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A Unified and Quality-Guaranteed Approach for Dubins Vehicle Path Planning With Obstacle Avoidance and Curvature Constraint

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Abstract—Robotic technologies and applications have recently witnessed remarkable advancements. A major challenge is the shortest-path planning problem of a curvature-bounded vehicle from the known starting configuration to visit a target point and finally return to the starting configuration in an obstacle environment. Spurred by this significant issue in robotic surveillance and patrolling applications, this paper proposed the Two-trip Obstacle-environment Relaxed Dubins Problem (TORDP). In TORDP, the vehicle’s target-visiting heading is a critical variable. Analytical approaches have existed for simpler scenarios than TORDP. However, these approaches are unavailable when solving the complex TORDP simultaneously with bounded curvature, variable target heading and unified ability to tackle with- or without- obstacle cases. Hence, we develop the mixed-integer piecewise-linear program (MIPWLP) approach, making the otherwise intractable complex scenario unifiedly solved with guaranteed good quality. Extensive experiments demonstrate that the proposed approach demonstrates effective performance. Furthermore, the objective approximation error in some cases was analyzed to achieve a length near the optimal length within $h^2/(2\sqrt{2})$ tolerance where h is the approximation piece length. The proposed MIPWLP approach could also offer a generalizable optimization framework for broader robotic path-planning applications in constrained environments.

Index Terms—Dubins vehicle, obstacle avoidance, mixed-integer piecewise-linear program, relaxed Dubins problem, intelligent system, path planning.

I. INTRODUCTION

A. Background

SIGNIFICANT progress has recently been witnessed in unmanned vehicle path planning for various platforms such as Unmanned Arial Vehicles (UAVs), cars, and Unmanned Surface Vehicles (USVs) [1], [2], [3]. These vehicles have been widely deployed in applications

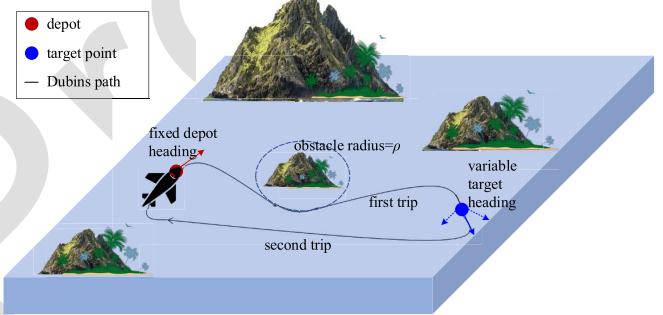


Fig. 1. Shortest-path planning of the curvature-bounded vehicle in environments with obstacles. Sometimes, the obstacles may not exist or exist but are ineffective, as in the second trip—the vehicle’s path does not necessarily touch the obstacle.

such as point-of-interest (POI) reconnaissance, surveillance, patrolling, search and rescue, and field exploration [4], [5]. However, shortest-path planning for these vehicles is often challenging because they are subject to physical constraints, specifically, a bound on the maximum turning curvature prohibiting the vehicle from turning sharply with a small radius [6]. These vehicles are categorized as Dubins vehicles.

This work deals with the vehicle’s curvature-bounded shortest-path planning problems, illustrated in Fig. 1. In the examined scenario, the vehicle, e.g., a fixed-wing UAV, is taking off at the depot with a fixed heading angle (the runway angle). The vehicle is tasked to visit a point of interest within the work area **with any heading angle** and finally return back to the depot with the fixed heading angle, ensuring the entire surveillance two-trip path is the shortest. Application examples may be that a UAV daily frequently visits a coral reef target with known coordinates, optimizing the target-reaching angle to shorten the go-and-back two-trip for energy saving [6]. This scenario is called the Two-trip Obstacle-environment Relaxed Dubins Problem (TORDP) by us.

The main challenge lies in that the **curvature-bounded** vehicle’s **visiting angle (heading)** at the target point is variable (i.e., free and should be optimized), and the environment may have an **obstacle** which represents an island or

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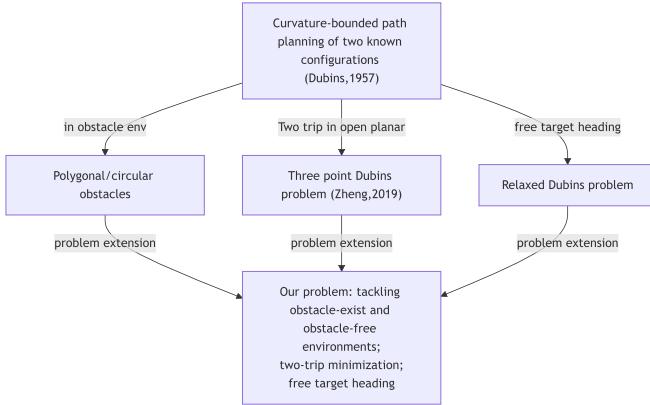


Fig. 2. Position and features of TORDP.

an adversarial area. Additionally, the obstacle is sometimes ineffective because it lies far from the shortest path or it does not exist. Hence, the proposed method requires a **unified tackling ability** for the effective or ineffective obstacle cases.

B. Related Work

TORDP is significant in various robotic applications and is complex, as shown in Fig. 2. Effective approaches exist for some simpler scenarios, e.g., considering only single-trip Dubins path [7], or fixed target heading [8], or obstacle-free scenarios [9]—some approaches are even analytical [10], [11], but we found that few of them have simultaneously considered the three features coming from robotic applications: (1) the entire two-trip Dubins path minimization, (2) the free target heading, and (3) the unification of obstacle and free-of-obstacle path. This missing is mainly because previous scenarios/applications did not encounter such a high complexity task; thus, their algorithm may not be capably applied to our scenario. The following is the detailed literature review.

In 1957, Dubins [10] conducted pioneering research on path planning under physical constraints [12]. He concluded that the shortest path between two positions with given heading angles, i.e., two configurations, must fall into one of the two path categories (each with no more than three path segments): CSC or subpaths, CCC or subpaths (where C represents minimum-radius circular-arc motion primitive, while S represents straight ahead motion primitive). Subsequently, researchers have followed the principles outlined by Dubins and Pontryagin [13], [14], [15] and have developed accelerated calculation methods for two known configurations [8], [16], [17]. When engaged with free target heading orientation angle (HOA) at the target point, it forms the Relaxed Dubins Problem (RDP) [18]. However, previous RDP solutions assumed obstacle-free environments and cannot be extended to the obstacle environment due to the difficulty of coupling the variable target heading and obstacles [9].

When obstacles exist in the work area, they impose a significant challenge in addressing the curvature-bounded vehicle [19], [20], [21]. Early studies focused on polygonal obstacles [22], [23], [24] for simple treatment of collision checking,

without being fit for curvature-constrained complex settings. Subsequently, researchers focused on circular obstacles, and later approaches incorporated geometrical or analytical solutions for fixed target heading angles [25], [26]. Since these methods need the target heading to be geometrically fixed, they cannot handle free target angles in the two-trip minimization problem effectively because a periodic variable target angle brings structural difficulty to previous methods. Recent extensions have explored time-optimal lengthening and shortening in circular or other environments [27], [28]. However, these algorithms cannot unify obstacle-existing and obstacle-free conditions.

When engaged with two-trip length minimization, there were studies of the three-point Dubins path problem (3PDP) [29], [30], which involves three points, i.e., depot, middle, and target. In 3PDP, the aim is to decide the heading at the middle point to minimize the time from the starting to the middle to the end configurations. Although analytic solutions based on geometry have been proposed for the necessary conditions, current 3PDP solutions can hardly apply to open-plane scenarios. Moreover, scenarios with angle optimization in obstacle environments complicate the models [31], i.e., new discrete or binary variables emerging from angle optimizations complicate the feasible spaces so closed-form solutions hardly exist. Particularly, the periodic property of angles (or the congruent equation constraints) leads to integers in the feasible spaces, making derivative-based programming approaches incapable in this complicated feasible space. Recent research suggests solutions for visiting problems in an open environment in three-dimensional or similar settings [32], [33], [34], [35]. These schemes utilize differential geometry or handcrafted adaptive algorithms. Still, the heading orientation angle minimization is seldom considered.

C. Contribution

The limitations of previous research are that: previous work based on derivatives was hindered by the periodic property of HOA. Other approaches were hindered by constraints involving the trigonometric sine and cosine functions and by the lack of parameterized motion primitive derivations. Unlike existing methods, the proposed solution managed to break through. This work proposes a unified, quality-guaranteed mixed-integer piecewise-linear program (MIPWLP) for the TORDP, which exhibits high performance to problems that previous work cannot tackle featuring simultaneously free HOA, obstacle-existing, curvature-bounded constraints. This work's contributions can be summarized as follows:

- 1) **We propose a novel TORDP problem and establish it as a mathematical program.** The resulting program introduces integer variables to derive the three basic motion primitives with complex numbers and can unify scenarios with or without obstacles. It is probably the first program for free HOA, obstacle-existing, curvature-bounded, two-trip shortest-path planning.
- 2) **We develop mixed-integer and piecewise-linear strategies for the mathematical constraints involving congruent equations, trigonometric equations and direction consistencies.** MIPWLP is derivative-free,

TABLE I
THE NOTATIONS AND EXPLANATIONS

Symbol	Explanation
C/S/D	minimum-curvature circular/straight/obstacle-circumventing path segment
L/R	belonging to C, meaning left turn/right turn circular path segment
x_I, y_I, α_I	the known coordinates and heading at the depot
x_G, y_G, θ	the known coordinates and the variable heading at the target, with $\theta \in [0, 2\pi)$
(x, y, α)	a variable configuration, for parameterized move deduction use. $x, y \in R, \alpha \in [0, 2\pi)$
(x', y', β)	a variable configuration after a certain move from (x, y, α) . $x', y' \in R, \beta \in [0, 2\pi)$
$\rho_{min}/\rho_1/\rho$	the normalized turn, the obstacle, and the obstacle-circumventing radii.
θ	heading $\in [0, 2\pi)$ at the target point. The coordinates of the target are also x', y'
ν	a move length or radian (the two will be the same because the turn radius is one)
f	$f=+1$ means vehicle's left turn; 0 means straight forward; -1 means right turn
$y = \sin[x]$	y is constrained with a piecewise-linear approximation of the sine function
cross-product $xp = \vec{a} \times \vec{b}$	$xp > 0$ if \vec{b} is in the left-hand half-plane of \vec{a} ; = 0 if co-linear; < 0 otherwise

making the program solvable by solvers. By approximating trigonometric functions with piecewise-linear functions and introducing integers for periodicity, we achieve an effective program. Moreover, several computational geometrical techniques, such as cross-product, dot-product, and directional consistencies, are employed to meet the constraints of the program.

3) **The approximation error (tolerance) between MIP-WLP to the analytic optimal length in some cases is successfully analyzed**, guaranteeing a tolerance smaller than $h^2/(2\sqrt{2})$ concerning the piece length. Thus, MIPWLP is a quality-guaranteed (near)-shortest path planning approach.

4) **Our algorithm exhibits high performance across different cases, offers highly accurate solutions and unifiedly tackles cases with and without obstacle**. Notably, some cases were never solvable by any other existing approaches. It requires a nominal time cost in seconds. High-fidelity UAV simulation demonstrates the effectiveness of our method on path planning.

The remainder of this paper is organized as follows. Section II formally describes TORDP, the derivation of three motion primitives, and presents the preliminary piecewise-linear formulation and its mathematical equations. Section III introduces the proposed method, including the overall detailed approach. Section IV consolidates the experimental results and analyzes relatively simple cases. Section V discusses important considerations about our approach, and finally, Section VI concludes this work and proposes future research directions.

II. PROBLEM DESCRIPTION AND PRELIMINARIES

This section first provides the basics for the Dubins vehicle and formulates the problem examined. Then, the parameterized motion primitives are deducted, followed by the basics of the piecewise-linear program. Table I lists and explains some notations that will be used throughout the paper.

A. Problem Description

The Dubins vehicle is defined as a vehicle that travels at a fixed and non-negative velocity with a curvature no larger than

some specified value [5] and can usually be viewed as a mass point without sizes for path planning. The shortest path of the Dubins vehicle between two configurations is of type CCC or CSC (or a substring thereof), where $C \in \{L, R\}$ denotes left (L) or right (R) turns with the smallest radius permitted, i.e., with shortest paths involving six types: LRL, RLR, RSR, RSL, LSL, or LSR [8].

Without loss of generality, a Dubins vehicle can be normalized to travel at a constant speed of $\mu_s = 1$ m/s with a normalized minimum turning radius of $\rho_{min} = 1$ meter. This TORDP involves optimizing a length function: the path length from the known starting configuration $q_I(x_I, y_I, \alpha_I)$ where 'I' means 'initial', to reach the target point $q_G(x_G, y_G, \theta)$ where 'G' means 'goal', and then return to q_I , with only the target heading $0 \leq \theta < 2\pi$ as the decision variable.

$$\begin{aligned} \min_{\theta} \text{Length} &= \int_0^{t_F} \sqrt{\dot{x}^2 + \dot{y}^2} dt = \int_0^{t_m} 1 dt + \int_{t_m}^{t_F} 1 dt \\ \text{s.t. } \dot{x} &= \mu_s \cos \theta \\ \dot{y} &= \mu_s \sin \theta \\ \dot{\theta} &= u \cdot \frac{\mu_s}{\rho_{min}} = u \\ Z(0) &= q_I, Z(t_m) = q_G, Z(t_F) = q_I \\ (Z(t, 0), Z(t, 1)) &\in \mathcal{R}^2 \setminus \mathcal{C}_{obs}, \end{aligned} \quad (1)$$

where the control $u \in \{+1, 0, -1\}$ represents turning left, going straight, or turning right and u can be viewed as direction variables; t_m is the moment to reach q_G , and t_F is the time backing to the depot configuration; $Z(t)$ is the vehicle's state vector concerning time, $Z(t, 0)$ and $Z(t, 1)$ denote for the x, y coordinates of state $Z(t)$, and \mathcal{C}_{obs} is the obstacle region.

B. Preliminaries

1) **Proposed Derivation of Three Motion Primitives via Complex Numbers and Direction Decision-Variables:** Our program involves vector rotations around a center with a certain radius. Therefore, it is fundamental to derive the parameterized rotation of the three motion primitives L, R, and S, illustrated in Fig. 3. The derivation is as follows: consider a vehicle at

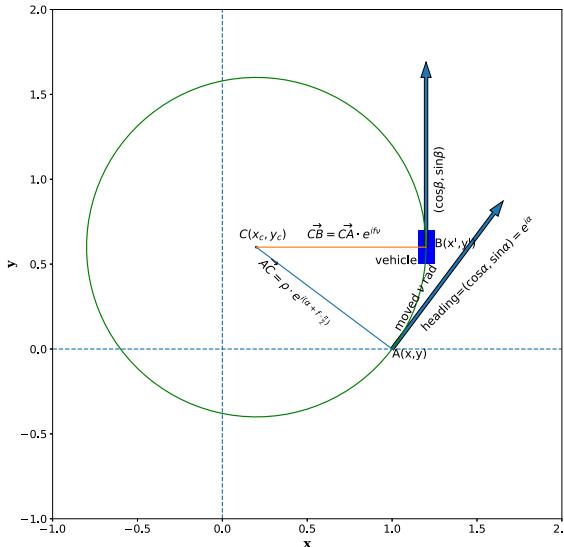


Fig. 3. A vehicle from the configuration $A(x, y, \alpha)$ to $B(x', y', \beta)$, via a parameterized rotating of v radians with the radius ρ perpendicular to the velocity all the time, along the direction f ($f=+1$ left and -1 right). The coordinate system is the complex coordinate system.

any known former configuration $A(x, y, \alpha)$ with the heading angle α . It then moves a distance of v radians on a circle with a radius of ρ . Note that $\rho \geq 1$ because the vehicle may merely conduct a normalized turn with $\rho_{min} = 0$ in an open environment, or may circumvent the obstacle with typical radius $\rho_1 \geq 1$ as assumed in [7]. The abstract of ρ contributes to unifying the obstacle-existing and obstacle-free conditions without explicitly distinguishing them individually. Let the configuration after the move be $B(x', y', \beta)$, where all three values are unknown. The derivation process utilizes complex numbers because of the convenience of representing rotation.

In Fig. 3, the current heading vector at point A is represented as a vector $(\cos \alpha, \sin \alpha)$ in Cartesian coordinates, or $e^{i\alpha}$ in complex number coordinates. Adding $\pi/2$ radians to the current heading yields a normal vector pointing from A toward the center of the left circle, while subtracting $\pi/2$ yields a normal vector pointing toward the right circle. These two normalized vectors are essential to derive the heading at point B . The normalized vector of \vec{AC} using a complex number representation will be the previous complex number $e^{i\alpha}$ multiplied by (the exponential of $i \cdot$ the rotated radian), i.e., $e^{i\alpha} \cdot e^{i\pi/2} = e^{i(\alpha+\pi/2)}$.

Let an indicator integer variable $f = +1$ denote the vehicle turns left and $f = -1$ right. Based on f , one can derive the new configuration B after traveling a radian of v from the current configuration A because (the complex number coordinates of B) = (coordinates of A) + $\vec{AC} + \vec{CB}$. So,

$$\begin{aligned} x' + iy' &= (x + iy) + \rho \cdot e^{i(\alpha+\frac{\pi}{2}f)} + \rho \left[-e^{i(\alpha+\frac{\pi}{2}f)} \right] \cdot e^{ifv} \\ &= \left[x + \rho \cos \left(\alpha + \frac{\pi}{2}f \right) - \rho \cos \left(\alpha + \frac{\pi}{2}f + fv \right) \right] \\ &\quad + i \cdot \left[y + \rho \sin \left(\alpha + \frac{\pi}{2}f \right) - \rho \sin \left(\alpha + \frac{\pi}{2}f + fv \right) \right], \end{aligned}$$

where ρ is the circular radius, which can be 1 in Dubins path turns, or ≥ 1 in obstacle circumventing situations as defined before.

Locating the real and imaginary parts of the complex number $x' + iy'$ and considering the property of the odd multiple of $\pi/2$ provides:

$$\begin{aligned} x' &= \left[x + \rho \cos \left(\alpha + \frac{\pi}{2}f \right) - \rho \cos \left(\alpha + \frac{\pi}{2}f + fv \right) \right] \\ &= x - f\rho \sin(\alpha) + f\rho \sin(\alpha + fv) \\ y' &= \left[y + \rho \sin \left(\alpha + \frac{\pi}{2}f \right) - \rho \sin \left(\alpha + \frac{\pi}{2}f + fv \right) \right] \\ &= y + f\rho \cos(\alpha) - f\rho \cos(\alpha + fv). \end{aligned}$$

Note that, if the vehicle travels straight for a distance of v meters, then $f = 0$. Incorporate it into the above equation as well, naturally:

$$x' = x - f\rho \sin(\alpha) + f\rho \sin(\alpha + fv) + (1 - f^2)v \cos(\alpha), \quad (2)$$

$$y' = y + f\rho \cos(\alpha) - f\rho \cos(\alpha + fv) + (1 - f^2)v \sin(\alpha). \quad (3)$$

It is evident that when turning left or right, $1 - f^2 = 0$. When $f = 0$, the x', y' expression is degraded to a straight movement of length v .

Together with the heading at B ,

$$\beta = \alpha + f \cdot v. \quad (4)$$

Hence, the left-turn/right-turn/straight motion primitives from (x, y, α) are derived using the equations Eq. (2)- (4). **These equations introduce as variables the directions, v and ρ , which can all be optimized in our optimization program.**

Remark 1: Prior work [8] provided a normalized setting of a left/right/straight movement with v radians/meters from an initial configuration (x, y, α) with radius 1, as follows:

$$\begin{aligned} L_v(x, y, \alpha) &= (x + \sin(\alpha + v) - \sin(\alpha), y - \cos(\alpha + v) \\ &\quad + \cos(\alpha), \alpha + v) \end{aligned}$$

$$\begin{aligned} R_v(x, y, \alpha) &= (x - \sin(\alpha - v) + \sin(\alpha), y + \cos(\alpha - v) \\ &\quad - \cos(\alpha), \alpha - v) \end{aligned}$$

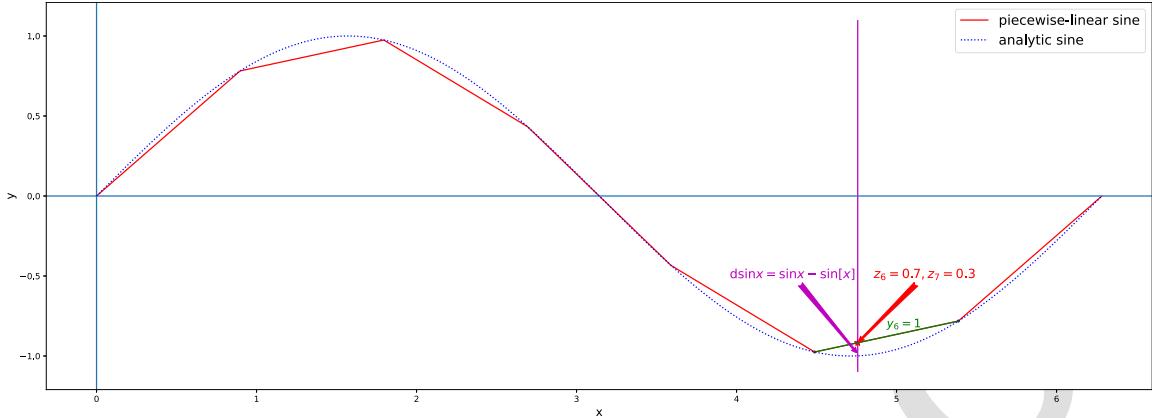
$$S_v(x, y, \alpha) = (x + v \cos(\alpha), y + v \sin(\alpha), \alpha).$$

However, our proposed derivation generalized [8], as equations (2), (3), (4) are parameterized with turning radius ρ and direction f , unifying the obstacle-existing and obstacle-free conditions without explicitly distinguishing them individually.

2) *Preliminaries of Mixed-Integer Piecewise-Linear Program:* The developed MIPWLP should handle sin and cos functions as constraints, which require special treatment to make them tractable by solvers.

Fig. 4 explains the piecewise-linear approximation concept for the sin function within $[0, 2\pi]$. It reveals the linear approximation of $y = \sin(x)$ in an interval by multiple line segments (here, the segment number is chosen as 7 for an example value so as to enhance visualizing the approximation error). Suppose the objective is to minimize $(x - 4.76)^2$ where $x \in [0, 2\pi]$ and y, x have a sine relationship.

A PWL constraint specifies that a relationship $y = f(x)$ must hold between variables x and y , where f is a piecewise-linear function defined by breakpoints. PWL approximation surrogates curves using small line segments, with the tolerance error related to the length of these line segments. The n -breakpoint piecewise-linear relationship $y = PWL(x, ptx, pty)$ between

349 Fig. 4. Coarse illustration of piecewise-linear approximation of $\sin(x)$ on the interval $[0, 2\pi]$ with piece length $2\pi/7 \approx 0.9$ to enhance visualizing the error.

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350 variables x and y , where ptx, pty are the breakpoints' x, y
 351 coordinates, can be constrained as [36]:

$$\begin{aligned} z_1 &\leq y_1 \\ z_i &\leq y_{i-1} + y_i, \quad \forall i = 2, 3, \dots, n-1 \\ z_n &\leq y_{n-1} \\ y_1 + y_2 + \dots + y_{n-1} &= 1 \\ z_1 + z_2 + \dots + z_n &= 1 \\ x &= z_1 \cdot ptx_1 + z_2 \cdot ptx_2 + \dots + z_n \cdot ptx_n \\ y &= z_1 \cdot pty_1 + z_2 \cdot pty_2 + \dots + z_n \cdot pty_n \\ y_i &\in \{0, 1\}, \quad \forall i = 1, 2, \dots, n-1 \\ z_i &\geq 0, \quad \forall i = 1, 2, \dots, n, \end{aligned}$$

361 where y_i can only take binary values, and their sum equals one,
 362 i.e., only one segment y_i will be 1 (selected). Additionally,
 363 the non-negative variables z_i form a convex combination
 364 of the two endpoints of the selected segment—a point between
 365 the segment's two endpoints. The z_i, y_i relationships ensure
 366 that only two z_i s are non-zero, and their sum equals 1.
 367 Solvers such as Gurobi [37] will automatically transform the
 368 $y = PWL(x, ptx, pty)$ constraints in the backend. These solvers
 369 tackle the mixed-integer piecewise-linear program via branch-
 370 and-bound or similar methods to find the best values for z_i
 371 and y_i , thus the best x and y , to optimize the objective. Our
 372 approach is not derivative-based. For our example, Gurobi
 373 uses a mixed-integer program that employs branch-and-bound
 374 techniques and finds the best (x, y) , i.e., the red star in Fig. 4
 375 at $x = 4.76$, $y = -0.92$, $y_6 = 1$ and $z_6 = 0.7$, $z_7 = 0.3$, while
 376 an analytic solution will be $x = 4.76$, $y = -0.99$.

377 An approximate error exists between the analytic sine and
 378 piecewise-linear sine $d \sin x = \sin(x) - \sin[x]$, namely the line
 379 length between the red star and bowstring. Although the piece
 380 length is quite large (0.9 here) for illustration visualization, the
 381 approximate error is much smaller. Naturally, the approxima-
 382 tion accuracy depends on the piece length and the breakpoints,
 383 i.e., the smaller the piece length, the smaller the error between
 384 the mathematical/analytical $\sin(x)$ and the PWL $\sin[x]$. The
 385 piece length should be selected considering the approximate
 386 accuracy and the computation time. An empirically typical
 387 value is 0.01.

III. THE MIPWLP APPROACH

389 In this section, we first provide a detailed approach to
 390 the Single-trip Obstacle-environment Relaxed Dubins Problem
 391 (SORDP), which focuses on finding the shortest path from the
 392 starting configuration to the target configuration, with only the
 393 angle in the target point as a variable. In fact, there were no
 394 previous methods for SORDP other than ours. Since SORDP
 395 serves as the foundation for the Two-trip Obstacle-environment
 396 Relaxed Dubins Problem (TORDP), understanding SORDP's
 397 formulation is crucial. Once the principles and techniques for
 398 solving SORDP are established, we extend them to address
 399 TORDP, which involves a few additional constraints due to the
 400 need to return to the starting configuration, also considering
 401 the target heading as a variable. This presentation structure
 402 ensures clarity in our MIPWLP approach for TORDP and
 403 preserves the paper at a reasonable length.

A. SORDP: The Shortest Path From the Starting Configuration (x_I, y_I, α_I) to the Target Configuration (x_G, y_G, θ)

405 Fig. 5 illustrates an instance of SORDP. Typically, the
 406 vehicle's turning radius is normalized to one meter, and the
 407 distances between the depot, obstacle, and target are several
 408 kilometers. Under this mild condition, Yang [7], [38] proved
 409 via Pontryagin's minimum principle that the optimal path type
 410 is the 5-letter path type CSDSC, i.e., an optimal path will be
 411 first turning L or R using the minimum radius, going straight,
 412 circumventing the obstacle, going straight and finally turning
 413 L or R.

414 The essential variable in SORDP is the target heading θ ,
 415 which should be optimized (see Alg. 1). Other variables such
 416 as t, p, q are secondary variables resulted from θ . The main
 417 idea for constraining SORDP is that the path CSDSC will
 418 sequentially pass through some critical points, such as tangent
 419 points and the target point, and constraints can be built on
 420 these critical points. After establishing the objectives and the
 421 constraints, MIPWLP solves SORDP through a comprehensive
 422 optimization framework.

423 Let $t, p, q, r, s \geq 0$ be the lengths of the five segments for a
 424 5-letter Dubins path that act as optimizing variables related to
 425 the target heading variable θ . The SORDP model will be the

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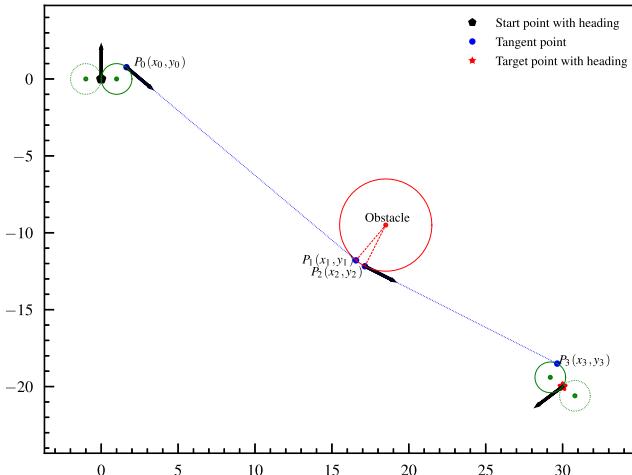


Fig. 5. SORDP's optimal path is CSDSC and the critical points of the 5-letter path, such as the depot, the tangent points and the target point, are shown. The angle at the target point is variable and should be optimized.

following equations with numbers in brackets. The objective of the SORDP is defined as follows:

$$\min_{t,p,q,r,s} \rho_{\min} t + p + \rho q + r + \rho_{\min} s, \quad (5)$$

where ρ_{\min} represents the turning circle radius of the unmanned vehicle, normalized to 1, and ρ represents the vehicle's actual turning radius corresponding to the obstacle. Usually, $\rho \geq \rho_1$ and $\rho \geq \rho_{\min}$ where ρ_1 is the radius of the obstacle. $\rho_1 \geq \rho_{\min} = 1$, as assumed in [7].

The piecewise-linear constraints of sine and cosine functions are denoted as follows:

$$\begin{aligned} \sin[x] &= \text{PWL}(x, ptx, ptys) \\ \cos[x] &= \text{PWL}(x, ptx, ptyc), \end{aligned}$$

where the PWL constraints are presented in Section II-B.

Note that $\sin[\cdot]$ ($\cos[\cdot]$) uses piecewise-linear approximation compared to the common brackets notation. The ptx represents the θ coordinate of the equally spaced interpolation points ($0 \leq ptx < 2\pi$), and the $ptys$ and $ptyc$ are the corresponding real values of $\sin(ptx)$ and $\cos(ptx)$. As mentioned before, the equal space of ptx should be selected considering the approximate accuracy and the computation time. Other radians outside the interval $0 \leq ptx < 2\pi$ can be transformed into this interval, as is well-known and will be seen later. Though MIPWLP only approximates the trigonometrical functions with linear pieces, as a consequence, the linear approximation will also impact the objective indirectly because there are equality relationships between the piecewise-linear variables in the constraints and normal variables (e.g., t , p , q) in the objective.

Assisted by the piecewise-linear constraints, the modulo operation, and computational geometry techniques, SORDP can be established as a tractable mixed-integer piecewise-linear program. The constraints, listed according to the chronological order of the vehicle's visit, are as follows:

Algorithm 1 MIPWLP Algorithm for Problem SORDP

- 1: **Input:**
 - 2: The starting configuration $q_I(x_I, y_I, \alpha_I) \in \mathbb{R}^2 \times \mathbb{S}$ and normalized turning radius ρ_{\min}
 - 3: end position $(x_G, y_G) \in \mathbb{R}^2$
 - 4: obstacle position $(ob_x, ob_y) \in \mathbb{R}^2$ and obstacle radius $\rho_1 \geq \rho_{\min}$
 - 5: Decision variables (continuous) the heading $\theta \in [0, 2\pi]$ at the end position, the $t, p, q, r, s \in [0, +\infty)$ and the intermediate configurations.
 - 6: Decision variables (discrete) $f_1, f_3, f_5 \in \{-1, 1\}$ for the first, third, and fifth segment direction (-1 for turning right)
 - 7: Constants M, ϵ for big-M and small threshold
 - 8: **Output:**
 - 9: Optimal solution $\theta^*, t^*, p^*, q^*, r^*, s^*$ and optimal (single-trip) length
 - 10: **Step 1: Set the objective:**
 - 11: $\min_{t,p,q,r,s,\rho} \rho_{\min} t + p + \rho q + r + \rho_{\min} s$
 - 12: **Step 2: Model the constraints** (modeled according to the chronological order of the vehicle's visit and the necessary property at each critical point and segment, for better logical understanding):
 - 13: **The first C segment constraints:** From the starting configuration q_I via f_1 turning a t radian, the vehicle will arrive at intermediate configuration $P_0(x_0, y_0, \theta_C)$.
 - 14: **The first S segment constraints:** From P_0 via going straight p length, with specific constraints in P_1 , the vehicle will arrive at intermediate configuration $P_1(x_1, y_1, \theta_C)$.
 - 15: **The D segment constraints:** From P_1 via f_3 turning a q radian, with specific constraints on f_3 and P_2 , the vehicle will arrive at intermediate configuration $P_2(x_2, y_2, \theta_D)$.
 - 16: **The second S segment constraints:** From P_2 via going straight r length, the vehicle will arrive at intermediate configuration $P_3(x_1, y_1, \theta_D)$, with specific constraints in P_3 .
 - 17: **The second C segment constraints:** From P_3 via f_5 turning a s radian, the vehicle will arrive at the **end configuration** (x_G, y_G, θ) , with specific constraints on f_5 and s .
 - 18: **Step 3: Optimize the program:** Solve the holistic mixed-integer program using an appropriate solver (e.g., Gurobi) and return the results constructed from the decision variables.
-

1) *The First C Segment:* Fig. 6(a)-(b) depict the initial setting of the scenario and the constraints/properties for the first C segment, the first S segment and the D segment. The intermediate variables $\theta_C, \theta_D \in [0, 2\pi]$ defined as the headings at points P_0, P_2 are presented in Fig. 5, respectively. Let the integer variables f_1, f_3, f_5 be the turning direction of the first, third, and fifth segments (the second and the fourth are straight already according to "CSDSC"). Thus, the following equation holds:

$$\theta_C = \alpha_I + f_1 \cdot t \pmod{2\pi}. \quad (6)$$

However, the mod operator cannot be handled directly to fit into solvers, causing periodicity and integers in the feasible

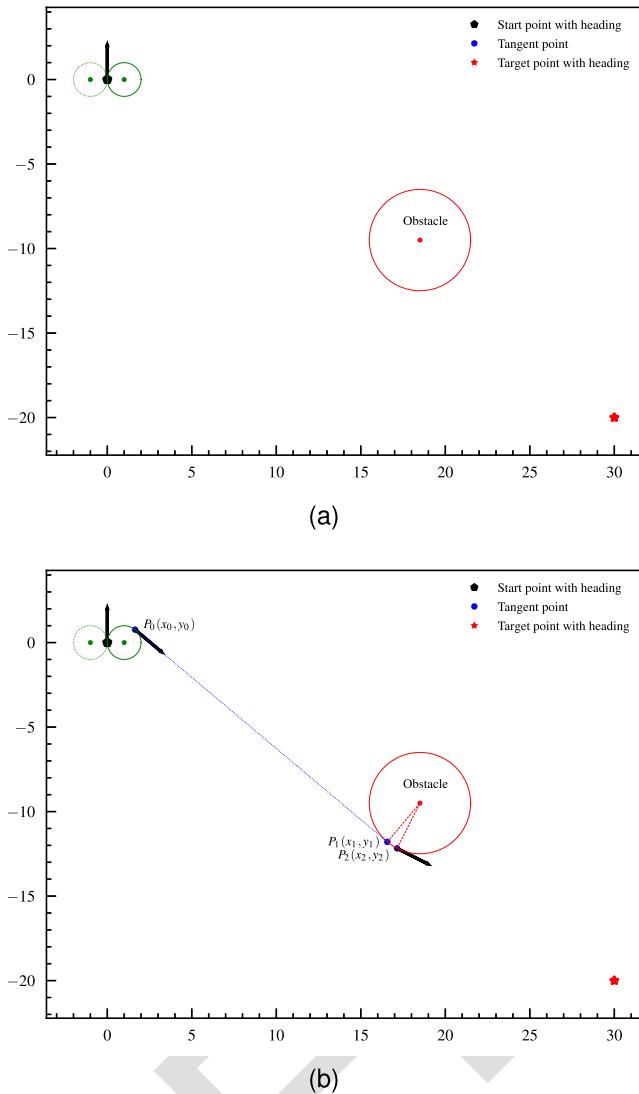


Fig. 6. The setting and the constraints for the first C, the first S and the D segments. The first C segment can turn left or right; the S segment will be tangent to both the turning circle and the obstacle-circumventing circle; the direction of D should be consistent with vector P_0P_1 , clockwise or counterclockwise along the obstacle boundary.

space. Therefore, a proper transformation is required for the solvers. Hence, equation 6 is transformed into: $\theta_C = (\alpha_I + f_1 \cdot t) - \left\lfloor \frac{\alpha_I + f_1 \cdot t}{2\pi} \right\rfloor \cdot 2\pi$. By introducing an auxiliary integer variable $k = \left\lfloor \frac{\alpha_I + f_1 \cdot t}{2\pi} \right\rfloor$,

$$\frac{\alpha_I + f_1 \cdot t}{2\pi} - 1 < k \leq \frac{\alpha_I + f_1 \cdot t}{2\pi}.$$

Hence, $\alpha_I + f_1 \cdot t - 2\pi < 2\pi k \leq \alpha_I + f_1 \cdot t$ and $\theta_C = \alpha + f_1 \cdot t - 2\pi k$.

According to Fig. 6 and the complex number derivation presented in Section II, the following relational constraints hold for the tangent point P_0 :

$$\begin{aligned} x_0 &= x_I + f_0 \cdot \rho_{min} \cdot \sin[\theta_C] - f_0 \cdot \rho_{min} \cdot \sin[\alpha] \\ y_0 &= y_I - f_0 \cdot \rho_{min} \cdot \cos[\theta_C] + f_0 \cdot \rho_{min} \cdot \cos[\alpha] \end{aligned} \quad (7)$$

Note that $[\cdot]$ denotes PWL constraints, which are essentially mixed-integer programs underneath, and solvers can use branch-and-bound techniques to solve them automatically.

2) *The First S Segment*: As shown in Fig. 6(b), a straight move from P_0 generates P_1 , hence:

$$\begin{aligned} x_1 &= x_0 + p \cos[\theta_C] \\ y_1 &= y_0 + p \sin[\theta_C]. \end{aligned} \quad (8)$$

P_1 also has two properties: ① its distance ρ to the obstacle center is no less than the obstacle radius ρ_1 . Therefore:

$$\begin{aligned} (b_x - x_1)^2 + (b_y - y_1)^2 &= \rho^2 \\ \rho &\geq \rho_1, \end{aligned} \quad (9)$$

② the shooting line from P_0 to P_1 is perpendicular to the line connecting the point P_1 and the obstacle center ob . Thus, their dot product is zero:

$$\cos[\theta_C] \cdot (ob_x - x_1) + \sin[\theta_C] \cdot (ob_y - y_1) = 0. \quad (10)$$

3) *The D Segment*: The directional tangent line P_0P_1 determines whether to follow the obstacle boundary clockwise or counterclockwise (see Fig. 6(b)), so the circumventing direction of the D segment is determined by the lateral position of the circle center with respect to the line P_0P_1 , where the cross-product can be utilized to constrain the lateral position. Let xp_1 be the cross-product of P_0P_1 respect to vector $P_1(x_1, y_1) - (ob_x, ob_y)$:

$$xp_1 = \cos[\theta_C] \cdot (ob_y - y_1) - \sin[\theta_C] \cdot (ob_x - x_1), \quad (11)$$

where (ob_x, ob_y) is the center of the obstacle.

The sign of xp_1 must equal to the binary variable f_3 , so it must hold:

$$f_3 = \text{sgn}(xp_1). \quad (12)$$

To transform the non-trivial sign function $g = \text{sgn}(\omega)$ to be tractable by solvers, one can use the big-M strategy:

$$\begin{aligned} -M(1 - z_1) &\leq g + 1 \leq M(1 - z_1), \quad \omega \leq -\epsilon + M(1 - z_1) \\ -M(1 - z_2) &\leq g \leq M(1 - z_2), \\ -M(1 - z_2) - \epsilon &\leq \omega \leq \epsilon + M(1 - z_2) \\ -M(1 - z_3) &\leq g - 1 \leq M(1 - z_3), \quad \omega \geq \epsilon - M(1 - z_3) \\ z_1 + z_2 + z_3 &= 1 \end{aligned}$$

where z_i are auxiliary binary variables, and M is a sufficiently large (but as small as possible) constant. In the experiments, $M = 1000$. ϵ is a small equality constant set to $1e-5$.

The constraint on θ_D is given by:

$$\theta_D = \theta_C + f_3 \cdot q \pmod{2\pi}, \quad (13)$$

which can be similarly transformed as the previous θ_C .

Circumventing (not necessarily touching) the obstacle boundary with q radian from P_1 (recall the complex number derivations), we reach P_2 , and thus have:

$$\begin{aligned} x_2 &= x_1 + f_2 \cdot \rho \cdot \sin[\theta_D] - f_2 \cdot \rho \cdot \sin[\theta_C] \\ y_2 &= y_1 + f_2 \cdot \rho \cdot \cos[\theta_D] + f_2 \cdot \rho \cdot \cos[\theta_C]. \end{aligned} \quad (14)$$

583 4) *The Second S Segment:* The second S segment of the
584 trip gives:

$$\begin{aligned} 585 \quad x_3 &= x_2 + r \cdot \cos[\theta_D] \\ 586 \quad y_3 &= y_2 + r \cdot \sin[\theta_D] \end{aligned} \quad (15)$$

587 5) *The Second C Segment:* Let decision variable $finalLR =$
588 1 represent the left unit circle of the target configuration
589 (x_G, y_G, θ) and $finalLR = -1$ represent the right unit circle.
590 The circle center $(final_x, final_y)$ of this unit circle is parameterized as follows:
591

$$\begin{aligned} 592 \quad (final_x, final_y) &= (x_G, y_G) + \rho_{min} \cdot finalLR \cdot \\ 593 \quad &\quad (-\sin[\theta], \cos[\theta]) \end{aligned} \quad (16)$$

594 Additional constraints mainly comprise clockwise or counter-clockwise **direction consistencies**. Hence: (1) P_3 lies on the
595 $finalLR$ circle passed the final point $(final_x, final_y)$ with a radius of $\rho_{min} = 1$. (2) The vector from P_2 to P_3 is perpendicular to
596 the vector P_3P_f . (3) The cross-product of P_2P_3 to P_3P_f equals $FinalLR$ and f_5 . (4) The fifth constraint is that the heading
597 value θ at the target point equals θ_D plus some circumventing
598 radians. These consistencies can be translated into:
599

$$\begin{aligned} 602 \quad (x_3 - final_x)^2 + (y_3 - final_y)^2 &= \rho_{min}^2 \\ 603 \quad (x_3 - final_x) \cdot \cos[\theta_D] + (y_3 - final_y) \cdot \sin[\theta_D] &= 0 \\ 604 \quad xp_2 &= \cos[\theta_D] \cdot (final_y - y_3) - \sin[\theta_D] \cdot (final_x - x_3) \\ 605 \quad f_5 &= \text{sgn}(xp_2) = FinalLR \\ 606 \quad \theta &= \theta_D + f_5 \cdot s(\text{mod } 2\pi). \end{aligned} \quad (17)$$

607 B. TORDP: Shortest Path From (x_I, y_I, α_I) to (x_G, y_G, θ) to 608 (x_I, y_I, α_I)

609 The equations (5)–(17) in the last subsection institute the
610 objective and constraints for SORDP, which is from configuration
611 (x_I, y_I, α_I) to configuration (x_G, y_G, θ) . The two trips of the TORDP can be treated as the trip from the starting configuration (x_I, y_I, α_I) to the target configuration (x_G, y_G, θ) and the trip from the target configuration (x_G, y_G, θ) to the starting configuration (x_I, y_I, α_I) , linked with the common θ variable at the target point.

617 Since the second trip of TORDP is planning a path from (x', y', θ) to (x, y, α) with CSDSC path type, we introduce an additional set of variables t_2, p_2, q_2, r_2, s_2 and extra constraints to the optimization objective. For TORDP, the objective and constraints of TORDP are:

$$\begin{aligned} 622 \quad \min \rho_{min} t &+ p + \rho q + r + \rho_{min} s \\ 623 \quad &+ rho_{min} t_2 + p_2 + \rho q_2 + r_2 + \rho_{min} s_2, \\ 624 \quad \text{s. t. constraints (as Eqs. (6) – (16)) for the path from} \\ 625 \quad \text{configuration } (x_I, y_I, \alpha_I) \text{ to configuration } (x_G, y_G, \theta); \\ 626 \quad \text{constraints with } t_2, p_2, q_2, r_2, s_2 \text{ for the path from} \\ 627 \quad \text{configuration } (x_G, y_G, \theta) \text{ to configuration } (x_I, y_I, \alpha_I), \\ 628 \quad \end{aligned} \quad (18)$$

629 where the decision variables are $t, p, q, r, s, t_2, p_2, q_2, r_2, s_2$,
630 with t_2 the immediate path segment after visiting the target
631 point. Similar to SORDP, all constraints of TORDP can be

TABLE II
INFORMATION ON THE DIFFERENT COMBINATIONS AND THE FOUR EXPERIMENTAL SETS FROM EIGHT COMBINATIONS

Trip	Obstacle	Heading	Available methods
one	without	fixed	Classic Dubins methods
one	without	variable	Classic RDP methods
one	with	fixed	Tangent line method; ours
one	with	variable	No previous methods; ours
two	without	fixed	Classic Dubins methods
two	without	variable	Chen's algorithm; ours
two	with	fixed	Tangent line method
two	with	variable	A modified heuristic; ours

635 established by following a chronological order of CSDSC-CSDSC path type and constraining the critical points—the two CSDSC trips are linked with the common θ variable and the two trips are holistically solved by GUROBI. The additional constraints of TORDP than SORDP are the constraints of the path from $q_G(x_G, y_G, \theta)$ to $q_I(x_I, y_I, \alpha_I)$, which highly resembles the Eqs.(6)–(17) because of the one-to-one correspondence between $x_I \leftrightarrow x_G, y_I \leftrightarrow y_G, \alpha_I \leftrightarrow \theta$.

636 *Remark 2:* Our mixed-integer piecewise-linear programming (MIP) approach differs fundamentally from traditional optimization techniques, such as the tangent-line method, Chen's method, and the angle-discretization heuristic: These traditional approaches rely on analytical formulations, which assume continuous and smooth solution spaces. While they work well for simpler Dubins path problems, they struggle to handle the complex constraints inherent in the TORDP with the three difficult features mentioned before.

IV. EXPERIMENTAL VALIDATIONS

652 The following experiments validate that MIPWLP is accurate, effective, and unified capable. Some combinations have been solved well among the eight combinations of different trip types, obstacles, and target headings. Therefore, we conducted and compared four sets of numerical experiments to evaluate the high accuracy and unification performance of the proposed algorithm (presented in bold in Table II): (1) Single-trip experiments in obstacle-constrained environments with a given target heading. (2) Single-trip experiments in obstacle-constrained environments with a variable target heading. (3) Two-trip experiments in obstacle-free environments with a given target heading. (4) Two-trip experiments in obstacle-constrained environments with a variable target heading.

653 All test cases have a starting configuration of $q_I(0, 0, \pi/2)$, as in [7]. We normalized the x, y coordinates to have a unit of one. The approximation piece length is 0.01, the first breakpoint in ptx is 0.00, and the last is 6.29 (nearly 2π). The cutoff time is set to 1 hour. However, many cases are in the order of seconds, and only the fourth group needs one hour. Although our approach is a path planning algorithm for a vehicle with a bounded curvature, we validated the path planning and path following performance using the high-fidelity UAV toolbox from MATLAB. After experimenting with MIPWLP, the final part of this section analyzes the approximation errors,

TABLE III
INFORMATION AND COMPARISON RESULTS BETWEEN OUR ALGORITHM AND THE TANGENT LINE METHOD [7]

Information			Tangent line method [7]	Our algorithm		
Case	target point	obstacle radius	length	our length	absolute error	ρ
a	(18.5, -9.5)	3	38.233	38.232	0.001	3.000
b	(19.5, -8.5)	3	38.003	38.004	0.001	3.000
c	(20.5, -7.5)	3	37.966	37.967	0.001	3.956
d	(18.5, -9.5)	2	38.063	38.064	0.001	2.218
e	(18.5, -9.5)	3	38.233	38.232	0.001	3.000
f	(18.5, -9.5)	4	38.525	38.524	0.001	4.045

679 presenting quality-guaranteed proofs. The experimental setup
 680 involves an Intel Core i7 and a GPU using CUDA with
 681 32 GB RAM. The operating system is Win10, and the solver
 682 is Gurobi 10.0 [37] with gurobipy interfaces.

683 A. Single-Trip Experiments in Obstacle Environments With 684 Given Target Heading

685 **Setup:** The first experiment aims to reveal whether MIP-
 686 WLP can obtain nearly optimal results compared to the
 687 analytical optimal results. In this scenario, the x, y coordinates
 688 and headings of the starting and target points are given, and the
 689 proposed algorithm is compared with the tangent line method
 690 [7]. The tangent line algorithm determines the positions of
 691 the tangent points and plans the collision-free path from
 692 the starting configuration to the ending configuration via the
 693 tangent points. The tangent line algorithm has proved to be
 694 optimal. Here, the target configuration adopts [7] and is fixed
 695 as $(30, -20, \text{atan}2(-4, -3))$.

696 This subsection conducted six experiments (among which
 697 the first and the fifth were identical—we added this duplicate
 698 to view the performance trends better). The obstacle center in
 699 the first three experiments is different (see Table III), while the
 700 radius of the obstacle in the last three experiments is different.

701 **Results:** Table III compares the results between the pro-
 702 posed algorithm and the tangent line method [7]. The
 703 comparison consists of different cases, i.e., the total length
 704 of the ground truth, our algorithm's length, the absolute error,
 705 and the circumventing radius that MIPWLP calculated. The
 706 shorter the total length, the better the algorithm's performance.
**The results highlight that the total length of the proposed
 708 algorithm is nearly the same as the optimal path length
 709 [7], with absolute errors smaller than 0.001.**

710 Fig. 7 illustrates the impact of the obstacle centers and the
 711 obstacle radii. The total length decreases as the obstacle is
 712 farther away from the main path, and the total length increases
 713 as the obstacle radius increases in the second-row settings. The
 714 subfigure (c) reveals that our approach has a natural unification
 715 of the 5-letter case and the “CSOSC” case in a single trip, i.e.,
 716 the length of q in the t, p, q, r, s vanishes to zero. In particular,
 717 in some cases where the starting and target points are far
 718 from obstacles, the length of some segments may be zero,
 719 so there will be one straight line rather than two passing near
 720 the obstacle.

721 Overall, in this experimental set, the proposed algo-
 722 rithm reaches the optimal, proving an effective and accurate

724 solution to the path planning problem with given headings. It
 725 unifiedly found the optimal 5-letter path and the “CSOSC” path
 726 naturally.

727 B. Single-Trip Experiments in Obstacle Environments With 728 Variable Target Heading

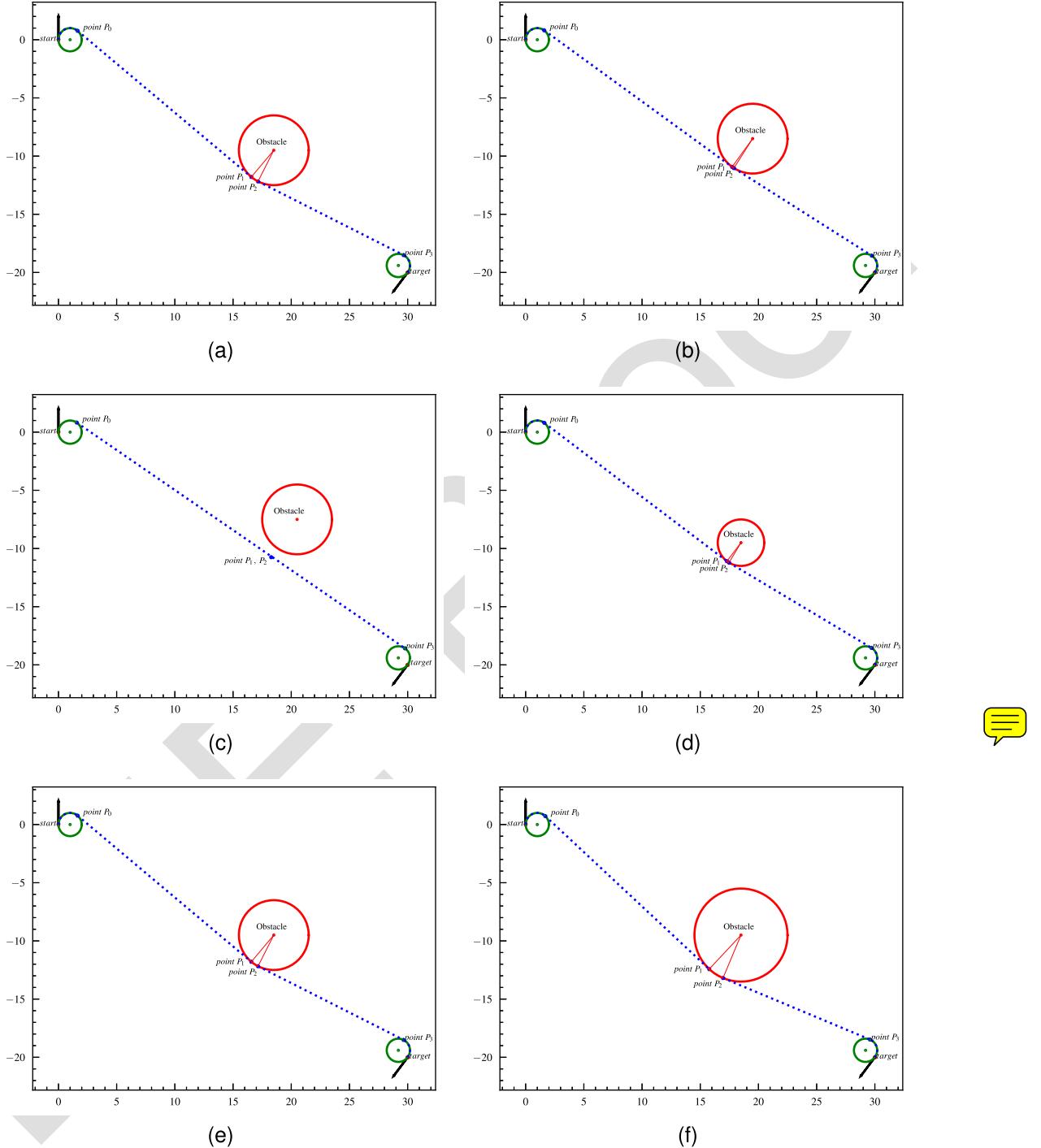
729 The first set of experiments verifies that the proposed algo-
 730 rithm provides the shortest path between the given start and
 731 target. This subsection verifies our algorithm's performance
 732 with a variable target heading in an obstacle environment.

733 **Setup:** The second set of experiments plans the shortest trip
 734 from a given starting configuration to the target point with a
 735 variable heading in an environment with obstacles. The depot's
 736 heading is given, but the target point heading is not fixed.
 737 Like Section IV-A, we conducted six experiments using the
 738 identical coordinates as in the experimental Set 1.

739 **Results:** We self-compare the results to those of Set 1 to
 740 preserve a reasonable article length. Table III highlights that
 741 the planned paths under variable target headings are always
 742 less than or equal to those with fixed headings since the
 743 unfixed target heading corresponds to a larger search space.
 744 For the “CSOSC” case, when the shortest path does not touch
 745 the obstacle, the length of q will be almost zero, and the
 746 circumventing radius will be larger than the obstacle radius,
 747 i.e., cases b and c in Table III. In addition, the planned paths
 748 are plotted in Fig. 8 to analyze and comprehend the results.
 749 The plotted paths infer that the algorithm chooses a proper θ
 750 that prevents the path from approaching the obstacle. Hence,
 751 the larger the obstacle radius, the longer the optimal path.

752 C. Two-Trip Experiments in Obstacle-Free Environments 753 With Variable Target Headings

754 **Setup:** The experimental Set 3 verifies our algorithm's
 755 performance in calculating the two-trip Dubins path within
 756 obstacle-free environments while eliminating the obstacle
 757 constraints in MIPWLP. TORDP passes through three predefined
 758 points: the depot, the target point, and the depot (as the final
 759 destination). In [30], the authors have proved that the shortest
 760 three-point Dubins path in an obstacle-free environment is
 761 obtained by finding the **provable** optimal heading of the target
 762 point via a roots-finding algorithm. Therefore, we adopted
 763 their strategy using the SymPy Python package and compared
 764 it against the proposed algorithm in four cases. In all cases, the
 765 starting and ending points are the same (i.e., $(0, 0, \pi/2)$), while



766 Fig. 7. The (near-)optimal solution of MIPWLP on the experimental set 1. It is a visual and detailed presentation of Table III.

767

TABLE IV

768 MIPWLP RESULTS ON THE EXPERIMENTAL SET 2, PRESENTING SHORTER PATHS THAN THE FIXED TARGET HEADING CASES. CASES B AND C HAVE A
769 “CSOSC” OPTIMAL PATH, WHERE Q IS ALMOST ZERO, AND THE CIRCUMVENTING RADIUS IS LARGER THAN THE OBSTACLE RADIUS

770
771

Case	t	p	q	r	s	optimal length (length in last table)	optimal target heading θ	ρ
a	2.271	19.506	0.155	15.131	0.147	37.519 (38.232)	5.590	3.003
b	2.203	19.947	0.000	15.264	0.001	37.417 (38.004)	5.650	3.078
c	2.203	20.164	0.000	15.050	0.000	37.417 (37.967)	5.651	4.471
d	2.220	19.685	0.037	15.436	0.008	37.423 (38.064)	5.680	2.000
e	2.271	19.506	0.155	15.131	0.147	37.519 (38.232)	5.590	3.003
f	2.322	19.274	0.270	14.887	0.168	37.731 (38.524)	5.970	4.000

770 the target points are $(30, -20), (30, 20), (-30, 20), (-30, -20)$,
771 respectively.

772 **Results:** Fig. 9 and Table V qualitatively compare the
773 proposed algorithm and [30]. The four cases are plotted within

774
775

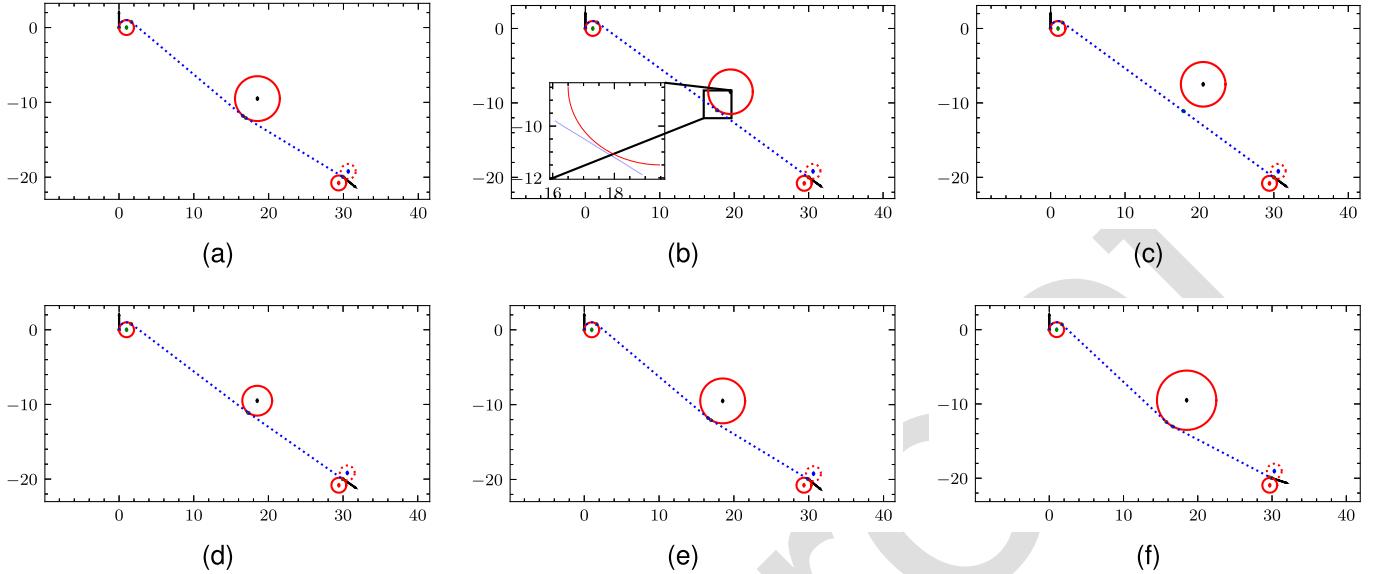


Fig. 8. Visual presentation of Table III, where the selected *finalLR* circles are depicted in solid line while the other circle is in dashed line. All solutions selected the right circle to pass the target point. (b) is zoomed in to view the details. The optimal heading at the target point is different from that of experimental Set 1.

TABLE V

COMPARISON OF OUR ALGORITHM AND CHEN'S ALGORITHM [30]. e_1 AND e_2 IS THE LENGTH AND θ ABSOLUTE ERROR, RESPECTIVELY

Case	Chen's algorithm						Our algorithm									
	t	p	q	q'	r	s	t	p	q	q'	r	s	length	e_1	e_2	T.(s)
BR	2.170	34.230	1.570	1.570	34.230	0.970	2.175	34.228	1.572	1.570	34.228	0.967	74.740	0.001	0.003	7.00
TR	0.970	34.230	1.570	1.570	34.230	2.170	0.967	34.228	1.566	1.575	34.228	2.174	74.740	0.001	0.000	2.98
TL	0.970	34.230	1.570	1.570	34.230	2.170	0.968	34.230	1.550	1.591	34.228	2.174	74.740	0.001	0.011	3.00
BL	2.170	34.230	1.570	1.570	34.230	0.970	2.175	34.227	1.563	1.578	34.228	0.967	74.739	0.000	0.006	3.00

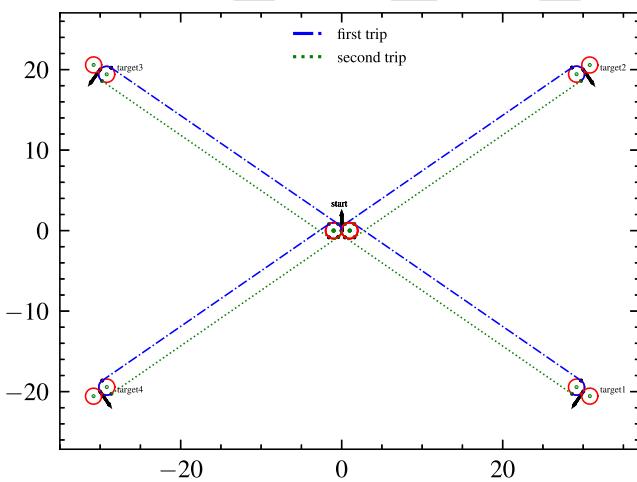


Fig. 9. Results of experimental Set 3 with four cases. The solutions are optimal and identical to Chen's ground-truth results [30]. The first and second trip in each TORDP is represented with '-' and '.' line styles, respectively.

one figure to save article space. Since the physical environment is assumed to be open, MIPWLP has been modified accordingly to eliminate the obstacle constraints. Fig. 9 illustrates the four cases, where the circular segment just before the target point is denoted as q and just after as q' . The planned paths reveal that the bottom left (BL) and bottom right (BR) cases

are symmetric and have obtained symmetric optimal results. The top left (TL) and top right (TR) cases are similar.

From the detailed data (Table V), it is evident that the proposed algorithm provides very close q and q' , which aligns with the findings from the optimality-proved algorithm [30]. In addition, the proposed algorithm has **nearly the same optimal θ and optimal path length as Chen's algorithm** [30], which has been theoretically proved optimal. However, the computational burden of our MIPWLP is less than 7 seconds in each case, which is quite quick. This experimental set demonstrated our algorithm's accuracy and efficiency in non-obstacle scenarios.

D. Two-Trip Experiments in Obstacle Environments With Variable Target Headings

Setup: Unlike the experimental Set 3, Set 4 verifies the performance of the proposed algorithm in obstacle environments. It resembles Set 3, but obstacles are added with radius=3 centered at (18.5,-9.5), (18.5,9.5), (-18.5,9.5), (-18.5,-9.5). This experimental set aims to reveal the performance of the developed scheme in this scenario and compare it with Set 3. Moreover, the results are compared with a prior heuristic that can be modified and applied to the free target heading optimization to prove the advantages of MIPWLP.

Results: Fig. 10 depicts the paths planned by our algorithm, where the first trip is in blue and the second trip is in dotted

817

TABLE VI
MIPWLP SOLUTIONS OF TORDPs FOR THE FOUR CASES

818

Case	t	p	q	r	s	ρ	t_2	p_2	q_2	r_2	s_2	ρ	length	θ
BR	2.270	19.510	0.221	14.043	1.605	3.000	1.645	14.393	0.034	19.761	0.950	3.454	74.961	4.200
TR	0.973	19.813	0.120	14.042	1.602	3.000	1.680	14.463	0.081	19.506	2.270	3.000	74.952	2.080
TL	0.972	19.814	0.135	14.009	1.798	3.020	1.483	14.454	0.098	19.502	2.271	3.020	75.006	0.880
BL	2.270	19.510	0.221	14.043	1.605	3.000	1.645	14.393	0.034	19.761	0.950	3.454	74.961	5.225

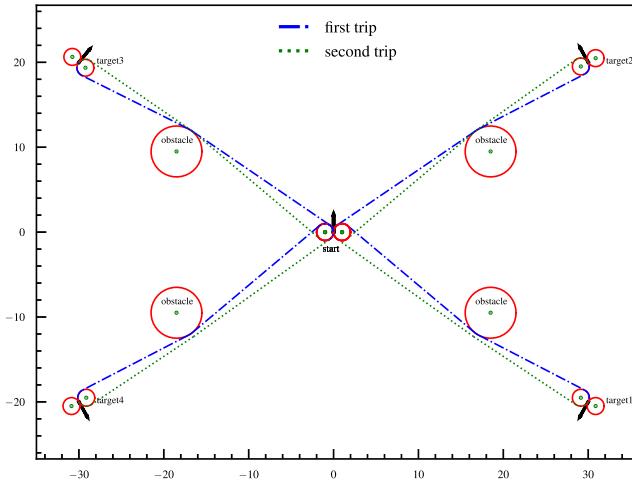


Fig. 10. Results of the experimental Set 4, with four cases, each with a target and obstacle. The symmetric cases have symmetrically identical solutions.

green. The two-trip length is smaller than twice the single-trip length. One may find the symmetries between the down-left and down-right cases, between the up-left and up-right cases—the cases will have the same ground-truth length, and MIPWLP indeed finds the same length and path.

Table VI presents the detailed length of each segment, where t_2 is the segment immediately after the target point and p_2 follows. The presence of obstacles increases the path length, leading to a slightly larger total length than the two-trip without obstacles—MIPWLP has provided quite good lengths.

Although no previous algorithms tackle TORDP, a heuristic (inspired by [39]) is used to validate the advancement of MIPWLP. The heuristic uniformly discretizes $[0, 2\pi]$ to L ($6 \leq L < 16$) samplings and then transforms a TORDP to two one-trip fixed target heading problems. The two problems can be easily and optimally solved using [7]. We implemented the heuristic, calculated the shortest two-trip length S_L among the L samples, and recorded the corresponding sampled angle as A_L . The BR and TR cases in Table VI were experimented with this heuristic.

In each case, the S_L is slightly longer (0.5 meters) than MIPWLP's length, and the length difference decreases as the discretization level increases. Fig. 11 illustrates the angle difference between the heuristic and MIPWLP, revealing that the angle difference is between 5 and 20 degrees. This finding suggests that the optimized heading of the heuristic and the MIPWLP to visit the target differs greatly and that the difference has no obvious trend regarding the number of discretizations. This is reasonable

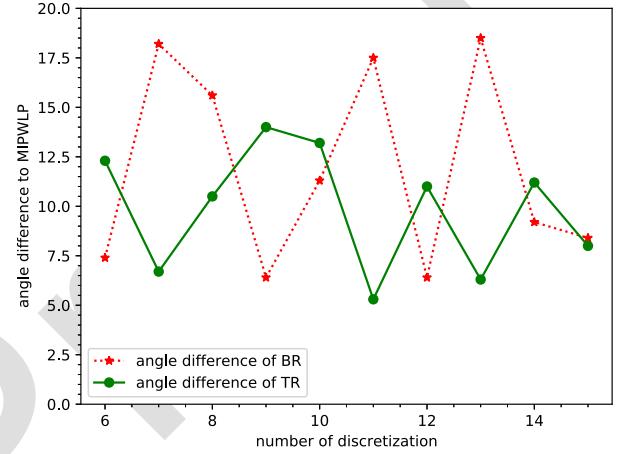


Fig. 11. The angle difference between the heuristic and MIPWLP, where the latter angles result in shorter path lengths.

because the performance of heuristics is hard to predict and guarantee.

E. Path Following Simulation Based on UAV Toolbox

This section validates the SORDP path-following performance of the paths planned by MIPWLP using a fixed-wing UAV model. The UAV Toolbox in MATLAB offers algorithms, simulation environments, and reference applications for designing, simulating, testing, and deploying unmanned aerial vehicle and drone applications. The UAV Guidance Model block includes fixed-wing UAV aerodynamics and an autopilot, approximating the kinematic behavior of a closed-loop system. The kinematics of a fixed-wing UAV at a constant height is that of a typical Dubins vehicle, offering a high-fidelity simulation. Since the UAV model is 3D, it is restricted to flying at a fixed height, turning it into a 2D Dubins vehicle to suit the MIPWLP algorithm. Samples with (x, y, θ) coordinates are generated along the MIPWLP planned (near-)shortest path and input to the UAV waypoint following modules for the path following.

The waypoint-follower configuration involves the Waypoint Follower block, which computes a desired heading for the UAV based on waypoints, current pose, and look-ahead distance inputs. The heading control block acts as a proportional controller to regulate the UAV's heading angle, with sliders available for tuning look-ahead distance and heading control values.

The simulated scenario represents a real scenario [7]: The initial and final configuration points are $(0, 1000)$ and $(3000, -1000)$ in the meter unit. We scale the coordinates to 1/100 in the simulation and translate the initial configuration to $(0, 0)$, so the final configuration is $(30, -20)$. The start heading is the

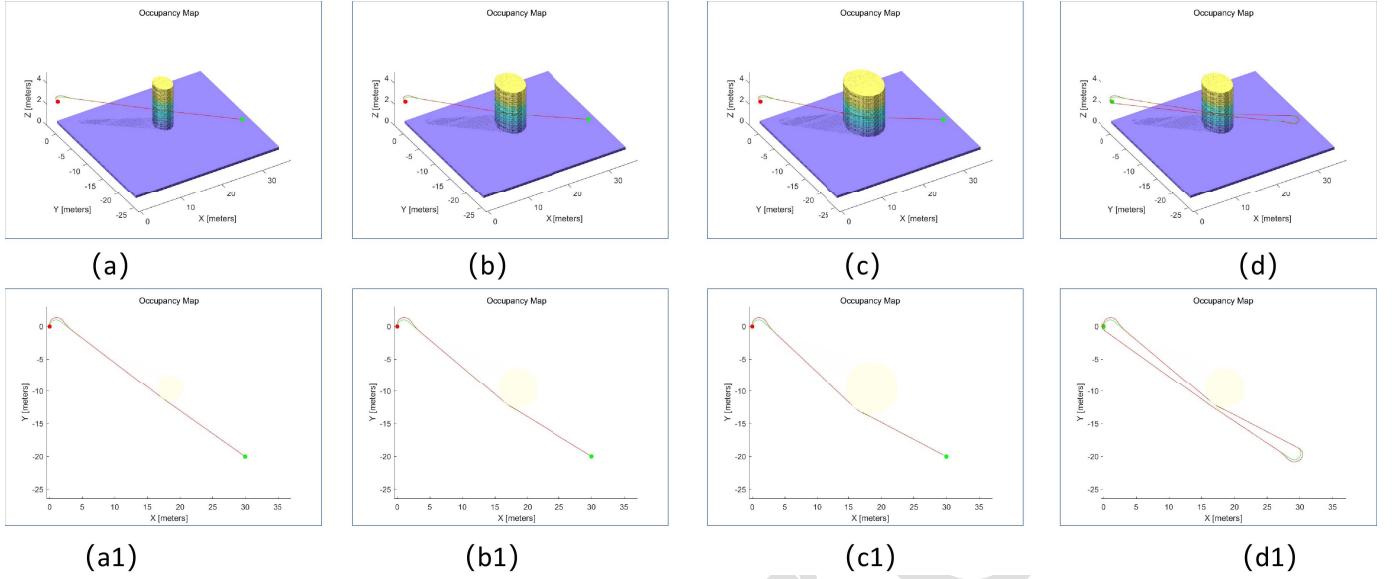


Fig. 12. Results of four simulation experiments. (a)-(d) are side views, and (a1)-(d1) are vertical views.

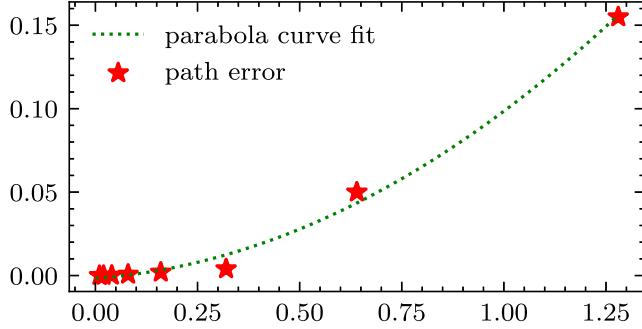


Fig. 13. Path length error of different piece lengths and a best-fit quadratic curve. The fit quadratic is about $0.07h^2$, showing better results than the worst-case quality guarantee analysis

TABLE VII

KEY PARAMETERS OF THE FIXED-WING UAVS

Key Parameter	Value
1 MaxRollAngle	0.10168 rad
2 AirSpeed	1m/s
3 FlightPathAngleLimit	[-0.0001 0.0001] rad
4 g	9.8 N/kg

same as the real scenario. To adapt fixed-wing UAVs to 2D Dubins vehicles, we fix the UAV's height (100 meters in the real scenario) and limit the FlightPathAngle to a range close to zero. Detailed UAV parameters are outlined in Table VII to normalize the vehicle's turning radius (200 meters) as one meter and velocity as 1 unit (30 m/s in the real scenario). The obstacle in the real scenario has similarly been normalized to (18.5, -9.5) with a radius of three. The obstacle in a coral reef reconnaissance application can be islands or dangers and it is treated as a cylinder, as depicted in Fig. 12.

Four validation cases have been conducted in this subsection. Fig. 12(a)-(c) correspond to Fig. 8(d)-(f), while the fourth case corresponds to Fig. 10 BR. Fig. 8 and Fig. 10 depict the shortest single-trip and two-trip from the given starting

configuration to the target point under a variable heading in an environment containing obstacles.

Fig. 12 presents the paths planned by MIPWLP (green paths) and the actual path followed by the UAV (red paths), highlighting that the two paths exhibit high coincidence. Indeed, there is only a slight deviation where the straight segment meets the arc due to the control change of the Dubins path at the junction. Future work could incorporate Clothoid curves to address curvature continuity and control smoothness at the junctions. These four experiments demonstrate that the fixed-wing UAV effectively follows the shortest path planned by MIPWLP.

F. Quality Guarantees

Next, the error (tolerance) between the optimal and MIPWLP solutions is theoretically analyzed in relatively simple cases. Given that comprehensively analyzing MIPWLP is quite complicated, thus it is left for future studies. Since the error estimation on the simple case can enhance our understanding of the proposed approach, this section demonstrates that the optimal length and the piecewise-linear approximated length have an absolute error less than $h^2/(2\sqrt{2})$, where h is the piece length. If $h = 0.01$, the length error is $1e-4$. Suppose the optimal length is 74. Then, the calculated approximate length may be 74.0001.

We consider a path planning case from (x, y, α) to (x', y', β) under the path type RS without obstacles and suppose the setting is feasible for RS. The floating-point numerical issues are not considered, as we suppose the computers are precise enough. Regarding the analytical/mathematical computation, this optimization problem is:

$$\begin{aligned} & \min_{t,p} t + p \\ & \text{s.t. } x - \sin(\alpha - t) + \sin(\alpha) + p \cos(\alpha - t) = x' \\ & \quad y + \cos(\alpha - t) - \cos(\alpha) + p \sin(\alpha - t) = y' \\ & \quad \beta = \alpha - t (\bmod 2\pi). \end{aligned} \tag{19}$$

Notably, variables t and p can be analytically calculated because $t \in [0, 2\pi]$ can be determined first, and then the first two constraints have only one unknown p . Thus:

$$\begin{aligned} p \cos(\alpha - t) &= x' - x + \sin(\beta) - \sin(\alpha) \\ p \sin(\alpha - t) &= y' - y - \cos(\beta) + \cos(\alpha), \end{aligned} \quad (20)$$

By squaring and summing the two equations, we can solve p , which is a function of the known $\sin(\beta), \sin(\alpha), \cos(\beta), \cos(\alpha), x, y, x', y'$.

For the PWL computation: Let \hat{t}, \hat{p} be the PWL solution. \hat{t} is computed similarly to the analytic t because the mod operator can be transformed via mixed-integer constraints. The PWL solution \hat{p} aims to find the solutions for these piecewise-linear constraints:

$$\begin{aligned} \hat{p} \cos[\alpha - t] &= x' - x + \sin[\beta] - \sin[\alpha] \\ \hat{p} \sin[\alpha - t] &= y' - y - \cos[\beta] + \cos[\alpha], \end{aligned} \quad (21)$$

where $[.]$ denotes piecewise approximation. Therefore, $\hat{p} = f(\sin \alpha, \cos \alpha, \sin \beta, \cos \beta, x, y, x', y')$ where f is a known function with piecewise-linear variables.

To derive the absolute error $|p - \hat{p}|$ requires the following lemma. From the Lagrangian interpolation and tolerance theories [40], if the domain of the function is within $[a, b]$, and there exists an interpolation polynomial $L_n(x) \approx f(x)$ on $[a, b]$, then the difference $R_n(x) = f(x) - L_n(x)$ is the interpolation remainder. In this paper, $n = 1$ in the Lagrangian interpolation polynomials L_n means linear interpolation.

Lemma 1: Let $f(x) \in C^{n+1}[a, b]$ (this denotes that $f(x)$ is continuously differentiable up to order $n+1$ on $[a, b]$), and let the interpolation points be $a \leq x_0 < x_1 < \dots < x_n \leq b$. Then, for the interpolation polynomial $L_n(x)$, for any $x \in [x_i, x_{i+1}]$, we have

$$|R_n(x)| = |f(x) - L_n(x)| = \left| \frac{f^{(n+1)}(\xi)}{(n+1)!} (x - x_i)(x - x_{i+1}) \right|,$$

where $x_i < \xi < x_{i+1}$.

This lemma allows deriving the absolute error $|p - \hat{p}|$. Let the difference between the curvy and piecewise-linear sine be $d \sin x = |\sin(x) - \sin[x]|$, as depicted in Fig. 4.

Then, according to Lemma 1,

$$\begin{aligned} d \sin x &\approx |\sin x - L_1(x)| \leq \frac{|f''(\xi_1)|}{2!} |(x - x_i)(x - x_{i+1})| \\ &\leq \frac{1}{2} \cdot |f''(\xi_1)| \frac{h^2}{4} = \frac{1}{2} \cdot \frac{h^2}{4}, \\ d \cos x &\approx |\cos x - p_1(x)| \leq \frac{1}{2} |g''(\xi_1)| \frac{h^2}{4} = \frac{1}{2} \cdot \frac{h^2}{4}, \end{aligned} \quad (22)$$

where h is the piece length, i.e., the x distance between two consecutive breakpoints. The second derivative functions of the sine and cosine functions are smaller than 1, and the rightmost \leq applies the arithmetic-mean–geometric-mean inequality.

We denote the symbol S as the square of \hat{p} , $\Delta x = x - x'$, $\Delta y = y - y'$, and symbols $K_1 = \Delta x + \sin \beta - \sin \alpha$, $K_2 = \Delta y - \cos \beta + \cos \alpha$, then, according to Eq.(21):

$$S = \hat{p}^2 = f^2(\sin \alpha, \cos \alpha, \sin \beta, \cos \beta, x, y, x', y')$$

$$= (\Delta x + \sin \beta - \sin \alpha)^2 + (\Delta y - \cos \beta + \cos \alpha)^2. \quad (23)$$

The tolerance of S is (similar as a difference of S):

$$\begin{aligned} dS &= \frac{\partial S}{\partial \sin \alpha} \cdot d \sin \alpha + \frac{\partial S}{\partial \cos \alpha} d \cos \alpha + \frac{\partial S}{\partial \sin \beta} d \sin \beta \\ &\quad + \frac{\partial S}{\partial \cos \beta} d \cos \beta = -2(\Delta x + \sin \beta - \sin \alpha)d \sin \alpha \\ &\quad + 2(\Delta y - \cos \beta + \cos \alpha)d \cos \alpha \\ &\quad + 2(\Delta x + \sin \beta - \sin \alpha)d \sin \beta \\ &\quad + (-2)(\Delta y - \cos \beta + \cos \alpha)d \cos \beta \\ &= -2K_1 d \sin \alpha + 2K_2 d \cos \alpha + 2K_1 d \sin \beta + 2K_2 d \cos \beta \\ &\leq 2K_1 d \sin \alpha + 2K_2 d \cos \alpha + 2K_1 d \sin \beta + 2K_2 d \cos \beta \\ &\leq 2K_1 \cdot \frac{1}{2} \cdot \frac{h^2}{4} + 2K_2 \cdot \frac{1}{2} \cdot \frac{h^2}{4} + 2K_1 \cdot \frac{1}{2} \cdot \frac{h^2}{4} + 2K_2 \cdot \frac{1}{2} \cdot \frac{h^2}{4} \\ &= (K_1 + K_2) \cdot \frac{h^2}{2}, \end{aligned}$$

where the last \leq applies Eq. 22.

From the relation $S = \hat{p}^2$, it induces $dS = 2\hat{p}d\hat{p}$, and thus the tolerance of p , i.e., $d\hat{p}$, is expressed as:

$$\begin{aligned} d\hat{p} &= \frac{dS}{2\hat{p}} = \frac{dS}{2\sqrt{(\Delta x + \sin \beta - \sin \alpha)^2 + (\Delta y - \cos \beta + \cos \alpha)^2}} \\ &= \frac{dS}{2\sqrt{K_1^2 + K_2^2}} \\ &\leq \frac{(K_1 + K_2) \cdot \frac{h^2}{2}}{2\sqrt{(K_1^2 + K_2^2)}} \\ &\leq \frac{\sqrt{\left(\frac{K_1^2 + K_2^2}{2}\right) \cdot \frac{h^2}{2}}}{\sqrt{K_1^2 + K_2^2}} = \frac{h^2}{2\sqrt{2}}. \end{aligned}$$

The last \leq applies the arithmetic-mean–quadratic-mean inequality.

This theoretical analysis demonstrates that the proposed algorithm achieves high precision with nominal error. The following empirical trials involve h taking piece lengths of 0.01, 0.02, 0.04,..., 0.64, and comparing the empirical errors to the theoretically analyzed errors. The start and target configurations are $(0, 0, \pi/2), (3, 1, 0)$. The absolute error of the length (the tolerance) is denoted as a star in the figure. The plotted length error demonstrates that the data fit a quadratic curve well, proving that the developed MIPWLP consistently provides quite accurate and guaranteed solutions, making it a reliable choice for the practical. Since setting piece length as 0.01 can generate paths with absolute error as small as 0.0001, and shorter segment lengths lead to excessive computational costs with diminishing returns, we recommend a piece length of 0.01 for path planning.

V. DISCUSSIONS

The numerical experiments and analysis confirm the proposed algorithm's effectiveness, accuracy, and efficiency for curvature-constrained path planning with obstacles and free target heading. Deriving parameterized L/S/R primitives, handling obstacles, and the piecewise-linear strategy contribute to

1043 our method's exceptional performance, making it applicable
 1044 and suitable for various surveillance and navigation applica-
 1045 tions. However, a few concerns should be further discussed.

- 1046 1) **Robustness, generalizability, and potential applica-**
tions: The MIPWLP approach demonstrates robustness
 1047 in handling complex path planning scenarios with
 1048 curvature-bounded vehicles, particularly in obstacle-free
 1049 and obstacle-existing environments. MIPWLP's general
 1050 ability to manage both discrete and continuous vari-
 1051 ables allows it to approximate complex trigonometric
 1052 and geometric constraints effectively. The approach can
 1053 be extended to various applications [41], including
 1054 UAV reconnaissance, autonomous ground vehicle nav-
 1055 igation, and unmanned surface vehicle (USV) patrolling,
 1056 where frequent mission planning with minimal energy
 1057 consumption is critical. These characteristics make
 1058 MIPWLP suitable for a wide range of path-planning
 1059 tasks, particularly where curvature constraints, obstacle
 1060 avoidance, and energy efficiency are essential. For real-
 1061 world application deployment, several steps are required:
 1062 First, the scenario should be modeled, and the vehicle's
 1063 speed should be normalized to one. The MIPWLP
 1064 computes the optimal heading after inputting the sce-
 1065 nario parameters (as demonstrated in the pseudocode)
 1066 into the MIPWLP solver. Subsequently, a PID or MPC
 1067 controller can follow the path (as tested in our fixed-
 1068 wing simulations). Given that the velocity is constant,
 1069 the planned path is both the shortest and the quickest.
 1070 Challenges may arise in the modeling and control, while
 1071 the planning can be quite trustworthy.
- 1072 2) This work has revealed intriguing findings in the path
 1073 planning domain. Specifically, we have explored the
 1074 challenges of finding the shortest path in environments
 1075 characterized by circular obstacles. Previous work has
 1076 proven that the decision version of the problem in poly-
 1077 gonal obstacle environments with a curvature-bounded
 1078 path length below a specific threshold belongs to the
 1079 NP-complete class [24]. This work considers circular
 1080 environments where a circle is considered a giganti-
 1081 cally large number of polygons, and thus it is also
 1082 NP-complete. The **NP-completeness** can also be viewed
 1083 from the discrete variables: different breakpoints and
 1084 selections can differ in the number of variables and
 1085 constraints [42].
- 1086 3) We employ mixed-integer programming (MIP) and
 1087 approximation strategy to at most one obstacle in the
 1088 two-trip. However, the path type and the target head-
 1089 ing variables make the problem challenging. As to
 1090 whether MIPWLP can effectively address scenarios with
 1091 **multiple obstacles**, we believe it can, by constraining
 1092 obstacles sequentially, identifying critical points, and
 1093 leveraging properties like the perpendicular effect and
 1094 the lateral cross-product. MIPWLP can also be promis-
 1095 ing to find the optimal path for **different vehicle types**
 1096 (**e.g., Reeds-Shepp vehicle**) and to cooperate with RRT
 1097 strategies [43].
- 1098 4) While this paper represents a significant step forward,
 1099 it is only a small step ahead, and many assumptions

1100 are **idealized**, e.g., the effect of wind current is not
 1101 considered. The proposed approach is based on the
 1102 distance between the circle centers larger than four,
 1103 whose optimal path type is CSDSC. However, for dense
 1104 or close circles, the optimal path type is yet an open
 1105 question. MIPWLP has the potential to tackle these
 1106 types (if determined) besides the "CSDSC" because it
 1107 can introduce segment direction variables f_2 and f_4 to
 1108 parameterize the path further.

- 1109 5) Although the performance of our approach has been
 1110 rigorously evaluated and compared against previous
 1111 optimal algorithms in some scenarios, it has demon-
 1112 strated close alignment with optimal values. We have
 1113 also successfully applied our algorithm to scenarios that
 1114 previous methods could not address. Combined with
 1115 a comprehensive quality analysis of relatively simple
 1116 cases, these findings instill trust in the algorithm's
 1117 robust performance. We will build and publish a certain
 1118 **benchmark** in the future.
- 1119 6) The **error analysis** for a 5-letter two-trip is not yet avail-
 1120 able because there are several path segments, resulting
 1121 in several stages, where the directions differ from each
 1122 other, and each stage may be parameterized by f and
 1123 t, p, q, r, s . Future work can draw inspiration from [44]
 1124 and [45], and it may require error propagation tools to
 1125 establish comprehensive quality guarantees in complex
 1126 scenarios.

VI. CONCLUSION

1128 This work presents a novel mathematical programming
 1129 approach to solve the challenging problem of reconnaissanc-
 1130 e vehicles operating in obstacle environments with curvature
 1131 constraints and free target heading. By formulating the recon-
 1132 naissance scenario as a mathematical program, we effectively
 1133 handle the challenging trigonometric constraints, periodic
 1134 properties, and the direction consistencies (clockwise or coun-
 1135 terclockwise) in this problem, which other approaches, such
 1136 as derivative-based approaches, can not address. Our approach
 1137 employs a piecewise-linear strategy to approximate these
 1138 constraints, and we propose a unified derivation encompassing all
 1139 possible primitive types and different numbers of segments to
 1140 tackle this challenging problem.

1141 The MIPWLP approach incorporates various techniques,
 1142 resulting in a final mathematical formulation with theoreti-
 1143 cal properties and a bounded error. We extensively test our
 1144 algorithm across four sets of scenarios and compare it with
 1145 several algorithms in four sets of experiments, demonstrating
 1146 its effectiveness (unification ability to tackle obstacle-free
 1147 and obstacle-existing scenarios), computational efficiency (a
 1148 few seconds), high accuracy and error bounds. Robustness,
 1149 NP-completeness and a few topics about MIPWLP are also
 1150 discussed thoroughly.

1151 Looking ahead, we aim to apply this algorithm to other
 1152 similar situations and explore online real-time or distributed
 1153 implementations by drawing inspiration from convex optimiza-
 1154 tion. Future work will enhance the algorithm's performance
 1155 and analysis in real-world surveillance applications.

1157

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