

Robust Servo LQR

1. Using the aircraft pitch-axis plant data from Example 5.2 (Eq. 5.58 page 119) without an actuator, DESIGN a RSLQR to command Az using a state feedback controller.

```
Ap = [Za_V    1.    Zd_V    0.;
      Ma      0.    Md      0;
      0.      0.    0.      1.;
      0.      0.   -w_act*w_act  -2*z_act*w_act]

Bp = [0.; 0.; 0.; w_act*w_act]

Cp = [Za 0. Zd 0.;
      eye(4)]
|
Dp = [ 0*Cp*Bp]

%Wiggle Model
Aw = [0. Za    0.;
      0. Za_V  1.;
      0. Ma     0.]
Bw = [Zd;
      Zd_V;
      Md]

qq=logspace(-6,0,100);
```

Ap =

-1.3046e+00	1.0000e+00	-2.1420e-01	0
4.7711e+01	0	-1.0483e+02	0
0	0	0	1.0000e+00
0	0	-4.7769e+03	-9.7729e+01

Bp =

0
0
0
4.7769e+03

Cp =

-1.1569e+03	0	-1.8995e+02	0
1.0000e+00	0	0	0
0	1.0000e+00	0	0
0	0	1.0000e+00	0
0	0	0	1.0000e+00

Dp =

0
0
0
0
0

Aw =

0	-1.1569e+03	0
0	-1.3046e+00	1.0000e+00
0	4.7711e+01	0

Bw =

-1.8995e+02
-2.1420e-01
-1.0483e+02

ip =

30

Q =

5.7224e-05	0	0
0	0	0
0	0	0

Eigen Values of Closed Loop System

-3.1297e+01 + 3.3953e+01i
-3.1297e+01 - 3.3953e+01i
-1.9517e+01 + 0.0000e+00i
-8.4613e+00 + 6.4480e+00i
-8.4613e+00 - 6.4480e+00i

cl_EigVec =

-8.6318e-04 + 5.2392e-04i	-8.6318e-04 - 5.2392e-04i	-1.4435e-02 + 0.0000e+00i	-5.9798e-03 + 6.1886e-03i	-5.9798e-03 - 6.1886e-03i
4.9751e-03 - 4.8411e-02i	4.9751e-03 + 4.8411e-02i	2.7236e-01 + 0.0000e+00i	1.2269e-03 - 8.3708e-02i	1.2269e-03 + 8.3708e-02i
-1.4587e-02 - 1.5826e-02i	-1.4587e-02 + 1.5826e-02i	4.4138e-02 + 0.0000e+00i	-7.7714e-03 - 4.0153e-03i	-7.7714e-03 + 4.0153e-03i
9.9388e-01 + 0.0000e+00i	9.9388e-01 + 0.0000e+00i	-8.6147e-01 + 0.0000e+00i	9.1647e-02 - 1.6135e-02i	9.1647e-02 + 1.6135e-02i
-1.7110e-02 - 9.5248e-02i	-1.7110e-02 + 9.5248e-02i	-4.2607e-01 + 0.0000e+00i	-9.9206e-01 + 0.0000e+00i	-9.9206e-01 + 0.0000e+00i

Classical Margins

ans =

struct with fields:

```
GainMargin: [3.5217e-01 3.4661e+00]  
GMFrequency: [6.9773e+00 6.0700e+01]  
PhaseMargin: 3.4716e+01  
PMFrequency: 2.1433e+01  
DelayMargin: 2.8269e-02  
DMFrequency: 2.1433e+01  
Stable: 1
```

SV Margins

```
RDu_nGM RDu_pGM RDu_Pha  
0.6484 2.1846 0.54913  
-3.7631 6.7876 31.463  
SRu_nGM SRu_pGM SRu_Pha  
0.47596 1.524 0.53023  
-6.4487 3.6599 30.3801
```

LQR Charts

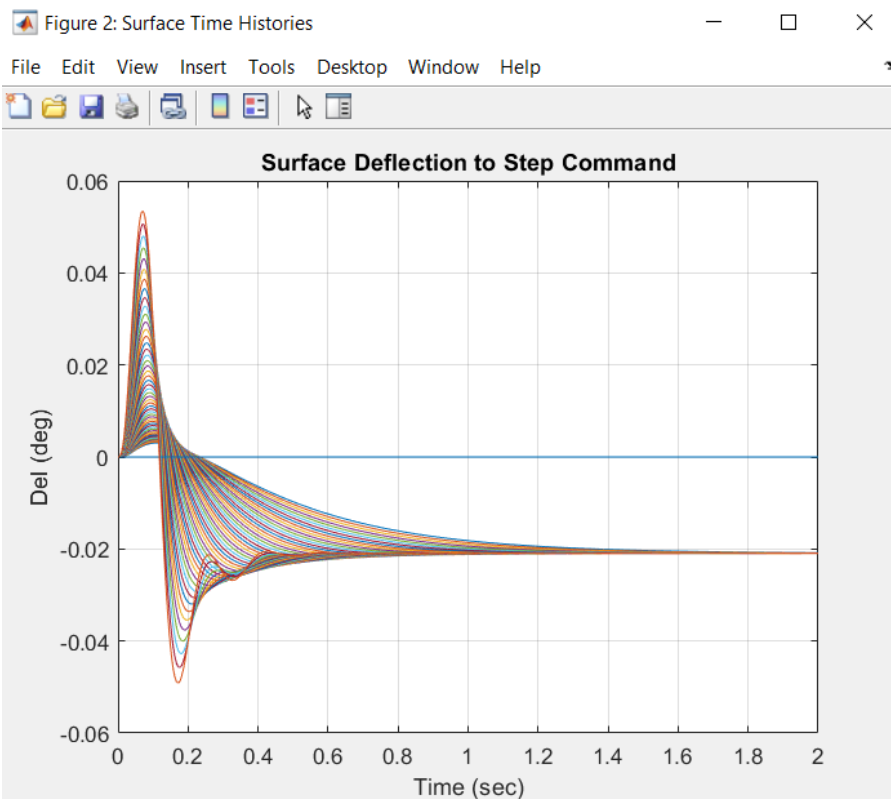
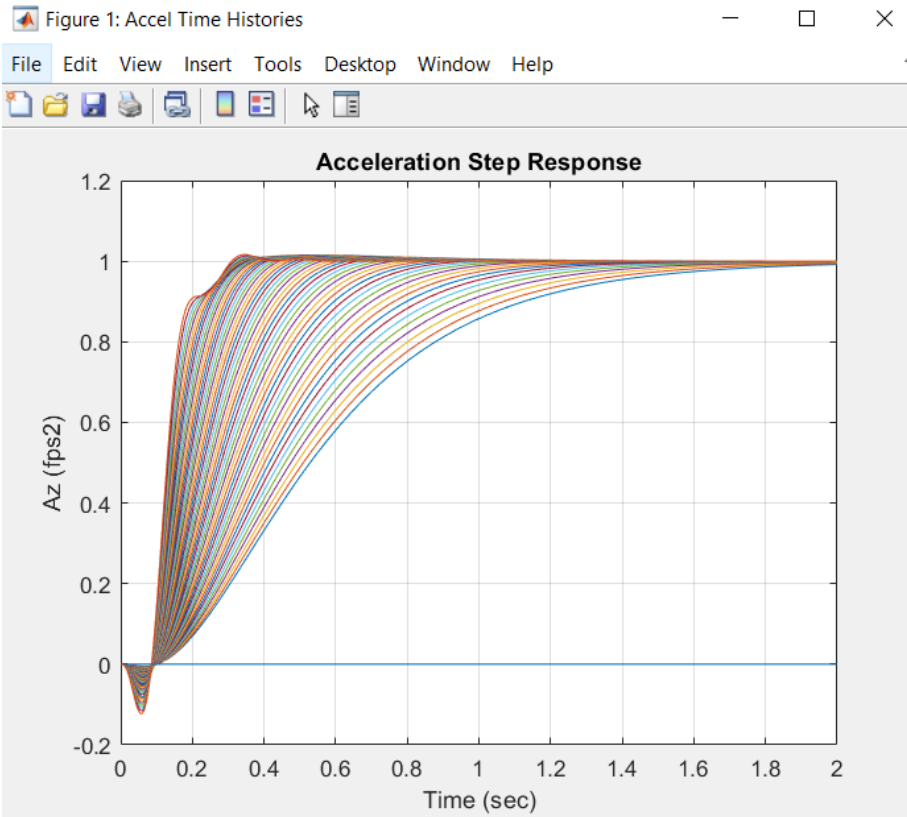


Figure 3: Surface Rate Time Histories

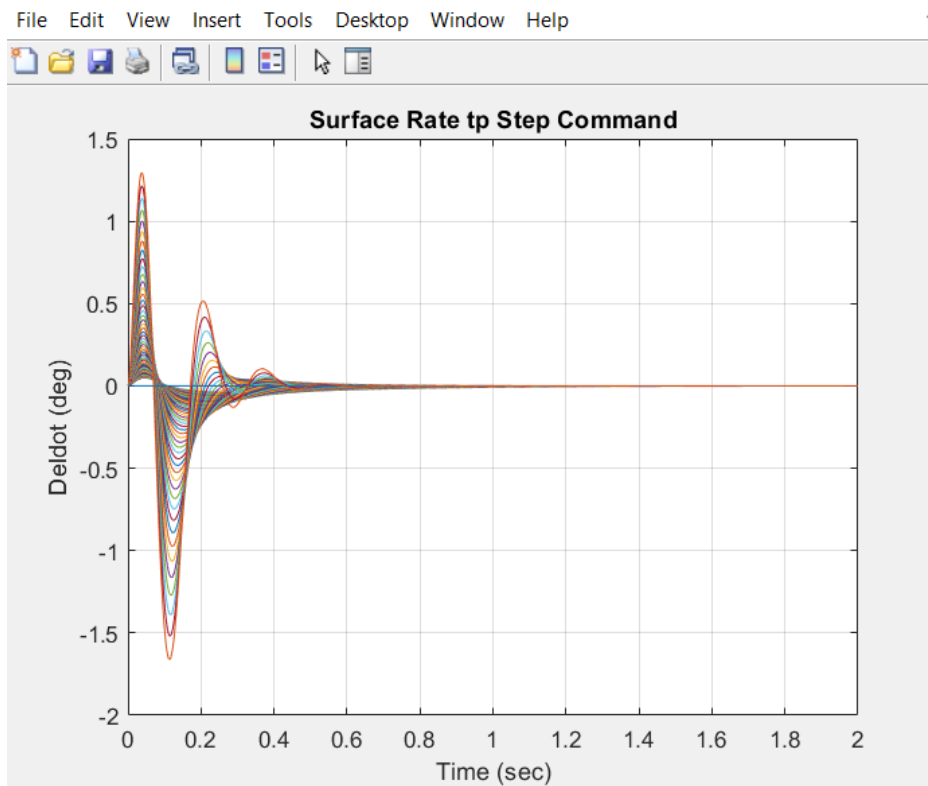
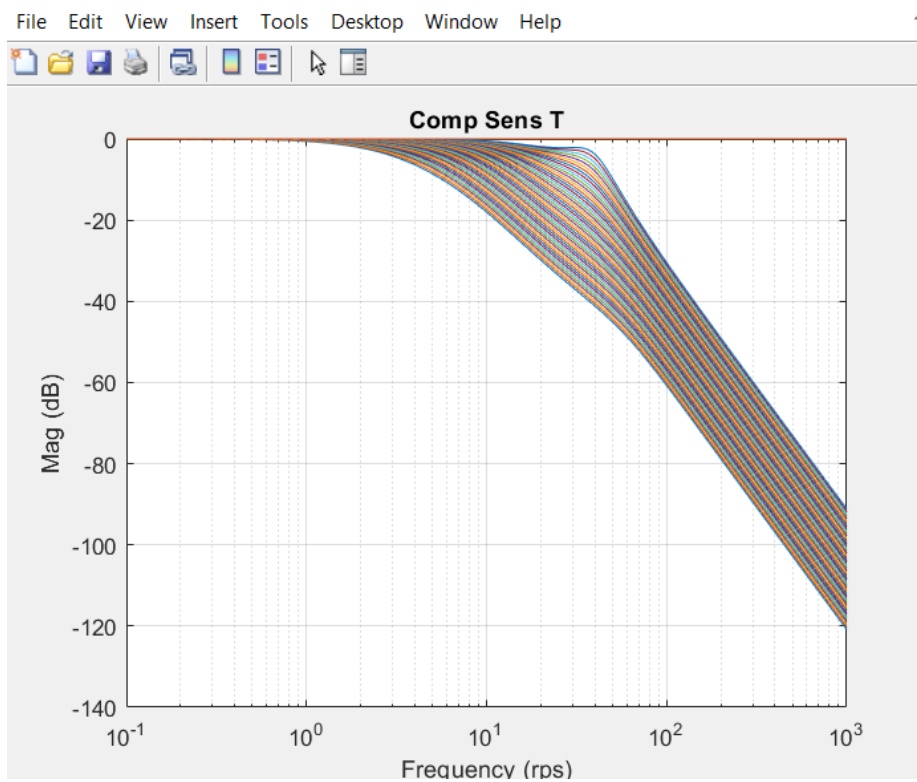
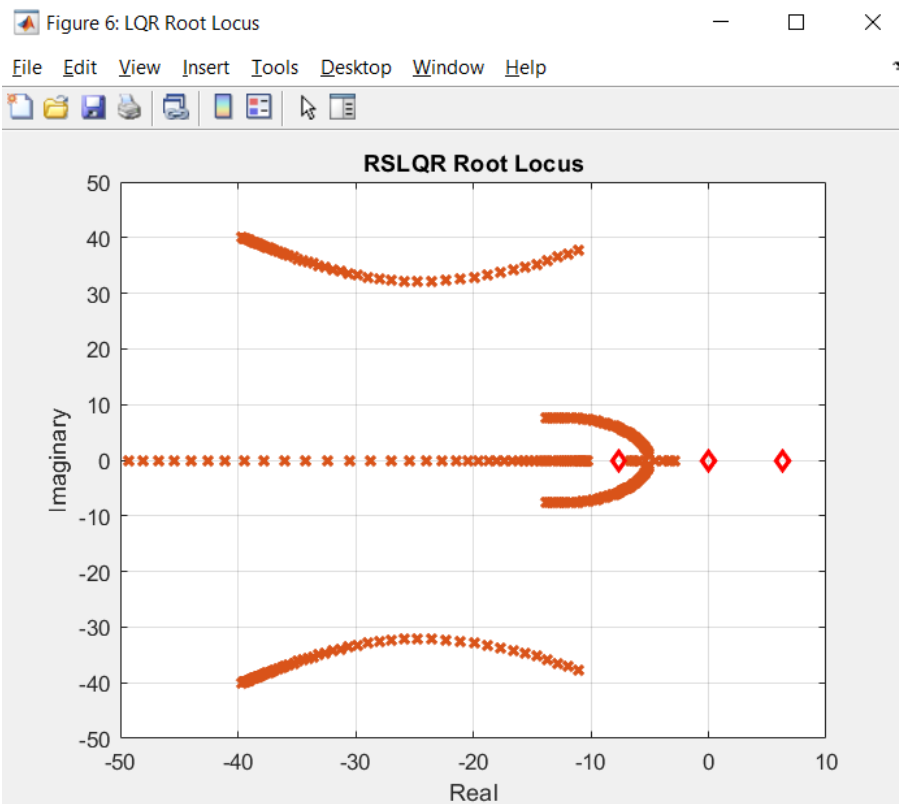
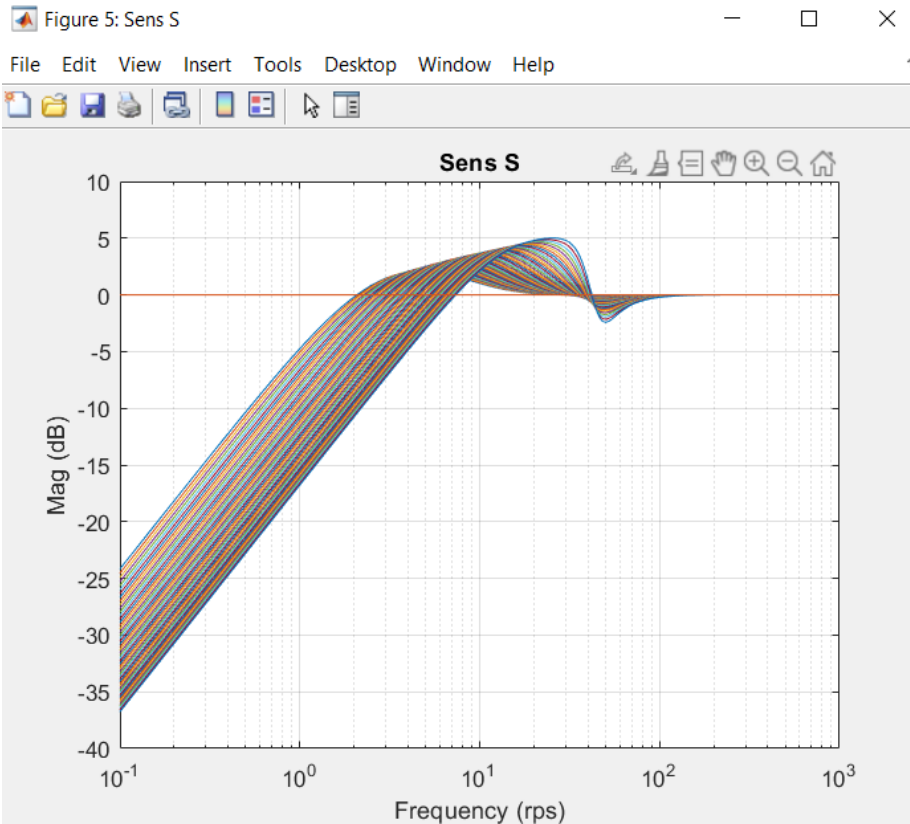
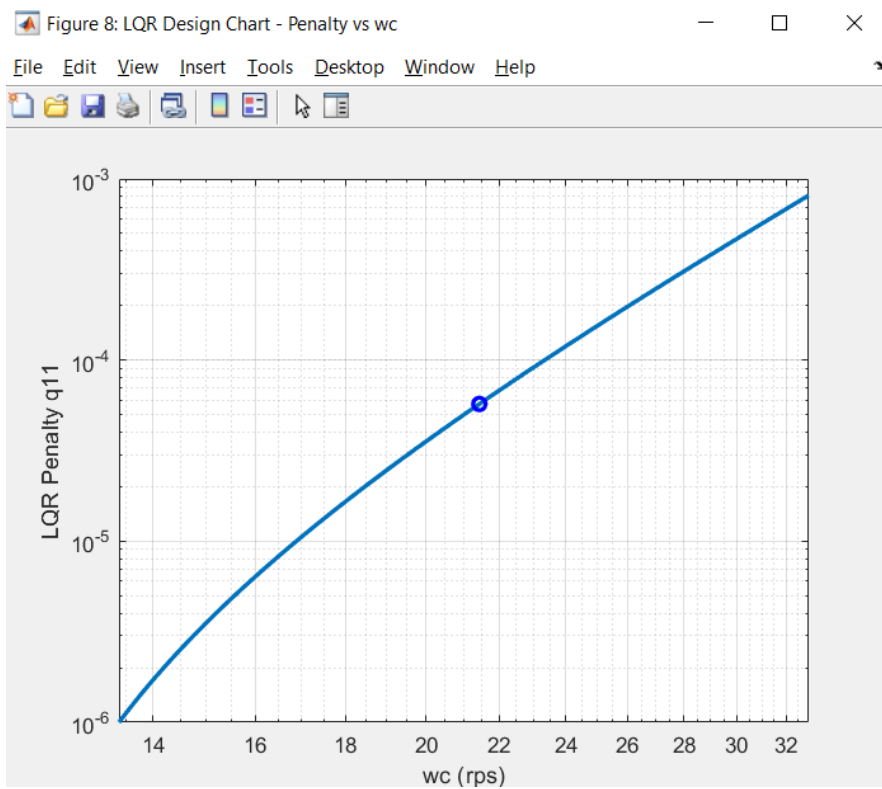
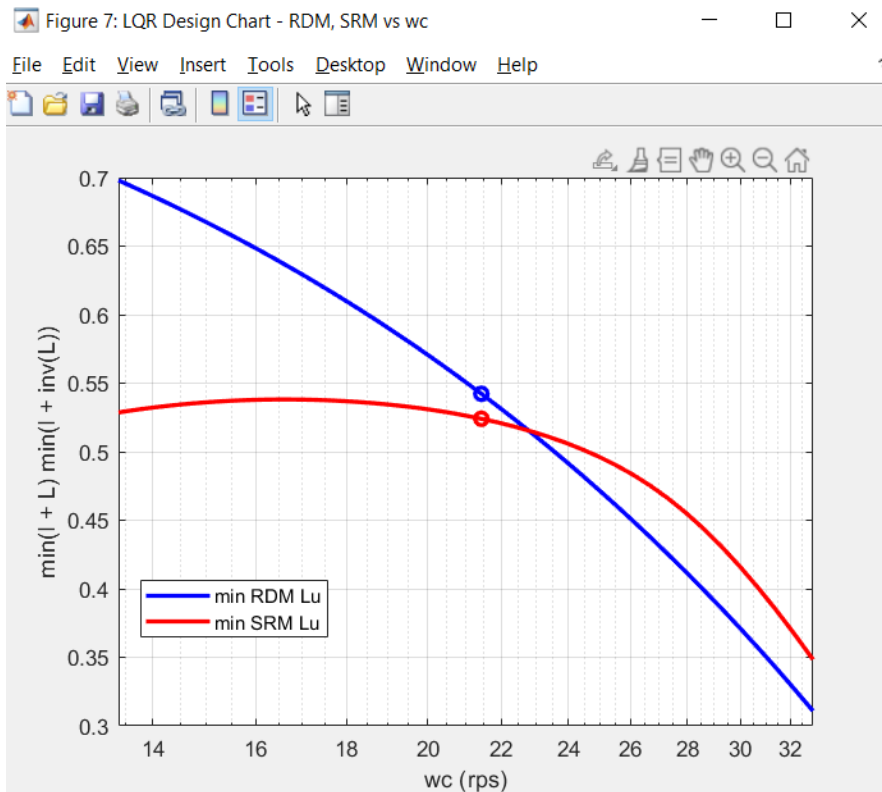
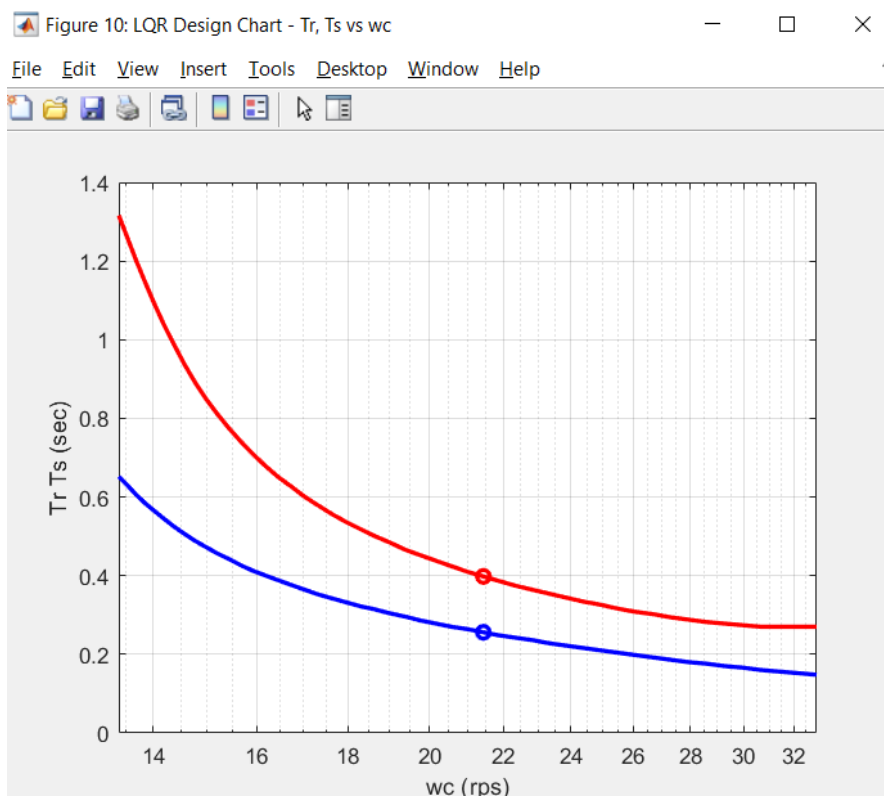
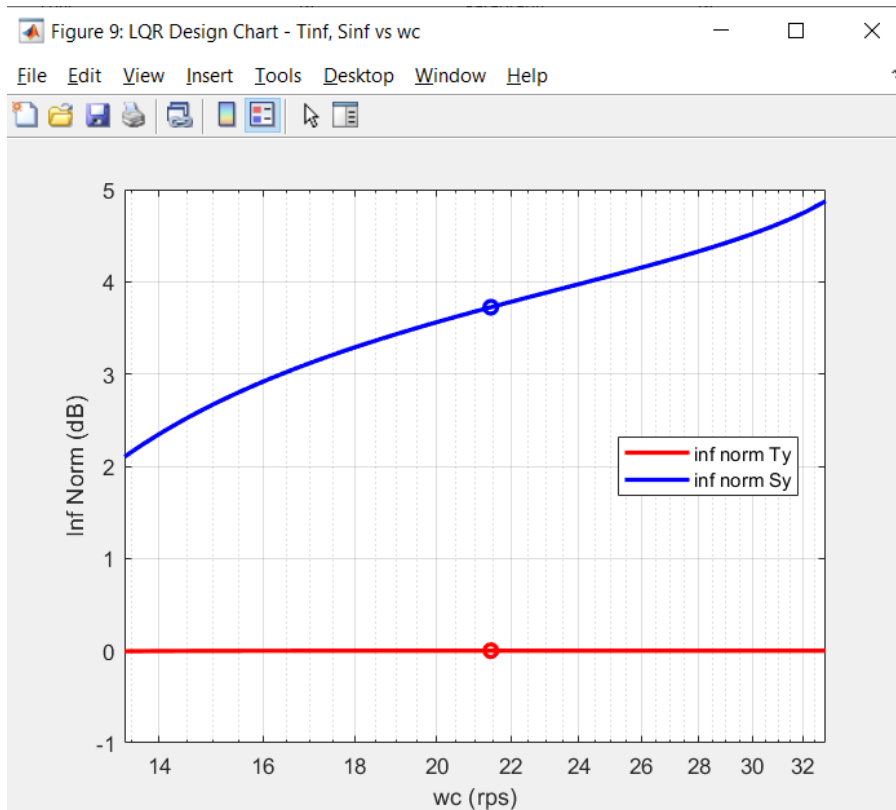


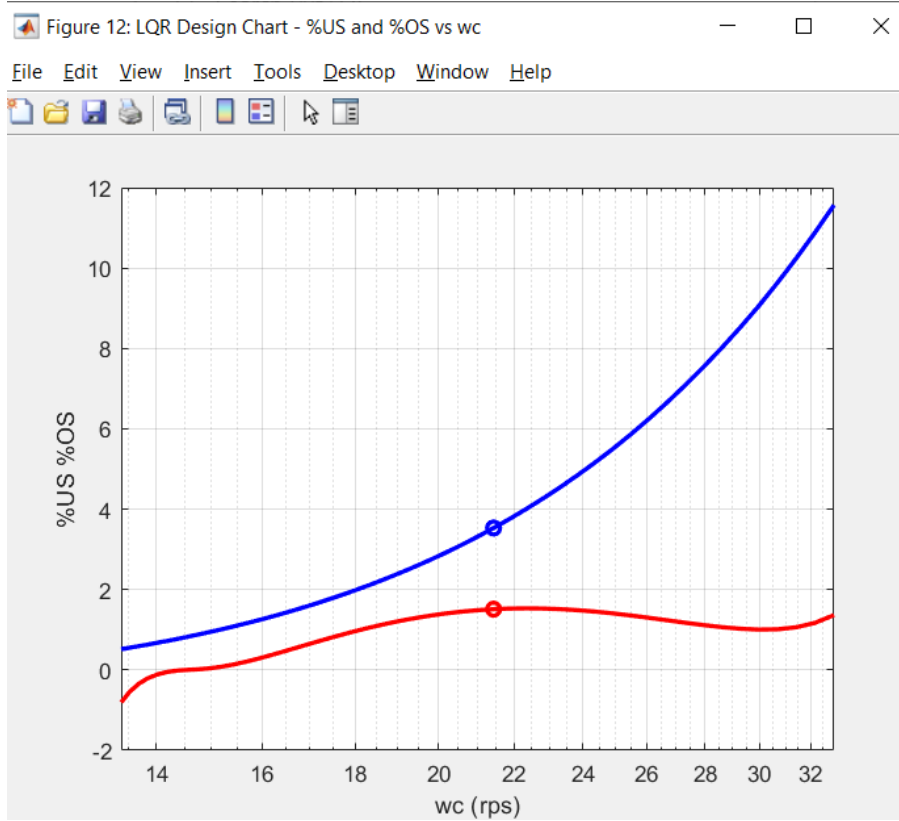
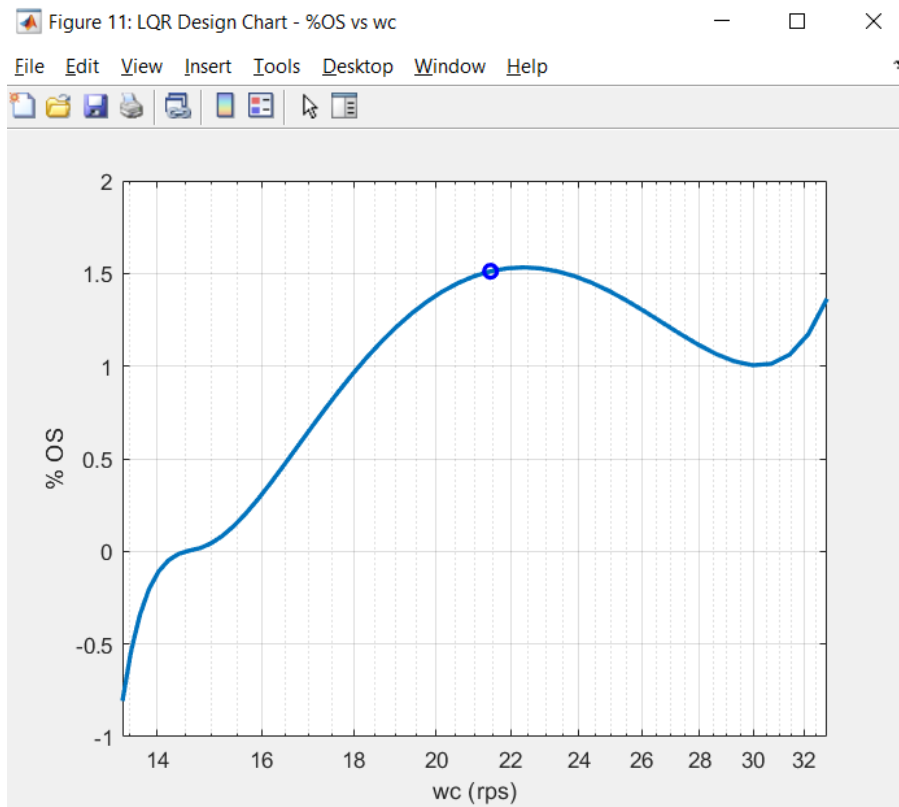
Figure 4: Comp Sens T

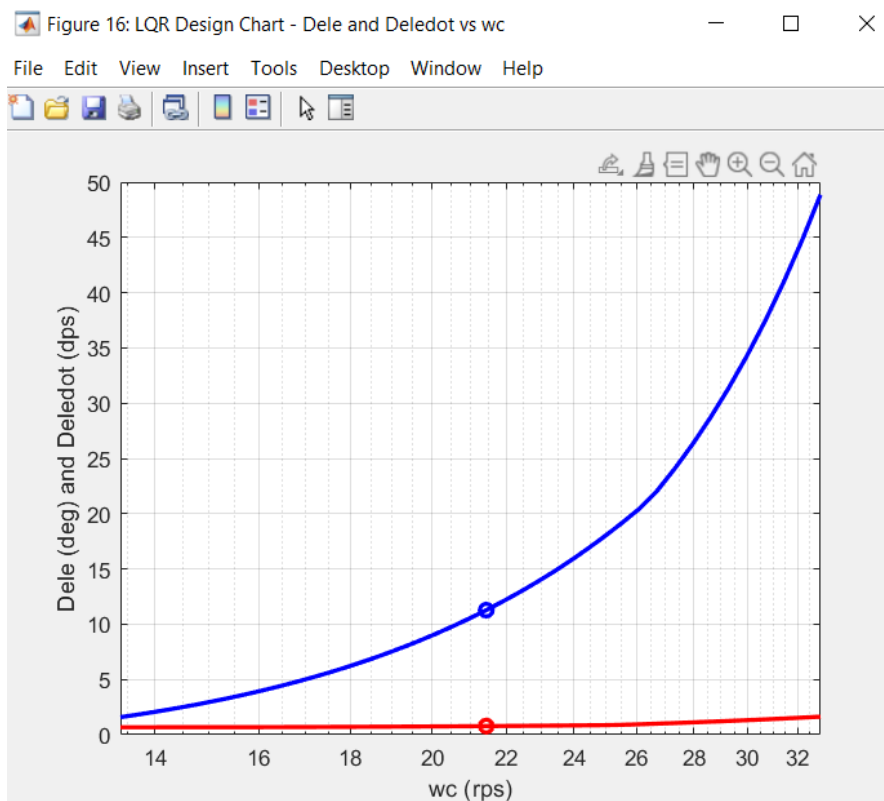
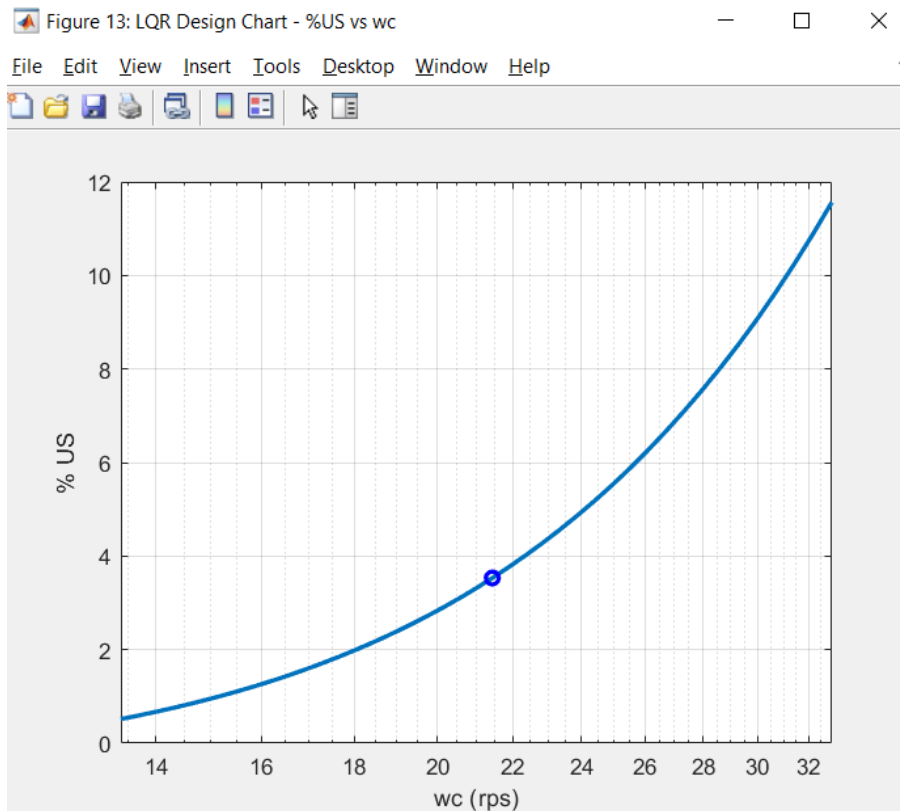












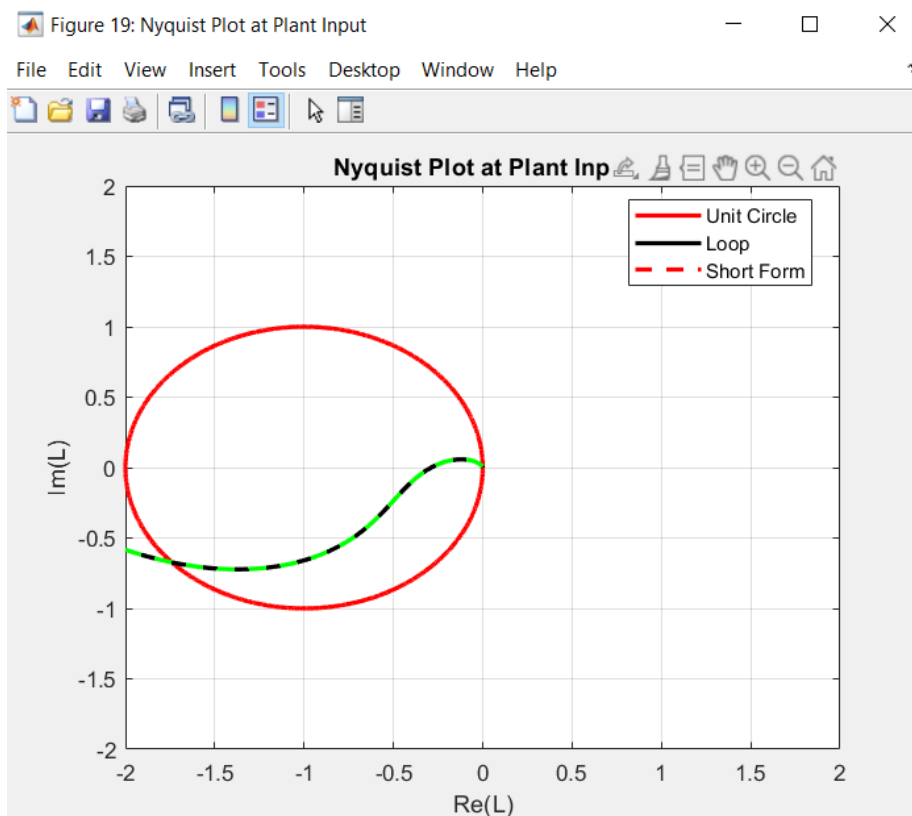
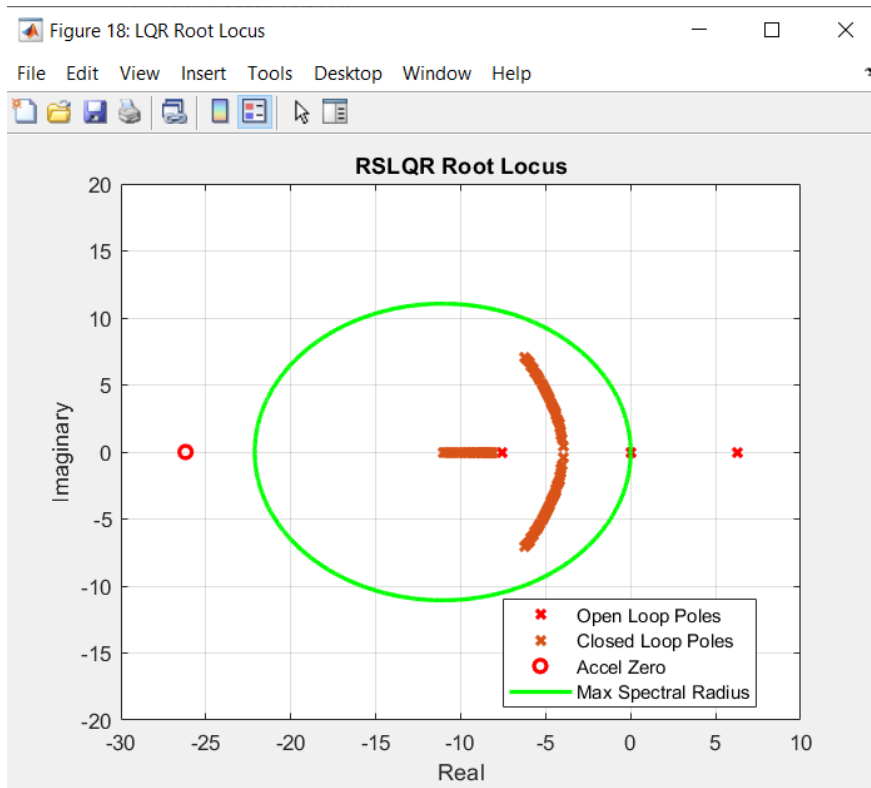
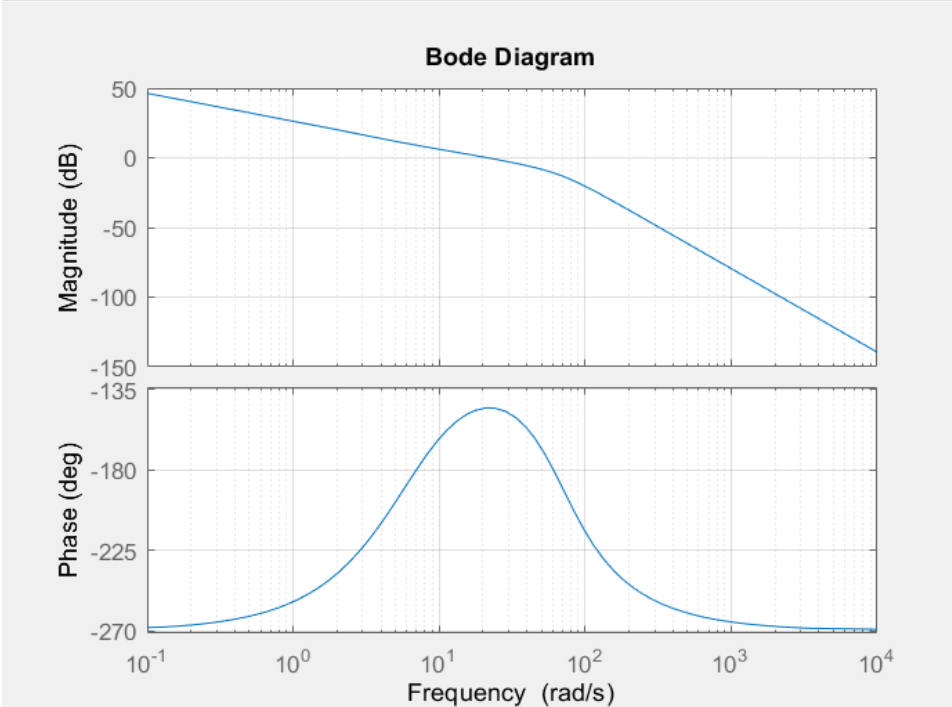


Figure 19: Nyquist Plot at Plant Input

File Edit View Insert Tools Desktop Window Help



GM =

3.5217e-01

PM_deg =

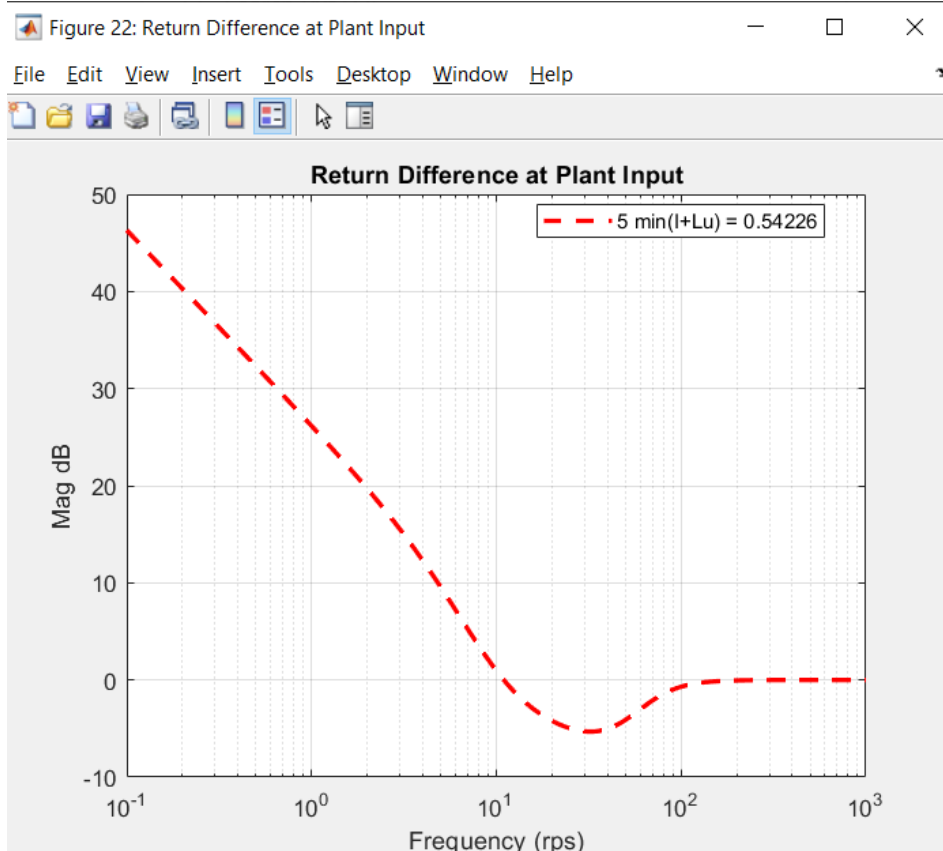
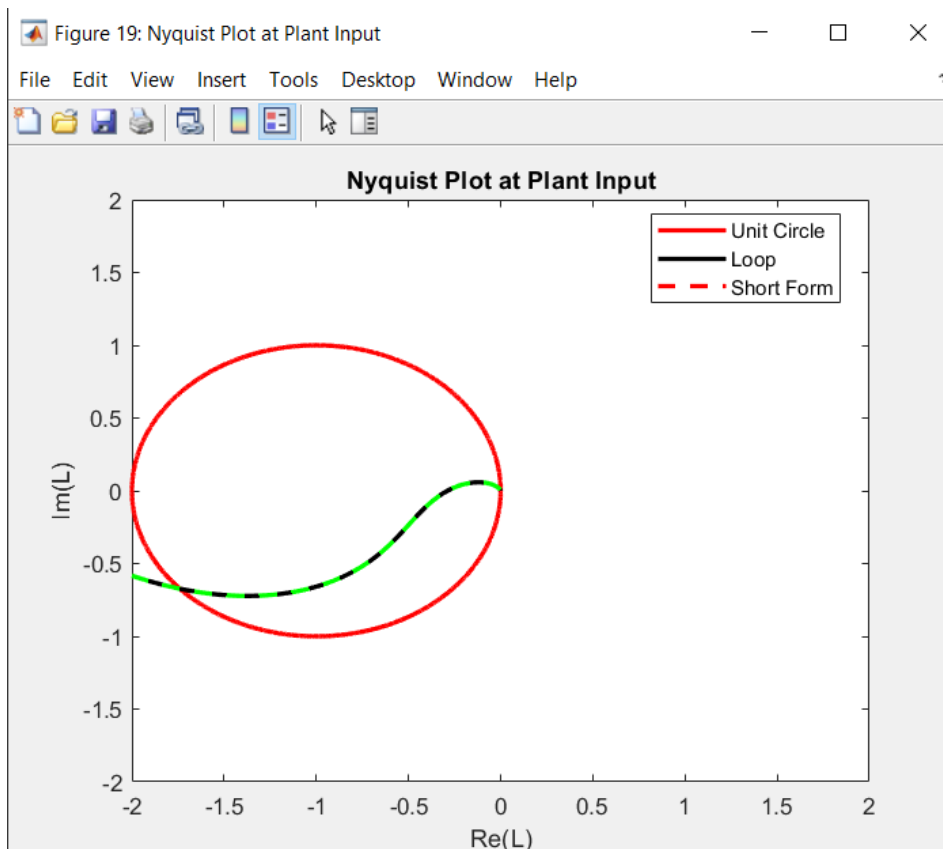
3.4716e+01

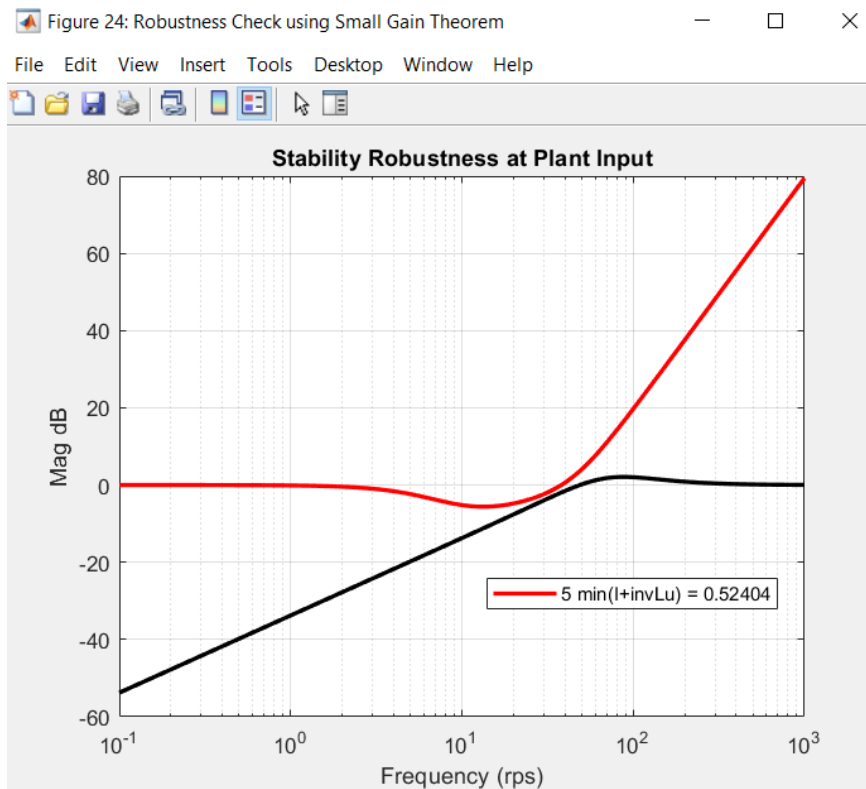
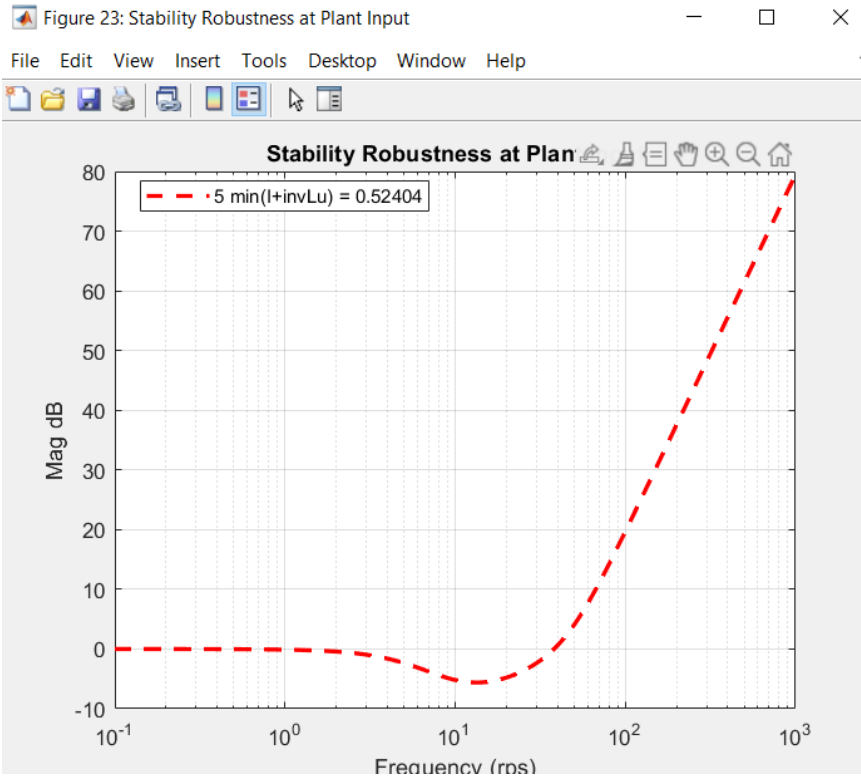
WC_GM =

6.9773e+00

WC_Pm =

2.1433e+01





Robust; using small gain theorem for actuator dynamics ($\omega_{act} = 11\text{Hz}$ & $\zeta_{act} = 0.707$)

2.

```
Ac = 0.;  
Bc1 = [1. 0. 0. 0. 0.];  
Bc2 = -1;  
Cc = -Kx_lqr(1);  
Dc1 = [0. -Kx_lqr(2:3) 0. 0.];  
Dc2 = 0.;
```

Ac =

0

Bc1 =

1 0 0 0 0

Bc2 =

-1

Cc =

-7.5646e-03

Dc1 =

0 2.3099e+00 2.2086e-01 0 0

Dc2 =

0