

PHYS 303 – Classical Mechanics of Particles and Waves II

Example Problem Set

Due: Not for Credit

Term: Fall 2024

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Readings

Read chapter 2 of Taylor.

Problems

Problem 1. [30 pts.] – Projectile Motion in Linear Resistive Medium

A projectile of mass m is launched from the surface of the Earth with an initial velocity \mathbf{v}_0 at time $t = 0$. The projectiles trajectory is such that the gravity field \mathbf{g} is assumed constant. The projectile moves in a medium which retards its motion with a magnitude proportional to its velocity, $\mathbf{F}_{\text{drag}} = -b\mathbf{v}$, with b being the positive drag coefficient.

- (a) [5 pts.] Determine the projectiles equations of motion. Show that the velocity \mathbf{v} as a function of time t is given by

$$\mathbf{v}(t) = \mathbf{g}\tau + (\mathbf{v}_0 - \mathbf{g}\tau) e^{-t/\tau},$$

where $\tau = m/b$. What are the physical dimensions of b and τ ?

- (b) [5 pts.] If the projectile is launched from the origin, show that its position as a function of time is

$$\mathbf{r}(t) = \mathbf{g}\tau t + \tau(\mathbf{v}_0 - \mathbf{g}\tau) \left(1 - e^{-t/\tau}\right).$$

Show that if the effect of drag is negligible, that is $b \sim \tau^{-1} \ll 1$, that $\mathbf{r}(t)$ reduces to the trajectory for a projectile in a uniform gravity field with no drag. *Hint:* Recall the Taylor expansion $e^x = 1+x+\mathcal{O}(x^2)$.

- (c) [10 pts.] Consider now a coordinate system such that $\mathbf{g} = (0, 0, -g)$ and $\mathbf{v}_0 = v_0(\cos\theta_0, 0, \sin\theta_0)$ where θ_0 is the launch angle with respect to the Earth's surface. Show that the equation for the range R of the projectile is of the form

$$\tan\theta_0 R + \frac{g\tau}{v_0 \cos\theta_0} R + g\tau^2 \ln\left(1 - \frac{R}{v_0\tau \cos\theta_0}\right) = 0.$$

This equation is transcendental in R , and thus cannot be solved analytically. An approximate solution can be made for systems where the effect of drag is small, that is $b \sim \tau^{-1} \ll 1$. Use the Taylor expansion $\ln(1+x) = x - x^2/2 + x^3/3 + \mathcal{O}(x^4)$ to generate the approximate algebraic equation

$$\tan\theta_0 - \frac{g}{2v_0^2 \cos\theta_0} R - \frac{g}{3\tau v_0^3 \cos^3\theta_0} R^2 + \mathcal{O}(\tau^{-2}) = 0.$$

- (d) [10 pts.] To solve the approximate equation for R in part (c), we construct a perturbative expansion for R of the form

$$R = R_0 - \alpha\tau^{-1} + \mathcal{O}(\tau^{-2}),$$

where R_0 is the range of the projectile when drag is ignored and α is a positive constant to be determined. Determine α , and the resulting shift in the range $\Delta R = R_0 - R$ in terms of v_0 , θ_0 , g , and τ . Verify that the shift decreases the range with respect to the vacuum case.

Physics 303 - Example Problem Set

Andrew Jackura

1. Projectile

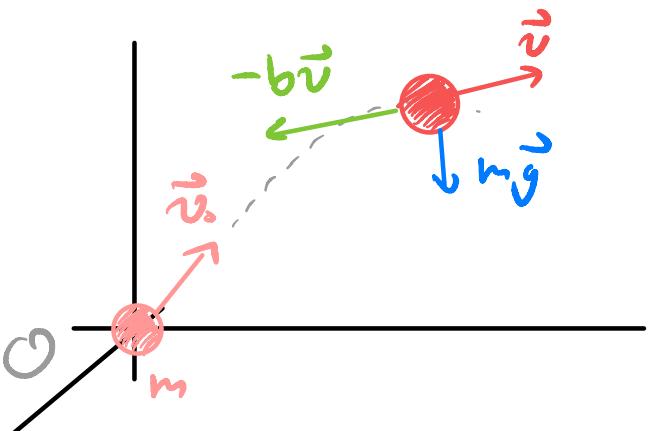
- mass, m

- initial velocity, \vec{v}_0 @ $t=0$

Forces

- gravity, $m\vec{g} = \text{const.}$

- drag, $-b\vec{v}$



a) Find $\vec{v}(t)$

$$\text{From N II : } m\ddot{\vec{r}} = m\vec{g} - b\vec{v}$$

Now, recall $\dot{\vec{r}} = \vec{v}$, so EOM is

$$\dot{\vec{v}} = \vec{g} - \frac{b}{m}\vec{v}$$

Define for convenience $\tau = m/b$

$$\text{Note that } [m] = M, \boxed{[b] = MT^{-1}}$$

$$\Rightarrow \boxed{[\tau] = [m][b]^{-1} = T}$$

So, we solve $\frac{d\vec{v}}{dt} + \frac{1}{\tau}\vec{v} = \vec{g}$ via integrating factor

$$\mu = e^{t/\tau} \text{ s.t. } \frac{1}{\tau} = \frac{d}{dt}e^{t/\tau} \Rightarrow \frac{d}{dt}(\mu \vec{v}) = \mu \dot{\vec{v}} + \frac{1}{\tau} \mu \vec{v}$$

$$\text{So, EOM is } \frac{d}{dt}(e^{t/\tau} \vec{v}) = \vec{g} e^{t/\tau}$$

We can integrate easily,

$$\Rightarrow \int dt \frac{d}{dt}(e^{t/\tau} \vec{v}) = \int dt \vec{g} e^{t/\tau}$$

$$\Rightarrow e^{t/\tau} \vec{v} = \tau \vec{g} e^{t/\tau} + \vec{c}$$

From I.C's, $\vec{v} = \vec{v}_0$ at $t=0$

$$\Rightarrow \vec{v}_0 = \tau \vec{g} + \vec{c} \Rightarrow \vec{c} = \vec{v}_0 - \tau \vec{g}$$

So, solution is

$$\boxed{\vec{v}(t) = \vec{g} \tau + (\vec{v}_0 - \vec{g} \tau) e^{-t/\tau}}$$

as required

b) We now want $\vec{r}(t)$ such that $\vec{r}(0) = \vec{0}$.

$$\text{Recall } \frac{d\vec{r}}{dt} = \vec{v}$$

$$\begin{aligned} \Rightarrow \vec{r}(t) &= \int_0^t dt' \vec{v}(t') \\ &= \int_0^t dt' \left[\vec{g} \tau + (\vec{v}_0 - \vec{g} \tau) e^{-t'/\tau} \right] \\ &= \vec{g} \tau t' \Big|_0^t + (\vec{v}_0 - \vec{g} \tau)(-\tau) e^{-t/\tau} \Big|_0^t \end{aligned}$$

$$\boxed{\vec{r}(t) = \vec{g} \tau t + \tau(\vec{v}_0 - \vec{g} \tau)(1 - e^{-t/\tau})}$$

If drag is negligible, $b \ll 1 \Rightarrow \tau^{-1} = \frac{b}{n} \ll 1$

Taylor expansion $e^x = 1 + x + \frac{1}{2}x^2 + \mathcal{O}(x^3)$

$$\Rightarrow e^{-t/\tau} = 1 - \frac{t}{\tau} + \frac{1}{2}\left(\frac{t}{\tau}\right)^2 + \mathcal{O}(\tau^{-3})$$

So, $\vec{r}(t) = \vec{v}_0 t$

$$+ \tau (\vec{v}_0 - \vec{g} \tau) \left[1 - \left(1 - \frac{t}{\tau} + \frac{1}{2} \frac{t^2}{\tau^2} + \dots \right) \right]$$

$$= \cancel{\vec{g} \tau t} + (\vec{v}_0 - \cancel{\vec{g} \tau}) \left[t - \frac{1}{2} \frac{t^2}{\tau} + \mathcal{O}(\tau^{-2}) \right]$$

$$= \vec{v}_0 t + \frac{1}{2} \vec{g} t^2 - \frac{1}{2} \vec{v}_0 \tau t^2 + \mathcal{O}(\tau^{-2})$$

$$= \vec{v}_0 t + \frac{1}{2} \vec{g} t^2 + \mathcal{O}(\tau^{-1})$$

$\boxed{\downarrow}$ trajectory for motion w/o drag.

c) choosing coordinate system $\vec{g} = (0, 0, -g)$
and $\vec{v}_0 = (v_0 \cos \theta_0, 0, v_0 \sin \theta_0)$

our equations for $\vec{r}(t)$ in this system are

$$\begin{cases} x(t) = \tau v_0 \cos \theta_0 (1 - e^{-t/\tau}) \\ y(t) = 0 \\ z(t) = -g \tau t + \tau (v_0 \sin \theta_0 + g \tau) (1 - e^{-t/\tau}) \end{cases}$$

Solving for $t = t(x)$,

$$\frac{x}{\tau v_0 \cos \theta_0} = 1 - e^{-t/\tau} \Rightarrow e^{-t/\tau} = 1 - \frac{x}{\tau v_0 \cos \theta_0}$$

$$\text{or, } t = -\tau \ln \left(1 - \frac{x}{\tau v_0 \cos \theta_0} \right)$$

so, $z = z(x)$ is

$$z(x) = g\tau^2 \ln \left(1 - \frac{x}{\tau v_0 \cos \theta_0} \right) + \frac{(v_0 \sin \theta_0 + g\tau)}{v_0 \cos \theta_0} x$$

When $x = R$, $z(R) = 0$

$$\Rightarrow 0 = \tan \theta_0 R + \frac{g\tau}{v_0 \cos \theta_0} R + g\tau^2 \ln \left(1 - \frac{R}{\tau v_0 \cos \theta_0} \right)$$

If drag is small, $b \sim \tau^{-1} \ll 1$,

Taylor expansion $\ln(1+x) = x - \frac{x^2}{2} + \frac{x^3}{3} + O(x^4)$

$$\begin{aligned} \Rightarrow 0 &= \tan \theta_0 R + \cancel{\frac{g\tau}{v_0 \cos \theta_0} R} \\ &\quad + g\tau^2 \left[\cancel{-\frac{R}{2\tau^2 v_0^2 \cos^2 \theta_0}} - \frac{R^2}{2\tau^2 v_0^2 \cos^2 \theta_0}, - \frac{R^3}{3\tau^3 v_0^3 \cos^3 \theta_0} + \dots \right] \end{aligned}$$

Since $R = 0$ is not physical,

$$\Rightarrow 0 = \tan \theta_0 - \frac{g}{2v_0^2 \cos^2 \theta_0} R - \frac{g}{3\tau v_0^3 \cos^3 \theta_0} R^2 + O(\tau^{-2})$$

d) Now, we construct expansion

$$R = R_0 - \alpha \tau^{-1} + O(\tau^{-2})$$

with $\alpha > 0$, & R_0 = large with no drag.

We want to determine $\Delta R = R_0 - R$.

Inserting $R = R_0 - \alpha \tau^{-1}$ into

$$O = \tan \theta_0 - \frac{g}{2v_0^2 \cos^2 \theta_0} R - \frac{g}{3\tau v_0^3 \cos^3 \theta_0} R^2 + O(\tau^{-2})$$

$$\Rightarrow O = \tan \theta_0 - \frac{g}{2v_0^2 \cos^2 \theta_0} (R_0 - \alpha \tau^{-1})$$

$$- \frac{g}{3\tau v_0^3 \cos^3 \theta_0} \frac{(R_0^2 - 2\alpha R_0 \tau^{-1} + \alpha^2 \tau^{-2})}{\downarrow \text{can ignore} - O(\tau^{-1})} + O(\tau^{-2})$$

$$\Rightarrow O = \left(\tan \theta_0 - \frac{g}{2v_0^2 \cos^2 \theta_0} R_0 \right)$$

$$+ \left(\frac{g}{2v_0^2 \cos^2 \theta_0} \alpha - \frac{g}{3v_0^3 \cos^3 \theta_0} R_0^2 \right) \tau^{-1} + O(\tau^{-2})$$

End coefficient of τ^{-1} must vanish

$$O(\tau^0) : \tan \theta_0 = \frac{g}{2v_0^2 \cos^2 \theta_0} R_0$$

$$\Rightarrow R_0 = \frac{2v_0^2}{g} \sin \theta_0 \cos \theta_0$$

\downarrow page with no drag

$$O(\tau^{-1}): \frac{g}{2v_0^2 \cos^2 \theta_0} \alpha - \frac{g}{3v_0^3 \cos^3 \theta_0} R_0^2 = 0$$

$$\begin{aligned}\Rightarrow \alpha &= \frac{2}{3v_0 \cos \theta_0} R_0^2 \\ &= \frac{2}{3v_0 \cos \theta_0} \left(\frac{4v_0^4}{g^2} \sin^2 \theta_0 \cos^2 \theta_0 \right) \\ &= \frac{8}{3} \frac{v_0^3}{g^2} \sin^2 \theta_0 \cos \theta_0\end{aligned}$$

Therefore $R = R_0 - \alpha \tau^{-1}$

$$\begin{aligned}\text{So, } \Delta R &= R_0 - R \\ &= \alpha \tau^{-1}\end{aligned}$$

$$\begin{aligned}\Rightarrow \Delta R &= \frac{8}{3} \frac{v_0^3}{g^2} \sin^2 \theta_0 \cos \theta_0 \\ &= \frac{4}{3} \frac{v_0^3}{g^2} \sin 2\theta_0 \cos \theta_0\end{aligned}$$

$$\sin 2\theta = 2 \sin \theta \cos \theta$$

Note that $v_0 > 0, g > 0, \sin \theta_0 > 0$ & $\cos \theta_0 > 0$ for $0 \leq \theta_0 \leq \frac{\pi}{2}$

So, $\Delta R > 0$, thus the shift decreases