

Homework 2 - Aircraft Flight Dynamics

1. Model the aircraft lateral and longitudinal dynamics using the state-space equations and stability derivatives provided in the reference section
2. Write functions that can be used to recall a specific set of dynamics, including:
 1. First-order roll dynamics (p)
 2. Second-order short period mode dynamics (q , α)
 3. Third-order Dutch roll mode dynamics (β , p , r)
 4. Fourth-order longitudinal dynamics (V , α , q , θ)
 5. Fourth-order lateral dynamics (β , p , r , ϕ)
3. Choose an aircraft and flight condition then show the dynamic modes for each of the five sets of dynamics

NOTE: There are many ways to structure your files and functions. One way of doing this is to combine the function arguments into a structure. For instance:

Define the aircraft stability derivatives as fields of the structure, such as:

```
air_data.C_m_q = C_m_q
```

Define the aircraft geometric parameters also as fields of that structure, such as:

```
air_data.b = b
```

You can put a practically unlimited number of fields into a structure, and you can also logically organize them into sub-fields, such as:

```
air_data.stability_derivatives.C_m_q = C_m_q
```

or

```
air_data.aircraft_geometry.b = b
```

If you setup your function like this:

```
output = compute_dynamics( air_data )
```

Then you can put whatever you want into `air_data` structure, and assign whatever you want into the output structure, such as:

```
output.sys_roll_mode or
```

```
output.roll_mode.A_matrix = [A]
```

In the next homework, we will look at using these models for digital control.

HW2 script for Navion:

```
v = 84; % (ft/s^2)

[sys_f,sys_n] = ME_401_0007_roll_dynamics(v);
[sys_f_sp,sys_n_sp] = ME_401_0007_second_period(v);
[sys_f_dr,sys_n_dr] = ME_401_0007_dutch_roll(v);
[sys_f_long,sys_n_long] = ME_401_0007_longitudinal(v);
[sys_f_lat,sys_n_lat] = ME_401_0007_lateral(v);
roll_dynamics_pole = pole(sys_n)
sys_n
second_period_poles = pole(sys_n_sp)
sys_n_sp
dutch_roll_poles = pole(sys_n_dr)
sys_n_dr
longitudinal_poles = pole(sys_n_long)
sys_n_long
lateral_poles = pole(sys_n_lat)
sys_n_lat
```

HW2 output:

ME_401_0007_HW2_final

roll_dynamics_pole =

-4.0083

sys_n =

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A =

x1

x1 -4.008

B =

u1

x1 -6.589

C =

x1

y1 1

D =

u1

y1 0

Continuous-time state-space model.

second_period_poles =

-0.9724 + 1.4149i

-0.9724 - 1.4149i

sys_n_sp =

A =

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	x1	x2
x1	-0.9542	1
x2	-2.002	-0.9907

B =

	u1
x1	-6.409
x2	-7.746

C =

	x1	x2
y1	1	1
y2	1	1

D =

	u1
y1	0
y2	0

Continuous-time state-space model.

dutch_roll_poles =

-3.9848
0.2274
-1.5941

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sys_n_dr =

A =

	x1	x2	x3
x1	-0.1212	0	-1
x2	-3.639	-4.008	1.046
x3	-1.037	-0.5621	-1.222

B =

	u1	u2
x1	0	0
x2	-6.589	5.262
x3	-0.0511	-1.051

C =

	x1	x2	x3
y1	1	1	1
y2	1	1	1
y3	1	1	1

D =

	u1	u2
y1	0	0
y2	0	0
y3	0	0

Continuous-time state-space model.

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longitudinal_poles =

0.0000 + 0.0000i

0.0000 + 0.0000i

-0.9724 + 1.4149i

-0.9724 - 1.4149i

sys_n_long =

A =

	x1	x2	x3	x4
x1	-0.9542	1	0	0
x2	-2.002	-0.9907	0	0
x3	-5.957	0	0	-32.17
x4	0	1	0	0

B =

	u1
x1	-6.409
x2	-7.746
x3	0
x4	0

C =

	x1	x2	x3	x4
y1	1	1	1	1
y2	1	1	1	1
y3	1	1	1	1

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y4 1 1 1 1

D =

u1

y1 0

y2 0

y3 0

y4 0

Continuous-time state-space model.

lateral_poles =

-4.0670 + 0.0000i

0.1485 + 0.5543i

0.1485 - 0.5543i

-1.5816 + 0.0000i

sys_n_lat =

A =

	x1	x2	x3	x4
x1	-0.1212	0	-1	0.383
x2	-3.639	-4.008	1.046	0
x3	-1.037	-0.5621	-1.222	0
x4	0	1	0	0

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B =

	u1	u2
x1	0	0.03374
x2	-6.589	5.262
x3	-0.0511	-1.051
x4	0	0

C =

	x1	x2	x3	x4
y1	1	1	1	1
y2	1	1	1	1
y3	1	1	1	1
y4	1	1	1	1

D =

	u1	u2
y1	0	0
y2	0	0
y3	0	0
y4	0	0

Continuous-time state-space model.