

# Home Work #1

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## 1 Question 1

### 1.1 a

$$\mathbf{h} = \mathbf{r} \times \mathbf{v} = \begin{vmatrix} i & j & k \\ 0 & 2 & 0 \\ \frac{\sqrt{2}}{2} & \frac{\sqrt{2}}{2} & 0 \end{vmatrix} = [0 \quad 0 \quad -\sqrt{2}]$$
$$\mathbf{C} = \dot{\mathbf{r}} \times \mathbf{h} - \mu \frac{\mathbf{r}}{r}$$

In Astronomical/Canonical Units:  $\mu = 1$

$$\frac{\mathbf{C}}{\mu} = \mathbf{e} \rightarrow \mathbf{e} = \frac{\mathbf{C}}{\mu} = [-1 \quad -1 \quad 0]$$
$$\mathbf{h} \cdot \mathbf{e} = [0 \quad 0 \quad -\sqrt{2}] \cdot [-1 \quad -1 \quad 0] = 0$$

### 1.2 b, c

$$r = \frac{P}{1 + e \cos(\theta)} \xrightarrow{P = \frac{h^2}{\mu}} r = \frac{h^2}{\mu} \frac{1}{1 + e \cos(\theta)} \rightarrow \theta = \arccos \left( \left( \frac{h^2}{\mu r} - 1 \right) / e \right)$$

Beacuse  $\mathbf{r} \cdot \mathbf{v} > 0$ ,  $\theta$  is in the range  $0 \leq \theta \leq \pi$

$$\rightarrow \theta = \pi/2$$

$$\varepsilon = \frac{v^2}{2} - \frac{\mu}{r} = 0 = \text{constant}$$

### 1.3 d

In  $r = 32DU$ ,  $\varepsilon = 0$  and  $\mathbf{h}$  = constant, then  $v$  and  $\theta$  calculated as below:

$$\varepsilon = 0 \rightarrow v = \sqrt{\frac{2\mu}{r}} = 0.25 \text{ DU/TU}$$

$$\theta = \arccos \left( \left( \frac{h^2}{\mu r} - 1 \right) / e \right) = 2.7862_{rad}$$

## 2 Question 4

### 2.1 a

$$h = \sqrt{2\mu} \sqrt{\frac{r_a r_p}{r_a + r_p}}$$
$$v = \frac{h}{r}$$

First orbit (circular):

$$r = 6570$$

For first circular orbit  $r_a = r_p$ .

$$h = 51174 \rightarrow v = 7.7891_{km/s}$$

Second orbit (elliptical):

$$r_p = 6570, \quad r_a = 42160$$

$$h = 67316 \rightarrow v_a = 10.2460_{km/s}, \quad v_p = 1.5967_{km/s}$$

Third orbit (circular):

$$h = 129634 \rightarrow v = 3.0748_{km/s}$$

Total delta change in velocity:

$$\Delta v_{\text{total}} = \Delta v_1 + \Delta v_2 = |10.2460 - 7.7891| + |1.5967 - 3.0748| = 3.9350_{km/s}$$

### 2.2 b

Time is half of the period.

$$\tau = 2\pi \sqrt{\frac{a^3}{\mu}} = 37850_{\text{sec}} \rightarrow t = 18925_{\text{sec}}$$

## Contents

<b>1</b>	<b>Question 1</b>	<b>1</b>
1.1	a . . . . .	1
1.2	b, c . . . . .	1
1.3	d . . . . .	1
<b>2</b>	<b>Question 4</b>	<b>2</b>
2.1	a . . . . .	2
2.2	b . . . . .	2

## List of Figures

## List of Tables