# CONTROL SYSTEMS (UE21EC241B) PROJECT ROLL ANGLE CONTROL

PROFESSOR: TIPPESWAMY.E

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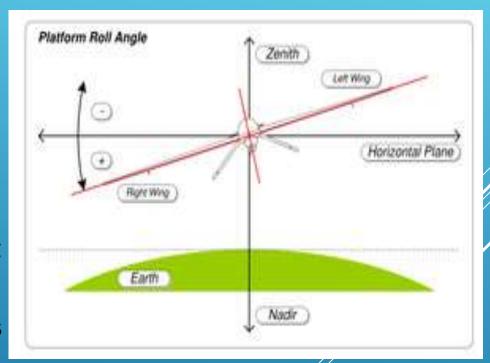
ABHISHEK A SHETTY(PES1UG21EC008)

AKASH C K(PES1UG21EC022)

SECTION: A

# INTRODUCTION

- ➤ Roll angle control (RAC) is required for the lateral stability of an aircraft. Lateral stability makes the aircraft more stable around the longitudinal axis.
- Roll angle control makes both the wings of the aircraft to be at the same level.
- If one of the wing dips below the other, then RAC tries to stabilize the system again



¿ Poll angle control3 Equalion of modion too a sioid body undergoing roll modion. It selates the soll angle output to the input control signal. moment of Inertice:

IXX ITYY IZZ Equation of motiontox sall mater. Ixx 0 "CD)+ CIZZ -Iyy)\* wyd + (DZ (4) = Z 200 + IXX STE SOCSH (IZZ IX) \* Wys) C By taking Laplace gransform) OCS ) / 1 | Z DC(S) - (IZZ -IV)

These the tronsfer firetion of can be worther as-Mcs)=0(s)/ (xcs)=1/ The open loop transfer function is -36.6 53+9252+13,45

▶ The open loop transfer function is

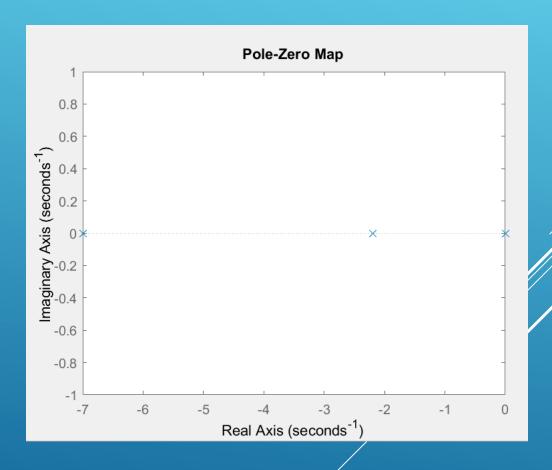
36.6

s^3 + 9.2 s^2 + 15.4 s

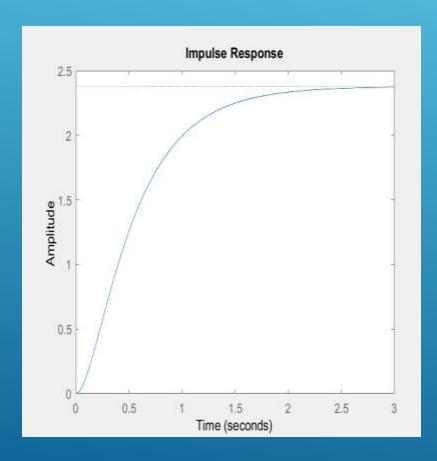
We first find the pole zero map first and it can be observed that the system is open loop stable

Poles are present at 0,-2,-7

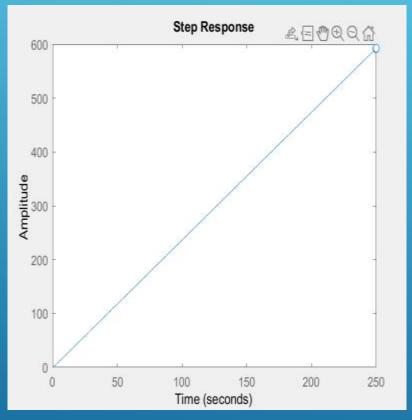
We apply different test signals and observe that the system is stable only for impulse input and unstable for step and ramp input



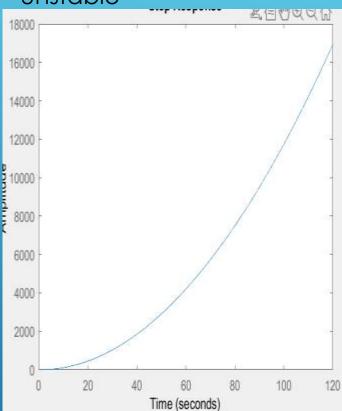
From the impulse response we can observe that the response settles to a value and hence we can call it stable



From the step response we can observe that the response is continuously increasing and unstable

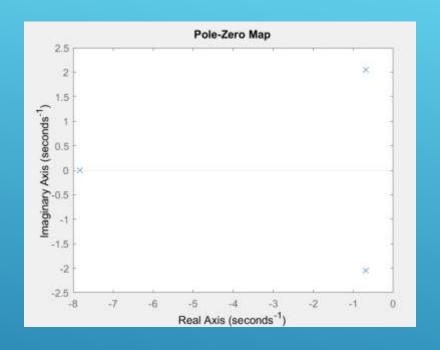


From the ramp response we can observe that the response is continuously increasing and unstable

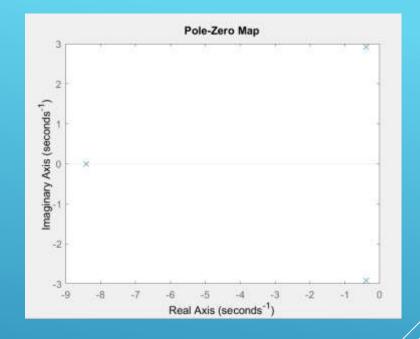


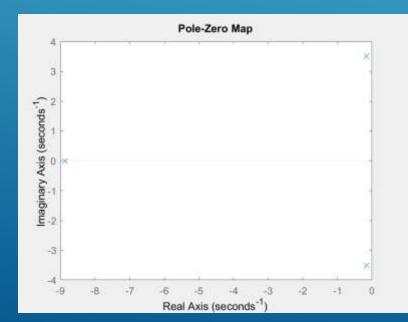
# Q2

- We apply a gain K in cascade with our initial transfer function
- The theoretical method is using the RH criterion, where we use the RH table to determine the value of K.
- ▶ Using this method we find that the for K < 3.87 the system is stable</p>
- To confirm this we use the pole zero plots
- > We apply different integer value of K in cascade with our transfer function
- ➤ The poles lie on LHP for K<=3
- > The system becomes unstable when we take K = 4

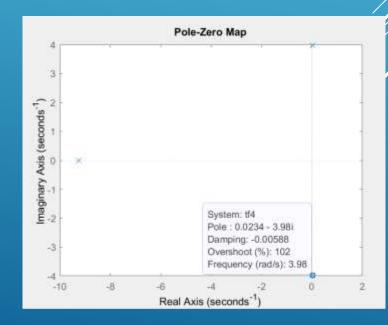












# Q3

- For the proportional controller we can take different values of K
- For k = 1, we obtain the following characteristics

RiseTime: 0.6592

TransientTime: 5.2933

SettlingTime: 5.2933

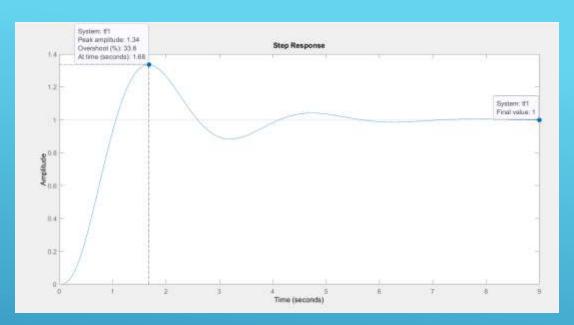
SettlingMin: 0.8825

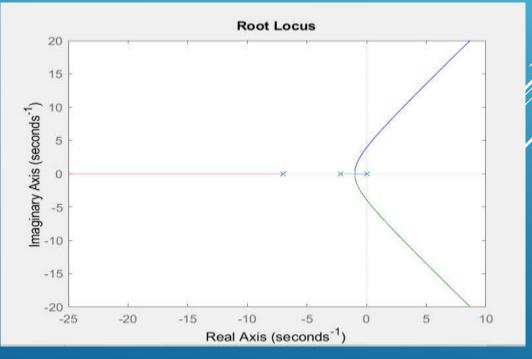
SettlingMax: 1.3359

Overshoot: 33.5871

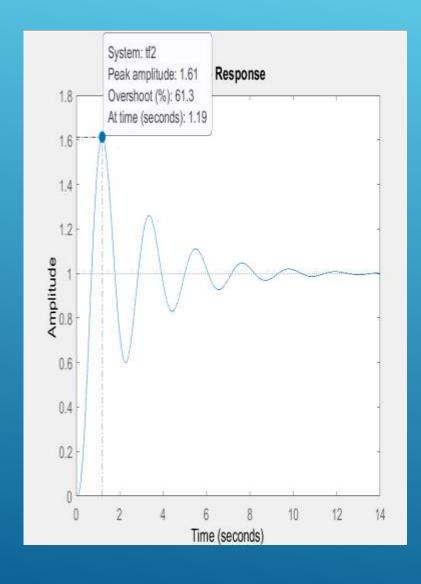
Undershoot: 0

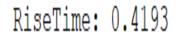
Peak: 1.3359





#### For K = 2





TransientTime: 9.8096

SettlingTime: 9.8096

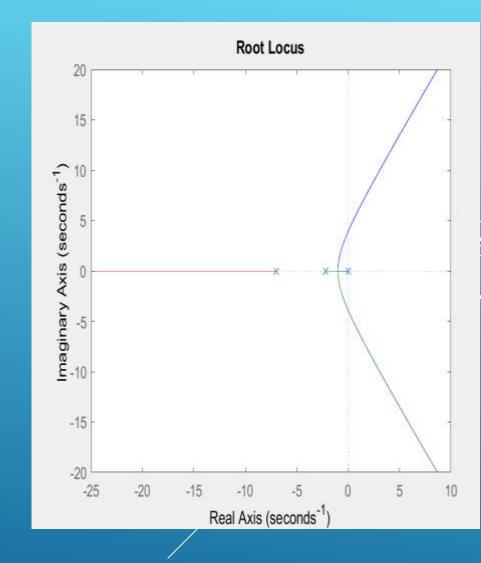
SettlingMin: 0.6003

SettlingMax: 1.6130

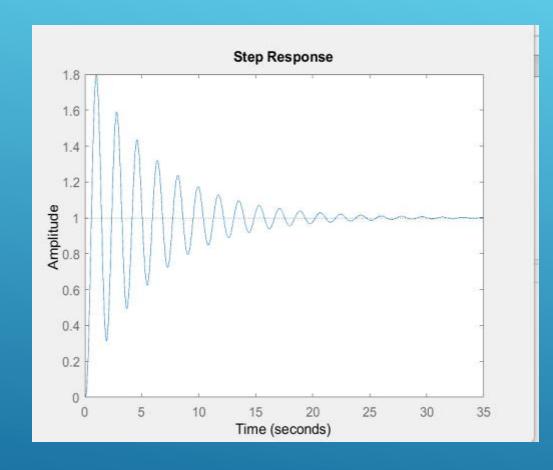
Overshoot: 61.3017

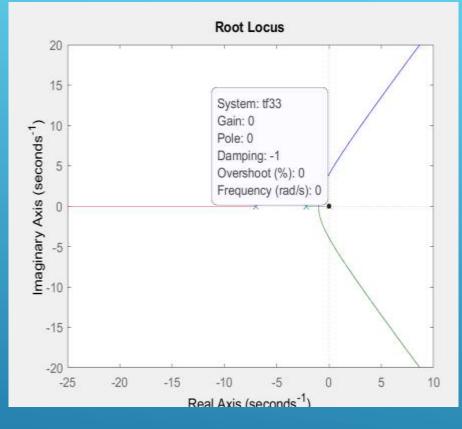
Undershoot: 0

Peak: 1.6130



For K = 3





RiseTime: 0.3330

TransientTime: 22.5317

SettlingTime: 22.5317

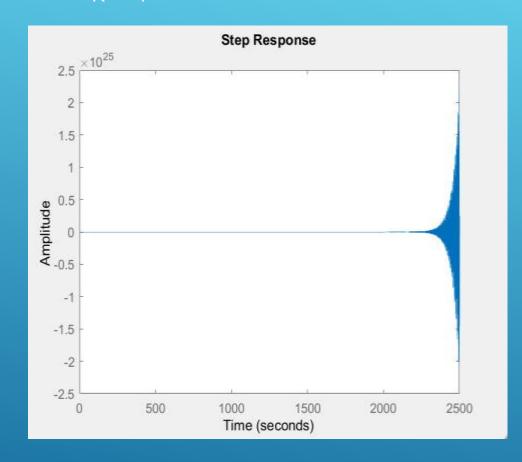
SettlingMin: 0.3151

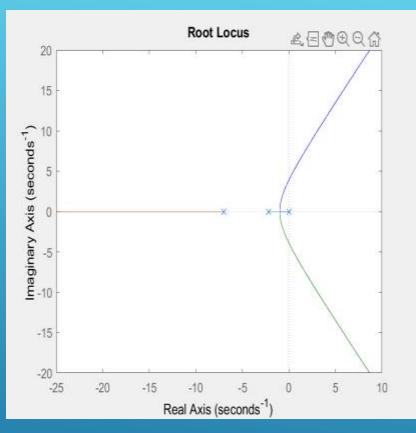
SettlingMax: 1.7970

Overshoot: 79.7043

Undershoot: 0

Peak: 1.7970





RiseTime: NaN

TransientTime: NaN

SettlingTime: NaN

SettlingMin: NaN

SettlingMax: NaN

Overshoot: NaN

Undershoot: NaN

Peak: Inf

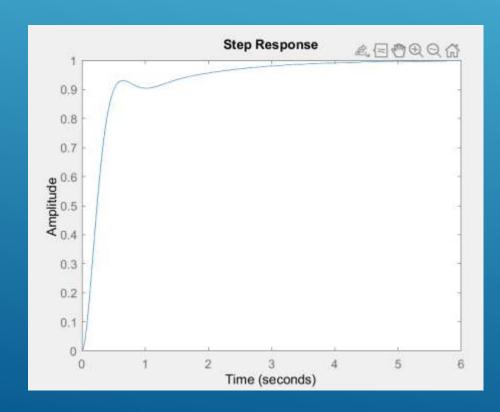
PeakTime: Inf

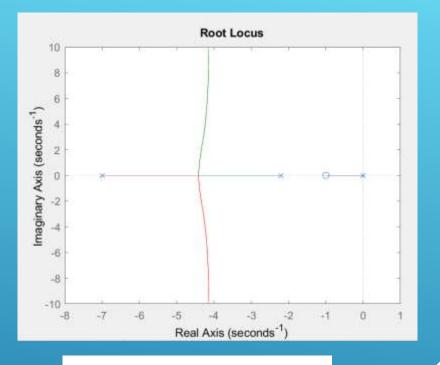
We observe that for K = 4, the system becomes unstable which is evident from observing the step response graph

We also observe that the rise time is least for K = 3 and hence it's the most suitable proportional controller for our system

# Q4

- We introduce a new PD controller into the system by cascading K(s+z) with our system. We choose different values of K and z and obtain the step response for each and determine which is more suitable for our system
- $\triangleright$  K = 1 and z = 1





RiseTime: 0.4233

TransientTime: 2.9633

SettlingTime: 2.9633

SettlingMin: 0.9043

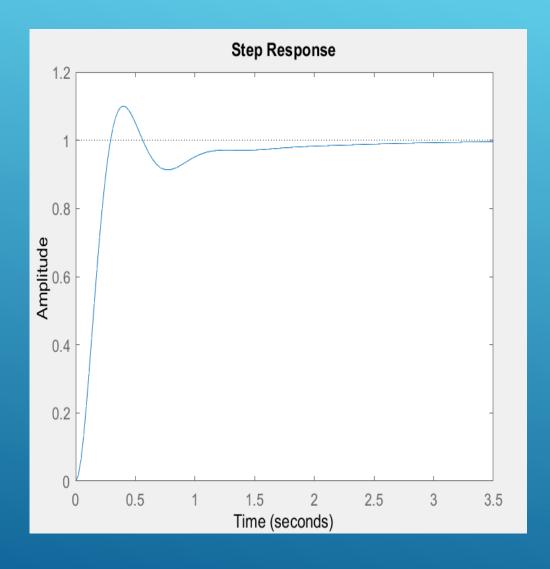
SettlingMax: 0.9985

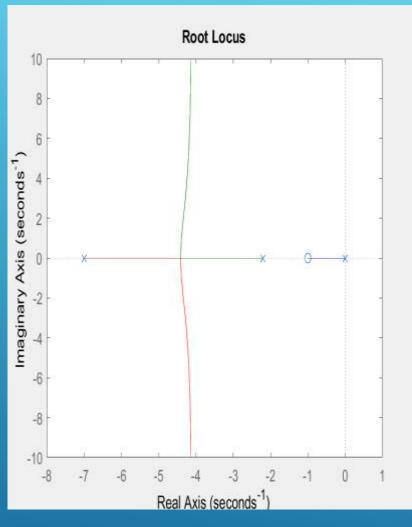
Overshoot: 0

Undershoot: 0

Peak: 0.9985

#### K = 2, z = 1





RiseTime: 0.1966

TransientTime: 1.8474

SettlingTime: 1.8474

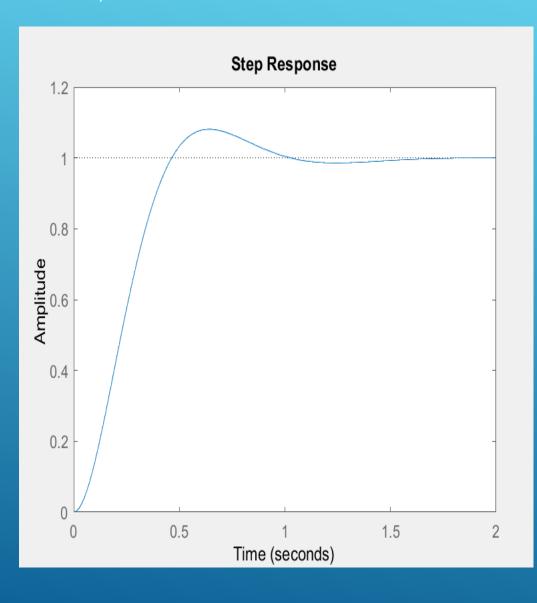
SettlingMin: 0.9132

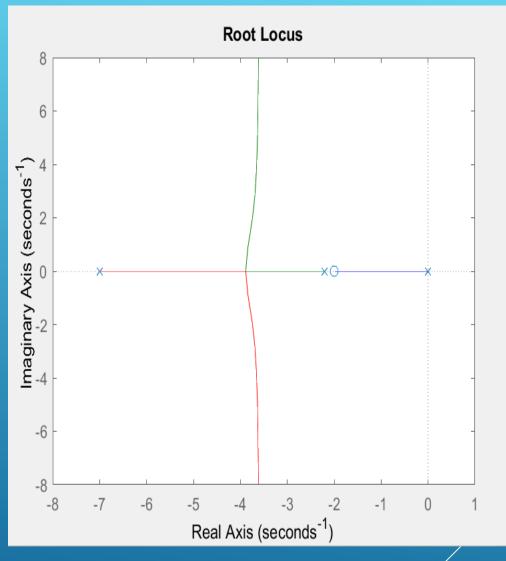
SettlingMax: 1.1007

Overshoot: 10.0727

Undershoot: 0

Peak: 1.1007





RiseTime: 0.3091

TransientTime: 0.9239

SettlingTime: 0.9239

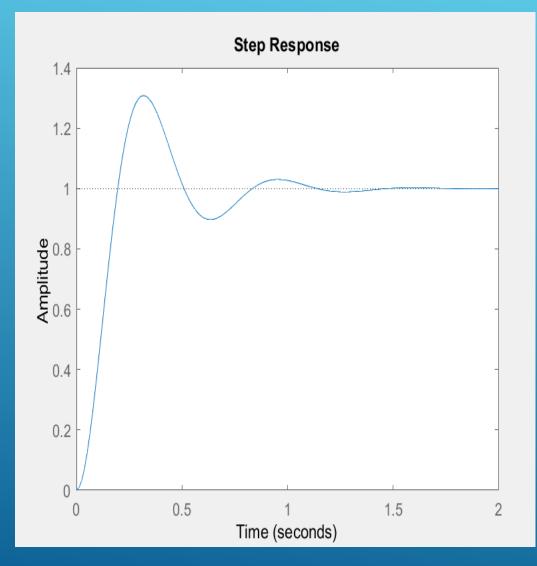
SettlingMin: 0.9028

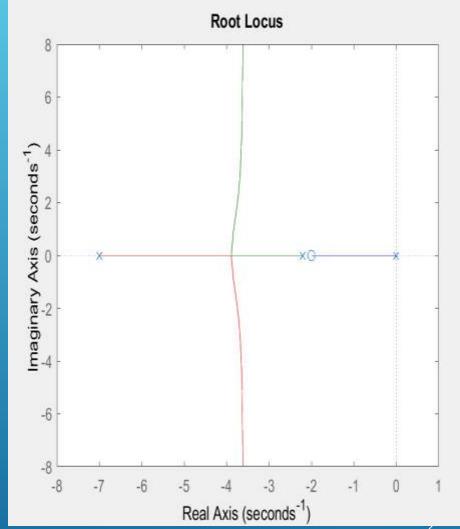
SettlingMax: 1.0810

Overshoot: 8.1000

Undershoot: 0

Peak: 1.0810





RiseTime: 0.1324

TransientTime: 1.0404

SettlingTime: 1.0404

SettlingMin: 0.8970

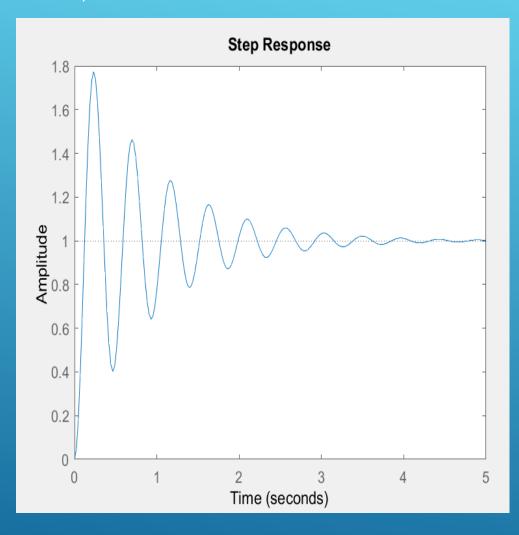
SettlingMax: 1.3084

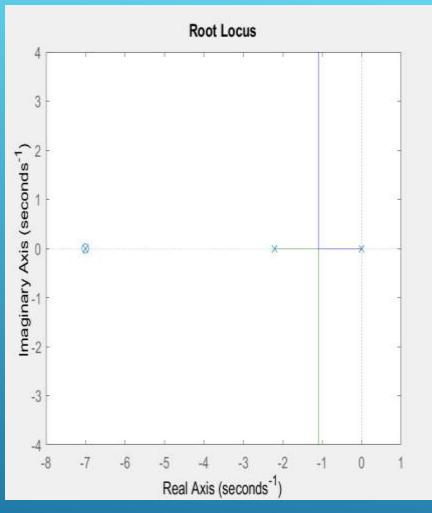
Overshoot: 30.8438

Undershoot: 0

Peak: 1.3084

#### K = 5, z = 7





RiseTime: 0.0821

TransientTime: 3.5198

SettlingTime: 3.5198

SettlingMin: 0.4012

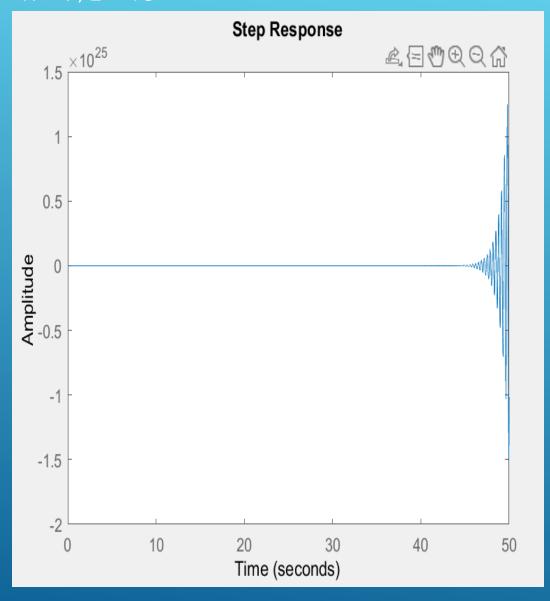
SettlingMax: 1.7739

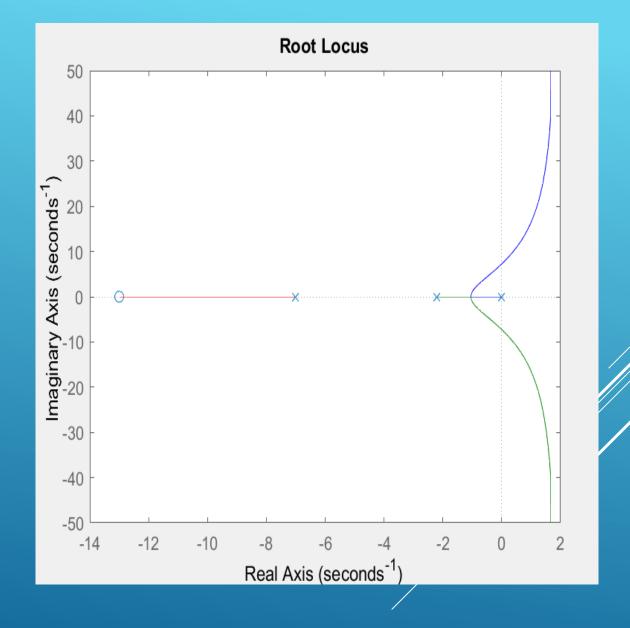
Overshoot: 77.3863

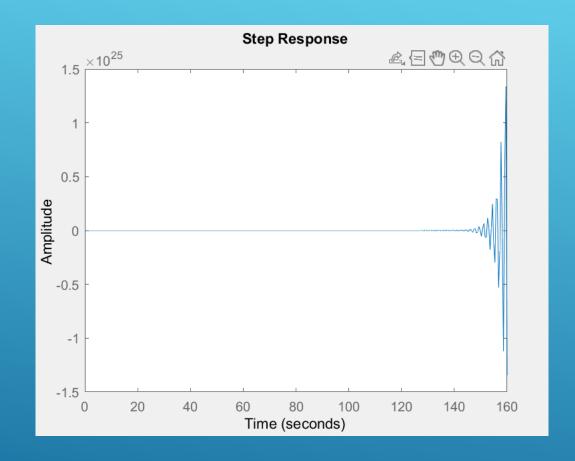
Undershoot: 0

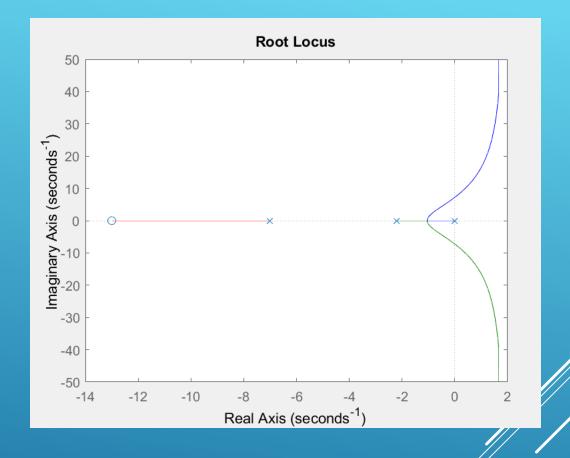
Peak: 1.7739

#### K = 9, z = 13



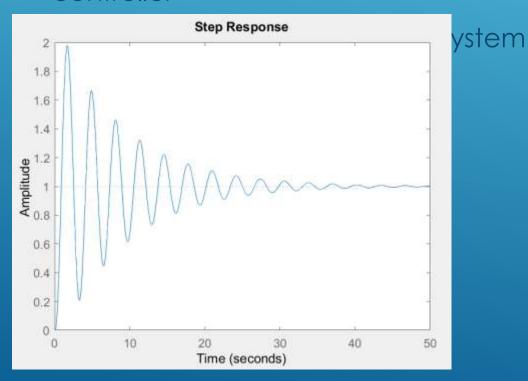


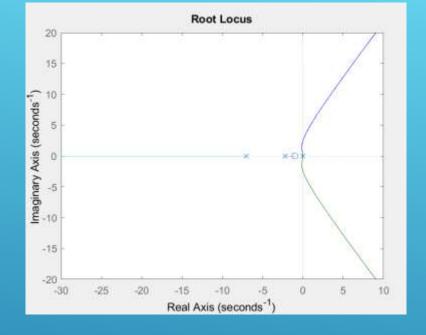




We observe that the introduction of zero allows for the system to have lower rise time for the same value of K. It makes the system more stable and allows higher values of K to be chosen, but there is a limit to this too. As we have observed from our examples, if the value of K is too high (in our example K = 9) the system tends to be unstable. Even in the case of the zero being non dominating, i.e, far away from the origin, the system ends up being unstable (as seen in our example z = 13)

 Adding a pole to our system is equivalent to have an integral controller

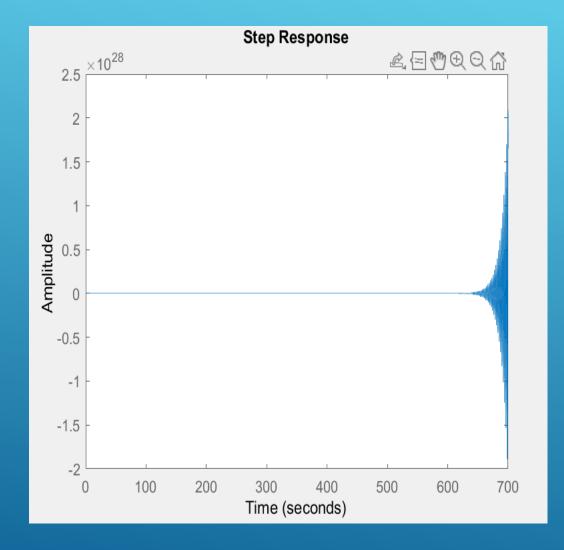


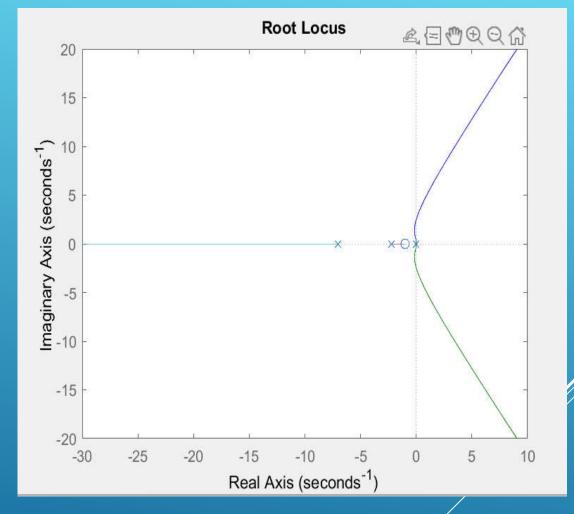


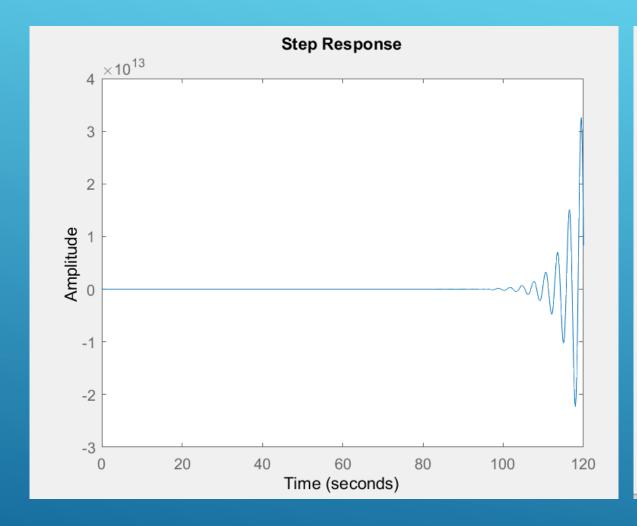
RiseTime: 0.5224
TransientTime: 35.5850
SettlingTime: 35.5850
SettlingMin: 0.2080
SettlingMax: 1.9781
Overshoot: 97.8142

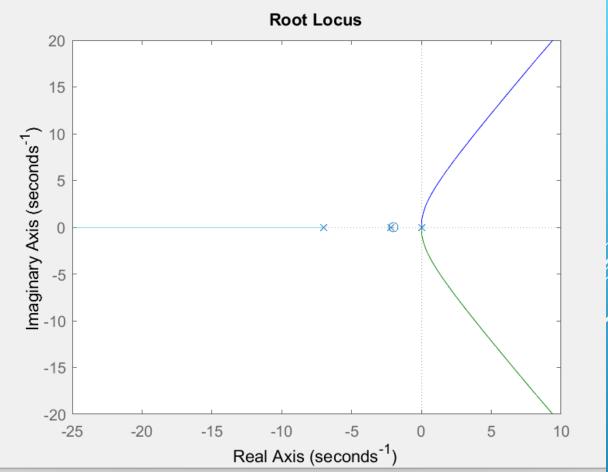
Undershoot: 0

Peak: 1.9781 PeakTime: 1.6454



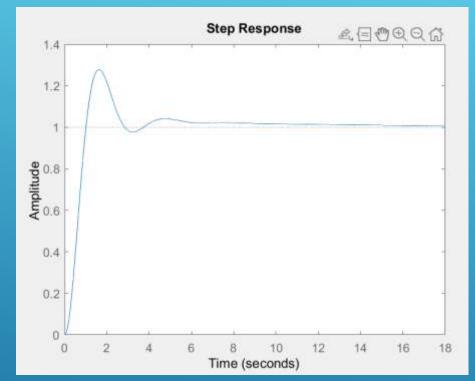


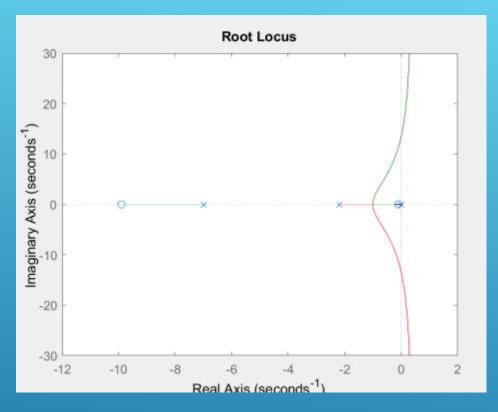




We observe from our experiments that the addition of a pole to our particular system ends up making it more unstable, rather than stable, limiting the value of K that can be chosen

# K1 = 1, K2 = 0.1, K3 = 0.1





RiseTime: 0.6738

TransientTime: 8.2502

SettlingTime: 8.2502

SettlingMin: 0.9471

SettlingMax: 1.2776

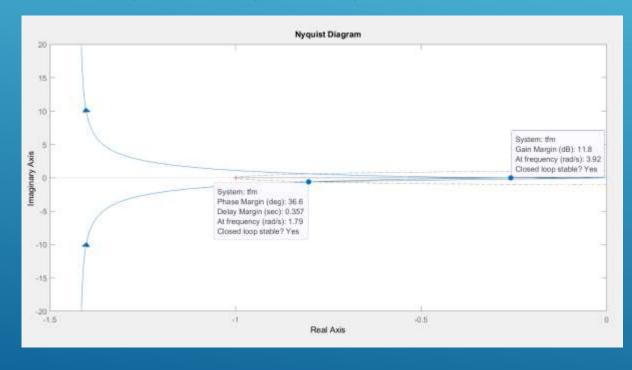
Overshoot: 27.7630

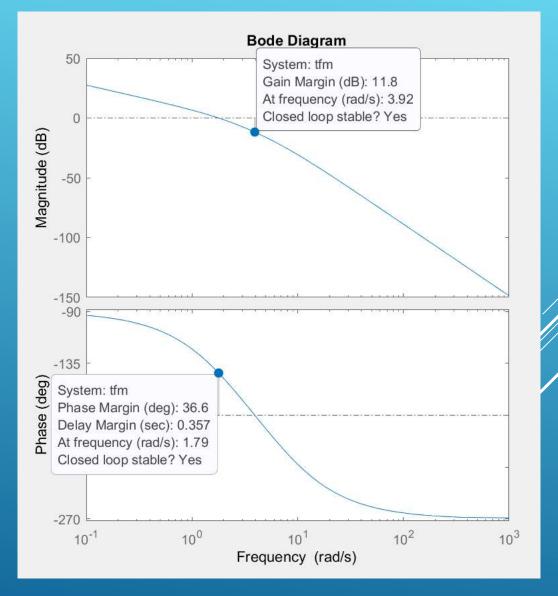
Undershoot: 0

Peak: 1.2776

## Q6

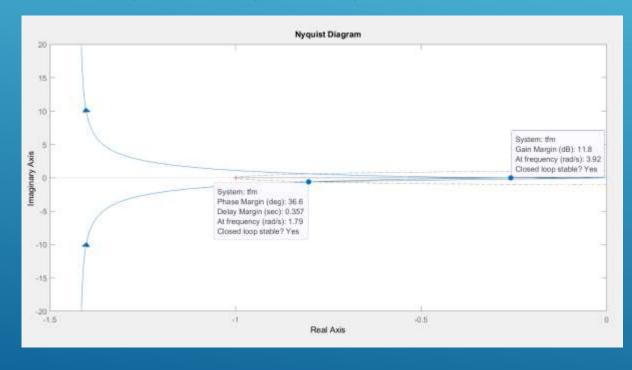
- Bode and Nyquist plots are used to obtain the stability of the system using the Open loop transfer function
- From these plots we observe that our system is open loop stable

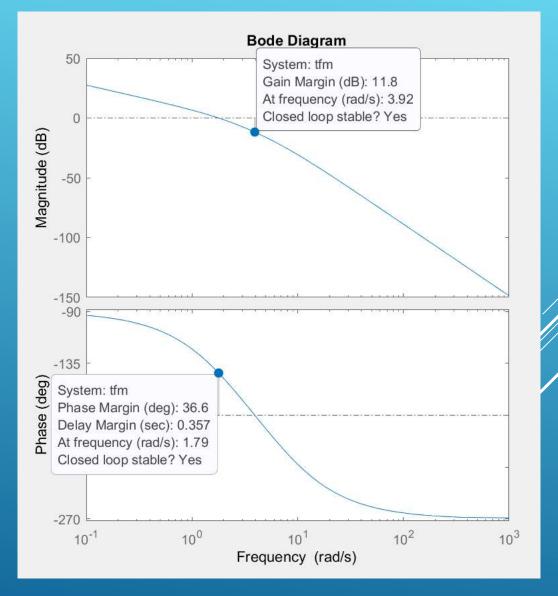




## Q6

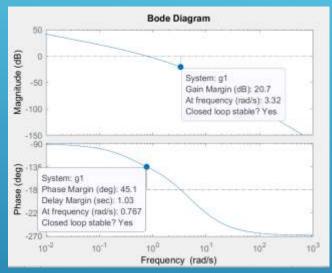
- Bode and Nyquist plots are used to obtain the stability of the system using the Open loop transfer function
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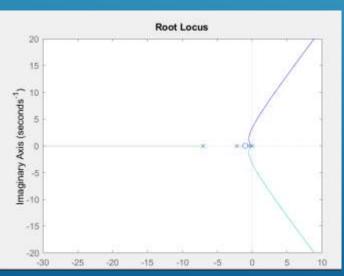


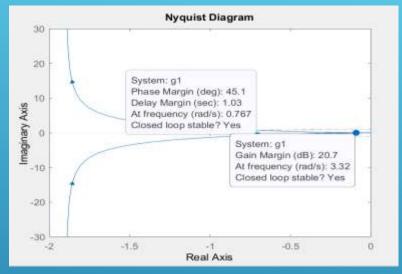


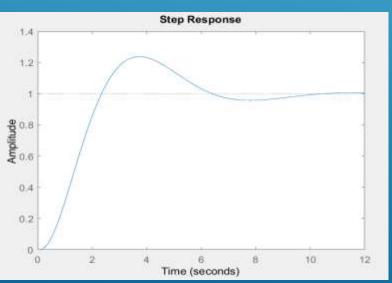
# LAG COMPENSATOR

#### Alpha = 2, Tau = 1









RiseTime: 1.5294

TransientTime: 9.3005

SettlingTime: 9.3005

SettlingMin: 0.9049

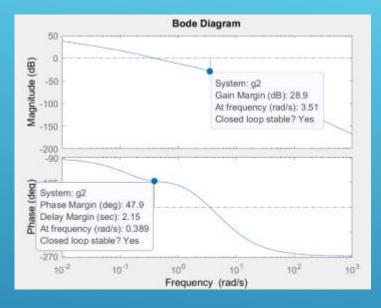
SettlingMax: 1.2381

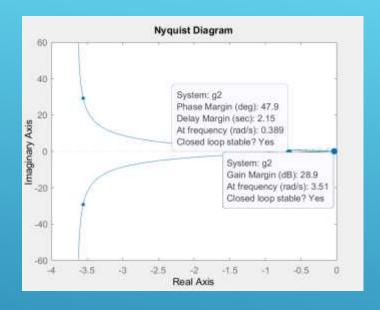
Overshoot: 23.8112

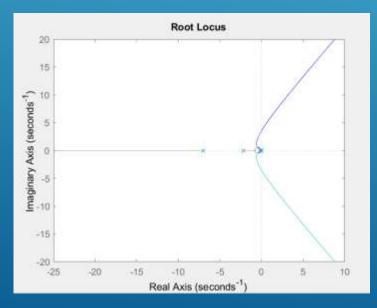
Undershoot: 0

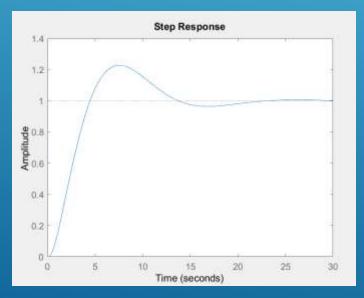
Peak: 1.2381

#### Alpha = 3, Tau = 2









RiseTime: 3.0675

TransientTime: 19.8752

SettlingTime: 19.8752

SettlingMin: 0.9002

SettlingMax: 1.2283

Overshoot: 22.8343

Undershoot: 0

Peak: 1.2283

#### Alpha = 4, Tau = 7

20

15

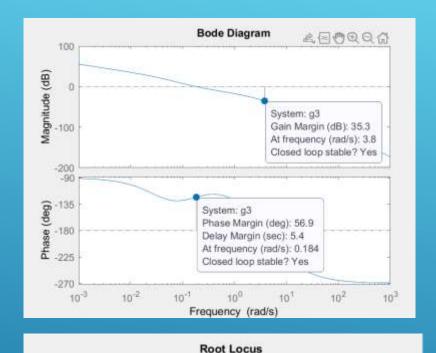
Imaginary Axis (seconds<sup>-1</sup>)

-15

-20

-25

-20



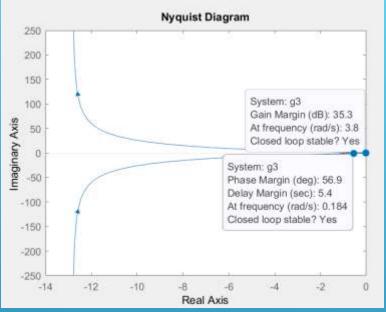
-10

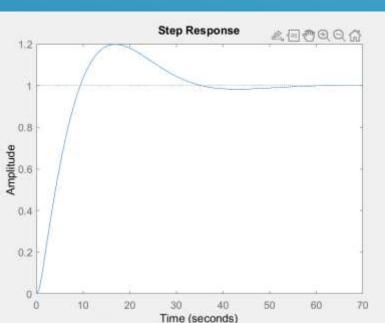
Real Axis (seconds<sup>-1</sup>

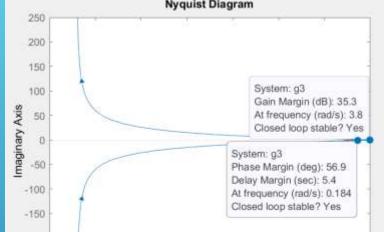
5

10

-15







RiseTime: 6.7323

TransientTime: 32.5073

SettlingTime: 32.5073

SettlingMin: 0.9320

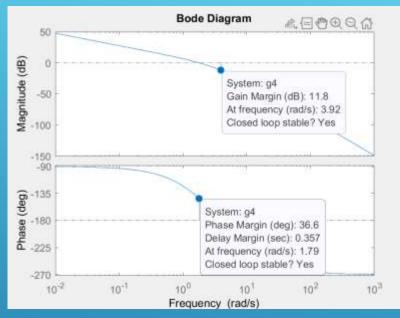
SettlingMax: 1.1982

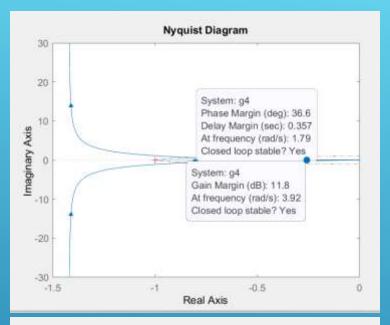
Overshoot: 19.8185

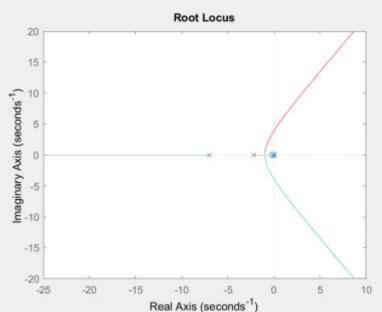
Undershoot: 0

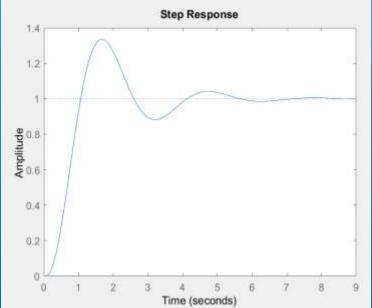
Peak: 1.1982

#### Alpha = 1, Tau = 7









RiseTime: 0.6592

TransientTime: 5.2933

SettlingTime: 5.2933

SettlingMin: 0.8825

SettlingMax: 1.3359

Overshoot: 33.5871

Undershoot: 0

Peak: 1.3359

#### Alpha = 9, Tau = 5

15

Imaginary Axis (seconds<sup>1</sup>)

-15

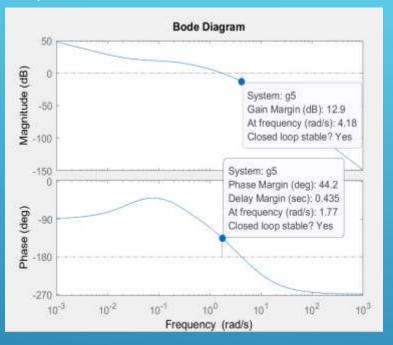
-20

-20

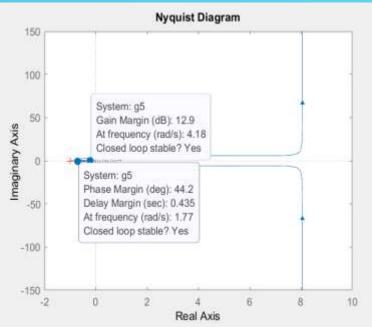
-15

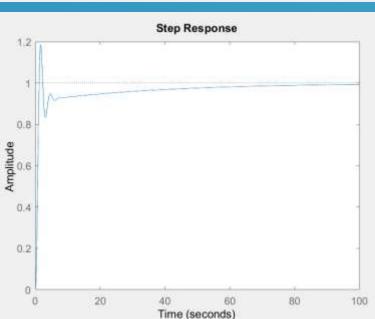
-10

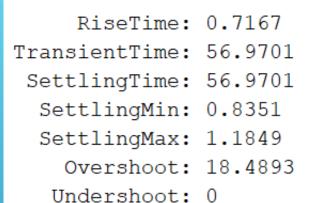
Real Axis (seconds<sup>-1</sup>



**Root Locus** 





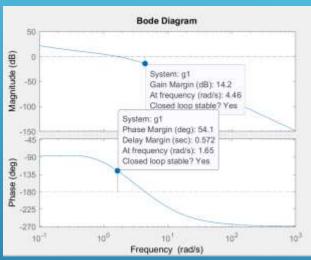


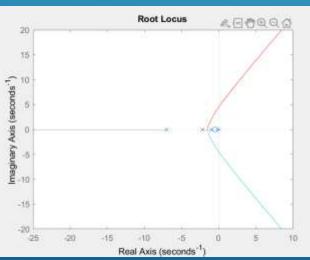
Peak: 1.1849 PeakTime: 1.6312

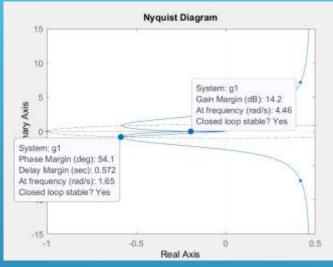
From these experiments we observe that the value of Tau must be large and Alpha be small for the best response (lowest rise time)

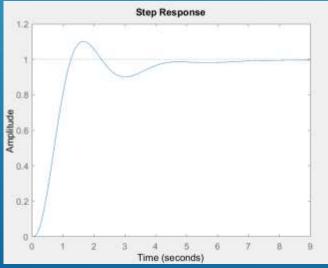
# LEAD COMPENSATOR

#### Alpha = 2, Tau = 1









RiseTime: 0.7877

TransientTime: 4.3211

SettlingTime: 4.3211

SettlingMin: 0.9001

SettlingMax: 1.1020

Overshoot: 10.2021

Undershoot: 0

Peak: 1.1020

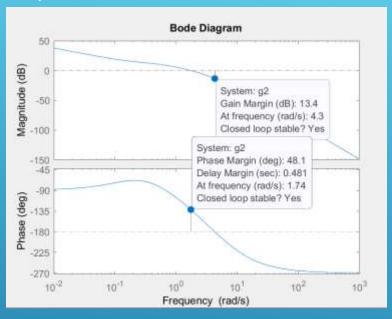
#### Alpha = 3, Tau = 2

15

Imaginary Axis (seconds<sup>-1</sup>)

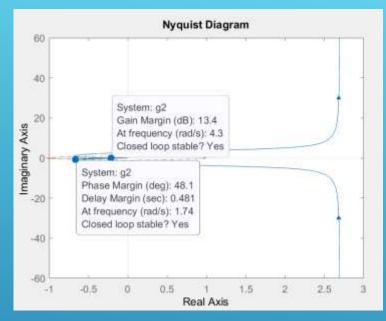
-15

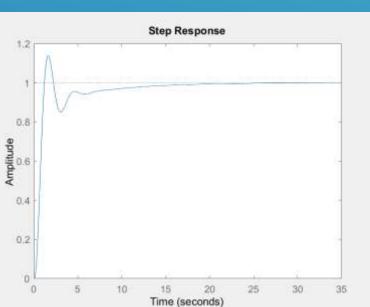
-20



**Root Locus** 

Real Axis (seconds<sup>-1</sup>)





RiseTime: 6.7323

TransientTime: 32.5073

SettlingTime: 32.5073

SettlingMin: 0.9320

SettlingMax: 1.1982

Overshoot: 19.8185

Undershoot: 0

Peak: 1.1982

#### Alpha = 4, Tau = 7

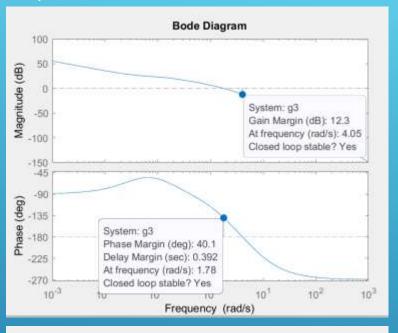
15

-15

-20

-25

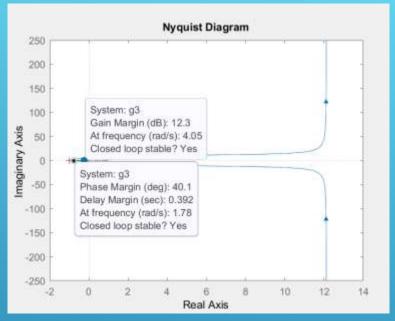
-20

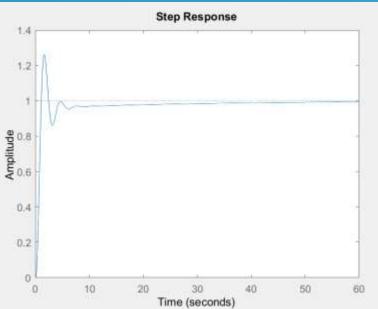


**Root Locus** 

-10

Real Axis (seconds<sup>-1</sup>)





RiseTime: 0.6826

TransientTime: 22.7317

SettlingTime: 22.7317

SettlingMin: 0.8611

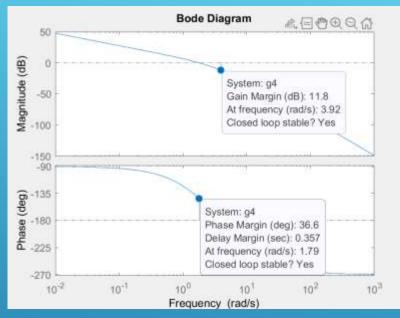
SettlingMax: 1.2611

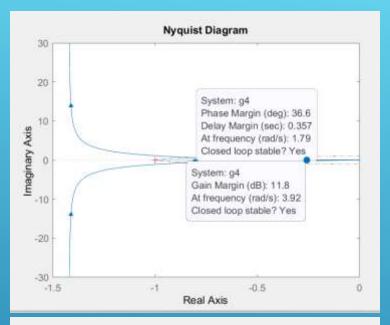
Overshoot: 26.1147

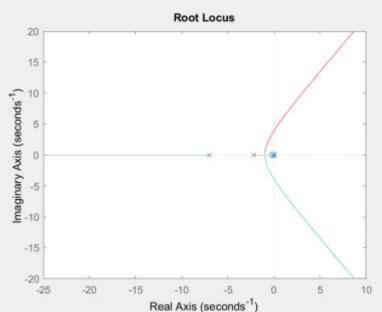
Undershoot: 0

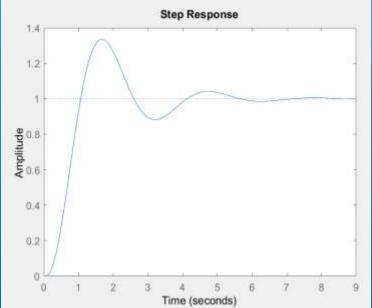
Peak: 1.2611

#### Alpha = 1, Tau = 7









RiseTime: 0.6592

TransientTime: 5.2933

SettlingTime: 5.2933

SettlingMin: 0.8825

SettlingMax: 1.3359

Overshoot: 33.5871

Undershoot: 0

Peak: 1.3359

#### Alpha = 9, Tau = 5

15

Imaginary Axis (seconds<sup>1</sup>)

-15

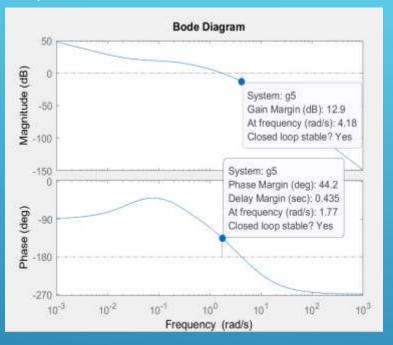
-20

-20

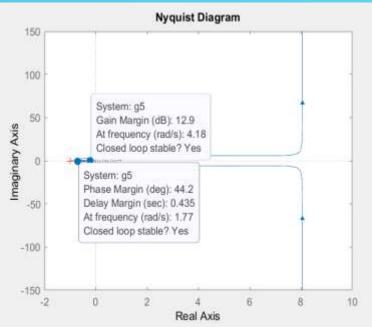
-15

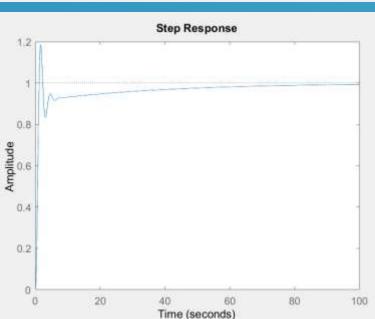
-10

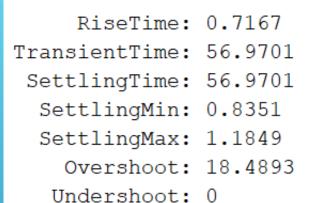
Real Axis (seconds<sup>-1</sup>



**Root Locus** 







Peak: 1.1849 PeakTime: 1.6312

From these experiments we observe that the value of Tau must be large and Alpha be small for the best response (lowest rise time)

# THANK YOU