

# Characterizing Heat Transfer From Impinging Jet Flow

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**Group TR9**

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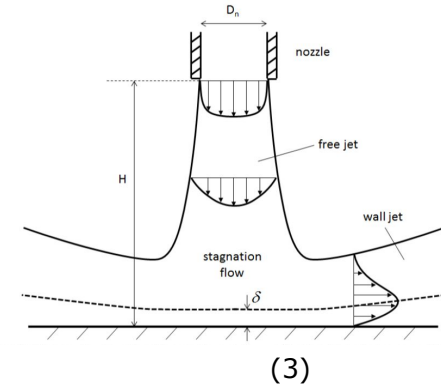
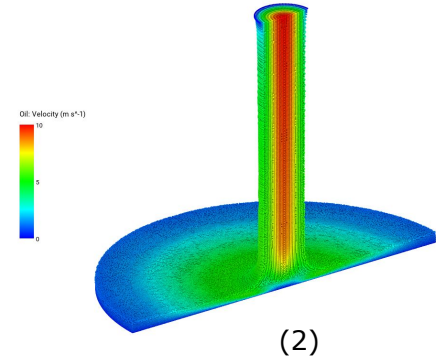
# Impinging Jet Flow Paramount in Industry

## Industrial Applications:<sup>1</sup>

- Temperature control and core cooling in nuclear power plants and foundries
- Temperature regulation in electronic devices

## Objectives:

- Determine how the heat-transfer coefficient,  $h$ , scales with jet velocity,  $v$
- Compare experimental heat transfer coefficient with theoretical values

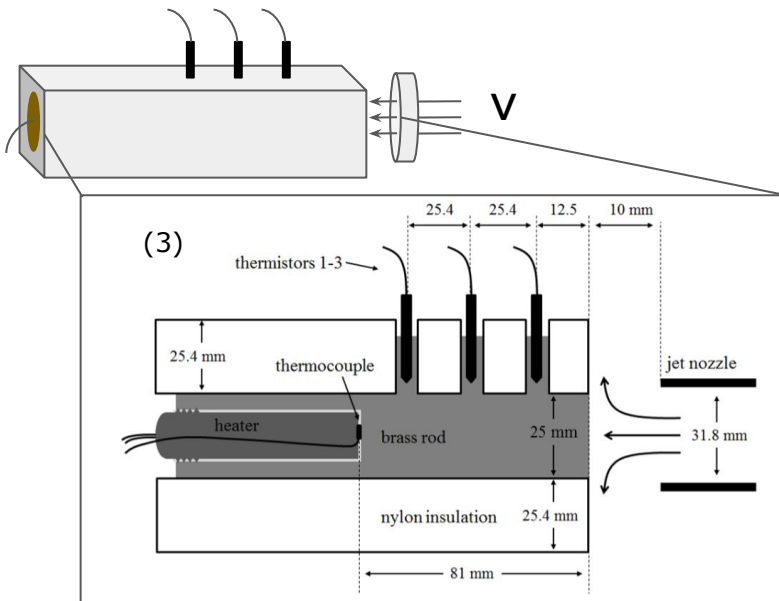


(1) Mills A.F., *Basic Heat and Mass Transfer*; Prentice Hall Inc.: Upper Saddle River, NJ, 1999.

(2) Wendling, L., & Marathe, S.. *Impinging jet benchmark for E-motor cooling applications*. 2010. <https://www.fifty2.eu/innovation/impinging-jet-benchmark-for-e-motor-cooling-applications/> (accessed March 6, 2025).

(3) Behesti-Pour, A. Katz. *Heat Transfer (HT) Revised 2025.docx*; University of California, Berkeley: Berkeley, California, 2025.

# Experimental Methods Uses Brass Rod Apparatus



$$(3) \quad h = \frac{-k_{\text{brass}} \left( \frac{dT}{dz} \right)}{(T_{\text{surface}} - T_{\text{air}})}$$

## Independent Variable

- Jet Velocity

## Constants

- Heater Temperature
- Air Temperature
- Fluid & Material Properties

## Dependent Variables

- Thermistor Temperatures

## Calculate

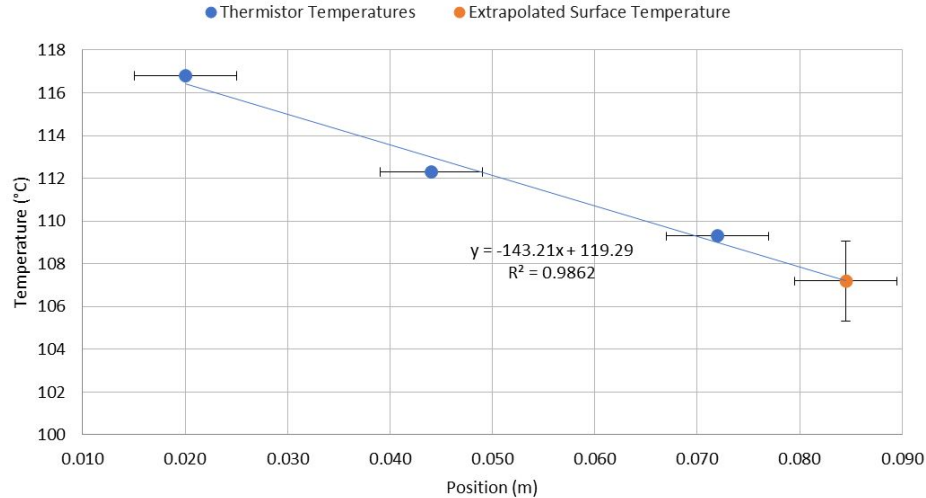
- Temperature Gradient (linear)
- Surface Temperature



Experimental  
Convective Heat  
Transfer Coefficient

# Determining Surface Temperature and Gradient

Temperature vs Position for  $v = 2.38 \pm 0.07$  m/s



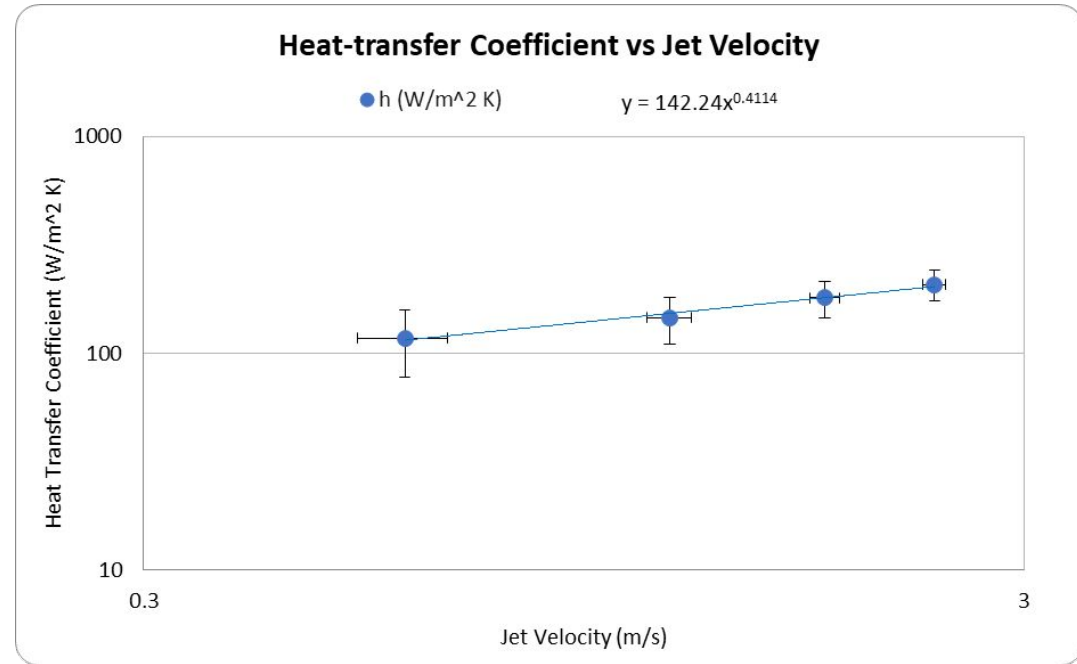
Jet Velocity (m/s)	dT/dz (C/m)	Surface Temperature (C)
$2.38 \pm 0.07$	$-140 \pm 20$	$107.2 \pm 1.9$
$1.78 \pm 0.07$	$-130 \pm 20$	$110 \pm 2$
$1.19 \pm 0.07$	$-110 \pm 30$	$112 \pm 2$
$0.59 \pm 0.07$	$-90 \pm 30$	$115 \pm 2$

Extrapolate to surface,  $x = 0.085$  m

# Convective Heat-transfer Coefficient scales with $v^{0.41}$

$$(3) \quad h = \frac{-k_{\text{brass}} \left( \frac{dT}{dz} \right)}{(T_{\text{surface}} - T_{\text{air}})}$$

Jet Velocity (m/s)	h (W/m <sup>2</sup> K)
2.38 ± 0.07	210 ± 30
1.78 ± 0.07	180 ± 30
1.19 ± 0.07	150 ± 40
0.59 ± 0.07	120 ± 40



**From Experimental Data:  $h = 142v^{0.41 \pm 0.04}$**

# Theoretical Heat Transfer Coefficient Model Predicts $\nu^{0.5}$

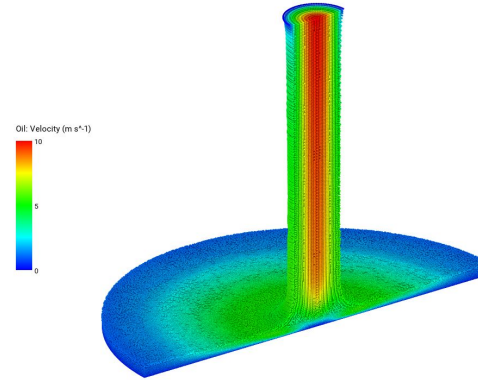
$$(4) \quad \frac{Nu}{Re^{1/2}Pr^{1/3}} = a_1 \left(\frac{z}{d}\right)^{-0.11} \left(1 - \frac{\left(\frac{r}{d}\right)^2 \left(\frac{z}{d}\right)^{-0.2}}{b_1}\right)^{1.2}$$

$$Nu_{(r/D=0)} = a_1 Re^{1/2} Pr^{1/3} \left(\frac{z}{d}\right)^{-0.11}$$

z/d	0.5	0.75	1.0	2.0	3.0	4.0	6.0	8.0
a <sub>1</sub>	5.3	5.1	4.6	3.6	3.2	3.2	2.9	2.3

$$Nu = \frac{hD_{\text{nozzle}}}{k_{\text{air}}} \quad Pr_{\text{air}} = \frac{\nu}{\alpha} = \frac{c_p \mu}{k} = 0.707$$

(2)



## Independent Variable

- Jet Velocity

## Constants

- Air Fluid Properties
- z/d
- r/d
- a<sub>1</sub>

## Dependent Variables

- Nu

## Calculate

- h



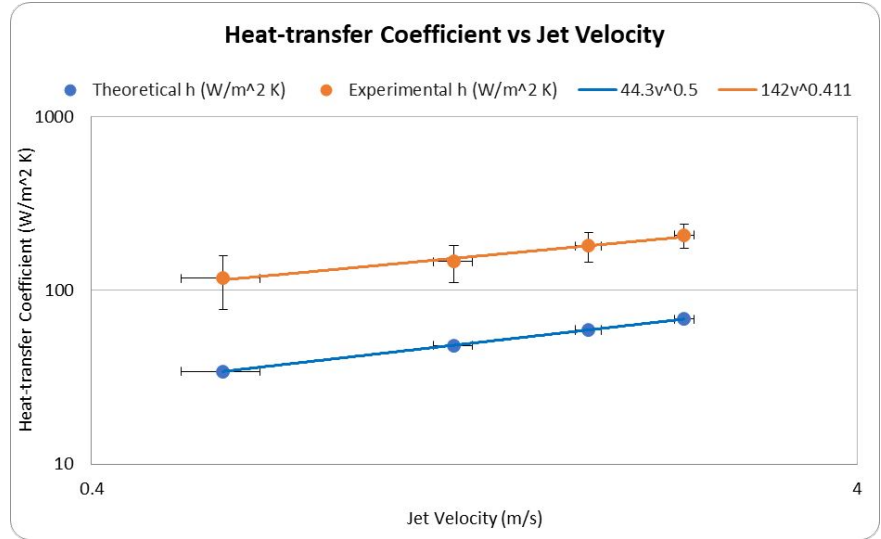
Theoretical Convective  
Heat Transfer  
Coefficient

(2) Wendling, L., & Marathe, S.. *Impinging jet benchmark for E-motor cooling applications*. 2010.  
<https://www.fifty2.eu/innovation/impinging-jet-benchmark-for-e-motor-cooling-applications/> (accessed March 6, 2025).

(4) Katti, V.; Prabhu, S. V. Experimental Study and Theoretical Analysis of Local Heat Transfer Distribution Between Smooth Flat Surface and Impinging Air Jet From a Circular Straight Pipe Nozzle. *Int. J. Heat Mass Transfer* 2008, *51* (17–18), 4480–4495. DOI: 10.1016/j.ijheatmasstransfer.2007.08.031.

# Theoretical Model Underestimates Experimental h

Jet Velocity (m/s)	Predicted h (W/m <sup>2</sup> K)
2.38 ± 0.07	68.3 ± 0.3
1.78 ± 0.07	59.1 ± 0.4
1.19 ± 0.07	48.3 ± 0.5
0.59 ± 0.07	34.1 ± 0.7



**From Experimental Data:  $h = 142v^{0.41 \pm 0.04}$**

**From Theoretical Predictions:  $h = 44.3v^{0.5}$**

Thermal boundary layer thickness decreases with  $v$ , resulting in increased heat transfer

# Radial Heat Transfer Coefficient larger than h

Volumetric Flow Rate (scfm)	U (W/m <sup>2</sup> /K)	h (W/m <sup>2</sup> /K)
4	713.31	207.82
3	701.98	180.22
2	689.11	146.10
1	649.49	117.41

$$q_{\text{heater}} = q_{\text{radial}} + q_{\text{axial}}$$
$$q_{\text{heater}} = \frac{P}{A_{\text{heater}}} \quad q_{\text{axial}} = \frac{T_{\text{surface}} - T_{\infty}}{\frac{L}{k} + \frac{1}{h}}$$
$$U = \frac{q_{\text{heater}} - q_{\text{axial}}}{T_{\text{surface}} - T_{\text{air}}}$$

## Possible reasons:

1. Insulation does not fully cover the rod.
2. Alternate conductive paths that bypass insulation.
3. Systematic errors when measuring temperatures.
4. Poor Nylon insulation efficiency at high temperatures.



# Axial Heat Flux Becomes More Prominent As Flow Rate Increases

Volumetric Flow Rate (scfm)	Radial Heat Flux (%)	Axial Heat Flux (%)
4	83.22	16.78
3	84.99	15.01
2	86.74	13.26
1	87.79	12.21

Thermal boundary layer thickness decreases with  $v$ , resulting in increased heat transfer

Volumetric Flow Rate (scfm)	$q_{\text{axial}} + q_{\text{radial}}$ (W/m <sup>2</sup> )	$q_{\text{heater}}$ (W/m <sup>2</sup> )	Difference (%)
4	89839.78	91953.92	2.32
3	91062.09	92724.85	1.80
2	87802.59	88950.29	1.29
1	80265.02	81041.26	0.96

The unaccounted energy loss could be due to **radiation** or **conduction losses** from structural supports.

# Conclusions

1. Convective heat transfer coefficient  $h$  increases as jet velocity  $v$  increases with a scaling of  $h = 142v^{0.41 \pm 0.04}$ .
2. The experimental scaling,  $h \propto v^{0.41}$ , is slightly lower than the theoretical correlation,  $h \propto v^{0.5}$ .
3. Radial heat transfer coefficient  $U$  is much higher than  $h$ , suggesting poor insulation performance.  $q_{\text{radial}} > q_{\text{axial}}$ .

## Appendix A: Error Analysis

Derived from: 
$$\delta y = \sqrt{\sum_{i=1}^N \left(\frac{\partial f}{\partial x_i}\right)^2 (\delta x_i)^2}$$

$$\Delta(T_{surface} - T_{air}) = \sqrt{(\Delta T_{surface})^2 + (\Delta T_{air})^2}$$

$$\Delta h = h \sqrt{\left(\frac{\Delta(dT/dz)}{(dT/dz)}\right)^2 + \left(\frac{\Delta(T_{surface} - T_{air})}{(T_{surface} - T_{air})}\right)^2}$$

# Appendix B: Theoretical Model Concepts

(4)

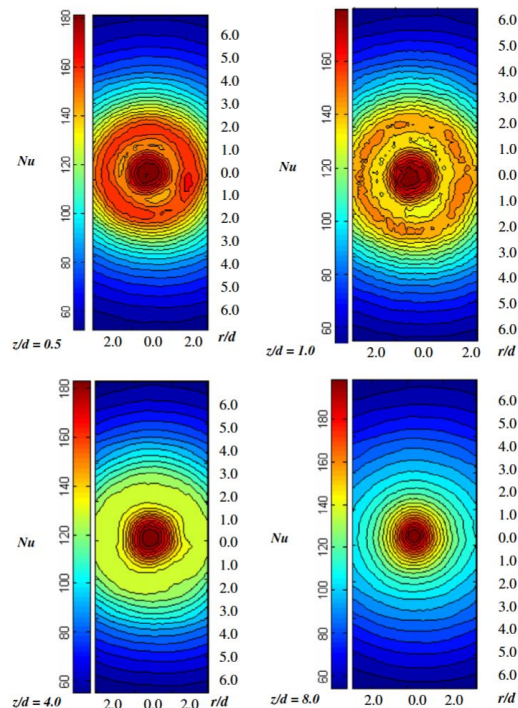


Fig. 6. Distribution of local Nusselt number at  $Re = 28,000$  for  $z/d = 0.5, 1.0, 4.0$  and  $8.0$ .

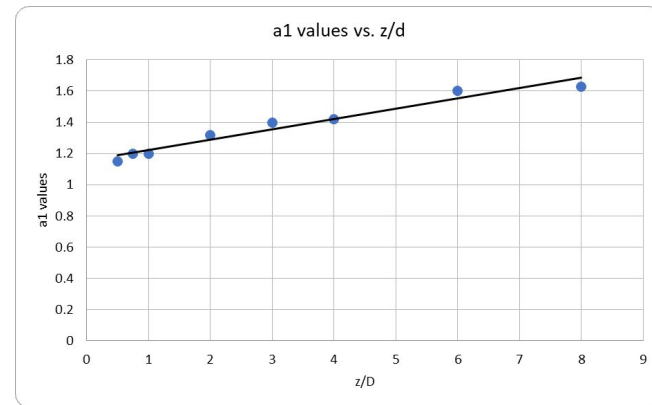


Table 1  
Values of constants  $a_1$  and  $b_1$  for different  $z/d$  used in Eq. (9)

$z/d$	0.5	0.75	1.0	2.0	3.0	4.0	6.0	8.0
$a_1$	1.15	1.2	1.2	1.32	1.4	1.42	1.6	1.63
$b_1$	5.3	5.1	4.6	3.6	3.2	3.2	2.9	2.3

## Appendix C: Definition of Constants

Variable	Value	Description
$k_{\text{brass}}$	120 W/(m*K)	Thermal Conductivity of Brass
$k_{\text{air}}$	0.025 W/(m*K)	Thermal Conductivity of Air
$d$	0.0318 m	Diameter of Jet Nozzle
$\alpha$	$2.062 * 10^{-5} \text{ kg/m}^3$	Thermal Diffusivity of Air
$\mu$	$1.66 * 10^{-5} \text{ Pa*s}$	Kinematic Viscosity of Air
$\rho$	$1.205 \text{ kg/m}^3$	Density of Air
$L$	0.081 m	Effective Length Along the Brass Rod
$z$	0.01 m	Distance Between Jet Nozzle and Brass Rod