

edX Robo4 Mini MS – Locomotion Engineering

Block 1 – Week 3 – Unit 1

Raibert Vertical Hopper

Video 6.1

Segment 6.0.1

Motivation & Agenda

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with

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July, 2017

Agenda

- Historical
 - core component of Raibert's Hoppers
 - inspired much further work
- Conceptual
 - 1 DoF vertical “heartbeat”
 - organizing rhythm around which other DoF are coordinated
- Technical
 - first look at hybrid dynamics
 - central feature of locomotion

Raibert Vertical Hopping

University of Michigan

<https://www.youtube.com/watch?v=XFXj81mvInc>

M. H. Raibert. Legged Robots That Balance. Cambridge: MIT Press, 1986

Active Vertical Batting

[Buehler & Koditschek; CSM '90]

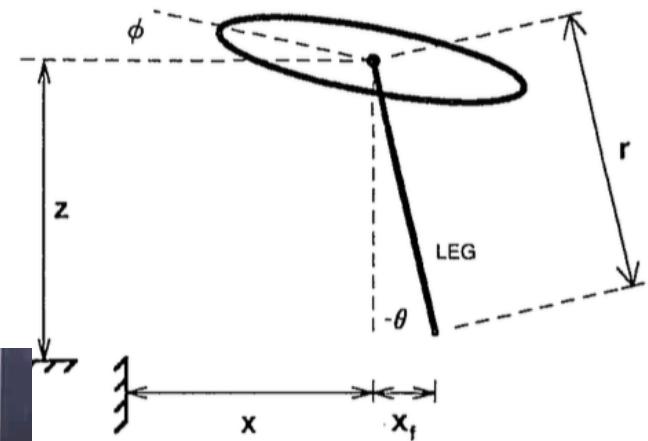
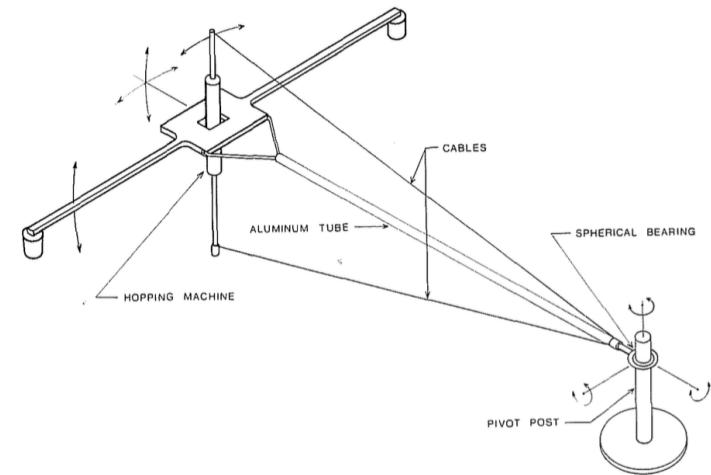
University of Michigan

M. Buehler, D. E. Koditschek, and P. J. Kindlmann, “A family of robot control strategies for intermittent dynamical environments,” *IEEE Control Systems Magazine*, vol. 10, pp. 16–22, 1990.

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Sagittal Plane Machine

- Raibert reduced complexity
 - initially, of control task
 - by mechanical restriction
 - to 2 dim. workspace
 - yields 3 DoF configuration space
- Control Ideas Generalize
 - to spatial machine in 3 dim
 - with 6 DoF
 - and beyond



figures from:
M. H. Raibert. Legged Robots That Balance. Cambridge: MIT Press, 1986

video from:
Plastic Pals. "Robots from MIT's Leg Lab". YouTube video, 02:28. Posted [October 31, 2011].

<https://www.youtube.com/watch?v=XEXj81mVnc>

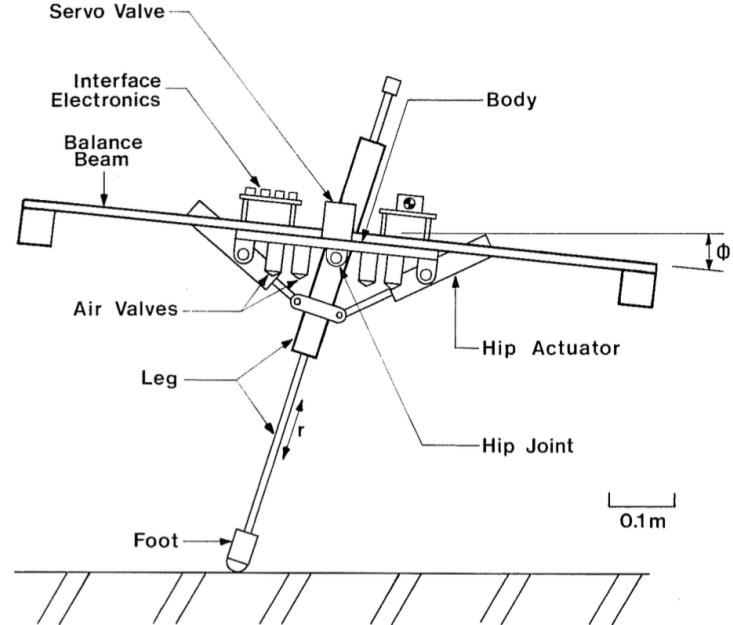
Robust Modular Robot Controller

- Problem setting
 - 3 DoF (RPR) kinematics
 - 6 dim dynamics

- highly nonlinear (and “hybrid”)
 - tightly coupled
 - underactuated

- Raibert’s control policies

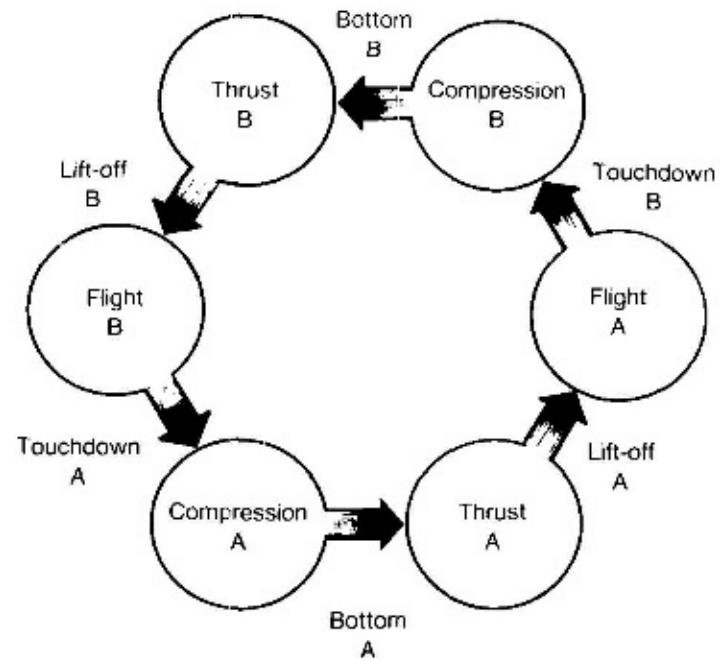
- use (mostly) “proportional-derivative” feedback loops
 - assume each DoF is decoupled from others
 - “time-share” the actuation resources
 - air valve adjustment of leg spring in stance for height control
 - hip actuator control of pitch in stance
 - hip actuator selection of leg angle in flight for fore-aft velocity



M. H. Raibert. Legged Robots That Balance. Cambridge: MIT Press, 1986

Control of Hybrid Dynamics

- legged locomotion results from
 - ground reaction forces (**GRF**)
 - imparted to the body mass center
 - through the limbs
- limbs make & break ground contact
 - introducing impulses
 - changing the GRF dynamics
 - yield hybrid dynamics
 - continuous-in-time force-driven
 - discrete-in-time event-driven
- control of hybrid dynamics
 - by actuation through limbs
 - must be coordinated in continuous time
 - and sensitive to the limb-contact events



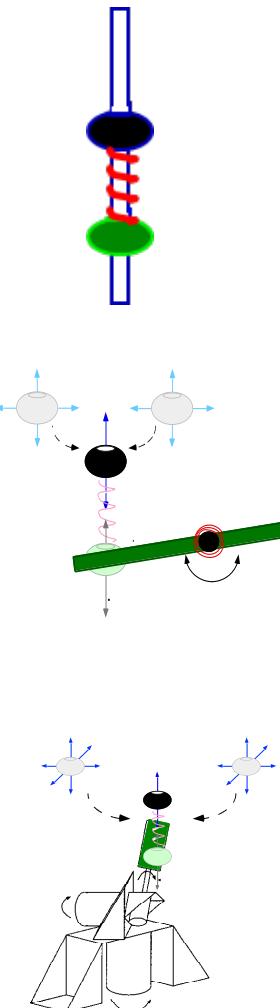
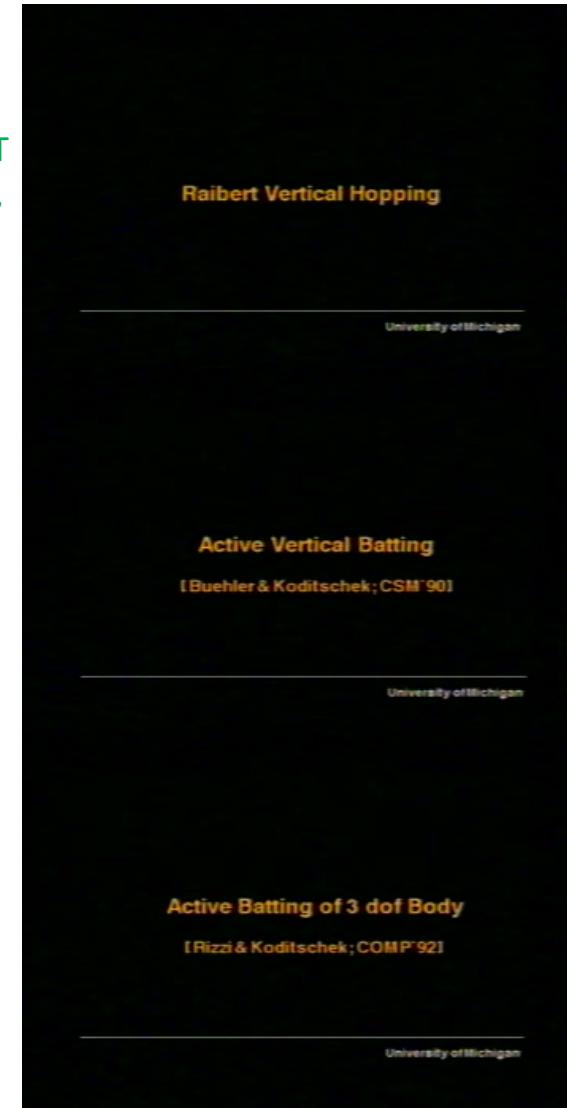
M. H. Raibert, "Legged Robots," *Commun. ACM*, vol. 29, no. 6, pp. 499–514, Jun. 1986.

Other Machines “Anchor” Vertical Hopper

- Vertical hopper
 - Can be used as a “template”
 - to be “anchored”
 - as the “heartbeat”
- In more complex systems
 - 2 DoF
 - 3 DoF
 - ... and beyond

Raibert, MIT
Press, 1986,
op. cit.

Buehler, et al.
*IEEE Control
Systems
Magazine,*
1990, *op. cit.*



A. Rizzi, L. Whitcomb, and D. Koditschek, “Real-Time Control of a Spatial Robot Juggler,” *IEEE Computer*, vol. 25, no. 5, pp. 12–24, 1992.

Robo4x 6.0.1 6

Initial Approach to Vertical Hopper

- Model Continuous time flows
 - each mode of contact
 - governed by different VF
- Model natural guard conditions
 - physical event interrupts mode
 - locomotion: typically LO/TD
- Study/Express mode map
- Model reset map
- Compose
 - mode map \circ reset map
 - further compose each composition in turn
- End up with return map

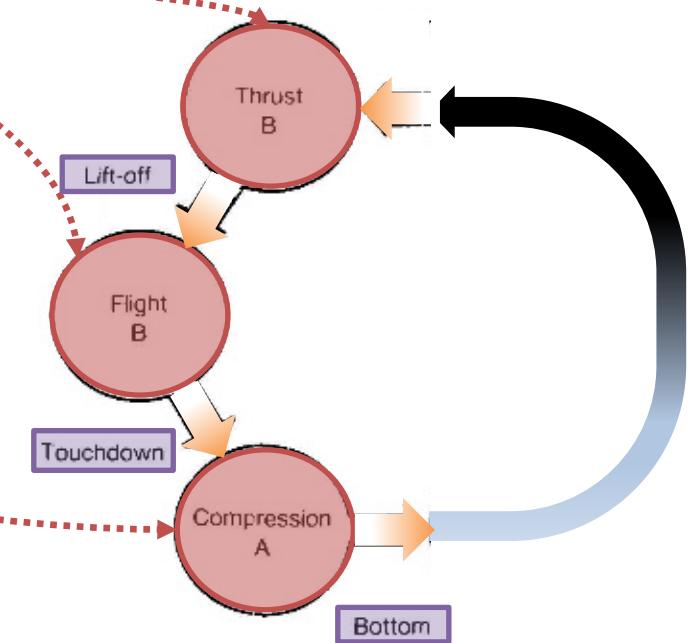


figure modified from
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Continuous Time Models

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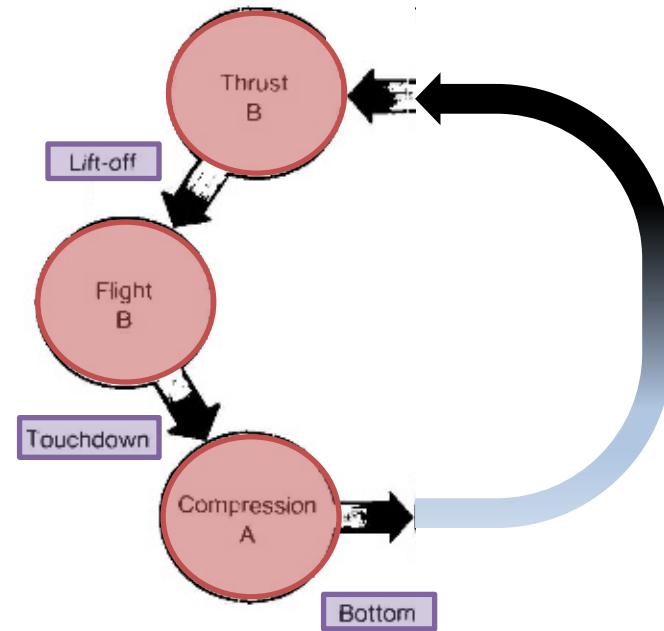


figure modified from
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Gravity Loaded Hooke's Law Spring

- Rework Seg.1.2.1.1-eqn (4)

$$\begin{aligned} m\ddot{\chi} &= \Phi_D(\dot{\chi}) + \Phi_{HS}(\chi - \chi_r) + \Phi_G(\chi) \\ &= -b\dot{\chi} - k(\chi - \chi_r) - mg \\ &=: \Phi_{GLH}(\mathbf{x}); \quad \mathbf{x} := \begin{bmatrix} \chi \\ \dot{\chi} \end{bmatrix} \end{aligned} \tag{1}$$

- Note differences from Φ_{DHO}
 - using variable χ to denote vertical extension
 - rather than b_y as in pendulum or $x := \chi - \chi_r$ as in DHO
 - to focus on role of relaxed spring position $\chi_r > 0$
 - addition of gravitational force, $\Phi_G := -mg$

Additional Modeling Assumptions

- Simplify physics
 - unit mass, $m:=1$; gravity-matched spring, $m \chi_r = g$
 - touchdown/liftoff
 - spring is disengaged (braked) below rest length: $\chi_{td} = \chi_{lo} = 0 < \chi_r$
 - to assure some spring potential in each stance
 - and (with gravity-matched spring) zero potential force at $\chi = 0$

- Yields familiar 1st order VF (f_{DHO} from Seg.1.2.1.2)

$$f_{GLH}(\mathbf{x}) := A_{GLH} \mathbf{x}$$

$$A_{GLH} := \begin{bmatrix} 0 & 1 \\ -\frac{k}{m} & -\frac{b}{m} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\omega^2(1 + \beta^2) & -2\omega\beta \end{bmatrix} \quad (2)$$

- with new parameters, $\omega, \beta > 0$ replacing physical ones in f_{DHO}
- introduced to assure under-damped system

- But now working with hybrid dynamics

- this VF only models “passive” stance mode $\chi < 0$
- actuation, introduced next, yields new “active” mode
- introduce formal guard and reset maps in next segment

Active Stance Mode: Energy Injection

- Raibert's original hopper had an “air-spring”
 - pneumatic pump adjusted pressure in air-chamber
 - controller added fixed energy (via pressure)
 - at “bottom”: creates new “active” mode with own VF
- Present model uses “series-elastic” actuator (SEA)
 - force control via spring rest-length adjustment
 - popularized by Pratt; still actively developed, e.g.
 - <https://www.youtube.com/watch?v=wJwKwLUUTjc>
- C. S. Knabe, V. Orekhov, M. A. Hopkins, B. Y. Lattimer, and D. W. Hong, “Two configurations of series elastic actuators for linearly actuated humanoid robots with large range of motion,” in *Humanoid Robots (Humanoids), 2014 14th IEEE-RAS International Conference on*, 2014, pp. 1096–1096
- present use: introduce constant upward thrust force
 - initiated at “bottom” as before (controlled to perform positive work)
 - restores energy lost to damping each successive stance

Active Stance Mode Control

- SEA force control: an “inverse statics” control policy
 - assume actuator can move spring rest length, χ_r
 - exact precision in position and timing
 - known inverse statics for gravity-loaded Hooke’s spring
$$\Phi_{\text{GHS}}(\chi) := \Phi_{\text{HS}}(\chi) + \Phi_{\text{G}}(\chi) = -k(\chi - \chi_r) - mg$$
$$\Rightarrow \Phi_{\text{GHS}}^{-1}(\Phi) = -(\Phi + mg - \chi)/k$$
 - given a desired reference force profile, $\Phi_{\text{ref}}(t)$
 - actuator rest length command
$$\chi_r^*(t) := \Phi_{\text{GHS}}^{-1} \circ \Phi_{\text{ref}}(t) = -(\Phi_{\text{ref}}(t) + mg - \chi)/k$$
 - injects exactly the right force at the right time
- Get new dynamics $m\ddot{\chi} = \Phi_{\text{D}}(\dot{\chi}) + \Phi_{\text{GHS}}^{-1} \circ \Phi_{\text{ref}}(t)$
$$= -b\dot{\chi} + \Phi_{\text{ref}}(t)$$

Constant Thrust Control Policy

- Stance control strategy
 - hold rest length fixed at nominal $\chi_r^*(t) \equiv \chi_r$
 - until compression ends at “bottom”
 - defined by vanishing of **guard** function $\gamma_{\text{comp}}(\mathbf{x}) := \dot{\chi}$
 - at which point inject constant upward thrust force
 $\Phi_{\text{ref}}(t) \equiv \tau$
 - for brief fixed time interval after bottom
 $t \in [t_b, t_b + \Delta_\tau]$
- Yields damped constant thrust VF

$$f_{\text{DCT}}(\mathbf{x}) = A_{\text{DCT}}\mathbf{x} + a_{\text{DCT}}$$

$$A_{\text{DCT}} := \begin{bmatrix} 0 & 1 \\ 0 & -b/m \end{bmatrix} \quad (3)$$

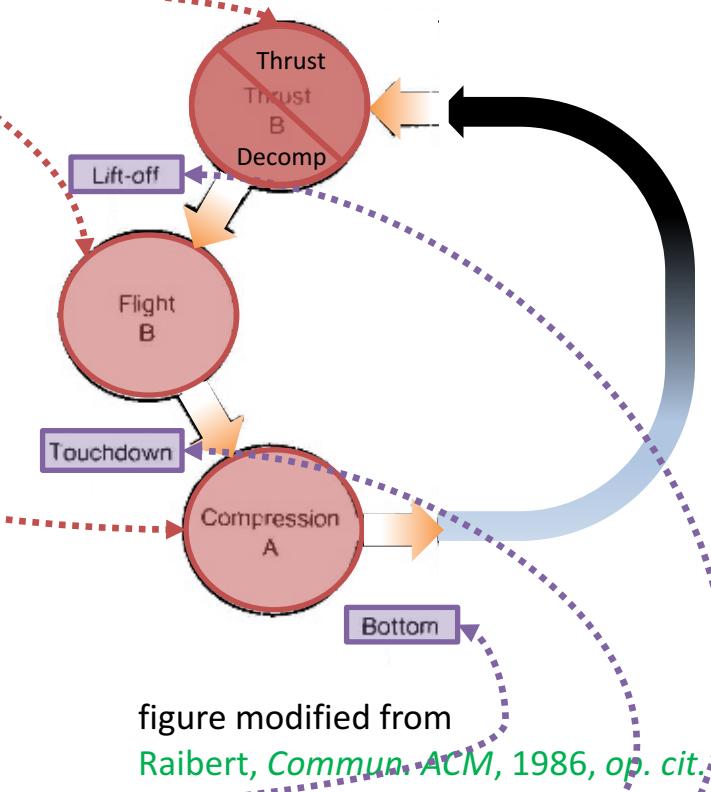
$$a_{\text{DCT}} := \begin{bmatrix} 0 \\ \tau/m \end{bmatrix}$$

Ballistic Flight Mode

- Decompression continues after end of thrust
 - rest length again fixed at nominal $\chi_r^*(t) \equiv \chi_r$
 - until decompression ends at “liftoff”
 - defined by vanishing of **guard** function $\gamma_{\text{decomp}}(\mathbf{x}) := \chi$
 - at which point ground reaction forces cease
 - only force is due to gravity
 - yields flight mode VF
- $$f_{BF}(\mathbf{x}) = A_{BF}\mathbf{x} + a_{BF}$$
- $$A_{BF} := \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad (4)$$
- $$a_{BF} := \begin{bmatrix} 0 \\ -g/m \end{bmatrix}$$

Moving Ahead

- Model Continuous time flows
 - each **mode** of contact
 - governed by different VF
- Model natural **guard** conditions
 - physical **event** interrupts mode
 - locomotion: typically LO/TD
 - had to introduce extra mode
 - decompression
 - model as timed event
 - to simplify analysis
- Next up:
 - study/express **mode** maps
 - model the **reset** maps



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Continuous Time Mode Maps

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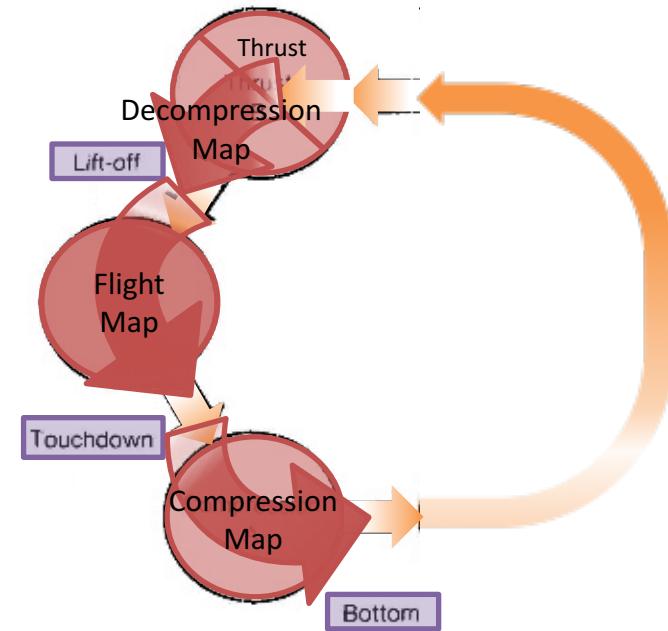


figure modified from
Raibert, *Commun. ACM*, 1986, *op. cit.*

Ballistic Flight Mode Flow Map Definition

- Exercises for this segment show

$$f_{BF}(\mathbf{x}) = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ -g \end{bmatrix}$$
$$\Rightarrow f_{BF}^t(\mathbf{x}_0) = \begin{bmatrix} x_0 + \dot{x}_0 t - \frac{1}{2} g t^2 \\ \dot{x}_0 - gt \end{bmatrix} \quad (5)$$

- Flight ends at Touchdown

- which, recall, occurs when $\chi = \chi_{td} := 0$
- represented by the vanishing of guard $\gamma_{BF}(\mathbf{x}) := \chi$

- Now solve the vanishing condition for t

$$0 = \gamma_{BF} \circ f_{BF}^t(\mathbf{x}_0) = \chi_0 + \dot{\chi}_0 t - \frac{1}{2} g t^2$$

- to get implicit function,

$$T_{td}(\mathbf{x}_0) := \frac{1}{g} \left(\dot{\chi}_0 + \sqrt{\dot{\chi}_0^2 - 2g\chi_0} \right)$$

- yielding mode flow map

$$\mathbf{x}_{td} = \boldsymbol{\varphi}_{BF}^{td}(\mathbf{x}_0) := f_{BF}^{T_{td}(\mathbf{x}_0)}(\mathbf{x}_0) = \begin{bmatrix} 0 \\ -\sqrt{\dot{\chi}_0^2 + 2g\chi_0} \end{bmatrix} \quad (6)$$



Gravity Loaded Hooke's Law Spring Flow

- Recall from Seg. 1.2.1.2 & exercises RC coordinates

$$\begin{bmatrix} \psi \\ \dot{\psi} \end{bmatrix} = \mathbf{y} = h_{\text{RC}}(\mathbf{x}) = S\mathbf{x}; \quad S := \begin{bmatrix} \omega\sqrt{1+\beta^2}\chi + \beta\dot{\chi}/\sqrt{1+\beta^2} \\ \dot{\chi}/\sqrt{1+\beta^2} \end{bmatrix}$$

- Reveals “oscillatory” (underdamped) GLH flow

$$\exp(tSA_{GLH}S^{-1}) = e^{-\beta t}R(\omega t); \quad R(\omega t) := \begin{bmatrix} \cos \omega t & -\sin \omega t \\ \sin \omega t & \cos \omega t \end{bmatrix}$$

- Simplify still further via nonlinear (“polar”) CC_D

$$\begin{bmatrix} \rho \\ \phi \end{bmatrix} := \mathbf{p} = h_{\text{P}}(\mathbf{y})\mathbf{p} = h_{\text{P}}(\mathbf{y}) := \begin{bmatrix} \psi^2 + \dot{\psi}^2 \\ \arctan \dot{\psi}/\psi \end{bmatrix} \quad (6)$$

(don't confuse with Seg. 1.2.2.1 vector $q = \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix}$!)

- Exercises: decoupled conjugate VF with simple flow

$$\dot{\mathbf{p}} = \begin{bmatrix} \dot{\rho} \\ \dot{\phi} \end{bmatrix} = \begin{bmatrix} -2\omega\beta\rho \\ -\omega \end{bmatrix} =: f_{\text{PRC}}(\mathbf{p}) \Rightarrow f_{\text{PRC}}^t(\mathbf{p}_0) = \begin{bmatrix} e^{-2\omega\beta t}\rho_0 \\ \phi_0 - \omega t \end{bmatrix}$$

Compression Flow Map Definition

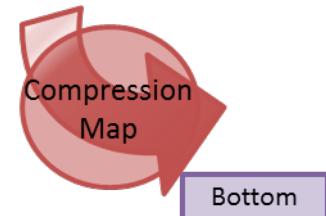
- Compression starts at touchdown, $\mathbf{x}_{td} = \begin{bmatrix} \chi_{td} \\ \dot{\chi}_{td} \end{bmatrix} = \begin{bmatrix} 0 \\ \dot{\chi}_{td} \end{bmatrix}$
- Compression ends at “bottom”
 - which, recall, occurs when $\dot{\chi} = 0$
 - represented by the vanishing of guard $\gamma_{\text{comp}}(\mathbf{x}) := \dot{\chi}$

preferably expressed in polar RC coordinates

$$\begin{aligned}\tilde{\gamma}_{\text{comp}}(\mathbf{p}) &:= \gamma_{\text{comp}} \circ h_{\text{PRC}}^{-1}(\mathbf{p}) = \gamma_{\text{comp}} \circ h_{\text{RC}}^{-1} \circ h_{\text{P}}^{-1}(\mathbf{p}) \\ &= \sqrt{1 + \beta^2} \dot{\psi}(\mathbf{p}) = \sqrt{1 + \beta^2} \rho \sin \phi\end{aligned}$$

- Solve vanishing condition for t in polar RC coords

$$n\pi = \phi_{td} - \omega t \Leftrightarrow t = T_{bot}(\mathbf{p}_{td}) := \frac{\phi_{td} - n\pi}{\omega}$$



- Yields mode flow map

$$\mathbf{p}_{bot} = f_{\text{PRC}}^{\text{bot}}(\mathbf{p}_{td}) := f_{\text{PRC}}^{T_{bot}(\mathbf{p}_{td})}(\mathbf{x}_{td}) = \left[e^{-2\beta(\phi_{td} - n\pi)} \rho_{td} \right] \quad (7)$$

Damped Constant Thrust Mode Flow

- Raibert's strategy of constant thrust from bottom
- Results in familiar 2nd order ODE $m\ddot{\chi} + \beta\dot{\chi} - \tau = 0$
- Yields 1st order VF $f_{DCT}(\mathbf{x}) = A_{DCT}\mathbf{x} + a_{DCT}$

$$A_{DCT} := \begin{bmatrix} 0 & 1 \\ 0 & -\beta/m \end{bmatrix} \quad (8)$$

$$a_{DCT} := \begin{bmatrix} 0 \\ \tau/m \end{bmatrix}$$

- With familiar flow derived from f_D in Seg.1.2.1.1
- Resulting in $\mathbf{x}_{et} = \mathbf{x}_{bot} + \mathbf{x}_t$

$$\mathbf{x}_t := \frac{\tau}{2\omega^2\beta^2} \begin{bmatrix} 2\omega\beta\Delta_\tau - (1 - e^{-2\beta\Delta_\tau}) \\ \omega\beta(1 - e^{-2\beta\Delta_\tau}) \end{bmatrix} \quad (9)$$

D. E. Koditschek and M. Bühler. Analysis of a simplified hopping robot. *The International Journal of Robotics Research*, 10(6):587–605, Dec 1991

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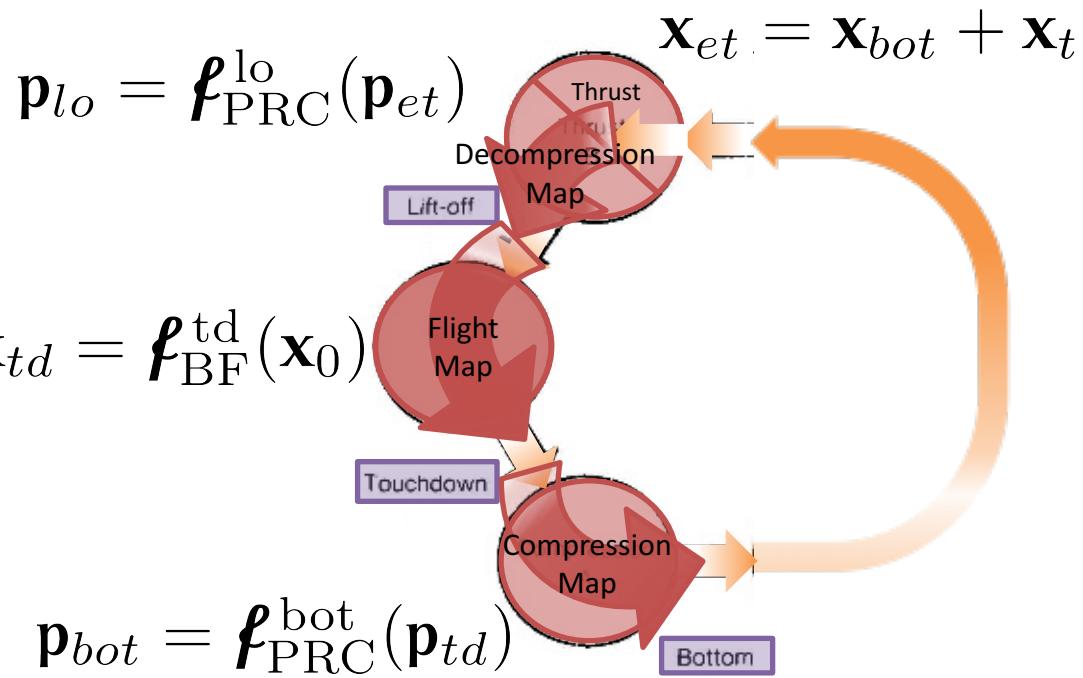


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