# Indian Institute of Technology Kanpur AE 322: Aircraft Control Systems

### Even Semester: 2019-20

#### Assignments 3 and 4 (Lock-down assignments)

#### Due on canvas on 15/06/20

#### Answer all the questions

Q1. Sketch the asymptotic Bode plot of the following transfer functions:

(1) 
$$G(s) = \frac{1}{0.1s + 1}$$
,

(2) 
$$G(s) = \frac{1}{s+10}$$
,

(3) 
$$G(s) = \frac{(s+1)}{(0.1s+1)}$$

(4) 
$$G(s) = \frac{(0.1s+1)}{(s+1)}$$

(5) 
$$G(s) = \frac{1}{(s+1)(0.1s+1)}$$

15 Marks

Q2. See the attached code for attitude stabilization and tracking using a PD control. The objective is to track attitude trajectory of appropriate frequency by choosing suitable gains (kp and kd) (a sample code is given however the code is not tracking pitch attitude). Find the maximum frequency that can be tracked using suitable gains. Support your results with (Bode) plot analysis. **25 Marks** 

Q3 Explain the Dutch-roll mode.

5 Marks

Q4. Assume that an aircraft exchange kinematic and potential energy in the phugoid mode, derive the expression of natural frequency of this mode.

10 Marks

Q5. Consider an aircraft in wind tunnel which is restricted to move in x-y plane.

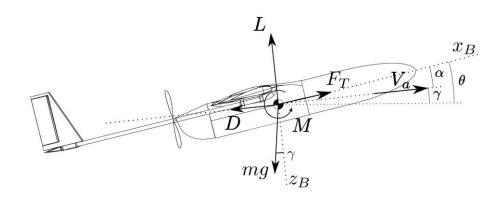
(i)	Derive the equations of motion	10 Marks
(ii)	Comment on static stability.	5 Marks
(iii)	Derive the expressions for damping and control derivatives.	10 Marks

Q6. Write short notes on any four

20 Marks

- (i) Dorsal fin
- (ii) Roll stability due to high and low wing configurations
- (iii) Significance of swept wing
- (iv) T-tail design
- (v) Attitude controller

## Q7. Consider motion of an aircraft in a longitudinal plane as shown below



- (I) Derive the equations of motion (Hint: find the rate of change in airspeed, flight path angle, pitch angle, and pitch rate, respectively). 10 Marks
- (II) Assuming that one can independently control airspeed using thrust force,  $F_T$ , design a controller to follow the given airspeed,  $V_r$ . 15 Marks