# Classical Mechanics: Assignment #5

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The Euler Equations of motion for rotation about principal axes with moments of inertia  $I_1, I_2, I_3$  and torques  $N_1, N_2, N_3$  are given by,

$$I_1 \dot{\omega}_1 - \omega_2 \omega_3 (I_2 - I_3) = N_1 \tag{1}$$

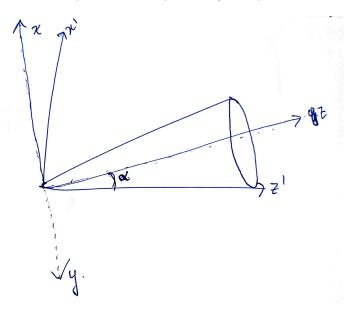
$$I_2 \dot{\omega}_2 - \omega_3 \omega_1 (I_3 - I_1) = N_2 \tag{2}$$

$$I_3\dot{\omega}_3 - \omega_1\omega_2(I_1 - I_2) = N_3 \tag{3}$$

# Problem 1

#### Part (a)

The moment of inertia tensor in the xyz coordinate system is given by,



$$I = \frac{3mh^2}{5} \begin{bmatrix} 1 + \frac{\tan^2 \alpha}{4} & 0 & 0\\ 0 & 1 + \frac{\tan^2 \alpha}{4} & 0\\ 0 & 0 & \frac{\tan^2 \alpha}{2} \end{bmatrix}$$

As I is a tensor, we rotate the current I to the get the moment of inertia I' in the x'y'z' frame,

$$I' = R(\alpha)IR^T(\alpha)$$

$$I' = \frac{3mh^2}{5} \begin{bmatrix} \left(\frac{\tan^2\alpha}{4} + 1\right)\cos^2\alpha + \frac{1}{2}\sin^2\alpha\tan^2\alpha & 0 & \frac{1}{2}\sin^2\alpha\tan\alpha - \cos\alpha\sin\alpha\left(\frac{\tan^2\alpha}{4} + 1\right) \\ 0 & \frac{\tan^2\alpha}{4} + 1 & 0 \\ \frac{1}{2}\sin^2\alpha\tan\alpha - \cos\alpha\sin\alpha\left(\frac{\tan^2\alpha}{4} + 1\right) & 0 & \left(\frac{\tan^2\alpha}{4} + 1\right)\sin^2\alpha + \frac{\sin^2\alpha}{2} \end{bmatrix}$$

The angular velocity is along the instantaneous axis of rotation, which in this case is the new z axis. The cone traces out a circle of its slant height  $l = \frac{h}{\cos \alpha}$ . Hence the angular velocity is just,

$$\Omega = \frac{2\pi l}{\tau h \tan \alpha} = \frac{2\pi}{\tau \sin \alpha}$$

Hence, the angular velocity vector  $\vec{\Omega} = \begin{bmatrix} 0 & 0 & \Omega \end{bmatrix}$ . The angular momentum  $\vec{L}$  is given by,

$$ec{\mathbf{L}} = I' ec{oldsymbol{\Omega}} = \left[ egin{array}{c} -rac{h^3 \pi 
ho (5\cos(2lpha) + 3) an^2(lpha)}{20 au} \ 0 \ rac{h^3 \pi 
ho (5\cos(2lpha) + 7) an^3(lpha)}{20 au} \end{array} 
ight]$$

and the Kinetic energy K is,

$$K = \vec{\Omega}^T \vec{\mathbf{L}} = \frac{\pi^2 h^3 \rho (5\cos(2\alpha) + 7) \tan^2(\alpha)}{10\tau^2}$$

In both the above expressions, we have used,

$$m = \rho \frac{\pi r^2 h}{3} = \rho \frac{\pi h^3 \tan^2 \alpha}{3}$$

#### Part (b)

The coordinates of the centre of mass of the rod are given by,

$$x = l \sin \theta$$
 and  $y = \alpha + l \cos \theta \implies \dot{x}^2 + \dot{y}^2 = \dot{\alpha}^2 + l^2 \dot{\theta}^2 - 2l \dot{\alpha} \dot{\theta} \sin \theta$ 

Let's write down the expressions for Kinetic Energy T and Potential Energy V,

$$T = \frac{Mv_{CM}^{2}}{2} + \frac{I_{CM}\dot{\theta}^{2}}{2}$$

$$= \frac{M(\dot{x}^{2} + \dot{y}^{2})}{2} + \frac{Ml^{2}\dot{\theta}^{2}}{6}$$

$$T = \frac{M(\dot{\alpha}^{2} + l^{2}\dot{\theta}^{2} - 2l\dot{\alpha}\dot{\theta}\sin\theta)}{2} + \frac{Ml^{2}\dot{\theta}^{2}}{6}$$

$$V = -Mgy_{CM} + \frac{k\alpha^{2}}{2}$$

$$= -Mg(\alpha + l\cos\theta) + \frac{k\alpha^{2}}{2}$$

We now write down the Lagrangian,

$$L = \frac{M(\dot{\alpha}^2 + l^2\dot{\theta}^2 - 2l\dot{\alpha}\dot{\theta}\sin\theta)}{2} + \frac{Ml^2\dot{\theta}^2}{6} + Mg(\alpha + l\cos\theta) - \frac{k\alpha^2}{2}$$

The equations of motion are,

$$\frac{\mathrm{d}}{\mathrm{d}t} \Big( M\dot{\alpha} - l\dot{\theta}\sin\theta \Big) = Mg - k\alpha \implies \ddot{\alpha} - \frac{l\ddot{\theta}\sin\theta + l\dot{\theta}^2\cos\theta}{M} - g + \frac{k\alpha}{M} = 0 \quad \text{and} \quad \frac{\mathrm{d}}{\mathrm{d}t} \left( Ml^2\dot{\theta} - l\dot{\alpha}\sin\theta + \frac{Ml^2\dot{\theta}}{3} \right) = -Mgl\sin\theta \implies \frac{4l\ddot{\theta}}{3} - \frac{\ddot{\alpha}\sin\theta + \dot{\alpha}\dot{\theta}\cos\theta}{M} + g\sin\theta = 0$$

#### Problem 2

Let  $f(q_i, t)$  be a function such that  $L'(q_i, \dot{q}_i, t) = L(q_i, \dot{q}_i, t) + \frac{\mathrm{d}f}{\mathrm{d}t}$ . We have seen that the equations of motion remain unchanged by this addition.

Let's first calculate the canonical momenta  $p'_i$ ,

$$p_i' = \frac{\partial L'}{\partial \dot{q}_i} = \frac{\partial L}{\partial \dot{q}_i} + \frac{\partial}{\partial \dot{q}_i} \left(\frac{\mathrm{d}f}{\mathrm{d}t}\right) = p_i + \frac{\partial}{\partial \dot{q}_i} \left(\frac{\mathrm{d}f}{\mathrm{d}t}\right) = p_i + \frac{\partial}{\partial \dot{q}_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial f}{\partial q_j} \dot{q}_j\right) = p_i + \frac{\partial f}{\partial q_i} \left(\sum_j \frac{\partial$$

Let's calculate the Hamiltonian  $H'(q_i, p_i, t)$ ,

$$\begin{split} H'(q_i,p_i,t) &= \sum_i \left( p_i + \frac{\partial f}{\partial q_i} \right) \dot{q}_i - L' \\ &= \sum_i p_i \dot{q}_i - L + \sum_i \dot{q}_i \frac{\partial f}{\partial q_i} - \frac{\mathrm{d}f}{\mathrm{d}t} \\ &= H(q_i,p_i,t) + \sum_i \dot{q}_i \frac{\partial f}{\partial q_i} - \sum_i \dot{q}_i \frac{\partial f}{\partial q_i} \\ H'(q_i,p_i,t) &= H(q_i,p_i,t) \end{split}$$

As the Hamiltonians are essentially the same, the equations of motion are same as well.

#### Part (b)

Given that,

$$\begin{split} L &= -mc^2 \sqrt{1 - \frac{v^2}{c^2}} - e\phi + \frac{e}{c} \vec{\mathbf{v}} \cdot \vec{\mathbf{A}} \\ &= -mc^2 \sqrt{1 - \frac{\dot{x}^2 + \dot{y}^2 + \dot{z}^2}{c^2}} - e\phi + \frac{e}{c} (\dot{x} A_x + \dot{y} A_y + \dot{z} A_z) \\ &= -\frac{m(c^2 - \dot{x}^2 + \dot{y}^2 + \dot{z}^2)}{\sqrt{1 - \frac{v^2}{c^2}}} - e\phi + \frac{e}{c} (\dot{x} A_x + \dot{y} A_y + \dot{z} A_z) \end{split}$$

We first write the canonical momenta,

$$p_{x} = \frac{\partial L}{\partial \dot{x}} = -mc^{2} \frac{\left(\frac{-2\dot{x}}{c^{2}}\right)}{2\sqrt{1 - \frac{v^{2}}{c^{2}}}} + \frac{e}{c}A_{x} = \frac{m\dot{x}}{\sqrt{1 - \frac{v^{2}}{c^{2}}}} + \frac{e}{c}A_{x}$$

Similarly,

$$p_y = \frac{m\dot{y}}{\sqrt{1 - \frac{v^2}{c^2}}} + \frac{e}{c}A_y$$
 and  $p_z = \frac{m\dot{z}}{\sqrt{1 - \frac{v^2}{c^2}}} + \frac{e}{c}A_z$ 

Let's first evaluate 
$$\gamma = \frac{1}{\sqrt{1 - \frac{v^2}{c^2}}}$$
.

$$\gamma = \frac{1}{\sqrt{1 - \frac{v^2}{c^2}}} = \frac{1}{\sqrt{1 - \frac{\sum_{i=x,y,z} \dot{x}_i^2}{c^2}}}$$

From the formulae of the canonical momenta, one can see that,

$$\begin{split} H &= p_x \dot{x} + p_y \dot{y} + p_z \dot{z} - L \\ &= \frac{p_x \left( p_x - \frac{eA_x}{c} \right)}{\gamma m} + \frac{p_y \left( p_y - \frac{eA_y}{c} \right)}{\gamma m} + \frac{p_z \left( p_z - \frac{eA_z}{c} \right)}{\gamma m} - \frac{e \left( \frac{A_x \left( p_x - \frac{eA_x}{c} \right)}{\gamma m} + \frac{A_y \left( p_y - \frac{eA_y}{c} \right)}{\gamma m} + \frac{A_z \left( p_z - \frac{eA_z}{c} \right)}{\gamma m} \right)}{c} + e\phi + mc^2 \gamma \\ &= \frac{-2ceA_x p_x - 2ceA_y p_y - 2ceA_z p_z + e^2(A_x^2 + A_y^2 + A_z^2) + c^2(p_x^2 + p_y^2 + p_z^2)}{c^2 \gamma m} + e\phi + mc^2 \gamma \\ H &= \frac{c^2 \sum_{i=x,y,z} (p_i - \frac{e}{c}A_i)^2}{c^2 \gamma m} + e\phi + mc^2 \gamma \end{split}$$

$$H = \sqrt{p^2c^2 + m^2c^4} + e\phi$$

where  $p^2 = \sum_{i=x,y,z} (p_i - \frac{e}{c}A_i)^2$ . This is  $\neq T + V$ . The Hamilton equations of motion can then be written as,

$$\dot{x}_i = \frac{p_i - \frac{e}{c} A_i}{\sqrt{p^2 c^2 + m^2 c^4}} \quad \text{and} \quad \dot{p}_i = \frac{e}{c} \frac{\left(p_i - \frac{e}{c} A_i\right) \frac{\partial A_i}{\partial x_i}}{\sqrt{p^2 c^2 + m^2 c^4}} - e \frac{\partial \phi}{\partial x_i}$$

## Problem 3

From the torque-free Euler equations of motion, one has,

$$I_1 \dot{\omega}_1 - \omega_2 \omega_3 (I_2 - I_3) = 0$$
  

$$I_2 \dot{\omega}_2 - \omega_3 \omega_1 (I_3 - I_1) = 0$$
  

$$I_3 \dot{\omega}_3 - \omega_1 \omega_2 (I_1 - I_2) = 0$$

Let's say that the body is undergoing steady motion. This means,

$$\omega_2 \omega_3 (I_2 - I_3) = \omega_3 \omega_1 (I_3 - I_1) = \omega_1 \omega_2 (I_1 - I_2) = 0$$

If  $I_1, I_2, I_3$  are all different, the above condition can be satisfied if and only if two of the  $\omega$ 's are zero. Hence, if the body is indeed undergoing uniform motion, it can do so only along one of the principal axes. (In general, one cannot reduce an overall non-uniform motion to uniform motion along one axis). Let  $\omega_1$  be the non-zero angular velocity. We apply a perturbation, such that  $\omega_1' = \omega_1 + \epsilon \Omega_1, \omega_2' = \epsilon \Omega_2, \omega_3' = \epsilon \Omega_3$ 

 $\epsilon\Omega_3$ . Substituting this in the Euler equations (upto first order in  $\epsilon$ ),

$$I_1\dot{\Omega}_1 = 0$$
  

$$I_2\dot{\Omega}_2 = \omega_1\Omega_3(I_3 - I_1)$$
  

$$I_3\dot{\Omega}_3 = \omega_1\Omega_2(I_1 - I_2)$$

The last two equations can be differentiated with respect to t and can be written as follows,

$$\ddot{\Omega}_2 = -\frac{\omega_1^2}{I_2 I_3} (I_1 - I_2) (I_1 - I_3) \Omega_2$$
  
$$\ddot{\Omega}_3 = -\frac{\omega_1^2}{I_2 I_2} (I_1 - I_3) (I_1 - I_2) \Omega_3$$

We see that the  $\Omega_1$  and  $\Omega_2$  will oscillate with the same frequency  $\alpha = \sqrt{\frac{\omega_1^2}{I_2 I_3}(I_1 - I_2)(I_1 - I_3)}$ . The motion will be stable only when  $\alpha > 0 \implies I_1 > I_2$  and  $I_1 > I_3$  or  $I_1 < I_2$  and  $I_1 < I_3$ .

# Problem 4

The sides are given to be 2a, 2b, 2c along x, y, z. Let's first calculate the moments of inertia,

$$I_{zz} = \rho \int_{-a}^{a} \int_{-a}^{a} \int_{-b}^{b} dx dy dz (x^{2} + y^{2})$$

$$= \rho(2b) \int_{-a}^{a} dx \left( x^{2}y + \frac{y^{2}}{3} \right) \Big|_{-a}^{a}$$

$$= \rho(2b) \left( \frac{x^{3}}{3} (2a) + x \frac{2a}{3} \right) \Big|_{-a}^{a}$$

$$= \frac{M}{8a^{2}b} (2b) \frac{8a^{4}}{3}$$

$$I_{zz} = \frac{2Ma^{2}}{3}$$

Similarly,  $I_{yy} = I_{xx} = \frac{M(a^2 + b^2)}{3}$ . Let's now calculate  $I_{xy}$ ,

$$I_{xy} = -\rho \int_{-a}^{a} \int_{-a}^{a} \int_{-b}^{b} dx dy dz \ xy$$
$$I_{xy} = 0$$

Similarly,  $I_{yz} = I_{zx} = 0$ . Hence  $I_{jj}$ 's are the moment of inertia about principal axes. The system has no forces acting on it. We can then write down the Euler equations as follows, with  $I_{xx} = I_1, I_{yy} = I_2, I_{zz} = I_3$ 

$$\dot{\omega}_1 - \omega_2 \omega_3 \left( 1 - \frac{2a^2}{a^2 + b^2} \right) = 0$$

$$\dot{\omega}_2 - \omega_3 \omega_1 \left( \frac{2a^2}{a^2 + b^2} - 1 \right) = 0$$

$$\dot{\omega}_3 = 0 \implies \omega_3 = constant$$

Taking the time derivative of the first equation and substituting for  $\dot{\omega}_2$  from the second equation, we get,

$$\ddot{\omega}_1 = -\left[\omega_3 \left(1 - \frac{2a^2}{a^2 + b^2}\right)\right]^2 \omega_1 \implies \text{periodic motion with frequency} \quad \omega = \omega_3 \left|1 - \frac{2a^2}{a^2 + b^2}\right|$$

As the problem is symmetric about the x and y axes, the motion will be periodic about the y axis too.