Assignment Report - 2

Vortex Lattice Method - used to find lift generated on a rudder

Aditya Narayanan (OE13S013) *

Abstract

Vortex Lattice Method is used to calculate numerically the lift generated on a flat rudder held at various angles of attack and is compared with the theoretical lift values obtained using the analytical equation for lift on an infinite plate.

1 Problem Statement

Consider a flat plate rudder of height (span) 5 m and width (chord) 1.5 m of a ship (as shown in Fig. 1). Let the density of water be $1000 \, [kg/m3]$. Neglecting interactions with the wave surface and the ship hull and using the vortex lattice method given in this notes, develop a computer program and determine the lift force generated by the rudder at forward speed Uo = $10 \, [m/s]$ for angle of attack 3, 5, 10, 15 and 20^{0} . Obtain the result for lattice size of N=9, 20, 50, 100 etc and study convergence. Discuss how the results compare with 2D flat plate analytical result which for this problem is:

 $2\pi\alpha[0.5\rho U_0^2 \times (Chordlength)] \times (Spanlength)$

Figure 1: Schematic of rudder (source: Notes, Prof. P Ananthakrishnan)

2 Methodology

The domain is discretized into a number of panels. Each panel has a horseshoe vortex whose strength is to be determined using the no flux condition. The velocity induced by the vortex is determined by the Biot-Savart law. The normal velocity of the fluid is known to be $U_0 sin(\alpha)$. The circulation is determined by solving the system of equations and the Kutta-Joukowski theorem is used to calculate the numerical lift force.

^{*}email: adityarn@gmail.com

The converged values of numerical lift (using 500 panels) is plotted along with the theoretical lift values for various angles of attack in Fig 2.	3 Results
	The converged values of numerical lift (using 500 panels) is plotted along with the theoretical lift values for various angles of attack in Fig 2.
Figure 2: Numerical and Theoretical Lift force for angles of attack (marked by the squares) of $3,5,10,15$, and 20°	Figure 2: Numerical and Theoretical Lift force for angles of attack (marked by the squares) of $3,5,10,15$, and 20° .