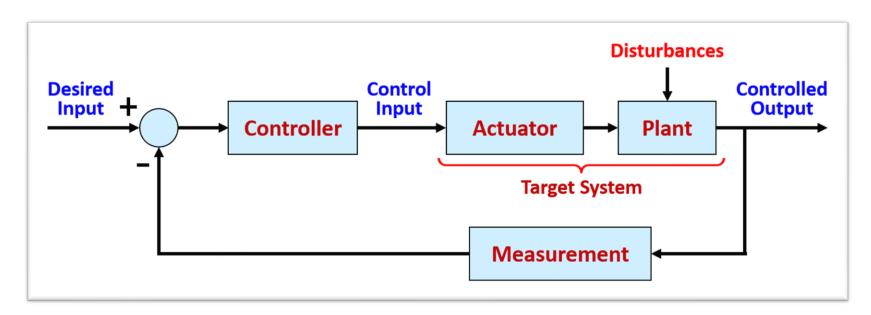
# **Analysis of Feedback 1**

#### **Lecture 6:**

- Overview of Feedback & PID Control
- Example of PID Control System



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### **New Arrived Robots from Boston Dynamics!!**

#### Parkour Atlas



https://www.youtube.com/watch?v=LikxFZZO2sk

## **Newly Arrived Robots from Boston Dynamics!!**

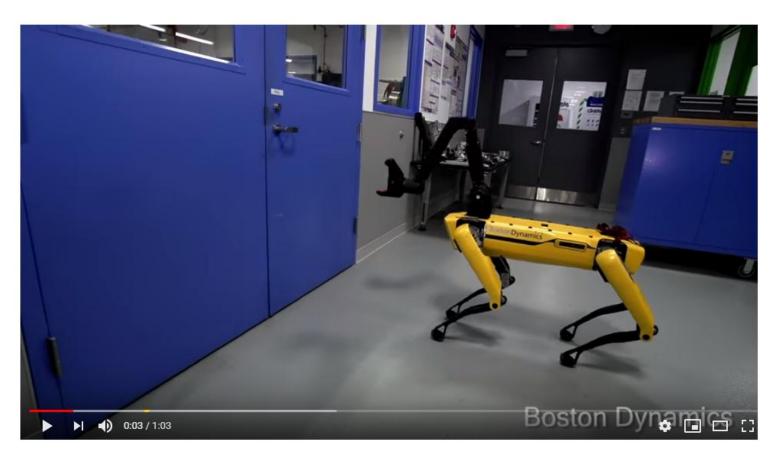
### SpotMini



https://www.youtube.com/watch?v=fUyU3IKzoio

## **Newly Arrived Robots from Boston Dynamics!!**

#### Robustness Test for Disturbances!!



https://www.youtube.com/watch?v=aFuA50H9uek

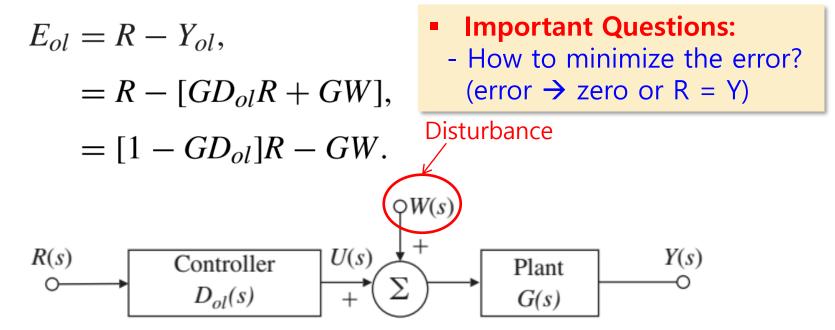
# **Basic Equations of Control**

#### Open-Loop System

The controlled output is given by

$$Y_{ol} = GD_{ol}R + GW,$$

Error, difference btw reference input (R) and output (Y) is given by



# **Basic Equations of Control (cont'd)**

#### Closed-Loop System

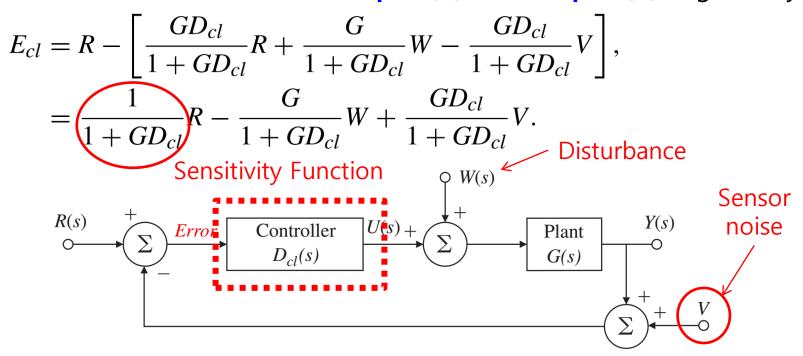
The controlled output is given by

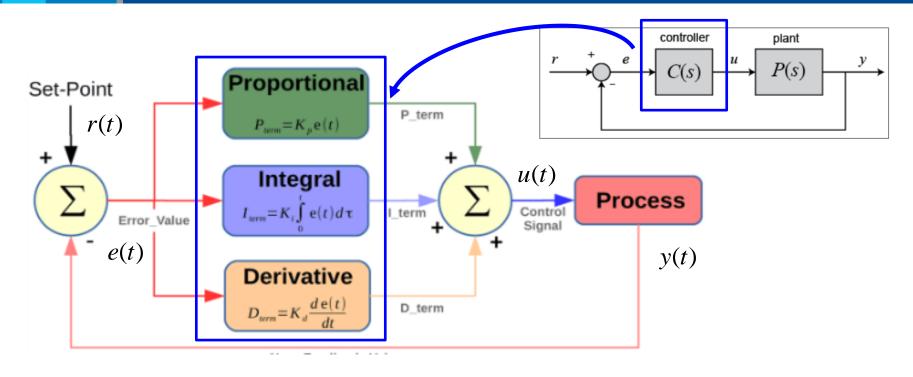
How to minimize the error?
 (Objectives: error → zero)

$$Y_{cl} = \underbrace{\frac{GD_{cl}}{1 + GD_{cl}}} R + \underbrace{\frac{G}{1 + GD_{cl}}}_{\text{C-L Transfer Function}} W - \underbrace{\frac{GD_{cl}}{1 + GD_{cl}}}_{\text{C-L Transfer Function}} V,$$



Error, difference btw reference input (R) and output (Y) is given by





❖ The control signal of a **PID controller** in the time-domain

$$u(t) = K_p e(t) + K_i \int e(t) dt + K_d \frac{de(t)}{dt}$$
where,
error value,  $e(t) = r(t) - y(t)$ 

# **Brief History of PID control**

- **❖** The First real PID-type Controlled by Elmer Sperry in 1911.
  - Gyroscopic compass & stabilizer application





Elmer Ambrose Sperry (1860~1930)

- ❖ The first theoretical analysis of a PID controller was published by Nicolas Minorsky in 1922.
  - Automatic steering systems for US Navy ships.
  - Developed by observation of a helmsman,
     steering ships based on not only current error (P-control)
     but also, past error (I) as well as the rate of change (D)

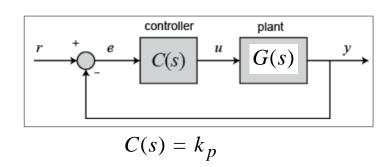


Nicolas Minorsky (1885~1970)



#### Proportional Control (P)

Linearly proportional to system error  $u(t) = k_P e(t)$ ,



Assuming plant transfer function can be given by

$$G(s) = \frac{A}{s^2 + a_1 s + a_2}.$$

natural frequency



$$G(s) = \frac{A}{s^2 + a_1 s + a_2}. \qquad \Rightarrow \frac{Y(s)}{R(s)} = T(s) = \frac{C(s)G(s)}{1 + C(s)G(s)} = \frac{k_p A}{s^2 + a_1 s + a_2 + k_p A}$$

**Applying model matching condition (i.e., comparing denominators)** 

$$s^2 + \underline{a_1 s} + \underline{a_2 + k_P A}$$
 Damping term including Natural frequency term

$$T(s) = \frac{\omega_n^2}{s^2 + 2\zeta \omega_n s + \omega_n^2}.$$

- **Important questions:** 
  - $\omega_n$  vs. rising time ??

# **Dynamic Response (Revisited)**

#### (5) Control Performance Index: Step Response (cont'd)

#### (3) Rise Time

$$y(t) = 1 - \frac{1}{\sqrt{1 - \zeta^2}} e^{-\zeta \omega_n t} \sin(\omega_d t + \phi)$$
 for  $\zeta < 1$ 

$$y(t) = 1$$
 @ rising time  $(T_r)$ 

$$\begin{array}{c} t_p \\ 0.9 \\ \hline \\ 0.1 \\ \hline \\ t_r \\ t_s \\ \end{array}$$

$$\sin(\omega_{d}T_{r} + \varphi) = 0$$

$$or$$

$$where,$$

$$\therefore T_{r} = \frac{\pi - \varphi}{\omega_{d}} \quad \omega_{d} = \omega_{n}\sqrt{1 - \varsigma^{2}}$$

$$\cos \varphi = \varsigma$$

$$i)\varsigma = 0.5$$

$$\Rightarrow \varphi = \cos^{-1}(0.5) = 1.05$$

$$\therefore T_{r} = \frac{3.14 - 1.05}{\omega_{n}\sqrt{1 - 0.5^{2}}} \approx \frac{2.4}{\omega_{n}}$$

$$\therefore T_{r} = \frac{3.14 - 0.8}{\omega_{n}\sqrt{1 - 0.7^{2}}} \approx \frac{3.3}{\omega_{n}}$$

$$i) \varsigma = 0.5$$

$$\Rightarrow \varphi = \cos^{-1}(0.5) = 1.05$$

$$\therefore T_r = \frac{3.14 - 1.05}{\omega_n \sqrt{1 - 0.5^2}} \approx \frac{2.4}{\omega_n}$$

$$i) \varsigma = 0.7$$

$$\Rightarrow \varphi = \cos^{-1}(0.7) \approx 0.8$$

$$\therefore T_r = \frac{3.14 - 0.8}{\omega_n \sqrt{1 - 0.7^2}} \approx \frac{3.3}{\omega_n}$$

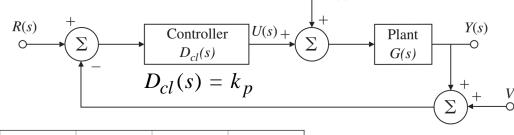
#### **Discussion & Question:**

- How does damping coefficient & natural frequency work for the rising time ??

#### Proportional Control (P) – cont'd

Closed-Loop transfer function of P-control

$$\frac{Y(s)}{R(s)} = T(s) = \frac{k_p G(s)}{1 + k_p G(s)} \xrightarrow{R(s)} \stackrel{R(s)}{\hookrightarrow} \stackrel{+}{\searrow}$$



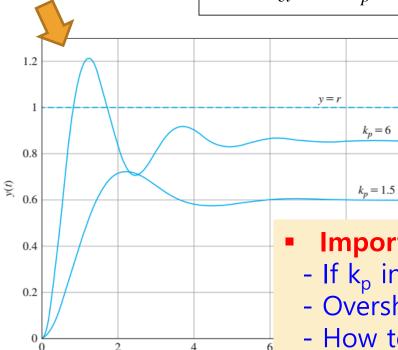


### MATLAB Simulation

$$G(s) = \frac{A}{s^2 + a_1 s + a_2}.$$

where,

$$a_1 = 1.4, a_2 = 1, A = 1$$



Time (sec)

**Important questions:** 

 $\bigcirc W(s)$ 

- If k<sub>p</sub> increases?
- Overshoot value?
- How to reject the error?

#### Proportional Control (P) – cont'd

Closed-loop TF

$$\frac{Y(s)}{R(s)} = T(s) = \frac{k_p G(s)}{1 + k_p G(s)}$$

Sensitivity TF



$$\frac{Y(s)}{R(s)} = T(s) = \frac{k_p G(s)}{1 + k_p G(s)}$$

$$\frac{E(s)}{R(s)} = S(s) = \frac{1}{1 + k_p G(s)}$$

■ Using final value theorem at G(0) = 1 & Step input R(s) = 1/s

$$\lim_{t\to\infty} f(t) = \lim_{s\to 0} sF(s) \quad \text{if all the poles of sF(s) are in} \\ \quad \text{the left half plane (LHP)}$$

$$y(\infty) = Y(0) = s \frac{k_p G(0)}{1 + k_p G(0)} \frac{1}{s} \qquad e(\infty) = E(0) = s \frac{1}{1 + k_p G(0)} \frac{1}{s}$$

- **Important Findings:** 
  - Proportional control (P) vs. regulation error ??
  - Can we have Zero SSE with P-control ?? (If not, How??)

#### Integral Control (I)

Control law of integral feedback is given by,

$$u(t) = k_{I} \int_{t_{0}}^{t} e(\tau) d\tau, \qquad \underbrace{\frac{U(s)}{E(s)}}_{0.8} = D_{cl}(s) = \frac{k_{I}}{s},$$

Area

$$u(t_{1}) = k_{I} \int_{t_{0}}^{t} e(\tau) d\tau, \qquad \underbrace{\frac{U(s)}{E(s)}}_{0.8} = D_{cl}(s) = \frac{k_{I}}{s},$$

Time (sec)

#### Integral Control (I) – cont'd

Closed-loop TF is given by,

$$\frac{Y(s)}{R(s)} = \mathcal{T}(s) = \frac{\frac{k_I}{s}G(s)}{1 + \frac{k_I}{s}G(s)} = \frac{k_IG(s)}{s + k_IG(s)}.$$
 From previous slide, 
$$y^{(\infty)} = Y(0) = \sqrt[k]{\frac{k_pG(0)}{1 + k_pG(0)}} \frac{1}{s}$$
 Sensitivity TF is given by,

Sensitivity TF is given by,

$$\frac{E(s)}{R(s)} = \frac{1}{1 + \frac{k_I}{s}G(s)} = \frac{s}{s + k_I G(s)},$$

$$y(\infty) = Y(0) = \sqrt{\frac{k_p G(0)}{1 + k_p G(0)}} \frac{1}{\sqrt{s}}$$

$$e(\infty) = E(0) = \sqrt{\frac{1}{1 + k_p G(0)}} \frac{1}{\sqrt{1 + k_p G(0)}}$$

Using final value theorem with G(0) = 1 & Step input R(s) = 1/s

$$y(\infty) = \frac{k_I G(0)}{0 + k_I G(0)} = 1,$$

$$e(\infty) = \frac{0}{0 + k_I G(0)} = 0,$$

- **Important Findings:** 
  - Integral control vs. SSE ??

#### Integral Control (I) – cont'd

#### Example:

$$C(s) = \frac{k_i}{s}$$
  $G(s) = \frac{b}{s+a}$ 

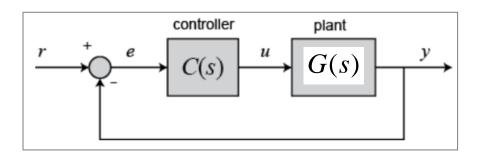
$$T(s)_{CL} = \frac{C(s)G(s)}{1 + C(s)G(s)}$$

$$= \frac{\frac{k_i}{s} \frac{b}{s + a}}{1 + \frac{k_i}{s} \frac{b}{s + a}}$$

$$= \frac{k_i b}{s^2 + as + k_i b}$$

Damping term including Natural natural frequency

Natural frequency term



$$T(s) = \frac{\omega_n^2}{s^2 + 2\zeta \omega_n s + \omega_n^2}.$$

#### Important questions:

- $\omega_n$  vs. rising time ??
- I-control vs. damping term ??
- I-control vs. rising time ??

#### Derivative Control (D)

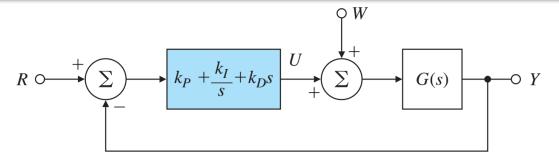
Control law of derivative feedback is given by,

$$u(t) = k_D \dot{e}(t), \quad \stackrel{\mathcal{L}}{\longrightarrow} \quad \frac{U(s)}{E(s)} = D_{cl}(s) = k_D s.$$

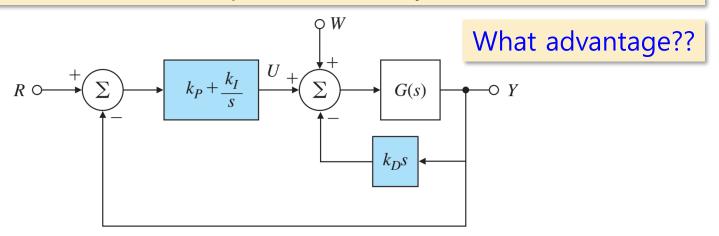
#### Important Findings:

- Derivative control is almost never used by itself !!. Why??
  - (1) The derivative does not supply information on the desired end state.
  - (2) if e(t) were to remain constant, the output of a derivative controller would be zero.
- Thus, a proportional or integral control would be needed to provide a control signal at this time.
- ※ An important effect of the derivative term is that it gives a sharp response to suddenly changing signals.

- Derivative Control (D) cont'd
  - Two types of derivative control applications PID vs. PIV
  - (1) D-term in the forward path General PID



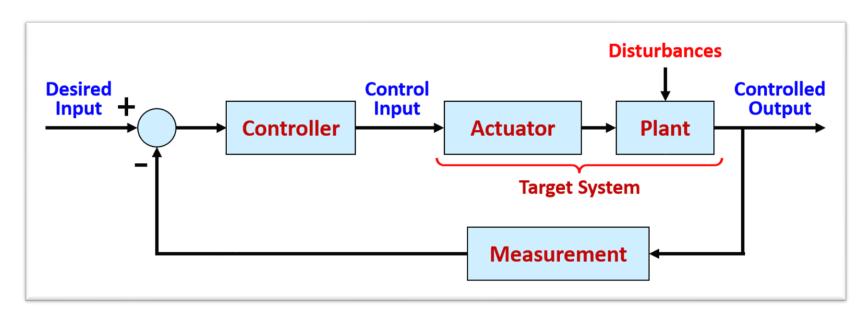
(2) D-term in the feedback path (or velocity feedback) - PIV



# **Analysis of Feedback 1**

#### **Lecture 6:**

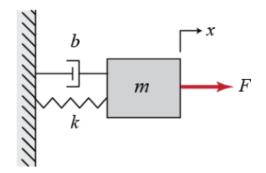
- Overview of Feedback & PID Control
- Example of PID Control System



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Example 1: Spring-Damper-Mass system



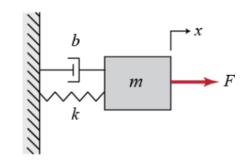
- Step 1 : Equation of Motion (EOM) of this system  $m\ddot{x} + b\dot{x} + kx = F$
- Step 2 : Taking Laplace transform of EOM  $(ms^2 + bs + k)X(s) = F(s)$
- Step 3 : TF btw the displacement {X(s)} and force {F(s)}

$$\frac{X(s)}{F(s)} = \frac{1}{ms^2 + bs + k}$$

#### Example 1: Spring-Damper-Mass system (cont'd)

Let suppose each parameter, and get TF

$$m = 1[kg]$$
 $b = 10[Ns/m]$ 
 $k = 20[N/m]$ 
 $M = 10[Ns/m]$ 
 $M = 10[N$ 

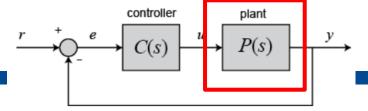


$$G(s) = \frac{1}{s^2 + 2\varsigma\omega_n s + {\omega_n}^2}$$

Objectives: Design PID controller to obtain

- 1. Rise time < 0.2 sec
- 2. Settling time < 0.5 sec
- 3. Maximum overshoot < 5%
- 4. No steady-state error < 0.01

# **Three-Term Controller: PID**



#### Example 1: Spring-Damper-Mass system (cont'd)

Check : Open-Loop Step Response

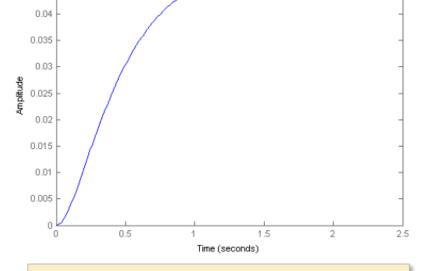


$$s = ti('s');$$
  
 $P = 1/(s^2 + 10*s + 20);$   
 $step(P)$ 

$$\frac{X(s)}{F(s)} = \frac{1}{s^2 + 10s + 20}$$



0.045



Step Response

#### **\*** Findings:

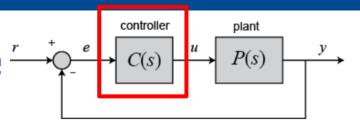
- Rise time is ~1.5sec
- Settling time is ~2sec
- Steady-state error is about 0.95



#### Design PID controller:

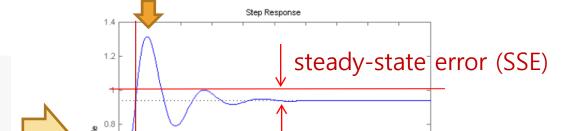
- reduce rise time (<0.2)</li>
- reduce settling time (<0.5)</li>
- eliminate SSE (<0.01)</li>

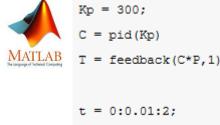
**❖** Example 1: Spring-Damper-Mas:



Case I : Proportional (P)

$$C(s) = K_p \quad P(s) = \frac{1}{s^2 + 10s + 20} \qquad \frac{X(s)}{F(s)} = \frac{C(s)P(s)}{1 + C(s)P(s)} = \frac{K_p}{s^2 + 10s + (20 + K_p)}$$







• Kp reduces rise time (1.5 → 0.1sec) and SSE (0.95 → 0.05), but overshoot (30%) !! Then, How to reduce overshoot ??

- Example 1: Spring-Damper-Mass system (cont'd)
  - Case II: Proportional + Derivative (PD)  $C(s)_{PD} = K_p + K_d s$

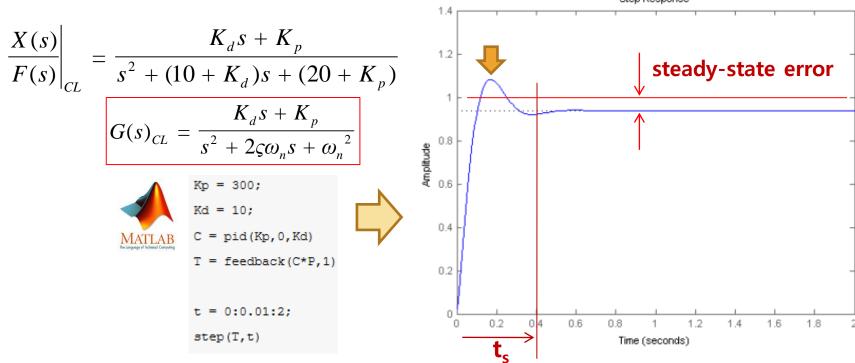
Step Response

Step Response

Steady-state error

$$0.8$$
 $0.8$ 
 $0.4$ 
 $0.2$ 
 $0.2$ 
 $0.4$ 
 $0.6$ 
 $0.8$ 
 $0.8$ 

Time (seconds)



■ PD reduces overshoot (30%  $\rightarrow$  15%) and settling time (1.0  $\rightarrow$  0.4sec) !! But, there is still the steady-state error (~0.05). How to eliminate SSE ??

- Example 1: Spring-Damper-Mass system (cont'd)
  - Case III: Proportional + Integral (PI)  $C(s)_{p_I} = K_{p_I} + \frac{K_i}{2}$

step(T,t)

$$\frac{X(s)}{F(s)}\Big|_{CL} = \frac{K_p s + K_i}{s^3 + 10s^2 + (K_p + 20)s + K_i}$$

$$Kp = 30;$$

$$Ki = 70;$$

$$C = pid(Kp, Ki)$$

$$T = feedback(C*P, 1)$$

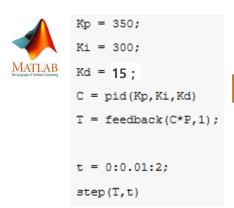
$$t = 0:0.01:2;$$

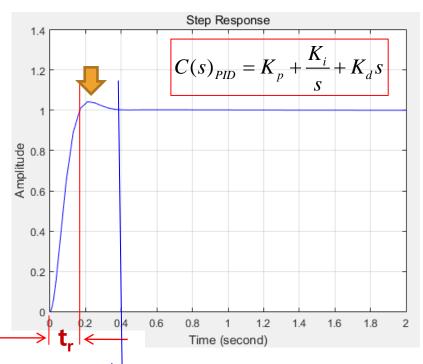
PI control eliminates the steady-state error  $(0.05 \rightarrow almost zero) !!$ But, rising time  $(0.1 \rightarrow 0.6)$  & settling time  $(0.4 \rightarrow 0.8)$  becomes longer !!

Time (seconds)

- Example 1: Spring-Damper-Mass system (cont'd)
  - Case IV : Proportional + Integral + Derivative (PID)

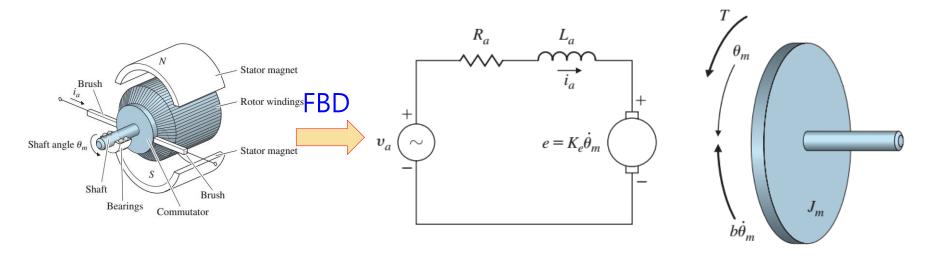
$$\frac{X(s)}{F(s)}\bigg|_{CL} = \frac{K_d s^2 + K_p s + K_i}{s^3 + (10 + K_d)s^2 + (20 + K_p)s + K_i}$$





- All goals are satisfied !!
  - 1. Fast rise time (t<sub>r</sub>)
- 2. Settling time
- □ less then 0.4 sec
- 3. Maximum overshoot less than 5%
- 4. No steady-state error almost zero (< 0.001)

#### Example 2: DC Motor – Speed Control



#### Part I: Mechanical motion

$$J_{m}\ddot{\theta}_{m} + b\dot{\theta}_{m} = K_{t}i_{a}.$$

$$\frac{\mathcal{L}}{I_{a}(s)} = \frac{K_{t}}{J_{m}s^{2} + bs}$$

#### Part II: Electric circuit equation

$$L_a \frac{di_a}{dt} + R_a i_a = v_a - K_e \dot{\theta}_m.$$

$$(L_a s + R_a) I_a(s) = V_a(s) - (K_e s \theta_m)(s)$$

#### **❖** Example 2: DC Motor – Speed Control (cont'd)

$$(L_a s + R_a) I_a(s) = V_a(s) - (K_e s \theta_m)(s)$$

$$I_a(s) = \frac{J_m s^2 + bs}{K_t} \theta(s)$$
Arranging
$$\frac{\Theta_m(s)}{V_a(s)} = \frac{K_t}{s[(J_m s + b)(L_a s + R_a) + K_t K_e]}.$$

But, practically  $L_a << J_m$  or  $R_a$ , Thus it is negligible !!

$$\frac{\Theta_m(s)}{V_a(s)} = \frac{\frac{K_t}{R_a}}{J_m s^2 + \left(b + \frac{K_t K_e}{R_a}\right) s} \qquad \text{Where,} \qquad \frac{\Omega(s)}{V_a(s)} = s \frac{\Theta_m(s)}{V_a(s)} = \frac{K}{\tau s + 1}.$$

$$K = \frac{K_t}{bR_a + K_t K_e}, \qquad \text{Velocity Model}$$

$$T = \frac{R_a J_m}{bR_a + K_t K_e}.$$

- Example 2: DC Motor Speed Control (cont'd)
  - Transfer function of DC motor speed control (output)

$$\frac{\Omega(s)}{V_a(s)} = s \frac{\Theta_m(s)}{V_a(s)} = \frac{K}{\tau s + 1}.$$

$$K = \frac{K_t}{bR_a + K_t K_e},$$

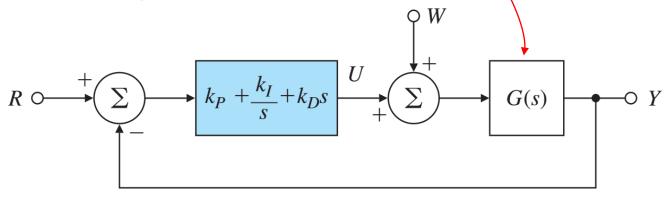
$$\tau = \frac{R_a J_m}{bR_a + K_t K_e}.$$

$$J_m = 1.13 \times 10^{-2} \qquad b = 0.028 \text{ N·m·sec/rad}, \quad L_a = 10^{-1} \text{ H},$$

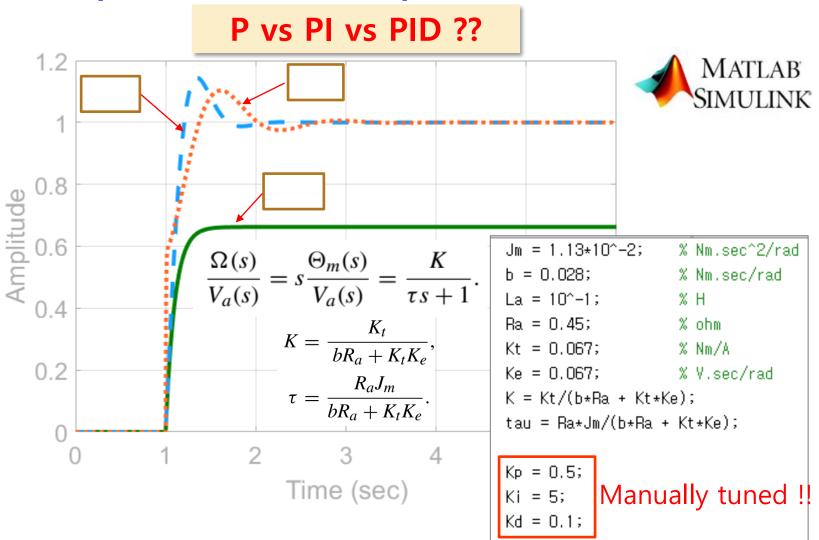
$$N \cdot m \cdot \sec^2 / \text{rad},$$

$$R_a = 0.45 \Omega, \qquad K_t = 0.067 \text{ N·m/amp}, \qquad K_e = 0.067 \text{ V·sec/rad}.$$

Block diagram of closed-loop with PID controller



#### Example 2: DC Motor – Speed Control (cont'd)



### PID Controller on Discrete Time

#### Implementation

- The analysis for designing a digital implementation of a PID controller in a microcontroller (MCU) or FPGA device requires the standard form of the PID controller to be discretized.
- The integral term is discretized, with a sampling time  $\Delta t$ , as follows,

$$\int_{0}^{t_{k}} e(t)dt = \sum_{k=1}^{n} e(t_{k}) \Delta t$$

The derivative term is approximated as,

$$\frac{de(t)}{dt} = \frac{e(t_k) - e(t_{k-1})}{\Delta t}$$

It is called "backward finite difference" (It is the nature of the feedback)



FPGA:FieldProgrammableGateArray

https://en.wikipedia.org/wiki/Field-programmable\_gate\_array

#### Implementation (cont'd)

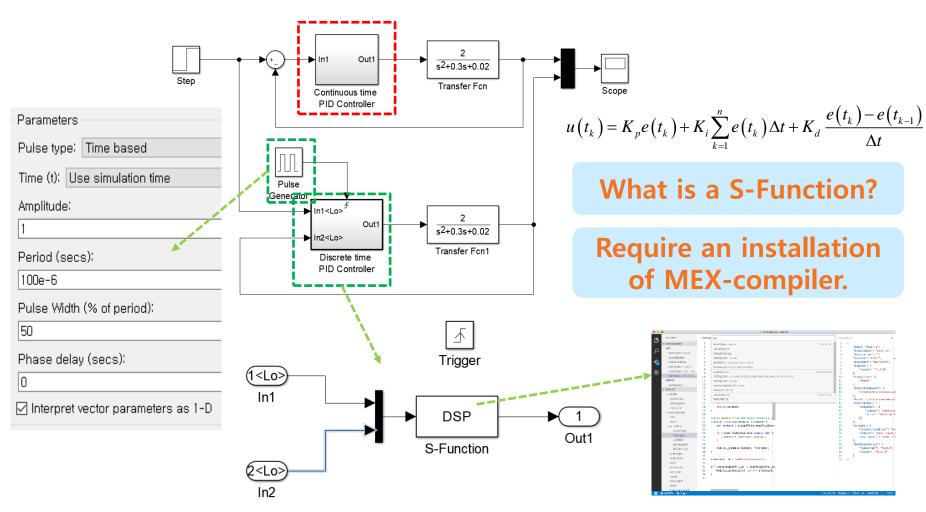
Thus, a PID algorithm for implementation of the **discretized PID controller in a MCU** is obtained by differentiating y(t), using the numerical definitions of the first and second derivative and solving for  $y(t_k)$  and finally obtaining:

$$u(t) = K_p e(t) + K_i \int e(t)dt + K_d \frac{de(t)}{dt}$$



$$u(t_k) = K_p e(t_k) + K_i \sum_{k=1}^n e(t_k) \Delta t + K_d \frac{e(t_k) - e(t_{k-1})}{\Delta t}$$

### **Example:** Implementation with C-code



#### PID pseudo code for the discrete-time,

$$u(t_k) = K_p e(t_k) + K_i \sum_{k=1}^n e(t_k) \Delta t + K_d \frac{e(t_k) - e(t_{k-1})}{\Delta t}$$

```
error_prior = 0
integral = 0
K_P = Some value you need to come up (see tuning section below)
K_1 = Some value you need to come up (see tuning section below)
K_D = Some value you need to come up (see tuning section below)
while(1) {
 error = desired value - actual value
 integral = integral + (error*iteration_time)
 derivative = (error - error_prior)/iteration_time
 output = K_P*error + K_I*integral + K_D*derivative + bias
 error_prior = error
 sleep(iteration_time)
```

PID C++ code for the discrete-time: PID + Saturation

```
double PIDImpl::calculate( double setpoint, double pv )
                                        Measured output (or filtered output)
   // Calculate error
   double error = setpoint - pv
   // Proportional term
   double Pout = Kp * error;
   // Integral term
   integral += error * _dt;
   double Iout = Ki * integral;
   // Derivative term
   double derivative = (error - pre error) / dt;
   double Dout = Kd * derivative;
   // Calculate total output
   double output = Pout + Iout + Dout;
   // Restrict to max/min
                                    Saturation (Restriction)
   if( output > _max )
      output = max;
   else if( output < min )</pre>
       output = min;
   // Save error to previous error
                                                        https://gist.github.com/bradley219/5373998
   pre error = error;
   return output;
```

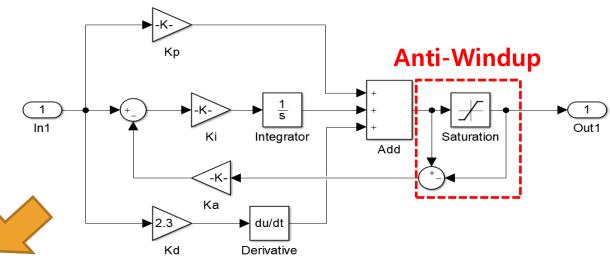
**Example:** PID with Anti-Windup in C-code (cont'd)

With Anti-Windup

C-language code

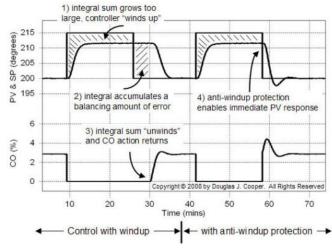
```
#define TS 100e-6
#define Out_MAX 10.0

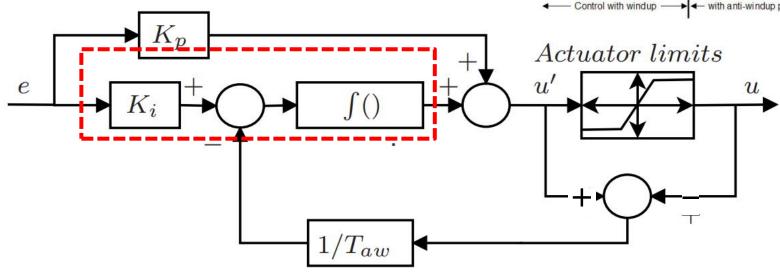
double err;
double err_old;
double out;
double kp;
double ki;
double kd;
double integ;
double ref;
double feed;
double out_sat;
```



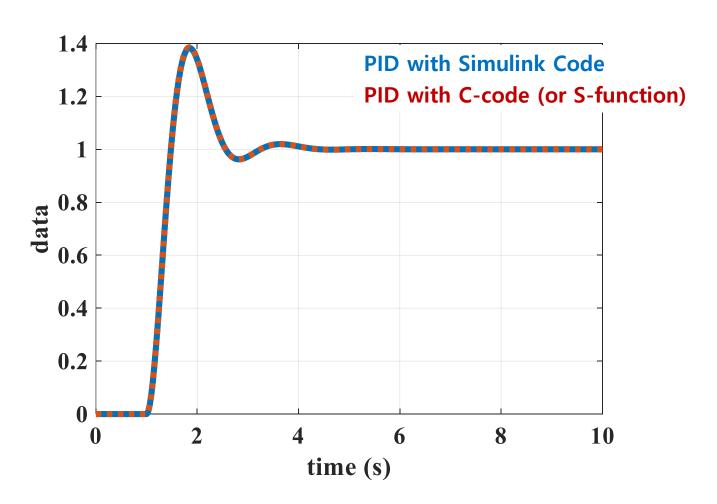
```
err = 0.0;
kp = 10.01;
ki = 11.12 * TS;
kd = 2.3 / TS;
ka = 1/(6*kp);
err = ref - feed;
integ += err - (ka*(out-out_sat));
out = (kp*err) + (ki*integ) + (kd*(err-err_old));
out_sat = (out > Out_MAX) ? Out_MAX : (out < -Out_MAX) ? -Out_MAX : out;
err_old = err;
```

- Example: Implementation with C-code (cont'd)
  - Anti-Windup for the Integral Control





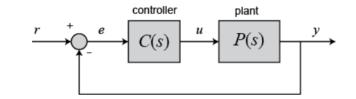
#### Example: Implementation with C-code (cont'd)



# **Summary: PID Controller**

#### Summary of PID

$$u(t) = K_p e(t) + K_i \int e(t)dt + K_d \frac{de(t)}{dt}$$



#### Proportional (P)

- reduce the rise time and reduce the steady-state error (SSE)
- but, never eliminate the SSE (e.g., certain amount of error will be actuated by input constraints)

#### Integral (I)

- eliminate the SSE for a constant or step input, (by accumulating error between r(t) and y(t)
- but, worse transient response(settling time or overshoot)

#### Derivative (D)

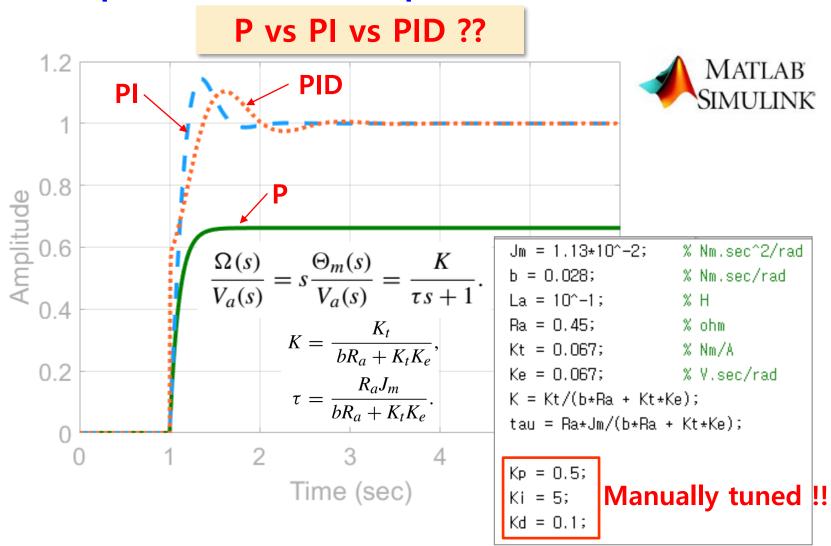
- Improve the transient response (settling time), reduce the overshoot
- but, amplify higher frequency measurement (sensor) or process noise

### **Limitations of PID & Modifications**

- PID is <u>Feedback</u> !! Thus, response is reactive!!
  - ✓ Feed-forward control with the knowledge about the process model will be able to increase responding performance!!
- PID has <u>constant</u> parameters (or gains) !!
  - Thus, if system variations exist, performance gets worse!!
  - ✓ Changing the control parameters (i.e., gains) based on the process variations (e.g., gain scheduling and adaptive control)
- PID control is <u>no direct knowledge</u> of the process !!
  - ✓ Identify the process model and optimize controller gain (e.g., by using Matlab tool(e.g., PID tuner), auto-tuning method, model matching condition and etc)
- Integral wind-up & High-frequency noise amplification on Derivative !!
  - (1) Anti-windup schemes (e.g., temporally stopping integral action)
  - (2) Low-pass-filter !! but, slow response is following.

# Three-Term Controller: PID Control (Again)

#### Example 2: DC Motor – Speed Control (cont'd)



# Summary

#### **Summary:**

- Transfer function of open- & closed-loop system.
- PID control design and how it works.
- PID Control with C-code.
- Limitation and modification of PID.