Variational Approach to Crack Propagation in a Cantilever Beam

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1 Linear Elasticity Basic Theory

We define:

$$\Omega \subset \mathbb{R}^d \ (d = 2, 3)
u : \Omega \to \mathbb{R}^d \ (\text{displacement})
e[v] := \frac{1}{2} (\nabla^T v + \nabla v^T) \ (\text{strain})
\nabla^T v := \begin{pmatrix} \partial_1 v_1 & \partial_2 v_2 \\ \partial_1 v_2 & \partial_2 v_1 \end{pmatrix}
\nabla v^T := (\nabla^T v)^T
\sigma[u] := Ce[u]
C = (C_{ijkl}) \begin{cases} C_{ijkl} = C_{klij} = C_{jikl} \\ (C_{\xi}) : \xi \geq C_* |\xi|^2 (\forall \xi \in \mathbb{R}^{d \times d}_{sym}) \end{cases}$$

Let's consider linear elasticity problem:

$$(**) \begin{cases} -div \ \sigma[u] = f(x), \text{ in } \Omega \\ u = g(x) \text{ on } \Gamma_D \\ \sigma[u]\nu = q(x) \text{ on } \Gamma_N \end{cases}$$
 (1)

$$f \in L^2(\Omega : \mathbb{R}^d, g \in H^1(\Omega : \mathbb{R}^d), q \in L^2(\Gamma_N : \mathbb{R}^d))$$

1.1 Strong Solution

$$u \in H^2(\Omega : \mathbb{R}^d)$$
 satisfies (**) then we call $u :$ a strong solution (2)

1.2 Weak Solution

Weak Solution
$$\begin{cases} \int_{\Omega} \sigma[u] : e[v] dx = \int_{\Omega} f \cdot v dx + \int_{\Gamma_N} q \cdot v ds \big(\forall v \in V := \{ v \in H^1 \ (\Omega : \mathbb{R}^d) \ |v|_{\Gamma_D} = 0 \} \big) \\ u \in V + g \end{cases}$$
(3)

1.3 Proposition

$$u:$$
 strong solution $\Leftrightarrow \begin{cases} u: \text{ weak solution} \\ u \in H^2 \ (\Omega: \mathbb{R}^d) \end{cases}$

Proof. (\Rightarrow) Assume we choose $v \in V := \{v \in H^1 \ (\Omega : \mathbb{R}^d) \ |v|_{\Gamma_D} = 0\}$, with v is a very smooth test function. Then we take integral over the domain for equation (1) on both side.

$$\begin{split} \int_{\Omega} -div\sigma[u] \cdot v dx &= \int_{\Omega} f \cdot v dx \\ \int_{\Omega} \sigma[u] : \nabla v dx - \int_{\Gamma} \sigma[u] \nu \cdot v ds &= \int_{\Omega} f \cdot v dx \text{ (by Divergence Formula)} \\ \int_{\Omega} \sigma[u] : \nabla v dx - \left(\int_{\Gamma_D} \sigma[u] \nu \cdot v ds + \int_{\Gamma_N} \sigma[u] \nu \cdot v ds \right) &= \int_{\Omega} f \cdot v dx \end{split}$$

From Boundary Condition we know that:

$$\begin{cases} v = 0 \text{ on } \Gamma_D \\ \sigma[u]\nu = q \text{ on } \Gamma_N \end{cases}$$

Hence, we have:

$$\int_{\Omega} \sigma[u] : e[v]dx - \int_{\Gamma_N} q \cdot v ds = \int_{\Omega} f \cdot v dx$$

$$\int_{\Omega} \sigma[u] : e[v]dx = \int_{\Omega} f \cdot v dx + \int_{\Gamma_N} q \cdot v ds$$
(4)

with:

$$\begin{split} X &:= H^1(\Omega:\mathbb{R}^d) \\ a(u,v) &:= \int_{\Omega} \sigma[u] : e[v] dx \\ l(v) &:= \int_{\Omega} f \cdot v dx + \int_{\Gamma_N} q \cdot v ds \\ &= a(v,u) \end{split}$$

Then we rewrite equation (4) in a bilinear and linear form:

$$a(u, v) = l(v)$$

 (\Leftarrow) Since $u \in H^2$ $(\Omega : \mathbb{R}^2)$, $div(\sigma[u]) \in L^2(\Omega : \mathbb{R}^2)$ and a(u,v) = 0 for all $v \in V$, we have:

$$\begin{aligned} 0 &= \int_{\Omega} \sigma[u] : e[v] dx \\ 0 &= \int_{\Omega} \begin{pmatrix} \sigma_{11} & \sigma_{12} \\ \sigma_{21} & \sigma_{22} \end{pmatrix} : \left(\nabla v_{1} & \nabla v_{2} \right) dx \\ 0 &= \int_{\Omega} \begin{pmatrix} \sigma_{11} \\ \sigma_{21} \end{pmatrix} \cdot \nabla v_{1} + \begin{pmatrix} \sigma_{12} \\ \sigma_{22} \end{pmatrix} \cdot \nabla v_{2} dx \\ 0 &= \int_{\Omega} div \begin{pmatrix} \sigma_{11} \\ \sigma_{21} \end{pmatrix} v_{1} + div \begin{pmatrix} \sigma_{12} \\ \sigma_{22} \end{pmatrix} v_{2} - \int_{\partial\Omega} \begin{pmatrix} \sigma_{11} \\ \sigma_{21} \end{pmatrix} \cdot \nu v_{1} + \begin{pmatrix} \sigma_{12} \\ \sigma_{22} \end{pmatrix} \cdot \nu v_{2} ds \\ 0 &= \int_{\Omega} (f + div\sigma[u]) \cdot v dx - \int_{\partial\Omega} (\sigma[u]\nu) \cdot v ds \\ 0 &= \int_{\Omega} f \cdot v dx + \int_{\Omega} div\sigma[u] \cdot v dx - \int_{\partial\Omega} (\sigma[u]\nu) \cdot v ds \end{aligned}$$

then, assume we choose $v \in C_0^{\infty}(\Omega) \subset V, v = 0$ near $\partial \Omega$ $f \in L^1(\Omega)$, then we have:

$$\int_{\Omega} (f + div\sigma[u]) \cdot v dx = 0 \ (\forall v \in C_0^{\infty}(\Omega, \mathbb{R}^2))$$
$$\therefore f + div\sigma[u] = 0 \text{ in } \Omega$$

then, $\forall v \in C_0^{\infty}(\bar{\Omega})$ s.t. $(supp(v) \cap \partial\Omega) \subset \Gamma_N$

$$\int_{\Gamma_N} (\sigma[u]\nu) \cdot v ds = 0$$

$$\sigma[u]\nu = 0 \text{ on } \Gamma_N$$

For $v \in V$

$$a(v,v) = \int_{\Omega} (Ce[v]) : e[v]dx$$

$$\geq C_* \int_{\Omega} |e[v]|^2 dx$$

$$\geq C_* ||v||_x^2$$

Properties 1. • $a(\cdot, \cdot)$ is bounded symmetric, bilinear form on $X \times X$.

- $a(\cdot, \cdot)$ is coercive on $V \times V$.
- \bullet l is bounded linear form on X.

Theorem 1. For any $g \in H^1(\Omega : \mathbb{R}^d)$,

$$\exists ! u : a \text{ weak solution of } (**), \text{ and } \left\{ u = \operatorname{argmin}_{w \in V + g} E(w) \right\}$$

2 Linear Elasticity with Crack Propagation Case

2.1 Strong Form

Let's consider Linear Elasticity with Crack Propagation Case:

$$\begin{cases}
-div\sigma[u] = f(x), & (x \in \Omega \setminus \Sigma(L)) \\
u = g & (x \in \Gamma_D) \\
\sigma[u]\nu = q & (x \in \Gamma_l) \\
\sigma[u]\nu = 0 & (x \in \Gamma_0 \cup \Sigma^+(L) \cup \Sigma^-(L))
\end{cases} \tag{5}$$

2.2 Weak Form

We define

$$V_L := v \in H^1(\Omega_L : \mathbb{R}^2) | v = 0 \text{ on } \Gamma_D$$

then the problem (5) becomes: Find $u \in V_L$ such that,

$$\int_{\Omega_L} \sigma[u] : e[v] dx = \int_{\Omega_L} fv dx + \int_{\Gamma_I} qv ds \ (\forall v \in V_L)$$
 (6)

then, we define elastic energy:

$$E(v,L) := \frac{1}{2}a_L(v,v) - l(v), \ (v \in V_L)$$
 (7)

3 Modelling and Simulation

In this simulation, we use a cantilever beam (Mild Steel Material) as the domain, which is a thin rectangular cross section introduced by Timoshenkol Goodier (1970), then we specify nondimensionalized parameter for simulation as shown in table ??.

Numerical Parameter	Typical Value [unit]	Nondimensionalized value
Young's modulus (E)	$2.1 \times 10^7 \; [Pa]$	210
Poisson's ratio (ν)	0.3 [-]	0.3
Gravity constant (f)	9.80655 [N/kg]	9.80655
Weight (q)	0 [N]	0
Length (L)	3.0 [m]	3.0
Depth (h)	0.2 [m]	0.2
Width (b)	0.25 [m]	0.25

Table 1: Material properties and numerical parameters

With the help of FreeFem++ software, we created a 2D and 3D model as shown in figure 1,

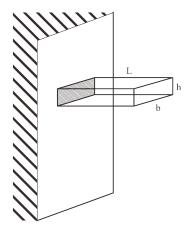


Figure 1: 3D Model of cantilever beam. We use gravity force as the body force \mathbf{f} and fixed the left part of the beam, and then we give a 1 Newton weight force act on the right part of the beam as the neumann boundary condition \mathbf{q} .

then we solve the displacement vector (u, v). After solving the displacement, we calculate σ which stand for stress force acting on surface of the cantilever beam using equation below:

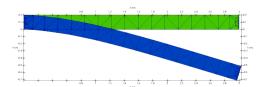
$$\sigma = (d\lambda^2 + 4\lambda\mu)div(u)^2 + (4\mu^2|e[u]|^2), \ d = 2,3$$

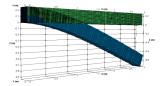
$$\lambda \text{ (Lame's first parameter) } := \frac{E\nu}{(1+\nu)(1-2\nu)}$$

$$\mu \text{ Lame's second parameter } := \frac{E}{2(1+\nu)}$$

4 Result and Discussion

We solved the problem in (3) by using P1 finite element method on FreeFEM++, where u and v calculated for division number of mesh = 32. In the figure 2 we can see the deformation of the cantilever beam in 2D and 3D graphics.

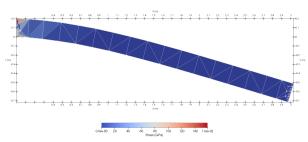


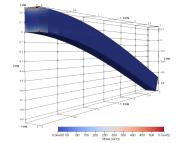


- (a) Deformation in 2D. Green line show condition before gravity and weight force applied to the domain. Red line show condition after we solve linear elasticity with gravity and weight force applied to the domain. Maximal Displacement $(u=0.03\ [m])$
- (b) Deformation in 3D. Green line show condition before gravity and weight force applied to the domain. Red line show condition after we solve linear elasticity with gravity and weight force applied to the domain. Maximal Displacement $(u = 0.05 \ [m])$

Figure 2: Deformation in 2D and 3D

While in the figure 3 we can see result from calculating the stress tensor on 2D and 3D case.





- (a) Calculated σ on 2D case. The value of σ on the domain, maped by the color in the picture with respect to the color palette on the lower side of the graph. Maximal stress given on the surface ($\sigma = 158.612 \ [GPa]$)
- (b) Calculated σ on 3D case. The value of σ on the domain, maped by the color in the picture with respect to the color palette on the lower side of the graph. Maximal stress given on the surface ($\sigma = 567.034 \ [GPa]$)

Figure 3: Stress Tensor in 2D and 3D

On the figure 4, we can see comparison of calculated surface stress on 2D and 3D case (sliced on the side).

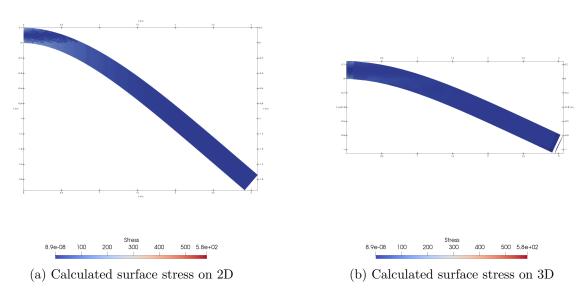


Figure 4: Comparison of Stress Tensor in 2D and 3D

On the figure 5 below, we can see the result from sliced view on 3D case, in this case, we sliced through the Y-normal plane of the 3D model.

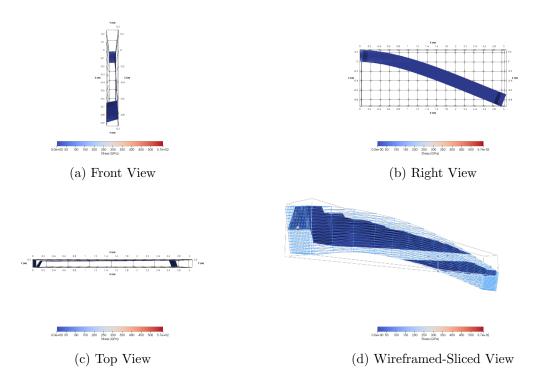


Figure 5: 3D Sliced View

4.1 Convergence Analysis

Since we don't have an exact solution for problem (1), we will define

$$u_h := u^k \iff \max |u^k - u^{k-1}| \le \epsilon$$

k is index of the current solution and ϵ is a small number, $\epsilon > 0$.

There are three types of error that we will compute, Infinity Error, $H^1(\Omega)$ and $L^2(\Omega)$, each of them defined by:

$$||u_h - u||_{\infty} = \max |u_h - u| \tag{8}$$

$$||u||_{H^1(\Omega)}^2 = \int_{\Omega} |u|^2 dx + \int_{\Omega} |\nabla u|^2 dx \tag{9}$$

$$||u||_{L^{2}(\Omega)}^{2} = \int_{\Omega} |u|^{2} dx \tag{10}$$

We can see the result from 2D and 3D case in the figure 6 below:

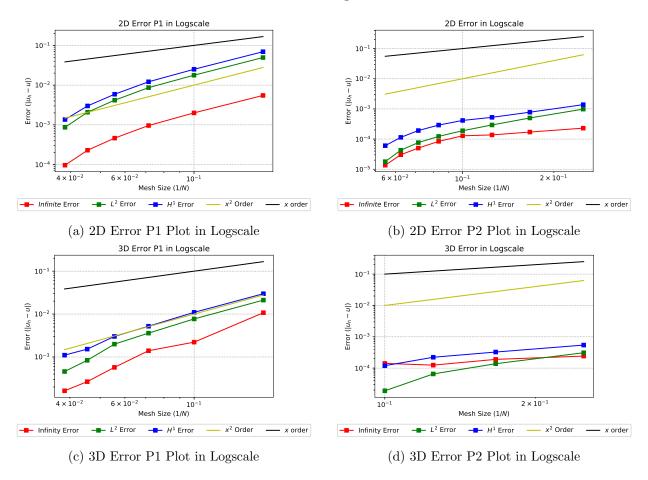
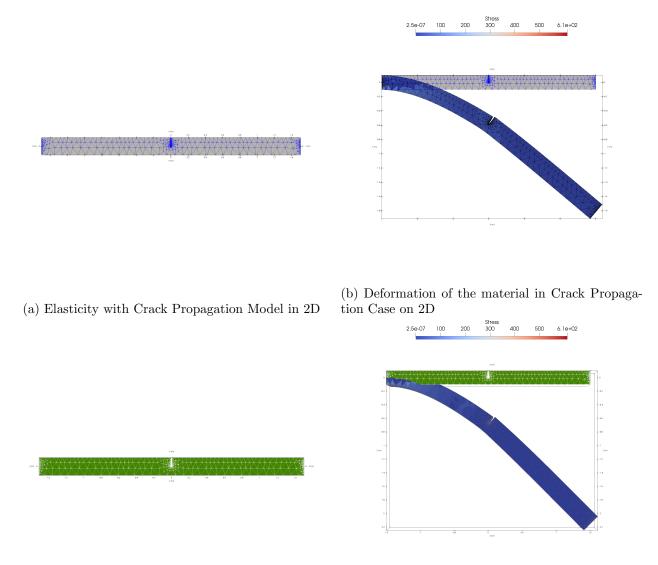


Figure 6: Error Plot in Logscale

5 Linear Elasticity with Crack Propagation

We solved problem (6) using same model as shown in figure 1 and numerical parameter like in table 1, the difference from linear elasticity problem is the domain. In the crack propagation case, we disturb the domain with adding some so called "crack path" into the domain. For this case, we use n(division number) = 32 and d(width of the crack) = 0.001 [m]. The result of our simulation shown the figure 7 for 2D case:



(d) Deformation of the material in Crack Propaga-(c) Elasticity with Crack Propagation Model in 3D tion Case on 3D

Figure 7: Elasticity with Crack Propagation Case in 2D and 3D

After that, we also compare our 2D and 3D model in crack propagation case, the result is shown in the figure 8

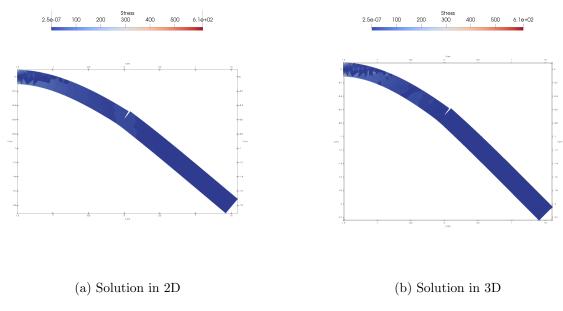


Figure 8: Elasticity with Crack Propagation Case in 2D and 3D

We also calculated the elastic energy using equation (7), below is the result of the energy profile from our model in figure 7.

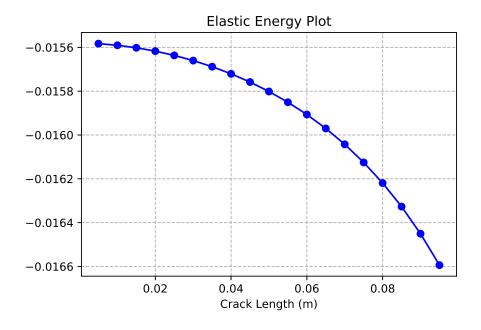


Figure 9: Elasticity Energy Profile Crack Problem Model in 2D