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□1. General System Health

☐ A. Inspect all thermal management subsystems for signs of wear or degradation.
☐ Surfaces exposed to thermal cycling, vacuum, or radiation show no signs of visible fatigue
→ If fatigue is observed, document affected zones, disassemble unit for close inspection, and test material strength before component replacement
☐ No discoloration can be seen
→ If discoloration is present, analyze for UV/radiation exposure effects and thermal oxidation; replace affected parts
☐ No cracking or surface fractures
→ If cracks are found, perform dye-penetrant or ultrasonic inspection; remove and replace component if crack depth exceeds safety threshold
☐ No corrosion or oxidation is visible
→ If corrosion is detected, clean using approved solvent method, inspect surrounding materials for electrochemical damage, and replace if pitting or thinning is present
☐ B. Inspect connectors for dust and security
☐ All mechanical and electronic connections are secure
→ If any connection is loose, torque to spec and verify retainer clips or fasteners are engaged
☐ No loose connectors

→ If movement is detected, inspect mating hardware and re-secure with thread-lock or anti-vibration clips
☐ No bent or misaligned pins
→ If bent pins are detected, attempt careful realignment under scope; replace connector if alignment fails or damage is permanent
☐ All connections inspected and confirmed free of dust/debris
→ If contamination is found, clean connector with antistatic brush or dry nitrogen. Replace if pin-to-socket contact resistance remains elevated
☐ Double-checked electrical connectors
→ If mismatch or incorrect mating found, cross-check cable routing, confirm part numbers, and update harness diagram if necessary
☐ Harnesses are clean and undamaged
→ If abrasion or cuts are found, replace harness or apply aerospace-rated sleeve. Validate insulation resistance
☐ Mounting brackets and mechanical fixtures are debris-free
→ If debris is observed, remove using cleanroom vacuum and inspect for fastening looseness or material delamination
☐ Dust/debris inspection completed using at least one of the following tools:
☐ Optical Microscope ☐ Scanning Floatron Microscope (SEM)
☐ Scanning Electron Microscope (SEM)☐ Atomic Force Microscope (AFM)
☐ Confocal Scanning Microscope
☐ Particle Impact Sensor
☐ Surface Acoustic Wave Sensor (SAW)
☐ Electrostatic Dust Sensor
→ If any instrument detects localized dust >10 μg/cm², perform controlled wipe-down and re-scan. Consider ionizing bar if buildup recurs

□ P:	articulate contamination confirmed below 10 μg/cm² threshold
	If threshold exceeded, pause all thermal operations, log contamination vent, and initiate cleaning protocol with documentation
	liagnostics on printer operational temperature ranges liagnostic test run conducted on onboard printer system
	If diagnostic fails to launch or crashes, validate firmware integrity and neck sensor communications
	☐ Nozzle operating temperature recorded
	→ If temperature reading exceeds spec or deviates >±3°C from model, recalibrate nozzle thermistor and inspect heater cartridge
	☐ Electronics and control board temperatures monitored
	→ If board temperature exceeds margin, verify cooling fan or spreader operation and inspect thermal paste
	☐ Motor temperatures measured during full duty cycle
	→ If motor temperature trends high, inspect for over-torque, drive board overcurrent, or insufficient ventilation
□ A	ll components remain within validated thermal margins
he	If any component nears thermal limit, reduce duty cycle, verify eatsink contact and airflow paths, and inspect for thermally degraded isulation
	ensor data matches thermal model predictions within ±3°C
	If sensor readings deviate from model, confirm thermal model oundary conditions and re-validate sensor calibration
□N	o component exceeds 80% of rated thermal limit
	If limit is exceeded, shut down operation, perform post-cooldown aspection, and revise control thresholds to include earlier warnings

□2. Passive Radiative Cooling Components

fy emissivity of external radiators and coating integrity Measure surface emissivity
→ If emissivity < 0.85 over >5% of surface, clean surface, inspect coating degradation, and consider recoating with approved high-emissivity paint (e.g., Z-93, AZ-93)
☐ Use calibrated IR thermography
→ If scan shows inconsistent readings, recalibrate IR sensor and verify emissivity input settings match material specs
\square Emissivity ≥ 0.85 across $\ge 95\%$ of surface area Scan surface for integrity
→ If micro-abrasions or delamination found, log area, compare to wear limits, and assess need for resurfacing or recoating
 ☐ Use 3D profilometry or optical reflectometry ☐ Confirm no micro-abrasion or coating delamination
→ If corrosion or discoloration appears, take spectroscopic sample and consider layer removal for deeper evaluation
Visual inspection under magnification (≥10x) ☐ No corrosion, discoloration, or peeling observed Tools Required
→ If any tool is nonfunctional or miscalibrated, halt inspection, notify metrology team, and swap for certified backup unit

☐ FLIR T1030sc or equivalent
☐ Keyence LJ-V7000 or equivalent
☐ B. Check radiator orientation and deployment mechanisms (if adjustable)
☐ Test deployment in TVAC chamber
→ If deployment time >30s or mechanism jams, inspect actuator geartrain lubricate moving joints, and verify motor torque spec
☐ Simulate thermal cycles: -100°C to +120°
→ If mechanical resistance increases at extremes, review thermal expansion tolerances and requalify mechanism materials
☐ Full deployment achieved in < 30 seconds ☐ Verify orientation and alignment
 ☐ Angular positioning accuracy within ±1.0° ☐ Use optical encoders or LIDAR tracking
→ If deviation >±1.0°, recalibrate encoders or realign deployment arms. If persistent, inspect hinge or encoder backlash
☐ Cycle test actuators
☐ Complete 100 deployment cycles without failure
→ If failure occurs before 100 cycles, isolate failed part, analyze fatigue/friction wear, and replace with high-cycle-rated component
☐ Tools Required
☐ Thermal Vacuum Chamber
☐ Leica Tracker / Encoder system
\square C. Inspect for micrometeoroid impact damage or surface contamination
☐ Inspect for micrometeoroid impact or structural damage
\square No punctures, pitting, or cracks > 0.3 mm
→ If damage >0.3 mm is found, record impact coordinates, assess structural impact, and apply bonded patch or replace panel
☐ Dust/contamination test
☐ Perform white-glove wipe and analyze particles per ISO 14644-1

☐ Dust level conforms to ISO Cla	ss 5
→ If results exceed ISO Class 5 inspect upstream dust control, a	_
☐ Color and surface reflectance test	
☐ Use spectrophotometer	
\square $\Delta E < 2.0$ from original standard	1
	reflectance over time, and verify if Replace if reflective loss exceeds
☐ D. Confirm thermal contact between heat so	ources and radiative elements
☐ Check heat transfer efficiency	
☐ Measure thermal contact resista	$nce < 0.1 \text{ K} \cdot \text{cm}^2/\text{W}$
☐ Use differential thermocouples	or heat flux sensors
	> 0.1 K·cm²/W, reapply or replace ping force, and repeat measurement
☐ Inspect thermal bonds	
☐ Use ultrasound or X-ray NDT	
☐ Confirm \geq 95% bond area cover	rage
\square No voids or delaminations > 2 r	nm
→ If bond area coverage < 95%	, disassemble and reapply adhesive
	must be patched or fully re-bonded
☐ Confirm use and condition of TIM	
Verify proper application and the	ickness
☐ Check for uniformity under visu	ual or infrared scan
→ If TIM layer is non-uniform surfaces, and reapply new TIM (e.g., shim gauge or film uniform	

□3. Pumped Fluid Loop (Low Boiling Point Fluids)

☐ A. Inspect flui	id integrity; check for discoloration, particulate contamination,
or phase sepa	ration
☐ Visual:	inspection of fluid sample
	No discoloration, clouding, or visible degradation
	No particulate matter visible under 10x magnification
	→ If discoloration or particulates detected, filter sample and test full loop. If repeat contamination appears, flush and refill with certified fluid batch
☐ Chemic	eal stability check (e.g., GC-MS or FTIR analysis)
	No unexpected molecular breakdown detected
	Two unexpected morecular orealization activities
	→ If breakdown products detected, isolate affected loop section, verify compatibility of wetted materials, and replace degraded fluid
☐ Phase u	niformity
	No visible phase separation after 24-hour storage at operational temperature extremes (0°C to 60°C)
	→ If phase separation occurs at temperature extremes, remove sample for analysis, verify shelf life, and replace with reformulated blend if needed
☐ B. Test pump	functionality and verify stable flow rates
☐ Measur	re steady-state flow rate
	Target: $10 \text{ mL/min} \pm 5\% \text{ under } 0.16 \text{ G}$
	Tool: Inline ultrasonic flow meter or Coriolis flow sensor
	→ If flow deviates >±5%, inspect pump inlet for blockage, check impeller or rotor condition, and recalibrate flow meter
☐ Motor o	efficiency and noise check
	Electrical draw within spec (e.g., <5W)

	coustic/vibration signature within baseline range (recorded uring ground testing)
	→ If electrical draw exceeds spec or acoustic signature changes, aspect bearing wear, rotor balance, and pump alignment
☐ Thermal	test
	Confirm pump maintains performance between -20°C to +60°C
in	→ If performance degrades at thermal extremes, review pump isulation, motor driver tuning, and inspect for thermal expansion interference
☐ C. Check seals.	valves, and joints for leakage or pressure loss
☐ Visual an	nd dye-enhanced inspection of seals, joints, and valves to visible leakage or moisture accumulation
	→ If leakage is detected, mark location, remove component, and eplace gasket or seal material. Log part number and incident time
☐ Pressure	test system
□ н	fold pressure at 150% nominal operating pressure for 30 minutes to pressure drop >2% during test
	→ If pressure drops >2% in test, isolate leak zone using segmental alve closure, retighten joints, and repeat pressurization
☐ Electroni	c leak detection (optional)
	se gas sniffer or sensor if integrated
	→ If gas sensor triggers, trace concentration gradient, increase entilation, and consider installing localized leak capture features
☐ D. Microgravity	y readiness and cavitation monitoring
☐ Simulate	d reduced gravity testing
	un system on parabolic flight or validated ground-based lunar ravity simulator
☐ Check fo	r cavitation / vapor lock indicators
□N	o air bubbles detected in return lines
\square N	o drop in pressure head or irregular pump RPM

	No spiking in acoustic/vibration signature
	\rightarrow If cavitation occurs, review fluid line routing, increase backpressure valve settings, and consider gas purge protocol prior to ops
☐ E. Redundan	cy and failover verification
☐ Backuj	p pump test
	Confirm backup unit activates within <2 seconds after primary pump shutdown
	Flow continues within nominal range after switch
	→ If backup activation exceeds 2s or flow is insufficient, test relay actuation and verify controller switching logic. Replace failed motor or switch
☐ Passive	e failsafe function (if present)
	Validate thermal siphon or gravity-assisted loop as fallback mode
	→ If thermal siphon fails to activate, test for loop blockage or height differential shortfall. Adjust geometry or verify fluid properties under gravity level
☐ Switch	-over protocol test
	Simulate fault event and log automatic recovery behavior Control system logs and sensor feedback confirm successful handoff
	→ If automatic recovery does not occur, inspect fault tree logic in control software, validate sensor inputs, and verify override pathways

□4. Phase Change Materials (PCMs)

☐ A. Verif	Ty PCM reservoir containment and structural integrity
	Inspect all PCM containers post-thermal cycling for cracks, bulging, seal deformation, or material fatigue
	→ If deformation or cracks detected, log failure mode, cut open sample container, and test welds or seam joints. Replace faulty batch
	Use high-sensitivity strain gauges and volumetric displacement sensors to detect micro-expansion due to repeated melt/freeze transitions
	→ If excessive displacement detected, analyze expansion pattern, revise fill volume margin, and apply expansion-tolerant container design
	Use ultrasonic thickness gauges to verify wall uniformity; acceptable deviation: <5% of nominal thickness
	→ If deviation >5%, perform targeted wall scan and correlate with thermal cycles. Reinforce or replace affected reservoirs
	Scan outer surfaces with optical profilometers for delamination or wear-induced surface anomalies
	Confirm no leakage through dye-penetrant or helium mass spectrometry eak test
	→ If dye/helium leak is detected, evacuate chamber, mark leakage point, and destructively examine seal quality. Do not reuse failed units
☐ B. Valid	late phase change behavior
	Compare actual thermal transition points to manufacturer specs using differential scanning calorimetry (DSC)
	→ If deviation >±5%, identify batch code and compare to reference PCM database. Re-test with fresh sample and flag batch for disqualification
	Run controlled heating/cooling cycles in a thermal vacuum chamber replicating lunar day/night extremes (100 K to 380 K)

		→ If enthalpy curve deviates or fails to stabilize, remove PCM for chemical reanalysis and confirm latent heat capacity via calorimeter
		Record temperature-time curves and ensure latent heat absorption/release aligns with predicted enthalpy values (±5% deviation allowed)
		Perform 10-cycle endurance test; phase transition point must not drift more than 2°C over test duration
		\rightarrow If drift >2°C observed, flag material for fatigue, compare against prior test results, and select alternate formulation for longer-duration missions
□ C.	Che	eck thermal interfaces
		Inspect TIM (thermal interface materials) between PCM and adjacent hardware
		→ If full contact not achieved, remove PCM module, clean surfaces, and reapply TIM per layer thickness spec. Ensure uniformity with pressure mapping film
		Confirm full contact area coverage using pressure-sensitive film or IR
		thermal mapping
		Use ultrasound or X-ray NDT to detect voids, inclusions, or dry spots in TIM layer
		→ If voids/inclusions found, assess if critical to thermal path. If so, rebuild interface or reapply compliant TIM with improved spread control
		Verify compressibility, thermal conductivity, and reusability of TIM per ASTM D5470
□ D .	Mo	nitor operational performance (as needed)
		Install Type-K or RTD sensors at PCM input/output junctions
		Log temperature differentials across each cycle to confirm proper heat exchange
		→ If dT falls outside expected range, inspect for thermal blockage, check sensor drift, and confirm heat flow directionality via heat flux mapping
		Cross-reference heat flux data with expected latent energy values (e.g. 200–300 J/g for paraffin-based PCMs)
		→ If lag or phase delay is detected, run in-situ DSC validation or replace PCM brick with tested alternate

	☐ If anomalous thermal lag or phase delay occurs, schedule PCM material replacement or requalification
□5. Eı	nvironmental Insulation and Radiation Shielding
	A. Test system insulation under lunar temperature swings ☐ Simulate full lunar day/night cycles in thermal vacuum environment (≥12-hour duration each)
	→ If cycle fails to complete due to system error, pause test, log anomaly code, inspect vacuum pump, heaters, and chamber seals. Rerun after confirming environmental control stability
	☐ Use internal array of thermocouples to track core-to-surface gradients
	→ If sensors report irregular or missing data, verify sensor placement and calibration. Replace faulty thermocouples, reinitialize data logging software, and repeat measurement run
	\square Insulation must hold internal ΔT within $\pm 5^{\circ}C$ of thermal baseline over full cycle
	$ ightarrow$ If ΔT exceeds threshold, evaluate insulation material interfaces, inspect for voids or compression, and re-model thermal conduction paths
	☐ Conduct cyclic thermal fatigue test (50+ day/night simulations) to verify long-term resilience
	→ If failure occurs before 50 cycles, identify mode (e.g., delamination, brittleness, thermal short). Disassemble insulation stack, analyze failed layer, and test alternate materials or bonding techniques

☐ B. Inspect radiation shielding	
☐ Use onboard dosimeters or TLDs to measure cumulative particle flux (target: <100 mSv/year exposure)	
→ If exposure exceeds threshold, re-evaluate shielding thickness composition, and configuration. Consider augmenting mass shielding or rerouting critical systems to lower-radiation zones	'>
☐ Perform borescope inspection of interior-facing shield surfaces for bubbling, cracking, or ablation	
→ If anomalies are detected, record damage size and location, classify damage type, and assess if local repair is feasible or if fu shield replacement is needed	11
☐ Inspect MLI blankets and embedded radiation deflectors for fiber degradation, separation, or conductivity loss	
→ If defects are found, replace damaged MLI sections, re-crimp stitch fiber interfaces, and verify continuity of conductive layers using multimeter tester	O
☐ Record radiation exposure profile and compare to maximum material do ratings per NASA-STD-6016	S€
→ If material dose limit is approached or exceeded, flag for accelerated life review and replace component or increase shielding before next exposure campaign	
☐ C. Simulate extended exposure cycles	
☐ Run chamber simulations up to 672 hours continuous operation (28 luna days)	r
→ If chamber operation halts or system fails, log shutdown timestamp and condition, inspect thermal control software for watchdog or thermal runaway errors	
☐ Observe for time-lagged temperature drift, material creep, or degradation of insulation layer reflectivity (via hemispherical reflectance meter)	1

	→ If reflectivity drops >15% or temperature drift exceeds threshold, remove insulation sample and conduct surface analysis (SEM, spectrometry). Replace layer or apply recoating
	IR thermography to detect loss of thermal homogeneity (>±7°C is idered failure)
	→ If nonuniformity is observed, localize hotspots or cold spots, examine thermal interfaces, and enhance thermal spreaders or insulation continuity in affected areas
☐ D. Validate	shielding response to particle events
	ze active particle counters to capture transient solar or GCR events
	→ If particle count spikes above expected levels, log event timestamp, correlate with solar data, and run real-time shielding stress mode
☐ Verif	by that critical systems maintain function without thermal breach
	→ If system behavior degrades, switch to backup thermal path (if available), initiate safe mode, and diagnose control loop integrity. Log temperatures and review logs for command execution errors
	nine for increased local surface temperatures or structural ploration near shielding faults
	→ If discoloration or hotspots are found, halt test, photograph affected regions, and sample material if possible. Assess for radiation-induced chemical changes or burn-through
	re cumulative shielding thickness >10 g/cm² for SPE (Solar Particle at) compliance
	→ If thickness is insufficient, reinforce shield stack-up with additional material or layered laminates. Recalculate shielding effectiveness using GEANT4 or NASA's OLTARIS tool

\Box 6. Software and Monitoring

	bration and Verification Il thermal sensors are within calibration date per manufacturer
	→ If calibration is expired, remove from service, send for recalibration, or replace with a certified unit
_	cision temperature source (e.g., dry block or fluid bath) to test each t multiple setpoints (e.g., -50°C, 0°C, +50°C)
i	→ If test cannot be completed, check for equipment malfunction or improper sensor contact. Re-seat sensor, confirm bath/stirring uniformity, and repeat test
☐ Compar	e readings to NIST-traceable reference sensor; record deviations
I	→ If deviation exceeds ±0.5°C or specified tolerance, log sensor ID, flag as failed, and remove it from the system. Investigate for drift or physical damage
_	d recalibrate/replace any sensors with deviations $> \pm 0.5$ °C or specified tolerance
ϵ	→ If recalibration fails or is not possible, dispose of the sensor per electronics waste protocols. Install new unit and re-run verification test
☐ Disconn	are and Dropout Simulation sect each sensor manually to simulate dropout; verify system logs d triggers fallback mode
ł	→ If fault not logged or fallback mode fails, inspect firmware error nandling routines. Add diagnostics to detect loss of signal and ensure graceful degradation of control logi

Inject false data or out-of-range values (e.g., -200°C or +200°C) and verify software response
→ If system accepts bad data without flag, update software to include input validation and hard-coded physical limits
Test for sensor lag or slow response using controlled temperature ramps
→ If lag exceeds spec, verify sensor thermal contact and firmware filter settings. Replace sensor or increase sampling rate
trol Algorithm Validation Test PID or custom control logic in simulated environment with thermal hardware model
→ If test fails or system becomes unstable, tune PID gains or refine model fidelity. Add bounds checking, rate limiting, or anti-windup logic to prevent runaway control
Run edge-case scenarios: Rapid external temperature shift (simulate sun/shade transition)
→ If overshoot or instability occurs, adjust response time constants, add feedforward terms, or thermal rate-of-change caps
☐ Sensor dropout or conflicting inputs
→ If software crashes or misbehaves, implement redundancy logic, sensor voting, or fallback default control profile
☐ Coolant pump failure
→ If thermal system exceeds limits, test emergency heat dump or shutdown routine. Verify alerts and ensure thermal margins are adequate under no-flow conditions
Verify system maintains safe thermal range without overshoot or oscillation

	→ If not, revise control loop parameters, consider additional damping, or implement hierarchical control structure with safety overrides
☐ Verify	& Data Logging Accuracy all sensor outputs and system states (valves, pumps, heater status) ged at required intervals (e.g., 1 Hz or mission-specific)
	→ If logging interval is inconsistent or missing data, check for CPU load issues, logger memory overflows, or incorrect task prioritization
☐ Cross-c	check raw logs vs real-time display in control software
	→ If mismatch found, inspect data pipeline from sensor to GUI, validate that buffers and converters (e.g., ADC to engineering units) are synchronized
	logs include timestamps with synchronized time (e.g., GPS time or n clock)
	→ If timestamp drift or loss occurs, inspect time sync protocol (e.g., NTP/GPS), reset system clock, or add watchdog for time integrity
☐ Induce	network latency or dropout; verify data buffering and loss recovery
	\rightarrow If data is lost or corrupted, improve buffer size, implement local storage fallback, and validate retransmission logic on reconnection
☐ Simula full du	te full mission scenario and ensure complete, lossless logging for ration
	→ If logging fails partway, diagnose storage write errors, thermal throttling of SSD, or file system faults. Implement log segmentation with redundancy and backup mechanism
	smission and Integrity
☐ Check per pac	uplink/downlink packet integrity: Run CRC or checksum validation ket

	→ If checksum fails, isolate affected subsystem, increase error correction redundancy, and ensure consistent byte framing
☐ Simula	te transmission interruptions (RF blackout, loss of signal)
	→ If data is lost or not resumed correctly, verify transmission state machine, increase retransmit buffer size, and add re-acknowledgment logic
☐ Verify connect	data retransmission and catch-up logic on re-establishment of etion
	→ If gaps persist, ensure log pointer synchronization and verify start-of-message indicators are not lost. Add persistent queueing and session resumption
	m secure, redundant storage of logs (e.g., onboard SSD + mirror storage)
	→ If redundancy fails, test hardware RAID or replication scripts, verify regular sync schedule, and restore from backup to confirm recoverability
☐ Compa	are transmitted vs received data sets: Ensure byte-for-byte integrity
	→ If mismatch found, trace packet loss locations, compare CRC results, and verify encoding schemes match on sender and received
□7. Emergency	Protocols
☐ Review	cy Plan: Passive-Only Operation v and update thermal models to predict system performance under e-only cooling (no pump operation)

1	→ If model predictions are inaccurate or non-converging, re-validate boundary conditions, heat load assumptions, and mesh fidelity. Cross-check with empirical thermal test data to recalibrate
	nulation of passive cooling under various mission thermal loads ternal heat sources + external fluxes)
(→ If system overheats in simulation, identify critical nodes and determine if heat paths can be improved. Add insulation, adjust radiator orientation, or redistribute heat-generating components
	ine and document survivability duration at each thermal load level naintain <50°C for 6 hours @ 80W load")
	→ If survivability duration is insufficient, flag mission timeline or operating mode for restriction. Modify passive element design (larger radiator, more PCM mass) and re-simulate
	backup passive elements (radiators, heat sinks, phase change ls) for integrity and capacity
(→ If physical degradation is found (e.g., cracked fins, PCM leaks), document failure type, remove component, and replace with verified spare. Re-run passive system validation after repair
☐ B. Low-Power	/ Pump-Out Survival Readiness
	e low-power state (<10% nominal system power) and verify l thermal protection functions remain operational
1	→ If thermal protection fails, identify which subsystem lost function. Evaluate battery reserve allocation and check autonomous transition logic. Reprioritize thermal protection in low-power hierarchy
	for autonomous switch to low-power cooling mode (e.g., PCM iator exposure adjustment)
j	→ If switch does not occur, debug system state logic and sensor inputs. Update firmware to ensure fallback logic initiates at correct thresholds

☐ Set and confirm passively regulation	n thermal load thresholds above which system cannot ate
upgrade	resholds are too low for mission success, explore hardware is (e.g., deployable radiators) or limit thermal output of ary systems
☐ Document shut	down time estimates once thresholds are exceeded
capacity	timates are inconsistent with test data, refine thermal models, validate initial conditions, and rerun failure simulations
☐ C. Automated Therm	al Shutdown & Alerts
•	e temperature monitoring triggers shutdown protocols at sholds (e.g., >75°C on CPU, >50°C on battery pack)
logic, ar	utdown does not initiate, check sensor mapping, software nd relay actuation path. Confirm firmware threshold ts are correctly defined and tested
☐ Test full chain of Component Sho	of action: → Sensor → Software → Shutdown Signal → utdown
	y link fails, isolate component and perform subsystem tics. Log fault, repair or reflash affected module, and re-run on
☐ Inject overheati confirm:	ing scenario in test bench or software simulation and
☐ Proper s	shutdown of non-critical systems
•	→ If critical systems are affected or non-essentials remain online, revise shutdown prioritization matrix in control logic
☐ Emerge	ncy alert transmitted to mission control
	→ If alert is missed or delayed, check communication buffer, telemetry bandwidth limits, and packet priority

☐ Telemetry log includes thermal event details
→ If event is not logged, verify telemetry schema and ensure logging service is not overloaded or improperly formatted
☐ Confirm reset logic: system only resumes when safe conditions are re-established or manual override is issued
→ If reset occurs prematurely, update state machine with stricter condition checks. Implement hysteresis or manual review before restart
☐ D. Failure Mode and Effects Analysis (FMEA)
Review current FMEA every 3 months, focusing on components with observed degradation or failures
→ If review is overdue or incomplete, escalate to systems lead, reschedule immediately, and add overdue reviews to audit logs
☐ Add new failure cases from testing, field data, or mission events
→ If data is missing or incomplete, retrieve anomaly logs from test sessions or mission reports. Assign responsible engineers to retroactively analyze and add to database
☐ Re-score risk levels (Severity × Occurrence × Detection) for all failure modes
→ If risk scoring is inconsistent or outdated, use cross-functional team to validate inputs. Re-assess based on most recent test statistics and mission criticality
☐ Identify high-priority risks and assign mitigation strategies
→ If no strategy exists, flag issue with risk owner and engineering management. Assign temporary mitigation while permanent solution is developed
☐ Verify backup or redundant components for each critical failure mode

perform manual failover check. Log readiness status in FMEA tracking tool ☐ E. Documentation & Mitigation Updates Update thermal operations manual with new test results, edge cases, and any revised procedures → If documentation is out of date, assign responsible author, set deadline for update, and send release notes to relevant teams ☐ Log all failure test outcomes (simulated or real) into anomaly database → If test results are missing from database, backfill data from test reports and ensure future tests use standardized logging format Document all new mitigation strategies, including firmware updates, hardware redundancies, or procedural changes → If not documented, require engineering signoff before mission integration. Review all undocumented fixes during readiness review ☐ Ensure engineering and operations teams are briefed on changes and mitigation timelines → If briefing was missed, organize immediate catch-up meeting or training session. Record attendance and upload briefing slides to central repository

→ If backup system is unverified or inoperable, schedule test or