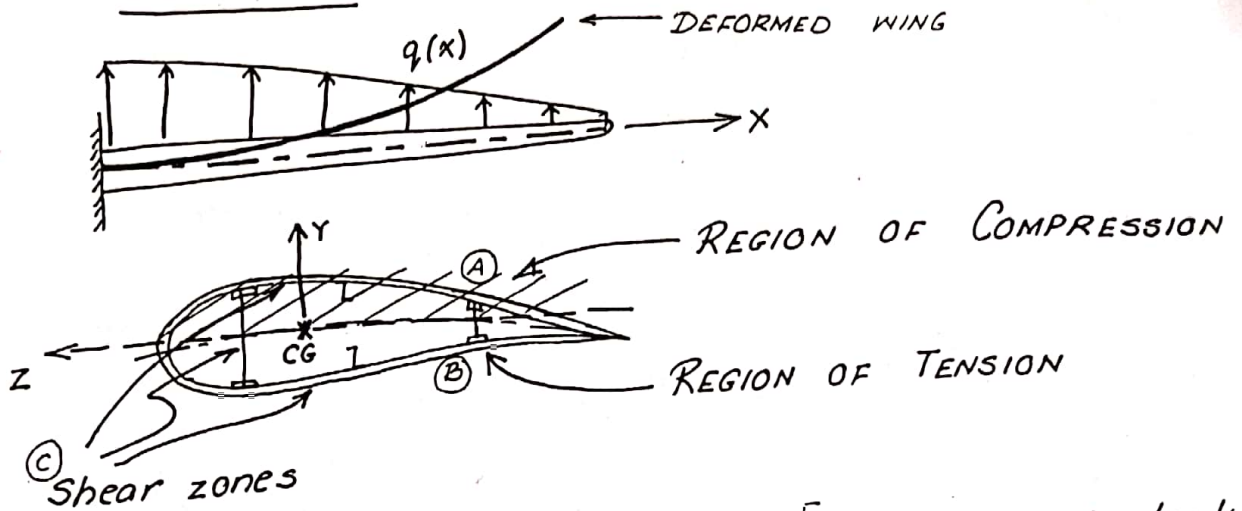


# BUCKLING



Region A : Compressive loads (direct)

Region B : Tensile loads (direct)

Region C : Shear load (dominant)

\* Regions A & C buckle or lose form at a value of stress  $\sigma_{rms} < \sigma_y$   
 equivalent  
 Mises stress

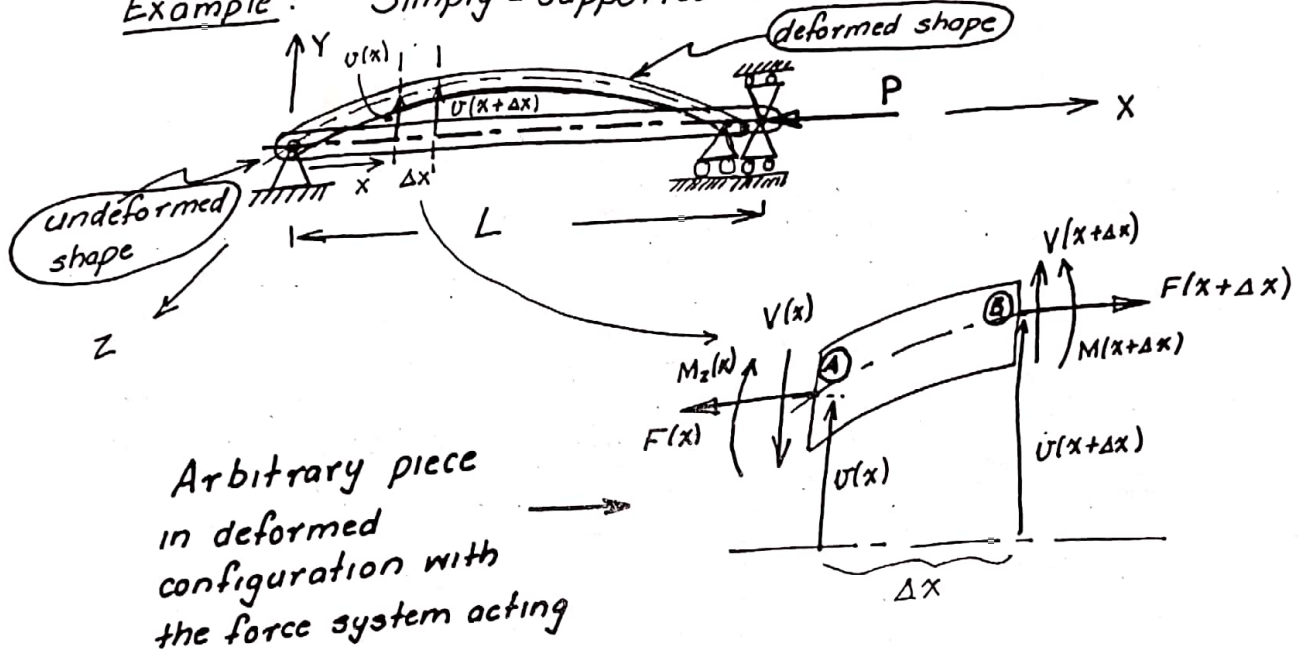
→ The deformation in transverse direction is LARGE, and hence cannot be treated as small deformation.

Q Infinitesimal assumption  $\Rightarrow$  equilibrium in undeformed coordinate system. How can axial load give rise to bending action?

Here will be our first excursion into finite elasticity

i.e. we look at equilibrium in the DEFORMED configuration.

Example: Simply-supported beam under axial load



\* We have assumed bending in the PLANE of the paper as the only possible alternate configuration

Now, we look at resultant force and moment balance here :

$$\sum F_x = 0 \Rightarrow F(x+\Delta x) - F(x) = 0 \Rightarrow \frac{dF(x)}{dx} = 0$$

$$\Rightarrow F(x) = \text{Constant} \Rightarrow F_0$$

$$\text{But } F(L) = -P < \text{from B.C.} > \Rightarrow \boxed{F(x) = -P} \text{ --- (a)}$$

$$\sum F_y = 0 \Rightarrow V_y(x+\Delta x) - V_y(x) = 0 \Rightarrow \boxed{\frac{dV_y}{dx} = 0} \text{ --- (b)}$$

\* Note that  $F(x) = EA \frac{du_0}{dx}$ , with  $u_0(x)$  is the stretch of centroidal line

$$\sum M_z|_A = 0$$

$$\Rightarrow M_z(x + \Delta x) - M_z(x) + V_y(x + \Delta x) \cdot \Delta x - F(x + \Delta x) \cdot (v(x + \Delta x) - v(x)) = 0$$

this is new, and arises due to the displacement

$$\Rightarrow \frac{dM_z}{dx} + V_y - F \cdot \frac{dv}{dx} = 0 \quad \text{--- (c1)}$$

$$\Rightarrow \frac{d^2 M_z}{dx^2} + \cancel{\frac{dV_y}{dx}} - \frac{d}{dx} \left( F \frac{dv}{dx} \right) = 0$$

$$\Rightarrow \boxed{\frac{d^2 M_z}{dx^2} + P \frac{d^2 v}{dx^2} = 0} \quad 0 < x < L \quad \text{--- (c)}$$

$$\text{From (c1)} \quad \boxed{V_y = - \left( \frac{dM_z}{dx} + P \frac{dv}{dx} \right)} \quad \text{--- (d)}$$

What are the boundary-conditions?

At  $x = 0$

$$u_0(0) = 0 \rightarrow \text{axial stretch}$$

$$v(0) = 0 \rightarrow \text{transverse displacement constrained}$$

$$M_z(0) = 0 \rightarrow \text{unconstrained rotation or zero } M_z$$

At  $x = L$

$$EA \frac{du_0}{dx} = -P \rightarrow \text{axial force specified}$$

$$v(L) = 0 \rightarrow \text{constrained transverse displacement}$$

$$M_z(L) = 0 \rightarrow \text{free rotation or zero } M_z$$

(e)

\* Note that (c) with boundary-conditions (e) is an EIGENVALUE problem, as neither  $P$  is known nor  $v(x)$  is known. We are looking for those values of  $P$ , i.e.  $P_i$ , for which the system of equations has a NON-TRIVIAL solution  $v_i(x)$ .

\* There are infinite distinct  $P_i$ 's possible and the corresponding  $v_i(x)$  eigenvalues  
mode shapes

How to get these?

Here, let  $v(x) = \sum_{i=1}^{\infty} a_i \sin\left(\frac{i\pi x}{L}\right)$

$$\Rightarrow v(0) = v(L) = 0$$

$$M_z(0) = M_z(L) = 0$$

$$\left| \begin{array}{l} M_z = EI_{zz} \frac{d^2 v}{dx^2} \\ (I_{yz} = 0) \leftarrow \text{ASSUME!} \end{array} \right.$$

$$\Rightarrow \frac{d^2}{dx^2} (EI_{zz} v'') + P v'' = 0$$

$$\Rightarrow +EI_{zz} \sum_{i=1}^{\infty} a_i \left(\frac{i\pi}{L}\right)^4 \sin\left(\frac{i\pi x}{L}\right) - P \sum_{i=1}^{\infty} a_i \left(\frac{i\pi}{L}\right)^2 \sin\left(\frac{i\pi x}{L}\right) = 0$$

$\Rightarrow$  As each of  $\sin \frac{i\pi x}{L}$  is independent, and  $a_i = 0$  gives the trivial solution,

$$a_i \left( EI_{zz} \left(\frac{i\pi}{L}\right)^4 - P \left(\frac{i\pi}{L}\right)^2 \right) = 0 \Rightarrow$$

$$P_i = \frac{EI_{zz} \left(\frac{i\pi}{L}\right)^4}{\left(\frac{i\pi}{L}\right)^2} = \frac{EI_{zz} \pi^2 i^2}{L^2}$$

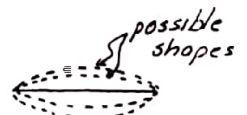
are the non-trivial, distinct loads at which buckling can happen.

$$P_{min} = \min_i (P_i) = P_1 = P_{CR} \leftarrow \text{CRITICAL BUCKLING LOAD}$$

$$\Rightarrow P_{CR} = \frac{\pi^2 EI_{zz}}{L^2}$$

$$v_{CR} = a_i \sin \pi x / L$$

unknown!  
can be anything



Here, we have solved the differential equation. We may not be so lucky always. Hence, we need a more generalized tool.

E.g. PRINCIPLE OF VIRTUAL WORK!

$$\int_0^L \left( \delta v_x \frac{d^2}{dx^2} (EI_{zz} v'') + P v'' \delta v \right) dx = \int_0^L (Q \delta v) dx \quad 0 < x < L$$

$$\Rightarrow \int_0^L \left( (EI_{zz} v'')'' \delta v + P v'' \delta v \right) dx = 0$$

$$\Rightarrow \int_0^L (EI_{zz} v'' \delta v'' - P v' \delta v') dx + (EI_{zz} v'')' \delta v \Big|_0^L + P v' \delta v \Big|_0^L - (EI_{zz} v'' \cdot \delta v') \Big|_0^L = 0$$

$$\Rightarrow \int_0^L EI_{zz} v'' \cdot \delta v'' dx = \int_0^L P v' \delta v' dx + \left( (EI_{zz} v'')' \delta v \right) \Big|_0^L - \left\{ \left( (EI_{zz} v'')' + P v' \right) \delta v \right\} \Big|_0^L$$

- V<sub>y</sub>

Now, at  $x=0 \rightarrow v=0 \Rightarrow \delta v=0$   
 $M_z = 0$  :

$x=L \rightarrow v=0 \Rightarrow \delta v=0$   
 $M_z = 0$

Now use  $v(x) = \sum_{i=1}^{\infty} a_i \sin \frac{i\pi x}{L} \Rightarrow \delta v(x) = \sum_{i=1}^{\infty} \delta a_i \sin \frac{i\pi x}{L}$

$$\Rightarrow \int_0^L EI_{zz} \sum_{i=1}^{\infty} a_i (i\pi/L)^2 \sin i\pi x/L \cdot \underbrace{(j\pi/L)^2 \sin j\pi x/L}_{\delta v = \sin j\pi x/L} dx = P \int_0^L \sum_{i=1}^{\infty} \frac{(i\pi/L) a_i \cos i\pi/L}{(j\pi/L) \cos j\pi/L} dx$$

$$\Rightarrow a_j \int_0^L EI_{zz} (j\pi/L)^4 \sin^2 j\pi x/L dx = P \cdot a_j \int_0^L (j\pi/L)^2 \cos^2 j\pi x/L dx$$

$\{a_j \neq 0\} \Rightarrow \left[ P_j = \frac{EI_{zz} j^2 \pi^2}{L^2} \right] \leftarrow \text{same as before!}$



In the generic case, with different boundary conditions buckling load will change. Note that buckling is a geometry and constraint dependent failure phenomenon.

$$\text{So, } P_{cr} = \frac{\pi^2 EI}{l_e^2} \leftarrow \text{effective length}$$

with  $(l_e) = \beta l \leftarrow$  where  $l$  is actual length.

B. C. Type

a) Both ends pinned  
( $v = M_2 = 0$ )

$\beta$

1



b) Both fixed  
( $v = v' = 0$ )

$1/2$



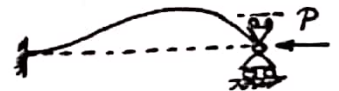
c) One fixed, other free  
 $v = v' = 0$        $M_2 = v_y = 0$

2



d) One fixed, other pinned  
 $v = v' = 0$        $v = M_2 = 0$

0.6998



Now, let us look at this problem from the point of view of virtual work.

$$\delta U \left( \frac{d^2}{dx^2} (EI_{zz} v'') + \underbrace{P v''}_{-\frac{d}{dx}(F(x)v')} \right) = (0) \times \delta U$$

$$\Rightarrow \int_0^L \{ (EI_{zz} v'')'' - (Fv')' \} \delta v dx = 0$$

$$\begin{aligned} \Rightarrow \int_0^L (EI_{zz} v'' \delta v'' + Fv' \delta v') dx &= \left\{ \underbrace{-(EI_{zz} v'')' + F}_{V_y} \right\} \delta v \Big|_0^L \\ &+ \underbrace{EI_{zz} v'' \delta v'}_{M_z} \Big|_0^L = V_y \cdot \delta v \Big|_0^L + M_z \cdot \delta v' \Big|_0^L \end{aligned}$$

In case no end-loads are applied:  $V_y = M_z = 0$  at free/pinned ends  
 $\delta v, \delta v' = 0$  at pinned/fixed ends

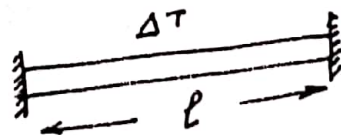
RHS = 0  $\Leftarrow$

$$\Rightarrow \boxed{\int_0^L EI_{zz} v'' \delta v'' dx = - \int_0^L F \cdot v' \delta v' dx} \quad \text{--- (a)}$$

For special case of end-load  $F(x) = -P$  (a)  
 becomes

$$\int_0^L EI_{zz} v'' \delta v'' dx = P \int_0^L v' \delta v' dx \quad \text{--- (b)}$$

For thermal loading of fixed column



$$\epsilon_{xx}^T = \alpha \Delta T \Rightarrow \epsilon_{xx} = \frac{1}{E} (\sigma_{xx}) + \epsilon_{xx}^T$$

$$\Rightarrow \text{For } \epsilon_{xx} = 0, \sigma_{xx} = -E \cdot \epsilon_{xx}^T = -E \cdot \alpha \Delta T$$

$$\Rightarrow F(x) = -EA \cdot \alpha \Delta T$$

$$\therefore E \alpha \Delta T_{cr} A = P_{cr} = \frac{4 \pi^2 E I_{zz}}{l^2}$$

$$\Rightarrow \Delta T_{cr} = \frac{4 \pi^2 I_{zz}}{\alpha \cdot l^2 \cdot A} \quad **$$

$$\alpha_{Al} \approx 22.2 \times 10^{-6} \text{ m/(m}^\circ\text{K)}$$

$\Rightarrow$  Buckling problem is:

$$\int_0^L E I_{zz} v'' \delta v'' dx = \Delta T \int_0^L E A \alpha v' \delta v' dx$$

Now we are looking for  $\Delta T_{cr} \rightarrow$  minimum eigenvalue of this problem!

Choice of basis functions:

For pinned-pinned:  $\frac{\sin i \pi x}{L}$ ;  $x(L-x)$ ,  $x(L-x)(\frac{1}{2}-x)$ ,  $x(L-x)(\frac{1}{2}-x)(\frac{1}{4}-x) \dots$

$$\Rightarrow v(x) = \sum_{i=1}^N \alpha_i \frac{\sin i \pi x}{L}$$

$$\delta v(x) = \sum_{i=1}^N \delta \alpha_i \frac{\sin i \pi x}{L}$$

For fixed-fixed: Use hermite polynomials  
e.g.  $x^2(L-x)^2$ ;  $x^2(L-x)^2(\frac{1}{2}-x)$ ;  $x^2(L-x)^2(\frac{1}{2}-x)(\frac{1}{4}-x)$

$$\alpha_{Al} = 22.2 \times 10^{-6} \text{ m/m}^\circ\text{K}$$

\*\* Let square section with  $a = b = 10 \text{ mm}$ ,  $L = 1 \text{ m} \Rightarrow I_{zz} = \frac{1}{12} \times 10^{-4} \times 10^{-12}$

$$\Rightarrow \Delta T_{cr} \approx \frac{\frac{40 \times 10^{-4}}{12 \times 22.2 \times 10^{-6}}}{\frac{1000}{66.6}} \approx \frac{1000}{2 \times 100} = \boxed{15^\circ\text{K}}$$



\* Example of Fixed - Fixed B.C.

Let 
$$v(x) \approx A \underbrace{x^2(L-x)^2}_{(L^2x^2 - 2Lx^3 + x^4)} \Rightarrow \delta v(x) \approx \delta A \cdot x^2(L-x)^2$$

$$\Rightarrow v' = A(2L^2x - 6Lx^2 + 4x^3)$$

$$v'' = A(2L^2 - 12Lx + 12x^2)$$

$$\Rightarrow \int_0^L EI_{zz} (2L^2 - 12Lx + 12x^2)^2 dx = P \int_0^L (2L^2x - 6Lx^2 + 4x^3)^2 dx$$

$$(4L^4x^2 - 24L^3x^3 + 36L^2x^4 - 48Lx^5 + 16x^6)$$

$$\Rightarrow EI_{zz} \left[ 4L^4x + \frac{144}{3}L^2x^3 + \frac{144}{5}x^5 - \frac{48}{2}L^3x^2 + \frac{48}{3}L^2x^3 - \frac{288}{4}Lx^4 \right]_0^L$$

$$EI_{zz} \left[ 4 + \frac{144}{5} + \frac{144}{5} - 24 + \frac{48}{3} - 72 \right] L^5$$

$$\frac{60 + 720 + 432 - 360 + 240 - 1080}{15}$$

$$= \frac{12}{15}$$

$$\Rightarrow P_{cr} \approx \frac{108}{2} \times \frac{12}{15} \frac{EI_{zz}}{L^2} \approx 42 \frac{EI_{zz}}{L^2} > \frac{4\pi^2 EI_{zz}}{L^2}$$

\* Note that the approximation will give a value higher than the actual, but nevertheless, is close.

$$v(x) \approx A(1 - \cos \frac{2\pi x}{L}) \Rightarrow v'(x) = A(\frac{2\pi}{L}) \sin \frac{2\pi x}{L};$$

$$v'' = A(\frac{2\pi}{L})^2 \cos \frac{2\pi x}{L} \Rightarrow EI_{zz} \int_0^L (\frac{2\pi}{L})^4 \cos^2 \frac{2\pi x}{L} dx = P \int_0^L (\frac{2\pi}{L})^2 \sin^2 \frac{2\pi x}{L} dx$$

$$\Rightarrow EI_{zz} \left( \frac{2\pi}{L} \right)^4 \cdot \frac{L}{2} = P \left( \frac{2\pi}{L} \right)^2 \cdot \frac{L}{2}$$

$$\Rightarrow P_{cr} \approx \frac{4\pi^2 EI_{zz}}{L^2} \leftarrow \text{EXACT! (WHY?)}$$