



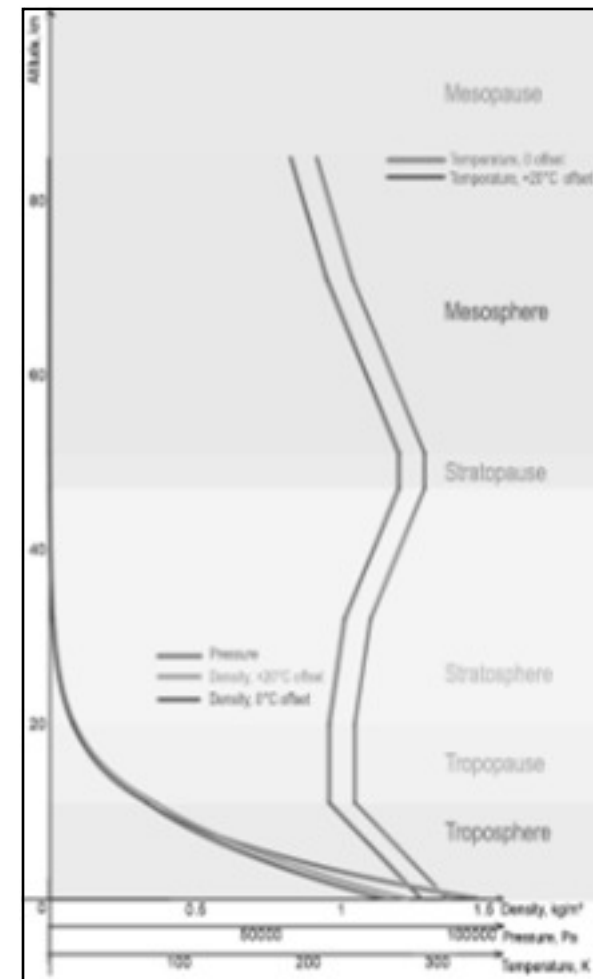
# *Atmospheric Density Model*



## *Atmospheric Models*

The most **commonly** employed atmospheric model is the **ISA**, shown **alongside**.

It is **adequate** for ascent mission modelling as **drag** is less than **1%** of weight beyond **75 km**.





## *Atmospheric Density & Pressure*

The **density** and pressure are the **primary** parameters that impact the **ascent** mission.

In the context of **pressure**, its impact is mainly on the  **$I_{sp}$**  which is specified for sea-level & **vacuum**.



## *Earth Geometric Model*



## *Flat Earth Model*

**Ascent** missions generally use a Cartesian **coordinate** system that is defined at the **launch** point.

In such a situation, when **motion** of vehicle is along a **radial** line, the local **tangent**, along with the **radial** line, can be used to **represent** a 2-D coordinate system.

This **approximation** results in constant **gravity direction**, but is restricted to small **distances** over Earth's surface. (E.g.  $1^\circ$  change in slope  $\cong$  **110 km**).



## ***Implication of Spherical Earth***

However, as rockets **typically** travel thousands of kilometers over **Earth's** surface, we need to **account** for earth's **curvature** as well as its **rotation**.

In most cases, a **polar** or spherical coordinate **system** is able to address the **issues** with reasonable **accuracy**.



## *Summary*

To **summarize**, both atmospheric and earth's geometric **representations** are fairly simplified **forms** of more complex **models** due to their small **order** of magnitude.