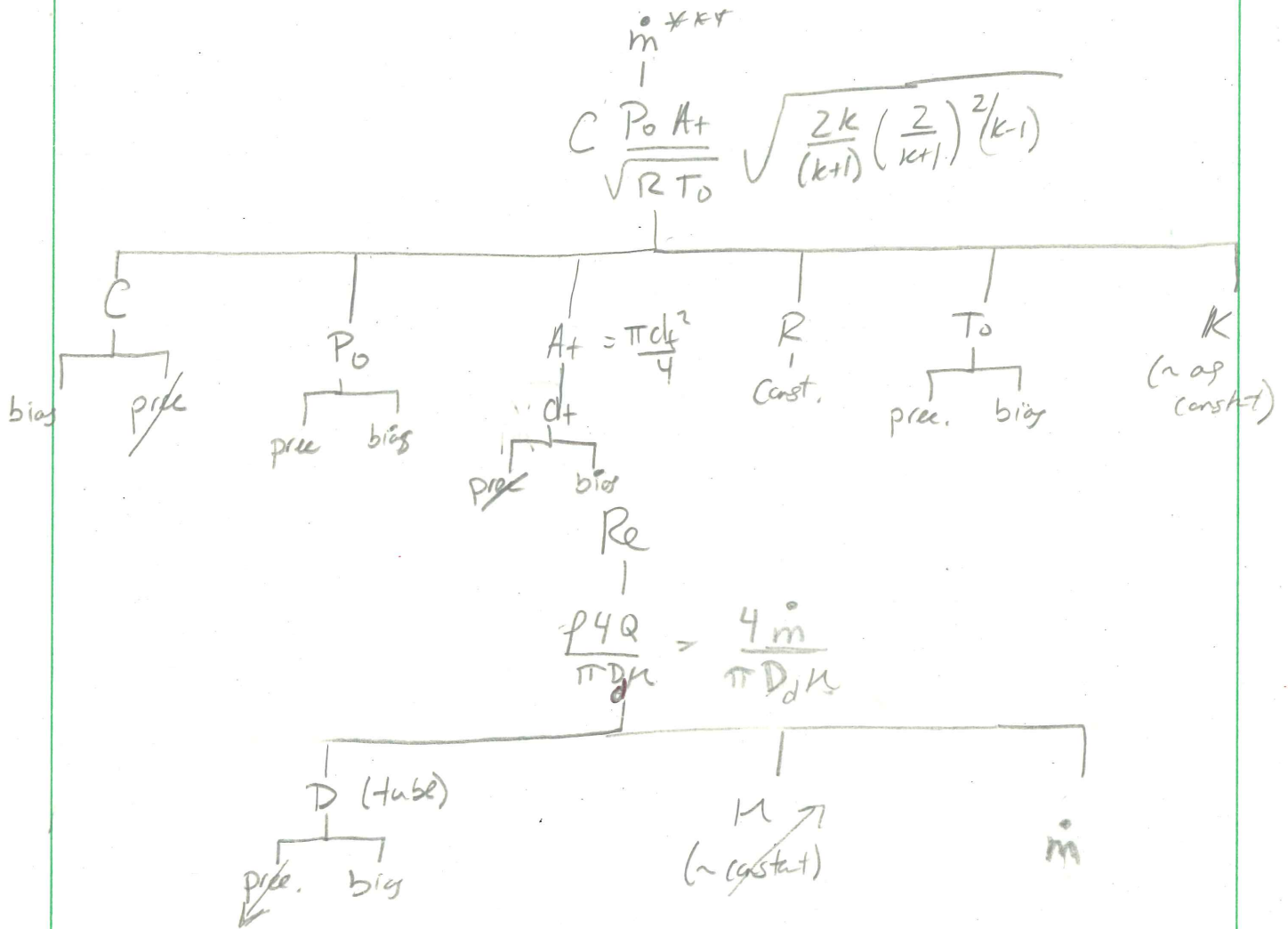


Example of Higher-order Uncertainty Analysis

Determine the uncertainty of the mass flow rate of air through a choked orifice (see governing equation on the laboratory 1 assignment) and the uncertainty in the Reynolds number of the same flow through a tube with a known diameter (0.5 ± 0.01 inches). Hypothetical values are supplied for the instrument uncertainty as well as from repeated measurements. Be sure to apply an uncertainty tree to help with the analysis.

In class on Thursday I will show the solution and answer questions, but we will not have time for me to work through the entire solution.



$$A_+ = \frac{\pi d^2}{4}$$

prec bias

Kline McClintock

$$\dot{m} = \frac{C P_0 A_t}{\sqrt{R T_0}} \underbrace{\sqrt{\frac{2k}{(k+1)} \left(\frac{2}{k+1}\right)^{\frac{2}{k-1}}}}_F$$

$$\frac{\partial \dot{m}}{\partial C} = \frac{P_0 A_t}{\sqrt{R T_0}} F \quad U_{\dot{m}, C} = F \frac{P_0 A_t}{\sqrt{R T_0}} U_C$$

$$\frac{\partial \dot{m}}{\partial P_0} = \frac{C A_t}{\sqrt{R T_0}} F \quad U_{\dot{m}, P_0} = \frac{F C A_t}{\sqrt{R T_0}} U_{P_0}$$

$$\frac{\partial \dot{m}}{\partial A_t} = \frac{C P_0}{\sqrt{R T_0}} F \quad U_{\dot{m}, A_t} = \frac{F C P_0}{\sqrt{R T_0}} U_{A_t}$$

$$\frac{\partial \dot{m}}{\partial T_0} = \frac{-C P_0 A_t F}{2 R^{1/2} T_0^{3/2}} \quad U_{\dot{m}, T_0} = \frac{-C P_0 A_t F}{2 R^{1/2} T_0^{3/2}} U_{T_0}$$

B.4 $U_{\dot{m}} = \sqrt{U_{\dot{m}, C}^2 + U_{\dot{m}, P_0}^2 + U_{\dot{m}, A_t}^2 + U_{\dot{m}, T_0}^2}$

$$Re = \frac{4 \dot{m}}{\pi D \mu}$$

$$\frac{\partial Re}{\partial \dot{m}} = \frac{4}{\pi D \mu} \quad U_{Re, \dot{m}} = \frac{4}{\pi D \mu} U_{\dot{m}}$$

$$\frac{\partial Re}{\partial D} = \frac{-4 \dot{m}}{\pi D^2 \mu} \quad U_{Re, D} = \frac{4 \dot{m}}{\pi D^2 \mu} U_D$$

$$U_{Re} = \sqrt{U_{Re, \dot{m}}^2 + U_{Re, D}^2}$$

$$A_t = \pi d^2 / 4$$

$$\frac{\partial A_t}{\partial d} = \frac{2 \pi d}{4}$$

$$U_{A_t} = \frac{2 \pi d}{4} U_d$$

(4)

Set	Po (PSIG), i.e., pressure tank (i.e., stagnation)	To (oC), i.e., temperature in tank (i.e., stagnation)
1	25	23.2
2	22	23.1
3	32.7	23.5
4	21.2	23.1
5	26	23.7
6	29.1	23.3
7	25.2	23.7

** Note that the values are hypothetical

student-t distribution (95% confidence)

2.447

Note that this value is for 6 degrees of freedom (i.e., 7 samples)

Stagnation Pressure in kPa absolute	Stagnation Temp. in Kelvin
273.4137931	296.2
252.7241379	296.1
326.5172414	296.5
247.2068966	296.1
280.3103448	296.7
301.6896552	296.3
274.7931034	296.7
Mean	Mean
279.5221675	296.3714286
Stdev	Stdev
27.45494221	0.262769136
Precision Uncertainty	Precision Uncertainty
25.3925013	0.243029673

Po (PSIG), i.e., pressure tank	To (oC), i.e., temperature in tank	C	D (in)
2.5 (linearity uncertainty)	0.4 (manufacturer)	0.99 ± 0.01	0.3 ± 0.01
1% of full-scale (up to 30 psi)			

Bias Uncertainty

Stagnation Pressure in kPa absolute	Stagnation Temp. in Kelvin	C (discharge coefficient)	D (m)
17.24137931	0.4		0.00762
2.068965517			
17.36507353	0.4	0.01	0.000254

** Note that the values are hypothetical

	Pressure (kPa)	Temp (K)	C	D (m)	area of throat (m^2)	Ratio of specific heats	Viscosity (kg-m/s)	Ideal gas constant (kJ/kg-K)
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Average value	279.5221675	296.3714286		0.99	0.00762	4.56037E-05	1.4	0.00001846
Uncertainty	30.76239427	0.468042116		0.01	0.000254	3.04024E-06	n/A	n/A

Evaluation of uncertainty for mass flow rate (kg/s)

Constant for m_dot 0.684731456

Sensitivity to independent variable Uncertainty for that independent variable

C 0.000946405 9.46405E-06

stag pressure 3.35194E-06

Area of throat 20.54530602

stag temp -1.58069E-06 -7.39828E-07

Uncertainty in mass flow rate 0.00012093

Average mass flow rate 0.000936941

Percent uncertainty 12.9069254

Evaluation of uncertainty in Re

Sensitivity to independent variable Uncertainty for that independent variable

m_dot 9051560.334

D -1112963.49 -282.6927264

Uncertainty in mass flow rate 1130.522995

Average Re 8480.781792

Percent uncertainty 13.33041013