

Bio-Inspired Distributed Sensing for Improved Flight Control

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Overview

Introduction

- Motivation

- Research Problem

Research at UoB

- Previous Research

- Current Research

Concluding Remarks

Further Work

Motivation: Why Bio-Inspired Distributed Sensing?

Amazing seagull!!!

<https://www.youtube.com/watch?v=MZQfxSn0FgA>

Motivation: Why Bio-Inspired Distributed Sensing?

✦ Current UAV autopilot technologies

✦ Challenges

✦ Potential use of force and flow information

Motivation: Why Bio-Inspired Distributed Sensing?

✶ Current UAV autopilot technologies

- Inertial
- Single point air speed
- GPS
- Vision

✶ Challenges

✶ Potential use of force and flow information

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✦ Current UAV autopilot technologies

✦ Challenges

✦ Potential use of force and flow information

Motivation: Why Bio-Inspired Distributed Sensing?

✶ Current UAV autopilot technologies

✶ Challenges

- Intrinsic nonlinear dynamics
- Classic control strategies limitations
- Limitations of inertial controls

✶ Potential use of force and flow information

Motivation: Why Bio-Inspired Distributed Sensing?

- ✦ Current UAV autopilot technologies
- ✦ Challenges
- ✦ Potential use of force and flow information

Motivation: Why Bio-Inspired Distributed Sensing?

✿ Current UAV autopilot technologies

✿ Challenges

✿ Potential use of force and flow information

- Availability of aerodynamic variables
 - Improved flight dynamics model
 - Stall detection
- Earlier gust detection
 - Gust rejection/alleviation
- Localised information
 - Localised control
 - Load tailoring

Research Problem

Use force and flow sensing to improve performance of UAVs flight control systems.

To achieve this we aim to:

- ✦ Develop distributed force and flow a sensing system for a small scale fixed wing UAV
- ✦ Integrate force and flow sensing into conventional flight control system architecture
- ✦ Measure response of systems to controlled and natural turbulence
- ✦ Develop advanced reflexive flight control system

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Previous Research at UoB: Strain sensing

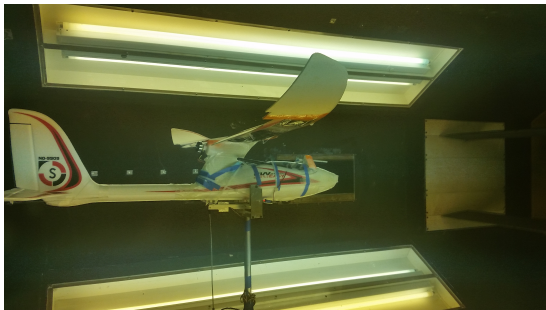


Figure: Strain sensing platform

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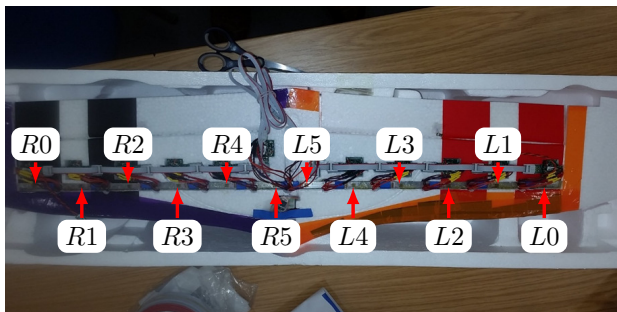


Figure: Strain sensing platform instrumentation

- 12 full-bridge strain gauges and amplifiers distributed along spar of wing
- Wind tunnel characterisation

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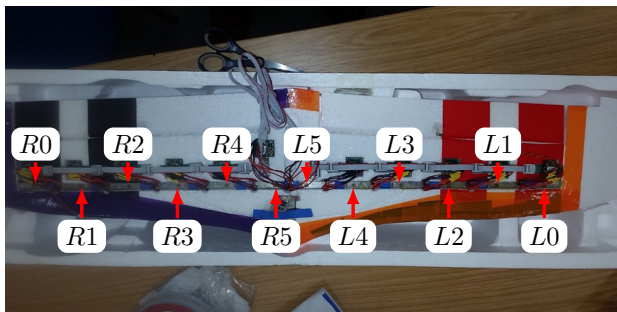


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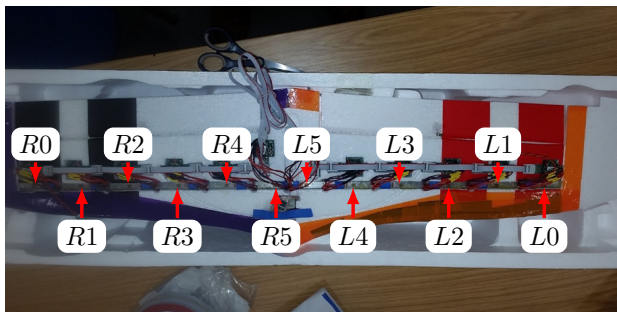


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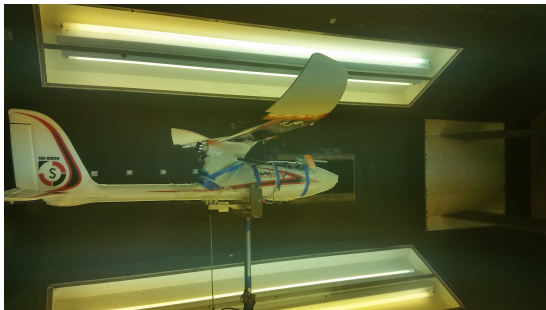


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- ✦ Open loop free flight
- ✦ Closed loop free flight
- ✦ Outdoor flight testing

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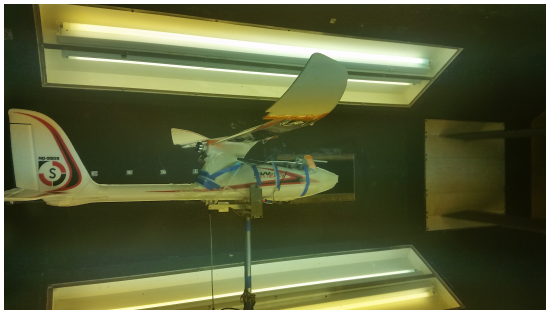


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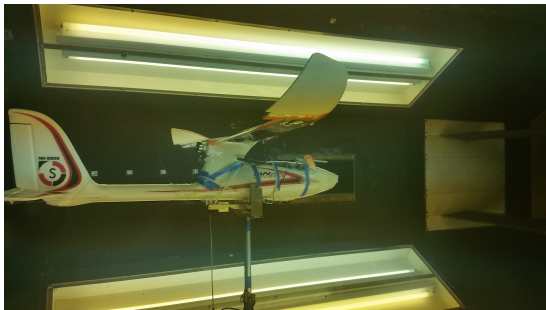


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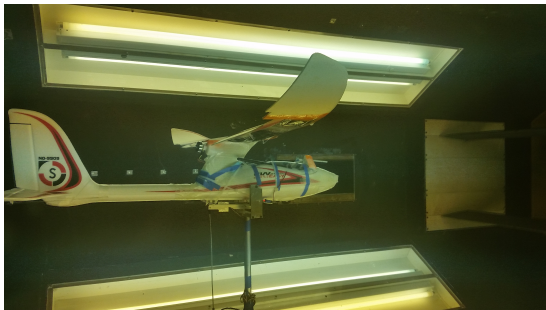


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What did we learned?

- ✦ Strain signal shows linear response with AoA
- ✦ Stall markers
- ✦ Strain-based roll control similar performance to IMU based control
- ✦ Information not available using IMU: AoA, stall, roll acceleration

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Figure: Pressure sensing platform

- ✦ 3-D printed insert on starboard wing
- ✦ 7 static-pressure ports distributed along wing-chord
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- ✦ Closed loop 1DOF WT testing
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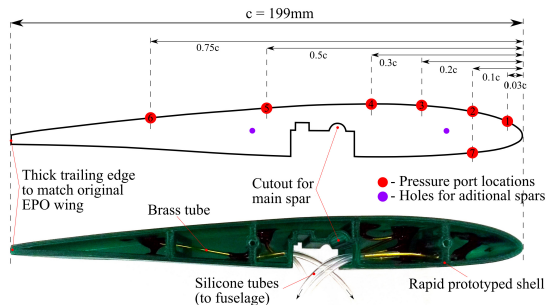


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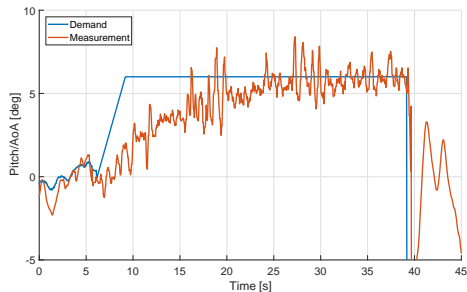


Figure: outdoors angle-of-attack tracking

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Current Research at UoB

- ✦ Experimental platform(s) with a distributed array of pressure and strain sensors
- ✦ Carry out calibration & characterisation (WT & outdoors)
- ✦ Design and implement closed loop control algorithms that use information from distributed array

Divided into two phases:

- ✦ Phase 1: Wind tunnel experiments using WT model
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Wing model instrumentation:

- ✦ Chord-wise array of 30 pressure ports in two sections
- ✦ Span-wise array with 16 strain gauges
- ✦ Servo actuated control surfaces
- ✦ MCU-based data acquisition system using, sampling 100 Hz
- ✦ 1-DOF pitch motion servo-driven system for automated motion

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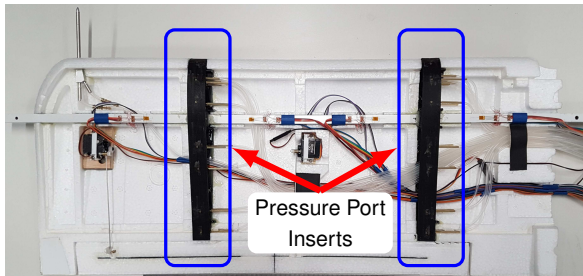


Figure: Wing model experimental platform

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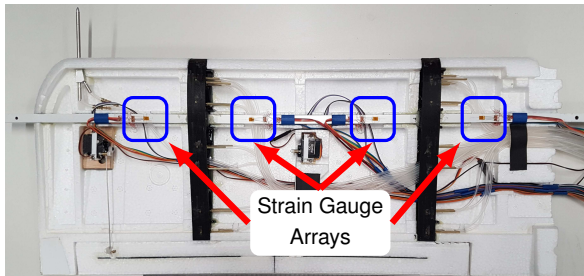


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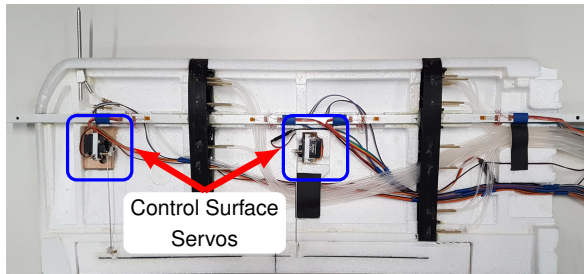


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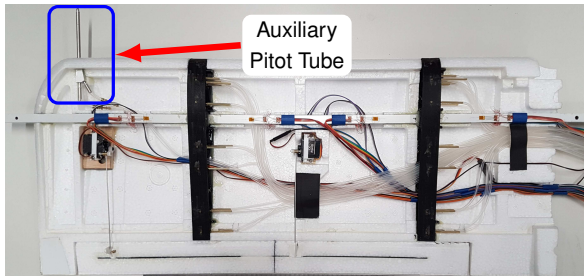


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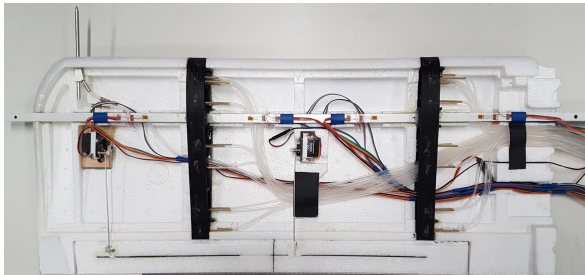


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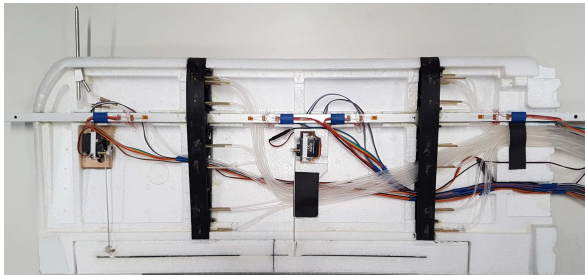
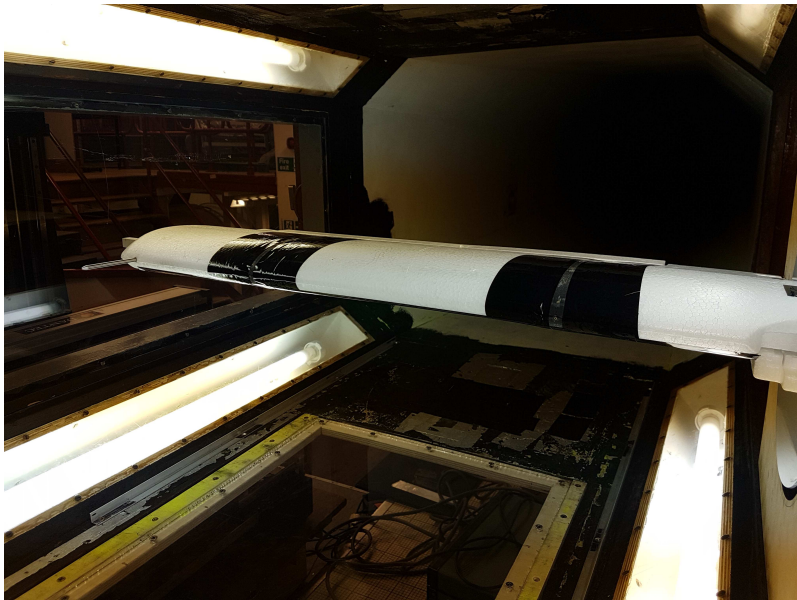


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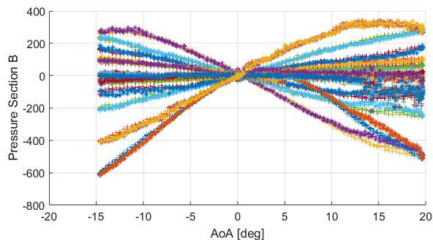


Figure: Pressure response for $q = \pm 20^\circ \text{ s}^{-1}$

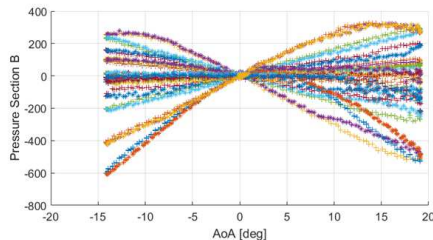


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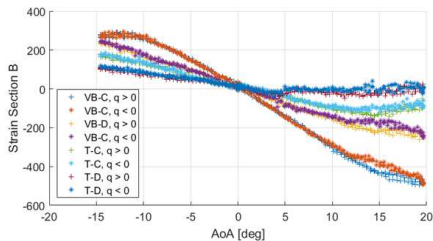


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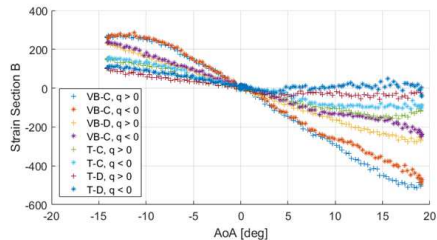


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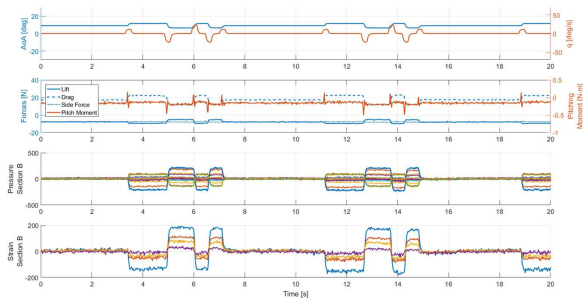


Figure: Pressure & strain response to dynamic input

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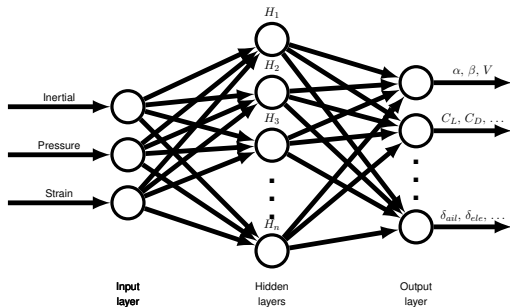


Figure: Possible UAV control strategies

Currently working on:

✶ Use strain and pressure signals to estimate

✶ Design and implement closed loop control.

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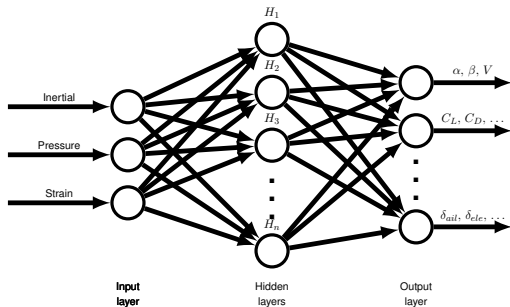


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- Aerodynamic loads

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- Classic control architecture (SISO)
- Algorithm using information from distributed array, (e.g. MISO, MIMO)

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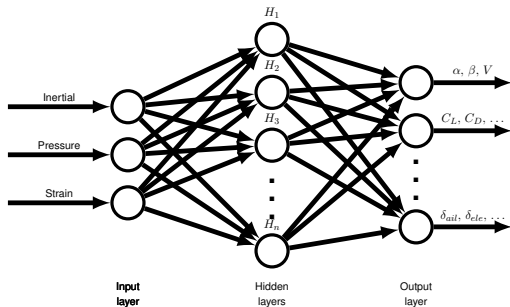


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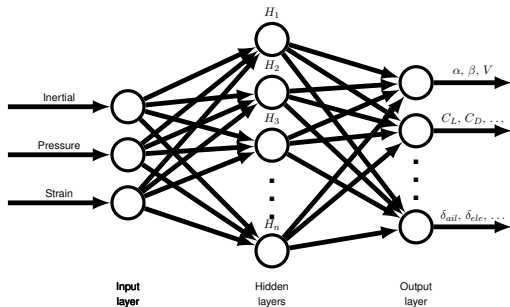


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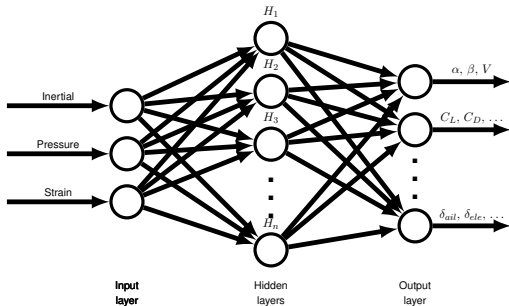


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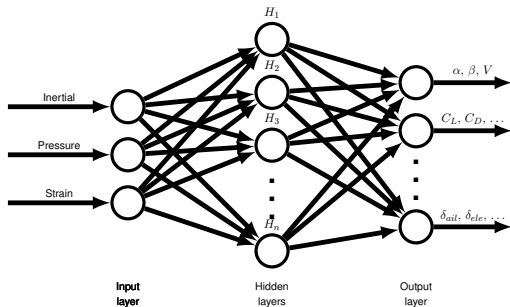


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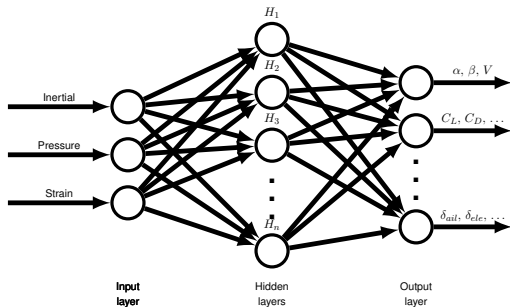


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Concluding Remarks

WT Characterisation experiments for strain and pressure signals

- Pressure and strain signal show linear relationship with ΔA

- Good detection of external

- Pressure and strain sensor similar performance to BCU based sensors

- Good pressure-based strain sensor to BCU based sensors

- Pressure sensor outperforming BCU based sensors in terms of

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✶ WT Characterisation experiments for strain and pressure signals

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Further Work

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 - Design and implement closed loop control algorithms
 - Carry out closed loop wind tunnel experiments

- ✿ Phase 2: Flying platform
 - Build and test a flying platform
 - Wind tunnel experiments
 - Outdoor flight tests

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Thank you