## 1 Kinematics

$$M_{RB}\dot{\boldsymbol{\nu}} + M_A\dot{\boldsymbol{\nu}_r} + C_A(\boldsymbol{\nu})\boldsymbol{\nu} + C_{RB}(\boldsymbol{\nu}_r)\boldsymbol{\nu}_r + D(\boldsymbol{\nu}_r)\boldsymbol{\nu}_r = \boldsymbol{\tau} + \boldsymbol{w}(t)$$
(1)

$$\dot{\boldsymbol{\eta}} = \boldsymbol{J}(\eta)\boldsymbol{\nu} \tag{2}$$

$$\tau = \tau_{control} \tag{3}$$

## 2 Thruster model

$$\tau_{control} = \begin{bmatrix} \tau_{control,linear} \\ \tau_{control.torque} \end{bmatrix}$$
(4)

Each propeller can rotate with an angle  $\xi_p$ . The thrust of each propeller can be describe as a vector by spilting it into components that align with the ship coordinate frame

$$t_p = T_{nn}n_p^2 + T_{nv}V_A n_p (5)$$

$$\boldsymbol{T}_{p} = \begin{bmatrix} \cos(\xi_{p}) \\ \sin(\xi_{p}) \\ 0 \end{bmatrix} t_{p} \tag{6}$$

where  $t_p$  is the magnitude of the thrust and  $T_p$  is the thrust vector. The linear control force is found as the sum of forces from each propeller p.

$$\tau_{control,linear} = \sum_{p}^{P} T_{p} \tag{7}$$

$$\tau_{control,torque} = \sum_{p}^{P} r_p \times T_p = \sum_{p}^{P} S(r_p) T_p$$
 (8)

where  $r_p$  is the position vector of thruster p and  $S(r_p)$  is the skew-symetric matrix of vector  $r_p$ .

In the case of two propellers we have

$$\tau_{control} = \begin{bmatrix} I & I \\ S(\mathbf{r}_1) & S(\mathbf{r}_2) \end{bmatrix} \begin{bmatrix} \mathbf{T}_1 \\ \mathbf{T}_2 \end{bmatrix}$$
(9)