480000 \tag{9}$$

sigma[theta, theta] :=
$$\frac{N}{A} + \frac{M(A - r \cdot A[m])}{A \cdot r(R \cdot A[m] - A)}$$

$$\sigma_{\theta, \theta} := -560.2253996 \tag{10}$$

2) 9.20-9.21. Ignore Bleich factors (section 9.4 is not covered in this course) **9.20.** A load P = 12.0 kN is applied to the clamp shown in Figure P9.20. Determine the circumferential stresses at points B and C, assuming that the curved beam formula is valid at that section.



Assumption: that the Neutral Axis and the centroid are the same distance form the axis of curvature.



$$yBar := \frac{\operatorname{Sigma} A \cdot y}{\operatorname{Sigma}[0] \cdot A[i]}$$

$$yBar := \frac{\sum A y}{\sum_{0} A_{i}}$$
 (1)

$$A[1] := (36 - 22) \cdot 36$$

$$A_1 := 504 \tag{2}$$

$$A[2] := 12 \cdot (68 - 36)$$

$$4_2 := 384$$
 (3)

$$A[3] := (76 - 68) \cdot 26$$

$$4_2 := 208 \tag{4}$$

$$yBar := evalf\left(\frac{A[1] \cdot (29) + A[2] \cdot (52) + A[3] \cdot (72)}{A[1] + A[2] + A[3]}\right)$$
$$yBar := 45.21897810$$
 (5)

R := yBar

$$R := 45.21897810 \tag{6}$$

$$r[B] := 22$$

$$r_B := 22 \tag{7}$$

$$r[C] := 76$$

$$r_C := 76 \tag{8}$$

$$Am[1] := evalf\left(36 \cdot \ln\left(\frac{36}{22}\right)\right)$$

 $Am_1 := 17.72915346$

$$Am_1 := 17.72915346 (9)$$

$$Am[2] := evalf\left(12 \cdot \ln\left(\frac{68}{36}\right)\right)$$
$$Am_2 := 7.631865202$$

$$\frac{\ln\left(\frac{36}{36}\right)}{36}\right)$$

$$Am_2 := 7.631865202$$
(10)

$$Am[3] := evalf\left(26 \cdot \ln\left(\frac{76}{68}\right)\right)$$

$$Am := Am[1] + Am[2] + Am[3]$$

$$Am := 28.25288518 \tag{12}$$

(13)

$$A[tot] := A[1] + A[2] + A[3]$$

 $A_{tot} := 1096$

#Sum of Forcs

 $P := 12 \cdot 1000$

$$P \coloneqq 12000 \tag{14}$$

N := P

$$N := 12000 \tag{15}$$

 $M := P \cdot (73 + 22)$

$$M := 1140000 \tag{16}$$

sigma[theta, theta
$$B$$
] := $\frac{N}{A[tot]} + \frac{M(A[tot] - r[B] \cdot Am)}{A[tot] \cdot r[B] \cdot (R \cdot Am - A[tot])}$

$$\sigma_{\theta, \theta, B} := 11.20930192$$
 (17)

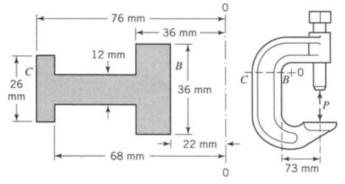
sigma[theta, theta
$$C$$
] :=
$$\frac{N}{A[tot]} + \frac{M(A[tot] - r[C] \cdot Am)}{A[tot] \cdot r[C] \cdot (R \cdot Am - A[tot])}$$

$$\sigma_{\theta, \theta, C} := 11.02428313$$
 (18)

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3) 9.21

9.21. Determine the radial stress at the junction of the web and inner flange of the curved beam portion of the clamp in Problem 9.20. Neglect stress concentrations.



$$A'_{m} = \int_{a}^{r} \frac{dA}{r}$$
 and $A' = \int_{a}^{r} dA$ (9.20)

$$\sigma_{rr} = \frac{AA'_m - A'A_m}{trA(RA_m - A)}M_x \tag{9.21}$$

Inner flange disst from O is 68mm

 $r := 68 \, \# \, mm$

$$r := 68 \tag{2}$$

$$A_prime[1] := 36 \cdot (14)$$

$$A_prime_1 := 504 \tag{3}$$

$$A_prime[m] := evalf \left(\int_{22}^{36} \frac{361}{r} dr \right) \#(9.20)$$

$$A_prime_m := 177.7840113$$
 (4)

 $Area := 14 \cdot 36 + 12 \cdot 32 + 8 \cdot 26$

$$Area := 1096 \tag{5}$$

t := 12

$$t := 12 \tag{6}$$

yBar := 45.21897810

$$yBar := 45.21897810$$
 (7)

R := yBar

$$R := 45.21897810 \tag{8}$$

 $M := 1140000 \,\#prob2 \,N{\cdot}m$

$$M := 1140000$$
 (9)

$$A[m] := 1096 \# prob2$$

$$A_m := 1096$$
 (10)

$$sigma[rr] := evalf\left(\frac{(Area \cdot A_prime[m] - A_prime[1] \cdot A[m])}{t \cdot r \cdot Area \cdot (R \cdot A[m] - Area)} \cdot M\right)$$

$$\sigma_{rr} := -9.403741443 \tag{11}$$

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3) 12.6

12.6. Three pinned-end columns each have a cross-sectional area of 2000 mm² and length of 750 mm. They are made of 7075-T5 aluminum alloy (E = 72.0 GPa and $\sigma_{PL} = 448$ MPa).

One of the columns has a solid square cross section. A second column has a solid circular cross section. The third has a hollow circular cross section with an inside diameter of 30.0 mm. Determine the critical buckling load for each of the columns.

Pined is kind of like a ball and socket

Simple support-simple support





Square Cross Section

$$I_C = \frac{bh}{12}(b^2 + h^2)$$

Circular Cross Section

$$I_C = \frac{\pi d^4}{32}$$

Hollow Circular Cross Section

$$I_C = \frac{\pi}{32}(d^4 - d_i^4)$$

solve
$$(2000 = b^2, \{b\})$$

 $\{b = -20\sqrt{5}\}, \{b = 20\sqrt{5}\}$

$$solve(2000 = Pi \cdot r^2, \{r\})$$

$$\left\{r = -\frac{20\sqrt{5}}{\sqrt{\pi}}\right\}, \left\{r = \frac{20\sqrt{5}}{\sqrt{\pi}}\right\}$$
 (2)

$$r[\mathit{inner}] := \frac{30}{2} \, \#\mathit{mm}$$

$$r_{inner} := 15 \tag{3}$$

 $solve(2000 = Pi \cdot r[out]^2 - Pi \cdot r[inner]^2, \{r[out]\})$

$$\left\{ r_{out} = -\frac{5\sqrt{\pi (9\pi + 80)}}{\pi} \right\}, \left\{ r_{out} = \frac{5\sqrt{\pi (9\pi + 80)}}{\pi} \right\}$$
 (4)

$$b := 20\sqrt{5}$$

$$b := 20\sqrt{5} \tag{5}$$

$$Iota[Sqr] := evalf\left(\frac{1}{12}b^4\right)\left(\frac{1}{1000^4}\right) \# m^4$$

$$I_{Sqr} := 3.333333333310^5$$
 (6)

$$d := 2 \cdot \left(\frac{20\sqrt{5}}{\sqrt{\pi}} \right)$$

$$d := \frac{40\sqrt{5}}{\sqrt{\pi}} \tag{7}$$

Iota[cir] := evalf
$$\left(\frac{\text{Pi} \cdot d^4}{32}\right) \left(\frac{1}{1000^4}\right) \# m^4$$

 $I_{\text{cir}} := 6.366197722 \cdot 10^5$ (8)

$$d[out] := evalf\left(2\left(\frac{5\sqrt{\text{Pi }(9 \text{ Pi} + 80)}}{\text{Pi}}\right)\right)$$

$$d_{out} := 58.70672098$$
(9)

$$d[inner] := (r[inner] \cdot 2)$$

$$d_{inner} := 30$$
(10)

$$Iota[HollC] := evalf\left(\frac{Pi \cdot \left(d[out]^4 - d[inner]^4\right)}{32}\right) \left(\frac{1}{1000^4}\right) # m^4$$

$$I_{HollC} := 0.000001086619772$$
(11)

$$E := 72 \cdot 10^9 \, \#Pa$$

$$E := 720000000000 \tag{12}$$

$$P[crit_sq] := \frac{\pi^2 E \cdot \text{Iota}[Sqr]}{(.75)^2} \# N$$

$$P_{crit_sq} := 4.211031214 \cdot 10^{17}$$
(13)

$$P[crit_ci] := \frac{\pi^2 E \cdot \text{Iota}[cir]}{(.75)^2} \# N$$

$$P_{crit_ci} := 8.042477196 \ 10^{17}$$
(14)

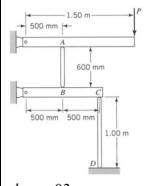
$$P[crit_Hci] := \frac{\pi^2 E \cdot \text{Iota}[HollC]}{(.75)^2} \# N$$

$$P_{crit_Hci} := 1.372736933 \cdot 10^6$$
(15)

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4) 12.8. Assume that the horizontal members are rigid and weightless.

12.8. In Figure P12.8, columns AB and CD have pinned ends, are made of an aluminum alloy (E = 72.0 GPa), and have equal rectangular cross sections of 20 mm by 30 mm. Determine the magnitude of P that will first cause one of the columns to buckle. Assume elastic conditions.



$$b := .03$$

$$b := 0.03 \tag{1}$$

$$h := .02$$

$$h := 0.02 \tag{2}$$

$$Iota[x] := \frac{1}{12} b \cdot h^3 \# m^4$$

$$I_{x} := 2.0000000000 \, 10^{-8} \tag{3}$$

b := .02

$$b \coloneqq 0.02 \tag{4}$$

$$h := .03$$

$$h := 0.03 \tag{5}$$

$$Iota[y] := \frac{1}{12}b \cdot h^3$$

$$I_{y} := 4.5000000000 \, 10^{-8} \tag{6}$$

$$L[AB] := .6 \# m$$
 $L_{AB} := 0.6$ (7)

$$L[CD] := 1 \# m$$

$$L_{CD} := 1$$
(8)

$$E := 72 \cdot 10^9$$

$$E := 72000000000$$
(9)

$$F[C, Ix] := \frac{\pi^2 E \cdot \text{Iota}[x]}{L[CD]^2}$$

$$F_{C, Ix} := 14212.23034$$
(10)

$$F[C, Iy] := \frac{\pi^2 E \cdot \text{Iota}[y]}{L[CD]^2}$$

$$F_{C, Iy} := 31977.51827$$
(11)

$$F[A, Ix] := \frac{\pi^2 E \cdot \text{Iota}[x]}{L[AB]^2}$$

$$F_{A, Ix} := 39478.41762$$
(12)

$$F[A, Iy] := \frac{\pi^2 E \cdot \text{Iota}[y]}{L[AB]^2}$$

$$F_{A, Iy} := 88826.43964$$
(13)

 $\#F_{C, Ix}$ is the Lowest

$$F[A_new] := \frac{F_{C, Ix}}{.5}$$
 $F_{A_new} := 28424.46068$ (14)

$$P := \frac{F_{A_new} \cdot (.5)}{1.5} \#N$$

$$P := 9474.820226$$
 (15)