Model Documentation of the 'Boeing B-747 aircraft'

1 Nomenclature

1.1 Nomenclature for Model Equations

- x state vector
- u control input vector
- w noise vector
- z regulated output vector
- y measurement vector

2 Model Equations

State Vector and Input Vector:

$$x \in \mathbb{R}^4 u$$
 $\in \mathbb{R}^2 w \in \mathbb{R}^4 z$ $\in \mathbb{R}^4 y \in \mathbb{R}^2$

System Equations:

$$\dot{x}(t) = Ax(t) + B_1 w(t) + Bu(t) \tag{1a}$$

$$z(t) = C_1 x(t) + D_{11} w(t) + D_{12} u(t)$$
(1b)

$$y(t) = Cx(t) + D21w(t)$$
(1c)

Outputs: z

2.1 Exemplary parameter values

| Symbol | Value | | | |
|----------|---|---------------------------------------|--------|---------|
| A | 0.9801 | 0.0003 | -0.098 | 0.0038 |
| | -0.3868 | 0.9071 | 0.0471 | -0.0008 |
| | 0.1591 | -0.0015 | 0.9691 | 0.0003 |
| | -0.0198 | 0.0958 | 0.0021 | 1.0 |
| В | -0.0001 | 0.0058 | | _ |
| | 0.0296 | 0.0153 | | |
| | 0.0012 | -0.0908 | | |
| | 0.0015 | 0.0008 | | |
| B_1 | -0.0001 | 0.0058 | | |
| | 0.0296 | 0.0153 | | |
| | 0.0012 | -0.0908 | | |
| | 0.0015 | 0.0008 | | |
| C_1 | [1.0 0] | $\begin{bmatrix} 0 & 0 \end{bmatrix}$ | | |
| | 0 1.0 | 0 0 | | |
| | 0 0 | 1.0 0 | | |
| | 0 0 | 0 1.0 | | |
| C | $\begin{bmatrix} 1.0 & 0 & 0 \end{bmatrix}$ | 0] [| | |
| | 0 0 0 | I | | |
| D_{11} | $[0 \ 0 \ 0]$ | 0 | | |
| | 0 0 0 | 0 | | |
| | $\begin{bmatrix} 0 & 0 & 0 \end{bmatrix}$ | 0 | | |
| | $\begin{bmatrix} 0 & 0 & 0 \end{bmatrix}$ | 0 | | |
| D_{12} | $\begin{bmatrix} 0 & 0 \end{bmatrix}$ | _ | | |
| | 0 0 | | | |
| | 1.0 0 | | | |
| | $\begin{bmatrix} 0 & 1.0 \end{bmatrix}$ | | | |
| D_{21} | $[0 \ 0 \ 0]$ | 0] | | |
| | $\begin{bmatrix} 0 & 0 & 0 \end{bmatrix}$ | 0 | | |

3 Derivation and Explanation

This model is part of the "'COMPleib"' - library and was automatically imported into ACKREP.

The original description was:

AC5 Boeing B-747 aircraft T. Ishihara, H.-J. Guo and H. Takeda, "A Design of Discrete-Time Integral Controllers with Computation Delays via Loop Transfer Recovery", AUTO, Vol. 28, Nr. 3, pp. 599-603, 1992

4 Simulation

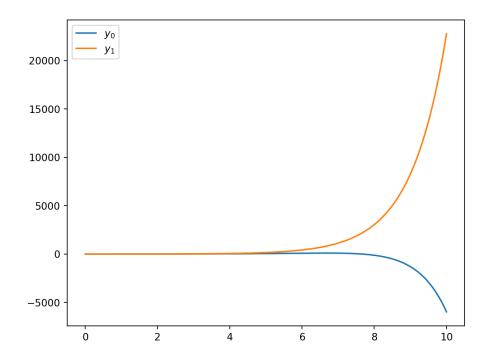


Figure 1: Simulation of the Boeing B-747 aircraft.

References

[1] . Ishihara, H.-J. Guo and H. Takeda, "A Design of Discrete-Time Integral Controllers with Computation Delays via Loop Transfer Recovery", AUTO, Vol. 28, Nr. 3, pp. 599-603, 1992