

Assign #2 Soln. Guide 2016

✓

+5 1. $\psi = U r \sin \theta \left[1 - \frac{a^2}{r^2} \right] \pm \frac{\Gamma}{2\pi} \ln r$

a = cylinder radius ($= .1 \text{ m}$)
 Γ = circulation $= 2\pi a v_\theta(r=a)$
 or $\Gamma = 2\pi a^2 \omega$

+5 $\phi = -U r \cos \theta \left[1 + \frac{a^2}{r^2} \right] \pm \frac{\Gamma \theta}{2\pi}$

U = freestream
 $\Gamma > 0 \quad v_\theta > 0$

in eqn. (+) rotation is clockwise
 (-) rotation is counter-clockwise

ψ - +5 2. Plot ψ & ϕ : conditions $U = 10 \text{ m/s}$

ϕ - +5

ω - +5

$\omega \rightarrow$ select to be $.6 \omega_{\max}$ where ω_{\max} is

when stag. pts. at $\theta = \frac{\pi}{2}$:

or $\sin \theta = 1 = \frac{\Gamma}{4\pi U a}$

$\omega_{\max} = \frac{2U}{a} = \frac{2(10)}{.1} = 200 \frac{\text{rad}}{\text{s}}$

so $\omega = .6(200) = 120 \frac{\text{rad}}{\text{s}}$ and $\Gamma = 2\pi(.1)^2(120) = 7.54$

• select rot. direction (or sign of vortex term) & plot.

Stag. pts. should be at π :

$\sin \theta = \frac{7.54}{4\pi(10)(.1)} = .6 \quad \theta = 36.9^\circ \text{ \& } 143.1^\circ$

3. $U = 4 \rightarrow 15 \text{ m/s}$: use $\sin 90^\circ = \frac{\Gamma}{2\pi U a} = 1$; $\Gamma = 2\pi a^2 \omega_{\max}$

+8 \rightarrow • plot $\omega_{\max} = \frac{2U}{a} = 20(U) \rightarrow$ straight, linear

+8 \rightarrow • derivation: stag. pt. at $\pi/2$: $v_\theta = 0$

$v_\theta = -\frac{\partial \psi}{\partial r} = -\left[U \sin \theta + U \sin \theta \frac{a^2}{r^2} + \frac{\Gamma}{2\pi} \frac{1}{r} \right]$

set $r=a$: $v_\theta = -2U \sin \theta + \frac{\Gamma}{2\pi a}$

for stag. pt. at $v_\theta = 0$ so $\sin \theta = \frac{\Gamma}{4\pi U a}$

4. Start with general Bern. eqn. & find P_s .

$$\int \underbrace{\frac{d\mathbf{g}}{dt}}_{(1)} \cdot d\mathbf{l} + \underbrace{\Delta \frac{g^2}{2}}_{(2)} + \underbrace{\Delta gh}_{(3)} + \underbrace{\int \frac{dP}{\rho}}_{(4)} = \underbrace{\Delta \pi}_{(5)} + \underbrace{F(t)}_{(6)}$$

steady, incomp., irrot. flows results in terms ① & ⑤ & ⑥ = 0
 $\oint \frac{dP}{\rho} = \Delta P/\rho$

apply eqn. between far upstream at $h=0$ (x axis through the center of the cylinder to any pt. on surface "s".

derivation
+8

$$\left(\frac{g^2}{2} + gh + \frac{P}{\rho} \right)_s = \left(\frac{g^2}{2} + gh + \frac{P}{\rho} \right)_\infty = \frac{U^2}{2} + 0 + \frac{P_\infty}{\rho}$$

$$P_s = P_\infty + \frac{U^2}{2} - \left(\frac{g^2}{2} \right)_s - gh_s$$

+5 {

where $g_s = v_\theta = -2U \sin \theta + \frac{\Gamma}{2\pi a}$

$h_s = a \sin \theta$

5. Plot P_s for $U(4 \rightarrow 15)$ this should be a number, say 4-6, values of U for P_s vs. θ (position around cylinder)

+10 (plots)

on a given plot illustrating effect of U on P_s

All shapes are the same, as $U \uparrow P_s$ should go up

+10 (discussion) (or down) relatively

\rightarrow 2 plots $\left\{ \begin{array}{l} \text{stag. pt. at } 45^\circ \\ \text{for the range of } U \text{ values} \end{array} \right. \rightarrow \omega = \frac{2U}{a} \sin 45^\circ = \frac{1.414 U}{a}$

so $\Gamma = 2\pi a^2 \omega = .89 U$

at $90^\circ \rightarrow \omega = \frac{2}{a} U ; \Gamma = 1.26 U$

Increase rotation or Γ will increase the relative Press. (more neg. values) creating greater lift.

6. Numerically integrate P vs. θ to obtain $\int_0^{2\pi} P_s \sin\theta \, d\theta$
 (+13) (integration) results in $L = \int_0^{2\pi} P_s \sin\theta \, d\theta$ for each U and for 2 values of Γ

as U increases L increases; as Γ increases L increases.

discussion

+5

Analytic soln. $L = \rho U \Gamma$

Compare num. integration results with analytic

7. Use plot of C_L given: C_L vs. aw/U

$$C_L = \frac{L}{\frac{1}{2} \rho U^2 (2a)} = \frac{\rho U \Gamma}{\frac{1}{2} \rho U^2 (2a)} = \frac{\rho U 2\pi a^2 \omega}{\frac{1}{2} \rho U^2 (2a)} = \frac{2\pi a \omega}{U}$$

(fig. eqn. for C_L is off by "2" although numbers are correct !!).

— use $2a$ in denominator since it is based on the frontal area per length.

+5 = Plot or make table of $L_{calc.}$ vs. L_{real} for range of U 4 to 15
 (includes discussion)

$$L_{calc.} = \int_0^{2\pi} P_s \sin\theta \, d\theta \text{ from \#6}$$

L_{real} = get data from plot for each aw/U

Compare $\frac{L_{calc.}}{L_{real}}$ for range of U & 2 values of Γ

• since $L_{calc.} \approx \rho U \Gamma$ then this $C_{L, calc.}$ should be close to Theory in the plot.
 (e.g. at $\frac{aw}{U} = 2$ $C_L \approx 12.5$)
 $_{calc.}$

• For fixed w : as $U \uparrow$ $\frac{aw}{U} \downarrow$ and $C_{L, calc.} \downarrow$ and $C_{L, real} \downarrow$ and difference \downarrow .

• For fixed U : as $w \uparrow$ $\frac{aw}{U} \uparrow$ and $C_{L, calc.} \uparrow$ and $C_{L, real} \uparrow$ and difference \uparrow .

+3 \rightarrow Plot $(\frac{1}{U})_{real}$ from fig. for range of U for 2 w 's.