

# MAC for Incompressible

## Navier - Stokes Flow

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The complete Navier Stokes equation, i.e., the momentum conservation equation, valid for inhomogeneous fluids as well, is:

$$\frac{\text{momentum}}{\text{density}} = \rho \vec{u} \quad \begin{matrix} \rightarrow \\ \uparrow \end{matrix} \quad \begin{matrix} \text{fluid} \\ \text{velocity} \end{matrix}$$

mass density

$$(\rho \vec{u})_t + \vec{\nabla} \cdot (\rho \vec{u} \vec{u}^T) + \nabla p =$$
$$\nabla \cdot (\overleftrightarrow{\sigma}_0) + \underbrace{\rho g}_{\text{viscous stress tensor}} \quad \begin{matrix} \leftarrow \text{pressure} \\ \text{(e.g.)} \\ \leftarrow \text{gravity} \end{matrix}$$

(1)

This needs to be supplemented with  
(at least) mass conservation:

$$S_t + \nabla \cdot (\rho \vec{u}) = 0$$

(continuity equation)

Equivalent form of momentum eq:

$$\rho \vec{u}_t + \rho \vec{u} \cdot \nabla \vec{u} = \nabla \cdot (\vec{\sigma}_v) + \rho \vec{g}$$

Where viscous stress tensor is:

$$\vec{\sigma}_v = \eta (\vec{\nabla} \vec{u} + \vec{\nabla}^T \vec{u})$$

$$\sigma_{ij}^{(v)} = \eta (\partial_i u_j + \partial_j u_i)$$

Often one can make two key

assumptions:

$$\left\{ \begin{array}{l} S = \text{const} \\ \eta = \text{const} \end{array} \right.$$

means flow is

constant viscosity

Low Mach number

②

This allows for some key simplifications of the equations:

$$\nabla \cdot (\rho \vec{u}) = \rho \nabla \cdot \vec{u} = 0 \Rightarrow \nabla \cdot \vec{u} = 0$$

(incompressible flow)

This leads to more simplifications:

$$\nabla \cdot \vec{\sigma} = \eta \nabla^2 \vec{u}$$

why:  $\partial_j \sigma_{ji} = \eta \partial_j^2 u_i + \eta \partial_i (\partial_j u_j)$

zero

$$\left\{ \begin{array}{l} \rho (\vec{u}_t + \vec{u} \cdot \nabla \vec{u}) + \nabla p = \eta \nabla^2 \vec{u} \\ \nabla \cdot \vec{u} = 0 \end{array} \right.$$

pressure is the Lagrange multiplier for incompressibility

Challenge:

1)  $\vec{u} \cdot \nabla \vec{u}$  is non linear

2) Incompressibility constraint

③

Formally, NS equations are a differential-algebraic (DAE) system of equations of index 2.

This is hard even for ODEs!

In periodic / unbounded domains, we can eliminate pressure using a projection operator. Start with Hodge decomposition theorem:

$$\nabla \vec{\varphi} = -\vec{\nabla} \varphi + \vec{\nabla} \times \vec{A}$$

$$= -\vec{\nabla} \varphi + \vec{u}$$

s.t.  $\vec{\nabla} \cdot \vec{u} = 0$

$$\vec{u} = \underbrace{\vec{P} \vec{\varphi}}_{\text{projection operator}} = \vec{\varphi} + \vec{\nabla} \varphi$$

(4)

How to compute  $\varphi$ ?

$$\vec{\nabla} \cdot \vec{u} = \nabla \cdot \vec{\varphi} + \nabla^2 \varphi = 0$$

$$\Rightarrow -\nabla^2 \varphi = \nabla \cdot \vec{\varphi} \leftarrow \begin{array}{l} \text{Poisson equation} \\ \text{for} \\ \text{pressure} \end{array}$$

(Solving the incompressible equations  
is always at least as hard as  
solving a Poisson equation (elliptic),  
so generally much more expensive  
than simple advection-diffusion).

In periodic/unbounded domains  
we can write NS in unconstrained  
form (c.f. vorticity-stream):

$$u_t = \mathcal{P} \left[ -u \cdot \nabla u + \underbrace{\nu \nabla^2 u}_{\nu = \eta/\sigma} + f \right]$$

↑  
gravity/  
wind/etc.

(5)

The problem is that this does not work with boundary conditions! First methods for NS were based on the **projection method** of A. Chorin, but I will present a different version that deals correctly with BCs.

Key point to remember:

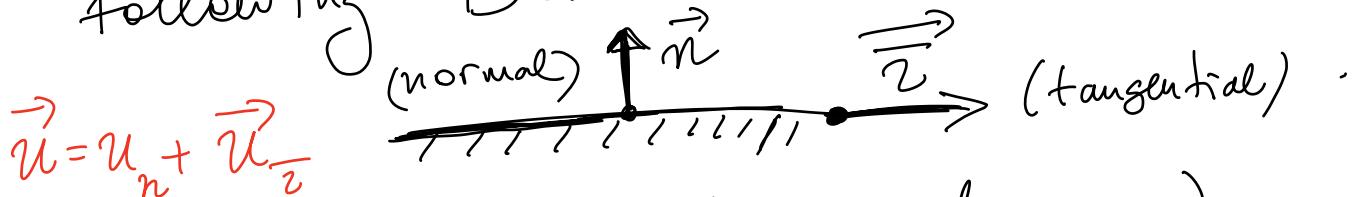
One cannot separate pressure & velocity when there are physical boundaries: must solve for both at the same time.

In other words, we will need to solve something like **Stokes equation** not Poisson equation to get evolution of  $\phi/p$ .

⑥

## BCs for incompressible NS

At a physical boundary, the following BCs are common.



Normal component (pick one)  
(also for Euler eqs)

① No penetration  
 $u_n = \vec{u} \cdot \vec{n} = 0$  (Dirichlet)  
 normal vector

② Specified normal stress:

$$(\text{Neumann}) \quad \vec{n} \cdot \vec{\sigma} \cdot \vec{n} = -p + 2\eta \frac{\partial u_n}{\partial n}$$

Tangential component (pick one)  
(only if  $n > 0$ )

① Specified slip (or no slip)

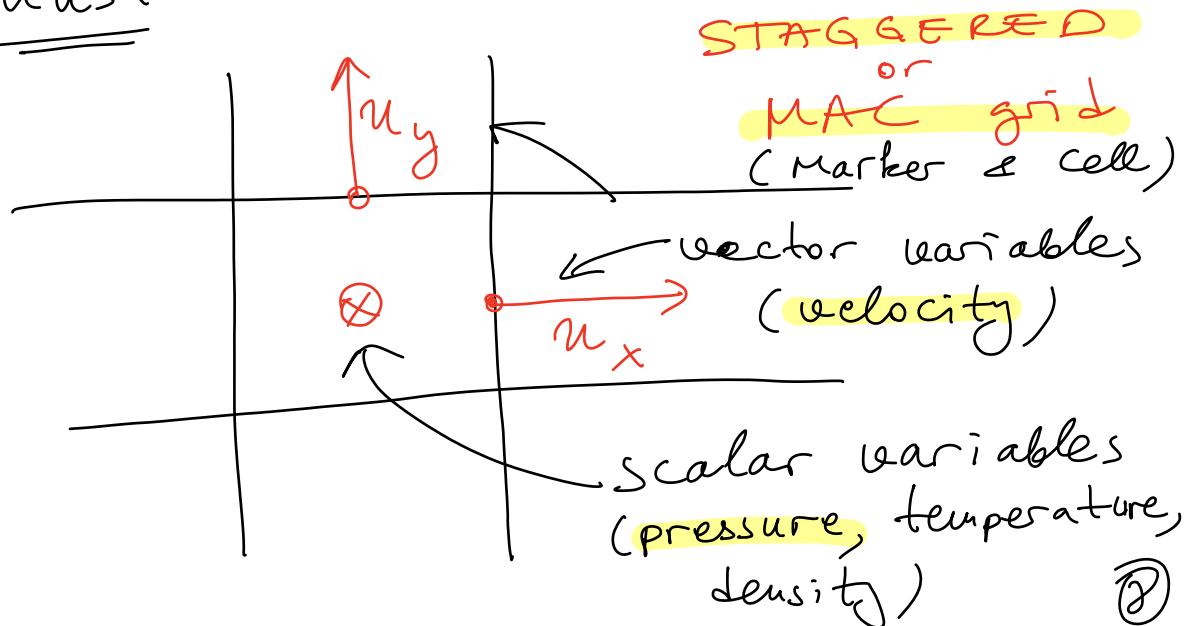
$$\vec{u}_z = \vec{n} - \vec{u} \cdot \vec{n} = 0 \quad (\text{Dirichlet})$$

② Specified traction

$$\vec{z} \cdot \vec{\sigma} \cdot \vec{n} = \eta \left[ \frac{\partial \vec{u}_z}{\partial n} + \frac{\partial \vec{u}_n}{\partial \vec{z}} \right] \quad (7)$$

## Numerical Method: MAC

What I will describe can be viewed as a finite difference but also (discontinuous Galerkin) finite element method. It is the simplest 2<sup>nd</sup> order method but only works for **regular grid**. I will illustrate in 2D but also works in 3D. Domain must be an orthogonal cuboid.



Scalars "live" on a standard finite volume / difference grid :

$\nabla \cdot \vec{v}$  or pressure  $\rightarrow$  cell centered

but vector fields like velocity have different components "living" on distinct grids

velocity  $\rightarrow$  face centered

Velocities are on a staggered grid.

Finite difference MAC operators / matrices

$$\vec{u} = (u, v)$$

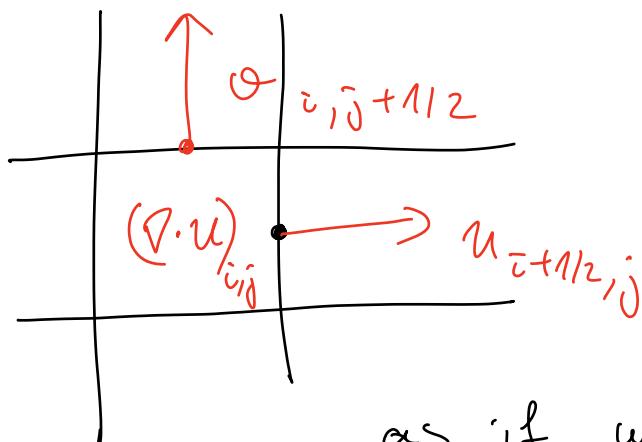
$$\textcircled{1} \quad \nabla \cdot u \longleftrightarrow D u$$

↓  
discrete divergence

$$(D u)_{i,j} = \frac{u_{i+1/2,j} - u_{i-1/2,j}}{\Delta x}$$

$$- \frac{v_{i,j+1/2} - v_{i,j-1/2}}{\Delta y}$$

⑨



This really is  
a divergence of  
a flux through  
faces, exactly

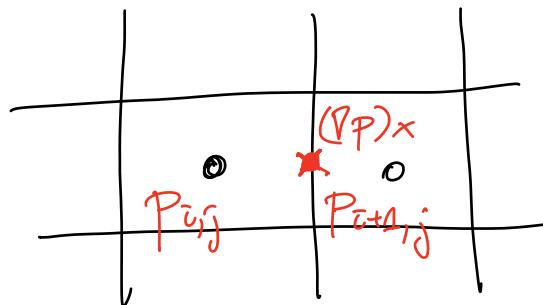
as if we were doing  
finite volume schemes  
for the scalar fields

$$\textcircled{2} \quad D_P \longleftrightarrow G_P$$

↑  
Discrete gradient

$$(G_P)_{\bar{i}+1/2, j}^x = \frac{P_{\bar{i}+1, j} - P_{\bar{i}, j}}{\Delta x}$$

$$(G_P)_{\bar{i}, \bar{j}+1/2}^y = \frac{P_{\bar{i}, \bar{j}+1} - P_{\bar{i}, \bar{j}}}{\Delta y}$$



$D = -G_T^{adjoint}$

\textcircled{10}

③ Scalar / vector Laplacian is standard  $S^+$  Laplacian, e.g.

$$(L_c P)_{i,j} = \frac{P_{i+1,j} - 2P_{i,j} + P_{i-1,j}}{\Delta x^2}$$

$$+ \frac{P_{i,j+1} - 2P_{i,j} + P_{i,j-1}}{\Delta y^2}$$

Important :  $L_c = DG$

Similarly

$$(L^x u)_{i+1/2,j} = \frac{u_{i+3/2,j} - 2u_{i+1/2,j} + u_{i-1/2,j}}{\Delta x^2}$$

$$+ \frac{u_{i+1/2,j+1} - 2u_{i+1/2,j} + u_{i+1/2,j-1}}{\Delta y^2}$$

(11)

With these operators we can write a discrete NS equation

$$\left\{ \begin{array}{l} S \frac{d\vec{u}}{dt} + GP = \eta \begin{pmatrix} L_x u \\ L_y \phi \end{pmatrix} - \vec{N}(\vec{u}) \\ D\vec{u} = 0 \end{array} \right.$$

where  $\vec{N}(\vec{u}) \Leftrightarrow \vec{u} \cdot \vec{\nabla} \vec{u}$

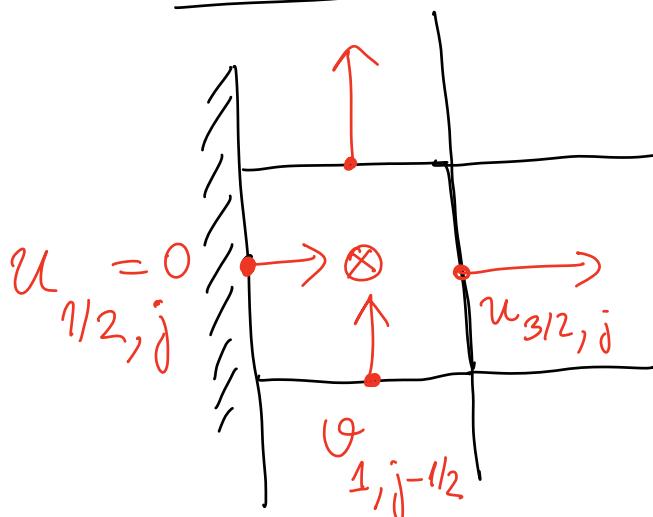
(DAE system of index 2)

Things we need:

- a) Temporal integrator
- b) How to discretize  $\vec{N}(\vec{u})$
- c) Boundary conditions

The "easiest" of these is BCs, especially for Dirichlet BCs on  $\vec{u}$  (e.g. no slip / no penetration)

## No-slip BCs



$$(D\vartheta)_{1,j} =$$

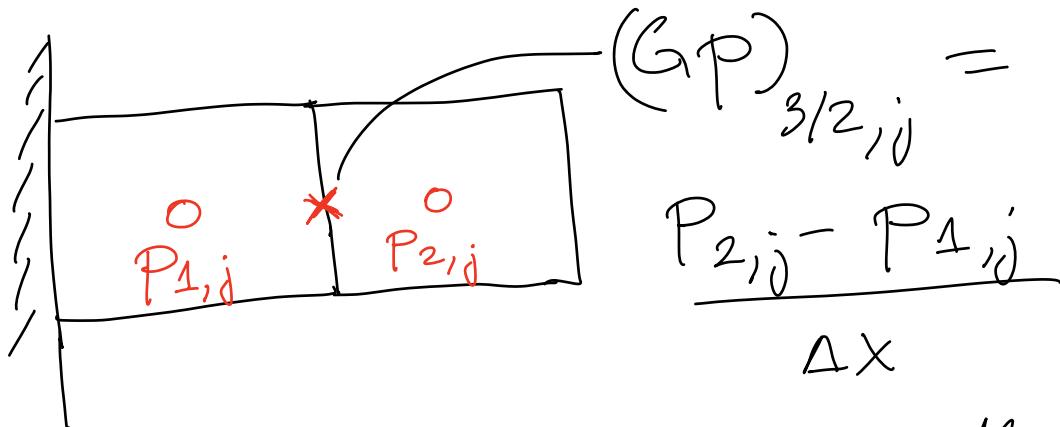
Discrete Divergence is easy to write:

$$\frac{u_{3/2,j} - u_{1/2,j}}{\Delta x}$$

$$+ \frac{\vartheta_{1,j+1/2} - \vartheta_{1,j-1/2}}{\Delta y}$$

Since normal velocity is given (zero for no-slip), the discrete divergence  $D$  maps from velocity on interior faces to cell centers.

The discrete gradient  $G = -D^*$   
 therefore maps from cell  
 centers to interior faces.



We do not need ghost cells  
 to define either  $D$  or  $G$  - this  
 is the real beauty of the  
 staggered grid and why it is  
 capturing the physics of  
 incompressible flow.

$L_c = D G$   
looks like a discrete Laplacian  
 with Neumann BCs for pressure

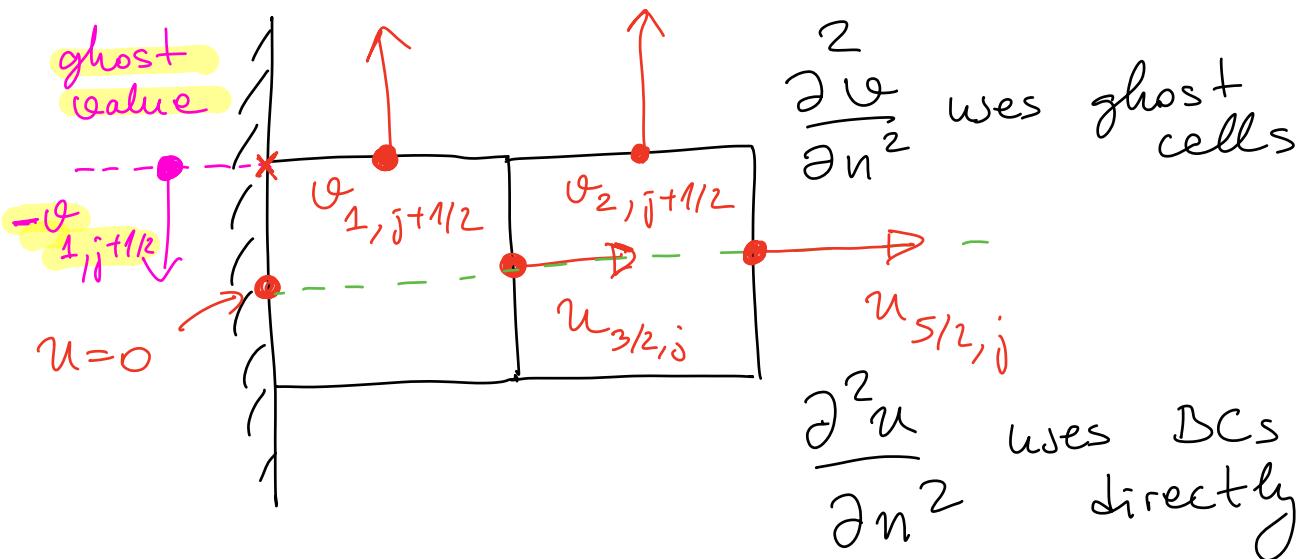
$$\frac{\partial P}{\partial n} = 0 \quad \text{at impermeable boundary}$$

But this is misleading! Viscosity couples pressure and velocity at boundaries. The MAC grid shows that no BCs are needed for the pressure at no-slip walls.

For the discrete viscous

Laplacian, we can reuse the things we did in 1D for advection-diffusion.

The idea is that perpendicular velocity is on a finite-difference (non-staggered) grid, but parallel velocity is on a finite-volume (staggered) grid.



This is done in the same way as if the  $x$  direction were the only dimension.

Corners are not that hard but they can lead to singularities (driven lid cavity test)

Stress-based BCs can also be handled but they are less natural than no slip - see article by Boyce Griffith linked on course webpage

## Adection

First the "easy" case. Often the velocity adects other quantities like concentration or temperature, assumed here to be scalar (even if multi-component) fields:

$$\partial_t c + \mathbf{u} \cdot \nabla c = D \nabla^2 c$$

This is exactly the finite volume setup we studied in 1D for advection-diffusion equations:  
for a face-centered velocity  $\mathbf{u}$  is a face-centered and  $c$  is a cell-centered conserved quantity  
One can generalize 2nd order Godunov methods to 2D / 3D (good final project) (17)

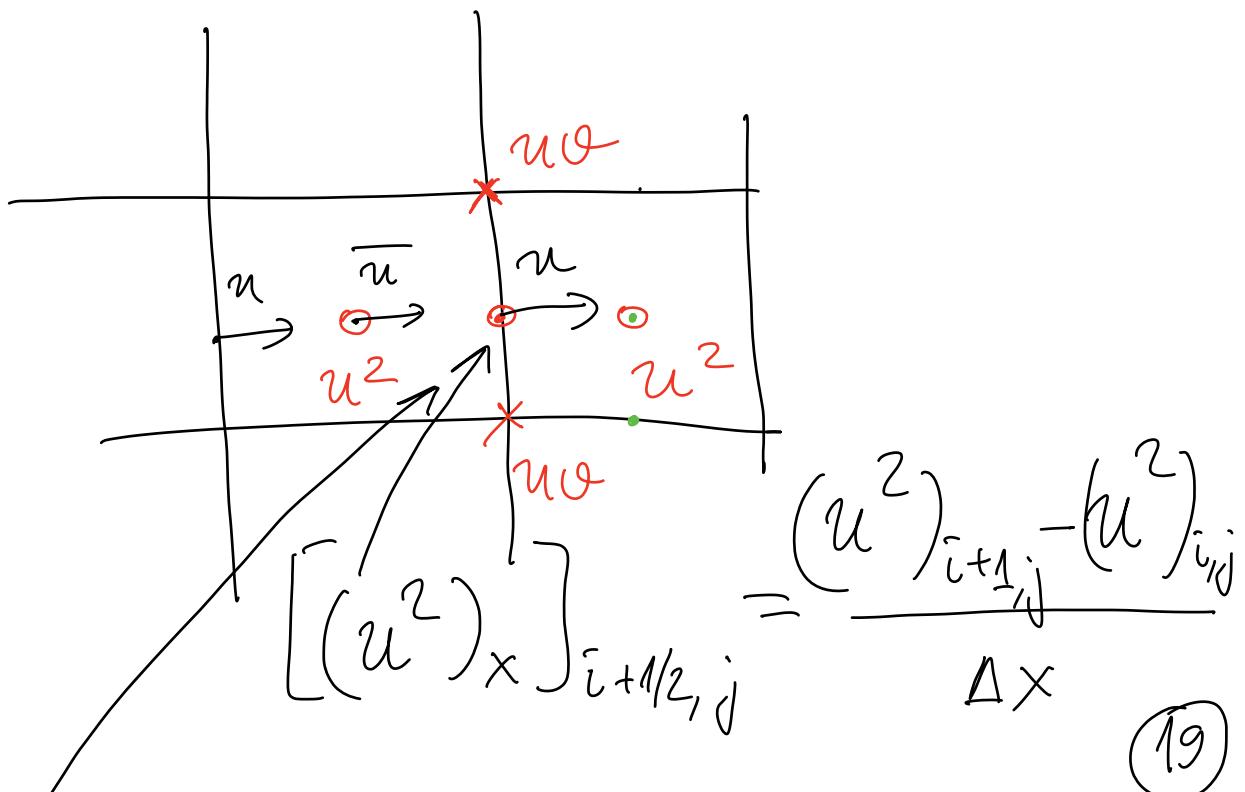
But what about Q.D.Q ?

One cannot do the extrapolation to midpoint in time for this term due to the staggered grid (it has been done for cell-centered velocities but this has other issues like BCs). So we will do MOL, with second order in space finite-difference discretization.

This is not great for high Reynolds number flows, but nothing really is that I know of. So lets do FD MOL 2<sup>nd</sup> order:

$$u \cdot \nabla u = \begin{bmatrix} (u^2)_x + (uv)_y \\ - - - \\ (uv)_x + (v^2)_y \end{bmatrix} \begin{matrix} \leftarrow u \\ \leftarrow v \end{matrix}$$

Idea: Place  $u^2$  and  $v^2$  on cell centers and  $uv$  on grid corners (In 3D, there are both corners/nodes & edges)



$$\left[ (uv)_y \right]_{\bar{i}+1/2, j} = \frac{(uv)_{\bar{i}+1/2, j+1/2}^{(y)} - (uv)_{\bar{i}+1/2, j-1/2}^{(y)}}{\Delta y}$$

To improve stability, we upwind to define  $u^2$ ,  $v^2$ , and  $uv$ :

$$[u^2]_{\bar{i}, j} = \overline{u}_{\bar{i}, j} \cdot u_{\underbrace{\bar{i}-1/2, j}_{\text{upwind}}}$$

If  $\overline{u}_{\bar{i}, j} \geq 0$

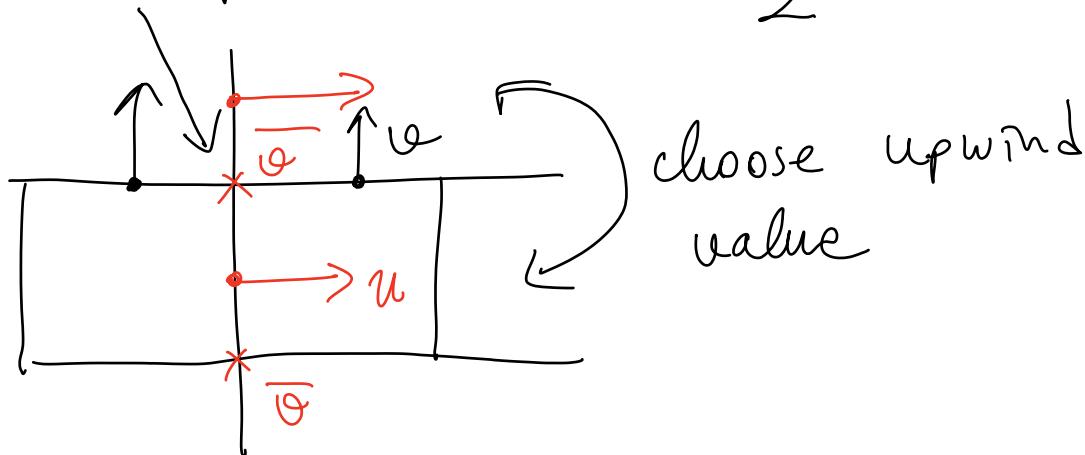
$$\overline{u}_{\bar{i}, j} = \frac{u_{\bar{i}+1/2, j} + u_{\bar{i}-1/2, j}}{2}$$

The notion here is to upwind in the direction of the derivative:

$$(uv)_{i+1/2, j+1/2}^y = \bar{\vartheta}_{i+1/2, j+1/2} \cdot u_{\underbrace{i+1/2, j}_{\text{upwind}}}$$

if  $\bar{\vartheta}_{i+1/2, j+1/2} \geq 0$

$$\bar{\vartheta}_{i+1/2, j+1/2} = \frac{\vartheta_{i+1, j+1/2} + \vartheta_{i, j+1/2}}{2}$$



For low Re numbers one may prefer centered discretizations, which are non dissipative but are dispersive.

One can combine in code:

$$(u^2)_{i,j} = \frac{u_{i+1/2,j} + u_{i-1/2,j}}{2} \underbrace{\left( \frac{1-\gamma}{2} u_{i+1/2,j} + \frac{1+\gamma}{2} u_{i-1/2,j} \right)}_{\bar{u}_{i,j}}$$

Choose + sign if  $\bar{u}_{i,j} \geq 0$

$\gamma=0$  for centered

$\gamma=1$  for upwinding

Adaptive  $\gamma$  would be a  
"poor person's" limiter.

There are better ways but  
none are very natural on  
the staggered grid.

## Time Stepping

The last piece is how to integrate in time. The standard / old approach has been to use projection methods, but I suggest a better alternative that is equally efficient (I learned from Boyce Griffith).

We want:

$$\left\{ \begin{array}{l} \frac{\mathbf{u}^{n+1} - \mathbf{u}^n}{\Delta t} + \nabla p^{n+1/2} = -[\mathbf{g} \cdot \nabla \mathbf{u}] \\ \text{Crank-Nicolson} \rightarrow + \gamma \nabla^2 \left( \frac{\mathbf{u}^n + \mathbf{u}^{n+1}}{2} \right) \\ \nabla \cdot \mathbf{u}^{n+1} = 0 \end{array} \right.$$

This is a so-called saddle-point linear system:

$$\begin{bmatrix} \frac{\rho}{\Delta t} - \frac{\eta}{2} L & | & G \\ \hline -D & | & \emptyset \end{bmatrix} \begin{bmatrix} u^{n+1} \\ P^{n+1/2} \end{bmatrix} = \begin{bmatrix} \left( \frac{\rho}{\Delta t} + \frac{\eta}{2} L_v \right) u^n \\ - [\rho u \cdot \nabla u]^{n+1/2} \\ \hline \emptyset \end{bmatrix}$$

↑  
Saddle-point

For Stokes flow,  $\text{Re} \rightarrow 0$ ,  
use L-stable backward Euler:

$$\begin{bmatrix} -\eta L & | & G \\ \hline -D & | & \emptyset \end{bmatrix} \begin{bmatrix} u^{n+1} \\ P^{n+1} \end{bmatrix} = \begin{bmatrix} \text{rhs} \\ \emptyset \end{bmatrix}$$

How do we solve this  
saddle-point linear system?  
(Harder than solving a Poisson  
equation for pressure but similar)

Use GMRES with a smart  
preconditioner (see webpage)  
(MULTIGRID method, more on  
this once we cover FEM)

One can prove (mostly in FEM  
literature) that the # of  
GMRES iterations is roughly  
independent of grid size if  
the preconditioner is chosen  
smartly.

For high Re numbers one can approximate the GMRES solution to second-order accuracy using a Projection method

① Time lag pressure and solve

$$\left\{ \begin{array}{l} \textcircled{1} \text{ Time lag pressure for } u^{n+1} \text{ only :} \\ \frac{u^{n+1,*} - u^n}{\Delta t} + G_p = \eta L_0 \left( \frac{u^n + u^{n+1,*}}{2} \right) \\ - [g u \cdot \nabla u]^{n+1/2} \\ u_{\partial \Omega}^{n+1,*} = 0 \quad (\text{no slip}) \end{array} \right.$$

This is just advection-diffusion for velocity

② Solve Poisson equation for

pressure:

$$L_c \psi^{n+1} = (DG) \psi^{n+1} = \frac{Du^{n+1,*}}{\Delta t}$$

Recall: No BCs required on  
staggered grid

③ Project velocity onto space of  
discretely divergence-free fields:

$$u^{n+1} = u^* - \Delta t G \psi^{n+1}$$

④ Correct pressure (hey!)

$$p^{n+1/2} = p^{n-1/2} + \psi^{n+1} - \frac{u \Delta t}{2G} \left( L_c \psi^{n+1} \right)$$

(27)

## Notes :

1) For periodic domain, this method is equivalent to the full GMRES solve approach

2) For non-periodic domains the projection method is a great preconditioner for the GMRES solver (use inexact sub-solvers), see paper by Boyce Griffith.

Only missing piece is how to get 2<sup>nd</sup> order (midpoint) estimate for  $[\bar{u}, \bar{v}]^{n+1/2}$ .

Let's show one approach on a relevant problem.

Boussinesq approximation:

$$\left\{ \begin{array}{l} \rho \partial_t u + \nabla \Pi + g u \cdot \nabla u = \\ \eta \nabla^2 u + \vec{g}^c \\ \text{gravity / buoyancy} \end{array} \right.$$
$$\partial_t c + u \cdot \nabla c = D \nabla^2 c$$

concentration / temperature / salinity

We have two options for the advection scalar  $c$ :

- 1) Discretize  $u \cdot \nabla c$  at a specific point in time, as in MOL schemes (e.g. something like third order upwind biased in 2d)
- 2) Solve the advection-diffusion equation using a second-order high-resolution Godunov method like Fromm + limiters + diffusion as you did in homework

Notation:

$\nabla \cdot F^{(\text{adve})} = \text{SpaceTime}(c^n, u, s, \Delta t)$   
computes advection fluxes over  
one timestep of duration  $\Delta t$ :

$$\partial_t c + u \cdot \nabla c \xrightarrow{\text{source}} S$$

kept fixed  
in time

$$\underbrace{[u \cdot \nabla c]}_{\text{accurate to } O(\Delta t)} \xrightarrow{n+1/2} \nabla \cdot F^{(\text{adve})}$$

Recall that this works even  
if  $D=0$  and is only limited  
in stability by advection CFL.  
By contrast, an RK2 MOL  
approach is unstable without  
sufficient diffusion.

## Algorithm : Boussinesq predictor-corrector

1. Solve using GMRES for  
 $u^{n+1,*}$  and  $p^{n+1/2,*}$  (predictor) :

$$\left\{ \begin{array}{l} S \frac{u^{n+1,*} - u^n}{\Delta t} + D p^{n+1/2,*} = -S[u \cdot \nabla u]^n \\ + \gamma D^2 \left( u^n + u^{n+1,*} \right) + c \vec{g} \\ D \cdot u^{n+1,*} = 0 \end{array} \right.$$

and define  
 $u^{n+1/2,*} = \frac{u^n + u^{n+1,*}}{2}$

② Update concentration

$$\frac{c^{n+1} - c^n}{\Delta t} = D D^2 \left( c^n + c^{n+1} \right)$$

$$- \left\{ \begin{array}{l} [u^{n+1/2,*} \cdot \nabla c]^n \text{ (mol approach)} \\ \text{Space Time } (c^n, u^{n+1/2,*}, D D^2 c^n, \Delta t) \end{array} \right.$$

(31)

③ Solve using GMRES for  
 $u^{n+1}$  and  $p^{n+1/2}$  (corrector)

$$\left\{ \begin{array}{l} \frac{g^{n+1} - g^n}{\Delta t} + D p^{n+1/2} = -g[u \cdot Du] \\ + \eta D^2 \left( u^n + \frac{u^{n+1}}{2} \right) + \left( \frac{C^n + C^{n+1}}{2} \right) g \\ D \cdot u^{n+1} = 0 \end{array} \right.$$

This is a one-step RK2 type method but requires two Stokes solves per step, which is the most expensive step by far. We can do cheaper by using Adams-Basforth for the  $u \cdot Du$  term:

(32)

# Algorithm Boussinesq AB2

① Predict concentration:

$$\frac{c^{n+1,*} - c^n}{\Delta t} = D \nabla^2 \left( \frac{c^n + c^{n+1,*}}{2} \right)$$

$$- \left\{ \begin{array}{l} [u^n \cdot \nabla c]^n \text{ (MOL approach)} \\ \text{SpaceTime } (c^n, u^n, D \nabla^2 c^n, \Delta t) \end{array} \right.$$

② Solve using GMRES

$$\left\{ \begin{array}{l} \frac{s^n - n}{\Delta t} + D p^{n+1/2} = \left( \frac{c^n + c^{n+1,*}}{2} \right) g \\ + \gamma \nabla^2 \left( u^n + \frac{n+1}{2} \right) + \\ \left( \frac{3}{2} [u \cdot \nabla u]^n - \frac{1}{2} [u \cdot \nabla u]^{n-1} \right) \end{array} \right. \quad AB2$$

(33)

③ Define  $u^{n+1/2} = \frac{u^n + u^{n+1}}{2}$  and  
correct concentration:

$$\frac{c^{n+1} - c^n}{\Delta t} = D \nabla^2 \left( \frac{c^n + c^{n+1}}{2} \right)$$

$$- \left\{ \begin{array}{l} \left[ u^{n+1/2} \cdot \nabla c \right]^n \text{ (MOL approach)} \\ \text{SpaceTime } (c^n, u^{n+1/2}, D \nabla^2 c^n, \Delta t) \end{array} \right.$$