

Quadrotor with a Cable-Suspended Load

July 18, 2013

$m_Q, m_L \in \mathbb{R}$	Mass of quadrotor, load
$J_Q \in \mathbb{R}$	Inertia of quadrotor
$f \in \mathbb{R}$	Magnitude of thrust for quadrotor
$M \in \mathbb{R}$	Magnitude of moment for quadrotor in body frame
$l \in \mathbb{R}$	Length of suspension cable
$T \in \mathbb{R}$	Magnitude of tension in cable
$\mathbf{x}_Q, \mathbf{x}_L \in \mathbb{R}^2$	Position vector of center of mass of quadrotor, load in inertial frame
$\mathbf{v}_Q, \mathbf{v}_L \in \mathbb{R}^2$	Velocity vector of center of mass of quadrotor, load in inertial frame
$\mathbf{p} \in S^1$	Unit vector from quadrotor to load, $\mathbf{p} = [\sin(\phi_L) \quad -\cos(\phi_L)]^T$
$\mathbf{R} \in SO(2)$	Rotation matrix of quadrotor from body to inertial frame, $\mathbf{R} = [\cos(\phi_Q) \quad -\sin(\phi_Q); \sin(\phi_Q) \quad \cos(\phi_Q)]$
$\mathbf{e}_2, \mathbf{e}_3$	Axes of the inertial frame, $\mathbf{e}_2 = [1 \ 0]^T$, $\mathbf{e}_3 = [0 \ 1]^T$
$\mathbf{b}_2, \mathbf{b}_3$	Axes of quadrotor body frame, $\mathbf{b}_2 = \mathbf{R}\mathbf{e}_2$, $\mathbf{b}_3 = \mathbf{R}\mathbf{e}_3$
$\phi_Q \in (-\pi, \pi]$	Angle of quadrotor counter-clockwise from horizontal
$\phi_L \in (-\pi, \pi]$	Angle of load counter-clockwise from vertical
$\dot{\phi}_Q, \dot{\phi}_L \in \mathbb{R}$	Angular velocity of the quadrotor, load

1 Equations of Motion

$$\Sigma : \begin{cases} \dot{\mathbf{x}}_1 = f_1(\mathbf{x}_1) + g_1(\mathbf{x}_1)\mathbf{u}_1, & \mathbf{x}_1 \notin \mathcal{S}_1 \\ \mathbf{x}_2^+ = \Delta_1(\mathbf{x}_1^-), & \mathbf{x}_1^- \in \mathcal{S}_1 \\ \dot{\mathbf{x}}_2 = f_2(\mathbf{x}_2) + g_2(\mathbf{x}_2)\mathbf{u}_2, & \mathbf{x}_2 \notin \mathcal{S}_2 \\ \mathbf{x}_1^+ = \Delta_2(\mathbf{x}_2^-), & \mathbf{x}_2^- \in \mathcal{S}_2 \end{cases}$$

$$\Sigma : \begin{cases} \dot{\mathbf{x}}_1 = \begin{bmatrix} \dot{\mathbf{x}}_L \\ \dot{\mathbf{v}}_L \\ \dot{\phi}_L \\ \dot{\phi}_L \\ \dot{\phi}_Q \\ \dot{\phi}_Q \end{bmatrix} = \begin{bmatrix} \mathbf{v}_L \\ \frac{-m_Q l \dot{\phi}_L^2}{(m_Q + m_L)} \mathbf{p} - g \mathbf{e}_3 \\ \dot{\phi}_L \\ 0 \\ \dot{\phi}_Q \\ 0 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ \frac{\cos(\phi_Q - \phi_L)}{(m_Q + m_L)} \mathbf{p} & 0 \\ 0 & 0 \\ \frac{\sin(\phi_Q - \phi_L)}{m_Q l} & 0 \\ 0 & 0 \\ 0 & \frac{1}{J_Q} \end{bmatrix} \begin{bmatrix} f \\ M \end{bmatrix}, & \mathbf{x}_1 \notin \{\mathbf{x}_1 \mid T \equiv \|m_L(\dot{\mathbf{v}}_L + g \mathbf{e}_3)\| = 0\} \\ \\ \mathbf{x}_2^+ = \begin{bmatrix} \mathbf{x}_L^+ \\ \mathbf{v}_L^+ \\ \mathbf{x}_Q^+ \\ \mathbf{v}_Q^+ \\ \phi_Q^+ \\ \phi_Q^+ \end{bmatrix} = \begin{bmatrix} \mathbf{x}_L^- \\ \mathbf{v}_L^- \\ \mathbf{x}_L^- - l \mathbf{p}^- \\ \mathbf{v}_L^- - l \dot{\mathbf{p}}^- \\ \phi_Q^- \\ \phi_Q^- \end{bmatrix}, & \mathbf{x}_1^- \in \{\mathbf{x}_1 \mid T = 0\} \\ \\ \dot{\mathbf{x}}_2 = \begin{bmatrix} \dot{\mathbf{x}}_L \\ \dot{\mathbf{v}}_L \\ \dot{\mathbf{x}}_Q \\ \dot{\mathbf{v}}_Q \\ \dot{\phi}_Q \\ \dot{\phi}_Q \end{bmatrix} = \begin{bmatrix} \mathbf{v}_L \\ -g \mathbf{e}_3 \\ \mathbf{v}_Q \\ -g \mathbf{e}_3 \\ \dot{\phi}_Q \\ 0 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 0 & 0 \\ \frac{1}{m_Q} \mathbf{b}_3 & 0 \\ 0 & 0 \\ 0 & \frac{1}{J_Q} \end{bmatrix} \begin{bmatrix} f \\ M \end{bmatrix}, & \mathbf{x}_2 \notin \{\mathbf{x}_2 \mid \|\mathbf{x}_Q - \mathbf{x}_L\| = L\} \\ \\ \mathbf{x}_1^+ = \begin{bmatrix} \mathbf{x}_L^+ \\ \mathbf{v}_L^+ \\ \phi_L^+ \\ \phi_L^+ \\ \phi_Q^+ \\ \phi_Q^+ \end{bmatrix} = \begin{bmatrix} \mathbf{x}_L^- \\ \frac{m_L \mathbf{v}_L^- + m_Q \mathbf{v}_Q^-}{(m_L + m_Q)} \\ \cos^{-1}\left(-\frac{\mathbf{x}_L^- - \mathbf{x}_Q^-}{l} \cdot \mathbf{e}_3\right) \\ 0 \\ \phi_Q^- \\ \phi_Q^- \end{bmatrix}, & \mathbf{x}_2^- \in \{\mathbf{x}_2 \mid \|\mathbf{x}_Q - \mathbf{x}_L\| = L\} \end{cases}$$

1.1 When cable is taut

Let $\mathbf{e}_2 = [1 \ 0]^T$ and $\mathbf{e}_3 = [0 \ 1]^T$ be unit vectors in the plane; let \mathbf{e}_1 be a unit vector out of the plane

Constraint: $\mathbf{x}_Q = \mathbf{x}_L - l\mathbf{p}$

Applied forces: $\mathbf{f}_1 = f\mathbf{b}_3$ at $\mathbf{r}_1 = \mathbf{x}_Q = \mathbf{x}_L - l\mathbf{p}$, $\mathbf{f}_2 = M\mathbf{e}_1$ at $\mathbf{r}_2 = \phi_Q\mathbf{e}_1$

Coordinates: $\mathbf{q} = [\mathbf{x}_L \ \phi_L \ \phi_Q]^T$

Forces: $Q_j = \sum_{i=0}^n \mathbf{f}_i \cdot \frac{\partial \mathbf{r}_i}{\partial q_j}$

$$Q_{\mathbf{x}_L} = f\mathbf{b}_3 \cdot \frac{\partial(\mathbf{x}_L - l\mathbf{p})}{\partial \mathbf{x}_L} + M\mathbf{e}_1 \cdot \frac{\partial \phi_Q \mathbf{e}_1}{\partial \mathbf{x}_L}$$

$$= f\mathbf{b}_3$$

$$Q_{\phi_L} = f\mathbf{b}_3 \cdot \frac{\partial(\mathbf{x}_L - l\mathbf{p})}{\partial \phi_L} + M\mathbf{e}_1 \cdot \frac{\partial \phi_Q \mathbf{e}_1}{\partial \phi_L}$$

$$= -f\mathbf{b}_3 \cdot L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix}$$

$$= Lf \sin(\phi_Q - \phi_L)$$

$$Q_{\phi_Q} = f\mathbf{b}_3 \cdot \frac{\partial(\mathbf{x}_L - l\mathbf{p})}{\partial \phi_Q} + M\mathbf{e}_1 \cdot \frac{\partial \phi_Q \mathbf{e}_1}{\partial \phi_Q}$$

$$= M$$

$$\mathcal{L} = \mathcal{T} - \mathcal{U}$$

$$\mathcal{T} = \frac{1}{2}m_Q \mathbf{v}_Q \cdot \mathbf{v}_Q + \frac{1}{2}m_L \mathbf{v}_L \cdot \mathbf{v}_L + \frac{1}{2}J_Q \dot{\phi}_Q^2$$

$$= \frac{1}{2}m_Q(\mathbf{v}_L - l\dot{\mathbf{p}}) \cdot (\mathbf{v}_L - l\dot{\mathbf{p}}) + \frac{1}{2}m_L \mathbf{v}_L \cdot \mathbf{v}_L + \frac{1}{2}J_Q \dot{\phi}_Q^2$$

$$= \frac{1}{2}(m_Q + m_L)(\mathbf{v}_L \cdot \mathbf{v}_L) - m_Q(\mathbf{v}_L \cdot l\dot{\mathbf{p}}) + \frac{1}{2}m_Q l^2(\dot{\mathbf{p}} \cdot \dot{\mathbf{p}}) + \frac{1}{2}J_Q \dot{\phi}_Q^2$$

$$= \frac{1}{2}(m_Q + m_L)(\mathbf{v}_L \cdot \mathbf{v}_L) - m_Q(\mathbf{v}_L \cdot l\dot{\phi}_L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix}) + \frac{1}{2}m_Q l^2 \dot{\phi}_L^2 + \frac{1}{2}J_Q \dot{\phi}_Q^2$$

$$\mathcal{U} = m_Q g \mathbf{e}_3 \cdot \mathbf{x}_Q + m_L g \mathbf{e}_3 \cdot \mathbf{x}_L$$

$$= m_Q g \mathbf{e}_3 \cdot (\mathbf{x}_L - l\mathbf{p}) + m_L g \mathbf{e}_3 \cdot \mathbf{x}_L$$

$$= (m_Q + m_L)g \mathbf{e}_3 \cdot \mathbf{x}_L + m_Q l g \mathbf{e}_3 \cdot \mathbf{p}$$

$$= (m_Q + m_L)g \mathbf{e}_3 \cdot \mathbf{x}_L - m_Q l g \cos(\phi_L)$$

$$\mathcal{L} = \frac{1}{2}(m_Q + m_L)(\mathbf{v}_L \cdot \mathbf{v}_L) - m_Q(\mathbf{v}_L \cdot l\dot{\phi}_L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix}) + \frac{1}{2}m_Q l^2 \dot{\phi}_L^2 + \frac{1}{2}J_Q \dot{\phi}_Q^2$$

$$- (m_Q + m_L)g \mathbf{e}_3 \cdot \mathbf{x}_L + m_Q l g \cos(\phi_L)$$

$$\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \mathbf{v}_L} - \frac{\partial \mathcal{L}}{\partial \mathbf{x}_L} = f\mathbf{b}_3$$

$$\frac{d}{dt} \left((m_Q + m_L)\mathbf{v}_L - m_Q l \dot{\phi}_L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} \right) - (m_Q + m_L)g \mathbf{e}_3 = f\mathbf{b}_3$$

$$(m_Q + m_L)\dot{\mathbf{v}}_L - m_Q l \ddot{\phi}_L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + m_Q l \dot{\phi}_L^2 \begin{bmatrix} -\sin(\phi_L) \\ \cos(\phi_L) \end{bmatrix} - (m_Q + m_L)g \mathbf{e}_3 = f\mathbf{b}_3$$

$$\begin{aligned}
\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\phi}_L} - \frac{\partial \mathcal{L}}{\partial \phi_L} &= Lf \sin(\phi_Q - \phi_L) \\
\frac{d}{dt} \left(-m_Q l \mathbf{v}_L \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + m_Q l^2 \dot{\phi}_L \right) - \left(m_Q l \dot{\phi}_L \mathbf{v}_L \cdot \begin{bmatrix} \sin \phi_L \\ -\cos(\phi_L) \end{bmatrix} + m_Q l g \sin(\phi_L) \right) &= Lf \sin(\phi_Q - \phi_L) \\
-m_Q l \dot{\mathbf{v}}_L \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + m_Q l \dot{\phi}_L \mathbf{v}_L \cdot \begin{bmatrix} \sin \phi_L \\ -\cos(\phi_L) \end{bmatrix} + m_Q l^2 \ddot{\phi}_L & \\
-m_Q l \dot{\phi}_L \mathbf{v}_L \cdot \begin{bmatrix} \sin \phi_L \\ -\cos(\phi_L) \end{bmatrix} - m_Q l g \sin(\phi_L) &= Lf \sin(\phi_Q - \phi_L) \\
-m_Q l \dot{\mathbf{v}}_L \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + m_Q l^2 \ddot{\phi}_L - m_Q l g \sin(\phi_L) &= Lf \sin(\phi_Q - \phi_L)
\end{aligned}$$

$$\begin{aligned}
\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\phi}_Q} - \frac{\partial \mathcal{L}}{\partial \phi_Q} &= M \\
\frac{d}{dt} (J_Q \dot{\phi}_Q) &= M \\
J_Q \ddot{\phi}_Q &= M
\end{aligned}$$

Decoupling equations:

$$\begin{aligned}
(m_Q + m_L)(\dot{\mathbf{v}}_L + g\mathbf{e}_3) &= (-f \cos(\phi_Q - \phi_L) - m_Q l \dot{\phi}_L^2) \mathbf{p} \\
m_Q l \ddot{\phi}_L &= f \sin(\phi_Q - \phi_L) \\
J_Q \ddot{\phi}_Q &= M
\end{aligned}$$

$$\begin{aligned}
\mathbf{x}_1 &= [\mathbf{x}_L \quad \mathbf{v}_L \quad \phi_L \quad \dot{\phi}_L \quad \phi_Q \quad \dot{\phi}_Q]^T \\
\dot{\mathbf{x}}_1 &= \begin{bmatrix} \dot{\mathbf{x}}_L \\ \dot{\mathbf{v}}_L \\ \dot{\phi}_L \\ \ddot{\phi}_L \\ \dot{\phi}_Q \\ \ddot{\phi}_Q \end{bmatrix} = \begin{bmatrix} \mathbf{v}_L \\ \frac{(f \cos(\phi_Q - \phi_L) - m_Q l \dot{\phi}_L^2)}{(m_Q + m_L)} \mathbf{p} - g\mathbf{e}_3 \\ \phi_L \\ \frac{f \sin(\phi_Q - \phi_L)}{m_Q l} \\ \phi_Q \\ \frac{M}{J_Q} \end{bmatrix}
\end{aligned}$$

Note the tension force in the cable can be explicitly calculated by taking Newton's Law on the body of the load:

$$\begin{aligned}
\mathbf{F} &= m_L \dot{\mathbf{v}}_L \\
-T\mathbf{p} - m_L g\mathbf{e}_3 &= m_L \dot{\mathbf{v}}_L \\
-T\mathbf{p} &= m_L \dot{\mathbf{v}}_L + m_L g\mathbf{e}_3 \\
\|T\| &= \|m_L \dot{\mathbf{v}}_L + m_L g\mathbf{e}_3\|
\end{aligned}$$

1.2 When cable is slack

Let $\mathbf{e}_2 = [1 \ 0]^T$ and $\mathbf{e}_3 = [0 \ 1]^T$ be unit vectors in the plane; let \mathbf{e}_1 be a unit vector out of the plane
 Applied forces: $\mathbf{f}_1 = f\mathbf{b}_3$ at $\mathbf{r}_1 = \mathbf{x}_Q$, $\mathbf{f}_2 = M\mathbf{e}_1$ at $\mathbf{r}_2 = \phi_Q\mathbf{e}_1$

Coordinates: $\mathbf{q} = [\mathbf{x}_L \ \mathbf{x}_Q \ \phi_Q]^T$

$$\text{Forces: } Q_j = \sum_{i=0}^n \mathbf{f}_i \cdot \frac{\partial \mathbf{r}_i}{\partial q_j}$$

$$Q_{\mathbf{x}_L} = f\mathbf{b}_3 \cdot \frac{\partial \mathbf{x}_Q}{\partial \mathbf{x}_L} + M\mathbf{e}_1 \cdot \frac{\partial \phi_Q \mathbf{e}_1}{\partial \mathbf{x}_L}$$

$$= 0$$

$$Q_{\mathbf{x}_Q} = f\mathbf{b}_3 \cdot \frac{\partial \mathbf{x}_Q}{\partial \mathbf{x}_Q} + M\mathbf{e}_1 \cdot \frac{\partial \phi_Q \mathbf{e}_1}{\partial \mathbf{x}_Q}$$

$$= f\mathbf{b}_3$$

$$Q_{\phi_Q} = f\mathbf{b}_3 \cdot \frac{\partial (\mathbf{x}_L - l\mathbf{p})}{\partial \phi_Q} + M\mathbf{e}_1 \cdot \frac{\partial \phi_Q \mathbf{e}_1}{\partial \phi_Q}$$

$$= M$$

$$\mathcal{L} = \mathcal{T} - \mathcal{U}$$

$$\mathcal{T} = \frac{1}{2}m_Q \mathbf{v}_Q \cdot \mathbf{v}_Q + \frac{1}{2}m_L \mathbf{v}_L \cdot \mathbf{v}_L + \frac{1}{2}J_Q \dot{\phi}_Q^2$$

$$\mathcal{U} = m_Q g \mathbf{e}_3 \cdot \mathbf{x}_Q + m_L g \mathbf{e}_3 \cdot \mathbf{x}_L$$

$$\mathcal{L} = \frac{1}{2}m_Q \mathbf{v}_Q \cdot \mathbf{v}_Q + \frac{1}{2}m_L \mathbf{v}_L \cdot \mathbf{v}_L + \frac{1}{2}J_Q \dot{\phi}_Q^2 - m_Q g \mathbf{e}_3 \cdot \mathbf{x}_Q - m_L g \mathbf{e}_3 \cdot \mathbf{x}_L$$

$$\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \mathbf{v}_L} - \frac{\partial \mathcal{L}}{\partial \mathbf{x}_L} = 0$$

$$\frac{d}{dt} (m_L \mathbf{v}_L) + m_L g \mathbf{e}_3 = 0$$

$$m_L \dot{\mathbf{v}}_L + m_L g \mathbf{e}_3 = 0$$

$$\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \mathbf{v}_Q} - \frac{\partial \mathcal{L}}{\partial \mathbf{x}_Q} = f\mathbf{b}_3$$

$$\frac{d}{dt} (m_Q \mathbf{v}_Q) + m_Q g \mathbf{e}_3 = f\mathbf{b}_3$$

$$m_Q \dot{\mathbf{v}}_Q + m_Q g \mathbf{e}_3 = f\mathbf{b}_3$$

$$\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\phi}_Q} - \frac{\partial \mathcal{L}}{\partial \phi_Q} = M$$

$$\frac{d}{dt} (J_Q \dot{\phi}_Q) = M$$

$$J_Q \ddot{\phi}_Q = M$$

$$\mathbf{x}_2 = [\mathbf{x}_L \quad \mathbf{v}_L \quad \mathbf{x}_Q \quad \mathbf{v}_Q \quad \phi_Q \quad \dot{\phi}_Q]^T$$

$$\dot{\mathbf{x}}_2 = \begin{bmatrix} \dot{\mathbf{x}}_L \\ \dot{\mathbf{v}}_L \\ \dot{\mathbf{x}}_Q \\ \dot{\mathbf{v}}_Q \\ \dot{\phi}_Q \\ \ddot{\phi}_Q \end{bmatrix} = \begin{bmatrix} \mathbf{v}_L \\ -g\mathbf{e}_3 \\ \mathbf{v}_Q \\ \frac{f}{m_Q}\mathbf{b}_3 - g\mathbf{e}_3 \\ \dot{\phi}_Q \\ \frac{M}{J_Q} \end{bmatrix}$$

2 Differential Flatness

2.1 \mathbf{x}_1 system:

2.1.1 Differential Flatness

Recall the equations of motion:

$$\begin{aligned}(m_Q + m_L)(\dot{\mathbf{v}}_L + g\mathbf{e}_3) &= (-f \cos(\phi_Q - \phi_L) - m_Q l \dot{\phi}_L^2)\mathbf{p} \\ m_Q l \ddot{\phi}_L &= f \sin(\phi_Q - \phi_L) \\ J_Q \ddot{\phi}_Q &= M\end{aligned}$$

Note the tension force in the cable can be explicitly calculated by taking Newton's Law on the body of the load:

$$\begin{aligned}\mathbf{F} &= m_L \dot{\mathbf{v}}_L \\ -T\mathbf{p} - m_L g\mathbf{e}_3 &= m_L \dot{\mathbf{v}}_L \\ -T\mathbf{p} &= m_L \dot{\mathbf{v}}_L + m_L g\mathbf{e}_3 \\ \|T\| &= \|m_L \dot{\mathbf{v}}_L + m_L g\mathbf{e}_3\|\end{aligned}$$

Choose flat outputs $\mathbf{y} = [\mathbf{x}_L]^T = [y_L \ z_L]^T$

Derive $\dot{y}_L = v_{yL}$, $\dot{z}_L = v_{zL}$, and all higher derivatives from differentiation of y_L , z_L

From equation of motion:

$$\begin{aligned}T\mathbf{p} &= -(m_L \ddot{\mathbf{x}}_L + m_L g\mathbf{e}_3) \\ T &= \|m_L \ddot{\mathbf{x}}_L + m_L g\mathbf{e}_3\| \\ &= m_L (\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{\frac{1}{2}} \\ \mathbf{p} &= \frac{-(m_L \ddot{\mathbf{x}}_L + m_L g\mathbf{e}_3)}{T} \\ &= -\frac{m_L}{T} \begin{bmatrix} \ddot{y}_L \\ \ddot{z}_L + g \end{bmatrix} = \begin{bmatrix} \sin(\phi_L) \\ -\cos(\phi_L) \end{bmatrix} \\ \phi_L &= \tan^{-1} \left(\frac{\mathbf{p} \cdot \mathbf{e}_2}{-\mathbf{p} \cdot \mathbf{e}_3} \right)\end{aligned}$$

Differentiating this equation of motion:

$$T\dot{\mathbf{p}} + \dot{T}\mathbf{p} = -m_L\ddot{\mathbf{x}}_L$$

$$\dot{\mathbf{p}} = -\frac{1}{T}(m_L\ddot{\mathbf{x}}_L + \dot{T}\mathbf{p})$$

Where:

$$\dot{T} = m_L \frac{\ddot{y}_L \ddot{y}_L + (\ddot{z}_L + g) \ddot{z}_L}{(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{\frac{1}{2}}}$$

From $\dot{\mathbf{p}}$, we can find the state $\dot{\phi}_L$:

$$\begin{aligned} \dot{\mathbf{p}} &= \dot{\phi}_L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} \\ \dot{\phi}_L &= -\frac{1}{T}(m_L\ddot{\mathbf{x}}_L + \dot{T}\mathbf{p}) \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} \\ &= -\frac{m_L}{T}(\ddot{y}_L \cos(\phi_L) + \ddot{z}_L \sin(\phi_L)) \\ &= -\frac{m_L}{T} \left(\frac{m_L \ddot{y}_L (\ddot{z}_L + g)}{T} - \frac{m_L \ddot{z}_L \ddot{y}_L}{T} \right) \\ &= -\frac{m_L^2}{T^2}(\ddot{y}_L (\ddot{z}_L + g) - \ddot{z}_L \ddot{y}_L) \\ &= \frac{\ddot{z}_L \ddot{y}_L - \ddot{y}_L (\ddot{z}_L + g)}{\ddot{y}_L^2 + (\ddot{z}_L + g)^2} \end{aligned}$$

Differentiating again to find higher derivatives of \mathbf{p} and ϕ_L :

$$2\dot{T}\dot{\mathbf{p}} + T\ddot{\mathbf{p}} + \ddot{T}\mathbf{p} = -m_L\mathbf{x}_L^{(4)}$$

$$\ddot{\mathbf{p}} = -\frac{1}{T}(m_L\mathbf{x}_L^{(4)} + \ddot{T}\mathbf{p} + 2\dot{T}\dot{\mathbf{p}})$$

Where:

$$\begin{aligned} \ddot{T} &= m_L \left(\frac{(\ddot{y}_L^2 + \ddot{y}_L \ddot{y}_L^{(4)} + \ddot{z}_L^{(4)}(\ddot{z}_L + g) + \ddot{z}_L^2)}{(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{\frac{1}{2}}} - \frac{(\ddot{y}_L \ddot{y}_L + (\ddot{z}_L + g) \ddot{z}_L)^2}{(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{\frac{3}{2}}} \right) \\ &= m_L((\ddot{y}_L^2 + \ddot{y}_L \ddot{y}_L^{(4)} + \ddot{z}_L^{(4)}(\ddot{z}_L + g) + \ddot{z}_L^2)(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{1}{2}} \\ &\quad - (\ddot{y}_L \ddot{y}_L + (\ddot{z}_L + g) \ddot{z}_L)^2 (\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{3}{2}}) \end{aligned}$$

From $\ddot{\mathbf{p}}$, we can find the state $\ddot{\phi}_L$:

$$\begin{aligned} \ddot{\mathbf{p}} &= \ddot{\phi}_L \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + \dot{\phi}_L^2 \begin{bmatrix} -\sin(\phi_L) \\ \cos(\phi_L) \end{bmatrix} \\ \ddot{\phi}_L &= \ddot{\mathbf{p}} \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} \end{aligned}$$

$$3\ddot{T}\dot{\mathbf{p}} + 3\dot{T}\ddot{\mathbf{p}} + T\ddot{\mathbf{p}} + \ddot{T}\mathbf{p} = -m_L\mathbf{x}_L^{(5)}$$

$$\ddot{\mathbf{p}} = -\frac{1}{T} \left(m_L\mathbf{x}_L^{(5)} + 3\ddot{T}\dot{\mathbf{p}} + 3\dot{T}\ddot{\mathbf{p}} + \ddot{T}\mathbf{p} \right)$$

Where:

$$\begin{aligned} \ddot{T} = m_L(& (3\ddot{y}_L y_L^{(4)} + \ddot{y}_L y_L^{(5)} + (\ddot{z}_L + g)z_L^{(5)} + 3\ddot{z}_L z_L^{(4)})(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{1}{2}} \\ & + 3(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{1}{2}}(\ddot{y}_L \ddot{y}_L + (\ddot{z}_L + g)\ddot{z}_L)(\ddot{y}_L^2 + \ddot{y}_L y_L^{(4)} + (\ddot{z}_L + g)z_L^{(4)} + \ddot{z}_L^2) \\ & + 3(\ddot{y}_L \ddot{y}_L + (\ddot{z}_L + g)\ddot{z}_L)^3(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{5}{2}}) \end{aligned}$$

From $\ddot{\mathbf{p}}$, we can find the state $\ddot{\phi}_L$:

$$\begin{aligned} \ddot{\mathbf{p}} &= (\ddot{\phi}_L - \dot{\phi}_L^3) \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + 3\dot{\phi}_L \ddot{\phi}_L \begin{bmatrix} -\sin(\phi_L) \\ \cos(\phi_L) \end{bmatrix} \\ \ddot{\phi}_L &= \ddot{\mathbf{p}} \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + \dot{\phi}_L^3 \end{aligned}$$

$$4\ddot{T}\dot{\mathbf{p}} + 6\dot{T}\ddot{\mathbf{p}} + 4\ddot{T}\ddot{\mathbf{p}} + T\mathbf{p}^{(4)} + T^{(4)}\mathbf{p} = -m_L\mathbf{x}_L^{(6)}$$

$$\mathbf{p}^{(4)} = -\frac{1}{T}(m_L\mathbf{x}_L^{(6)} + 4\ddot{T}\dot{\mathbf{p}} + 6\dot{T}\ddot{\mathbf{p}} + 4\ddot{T}\ddot{\mathbf{p}} + T^{(4)}\mathbf{p})$$

$$\begin{aligned} T^{(4)} = m_L(& -\frac{15}{8}((\ddot{z}_L + g)\ddot{z}_L + \ddot{y}_L \ddot{y}_L)^4(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{7}{2}} \\ & + 9(\ddot{y}_L \ddot{y}_L + (\ddot{z}_L + g)\ddot{z}_L)^2((\ddot{z}_L + g)z_L^{(4)} + \ddot{y}_L y_L^{(4)} + \ddot{y}_L^2 + \ddot{z}_L^2)(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{5}{2}} \\ & - 3((\ddot{z}_L + g)z_L^{(4)} + \ddot{y}_L y_L^{(4)} + \ddot{y}_L^2 + \ddot{z}_L^2)^2(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{3}{2}} \\ & - 4((\ddot{z}_L + g)\ddot{z}_L + \ddot{y}_L \ddot{y}_L)(z_L^{(5)}(\ddot{z} + g) + \ddot{y}_L y_L^{(5)} + 3y_L^{(4)}\ddot{y}_L + 3\ddot{z}_L z_L^{(4)})(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{3}{2}} \\ & + (z_L^{(6)}(\ddot{z}_L + g) + \ddot{y}_L y_L^{(6)} + 3y_L^{(4)2} + 4y_L^{(5)}\ddot{y}_L + 3z_L^{(4)2} + 4z_L^{(5)}\ddot{z}_L)(\ddot{y}_L^2 + (\ddot{z}_L + g)^2)^{-\frac{1}{2}}) \end{aligned}$$

From $\mathbf{p}^{(4)}$, we can find $\phi_L^{(4)}$:

$$\begin{aligned} \mathbf{p}^{(4)} &= (\phi_L^{(4)} - 6\dot{\phi}_L^2 \ddot{\phi}_L) \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + (3\ddot{\phi}_L^2 - \dot{\phi}_L^4) \begin{bmatrix} -\sin(\phi_L) \\ \cos(\phi_L) \end{bmatrix} \\ \phi_L^{(4)} &= \mathbf{p}^{(4)} \cdot \begin{bmatrix} \cos(\phi_L) \\ \sin(\phi_L) \end{bmatrix} + 6\dot{\phi}_L^2 \ddot{\phi}_L \end{aligned}$$

Using Newton's Equations on the quadrotor and the load to eliminate $T\mathbf{p}$ and solve for $f\mathbf{b}_3$:

$$\begin{aligned} m_Q \ddot{\mathbf{x}}_Q &= f\mathbf{b}_3 - m_Q g\mathbf{e}_3 + T\mathbf{p} \\ m_L \ddot{\mathbf{x}}_L &= -T\mathbf{p} - m_L g\mathbf{e}_3 \\ f\mathbf{b}_3 &= m_Q \ddot{\mathbf{x}}_Q + m_L \ddot{\mathbf{x}}_L + m_Q g\mathbf{e}_3 + m_L g\mathbf{e}_3 \end{aligned}$$

Using the constraint $\ddot{\mathbf{x}}_Q = \ddot{\mathbf{x}}_L - l\ddot{\mathbf{p}}$:

$$\begin{aligned} f\mathbf{b}_3 &= m_Q(\ddot{\mathbf{x}}_L - l\ddot{\mathbf{p}}) + m_L \ddot{\mathbf{x}}_L + m_Q g\mathbf{e}_3 + m_L g\mathbf{e}_3 \\ &= (m_Q + m_L)(\ddot{\mathbf{x}}_L + g\mathbf{e}_3) - m_Q l\ddot{\mathbf{p}} \end{aligned}$$

$$\mathbf{b}_3 = \frac{(m_Q + m_L)(\ddot{\mathbf{x}}_L + g\mathbf{e}_3) - m_Q l\ddot{\mathbf{p}}}{\|(m_Q + m_L)(\ddot{\mathbf{x}}_L + g\mathbf{e}_3) - m_Q l\ddot{\mathbf{p}}\|} = \begin{bmatrix} -\sin(\phi_Q) \\ \cos(\phi_Q) \end{bmatrix}$$

$$f = ((m_Q + m_L)(\ddot{\mathbf{x}}_L + g\mathbf{e}_3) - m_Q l\ddot{\mathbf{p}}) \cdot \mathbf{b}_3$$

$$\phi_Q = \tan^{-1} \left(\frac{f\mathbf{b}_3 \cdot \mathbf{e}_2}{f\mathbf{b}_3 \cdot \mathbf{e}_3} \right)$$

$$= \tan^{-1} \left(\frac{-(m_Q + m_L)\ddot{y}_L + m_Q l\ddot{\mathbf{p}} \cdot \mathbf{e}_2}{(m_Q + m_L)(\ddot{z}_L + g) - m_Q l\ddot{\mathbf{p}} \cdot \mathbf{e}_3} \right)$$

$$\begin{aligned} \dot{\phi}_Q &= ((lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)((m_Q + m_L)(\ddot{z} + g) - lm_Q \ddot{p}_2) + (lm_Q \ddot{p}_2 - (m_Q + m_L)\ddot{z}_L)(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)) \\ &\quad \left((lm_Q \ddot{p}_2 - (m_Q + m_L)(\ddot{z}_L + g))^2 \left(\frac{(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)^2}{(lm_Q \ddot{p}_2 + (m_Q + m_L)(\ddot{z}_L + g))^2} + 1 \right) \right)^{-1}, \text{ where } \mathbf{p}^{(k)} = \begin{bmatrix} p_1^{(k)} \\ p_2^{(k)} \end{bmatrix} \end{aligned}$$

$$\begin{aligned} \ddot{\phi} &= \left[\left(\frac{lm_Q p_1^{(4)} - (m_Q + m_L)y_L^{(4)}}{(m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2} - \frac{2(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)((m_Q + m_L)\ddot{z}_L - lm_Q \ddot{p}_2)}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} \right. \right. \\ &\quad \left. \left. - \frac{((m_Q + m_L)\ddot{z}_L^{(4)} - lm_Q p_2^{(4)})(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} + \frac{2((m_Q + m_L)\ddot{z}_L - lm_Q \ddot{p}_1)^2(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^3} \right) \right. \\ &\quad \left. \left(\frac{(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)^2}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} + 1 \right)^{-1} \right] \\ &\quad - \left[\left(\frac{2(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} - \frac{2((m_Q + m_L)\ddot{z}_L - lm_Q \ddot{p}_2)(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)^2}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^3} \right) \right. \\ &\quad \left. \left(\frac{lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} - \frac{((m_Q + m_L)\ddot{z}_L - lm_Q \ddot{p}_2)(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} \right) \right. \\ &\quad \left. \left(\frac{(lm_Q \ddot{p}_1 - (m_Q + m_L)\ddot{y}_L)^2}{((m_Q + m_L)(\ddot{z}_L + g) - lm_Q \ddot{p}_2)^2} + 1 \right)^{-2} \right] \end{aligned}$$

2.1.2 Control Laws

Using the control law from the paper:

Load Position Control

$$\begin{aligned}
f &= -k_p(e_{\mathbf{x}}) - k_d(\dot{e}_{\mathbf{x}}) + m_L \ddot{\mathbf{x}}_L^d + m_L g \mathbf{e}_3 + m_Q \ddot{\mathbf{x}}_Q^d + m_Q g \mathbf{e}_3 \\
&= -k_p(\mathbf{x}_L - \mathbf{x}_T) - k_d(\dot{\mathbf{x}}_L - \dot{\mathbf{x}}_T) + m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3 + m_Q \ddot{\mathbf{x}}_Q^d + m_Q g \mathbf{e}_3 \\
&= -k_p(\mathbf{x}_L - \mathbf{x}_T) - k_d(\dot{\mathbf{x}}_L - \dot{\mathbf{x}}_T) + m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3 + m_Q(\ddot{\mathbf{x}}_T - l\ddot{\mathbf{p}}) + m_Q g \mathbf{e}_3 \\
\mathbf{p}^d &= -\frac{(-k_p(\mathbf{x}_L - \mathbf{x}_T) - k_d(\dot{\mathbf{x}}_L - \dot{\mathbf{x}}_T) + m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3)}{\| -k_p(\mathbf{x}_L - \mathbf{x}_T) - k_d(\dot{\mathbf{x}}_L - \dot{\mathbf{x}}_T) + m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3 \|} \\
\phi_L^d &= \tan^{-1} \left(-\frac{\mathbf{p}^d \cdot \mathbf{e}_2}{\mathbf{p}^d \cdot \mathbf{e}_3} \right)
\end{aligned}$$

Load Attitude Control

$$\begin{aligned}
T_{nom} \mathbf{p}_{nom} &= -(m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3) \\
\mathbf{p}_{nom} &= \frac{-(m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3)}{\| -(m_L \ddot{\mathbf{x}}_T + m_L g \mathbf{e}_3) \|} \\
T_{nom} &= T_{nom} \cdot \mathbf{p}_{nom} \\
\phi_{L_{nom}} &= \tan^{-1} \left(\frac{T_{nom} \mathbf{p}_{nom} \cdot \mathbf{e}_2}{-T_{nom} \mathbf{p}_{nom} \cdot \mathbf{e}_3} \right) \\
\dot{\mathbf{p}}_{nom} &= -\frac{1}{T_{nom}} (m_L \ddot{\mathbf{x}}_T + \dot{T}_{nom} \mathbf{p}_{nom}), \dot{T}_{nom} = m_L \frac{\ddot{y}_T \ddot{y}_T + (\ddot{z}_T + g) \ddot{z}_T}{(\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{\frac{1}{2}}} \\
\dot{\phi}_L^d &= \dot{\mathbf{p}}_{nom} \cdot \begin{bmatrix} \cos(\phi_{L_{nom}}) \\ \sin(\phi_{L_{nom}}) \end{bmatrix} \\
\ddot{\mathbf{p}}_{nom} &= -\frac{1}{T_{nom}} (m_L \mathbf{x}^{(4)}_T + \ddot{T}_{nom} \mathbf{p}_{nom} + 2\dot{T}_{nom} \dot{\mathbf{p}}_{nom}), \\
\ddot{T}_{nom} &= m_L \left(\frac{(\ddot{y}_T^2 + \ddot{y}_T \ddot{y}_T^{(4)} + z_T^{(4)} (\ddot{z}_T + g) + \ddot{z}_T^2)}{(\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{\frac{1}{2}}} - \frac{(\ddot{y}_T \ddot{y}_T + (\ddot{z}_T + g) \ddot{z}_T)^2}{(\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{\frac{3}{2}}} \right) \\
\ddot{\phi}_L^d &= \ddot{\mathbf{p}}_{nom} \cdot \begin{bmatrix} \cos(\phi_{L_{nom}}) \\ \sin(\phi_{L_{nom}}) \end{bmatrix} \\
f_{nom} \mathbf{b}_{3_{nom}} &= (m_Q + m_L)(\ddot{\mathbf{x}}_T + g \mathbf{e}_3) - m_Q l \ddot{\mathbf{p}}_{nom} \\
\mathbf{b}_{3_{nom}} &= \frac{(m_Q + m_L)(\ddot{\mathbf{x}}_T + g \mathbf{e}_3) - m_Q l \ddot{\mathbf{p}}_{nom}}{\| (m_Q + m_L)(\ddot{\mathbf{x}}_T + g \mathbf{e}_3) - m_Q l \ddot{\mathbf{p}}_{nom} \|} \\
\phi_{Q_{nom}} &= \tan^{-1} \left(\frac{-f_{nom} \mathbf{b}_{3_{nom}} \cdot \mathbf{e}_2}{f_{nom} \mathbf{b}_{3_{nom}} \cdot \mathbf{e}_3} \right) \\
f_{nom} &= f_{nom} \mathbf{b}_{3_{nom}} \cdot \mathbf{b}_{3_{nom}} \\
\phi_Q^d &= \phi_{L_{nom}} + \sin^{-1} \left(-k_p^L e_L - k_d^L \dot{e}_L + \frac{\ddot{\phi}_L^d m_Q l}{f_{nom}} \right) \\
&= \phi_{L_{nom}} + \sin^{-1} \left(-k_p^L (\phi_L - \phi_L^d) - k_d^L (\dot{\phi}_L - \dot{\phi}_L^d) + \frac{\ddot{\phi}_L^d m_Q l}{f_{nom}} \right)
\end{aligned}$$

$$\begin{aligned}
 \ddot{\mathbf{p}}_{nom} &= -\frac{1}{T_{nom}} \left(m_L \mathbf{x}_T^{(5)} + 3\ddot{T}_{nom} \dot{\mathbf{p}}_{nom} + 3\dot{T}_{nom} \ddot{\mathbf{p}}_{nom} + \ddot{T}_{nom} \mathbf{p}_{nom} \right), \\
 \ddot{T}_{nom} &= m_L (\\
 &\quad (3\ddot{y}_T y_T^{(4)} + \ddot{y}_T y_T^{(5)} + (\ddot{z}_T + g) z_T^{(5)} + 3\ddot{z}_T z_T^{(4)}) (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{1}{2}} \\
 &\quad + 3(\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{1}{2}} (\ddot{y}_T \ddot{y}_T + (\ddot{z}_T + g) \ddot{z}_T) (\ddot{y}_T^2 + \ddot{y}_T y_T^{(4)} + (\ddot{z}_T + g) z_T^{(4)} + \ddot{z}_T^2) \\
 &\quad + 3(\ddot{y}_T \ddot{y}_T + (\ddot{z}_T + g) \ddot{z}_T)^3 (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{5}{2}}) \\
 \ddot{\phi}_{L_{nom}} &= \ddot{\mathbf{p}}_{nom} \cdot \begin{bmatrix} \cos(\phi_{L_{nom}}) \\ \sin(\phi_{L_{nom}}) \end{bmatrix} + \dot{\phi}_{L_{nom}}^3 \\
 \mathbf{p}_{nom}^{(4)} &= -\frac{1}{T_{nom}} (m_L \mathbf{x}_T^{(6)} + 4\ddot{T}_{nom} \dot{\mathbf{p}}_{nom} + 6\dot{T}_{nom} \ddot{\mathbf{p}}_{nom} + 4\ddot{T}_{nom} \ddot{\mathbf{p}}_{nom} + T^{(4)} \mathbf{p}_{nom}) \\
 T_{nom}^{(4)} &= m_L (\\
 &\quad -\frac{15}{8} ((\ddot{z}_T + g) \ddot{z}_T + \ddot{y}_T \ddot{y}_T)^4 (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{7}{2}} \\
 &\quad + 9(\ddot{y}_T \ddot{y}_T + (\ddot{z}_T + g) \ddot{z}_T)^2 ((\ddot{z}_T + g) z_T^{(4)} + \ddot{y}_T y_T^{(4)} + \ddot{y}_T^2 + \ddot{z}_T^2) (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{5}{2}} \\
 &\quad - 3((\ddot{z}_T + g) z_T^{(4)} + \ddot{y}_T y_T^{(4)} + \ddot{y}_T^2 + \ddot{z}_T^2)^2 (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{3}{2}} \\
 &\quad - 4((\ddot{z}_T + g) \ddot{z}_T + \ddot{y}_T \ddot{y}_T) (z_T^{(5)} (\ddot{z}_T + g) + \ddot{y}_T y_T^{(5)} + 3y_T^{(4)} \ddot{y}_T + 3\ddot{z}_T z_T^{(4)}) (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{3}{2}} \\
 &\quad + (z_T^{(6)} (\ddot{z}_T + g) + \ddot{y}_T y_T^{(6)} + 3y_T^{(4)2} + 4y_T^{(5)} \ddot{y}_T + 3z_T^{(4)2} + 4z_T^{(5)} \ddot{z}_T) (\ddot{y}_T^2 + (\ddot{z}_T + g)^2)^{-\frac{1}{2}}) \\
 \phi_{L_{nom}}^{(4)} &= \mathbf{p}_{nom}^{(4)} \cdot \begin{bmatrix} \cos(\phi_{L_{nom}}) \\ \sin(\phi_{L_{nom}}) \end{bmatrix} + 3\dot{\phi}_{L_{nom}}^2 \ddot{\phi}_{L_{nom}} + \ddot{\phi}_{L_{nom}} \dot{\phi}_{L_{nom}}^2 + 2\dot{\phi}_{L_{nom}}^2 \ddot{\phi}_{L_{nom}} \\
 \dot{f}_{nom} &= ((m_Q + m_L) \ddot{\mathbf{x}}_T - m_Q l \ddot{\mathbf{p}}_{nom}) \cdot \mathbf{b}_{3_{nom}} + \dot{\phi}_{L_{nom}} ((m_Q + m_L) (\ddot{\mathbf{x}}_T + g \mathbf{e}_3) - m_Q l \ddot{\mathbf{p}}_{nom}) \cdot \mathbf{b}_{2_{nom}} \\
 \ddot{f}_{nom} &= ((m_Q + m_L) \mathbf{x}_T^{(4)} - m_Q l \mathbf{p}_{nom}^{(4)} + \dot{\phi}_{L_{nom}}^2 ((m_Q + m_L) (\ddot{\mathbf{x}}_T + g \mathbf{b}_{3_{nom}}) - m_Q l \ddot{\mathbf{p}}_{nom})) \cdot \mathbf{b}_{3_{nom}} \\
 &\quad + \ddot{\phi}_{L_{nom}} ((m_Q + m_L) (\ddot{\mathbf{x}}_T + g \mathbf{e}_3) - m_Q l \ddot{\mathbf{p}}_{nom}) \cdot \mathbf{b}_{2_{nom}} \\
 \dot{\phi}_Q^d &= \frac{m_Q l \ddot{\phi}_{L_{nom}} - \dot{f}_{nom} \sin(\phi_{Q_{nom}} - \phi_{L_{nom}})}{f_{nom} \cos(\phi_{Q_{nom}} - \phi_{L_{nom}})} + \dot{\phi}_{L_{nom}} \\
 \ddot{\phi}_Q^d &= \frac{m_Q l \phi_{L_{nom}}^{(4)} - \ddot{f}_{nom} \sin(\phi_{Q_{nom}} - \phi_{L_{nom}}) - 2\dot{f}_{nom} \cos(\phi_{Q_{nom}} - \phi_{L_{nom}}) (\dot{\phi}_{Q_{nom}} - \dot{\phi}_{L_{nom}})}{\cos(\phi_{Q_{nom}} - \phi_{L_{nom}})} \\
 &\quad + \frac{\dot{f}_{nom} \sin(\phi_{Q_{nom}} - \phi_{L_{nom}}) (\dot{\phi}_{Q_{nom}} - \dot{\phi}_{L_{nom}})^2}{\cos(\phi_{Q_{nom}} - \phi_{L_{nom}})} \\
 &\quad + \ddot{\phi}_{L_{nom}} \\
 M &= J_Q (-k_p^Q e_Q - k_d^Q \dot{e}_Q + \ddot{\phi}_Q^d) \\
 &= J_Q (-k_p^Q (\phi_Q - \phi_Q^d) - k_d^Q (\dot{\phi}_Q - \dot{\phi}_Q^d) + \ddot{\phi}_Q^d)
 \end{aligned}$$

2.2 \mathbf{x}_2 system:

$$\begin{aligned} m_L \dot{\mathbf{v}}_L + m_L g \mathbf{e}_3 &= 0 \\ m_Q \dot{\mathbf{v}}_Q + m_Q g \mathbf{e}_3 &= f \mathbf{b}_3 \\ J_Q \ddot{\phi}_Q &= M \end{aligned}$$

2.2.1 Differential Flatness

Flat outputs $\mathbf{y} = [\mathbf{x}_Q]^T = [y_Q \ z_Q]^T$

\mathbf{x}_L and \mathbf{v}_L are known from initial conditions because load is in free fall:

$$\begin{aligned} \dot{\mathbf{x}}_L &= \mathbf{v}_L \\ \dot{\mathbf{v}}_L &= -g \mathbf{e}_3 \end{aligned}$$

Derive $\dot{y}_Q = v_{yQ}$, $\dot{z}_Q = v_{zQ}$, and all higher derivatives from differentiation of y_Q , z_Q

From equation of motion:

$$\begin{aligned} f \mathbf{b}_3 &= m_Q \ddot{\mathbf{x}}_Q + m_Q g \mathbf{e}_3 \\ f &= \|m_Q \ddot{\mathbf{x}}_Q + m_Q g \mathbf{e}_3\| \\ &= m_Q (\ddot{y}_Q^2 + (\ddot{z}_Q + g)^2)^{\frac{1}{2}} \\ \mathbf{b}_3 &= \begin{bmatrix} -\sin(\phi_Q) \\ \cos(\phi_Q) \end{bmatrix} = \frac{m_Q \ddot{\mathbf{x}}_Q + m_Q g \mathbf{e}_3}{\|m_Q \ddot{\mathbf{x}}_Q + m_Q g \mathbf{e}_3\|} \\ \sin(\phi_Q) &= -\frac{m_Q}{f} \ddot{y}_Q \\ \cos(\phi_Q) &= \frac{m_Q}{f} (\ddot{z}_Q + g) \\ \phi_Q &= \tan^{-1} \left(\frac{-f \mathbf{b}_3 \cdot \mathbf{e}_2}{f \mathbf{b}_3 \cdot \mathbf{e}_3} \right) \\ &= \tan^{-1} \left(\frac{-\ddot{y}_Q}{\ddot{z}_Q + g} \right) \end{aligned}$$

Differentiating the equation of motion:

$$\begin{aligned} m_Q \ddot{\mathbf{x}}_Q &= \dot{f} \mathbf{b}_3 + f (\mathcal{I} \omega^B \times \mathbf{b}_3) \\ &= \dot{f} \mathbf{b}_3 + f (\dot{\phi}_Q \mathbf{b}_1 \times \mathbf{b}_3) \\ &= \dot{f} \mathbf{b}_3 - f \dot{\phi}_Q \mathbf{b}_2 \\ \dot{\phi}_Q &= -\frac{m_Q}{f} (\ddot{\mathbf{x}}_Q \cdot \mathbf{b}_2) \\ &= -\frac{m_Q}{f} (\ddot{y}_Q \cos(\phi_Q) + \ddot{z}_Q \sin(\phi_Q)) \\ &= -\frac{m_Q^2}{f^2} (\ddot{y}_Q (\ddot{z}_Q + g) - \ddot{z}_Q \ddot{y}_Q) \\ &= \frac{(\ddot{z}_Q \ddot{y}_Q - \ddot{y}_Q (\ddot{z}_Q + g))}{(\ddot{y}_Q^2 + (\ddot{z}_Q + g)^2)} \end{aligned}$$

Differentiating the equation of motion again:

$$\begin{aligned}
m_Q \ddot{\mathbf{x}}_Q &= \ddot{f} \mathbf{b}_3 + \dot{f} (\mathcal{I} \omega^B \times \mathbf{b}_3) - \left((\dot{f} \dot{\phi}_Q + f \ddot{\phi}_Q) \mathbf{b}_2 + f \dot{\phi}_Q (\mathcal{I} \omega^B \times \mathbf{b}_2) \right) \\
&= \ddot{f} \mathbf{b}_3 + \dot{f} (\dot{\phi}_Q \mathbf{b}_1 \times \mathbf{b}_3) - \left((\dot{f} \dot{\phi}_Q + f \ddot{\phi}_Q) \mathbf{b}_2 + f \dot{\phi}_Q (\dot{\phi}_Q \mathbf{b}_1 \times \mathbf{b}_2) \right) \\
&= \ddot{f} \mathbf{b}_3 - \dot{f} \dot{\phi}_Q \mathbf{b}_2 - \left((\dot{f} \dot{\phi}_Q + f \ddot{\phi}_Q) \mathbf{b}_2 + f \dot{\phi}_Q^2 \mathbf{b}_3 \right) \\
&= -(2\dot{f} \dot{\phi}_Q + f \ddot{\phi}_Q) \mathbf{b}_2 + (\ddot{f} - f \dot{\phi}_Q^2) \mathbf{b}_3 \\
\ddot{\phi}_Q &= -\frac{m_Q}{f} (\ddot{\mathbf{x}}_Q \cdot \mathbf{b}_2) - 2 \frac{\dot{f} \dot{\phi}_Q}{f}
\end{aligned}$$

Where:

$$\begin{aligned}
\dot{f} &= m_Q^2 \frac{(\ddot{y}_Q \ddot{y}_Q + (\ddot{z}_Q + g) \ddot{z}_Q)}{f} \\
\ddot{\phi}_Q &= -\frac{m_Q}{f} (\ddot{y}_Q \cos(\phi_Q) + \ddot{z}_Q \sin(\phi_Q)) \\
&\quad - \frac{2m_Q^2 (\ddot{y}_Q \ddot{y}_Q + (\ddot{z}_Q + g) \ddot{z}_Q)}{f^2} \dot{\phi}_Q \\
&= -\frac{m_Q^2}{f^2} (\ddot{y}_Q (\ddot{z}_Q + g) - \ddot{z}_Q \ddot{y}_Q) \\
&\quad + \frac{2m_Q^4}{f^4} (\ddot{y}_Q \ddot{y}_Q + (\ddot{z}_Q + g) \ddot{z}_Q) (\ddot{y}_Q (\ddot{z}_Q + g) - \ddot{z}_Q \ddot{y}_Q) \\
&= \frac{(\ddot{z}_Q \ddot{y}_Q - \ddot{y}_Q (\ddot{z}_Q + g))}{\left(\ddot{y}_Q^2 + (\ddot{z}_Q + g)^2 \right)} \\
&\quad + \frac{2(\ddot{y}_Q \ddot{y}_Q + (\ddot{z}_Q + g) \ddot{z}_Q) (\ddot{y}_Q (\ddot{z}_Q + g) - \ddot{z}_Q \ddot{y}_Q)}{\left(\ddot{y}_Q^2 + (\ddot{z}_Q + g)^2 \right)^2}
\end{aligned}$$

The moment input be found from:

$$M = J_Q \ddot{\phi}_Q$$

2.2.2 Control Laws

Use control laws from paper with desired trajectory: $\sigma_T(t) = [\mathbf{x}_T(t)] = [y_T(t) \ z_T(t)]^T$:

$$\begin{aligned}\mathbf{F1} &= m_Q g \mathbf{e}_3 + m_Q \ddot{\mathbf{x}}_T \\ \mathbf{F} &= -K_p (\mathbf{x} - \mathbf{x}_T) - K_d (\dot{\mathbf{x}} - \dot{\mathbf{x}}_T) + \mathbf{F1} \\ f &= \mathbf{F} \cdot \mathbf{b}_3 \\ \phi_Q^d &= \tan^{-1} \left(\frac{-\mathbf{F} \cdot \mathbf{e}_2}{\mathbf{F} \cdot \mathbf{e}_3} \right)\end{aligned}$$

$$\begin{aligned}f_{nom} \mathbf{b}_{3_{nom}} &= m_Q \ddot{\mathbf{x}}_T + m_Q g \mathbf{e}_3 \\ \mathbf{b}_{3_{nom}} &= \frac{m_Q \ddot{\mathbf{x}}_T + m_Q g \mathbf{e}_3}{\|m_Q \ddot{\mathbf{x}}_T + m_Q g \mathbf{e}_3\|} \\ \phi_{Q_{nom}} &= \tan^{-1} \left(\frac{f_{nom} \mathbf{b}_{3_{nom}} \cdot \mathbf{e}_2}{f_{nom} \mathbf{b}_{3_{nom}} \cdot \mathbf{e}_3} \right) \\ f_{nom} &= f_{nom} \mathbf{b}_{3_{nom}} \cdot \mathbf{b}_{3_{nom}} \\ \dot{\phi}_Q^d &= -\frac{m_Q}{f_{nom}} (\ddot{\mathbf{x}}_T \cdot \mathbf{b}_{2_{nom}}), \text{ where } \mathbf{b}_{2_{nom}} = \begin{bmatrix} \cos(\phi_{Q_{nom}}) \\ \sin(\phi_{Q_{nom}}) \end{bmatrix} \\ \ddot{\phi}_Q^d &= -\frac{m_Q}{f_{nom}} (\ddot{\mathbf{x}}_T \cdot \mathbf{b}_{2_{nom}}) - 2 \frac{\dot{f} \dot{\phi}_Q^d}{f_{nom}}, \dot{f} = \frac{(\ddot{y}_T \ddot{y}_T + (\ddot{z}_T + g) \ddot{z}_T)}{f_{nom}}\end{aligned}$$

$$\begin{aligned}\mathbf{M}^d &= J_Q \ddot{\phi}_Q^d \\ \mathbf{M} &= J_Q (-K_{p_\phi} (\phi_Q - \phi_Q^d) - K_{d_\phi} (\dot{\phi}_Q - \dot{\phi}_Q^d)) + \mathbf{M}^d\end{aligned}$$

2.3 Differential Flatness of Hybrid System

“A Differentially-Flat Hybrid System is a hybrid system where each subsystem is differentially-flat, the switching surfaces are functions of the flat outputs and their derivatives, and moreover the flat outputs map from one subsystem to a subsequent subsystem through the sufficiently smooth transition maps.”

The switching surfaces are $\mathcal{S}_1 = \{\mathbf{x}_1 \mid T \equiv \|m_L(\dot{\mathbf{v}}_L + g\mathbf{e}_3)\| = 0\}$, which is in terms of the flat outputs \mathbf{x}_L . The second switching surface is $\mathcal{S}_2 = \{\mathbf{x}_2 \mid \|\mathbf{x}_Q - \mathbf{x}_L\| = l\}$, which is in terms of the flat output \mathbf{x}_Q and \mathbf{x}_L which can be determined from initial conditions.