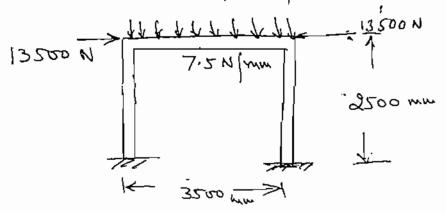
#1 The frame shows conords of 3 members. Given  $E = 200 \times 10^9 \text{ Pa}$ ,  $A = 2450 \text{ mm}^2$ ,  $T = 27 \times 10^6 \text{ mm}^4$  for all members.

- and with of beam. Assume rectangular x-section
  - b) Using symmetry write the Boundary Conditions
  - c) Set up the stiffness matrix (for symmetric case)
  - d) set up the board watroise (",")
  - e) Impose BCs and unite the final stiffness and load matrix after elimination



<del>†</del>2\_

LEB PEB LE L'ON (Forque)

AB is a begin

BE and cf are and

bares

C D4

C torque

EB

Write the Alifness matrix for element BE

a) write the Alfred matrix for element Bt.
b) It the X-section of beam to is circular what is the relation between Polar M. I (3) of the X-section and Arrea M. I. Tyy= Icz

Write the load rector for element co ) If the rod AD is very stiff how will you solve the broppen (Brief Hurman).

3) For the element shows

a) determine the stiffness (undustivity) matrix

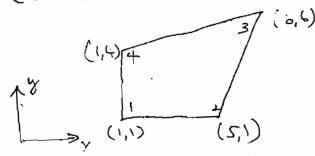
b) " the head board rector

h, To 1 1 0.4m

P= 20 M/m5 G Ta = 25 C Kx = 1.5 W/mC Ky = 1.0 W/mC

( For (a) do not ferterm matrix multiplications)

4). Follow the steps fiven below to numerically integrate I (xy +y2) dx dy over the domain of teg.



write a and y in terms of I and of

What are the values of g and of for me point り)

formula Determine the values of x and y at the C) Gauss point usung 3 and of determined in (b)

Determine the Jacobsan using the equation T)

#5 For the axial boar shown determine the natural trequencies and the corresponding mode shapes Draw the shapes on n-4 fragh (x - cound of node and is axial displacement)

A = 600 mm<sup>2</sup>  $E = 200 \times 10^{3} \text{ N/m}^{2}$   $CM \quad CM$   $CM \quad CM$ 

#6A = 600 mm²

= 25 cm > A = 600 mm²

= 25 cm > M = 600 mm²

The initial relocates of node 2 is 2 m/s wode 1 is fixed and initial displacement of nodes 2 and 3 is zero. A linearly varying force  $F = 10^{7} \pm 10^{12}$  is applied at node 2. Using  $\Delta t = 10^{-5}$  sec

(b) Determine duplacement of wade 2 at t=10 sec

(c) velocity of node 2 at t=10-5 s

(a) Using Control Difference scheme show that for no damping case  $\frac{2u_{1}^{2}}{1} = \left(\frac{M}{\Delta t^{2}}\right)^{-1} \left(\frac{2M}{(\Delta t)^{2}} - K^{2}_{1} \left[\frac{M}{(\Delta t)^{2}}\right] \left[\frac{M}{(\Delta t)^{2}}\right] \left[\frac{M}{(\Delta t)^{2}}\right]$ 

(C) For a torangular plane stress element write the expression for k.E. The element has 6 dyrees of freedom. The expression should contain these days and shape fus  $K \cdot E = \frac{1}{2} [\ ][\ ] tdh of$