```
function [maxHeight,maxRange,tMaxHeight,tMaxRange] = AirResistanceFunction(params)
   % Extract parameters
   % Here, I'm setting up the input parameters from the `params` structure.
   \% `theta` is the launch angle, `mu` is the coefficient of air resistance,
   \% and I'm using \ \hat{\ }g\ \hat{\ } for gravity and \ \hat{\ }vInitial\ \hat{\ } for the initial velocity.
   theta = params.Theta; % Launch angle in degrees
                        % Air resistance coefficient
   mu = params.Mu;
                          % Gravitational acceleration in m/s<sup>2</sup>
   g = 9.81;
   vInitial = 300;
                          % Initial velocity in m/s
   % Define time bounds
   \ensuremath{\text{\%}} I'm calculating the total flight time for the projectile without air resistance
   \% as an estimate for the simulation duration. I added 1 second to ensure full capture.
   tInitial = 0;
   tFinal = 2 * vInitial * sind(theta) / g + 1;
   % Define initial conditions
   \% Initial position is (0,0) with horizontal and vertical velocity components split
   % using cosd and sind for the angle in degrees. This ensures correct trigonometric calculations.
   yInitial = [0; 0; vInitial * cosd(theta); vInitial * sind(theta)];
   % Choose the air resistance model
   % Here, I'm using a `switch` to define the differential equations based on the air resistance model.
   % For no resistance, forces only include gravity. For "Stokes" and "Newton," I include velocity-dependent terms.
   switch params.Model
        case "None"
           % No air resistance
           dydt = @(t,y) [y(3); y(4); 0; -g];
        case "Stokes
            % Stokes drag: air resistance proportional to velocity
           dydt = @(t,y) [y(3); y(4); -mu * y(3); -g - mu * y(4)];
        case "Newton'
            \ensuremath{\mathrm{\%}} Newton drag: air resistance proportional to velocity squared
            dydt = @(t,y) [y(3); y(4); -mu * y(3) * sqrt(y(3)^2 + y(4)^2); ...
                           -g - mu * y(4) * sqrt(y(3)^2 + y(4)^2);
        otherwise
            \% Error handling for invalid models
            error("Invalid air resistance model");
    end
   % Solve for maximum height
   % I'm using `ode45` to integrate the equations until the projectile reaches the peak.
    options = odeset('Events', @endOfAscent); % Stop integration at the peak
    [\tt \sim, yout, te, ye, \tt \sim] = ode45(dydt, [tInitial tFinal], yInitial, options);
                        % Time to reach max height
    tMaxHeight = te;
   maxHeight = ye(2);
                        % Maximum height
   % Solve for maximum range
   % After reaching the peak, I update the initial conditions and continue integration until the projectile hits the ground.
   tInitial = te;
   yInitial = ye';
   options = odeset('Events', @endOfDescent); % Stop integration when it hits the ground
   [~, y, te, ye, ~] = ode45(dydt, [tInitial tFinal], yInitial, options);
   yout = [yout; y(2:end,:)]; % Combine the results
   tMaxRange = te;
                              % Time to reach max range
   maxRange = ye(1);
                              % Maximum range
   % Plotting the results
   % I want to visualize the trajectory and compare it to an ideal parabolic path.
    figure(Name="Projectile Trajectory");
   hold on:
   % Plot the actual trajectory
   plot(yout(:,1), yout(:,2), 'LineWidth', 2);
   \% Plot the parabolic path for comparison
   X = maxRange * (0:0.05:1);
   Y = 4 * maxHeight * X .* (maxRange - X) / maxRange^2;
   plot(X, Y, '-.');
    % Add titles, labels, and a legend for clarity
   title("Comparison of Trajectory and Parabolic Path");
   xlabel("Horizontal Distance (m)");
   ylabel("Vertical Distance (m)");
   legend("Trajectory", "Parabolic Path");
   axis tight;
   hold off;
end
% Helper function: Detect when the projectile reaches its peak
function [value, isterminal, direction] = endOfAscent(~, y)
```

```
% I stop integration when the vertical velocity (y(4)) is zero.
   value = y(4);
                         % Vertical velocity
    isterminal = 1;
                         % Stop the integration
    direction = 0;
                         \% Detect zero crossing in any direction
% Helper function: Detect when the projectile hits the ground
function [value, isterminal, direction] = endOfDescent(~, y)
   \% I stop integration when the height (y(2)) is zero.
   value = y(2);
                         % Height
   isterminal = 1;
                         % Stop the integration
    direction = 0;
                        % Detect zero crossing in any direction
params.Theta = 45;
                        % Launch angle in degrees
params.Mu = 0.02;
                         % Air resistance coefficient
params.Model = "Stokes"; % Air resistance model: "None", "Stokes", or "Newton"
[maxHeight, maxRange, tMaxHeight, tMaxRange] = AirResistanceFunction(params);
```

Comparison of Trajectory and Parabolic Path

