# Model description for the implementation of System Identification methodologies on the rotorcraft SH09

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| L        | ist of symbols  |          |   |
| $\alpha$ | Angle of attack   | $[\deg]$ |   |

# 1 Introduction

Bibliography: [1], [2], [3], [4]

## 2 Model description

#### 2.1 Model forces and moments coefficients

F matrix Forces:

$$X_u, X_v, X_w, X_p, X_q, X_rY_u, Y_v, Y_w, Y_p, Y_q, Y_rZ_u, Z_v, Z_w, Z_p, Z_q, Z_r$$

Moments:

$$L_u, L_v, L_w, L_p, L_q, L_r M_u, M_v, M_w, M_p, M_q, M_r N_u, N_v, N_w, N_p, N_q, N_r$$

Controllability, G matrix: Forces

$$X_{\delta_{\mathrm{lon}}}, X_{\delta_{\mathrm{lat}}}, X_{\delta_{\mathrm{ped}}}, X_{\delta_{\mathrm{col}}} Y_{\delta_{\mathrm{lon}}}, Y_{\delta_{\mathrm{lat}}}, Y_{\delta_{\mathrm{ped}}}, Y_{\delta_{\mathrm{col}}} Z_{\delta_{\mathrm{lon}}}, Z_{\delta_{\mathrm{lat}}}, Z_{\delta_{\mathrm{ped}}}, Z_{\delta_{\mathrm{col}}}$$

Moments

$$M_{\delta_{\mathrm{lon}}}, M_{\delta_{\mathrm{lat}}}, M_{\delta_{\mathrm{ped}}}, M_{\delta_{\mathrm{col}}} N_{\delta_{\mathrm{lon}}}, N_{\delta_{\mathrm{lat}}}, N_{\delta_{\mathrm{ped}}}, N_{\delta_{\mathrm{col}}} N_{\delta_{\mathrm{lon}}}, N_{\delta_{\mathrm{lat}}}, N_{\delta_{\mathrm{ped}}}, N_{\delta_{\mathrm{col}}}$$

Time delays

$$\tau_{\mathrm{lon}}, \tau_{\mathrm{lat}}, \tau_{\mathrm{ped}}, \tau_{\mathrm{col}}$$

# References

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- [3] G. Morelli and S. Derry, "System Identification Methods for Aerodynamic Modeling and Validation using Flight Data," 2011.
- [4] G. Morelli, "Aircraft System Identification," 2011.