# Ch4: Linear and Nonlinear Systems

Amin Fakhari, Spring 2025

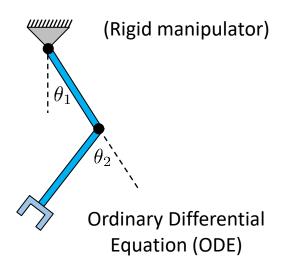
State-Space Representation Linear Systems Nonlinear Systems Equilibrium Points Linearization 000 00000000 0000 0000

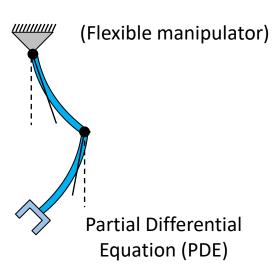
# **System Classifications**

System Classifications

### **System Description (Analytical)**

For analysis and design of control systems, the dynamic systems (i.e., mechanical, electrical, thermal, economic, biological, ...) must be **mathematically modeled** in terms of **differential equations** using fundamental **physical laws** (e.g., Newton-Euler's laws for mechanical systems and Kirchhoff's laws for electrical systems).



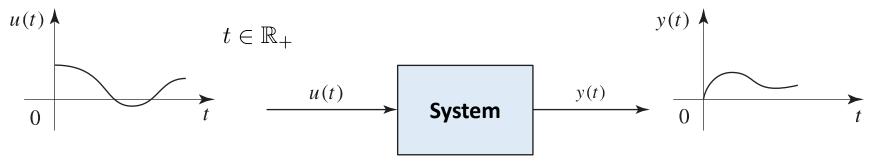


In obtaining a mathematical model, we must make a **compromise** between the **simplicity** of the model and the **accuracy** of the results of the analysis. We may simplify the system model in order to design a relatively simple controller.

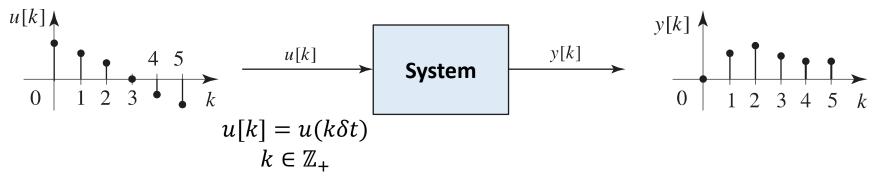
System Classifications

### Continuous-time vs. Discrete-time Systems

A system is called a **continuous-time** system if it accepts continuous-time signals as its input and generates continuous-time signals as its output.



A system is called a **discrete-time** system if it accepts discrete-time signals as its input and generates discrete-time signals as its output.



### Single-variable vs. Multivariable Systems

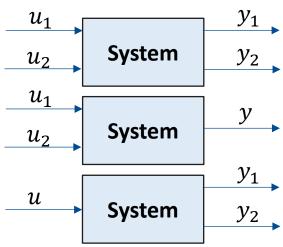
A system with only one input and only one output is called a **single-variable** system or a **single-input single-output** (SISO) system.



A system with two or more inputs and/or two or more outputs is called a multivariable

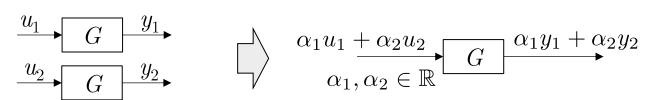
system.

- Multi-input Multi-output (MIMO)
- Multi-input Single-output (MISO)
- Single-input Multi-output (SIMO)



### Linear vs. Nonlinear Systems

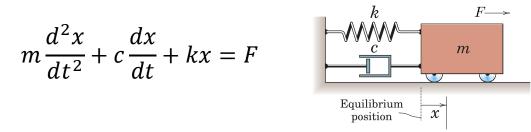
A system is **Linear** w.r.t. its inputs and outputs iff it obeys the **Principle of Superposition**:



Note: A differential equation is linear if the coefficients are constants or functions only of the independent variable.

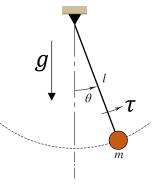
$$a_0(t)y + a_1(t)\dot{y} + a_2(t)\ddot{y} + \dots + a_n(t)y^{(n)} = u(t)$$

$$m\frac{d^2x}{dt^2} + c\frac{dx}{dt} + kx = F$$



A system is **Nonlinear** if the **principle of superposition** does **not** apply.

Example: Pendulum 
$$ml^2 \frac{d^2\theta}{dt^2} + mgl \sin\theta = \tau$$

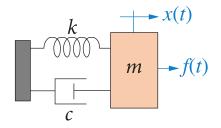


### Time-Invariant vs. Time Varying Systems

A system is said to be **Time-Invariant** (or **Autonomous** for nonlinear systems) if the relationship between the input and output is **independent** of time.

• If the response to u(t) is y(t), then the response to  $u(t-t_0)$  is  $y(t-t_0)$ .

**Ex.** A mass-spring-damper system which its physical parameters remains constant.



A system is said to be **Time-Varying/Varient** (or **Non-Autonomous** for nonlinear systems) if the relationship between the input and output is **dependent** of time.

**Ex.** A spacecraft system which its mass changes due to fuel consumption.

LTI: Linear Time-Invariant LTV: Linear Time-Varying

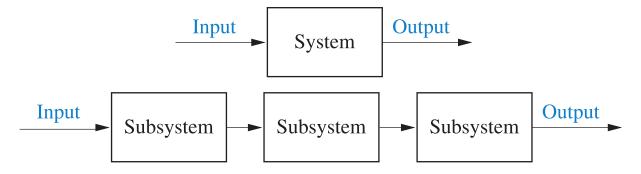


System Classifications

### **System Representation**

The ODE is not a satisfying representation because the system input u(t) and output y(t) appear throughout the equation. It is preferred a mathematical representation which the input, output, and system are separate parts and it can be modeled as a block diagram.

e.g., a SISO LTI system: 
$$\frac{d^n y(t)}{dt^n} + a_{n-1} \frac{d^{n-1} y(t)}{dt^{n-1}} + \dots + a_0 y(t) = b_m \frac{d^m u(t)}{dt^m} + b_{m-1} \frac{d^{m-1} u(t)}{dt^{m-1}} + \dots + b_0 u(t)$$
 
$$n \geq m$$



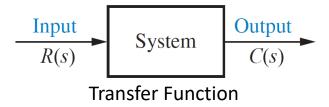
Two methods for representation of mathematical models of dynamic systems:

- (1) Transfer Function (TF) in the Frequency Domain,
- (2) State-Space (SS) Representation in the Time Domain.

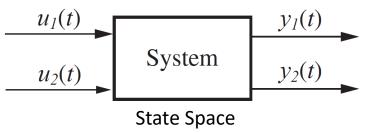


### Classical vs. Modern Control Theory

Classical (or Frequency-Domain or Transfer-Function) approach [Since 1920s] can be applied only to linear, time-invariant (LTI), SISO (Single-Input Single-Output) systems with zero initial conditions, or systems that can be approximated as such. It does not use any knowledge of the interior structure of the system.



Modern (or Time-Domain or State-Space) approach [Since 1960s] can be applied to a wide range of systems including nonlinear, time variant (non-autonomous), MIMO (Multi-Input Multi-Output) systems with nonzero initial conditions and also LTI systems modeled by the classical approach.



## **State-Space Representation**

**System Classifications** 

### **Some Definitions**

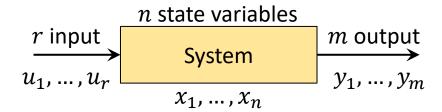
**Linear Combination**: A linear combination of n variables,  $x_i$ , is given by

$$x_{n+1} = k_1 x_1 + k_2 x_2 + \dots + k_n x_n$$
,  $k_i = \text{constant} (i = 1, \dots, n)$ 

**Linear Independence**: A set of variables is said to be linearly independent if none of the variables can be written as a linear combination of the others.

**System Variable**: Any variable that responds to an input or initial conditions in a system.

**State Variables**: The **smallest set of linearly independent** system variables  $(x_1, ..., x_n)$  such that knowledge of these variables at  $t=t_0$ , together with knowledge of the input  $\boldsymbol{u}(t)$  for  $t\geq t_0$ , completely determines the behavior of the system  $\boldsymbol{y}(t)$  for any time  $t\geq t_0$ .



### **State-Space Representation**

**State-space Representation** is a mathematical model of a physical system as a set of input  $u(t) \in \mathbb{R}^r$ , output  $y(t) \in \mathbb{R}^m$ , and state variables  $x(t) \in \mathbb{R}^n$  related by n simultaneous first-order differential equations.

$$\dot{m{x}}(t) = m{f}(m{x}(t), m{u}(t), t)$$
 State Equation  $m{y}(t) = m{g}(m{x}(t), m{u}(t), t)$  Output Equation

f and g are vector functions.

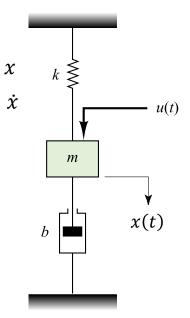


The number of states (n) is the **order** of the system.

State Vector:  $x(t) \in \mathbb{R}^n$ 

**Note**: State variables need not be physically measurable or observable quantities.

**Note**: The choice of state variables of a system is not unique, but the number of states is unique. For all invertible  $T \in \mathbb{R}^{n \times n}$ ,  $\overline{x}(t) = \mathbf{T}x(t)$  can be also the system state variables.



### **State-Space Representation**

#### **General Form:**

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MIMO, Nonlinear, Time Variant (General Form )

$$\dot{\boldsymbol{x}}(t) = \boldsymbol{f}(\boldsymbol{x}, \boldsymbol{u}, t)$$

$$\mathbf{y}(t) = \mathbf{g}(\mathbf{x}, \mathbf{u}, t)$$

MIMO, Linear, Time Variant

$$\dot{x}(t) = A(t)x(t) + B(t)u(t)$$

$$y(t) = C(t)x(t) + D(t)u(t)$$

MIMO, Linear, Time Invariant

$$\dot{x}(t) = Ax(t) + Bu(t)$$

$$y(t) = Cx(t) + Du(t)$$

A: State matrix,

C: Output matrix,

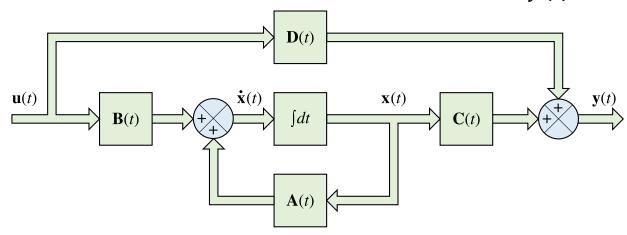
**B**: Input matrix

D: Feedforward matrix

SISO, Linear, Time Invariant

$$\dot{\boldsymbol{x}}(t) = \boldsymbol{A}\boldsymbol{x}(t) + \boldsymbol{B}\boldsymbol{u}(t)$$

$$y(t) = \mathbf{C}\mathbf{x}(t) + \mathbf{D}u(t)$$





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### **State-Space Representation of Linear Systems**

Consider a general, nth-order, linear differential equation with constant coefficients:

$$\frac{d^n x}{dt^n} + a_{n-1} \frac{d^{n-1} x}{dt^{n-1}} + \dots + a_0 x = b_0 u$$

- An nth-order differential equation can be converted to n simultaneous **first-order differential equations**.
- There are many ways to do this conversion and obtain state-space representations of systems, such as **phase-variable** form, **controllable** canonical form, **observable** canonical form, **diagonal** canonical form, and **Jordan** canonical form.

A convenient way to choose state variables is to choose x(t) and its (n-1) derivatives as the state variables, which are called **phase variables**.



### State-Space Representation of LTI Systems

$$\frac{d^n x}{dt^n} + a_{n-1} \frac{d^{n-1} x}{dt^{n-1}} + \dots + a_0 x = b_0 u$$

$$x_{1} = x$$

$$x_{2} = \frac{dx}{dt}$$

$$\vdots$$

$$x_{n} = \frac{d^{n-1}x}{dt^{n-1}}$$

$$\dot{x}_{1} = \frac{dx}{dt}$$

$$\dot{x}_{1} = x_{2}$$

$$\dot{x}_{2} = x_{3}$$

$$\vdots$$

$$\dot{x}_{n-1} = x_{n}$$

$$\dot{x}_{n} = -a_{n-1}x_{n} - \dots - a_{0}x_{1} + b_{0}u$$

$$\dot{x}_{n} = \frac{d^{n}x}{dt^{n}}$$

$$\dot{x}_{n} = Ax + B$$
Vector-Matrix Form
$$\dot{x} = Ax + B$$

$$\dot{x}_{n} = \frac{a \cdot x}{dt^{n}}$$

$$\dot{x}_{n} = \frac{a \cdot x}{dt^{n}}$$

$$x = \begin{bmatrix} x_{1} \\ x_{2} \\ \vdots \\ x_{n} \end{bmatrix}, \quad A = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ \vdots & \vdots & \ddots & \vdots & \vdots \\ 0 & 0 & 0 & \cdots & 1 \\ -a_{0} & -a_{1} & -a_{2} & \cdots & -a_{n-1} \end{bmatrix}, \quad B = \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ b_{0} \end{bmatrix}, \quad C = \begin{bmatrix} 1 & 0 & \cdots & 0 \end{bmatrix}$$
(Output can be the first state)

$$\dot{x}_1 - x_2 
\dot{x}_2 = x_3 
\vdots 
\dot{x}_{n-1} = x_n 
\dot{x}_n = -a_{n-1}x_n - \dots - a_0x_1 + b_0u$$

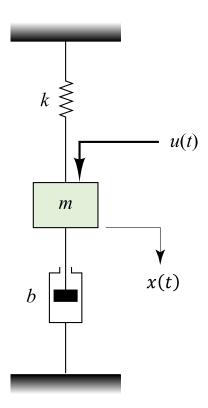
, 
$$extbf{\emph{C}} = [1 \quad 0 \quad \cdots \quad 0]$$
 (Output can be the first state)

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### **Example**

The external force u(t) is the input to the system, and the displacement x(t) of the mass, measured from the equilibrium position in the absence of the external force, is the output. Find the state equations.

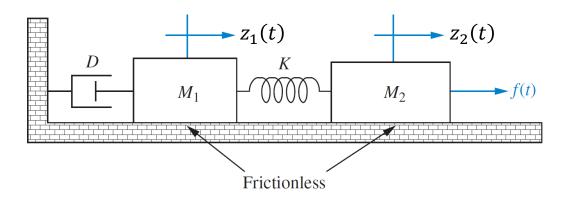


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### **Example**

Find the state equations. What is the output equation if the output is  $z_1(t)$ ?



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**Transfer Function Matrix** 

### Converting from SS to a TF

Deriving the transfer function from the state-space equations:

Laplace transform assuming zero initial conditions 
$$\mathbf{X}(s) = A\mathbf{X}(s) + B\mathbf{U}(s) \rightarrow \mathbf{X}(s) = (s\mathbf{I} - A)^{-1}B\mathbf{U}(s)$$

$$\mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) + \mathbf{D}\mathbf{U}(s) \qquad (\mathbf{I} \text{ is the identity matrix})$$

$$\mathbf{Y}(s) = [\mathbf{C}(s\mathbf{I} - A)^{-1}\mathbf{B} + \mathbf{D}]\mathbf{U}(s)$$

Transfer Function for a SISO system which U(s) = U(s) and Y(s) = Y(s):

$$G(s) = \frac{Y(s)}{U(s)} = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + D$$

### **Example**

Obtain the transfer function Y(s)/U(s) from the state-space equations of the system shown in the previous example.

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\frac{k}{m} & -\frac{b}{m} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u$$

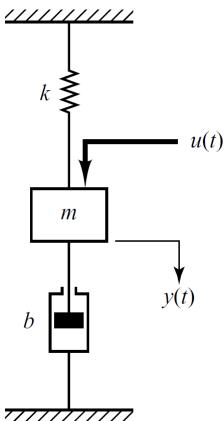
$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

#### Solution:

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$$G(s) = \frac{Y(s)}{U(s)} = C(sI - A)^{-1}B + D$$

$$G(s) = \frac{1}{ms^2 + bs + k}$$



### **Converting a TF to SS**

To convert a transfer function into state-space equations in phase-variable form, first convert the transfer function to a **differential equation** by cross-multiplying and taking the inverse Laplace transform, assuming zero initial conditions.

$$\frac{Y(s)}{U(s)} = \frac{b_0}{s^n + a_{n-1}s^{n-1} + \dots + a_0} \longrightarrow \frac{d^n y}{dt^n} + a_{n-1}\frac{d^{n-1}y}{dt^{n-1}} + \dots + a_0 y = b_0 u$$

Then, convert this nth-order differential equation to n simultaneous first-order differential equations.

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### **Example**

Find the state-space representation in phase-variable form.

#### **Solution:**

$$\frac{R(s)}{s^3 + 9s^2 + 26s + 24} \qquad C(s)$$

$$(s^3 + 9s^2 + 26s + 24)C(s) = 24R(s)$$

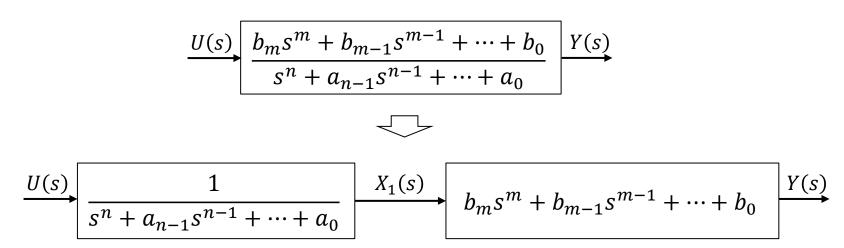
$$\ddot{c} + 9\ddot{c} + 26\dot{c} + 24c = 24r$$

$$x_1 = c$$
  
 $x_2 = \dot{c}$   
 $x_3 = \ddot{c}$   
 $\dot{x}_1 = x_2$   
 $\dot{x}_2 = x_3$   
 $\dot{x}_3 = -24x_1 - 26x_2 - 9x_3 + 24r$   
 $y = c = x_1$ 

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -24 & -26 & -9 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 24 \end{bmatrix} r \qquad y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

### **Converting a TF to SS**

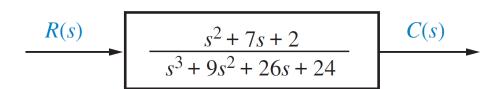
If a transfer function has a polynomial in *s* in the numerator, separate the transfer function into two cascaded transfer functions; the first is the denominator and the second is just the numerator.



- The first transfer function with just the denominator is converted to the phase-variable representation in state space.
- The second transfer function with just the numerator yields the output equation.

### **Example**

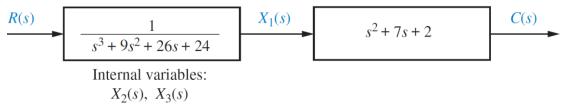
Find the state-space representation of the transfer function.



#### **Solution:**

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From previous example:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -24 & -26 & -9 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} r$$

$$C(s) = (s^2 + 7s + 2)X_1(s)$$

$$x_{1} = x_{1}$$

$$\dot{x}_{1} = x_{2}$$

$$\ddot{x}_{1} = x_{3}$$

$$\ddot{x}_{1} = x_{3}$$

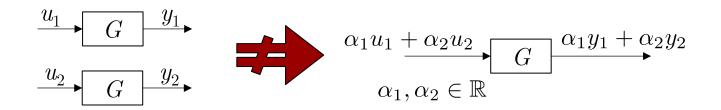
$$y = c(t) = x_{3} + 7x_{2} + 2x_{1} \longrightarrow y = \begin{bmatrix} 2 & 7 & 1 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ x_{3} \end{bmatrix}$$



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### **Nonlinear Systems**

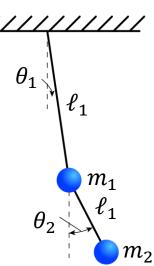
A system is nonlinear if the **principle of superposition** does **not** apply.



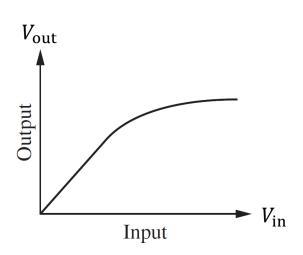
For example, in the **dynamic equations of robots** usually the nonlinear terms sin, cos, and squares of velocities appears.

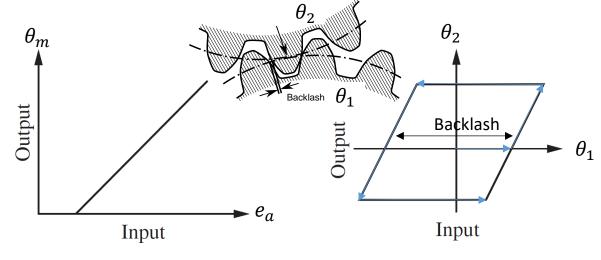
Double-Pendulum:

$$\begin{split} (m_1 + m_2)\ell_1\ddot{\theta}_1 + m_2\ell_2\ddot{\theta}_2\cos(\theta_1 - \theta_2) + m_2\ell_2\dot{\theta}_2^2\sin(\theta_1 - \theta_2) + g(m_1 + m_2)\sin\theta_1 &= 0 \\ m_2\ell_2\ddot{\theta}_2 + m_2\ell_1\ddot{\theta}_1\cos(\theta_1 - \theta_2) - m_2\ell_1\dot{\theta}_1^2\sin(\theta_1 - \theta_2) + m_2g\sin\theta_2 &= 0 \end{split}$$



### **Examples of Physical Nonlinearities**





#### **Amplifier Saturation**

An electronic amplifier is linear over a specific range but exhibits the nonlinearity called saturation at high input voltages.

#### **Motor Dead Zone**

A motor that does not respond at very low input voltages due to frictional forces exhibits a nonlinearity called dead zone.

#### **Backlash in Gears**

Gears that do not fit tightly exhibit a nonlinearity called backlash which the input moves over a small range without the output responding.

• Nonlinearities can be classified in terms of their mathematical properties, as **continuous** and **discontinuous**. Because discontinuous nonlinearities cannot be locally approximated by linear functions, they are also called **hard nonlinearities** (e.g., backlash, hysteresis, or stiction).

System Classifications

### Nonlinear System Behavior: Step Response

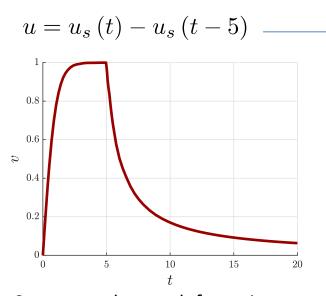
A simplified model of the motion of an underwater vehicle (ROV):

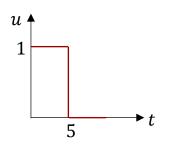
$$\dot{v} + |v| \, v = u$$

v: vehicle velocity

u: control input (thrust provided by a propeller)

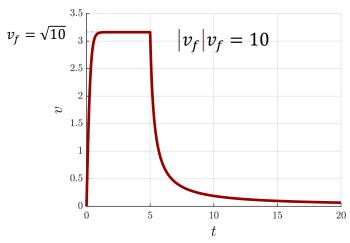
|v|v: nonlinearity due to square-law drag





$$u_s(t) = \begin{cases} 1 & t \ge 0 \\ 0 & t < 0 \end{cases}$$

$$u = 10(u_s(t) - u_s(t-5))$$



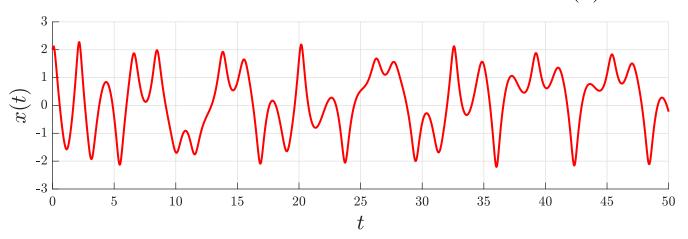
Settling speed is not 10 times, as it would be in a linear system!

System settles much faster in response to the positive unit step than it does in response to the subsequent negative unit step.

### **Nonlinear System Behavior: Chaos**

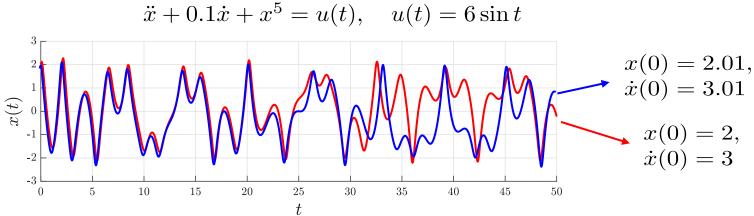
In the steady state, **sinusoidal inputs** to a stable LTI system generate a sinusoidal outputs of the same frequency (but different in amplitude and phase angle from the input). By contrast, the output of a nonlinear system may display sinusoidal, periodic, or chaotic behaviors.

$$\ddot{x} + 0.1\dot{x} + x^5 = u(t), \quad u(t) = 6\sin t$$
  $\begin{aligned} x(0) &= 2 \\ \dot{x}(0) &= 3 \end{aligned}$ 

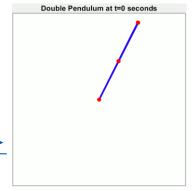


### **Nonlinear System Behavior: Chaos**

• For stable linear systems, small differences in initial conditions can only cause small differences in output. Strongly nonlinear systems, however, can display a phenomenon called **chaos**, i.e., the system output is extremely sensitive to **initial conditions**.



• Starting the pendulum from a slightly different initial condition would result in a vastly different trajectory.

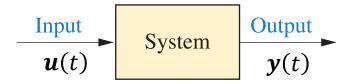


### State-Space Representation of Nonlinear Systems

In nonlinear systems, e.g., robotic manipulators, the underlying physical behavior is described by **nonlinear differential equations**. Although the **state-space representation** is capable of handling these systems, the **transfer function methods fail**.

A general nth-order nonlinear, continuous-time, TIV, SISO system is described by a nonlinear, scalar, constant-coefficient ODE:

$$\frac{d^n x(t)}{dt^n} = f\left(x(t), x^{(1)}(t), \dots, x^{(n-1)}(t), u(t), u^{(1)}(t), \dots, u^{(m)}(t)\right) \qquad n \ge m$$



### State-Space Representation of Nonlinear Systems

$$\frac{d^n x(t)}{dt^n} = f\left(x(t), x^{(1)}(t), \dots, x^{(n-1)}(t), u(t), u^{(1)}(t), \dots, u^{(m)}(t)\right) \qquad n \ge m$$

$$x_1(t) = x(t)$$

$$x_2(t) = \dot{x}(t)$$

$$\vdots$$

$$x_n(t) = x^{(n-1)}(t)$$

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$$\dot{x}_1(t) = x_2(t)$$

$$\dot{x}_2(t) = x_3(t)$$
:

$$\dot{x}_n(t) = f\left(x(t), x^{(1)}(t), \cdots, x^{(n-1)}(t), u(t), u^{(1)}(t), \cdots, u^{(m)}(t)\right)$$

$$\dot{\boldsymbol{x}}(t) = \mathbf{f}(\boldsymbol{x}(t), \boldsymbol{u}(t))$$
  $\boldsymbol{u}(t) = \left[u(t), u^{(1)}(t), \cdots, u^{(m)}(t)\right]^T \in \mathbb{R}^m$ 

If we choose 
$$y(t) = x(t)$$
, then  $y(t) = \mathbf{C}x(t)$   $\mathbf{C} = [1, 0, \dots, 0] \in \mathbb{R}^{1 \times n}$ 

General Form:

$$\dot{\boldsymbol{x}}(t) = \mathbf{f}(\boldsymbol{x}(t), \boldsymbol{u}(t), t)$$
 $\boldsymbol{y}(t) = \mathbf{g}(\boldsymbol{x}(t), \boldsymbol{u}(t), t)$ 

State Equation

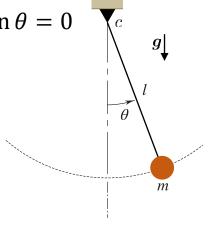
**Output Equation** 

**f** and **g** are nonlinear vector functions.



### **Examples**

Find the state equations of a damped pendulum.  $ml^2\ddot{\theta}+c\dot{\theta}+mgl\sin{\theta}=0$ 



Find the state equations of a rigid manipulator.  $M(\theta)\ddot{\theta} + c(\theta,\dot{\theta}) + g(\theta) = au$ 

$$m{ heta} \in \mathbb{R}^{n \times 1}$$
 $m{M} \in \mathbb{R}^{n \times n}$ 
 $m{c} \in \mathbb{R}^{n \times 1}$ 
 $m{g} \in \mathbb{R}^{n \times 1}$ 
 $m{\tau} \in \mathbb{R}^{n \times 1}$ 

System Classifications



### **Autonomous & Non-Autonomous Systems**

- Dynamic of a nonlinear system  $\dot{x}(t) = \mathbf{f}(x(t), u(t), t)$  when  $u(t) = \mathbf{0}$  can be represented as  $\dot{x}(t) = \mathbf{f}(x(t), t)$
- Moreover, the closed-loop dynamics of a feedback control system when u(t) = k(x, t) can be also represented as

$$\dot{x}(t) = \mathbf{f}(x(t), \mathbf{u}(t), t) \longrightarrow \dot{x}(t) = \mathbf{f}(x(t), \mathbf{k}(x, t), t) \longrightarrow \dot{x}(t) = \mathbf{f}(x(t), t)$$

A nonlinear system of the form  $\dot{x}(t) = f(x(t), t)$  is said to be **Autonomous** (or **Time-Invariant**) if the function f does not depend <u>explicitly</u> on time, i.e.,

$$\dot{\mathbf{x}}(t) = \mathbf{f}(\mathbf{x}(t))$$

otherwise, the system is called **Non-autonomous** (or **Time-Varying/Varient**).

### **Autonomous & Non-Autonomous Systems**

 A special class of nonlinear systems are linear systems. LTI systems are autonomous and LTV systems are non-autonomous.

$$\dot{m{x}}(t) = \mathbf{A}(t) m{x}(t)$$

 $\dot{\boldsymbol{x}}(t) = \mathbf{A}\boldsymbol{x}(t)$ 

• The non-autonomous nature of a control system may be due to a time-variation either in the plant or in the control law, e.g., trajectory trackers or adaptive controllers (adaptive controllers for linear time-invariant plants usually make the closed-loop control systems nonlinear and non-autonomous).

$$\dot{x}(t) = \mathbf{f}(x(t), t)$$

**Example**: the closed-loop system of the simple plant

$$\dot{x} = -x + u$$

by choosing u

System Classifications

$$u(t) = -x^2 \sin t$$

as is nonlinear and non-autonomous.

$$\dot{x} = -x - x^2 \sin t$$



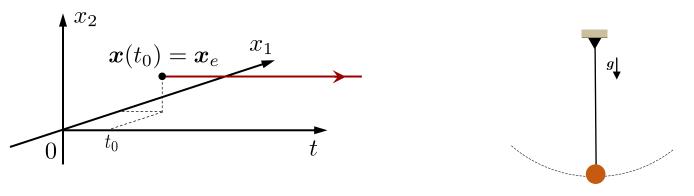
System Classifications

### **Equilibrium Points**

A state  $x_e$  is an **Equilibrium Point** (or **Equilibrium State**) of the system if once  $x = x_e$ , it remains equal to  $x_e$  for all future time.

$$\dot{m{x}}=\mathbf{f}(m{x}_e)=m{0} \quad orall t\geq 0$$
  $\dot{m{x}}=\mathbf{f}(m{x}_e,t)=m{0} \quad orall t\geq t_0$  (For Autonomous Systems)

i.e., a point for which if the system starts there (initial state  $x(t_0) = x_e$ ) it will remain there for all future time.



Stability of a system 

 = stability of systems at equilibrium points. Thus, many stability problems are naturally formulated with respect to equilibrium points.

### **Example**

Consider the system 
$$\dot{x} = x^2 - 1$$

This autonomous system has two equilibrium points  $\pm 1$ .

Consider the system 
$$\dot{x} = tx^2 - 1$$

This non-autonomous system has no equilibrium points. Although it might seem that it has 2 equilibrium points  $x_e=\pm 1$  at time  $t_0=1$ . However, these are not equilibrium points for all  $t\geq 1$ .

System Classifications

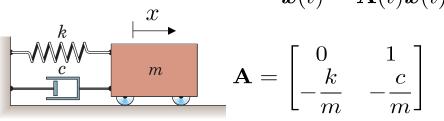
### **Isolated Equilibrium Points**

An equilibrium point  $x_e$  (at  $t_0$ ) of  $\dot{x} = \mathbf{f}(x,t)$  is said to be **Isolated** if there exists a real positive number r such that there may not be any equilibrium point other than  $x_e$  in  $\Omega$ , where

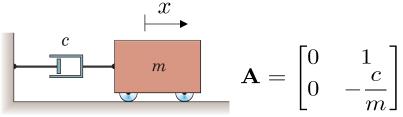
$$\Omega = \{ \boldsymbol{x} \in \mathbb{R}^n : \|\boldsymbol{x} - \boldsymbol{x}_e\| < r \}$$

In the case that there does not exist any r that satisfies the above then the equilibrium point  $x_e$  is **not isolated**.

• A linear system (LTI or LTV) has a **single** isolated equilibrium point at the origin  $\mathbf{0}$  **if**  $\mathbf{A}$  is nonsingular. However, a **nonlinear** system often has **more than one** isolated equilibrium point.  $\dot{\boldsymbol{x}}(t) = \mathbf{A}(t)\boldsymbol{x}(t)$ 



$$\dot{x} = Ax = 0$$



All *x* are equilibrium points!



### Shifting an Equilibrium Point to Origin

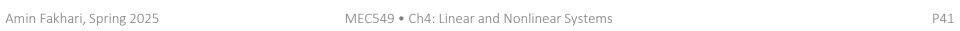
If the equilibrium point of interest is not at the origin, by defining the difference between the original state x and the specific equilibrium point  $x_e$  as a new set of state variables, one can always shift the equilibrium point to the origin  $\mathbf{0}$  (for analytical simplicity).

$$\dot{x} = \mathbf{f}(x) \xrightarrow{x = y + x_e} \dot{y} = \mathbf{f}(y + x_e) \qquad (y = \mathbf{0} \leftrightarrow x = x_e)$$

Therefore, instead of studying the behavior of  $\dot{x} = f(x)$  in the neighborhood of  $x_e$ , one can equivalently study the behavior of  $\dot{y} = f(y + x_e)$  in the neighborhood of the origin 0.

**System Classifications** 



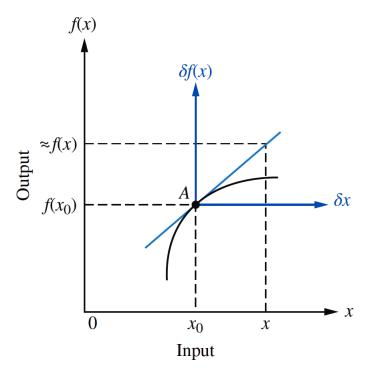




### **Linearization of Nonlinear Systems**

In control engineering, a normal operation of the system may be around an **equilibrium point** or **a limited operating range**. Therefor, it is possible to approximate the nonlinear system by an equivalent linear system within the limited operating range.

• Linear approximations simplify the analysis and design of a system.

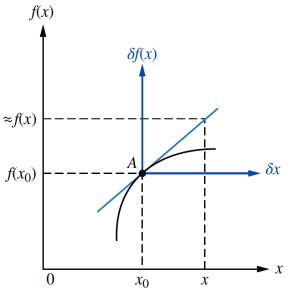


System Classifications

### **Linear Approximation of Nonlinear Systems**

The linearization procedure is based on (1) the expansion of nonlinear function f(x) into a **Taylor Series** about the operating point  $A(x_0, y_0 = f(x_0))$  and (2) the retention of only the linear term.

**Note**: Since the variables deviate only slightly from the operating condition  $(x - x_0)$ , higher-order terms of the Taylor series expansion can be neglected.



$$y = f(x)$$
  
$$f(x) = f(x_0) + \frac{f'(x_0)}{1!}(x - x_0) + \frac{f''(x_0)}{2!}(x - x_0)^2 + \cdots$$

$$f(x) \approx f(x_0) + f'(x_0)(x - x_0)$$

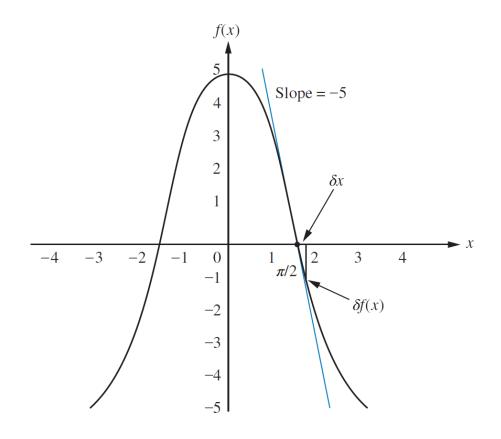
(A straight-line relationship)

Expressing this straight line in frame 
$$\delta x - \delta f(x)$$
: 
$$\begin{cases} \delta x = x - x_0 \\ \delta f(x) = f(x) - f(x_0) \end{cases} \Rightarrow \delta f(x) = f'(x_0) \delta x$$



### **Example**

Linearize  $f(x) = 5 \cos x$  about  $x = \pi/2$ .



### **Example**

Linearize  $\ddot{x} + 2\dot{x} + \cos x = 0$  for small deviations about  $x = \pi/4$ .

\* Note: If  $\delta x$  is a small variable:  $\sin \delta x \approx \delta x$ ,  $\cos \delta x \approx 1$ ,  $\delta x^2 \approx 0$ 

System Classifications