

# Life Extension of the P-3 Aircraft by Robotic Applications of LPB

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# OUTLINE



- **Statement of Problem**
- **LPB Application to Floor Beam**
- **LPB Application to Propeller Bore**
- **Summary/Conclusions**

# STATEMENT OF PROBLEM

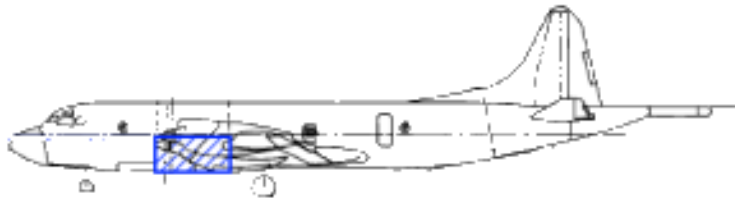
- **Fatigue cracking and SCC cause:**
  - reduced service life
  - increased O&M costs
  - reduced fleet readiness
- **Floor Beam Fatigue**
  - FSFT reveals fatigue failures in 7075-T6 Floor Beam
- **Propeller Bore SCC/Fatigue**
  - Heavy Shot peening and reaming done to mitigate SCC
  - Blades scrapped after 3 SP + re-machining cycles

## Goals:

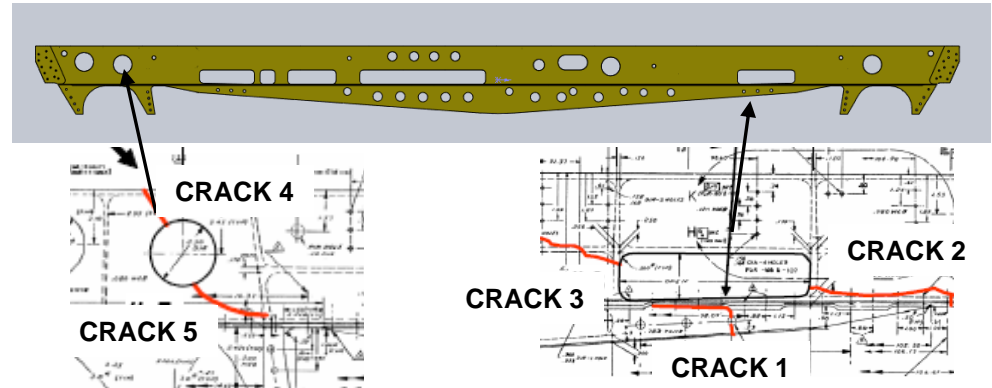
**Use LPB to enhance floor beam fatigue life**  
**Substitute LPB for SP in propeller bore**

# LPB Application to Floor Beam

# Statement of Problem

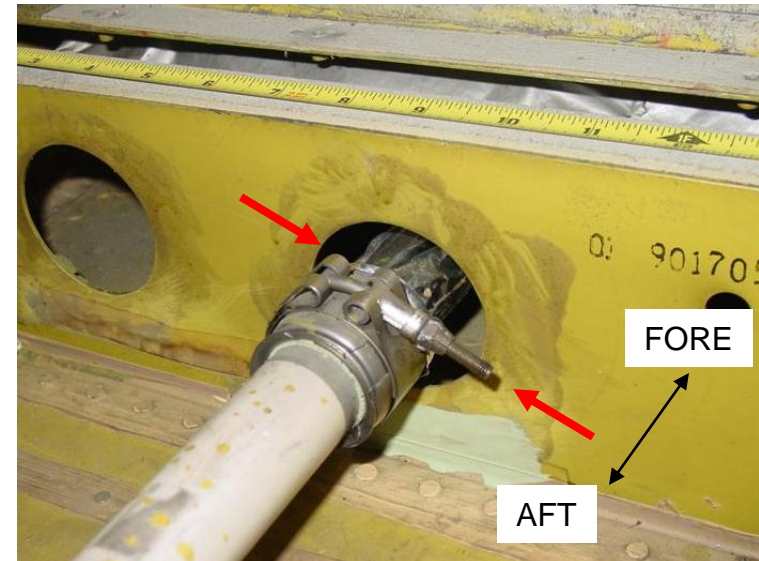
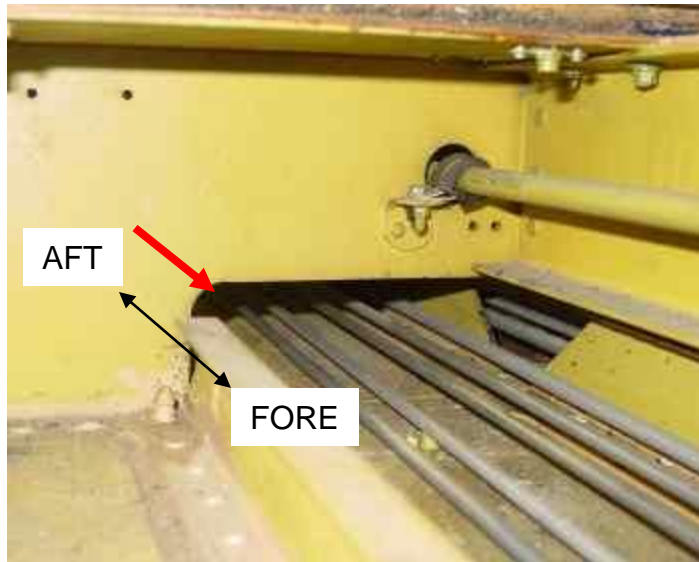


\* Images from LG04ER0112\_VolIII Report



- During FSFT, floor beams failed at combined 20,500 simulated flight hours and simulated fuselage pressure cycles
- Without proper repair fleet may not meet required service life
- Inspection can be proven costly
- In-flight failure could be catastrophic resulting in loss of aircraft and life

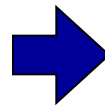
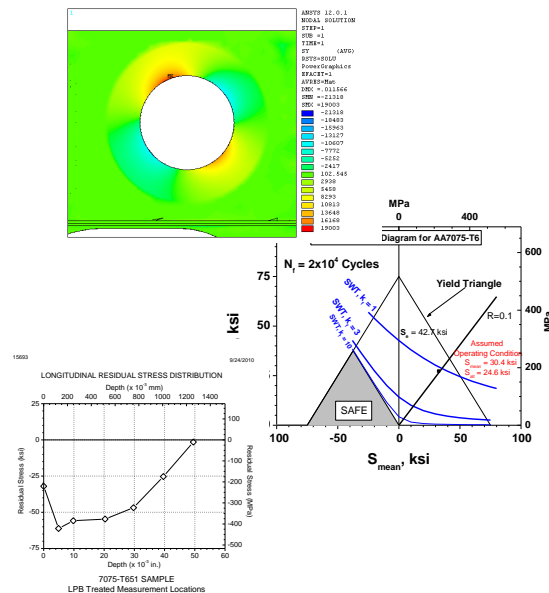
# Fatigue Crack-Prone Floor Beams



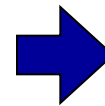
**Cut outs in the FS 515 floor beam prone to fatigue cracking**

# Proposed LPB Solution

- Turnkey robotic system for in-situ maintenance
- Treat the fatigue critical areas



**PHASE 2 - ROBOTIC  
LPB SYSTEM**

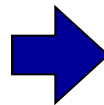
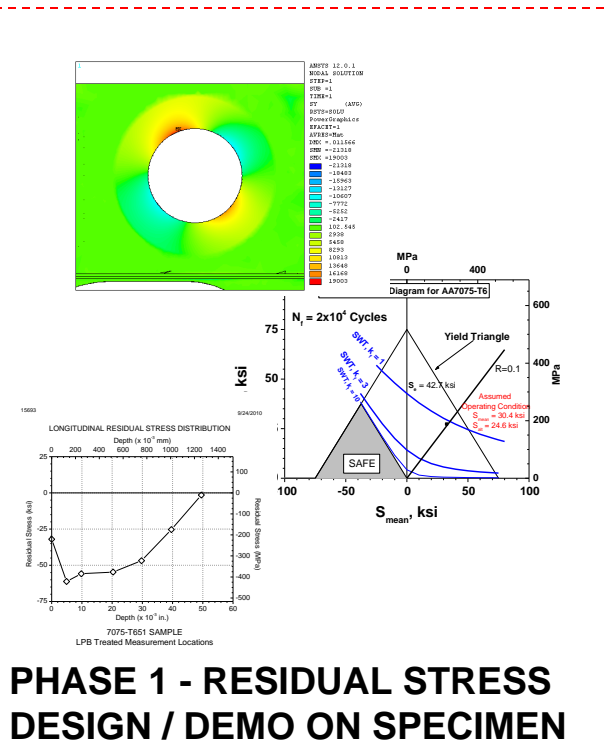


**PHASE 3 - INSTALLATION AT  
DEPOT**

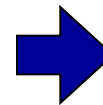


# Proposed LPB Solution

- Turnkey robotic system for in-situ maintenance
- Treat the fatigue critical areas



PHASE 2 - ROBOTIC LPB SYSTEM

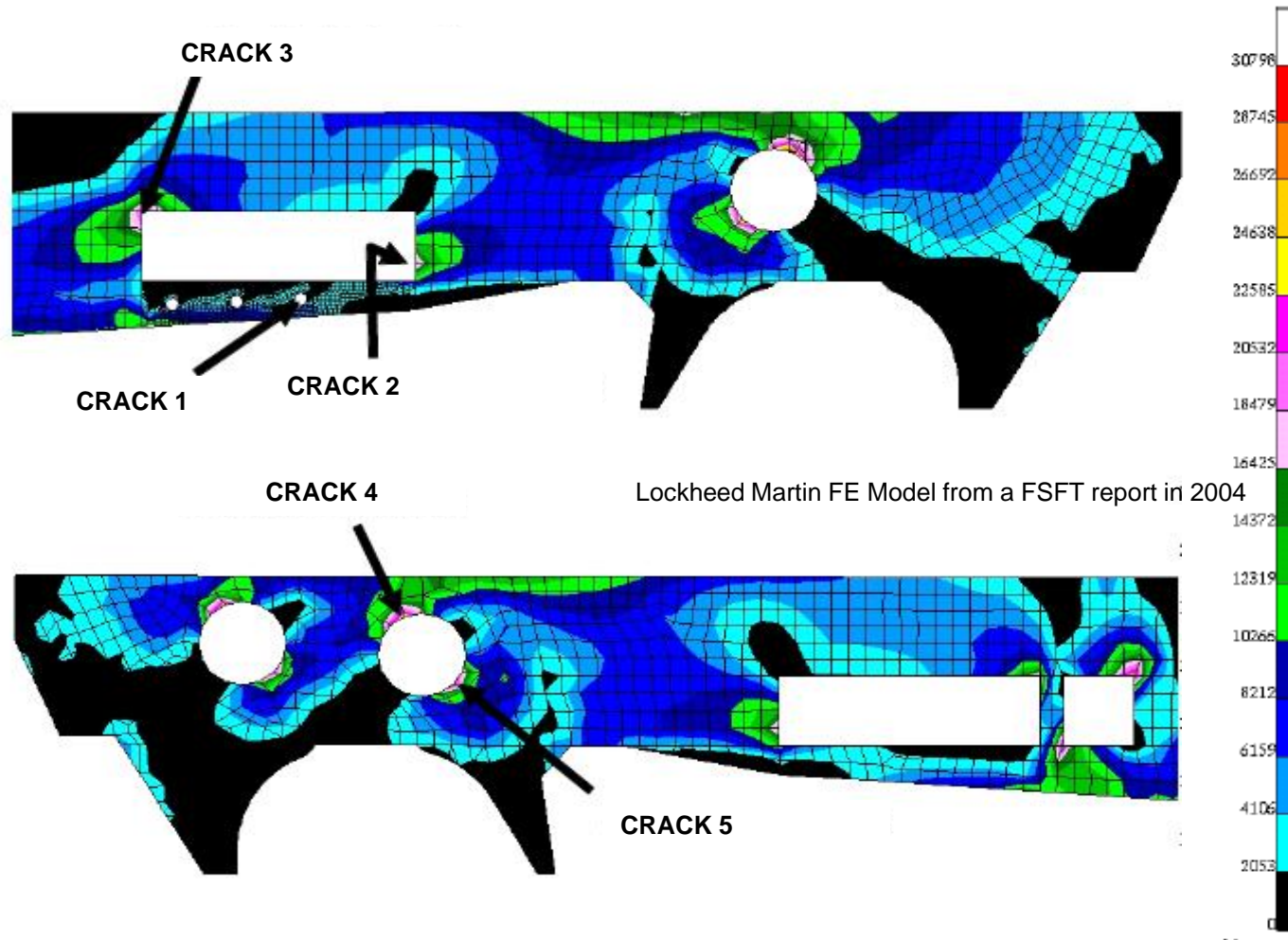


PHASE 3 - INSTALLATION AT DEPOT

At present, RS design and proof of concept on feature specimen

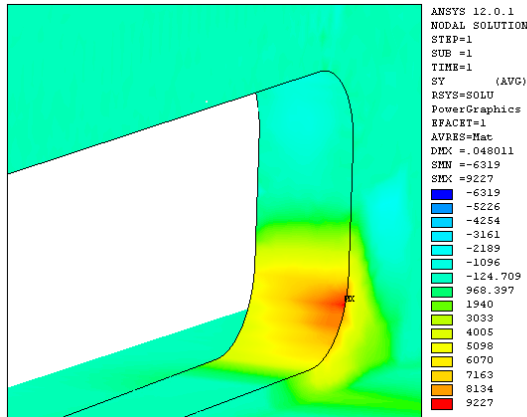


# Applied Stresses

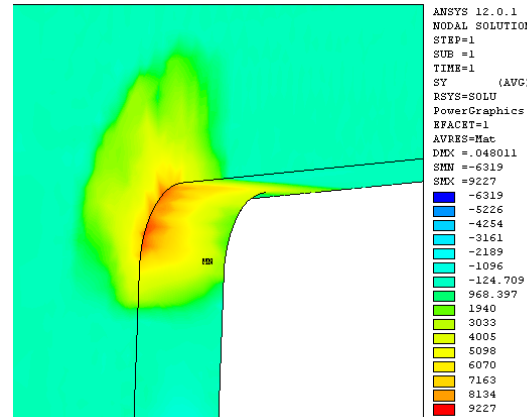


# Applied Stresses

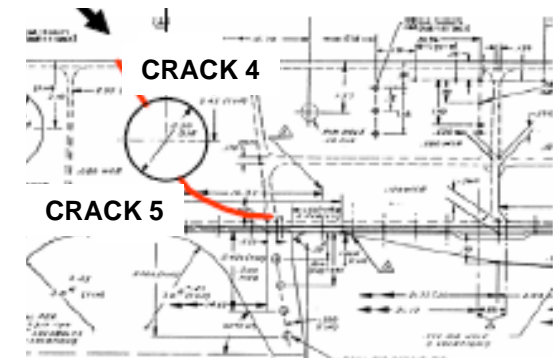
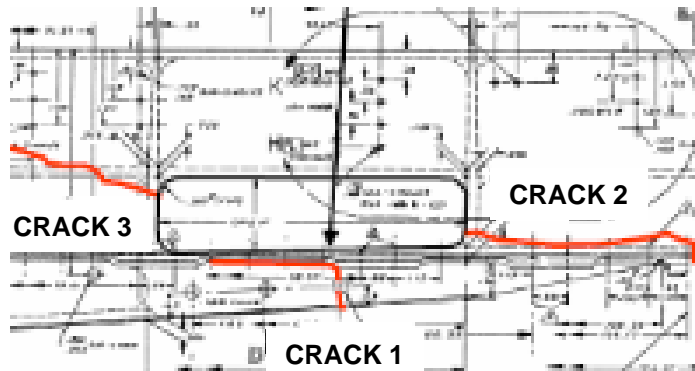
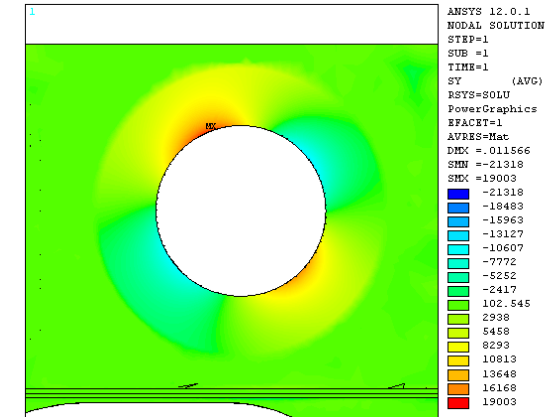
## Crack 2



## Crack 3

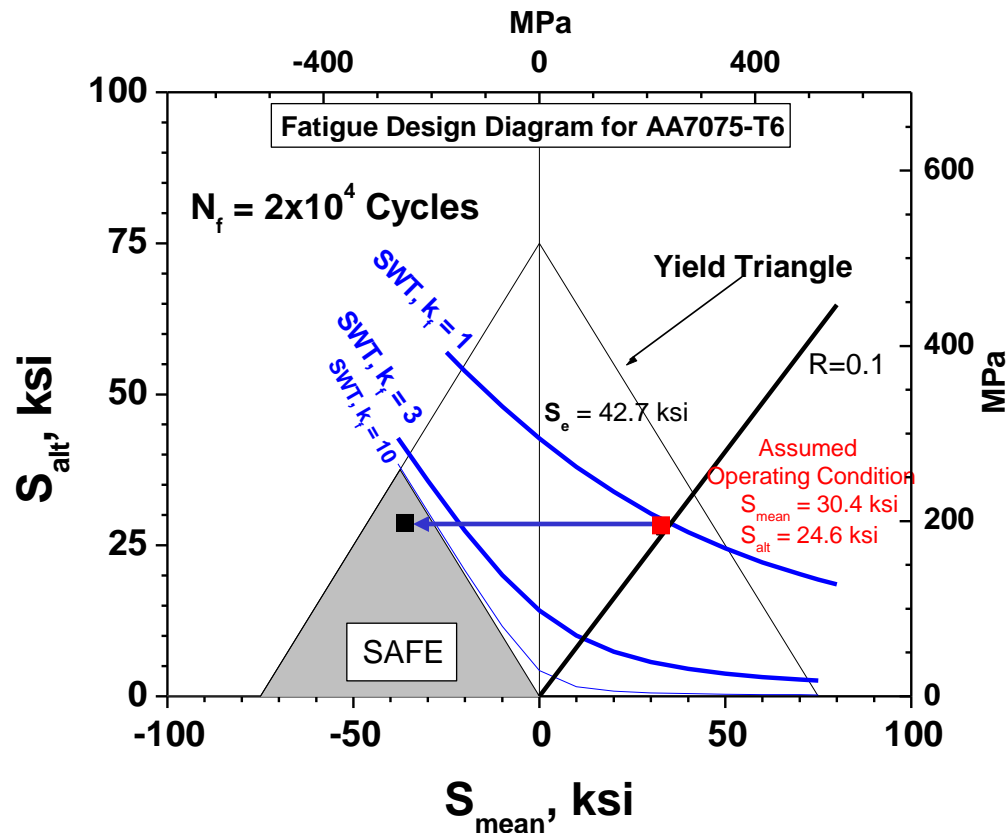


## Cracks 4 and 5



- Peak stress magnitude 45-55 ksi
- Model correlates well to the FSFT

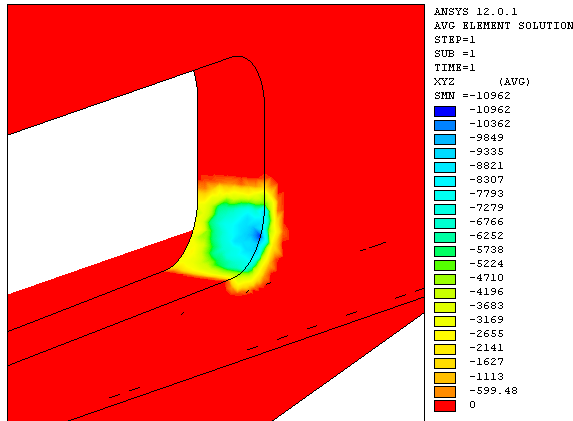
# Fatigue Design Diagram



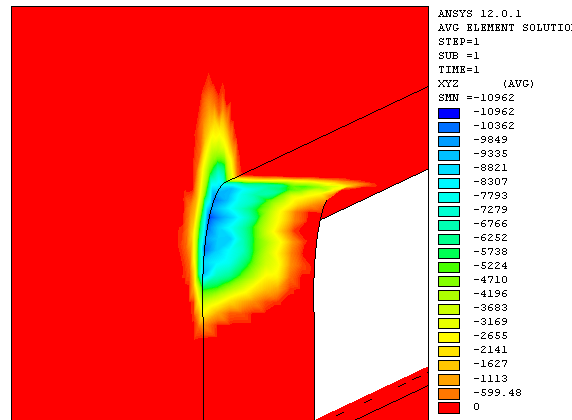
**Residual stress of -60 ksi will lead to SAFE zone.  
Fatigue life will be extended.**

# Residual Stress Goals

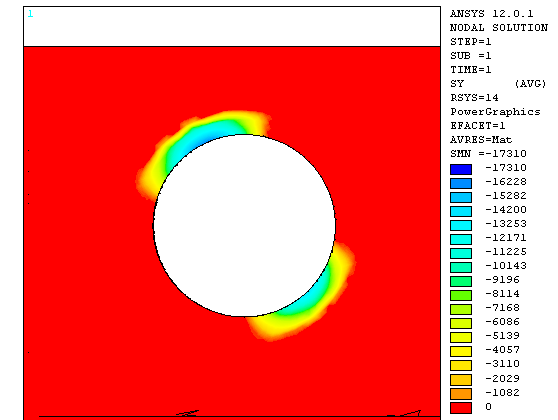
Crack 2



Crack 3



Cracks 4 and 5



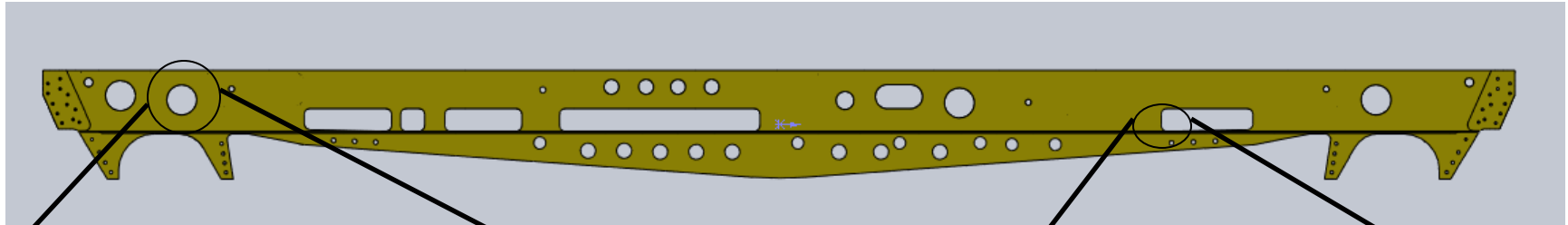
## Cracks 2 and 3

- Process inside face rectangular cutout radius
- Achieve -60 ksi residual stress in LPB zone

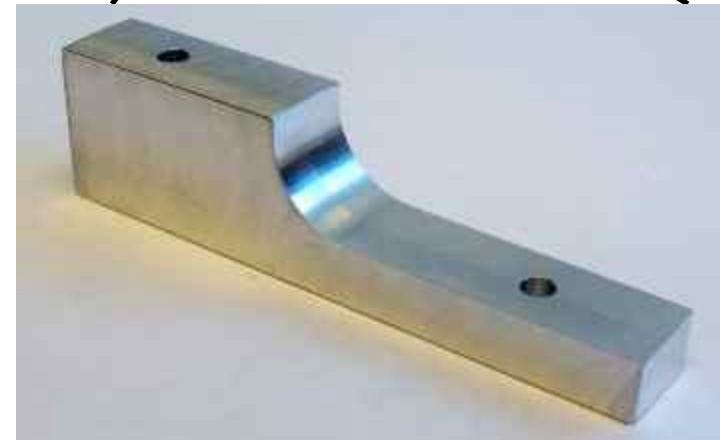
## Cracks 4 and 5

- Process front and back face of the 2.5"
- Achieve -60 ksi residual stress in LPB zone
- Achieve through-thickness compression

# Feature Specimen Design



Cracks 4 and 5



Cracks 2 and 3

**Simulate geometry of the fatigue critical areas**

# Application of LPB Process



**Process design completed**

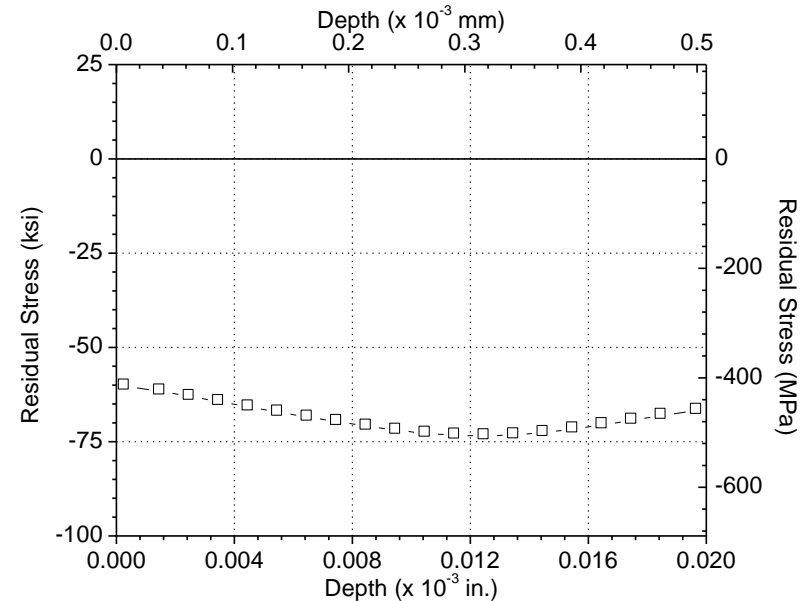
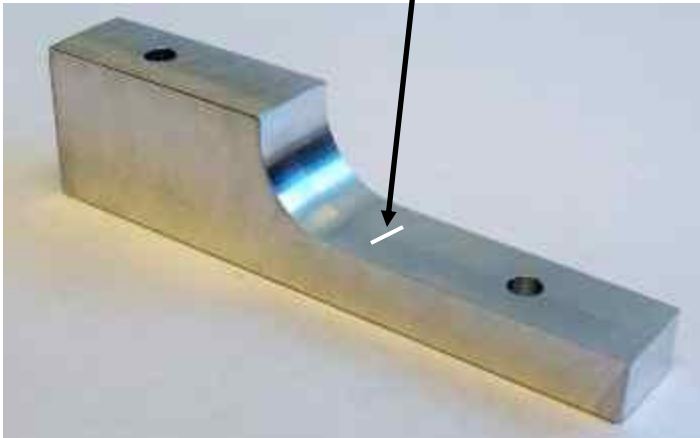
- **Tool design**
- **Process parameters**
- **Processing codes**



# RS Measurement

## Cracks 2 and 3 Feature Specimen

MEASUREMENT  
LOCATION



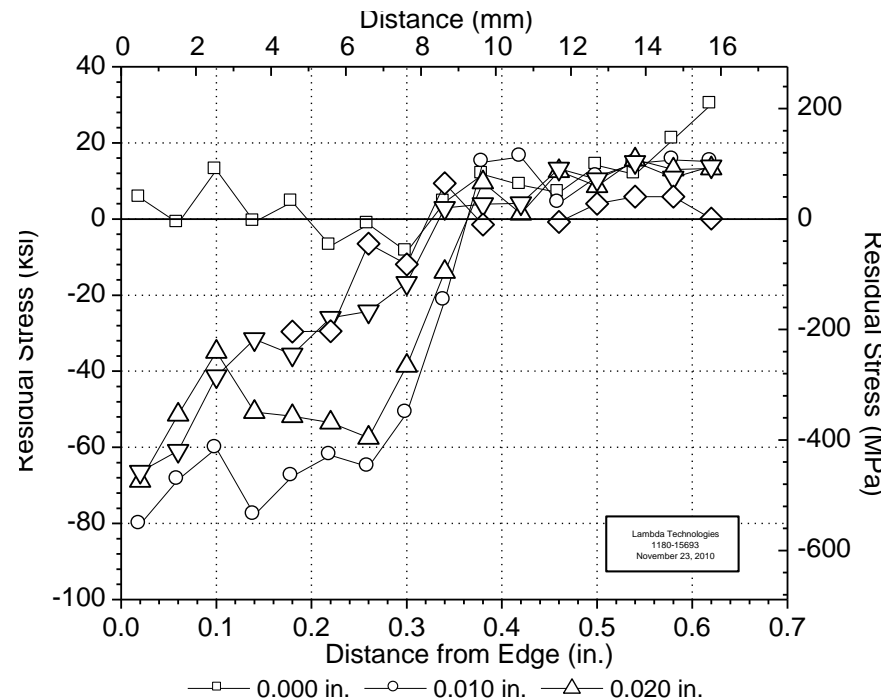
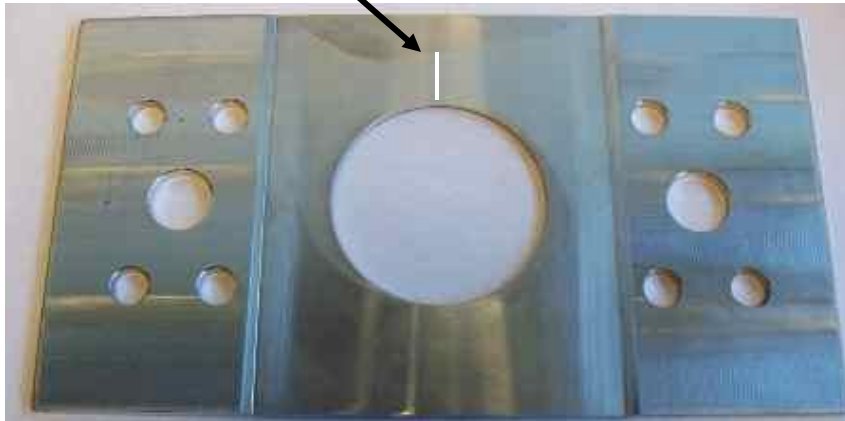
**Achieved -75 ksi at a depth of 0.010"**  
**Compared to the required -60 ksi**



# RS Measurement

## Crack 4-5 Feature Specimen

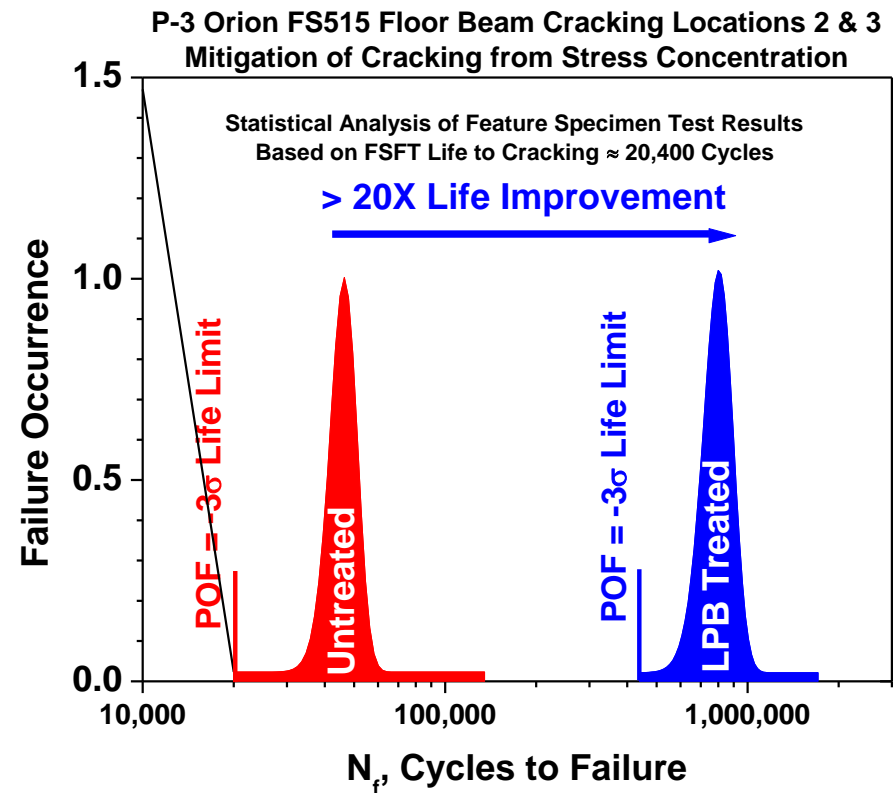
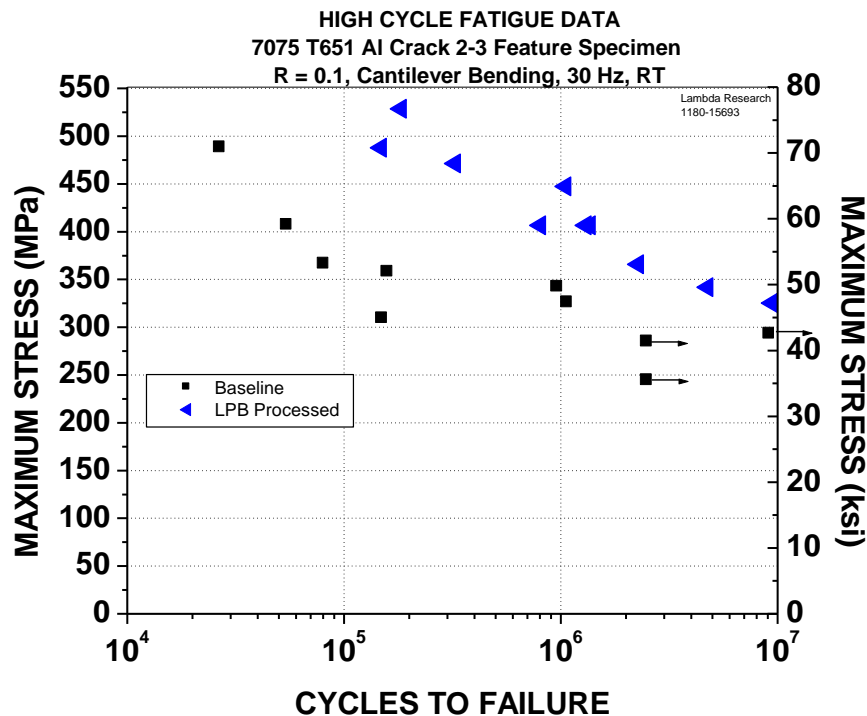
MEASUREMENT  
LOCATION



**Achieved -80 ksi at a depth of 0.010"**  
**Through-thickness compression**

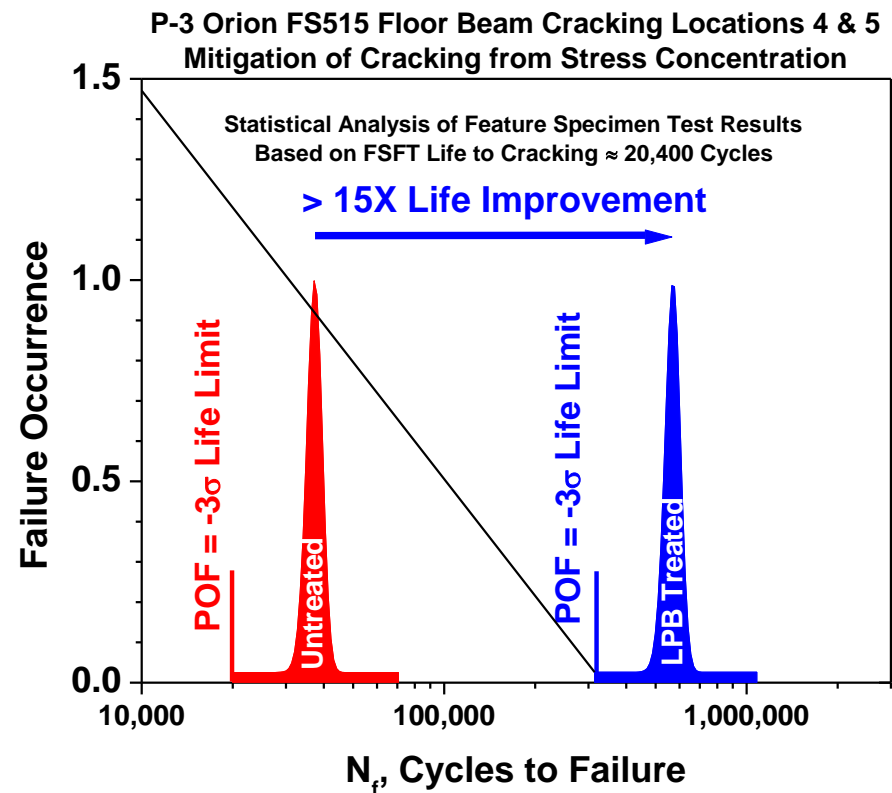
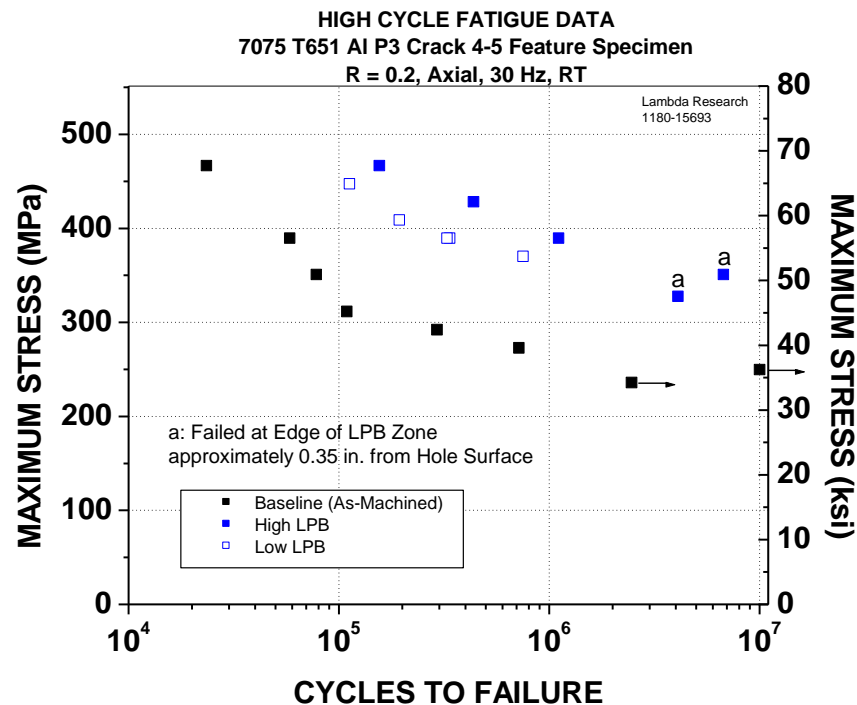
# Fatigue Performance Cracks 2 and 3

## 20x Fatigue Life Improvement



# Fatigue Performance Cracks 4 and 5

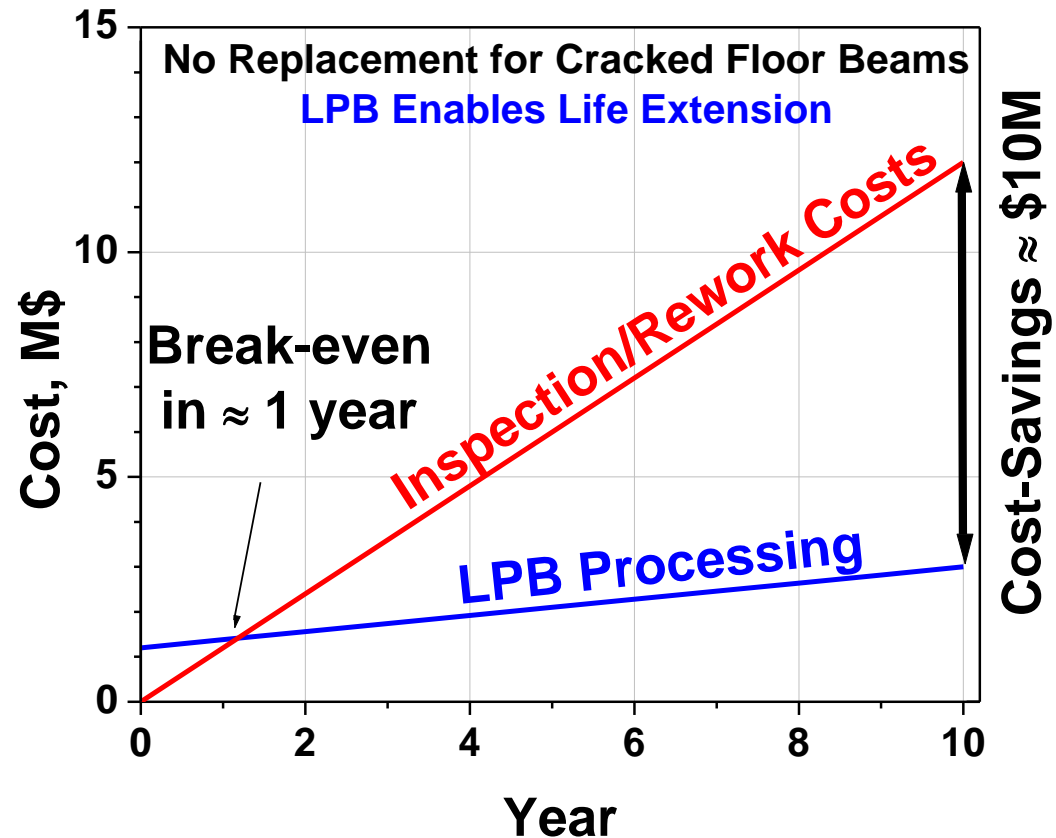
## 15x Fatigue Life Improvement



# Cost-Benefit Analysis

## ASSUMPTIONS:

- 150 aircraft in the fleet
- Maintenance Costs (estimate)
  - 4 inspections till end of life of a/c at \$20,000 per inspection
  - No replacement for cracked floor beams
- LPB Processing Costs (estimate)
  - 15 a/c to be LPB treated per year at \$12,000 per a/c
  - Initial Non-recurring Costs = \$1.2M



**Projected Cost Savings ≈ \$10M**

# Summary – Floor Beam

## LPB Processing

- 15-20X Fatigue Life Improvement
- Extended Service Life Would Reduce or Eliminate Need for Inspections
- Estimated **Savings of \$10M** on Maintenance

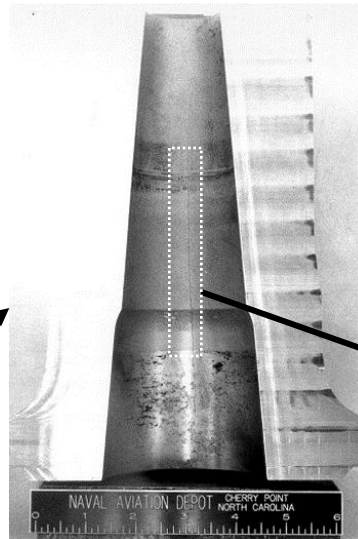
**LPB Program Phase I complete**  
**Ready for transition to production**

# LPB Application to Propeller Bore

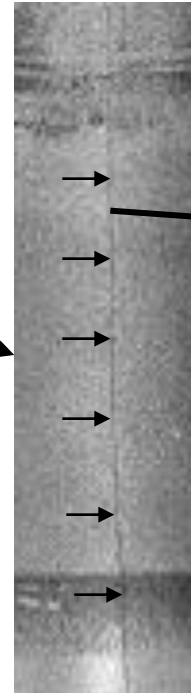
# SCC in P-3 Propeller Blade Bore



Propeller blade

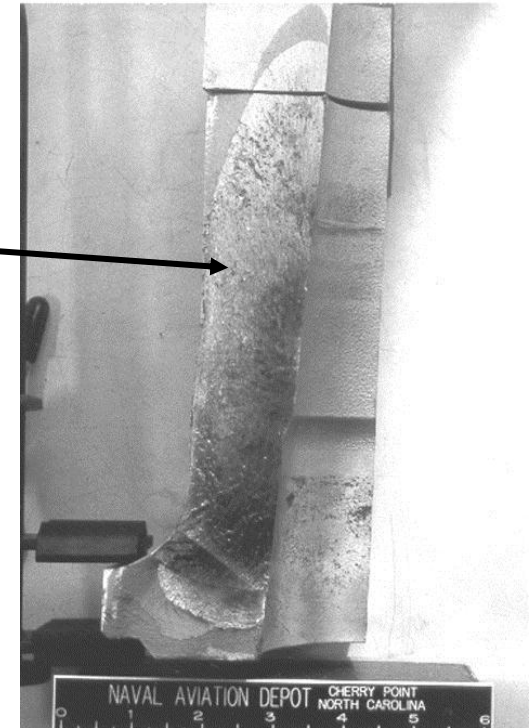


Cross sectional view of the bore prone to SCC



Close-up view of the crack (arrows)

Arrows Indicate the Fracture Face Shown on the Right



Macro fractograph of the fracture surface

## Current Repair/Overhaul Practice:

Repeated Heavy Shot peening +  
Reaming + Re-machining

**Blades Scrapped after 3 SP + Reaming Overhaul ~ \$35,000 per Blade**

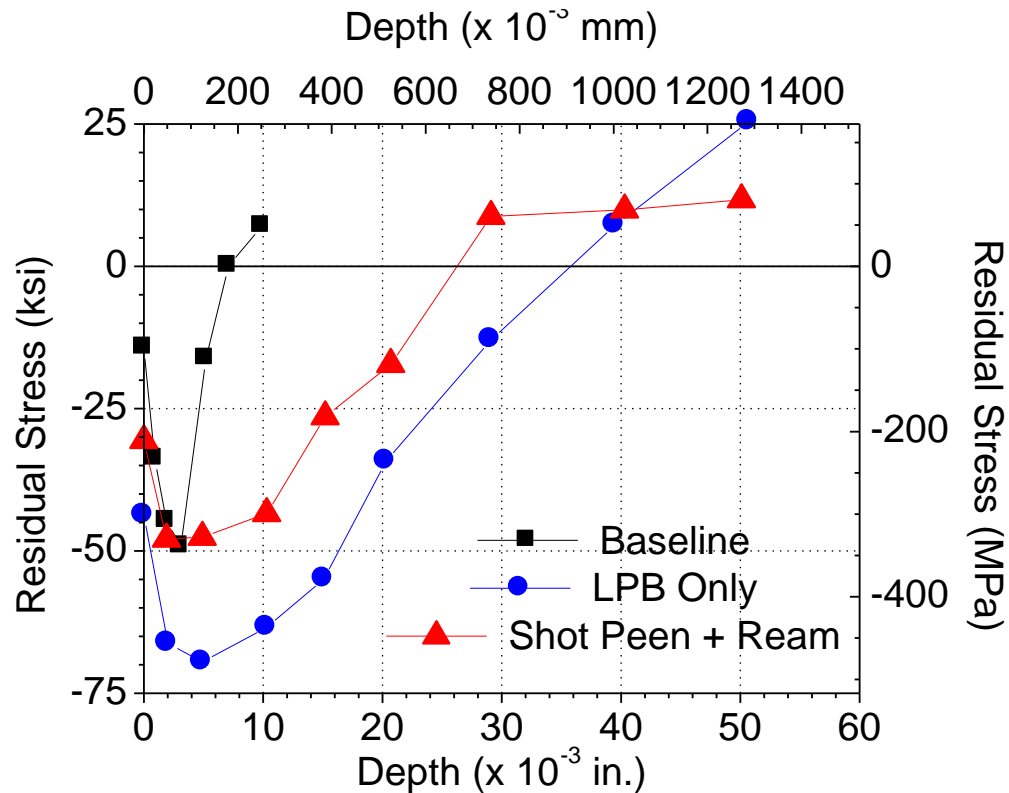
**Estimated Cost of Current Overhaul Practice : >\$1,000 per Blade**



# Robotic LPB Treatment of P-3 Propeller Bore

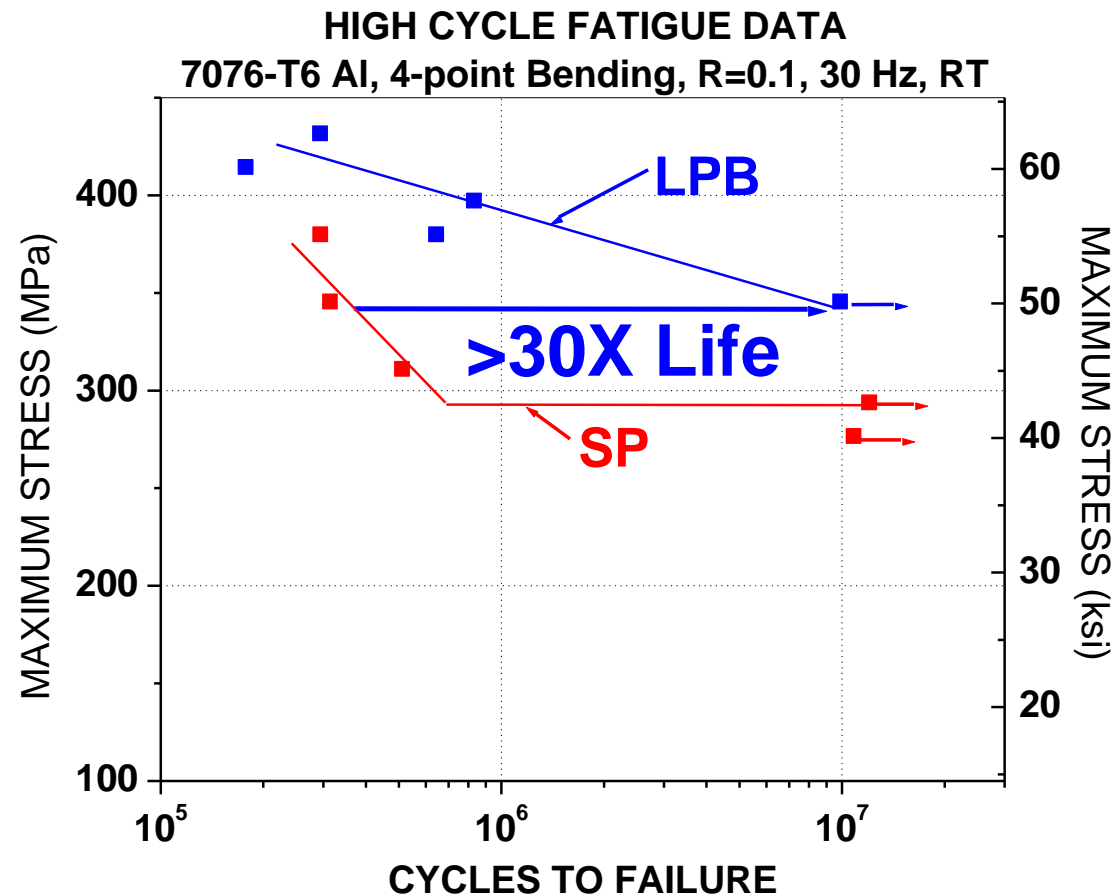


# Residual Stress



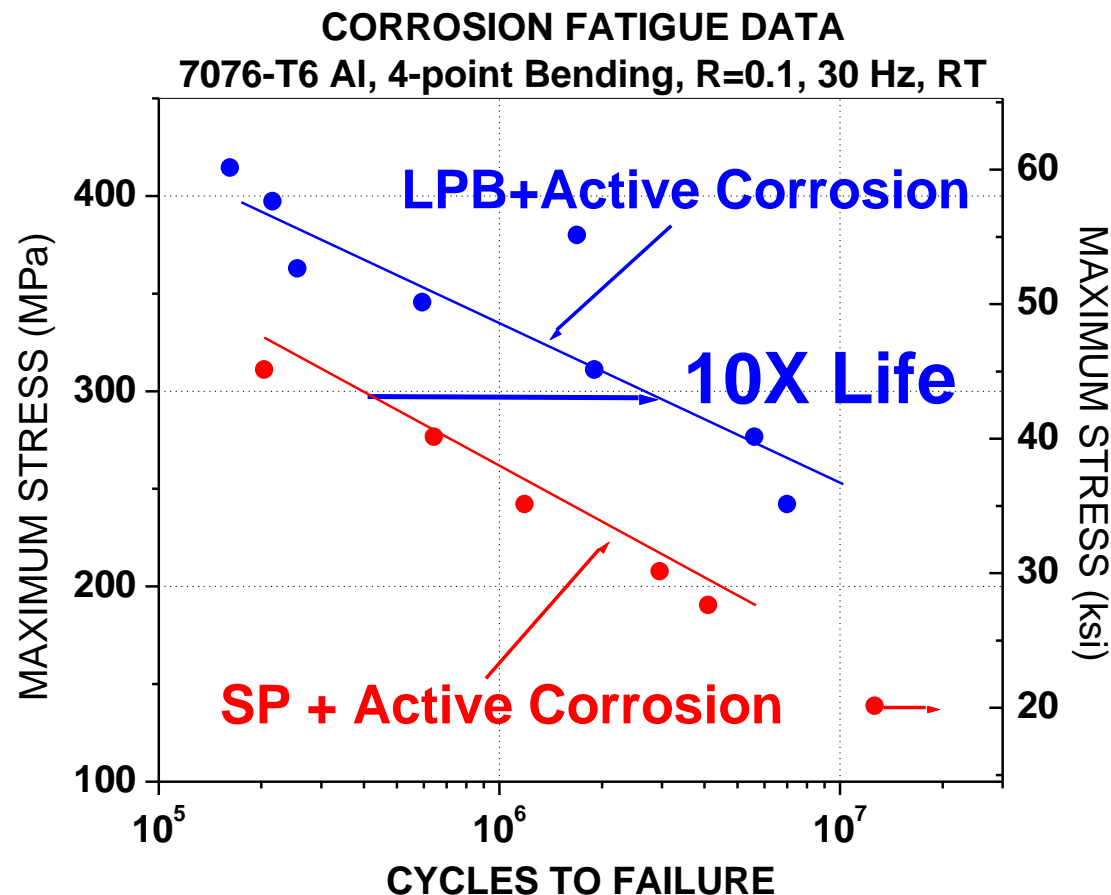
**LPB PRODUCED DEEPER AND HIGHER MAGNITUDE COMPRESSION THAN SHOT PEENING**

# Baseline Fatigue Performance



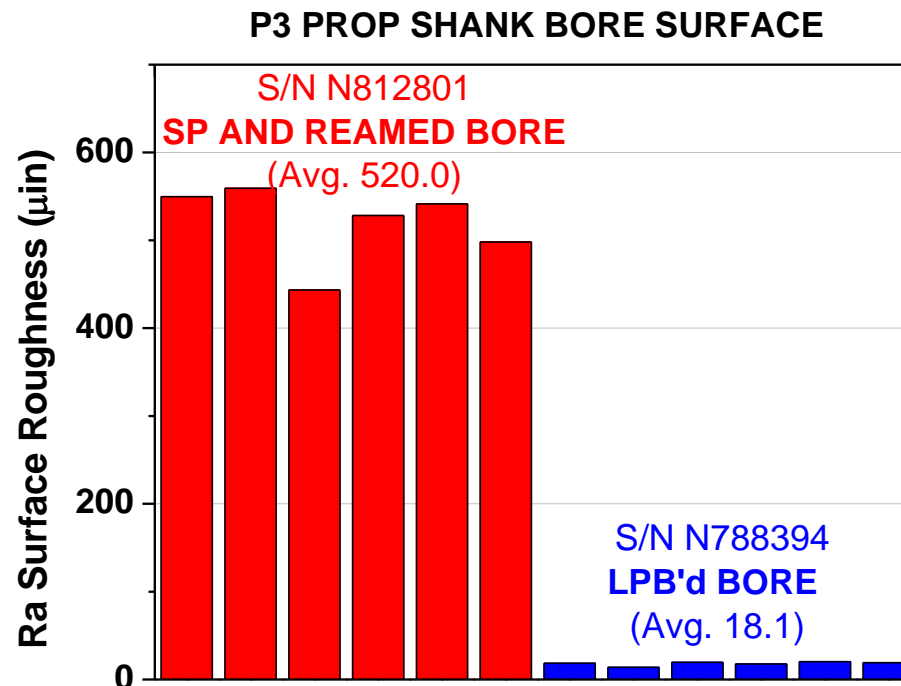
**LPB PRODUCED >30X LIFE IMPROVEMENT**

# Active Corrosion Fatigue Performance



**LPB PRODUCED 10X LIFE IMPROVEMENT**

# Surface Roughness



**MIRROR-LIKE SURFACE FINISH FROM LPB ELIMINATES  
THE NEED FOR REAMING AND RE-MACHINING**

# Substitution of LPB for SP

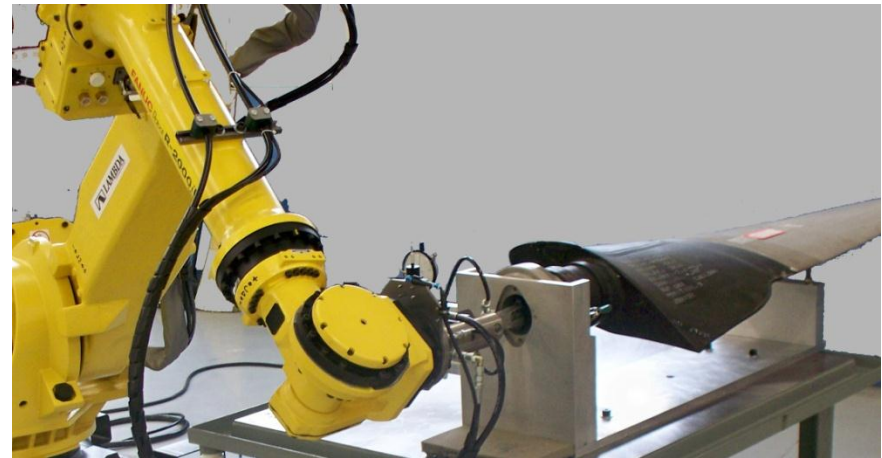
- 30X improved baseline fatigue life
- 10X improved corrosion fatigue life
- Mirror-like surface finish
- One-step process with no loss of material

**Saves \$1,000 per blade processed**

**Eliminates 3-cycle scrap rate for \$35,000 blade**

# Integration Into Production

1. NAVAIR – Cherry Point
  - P-3 Overhaul
2. Pacific Propeller Intl.
  - P-3 & C-130 Overhaul
3. Robins AFB
  - C-130 Overhaul



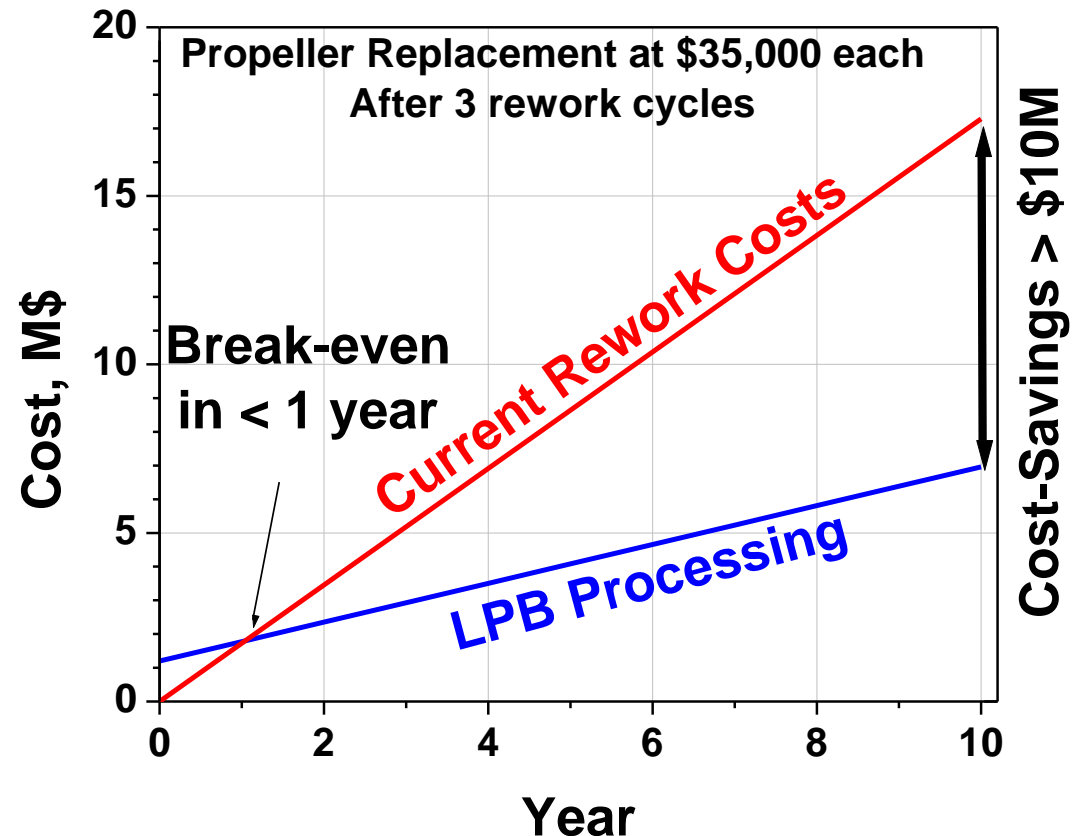
**Robotic LPB systems were placed into three MRO facilities.  
Over 1,000 blades are expected to be processed each year.**



# Cost-Benefit Analysis

## ASSUMPTIONS:

- Maintenance
  - Repeatedly Shot peened and reamed
    - Blade retired and replaced after 3 rework cycles
- Costs
  - \$1,500 per blade
  - \$35,000 per blade for replacement
- LPB process costs per year
  - 1,152 Blades per year
  - Initial Non-recurring Costs = \$1.2M



**Projected Cost Savings = \$10M**

**Projected Cost Avoidance Through Replacement = \$35,000 per blade**

# Summary – Propeller Bore

**LPB reduces maintenance costs by an estimated **\$1M annually****

**Elimination of 3-cycle scrap rate is estimated to yield **\$10M in cost avoidance each year****