# **AER1517 Control for Robotics**

Assignment 3: Path Integral Policy Improvement

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#### General Information



- Download handout and script templates from Quercus
  - Problem 3.1 Quadrotor Agile Maneuver with Path Integral Policy Improvement (PI2)
- Due on April 13 (Monday) 23:59
- Submission through Gradescope
  - A single PDF with solutions and requested scripts
  - Both typed and scanned handwritten solutions are accepted
- Office hours: April 2 and April 9 (Thursdays) 14:00 via Zoom
- Send emails for any questions not covered in the office hours



### Problem 3.1 Agile Quadrotor Maneuver w/ PI2 | Overview



- Recap: In Assignment 1, we implemented a model-based optimal controller, ILQC, that enabled a quadrotor to fulfill the task of flying to a goal state and/or passing through a predefined via-point
  - Assumed perfect knowledge of system dynamics and saw good performance in simulation
  - In reality, we typically do not have perfect knowledge about the system dynamics, and this often results in sub-optimal performance
- In this assignment, we will explore a learning-based method that allows the quadrotor to achieve similar goals as in Assignment 1 but *not* relying on perfect knowledge of the system dynamics



#### Problem 3.1 Agile Quadrotor Maneuver w/ PI2 | Overview



- Path Integral Policy Improvement (PI2)
  - Provides reference trajectory and controller
  - Does not require and use domain knowledge (e.g., model of the system)
  - But, performance relies on good initialization
- Overall idea: Combine model-based approach (i.e., ILQC) and learning-based approach (i.e., PI2)
  - Initialize PI2 with a trajectory and controller obtained from ILQC
  - Improve performance using PI2 algorithm

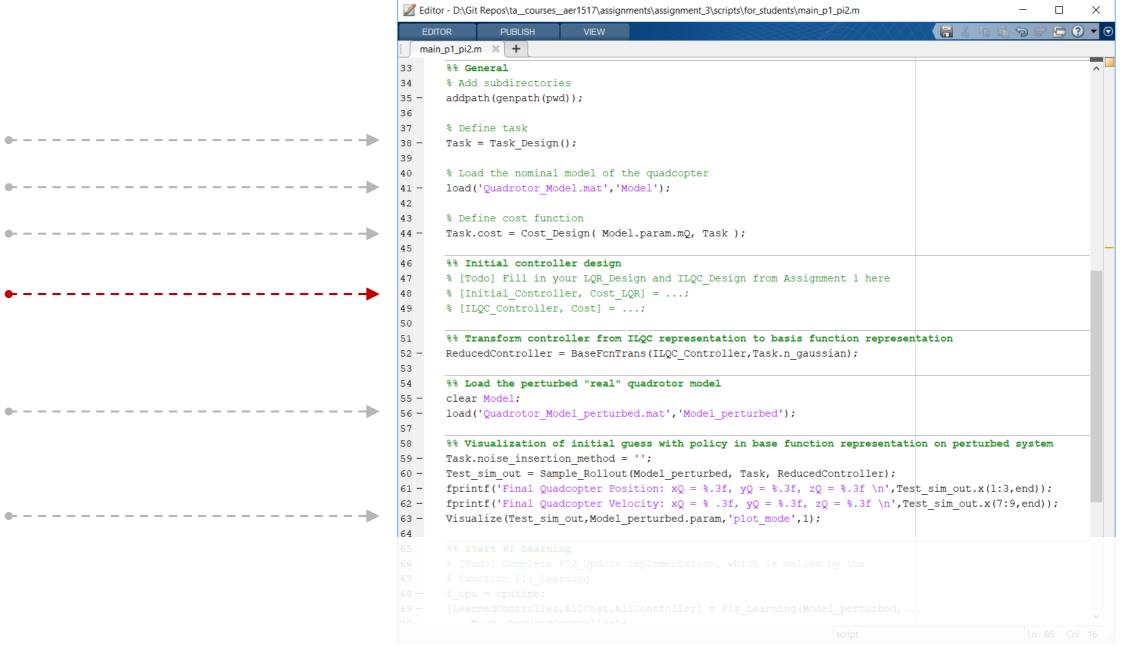


# Problem 3.1 Agile Quadrotor Maneuver w/ PI2 | Code Structure



#### Initialization in main\_p1\_pi2.m

- Define task
- Load nominal model
- Define cost
- LQR and ILQC design (from Assignment 1)
- Load test model (model w/ perturbed parameters)
- Visualization (ILQC)

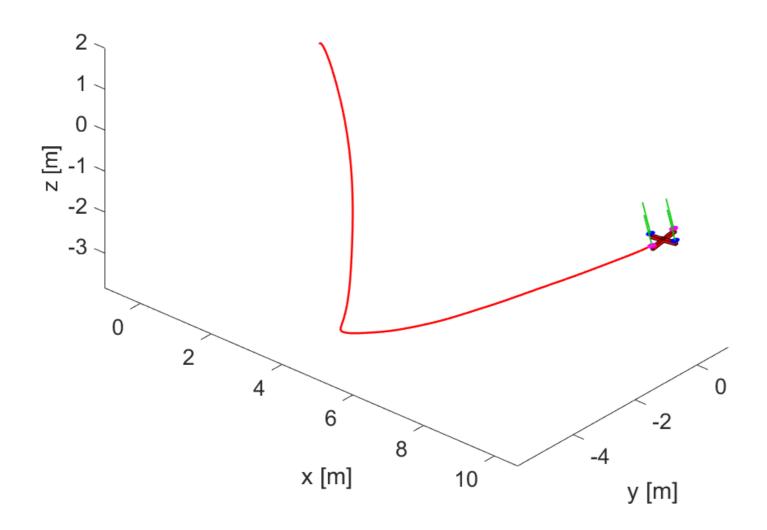




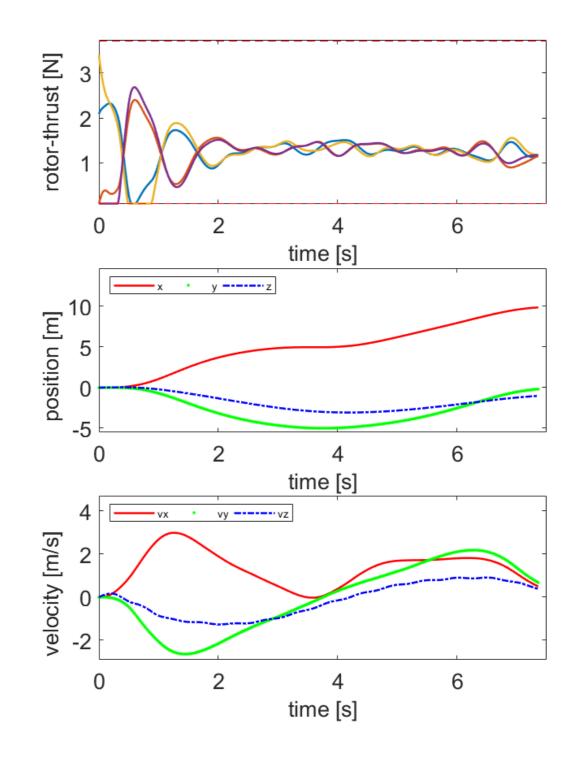
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#### **Quadrotor Flight Simulation**



**ILQC** Initialization



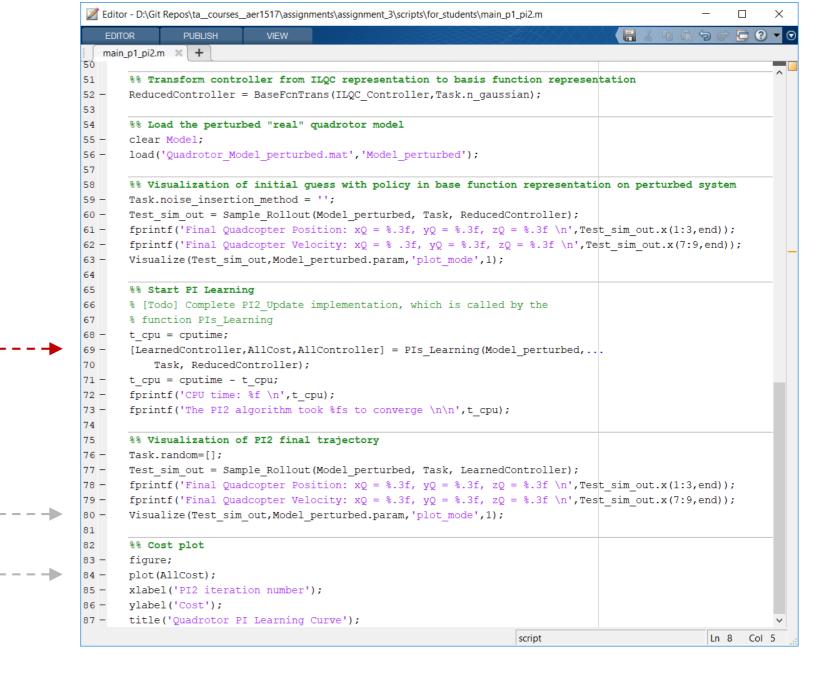


# Problem 3.1 Agile Quadrotor Maneuver w/ PI2 | Code Structure



#### PI2 starter code in main\_p1\_pi2.m

- Pls\_Learning (Todo: Pl2\_Update)
- Visualization (PI2)
- Plot cost





### Problem 3.1 Agile Quadrotor Maneuver w/ PI2 | Code Structure



#### PI2 Algorithm

#### Pls\_Learning (Provided)

```
Algorithm 10 General PI2 Algorithm

given

The cost function J = \Phi(\mathbf{x}(t_f)) + \int_s^{t_f} \left(q(t, \mathbf{x}), \frac{1}{2}\mathbf{u}^T\mathbf{R}\mathbf{u}\right) \, dt

A linear model for function approximation
Initialize \boldsymbol{\theta}_i for each input i
Initialize exploration noise standard deviation c

while maximum number of iteration not reached \mathbf{do}

Create K rollouts of the system with the perturbed parameter \boldsymbol{\theta}_i + \boldsymbol{\epsilon}_i for each input i, where \boldsymbol{\epsilon}_{i,j} \sim \mathcal{N}(0,c^2)

Parameter update

\Delta \boldsymbol{\theta} \leftarrow \text{PI2\_Update}(\text{Task}, \text{ batch\_sim\_out}, \text{ batch\_cost})
\boldsymbol{\theta} \leftarrow \boldsymbol{\theta} + \omega \Delta \boldsymbol{\theta}

Decrease c for noise annealing
```

#### PI2\_Update (To be Implemented)

```
for the ith control input {\bf do}

for each time s {\bf do}

Calculate the Return starting from s for the kth rollout

R(\tau^k(s)) = \Phi({\bf x}(t_f)) + \int_{s}^{t_f} \left(q(t,{\bf x}), \frac{1}{2}{\bf u}^T {\bf R} {\bf u}\right) \, dt

Calculate \alpha starting from s for the kth rollout

\alpha^k(s) = \exp\left(-\frac{1}{\lambda}R(\tau^k(s))\right) / \sum_{k'=1}^K R(\tau^{k'}(s))

Calculate the time-varying parameter increment \Delta \theta_i(s)

\Delta \theta_i(s) = \sum_{k=1}^K \alpha^k(s) \frac{{\bf \Upsilon}(s) {\bf \Upsilon}(s)^T}{{\bf \Upsilon}(s)^T {\bf \Upsilon}(s)} \epsilon_i^k(s)

end for

for the jth column of \Delta \theta_i matrix, \Delta \theta_{i,j}, {\bf do}

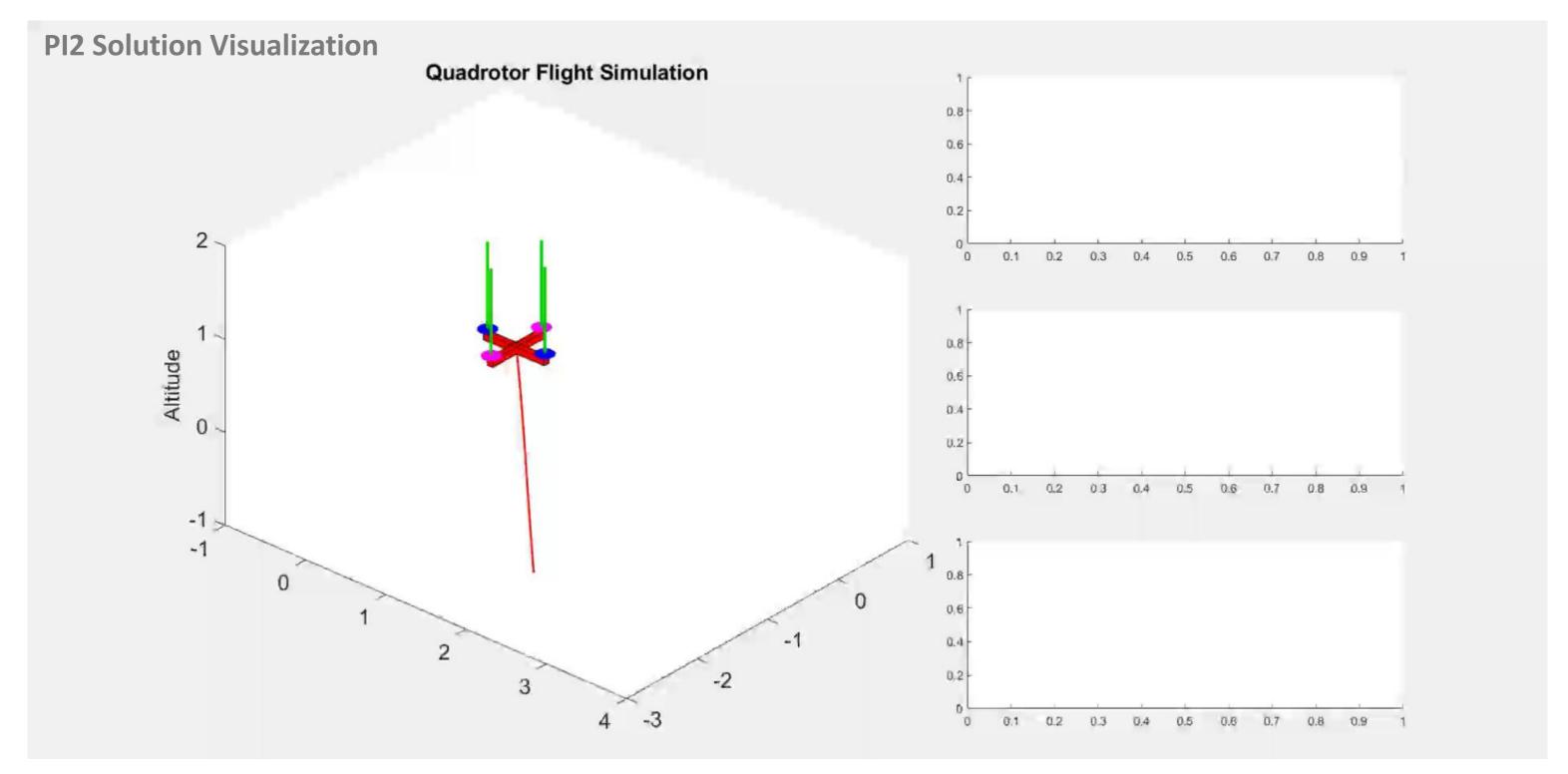
Calculate the time-averaged parameter vector

\Delta \theta_{i,j} = \left(\int_{t_0}^{t_f} \Delta \theta_{i,j} \circ {\bf \Upsilon}(s) \, ds\right) \cdot / \int_{t_0}^{t_f} {\bf \Upsilon}(s) \, ds,

where \circ and \cdot/ denote element-wise multiplication and division. end for end for
```

end while





Link to video: <a href="https://drive.google.com/file/d/155NH1ptAQOZCHrX4SaGbrKXTPVmkysX3/view?usp=sharing">https://drive.google.com/file/d/155NH1ptAQOZCHrX4SaGbrKXTPVmkysX3/view?usp=sharing</a>

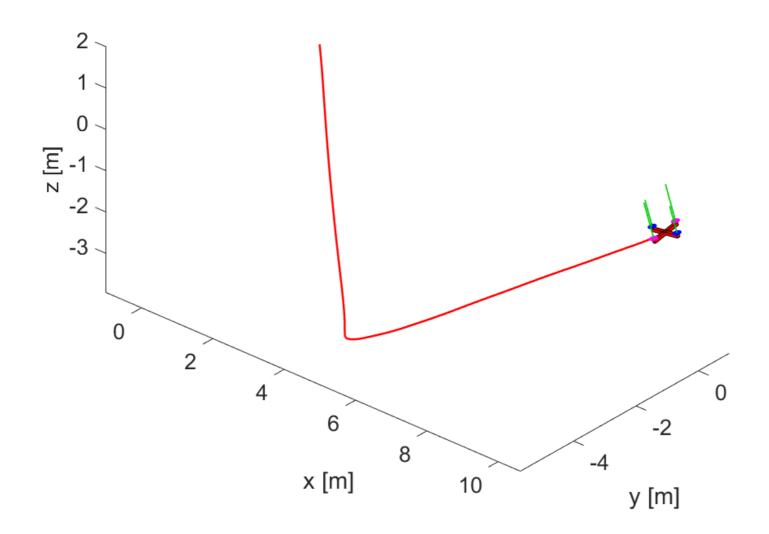


rotor-thrust [N]



10

#### **Quadrotor Flight Simulation**



time [s]

6

6

**PI2 Solution** 

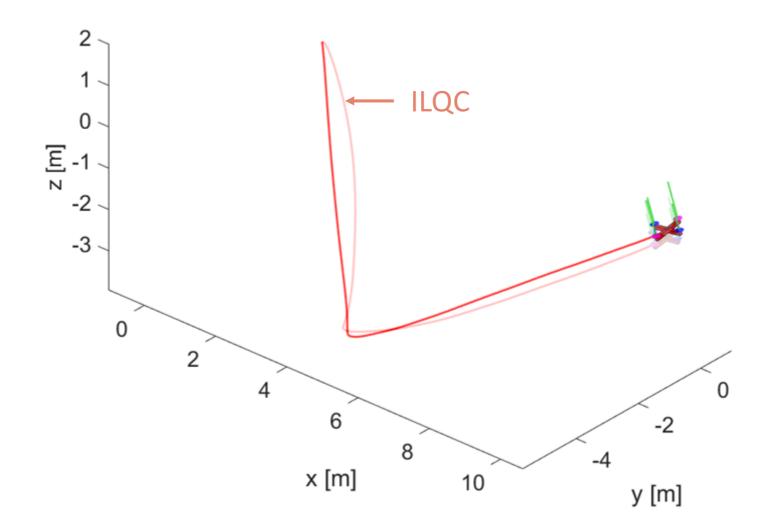


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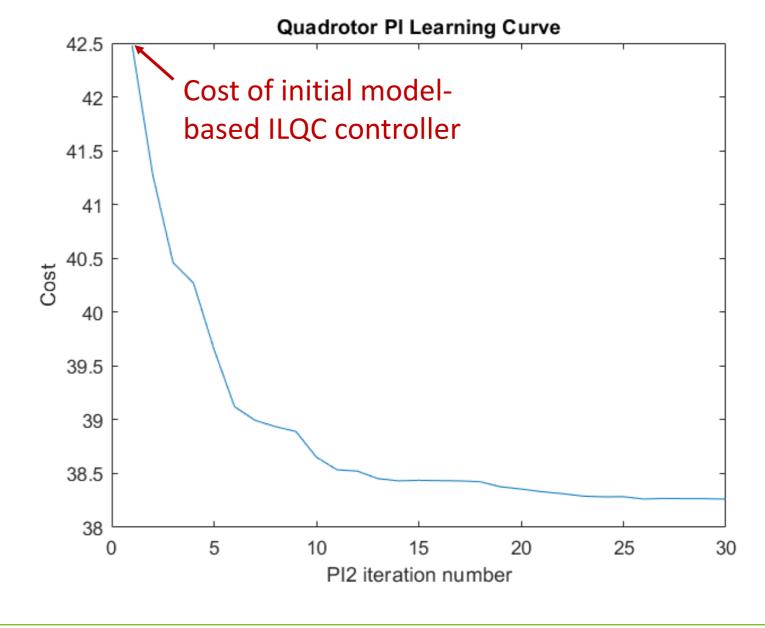
time [s]



#### **Quadrotor Flight Simulation**



# Performance improvement of PI2 controller over 30 iterations





**PI2 Solution** 

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