

HI MY FELLOW VIEWERS,
I WRITE π AS A
LOWER-CASE X. DON'T
BE CONFUSED

THE REGULATION (OR SET-POINT STABILIZATION) PROBLEM

Consider the system (all functions are assumed sufficiently smooth)

$$\begin{cases} \dot{x} = f(x, u) \\ y_r = h_r(x) \\ y_m = h_m(x) \end{cases}$$

state space of state space of input
 $x(t) \in \mathbb{R}^{n_x}$ (STATE), $u(t) \in \mathbb{R}^{n_u}$ (control)
 $y_r(t) \in \mathbb{R}^{n_r}$ (REGULATED OUTPUT), $y_m(t) \in \mathbb{R}^{n_m}$ (MEASURED OUTPUT)

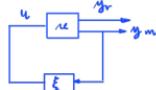
variable on which we have performance specifications

Check Khalil's book

Let $y_r^* \in \mathbb{R}^{n_r}$ be a desired value (SET-POINT) for $y_r(t)$

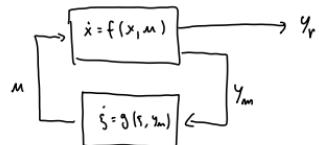
PROBLEM: design a controller of the form (still autonomous)

$$\begin{cases} \dot{\xi} = g(\xi, y_m) \\ M = \gamma(\xi, y_m) \end{cases} \quad \xi(t) \in \mathbb{R}^{n_\xi} \quad (\xi = \text{CONTROLLER'S STATE})$$



such that the closed-loop system

$$\begin{cases} \dot{x} = f(x, \gamma(\xi, h_m(x))) \\ \dot{\xi} = g(\xi, h_m(x)) \\ y_r = h_r(x) \end{cases} \quad \text{(*)} \quad \text{we want to have this to be steady-state}$$



Satisfies the following:

1. There exists an equilibrium point $(x^*, \xi^*) \in \mathbb{R}^{n_x} \times \mathbb{R}^{n_\xi}$ of (*) such that:
 solutions must converge

$$y_r^* = h_r(x^*)$$

\leftarrow The regulated output y_r equals y_r^* at the equilibrium

2. The equilibrium (x^*, ξ^*) is (LOCALLY / GLOBALLY / ...) asymptotically stable

local stability is a local property
 Global stability is only feasible on linear systems

local non-local
 global
 asymptotically stable
 robustness: a small push will not explode to infinity

If this problem is solved, then all solutions of the closed-loop system converge to a steady state (x^*, ξ^*) at which y_r attains its desired value y_r^*

This problem will be our main focus throughout the module

BIBLIOGRAPHY: H. KHALIL, Nonlinear systems (Chap. 12.2)

Solving steps:

1. Using Local Linear Control (locality in Lyapunov).
 2. Using Robust Linear Control via integral action.
 3. Using Non-Local which is Gain Scheduling
 4. Using Non-Local / Global (non-linear control).
Sometimes it is non-robust or fragile
 5. Using Robust Control via "high-gain." (we will not reach this in this course)
-

We want to see the equilibrium conditions in
 (x^*, ξ^*)

$$\begin{cases} 0 = f(x^*, \gamma(\xi^*, h_m(x^*))) \\ 0 = g(\xi^*, h_m(x^*)) \\ y_r^* = h_r(x^*) \end{cases}$$

Call a dummy variable $u^* = \gamma(\xi^*, h_m(x^*))$ and split to two sets of equations.

$$\begin{cases} 0 = f(x^*, u^*) \\ y_r^* = h_r(x^*) \end{cases} \quad \exists x^*, u^* \text{ solvable equations}$$

$$\begin{cases} 0 = g(\xi^*, h_m(x^*)) \\ u^* = \gamma(\xi^*, h_m(x^*)) \end{cases}$$

• NECESSARY CONDITION: If points 1 and 2 are satisfied, then there exist $(x^*, u^*) \in \mathbb{R}^{n_x} \times \mathbb{R}^{n_u}$ such that :

Plant $\left\{ \begin{array}{l} 0 = f(x^*, u^*) \rightarrow \text{starting point} \\ y_r^* = h_r(x^*) \end{array} \right.$

what to define the system
and the point g^* satisfies

controller $\left\{ \begin{array}{l} 0 = g(g^*, h_m(x^*)) \\ M^* = Y(g^*, h_m(x^*)) \end{array} \right.$

what we
should have
to control the system

CAN BE FOUND IN
CHAPTER 12 OF
KHALIL
(SOLVABILITY EQ.)

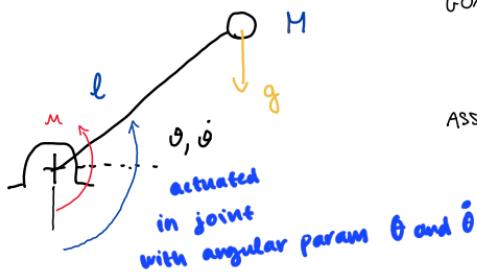
↑
Property of the plant that tells
us when a problem

(REGULATOR EQ.)



Prop. of the controller that tells
us what must be the steady-state
behavior of the controller

EXAMPLE: ACTUATED PENDULUM



Pendulum equations: $(x_1 = \theta, x_2 = \dot{\theta})$

$$\left\{ \begin{array}{l} \dot{x}_1 = x_2 \\ \dot{x}_2 = -\frac{g}{l} \sin x_1 - \frac{f}{l e^2} x_2 + \frac{1}{l e^2} u \end{array} \right. \quad \begin{array}{l} \text{β as the friction} \\ \text{u as the torque input} \end{array}$$

$$\left\{ \begin{array}{l} y_m = x_1 \\ y_r = x_1 \end{array} \right.$$

regulation goal:
drive the pendulum to
 $\theta = 45^\circ$

SET-POINT: $y_r^* = \theta^*$

There must exist (x^*, u^*)

Let us solve the SOLVABILITY EQUATIONS: we look for (x^*, u^*) such that

what we found on the previous page

$$\left\{ \begin{array}{l} 0 = x_2^* \\ 0 = -\frac{g}{l} \sin x_1^* - \frac{B}{Ml^2} x_2^* + \frac{1}{Ml^2} u^* \\ y_r^* = x_1^* \end{array} \right. \quad \theta^* = Ml^2 x_2^* \rightarrow \left\{ \begin{array}{l} x_1^* = \theta^*, \quad x_2^* = 0 \\ M^* = Mgl \sin \theta^* \end{array} \right.$$

$u^* = Mgl \sin \theta$

REGULATOR must give us u^*

the controller does not switch off at steady-state except when $\theta^* = k\pi$ ($k \in \mathbb{N}$)

the REGULATOR Eqs. tell us that only controller solving the problem must provide this control action at steady-state

LOCAL STATE-FEEDBACK SOLUTION TO THE REGULATION PROBLEM

- STATE-FEEDBACK : $y_m(x) = x$ (we measure the full state)
- SOLUTION APPROACH BY LINEAR FEEDBACK + LYAPUNOV INDIRECT METHOD

STEP 1) given y_r^* , we solve the SOLVABILITY Eqs.:

$$\left\{ \begin{array}{l} 0 = f(x^*, u^*) \\ y_r^* = h_r(x^*) \end{array} \right. \xrightarrow{\text{linearized}} \text{What is this?}$$

STEP 2) we LINEARIZE the system $\dot{x} = f(x, u)$ around (x^*, u^*) by defining the matrices

$$A \doteq \frac{\partial f}{\partial x}(x^*, u^*)$$

$$B \doteq \frac{\partial f}{\partial u}(x^*, u^*)$$

We're now dealing with LOCAL-LINEAR CONTROL.
It's also STATE-FEEDBACK.

Local

$$\begin{cases} \dot{x} = f(x, u) \\ y_m = h_m(x) = x \\ y_r = h_r(x) \end{cases}$$

y_r^* is given

① STEP 1: Call the solvability equation

reference eq. point

$\exists (x^*, u^*) :$

$$\begin{cases} 0 = f(x^*, u^*) \\ y_r^* = h_r(x^*) \end{cases}$$

② STEP 2:

We know $f(x, u) = A(x - x^*) + B(u - u^*) + f_{HOT}(x, u)$
and now we linearize it by approximation)

$$A = \left. \frac{\partial f}{\partial x} (x, u) \right|_{\substack{x=x^* \\ u=u^*}} = \frac{\partial f}{\partial x} (x^*, u^*)$$

$$B = \left. \frac{\partial f}{\partial u} (x^*, u^*) \right|_{\substack{x=x^* \\ u=u^*}}$$

If it's stabilizable, proceed to
STEP 3

③ STEP 3:

Choose $u(t) = u^* + K(x(t) - x^*)$. So $A + BK$ is Hurwitz. When we apply all the $u(t)$, it because

$$f(x, u) = (A + BK)(x - x^*) + f_{HOT}(x, u^* + K(x - x^*))$$

$K \in \mathbb{R}^{n \times n}$

$$u(t) = u^* + K(x(t) - x^*)$$

Touching the HOT is a problem!

STEP 3) We choose the controller $n_u \times n_{x^*}$

If A is Hurwitz already
then $u(t) = u^*$ is enough

$$M(t) = u^* + K(x(t) - x^*) \quad (*)$$

(u^* = FEED FORWARD ACTION)

where K is chosen so that $\underbrace{A+BK}$ is Hurwitz

\downarrow
we need (A, B) stabilizable!

RESULT. With the controller $(*)$, the equilibrium x^* is LAS

Hence, there exists an open set $\Omega \subset \mathbb{R}^n$ such that $x^* \in \Omega$, and

$$\forall x(0) \in \Omega, \quad \lim_{t \rightarrow \infty} y_r(t) = y_r^*$$

Remember that : $\dot{\xi} = g(\xi, y_m)$

PROOF.

We can write the plant as

$$\dot{x} = f(x, u^* + K(x - x^*))$$

$$\dot{x} = A\tilde{x} + B\tilde{u} + f_{hor}(x, u)$$

$$y(x) = u^* + K(x - x^*)$$

$$\left(\begin{array}{l} \tilde{x} = x - x^*, \quad \tilde{u} = u - u^* \end{array} \right)$$

$$\begin{aligned} \dot{x} &= f(x, u^* + K(x - x^*)) \\ &= F(x) \text{ where } x = \tilde{x} + x^* \end{aligned}$$

$$\text{So, } F(x^*) = f(x^*, u^*) = 0$$

satisfies

$$\frac{\partial f_{hor}}{\partial x}(x^*, u^*) = \left[\frac{\partial f_{hor}}{\partial x}(x^*, u^*) - A : \frac{\partial f_{hor}}{\partial u}(x^*, u^*) - B \right] = 0$$

\downarrow
we have
to proof that

$$\bar{A} = \frac{\partial F}{\partial x}(x^*) \text{ is Hurwitz}$$

We can rewrite $(*)$ as

$$\tilde{u} = K\tilde{x}$$

Plugging this into $(*)$ leads to the closed-loop system:

$$\dot{x} = (A + BK)\tilde{x} + f_{hor}(x, u^* + K\tilde{x}) \doteq F(x)$$

$$\frac{dF(\nu e)}{d\nu e} = \frac{d}{d\nu e} f(\nu e, u^*) + K(\nu e - \nu e^*)$$

$$= \frac{\partial}{\partial \nu e} f(\nu e, u^*) + K(\nu e - \nu e^*) +$$

$$\frac{\partial}{\partial u} f(\nu e, u^*) + K(\nu e - \nu e^*) \cdot \frac{\partial}{\partial \nu e} (u^* + K(\nu e - \nu e^*))$$

$$\bar{A} = \frac{\partial F(\nu e^*)}{\partial \nu e} = \underbrace{\frac{\partial}{\partial \nu e} f(\nu e^*, u^*)}_{A} + \underbrace{\frac{\partial}{\partial u} f(\nu e^*, u^*) K}_{BK}$$

It is proven that $\bar{A} = A + BK$ is Hurwitz and implies $\nu e = \nu e^*$ is linear asymptotically stable. for the closed-loop system with K .

Addendum : $u = u^* + K(\nu e - \nu e^*)$.

What if $f(\nu e, u) = A(\nu e - \nu e^*) + B(u - u^*) + f_{HOT}(\nu e, u)$ already Hurwitz?

Hence, $u(t) = u^*$, which is an open-loop system.

The linearization of $\dot{x} = F(x)$ around x^* is

$$\begin{aligned}\frac{\partial F}{\partial x}(x^*) &= \left. \frac{\partial}{\partial x} \left[(A + BK)(x - x^*) \right] \right|_{x=x^*} + \left. \frac{\partial}{\partial x} f_{\text{hor}}(x, u^* + Kx^*) \right|_{x=x^*} \\ &= A + BK + \cancel{\frac{\partial}{\partial x} f_{\text{hor}}(x^*, u^*)} = 0 \quad (\text{see above})\end{aligned}$$

Hence:

$$\frac{\partial F}{\partial x}(x^*) = A + BK$$

Since $A + BK$ is Hurwitz by design, the result follows from Lyapunov's indirect theorem.

□

EXAMPLE:

consider the system

$$\begin{cases} \dot{x} = x^3 + u^2 \\ y_r = y_m = x \end{cases} \quad \text{we want to regulate } u$$

CASE 1 : Regulation to $y_r^* = 1$

The SOLVABILITY CONDITIONS read

$$\begin{cases} 0 = x^{*3} + M^{*2} \\ 1 = x^* \end{cases} \Rightarrow M^{*2} = -1 \quad \text{IMPOSSIBLE}$$

We cannot solve case 1.

CASE 2. Regulation to $y_r^* = 0$

the SOLVABILITY CONDITIONS are:

$$\begin{cases} 0 = x^{*3} + M^{*2} \\ 0 = x^* \end{cases} \Rightarrow \begin{cases} x^* = 0 \\ M^* = 0 \end{cases}$$

Case 1: Regulate $y_r^* = 1$

Case 2: Regulate $y_r^* = 0$

Case 3: Regulate $y_r^* = -1$

CASE 1

① Step 1 : Solvability eq.

$$\begin{cases} 0 = (\lambda e^*)^3 + (u^*)^2 \\ \lambda e^* = 1 \end{cases}$$

No way to solve this ^{in real number} because : $(u^*)^2 = -1$

CASE 2

① Step 1 : Solvability eq.

$$\begin{cases} 0 = (\lambda e^*)^3 + (u^*)^2 \\ \lambda e^* = 0 \end{cases}$$

We found $u^* = 0,$

② Step 2 :

$$A = \frac{\partial}{\partial \lambda} (\lambda e^3 + u^2) \Big|_{\substack{\lambda = \lambda^* \\ u = u^*}} = 3 \lambda e^2 \Big|_{\lambda = \lambda^*} = 0$$

$$B = \frac{\partial}{\partial u} (\lambda e^3 + u^2) \Big|_{u=0} = 0$$

A and B are not stabilizable because it is impossible to put eigenval on 0.

CASE 3

① Step 1: Solvability

$$\begin{cases} 0 = (\lambda e^*)^3 + (u^*)^2 \\ \lambda e^* = -1 \end{cases}$$

$$u^* = \pm 1 \text{ (pick } u^* = 1\text{)}$$

② Step 2:

$$A = \left. \frac{\partial}{\partial \lambda e} (\lambda e^3 + u^2) \right|_{u=u^*=1} = 3$$

$$B = \left. \frac{\partial}{\partial u} (\lambda e^3 + u^2) \right|_{u=u^*=1} = 2$$

$$A + BK = 3 + 2K < 0 \Leftrightarrow K < -\frac{3}{2}$$

How does it affect our control eq.?

$$\begin{cases} \dot{\lambda e} = \lambda e^3 + u^2 \\ u = 1 + K(\lambda e + 1) \end{cases}$$

Substitute u , we get $\dot{\lambda e} = \lambda e^3 + (1+K(\lambda e + 1))^2 = F(\lambda e)$.

Call $\tilde{\lambda e} = \lambda e - \lambda e^* = \lambda e + 1$, so we get $\lambda e = \tilde{\lambda e} - 1$

$$\begin{aligned} \tilde{\lambda e} &= \dot{\lambda e} = (\tilde{\lambda e} - 1)^3 + (1 + K\tilde{\lambda e})^2 \\ &= \tilde{\lambda e}^3 - 2\tilde{\lambda e}^2 + \tilde{\lambda e} - \tilde{\lambda e}^2 + 2\tilde{\lambda e} - 1 + 1 + 2K\tilde{\lambda e} + K^2\tilde{\lambda e}^2 \\ \dot{\tilde{\lambda e}} &= \tilde{\lambda e}^3 + (K^2 - 3)\tilde{\lambda e}^2 + (3 + 2K)\tilde{\lambda e} \\ &= [\tilde{\lambda e}^2 + (K^2 - 3)\tilde{\lambda e} + (3 + 2K)]\tilde{\lambda e} \end{aligned}$$

$$V(\tilde{e}) = (\tilde{e} - \tilde{e}^*)^\top P (\tilde{e} - \tilde{e}^*) = \tilde{e}^\top P \tilde{e}$$

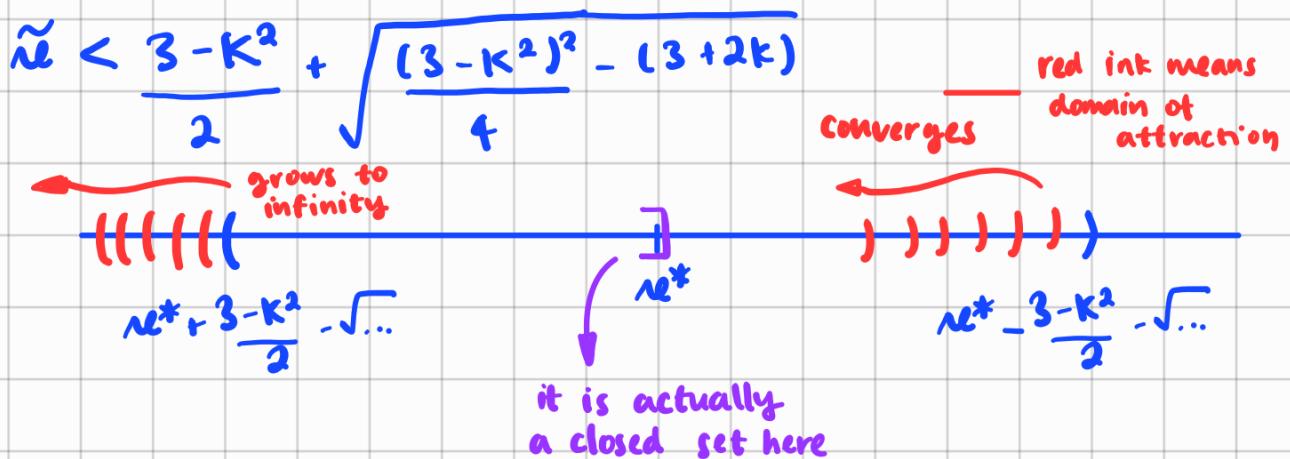
$$= \frac{1}{2} \tilde{e}^2$$

$$V(\tilde{e}) = \frac{\partial V(\tilde{e})}{\partial \tilde{e}} F(\tilde{e}) = [\tilde{e}^2 + (u^2 - 3)\tilde{e} + (3 + 2k)] \tilde{e}^2$$



Take the Lyapunov eq. $\tilde{e}^2 + (K^2 - 3)\tilde{e} + (3 + 2K) < 0$
and the solutions are :

$$\tilde{e} > \frac{3 - K^2}{2} - \sqrt{\frac{(3 - K^2)^2 - (3 + 2K)}{4}}$$



Other example:

$$\begin{cases} \dot{n}_e = n_e u + u \\ y_r = n_e \end{cases} \rightarrow y_r^* = 0$$

① Step 1

$$\begin{cases} 0 = n_e^* u^* + u^* \rightarrow u^* = 0 \\ 0 = n_e^* \end{cases}$$

② Step 2

$$A = \frac{\partial}{\partial n_e} (n_e u + u) \Big|_{\substack{n_e=0 \\ u=0}} = 0$$

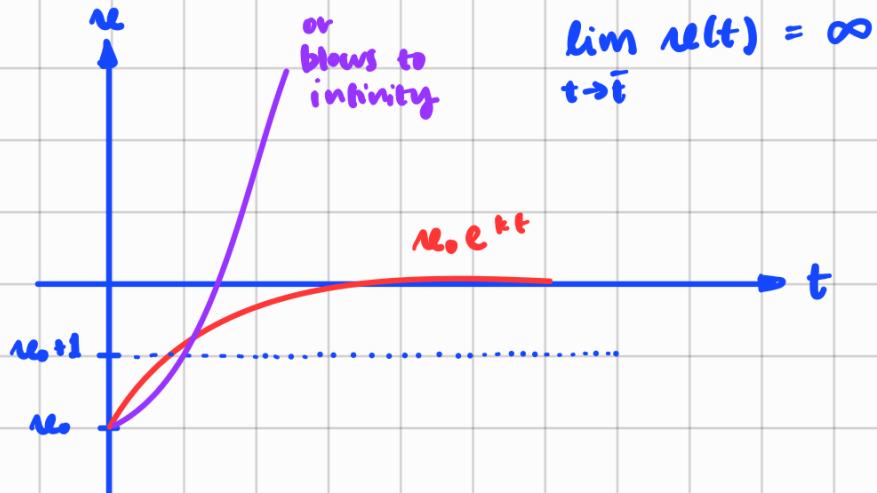
$$B = \frac{\partial}{\partial u} (n_e u + u) \Big|_{\substack{n_e=0 \\ u=0}} = n_e + 1 \Big|_{n_e=0} = 1$$

From this, we find $A + BK = K$. Thus $u(t) = K n_e$

$$\ddot{n}_e = K n_e^2 + K n_e = K n_e (n_e + 1) \Rightarrow n_e(t) = \frac{n_{e0} e^{Kt}}{n_{e0} + 1 - n_{e0} e^{Kt}}$$

$$n_{e0} < -1$$

What's weird from this solution?



$$\lim_{t \rightarrow \bar{t}} n_e(t) = \infty$$

$$N_0 e^{kt} = N_0 + 1$$

$$kt = \log\left(\frac{N_0 + 1}{N_0}\right)$$

$$t = \frac{1}{k} \log\left(\frac{N_0 + 1}{N_0}\right)$$

$$= \frac{1}{|k|} \log\left(\frac{|N_0|}{|N_0 + 1|}\right) = \bar{t}$$

LOCALITY:

$$\begin{aligned} f(N_e, u) &= A(N_e - N_e^*) + B(u - u^*) + f_{HOT}(N_e, u) \\ &= (A + B)(N_e - N_e^*) + f_{HOT}(N_e, u^* + K(N_e - N_e^*)) \end{aligned}$$

To define the control law we first need to linearize the system around (x^*, u)

$$A = \frac{\partial f}{\partial x}(x^*, u^*) = \left. \frac{\partial}{\partial x} (x^3 + u^2) \right|_{\substack{x=0 \\ u=0}} = 3x^2 \Big|_{x=0} = 0$$

$$B = \frac{\partial f}{\partial u}(x^*, u^*) = \left. \frac{\partial}{\partial u} (x^3 + u^2) \right|_{\substack{x=0 \\ u=0}} = 2u \Big|_{u=0} = 0$$

$\rightarrow A=0, B=0 \Rightarrow (A, B)$ IS NOT STABILIZABLE

\Rightarrow we cannot solve $\dot{x} = Ax + Bu$ with a linear controller

CASE 3. Regulation to $y_r^* = -1$

the solvability conditions are:

$$\begin{cases} 0 = x^*{}^3 + u^*{}^2 \\ -1 = x^* \end{cases} \Rightarrow \begin{cases} x^* = -1 \\ u^* = 1 \end{cases}$$

The linearization matrices are

$$A = \left. 3x^2 \right|_{x=-1} = 3 \quad \Rightarrow \quad (A, B) \text{ is controllable}$$

$$B = \left. 2u \right|_{u=1} = 2$$

For every $K < -\frac{3}{2}$, the matrix $A+BK$ is Hurwitz since:

$$A+BK = 3+2K < 0$$

then, for every $K < -\frac{3}{2}$, the controller

$$u(t) = 1 + K(x(t) + 1) \quad (\bullet)$$

locally stabilizes the eq. point where $y^* = -1$.

What about the domain of attraction?

Plugging (\bullet) into the system's equations leads to

$$\dot{x} = x^3 + (1 + K(x+1))^2 = x^3 + 1 + k^2(x+1)^2 + 2K(x+1)$$

changing coordinates from x to $\tilde{x} := x - x^* = x + 1$ leads to

$$\begin{aligned}\dot{\tilde{x}} &= \dot{x} + \dot{1} = \dot{x} = x^3 + 1 + k^2(x+1)^2 + 2k(x+1) \\ &= (\tilde{x}-1)^3 + 1 + k^2\tilde{x}^2 + 2k\tilde{x} \\ &= \tilde{x}^3 - 3\tilde{x}^2 + 3\tilde{x} + k^2\tilde{x}^2 + 2k\tilde{x} \\ &= \tilde{x}(3+2k + (k^2-3)\tilde{x} + \tilde{x}^2)\end{aligned}$$

this is suggested
by Lyapunov's indirect
method

Considering the Lyapunov candidate $V(x) = (x+1)^2 = \tilde{x}^2$, we get

$$\begin{aligned}\frac{\partial V}{\partial x}(x) \cdot f(x) &= 2\tilde{x}\dot{x} = 2\tilde{x}^2(3+2k + (k^2-3)\tilde{x} + \tilde{x}^2) \\ &= 2V(x) \cdot \underbrace{[3+2k + (k^2-3)\tilde{x} + \tilde{x}^2]}_{\text{we want this to be negative}}\end{aligned}$$

We want this to be negative

$$\Rightarrow \frac{\partial V}{\partial x}(x) f(x) < 0 \quad \text{as long as } 3+2k + (k^2-3)\tilde{x} + \tilde{x}^2 < 0$$

so, we need to solve $\tilde{x}^2 + (k^2-3)\tilde{x} + 3+2k < 0$, which implies:

$$\tilde{x} < \frac{3-k^2}{2} + \sqrt{\frac{(3-k^2)^2}{4} - (3+2k)} \quad \vee \quad \tilde{x} > \frac{3-k^2}{2} - \sqrt{\frac{(3-k^2)^2}{4} - (3+2k)}$$

Thus, the domain of attraction is

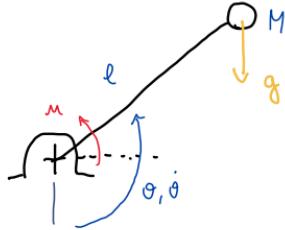
$$(1) = \left(\underset{-1}{\underset{\cup}{x'}} + \frac{3-k^2}{2} - \sqrt{\frac{(3-k^2)^2}{4} - (3+2k)}, \underset{-1}{\underset{\cup}{x''}} + \frac{3-k^2}{2} + \sqrt{\frac{(3-k^2)^2}{4} - (3+2k)} \right)$$

notice that as $|k|$ grows to infinity (1) converges to

$$\lim_{\substack{|k| \rightarrow \infty \\ k < 0}} (1) = [-\infty, x^*]$$

Thus, by "highering" the control gain ($|k| \uparrow$) enlarges the domain of attraction on the left, but reduces it on the right

EXAMPLE : ACTUATED PENDULUM



GOAL: stabilize the pendulum to a given position θ^*

Pendulum equations: $(x_1 = \theta, x_2 = \dot{\theta})$

$$\left\{ \begin{array}{l} \dot{x}_1 = x_2 \\ \dot{x}_2 = -\frac{g}{l} \sin x_1 - \frac{\beta}{Ml^2} x_2 + \frac{1}{Ml^2} u \\ y_m = x_1 \\ y_r = x_1 \\ y^* = \theta^* \end{array} \right.$$

We saw before that the SOLVABILITY Eqs. have the solution

$$\left\{ \begin{array}{l} x_1^* = \theta^*, \quad x_2^* = 0 \\ M^* = Mg l \sin \theta^* \end{array} \right. \quad [\text{in short: } x^* = (\theta^*, 0)]$$

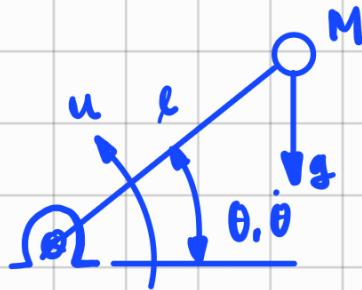
Next, we linearize the system around (x^*, M^*) . We have

$$\frac{\partial f}{\partial x} = \begin{pmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} \end{pmatrix} \quad \text{where} \quad \left\{ \begin{array}{l} f_1(x, u) = x_2 \\ f_2(x, u) = -\frac{g}{l} \sin x_1 - \frac{\beta}{Ml^2} x_2 + \frac{1}{Ml^2} u \end{array} \right.$$

we have:

$$\frac{\partial f_1}{\partial x_1}(x, u) = 0, \quad \frac{\partial f_1}{\partial x_2}(x, u) = 1, \quad \frac{\partial f_2}{\partial x_1}(x, u) = -\frac{g}{l} \cos x_1, \quad \frac{\partial f_2}{\partial x_2}(x, u) = -\frac{\beta}{Ml^2}$$

Example:



$$\left\{ \begin{array}{l} \dot{\nu}_2 = \nu_2 \\ \ddot{\nu}_2 = -\frac{g}{l} \sin \nu_2 - \frac{\beta}{Ml^2} \nu_2 + \frac{1}{Ml^2} u \\ y_m = \nu_2 \\ y_r = \theta \quad y_r^* = \theta^* \end{array} \right.$$

① Step 1:

$$\left\{ \begin{array}{l} \nu^* = (\theta^*, 0) \\ u^* = Mg l \sin \theta^* \end{array} \right. \rightarrow f(\nu, u) = \begin{pmatrix} f_1(\nu, u) \\ f_2(\nu, u) \end{pmatrix}$$

② Step 2:

$$A = \frac{\partial f}{\partial u}(\nu^*) = \begin{bmatrix} 0 & 1 \\ -g/l \cos \theta^* & -\beta/Ml^2 \end{bmatrix}$$

$$B = \begin{bmatrix} 0 \\ 1/Ml^2 \end{bmatrix}$$

Based on controllability canonical form, AB is controllable.

We then have to find K so that $K = [K_1 \ K_2]$

$$A + BK = \begin{bmatrix} 0 & 1 \\ -g/l \cos \theta^* & -\beta/Ml^2 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ \frac{K_1}{Ml^2} & \frac{K_2}{Ml^2} \end{bmatrix} \quad \text{cont...}$$

$$A + BK = \begin{bmatrix} 0 & 1 \\ -\frac{g}{l} \cos \theta^* + \frac{K_2}{Ml^2} & \frac{K_2 - \beta}{Ml^2} \end{bmatrix}$$

The eigenval charact. polynom. is:
 $\lambda^2 + d_2 \lambda + a_0$
 $d_2 > 0$
 $a_0 > 0$

$$\frac{K_2 - \beta}{Ml^2} < 0, \quad \cos \theta^* > 0,$$

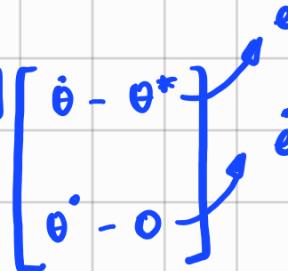
$$\frac{K_1}{Ml^2} - \frac{g}{l} \cos \theta^* < 0,$$

$$K_1 < Mgl \underbrace{\cos \theta^*}_{-1 \leftarrow \text{worst case}}$$

$$\forall K_2 < 0$$

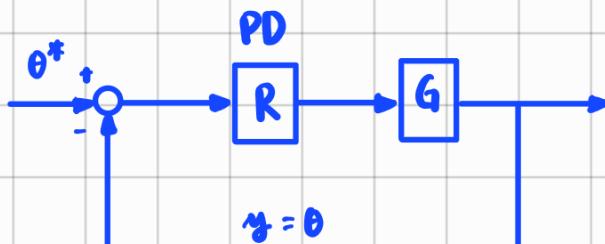
④ Step 3:

Pick $u = Mgl \sin \theta^* + [K_1 \ K_2] \begin{bmatrix} \dot{\theta} - \theta^* \\ \ddot{\theta} - 0 \end{bmatrix}$



$$u = Mgl \sin \theta^* + K_2 e + K_2 \dot{e}$$

it's a PD controller



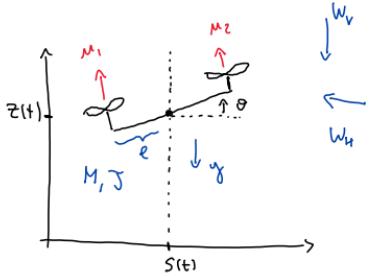
Thus, we obtain:

$$A = \frac{\partial f}{\partial x}(x^*, u^*) = \begin{pmatrix} 0 & 1 \\ -\frac{g}{l} \cos \theta^* & -\frac{\beta}{M l^2} \end{pmatrix} \quad \leftarrow \begin{cases} \text{If } \theta^* > \frac{\pi}{2}, \text{ the forced} \\ \text{equilibrium } x^* \text{ is unstable} \end{cases}$$

$$B = \frac{\partial f}{\partial u}(x^*, u^*) = \begin{pmatrix} \frac{\partial f_1}{\partial u}(x^*, u^*) \\ \frac{\partial f_2}{\partial u}(x^*, u^*) \end{pmatrix} = \begin{pmatrix} 0 \\ \frac{1}{M l^2} \end{pmatrix}$$

. Since (A, B) is controllable, we can always design K so that $A+BK$ is Hurwitz, for instance: $K = [M \lg(\cos \theta^* - \varepsilon) \quad -\mu]$ for any $\varepsilon > 0$ and $\mu \geq 0$ (PD controller)

EXAMPLE: PLANAR DRONE



$$\begin{cases} M \ddot{s} = -(m_1 + m_2) \sin \theta - w_H - \delta \dot{z} \\ M \ddot{z} = (m_1 + m_2) \cos \theta - Mg - w_v - \delta \dot{s} \\ J \ddot{\theta} = -\beta \dot{\theta} + l(m_2 - m_1) \end{cases}$$

β, δ = friction coefficients
 $w_H \geq 0$ and $w_v \geq 0$ = (constant) wind

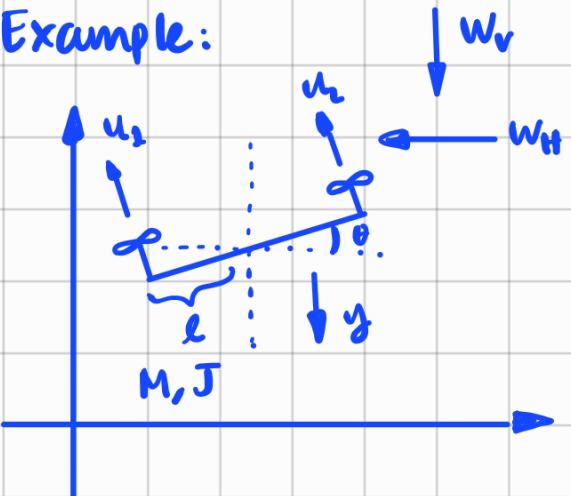
GOAL: regulate both velocities \dot{s} and \dot{z} to zero

$$\hookrightarrow y_r = \begin{pmatrix} \dot{s} \\ \dot{z} \end{pmatrix} \quad \text{and} \quad y_r^* = \begin{pmatrix} 0 \\ 0 \end{pmatrix}$$

We neglect the vertical position and call

$$x_1 = \dot{s}, \quad x_2 = \dot{z}, \quad x_3 = \theta, \quad x_4 = \dot{\theta}$$

Example:



Model Eq:

$$\begin{cases} M\ddot{s} = -(u_1 + u_2) \sin\theta - \delta \dot{s} - w_H \\ M\ddot{z} = (u_1 + u_2) \cos\theta - \delta \dot{z} - w_v - Mg \\ J\ddot{\phi} = -\beta \dot{\phi} + l(u_2 - u_1) \end{cases}$$

neglect the position dynamics

$$\alpha_{e_3} = \dot{s}, \quad \alpha_{e_2} = \dot{z}, \quad \alpha_{e_3} = \theta \quad \alpha_{e_4} = \dot{\theta}$$

$$\begin{cases} \dot{\alpha}_{e_3} = -\frac{1}{M}(u_1 + u_2) \sin \alpha_{e_3} - \frac{\delta}{M} \alpha_{e_2} - \frac{w_H}{M} \\ \dot{\alpha}_{e_2} = \frac{1}{M}(u_1 + u_2) \cos \alpha_{e_3} - \frac{\delta}{M} \alpha_{e_2} - \frac{w_v}{M} - g \\ \ddot{\alpha}_{e_3} = \alpha_{e_4} \\ \dot{\alpha}_{e_4} = -\frac{\beta}{J} \alpha_{e_4} + \frac{l}{J} (u_2 - u_1) \\ \gamma_r = (\alpha_{e_1}, \alpha_{e_2}) \quad \gamma_r^* = 0 \end{cases}$$

① Step 1:

$$\begin{cases} 0 = -(u_1^* + u_2^*) \sin \alpha_{e_3}^* - \delta \alpha_{e_1}^* - w_H \\ 0 = (u_1^* + u_2^*) \cos \alpha_{e_3}^* - \delta \alpha_{e_2}^* - w_v - Mg \\ 0 = \alpha_{e_4}^* \\ 0 = \frac{l}{J} (u_2^* - u_1^*) \quad u_3^* = u_2^* = U^* \\ \alpha_{e_1}^* = \alpha_{e_2}^* = 0 \end{cases} \quad \textcircled{#}$$

#

$$2V^* \sin \alpha_3^* = -W_H$$

$$2V^* \cos \alpha_3^* = W_v + Mg$$

$$\tan \alpha_3^* = -\frac{W_H}{W_v + Mg} \Leftrightarrow \alpha_3^* = -\tan^{-1}\left(\frac{W_H}{W_v + Mg}\right)$$



$$(2V^*)^2 \underbrace{(\sin^2 \alpha_3^* + \cos^2 \alpha_3^*)}_{=1} = W_H^2 + (W_v + Mg)^2$$

$$V^* = \frac{1}{2} \sqrt{W_H^2 + (W_v + Mg)^2}$$

$$\alpha^* = \begin{bmatrix} 0 \\ 0 \\ -\tan^{-1}\left(\frac{W_H}{W_v + Mg}\right) \\ 0 \end{bmatrix}$$

① Step 2:

$$A = \left[\begin{array}{cc|cc} -\frac{\sigma}{M} & 0 & -\frac{2V^* \cos \alpha_3^*}{M} & 0 \\ 0 & -\frac{\delta}{M} & -\frac{2V^* \sin \alpha_3^*}{M} & 0 \\ \hline 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & -\frac{\rho}{J} \end{array} \right]$$

$$B = \begin{bmatrix} -\frac{\sin \alpha_3^*}{M} & -\frac{\sin \alpha_3^*}{M} \\ \frac{\cos \alpha_3^*}{M} & \frac{\cos \alpha_3^*}{M} \\ 0 & 0 \\ -\frac{\rho}{J} & \frac{\rho}{J} \end{bmatrix}$$

$$K = \left[\begin{array}{cc|cc} 0 & 0 & \alpha & \alpha \\ 0 & 0 & \alpha & -\alpha \end{array} \right]$$

④ Step 3:

$$u(t) = \begin{pmatrix} u^* \\ u^* \end{pmatrix} + K \left(u - \begin{bmatrix} u^* \\ 0 \\ 0 \\ -\tan\left(\frac{w_h}{w_r + u_0}\right) \\ 0 \end{bmatrix} \right)$$

$$U^* = \sqrt{w_h^2 + (w_r + Mz)^2}$$

The model in the state space reads

$$\left\{ \begin{array}{l} \dot{x}_1 = -\frac{1}{M}(m_1 + m_2) \cdot \sin x_3 - \frac{w_H}{M} - \frac{\delta}{M} x_1 \\ \dot{x}_2 = \frac{1}{M}(m_1 + m_2) \cdot \cos x_3 - g - \frac{w_v}{M} - \frac{\delta}{M} x_2 \\ \dot{x}_3 = x_4 \\ \dot{x}_4 = -\frac{\beta}{J} x_4 + \frac{\ell}{J}(m_2 - m_1) \\ y_m = x \\ y_r = (x_1, x_2) \end{array} \right. \quad \text{GOAL: } y_r^* = 0$$

SOLVABILITY EQUATIONS:

$$\left\{ \begin{array}{l} 0 = - (m_1^* + m_2^*) \sin x_3^* - w_H - \delta x_1^* \\ 0 = (m_1^* + m_2^*) \cos x_3^* - Mg - w_v - \delta x_2^* \\ 0 = x_4^* \\ 0 = -\beta x_4^* + \ell(m_2^* - m_1^*) \\ 0 = x_1^* = x_2^* \end{array} \right.$$

$$\rightarrow \left\{ \begin{array}{l} \tan x_3^* = -\frac{w_H}{Mg + w_v} \Rightarrow x_3^* = -\arctan \left(\frac{w_H}{Mg + w_v} \right) \\ m_1^* = m_2^* = U^*, \quad U^* \doteq \sqrt{w_H^2 + (Mg + w_v)^2} \\ x_1^* = x_2^* = x_4^* = 0 \end{array} \right.$$

Let us linearize the system about (x^*, μ^*) . We have

$$\dot{x} = f(x, \mu) \quad f(x, \mu) = \begin{pmatrix} f_1(x, \mu) \\ f_2(x, \mu) \\ f_3(x, \mu) \\ f_4(x, \mu) \end{pmatrix}$$

with

$$f_1(x, \mu) = -\frac{1}{M} (\mu_1 + \mu_2) \cdot \sin x_3 - \frac{\omega_h}{M} - \frac{\delta}{M} x_1$$

$$f_2(x, \mu) = \frac{1}{M} (\mu_1 + \mu_2) \cdot \cos x_3 - g - \frac{\omega_v}{M} - \frac{\delta}{M} x_2$$

$$f_3(x, \mu) = x_4$$

$$f_4(x, \mu) = -\frac{\beta}{J} x_4 + \frac{\ell}{J} (\mu_2 - \mu_1)$$

Then:

$$\frac{\partial f_1}{\partial x}(x^*, \mu^*) = \begin{bmatrix} 0 & -\frac{\delta}{M} & -\frac{1}{M} 2U^* \cdot \cos x_3^* & 0 \end{bmatrix}$$

$$\frac{\partial f_2}{\partial x}(x^*, \mu^*) = \begin{bmatrix} 0 & -\frac{\delta}{M} & -\frac{1}{M} 2U^* \sin x_3^* & 0 \end{bmatrix}$$

$$\frac{\partial f_3}{\partial x}(x^*, \mu^*) = \begin{bmatrix} 0 & 0 & 0 & 1 \end{bmatrix}$$

$$\frac{\partial f_4}{\partial x}(x^*, \mu^*) = \begin{bmatrix} 0 & 0 & 0 & -\frac{\beta}{J} \end{bmatrix}$$

$$\frac{\partial f}{\partial u}(x^*, u^*) = \begin{bmatrix} -\frac{1}{M} \sin x_3^* & -\frac{1}{\pi} \sin x_3^* \\ \frac{1}{\pi} \cos x_3^* & \frac{1}{\pi} \cos x_3^* \\ 0 & 0 \\ -\frac{\ell}{J} & \frac{\ell}{J} \end{bmatrix}$$

Thus, the linearization is given by the matrices

$$A = \begin{bmatrix} -\frac{\delta}{M} & 0 & -\frac{1}{M} 2U^* \cdot \cos x_3^* & 0 \\ 0 & -\frac{\delta}{\pi} & -\frac{1}{\pi} 2U^* \sin x_3^* & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & -\frac{\beta}{J} \end{bmatrix}$$

$$B = \begin{bmatrix} -\frac{1}{M} \sin x_3^* & -\frac{1}{\pi} \sin x_3^* \\ \frac{1}{\pi} \cos x_3^* & \frac{1}{\pi} \cos x_3^* \\ 0 & 0 \\ -\frac{\ell}{J} & \frac{\ell}{J} \end{bmatrix}$$

where $x_3^* = -\arctan\left(\frac{w_H}{Mg + w_V}\right)$ and $U^* = \frac{1}{2} \sqrt{w_H^2 + (Mg + w_V)^2}$

The pair (A, B) is always stabilizable. Indeed, the matrix

$$K = \begin{bmatrix} 0 & 0 & \alpha & \alpha \\ 0 & 0 & -\alpha & -\alpha \end{bmatrix} \quad \alpha > 0$$

is such that $A+BK$ is Hurwitz. It can be also shown that (A, B) is controllable whenever $x_3^* \neq K \frac{\pi}{2}$, $K \in \mathbb{N}$

The control design is then concluded by taking K such that $A+BK$ is Hurwitz and choosing

$$u(t) = u^* + K(x(t) - x^*)$$

that in this case reads: (K_{ij} is the (i,j) -th entry of K)

$$M_1 = \frac{1}{2} \sqrt{w_u^2 + (M_g + w_v)^2} + K_{11}x_1 + K_{12}x_2 + K_{13} \left(x_3 + \alpha \tan \left(\frac{w_u}{M_g + w_v} \right) \right) + K_{14}x_4$$

$$M_2 = \frac{1}{2} \sqrt{w_u^2 + (M_g + w_v)^2} + K_{21}x_1 + K_{22}x_2 + K_{23} \left(x_3 + \alpha \tan \left(\frac{w_u}{M_g + w_v} \right) \right) + K_{24}x_4$$



FEED FORWARD ACTION



ERROR - FEEDBACK

- EXERCISE:**
- . study the controllability properties of the system
 - . design K to place the eigenvalues where desired
(whenever possible)

DESIGNING AN OBSERVER

Let a model equation:

$$\begin{cases} \dot{x} = f(x, u) \\ y_m = h_m(x) \neq x \\ y_r = h_r(x) \underset{x}{\sim} \end{cases}$$

$$u = u^* + K(\overbrace{x - x^*})$$

Consider the state equation where:

$$\dot{x} = A(x - x^*) + B(u - u^*) + f_{hot}(x, u)$$

$$\overset{\text{||}}{\underset{\sim}{\sim}}$$

$$\dot{\tilde{x}} = A\tilde{x} + B\tilde{u} + f_{hot}(x, u)$$

To create a Luenberger Observer, we just copy the dynamic. Hence,

$$u = u^* + K\hat{x}$$

$$\dot{\hat{x}} = A\hat{x} + B\tilde{u} + L[C_m\hat{x} - \tilde{y}_m]$$

So $A + LC_m$ is Hurwitz

We can linearize the output:

$$\begin{aligned} \tilde{y}_m &= y_m - h_m(x^*) = h_m(x) - h_m(x^*) \\ &= C_m(x - x^*) + h_{m, hot}(x) \\ \text{where } C_m &= \frac{\partial h_m}{\partial x}(x^*) \end{aligned}$$

After we linearize the estimation:

$$\begin{cases} \dot{\hat{x}} = A\hat{x} + BK\hat{x} + L[C_m\hat{x} - \tilde{y}_m] \\ u = u^* + K\hat{x} \end{cases} \quad \parallel \quad \tilde{y}_m - u_m(u^*)$$

Remember we introduced:

$$\begin{aligned} \xi &= g(\xi, y_m), \quad \xi = \hat{x} \\ u &= \delta(\xi, y_m) \end{aligned}$$

What is \hat{x} considering (u^*, ξ^*) where $x \rightarrow x^*$?
It is 0 because $\hat{x} = x - x^*$.

NB: we don't change the coordinate in this case.

Let the controller dynamic as:

$$\begin{cases} \dot{x} = A(x - x^*) + BK\hat{x} + f_{hor}(x, u) \\ \dot{\hat{x}} = (A + BK + LC_m)\hat{x} - LC_m(x - x^*) - Lh_{m, hor}(x) \end{cases}$$

Remember:

$$-L\tilde{y}_m = -L(C_m\hat{x} + h_{m, hor}(x))$$

$$\begin{bmatrix} f_{hor}(x, u^* + K\hat{x}) \\ -Lh_{m, hor}(x) \end{bmatrix}$$

It's a tradeoff
between speed of
conv. and stability.

In matrix form:

$$\begin{bmatrix} \dot{x} \\ \dot{\hat{x}} \\ \dot{x} \end{bmatrix} = \underbrace{\begin{bmatrix} A & BK \\ -LC_m & A + BK + LC_m \end{bmatrix}}_{\bar{A}} \begin{bmatrix} x - x^* \\ \hat{x} \end{bmatrix} + \begin{bmatrix} f_{hor}(x, u) \\ -Lh_{m, hor}(x) \end{bmatrix}$$

LOCALITY PROBLEM

LOCAL OUTPUT - FEEDBACK SOLUTION

In this case $y_m(x) \neq X$, typically we only measure a part of the state

We can resort to the linear output-feedback theory, by which we implement

$$M(t) = M^* + K \hat{\tilde{x}}(t) \quad \left(\text{instead of } M = M^* + K(x - x^*) \right)$$

in which \hat{X} is an estimate of $X \div x - x^*$.

We can write the plant as before as

$$\tilde{x} = A\tilde{x} + B\tilde{u} + f_{\text{hot}}(x, u)$$

where

$$\hat{m} = m - m^*, \quad A = \frac{\partial f}{\partial x}(x^*, m^*), \quad B = \frac{\partial f}{\partial u}(x^*, m^*)$$

$$f_{\text{HOT}}(x, \mu) = f(x, \mu) - A\hat{x} - B\hat{\mu} \quad \rightarrow \quad \text{HIGHER-ORDER TERMS}$$

we add the output equation

$$\tilde{y}_m = h_m(x) - h_m(x^*) = C \tilde{x} + h_{m, \text{not}}(x)$$

where

$$C \doteq \frac{\partial h_m}{\partial x}(x^*) \quad \text{and} \quad h_{m,\text{HOT}}(x) = h_m(x) - h_m(x^*) - C \hat{x}$$

C
= HIGHER ORDER TERM

If (C, A) is detectable, we can find L such that $A + LC$ is Hurwitz,

and we can define the Luenberger observer

$$\dot{\tilde{x}} = A\hat{\tilde{x}} + B\tilde{m} + L(c\hat{\tilde{x}} - \tilde{y}_m)$$

$$= A \hat{\tilde{x}} + B K \hat{\tilde{x}} + L (c \hat{\tilde{x}} - y_m + h_m(x^*))$$

The overall controller is then

$$(▲) \quad \begin{cases} \dot{\hat{x}} = (A + BK + LC)\hat{x} - L\tilde{y}_m \\ u = u^* + K\hat{x} \end{cases}$$

\rightarrow of the form (see page 1)
 $\begin{cases} \dot{\xi} = g(\xi, y_m) \\ u = \gamma(\xi, y_m) \end{cases}$
 with
 $\xi = \hat{x}$

RESULT. suppose that (A, B) is stabilizable and (C, A) detectable.

The dynamic controller (▲) Locally stabilizes the equilibrium $(x, \hat{x}) = (x^*, 0)$. Therefore, there exists an open set $\mathcal{O} \subset \mathbb{R}^n \times \mathbb{R}^n$ containing $(x^*, 0)$ such that

$$\forall (x(0), \hat{x}(0)) \in \mathcal{O}, \quad \lim_{t \rightarrow \infty} y_r(t) = y_r^*$$

PROOF.

Let us analyze the closed-loop system, which has state (\hat{x}, \hat{x})
 Changing variables from \hat{x} to

$$e := \hat{x} - x$$

leads to the closed-loop system

$$\dot{\hat{x}} = Ax + B\hat{x} + f_{hor}(x, u) = (A + BK)\hat{x} + BKe + f_{hor}(x, u)$$

$$\hat{x} = K\hat{x} = K(e + \hat{x})$$

$$\begin{aligned} \dot{e} &= \dot{\hat{x}} - \dot{x} = (A + BK + LC)\hat{x} - L\tilde{y}_m - (A + BK)\hat{x} - BKe - f_{hor}(x, u) \\ &= (A + BK + LC)(e + \hat{x}) - (A + BK)\hat{x} - BKe - L(C\hat{x} + h_{m, hor}(x)) - f_{hor}(x, u) \\ &= (A + LC)e - (f_{hor}(x, u) + Lh_{m, hor}(x)) \end{aligned}$$

Proof (in more direct way)

$$T = \begin{bmatrix} I & 0 \\ -I & I \end{bmatrix} \Leftrightarrow T^{-1} = \begin{bmatrix} I & 0 \\ I & I \end{bmatrix}$$

Locality problem

So,

$$T \begin{bmatrix} \alpha \\ \hat{\alpha} \end{bmatrix} = \begin{bmatrix} \alpha \\ \hat{\alpha} - \alpha \end{bmatrix}$$

$$\bar{A} = \begin{bmatrix} A & BK \\ -LC_m & A+BK+LC_m \end{bmatrix}$$

What is $T\bar{A}T^{-1}$?

$$T\bar{A} = \begin{bmatrix} A & BK \\ -A-LC_m & A+LC_m \end{bmatrix}$$

$$T\bar{A}T^{-1} = \begin{bmatrix} A+BK & BK \\ \hline 0 & ALC_m \end{bmatrix}$$

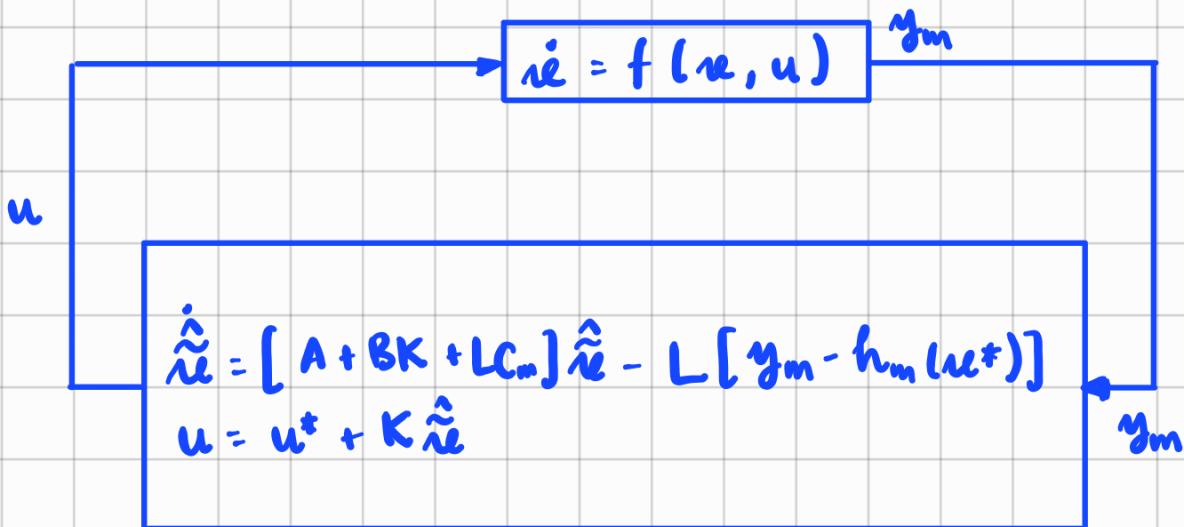
The robustness problem

$$u^*, h_m(x^*)$$

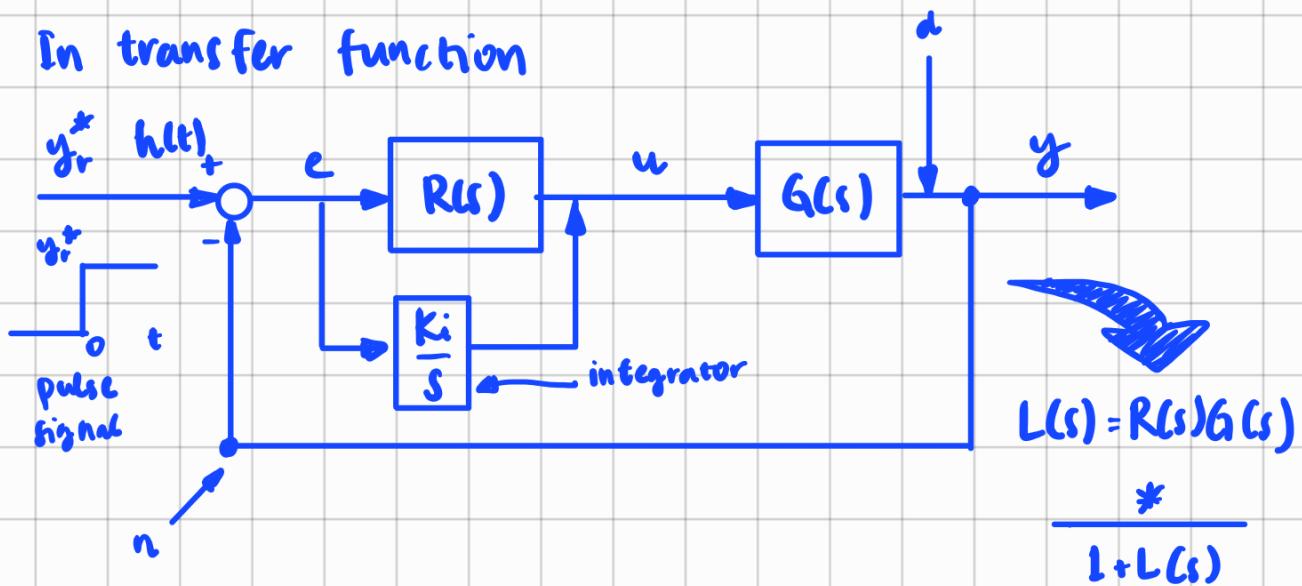
$$\begin{matrix} \parallel \\ y_r^* \\ \parallel \\ y_m \end{matrix}$$

Assume:

$$y_m = y_r$$



In transfer function



$$L(s) = \frac{1}{s} L'(s) \quad \text{where } L'(0) \neq 0$$

$$E(j\omega) = \frac{1}{1 + L(j\omega)} \cdot Y_R^*(j\omega)$$

The error function :

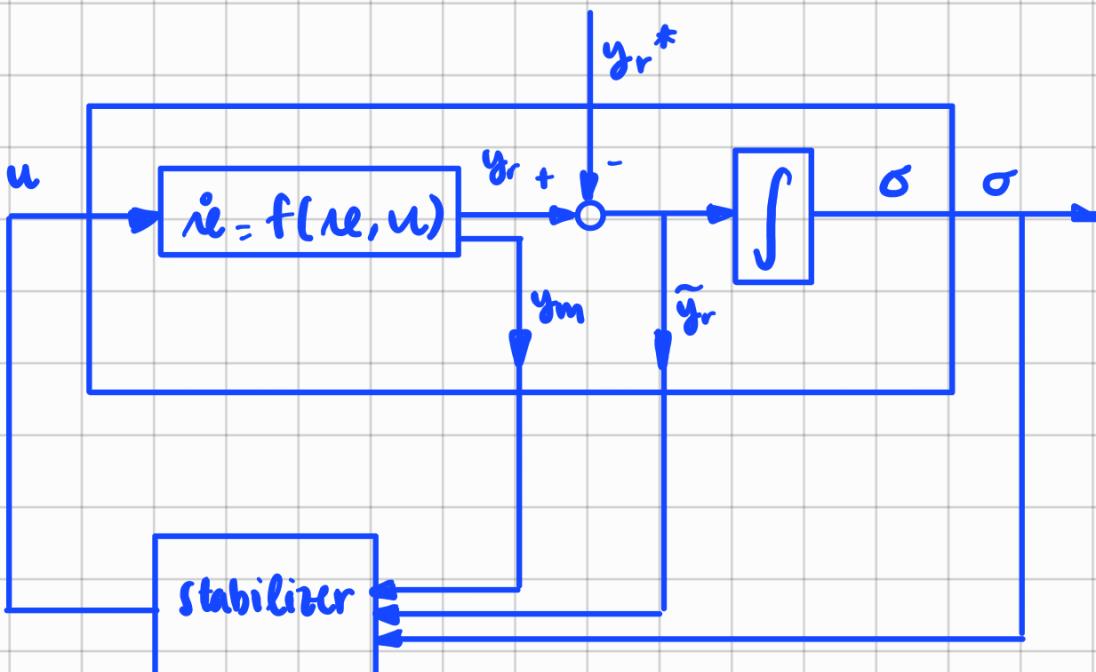
$$e(t) \approx \left| \frac{1}{1+L(j\omega)} \right| \cdot Y_r^* = 0 \Rightarrow L(j\omega) \xrightarrow{\omega \rightarrow 0} \infty$$

$$Y_r^*(t) = Y_r^*$$

Say we have $\frac{1}{s} E(s) \rightarrow \int_0^t e(\tau) d\tau = \sigma(t)$

$$\Sigma(s) = \frac{1}{s} E(s) \rightarrow \int_0^t e(\tau) d\tau = \sigma(t) \quad \dot{\sigma} = 0$$

In non-linear system:



$$\star \quad \dot{\boldsymbol{y}}_m = \begin{bmatrix} \boldsymbol{x} \\ \tilde{\boldsymbol{y}}_r \end{bmatrix}$$

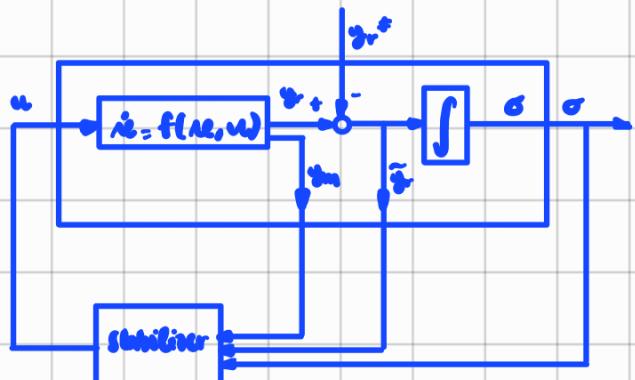
$$\star \quad n_u = n_r \quad \begin{cases} n_u = \dim(u) \\ n_r = \dim(y_r) \end{cases}$$

$$\dot{\boldsymbol{x}} = f(\boldsymbol{x}, u)$$

$$\tilde{\boldsymbol{y}}_r = h_r(\boldsymbol{x}), \quad \tilde{\boldsymbol{y}}_r^* = \boldsymbol{y}_r - \boldsymbol{y}_r^*$$

$$\begin{cases} \dot{\sigma} = \tilde{\boldsymbol{y}}_r \\ u = \hat{u}^* + K_1(\boldsymbol{x} - \hat{\boldsymbol{x}}^*) + K_2 \sigma \end{cases}$$

↓ modifiable ↓ modifiable



$$\dot{\sigma}_i = \tilde{y}_{r,i} \quad \forall i = 1, \dots, n_r$$

Why with wrong estimate $\hat{\boldsymbol{x}}^*$ can still lead to stability?

$$\dot{\boldsymbol{x}} = A(\boldsymbol{x} - \boldsymbol{x}^*) + B(u - u^*) + f_{hot}(\boldsymbol{x}, u)$$

$$\dot{\sigma} = h_r(\boldsymbol{x}) - \boldsymbol{y}_r^* = h_r(\boldsymbol{x}) - h_r(\boldsymbol{x}^*) \\ = C_r(\boldsymbol{x} - \boldsymbol{x}^*) + h_{r, hot}(\boldsymbol{x})$$

$$u = \hat{u}^* + K_1(\boldsymbol{x} - \boldsymbol{x}^* + \boldsymbol{x}^* - \hat{\boldsymbol{x}}^*) + K_2 \sigma$$

$$\dot{\boldsymbol{x}} = A(\boldsymbol{x} - \boldsymbol{x}^*) + B K_1 (\boldsymbol{x} - \boldsymbol{x}^*) + B K_2 \sigma \\ + B(\hat{u}^* - u^* + K_1(\boldsymbol{x}^* - \hat{\boldsymbol{x}}^*)) + f_{hot}(\boldsymbol{x}, u)$$

$$\dot{\sigma} = C_r(\boldsymbol{x} - \boldsymbol{x}^*) + h_{r, hot}(\boldsymbol{x})$$

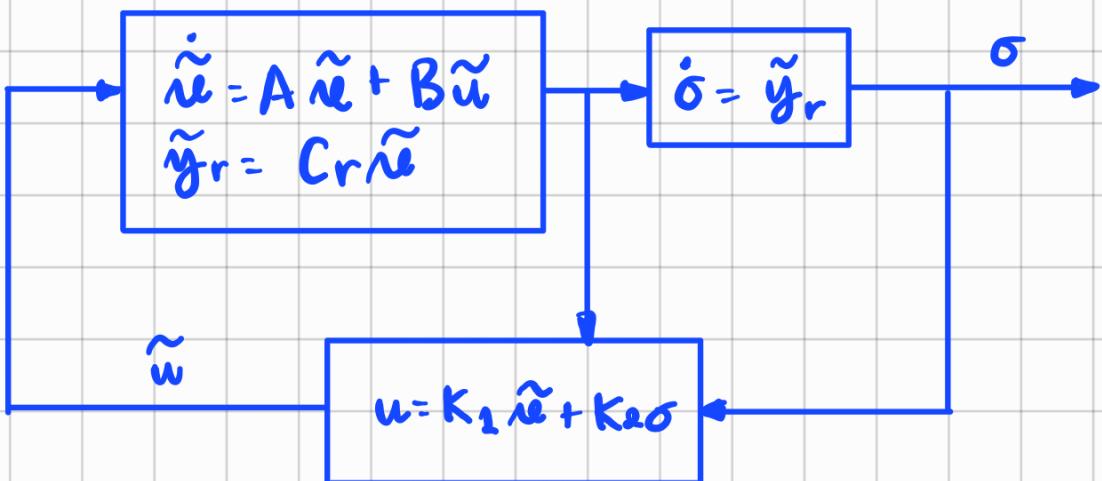
The matrix form:

$$\begin{bmatrix} \dot{\mathbf{x}} \\ \dot{\sigma} \end{bmatrix} = \begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix} \begin{bmatrix} \mathbf{x} - \mathbf{x}^* \\ \sigma \end{bmatrix} + \begin{bmatrix} f_{\text{hor}}(\mathbf{x}, u) \\ h_r(\mathbf{x}) \end{bmatrix}$$

$$+ \begin{bmatrix} B \\ 0 \end{bmatrix} [\hat{u}^* - u^* + K_1(\mathbf{x}^* - \hat{\mathbf{x}}^*)]$$



Consider
a linear system



This cascade is stabilizable if:

$$\bar{A} = \begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix} \quad \exists K_1, K_2 : \bar{A} \text{ is Hurwitz}$$

1) AB are stabilizable

2) $\text{rank} \begin{bmatrix} A & B \\ Cr & 0 \end{bmatrix} = n_x + n_r$
(non-resonance cond.)

Proof using PBH-test

$$\begin{bmatrix} A + BK_2 & BK_2 \\ Cr & 0 \end{bmatrix} = \underbrace{\begin{bmatrix} A & 0 \\ Cr & 0 \end{bmatrix}}_{A_0} + \underbrace{\begin{bmatrix} B \\ 0 \end{bmatrix}}_{B_0} \begin{bmatrix} K_1 & K_2 \end{bmatrix}$$

↙ this is stabilizable
if A_0 and B_0 stabilizable

$$\begin{array}{|c|} \hline \dot{\tilde{e}} = A\tilde{e} + B\tilde{u} \\ \tilde{y}_r = Cr\tilde{e} \\ \hline \end{array} \rightarrow \dot{e} = \tilde{y}_r$$

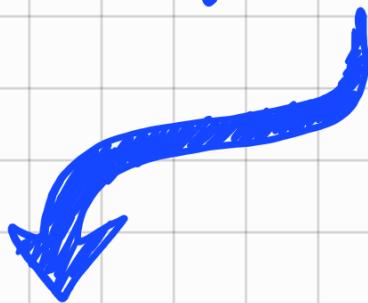
We found that

$$\begin{cases} \dot{\epsilon} = f(\epsilon, u) \\ y_r = h_r(\epsilon), \quad y_r^* \\ y_m = (\epsilon, \underbrace{y_r - y_r^*}_{\tilde{y}_r}) \end{cases}$$



$$\begin{cases} \dot{\epsilon} = \tilde{y}_r \\ u = \hat{u}^* + K_1(\epsilon - \hat{\epsilon}^*) \\ \quad + K_2 \sigma \end{cases}$$

where $n_u = n_r$



From those expressions above, we can form:

$$\begin{bmatrix} \dot{\epsilon} \\ \dot{\sigma} \end{bmatrix} = \underbrace{\begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix}}_{\bar{A}} \begin{bmatrix} \epsilon - \epsilon^* \\ \sigma \end{bmatrix} + \begin{bmatrix} f_{m1}(\epsilon, u) \\ h_{f, H_r}(\epsilon) \end{bmatrix} + \begin{bmatrix} B \\ 0 \end{bmatrix} [\hat{u}^* - u^* + K_1(\epsilon^* - \hat{\epsilon}^*)]$$

where $A = \frac{\partial f}{\partial \epsilon}(\epsilon^*, u^*), \quad B = \frac{\partial f}{\partial u}(\epsilon^*, u^*), \quad Cr = \frac{\partial h_r}{\partial \epsilon}(\epsilon^*)$

The result: $\exists K_1, K_2$ such that \bar{A} is Hurwitz if:

1) (A, B) is stabilizable

2) Rank $\begin{bmatrix} A & B \\ Cr & 0 \end{bmatrix} = \text{n. of rows} = n_{\text{net}} + n_r$

NON-RESONANCE CONDITION

Proof: $\bar{A} = \underbrace{\begin{bmatrix} A & 0 \\ Cr & 0 \end{bmatrix}}_{A_0} + \underbrace{\begin{bmatrix} B \\ 0 \end{bmatrix} \begin{bmatrix} K_1 & K_2 \end{bmatrix}}_{B_0 K}$

$\exists K_1, K_2$ such that
 \bar{A} is Hurwitz iff
 (A_0, B_0) stabilizable

PBH — Controllability \rightarrow rank $[\lambda I - A_0 \quad B_0]$
 = n. of rows $\forall \lambda$ (full)

Stabilizability \rightarrow rank $[\lambda I - A_0 \quad B_0]$ (full)
 = n. of rows $\forall \lambda : \text{Real}[\lambda] \geq 0$

recommended to study the geometry

The rank of $\begin{bmatrix} \lambda I - A & 0 & B \\ -C_r & \lambda I & 0 \end{bmatrix}$ is:

* CASE 1: $\lambda \neq 0$

$$\text{rank} \begin{bmatrix} \lambda I - A & 0 & B \\ -C_r & \lambda I & 0 \end{bmatrix} = \text{n. of rows (full rank)}$$

* CASE 2: $\lambda = 0$

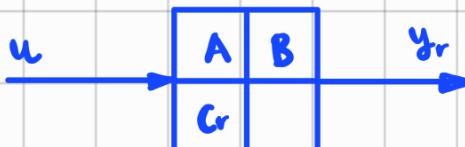
$$\text{rank} \begin{bmatrix} -A & 0 & B \\ -C_r & 0 & 0 \end{bmatrix} = \text{rank} \begin{bmatrix} A & B & 0 \\ C_r & 0 & 0 \end{bmatrix}$$

switch the col. to see the matrix is full rank n. of rows (full rank)

switch the col.

= n. of rows (full rank)

It indicates:



Back to the closed-loop system:

$$\begin{bmatrix} \dot{e} \\ \dot{o} \end{bmatrix} = \underbrace{\begin{bmatrix} A+BK_2 & BK_2 \\ Cr & 0 \end{bmatrix}}_{\bar{A}} \begin{bmatrix} e - e^* \\ \sigma \end{bmatrix} + \begin{bmatrix} f_{hor}(e, u) \\ h_{hor}(e) \end{bmatrix}$$

$$= \bar{A} + \begin{bmatrix} B \\ 0 \end{bmatrix} [\hat{e}^* - e^* + K_2(e^* - \hat{e}^*)] \star - K_2 \sigma^*$$

Fix K_1, K_2 such that A is Hurwitz:

Set $\alpha_e = \alpha_e^*$, $\sigma = 0$, we got

$$u = \hat{u}^* + K_1(\alpha_e - \hat{\alpha}_e^*) + K_2\sigma = \hat{u}^* + K_1(\alpha_e^* - \hat{\alpha}_e^*) \neq u^*.$$

We find that $\sigma \neq 0$!

Remember regulator $\begin{cases} \dot{\xi} = g(\xi, y_m) \\ u = \gamma(\xi, y_m) \end{cases}$ ★

(α_e^*, ξ^*) asymptot. stable equilibrium,
 $h_r(\alpha_e^*) = y_r^*$

Solv. eq:

$$\begin{cases} 0 = f(\alpha_e^*, u^*) \\ y_r^* = h_r(\alpha_e^*) \end{cases}$$

Regulator eq.

$$\begin{cases} 0 = g(\xi^*, y_m^*) \\ u^* = \gamma(\xi^*, y_m^*) \end{cases}$$
 ★

Make $\xi = \sigma$, $g(\xi, y_m) = \hat{y}_r$, $\gamma(\xi, y_m) =$
 $\hat{u}^* + K_1(\alpha_e - \hat{\alpha}_e^*) +$
 $K_2\sigma$

★ $\begin{cases} \dot{\sigma} = \hat{y}_r \\ u = \hat{u}^* + K_1(\alpha_e - \hat{\alpha}_e^*) + K_2\sigma \end{cases}$

★ $\begin{cases} 0 = 0 \\ u^* = \hat{u}^* + K_1(\alpha_e^* - \hat{\alpha}_e^*) + \underbrace{K_2\sigma^*}_{\text{---}} \end{cases}$

$\rightarrow K_2\sigma^* = u^* - \hat{u}^* + K_2(\hat{\alpha}_e^* - \alpha_e^*)$ ★

\uparrow \uparrow
 $n_r = n_u$

$$\begin{bmatrix} \dot{\epsilon} \\ \dot{\sigma} \end{bmatrix} = \begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix} \begin{bmatrix} \epsilon - \epsilon^* \\ \sigma \end{bmatrix} - \underbrace{\begin{bmatrix} B \\ 0 \end{bmatrix} K_2 \sigma^*}_{\begin{bmatrix} 0 & -BK_2 \\ 0 & 0 \end{bmatrix}} \begin{bmatrix} f_{HOT}(\epsilon, u) \\ f_{r, HOT}(\epsilon, u) \end{bmatrix}$$

$$= \begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix} \begin{bmatrix} \epsilon - \epsilon^* \\ \sigma - \sigma^* \end{bmatrix} - \begin{bmatrix} f_{HOT}(\epsilon, u) \\ f_{r, HOT}(\epsilon, u) \end{bmatrix}$$

$$u \mid \begin{array}{l} \epsilon = \epsilon^* \\ \sigma = \sigma^* \end{array} = \hat{u}^* + K_2(\epsilon - \hat{\epsilon}^*) + K_2 \sigma$$

$$= \hat{u}^* + K_2(\epsilon^* - \hat{\epsilon}^*) + K_2 \sigma^*$$

$$= u^*$$

The good news is: result if $\begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix}$ Hurwitz,
then K_2 is invertible!

Proofing by contradiction:

Assume $\begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix}$ Hurwitz and K_2 is not invertible.

K_2 is not invertible $\Rightarrow \exists w \neq 0$ such that $K_2 w = 0$

$$\begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix} \begin{bmatrix} 0 \\ w \end{bmatrix} = \begin{bmatrix} BK_2 w \\ 0 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} = 0 \begin{bmatrix} 0 \\ w \end{bmatrix} \rightarrow \text{eigenv}\text{c!}$$

$$\begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix} v = \lambda v \Leftrightarrow \lambda = 0, v = \begin{bmatrix} 0 \\ w \end{bmatrix} \neq 0$$

A cannot be Hurwitz if K_2 is not invertible

All in all, it results

$$\begin{cases} \dot{\sigma} = \tilde{y}_r \\ u = \hat{u}^* + K_1(\text{re} - \hat{\text{re}}^*) + K_2\sigma \end{cases}$$

where K_1, K_2 , such that $\begin{bmatrix} A+BK_1 & BK_2 \\ Cr & 0 \end{bmatrix}$ is Hurwitz,

where $A = \frac{\partial f}{\partial x}(x^*, u^*)$, $B = \frac{\partial f}{\partial u}(x^*, u^*)$, $C_r = \frac{\partial h_r}{\partial x}(x^*)$

is such that $\exists \delta^*$ such that (x^*, σ^*) is a Lyapunov Asymp. Stable equilibrium for the closed loop system.

Hence $\exists \mathcal{O} \subset \mathbb{R}^{n_x} \times \mathbb{R}^{n_y}$ open continuity (x^*, σ^*) such that $\forall (x(0), \sigma(0)) \in \mathcal{O}, \lim_{t \rightarrow \infty} y_r(t) = y_r^*$

No matter how badly we estimate \hat{u}^* and \hat{x}^* , can it affect at all? Not quite really because K_1 and K_2 are based on K_1 and K_2 .

$$\begin{cases} \dot{\sigma} = \tilde{y}_r, \sigma(0) = 0 \\ u = \hat{u}^* + K_1(x - \hat{x}^*) + K_2\sigma \end{cases}$$

\downarrow \downarrow
 u^* x^*

Because we need $\underline{\sigma(0) - \sigma^*}$ small enough
 $\quad \quad \quad = -\sigma^*$

Then, what we can do if we already knew u^* and x^* ?

$$u = u^* + K(x - x^*) \text{ from } K_2\sigma^* = u^* - \hat{u}^* + K_2(\hat{x}^* - x^*)$$

will result:

$$\begin{aligned} u &= u^* + K_1 (\pi_e - \pi_e^*) + K_2 (\sigma - \sigma^*) \\ &= u^* - K_2 \sigma^* + K_1 (\cancel{\pi_e^*} - \pi_e^*) + K_1 (\pi_e - \cancel{\pi_e^*}) + K_2 \sigma \\ &= u^* + K_1 (\pi_e - \pi_e^*) + K_2 (\sigma - \sigma^*) \end{aligned}$$

Because of the integration until these parameters

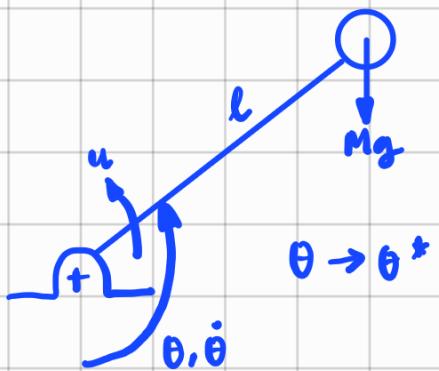
- 1) σ approaches σ^*
- 2) π_e approaches π_e^*
- 3) u approaches u^*

$$\begin{bmatrix} \dot{\pi}_e \\ \dot{\sigma} \end{bmatrix} = \begin{bmatrix} * \\ * \end{bmatrix} \begin{bmatrix} \pi_e - \pi_e^* \\ \sigma - \sigma^* \end{bmatrix} + \begin{bmatrix} f_{HOT}(\pi_e, u) \\ h_{r, HOT}(\pi_e) \end{bmatrix}$$

$$f_{HOT}(\pi_e, u^* + K_1 (\pi_e - \pi_e^*) + K_2 (\sigma - \sigma^*))$$

if we
estimate u^*
and π_e^* wrong
enough, it messes the
HOT!

PENDULUM EXAMPLE



$$u = Mgl \cos \theta^* + K(\alpha e - [\theta^*])$$

$$\alpha e^* = \begin{bmatrix} \theta^* \\ 0 \end{bmatrix} \quad u^* = Mgl \cos \theta^*$$

$$A = \frac{\partial f}{\partial \alpha e} (\alpha e^*, u^*) = \begin{bmatrix} 0 & 1 \\ -\frac{g}{l} \cos \theta^* & -\frac{B}{Ml^2} \end{bmatrix}$$

$$B = \frac{\partial f}{\partial u} (\alpha e^*, u^*) = \begin{bmatrix} 0 \\ \frac{1}{Ml^2} \end{bmatrix}$$

The error equation would be:

$$\begin{cases} \dot{\alpha} = \alpha e_2 - \theta^* \\ u = K_2 (\alpha e - [\theta^*]) + K_3 \alpha \end{cases}$$

$$\hat{u}^* = 0, \quad \hat{\alpha}^* = \alpha e^* = \begin{bmatrix} \theta^* \\ 0 \end{bmatrix}$$

Design K_2 and K_3 so the system is stabilizable.

Check rank $\begin{bmatrix} A & B \\ C_r & 0 \end{bmatrix} = 3$ where $C_r = [1 \ 0]$. full rank.

$$\left[\begin{array}{cc|c} 0 & 1 & 0 \\ -\frac{g}{l} \cos \theta^* & -\frac{B}{Ml^2} & \frac{1}{Ml^2} \\ 1 & 0 & 0 \end{array} \right]$$

→ proved it is full rank

Form a matrix $\begin{bmatrix} A + BK_1 & BK_2 \\ Cr & 0 \end{bmatrix}$

$$= \begin{bmatrix} 0 & 1 & 0 \\ -\frac{g \cos \theta^*}{L} + \frac{K_{11}}{ML^2} & \frac{K_{12} - \beta}{ML^2} & \frac{K_2}{ML^2} \\ 1 & 0 & 0 \end{bmatrix}$$

Find the determinant of eigenval. matrix

$$\det \begin{bmatrix} \lambda & -1 & 0 \\ \frac{g \cos \theta^* - K_{11}}{ML^2} & \lambda - \frac{K_{12} - \beta}{ML^2} & -\frac{K_{22}}{ML^2} \\ -1 & 0 & \lambda \end{bmatrix}$$

$$= \lambda^2 \left(\lambda + \frac{\beta - K_{22}}{ML^2} \right) + \lambda \left(\frac{g \cos \theta^* - K_{11}}{ML^2} \right) - \frac{K_2}{ML^2}$$

$$= \underbrace{\lambda^3 + \frac{\beta - K_{22}}{ML^2} \lambda^2}_{d_2} + \underbrace{\left(\frac{g \cos \theta^* - K_{11}}{ML^2} \right) \lambda}_{d_1} - \underbrace{\frac{K_2}{ML^2}}_{d_0}$$

$$\begin{cases} d_2 > 0, \quad d_1 > 0, \quad d_0 > 0 \\ d_1 \cdot d_2 > d_0 \end{cases} \Rightarrow \text{Real}[\lambda] < 0$$

it's

ROUTH-HURWITZ

Because *) $d_0 > 0, -\frac{K_2}{ML^2} > 0 \Leftrightarrow K_2 < 0 //$

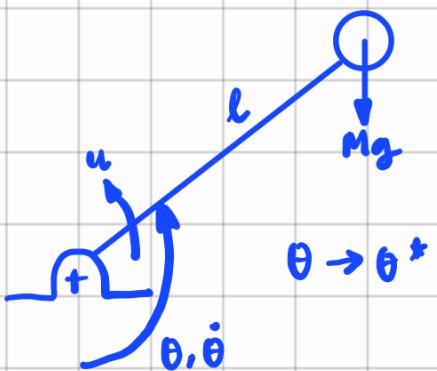
) $\frac{g \cos \theta^ - K_{11}}{ML^2} > 0 \Leftrightarrow K_{11} < M g l \cos \theta^* //$

$$*) \frac{\beta - K_{22}}{M\ell^2} > 0 \iff K_{12} < \beta,$$

Then, $\left(\frac{g \cos \theta^*}{\ell} - \frac{K_{11}}{M\ell^2} \right) \cdot \underbrace{\frac{\beta - K_{22}}{M\ell^2}}_{>0} > \underbrace{\frac{K_2}{M\ell^2}}_{>0}$

This implies K_{11} with any negative number

So, we can fix $K_2 < 0$, $K_{12} < 0$, $K_{22} < 0$ with $|K_{22}|$ is VERY LARGE!



Last time we arrived at

$$\begin{cases} \dot{\theta} = \alpha_2 - \theta^* \\ u = K_{22}(\alpha_2 - \theta^*) + K_{12}\alpha_2 + K_2\theta \end{cases}$$

where $\alpha_1 = \theta$, $\alpha_2 = \dot{\theta}$

We can set $K_2 < 0$, $K_{22} < 0$, $K_{12} < 0$ and $|K_{22}|$ could be LARGE.

$$\rightarrow \begin{bmatrix} A+BK_2 & BK_2 \\ C_r & 0 \end{bmatrix}$$

So, $\sigma^* = \frac{1}{K_2} u^* = \frac{1}{K_2} Mgl \sin \theta^*$

$$e(t) = \tilde{y}_r(t) = \eta_r(t) - \theta^*$$

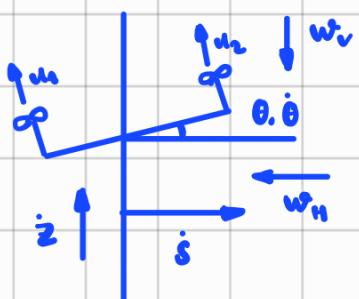
$$\dot{e}(t) = \eta_{\dot{r}}(t)$$

$$\sigma(t) = \int_0^t e(\tau) d\tau$$

$$u = K_{11} e + K_{12} \dot{e} + K_2 \int_0^t e(\tau) d\tau$$

P D I

DRONE EXAMPLE



$$y_r = (\dot{s}, \dot{z}) \rightarrow y_r^* = 0$$

$$\begin{aligned}\eta_1 &= \dot{s} & \eta_3 &= \theta \\ \eta_2 &= \dot{z} & \eta_4 &= \dot{\theta}\end{aligned}$$

The model equation is:

$$u(t) = \begin{bmatrix} \frac{1}{2} \sqrt{w_h^2 + (Mg + w_v)^2} \\ \frac{1}{2} \sqrt{w_h^2 + (Mg + w_v)^2} \end{bmatrix} + K \begin{bmatrix} \eta_1 \\ \eta_2 \\ \eta_3 + \tan^{-1}\left(\frac{w_h}{Mg + w_v}\right) \\ \eta_4 \end{bmatrix}$$

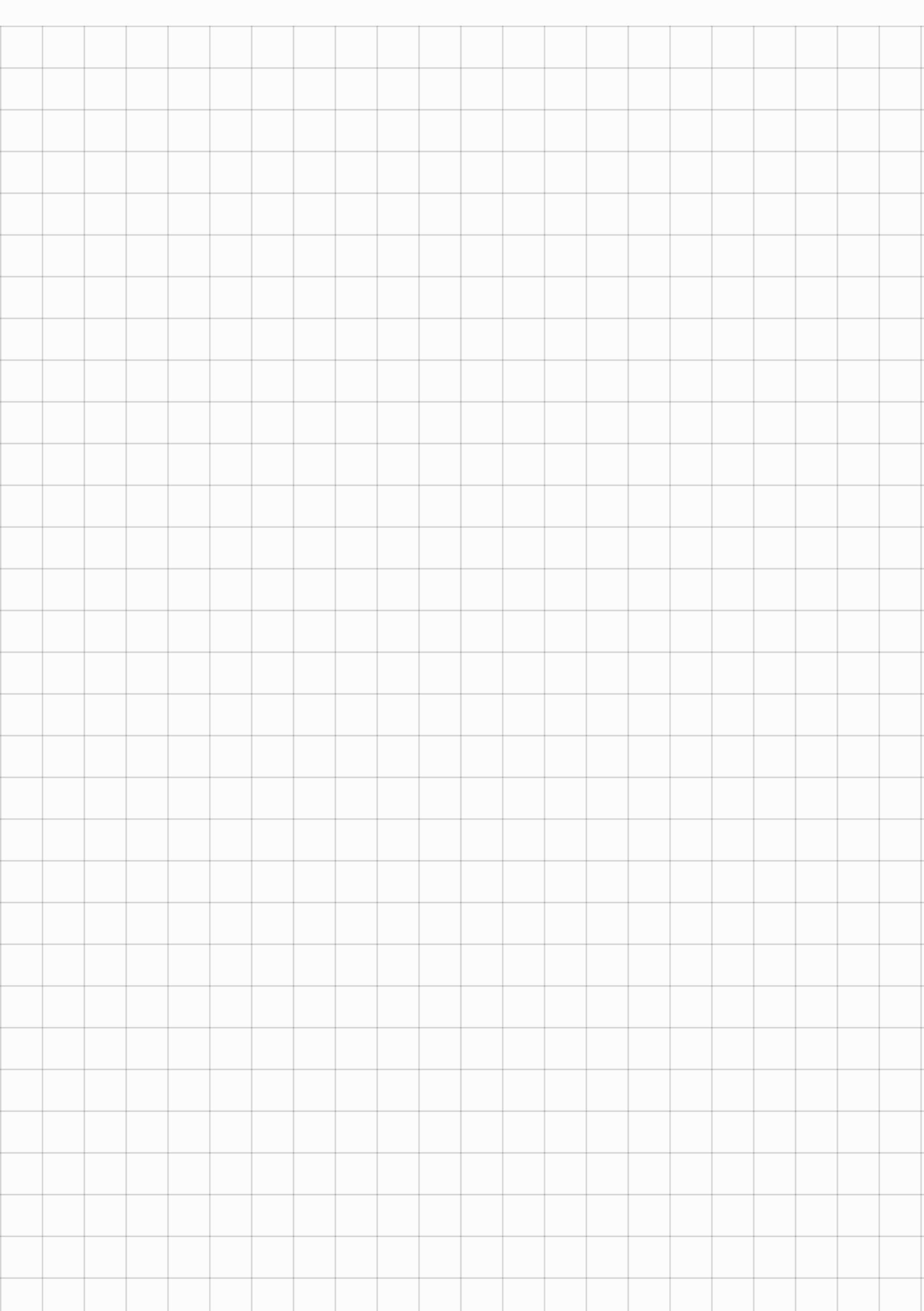
$$\dot{\sigma} = \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix}, \quad \sigma \in \mathbb{R}^2$$

estimate
no wind

$$u = \begin{bmatrix} \frac{1}{2} \hat{M} \hat{g} \\ \frac{1}{2} \hat{M} \hat{g} \end{bmatrix} + K_1 \eta + K_2 \sigma$$

$\uparrow \quad \uparrow$
 $\mathbb{R}^{2 \times q} \quad \mathbb{R}^{2 \times 2}$

$\hat{u}^* = 0$



hence:

$$\begin{pmatrix} \dot{\hat{x}} \\ \dot{e} \end{pmatrix} = \underbrace{\begin{pmatrix} A+BK & BK \\ 0 & A+LC \end{pmatrix}}_{\text{LINEAR PART}} \begin{pmatrix} \hat{x} \\ e \end{pmatrix} + \underbrace{\begin{pmatrix} f_{\text{hor}}(x_m) \\ -f_{\text{hor}}(x_m) - Lh_{m,\text{hor}}(x) \end{pmatrix}}_{\text{HIGHER ORDER TERMS}}$$

The linear part is Hurwitz
(recall the SEPARATION PRINCIPLE)

By Lyapunov indirect theorem we conclude that $(\hat{x}, e) = 0$ is LAS

Thus we conclude that the equilibrium point $(x^*, 0)$ is LAS for (x, \hat{x})

PROBLEMS OF CONTROL VIA LINEARIZATION:

P1) LOCALITY the guarantees are given by Lyapunov's indirect Theorem that, however, only guarantees the existence of an open non-empty domain of attraction, but it does not say anything about its size

sometimes taking large gains K enlarges the domain of attraction. But this is not always the case.



this can be seen in the drone example and in the examples below.

EXAMPLE consider the system

$$\dot{x} = xm + m$$

and let us try to stabilize $x^* = 0$ (requiring $m^* = 0$).
The linear part of the dynamics is defined as

$$A = m^* = 0$$

$$B = x^* + 1 = 1$$

The state-feedback controller is: $u(t) = u^* + K(x(t) - x^*) = Kx(t)$ for some $K < 0$

This leads to

$$\dot{x} = (x+1)Kx = K(x^2 + x) \quad (+)$$

No matter how large is $|K|$, for all $x(0) < -1$ the closed-loop trajectories are divergent.

\Rightarrow we cannot enlarge the domain of attraction

In particular the solutions of (+) are (recall, $K < 0$)

$$x(t) = x_0 \frac{e^{kt}}{x_0 + 1 - x_0 e^{kt}} \rightarrow \text{if } x_0 < -1, \exists \bar{t} = \bar{t}(x_0) > 0 \text{ s.t.} \\ \lim_{t \rightarrow \bar{t}} x(t) = \infty$$

P2. ROBUSTNESS ISSUE The controller depends on the quantities x^* and u^* that are highly sensitive to model uncertainties

To find x^* and u^* we need to solve the solvability Eqs.

$$\begin{cases} 0 = f(x^*, u^*) \\ y_r^* = h_r(x^*) \end{cases}$$

Any slight uncertainty in the knowledge of f , h_m , and h_r may produce some wrong values of x^* and u^*



A control law that depends so critically from x^* and u^* is FRAGILE and its implementation in practice may lead to problems

(TYPICAL PROBLEM OF FEEDFORWARD CONTROL)

EXAMPLE

- In the previous example regarding the actuated pendulum we had:

$$\left\{ \begin{array}{l} x^* = (\theta^*, 0) \\ M^* = Mg l \sin \theta^* \end{array} \right.$$

$\rightarrow M^*$ depends on $\left\{ \begin{array}{l} \cdot \text{mass } M \\ \cdot \text{length } l \\ \cdot \text{gravity } g \end{array} \right.$ ALL UNCERTAIN QUANTITIES!

- In the previous example of the planar drone, we had:

$$\left\{ \begin{array}{l} X_3^* = - \operatorname{atan} \left(\frac{w_H}{Mg + w_V} \right) \\ M_i^* = M_i^* = \frac{1}{2} \sqrt{w_H^2 + (Mg + w_V)^2} \end{array} \right.$$

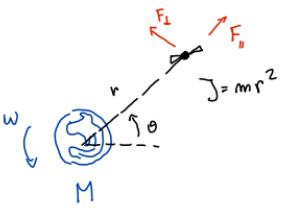
\rightarrow both X^* and M^* depend on the wind (w_H and w_V), the gravity (g) and the mass (M), all uncertain quantities!

Some of the effects of using wrong x^* and M^* are:

- wrong set-point: the state may not converge to x^*
 $\rightarrow \lim_{t \rightarrow \infty} y_r(t) \neq y_r^*$
- destabilization: $x(t)$ may exit the stability domain

EXAMPLE : GEOSTATIONARY SATELLITE STABILIZATION

MODEL (relative to a ref. frame rotating with the Earth)



$$\left\{ \begin{array}{l} m \ddot{r} = - \frac{GmM}{r^2} + m r (\dot{\theta} + \omega)^2 + F_{\parallel}, \\ J \ddot{\theta} = - 2mr \dot{r} (\dot{\theta} + \omega) + r F_{\perp} \end{array} \right.$$

gravity

centrifugal force

thrust

where:

$$G = \text{Gravitational constant} \approx 6.675 \cdot 10^{-11} \frac{\text{m}^3}{\text{kg} \cdot \text{s}^2}$$

$$M = \text{Earth's mass} \approx 5,972 \cdot 10^{24} \text{ kg}$$

$$w = \text{Earth's rotational speed} \approx 2\pi \frac{\text{rad}}{\frac{24}{1} \text{ h}} \approx 7.3 \cdot 10^{-5} \frac{\text{rad}}{\text{s}}$$

GOAL: stabilize the satellite to a fixed angle θ^* relative to the Earth

$$\text{Let } x_1 = r, \quad x_2 = \dot{r}, \quad x_3 = \theta, \quad x_4 = \dot{\theta}, \quad M_1 = F_{\parallel}, \quad M_2 = F_{\perp}$$

Then, we obtain the following model in the state space

$$\begin{cases} \dot{x}_1 = x_2 \\ \dot{x}_2 = -\frac{GM}{x_1^2} + x_1(x_4 + \omega)^2 + \frac{1}{m}M_1 \\ \dot{x}_3 = x_4 \\ \dot{x}_4 = -2\frac{x_2}{x_1}(x_4 + \omega) + \frac{1}{mx_1}M_2 \\ y_r = x_3 \end{cases}$$

with objective

$$y_r^* = \vartheta^*$$

Let us solve the solvability equations:

$$\left\{ \begin{array}{l} 0 = x_2^* \\ 0 = -\frac{Cm}{x_i^*} + x_i^* (x_4^* + w)^2 + \frac{1}{m} M_i^* \\ 0 = x_4^* \\ 0 = -2 \frac{x_2^*}{x_i^*} (x_4^* + w) + \frac{1}{m x_i^*} M_i^* \\ \Theta^* = X_3^* \end{array} \right.$$

These give:

$$\left\{ \begin{array}{l} x_2^* = 0 \\ x_3^* = \Theta^* \\ x_4^* = 0 \\ M_i^* = 0 \\ 0 = -\frac{Cm}{x_i^*} + x_i^* w^2 + \frac{1}{m} M_i^* \end{array} \right.$$

Note: for every value x_i^* for the altitude of the satellite we can find M_i^* for which the problem is solvable

Conversely, given a value M_i^* for the steady-state control action, we can find an altitude x_i^* for which the problem is solvable

To save as much energy as possible let us take $M_i^* = 0$

The solvability conditions give:

$$x_1^* = \sqrt[3]{\frac{GM}{\omega^2}} \approx 42241 \text{ km}$$

At this height, the centrifugal effect balances gravity, and we can keep a satellite in a geostationary orbit "for free"

↓

It is important though to stabilize such an orbit as otherwise any small perturbation would result in a trajectory that either falls toward the Earth, or diverges into deep space.

Let us compute the linearization around $x^* = (x_1^*, 0, 0, 0)$, $M^* = 0$

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 2 \frac{GM}{(x_1^*)^3 + \omega^2} & 0 & 0 & 2x_1^*\omega \\ 0 & 0 & 0 & 1 \\ 0 & -2 \frac{\omega}{x_1^*} & 0 & 0 \end{bmatrix}$$

$$= \begin{bmatrix} 0 & 1 & 0 & 0 \\ 3\omega^2 & 0 & 0 & 2R\omega \\ 0 & 0 & 0 & 1 \\ 0 & -2 \frac{\omega}{R} & 0 & 0 \end{bmatrix}$$

$$B = \begin{pmatrix} 0 & 0 \\ 1/m & 0 \\ 0 & 0 \\ 0 & \frac{1}{m\omega_i^2} \end{pmatrix} = \begin{pmatrix} 0 & 0 \\ 0 & 0 \\ 0 & 0 \\ 0 & \frac{1}{mR} \end{pmatrix}$$

The reachability matrix reads

$$R = \begin{bmatrix} 0 & 0 & \frac{1}{m} & 0 \\ \frac{1}{m} & 0 & 0 & \frac{2\omega}{m} \\ 0 & 0 & 0 & \frac{1}{mR} \\ 0 & \frac{1}{mR} & -\frac{2\omega}{mR} & 0 \end{bmatrix} \quad \dots \quad \dots$$

B AB

we can stop here as this is full rank

rank $R = 4 \Rightarrow$ the system is controllable

Hence, we can find K so as $A+BK$ is Hurwitz, or a state-feedback control law is

$$u(t) = K(x(t) - x^*) \quad \text{with} \quad x^* = \begin{pmatrix} \sqrt[3]{\frac{G}{m}} \\ 0 \\ 0 \\ 0 \end{pmatrix}$$

Can we stabilize the geostationary orbit via OUTPUT-FEEDBACK from the output $y_m = \theta$?

In this case we have

$$C = \begin{bmatrix} 0 & 0 & 1 & 0 \end{bmatrix}$$

and the observability matrix is

$$O = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & -\frac{2w}{\pi} & 0 & 0 \\ -\frac{6w^3}{\pi} & 0 & 0 & -4w^2 \end{bmatrix}$$

O is full-rank! so we can build a local output-feedback control system as detailed by the theory:

$$\begin{cases} \dot{\hat{x}} = (A + BK + LC) \hat{x} - L y_m \\ M = K \hat{x} \end{cases} \quad (y_m^* = x_3^* = 0)$$

where L is such that $A + LC$ is Hurwitz

EXERCISE: find a possible value for K and L

