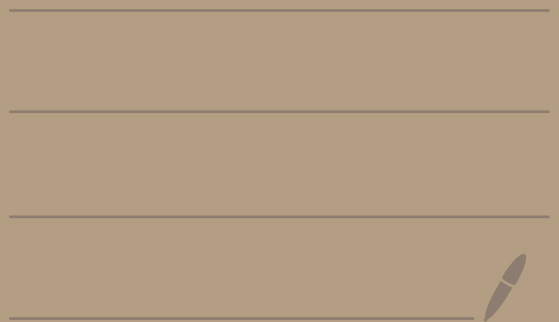


## II. Elementary viscous flow



2.2 The problem of 2-D steady viscous flow past a circular cylinder of radius  $a$  involves finding a velocity field  $\bar{u} = [u(x, y), v(x, y), 0]^T$  which satisfies

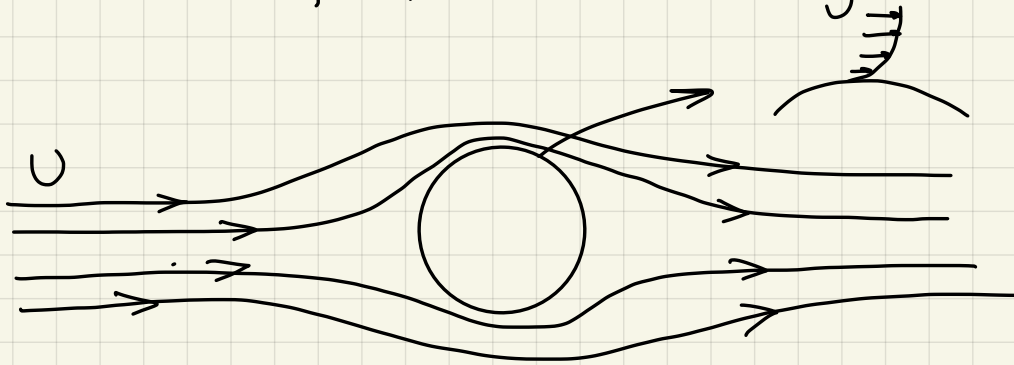
$$(\bar{u} \cdot \nabla) \bar{u} = -\frac{1}{\rho} \nabla p + \nu \nabla^2 \bar{u},$$

$$\nabla \cdot \bar{u} = 0,$$

together with the boundary conditions

$$\bar{u} = 0 \quad \text{on} \quad x^2 + y^2 = a^2,$$

$$\bar{u} \rightarrow (U, 0, 0) \quad \text{as} \quad x^2 + y^2 \rightarrow \infty.$$



Rewrite this problem in dimensionless form by using the dimensionless variables

$$\bar{x}' = \bar{x} / a, \quad \bar{u}' = \bar{u} / U, \quad p' = p / \rho U^2$$

Without attempting to solve the problem, show that the streamline pattern can depend on  $\nu$ ,  $a$  and  $U$  only in the combination, so that the flows at equal Reynolds numbers are geometrically similar.

$$(\bar{u} \cdot \nabla) \bar{u} = -\frac{1}{\rho} \nabla p + \nu \nabla^2 \bar{u}$$

$$\Rightarrow \frac{U^2}{a} (\bar{u}' \cdot \nabla') \bar{u}' = -\frac{U^2}{a} \nabla' p' + \frac{\nu U}{a^2} \nabla'^2 \bar{u}'$$

$$\frac{U}{a} \nabla' \cdot \bar{u}' = 0$$

$$\left\{ \begin{array}{l} (\bar{u}' \cdot \nabla') \bar{u}' = -\nabla' p' + \frac{\nu}{Ua} \nabla'^2 \bar{u}' \\ \nabla' \cdot \bar{u}' = 0 \end{array} \right.$$

$\uparrow$   $1/Re$

$$Re = \frac{UL}{\nu}$$

$$L \gg 1$$

$$V \ll 1$$

$$\Rightarrow R_{\text{exp}} = R_{\text{real}}$$

2.3. (i) Viscous fluid flows between two stationary rigid boundaries  $y = \pm h$  under a constant pressure gradient  $P = -dp/dx$ . Show that

$$u = \frac{P}{2\mu} (h^2 - y^2), \quad v = w = 0.$$

\_\_\_\_\_  $y = h$   $\bar{u} = (u(y), 0, 0)$



\_\_\_\_\_  $y = -h$   $\begin{matrix} y \\ \uparrow \\ \circ \rightarrow x \end{matrix}$

$$\underbrace{(\bar{u} \cdot \nabla)}_{=0} u = \underbrace{\frac{P}{\rho}}_{\text{const}} + \nu \frac{d^2 u}{dy^2}$$

$\nu = \mu/\rho$

$$u'' = -\frac{P}{\rho \nu} = -\frac{P}{\mu}$$

$$u(y) = -\frac{P}{2\mu} y^2 + C_1 y + C_2$$

$$u(-h) = u(h) = 0$$

$$u(h) = -\frac{P}{2\mu} h^2 + C_1 h + C_2 = 0$$

$$u(-h) = -\frac{P}{2\mu} h^2 - C_1 h + C_2 = 0$$

$$\begin{aligned} 1) \quad u(h) + u(-h) &= -\frac{P}{\mu} h^2 + 2C_2 = 0 \\ \Rightarrow C_2 &= \frac{P}{2\mu} h^2 \end{aligned}$$

$$2) \quad u(h) - u(-h) = 2C_1 h = 0 \Rightarrow C_1 = 0$$

$$u(y) = -\frac{P}{2\mu} y^2 + \frac{P}{2\mu} h^2 = \frac{P}{2\mu} (h^2 - y^2)$$

The plane Poiseuille flow.

(II) Viscous fluid flows down a pipe of circular cross-section  $r=a$  under a constant pressure gradient  $P = -dp/dz$ . Show that

$$u_z = \frac{P}{4\mu} (a^2 - r^2)$$

Equivalent solution but in cylindrical coordinate system.

$$\bar{u} = (0, 0, u_z(r))$$

pipe Poiseuille flow.

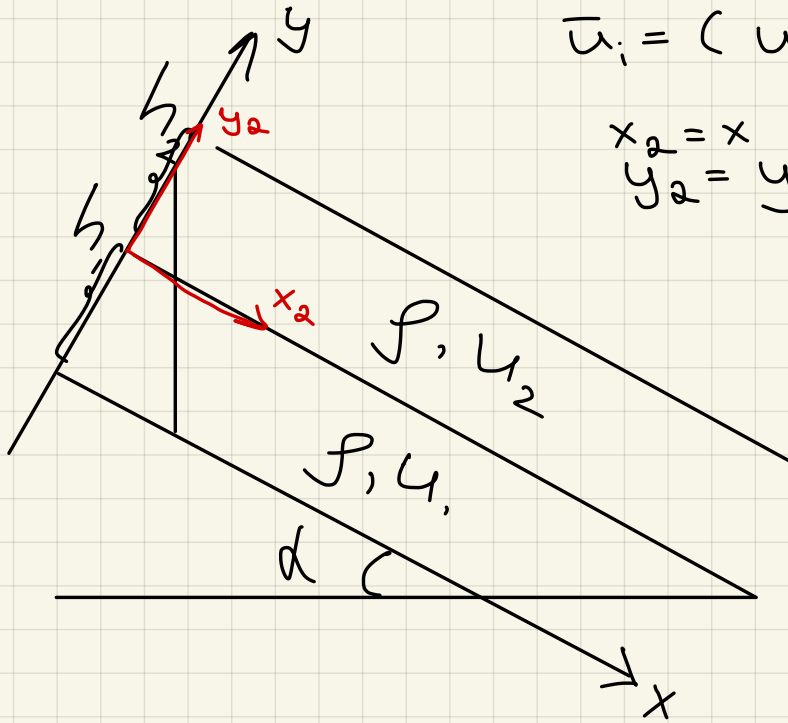
2.4 Two incompressible viscous fluids of the same density  $\rho$  flow, one on top of the other, down an inclined plane making an angle  $\alpha$  with the horizontal. Their viscosities are  $\mu_1$ , and  $\mu_2$ , the lower fluid is of depth  $h_1$ , and the upper fluid is of depth  $h_2$ . Show that

$$u_1(y) = \left[ (h_1 + h_2) y - \frac{1}{2} y^2 \right] \frac{g \sin \alpha}{\nu_1}$$

$$\bar{u}_i = (u_i(y_i), 0, 0), \quad i = 1, 2$$

$$x_2 = x \\ y_2 = y - h_1$$

$$\bar{g} = (g \sin \alpha, -g \cos \alpha)$$



$\mu_1 / g$

$$\begin{cases} (\bar{u}_i, \bar{\nabla}_i) u_i = -\frac{1}{g} \frac{\partial p_i}{\partial x} + \nu_i u_i'' + g \sin \alpha \\ \frac{1}{g} \frac{\partial p_i}{\partial y_i} = -g \cos \alpha \end{cases}$$

$$p_i = -g y \cos \alpha + f_i(x)$$

$$-\frac{1}{g} f_i' + \nu_i u_i'' + g \sin \alpha = 0$$

$$\frac{1}{g} \frac{df}{dx} = \nu_i \frac{d^2 u_i}{dy_i^2} + g \sin \alpha = \text{func}(x)$$

$$\frac{df}{dx} = \text{const}$$

$$\nu_i u_i'' = \underbrace{-g \sin \alpha - K}_{\text{const}} = -H$$

$$u_i'' = -\frac{H_i}{2} = -H_i$$

$$u_i = -\frac{H_i}{2} y_i^2 + C_{1,i} y_i + C_{2,i}$$

(i) "No-slip" at rigid bud:

$$u_1(0) = 0$$

$$u_1(0) = C_{2,1} = 0$$

(ii) "Free-surf." cond.:

$$u_2'(y_2 = h_2) = 0$$

$$u_2'(h_2) = -H_2 h_2 + C_{1,2} = 0$$

$$C_{1,2} = H_2 h_2$$

(iii) "No-slip" between 2 fluids:

$$u_1 = u_2 \text{ on } y = h_1 \text{ or } y_2 = 0$$

$$-\frac{H_1}{2} h_1^2 + C_{1,1} h_1 = C_{2,2}$$

(iv) The shear stress exerted on the fluid 1 by the fluid 2 = the sh. str. ex. on the fluid 2 by the fluid 1.

$$\nu_1 u_1' = \nu_2 u_2' \text{ on } y = h_1 \text{ or } y_2 = 0$$

2

$$\tau_1 \delta \bar{S} - \tau_2 \delta \bar{S} = \rho \delta V \bar{a}$$

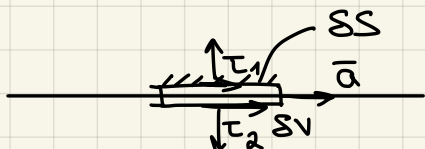
$$\bar{a} = \frac{\tau_1 \delta \bar{S} - \tau_2 \delta \bar{S}}{\rho \delta V}$$

$$\delta \bar{S}$$

$$\downarrow$$

$$\Rightarrow \bar{a} \rightarrow \infty$$

$$\Rightarrow \tau_1 = \tau_2$$



1

$$-\nu_1 H_1 h_1 + \nu_1 C_{1,1} = \nu_2 C_{1,2} = \nu_2 H_2 h_2$$

$$C_{1,1} = \frac{\nu_1 H_1 h_1 + \nu_2 H_2 h_2}{\nu_1} = \frac{H(h_1 + h_2)}{\nu_1}$$

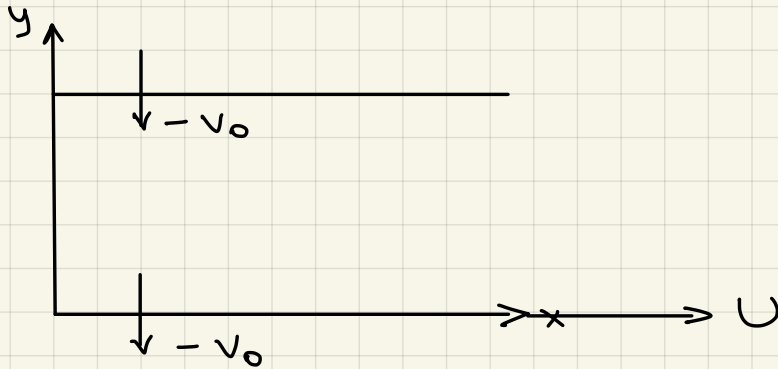
$$u_1(y) = -\frac{H}{2v_1} y^2 + \frac{H(h_1 + h_2)}{v_1} y$$

$$= \frac{(g \sin \alpha + \cancel{K})}{v_1} \left[ (h_1 + h_2) y - \frac{1}{2} y^2 \right]$$



2.6 Viscous fluid flows between two rigid boundaries  $y=0, y=h$ , the lower boundary moving in the  $x$ -direction with constant speed  $U$ , the upper boundary being at rest. The boundaries are porous, and the vertical velocity  $v$  is  $-v_0$  at each one,  $v_0$ , being a given constant (so that there is an imposed flow across the system). Show that the resulting flow is

$$u = U \left( \frac{e^{-v_0 y / \nu} - e^{-v_0 h / \nu}}{1 - e^{-v_0 h / \nu}} \right), \quad v = -v_0$$



$$\bar{u} = (u(y), -v_0, 0)$$

$$\begin{cases} (\bar{u} \cdot \nabla) u = \nu u'' \\ u(0) = U, \quad u(h) = 0 \\ -v_0 u' = \nu u'' \end{cases}$$

$$\frac{u''}{u'} = -\frac{v_0}{\nu} = (\ln u')'$$

$$\ln u' = -\frac{v_0}{\nu} y + \tilde{c}_1$$

$$u' = c_1 e^{-v_0 y / \nu}$$

$$u(y) = -\frac{\nu}{v_0} c_1 e^{-v_0 y / \nu} + c_2$$

$$u(0) = -\frac{\nu}{v_0} c_1 + c_2 = U$$

$$u(h) = -\frac{\nu}{v_0} c_1 e^{-v_0 h / \nu} + c_2 = 0$$

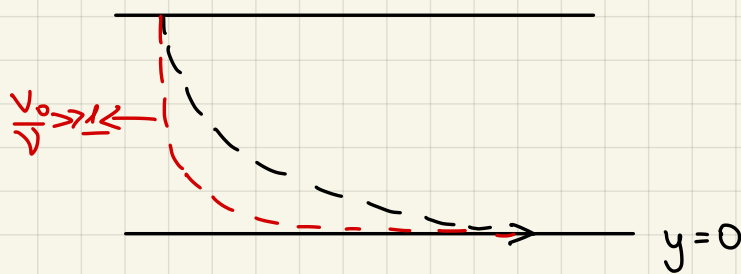
$$c_2 = \frac{\nu}{v_0} c_1 e^{-v_0 h / \nu}$$

$$-\frac{\nu}{v_0} c_1 + \frac{\nu}{v_0} c_1 e^{-v_0 h / \nu} = U$$

$$C_1 = \frac{U V_0}{\gamma} \frac{1}{e^{-V_0 h / \gamma} - 1}$$

$$u(y) = - \frac{\gamma}{V_0} \cdot \frac{U V_0}{\gamma} \frac{1}{e^{-V_0 h / \gamma} - 1} e^{-V_0 y / \gamma} + \frac{\gamma}{V_0} \cdot \frac{U V_0}{\gamma} \frac{e^{-V_0 h / \gamma}}{e^{-V_0 h / \gamma} - 1}$$

$$u(y) = \frac{e^{-V_0 y / \gamma} - e^{-V_0 h / \gamma}}{1 - e^{-V_0 h / \gamma}} U \sim e^{-V_0 y / \gamma}$$



rate of decay  $\sim \frac{V_0}{\gamma}$

$$V_0 / \gamma \gg 1$$

1)  $V_0 \gg 1 \rightarrow$  fast movement through the walls

$$2) \gamma < 1$$

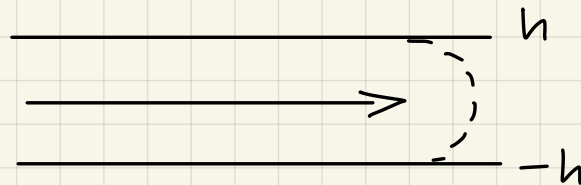
$$Re = \frac{V_0 h}{\gamma}$$

$$u(y) = \frac{e^{-Re(\frac{y}{h})} - e^{-Re}}{1 - e^{-Re}}$$

2.5 Viscous fluid is at rest in a two-dimensional channel between two stationary rigid walls  $y = \pm h$ . For  $t > 0$ , a constant pressure gradient  $P = -\frac{\partial p}{\partial x}$  is imposed. Show that  $u(y, t)$  satisfies

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2} + \frac{P}{\rho},$$

and give suitable initial and boundary conditions. Find  $u(y, t)$  in the form of a Fourier series, and show that the flow approximates to steady channel flow when  $t \gg h^2/\nu$ .



Plane Poiseuille flow:

$$u(y) = \frac{P}{2\mu} (h^2 - y^2)$$

$$t = 0$$

$$\bar{u} = 0$$

$$0 < t < t_f$$

$$\bar{u}(y, t) = u(y, t) \bar{e}_x$$

$$t \geq t_f$$

$$\bar{u}(y)$$

Eq- n of motion:

$$\frac{\partial u}{\partial t} = \frac{P}{\rho} + \nu \frac{\partial^2 u}{\partial y^2} \quad (*)$$

BC:

$$u(-h, t) = u(h, t) = 0$$

Init. cond.:

$$u(y, 0) = 0.$$

$$u(y, t) = \underbrace{u_0(y)}_{\text{Pois. flow}} + u_1(y, t)$$

$$\frac{\partial u_1}{\partial t} = \frac{P}{\rho} + \nu \frac{\partial^2 u_0}{\partial y^2} + \nu \frac{\partial^2 u_1}{\partial y^2}$$

$$\frac{\partial u_1}{\partial t} = \nu \frac{\partial^2 u_1}{\partial y^2} \quad (**)$$

$$u_1(-h, t) = u_1(h, t) = 0$$

$$\begin{aligned} u_1(y, 0) &= u(y, 0) - u_0(y) = 0 - \frac{P}{2\mu}(h^2 - y^2) \\ &= -\frac{P}{2\mu}(h^2 - y^2) \end{aligned}$$

$$u_1(y, t) = Y(y)T(t)$$

$$YT' = \nu Y''T$$

$$\Rightarrow \frac{Y''}{Y} = \frac{T'}{T} \frac{1}{\nu} = C$$

$$\begin{cases} Y'' - CY = 0 \\ T' - \nu CT = 0 \end{cases}$$

$$1) Y'' - CY = 0$$

- $C > 0, C = \lambda^2$

$$Y = K_1 e^{\lambda y} + K_2 e^{-\lambda y}$$

$$Y(-h) = K_1 e^{-\lambda h} + K_2 e^{\lambda h} = 0 \rightarrow \text{div. by } e^{-\lambda h}$$

$$Y(h) = K_1 e^{\lambda h} + K_2 e^{-\lambda h} = 0$$

$$\Rightarrow K_1 + K_2 e^{2\lambda h} = 0$$

$$-K_2 e^{2\lambda h} e^{\lambda h} + K_2 e^{-\lambda h} = -K_2 e^{-\lambda h} (e^{4\lambda h} - 1) = 0$$

$$\Rightarrow K_2 = K_1 = 0 \rightarrow \text{only triv. solut.}$$

- $C = 0 \Rightarrow Y'' = 0$

$$\Rightarrow Y = K_1 + K_2 y$$

$$\begin{aligned} Y(-h) &= K_1 - K_2 h = 0 \\ Y(h) &= K_1 + K_2 h = 0 \end{aligned}$$

$$K_1 = K_2 = 0 \rightarrow \text{only triv. sol.}$$

- $C < 0, C = -\lambda^2 < 0$

$$Y = K_1 \cos \lambda y + K_2 \sin \lambda y$$

$$\begin{aligned} y(-h) &= K_1 \cos \lambda h - K_2 \sin \lambda h = 0 \\ y(h) &= K_1 \cos \lambda h + K_2 \sin \lambda h = 0 \end{aligned}$$

$$\begin{aligned} K_1 \cos \lambda h &= 0 \\ K_2 \sin \lambda h &= 0 \end{aligned}$$

$$\underline{\cos \lambda h = 0}, \quad K_2 = 0$$

$$\rightarrow \lambda h = \left(n + \frac{1}{2}\right) \pi, \quad n = 0, 1, 2, \dots$$

$$y_n = K_{1,n} \cos \left[ \left(n + \frac{1}{2}\right) \pi \frac{y}{h} \right]$$

$$2) \quad T' - \nu C_n T = 0, \quad C_n < 0$$

$$T_n = M_n \exp \left[ - \left(n + \frac{1}{2}\right)^2 \frac{\pi^2}{h^2} \nu t \right]$$

$$u_{1,n}(y,t) = \underbrace{A_n}_{K_{1,n} M_n} \cos \left[ \left(n + \frac{1}{2}\right) \pi \frac{y}{h} \right] \exp \left[ - \left(n + \frac{1}{2}\right)^2 \frac{\pi^2}{h^2} \nu t \right]$$

$$u_1(y,t) = \sum_{n=0}^{\infty} A_n \cos \left[ \left(n + \frac{1}{2}\right) \pi \frac{y}{h} \right] \exp \left[ - \left(n + \frac{1}{2}\right)^2 \frac{\pi^2}{h^2} \nu t \right]$$

$$1) \quad u_1(y,0) = \sum_n A_n \cos \left[ \left(n + \frac{1}{2}\right) \pi \frac{y}{h} \right]$$

Multiply the LHS and the RHS by  $\cos \left[ \left(m + \frac{1}{2}\right) \pi \frac{y}{h} \right]$   
 $m = n \quad \text{or} \quad m \neq n$

$$2) \quad \text{Integr. } \int_{-h}^h dy$$

$\Rightarrow$  on the RHS the only non-zero term  
for  $n = m$

$$\begin{aligned} \int_{-h}^h \frac{P}{2\mu} \left( y^2 - \frac{1}{2} h^2 \right) \cos \left[ \left(m + \frac{1}{2}\right) \frac{\pi}{h} y \right] dy \\ = \int_{-h}^h A_m \cos^2 \left[ \left(m + \frac{1}{2}\right) \frac{\pi}{h} y \right] dy \end{aligned}$$

2.7 Incompressible fluid occupies the space  $0 < y < \infty$  above a plane rigid boundary  $y = 0$  which oscillates to and fro in the  $x$ -direction with velocity  $U \cos \omega t$ . Show that the velocity field  $\bar{u} = [u(y, t), 0, 0]$  satisfies

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2}$$

(there being no applied pressure gradient), and by seeking a solution of the form

$$u = \text{Re}[f(y) e^{i\omega t}],$$

where  $\text{Re}$  denotes 'real part of', show that

$$u(y, t) = U e^{-ky} \cos(ky - \omega t),$$

where  $k = (\omega/2\nu)^{1/2}$ .



$$\bar{u}_\infty = U \cos \omega t \bar{e}_x \rightarrow \bar{u} = [u(y, t), 0, 0]$$

$$\text{BC: } u(0, t) = U \cos \omega t,$$

$$u(\infty, t) = 0.$$

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2} \quad (*)$$

$$u = \underbrace{\text{Re}}_{\text{real part}} [f(y) e^{i\omega t}] \rightarrow \text{normal mode sol.}$$

$$i\omega f e^{i\omega t} = \nu f'' e^{i\omega t}$$

$$\Rightarrow f'' = \frac{i\omega}{\nu} f$$

$$\text{General sol.: } f = A e^{\sqrt{\frac{i\omega}{\nu}} y} + B e^{-\sqrt{\frac{i\omega}{\nu}} y}$$

$$\sqrt{i} = \pm \frac{1}{\sqrt{2}} (1 + i).$$

$$k = \sqrt{\frac{\omega}{2\nu}}$$

$$\Rightarrow f(y) = A e^{(1+i)ky} + B e^{-(1+i)ky}$$

$$BC: f(y \rightarrow \infty) = A e^{(1+i)k\infty} = 0 \Rightarrow A = 0$$

$$\begin{aligned} u(y, t) &= \operatorname{Re} [B e^{-(1+i)ky} e^{i\omega t}] \\ &= \operatorname{Re} [B e^{-ky} e^{i(\omega t - ky)}] \\ &= B e^{-ky} \cos(\omega t - ky) \end{aligned}$$

$$u(0, t) = B \cos \omega t = U \cos \omega t$$

$$\Rightarrow B = U$$

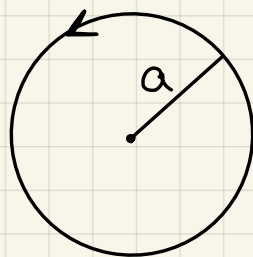
$$u(y, t) = U e^{-ky} \cos(\omega t - ky)$$

2.8 A circular cylinder of radius  $a$  rotates with constant angular velocity  $\Omega$  in a viscous fluid. Show that the line vortex flow

$$\bar{u} = \frac{\Omega a^2}{r} \bar{e}_\theta, \text{ for } r \geq a$$

is an exact solution of the equations and boundary conditions.

Describe roughly how the vorticity changes with time when the cylinder is suddenly started into rotation from a state of rest. Likewise, discuss the case in which an outer cylinder  $r=b$  is simultaneously given an angular velocity  $\Omega a/b$ .



surrounded by fluid.

$$\bar{u}(r, t) = u(r, t) \bar{e}_\theta$$

$$BC: u(\infty, t) = 0, \quad u(a, t) = \Omega a$$

$$\frac{\partial u}{\partial t} = \nu \left( \frac{\partial^2 u}{\partial r^2} + \frac{1}{r} \frac{\partial u}{\partial r} - \frac{u}{r^2} \right)$$

$$\bar{u}_v = \frac{\Omega a^2}{r} \bar{e}_\theta \text{ for } r \geq a.$$

$$u_v(a) = \Omega a$$

$$\frac{\partial u_v}{\partial t} = 0$$

$$+ \frac{2 \Omega a^2}{r^3} - \frac{\Omega a^2}{r^3} - \frac{\Omega a^2}{r^3} = 0$$

$$\bar{\omega} = \nabla \times \bar{u} = \bar{\omega}(r, t)$$

$$\Rightarrow \frac{\partial \bar{\omega}}{\partial t} = \nu \nabla^2 \bar{\omega}$$

$$\bar{\omega}_v = \bar{e}_z \left( \frac{\partial u_v}{\partial r} + \frac{u_v}{r} \right) = 0$$