

SCIENCE & TECHNOLOGY

SATELLITE LAUNCH VEHICLES OF INDIA





SATELLITE LAUNCH VEHICLES OF INDIA

- ✓ Active operational launch vehicles:
 - Geosynchronous Satellite Launch Vehicle (GSLV)
 - o Geosynchronous Satellite Launch Vehicle Mk-III (LVM3)
 - o Polar Satellite Launch Vehicle (PSLV)
 - Sounding Rockets
- ✓ Launchers under development:
 - o Human Rated Launch Vehicle (HRLV)
 - o Small Satellite Launch Vehicle (SSLV)
 - o Reusable Launch Vehicle Technology Demonstrator (RLV-TD)
 - Scramjet Engine TD
- ✓ Retired:
 - o Satellite Launch Vehicle-3 (SLV-3)
 - o Augmented Satellite Launch Vehicle (ASLV)

GEOSYNCHRONOUS SATELLITE LAUNCH VEHICLE MARK II (GSLV Mk II)

- ✓ **Fourth generation** launch vehicle
- ✓ Developed to launch communication satellites using cryogenic third stage
- ✓ Capability of placing up to 6000 kg in Low Earth Orbits and 2,250 kg in Geosynchronous Transfer Orbits
- ✓ Three stage vehicle:
 - First Stage:
 - Solid rocket motor is augmented by 4 liquid strap-ons.
 - Second Stage:
 - Uses a liquid rocket engine called the Vikas engine, developed by Liquid Propulsion Systems Centre
 - Third stage:
 - Uses the cryogenic engine developed by the Liquid Propulsion Systems Centre
- ✓ Primary payloads are INSAT class of communication satellites that operate from Geostationary orbits

CRYOGENIC ENGINE:

- ✓ In physics, cryogenics is the production and behaviour of materials at **very low temperatures**.
- ✓ Cryogenic engines use of Liquid Oxygen (LOX) and Liquid Hydrogen (LH2) as propellants, which liquefy at -183 °C and -253 °C respectively.
- ✓ The **United States, Russia, Japan, India, France and China** are the only countries that have operational cryogenic rocket engines.
- ✓ ISRO's **Cryogenic Upper Stage Project** (CUSP) envisaged the design and development of the indigenous Cryogenic Upper Stage.



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GEOSYNCHRONOUS SATELLITE LAUNCH VEHICLE MARK III (LVM3)

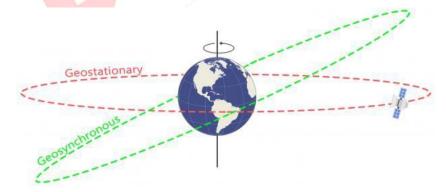
- Heavy lift launch vehicle of ISRO. Used for Chandrayaan 3.
- Capable of placing the **4 tonne class satellites** into Geosynchronous Transfer Orbits and 8000 kg in Low Earth Orbits of 600 km altitude
- ✓ Three stage vehicle:
 - o First Stage:
 - Solid Rocket Boosters:
 - Uses two solid rocket boosters to provide the huge amount of thrust required for lift off
 - Core Stage: Liquid Stage
 - Powered by two Vikas engines
 - Cryogenic Upper Stage:
 - Powered by CE-20, India's largest cryogenic engine





GEOSTATIONARY AND GEOSYNCHRONOUS ORBITS:

- When a satellite reaches about 36,000 kilometers from Earth's surface, it enters a "sweet spot" in which its orbit matches Earth's rotation. This high Earth orbit is called geosynchronous.
- A spacecraft in geosynchronous orbit appears to remain above Earth at a constant longitude.
- Geosynchronous satellites are useful for telecommunications and other remote sensing applications.
- ➤ Geostationary orbits fall in the same category as geosynchronous orbits, but the **orbit lies on** the same plane as the equator.
- Weather monitoring satellites use geostationary orbits because it gives a constant view of the same area.





POLAR AND SUN-SYNCHRONOUS ORBITS:

- > Satellites in polar orbits travel past Earth from north to south, passing roughly over Earth's poles.
- > Polar orbits are a type of **low Earth orbit**
- Sun-synchronous orbit (SSO) is a particular kind of polar orbit.
- Satellites in SSO are synchronised to always be in the same 'fixed' position relative to the Sun. Hence, the satellite always visits the same spot at the same local time.

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POLAR SATELLITE LAUNCH VEHICLE (PSLV): 'The Workhorse of ISRO'

- Third generation launch vehicle of India
- First Indian launch vehicle to be equipped with liquid stages
- 4 stage vehicle:
 - First Stage:
 - uses the S139 solid rocket motor and 6 solid strap-on boosters
 - **Second Stage:**
 - Uses a **liquid rocket engine** called the **Vikas engine**, developed by Liquid Propulsion Systems Centre
 - Third Stage:
 - Uses a **solid rocket motor**
 - Fourth Stage:
 - Uses two liquid engines
- Multiple satellite launch capability and multiple orbit capability
- Can take up to 1,750 kg of payload to Sun-Synchronous Polar Orbits of 600 km altitude. Used to launch various satellites into Geosynchronous and **Geostationary orbits**
 - Notable launches: Chandrayaan-1, Mars Orbiter Mission
- The XL version is launched with additional propellant in the strap-on solid rocket boosters, allowing for a larger payload capacity. The PSLV XL rocket was used for the launch of the Aditya L1 Mission.

HUMAN RATED LAUNCH VEHICLE (HRLV)

- **LVM3 rocket** is identified as the launch vehicle for **Gaganyaan mission**.
- ✓ All systems in LVM3 launch vehicle are re-configured to meet human rating requirements and christened **Human Rated LVM3**.
- ✓ HLVM3 will be capable of launching the **Orbital Module** to an intended **Low** Earth Orbit of 400 km.
- Orbital Module will **orbit the Earth** and is equipped with state-of-the-art avionics systems.
- Orbital Module comprises of:
 - o **Crew Module:** the habitable space with Earth like environment in space for the crew
 - Service Module: will be used for providing necessary support to CM while in orbit
- HLVM3 consists of Crew Escape System (CES) which ensures that Crew Module along with crew is taken to a safe distance in case of any emergency either at launch pad or during ascent phase.



🕰 GAGANYAAN:

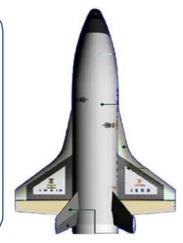
- India's maiden human space flight mission it envisages demonstration of human spaceflight capability by launching crew of 3 members to an orbit of 400 km for a 3 days mission and bring them back safely to earth, by landing in Indian sea waters.
- It is targeted to be launched in 2025 using an LVM3 Rocket.

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REUSABLE LAUNCH VEHICLE - TECHNOLOGY DEMONSTRATOR (RLV-TD)

- ✓ RLV-TD is an endeavour of ISRO towards developing essential technologies for a fully reusable launch vehicle to enable low-cost access to space
- ✓ Configuration is **similar to that of an aircraft** and combines the **complexity of both launch vehicles and aircraft**.
- ✓ The winged RLV-TD will be used to develop technologies like hypersonic flight (HEX), autonomous landing (LEX) and powered cruise flight.
- ✓ In the future, this vehicle will be scaled up to become the first stage of India's **reusable two-stage orbital (TSTO) launch vehicle**.



SMALL SATELLITE LAUNCH VEHICLE (SSLV)

- ✓ 3 stage Launch Vehicle configured with three Solid Propulsion Stages and liquid propulsion-based Velocity Trimming Module (VTM) as a terminal stage
- ✓ Payload Capability
 - o Single/Multi Satellites Nano, Micro and Mini satellites
 - o Single Satellite up to 500kg in 500km planar orbit
 - Three Multiple satellites ranging from 10kg to 300kg into
 500km Planar Orbit
- ✓ Features of SSLV:
 - Launch on demand feasibility
 - Low cost, with low turn-around time
 - Flexibility in accommodating multiple satellites
 - Minimal launch infrastructure requirements
- ✓ In Feb 2024, the PM laid the foundation stone for a **new spaceport at Kulasekarapattinam** in southern Tamil Nadu. It will focus on the **launch of SSLVs on a commercial basis for the private sector.** This spaceport will help save fuel for small rocket launches because it **can launch rockets directly south over the Indian Ocean without requiring crossing landmasses.**
- ✓ This is unlike the existing launch site at the Satish Dhawan Space Centre, which adds more fuel requirements for launching into a polar orbit as rockets need to follow a curved path to the south to avoid Sri Lanka's landmass.

SOUNDING ROCKETS

- ✓ One or two stage solid propellant rockets
- ✓ Used for probing the upper atmospheric regions and for space research
- ✓ Also serve as **affordable platforms to test prototypes** intended for use in launch vehicles and satellites
- ✓ Launch of the first sounding rocket 'Nike Apache' from Thumba in Kerala on 21 November 1963, marked the beginning of the Indian Space Programme.





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✓ Currently, three versions are offered as operational sounding rockets, which cover a payload range of 8-100 Kg and a range of 80-475 km.

SCRAMJET ENGINE – TECHNOLOGY DEMONSTRATOR

- ✓ **Air Breathing Propulsion System** has the potential to bring a significant shift in the launch vehicle design. The first experimental mission of ISRO's Scramjet Engine was successfully conducted on August 28, 2016.
- ✓ ISRO's Scramjet engine uses **Hydrogen as fuel** and **Oxygen from atmosphere as oxidiser.**
- ✓ **India is the fourth country** to demonstrate the flight testing of a Scramjet Engine.

RAMJET AND SCRAMJET:

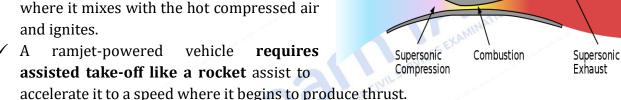
✓ Next generation launch vehicles are expected to have propulsion systems which can utilise the atmospheric oxygen during their flight.

Inlet body

Fuel injection

Nozzle

✓ A ramjet is a form of air-breathing jet engine that uses the vehicles forward motion to compress incoming air for combustion without a rotating compressor. Fuel is injected in the combustion chamber where it mixes with the hot compressed air and ignites.



- ✓ Ramjets work most efficiently at **supersonic speeds around Mach 3**.
- ✓ A Supersonic Combustion Ramjet, or Scramjet, is an improvement over the ramjet engine as it efficiently operates at hypersonic speeds (Mach \geq 5). A Scramjet engine performs the combustion itself at supersonic speeds (Mach \geq 1).

PSLV ORBITAL EXPERIMENTAL MODULE (POEM)

- ✓ POEM is an experimental mission which performs in-orbit scientific experiments during the fourth stage of the PSLV launch vehicle as an orbital platform.
- ✓ The PSLV is a four-stage rocket where the first three spent stages fall back into the ocean, and the final stage (PS4) after launching the satellite into orbit ends up as space junk.
- ✓ However, in POEM mission, the spent final stage will be utilised as a "stabilised platform" to perform experiments.
- ✓ POEM has a dedicated Navigation Guidance and Control (NGC) system for attitude stabilization, controlling the orientation of any aerospace vehicle within permitted limits.



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NEXT-GENERATION LAUNCH VEHICLE (NGLV)

- ✓ ISRO is developing a NGLV to **replace operational systems like the PSLV**.
- ✓ In NGLV, ISRO is looking at a cost-efficient, **three-stage to orbit**, **reusable heavy-lift vehicle** with a payload capability of ten tonnes to **Geostationary Transfer Orbit (GTO)**.
- ✓ It will feature **semi-cryogenic propulsion** (refined kerosene as fuel with liquid oxygen as oxidizer) for the booster stages. It will have 3 times the present payload capability with 1.5 times the cost of LVM3, resulting in **low-cost missions and modular propulsion systems**.
- ✓ This project will be **implemented with maximal participation from the Indian industry**, which is also expected to invest in the
 manufacturing capacity at the outset itself, thereby allowing a seamless
 transition to the operational phase subsequent to the development.
- ✓ NGLV will be demonstrated with three development flights (D1, D2 & D3) with a target of 96 months (8 years) for the completion of the development phase. It will enable national & commercial missions including launch of human spaceflight missions to the planned Bharatiya Antariksh Station.



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