# Project, structural optimization Structural optimization, FHLN01

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## Contents

1	Introduction	1			
	1.1 Problem formulation	1			
2	Optimization problem	1			
3	CONLIN and MMA				
4	Solid Isotropic Material Penalization, (SIMP)				
5	Filters				
	5.1 Density filter	5			
	5.2 Helmholtz' PDE based filter	6			
6	Result and discussion	7			
	6.1 CONLIN and MMA	7			
	6.2 SIMP	8			
	6.3 Density filter				
	6.4 PDE filter				
7	Appendix	12			
	7.1 Pseudo code	12			
	7.2 Inspire	13			
	7.3 Software	14			

## Nomenclature

C	Compliance	$\boldsymbol{K}$	Stiffness matrix
D	Tangent stiffness tensor	$oldsymbol{u}$	Nodal displacements
$m{k}^0$	Normalized stiffness matrix	E	Young's modulus
R	Radius	r	Length scale parameter
$\lambda$	Lagrangian multiplier	V	Volume
l	Length of a bar	$\rho$	Density
p	Penalization factor	$\nu$	Poisson's ratio
M	Mass matrix	$\boldsymbol{a}$	Area vector
$\widetilde{\nabla}$	Linear elasticity operator	$\widetilde{ ho}$	Filtered density vector
$\boldsymbol{F}$	External force vector	$\boldsymbol{y}$	Intervening variable
$oldsymbol{N}_e$	Neighborhood vector	$\boldsymbol{w}$	Weight function vector
$oldsymbol{v}$	Volume vector	$\boldsymbol{x}$	Design variable
$L_i$	Moving asymptote		-

#### 1 Introduction

#### 1.1 Problem formulation

The objective consists of analyzing different optimization methods of the structure depicted in figure 1. The optimization methods include CONLIN, MMA and SIMP. Two different filters are implemented and discussed regarding the SIMP method. The equipment provided for analyzing the assignment is MATLAB software and CALFEM toolbox. The compliance of the beam illustrated in figure 1 should be minimized where the material in the beam is linear elastic, homogeneous and isotropic.

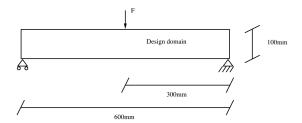


Figure 1: Represented is a illustration of the geometry and boundary condition, i.e. the supports in the corners of the beam used in the analyze.

The first task is to perform a size optimization where the beam is built up with a truss structure of circular bars. The maximum cross section diameter is  $d_{max} = 20mm$  and the maximum volume of the structure is  $V_{max} = 2000mm^3$ . The methods used to optimize the cross sectional areas of the bars in the provided geometry are the two similar algorithms CONLIN and MMA for a range of  $d_{min}$  and initial values. The second task is a topology optimization that should be solved for two different finite element discretization. Both are built up with isoparametric four node elements, one coarser and one finer. The maximum volume of the structure is  $V_{max} = 0.4V_{box}$  where  $V_{box}$  is the volume defined by figure 1, with a thickness of  $20 * 10^{-3}m$ . Poisson's ratio  $\nu$  is set to 0.3 and plane stress is used. To solve this task the SIMP algorithm will be used with and without filters. The filters that will be used are a density filter and the Helmholtz' PDE based filter. For simplicity the tasks uses a symmetry plane at x = 300mm. This means that only one half of the geometry will be analyzed using boundary conditions connecting to the second part, where the second part consist of the rolling support.

## 2 Optimization problem

The compliance which is a good measurement of the stiffness in the structure should be minimized. The simultaneous optimization problem for task one, where a truss structure is to be analyzed, becomes

$$(\mathbf{P})_{sf} = \begin{cases} \min_{\mathbf{x} \in \chi} & C = \mathbf{F}^T \mathbf{u} \\ s.t. & \begin{cases} \mathbf{K}(\mathbf{x})\mathbf{u} = \mathbf{F} \\ \sum l_j x_j - V_{max} \le 0 \\ \mathbf{x} \in \chi = x_j^{min} \le x_j \le x_j^{max}, j = 1, ..., n \end{cases}$$
(1)

Where  $C, \mathbf{F}, \mathbf{u}, \mathbf{K}$ , is the compliance, the external force, the nodal displacement and the stiffness matrix respectively.  $x_j$  is the design variable as a function of the cross sectional area for the trusses in the structure and  $l_j$  is the length of the bar. The simultaneous formulation is a disadvantage for large scale problems due to the number of constraints from the equilibrium equations. In the case where the stiffness matrix is nonsingular, i.e. there exist an inverse to the matrix, the displacement can be rewritten as a function of the design variables. This provides the nested version of the problem and becomes

$$(\mathbf{P})_{nf} = \begin{cases} \min_{\mathbf{x} \in \chi} & C = \mathbf{F}^T \mathbf{u}(\mathbf{x}) \\ s.t. & \begin{cases} \sum l_j x_j - V_{max} \le 0 \\ \mathbf{x} \in \chi = x_j^{min} \le x_j \le x_j^{max}, j = 1, ..., n \end{cases}$$
 (2)

The simultaneous formulation is not convex however the nested formulation is. In task two a topology optimization is to be performed. The design variable when solving task two where a

topology optimization is to be performed will be a density-type parameter  $\rho$ , and can be related to the thickness. As well as for the truss structure there will be an upper bound  $\rho_{max}$  and a lower bound  $\rho_{min}$ , where the lower bound will be close to zero. When identifying global minimums of optimization problems the Lagrangian function is used and is defined as

$$\mathcal{L}(\boldsymbol{x}, \boldsymbol{\lambda}) = g_0(\boldsymbol{x}) + \sum_{i=1}^{l} \lambda_i g_i(\boldsymbol{x})$$
(3)

 $\lambda_i$  are called Lagrange multipliers and the KKT, Karush-Kuhn-Tucker, conditions are defined as

$$\frac{\partial \mathcal{L}(\boldsymbol{x}, \boldsymbol{\lambda})}{\partial x_{j}} \leq 0 \qquad \text{if} \qquad x_{j} = x_{j}^{max}$$

$$\frac{\partial \mathcal{L}(\boldsymbol{x}, \boldsymbol{\lambda})}{\partial x_{j}} = 0 \qquad \text{if} \qquad x_{j}^{min} \leq x_{j} \leq x_{j}^{max}$$

$$\frac{\partial \mathcal{L}(\boldsymbol{x}, \boldsymbol{\lambda})}{\partial x_{j}} \geq 0 \qquad \text{if} \qquad x_{j} = x_{j}^{min}$$

$$\lambda_{i} g_{i}(\boldsymbol{x}) = 0 \qquad g_{i}(\boldsymbol{x}) \leq 0 \qquad \lambda_{i} \geq 0 \qquad \boldsymbol{x} \in \chi$$
(4)

For convex problems the KKT points provide global optima, however for nonconvex problems these may only be local optima. For large scale structures it might be time consuming to solve the KKT conditions defined in equation (4) instead a method called Lagrangian Duality is used. The optimization problem can solved with

$$\min_{\boldsymbol{x} \in \chi} \max_{\boldsymbol{\lambda} \ge 0} \mathcal{L}(\boldsymbol{x}, \boldsymbol{\lambda}) = \min_{\boldsymbol{x} \in \chi} \max_{\boldsymbol{\lambda} \ge 0} \left( g_0(\boldsymbol{x}) + \sum_{i=1}^{l} \lambda_i g_i(\boldsymbol{x}) \right)$$
(5)

The Lagrangian  $\mathcal{L}$  is to be maximized with respect to  $\lambda \geq 0$  for a given  $\boldsymbol{x}$  which is after minimized with respect to  $\boldsymbol{x}$ . The objective function and constraints are often approximated in such a way where it will be more computationally efficient to solve the so called dual Lagrangian problem which corresponds to equation (5). The dual objective function is defined as

$$\varphi(\lambda) = \min_{x \in \chi} \mathcal{L}(x, \lambda) \tag{6}$$

 $\varphi$  is always concave which means that it is easy to maximize. If  $\min_{x \in \chi} \mathcal{L}(x, \lambda)$  has one solution for a given  $\lambda$ ,  $\varphi$  is differentiable at  $\lambda$  and can be solved

$$\frac{\partial \varphi(\boldsymbol{\lambda})}{\partial \lambda_i} = g_i(\boldsymbol{x}^*(\boldsymbol{\lambda})) \qquad i = 1, ...l \quad \text{where} \quad \boldsymbol{x}^*(\boldsymbol{\lambda}) = \min_{\boldsymbol{x} \in \chi} \mathcal{L}(\boldsymbol{x}, \boldsymbol{\lambda})$$
 (7)

The dual Lagrangian problem is especially convenient to use for CONLIN and MMA. See reference [1].

#### 3 CONLIN and MMA

A truncated Taylor approximation of the objective function (Compliance C) and the volume constraint function,  $g_1(\mathbf{x})$ , are written as

$$C(\boldsymbol{x}) \approx C(\boldsymbol{x}^k) + \sum_{e=1}^n \frac{\partial C}{\partial y_e} \bigg|_{\boldsymbol{x}=\boldsymbol{x}^k} (y_e - y_e^k)$$
 (8)

In CONLIN the intervening variable  $y_e(x_i)$  are defined as

$$\begin{cases} y_j = x_j & \text{if} & \frac{\partial C}{\partial x_j} > 0\\ y_j = \frac{1}{x_j} & \text{if} & \frac{\partial C}{\partial x_j} \le 0 \end{cases}$$
(9)

Where  $\frac{\partial C(\boldsymbol{x})}{\partial y_j}$  becomes

$$\frac{\partial C(\mathbf{x})}{\partial y_j} = \frac{\partial C(\mathbf{x})}{\partial x_j} \frac{\partial x_j}{\partial y_j} \tag{10}$$

The derivative of the compliance C with respect to the design variable becomes

$$\frac{\partial C(\boldsymbol{x})}{\partial x_j} = -\boldsymbol{u}(\boldsymbol{x})^T \boldsymbol{k}_j^0 \boldsymbol{u}_j(\boldsymbol{x})$$
 (11)

Equation (11) is negative since  $k_j^0$  is positive semidefinite, in turn with a objective function  $g_0^{C,k}$ , i.e. the compliance, the CONLIN approximation becomes

$$g_0^{C,k}(\boldsymbol{x}) = \sum_{j=1}^n \frac{\partial g_0(\boldsymbol{x}^k)}{\partial x_j} \frac{x_j^k(x_j - x_j^k)}{x_j}$$
(12)

The intervening variable is  $y_j = \frac{1}{x_j}$  from equation (9). Then,  $(P)_{nf}$  becomes

$$(\mathbf{P})_{nf}^{C,k} = \begin{cases} \min_{\mathbf{x}} & g_0^{C,k}(\mathbf{x}) \\ s.t. & \begin{cases} g_1(\mathbf{x}) = \sum\limits_{j=1}^n l_j x_j - V_{max} \le 0 \\ \mathbf{x} \in \chi \end{cases} \end{cases}$$
(13)

 $g_0^{C,k}$  is strictly convex if the strain energy of all bars is nonzero. Solving the subproblem with Lagrangian duality and evaluating the partial derivative  $\frac{\partial \mathcal{L}}{\partial x_j} = 0$  provides

$$x_{j}^{t} = \sqrt{\frac{(x_{j}^{k})^{2} \boldsymbol{u}_{j}(\boldsymbol{x}^{k})^{T} \boldsymbol{k}_{j}^{0} \boldsymbol{u}_{j}(\boldsymbol{x}^{k})}{\lambda l_{j}}} \qquad \text{where} \qquad x_{j}^{t} = \begin{cases} \alpha_{j}^{k} & \text{if } x_{j}^{t} < \alpha_{j}^{k} \\ x_{j}^{max} & \text{if } x_{j}^{t} > x_{j}^{max} \\ x_{j}^{t} & \text{otherwise} \end{cases}$$
(14)

Solving the dual problem, where the gradient of  $\varphi^k$  is the constraint condition  $g_1$  in equation (13).

$$\frac{\partial \varphi^k(\lambda)}{\partial \lambda} = \sum_{j=1}^n l_j x_j^*(\lambda) - V_{max}$$
 (15)

If the CONLIN approximation converges slowly or not at all MMA, method of moving asymptotes, is an alternative. The name explains the method quite well, the asymptotes are altered during the interactions which can improve the performance. The intervening variable  $y(x_i)$  is defined as

$$\begin{cases} y_j = \frac{1}{x_j - L_j} & \text{if} & \frac{\partial C}{\partial x_j} \ge 0\\ y_j = \frac{1}{U_j - x_j} & \text{if} & \frac{\partial C}{\partial x_j} < 0 \end{cases}$$
 (16)

Continuing in the same manner as for CONLIN, the derivative of the compliance C for MMA becomes

$$\frac{\partial C(\boldsymbol{x})}{\partial x_i} = -\boldsymbol{u}(\boldsymbol{x})^T \boldsymbol{k}_j^0 \boldsymbol{u}_j(\boldsymbol{x})$$
 (17)

Equation (17) is negative since  $\mathbf{k}_{j}^{0}$  is positive semidefinite, in turn with a objective function  $g_{0}^{M,k}$ , i.e. the compliance, the MMA approximation becomes

$$g_0^{M,k}(\mathbf{x}) = r_0^k + \sum_{j=1}^n \frac{q_{0j}^k}{x_j - L_j^k}$$
(18)

where  $q_{0j}^k$  and  $r_0^k$  are defined as

$$q_{0j}^k = (x_j^k - L_j^k)^2 \boldsymbol{u}_j(\boldsymbol{x}^k)^T \boldsymbol{k}_j^0 \boldsymbol{u}_j(\boldsymbol{x}^k)$$
(19)

$$r_0^k = g_0(\boldsymbol{x}^k) - \sum_{j=1}^n (x_j - L_j^k) \boldsymbol{u}_j(\boldsymbol{x}^k)^T \boldsymbol{k}_j^0 \boldsymbol{u}_j(\boldsymbol{x}^k)$$
(20)

The intervening variable is  $y_j = \frac{1}{x_j - L_j}$  according to (16). Then,  $(P)_{nf}$  becomes

$$(\mathbf{P})_{nf}^{M,k} = \begin{cases} \min_{\mathbf{x}} & g_0^{M,k}(\mathbf{x}) \\ s.t. & \begin{cases} g_1(\mathbf{x}) = \sum\limits_{j=1}^n l_j x_j - V_{max} \le 0 \\ \alpha_j^k \le x_j \le x_j^{max}, j = 1, ..., n \end{cases}$$

$$(21)$$

The move limit  $\alpha_j^k$  is implemented as  $\alpha_j^k = \max(x_j^{min}, L_j^k + \mu(x_j^k - L_j^k))$ . Note that  $\mu$  is restricted to  $\mu \in (0,1)$ .  $g_0^{M,k}$  is strictly convex if the strain energy of all bars is nonzero. The asymptote  $L_j$  is for the first two iterations calculated with

$$L_j = x_j - s_{init}(x_j^{max} - x_j^{min}) \tag{22}$$

A typical value for the variable  $s_{init} = 0.1$ , however this is based on the structural optimization problem. For the following iterations when calculating  $L_j$  the product of  $Z = (x_j^k - x_j^{k-1})(x_j^{k-1} - x_j^{k-2})$  are studied and  $L_j$  is calculated with

$$\begin{cases}
L_j^k = x_j^k - s_{slower}(x_j^{k-1} - L_j^{k-1}) & \text{if} & Z \le 0 \\
L_i^k = x_j^k - s_{faster}(x_j^{k-1} - L_j^{k-1}) & \text{if} & Z > 0
\end{cases}$$
(23)

Typical values for  $s_{slower} = 0.6$  and  $s_{faster} = 1.1$  however this is as well based on the structural problem and could be changed. If  $L_j$  becomes negative it should be set to  $L_j = 0$ . Solving the subproblem with Lagrangian duality and evaluating the partial derivative  $\frac{\partial \mathcal{L}}{\partial x_i} = 0$  provides

$$x_{j}^{t} = L_{j}^{k} + \sqrt{\frac{q_{0j}^{k}}{\lambda l_{j}}} \qquad \text{where} \qquad x_{e}^{*}(\lambda) = \begin{cases} \alpha_{j}^{k} & \text{if } x_{j}^{t} < \alpha_{j}^{k} \\ x_{j}^{max} & \text{if } x_{j}^{t} > x_{j}^{max} \\ x_{j}^{t} & \text{otherwise} \end{cases}$$
 (24)

Solving the dual problem, where the gradient of  $\varphi^k$  is the constraint condition  $g_1$ .

$$\frac{\partial \varphi^k(\lambda)}{\partial \lambda} = \sum_{j=1}^n l_j x_j^*(\lambda) - V_{max}$$
 (25)

Reference for theory regarding MMA and CONLIN approximations is found in [1].

## 4 Solid Isotropic Material Penalization, (SIMP)

The same method as for CONLIN and MMA are used. A truncated Taylor approximation of the objective function provides the following relation

$$C(\boldsymbol{x}) \approx C(\boldsymbol{x}^k) + \sum_{e=1}^n \frac{\partial C}{\partial y_e} \bigg|_{\boldsymbol{x}=\boldsymbol{x}^k} (y_e - y_e^k)$$
 (26)

The intervening variable is  $y_e = x_e^{-\alpha}$ . Where  $\alpha > 0$ , and  $\alpha = 1$  represents the CONLIN linearizion. Using this, the derivative of the compliance becomes

$$\frac{\partial C(\boldsymbol{x})}{\partial x_e} = -(\boldsymbol{u}_e^k)^T \boldsymbol{k}_e^0 \boldsymbol{u}_e^0 \quad \text{at } \boldsymbol{x} = \boldsymbol{x}^k$$
 (27)

where

$$\boldsymbol{u}(\boldsymbol{x}^k) = \boldsymbol{K}(\boldsymbol{x}^k)^{-1} \boldsymbol{F} \tag{28}$$

Constitutive matrix in SIMP uses Hooke's law with penalization implemented.

$$\mathbf{D} = \frac{\rho^p E}{(1 - \nu^2)} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{(1 - \nu)}{2} \end{bmatrix}$$
 (29)

A typical value of the constant parameter p is p=3 and  $\rho^p E$  is the effective Young's modulus. In the SIMP method  $\underline{\rho}\approx 0$  and  $\overline{\rho}=1$  is set in equation (33). The result is illustrated in figure 2 for different values of  $\underline{\rho}$ . The optimal solution will provide a almost zero to one solution of the problem which practically means areas of holes where  $\rho=0$  and regions where Young's modulus is E where  $\rho=1$ .

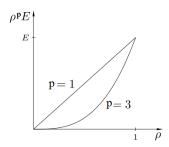


Figure 2: Youngs modulus as a function of  $\rho$  for different values of p.

The global stiffness matrix in equation (28) is calculated with equation (29) and becomes

$$K(\mathbf{x}) = \sum_{e=1}^{n} x_e^p K_e^0 \tag{30}$$

Evaluation of  $\frac{\partial C}{\partial u_e}$  with equation (27) and (26) provides the subproblem:

$$\mathbf{P} = \begin{cases} \min_{\mathbf{x}} \sum_{e=1}^{n} b_{e}^{k} x_{e}^{-\alpha} \\ s.t \end{cases} \begin{cases} \mathbf{x}^{T} \mathbf{a} = V \\ \underline{\rho} \leq x_{e} \leq \overline{\rho}, \quad e = 1, ..., n \end{cases}$$
(31)

where

$$b_e^k = \frac{(x_e^k)^{1+\alpha}}{\alpha} ((\boldsymbol{u}_e^k)^T p(x_e^k)^{p-1} \boldsymbol{k}_e^0 \boldsymbol{u}_e^k)$$
(32)

P is a convex problem and using Lagrangian duality one arrives after finding a stationary point of  $\frac{\partial \varphi_e}{x_e}$  at

$$x_{e} = \left(\frac{\alpha b_{e}^{k}}{\lambda a_{e}}\right)^{\frac{1}{1+\alpha}} \qquad \text{where} \qquad x_{e}(\lambda) = \begin{cases} \underline{\rho} & \text{if } \left(\frac{\alpha b_{e}^{k}}{\lambda a_{e}}\right)^{\frac{1}{1+\alpha}} < \underline{\rho} \\ \left(\frac{\alpha b_{e}^{k}}{\lambda a_{e}}\right)^{\frac{1}{1+\alpha}} & \text{if } \underline{\rho} \leq \left(\frac{\alpha b_{e}^{k}}{\lambda a_{e}}\right)^{\frac{1}{1+\alpha}} \leq \overline{\rho} \end{cases}$$
(33)

Investigation of stationary point of  $\varphi(\lambda)$  provides

$$\frac{\partial \varphi(\lambda)}{\partial \lambda} = \sum_{e=1}^{n} a_e x_e(\lambda) - V = 0 \tag{34}$$

Equation (34) is the volume constraint. Optimization iteration scheme and pseudo code is found in appendix, subsection 7.1. Reference for theory regarding the SIMP is found in [1].

### 5 Filters

To ensure a more mesh independent solution a filter may be added to the structural optimization algorithms. A density filter and the Helmholtz' PDE based filter will be introduced and further discussed below.

#### 5.1 Density filter

Computing neighbor elements to an element  $x_e$  with a prescribed filter radius R and where the neighbor elements are defined as  $x_i$ , results in the following relation

$$N_e = \{i | ||\boldsymbol{x}_i - \boldsymbol{x}_e|| \le R\} \tag{35}$$

Filtered density is calculated by

$$\widetilde{\rho}_e = \frac{\sum_{i \in N_e} w(\boldsymbol{x}_i) v_i \rho_i}{\sum_{i \in N_e} w(\boldsymbol{x}_i) v_i}$$
(36)

 $v_i$  is the volume. A constant weighting function is represented to the left in equation (37) and a cone-shaped weighting function is represented to the right.

$$w(\boldsymbol{x}_i) = 1 \qquad , w(\boldsymbol{x}_i) = R - ||\boldsymbol{x}_i - \boldsymbol{x}_e|| \tag{37}$$

The sensitivity of the objective function with respect to the design variables and use of the chain rule becomes

$$\frac{\partial C}{\partial \rho_e} = -p\widetilde{\rho}_e^{p-1} \boldsymbol{u}_e^T \boldsymbol{k}_e^0 \boldsymbol{u}_e \frac{w(\boldsymbol{x}_e) v_e}{\sum\limits_{j \in N_i} w(\boldsymbol{x}_j) v_j}$$
(38)

i in equation (38) represents neighbor elements. Equation (32) is instead computed as

$$b_e^k = \frac{(x_e^k)^{1+\alpha}}{\alpha} ((\boldsymbol{u}_e^k)^T p(\widehat{\rho}_e^k)^{p-1} \boldsymbol{k}_e^0 \boldsymbol{u}_e^k) \frac{w(\boldsymbol{x}_e) v_e}{\sum\limits_{j \in N_i} w(\boldsymbol{x}_j) v_j}$$
(39)

Equations and information used regarding the density filter are acquired from reference [3].

A control of the implemented algorithm can be performed by calculating  $C_1 = F^T * a(\rho)$  and  $C_2 = F^T * a(\rho + e_e \epsilon)$  for a chosen element. The derivative of the compliance for the same element should be  $\frac{\partial C}{\partial \rho_e} = \frac{C_2 - C_1}{\epsilon}$ . Where  $e_e$  is a basis vector and  $\epsilon$  is a small value.

#### 5.2 Helmholtz' PDE based filter

Helmholtz equation is defined as

$$\nabla^{T} \boldsymbol{K}_{d} \nabla \widetilde{\rho} + \widetilde{\rho} = \rho \qquad \text{where} \qquad \boldsymbol{K}_{d} = \sum_{i=1}^{d} r_{i}^{2} \boldsymbol{v}_{i} \boldsymbol{v}_{i}^{T}$$

$$\tag{40}$$

Note that the vector  $\mathbf{v}_i$  is represented as the direction of the length scale  $r_i$  in equation (40) and the number of dimensions are defined as d. If  $r_i$  has different values for different i, then anisotropy is introduced. The implemented filter regarding the assignment is isotropic, i.e.  $r_i = r_{i+1} = r_{i+d}$  The optimization problem, equation (40) is solved using FEM. In turn the information about neighbor cells is not required in the PDE filter compared to the Density filter. This is because the PDE filter has properties which are parallel to the implementation of FEM.

$$\widetilde{\rho}_e = N_e \widetilde{\rho}_e \tag{41}$$

 $\tilde{\rho}_e$  is the filtered density per element while  $\tilde{\rho}_e$  is a vector with the nodal values for element e.  $N_e$  consists of finite element interpolation functions. The filtered field is represented by equation (42) and is obtained by using Green's formula, multiplying the weight functions to the PDE filter and integrating over the design domain.

$$\sum_{i \in N_e} \int_{\Omega_i} \left[ \nabla \boldsymbol{N}_e^T \boldsymbol{K}_d \nabla \boldsymbol{N}_e + \boldsymbol{N}_e^T \boldsymbol{N}_e \right] d\Omega \widetilde{\boldsymbol{\rho}} = \sum_{i \in N_e} \rho_i \int_{\Omega_e} \boldsymbol{N}_e^T d\Omega$$
 (42)

Note that the stiffness matrix  $K_f$  is positive definite. Rewriting this with  $K = \int_{\Omega_i} \nabla N_e^T K_d \nabla N_e d\Omega$ ,

$$\pmb{M} = \int\limits_{\Omega_i} \pmb{N}_e^T \pmb{N}_e d\Omega$$
 and  $\pmb{T} = \int\limits_{\Omega_e} \pmb{N}_e^T d\Omega$  becomes

$$(K + M)\widetilde{\rho} = T\rho \tag{43}$$

By utilizing the adjoint method the sensitivity of the objective function, i.e. the compliance, is represented as

$$\frac{\partial C}{\partial \rho_e} = \frac{\partial C}{\partial \widetilde{\rho}_i} \frac{\partial \widetilde{\rho}_i}{\partial \rho_e} \tag{44}$$

Where

$$\frac{\partial C}{\partial \widetilde{\rho}_{i}} = \sum_{i \in N_{e}} -p \int_{\Omega_{i}} \mathbf{N}_{e} \widetilde{\rho}_{i}^{p-1} \mathbf{u}_{e}^{T} \mathbf{B}^{T} \mathbf{D} \mathbf{B} \mathbf{u}_{e} d\Omega \quad , \mathbf{B} = \widetilde{\nabla} \mathbf{N} \quad \text{and} \quad \frac{\partial \widetilde{\rho}_{i}}{\partial \rho_{e}} = (\mathbf{K} + \mathbf{M})^{-1} \mathbf{T} \quad (45)$$

T is a  $[nnod\ x\ nelm]$  matrix and i is the degree of freedom for one element.  $\widetilde{\nabla}$  is the linear elasticity operator.  $\frac{\partial \widetilde{\rho}_i}{\partial \rho_e} = constant$  and is calculated only one time regarding the optimization loop. Equations and information used regarding the PDE filter are acquired from reference [2].

#### 6 Result and discussion

#### 6.1 CONLIN and MMA

Represented in figure 3a is the structural optimization performed with CONLIN. In figure 3b the objective function and constraint function is observed to alter with increasing iterations. Only one result regarding the actual structure as observed in figure 3a is included in the result section since similar results were obtained with different values of  $Area_{min}$  and  $Area_{initial}$ . The Area parameter is used instead of diameter d and is defined as the cross sectional area of the bars. Note that a Young's modulus E = 1Pa is used regarding the CONLIN and MMA calculations.

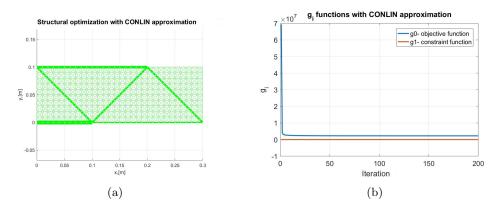


Figure 3: (a) Represents the structure after the structural optimization with CONLIN approximation, where  $Area_{min} = 3.14 * 10^{-14}$  and  $Area_{initial} = 10^{-8}$ . (b) Represents the corresponding objective function and constraint function.

The most interesting aspect is the similarity between CONLIN and MMA with identical initial values regarding  $Area_{min}$  and  $Area_{initial}$ , which is observed by comparing figure 3b and 4b. Comparing the time it takes to converge with a tolerance (tolerance equals the change from an optimized area from the previous iteration to the current iteration) of  $10^{-9}$  between CONLIN and MMA with identical initial values presents a time difference of 16.7 seconds, where MMA is less time consuming. Although solving the equilibrium equation (28) is the most time consuming part. Therefore, MMA is more preferable to use regarding the performed optimization. Observed in figure 4a is the result when altering the initial area to  $Area_{min} = 10^{-10}$ . An altering of the parameter  $Area_{min}$  in the CONLIN approximation provides a very similar behavior regarding structural visualization, objective function and constraint function results when comparing to the provided results for CONLIN. Therefore the result is not provided in this report. As some of the cross-sectional areas of the elements equals either  $Area_{min}$  or  $Area_{max}$  the stresses will vary in the trusses and in turn not result in a fully stressed design.

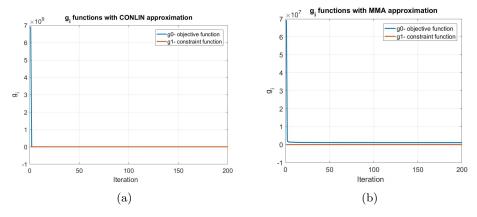


Figure 4: (a) Represents the objective function and constraint function for CONLIN where,  $Area_{min}=3.14*10^{-14}$  and  $Area_{initial}=10^{-10}$ . (b) Represents the objective function and constraint function for MMA, where  $Area_{min}=3.14*10^{-14}$  and  $Area_{initial}=10^{-8}$ 

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#### 6.2 SIMP

The colorbar represents the distribution of density in the result section for SIMP, density filter and PDE filter, i.e. red/brown color corresponds to high density elements while blue color corresponds to low density elements. Also a Young's modulus of  $E = 200 * 10^9 Pa$  is used regarding the SIMP result and the filters. Initial parameters used regarding the SIMP method are  $\rho_{min} = 10^{-3}$  and  $\rho_{initial} = 10^{-3}$ . The SIMP method provides an almost zero to one solution. A significant disadvantage using the SIMP method is the numerical difficulties that may occur. The main concern is the mesh-dependency which means if the resolution of a design is altered. For instance with a finer mesh, and optimized again the result could provide another design comparing with a coarser mesh. Figure 5a and figure 5b illustrates a mesh dependent algorithm where all parameters are the same but figure 5b consists of a finer mesh than figure 5a. The two results are similar but not entirely similar. What could also be observed are so called checkerboarders which is represented as designs with alternating solid and void cells ordered in checkerboarder-like patterns. These checkerboarders are usually not acceptable and they exist due to bad numerical modeling which overestimates the stiffness of the checkerboarders.

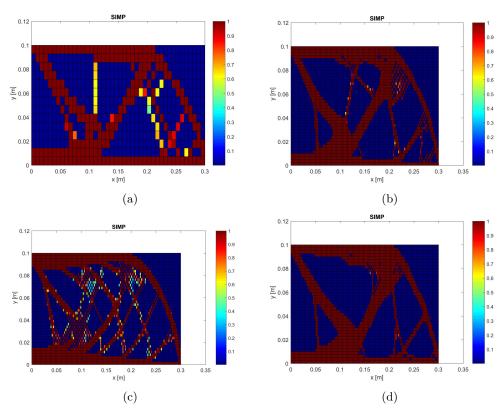


Figure 5: The optimization problem is solved using the SIMP method. (a) A coarse mesh with  $p=3, \alpha=3$ . (b) A fine mesh with  $p=3, \alpha=3$ . (c) A fine mesh with  $p=2, \alpha=3$ . (d) A fine mesh with  $p=3, \alpha=3$ .

One more difficulty using the SIMP-method is the parameter p can make the convex problem of the

thickness of the sheet into a nonconvex problem. This can result in where the algorithm terminates to different optimas for different starting values. To solve this the algorithm can be exectued several times for different starting values or increasing the value of p gradually from one which provides a convex solution to higher which will provide a more zero to one solution. Figure 5c and figure 5b illustrates this phenomenon. A higher p-value provides a more zero to one solution than with a lower p-value as there is less "half values" for a higher p-value. The exponent in equation (33) is called the damping factor,  $\eta$ . The name is derived from where elements with high strain energy is

called the damping factor,  $\eta$ . The name is derived from where elements with high strain energy is expected to have low stiffness and this leads to that they become thicker. When  $\eta$  is less than unity this modification is damped, thus the name damping factor.  $\alpha = 1$  corresponds to the CONLIN linearization, illustrated in figure 5d. When comparing different values of  $\alpha$ , a  $\alpha = 3$  yields more checkerboarder structure than  $\alpha = 1$ .

#### 6.3 Density filter

Using the SIMP method with the results in figure 5 as discussed above some difficulties have been noticed, such as checkerboarders and mesh dependency. What should be noted by investigating the filtered result is that the filtered densities,  $\tilde{\rho}$ , are plotted in the following figures. The density filter used in figure 6 and figure 7 improves the result from the SIMP algorithm by removing checkerboarders and mesh dependency. This is obtained with no use of extra constraints, a zero to one solution, stable and fast convergence and a relatively easy implementation. Density filtering makes use of preserving the volume, this means that the volume of material should be the same before and after being filtered. This would have been necessary if the structure would contain details smaller than the neighborhood area. Calculating this gave a sufficiently close result for the filtered and non-filtered elements. Initial parameters used regarding the density filter are  $\rho_{min} = 10^{-3}$ ,  $\rho_{initial} = 10^{-1}$  and  $\alpha = 3$ .

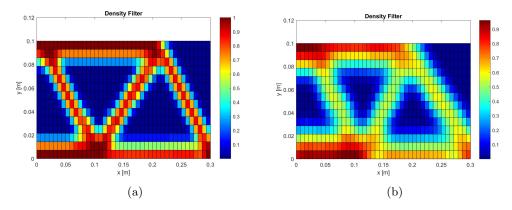


Figure 6: The optimization problem is solved using the SIMP method with a density filter. (a) Coarse mesh at a length scale r = 2\* element side, where one element side regarding the coarse mesh is  $\approx 0.0071$  (b) Coarse mesh at a length scale of r = 3.5\*element side is used.

What can be noticed comparing figure 6a with figure 7a is some kind of mesh dependency. However what is meant with mesh dependency is described as where the details are defined by the radius used to define the neighbors. The "bridges" present for the fine mesh in figure 7a can easily be removed by increasing the radius and this is performed in figure 7b. However using the same length scale, r = 3.5\*element side, for the coarse mesh will filter the structure too much which will result in a blurry image, this is illustrated in figure 6b. This is due to the density being defined as a weighted average of the densities in a mesh independent neighborhood which in turn are defined by the radius used. The effect of the weighting is highest for a constant weighting and less for the linear weighting (cone-shaped). However the constant weighting function does result in a more computationally efficient algorithm.

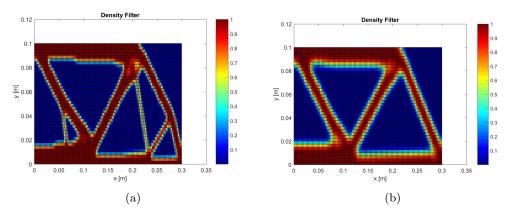


Figure 7: The optimization problem is solved using the SIMP method with a density filter. (a) Fine mesh at a length scale r = 2\* element side, where one element side regarding the fine mesh is  $\approx 0.0025$  (b) Fine mesh at a length scale of r = 3.5\*element side is used.

#### 6.4 PDE filter

The PDE filter is applied as a density filter according to [2]. The main difference with the PDE method is to avoid large matrices storing neighborhood elements. As well as for the density filter the PDE filter is volume preserving. Initial parameters used regarding the PDE filter are  $\rho_{min} = 10^{-5}$  (lower bound),  $\alpha = 2$  and  $\rho_{initial} = 10^{-3}$ .

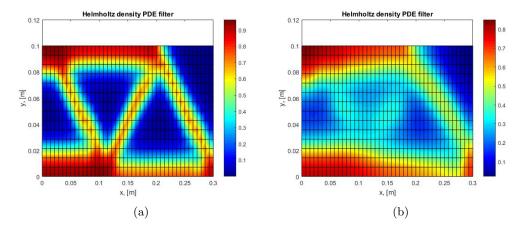


Figure 8: (a) The use of PDE filter for a coarse mesh at a length scale r = 1\* element side, where one element side regarding the coarse mesh is  $\approx 0.0071$ . At figure (b) a length scale of r = 2\* element side is used. Both figure (a) and (b) has the same tolerance regarding convergence.

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In figure 8a the use of a less sizable length scale r is used compared to in figure 8b regarding the coarse mesh. In a similar manner, the length scale r less sizable in figure 9a than in figure 9b. Comparing the result in figure 8a and figure 8b represents the importance to designate a not to great r-value when applying a PDE filter to the structural optimization. Where a too big r-value provides a very blurry result of the density distribution, this phenomenon is depicted in figure 8b. Comparing figure 9a and 9b depicts the difference of different values for r, although with a finer mesh it is not as evident that figure 9b is more blurry. It might be important to further investigate the r-value more. If the r-value is chosen as to small, indicating a smaller surrounding of the element thus closer to the boarders of the element, it might be to small and switch over to the negative side. If it becomes to small the derivative of the compliance with respect to the design variable,  $\rho$ , becomes positive and the algorithm will be incorrect. By examine the sign of the derivative of the compliance with respect to the design variable this may be detected.

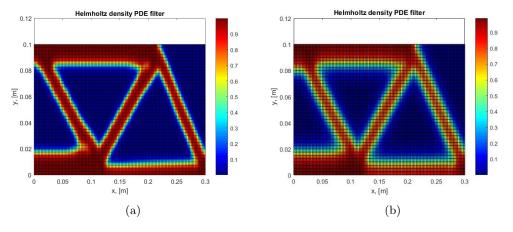


Figure 9: (a) The use of PDE filter for a fine mesh at a length scale r=1\* element side, where one element side regarding the fine mesh is  $\approx 0.0071$ . At figure (b) a length scale of r=2\*element side is used. Both figure (a) and (b) has the same tolerance regarding convergence.

In order to speed up the algorithms sparse-matrices and scalar products have been used. However there could be more done in order to optimize the algorithm. This is mainly necessary in the filtering algorithms which for the fine mesh was very time consuming.

## References

- [1] Peter W. Christensen Anders Klarbring. An Introduction to Structural Optimization, volume 153. 2009.
- [2] B. S. Lazarov and O. Sigmund. Filters in topology optimization based on helmholtz-type differential equations. *International Journal for Numerical Methods in Engineering*, 86(6):765–781, 2011.
- [3] Ole Sigmund. Morphology-based black and white filters for topology optimization. Structural and Multidisciplinary Optimization, 33(4):401–424, 2007.

## 7 Appendix

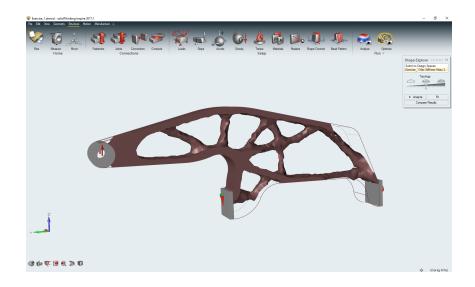
#### 7.1 Pseudo code

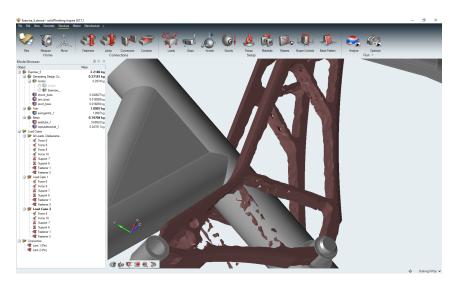
#### Optimization procedure

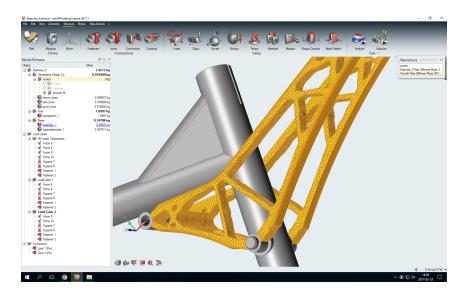
- Initiation of quantities
- Iteration k = 0, 1, 2... until the change is smaller then a set tolerance, i.e  $change = norm(\boldsymbol{x}^{k+1} \boldsymbol{x}^k)$ 
  - Calculate the tangent stiffness matrix  $\boldsymbol{D}$  with the CALFEM function hooke.
  - Calculate the element stiffness matrix  $\boldsymbol{K}$ , with e.g. CALFEM function plani4e for a four node element.
  - Calculate the displacement increment  $\boldsymbol{u}$  with CALFEM function solveq.
  - Extract displacements with CALFEM function extract.
  - Compute sensitivities of the compliance with respect to the design variable per element  $x_e$ ,  $\frac{\partial C}{\partial x_e}$ .
  - Solve the dual problem by computing  $\frac{\partial \varphi(\lambda)}{\partial \lambda}$  and obtain an optimal value  $\lambda^*$ . Subroutine textttdphidlambda can be used.
  - Compute an optimal design variable per element j,  $x_j^*(\lambda^*)$ .  $\boldsymbol{x}^* = \boldsymbol{x}^{k+1}$ . Subroutine textttgetAstar can be used.
  - Compute stresses if needed with e.g. CALFEM function bar2s for a bar element.
  - Check the change between current and previous iteration design variables and compare with tolerance.
- End iteration loop
- Accept quantities
- End load step loop

Table 1: Optimization iteration scheme. Depending on what filter and optimization algorithm used, the pseudo code presented may be altered to the desired algorithm provided in the report.

## 7.2 Inspire







#### 7.3 Software

```
CONLIN
Main
clc
clear all
close all
format long
load geomSO.mat
%Material parameters
```

E = 1; %Young's modulus Ae  $\max = (((20*10^-3)^2)*pi)/4;$ Maximum cross sectional area Ae min =  $(((20*10^-8)^2)*pi)/4$ ; Minimum cross sectional area V max =  $(2*10^-6)/2$ ; % Maximum volume allowed

%Number of elements

nelm = max(edof(:,1));

%Initiate

K = sparse(ndof,ndof); %Introduce stiffness matrix, Ke [8x8] A\_global = ones(nelm,1)\*1\*10^-10; %Introduce element Area vector, with all Area being

%Initate Global cell for normalized stiffness matrix Cell K0e = cell(nelm, 1);

for n=1:nelm

%Initate constant norm element stiffness matrix  $Cell_K0e\{n\} = bar2e(ex(n,:), ey(n,:), [E 1]);$ 

%Obtain a global length vector for each element. ec = [ex(n,:), ey(n,:)];

$$l_0 = sqrt(x_0*x_0);$$

%Global length vector  $1 \ 0 \ global(n,:) = 1 \ 0;$ 

end

 $V = A_global'*l_0_global;$ 

%Initate Tolerance  $Tol = 10^-9;$ 

%Initate quantaties

iter = 0;

Nres = 0;

q0 = 0;

%Input min and max values for fzero function.

```
lambda \min=10^10;
 lambda \max=10^15;
%Optimization loop
while Nres > Tol \mid \mid iter = 0
    iter = iter + 1;
    disp (['Iteration', num2str(iter),'-----'])
    %Reset global stiffness matrix
     K=sparse (ndof, ndof);
    %Element loop to assemble global stiffness matrix
       for i=1:nelm
           ep = [E A global(i,:)];
           Ke = bar2e(ex(i,:), ey(i,:), ep);
          %Assemble element matrices to global matrices:
               indx = edof(i, 2: end);
               K(indx, indx) = K(indx, indx) + Ke;
       end
      %Obtain global displacements
           a = solveq(K, f, bc);
           ed = extract(edof, a);
    for j = 1:nelm
      %Compute q0j according to eq 5.37, applied for CONLIN approx
       q0e(j,:) = (ed(j,:)*Cell_K0e\{j\}*ed(j,:)'*(A_global(j,:)^2));
    end
         %Solve the dual function
         lambdastar = fzero(@(lambda) dphidlambda...
         (lambda, l 0 global, V max, Ae min, Ae max, nelm, q0e)...
          ,[lambda min lambda max]);
for z=1:nelm
        %Find new rho value per element based on new lambda from the dual
        Ae_star(z,:) = getAstar(lambdastar, Ae_min, Ae_max,...
                                     l_0_{\text{global}}(z,:), q0e(z,:);
        %Compute element forces
        \operatorname{stress}(z,:) = \operatorname{bar2s}(\operatorname{ex}(z,:), \operatorname{ey}(z,:), [E \operatorname{Ae} \operatorname{star}(z,:)], \operatorname{ed}(z,:)) / \operatorname{Ae} \operatorname{star}(z,:);
end
     K=sparse (ndof, ndof);
     %Element loop to assemble global stiffness matrix
       for i=1:nelm
           ep = [E A global(i,:)];
```

```
Ke = bar2e(ex(i,:), ey(i,:), ep);
         %Assemble element matrices to global matrices:
              indx = edof(i, 2: end);
             K(indx, indx) = K(indx, indx) + Ke;
      end
      %Obtain displacement
          a = solveq(K, f, bc);
          ed = extract (edof, a);
 for z=1:nelm
     \% Compute\ element\ stress
     stress(z,:) = bar2s(ex(z,:), ey(z,:), [E Ae_star(z,:)], ed(z,:)) / Ae_star(z,:);
 end
    %Compute the change between current and previous iteration
    %regarding the updated area.
     Nres = norm(Ae star - A global);
     A global = Ae star;
     %Plot vectors
     Ae_star_plot(:,iter) = Ae_star;
     Iter plot(:, iter) = iter;
     g0(:, iter) = f'*a;
     stress global(:, iter) = stress;
     g1(:,iter) = A global'*l 0 global-V max;
     V = A global'*1 0 global;
     fprintf('Iter %i Res %e\n', iter, Nres)
     end
%%
figure
                    %Magnification of Cross Section area.
fac = 4*10^6;
magnfac = 1*10^-10; %Magnification of Deformed structure.
myeldisp2 (ex, ey, ed, [1 2 0], magnfac, A global, fac);
grid on
xlabel ('x, [m]')
ylabel('y,[m]')
title (['Structural optimization with CONLIN approximation'], 'FontSize', 14)
figure
plot(Iter plot, g0, 'LineWidth', 2)
plot (Iter plot, g1, 'LineWidth', 2)
grid on
xlabel('Iterationer', 'FontSize', 14)
ylabel ('g_i', 'FontSize', 14)
legend('g0- objective function', 'g1- constraint function')
title ('g_i functions with CONLIN approximation', 'FontSize', 14)
set (gca, 'FontSize', 14)
```

```
%Check Area fulfills Box constraint
A_{max\_check} = max(A_{global});
A \min \text{ check} = \min(A \text{ global});
           if A max check <= Ae max
                     disp ('Valid Box constraint regarding A max')
                     disp ('Box constraint not fulfilled')
           end
              if A_{\min}_{\operatorname{check}} >= Ae_{\min}
                     disp ('Valid Box constraint regarding A_min')
                     disp ('Box constraint not fulfilled')
             end
     %Check procentage of elements in A global which have the values Ae min
     %and Ae max.
      procent_min = length(find(A_global == Ae_min))/nelm
      procent max = length(find(A global == Ae max))/nelm
Dphidlambda
function \ [ \ dphidlambda \ ] \ = \ dphidlambda ( \ lambda \ , l\_0\_global \ , Vmax \ , Ae\_min \ , \ Ae\_max \ , nelm \ , q0e^{-1} \ ,
%Compute dphidlambda, solving the dual function.
%Input: Lambda
                     l\_0 , Global length vector
%
%
                     Vmax, Maximum volume
%
                     Ae min, Minimum area
%
                     Ae max, Maximum area
                                  , qo array, computed in the main m-file.
%Output: dphidlambda
             h = 0;
           for z=1:nelm
                     Ae star(z,:) = getAstar(lambda, Ae min, Ae max, l 0 global(z,:), q0e(z,:));
                     he = l_0_global(z,:)*Ae_star(z,:);
                     h = he + h;
           end
%Constraint function
dphidlambda = h-Vmax;
end
GetAstar
function [ Ae star ] = getAstar( lambda, Ae min, Ae max, 1 0, q0e )
%Calculate A* for a given lambda.
%Input: lambda
                     Ae min, Minimum area
                     Ae max, Maximum area
%
                     l_0 , Global length vector
                     q0e , qo(e) (per element), computed in the main m-file.
```

```
%Output: A*
Ae\_trial = sqrt(q0e/(lambda*l\_0));
   if Ae_trial < Ae_min;
          Ae star = Ae \min;
   \verb|elseif| Ae trial| > Ae max|
         Ae\_star = Ae\_max;
   else
         Ae_star = Ae_trial;
   end
end
MMA
Main
clc
clear all
close all
format long
load geomSO.mat
%Material parameters
E = 1;
            %Young's modulus
Ae_{max} = (((20*10^{-3})^{2})*pi)/4;
                                      Maximum cross sectional area
Ae_{min} = (((20*10^--8)^2)*pi)/4;
                                     %Minimum cross sectional area
                                      %Maximum allowed volume
V_{max} = 2*10^-6;
%Number of elements
nelm = max(edof(:,1));
%Initiate
K = sparse(ndof, ndof);
A global = ones(nelm,1)*10^--8; %Introduce element Area vector
%Initate Global cell for normalized stiffness matrix
Cell_K0e = cell(nelm, 1);
%Initiate MMA parameters
s slower = 0.6;
s_faster = 1.1;
s_init = 0.1;
A_last = 0;
A lastlast = 0;
my = 0.5;
```

```
%Initate constant norm element stiffness matrix
    Cell_K0e\{n\} = bar2e(ex(n,:), ey(n,:), [E 1]);
    %Obtain a global length vector for each element.
    ec = [ex(n,:), ey(n,:)];
    x_0 = [ec(1,2) - ec(1,1);
        ec(1,4) - ec(1,3);
    1 \ 0 = sqrt(x \ 0'*x \ 0);
    %Global length vector
    l_0_{global}(n,:) = l_0;
end
V = A_global'*l_0_global;
%Initate Tolerance
Tol = 10^-9;
%Initate quantaties
iter = 0;
Nres = 0;
q0 = 0;
lambda min=10^10;
lambda \max=10^15;
%Optimization loop
while Nres > Tol \mid \mid iter == 0
    iter = iter + 1;
    disp (['Iteration', num2str(iter),'-----'])
    K=sparse (ndof, ndof);
    %Element loop to assemble global stiffness matrix
    for \quad i = 1 : nelm
        ep = [E A global(i,:)];
        Ke = bar2e(ex(i,:), ey(i,:), ep);
        %Assemble element matrices to global matrices:
        indx = edof(i, 2: end);
        K(indx, indx) = K(indx, indx) + Ke;
    end
    %Obtain displacement
    a = solveq(K, f, bc);
    ed = extract(edof,a);
```

for n=1:nelm

```
for j=1:nelm
    Modifying the asymptotes during the iterations
    %according to page 68 in the course literature.
     if iter < 3
         Lj(j,:) = A_global(j,:) - s_init*(Ae_max-Ae_min);
     else
         sign1 = A_global(j,:) - A_last(j,:);
         sign2 = A last(j,:) - A lastlast(j,:);
         if sign1*sign2 > 0
              Lj(j,:) = A_global(j,:) - s_faster*(A_last(j,:)-Lj(j,:));
         else
             Lj(j,:) = A global(j,:) - s slower*(A last(j,:) - Lj(j,:));
         end
    end
    %If Lj is negative, give it a value of zero.
    if sign(Lj(j,:)) = 1
         Lj(j,:) = 0;
         %disp('Negative Lj')
    end
    %Update alpha j per element
    alpha j(j,:) = max([Ae min, Lj(j,:)+my*(A global(j,:)-Lj(j,:))]);
    %Compute q0j according to eq 5.37
    q0e(j,:) = ((A_global(j,:) - Lj(j,:))^2) * ed(j,:) * Cell_K0e\{j\} * ed(j,:)';
end
%Solve the dual function
lambdastar = fzero(@(lambda) dphidlambda...
     ,[lambda min lambda max]);
for z=1:nelm
    %Find new rho value per element based on new lambda from the dual
    Ae\_star\left(z\,,:\right) = \ getAstar\left(\ lambdastar\ ,\ Ae\_min,\ Ae\_max,l\_0\_global\left(z\,,:\right)\,,q0e\left(z\,,:\right)\,,
    %Compute element forces
    \operatorname{stress}(z,:) = \operatorname{bar2s}(\operatorname{ex}(z,:), \operatorname{ey}(z,:), [E \operatorname{Ae} \operatorname{star}(z,:)], \operatorname{ed}(z,:)) / \operatorname{Ae} \operatorname{star}(z,:);
end
%Compute the change between current and previous iteration
%regarding the updated area.
res=Ae star-A global;
Nres = norm(res);
```

 $A_{lastlast} = A_{last};$ 

```
A last = A global;
    A global = Ae star;
    %Plot vectors
    Ae_star_plot(:, iter) = Ae_star;
    Iter_plot(:, iter) = iter;
    g0(:,iter) = f'*a;
    stress global(:,iter) = stress;
    res plot(:, iter) = Nres;
    g1(:,iter) = A global'*l 0 global-V max;
    V = A \text{ global} * 1 0 \text{ global};
    fprintf('Iter %i Res %e\n', iter, Nres)
end
figure
plot(Iter_plot, res_plot, 'LineWidth', 1.5)
xlabel('Iteration')
ylabel ('Residual')
%%
fac = 4*10^6;
                    %Magnification of Cross Section area.
magnfac = 1*10^-10; %Magnification of Deformed structure.
myeldisp2 (ex, ey, ed, [1 2 0], magnfac, A global, fac);
grid on
xlabel('x,[m]')
ylabel('y,[m]')
title (['Structural optimization with MMA approximation'], 'FontSize', 14)
figure
plot (Iter_plot, g0, 'LineWidth', 2)
hold on
plot (Iter plot, g1, 'LineWidth', 2)'
grid on
xlabel ('Iterationer', 'FontSize', 14)
ylabel ('g_i', 'FontSize', 14)
legend('g0- objective function', 'g1- constraint function')
title ('g_i functions with MMA approximation', 'FontSize', 14)
set (gca, 'FontSize', 14)
xlim ([0 200])
%Check Area fulfills Box constraint
A \max \text{ check} = \max(A \text{ global});
A min check = min(A global);
i\,f\ A\_max\_check <= Ae\_max
    disp ('Valid Box constraint regarding A max')
else
    disp ('Box constraint not fulfilled')
```

end

```
i\,f\ A\ \min\ check >= Ae\ \min
    disp ('Valid Box constraint regarding A_min')
    disp ('Box constraint not fulfilled')
end
Dphidlambda
function [ dphidlambda ] = dphidlambda ( lambda , l 0 global , Vmax , Ae min . . .
, Ae max, nelm, q0e, Lj, alpha j )
%Compute dphidlambda, solving the dual function.
%
%Input: Lambda
%
        l\_0 , Global length vector
%
        Vmax, Maximum volume
%
        Ae min, Minimum area
%
        Ae max, Maximum area
%
             , qo array, computed in the main m-file.
               , Moving asymptote,
%
        alpha j, move limit
%Output: dphidlambda
     h = 0;
    for z=1:nelm
        Ae_star(z,:) = getAstar(lambda, Ae_min,...
        Ae_{max}, l_0_{global}(z,:), q0e(z,:), Lj(z,:), alpha_j(z,:));
        he = 1 \ 0 \ global(z,:) * Ae \ star(z,:);
        h = he + h;
    end
%Constraint function
dphidlambda = h-Vmax;
end
GetAstar
function [ Ae_star ] = getAstar( lambda, Ae_min, Ae_max, l_0, q0e, Lj, alpha_j )
%Calculate A* for a given lambda.
%Input: lambda
%
        Ae min, Minimum area
        Ae max, Maximum area
%
%
        l_0 , Global length vector
        q0e , qo(e) (per element), computed in the main m-file.
%
%
        Lj , Moving asymptote,
        alpha_j, move limit
%Output: A*
Ae trial =Lj + sqrt(q0e/(lambda*l 0));
      Ae trial < alpha j;
   i f
          Ae star = alpha j;
   elseif Ae trial > Ae max
```

```
Ae\_star = Ae\_max;
   else
         Ae_star = Ae_trial;
   end
end
SIMP
Main
clc
clear all
close all
format long
el\_length\_Fine = 0.0025;
el length Coarse = 0.0071;
%Selection of coarse mesh or fine mesh.
prompt = ['Select Coarse Mesh or Fine Mesh:\n'...
    'A1: Coarse Mesh \nB1: Fine Mesh \n'];
A1 = 1;
B1 = 2;
selection = input(prompt)
if selection = 1
    load MBBCoarseMesh
   %Selection of what length scale R to use.
    prompt = ['Select \ a \ length \ scale \ R \ ' \dots ]
        'C2: R = Element Side*2.5 \ n'];
   A2 = 1;
   B2 = 2;
    C2 = 3;
    selection 2 = input (prompt)
    if selection 2 = 1
       R = el_length_Coarse;
    elseif selection 2 = 2
       R = el length Coarse *2;
    elseif selection 2 == 3
       R = el_length_Coarse *2.5;
```

```
^{
m end} ^{
m elseif} ^{
m selection} = 2 ^{
m load} ^{
m MBBFineMesh}
```

```
load MBBFineMesh
    %Selection of what length scale R to use.
    prompt = ['Select a length scale R\n'...
        'C2: R = Element Side *2.5 \ n';
    A2 = 1;
    B2 = 2;
    C2 = 3;
    selection 2 = input (prompt)
    if selection 2 = 1
        R = el length Fine;
    elseif selection 2 = 2
        R = el length Fine*2;
    elseif selection 2 = 3
        R = el length Fine *2.5;
    end
end
%Input min and max values for fzero function.
lambda_min=10^-8;
lambda_max=10^-1;
nen = 4; %Number of nodes per element
%Extract x and y coordinates
[exC, eyC] = coordxtr(edof, coord, dof, nen);
ec = [exC, eyC];
                       %Number of elements
nelm = \max(edof(:,1));
ndof = max(max(dof));
                       %Number of degree of freedom, displacement part
nnod = ndof/2;
                       %Number of nodes
%Material parameters
E = 200e9;
                        %Young's modulus
rho \max = 1;
                         Maximum density factor
rho min = 10^-5;
                         %Minimum density factor
thickness = 20*10^-3;
                         %Thickness
V box = 300*100*(10^-6)*thickness;
                         %Maximum allowed volume
V \max = 0.4 * V \text{ box};
v = 0.3;
                         %Poissons tal
eq = 1;
```

```
%Initiate
x_e = ones(nelm,1)*10^-3; %Introduce element Area vector, with all Area being
                           %Introduce Area vector
a e = zeros(nelm, 1);
T = sparse(nnod, nelm);
                           %Introduce T matrix, [nnod x nelm]
M = sparse(nnod, nnod);
                           %Introduce Mass matrix
K_{dens} = sparse(nnod, nnod); %Introduce constant stiffness matrix, Ke [4x4]
K rho = sparse(ndof, ndof); %Introduce stiffness matrix, Ke [8x8]
dgdrhotilde = sparse(nnod,1);
                           %Penalization factor
p = 3;
ptype = 1; %Plane stress
ep = [ptype thickness 2];
%Hooke matrix
D0 = hooke(ptype, E, v);
%Parameters for flw2i4e
ep2 = [thickness 2];
D2 = [1 \ 0;
    0 \ 1;
for n=1:nelm
    a e(n,:) = ElementArea(ec(n,:)); %Area per element
    %Compute T matrix
    [Ke, Te] = flw2i4e(exC(n,:), eyC(n,:), ep2, D2, eq);
    %Assemble element matrices to global matrices:
    indx = enod(n, 2 : end);
    T(indx, n) = T(indx, n) + Te;
    indx = enod(n, 2 : end);
    K_{dens}(indx, indx) = K_{dens}(indx, indx) + Ke.*R^2;
    %Compute Mass matrix
    Me = flw2i4m(exC(n,:), eyC(n,:), thickness);
    %Assemble element matrices to global matrices:
    indx = enod(n, 2:end);
    M(indx, indx) = M(indx, indx) + Me;
end
%Compute constant drhoTilde drho matrix.
drhoTilde drho = (K dens + M) \setminus T;
%Initate Tolerance
Tol = 10^-4;
%Initate quantaties
iter = 0;
Nres = 0;
```

```
alpha = 2;
%Optimization loop
while Nres > Tol \mid \mid iter == 0
    iter = iter + 1;
    disp (['Iteration', num2str(iter),'----'])
    %Reset global matrices
    K rho = sparse(ndof, ndof);
    dgdrhotilde = sparse(nnod,1);
    %Compute filtered density
    x_e_tilde = drhoTilde_drho*x_e;
    %Extract the densities at the nodal points.
    ed_rho=extract(enod,x_e_tilde);
    %Element loop to assemble global stiffness matrix
    for i=1:nelm
        Ke rho=plani4e rho(exC(i,:), eyC(i,:), ep,D0,ed rho(i,:),p);
        %Assemble element matrices to global matrices:
        indx = edof(i, 2: end);
        K_{rho}(indx, indx) = K_{rho}(indx, indx) + Ke_{rho};
    end
    %Obtain displacement
    a = solveq(K rho,F,bc);
    ed = extract(edof, a);
    for j=1:nelm
        %Compute dgdrhotilde
        dgdrhotilde e=getdgdrhotilde el(exC(j,:),eyC(j,:),ep,D0,ed(j,:)...
             , ed rho(j, :), p);
        %Assemble element matrices to global matrices:
        indx = enod(j, 2: end);
        dgdrhotilde(indx,1) = dgdrhotilde(indx,1) + dgdrhotilde e;
    end
    %Compute global dg drho
    dg drho = dgdrhotilde '*drhoTilde drho;
    b = (1/alpha).*dg drho'.*(x e.^(1+alpha));
    %Solve the dual function
    lambdastar = fzero(@(lambda) dphidlambda b...
        (lambda, rho_max, rho_min, alpha, b, a_e, V_max, nelm, thickness)...
        ,[lambda_min lambda_max]);
```

```
for \ z\!=\!1\!:\!nelm
         %Find new rho value per element based on new lambda from the dual
         %function
         xe\_star\left(z\,,:\right) = \ getx\_e\_star\left(\ lambdastar\ ,rho\_max\,,rho\_min\,,alpha\,,b\left(z\,,:\right)\,,a\_e\left(z\,,:\right)
     end
    %Compute the change between current and previous iteration
    %regarding the updated density.
     Nres = norm(xe star - x e);
    x e = xe star;
    %Plot vectors
     Iter_plot(:, iter) = iter;
     g0(:, iter) = F'*a;
     g1(:,iter) = x_e'*a_e*thickness-V_max;
     fprintf('Res %e\n', Nres)
     if iter == 200
         fprintf('Force break optimization loop at iteration: %i \n', iter)
         break
     \quad \text{end} \quad
\quad \text{end} \quad
%%
ed el = extract (enod, x e tilde);
figure
fill (exC', eyC', ed_el')
title ('Helmholtz density PDE filter')
xlabel('x, [m]')
ylabel('y, [m]')
colormap ('Jet')
%%
figure
plot (Iter_plot, g0, 'LineWidth', 2)
grid on
xlabel('Iterationer')
ylabel ('g 0')
title ('Compliance history')
figure
plot (Iter_plot, g1, 'LineWidth', 2)
xlabel('Iterationer')
ylabel ('g 1')
grid on
title ('Weight history')
Dphidlambda
function [ dphidlambda ] = dphidlambda_b(lambda,rho_max,rho_min,alpha,...
be, a_e, V_max, nelm, thickness)
%Compute dphidlambda, solving the dual function.
```

```
%
%Input: Lambda
        l_0 , Global length vector
%
        Vmax, Maximum volume
%
        rho min, Minimum density
%
        rho_max, Maximum density
%
        alpha, set parameter in the main m-file, connected to the
%
        intervening variable in the SIMP OC method.
%
            , b array, computed in the main m-file.
%
        a_e , area of each element
%
        nelm , number of elements
%
        thickness
%Output: dphidlambda
     h = 0;
    for z=1:nelm
        xe_star(z,:) = getx_e_star(lambda, rho_max, rho_min, alpha, be(z,:), a_e(z,:));
        he = a e(z,:) * xe star(z,:);
        h = he + h;
    end
%Constraint function
dphidlambda = h*thickness-V max;
end
getx-e-star
function [ xe_star ] = getx_e_star( lambda,rho_max,rho_min,alpha,be,a_e )
%Calculate x* (= rho*) for a given lambda.
%Input: lambda
        rho min, Minimum density
%
        rho max, Maximum density
%
        l_0 , Global length vector
%
        be , b array, computed in the main m-file.
%
        alpha, set parameter in the main m-file, connected to the
        intervening variable in the SIMP OC method.
        a_e , area of each element
%Output: x*
xe trial = ((alpha*be)/(lambda*a e))^(1/(1+alpha));
   if xe trial < rho min;
          xe star = rho min;
   elseif xe trial > rho max
         xe star = rho max;
   else
         xe star = xe trial;
```

end

```
ElementArea
```

```
function [ A0 ] = ElementArea ( ec )
%Compute element area
%Input: ec, coordinate matrix,
        where ec(e) = [1x8]
%Output: A0, the area of one element.
x1 = ec(1,1);
x2 = ec(1,2);
x3 = ec(1,3);
x4 = ec(1,4);
y1 = ec(1,5);
y2 = ec(1,6);
y3 = ec(1,7);
y4 = ec(1,8);
%Element Area.
A0 = (1/2)*abs(x1*y2+x2*y3+x3*y4+x4*y1-x2*y1-x3*y2-x4*y3-x1*y4);
end
```

#### Density filter

#### Main, constant weight function

```
clc
clear all
close all
format long
tic
% load MBBFineMesh
load MBBCoarseMesh
nen = 4;
[ex,ey]=coordxtr(edof,coord,dof,nen);
ec = [ex, ey];
%Number of elements
nelm = max(edof(:,1));
ndof = max(max(dof));
\% Material parameters
E = 200e9;
                                  % Young's modulus
rho max = 1;
                                  % Maximum density factor
rho_min = 10^-5;
                                 % Minimum density factor
thickness = 20*10^-3;
                                 % Thickness
V box = 300*100*(10^-6)*thickness;
V \max = 0.4 * V \text{ box};
                                 % Maximum allowed volume
v = 0.3;
                                 % Poissons tal
```

```
%Filter radius
\Re R = 0.0025;
R = 0.0071*2;
%Initiate global matrix
K = sparse(ndof, ndof);
                                   % Stiffness
K0=sparse (ndof, ndof);
                                   % Normalized stiffness
Cell K0e = cell(nelm, 1);
                                   % Normalized stiffness cell
                                   % Initial design parameter
x e = ones(nelm,1)*10^-1;
a e = zeros(nelm, 1);
                                   % Area vector
                                   % Penalization factor
p = 3;
                                   % Plane stress
ptype = 1;
ep = [ptype thickness 2];
% Constitutive matrix
D0 = hooke(1,E,v);
for n=1:nelm
    % Calculate area and volume of each element
    a e(n,:) = ElementArea(ec(n,:));
    \overline{\text{Vol}} = \text{lickness} * \text{ElementArea}(\text{ec}(n,:));
    %Initate constant norm element stiffness matrix
    [Cell K0e\{n\}] = plani4e (ex(n,:), ey(n,:), ep,D0);
    % Calculate centrum of current element and put in a vector
    y centrum=\min(ey(n,:)) + (\max(ey(n,:)) - \min(ey(n,:))) / 2;
    x = \min(ex(n,:)) + (\max(ex(n,:)) - \min(ex(n,:))) / 2;
    kord centrum = [x centrum, y centrum];
    % Neigbour matrix
    N(:,1) = edof(:,1);
    var = 0;
    for lite=1:nelm
        % Calculate centrum and check if neighbour
         y centrum test=\min(\text{ey}(\text{lite},:)) + (\max(\text{ey}(\text{lite},:)) - \min(\text{ey}(\text{lite},:))) / 2;
         x centrum test=min(ex(lite,:))+(max(ex(lite,:))-min(ex(lite,:)))/2;
         kord test=[x centrum test, y centrum test];
         if norm(kord test-kord centrum) < R % lite ~= n
             % Update neigbouring matrix
             N(n, lite) = lite;
             % Distance from centrum of current element cone-shaped function
             weights (n, lite)=1;
             % Global M-matrix
             M(n, lite) = weights(n, lite) * Vol el(n, :);
             var=weights(n, lite)*Vol el(n,:)+var;
         end
    end
    M(n,:)=M(n,:) / var;
end
%Initate Tolerance
```

```
Tol = 10^-2;
%Initate quantaties
iter = 0;
Nres = 0;
alpha = 3;
\% Interval to fzero
lambda \min=10^-20;
lambda \max=10^20;
%Optimization loop
while Nres > Tol \mid \mid iter = 0
    iter = iter + 1;
    disp (['Iteration', num2str(iter),'-----'])
    K = sparse(ndof, ndof);
    % Element loop to assemble global stiffness matrix
    for i=1:nelm
         x \in \text{tilde}(i,:) = \text{weights}(i,:) *x \in \text{sum}(\text{weights}(i,:));
         % Constitutive matrix
         D = (x e tilde(i,:)^p)*D0;
         \% Stiffness matrix
         Ke = plani4e(ex(i,:), ey(i,:), ep,D);
         %Assemble element matrices to global matrices:
         indx = edof(i, 2: end);
         K(indx, indx) = K(indx, indx) + Ke;
    end
    %Obtain displacement
    a = solveq(K, F, bc);
    ed = extract(edof, a);
    % Assemble and calculate the sensitivities
    for j=1:nelm
         dC \operatorname{drhotilde}(j,:) = \operatorname{ed}(j,:) *(-p*x \ e \ \operatorname{tilde}(j,:)^{(p-1)}) * \operatorname{Cell} \operatorname{K0e}\{j\} * \operatorname{ed}(j,:)';
    end
    dC drho=dC drhotilde '*M;
    be= -(1/alpha)*x_e.^(1+alpha).*dC_drho';
    % Solve the dual Lagrangian function
    lambdastar = fzero(@(lambda) dphidlambda_b...
         (lambda, rho\_max, rho\_min, alpha, be, a\_e, V\_max, nelm, thickness)...
          ,[lambda min lambda max]);
    % Find KKT point
    for z=1:nelm
         xe_star(z,:) = getx_e_star(lambdastar_,rho_max,rho_min,alpha,be(z,:),a_e(z,:)
    end
    % Calculate the change
```

```
Nres = norm(xe star - x e);
    x_e = xe_star;
    %Plot vectors
    Iter plot(:,iter) = iter;
    g0(:,iter) = F'*a;
    g1(:,iter) = x_e_tilde'*a_e*thickness-V_max;
    fprintf ('Res %e\n', Nres)
    if iter = 1000
         fprintf('Force break optimization loop at iteration: %i \n', iter)
         break
    end
    \% Check if R is to small
    if any(dC drho<0)==0
         fprintf('R to small')
         break
    end
end
%%
figure
plot (Iter_plot, g0, 'LineWidth', 2)
grid on
xlabel('Iterationer')
ylabel('g_0')
title('Compliance history')
figure
plot (Iter_plot, g1, 'LineWidth', 2)
xlabel('Iterationer')
ylabel('g_1')
grid on
title ('Weight history')
figure
colormap('jet')
fill (ex', ey', x_e_tilde)
title ('Density Filter')
xlabel('x [m]')
ylabel ('y [m]')
Main, cone-shaped weight function
clc
clear all
close all
format long
tic
% load MBBFineMesh
load MBBCoarseMesh
nen = 4;
[ex, ey] = coordxtr(edof, coord, dof, nen);
ec = [ex, ey];
```

```
%Number of elements
nelm = max(edof(:,1));
ndof = max(max(dof));
\% Material parameters
E = 200e9;
                                  % Young's modulus
rho_max = 1;
                                  % Maximum density factor
rho min = 10^-5;
                                  % Minimum density factor
thickness = 20*10^-3;
                                  % Thickness
V box = 300*100*(10^-6)*thickness;
                                  % Maximum allowed volume
V \max = 0.4 * V \text{ box};
                                  % Poissons tal
v = 0.3;
%Filter radius
\Re R = 0.0025;
R = 0.0071*3.5;
%Initiate global matrix
                                  % Stiffness
K = sparse(ndof, ndof);
                                  % Normalized stiffness
K0=sparse(ndof,ndof);
Cell K0e = cell(nelm, 1);
                                  % Normalized stiffness cell
x e = ones(nelm,1)*10^-1;
                                  % Initial design parameter
                                  % Area vector
a e = zeros(nelm, 1);
                                  % Penalization factor
p = 3;
                                  % Plane stress
ptype = 1;
ep = [ptype thickness 2];
% Constitutive matrix
D0 = hooke(1,E,v);
for n=1:nelm
    % Calculate area and volume of each element
    a e(n,:) = ElementArea(ec(n,:));
    Vol el(n,:) = thickness*ElementArea(ec(n,:));
    %Initate constant norm element stiffness matrix
    [Cell_K0e\{n\}] = plani4e(ex(n,:), ey(n,:), ep, D0);
    % Calculate centrum of current element and put in a vector
    y_{entrum}=\min(ey(n,:))+(\max(ey(n,:))-\min(ey(n,:)))/2;
    x_{entrum}=min(ex(n,:))+(max(ex(n,:))-min(ex(n,:)))/2;
    kord\_centrum = [x\_centrum \,, y\_centrum \,] \ ';
    % Neigbour matrix
    N(:,1) = edof(:,1);
    var = 0;
    for lite=1:nelm
        % Calculate centrum and check if neighbour
         y centrum test=\min(\text{ey}(\text{lite},:)) + (\max(\text{ey}(\text{lite},:)) - \min(\text{ey}(\text{lite},:))) / 2;
         x_{entrum_test=min(ex(lite,:))+(max(ex(lite,:))-min(ex(lite,:)))/2;
         kord_test=[x_centrum_test, y_centrum_test]';
         if norm(kord test-kord centrum) < R % lite ~= n
             % Update neigbouring matrix
```

```
N(n, lite) = lite;
            % Distance from centrum of current element cone-shaped function
             weights(n, lite)=R - norm(kord_test-kord_centrum);
            % Global M-matrix
            M(n, lite) = weights(n, lite) * Vol_el(n, :);
             var=weights(n, lite)*Vol_el(n,:)+var;
        end
    end
    M(n,:) = M(n,:) / var;
end
%Initate Tolerance
Tol = 10^-2;
%Initate quantaties
iter = 0;
Nres = 0;
alpha = 3;
% Interval to fzero
lambda min=10^-20;
lambda \max=10^20;
%Optimization loop
while Nres > Tol \mid \mid iter = 0
    iter = iter + 1;
    disp (['Iteration', num2str(iter),'-----'])
    K = sparse(ndof, ndof);
    % Element loop to assemble global stiffness matrix
    for i=1:nelm
        x \in \text{tilde}(i,:) = \text{weights}(i,:) *x \in \text{sum}(\text{weights}(i,:));
        % Constitutive matrix
        D = (x_e_tilde(i,:)^p)*D0;
        % Stiffness matrix
        Ke = plani4e(ex(i,:), ey(i,:), ep,D);
        %Assemble element matrices to global matrices:
        indx = edof(i, 2:end);
        K(indx, indx) = K(indx, indx) + Ke;
    end
    %Obtain displacement
    a = solveq(K, F, bc);
    ed = extract(edof, a);
    % Assemble and calculate the sensitivities
    for j=1:nelm
        dC_{drhotilde(j,:)} = ed(j,:)*(-p*x_e_{tilde(j,:)}^(p-1))*Cell_K0e\{j\}*ed(j,:)';
    end
    dC_drho=dC_drhotilde'*M;
```

```
be= -(1/alpha)*x e.^(1+alpha).*dC drho';
    % Solve the dual Lagrangian function
    lambdastar = fzero(@(lambda) dphidlambda b...
        (lambda, rho\_max, rho\_min, alpha, be, a\_e, V\_max, nelm, thickness)...
         ,[lambda min lambda max]);
    % Find KKT point
    for z=1:nelm
        xe star(z,:) = getx e star(lambdastar, rho max, rho min, alpha, be(z,:), a e(z,:)
    end
    % Calculate the change
    Nres = norm(xe\_star-x\_e);
    x_e = xe_star;
    %Plot vectors
    Iter_plot(:,iter) = iter;
    g0(:,iter) = F'*a;
    g1(:,iter) = x e tilde'*a e*thickness-V max;
    fprintf('Res %e\n', Nres)
    if iter == 1000
         fprintf('Force break optimization loop at iteration: %i \n', iter)
        break
    end
    \% Check if R is to small
    if any(dC drho<0)==0
         fprintf('R to small')
        break
    end
end
%%
figure
plot(Iter_plot,g0,'LineWidth',2)
grid on
xlabel('Iterationer')
ylabel ('g 0')
title('Compliance history')
figure
plot (Iter plot, g1, 'LineWidth', 2)
xlabel('Iterationer')
ylabel('g_1')
grid on
title('Weight history')
figure
colormap('jet')
fill (ex', ey', x e tilde)
title ('Density Filter')
xlabel('x [m]')
ylabel ('y [m]')
toc
```

## PDE filter

```
Main
```

```
clc
clear all
close all
format long
el length Fine = 0.0025;
el length Coarse = 0.0071;
%Selection of coarse mesh or fine mesh.
prompt = ['Select Coarse Mesh or Fine Mesh:\n'...
    'A1: Coarse Mesh \nB1: Fine Mesh \n'];
A1 = 1;
B1 = 2;
selection = input(prompt)
if selection = 1
   load MBBCoarseMesh
   %Selection of what length scale R to use.
   prompt = ['Select a length scale R \setminus n' \dots
       'C2: R = Element Side *2.5 \ n';
   A2 = 1;
   B2 = 2;
   C2 = 3;
   selection 2 = input (prompt)
   if selection 2 = 1
       R = el length Coarse;
    elseif selection 2 = 2
       R = el length Coarse *2;
    elseif selection 2 == 3
       R = el length Coarse *2.5;
   end
elseif selection = 2
   load MBBFineMesh
   %Selection of what length scale R to use.
   prompt = ['Select a length scale R \ '...
       'C2: R = Element Side *2.5 \ n';
   A2 = 1;
   B2 = 2;
   C2 = 3;
```

```
selection 2 = input (prompt)
    if selection 2 = 1
        R = el length Fine;
    elseif selection 2 = 2
        R = el length Fine*2;
    elseif selection 2 == 3
        R = el length Fine *2.5;
    end
end
%Input min and max values for fzero function.
lambda\_min\!=\!10^-8;
lambda \max=10^-1;
nen = 4; %Number of nodes per element
%Extract x and y coordinates
[exC, eyC] = coordxtr(edof, coord, dof, nen);
ec = [exC, eyC];
                         %Number of elements
nelm = max(edof(:,1));
                         %Number of degree of freedom, displacement part
ndof = max(max(dof));
nnod = ndof/2;
                         %Number of nodes
%Material parameters
E = 200e9;
                          %Young's modulus
rho max = 1;
                           Maximum density factor
rho min = 10^-5;
                           %Minimum density factor
thickness = 20*10^-3;
                           %Thickness
V box = 300*100*(10^-6)*thickness;
V_{max} = 0.4 * V_{box};
                           %Maximum allowed volume
v = 0.3;
                           %Poissons tal
eq = 1;
%Initiate
x e = ones(nelm,1)*10^-3; %Introduce element Area vector, with all Area being
a e = zeros(nelm, 1);
                           %Introduce Area vector
T = sparse(nnod, nelm);
                           %Introduce T matrix, [nnod x nelm]
M = sparse(nnod, nnod);
                           %Introduce Mass matrix
K dens = sparse(nnod, nnod); %Introduce constant stiffness matrix, Ke [4x4]
K rho = sparse (ndof, ndof); %Introduce stiffness matrix, Ke [8x8]
dgdrhotilde = sparse(nnod,1);
                           %Penalization factor
p = 3;
ptype = 1; %Plane stress
ep = [ptype thickness 2];
```

```
%Hooke matrix
D0 = hooke(ptype, E, v);
%Parameters for flw2i4e
ep2 = [thickness 2];
D2 = [1 \ 0;
    0 \ 1;
for n=1:nelm
    a\ e(\,n\,,:\,)\ =\ ElementArea\,(\ ec\,(\,n\,,:\,)\ )\,; \qquad \% Area\ per\ element
    %Compute T matrix
     [Ke, Te] = flw2i4e(exC(n,:), eyC(n,:), ep2, D2, eq);
    %Assemble element matrices to global matrices:
     indx = enod(n, 2 : end);
    T(indx, n) = T(indx, n) + Te;
     indx = enod(n, 2 : end);
     K \operatorname{dens}(\operatorname{indx}, \operatorname{indx}) = K \operatorname{dens}(\operatorname{indx}, \operatorname{indx}) + Ke.*R^2;
    %Compute Mass matrix
    Me = flw2i4m(exC(n,:), eyC(n,:), thickness);
    %Assemble element matrices to global matrices:
    indx = enod(n, 2: end);
    M(indx, indx) = M(indx, indx) + Me;
end
\% Compute \ constant \ drhoTilde\_drho \ matrix\,.
drhoTilde_drho = (K_dens + M) \ T;
%Initate Tolerance
Tol = 10^-4;
%Initate quantaties
iter = 0;
Nres = 0;
alpha = 2;
%Optimization loop
while Nres > Tol | | iter == 0
     iter = iter + 1;
     disp (['Iteration', num2str(iter),'-----'])
    %Reset global matrices
    K rho = sparse(ndof, ndof);
     dgdrhotilde = sparse(nnod, 1);
    %Compute filtered density
```

```
x e tilde = drhoTilde drho*x e;
%Extract the densities at the nodal points.
ed_rho=extract(enod,x_e_tilde);
%Element loop to assemble global stiffness matrix
for i=1:nelm
    Ke rho=plani4e rho(exC(i,:), eyC(i,:), ep, D0, ed rho(i,:), p);
    %Assemble element matrices to global matrices:
    indx = edof(i, 2: end);
    \label{eq:K_rho} {\rm (indx\,,indx\,)} \; = \; {\rm K\_rho}({\rm indx\,,indx}) + {\rm Ke} \;\; {\rm rho}\,;
end
%Obtain displacement
a = solveq(K rho,F,bc);
ed = extract(edof,a);
for j=1:nelm
    %Compute dgdrhotilde
    dgdrhotilde = getdgdrhotilde = el(exC(j,:), eyC(j,:), ep, D0, ed(j,:)...
         , ed_{rho(j,:),p);
    %Assemble element matrices to global matrices:
    indx = enod(j, 2: end);
    dgdrhotilde(indx,1) = dgdrhotilde(indx,1)+dgdrhotilde e;
end
%Compute global dg_drho
dg_drho = dgdrhotilde '* drhoTilde_drho;
b = (1/alpha).*dg drho'.*(x e.^(1+alpha));
%Solve the dual function
lambdastar = fzero(@(lambda) dphidlambda_b...
    (lambda, rho\_max, rho\_min, alpha, b, a\_e, V\_max, nelm, thickness)...
     , [lambda min lambda max]);
for z=1:nelm
    %Find new rho value per element based on new lambda from the dual
    %function
    xe_star(z,:) = getx_e_star(lambdastar, rho_max, rho_min, alpha, b(z,:), a_e(z,:)
end
%Compute the change between current and previous iteration
%regarding the updated density.
Nres = norm(xe star - x e);
x_e = xe_star;
```

```
%Plot vectors
    Iter_plot(:, iter) = iter;
    g0(:,iter) = F'*a;
    g1(:,iter) = x_e'*a_e*thickness-V_max;
    fprintf ('Res %e\n', Nres)
    if iter == 200
         fprintf('Force break optimization loop at iteration: %i \n', iter)
        break
    end
end
%%
ed_el = extract(enod, x_e_tilde);
figure
fill(exC',eyC',ed_el')
title ('Helmholtz density PDE filter')
xlabel('x, [m]')
ylabel('y, [m]')
colormap ('Jet')
%%
figure
plot(Iter plot, g0, 'LineWidth', 2)
grid on
xlabel('Iterationer')
ylabel('g_0')
title ('Compliance history')
figure
plot (Iter_plot, g1, 'LineWidth', 2)
xlabel('Iterationer')
ylabel('g_1')
grid on
title('Weight history')
Extra
myeldisp2
function [magnfac] = myeldisp2 (ex, ey, ed, plotpar, magnfac, Area, fac)
%
%
% PURPOSE
    Draw the deformed 2D mesh for a number of elements of
%
    the same type. Supported elements are:
%
%
             1) -> bar element
                                              2) \rightarrow \text{beam el}.
%
             3) -> triangular 3 node el.
                                            4) -> quadrilateral 4 node el.
%
             5) -> 8-node isopar. element
%
  INPUT
%
                               number of element nodes
     ex, ey : \dots nen :
%
                               number of elements
                        nel:
```

```
%
      ed:
              element displacement matrix
%
%
      plotpar=[ linetype , linecolor , nodemark]
%
%
                linetype=1 -> solid
                                         linecolor=1 -> white
%
                         2 \rightarrow dashed
                                                2 \rightarrow green
%
                         3 \rightarrow dotted
                                                    3 -> yellow
%
                                                    4 \rightarrow red
%
                nodemark=1 -> circle
%
                         2 \rightarrow star
%
                         0 \rightarrow \text{no mark}
%
%
      magnfac: magnification factor for displacements
%
%
     Rem. Default if magnfac and plotpar is left out is auto magnification
           and dashed white lines with circles at nodes \rightarrow plotpar=[2 1 1]
%
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                   Department of Solid Mechanics.
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%-
   if ((nargin == 3) | (nargin == 4) | (nargin == 5))
       error ('??? Wrong number of input arguments!')
%
% end
nargin = 5;
 a=size(ex); b=size(ey);
 if (a-b) = [0 \ 0]
    nen=a(2);
 else
    error ('??? Check size of coordinate input arguments!')
 end
 c=size(ed);
 if (c(1) = a(1))
    error ('??? Check size of displacement input arguments!')
 end
 ned=c(2);
 dxmax = max(max(ex') - min(ex')); dymax = max(max(ey') - min(ey'));
 dlmax=max(dxmax, dymax);
 edmax=max(max(abs(ed)));
 krel=0.1;
 if nargin == 3;
       plotpar=[2 1 1]; magnfac=krel*dlmax/edmax;
 elseif nargin==4
       magnfac=krel*dlmax/edmax;
 end
 [s1, s2] = pltstyle(plotpar);
 k=magnfac;
```

```
% ****** Bar or Beam elements *******
    if nen==2
       if ned==4 % ----- Bar elements -----
           x = (ex + k * ed(:, [1 \ 3]));
           y = (ey + k * ed (:, [2 \ 4]));
           xc=x;
           yc=y;
        elseif ned==6 % ----- Beam elements -----
           x = (ex + k * ed (:, [1 \ 4]));
           y=(ey+k*ed(:,[2 5]))';
           [exc, eyc] = beam2crd(ex, ey, ed, k);
           xc=exc;
           vc=evc ';
       end
% ****** 2D triangular elements *******
    elseif nen==3
       x = (ex + k * ed (:, [1 \ 3 \ 5]));
       y=(ey+k*ed(:,[2\ 4\ 6]))';
       xc = [x; x(1,:)];
       yc = [y; y(1,:)];
% ****** 2D quadrilateral elements ******
    elseif nen==4
       x = (ex + k * ed (:, [1 \ 3 \ 5 \ 7]));
       y=(ey+k*ed(:,[2\ 4\ 6\ 8]))';
       xc = [x; x(1,:)];
       yc = [y; y(1,:)];
\% ******* 2D 8-node quadratic elements *******
    elseif nen==8
       x=(ex+k*ed(:,[1 \ 3 \ 5 \ 7 \ 9 \ 11 \ 13 \ 15]));
       y=(ey+k*ed(:,[2\ 4\ 6\ 8\ 10\ 12\ 14\ 16]));
%
        xc = [x(1); x(5); x(2); x(6); x(3); x(7); x(4); x(8); x(1)];
%
        yc = [y(1); y(5); y(2); y(6); y(3); y(7); y(4); y(8); y(1)];
\% isoparametric elements
    t = -1;
    n=0;
    for s = -1:0.4:1
      n=n+1;
      N1 = -1/4*(1-t)*(1-s)*(1+t+s);
      N2=-1/4*(1+t)*(1-s)*(1-t+s);
      N3=-1/4*(1+t)*(1+s)*(1-t-s);
      N4 = -1/4*(1-t)*(1+s)*(1+t-s);
      N5=1/2*(1-t*t)*(1-s);
      N6=1/2*(1+t)*(1-s*s);
      N7=1/2*(1-t*t)*(1+s);
      N8=1/2*(1-t)*(1-s*s);
      N=[N1, N2, N3, N4, N5, N6, N7, N8];
      x1(n,:)=N*x';
      y1(n,:)=N*y';
    end;
    xc = [xc \ x1];
    yc = [yc \ y1];
    clear x1
    clear y1
%
    s=1;
    n=0;
```

```
for t = -1:0.4:1
      n=n+1;
       N1 = -1/4*(1-t)*(1-s)*(1+t+s);
       N2 = -1/4*(1+t)*(1-s)*(1-t+s);
       N3=-1/4*(1+t)*(1+s)*(1-t-s);
       N4 = -1/4*(1-t)*(1+s)*(1+t-s);
      N5=1/2*(1-t*t)*(1-s);
      N6=1/2*(1+t)*(1-s*s);
      N7=1/2*(1-t*t)*(1+s);
      N8=1/2*(1-t)*(1-s*s);
      N=[N1, N2, N3, N4, N5, N6, N7, N8];
       x1(n,:)=N*x';
       y1(n,:)=N*y';
    end;
    xc = [xc \ x1];
    yc = [yc \ y1];
    clear x1
    clear y1
%
    t=1;
    n=0:
    for s=1:-0.4:-1
       n=n+1;
       N1 = -1/4*(1-t)*(1-s)*(1+t+s);
       N2=-1/4*(1+t)*(1-s)*(1-t+s);
      N3 = -1/4*(1+t)*(1+s)*(1-t-s);
      N4 = -1/4*(1-t)*(1+s)*(1+t-s);
      N5=1/2*(1-t*t)*(1-s);
      N6=1/2*(1+t)*(1-s*s);
      N7=1/2*(1-t*t)*(1+s);
      N8=1/2*(1-t)*(1-s*s);
      N=[N1, N2, N3, N4, N5, N6, N7, N8];
       x1(n,:)=N*x';
       y1(n,:)=N*y';
    end;
    xc = [xc \ x1];
    yc = [yc \ y1];
    clear x1
    clear y1
%
    s = -1;
    n=0;
    for t=1:-0.4:-1
       n=n+1;
       N1 = -1/4*(1-t)*(1-s)*(1+t+s);
       N2=-1/4*(1+t)*(1-s)*(1-t+s);
      N3 = -1/4*(1+t)*(1+s)*(1-t-s);
      N4 {=}\, {-}1/4 {*} (1 {-}\, t\,) {*} (1 {+}\, s\,) {*} (1 {+}\, t\, {-}\, s\,)\,;
       N5=1/2*(1-t*t)*(1-s);
      N6=1/2*(1+t)*(1-s*s);
      N7 = 1/2*(1-t*t)*(1+s);
      N8=1/2*(1-t)*(1-s*s);
      N=[N1, N2, N3, N4, N5, N6, N7, N8];
       x1(n,:)=N*x';
       y1(n,:)=N*y';
    end;
    xc = [xc \ x1];
    yc = [yc y1];
```

```
clear x1
    clear v1
%
error ('Sorry, this element is currently not supported!')
    end
% ******* plot commands **********
    axis ('equal')
    hold on
    for el=1: size(ex,1)
        w = Area(el)*fac;
        plot(xc(:,el),yc(:,el),s1,'linewidth',w)
    end
    if s2 = ',
     plot(x,y,s2)
    end
    hold off
           -----end-----
function myeldraw2 (ex, ey, plotpar, Area, fac)
% PURPOSE
    Draw the undeformed 2D mesh for a number of elements of
    the same type. Supported elements are:
%
%
    1) -> bar element
                                  2) \rightarrow \text{beam el}.
    3) -> triangular 3 node el. 2) -> deam el. 3) -> quadrilateral 4 node el.
%
    5) -> 8-node isopar. elemen
%
% INPUT
%
     ex, ey:.... nen: number of element nodes
%
                      nel: number of elements
%
     plotpar=[ linetype , linecolor , nodemark]
%
%
              linetype=1 -> solid linecolor=1 -> white
%
                       2 \rightarrow dashed 2 \rightarrow green
%
                       3 \rightarrow dotted
                                              3 -> yellow
%
                                               4 \rightarrow red
%
              nodemark=1 -> circle
%
%
                       2 \rightarrow star
%
                       0 \rightarrow \text{no mark}
%
%
     elnum=edof(:,1); i.e. the first column in the topology matrix
%
%
     Rem. Default is solid white lines with circles at nodes.
%
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                Department of Solid Mechanics.
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%-
\% if ((nargin == 2) | (nargin == 3) | (nargin == 4))
      disp('??? Wrong number of input arguments!')
```

```
%
      break
%
      error ('??? Wrong number of input arguments!')
%
  end
nargin = 3;
 a=size(ex); b=size(ey);
 if (a-b) = [0 \ 0]
     nel=a(1); nen=a(2);
 else
    error ('??? Check size of coordinate input arguments!')
   %disp('??? Check size of coordinate input arguments!')
  \%break
 end
 if \quad nargin == 2;
      plotpar = [1 \ 1 \ 1];
 [s1,s2]=pltstyle(plotpar);
% *******************
% ******* plot coordinates ************
x0=sum(ex')/nen; y0=sum(ey')/nen;
% ****** Bar or Beam elements ******
 if nen==2
   x=ex;
    y=ey ';
    xc=x ; yc=y;
% ****** 2D triangular elements *******
 elseif nen==3
   x=ex;
    y=ev ';
    xc = [x ; x(1,:)]; yc = [y ; y(1,:)];
% ****** 2D quadrilateral elements ******
 elseif nen==4
   x=ex;
    y=ey ';
    xc = [x ; x(1,:)]; yc = [y ; y(1,:)];
% ****** 2D 8 node quadratic elements ******
 elseif nen== 8
   x=ex;
  %xc = [x(1); x(5); x(2); x(6); x(3); x(7); x(4); x(8); x(1)];
  \text{%yc} = [y(1); y(5); y(2); y(6); y(3); y(7); y(4); y(8); y(1)];
xc = [];
yc = [];
%
% isoparametric elements
    t = -1:
   n=0;
    for s = -1:0.4:1
     n=n+1;
     N1 = -1/4*(1-t)*(1-s)*(1+t+s);
     N2=-1/4*(1+t)*(1-s)*(1-t+s);
     N3 = -1/4*(1+t)*(1+s)*(1-t-s);
     N4 = -1/4*(1-t)*(1+s)*(1+t-s);
```

```
N5=1/2*(1-t*t)*(1-s);
       N6=1/2*(1+t)*(1-s*s);
       N7=1/2*(1-t*t)*(1+s);
       N8=1/2*(1-t)*(1-s*s);
       N=[ N1, N2, N3, N4, N5, N6, N7, N8 ];
       x1(n,:)=N*x';
       y1(n,:)=N*y';
     end;
     xc = [xc \ x1];
     yc = [yc y1];
     clear x1
     clear y1
%
     s=1;
     n=0;
     for t = -1:0.4:1
       n=n+1;
       N1\!=\!-1/4\!*\!(1\!-\!t\,)\!*\!(1\!-\!s\,)\!*\!(1\!+\!t\!+\!s\,)\,;
       N2 {=} {-}1/4 {*} (1 {+}\, t\,) {*} (1 {-}\, s\,) {*} (1 {-}\, t {+} s\,)\,;
       N3=-1/4*(1+t)*(1+s)*(1-t-s);
       N4 = -1/4*(1-t)*(1+s)*(1+t-s);
       N5=1/2*(1-t*t)*(1-s);
       N6=1/2*(1+t)*(1-s*s);
       N7=1/2*(1-t*t)*(1+s);
       N8=1/2*(1-t)*(1-s*s);
       N=[N1, N2, N3, N4, N5, N6, N7, N8];
       x1(n,:)=N*x';
       y1(n,:)=N*y';
     end;
     xc = [xc \ x1];
     yc = [yc \ y1];
     clear x1
     clear y1
%
     t=1;
     n=0;
     for s=1:-0.4:-1
       n=n+1;
       N1 = -1/4*(1-t)*(1-s)*(1+t+s);
       N2=-1/4*(1+t)*(1-s)*(1-t+s);
       N3 = -1/4*(1+t)*(1+s)*(1-t-s);
       N4 = -1/4*(1-t)*(1+s)*(1+t-s);
       N5=1/2*(1-t*t)*(1-s);
       N6=1/2*(1+t)*(1-s*s);
       N7=1/2*(1-t*t)*(1+s);
       N8=1/2*(1-t)*(1-s*s);
       N=[N1, N2, N3, N4, N5, N6, N7, N8];
       x1(n,:)=N*x';
       y1(n,:)=N*y';
     end;
     xc = [xc \ x1];
     yc = [yc y1];
     clear x1
     clear y1
%
     s = -1;
     n=0;
     for t=1:-0.4:-1
```

```
n=n+1;
     N1 = -1/4*(1-t)*(1-s)*(1+t+s);
     N2=-1/4*(1+t)*(1-s)*(1-t+s);
     N3 = -1/4*(1+t)*(1+s)*(1-t-s);
     N4 = -1/4*(1-t)*(1+s)*(1+t-s);
     N5=1/2*(1-t*t)*(1-s);
     N6=1/2*(1+t)*(1-s*s);
     N7 = 1/2*(1-t*t)*(1+s);
     N8=1/2*(1-t)*(1-s*s);
     N=[N1, N2, N3, N4, N5, N6, N7, N8];
     x1(n,:)=N*x';
     y1(n,:)=N*y';
   end;
   xc = [xc \ x1];
   yc = [yc \ y1];
   clear x1
   clear v1
%******************
 else
   error ('!!!! Sorry, this element is currently not supported!')
   %disp('!!!! Sorry, this element is currently not supported!')
   %break
 end
% ******** plot commands **********
axis ('equal')
 hold on
 plot(xc,yc,s1)
 for el=1:nel
    w = Area(el)*fac;
    plot(xc(:, el), yc(:, el), s1, 'linewidth', w)
end
 if s2 = ',
  plot(x,y,s2)
end
 if nargin==4
   for i=1:nel
       h=text(x0(i),y0(i),int2str(elnum(i)));
       set(h, 'fontsize', 8);
   end
end
 xlabel('x'); ylabel('y');
 hold off
\%------end--------
plani4e-rho
function Ke=plani4e_rho(ex, ey, ep, D, ed_rho, q)
%-----
% PURPOSE
% Calculate the stiffness matrix for a 4 node isoparametric
\% element in plane strain or plane stress with a distributed
\% density field.
\% INPUT: ex = [x1 x2 x3 x4] element coordinates
        ey = [y1 \ y2 \ y3 \ y4]
```

```
%
%
         ep = [ptype t ir]
                           element property
%
                              ptype: analysis type
%
                              ir: integration rule
%
                              t : thickness
%
%
         D
                             constitutive matrix
%
%
                            bx: body force in x direction
         eq = [bx; by]
%
                            by: body force in y direction
%
%
         ed rho = [rho1, rho2, rho3, rho4] : The densities at the
%
                                              nodal points of the element.
% OUTPUT: Ke: element stiffness matrix (8 x 8)
         fe : equivalent nodal forces (8 x 1)
\% LAST MODIFIED: M Ristinmaa 1995–10–25
                Eric Borgqvist 2014-03-04
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                Department of Solid Mechanics.
%
                Lund Institute of Technology
%-----
  ptype=ep(1); t=ep(2); ir=ep(3); ngp=ir*ir;
%----- gauss points ------
  i\,f\quad i\,r\!=\!\!=\!1
    g1 = 0.0; w1 = 2.0;
    gp = [g1 g1 ]; w = [w1 w1];
  elseif ir == 2
    g1 = 0.577350269189626; w1 = 1;
    gp(:,1) = [-g1; g1; -g1; g1]; gp(:,2) = [-g1; -g1; g1; g1];
   w(:,1) = [w1; w1; w1; w1]; w(:,2) = [w1; w1; w1; w1; w1];
  elseif ir == 3
    g1 = 0.774596699241483; g2 = 0.;
    gp(:,1) = [-g1;-g2; g1;-g1; g2; g1;-g1; g2; g1];
   gp(:,2) = [-g1;-g1;-g1; g2; g2; g2; g1; g1; g1; g1];
   w(:,1) = [w1; w2; w1; w1; w2; w1; w1; w2; w1];
   w(:,2) = [w1; w1; w1; w2; w2; w2; w1; w1; w1];
    disp ('Used number of integration points not implemented');
    return
  end
  wp=w(:,1).*w(:,2);
  xsi=gp(:,1); eta=gp(:,2); r2=ngp*2;
%----- shape functions ------
 N(:,1) = (1-xsi).*(1-eta)/4; N(:,2) = (1+xsi).*(1-eta)/4;
 N(:,3) = (1 + xsi).*(1 + eta)/4; N(:,4) = (1 - xsi).*(1 + eta)/4;
  dNr(1:2:r2,1) = -(1-eta)/4;
                              dNr(1:2:r2,2) = (1-eta)/4;
                              dNr(1:2:r2,4) = -(1+eta)/4;
  dNr(1:2:r2,3) = (1+eta)/4;
  dNr(2:2:r2+1,1) = -(1-xsi)/4; dNr(2:2:r2+1,2) = -(1+xsi)/4;
  dNr(2:2:r2+1,3) = (1+xsi)/4; dNr(2:2:r2+1,4) = (1-xsi)/4;
  Ke=zeros(8,8);
  fe=zeros(8,1);
  JT=dNr*[ex;ey]';
```

```
\%----- plane stress -----
if ptype==1
  colD = size(D, 2);
  if colD > 3
    Cm=inv(D);
    Dm=inv(Cm([1 \ 2 \ 4],[1 \ 2 \ 4]));
    Dm=D:
  end
  for i=1:ngp
       indx = [2*i-1; 2*i];
       detJ=det(JT(indx,:));
       if \det J < 10*eps
         disp ('Jacobideterminant equal or less than zero!')
       JTinv=inv(JT(indx,:));
      dNx=JTinv*dNr(indx,:);
      B(1,1:2:8-1) = dNx(1,:);
      B(2,2:2:8) = dNx(2,:);
      B(3,1:2:8-1) = dNx(2,:);
      B(3,2:2:8) = dNx(1,:);
      N2\,(\,1\,\,,1\!:\!2\!:\!8\,-1\,)\!=\!N(\,i\,\,,:\,)\;;
      N2(2,2:2:8) = N(i,:);
      N1 = N(i, :);
       rhogp = N1*ed rho';
      Ke=Ke+rhogp^q*B'*Dm*B*detJ*wp(i)*t;
      ---- plane strain ------
elseif ptype==2
  colD = size(D, 2);
  if colD > 3
    Dm = D([1 \ 2 \ 4], [1 \ 2 \ 4]);
    Dm=D;
  end
  for i=1:ngp
       indx\!=\!\!\left[\begin{array}{cc}2\!*i-\!1;&2\!*i\end{array}\right];
       \det J = \det (JT(indx,:));
       if \det J < 10*eps
         disp ('Jacobideterminant equal or less than zero!')
       JTinv=inv(JT(indx,:));
      dNx=JTinv*dNr(indx,:);
      B(1,1:2:8-1) = dNx(1,:);
      B(2,2:2:8) = dNx(2,:);
      B(3,1:2:8-1) = dNx(2,:);
      B(3,2:2:8) = dNx(1,:);
      N2(1,1:2:8-1)=N(i,:);
      N2(2,2:2:8) = N(i,:);
      N1 = N(i, :);
```

```
rhogp = N1*ed rho';
     Ke=Ke+rhogp^q*B*Dm*B*detJ*wp(i)*t;
  end
else
   error ('Error ! Check first argument, ptype=1 or 2 allowed')
   return
end
\%-----end------
getdgdrhotilde-el
function fe=getdgdrhotilde el(ex,ey,ep,D,ed u,ed rho,q)
% PURPOSE
    Computes the sensitivity of the compliance with respect to a
    continues density field (rhotilde)
\% INPUT: ex = [x1 x2 x3 x4] element coordinates
%
         ey = [y1 \ y2 \ y3 \ y4]
%
%
         ep =[ptype t ir]
                            element property
%
                              ptype: analysis type
%
                              ir: integration rule
%
                              t: thickness
%
%
         D
                             constitutive matrix
%
%
                       Element displacement vector = [u1, u2, ... u8];
     ed u:
%
%
     ed rho:
                       Element density vector = [rho1, rho2, rho3, rho4]
%
%
                       Exponent in density field.
     q:
%
%
% OUTPUT: fe : Nodal sensitivites (4 x 1)
% LAST MODIFIED: M Ristinmaa 1995-10-25
                Eric Borgqvist 2014-03-04
% Copyright (c) Division of Structural Mechanics and
                Department of Solid Mechanics.
%
          Lund Institute of Technology
%-----
  ptype=ep(1); t=ep(2); ir=ep(3); ngp=ir*ir;
%----- gauss points -----
  if ir ==1
    g1 = 0.0; w1 = 2.0;
    gp = [g1 g1]; w = [w1 w1];
  elseif ir == 2
    g1 = 0.577350269189626; w1 = 1;
    gp\,(:\,,1)\!=\![\,-\,g1\,;\ g1\,;\,-\,g1\,;\ g1\,]\,;\quad gp\,(:\,,2)\!=\![\,-\,g1\,;\,-\,g1\,;\ g1\,]\,;
   w(:,1) = [w1; w1; w1; w1; w1]; w(:,2) = [w1; w1; w1; w1; w1];
  elseif ir==3
    g1 = 0.774596699241483; g2 = 0.;
    gp(:,1) = [-g1;-g2; g1;-g1; g2; g1;-g1; g2; g1];
   gp(:,2) = [-g1;-g1;-g1; g2; g2; g2; g1; g1; g1; g1];
   w(:,1) = [ w1; w2; w1; w1; w2; w1; w1; w2; w1];
   w(:,2) = [w1; w1; w1; w2; w2; w2; w1; w1; w1];
```

```
else
    disp ('Used number of integration points not implemented');
    return
  end
  wp=w(:,1).*w(:,2);
  xsi=gp(:,1); eta=gp(:,2); r2=ngp*2;
%----- shape functions -----
 N(:,1) = (1-xsi).*(1-eta)/4; N(:,2) = (1+xsi).*(1-eta)/4;
 N(:,3)=(1+xsi).*(1+eta)/4; N(:,4)=(1-xsi).*(1+eta)/4;
 dNr(1:2:r2,1) = -(1-eta)/4;
                               dNr(1:2:r2,2) = (1-eta)/4;
 dNr(1:2:r2,3) = (1+eta)/4;
                               dNr(1:2:r2,4) = -(1+eta)/4;
                               dNr(2:2:r2+1,2) = -(1+xsi)/4;
 dNr(2:2:r2+1,1) = -(1-xsi)/4;
 dNr(2:2:r2+1,3) = (1+xsi)/4;
                               dNr(2:2:r2+1,4) = (1-xsi)/4;
  fe=zeros(4,1);
 JT=dNr*[ex;ey]';
%----- plane stress ------
if ptype==1
  colD = size(D, 2);
  if colD > 3
   Cm=inv(D);
   Dm=inv(Cm([1 \ 2 \ 4],[1 \ 2 \ 4]));
  else
   Dm=D;
  end
  for i=1:ngp
      indx = [2*i-1; 2*i];
      detJ=det(JT(indx,:));
      if \det J < 10 * eps
        disp ('Jacobideterminant equal or less than zero!')
      JTinv=inv(JT(indx,:));
     dNx=JTinv*dNr(indx,:);
     B(1,1:2:8-1) = dNx(1,:);
     B(2,2:2:8) = dNx(2,:);
     B(3,1:2:8-1) = dNx(2,:);
     B(3,2:2:8) = dNx(1,:);
     N2(1,1:2:8-1)=N(i,:);
     N2(2,2:2:8) = N(i,:);
     N1 = N(i, :);
      rhogp = N1*ed rho';
      fe=fe+N1'*q*rhogp^(q-1)*(ed u*B'*Dm*B*ed u')*detJ*wp(i)*t;
 end
%----- plane strain -----
elseif ptype==2
  colD = size(D, 2);
  if colD > 3
   Dm = D([1 \ 2 \ 4], [1 \ 2 \ 4]);
  else
   Dm=D;
```

```
end
 for i=1:ngp
     indx = [2*i-1; 2*i];
     \det J = \det (JT(indx,:));
     if \det J < 10*eps
      disp ('Jacobideterminant equal or less than zero!')
     JTinv=inv(JT(indx,:));
    dNx=JTinv*dNr(indx,:);
    B(1,1:2:8-1) = dNx(1,:);
    B(2,2:2:8) = dNx(2,:);
    B(3,1:2:8-1) = dNx(2,:);
    B(3,2:2:8) = dNx(1,:);
    N2(1,1:2:8-1)=N(i,:);
    N2(2,2:2:8) = N(i,:);
    N1 = N(i, :);
     rhogp = N1*ed rho';
     fe=fe+N1'*q*rhogp^(q-1)*(ed u*B'*Dm*B*ed u')*detJ*wp(i)*t;
 \quad \text{end} \quad
  error ('Error ! Check first argument, ptype=1 or 2 allowed')
  return
%-----end------end------
flw2i4m
function Me = flw2i4m(ex, ey, x)
\% Me=flw2i4m (ex, ey, x)
%_____
% PURPOSE
\% Compute the quantity: Ce=x*int(N^T*N)dA with full integration, i.e. 3x3
% gauss points
% INPUT: ex, ey; Element coordinates
%
%
\% OUTPUT: Theta : Matix 4 x 4
gp1 = -0.774596669241483;
gp2\ =\ 0\,;
gp3 = -gp1;
gp_m = [gp1 gp1;
   gp2 gp1;
   gp3 gp1;
   gp1 gp2;
   gp2 gp2;
   gp3 gp2;
   gp1 gp3;
   gp2 gp3;
   gp3 gp3;];
```

```
Me = zeros(4);
for gp=1:9
     xsi = gp_m(gp, 1);
     eta = gp_m(gp, 2);
     dN1dxsi = 0.25*(eta-1);
     dN1deta = 0.25*(xsi-1);
     dN2dxsi = -0.25*(eta - 1);
     dN2deta = -0.25*(xsi+1);
     dN3dxsi = 0.25*(eta+1);
     dN3deta = 0.25*(xsi+1);
     dN4dxsi = -0.25*(eta+1);
     dN4deta = -0.25*(xsi-1);
     dNdxsi = [dN1dxsi, dN2dxsi, dN3dxsi, dN4dxsi];
     dNdeta = [dN1deta, dN2deta, dN3deta, dN4deta];
     ax = [ex(1); ex(2); ex(3); ex(4)];
     ay = [ey(1); ey(2); ey(3); ey(4)];
    dxdxsi = dNdxsi*ax; dxdeta= dNdeta*ax;
    dydxsi = dNdxsi*ay; dydeta = dNdeta*ay;
    J = [dxdxsi, dxdeta; dydxsi, dydeta];
    det J = det(J); %ska vara lika med Areaan av elementen *4
  N1 = 1/4*(xsi-1)*(eta-1);
   N2 = -1/4*(xsi+1)*(eta-1);
  N3 = 1/4*(xsi+1)*(eta+1);
  N4 = -1/4*(xsi-1)*(eta+1);
  N = [N1, N2, N3, N4];
  Me = Me + N'*N*detJ*x*w1(gp)*w2(gp);
```

end