Homework 2:

20/20

Deadline: Thursday 28 November 2024 (by 19h00)

Credits: 20 points

Really excellent code. It runs smoothly and is efficient.

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Instructions:

- When you finish, please send me the .ipynb file via email to wbanda@yachaytech.edu.ec
- This homework can be submitted **individually or in pairs**. Please include your name/s in the notebook.
- Within a **single python notebook**, solve the following problem:

Shocks in subsonic and supersonic turbulence

We want to study the properties of subsonic and supersonic turbulent flows in 2D. Shock waves are flow discontinuities that arise in turbulent gases when the local flow speed exceeds the sound speed of the gas. Let us consider the following high-resolution simulations of turbulence-in-a-box models:

Supersonic turbulence simulation:

https://yachaytecheduec-

 $my. share point.com/:f:/g/personal/wbanda_yachaytech_edu_ec/EhXQXyn8GudCnZ15af_fLdEBcbpt7h\\ e=IlgN7g$

Subsonic turbulence simulation:

https://yachaytecheduec-

my.sharepoint.com/:f:/g/personal/wbanda_yachaytech_edu_ec/EhNeXIEHX3JAjUYV44981LgBo9tn_E e=350N8b

As explained in class, these simulations introduce stochastic force fields to generate turbulent motions in an isothermal (\$\gamma=1\$) gas, which is initially at rest. Turbulence is continuously generated throughout the simulations.

The simulation folders contain 101 VTK files, jointly with:

- a units.out file that contains the CGS normalisation values and the isothermal sound speed (\$c_{\rm iso}\$).
- a vtk.out file whose second column contains the times in code units.

Each VTK file stores the following fields:

• density (rho)

- velocity_x (vx1)
- velocity_y (vx2)
- magnetic_field_x (Bx1)
- magnetic_field_y (Bx2)

Remember that you can use VisIt to inspect the data and check if your code produces consistent outputs.

1. (3 points) Data I/O functions:

Create a set of Python functions that:

(a) reads the **units.out** file, and returns the normalisation values for length, velocity, density, magnetic field, time, and isothermal sound speed (\$c_{\rm iso}\$) into callable objects.

Note: The normalisation values for the magnetic field ($B_0=\sqrt{4\,\nu_0\,\nu_0^2}$), and time ($L_0=\sqrt{\nu_0}$) can be derived from the length, velocity, and density values.

(b) opens any VTK file, reads the data arrays, and returns the 2D, CGS-normalised arrays for:

- density (rho)
- velocity_x (vx1)
- velocity_y (vx2)
- magnetic_field_x (Bx1)
- magnetic_field_y (Bx2)

Note: Use the normalisation values returned by the function from part (a) to convert from code units to CGS units.

(c) reads the 2D arrays returned by the function above, interpolates them into a CGS-normalised meshgrid created with the mesh information stored in the VTK files, and exports 3 figures containing maps of:

- density
- velocity vector field, scaled by its magnitude, \$v=\sqrt{v_x^2+v_y^2}\$.
- magnetic vector field, scaled by its magnitude: \$B=\sqrt{B_x^2+B_y^2}\$.

Notes:

- Choose different perceptually-uniform colour schemes for each of the above quantities.
- Since these are high-resolution models, one way to improve the visualisation of 2D vector fields is to rescale them onto a coarser grid.

2. (4 points) Data visualisation and density comparison:

Use python to carry out the following analysis:

- (d) Call the above functions for VTK file # 50 of each simulation, and make the following maps using the correct mesh coordinates and dimensions:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) density, \$\rho\$.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) velocity vector field, \$\vec{v}\$.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) magnetic vector field, \$\vec{B}\$\$.
- (e) Compute 1D histograms of the density and the velocity magnitude, and make the following figures:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) 1D histograms of the density.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) 1D histograms of the velocity magnitude.
- (f) What distributions do the density and velocity fields have? Do you see differences between the supersonic and subsonic distributions? Why?

3. (6 points) Numerical differentiation and shock candidates:

Create a set of Python functions that:

- (g) Isolate candidate shocked cells in binary fields, based on the following methods:
 - Method 1: Read a 2D velocity vector field. Compute the divergence of the velocity field and isolate the cells where there are convergent flows (i.e. where \$\vec\nabla\cdot \vec v <\alpha\$, where \$\alpha=0\$). Cells with convergent flows are candidate shocked cells. Try with \$\alpha\$ values slightly lower than \$0\$ for better results.
 - **Method 2:** Calculate the 2D pressure field using the equation of state of isothermal gas, i.e. \$p=\rho\,c_{\rm iso}^2\$. Compute the gradient of the pressure and isolate the cells with large pressure gradients (i.e. where \$\frac{|\vec\nabla P|}{P}>0.01\max{\left(\frac{|\vec\nabla P|}{P}\right)}\$). Such cells are candidate shocked cells.
- (h) Call your shock-finding function/s from (g) for VTK file # 50 of each simulation, and make binary maps of the resulting candidate shock cells from both methods and for both models. Show the results as follows:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 1.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 2.
- (i) Compute 1D histograms of the shock cell candidates from both methods and for both models, and make the following figures:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 1.

- A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 2.
- (j) Analyse your results, in particular:
 - Do the shock candidate results of methods 1 and 2 agree? Why?
 - Do you find similar distributions of shock candidates in both turbulence models? Which model has more shocks? Why?

4. (6 points) Numerical integration and custom outputs:

- (k) Create a python function that loops over all VTK files in a simulation, and saves maps (in both PNG and VTK-like format) of the shock cell candidates for all times. Add time stamps in physical units to the maps. Please don't attach any output files to your emails. Your codes should produce them when I run them locally.
- (I) Create a python function that loops over all VTK files, and computes the following integrated quantities for each time:
 - the total number of shock candidates, \$N_{\rm shocks}\$ computed from each method 1 and 2,
 - the (volume-weighted) average velocity dispersion, \$\sigma_v = \sqrt{[v^2] [v]^2}\$,
 - the rms Mach number, \${\cal M}_{\rm rms}\$, for which you need \$c_{\rm iso}\$ given in units.out,

and returns:

- a CSV file with 5 columns, time on the first column, and the above quantities in the next ones. There is no need to attach the CSV file to your emails, your code should produce the file locally when I run it.
- (m) Call your function from (l) for each simulation set (supersonic and subsonic) and use the CSV files to make the following plots:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) \$N_{\rm shocks}\$ time.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) \${\cal M}_{\rm rms}\$ versus time.
- (n) Analyse your results, in particular:
 - Does the flow reach steady state in both models? At what times?
 - Is there a relation between \$N_{\rm shocks}\$ and \${\cal M}_{\rm rms}\$?

5. (1 point) Shock animation:

- (o) Create a python function that returns movies for each simulation (supersonic and subsonic) showing the time evolution of:
 - maps of the shocks (printed in k), jointly with
 - the total number of shock candidates, \$N_{\rm shocks}\$, computed in (I).

Solution

1. (3 points) Data I/O functions

Create a set of Python functions that:

(a) reads the **units.out** file, and returns the normalisation values for length, velocity, density, magnetic field, time, and isothermal sound speed (\$c_{\rm iso}\$) into callable objects.

Note: The normalisation values for the magnetic field ($B_0=\sqrt{4\,\nu_0\,\nu_0^2}$), and time ($L_0=\sqrt{L_0}$) can be derived from the length, velocity, and density values.

```
In [1]: #Third party libraries
        import numpy as np
        import matplotlib.pyplot as plt
        import pandas as pd
        import pyvista as pv
        import os
        from skimage.transform import resize
        from pyevtk.hl import gridToVTK
        import scienceplots
        from PIL import Image
        import glob
        from IPython import display
        import scipy.optimize as opt
In [2]: #Define the paths to input simulation files
        path = "data_simulation" # This is the directory where the imulation files are
        sub_directory = "/TURB_DRIVE_SUB_hr" # Directory for subsonic simulation VTK files
        sup_directory = "/TURB_DRIVE_SUP_hr" # Directory for supersonic simulation VTK file
In [3]: plt.style.use(['science', 'notebook', 'no-latex']) #Use a specific style for figure
In [4]: #Function
        def io_norm_values(path):
            Function to read .cvs file and get normalization constants (CGS units)
            for a gas simulation.
            Input:
                File path (str)
            Output:
                Normalization constants for (CGS units):
                    rho: density
                    v: velocity
                    L: length
                    B: magnetic field
                    c_iso: isothermal sound speed
                    t: time
            Author: Alan Palma
            #Read with pandas
            data = pd.read_csv(path + "/units.out", sep = ",")
            #Put into pandas objects
            rho = np.array(data.loc[data["variable"]=="rho_0"]["normalisation"])
            v = np.array(data.loc[data["variable"]=="v_0"]["normalisation"])
            L = np.array(data.loc[data["variable"]=="L_0"]["normalisation"])
```

```
c_iso = np.array(data.loc[data["variable"]=="c_iso"]["normalisation"])

#Derive other normalization constants
B = np.sqrt(4*np.pi*rho*v**2) #magnetic field
t = L/v #time

return rho, v, L, B, c_iso, t
```

```
In [5]: # Call the function to get normalization units
rho_0, v_0, L_0, B_0, c_iso_0, t_0 = io_norm_values(path + sub_directory)
```

- (b) opens any VTK file, reads the data arrays, and returns the 2D, CGS-normalised arrays for:
 - density (rho)
 - velocity_x (vx1)
 - velocity_y (vx2)
 - magnetic_field_x (Bx1)
 - magnetic_field_y (Bx2)

Note: Use the normalisation values returned by the function from part **(a)** to convert from code units to CGS units.

```
In [6]: def io_time_cgs(path, t_0):
    """
    Fuction to get the time simulation in CGS untis from the vtk.out file Input:
        Path: file path (str)
        t_0: time normalization constant (float)
    Output:
        t_cgs: time array with time simualtion (1D array, float)
    Author: Alan Palma
    """
    data = pd.read_csv(path + "/vtk.out", sep = "\s+", header = None)

# Get the second column
    time_code = np.array(data.iloc[:,1], dtype = float)

# Convert this to CGS units
    t_cgs = time_code*t_0

return t_cgs
```

```
In [7]:

def io_vtk_file(path, time_arr):
    """"
    Function to read a vtk file and extract the data of simulation (in CGS untis).
    Input:
        Path: file directory (str)
        time_arr: 1D time array for all simulation data (float)
    Output:
        mesh: pvista object with the vtk file
        rho_cgs_2D: 2D gas density array in CGS units (float)
        vx1_cgs_2D: 2D x velocity array in CGS units (float)
        vx2_cgs_2D: 2D y velocity array in CGS units (float)
        Bx1_cgs_2D: 2D x magnetic field array in CGS units (float)
        Bx2_cgs_2D: 2D y magnetic field array in CGS units (float)
        time: information time correponding to simulation in CGS untis (float)
    Author: Alan Palma
    """
    mesh = pv.read(path)
```

```
#Arrays in code units
rho = pv.get_array(mesh, "rho", preference = 'cell')
vx1 = pv.get_array(mesh, "vx1", preference = 'cell')
vx2 = pv.get_array(mesh, "vx2", preference = 'cell')
Bx1 = pv.get_array(mesh, "Bx1", preference = 'cell')
Bx2 = pv.get_array(mesh, "Bx2", preference = 'cell')
#Arrays in CGS units
rho_cgs = rho*rho_0
vx1\_cgs = vx1*v\_0
vx2\_cgs = vx2*v\_0
Bx1\_cgs = Bx1*B\_0
Bx2\_cgs = Bx2*B\_0
# 2D arrays in CGS units
rho_cgs_2D = rho_cgs.reshape(mesh.dimensions[0] - 1, mesh.dimensions[1] - 1)
vx1\_cgs\_2D = vx1\_cgs.reshape(mesh.dimensions[0] - 1, mesh.dimensions[1] - 1)
vx2\_cgs\_2D = vx2\_cgs.reshape(mesh.dimensions[0] - 1, mesh.dimensions[1] - 1)
Bx1\_cgs\_2D = Bx1\_cgs.reshape(mesh.dimensions[0] - 1, mesh.dimensions[1] - 1)
Bx2\_cgs\_2D = Bx2\_cgs.reshape(mesh.dimensions[0] - 1, mesh.dimensions[1] - 1)
#Get the time information
file name = path[-13:]
indx = int(file_name[6:9]) #Indexing from file_name
time = time_arr[indx]
return mesh, rho_cgs_2D, vx1_cgs_2D, vx2_cgs_2D, Bx1_cgs_2D, Bx2_cgs_2D, time,
```

(c) reads the 2D arrays returned by the function above, interpolates them into a CGS-normalised meshgrid created with the mesh information stored in the VTK files, and exports 3 figures containing maps of:

- density
- velocity vector field, scaled by its magnitude, \$v=\sqrt{v_x^2+v_y^2}\$.
- magnetic vector field, scaled by its magnitude: \$B=\sqrt{B_x^2+B_y^2}\$.

Notes:

- Choose different perceptually-uniform colour schemes for each of the above quantities.
- Since these are high-resolution models, one way to improve the visualisation of 2D vector fields is to rescale them onto a coarser grid.

```
if os.path.isdir("output_data"):
    print("Directory already exists.")
else:
    print("Directory has been created.")
    os.mkdir("output_data")
```

Directory has been created.

```
vx: 2D array x component of the vector field (float)
                 vy: 2D array y component of vector field(float)
                 mesh: pvista object with the vtk file
                 fac_red: reduction factor that determines the new size of the array
                          in base of the original size (float)
             Outputs:
                 vx_new: 2D array x component of the vector field interpolated (float)
                 vy_new: 2D array xycomponent of the vector field interpolated (float)
                 xx_2d: x normalized meshgrid with the new dimentions (float)
                 yy_2d: y normalized meshgrid with the new dimentions (float)
             Author: Alan Palma
             #Set the new dimentions
             x_{dim} = int((mesh.dimensions[1] - 1)//fac_red)
             y_dim = int((mesh.dimensions[0] - 1)//fac_red)
             #Interpolate each component
             vx_new = resize(vx, (x_dim, y_dim))
             vy_new = resize(vy, (x_dim, y_dim))
             # Create normalized coordinate vectors for the interpolated vector field:
             xx = np.linspace(mesh.bounds[0], mesh.bounds[1], x_dim)*L_0
             yy = np.linspace(mesh.bounds[2], mesh.bounds[3], y_dim)*L_0
             # Generate Grid
             xx_2d, yy_2d = np.meshgrid(xx, yy)
             return vx_new, vy_new, xx_2d, yy_2d
In [10]: | def get_figures_from_vtk(mesh_1, rho_cgs_2D_1, vx1_cgs_2D_1, vx2_cgs_2D_1,
                             Bx1_cgs_2D_1, Bx2_cgs_2D_1, time_1, indx_1,
                             mesh_2, rho_cgs_2D_2, vx1_cgs_2D_2, vx2_cgs_2D_2,
                             Bx1_cgs_2D_2, Bx2_cgs_2D_2, time_2, indx_2):
             Function to generate density, velocity vector field and magnetic vector field m
             data (supersonic->1 and subsonic->2).
             Inputs:
                   mesh: pvista object with the vtk file
                   rho_cgs_2D_#: 2D gas density array in CGS units (float)
                   vx1_cgs_2D_#: 2D x velocity array in CGS units (float)
                   vx2_cgs_2D_#: 2D y velocity array in CGS units (float)
                   Bx1_cgs_2D_#: 2D x magnetic field array in CGS units (float)
                   Bx2_cgs_2D_#: 2D y magnetic field array in CGS units (float)
                   time_#: information time correponding to simulation in CGS untis (float)
                   indx_#: index corresponding to the vtk file opened
              Outputs:
                   Two panel figures for (Supersonic -> Right, Subsonic -> Left):
                        Gas density map
                        Velocity vector field
                        Magnetic vector field map
              Author: Alan Palma
             #Create a normalized meshgrid for both simuations (They are the same for both s
             # Create coordinate vectors:
             x = np.linspace(mesh_1.bounds[0], mesh_1.bounds[1], mesh_1.dimensions[1] - 1)*L
             y = np.linspace(mesh_1.bounds[2], mesh_1.bounds[3], mesh_1.dimensions[0] - 1)*L
             # Generate a grid
             x_2d, y_2d = np.meshgrid(x, y)
             #Gas density two-panel
             fig, ax = plt.subplots(nrows = 1, ncols = 2, figsize = (16, 6))
             ax1, ax2 = ax.flatten()
```

```
map1 = ax1.pcolormesh(x 2d, y 2d, np.log10(rho cgs 2D 1), cmap = "viridis", vmi
ax1.set_title(f"Gas Density: VTK File {indx_1} \n Time: {'%.2e' % time_1} s")
ax1.set_xlabel("x [cm]")
ax1.set_ylabel("y [cm]")
ax1.text(0.92*6.17e+18, 0.92*6.17e+18, "Supersonic", ha='right', va='top', font
         bbox=dict(facecolor='white', alpha=0.5))
plt.colorbar(map1, label = "$log_{10}(\\rho)$ [$g/cm^{3}$]")
map2 = ax2.pcolormesh(x_2d, y_2d, np.log10(rho_cgs_2D_2), cmap = "viridis", vmi
ax2.set_title(f"Gas Density: VTK File {indx_2} \n Time: {'%.2e' % time_2} s")
ax2.set_xlabel("x [cm]")
ax2.set_ylabel("y [cm]")
ax2.text(0.92*6.17e+18, 0.92*6.17e+18, "Subsonic", ha='right', va='top', fontsi
                   bbox=dict(facecolor='white', alpha=0.5))
plt.colorbar(map2, label = "$log_{10}(\\rho)$ [$g/cm^{3}$]")
plt.savefig("output_data/gas_density.{:03d}.png".format(indx_1))
plt.close()
#Velocity vector fields
#Interpolate the vector fields
#Supersonic:
fac_red1 = 8. #Determine a factor to reduce the dimetions
vx1_2d_int1, vx2_2d_int1, xx_2d_int1, yy_2d_int1 = interpolate(vx1_cgs_2D_1, v
#Calculate the modulus of the velocity vector field
v_norm_int1 = np.sqrt(vx1_2d_int1**2 + vx2_2d_int1**2) #Get the modulus of the
#Subsonic:
vx1_2d_int2, vx2_2d_int2, xx_2d_int2, yy_2d_int2 = interpolate(vx1_cgs_2D_2, v
#Calculate the modulus of the velocity vector field
v_norm_int2 = np.sqrt(vx1_2d_int2**2 + vx2_2d_int2**2) #Get the modulus of the
#Plotting vector velocity field interpolated
fig, axes = plt.subplots(nrows = 1, ncols = 2, figsize=(16, 6))
ax1, ax2 = axes.flatten()
map1 = ax1.quiver(xx_2d_int1, yy_2d_int1, vx1_2d_int1, vx2_2d_int1, np.log10(vx1_2d_int1) + vx2_2d_int1) + vx2_2d_int1 + vx2_2
ax1.set_title(f"Velocity Vector Field: VTK File {indx_1} \n Time: {'%.2e' % tim
ax1.set_xlabel("x [cm]")
ax1.set_ylabel("y [cm]")
ax1.text(0.95*6.17e+18, 1*6.17e+18, "Supersonic", ha='right', va='top', fontsiz
                   bbox=dict(facecolor='white', alpha=0.8))
plt.colorbar(map1, label = "$log_{10}(v)$ [$cm/s$]")
map2 = ax2.quiver(xx_2d_int2, yy_2d_int2, vx1_2d_int2, vx2_2d_int2 , np.log10(v
ax2.set_xlabel("x [cm]")
ax2.set_ylabel("y [cm]")
```

```
ax2.text(0.95*6.17e+18, 1*6.17e+18, "Subsonic", ha='right', va='top', fontsize=
                      bbox=dict(facecolor='white', alpha=0.8))
plt.colorbar(map2, label = "$log_{10}(v)$ [$cm/s$]")
plt.savefig("output_data/velocity_vector_field.{:03d}.png".format(indx_1))
plt.close()
#Magnetic vectors fields
#Interpolate the vector fields
#Supersonic:
fac_red2 = 9. #Determine a factor to reduce the dimetions
Bx1_2d_int1, Bx2_2d_int1, xx_2d_int1, yy_2d_int1 = interpolate(Bx1_cgs_2D_1, E
#Calculate the modulus of the magnetic vector field
B_norm_int1 = np.sqrt(Bx1_2d_int1**2 + Bx2_2d_int1**2) #Get the modulus of the
#Subsonic:
Bx1_2d_int2, Bx2_2d_int2, xx_2d_int2, yy_2d_int2 = interpolate(Bx1_cgs_2D_2, E
#Calculate the modulus of the magnetic vector field
B_norm_int2 = np.sqrt(Bx1_2d_int2**2 + Bx2_2d_int2**2) #Get the modulus of the
#Plotting magnetic vector field interpolated
fig, axes = plt.subplots(nrows = 1, ncols = 2, figsize=(16, 6))
ax1, ax2 = axes.flatten()
map1 = ax1.quiver(xx_2d_int1, yy_2d_int1, Bx1_2d_int1, Bx2_2d_int1 , np.log10(E
ax1.set title(f"Magnetic Vector Field: VTK File {indx 1} \n Time: {'%.2e' % tim
ax1.set_xlabel("x [cm]")
ax1.set_ylabel("y [cm]")
ax1.text(0.95*6.17e+18, 1*6.17e+18, "Supersonic", ha='right', va='top', fontsiz
                     bbox=dict(facecolor='white', alpha=0.8))
plt.colorbar(map1, label = "$log_{10}(B)$ [$Gauss$]")
map2 = ax2.quiver(xx_2d_int2, yy_2d_int2, Bx1_2d_int2, Bx2_2d_int2 , np.log10(E
ax2.set_title(f"Magnetic Vector Field: VTK File {indx_2} \n Time: {'%.2e' % times times times to a continuous times to a continuous times times times times to a continuous times ti
ax2.set_xlabel("x [cm]")
ax2.set_ylabel("y [cm]")
ax2.text(0.95*6.17e+18, 1*6.17e+18, "Subsonic", ha='right', va='top', fontsize=
                     bbox=dict(facecolor='white', alpha=0.8))
plt.colorbar(map2, label = "$log_{10}(B)$ [$Gauss$]")
plt.savefig("output data/magnetic vector field.{:03d}.png".format(indx 1))
plt.close()
return print("The figures were saved successfully in 'output data'")
```

2. (4 points) Data visualisation and density comparison:

Use python to carry out the following analysis:

(d) Call the above functions for VTK file # 50 of each simulation, and make the following maps using the correct mesh coordinates and dimensions:

- A 2-panel figure showing the supersonic (left) and subsonic (right) density, \$\rho\$.
- A 2-panel figure showing the supersonic (left) and subsonic (right) velocity vector field, \$\vec{v}\$.
- A 2-panel figure showing the supersonic (left) and subsonic (right) magnetic vector field, \$\vec{B}\$\$.

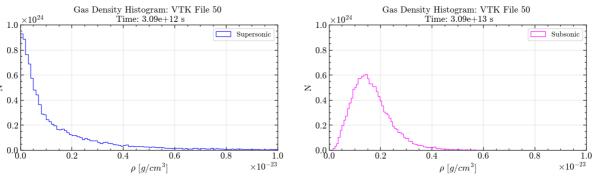
```
In [11]: #Get time arrays of each simulation
         time_arr_SUP = io_time_cgs(path + sup_directory, t_0)
         time_arr_SUB = io_time_cgs(path + sub_directory, t_0)
         #print(time_arr_SUP.shape, time_arr_SUB.shape)
         #print(time arr SUP[0])
In [12]: #Read the vtk files
         file = "/data.0050.vtk"
         mesh_SUP, rho_SUP, vx1_SUP, vx2_SUP, \
             Bx1_SUP, Bx2_SUP, time_SUP, ind_SUP = io_vtk_file(path + sup_directory + file,
         mesh_SUB, rho_SUB, vx1_SUB, vx2_SUB, \
             Bx1_SUB, Bx2_SUB, time_SUB, ind_SUB = io_vtk_file(path + sub_directory + file,
         # print(rho_SUB.shape, vx1_SUP.shape)
In [13]: #Call the function to get the 2-panel figures
         get_figures_from_vtk(mesh_SUP, rho_SUP, vx1_SUP, vx2_SUP, \
             Bx1_SUP, Bx2_SUP, time_SUP, ind_SUP,\
             mesh_SUB, rho_SUB, vx1_SUB, vx2_SUB, \setminus
             Bx1_SUB, Bx2_SUB, time_SUB, ind_SUB)
```

The figures were saved successfully in 'output_data'

- (e) Compute 1D histograms of the density and the velocity magnitude, and make the following figures:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) 1D histograms of the density.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) 1D histograms of the velocity magnitude.

```
#Flatten the velocity magnitude arrays
v_SUP_1D = v_SUP_2d.flatten()
v_SUB_1D = v_SUB_2d.flatten()
# print(v_SUB_1d.shape)
```

```
In [16]: #Plot the 2-Panel for gas density
         plt.style.use(['science', 'notebook', 'no-latex']) #Use a specific style
         fig, axes = plt.subplots(nrows = 1, ncols= 2, figsize = (18,4))
         ax1, ax2 = axes.flatten()
         n_rho_SUP, bins_rho_SUP, _ = ax1.hist(rho_SUP_1D, histtype = "step", bins = "auto",
         ax1.grid(True, alpha = 0.3)
         ax1.set_title(f"Gas Density Histogram: VTK File {ind_SUP} \n Time: {'%.2e' % time_S
         ax1.set_xlabel("$\\rho$ [$g/cm^{3}$]")
         ax1.set_ylabel("N")
         ax1.set_xlim(0.0, 1.0e-23)
         ax1.set_ylim(0., 10.0e23)
         ax1.legend(frameon = True, loc = 1, fontsize = 13)
         n_rho_SUB, bins_rho_SUB, _ = ax2.hist(rho_SUB_1D, histtype = "step", bins = 100, de
         ax2.grid(True, alpha = 0.3)
         ax2.set_title(f"Gas Density Histogram: VTK File {ind_SUB} \n Time: {'%.2e' % time_S
         ax2.set_xlabel("$\\rho$ [$g/cm^{3}$]")
         ax2.set_ylabel("N")
         ax2.set_xlim(0.0, 1.0e-23)
         ax2.set_ylim(0., 10.e23)
         ax2.legend(frameon = True, loc = 1, fontsize = 13)
         plt.show()
```



```
In [17]: #Plot the 2-Panel for velocity magnitude

fig, axes = plt.subplots(nrows = 1, ncols= 2, figsize = (18,4))
ax1, ax2 = axes.flatten()

n_v_SUP, bins_v_SUP, _ = ax1.hist(v_SUP_1D, histtype = "step", bins = 100, density

ax1.grid(True, alpha = 0.3)
ax1.set_title(f"Velocity Magnitude Histogram: VTK File {ind_SUP} \n Time: {'%.2e' % ax1.set_xlabel("$v$ [$cm/s$]")
ax1.set_ylabel("N")

ax1.set_ylabel("N")
```

```
ax1.legend(frameon = True, loc = 1, fontsize = 13)

n_v_SUB, bins_v_SUB, _ = ax2.hist(v_SUB_1D, histtype = "step", bins = 100, density

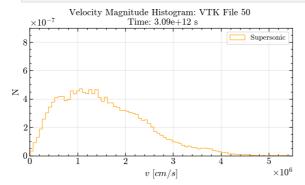
ax2.grid(True, alpha = 0.3)
ax2.set_title(f"Velocity Magnitude Histogram: VTK File {ind_SUP} \n Time: {'%.2e' % ax2.set_xlabel("$v$ [$cm/s$]")
ax2.set_ylabel("N")

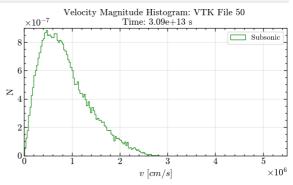
ax2.set_ylabel("N")

ax2.set_ylim(0.0, 5.5e6)
ax2.set_ylim(0.0, 9.0e-7)

ax2.legend(frameon = True, loc = 1, fontsize = 13)

plt.show()
```





To properly analyze the obtained distributions, the data will be fitted to proposed models.

Density distribution analysis

The proposed model is a log-normal distribution

 $\$ \rho_x=\frac 1 {x\sigma\sqrt{2\pi}}\ e^{- \frac{1}{2}(\frac{\ln(x) - \frac{\ln(x) - \frac{1}{2}}, \$\$ where \$\simeq \$ is the standar deviation and \$\mu\$ the mean of the distribution.

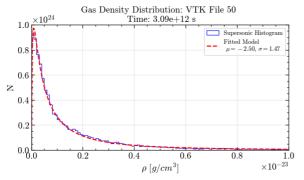
```
n fit norm: 1D array of the normalized counts of the model evaluation (floa
                 coef: 1D array with the free parameters [mu, s] found to fit the data (floa
             Author: Alan Palma
             # Shift array:
             x = 0.5 * (bins[1:] + bins[:-1])
             # Renormalize the axis
             x_0 = x*norm_factor
             n_0 = n/norm_factor
             # Fitting
             coef, cova = opt.curve_fit(log_normal, x_0, n_0)
             # Array to evaluate the model
             x_{arr} = np.linspace(1.e-6, x_0[-1], 1000)
             #Evaluate the model with the fitting parameters for both simualtions
             n_fit = log_normal(x_arr, *coef)
             #Normalize again the data obtained from evaluating the model
             x_fit_norm = x_arr/norm_factor
             n_fit_norm = n_fit*norm_factor
             return x_fit_norm, n_fit_norm, coef
In [20]: #Call the function the fit the data to a log_normal distribution
         rho_SUP_fit, n_rho_SUP_fit, coef_rho_SUP = fit_log_normal(n_rho_SUP, bins_rho_SUP,)
         rho_SUB_fit, n_rho_SUB_fit, coef_rho_SUB = fit_log_normal(n_rho_SUB, bins_rho_SUB,
In [21]: plt.style.use(['science', 'notebook', 'no-latex']) #Use a specific style
         fig, axes = plt.subplots(nrows = 1, ncols= 2, figsize = (18,4))
         ax1, ax2 = axes.flatten()
         n rho SUP, bins rho SUP, = ax1.hist(rho SUP 1D, histtype = "step", bins = "auto",
         ax1.plot(rho_SUP_fit, n_rho_SUP_fit, color = "red", linestyle = "--",
                  label = f"Fitted Model \n $\\mu ={'%.2f' % coef_rho_SUP[0]}$, $\\sigma={'%}
         ax1.grid(True, alpha = 0.3)
         ax1.set_title(f"Gas Density Distribution: VTK File {ind_SUP} \n Time: {'%.2e' % tim
         ax1.set_xlabel("$\\rho$ [$g/cm^{3}$]")
         ax1.set_ylabel("N")
         ax1.set_xlim(0.0, 1.0e-23)
         ax1.set_ylim(0., 10.e23)
         ax1.legend(frameon = True, loc = 1, fontsize = 11)
         n_rho_SUB, bins_rho_SUB, _ = ax2.hist(rho_SUB_1D, histtype = "step", bins = 100, de /
         ax2.plot(rho_SUB_fit, n_rho_SUB_fit, color = "indigo", linestyle = "--",
                  label = f"Fitted Model \n \ \mu ={'%.2f' % coef_rho_SUB[0]}$, \ \sigma={'% coef_rho_SUB[0]}$, $\
         ax2.grid(True, alpha = 0.3)
         ax2.set_title(f"Gas Density Histogram Distribution: VTK File {ind_SUB} \n Time: {'%
         ax2.set_xlabel("$\\rho$ [$g/cm^{3}$]")
         ax2.set_ylabel("N")
```

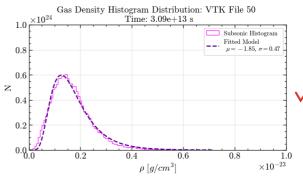
x_fit_norm: 1D array of the normalized x-values of the model evaluation (f

```
ax2.set_xlim(0.0, 1.0e-23)
ax2.set_ylim(0., 10.e23)

ax2.legend(frameon = True, loc = 1, fontsize = 11)

plt.show()
```

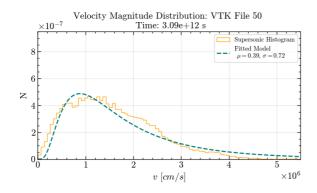


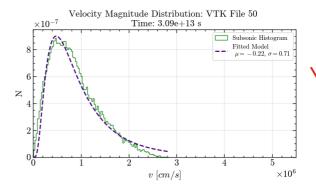


Velocity magnitud distribution analysis

The model used for velocity distribution also will be a log-normal distribution. Therefore the function created for gas density distribution analysis will also be used here.

```
In [22]: #Call the function the fit the data to a log_normal distribution
         v_SUP_fit, n_v_SUP_fit, coef_v_SUP = fit_log_normal(n_v_SUP, bins_v_SUP, 1.e-6)
         v_SUB_fit, n_v_SUB_fit, coef_v_SUB = fit_log_normal(n_v_SUB, bins_v_SUB, 1.e-6)
In [23]: #Plot the 2-Panel for velocity magnitude
         fig, axes = plt.subplots(nrows = 1, ncols= 2, figsize = (18,4))
         ax1, ax2 = axes.flatten()
         n_v_SUP, bins_v_SUP, _ = ax1.hist(v_SUP_1D, histtype = "step", bins = 100, density
         ax1.plot(v_SUP_fit, n_v_SUP_fit, color = "teal", linestyle = "--",
                  label = f"Fitted Model \n $\\mu ={'%.2f' % coef_v_SUP[0]}$, $\\sigma={'%.2
         ax1.grid(True, alpha = 0.3)
         ax1.set_title(f"Velocity Magnitude Distribution: VTK File {ind_SUP} \n Time: {'%.2e
         ax1.set_xlabel("$v$ [$cm/s$]")
         ax1.set_ylabel("N")
         ax1.set_xlim(0.0, 5.5e6)
         ax1.set ylim(0.0, 9.5e-7)
         ax1.legend(frameon = True, loc = 1, fontsize = 11)
         n_v_SUB, bins_v_SUB, _ = ax2.hist(v_SUB_1D, histtype = "step", bins = 100, density
         ax2.plot(v_SUB_fit, n_v_SUB_fit, color = "indigo", linestyle = "--",
                  label = f"Fitted Model \n $\\mu ={'%.2f' % coef_v_SUB[0]}$, $\\sigma={'%.2
         ax2.grid(True, alpha = 0.3)
         ax2.set_title(f"Velocity Magnitude Distribution: VTK File {ind_SUP} \n Time: {'%.2e
         ax2.set_xlabel("$v$ [$cm/s$]")
         ax2.set_ylabel("N")
         ax2.set_xlim(0.0, 5.5e6)
         ax2.set_ylim(0.0, 9.5e-7)
         ax2.legend(frameon = True, loc = 1, fontsize = 11)
         plt.show()
```





(f) What distributions do the density and velocity fields have? Do you see differences between the supersonic and subsonic distributions? Why?

As observed in the fit models, all distributions (density and velocity) follow a log-normal distribution. However, they do not share the same shape.

• In the density distribution:

It is observed that the subsonic regime (\$\sigma = 0.47\$) has a narrower distribution than the supersonic regime (\$\sigma = 1.47\$). This indicates that, in the supersonic case, there are both very low-density regions and very high-density regions, with a log tail extending to higher densities. This agrees with the number of shock candidates, which will be analyzed in the next section.

The mean of the distribution suggests that the supersonic case (\$\mu = -2.5\$) is more centered on smaller density values than the subsonic case (\$\mu = -1.85\$). These results can be explained by the strong compressibility in the supersonic simulation, as large voids are created between shocks, leading to a broader range of densities. In contrast, the subsonic regime exhibits a more uniform density distribution.

In the velocity distribution:

In this case, both distributions are moderately spread, as they have nearly the same standard deviation values. However, outside the model, the velocity magnitudes of the supersonic regime extend to higher velocities compared to the subsonic case.

The mean of the distributions shows that the subsonic simulation ($\mu = -0.22$) is skewed toward lower velocities than the supersonic simulation ($\mu = 0.39$). Therefore, on average, the supersonic simulation reaches higher velocities, which aligns with expectations.

3. (6 points) Numerical differentiation and shock candidates:

Create a set of Python functions that:

- (g) Isolate candidate shocked cells in binary fields, based on the following methods:
- Method 1: Read a 2D velocity vector field. Compute the divergence of the velocity field and isolate the cells where there are convergent flows (i.e. where \$\vec\nabla\cdot \vec v <\alpha\$, where \$\alpha=0\$). Cells with convergent flows are candidate shocked cells. Try with \$\alpha\$ values slightly lower than \$0\$ for better results.
- Method 2: Calculate the 2D pressure field using the equation of state of isothermal gas, i.e.
 \$p=\rho\,c_{\rm iso}^2\$. Compute the gradient of the pressure and isolate the cells with

large pressure gradients (i.e. where $\frac{|| e^{||} P|}{P} \$ \alpha \max{\left(\frac{||} P|){P}\right)}\$). Such cells are candidate shocked cells.

Method 1: Divergence of vector field

```
In [24]: def shock_cell_1(vx, vy, alpha, dx):
             Function to found cells where there are covergent flows using the divergence of
             Cells with convergent flows are candidate shocked cells.
             Inputs:
                 vx1_cgs_2D: 2D x velocity array in CGS units (float)
                 vx2_cgs_2D: 2D y velocity array in CGS units (float)
                 dx: array spacing (float)
                 alpha: threshold value (float)
             Output:
                 divt_clean: 2D array of the selected cells from divergence of the
                             velocity vector field (float)
             Author: Alan Palma
             #Compute the divergence
             div1 = np.gradient(vx, dx, axis = 1) #Derivative in x direction
             div2 = np.gradient(vy, dx, axis = 0) #Derivative in y direction
             divt = div1 + div2 #Total divergence
             #Use a threshold value to isolate the cells desired
             divt_clean = np.where(divt < alpha, divt , 0. )</pre>
             return divt_clean
```

Method 2: Pressure gradient

```
In [25]: def shock_cell_2(rho, c_iso, c, dx):
             Function to found cell where there are covergent flows using gradient of the pr
             Cells with convergent flows are candidate shocked cells.
             Inputs:
                 rho: 2D gas density array in CGS units (float)
                 c_iso: isothermal constant in CGS untis (float)
                 c: threshold value (float)
                 dx: array spacing (float)
                 ratio_clean: 2D array of the selected cells from gradient
                              of the pressure and pressure ratio (float)
             Author: Alan Palma
             #Compute the thermal pressure
             p = rho * c_iso**2
             #Calculate the gradient
             grad_p = np.gradient(p, dx)
             xgrad_p = grad_p[0] # x component of gradient
             ygrad_p = grad_p[1] # y component of gradient
             grad_p_magnitude = np.sqrt(xgrad_p**2 + ygrad_p**2) #Magnitude of gradient pres
```

```
#Relation between pressure and gradient pressure magnitude

ratio = grad_p_magnitude/p

#Select the shock candidates

ratio_clean = np.where(ratio > c*np.max(ratio), ratio , 0. )

return ratio_clean
```

- (h) Call your shock-finding function/s from (g) for VTK file # 50 of each simulation, and make binary maps of the resulting candidate shock cells from both methods and for both models. Show the results as follows:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 1.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 2.

The shock cell candidates with method 1 were isolated with the condition:

```
$$ \vec\nabla\cdot \vec v <\alpha $$
```

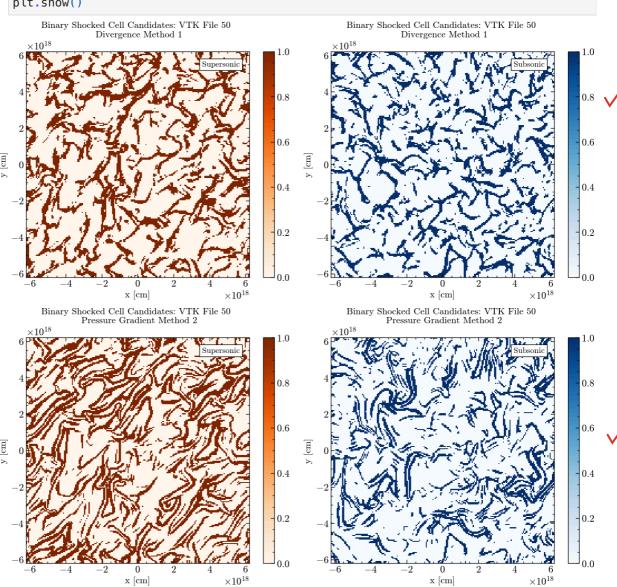
where the value that works better is $\alpha = 10^{-12}$ \$.

print(np.unique(shock_grid1_SUB_bin))

```
In [26]: # Method 1
         #Get the spacing
         x = np.linspace(mesh_SUB.bounds[0], mesh_SUB.bounds[1], mesh_SUB.dimensions[1] - 1)
         y = np.linspace(mesh\_SUB.bounds[2], mesh\_SUB.bounds[3], mesh\_SUB.dimensions[0] - 1)
         # Generate a grid
         x_2d, y_2d = np.meshgrid(x, y)
         dx = x[1] - x[0]
         dy = x[1] - x[0]
         # alpha = 0.0
         alpha = -1.e-12 #Define the threshold (Remarkable difference between order -11 and
         #Call the function
         #Evaluate supersonic simulation
         shock_grid1_SUP = shock_cell_1(vx1_SUP, vx2_SUP, alpha, dx)
         #Evaluate subsonic simulation
         shock_grid1_SUB = shock_cell_1(vx1_SUB, vx2_SUB, alpha, dx)
         # print(shock_grid_1.shape)
In [27]: #Generate binary images
         shock_grid1_SUP_bin = shock_grid1_SUP != 0.
         shock_grid1_SUB_bin = shock_grid1_SUB != 0.
         # print(np.unique(shock_grid1_SUP_bin))
```

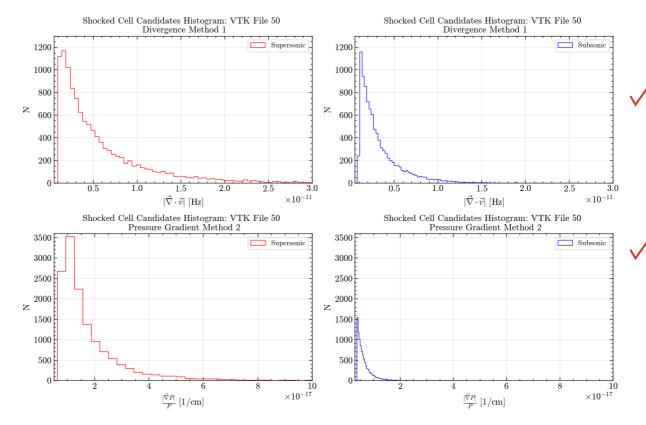
The threshold value used to select the shock candidate cells in Method 2 differs are $c_{SUP} = 0.025$ for supersonic and $c_{SUB} = 0.090$ for subsonic regime.

```
In [28]: #Method 2
         c_SUP = 0.025
         c_SUB = 0.090
         #Call the function
         #Evaluate supersonic simulation
         shock_grid2_SUP = shock_cell_2(rho_SUP, c_iso_0, c_SUP, dx)
         #Evaluate subsonic simulation
         shock_grid2_SUB = shock_cell_2(rho_SUB, c_iso_0, c_SUB, dx)
         # print(shock_grid1_SUB.shape)
In [29]: #Generate binary images
         shock_grid2_SUP_bin = shock_grid2_SUP != 0.
         shock_grid2_SUB_bin = shock_grid2_SUB != 0.
         # print(np.unique(shock_grid2_SUP_bin))
         # print(np.unique(shock_grid2_SUB_bin))
In [30]: # print(np.sum(shock_grid2_SUP_bin))
         # print(np.sum(shock grid2 SUB bin))
In [31]: #Shock cell cadidates for both methods
         fig, ax = plt.subplots(nrows = 2, ncols = 2, figsize = (14, 13))
         ax1, ax2, ax3, ax4 = ax.flatten()
         map1 = ax1.pcolormesh(x_2d, y_2d, shock_grid1_SUP_bin, cmap = "Oranges")
         ax1.set_title(f"Binary Shocked Cell Candidates: VTK File {ind_SUP} \n Divergence Me
         ax1.set_xlabel("x [cm]", fontsize = 15)
         ax1.set_ylabel("y [cm]", fontsize = 15)
         ax1.text(0.92*6.17e+18, 0.92*6.17e+18, "Supersonic", ha='right', va='top', fontsize
                 bbox=dict(facecolor='white', alpha=1.0))
         plt.colorbar(map1)
         map2 = ax2.pcolormesh(x_2d, y_2d, shock_grid1_SUB_bin, cmap = "Blues")
         ax2.set_title(f"Binary Shocked Cell Candidates: VTK File {ind_SUB} \n Divergence Me
         ax2.set_xlabel("x [cm]", fontsize = 15)
         ax2.set_ylabel("y [cm]", fontsize = 15)
         ax2.text(0.92*6.17e+18, 0.92*6.17e+18, "Subsonic", ha='right', va='top', fontsize=1
                     bbox=dict(facecolor='white', alpha=1.0))
         plt.colorbar(map2)
         map3 = ax3.pcolormesh(x_2d, y_2d, shock_grid2_SUP_bin, cmap = "Oranges")
         ax3.set title(f"Binary Shocked Cell Candidates: VTK File {ind SUP} \n Pressure Grad
         ax3.set_xlabel("x [cm]", fontsize = 15)
         ax3.set_ylabel("y [cm]", fontsize = 15)
         ax3.text(0.92*6.17e+18, 0.92*6.17e+18, "Supersonic", ha='right', va='top', fontsize
                 bbox=dict(facecolor='white', alpha=1.0))
         plt.colorbar(map3)
```



- (i) Compute 1D histograms of the shock cell candidates from both methods and for both models, and make the following figures:
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 1.
 - A 2-panel figure showing the supersonic (left) and subsonic (right) shocked cell candidates, computed from method 2.

```
shock_grid1_SUP_1D = shock_grid1_SUP.flatten()
         shock grid1 SUB 1D = shock grid1 SUB.flatten()
         #Method 2
         shock_grid2_SUP_1D = shock_grid2_SUP.flatten()
         shock_grid2_SUB_1D = shock_grid2_SUB.flatten()
         # print(shock_grid2_SUB_1D.shape)
In [33]: #Plot the 2-Panel histogram for pressure gradient method 1
         fig, axes = plt.subplots(nrows = 2, ncols= 2, figsize = (15, 10))
         ax1, ax2, ax3, ax4 = axes.flatten()
         n, bins, _ = ax1.hist(np.abs(shock_grid1_SUP_1D), histtype = "step", bins = 100, cd
         ax1.grid(True, alpha = 0.3)
         ax1.set_title(f"Shocked Cell Candidates Histogram: VTK File {ind_SUP} \n Divergence
         ax1.set_xlabel("$|\\vec \\nabla \\cdot \\vec v|$ [Hz]")
         ax1.set_ylabel("N")
         ax1.set_ylim(0., 1300.)
         ax1.set_xlim(0.5e-12, 3.e-11)
         ax1.legend(frameon = True, loc = 0, fontsize = 13)
         n, bins, _ = ax2.hist(np.abs(shock_grid1_SUB_1D), histtype = "step", bins = 100, cc 🗸
         ax2.grid(True, alpha = 0.3)
         ax2.set_title(f"Shocked Cell Candidates Histogram: VTK File {ind_SUB} \n Divergence
         ax2.set_xlabel("$|\\vec \\nabla \\cdot \\vec v|$ [Hz]")
         ax2.set_ylabel("N")
         ax2.set_ylim(0., 1300.)
         ax2.set_xlim(0.5e-12, 3.e-11)
         ax2.legend(frameon = True, loc = 0, fontsize = 13)
         n, bins, _ = ax3.hist(shock_grid2_SUP_1D, histtype = "step", bins = 100, color = "r
         ax3.grid(True, alpha = 0.3)
         ax3.set_title(f"Shocked Cell Candidates Histogram: VTK File {ind_SUP} \n Pressure @
         ax3.set_xlabel("$\\frac{|\\vec\\nabla P|}{P}$ [1/cm]")
         ax3.set_ylabel("N")
         ax3.set_ylim(0., 3600.)
         ax3.set_xlim(0.5e-17, 1.e-16)
         ax3.legend(frameon = True, loc = 1, fontsize = 13)
         n, bins, _ = ax4.hist(shock_grid2_SUB_1D, histtype = "step", bins = 100, color = "b
         ax4.grid(True, alpha = 0.3)
         ax4.set title(f"Shocked Cell Candidates Histogram: VTK File {ind SUB} \n Pressure @
         ax4.set_xlabel("$\frac{|\vec\nabla P|}{P}$ [1/cm]")
         ax4.set_ylabel("N")
         ax4.set_ylim(0., 3600.)
         ax4.set_xlim(0.3e-17, 1.e-16)
         ax4.legend(frameon = True, loc = 1, fontsize = 13)
         plt.tight_layout()
         plt.show()
```



- (j) Analyse your results, in particular:
 - Do the shock candidate results of methods 1 and 2 agree? Why?
 - Do you find similar distributions of shock candidates in both turbulence models? Which model has more shocks? Why?
- The shape of the distribution appears to be correct since it is similar in both methods.
 However, method 1 seems to be more accurate, as the threshold values around zero for both simulations provide a good basis for comparison. In contrast, method 2 determines the threshold based on a percentage of the maximum value of the ratio of pressure gradient magnitude to pressure, which is not consistently defined and must be adjusted for each regime (supersonic and subsonic).
- The distribution is very similar for both simulations in method 1 and method 2. Mowever, method 2 shows a higher peak in the number of shock candidates compared to method 1. As mentioned earlier, this can be adjusted by modifying the threshold value. Nonetheless, both methods agree that the number of shock candidates is higher in the supersonic simulation than in the subsonic one. This is expected, as the supersonic regime generates more turbulent flows, leading to more shocks.

4. (6 points) Numerical integration and custom outputs:

(k) Create a python function that loops over all VTK files in a simulation, and saves maps (in both PNG and VTK-like format) of the shock cell candidates for all times. Add time stamps in physical units to the maps. Please don't attach any output files to your emails. Your codes should produce them when I run them locally.

```
print("Directory already exists.")
         else:
             print("Directory has been created.")
             os.mkdir("output_data/fig_shock_cell")
         if os.path.isdir("output_data/VTK_shock_cell"):
             print("Directory already exists.")
         else:
             print("Directory has been created.")
             os.mkdir("output_data/VTK_shock_cell")
        Directory has been created.
        Directory has been created.
In [35]: # Create directories for each method
         if os.path.isdir("output_data/fig_shock_cell/method_1") or os.path.isdir("output_da
             print("Directories already exist.")
             print("Directories have been created.")
             os.mkdir("output_data/fig_shock_cell/method_1")
             os.mkdir("output_data/fig_shock_cell/method_2")
         if os.path.isdir("output_data/VTK_shock_cell/method_1") or os.path.isdir("output_da
             print("Directories already exists.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/VTK_shock_cell/method_1")
             os.mkdir("output_data/VTK_shock_cell/method_2")
        Directories have been created.
        Directories have been created.
In [36]: def shock1_cell_vtk(path, time_arr, alpha, simulation):
             Function to loops over all VTK files and saves maps of shock cell candidates for
             This function uses divergece method to get shock cell cantidates.
                 path: directory where VTK files are stored (str)
                 time arr: 1D time array for each simulation data (float)
                 alpha: threshold value to compute shock cell candidates (float)
                 simulation: simulation type specification (str) -> "Supersonic" or "Subsoni
             Output:
                 Shock cell candidates maps for all times
                 Shock cell candidates VTK files for all times
             Author: Alan Palma
             for j in range(0, len(time_arr)):
                 filename = "/data.0{:03d}.vtk".format(j)
                 #Call the function to read the vtk file
                 _, _, vx1_2D, vx2_2D, \
                 _, _, time, ind = io_vtk_file(path + filename, time_arr)
                 #Call the function to get shock cell cantidates
                 shock_grid = shock_cell_1(vx1_2D, vx2_2D, alpha, dx)
                 #Generate and save the map figures
                 fig = plt.figure(figsize = (7, 6))
                 map1 = plt.pcolormesh(x_2d, y_2d, shock_grid, cmap = "Oranges_r", clim = (-
                 plt.title(f"Shocked Cell Candidates: VTK File {ind} \n Divergence Method \n
```

plt.xlabel("x [cm]")

```
plt.ylabel("y [cm]")
                 plt.colorbar(map1, label = "$\\vec \\nabla \\cdot \\vec v$ [Hz]")
                 plt.text(0.92*6.17e+18, 0.92*6.17e+18, simulation, ha='right', va='top', fc
                         bbox=dict(facecolor='white', alpha=0.7))
                 plt.text(1.92e18, 1.18*6.17e+18, f"Time: {'%.2e' % time} s" , ha='right', v
                 plt.savefig(f"output_data/fig_shock_cell/method_1/{simulation}" + "/shock_a
                 plt.close()
                 #Reshape the x and y axis to get len(x) + 1 dimentions
                 x_{to} = resize(x, (len(x)+1,))
                 y_{to} = resize(y, (len(y)+1,))
                 #Artifitial third axis
                 z to vtk = np.zeros(1)
                 #Save a VTK-like format file
                 gridToVTK(f"output_data/VTK_shock_cell/method_1/{simulation}" + "/shock_cel
                           cellData = {"shock_grid" : shock_grid.flatten()})
             return print("The map figures and VTK files were correctly generated")
In [37]: def shock2_cell_vtk(path, time_arr, c, simulation):
             Function to loops over all VTK files and saves maps of shock cell candidates for
             This function uses gradient pressure method to get shock cell cantidates.
                 path: directory where VTK files are stored (str)
                 time_arr: 1D time array for each simulation data (float)
                 alpha: threshold value to compute shock cell candidates (float)
                 simulation: simulation type specification (str) -> "Supersonic" or "Subsoni
             Output:
                 Shock cell candidates maps for all times
                 Shock cell candidates VTK files for all times
             Author: Alan Palma
             for j in range(0, len(time_arr)):
                 filename = "/data.0{:03d}.vtk".format(j)
                 #Call the function to read the vtk file
                 _, rho_2D, _, _, \
                 _, _, time, ind = io_vtk_file(path + filename, time_arr)
                 #Call the function to get shock cell cantidates
                 shock_grid = shock_cell_2(rho_2D, c_iso_0, c, dx)
                 #Generate and save the map figures
                 fig = plt.figure(figsize = (7, 6))
                 map1 = plt.pcolormesh(x_2d, y_2d, shock_grid, cmap = "Blues", clim = (0.0,
                 plt.title(f"Shocked Cell Candidates: VTK File {ind} \n Pressure Gradient Me
                 plt.xlabel("x [cm]")
                 plt.ylabel("y [cm]")
                 plt.colorbar(map1, label = "$\\frac{|\\vec\\nabla P|}{P}$ [1/cm]")
```

```
bbox=dict(facecolor='white', alpha=0.7))
                 plt.text(1.92e18, 1.18*6.17e+18, f"Time: {'%.2e' % time} s" , ha='right', v
                 plt.savefig(f"output_data/fig_shock_cell/method_2/{simulation}" + "/shock_c
                 plt.close()
                 #Reshape the x and y axis to get len(x) + 1 dimentions
                 x_{to} = resize(x, (len(x)+1,))
                 y_{to} = resize(y, (len(y)+1,))
                 #Artifitial third axis
                 z to vtk = np.zeros(1)
                 #Save a VTK-like format file
                 gridToVTK(f"output_data/VTK_shock_cell/method_2/{simulation}" + "/shock_cel
                           cellData = {"shock_grid" : shock_grid.flatten()})
             return print("The map figures and VTK files were correctly generated")
In [38]: #Create a directories for supersonic and subsonic simulations
         # Method 1:
         if os.path.isdir("output_data/fig_shock_cell/method_1/Supersonic") or os.path.isdir
             print("Directory already exists.")
             print("Directories have been created.")
             os.mkdir("output_data/fig_shock_cell/method_1/Supersonic")
             os.mkdir("output data/fig shock cell/method 1/Subsonic")
         if os.path.isdir("output_data/VTK_shock_cell/method_1/Supersonic") or os.path.isdir
             print("Directories already exist.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/VTK_shock_cell/method_1/Supersonic")
             os.mkdir("output_data/VTK_shock_cell/method_1/Subsonic")
         # Method 2:
         if os.path.isdir("output_data/fig_shock_cell/method_2/Supersonic") or os.path.isdir
             print("Directories already exist.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/fig_shock_cell/method_2/Supersonic")
             os.mkdir("output_data/fig_shock_cell/method_2/Subsonic")
         if os.path.isdir("output_data/VTK_shock_cell/method_2/Supersonic") or os.path.isdir
             print("Directories already exist.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/VTK_shock_cell/method_2/Supersonic")
             os.mkdir("output data/VTK shock cell/method 2/Subsonic")
        Directories have been created.
        Directories have been created.
        Directories have been created.
        Directories have been created.
In [39]: # Method 1:
         #Call the function for supersonic simulation
```

plt.text(0.92*6.17e+18, 0.92*6.17e+18, simulation, ha='right', va='top', fc

```
shock1_cell_vtk(path + sup_directory, time_arr_SUP, alpha, "Supersonic")

The map figures and VTK files were correctly generated

In [40]: #Call the function for subsonic simulation

shock1_cell_vtk(path + sub_directory, time_arr_SUB, alpha, "Subsonic")

The map figures and VTK files were correctly generated

In [41]: # Method 2:

#Call the function for supersonic simulation

shock2_cell_vtk(path + sub_directory, time_arr_SUP, c_SUP, "Supersonic")

The map figures and VTK files were correctly generated

In [42]: #Call the function for subsonic simulation

shock2_cell_vtk(path + sub_directory, time_arr_SUB, c_SUB, "Subsonic")

The map figures and VTK files were correctly generated

(I) Create a python function that loops over all VTK files, and computes the following integrated guaratities for each time:
```

- quantities for each time:
 - the total number of shock candidates, \$N_{\rm shocks}\$ computed from each method 1 and 2,
 - the (volume-weighted) average velocity dispersion, $\sigma_v = \sqrt{v^2} [v]^2$,
 - the rms Mach number, \${\cal M}_{\rm rms}\$, for which you need \$c_{\rm iso}\$ given in units.out,

and returns:

• a CSV file with 5 columns, time on the first column, and the above quantities in the next ones. There is no need to attach the CSV file to your emails, your code should produce the file locally when I run it.

The rms Mach number is defined as follow:

```
\ {\cal M}_{\rm rms} = \frac{v_{mrs}}{c_{iso}}, $$ where $v_{rms} = \sqrt{v^2}$.
```

```
In [43]: def io data quatities(path, time arr, c, simulation):
             0.00
             Function that computes the total number of shock candidates, the volume weighte
             the rms Mach number form a set of VTK files and returns a CVS file containing t
             Inputs:
                 path: directory where VTK files are stored (str)
                 time_arr: 1D array storing time of each simulation data (float)
                 c: threshold value to compute shock cell candidates with method 2 (float)
                 simulation: simulation type specification (str) -> "Supersonic" or "Subsoni
             Outputs:
                 CVS file storing time array, and the following computed quantities:
                     - 1D array with the total number of shock candidates computed with meth
                     - 1D array with the total number of shock candidates computed with meth
                     - 1D array with average velocity dispersion
                     - 1D array with the rms Mach number
             Author: Alan Palma
```

```
#Empty lists to store computed quantities
N_{shocks_1_list} = []
N_{shocks_2_list} = []
std_v_list = []
rms_M_list = []
for j in range(0, len(time_arr)):
    filename = "/data.0{:03d}.vtk".format(j)
    #Call the function to read the vtk file
    _, rho_2D, vx1_2D, vx2_2D, \
   _, _, time, ind = io_vtk_file(path + filename, time_arr)
    #1. Calculate the shock candidates
    #Method1
    shock_grid1 = shock_cell_1(vx1_2D, vx2_2D, alpha, dx)
    shock_grid1_bin = shock_grid1 != 0. #Binary array
   N_shocks_1 = np.sum(shock_grid1_bin) #Count all shock candidates
   N_shocks_1_list.append(N_shocks_1)
    #Method 2
    shock_grid2 = shock_cell_2(rho_2D, c_iso_0, c, dx)
    shock_grid2_bin = shock_grid2 != 0. # Binary array
   N_shocks_2 = np.sum(shock_grid2_bin) #Count all shock candidates
   N_shocks_2_list.append(N_shocks_2) #Append to the list
   #2. Calculate the average velocity dispersion
    # Calculate the magnitude
    vx_2D = np.sqrt(vx1_2D**2 + vx2_2D**2)
    sgm_v = np.std(vx_2D)
    std_v_list.append(sgm_v)
    #3. Calculate the rms Mach number
    rms_M = np.sqrt(np.mean(vx_2D**2))/c_iso_0
    rms_M_list.append(rms_M[0])
#Create a pandas data frame
data frame = pd.DataFrame({ "Time" : time arr,
                            "N shocks (Method 1)" : np.array(N_shocks_1_list),
                           "N shocks (Method 2)" : np.array(N_shocks_2_list),
                           "Std. velocity" : np.array(std_v_list),
                           "rms Mach" : np.array(rms_M_list) })
data_frame.to_csv(f"output_data/final_quantities_{simulation}.csv", sep=',', fl
return print("The CSV file was saved correctly")
```

(m) Call your function from (l) for each simulation set (supersonic and subsonic) and use the CSV files to make the following plots:

- A 2-panel figure showing the supersonic (left) and subsonic (right) \$N_{\rm shocks}\$ time.
- A 2-panel figure showing the supersonic (left) and subsonic (right) \${\cal M}_{\rm rms}\$ versus time.

```
In [44]: #Call the function to get generate the CSV file for Supersonic simulation
         io data quatities(path + sup directory, time arr SUP, c SUP, "Supersonic")
        The CSV file was saved correctly
In [45]: #Call the function to get generate the CSV file for Subsonic simulation
         io_data_quatities(path + sub_directory, time_arr_SUB, c_SUB, "Subsonic")
        The CSV file was saved correctly
In [46]: #Read data from CSV files
         data_SUP = pd.read_csv("output_data/final_quantities_Supersonic.csv", sep = ",")
         data_SUB = pd.read_csv("output_data/final_quantities_Subsonic.csv", sep = ",")
         # print(data SUB)
In [47]: #Extract colums needed
         #Supersonic simulation
         N_shocks1_SUP = np.array(data_SUP["N shocks (Method 1)"])
         N_shocks2_SUP = np.array(data_SUP["N shocks (Method 2)"])
         rms_M_SUP = np.array(data_SUP["rms Mach"])
         #Subsonic Simulation
         N_shocks1_SUB = np.array(data_SUB["N shocks (Method 1)"])
         N_shocks2_SUB = np.array(data_SUB["N shocks (Method 2)"])
         rms_M_SUB = np.array(data_SUB["rms Mach"])
In [48]: #Plot the number of shock candidates for Method 1
         fig, ax = plt.subplots(nrows = 2, ncols = 2, figsize = (18,10))
         ax1, ax2, ax3, ax4 = ax.flatten()
         ax1.plot(time_arr_SUP, N_shocks1_SUP, color = "crimson", label = "Supersonic")
         ax1.set_xlabel("Time [s]")
         ax1.set_ylabel("N")
         ax1.set_title("Number of Shock Candidates \n Divergece Method 1")
         ax1.set_ylim(0, 13.e3)
         ax1.grid(True, alpha = 0.3)
         ax1.legend(frameon = True, fontsize = 12, loc = 4)
         ax2.plot(time_arr_SUB, N_shocks1_SUB, color = "cadetblue", label = "Subsonic")
         ax2.set_xlabel("Time [s]")
         ax2.set_ylabel("N")
         ax2.set_title("Number of Shock Candidates \n Divergece Method 1")
         ax2.set_ylim(0, 13.e3)
         ax2.grid(True, alpha = 0.3)
         ax2.legend(frameon = True, fontsize = 12, loc = 4)
```

```
ax3.plot(time arr SUP, N shocks2 SUP, color = "salmon", label = "Supersonic")
           ax3.set xlabel("Time [s]")
           ax3.set_ylabel("N")
           ax3.set_title("Number of Shock Candidates \n Pressure Gradient Method 2")
           ax3.set_ylim(0. , np.max( [np.max(N_shocks2_SUP), np.max(N_shocks2_SUB)] ) + 1000)
           ax3.grid(True, alpha = 0.3)
           ax3.legend(frameon = True, fontsize = 12)
           ax4.plot(time_arr_SUB, N_shocks2_SUB, color = "khaki", label = "Subsonic")
           ax4.set_xlabel("Time [s]")
           ax4.set_ylabel("N")
           ax4.set title("Number of Shock Candidates \n Pressure Gradient Method 2")
           ax4.set_ylim(0. , np.max( [np.max(N_shocks2_SUP), np.max(N_shocks2_SUB)] ) + 1000)
           ax4.grid(True, alpha = 0.3)
           ax4.legend(frameon = True, fontsize = 12)
           plt.tight_layout()
           plt.show()
                            Number of Shock Candidates
Divergece Method 1
                                                                              Number of Shock Candidates
Divergece Method 1
          12000
                                                             12000
          8000
                                                              8000
          6000
                                                             6000
           4000
                                                              4000
           2000
                                                              2000
                                                        \times 10^{12}
                                                                                                          \times 10^{13}
                                  Time [s]
                                                                                     Time [s]
                            Number of Shock Candidates
Pressure Gradient Method 2
                                                                               Number of Shock Candidates
Pressure Gradient Method 2
                                                                                                         Subsonic
          25000
                                                             20000
          20000
        z 15000
                                                             15000
          10000
                                                             10000
          5000
                                                              5000
                                  Time [s]
                                                       \times 10^{12}
                                                                                     Time [s]
In [49]: #Plot the rms Mach number
           fig, ax = plt.subplots(nrows = 1, ncols = 2, figsize = (18,5))
           ax1, ax2 = ax.flatten()
           ax1.plot(time_arr_SUP, rms_M_SUP, color = "deeppink", label = "Supersonic")
           ax1.set_xlabel("Time [s]")
           ax1.set_ylabel("${\cal M}_{rms}$")
           ax1.set title("RMS Mach Number in Function of Time")
           ax1.set_ylim(0.0, 1.9)
           ax1.grid(True, alpha = 0.3)
           ax1.legend(frameon = True, fontsize = 12, loc = 4)
           ax2.plot(time_arr_SUB, rms_M_SUB, color = "limegreen", label = "Subsonic")
           ax2.set_xlabel("Time [s]")
           ax2.set_ylabel("${\cal M}_{rms}$")
           ax2.set_title("RMS Mach Number in Function of Time")
```

```
ax2.set_ylim(0.0, 1.9)
            ax2.grid(True, alpha = 0.3)
            ax2.legend(frameon = True, fontsize = 12)
            plt.show()
                         RMS Mach Number in Function of Time
                                                                                    RMS Mach Number in Function of Time
                                                                                                                      Subsonio
            1.75
                                                                       1.75
            1.50
            1.25
                                                                       1.25
            1.00
            0.75
                                                                       0.75
            0.50
                                                                       0.50
            0.25
                                                                       0.25
                                                                       0.00
                                                             \times 10^{12}
                                                                                                                        \times 10^{13}
                                     Time [s]
                                                                                                Time [s]
In [50]: # print(np.mean(N_shocks2_SUP))
            # print(np.mean(N_shocks2_SUB))
```

- (n) Analyse your results, in particular:
 - Does the flow reach steady state in both models? At what times?
 - Is there a relation between \$N_{\rm shocks}\$ and \${\cal M}_{\rm rms}\$?
 - Yes, a steady state is reached in both simulations, as observed in the RMS Mach number figure. For the supersonic simulation, this occurs at approximately \$ 1.8 \times 10^{12}\$ s, and for the subsonic simulation, it appears to be at \$0.15 \times 10^{13} s\$. These results are actually very similar.
 - It is also evident that the RMS Mach number agrees with the expected results. The RMS
 Mach number is below 1 in the subsonic simulation and higher than 1 in the supersonic
 regime.
 - Additionally, the shapes of both figures are very similar, and it is observed that the number of shock candidates (in method 1) reaches a state where the number of shocks remains constant over time for both simulations. The time at which this steady state is reached aligns with the RMS Mach number results.

5. (1 point) Shock animation:

- (o) Create a python function that returns movies for each simulation (supersonic and subsonic) showing the time evolution of:
 - maps of the shocks (printed in k), jointly with
 - the total number of shock candidates, \$N_{\rm shocks}\$, computed in (I)

```
In [51]: #Create a directory to store number of shock candidates in function of time

if os.path.isdir("output_data/N_shock_time"):
    print("Directory already exists.")
else:
```

```
os.mkdir("output data/N shock time")
        Directory has been created.
In [52]: # Directories to separate Method 1 and Method 2
         if os.path.isdir("output_data/N_shock_time/method_1") or os.path.isdir("output_data
             print("Directories already exist.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/N_shock_time/method_1")
             os.mkdir("output_data/N_shock_time/method_2")
        Directories have been created.
In [53]: # Create a directories for supersonic and subsonic simulations
         # Method 1
         if os.path.isdir("output_data/N_shock_time/method_1/Supersonic") or os.path.isdir("
             print("Directories already exist.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/N_shock_time/method_1/Supersonic")
             os.mkdir("output_data/N_shock_time/method_1/Subsonic")
         # Method 2
         if os.path.isdir("output_data/N_shock_time/method_2/Supersonic") or os.path.isdir("
             print("Directories already exist.")
         else:
             print("Directories have been created.")
             os.mkdir("output_data/N_shock_time/method_2/Supersonic")
             os.mkdir("output_data/N_shock_time/method_2/Subsonic")
        Directories have been created.
        Directories have been created.
In [54]: def animate_shock_cadidates(simulation, method, N_shocks, time_arr):
             Function to create a movie of computed shock candidates with divergence method
             jointly with the number of number of shock candidates in each time.
             Inputs:
                 simulation: simulation type specification (str) -> "Supersonic" or "Subsoni
                 method: method specification (str) -> "method_1" or "method_2"
                 N shocks: 1D array storing the number of shock candidates for each time
                 time arr: 1D array storing time of each simulation data (float)
                 Movie of computed shock candidates maps with the number of shock candidates
             Author: Alan Palma
             1111111
             # For loop to generate all figures for the movie
             for j in range(0, len(time_arr)):
                 #Read the images generated before
                 img_path = f"output_data/fig_shock_cell/{method}/{simulation}" + "/shock_ce
                 img = plt.imread(img_path)
                 #Conditional to adjust the figure
                 if method == "method_1":
                     title = "Divergence Method 1"
                 else:
```

title = "Pressure Gradient Method 2"

print("Directory has been created.")

```
#Plotting maps and number of shock candidates showing the time evolution
                 fig, ax = plt.subplots(nrows = 1, ncols = 2, figsize = (20,5))
                 ax1, ax2 = ax.flatten()
                 ax1.imshow(img)
                 ax1.axis("off")
                 ax1.set_position([0.0, 0.21, 1., 1.])
                 ax2.plot(time_arr[:j+1], N_shocks[:j+1], color = "cadetblue", marker = ".",
                 ax2.set_xlabel("Time [s]", fontsize = 13.)
                 ax2.set_ylabel("N", fontsize = 13.)
                 ax2.set_title(f"Number of Shock Candidates: VTK File {j} \n {title}", fonts
                 ax2.grid(True, alpha = 0.3)
                 ax2.legend(frameon = True, fontsize = 12, loc = 4)
                 ax2.set_xlim(0., time_arr[-1] + 0.5e12)
                 ax2.set_ylim(0., np.max(N_shocks) + 3000)
                 #Modify the font size of tick labels
                 ax2.tick_params(axis='both', which='major', labelsize=13)
                 ax2.tick_params(axis='both', which='minor', labelsize=13)
                 ax2.set_position([0.67, 0.31, 0.4, 0.72])
                 plt.savefig(f"output_data/N_shock_time/{method}/{simulation}" + "/N_shock_i
                 plt.close()
             #Read all the generated figures to create the movie
             #Define the input directory
             images_input = f"output_data/N_shock_time/{method}/{simulation}/N_shock_in_time
             # Collect the images
             imgs = (Image.open(f) for f in sorted(glob.glob(images_input)))
             img = next(imgs)
             #Define the output directory
             imgif_output = f"output_data/N_shock_time_{simulation}_{method}.gif"
             img.save(fp = imgif_output, format="GIF", append_images=imgs,\
                     save_all=True, duration = 100, loop = 0)
             return print("The movie was generated correctly")
In [55]: # Method 1
         #Call the function for supersonic simulation
         animate_shock_cadidates("Supersonic", "method_1", N_shocks1_SUP, time_arr_SUP)
        The movie was generated correctly
In [56]: #Call the function for subsonic simulation
         animate_shock_cadidates("Subsonic", "method_1", N_shocks1_SUB, time_arr_SUB)
        The movie was generated correctly
In [57]: # Method 2
         #Call the function for supersonic simulation
```

