

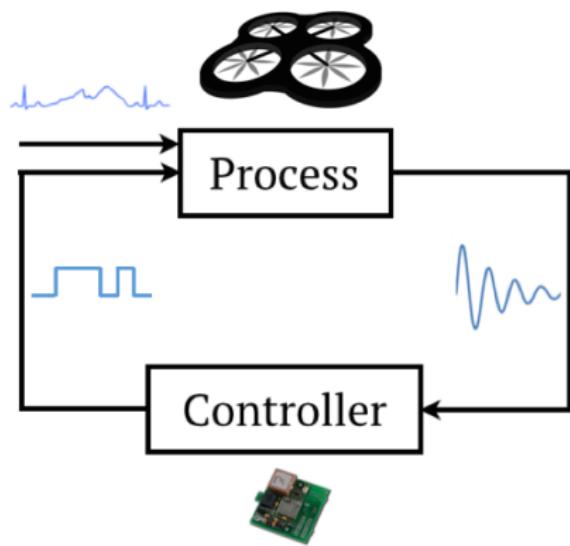
# Control of Large Scale Cyberphysical Systems

Abhishek Halder

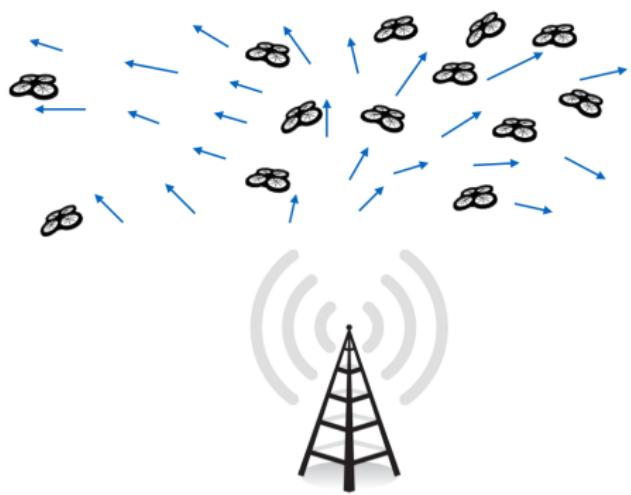
Department of Mechanical and Aerospace Engineering  
University of California, Irvine  
Irvine, CA 92697-3975

# Motivation: Drone Traffic Management

Controlling A Drone

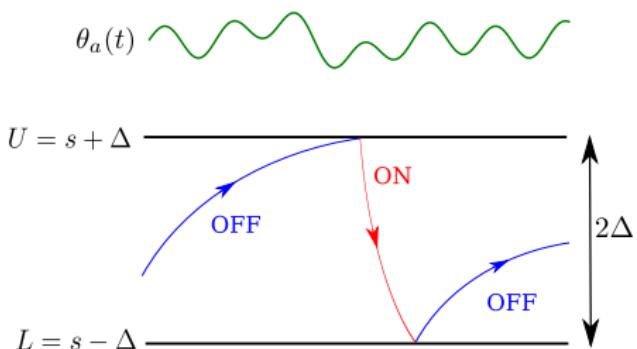


Controlling Swarm of Drones

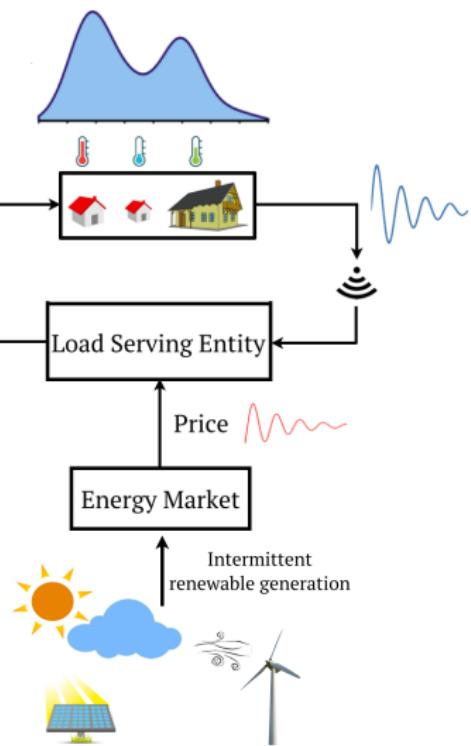


# Motivation: Smart Grid Demand Response

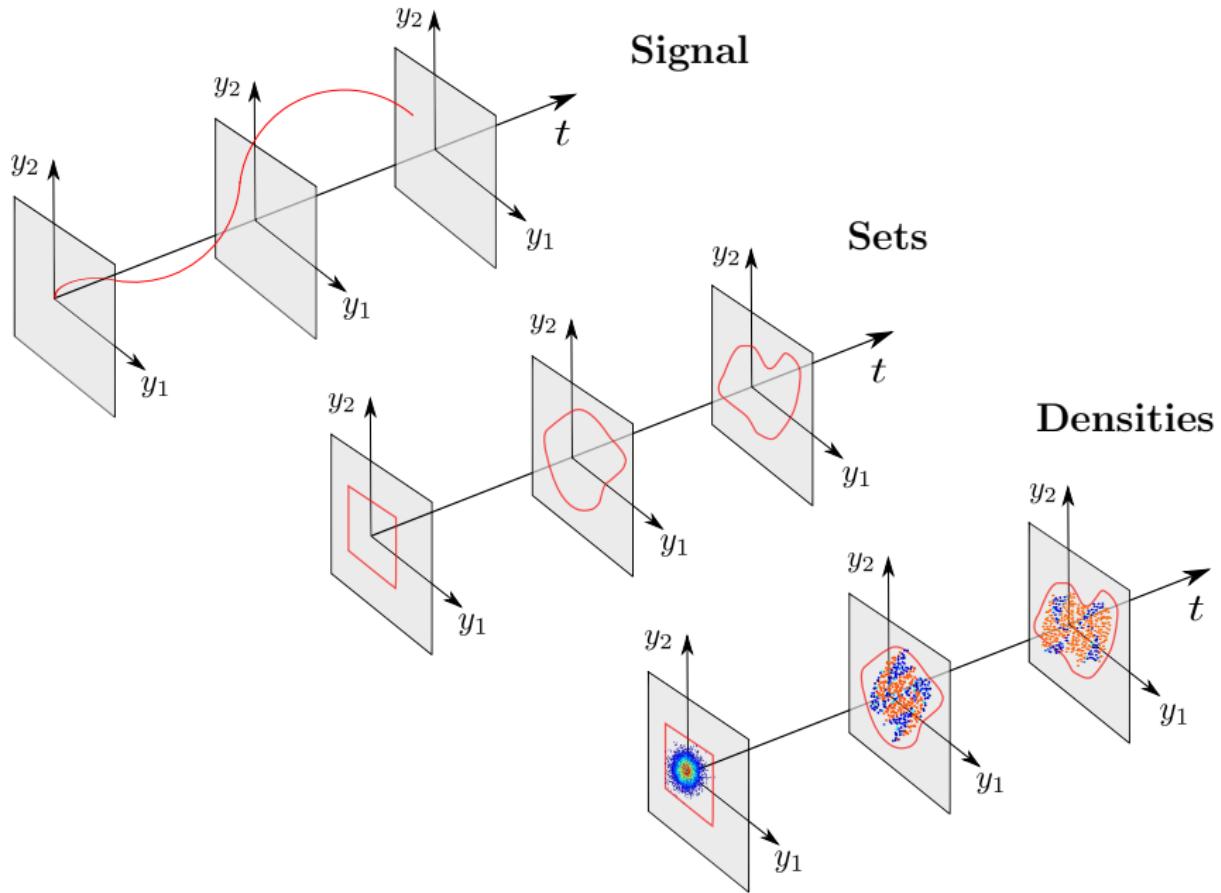
Controlling An AC



Controlling Population of ACs

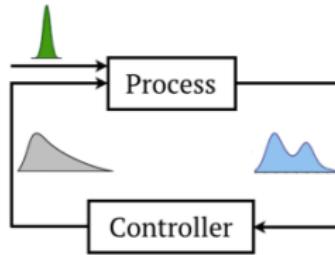


# What to Control

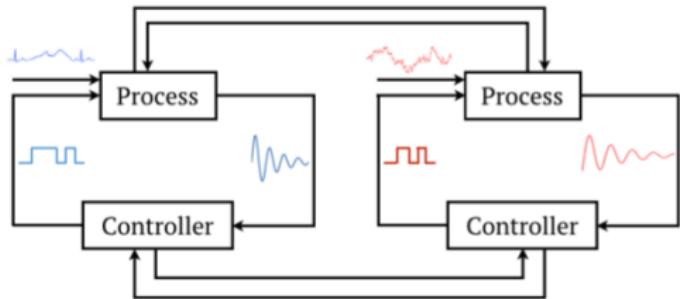


# Outlook

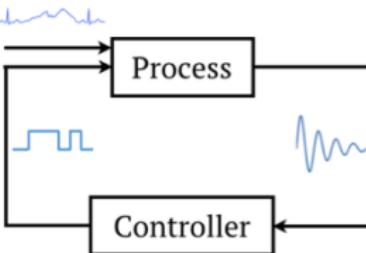
Continuum of systems



Finitely many systems



One system



# **Outline of Today's Talk**

## **Part I: An Application**

Controlling Air Conditioners

## **Part II: A Theory**

Controlling Density

## **Part III: Ongoing and Future Research**

Unmanned Aerial Systems Traffic Management

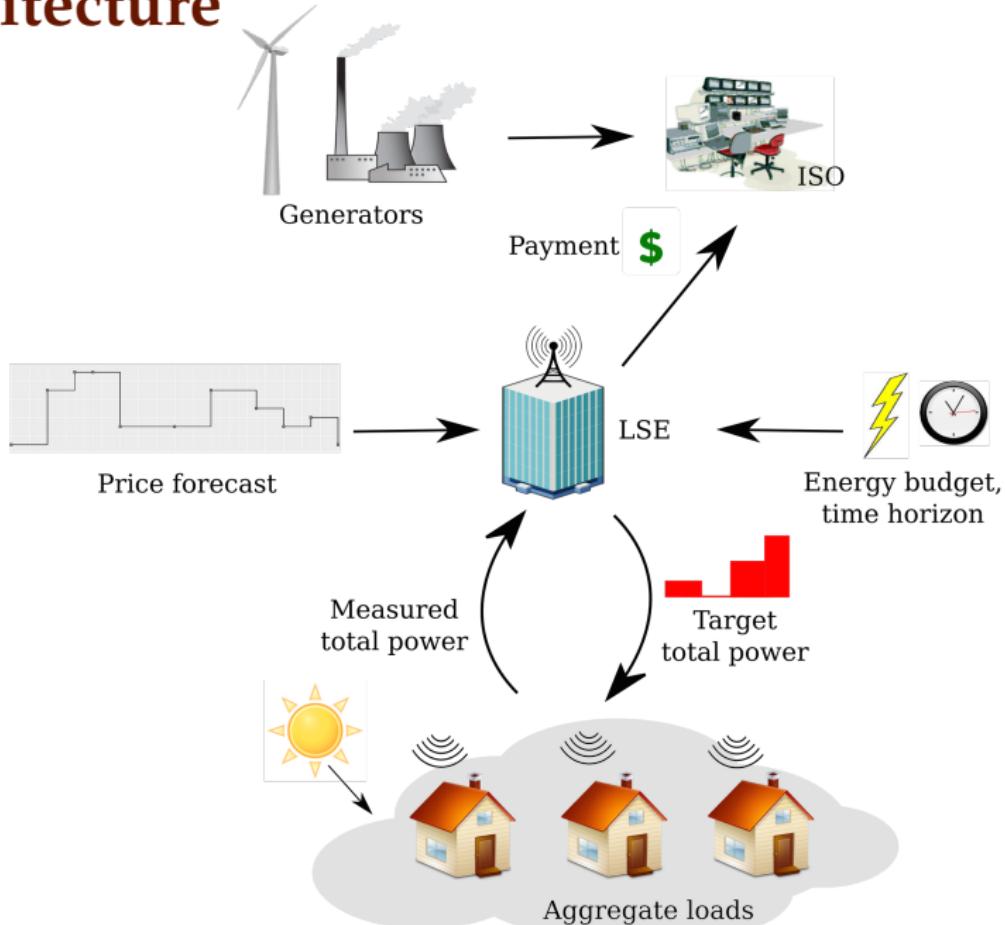
# Part I. An Application

## Controlling Air Conditioners

Direct Control for Demand Response

Joint work with X. Geng, F.A.C.C. Fontes, P.R. Kumar, and L. Xie

# Architecture



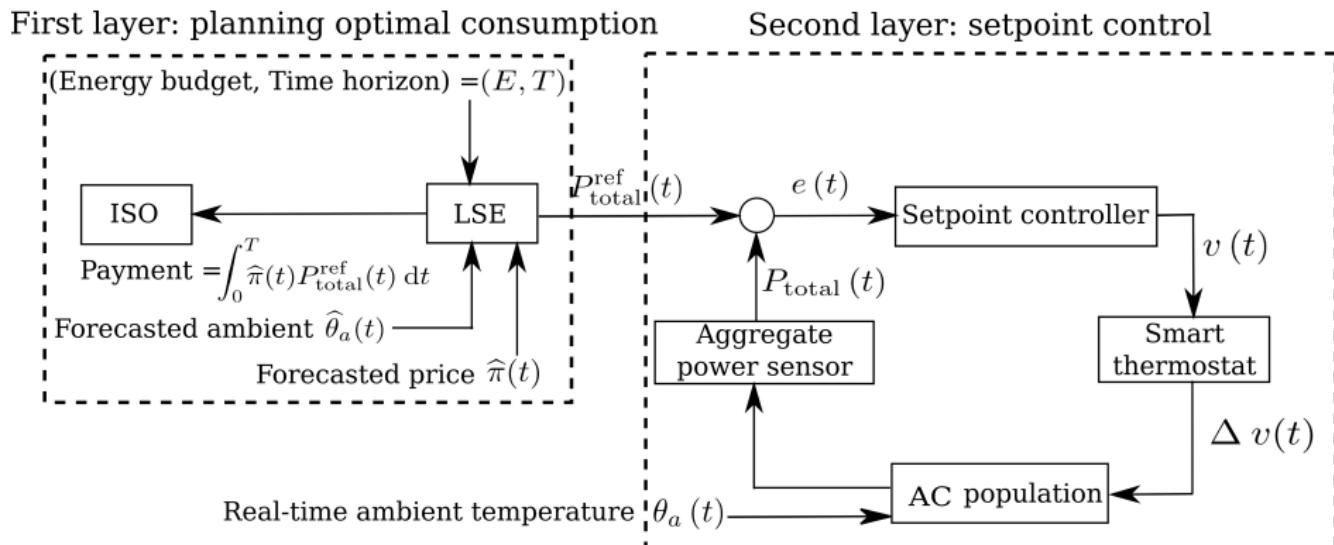
# Research Scope

**Objective:** A theory of operation for the LSE

**Challenges:**

1. How to design the **target consumption as a function of price**?
2. How to control so as to preserve **privacy** of the loads' states?
3. How to respect loads' **contractual obligations** (e.g. comfort range width  $\Delta$ )?

# Two Layer Block Diagram



# First Layer: Planning Optimal Consumption

$$\underset{\{u_1(t), \dots, u_N(t)\} \in \{0,1\}^N}{\text{minimize}} \quad \int_0^T \frac{P}{\eta} \left| \widehat{\pi}(t) \right| (u_1(t) + u_2(t) + \dots + u_N(t)) \, dt$$

subject to

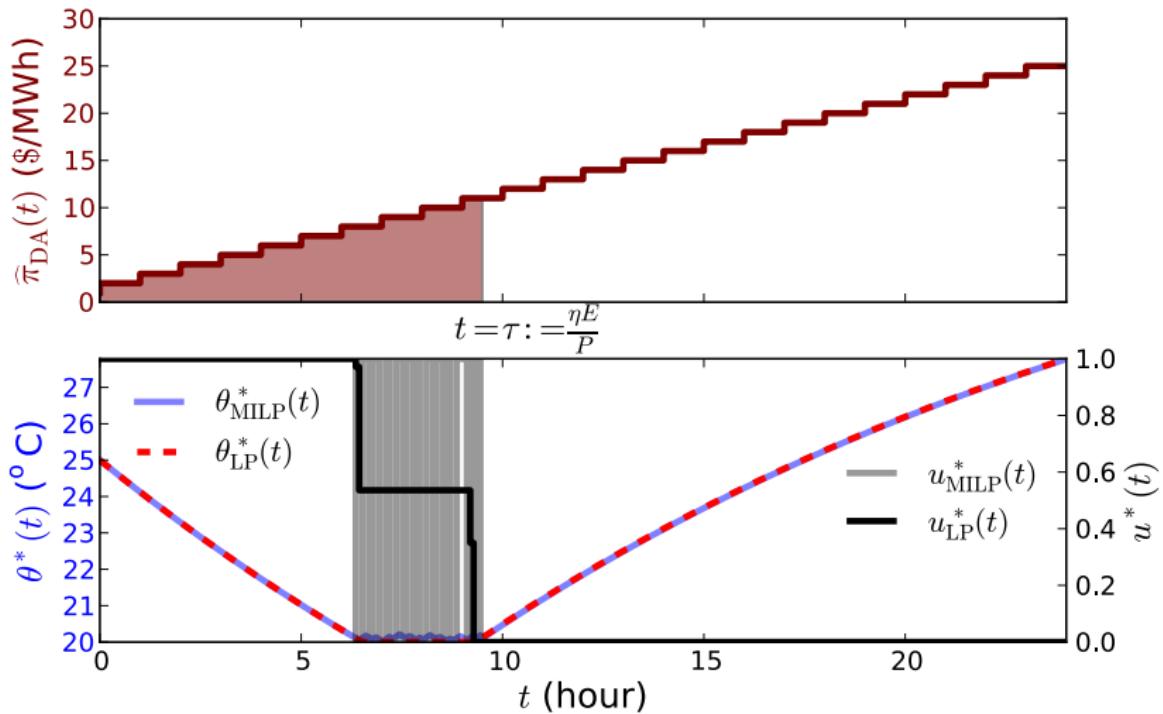
$$(1) \quad \dot{\theta}_i = -\alpha_i \left( \theta_i(t) - \widehat{\theta}_a(t) \right) - \beta_i P u_i(t) \quad \forall i = 1, \dots, N,$$

$$(2) \quad \int_0^T (u_1(t) + u_2(t) + \dots + u_N(t)) \, dt = \tau \doteq \frac{\eta E}{NP} (< T, \text{given})$$

$$(3) \quad L_{i0} \leq \theta_i(t) \leq U_{i0} \quad \forall i = 1, \dots, N.$$

**Optimal consumption:**  $P_{\text{ref}}^*(t) = \frac{P}{\eta} \sum_{i=1}^N u_i^*(t)$

# First Layer: "discretize-then-optimize"



Numerical challenges for MILP and LP

Solution: continuous time  $\rightsquigarrow$  PMP w. state inequality constraints

# Second Layer: Real-time Setpoint Control

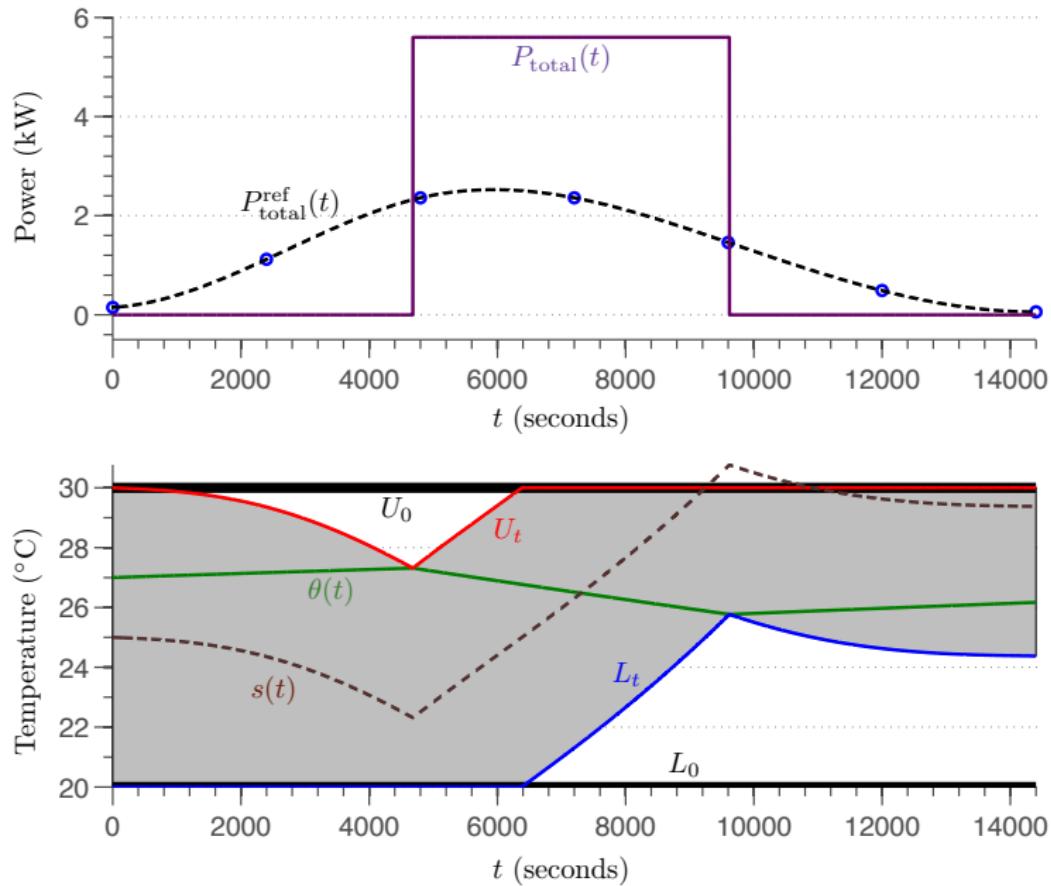
$$\begin{array}{c} \text{optimal reference} \\ | \\ P_{\text{ref}}^*(t) = \frac{P}{\eta} \sum_{i=1}^N u_i^*(t), \rightsquigarrow \end{array} \quad \begin{array}{c} \text{error} \\ | \\ e(t) = P_{\text{ref}}^*(t) - P_{\text{total}}(t), \end{array}$$

$$\begin{array}{c} \text{PDE based velocity control} \\ | \\ v(t) = \gamma_{\text{tracking}}(e(t)), \end{array} \quad \begin{array}{c} \text{gain} \\ | \\ \frac{ds_i}{dt} = \Delta_i \end{array} \quad \begin{array}{c} \text{broadcast} \\ | \\ v(t) \end{array},$$

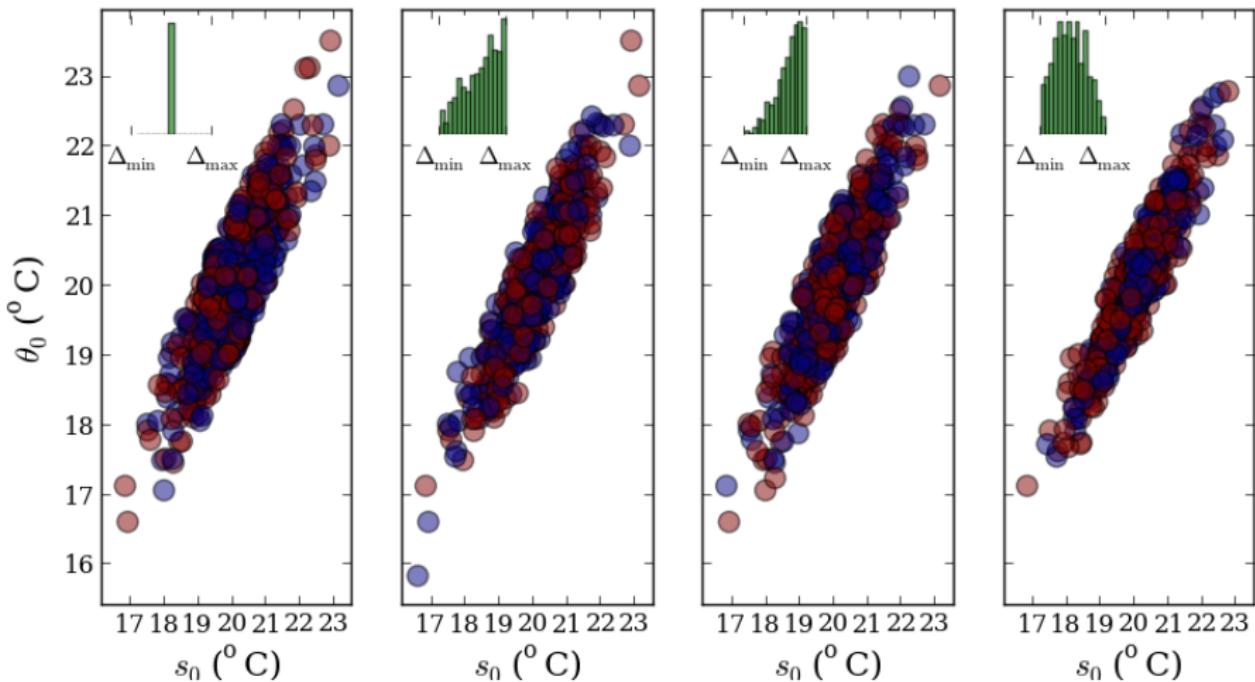
$$\begin{array}{c} \text{Moving lower boundary} \\ | \\ L_{it} = U_{i0} \wedge [L_{i0} \vee (s_i(t) - \Delta_i)], \end{array}$$

$$\begin{array}{c} \text{Moving upper boundary} \\ | \\ U_{it} = L_{i0} \vee [U_{i0} \wedge (s_i(t) + \Delta_i)]. \end{array}$$

# Boundary Control: Deadband $\rightarrow$ Liveband

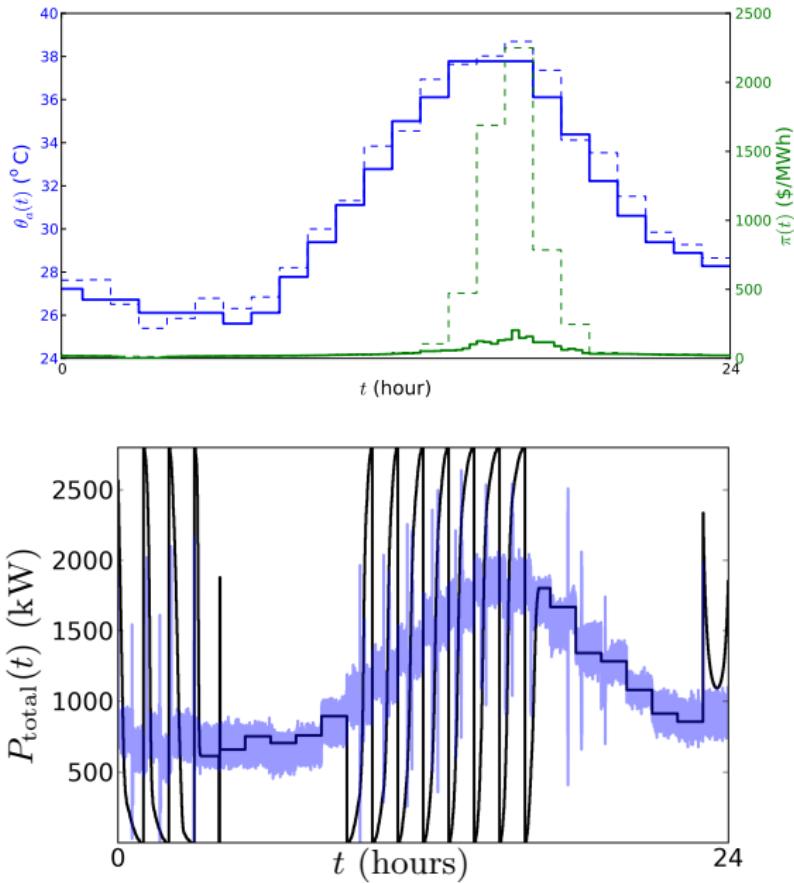


# Initial Condition and $\Delta$ Distribution for 500 Homes

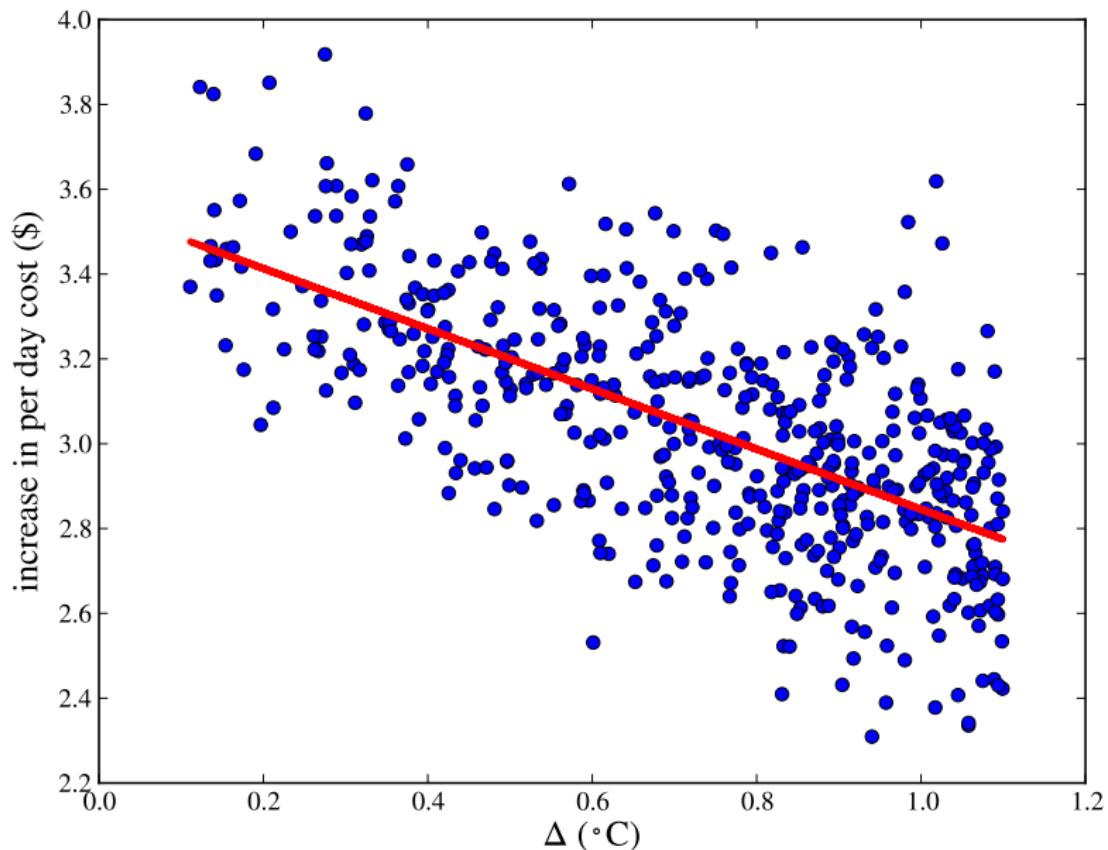


# Houston Temperature + Market Price

- forecasted ambient
- actual ambient
- forecasted price
- actual price
- target consumption
- actual consumption



# How Can the LSE Price A Contract



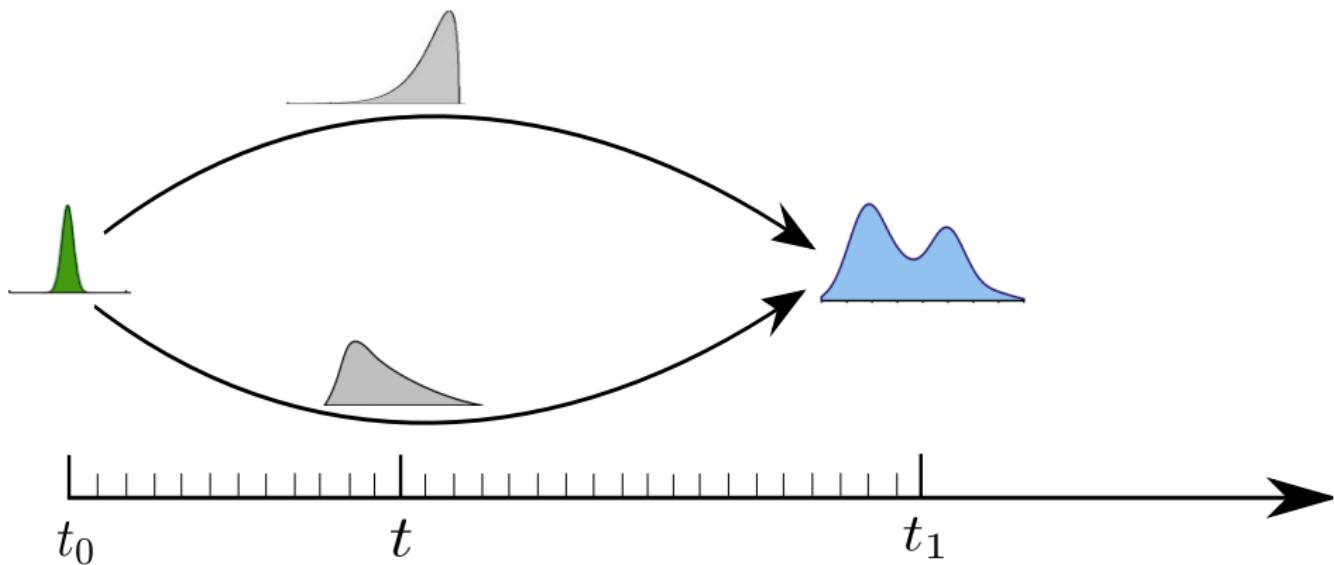
## Part II. A Theory

### Controlling Density

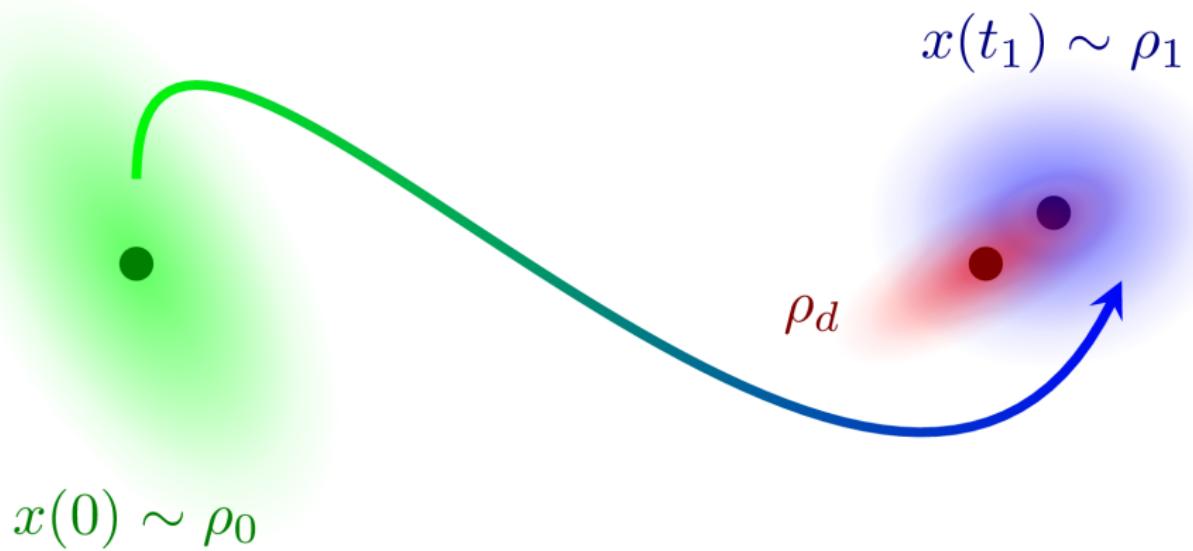
Finite Horizon LQG Density Regulator

Joint work with E.D.B. Wendel (Draper Laboratory)

# How to Go from One Density to Another



# or Close to Another



# LQG State Regulator

$$\min_{u \in \mathcal{U}} \phi(x_1, x_d) + \mathbb{E}_x \left[ \int_0^{t_1} (x^\top Q x + u^\top R u) dt \right]$$

$$dx(t) = Ax(t) dt + Bu(t) dt + F dw(t),$$

$x(0) = x_0$  given,  $x_d$  given,  $t_1$  fixed,

**Typical terminal cost: MSE**

$$\phi(x_1, x_d) = \mathbb{E}_{x_1} [(x_1 - x_d)^\top M (x_1 - x_d)]$$

# LQG Density Regulator

$$\min_{u \in \mathcal{U}} \varphi(\rho_1, \rho_d) + \mathbb{E}_x \left[ \int_0^{t_1} (x^\top Q x + u^\top R u) dt \right]$$

$$dx(t) = Ax(t) dt + Bu(t) dt + F dw(t),$$

$x(0) \sim \rho_0$  given,  $x_d \sim \rho_d$  given,  $t_1$  fixed,

**Proposed terminal cost: MMSE**

$$\varphi(x_1, x_d) = \inf_{y \sim \rho \in \mathcal{P}_2(\rho_1, \rho_d)} \mathbb{E}_y [(x_1 - x_d)^\top M (x_1 - x_d)],$$

where  $y := (x_1, x_d)^\top$

# Formulation: LQG Density Regulator

$$\varphi(\rho_1,\rho_d)$$

|

$$\begin{aligned} \min_{u \in \mathcal{U}} \quad & \inf_{y \sim \rho \in \mathcal{P}_2(\rho_1, \rho_d)} \mathbb{E}_y \left[ (x_1 - x_d)^\top M (x_1 - x_d) \right] \\ & + \mathbb{E}_x \left[ \int_0^{t_1} (x^\top Q x + u^\top R u) \, dt \right] \end{aligned}$$

$$\mathrm{d}x(t)=Ax(t)\,\mathrm{d}t+Bu(t)\,\mathrm{d}t+F\,\mathrm{d}w(t),$$

$$x(0) \sim \rho_0 = \mathcal{N}\left(\mu_0, S_0\right), \quad x_d \sim \rho_d = \mathcal{N}\left(\mu_d, S_d\right),$$

$$t_1 \text{ fixed}, \quad \mathcal{U} = \{u \, : \, u(x,t) = K(t)x + v(t)\}$$

**However,  $\varphi(\mathcal{N}(\mu_1, S_1), \mathcal{N}(\mu_d, S_d))$  equals**

$$(\mu_1 - \mu_d)^\top M (\mu_1 - \mu_d) +$$

$$\min_{C \in \mathbb{R}^{n \times n}} \text{tr}((S_1 + S_d - 2C)M) \text{ s.t. } \begin{bmatrix} S_1 & C \\ C^\top & S_d \end{bmatrix} \succeq 0$$

**However,  $\varphi(\mathcal{N}(\mu_1, S_1), \mathcal{N}(\mu_d, S_d))$  equals**

$$(\mu_1 - \mu_d)^\top M (\mu_1 - \mu_d) +$$

$$\min_{C \in \mathbb{R}^{n \times n}} \text{tr}((S_1 + S_d - 2C)M) \quad \text{s.t.} \quad \begin{bmatrix} S_1 & C \\ C^\top & S_d \end{bmatrix} \succeq 0$$

$\Updownarrow$

$$\max_{C \in \mathbb{R}^{n \times n}} \text{tr}(CM) \quad \text{s.t.} \quad CS_d^{-1}C^\top \succeq 0$$

$\Updownarrow$

$$C^* = S_1 S_d^{\frac{1}{2}} \left( S_d^{\frac{1}{2}} S_1 S_d^{\frac{1}{2}} \right)^{-\frac{1}{2}} S_d^{\frac{1}{2}}$$

This gives

$$\begin{aligned}\varphi(\mathcal{N}(\mu_1, S_1), \mathcal{N}(\mu_d, S_d)) &= (\mu_1 - \mu_d)^\top M (\mu_1 - \mu_d) \\ &+ \text{tr} \left( MS_1 + MS_d - 2 \left[ (\sqrt{S_d} MS_1 \sqrt{S_d}) (\sqrt{S_d} S_1 \sqrt{S_d})^{-\frac{1}{2}} \right] \right)\end{aligned}$$

Applying maximum principle:

$$K^o(t) = R^{-1} B^\top P(t),$$

$$v^o(t) = R^{-1} B^\top (z(t) - P(t)\mu(t))$$

$$\infty \textbf{ dim. TPBVP} \rightsquigarrow \left( n^2 + 3n \right) \textbf{ dim. TPBVP}$$

$$\begin{pmatrix}\dot{\mu}(t)\\ \dot{z}(t)\end{pmatrix} = \begin{pmatrix}A & BR^{-1}B^\top \\ Q & -A^\top\end{pmatrix}\begin{pmatrix}\mu(t)\\ z(t)\end{pmatrix},$$

$$\dot{S}(t)=(A+BK^o)S(t)+S(t)(A+BK^o)^{\top}+FF^{\top},$$

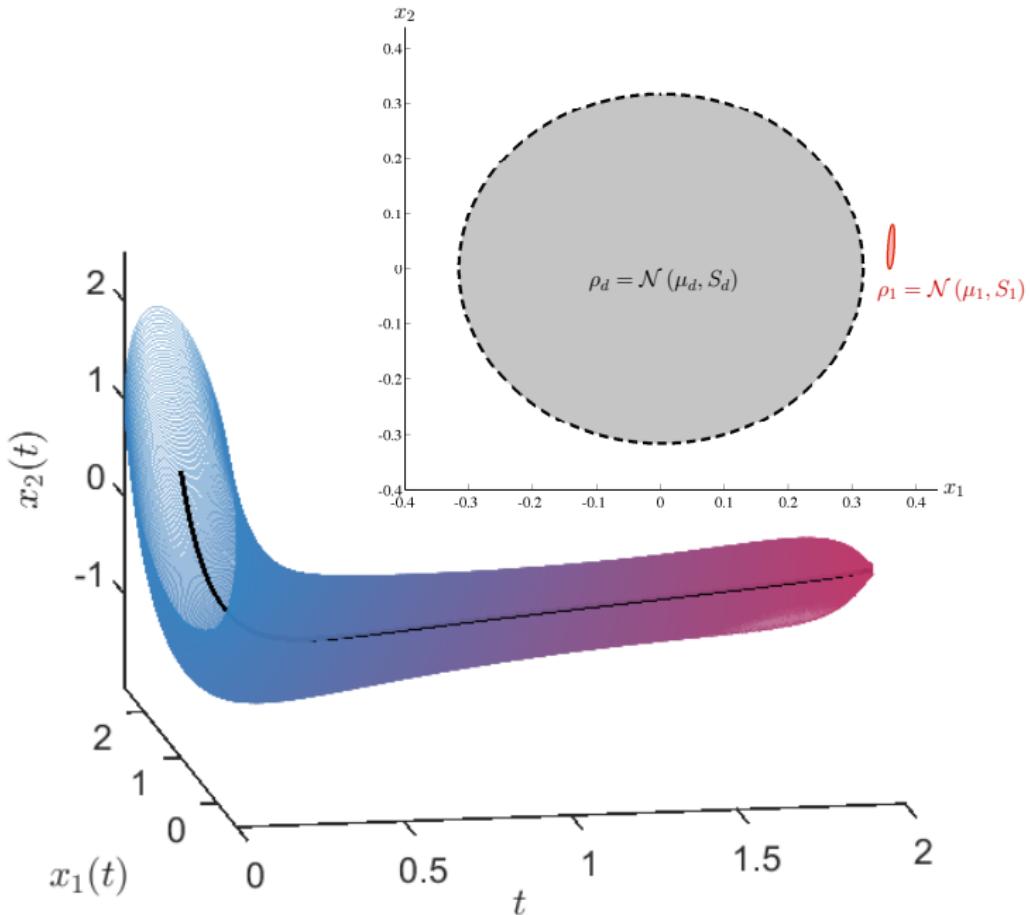
$$\dot{P}(t)=-A^{\top}P(t)-P(t)A-P(t)BR^{-1}B^{\top}P(t)+Q,$$

$$\textbf{Boundary conditions:}$$

$$\mu(0)=\mu_0,\quad z(t_1)=M(\mu_d-\mu_1),$$

$$S(0)=S_0,\quad P(t_1)=\,\,\boxed{\left(S_d\,\#\,S_1^{-1}-I_n\right)M}$$

# Controlled State Covariance



# Application: Active Control for Mars EDL

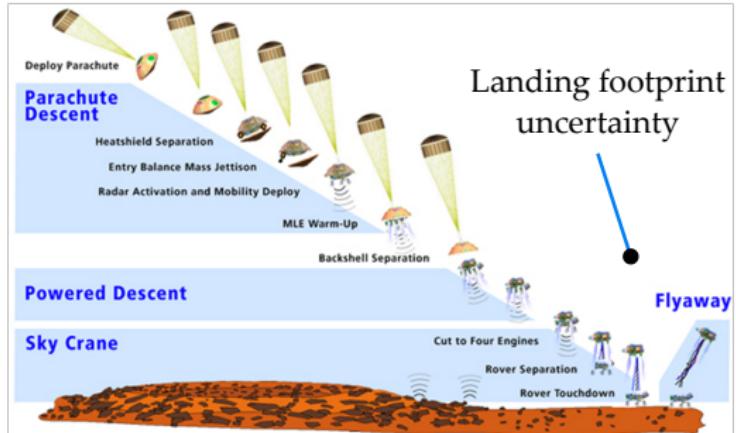
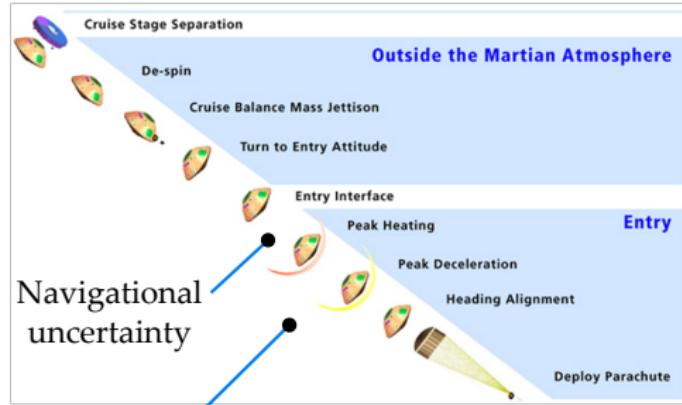


Image credit: NASA JPL

# Part III. Ongoing and Future Research

## UTM

Unmanned Aerial Systems Traffic Management

# Vision for UAS Traffic Management (UTM)

Class G airspace extends up to 1200 ft AGL

500 ft AGL



Weight no more than 55 lbs



200 ft AGL

- Requires:**
- Automated V2V separation management
  - Yield manned traffic
  - Avoid obstacles (buildings, towers etc.)

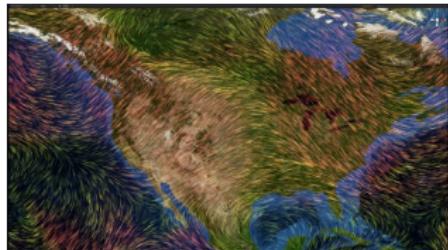
# Technical Challenges

## Dynamic Geofencing

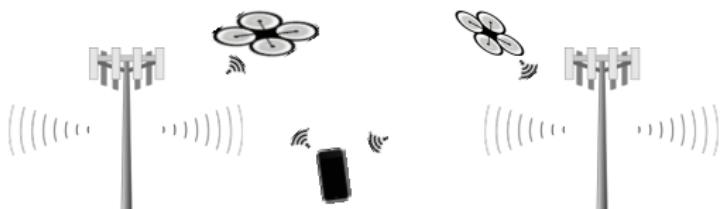


*Image credit: NASA Ames Research Center*

## Wind Uncertainty



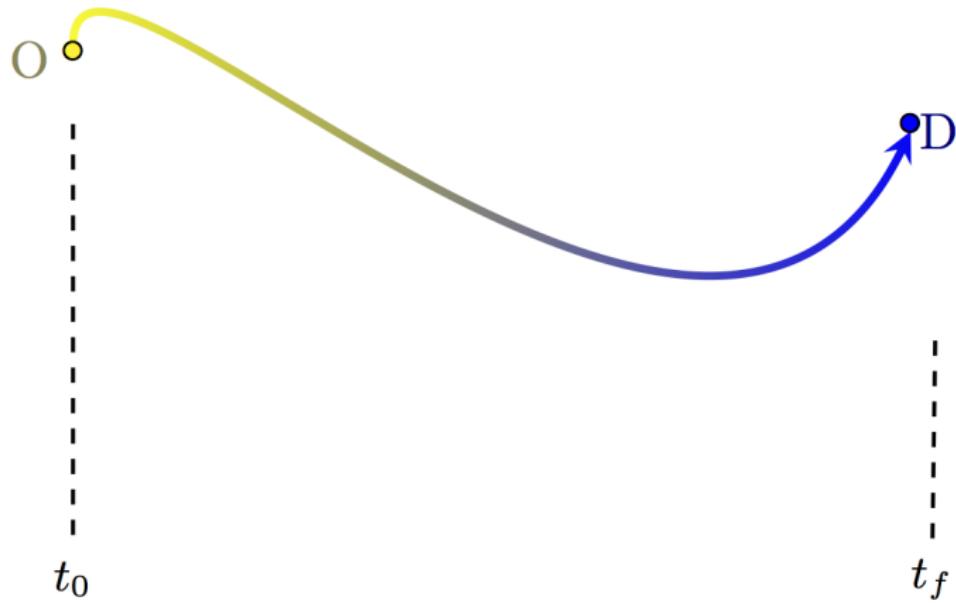
## Control over LTE



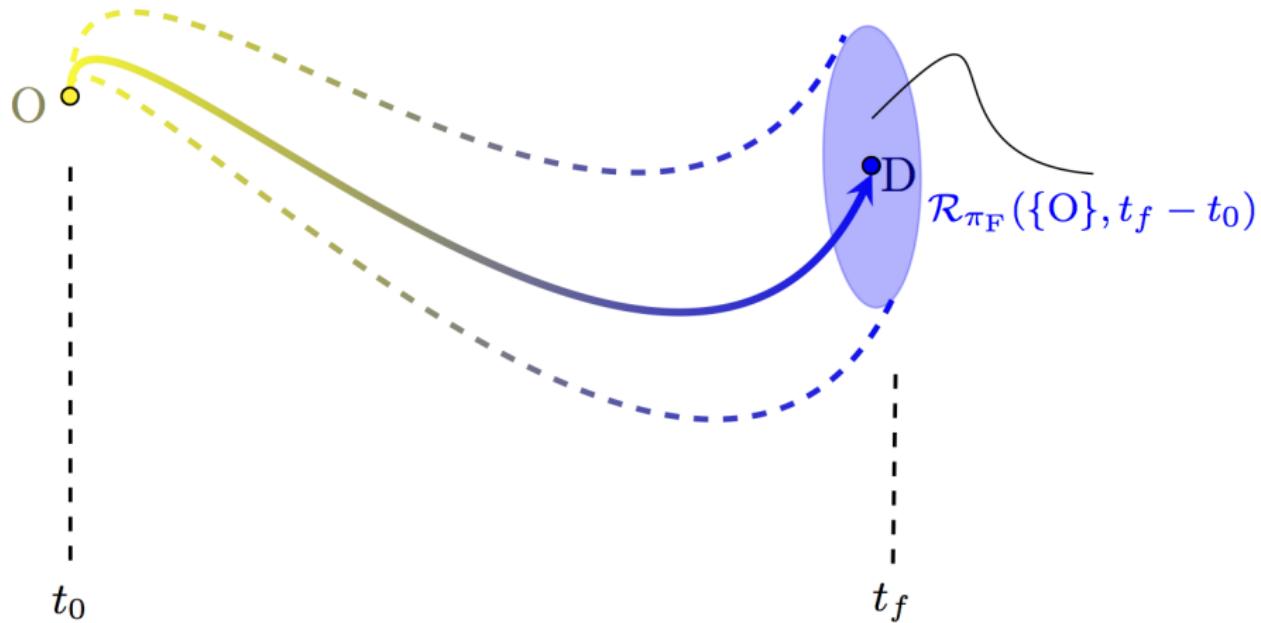
## Provable Safety



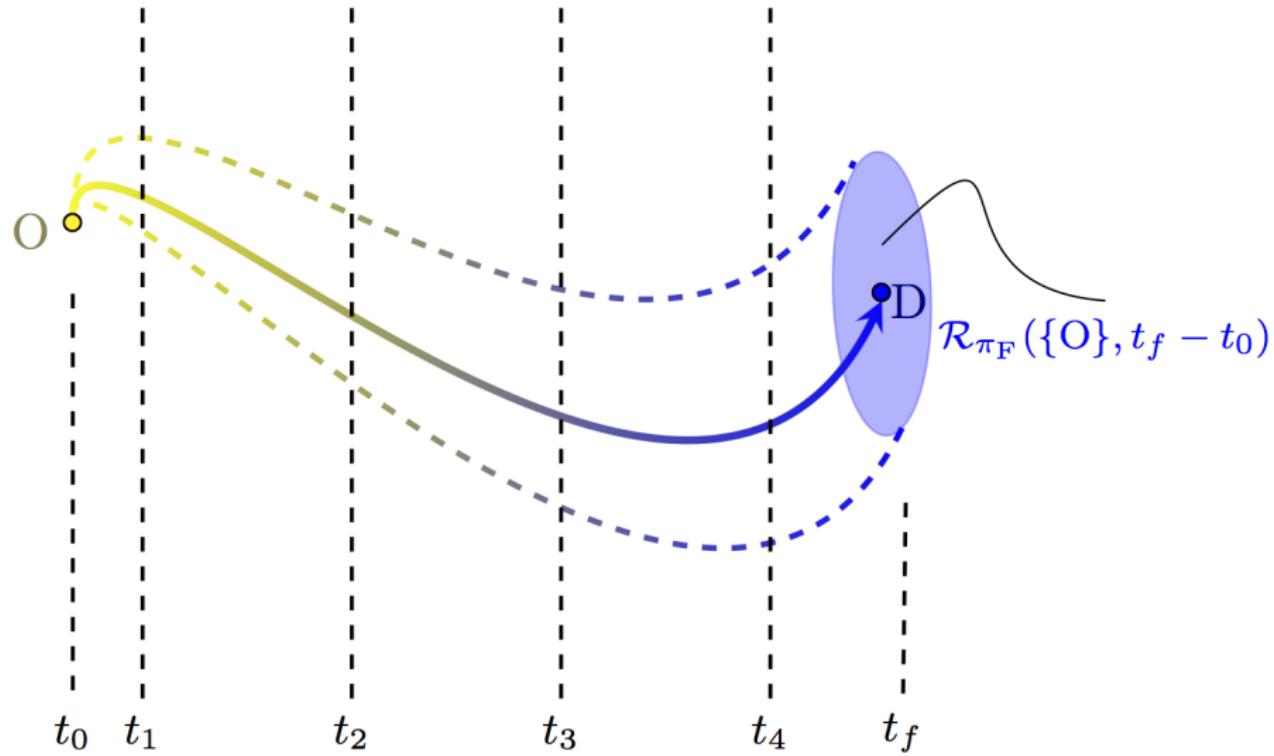
# Input: Approved Flight Path



# Reach Set Evolution due to Wind Uncertainty

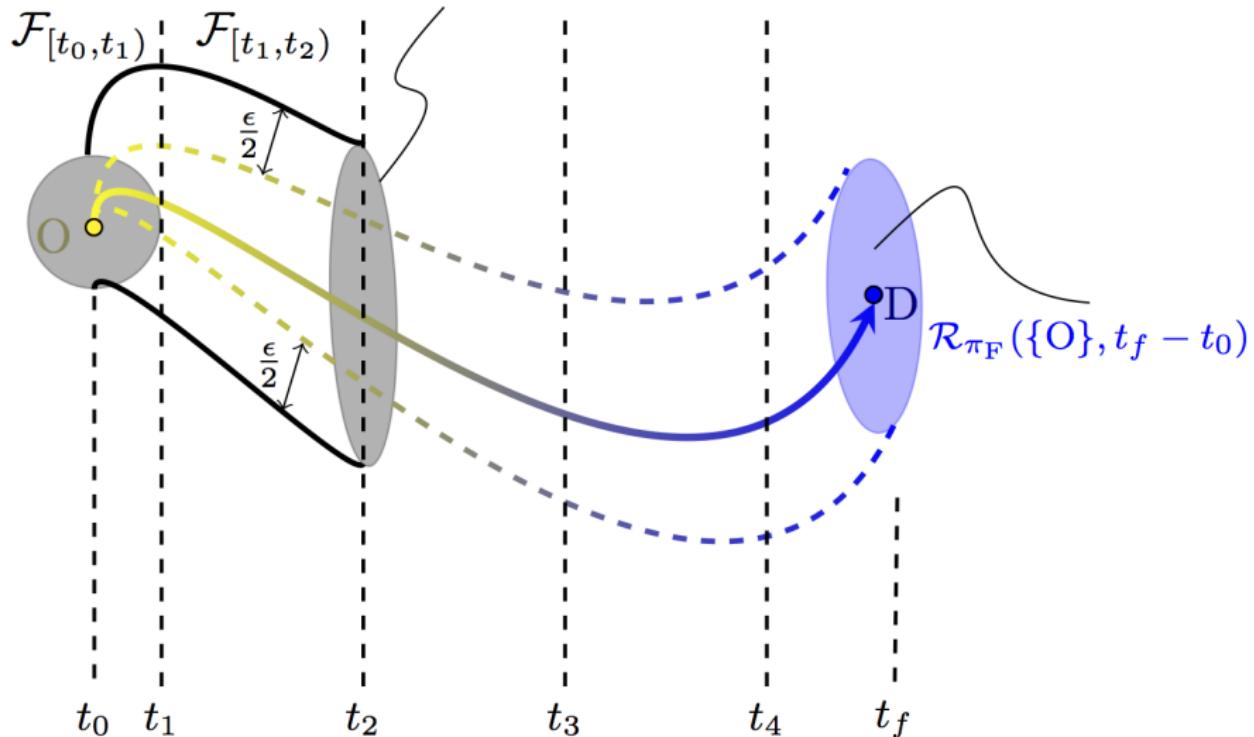


# Discrete Decision Making Instances

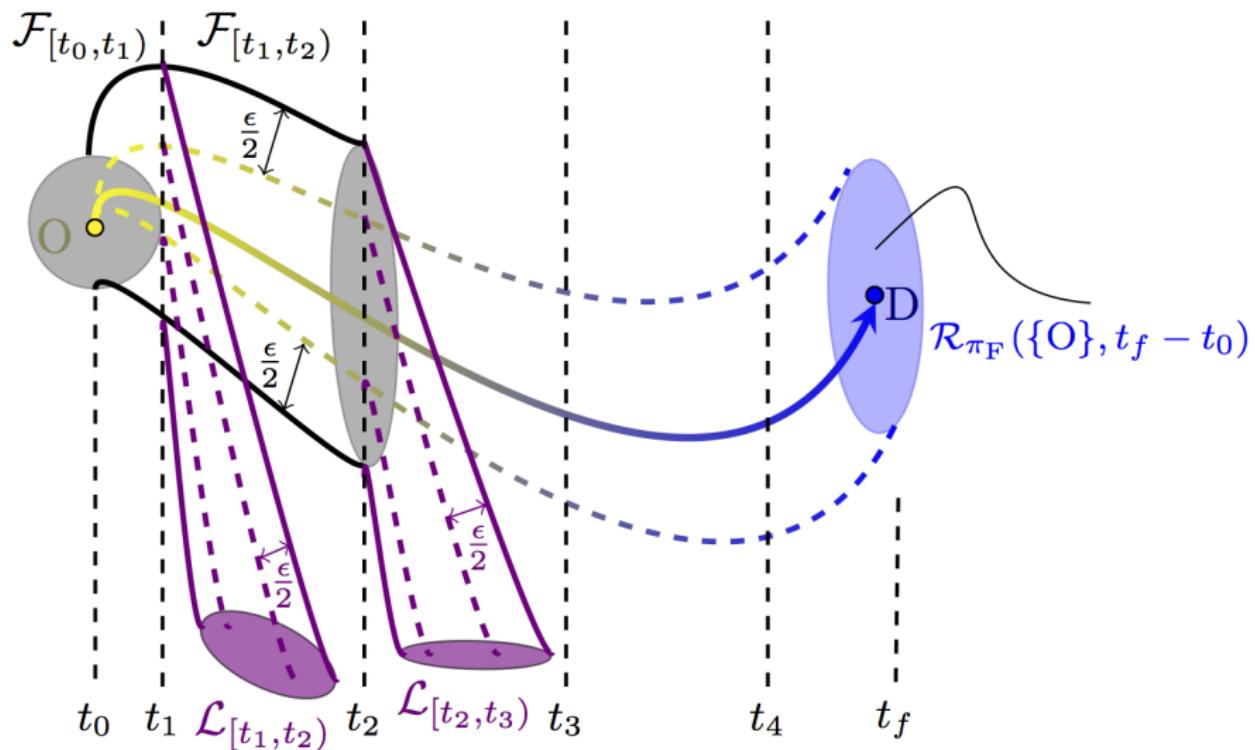


# 4D Flight Tubes $\mathcal{F}_{[t_j, t_{j+1})}$

Reach set enclosed with safety annulus

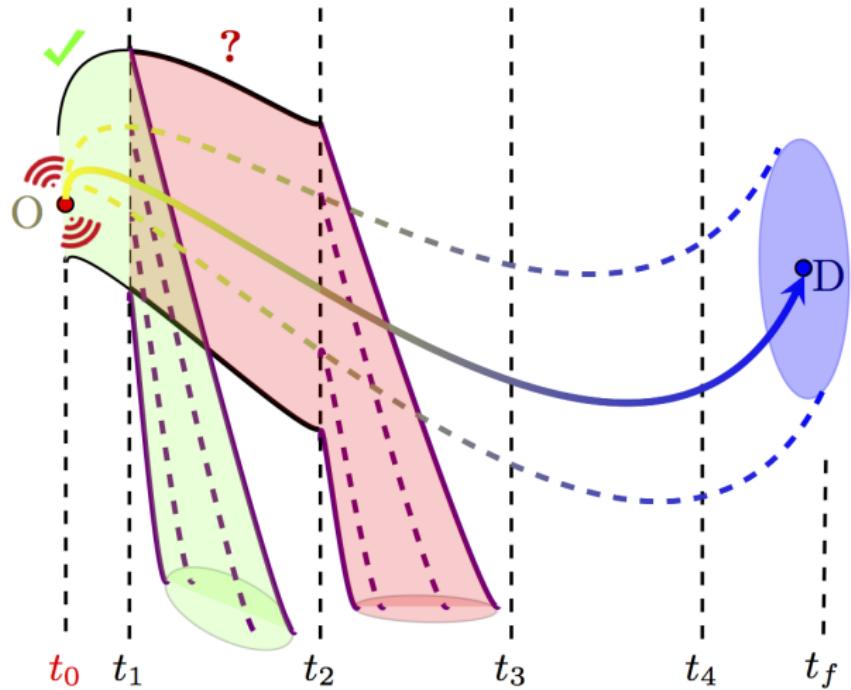


# 4D Flight + Landing Tubes $\{\mathcal{F}_{[t_j, t_{j+1})}, \mathcal{L}_{[t_{j+1}, t_{j+2})}\}$



# Motion Protocol: $t = t_0$

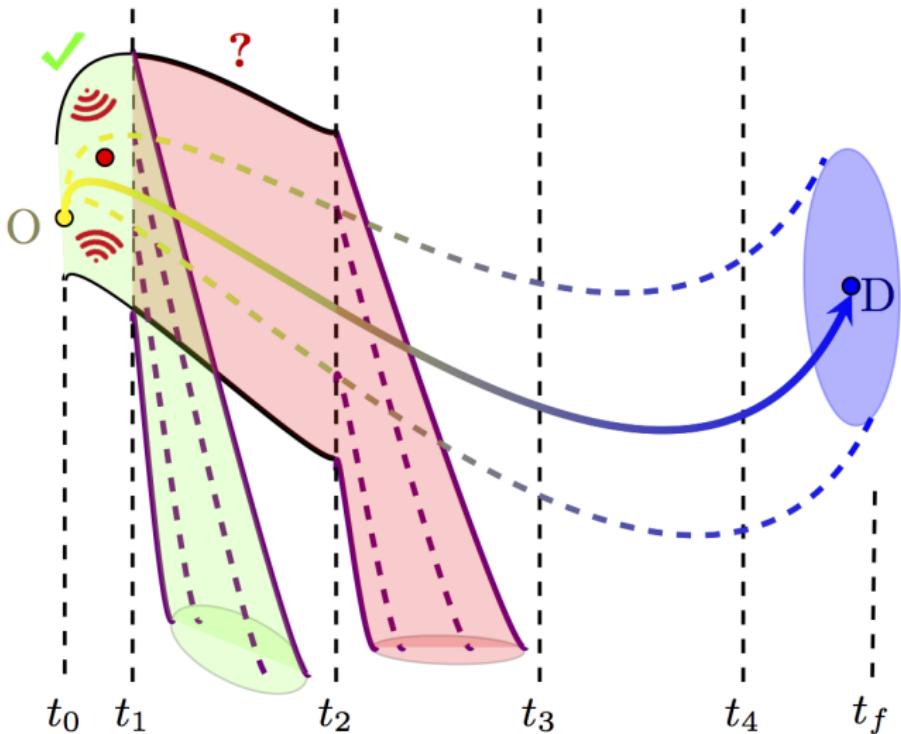
**IF:** Have all + ACKs for  $\{\mathcal{F}_{[t_0, t_1)}, \mathcal{L}_{[t_1, t_2)}\}$  **AND**  $D \in \mathcal{R}_{\pi_F}(\{O\}, t_f - t_0)$



**THEN:** Take-off **AND** broadcast req. for  $\{\mathcal{F}_{[t_1, t_2)}, \mathcal{L}_{[t_2, t_3)}\}$

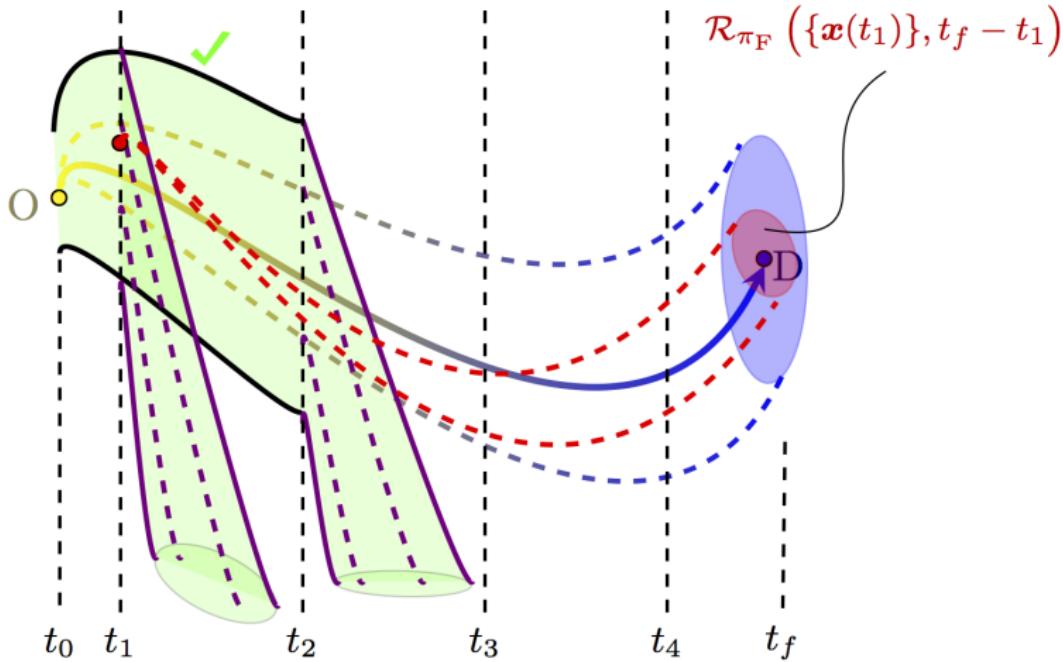
# Motion Protocol: $t \in [t_0, t_1)$

Listening for  $\pm$  ACKs,  $\boldsymbol{x}(t) \in \mathcal{F}_{[t_0, t_1)}$



# Motion Protocol: $t = t_1$

IF: All + ACKs AND  $D \in \mathcal{R}_{\pi_F} (\{\mathbf{x}(t_1)\}, t_f - t_1)$

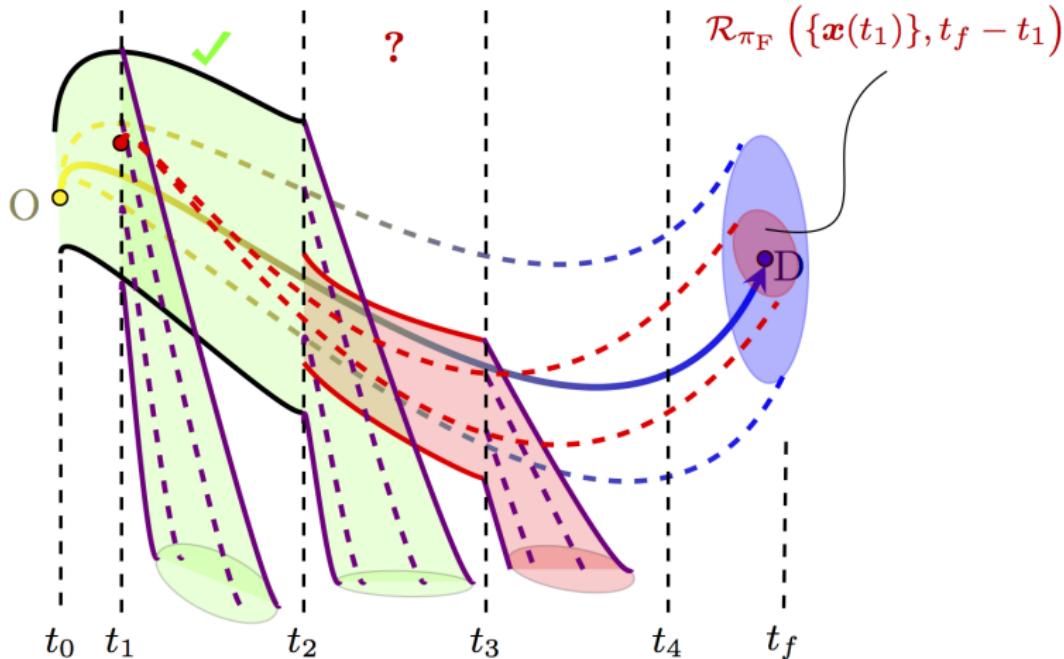


THEN: Continue in  $\mathcal{F}_{[t_1, t_2)}$  AND broadcast req. for  $\{\mathcal{F}_{[t_2, t_3)}, \mathcal{L}_{[t_3, t_4)}\}$

ELSE: Abort mission via  $\mathcal{L}_{[t_1, t_2)}$

# Motion Protocol: $t = t_1$

IF: All + ACKs AND  $D \in \mathcal{R}_{\pi_F} (\{\mathbf{x}(t_1)\}, t_f - t_1)$

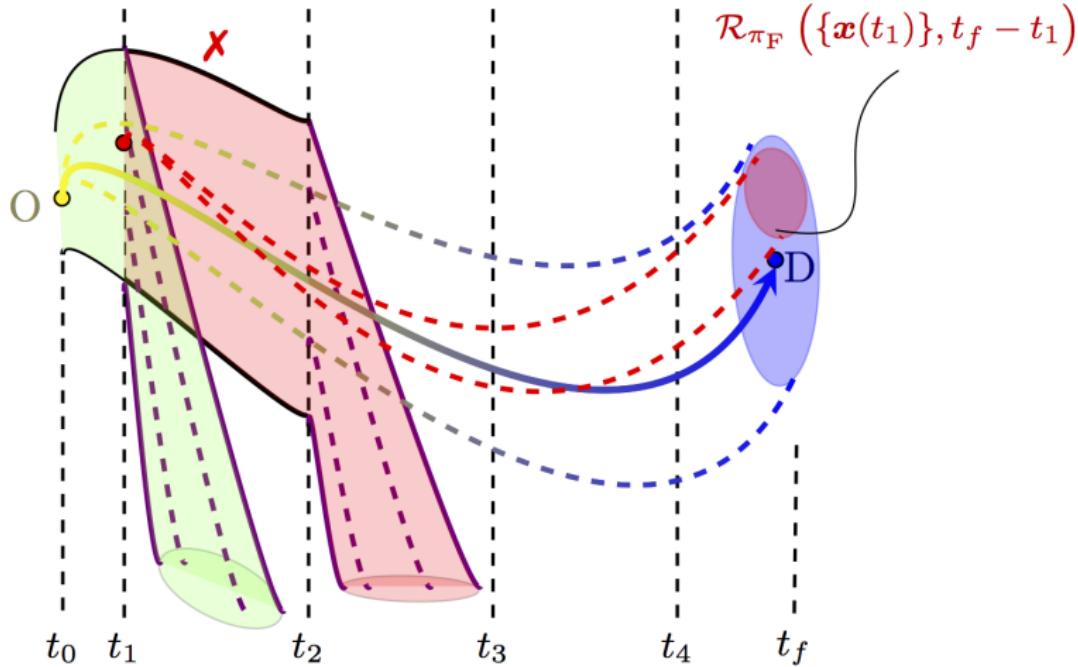


THEN: Continue in  $\mathcal{F}_{[t_1, t_2)}$  AND broadcast req. for  $\{\mathcal{F}_{[t_2, t_3)}, \mathcal{L}_{[t_3, t_4)}\}$

ELSE: Abort mission via  $\mathcal{L}_{[t_1, t_2)}$

# Motion Protocol: $t = t_1$

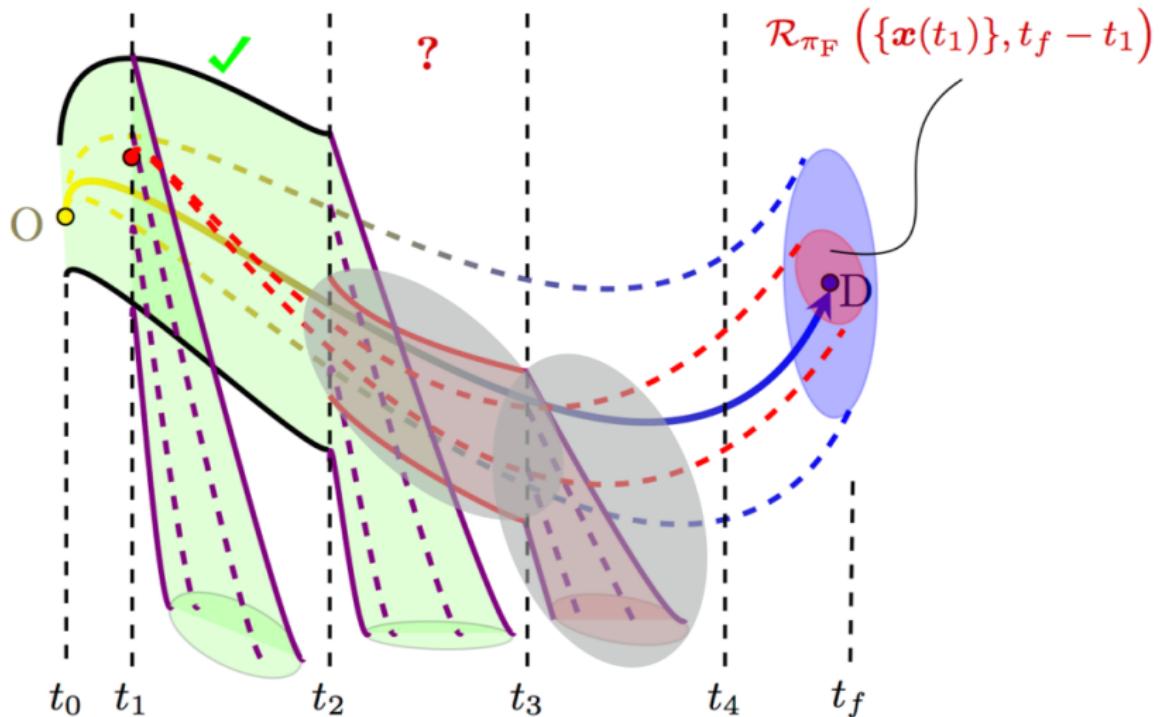
IF: All + ACKs AND D  $\notin \mathcal{R}_{\pi_F} (\{\mathbf{x}(t_1)\}, t_f - t_1)$



THEN: Continue in  $\mathcal{F}_{[t_1, t_2)}$  AND broadcast req. for  $\{\mathcal{F}_{[t_2, t_3)}, \mathcal{L}_{[t_3, t_4)}\}$

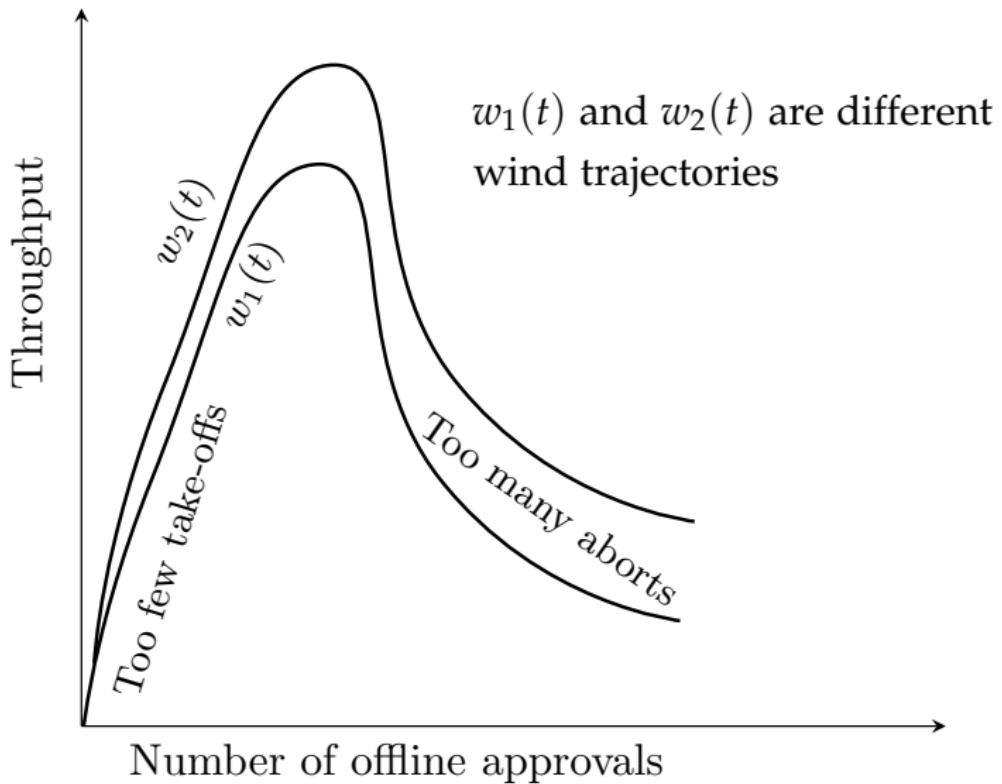
ELSE: Abort mission via  $\mathcal{L}_{[t_1, t_2)}$

# Algorithms for Motion Protocol

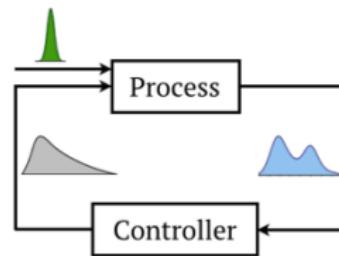


Compute minimum volume outer ellipsoids: SDP

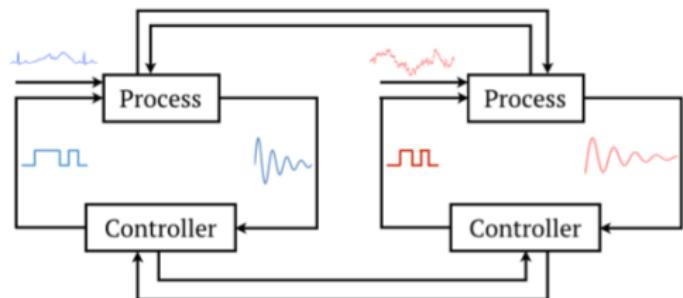
# Proposed Architecture: Performance



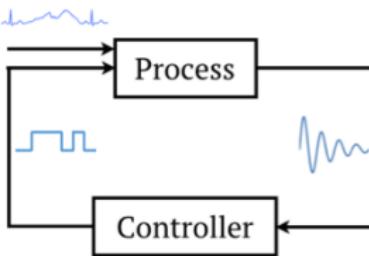
Continuum of systems



Finitely many systems



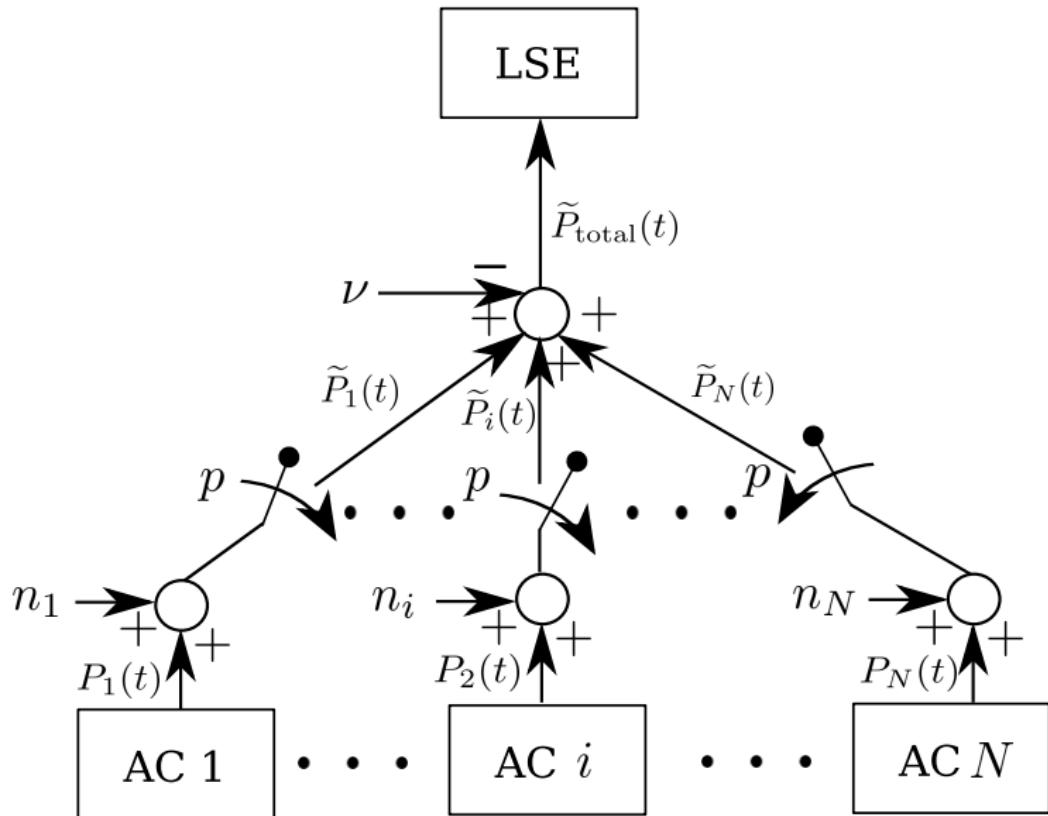
One system



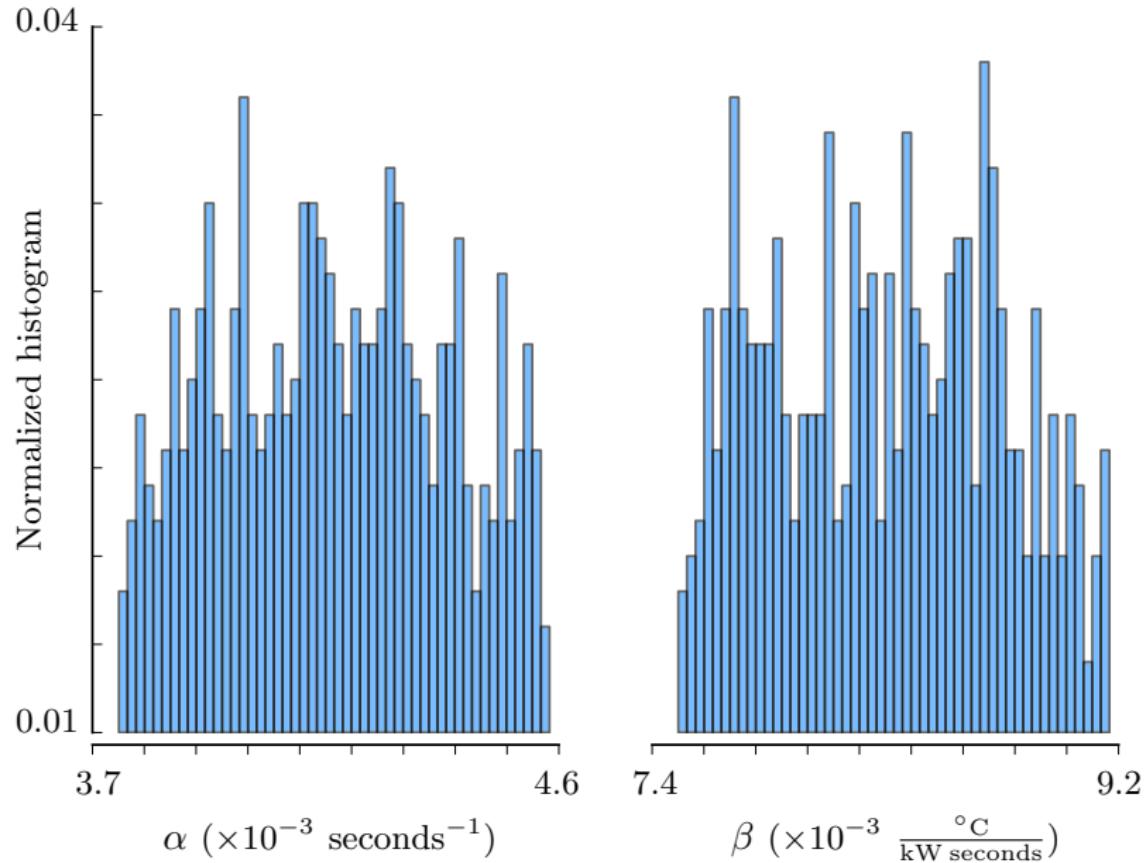
**Thank You**

# **Backup Slides for Part I**

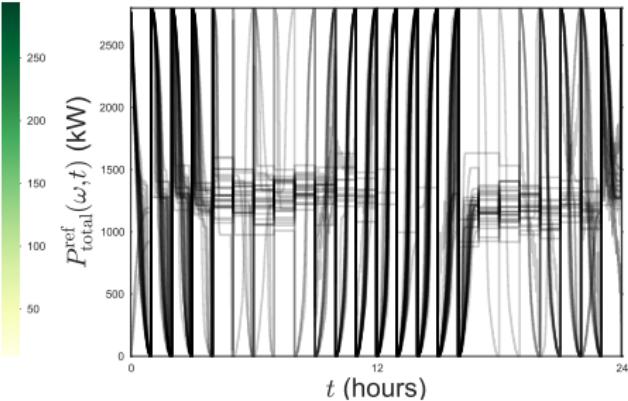
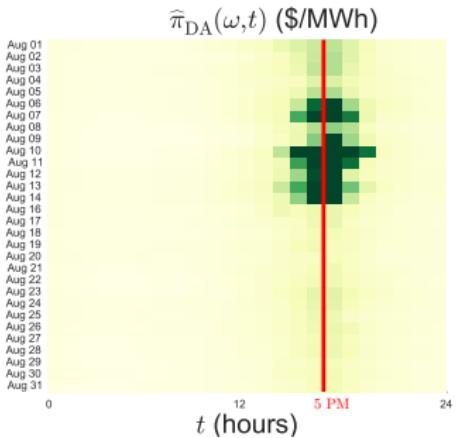
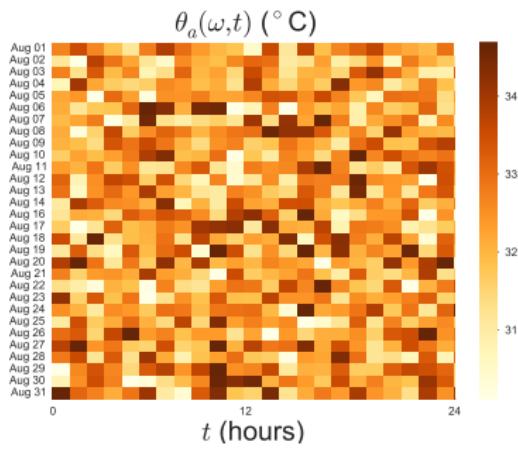
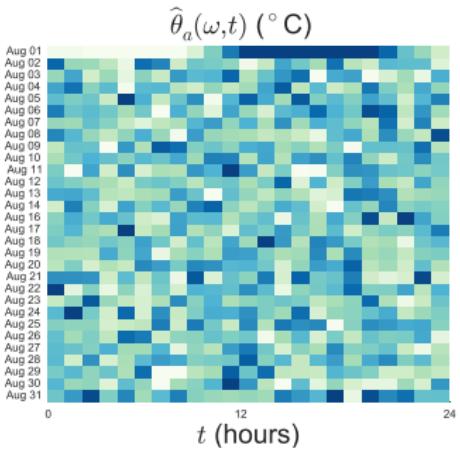
# Differential Privacy Preserving Sensing



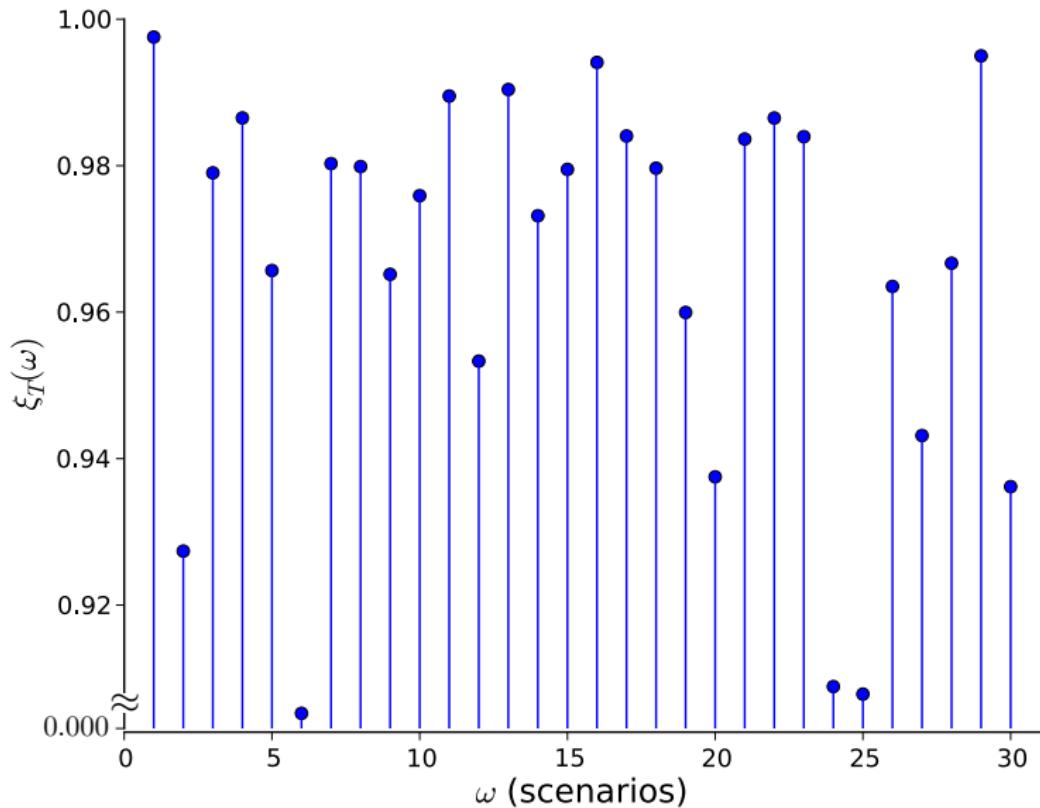
# Distribution of Parameters $\alpha$ and $\beta$



# Houston Data for August 2015



# Limits of Control Performance



# **Backup Slides for Part II**

# Matrix Geometric Mean

The minimal geodesic  $\gamma^* : [0, 1] \mapsto \mathbf{S}_n^+$  connecting  $\gamma(0) = S_d$  and  $\gamma(1) = S_1^{-1}$ , associated with the Riemannian metric  $g_A(S_d, S_1^{-1}) = \text{tr} (A^{-1} S_d A^{-1} S_1^{-1})$ , is

$$\begin{aligned}\gamma^*(t) &= S_d \#_t S_1^{-1} = S_d^{\frac{1}{2}} \left( S_d^{-\frac{1}{2}} S_1^{-1} S_d^{-\frac{1}{2}} \right)^t S_d^{\frac{1}{2}} \\ &= S_1^{-1} \#_{1-t} S_d = S_1^{-\frac{1}{2}} \left( S_1^{\frac{1}{2}} S_d S_1^{\frac{1}{2}} \right)^{1-t} S_1^{-\frac{1}{2}}\end{aligned}$$

## Geometric Mean:

$$\gamma^* \left( \frac{1}{2} \right) = S_d \#_{\frac{1}{2}} S_1^{-1} = S_1^{-1} \#_{\frac{1}{2}} S_d$$

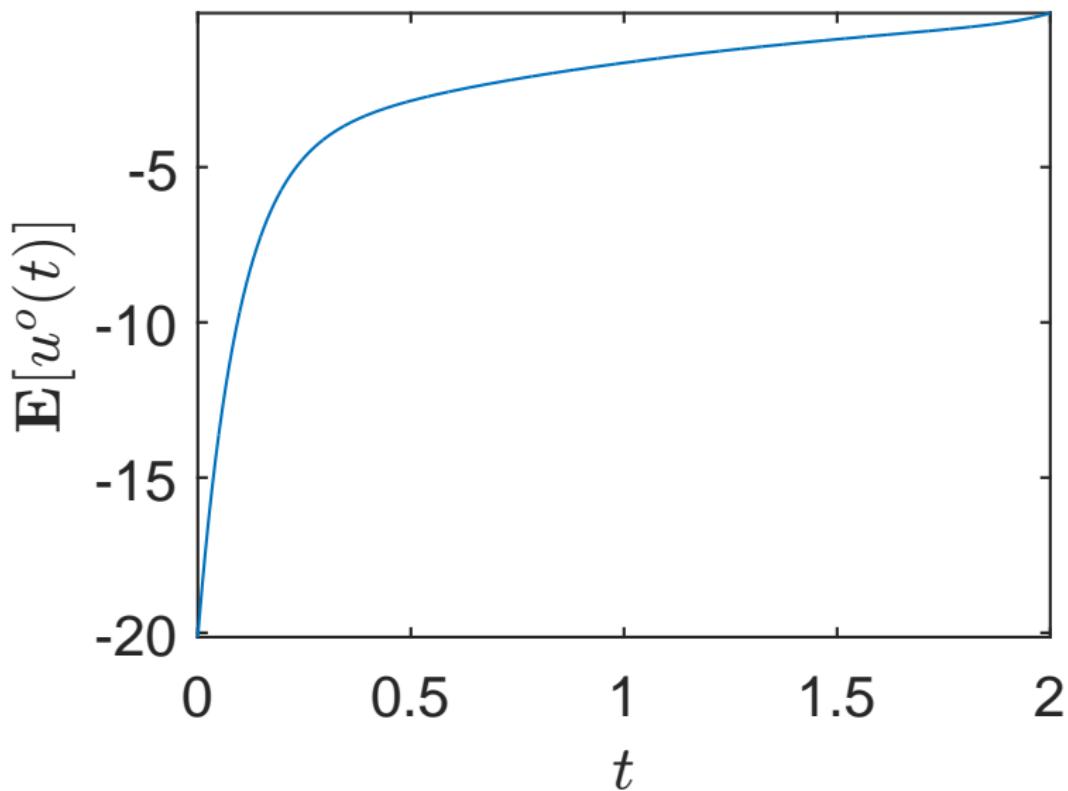
## Example

$$\begin{pmatrix} dx_1 \\ dx_2 \end{pmatrix} = \begin{bmatrix} 0 & 1 \\ 2 & -3 \end{bmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} dt + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u dt + \begin{bmatrix} 0.01 \\ 0.01 \end{bmatrix} dw$$

$$\rho_0 = \mathcal{N} \left( (1, 1)^\top, I_2 \right), \quad \rho_d = \mathcal{N} \left( (0, 0)^\top, 0.1 I_2 \right),$$

$$Q = 100 I_2, \quad R = 1, \quad M = I_2, \quad t_1 = 2$$

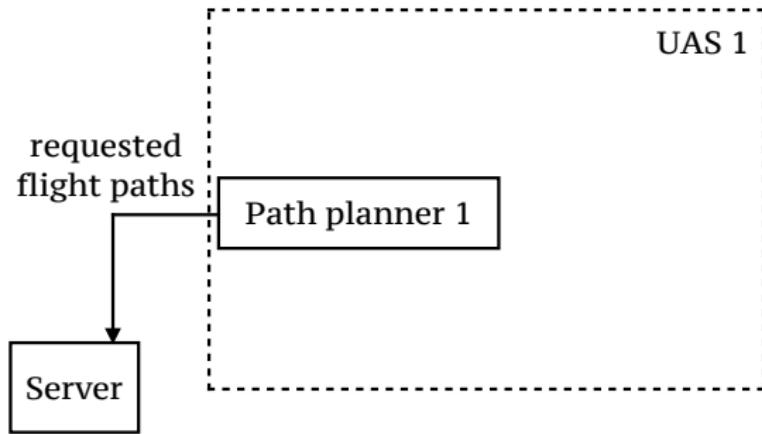
# Expected Optimal Control



# **Backup Slides for Part III**

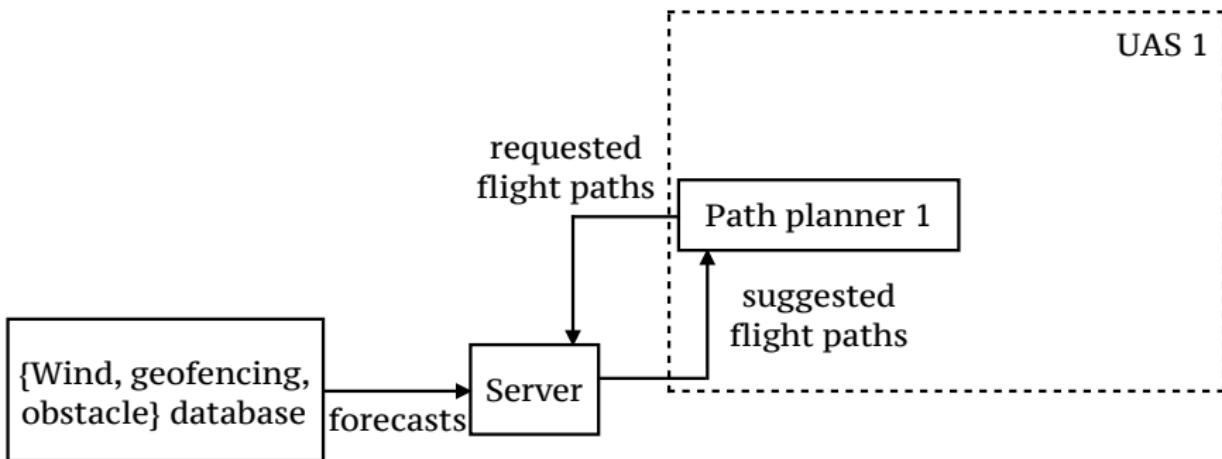
# Our Proposed Architecture

## Offline Path Planning and Conflict Resolution



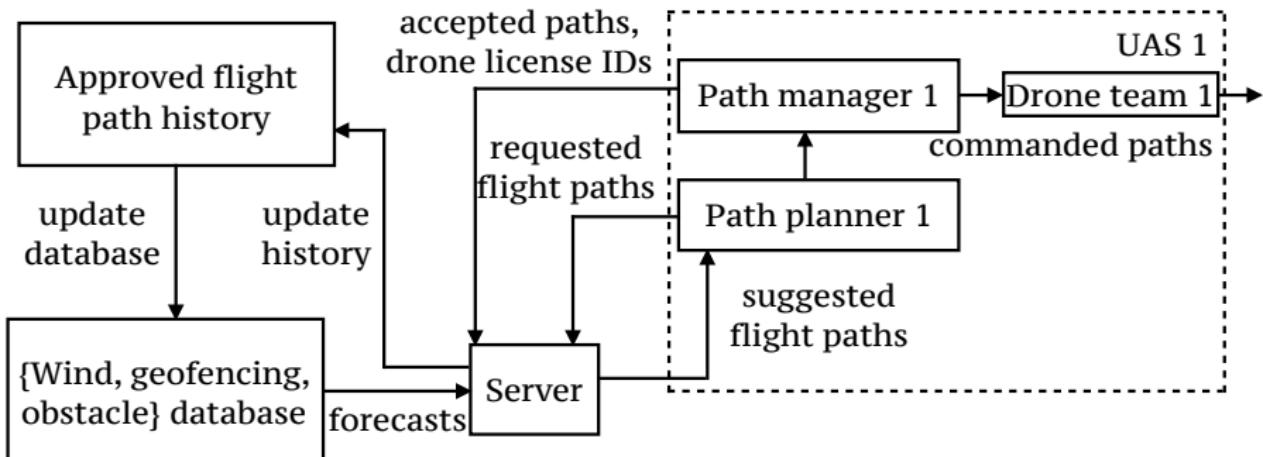
# Proposed Architecture

## Offline Path Planning and Conflict Resolution



# Proposed Architecture

## Offline Path Planning and Conflict Resolution



# Proposed Architecture

## Offline Path Planning and Conflict Resolution

