CHAPTER

POWER PLANT





737-600/700/800/900 TASK CARDS

CHAPTER 71 POWER PLANT

Subj	ect/Page	Date	COC	Subj	ect/Page	Date	COC	Subject/Page	Date	COC
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	1	JUN 15/2016		R	3	Jun 15/2016				
	2	BLANK		0	4	Jun 15/2016				
71-0	10-01-01	SYS			5	Feb 15/2015				
	1	Jun 15/2015			6	Oct 15/2014				
	2	Oct 15/2014		R	7	Jun 15/2016				
	3	Oct 15/2014		R	8	Jun 15/2016				
R	4	Jun 15/2016		R	9	Jun 15/2016				
R	5	Jun 15/2016		R	10	Jun 15/2016				
71-0	10-02-01	SYS		R	11	Jun 15/2016				
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R	5	Jun 15/2016		R	16	Jun 15/2016				
71-0	40-01-01	SYS		R	17	Jun 15/2016				
R	1	Jun 15/2016			18	Oct 15/2015				
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R	11	Jun 15/2016								
R	12	Jun 15/2016								
R	13	Jun 15/2016								
R	14	Jun 15/2016								
R	15	Jun 15/2016								
R	16	Jun 15/2016								
R	17	Jun 15/2016								
	18	Oct 15/2015								
71-0	40-02-01	SYS								
R	1	Jun 15/2016								
R	2	Jun 15/2016								

A = Added, R = Revised, D = Deleted, O = Overflow, C = Customer Originated Change

71-EFFECTIVE PAGES



737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	INLET	TITLE COWL - INNER SU		BOEING CARD NO. 71-010-01-01		
DATE	INSPECTION - DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLIO	ADILITY	
	L INLET COWL 1.1 2500 FH 2500 FH		2500 FH		ABILITY		
STATION	SKILL	-			AIRPLANE	ENGINE	
	AIRPL				ALL	ALL	
		ACCESS			ZONE		
					412		
		1					

Detailed inspection of the left inlet cowl's inner surface.

A. References

Reference	Title
AMM 71-11-01-300-801-F01	Replace the Thermal Anti-Ice (TAI) Exhaust Duct (P/B 801)
SRM 54-10-01	Structural Repair Manual

EFFECTIVITY AKS ALL	SOURCE MRB	INLET COWL - INNER SURFACE	
		D633A109-AKS 71-010-01-01	Page 1 of 5 Jun 15/2015



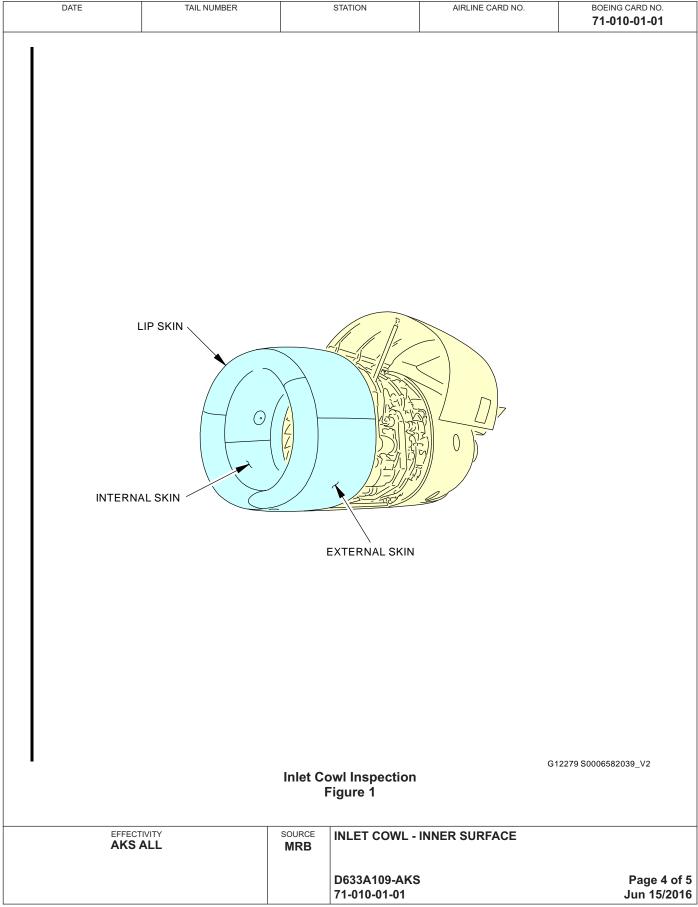
737-600/700/800/900 TASK CARDS

						IAS	K CARDS					
	1	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO 71-010-01-0			
l.	Inle		vl Insp		-801-F00 on				MECI	H INS		
	A.											
		(1)	This	task	examines the skir	n of the inle	t cowl for dam	age.				
	В.	3. Procedure										
	subtask 71-11-01-210-001-F00 (1) Examine the internal skin, the external skin and the lip skin of the inlet cowl for the damage that follows:											
			(a)	Crac	cks							
			(b)	Nick	s, gouges, scratc	hes and co	rrosion					
			(c)	Den	ts							
			(d)	Hole	es							
			(e)	Eros	sion (on the lip ski	n)						
			(f)	Miss	sing sealant betwe	een the lip s	skin segments	(3 locations)				
				1)	Missing sealant re-apply the sea		d between the	segments. It is not necess	sary to			
		suвт/ (2)	ASK 71-11 If you			this referen	nce (SRM 54-	10-01) for the permitted lir	nits.			
		 (3) If you find blockage in the acoustic panel holes in the inner barrel, use the limits for acoustic area loss (Figure 2) (a) Acoustic area loss is usually counted when more than half the holes in an area are blocked and the area is more than one square inch (6.45 square cm). 										
				1)	Holes are count fibers, paint or o		•	ot fully filled with adhesive	, resin,			
				2)	Holes which are considered block	•	are otherwise	clear are permitted and r	not			
	 Some small areas of perforation blockage are part of the new part proc bonding process. 											
		SUBTA	ASK 71-11	-01-210	-003-F00							
		(4)	Visua louve	-	xamine the cowl t	hermal anti	-ice (TAI) duct	louver for signs of cracks	or missing			
			NOT	<u>E</u> : T	he TAI duct louve	r is found a	t the 6 o'clock	position of the inlet cowl.				
			NOT	R T	eplace the Therm	al Anti-Ice	(TAI) Exhaust	luct from the inlet cowl. Do Duct, AMM sary to remove the inlet c				
			(a)		e or more missing nitted to return the	-		o cracks in the slotted are these conditions:	eas, it is			
				CTIVITY		SOURCE MRB	INLET COWL	- INNER SURFACE				

D633A109-AKS 71-010-01-01 Page 2 of 5 Oct 15/2014









737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 71-010-01-01 **ACOUSTIC AREA ACOUSTIC AREA LOST ACOUSTIC AREA NOT LOST** 1 MORE THAN 1 SQUARE INCH (6.45 SQUARE cm) AND MORE THAN 50 OUT OF 100 ADJACENT HOLES BLOCKED. MORE THAN 1 SQUARE INCH (6.45 SQUARE cm) AND LESS THAN 51 OUT OF 100 ADJACENT HOLES BLOCKED. 3 LESS THAN 1 SQUARE INCH (6.45 SQUARE cm). 2016566 S0000398527_V2 **Inner Barrel Acoustic Panel Inspection** Figure 2 **EFFECTIVITY** SOURCE **INLET COWL - INNER SURFACE AKS ALL MRB** D633A109-AKS Page 5 of 5 71-010-01-01 Jun 15/2016



737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	INLET	COWL - INNER SU	BOEING CARD NO. 71-010-02-01		
DATE	INSPECTION - DETAILED				RELATE	D CARD
TAIL NUMBER			THRESHOLD	REPEAT	APPLIC	ABILITY
	R INLET COWL	1.1	2500 FH	2500 FH	AIRPLANE	ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS			ZONE 422	

Detailed inspection of the right inlet cowl's inner surface.

A. References

Reference	Title
AMM 71-11-01-300-801-F01	Replace the Thermal Anti-Ice (TAI) Exhaust Duct (P/B 801)
SRM 54-10-01	Structural Repair Manual

EFFECTIVITY AKS ALL	SOURCE MRB	INLET COWL - INNER SURFACE	
		D633A109-AKS 71-010-02-01	Page 1 of 5 Jun 15/2015



737-600/700/800/900 TASK CARDS

	D	ATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CAR 71-010-02		
	ΤΔςι	K 71.	.11_01		-801-F00					иЕСН	INS
			ıl Insp								
		re 1)			<u></u>						
I	Α.	Gen	eral								
		(1)	This	task	examines the ski	n of the inle	t cowl for dam	age.			
ļ	В.	Prod	cedur								
		SUBTA	SK 71-11	-01-210	-001-F00						
		(1)			the internal skin, that follows:	the external	skin and the l	ip skin of the inlet cowl fo	or the		
			(a)	Crac	cks						
			(b)	Nick	s, gouges, scrato	hes and co	rosion				
			(c)	Den	ts						
			(d)	Hole	es						
			(e)	Eros	sion (on the lip sk	in)					
			(f)	Miss	sing sealant betwe	een the lip s	kin segments	(3 locations)			
	Missing sealant is permitted between the segments. It is not necessary to re-apply the sealant. SUBTASK 71-11-01-220-001-F00						sary to				
	(2) If you find damage, refer to this reference (SRM 54-10-01) for the permitted limits.					mits.					
		(3)	-	น find	I blockage in the a		nel holes in the	e inner barrel, use the lim	its for		
			, ,		area loss (Figure	,					
			(a)			-		re than half the holes in a inch (6.45 square cm).	n area are		
				1)	Holes are count fibers, paint or c		•	ot fully filled with adhesive	e, resin,		
				2)	Holes which are considered bloc		are otherwise	clear are permitted and i	not		
				3)	Some small are bonding process	•	ation blockage	are part of the new part	production		
		SUBTA	SK 71-11	-01-210	-003-F00						
		(4)	Visua louve	-	xamine the cowl	thermal anti	-ice (TAI) duct	louver for signs of cracks	s or missing		
			NOT	<u>E</u> : T	he TAI duct louve	r is found a	the 6 o'clock	position of the inlet cowl.			
			NOT		•			uct from the inlet cowl. D	o this task:		
	Replace the Thermal Anti-Ice (TAI) Exhaust Duct, AMM TASK 71-11-01-300-801-F01. It is not necessary to remove the inlet cowl to do this repair.						cowl to do				
			(a)	If on	•	-		o cracks in the slotted are	eas, it is		
			EFFE(CTIVITY		SOURCE		- INNER SURFACE			
				ALL		MRB	HALLI COAAL	- II4I4LIX GUIVFACE			

D633A109-AKS 71-010-02-01 Page 2 of 5 Oct 15/2014



DATE	TAIL NUMBER	S	STATION	AIRLINE CARD NO.	BOEING (71-010		
				do not exceed 1.0 inch		MECH	INSP
	in length, do these a) Do a stop dri			pair: .1285 inch dia.) at the e	end of the		
	cracks.			,			
	b) Return the a			and in 2000 has after you			
	Missing louvers ar missing louvers or			red in 800 hrs after you ry repair.	note the		
	a) Refer to com	nponent ma	nual for the ap	plicable repair.			
				25.40 mm) in length in a t the vendor (Goodrich			
		END OF T	ASK ——				
EFFECT AKS		SOURCE MRB	NLET COWL - I	NNER SURFACE			
			D633A109-AKS 71-010-02-01			Page 3 Oct 15/	



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 71-010-02-01
INTERNA	LIP SKIN O		EXTERNAL SKIN		
EFFECT AKS	TIVITY	Inlet Co	owl Inspection Figure 1	INNER SURFACE	G12279 S0006582039_V2
ANO			D633A109-AKS 71-010-02-01		Page 4 of 5 Jun 15/2016



737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 71-010-02-01 **ACOUSTIC AREA ACOUSTIC AREA LOST ACOUSTIC AREA NOT LOST** 1 MORE THAN 1 SQUARE INCH (6.45 SQUARE cm) AND MORE THAN 50 OUT OF 100 ADJACENT HOLES BLOCKED. MORE THAN 1 SQUARE INCH (6.45 SQUARE cm) AND LESS THAN 51 OUT OF 100 ADJACENT HOLES BLOCKED. 3 LESS THAN 1 SQUARE INCH (6.45 SQUARE cm). 2016566 S0000398527_V2 **Inner Barrel Acoustic Panel Inspection** Figure 2 **EFFECTIVITY** SOURCE **INLET COWL - INNER SURFACE AKS ALL MRB** D633A109-AKS Page 5 of 5 71-010-02-01 Jun 15/2016





737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	CHECK I	TITLE LEFT ENGINE DRA	BOEING CARD NO. 71-040-01-01		
DATE	TASK OPERATIONAL				RELATE	D CARD
TAIL NUMBER	WORK AREA LEFT ENGINE	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 413 414 431AT			ZONE 411	

Operationally check left engine all drain lines.

A. References

Reference	Title
AMM 71-00-00-800-811-F00	Power Plant Test Reference Table (P/B 501)
AMM 71-11-02-010-801-F00	Open the Fan Cowl Panels (P/B 201)
AMM 71-11-02-410-801-F00	Close the Fan Cowl Panels (P/B 201)
AMM 78-31-00-010-801-F00	Open the Thrust Reverser (Selection) (P/B 201)
AMM 78-31-00-010-804-F00	Close the Thrust Reverser (Selection) (P/B 201)

B. Consumable Materials

Reference	Description	Specification	
D00601	High-temperature graphite compound	SAE AMS 2518	
[CP2101]			

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1280	Source - Air, Regulated, Dry Filtered, 0-30 PSIG
STD-5497	Plug/Cap - To block each port

EFFECTIVITY	SOURCE	CHECK LEFT ENGINE DRAIN LINES	
AKS ALL	MRB		
		D633A109-AKS	Page 1 of 18
		71-040-01-01	Jun 15/2016



	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 71-040-		
ΤΛ	CV 71 :	71 00	700	-801-F00					MECH	IN
				tion (Operationa	l Check)					
			_	and Figure 3)	<u> </u>					
Α.	Gene	eral								
74.	(1)				nance task	which does a	n operational check of the er	ngine		
В.	Draiı	n Line	es In	spection (Opera	tional Chec	ck)				
	SUBTAS	SK 71-71	-00-010	-013-F00						
	(1)	Do th	nis ta	sk: Open the Fan	Cowl Pane	els, AMM TASI	K 71-11-02-010-801-F00.			
	SUBTAS	SK 71-71	-00-010	-010-F00						
			DI RI PA	EACTIVATION PE	ROCEDURE R GROUND OO NOT OB	ES FOR THE MAINTENAN EY THE ABO	CT THE LEADING EDGE, I LEADING EDGE AND THE T ICE), AND OPEN THE FAN (VE SEQUENCE, INJURIES CAN OCCUR.	THRUST COWL		
	(2) Do this task: Open the Thrust Reverser (Selection), AMM TASK 78-31-00-010-801-F00.					01-F00.				
		SK 71-71								
	(3)			steps to prepare	·					
		` '		the drain lines fo		, ,				
		(b)	(Fig	ure 2 and Figure	3).		e engine component to disco			
			NOT	<u>E</u> : Some conne component.	ctions are f	ound at a diffe	erent location than the engine	Э		
			1)	The forward sun	np drain is d	on the fan cas	e (rear) at the 5:00 o'clock p	osition.		
			2)	The TBV drain is						
			3)	The VSV drain i	s on the rod	l-end and hea	d-end of the actuators.			
				-001-F00						
	(4)			erational check b						
		(a)		not necessary to		_	·			
		(b)		onnect the applic y from the compo		ine from the e	ngine component and push t	the line		
			1)	It is not necessa	ry to discor	nnect the oil ta	ınk scupper drain.			
			2)	It can be necess	sary to disco	onnect the cla	mps on some of the drain lin	es.		
	AKS	ALL I	PRE :	SB CFM56-7B 73-	-44					
			3)				sconnect the drain lines from on the drain lines to do the c			
			CTIVITY		SOURCE	CHECK LEFT	ENGINE DRAIN LINES			



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CAF 71-040-0		
AKS ALL	POST SB CFM56-7E	3 73-44				МЕСН	INSP
	a) The BS	V is not install	ed.				
AKS ALL							
	4) For the HPTA	CC valve drai	n, do the applic	able step to get access	to the drain.		
AKS ALL	PRE SB 737-CFM56	-7B-73-045					
	a) Disconn	ect the drain I	ine at the fuel n	nanifold.			
AKS ALL	POST SB 737-CFM5	66-7B-73-045					
	b) Remove	the four bolts	at the fuel mar	nifold.			
AKS ALL							
(c)	Connect an 0-30 p the applicable drain		regulated air so	ource, STD-1280 to the	bottom of		
(d)	Make sure that the	air flows freel	y through the lir	ne.			
(e)	If you find blockage	e, remove the	blockage or rep	lace the drain line.			
(f)	Re-connect the ap	plicable drain	line at the engir	ne component.			
	1) For the HPTA	CC valve drai	n, do the applic	able step to connect the	e drain:		
AKS ALL	PRE SB 737-CFM56	-7B-73-045					
	a) Connec	t the drain line	at the fuel mar	nifold.			
AKS ALL	POST SB 737-CFM5	6-7B-73-045					
	b) Connec	t the fuel man	fold.				
		ubricate the the 1000601 [CP21		ur bolts with graphite co	empound,		
		nstall the gask alve.	et between the	fuel manifold and the H	PTACC		
	<u>N</u>		t the gasket pric amaged or defo	or to installation. Replac ormed.	e the gasket		
	<3> lı	nstall the four	bolts.				
	<4> T	ighten the fou	r bolts to 62-68	pound-inches (7-8 New	/ton-meters).		
AKS ALL							
(g)	Connect the clamp	-					
(h)	of the applicable di		regulated air s	source, STD-1280 from	the bottom		
SUBTASK 71-7	1-00-410-003-F00						
WARNIN	G: OBEY THE INST	RUCTIONS II	N THE PROCE	DURE TO CLOSE THE	THRUST		
				STRUCTIONS, INJURI	ES TO		
(E) Do t	PERSONS AND				0 004 500		
, ,	.nis task: Close the 1 1-00-410-008-F00	must reverse	a (Selection), A	MM TASK 78-31-00-01	U-0U4-FUU.		
		an Cowl Pane	els, AMM TASK	71-11-02-410-801-F00			
EFFE	ECTIVITY S ALL	SOURCE MRB	· I	ENGINE DRAIN LINES			
			D633A109-AKS	3		ge 3 (

71-040-01-01

Jun 15/2016



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	71-040		
AKS ALL P	OST SB 737-CFM56-7B	-73-045				MECH	INSP
SUBTASK 71-71-0							
(7) Do a l	eak check of the HPTA	CC valve	fuel manifold (A	AMM TASK 71-00-00-80	00-811-F00).		
AKS ALL							
		END OF	TASK ———				
EFFECT	FIVITY	SOURCE	CHECK LEET E	NGINE DRAIN LINES			
AKS	ALL	MRB	SIILON LEFT E	INGINE DRAIN LINES			
			D633A109-AKS	;	P	age 4	of 18
			71-040-01-01		.1	un 15/	2016



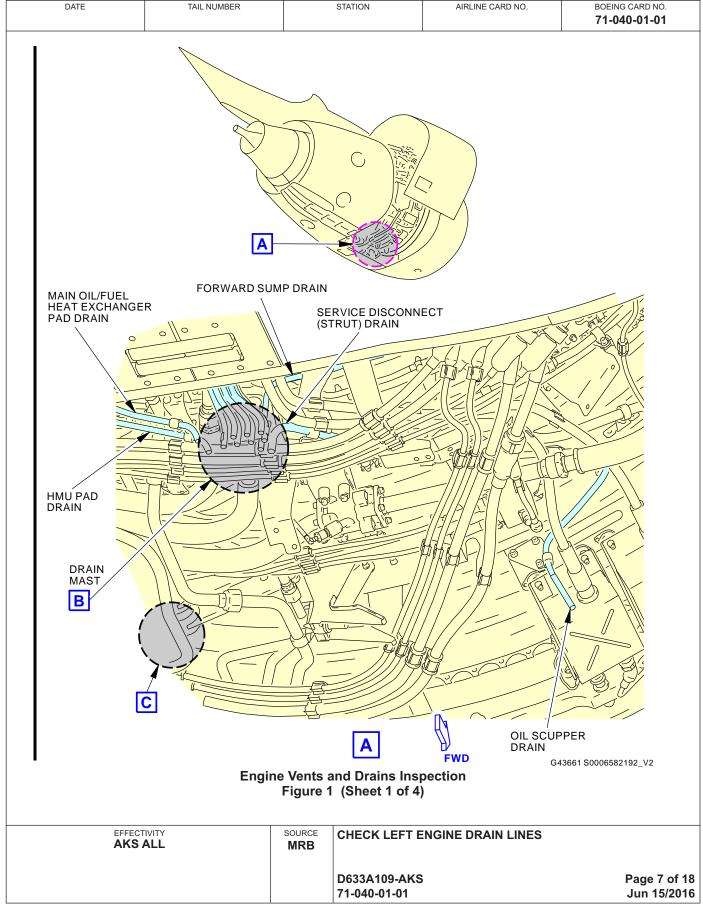
	С	ATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CA 71-040-		
	TAS	K 54.	-55-01-200-8	R01				MECH	INS
				oort Beam Drain -	Operational Test				
•	Figu				oporational root				
	Α.	Gen	oral						
	Λ.	(1)		ives stens to do an	operational test of the	Fan Cowl Support Beam	n Drain		
	В.	()	_	Operational Test	operational tool of the	r an com capport Boan	. 2		
	υ.		ASK 54-55-01-010-0						
				e access panels:					
		()	Number	Name/Location					
			431AT	Forward Strut Fa	iring, Thumbnail Fairin	g, Strut 1			
			441AT	Forward Strut Fa	iring, Thumbnail Fairin	g, Strut 2			
	C.	Stru	ıt Fan Cowl	Support Beam Dr	ain - Operational Test				
		SUBTA	ASK 54-55-01-210-0	02					
		(1)	Make sure material.	that the Fan Cowl	Support Beam and drai	n inlets are free of unwa	nted		
		SUBTA	ASK 54-55-01-710-0	03					
		(2)	Use a Plug	/Cap, STD-5497 to	plug one drain inlet.				
		SUBTA	ASK 54-55-01-710-0	01					
		WAI	PR YO	OTECTION. DO N	OT POINT THE NOZZL	T ON GOGGLES FOR E E AT OTHER PERSONI S, INJURIES TO PERSO	NEL. IF		
		(3)	Use a 0-30 applicable		gulated air source, STE	0-1280 to blow into the to	op of the		
		SUBTA	ASK 54-55-01-710-0	02					
		(4)	Make sure	that the air flows fr	eely through the drain l	ine.			
		SUBTA	ASK 54-55-01-710-0	04					
		(5)	Remove th	e Plug/Cap, STD-5	497.				
			ASK 54-55-01-710-0						
		(6)	Use a Plug	/Cap, STD-5497 to	plug the other drain inl	et.			
			ASK 54-55-01-710-0						
		WAI	PR YO	OTECTION. DO N	OT POINT THE NOZZL	T ON GOGGLES FOR E E AT OTHER PERSONI S, INJURIES TO PERSO	NEL. IF		
		(7)	Use a 0-30 applicable		gulated air source, STE	0-1280 to blow into the to	op of the		
		SUBTA	ASK 54-55-01-710-0						
		(8)	Make sure	that the air flows fr	eely through the drain I	ine.			

EFFECTIVITY AKS ALL	SOURCE MRB	CHECK LEFT ENGINE DRAIN LINES		
			Page 5 e Feb 15/	

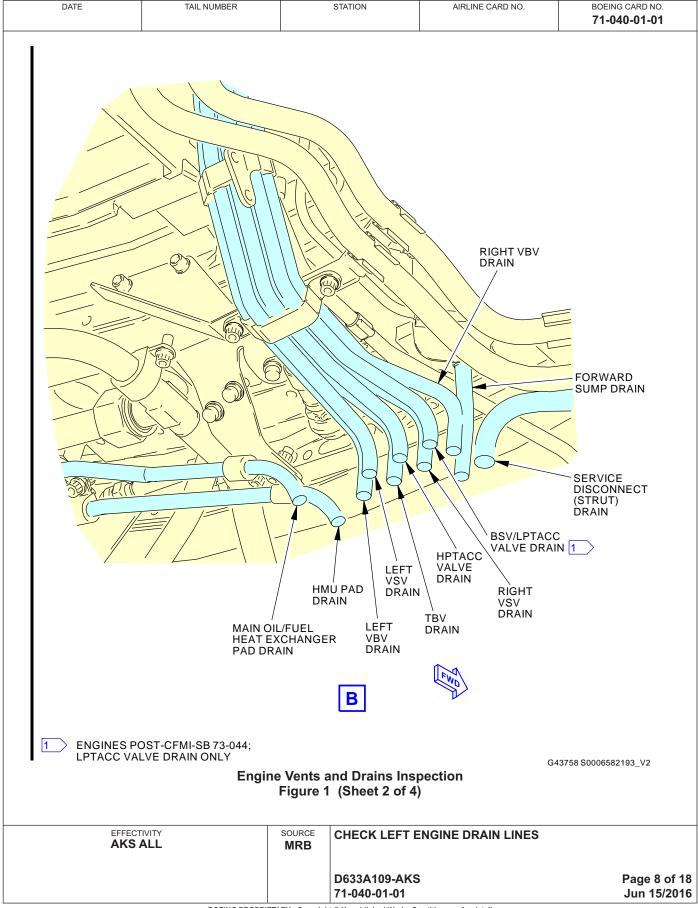


	DA	ΓE	TAIL	. NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ 71-040-		
			1-55-01-710-008 emove the Plu	n/Can_STD- ^r	5497	,			MECH	INSP
			Airplane Bac			tion				
١.			1-55-01-410-001	K to its osu	ai Collait					
			ose the panels	s removed for	r access.					
		(a)	Close these	e access pan	els:					
			<u>Number</u>	Name/Loc						
			431AT 441AT			ng, Thumbnail Fa ng, Thumbnail Fa				
			44 1/1				airing, Strut 2			
					END OF	TASK ———				
		EI A	FECTIVITY KS ALL		SOURCE MRB	CHECK LEFT E	NGINE DRAIN LINES			
						D633A109-AKS		Pa	age 6	of 18
						71-040-01-01		C	ct 15/	2014

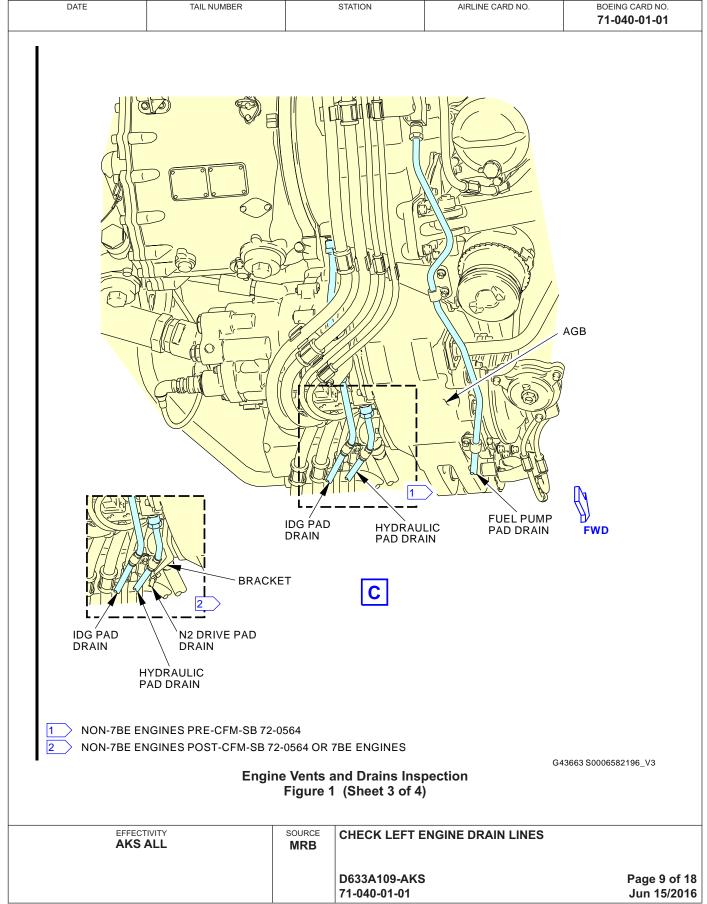




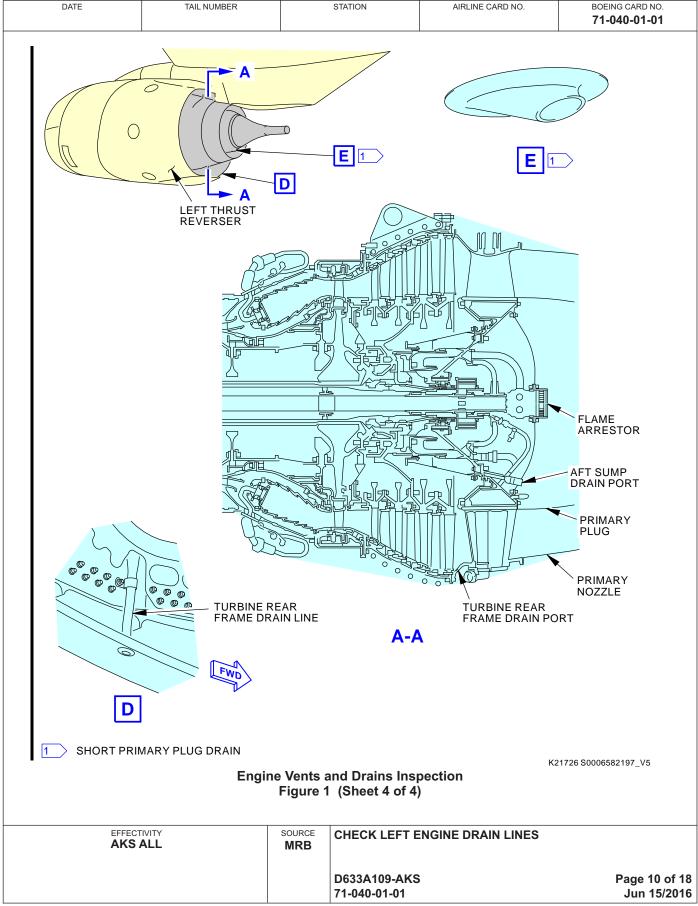




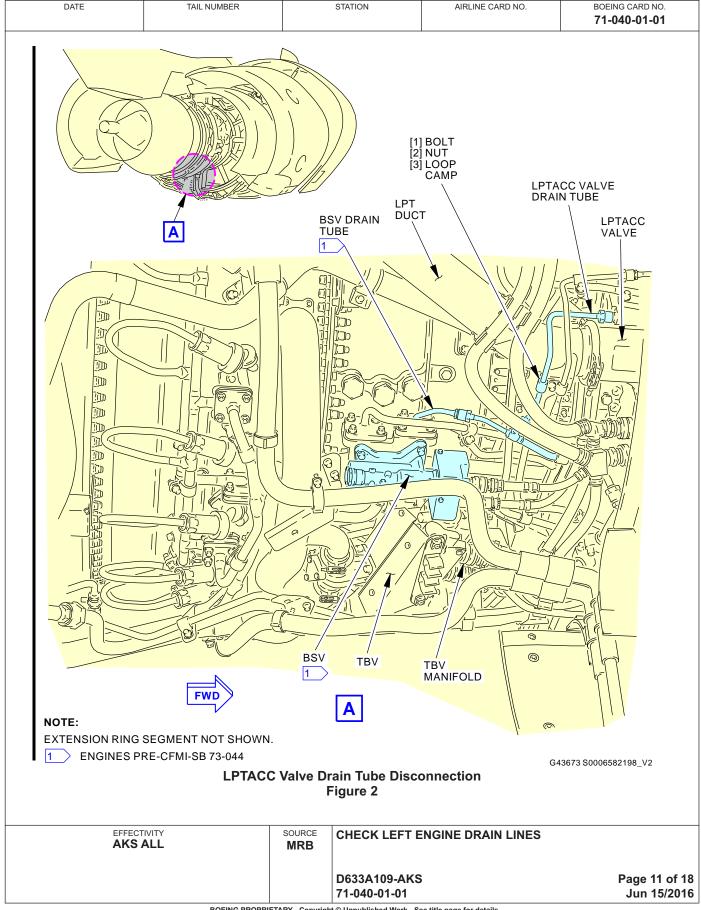




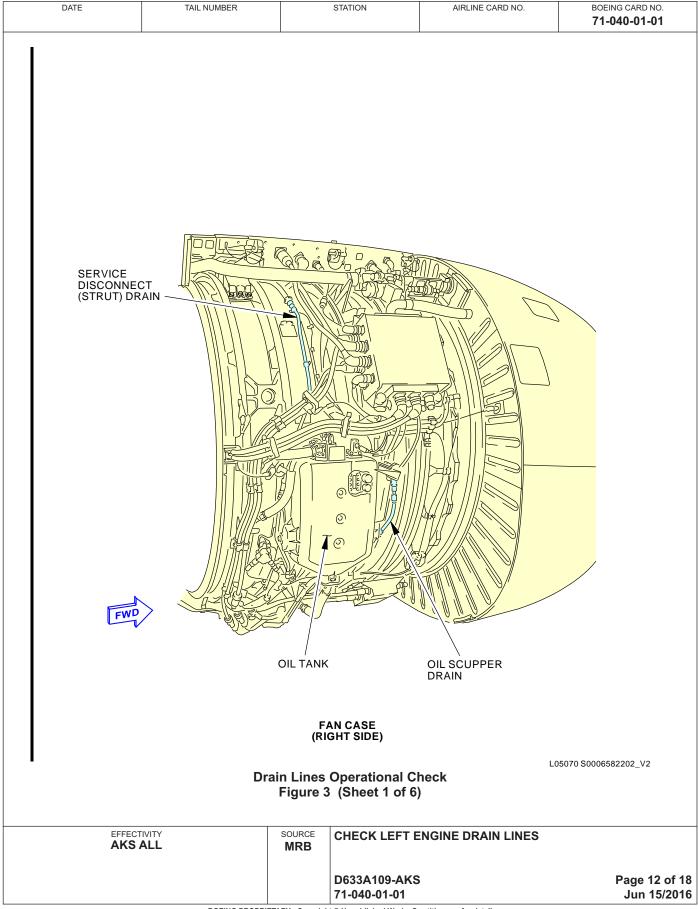




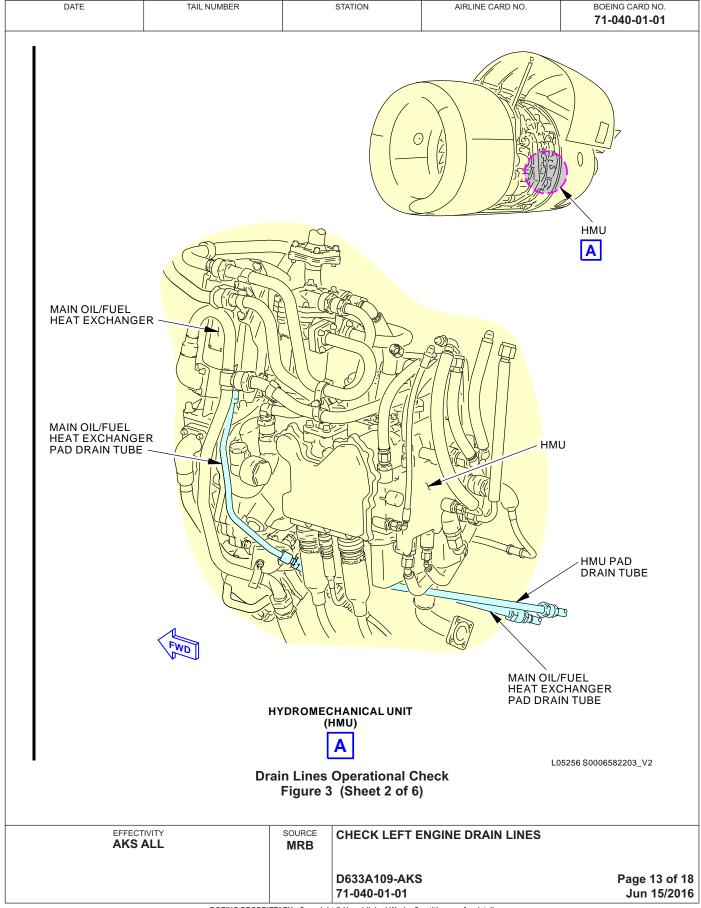




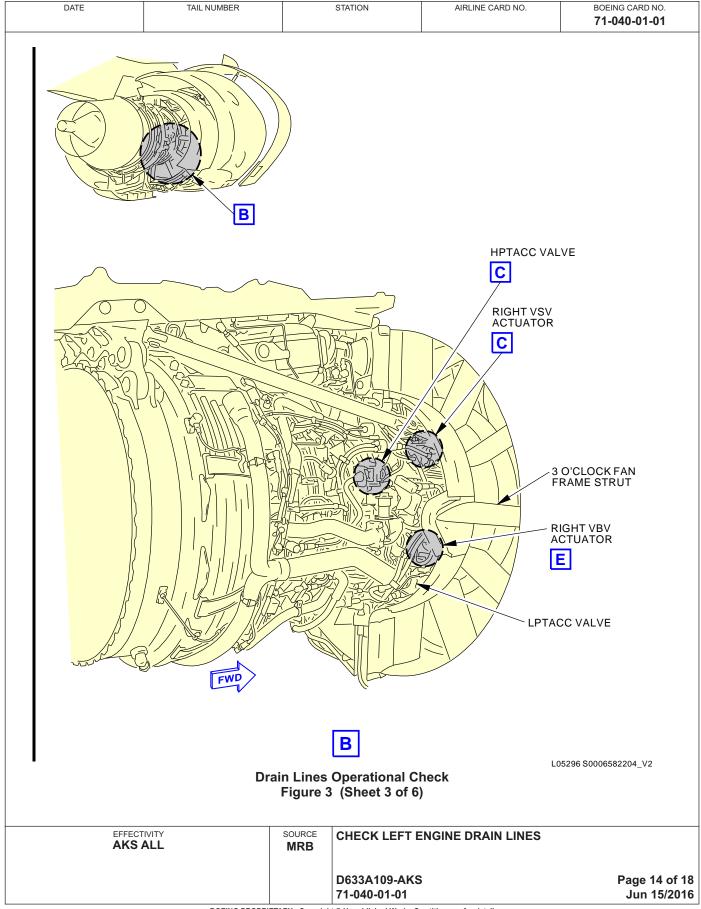




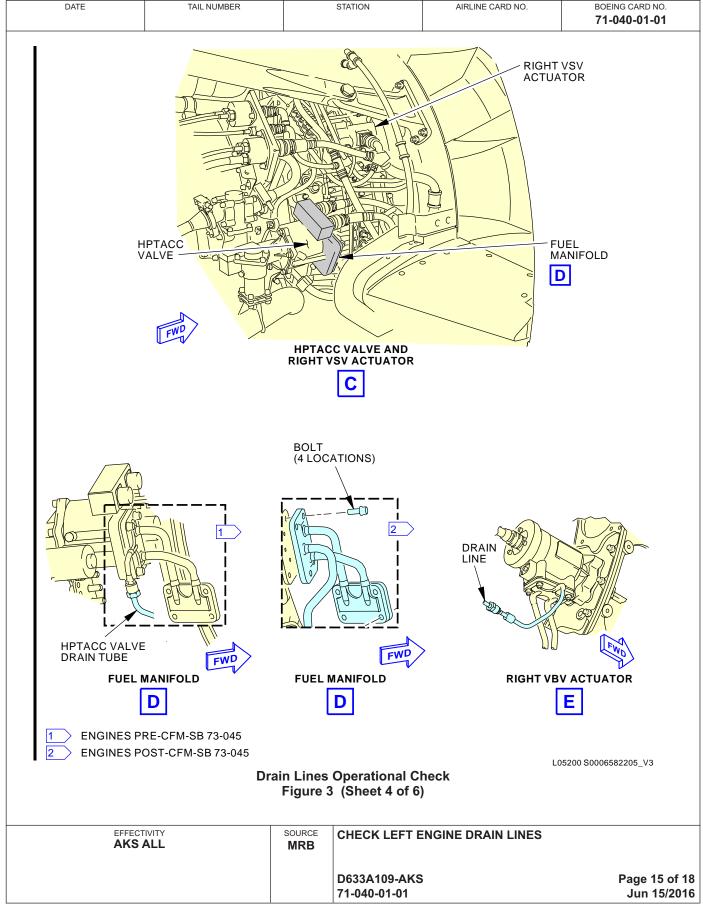




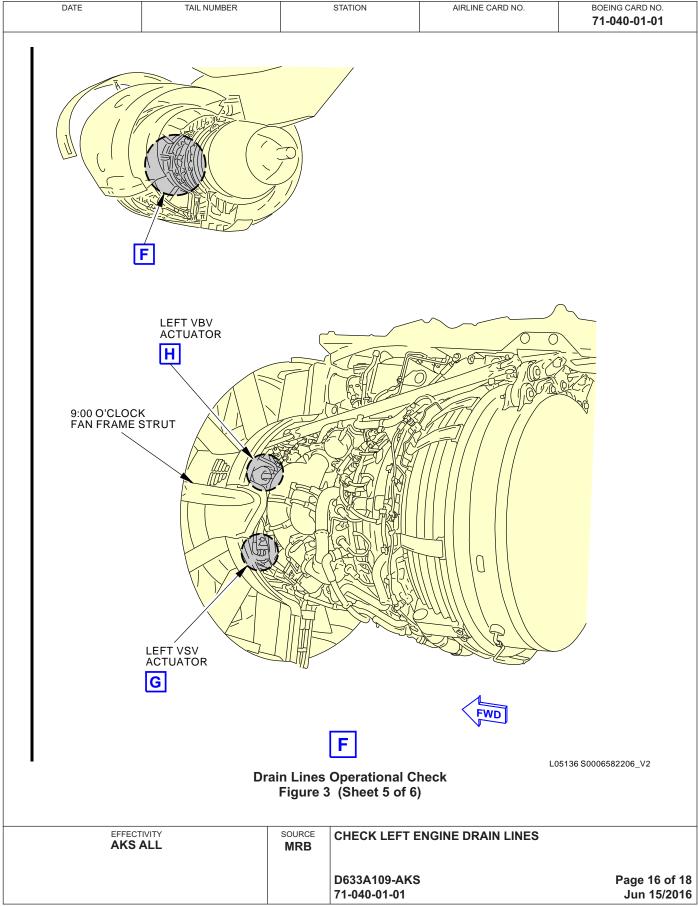




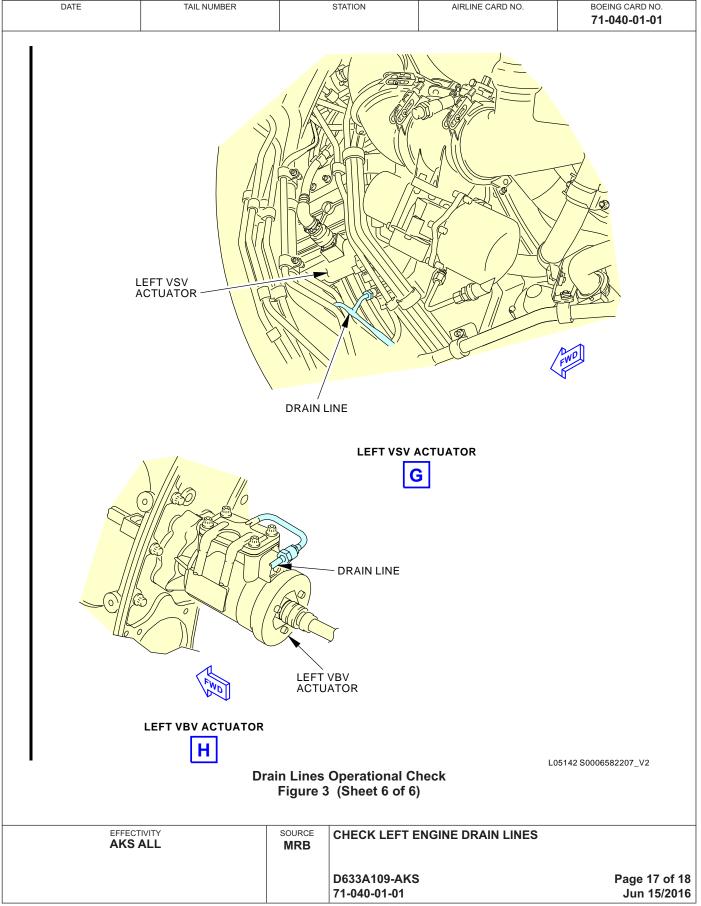




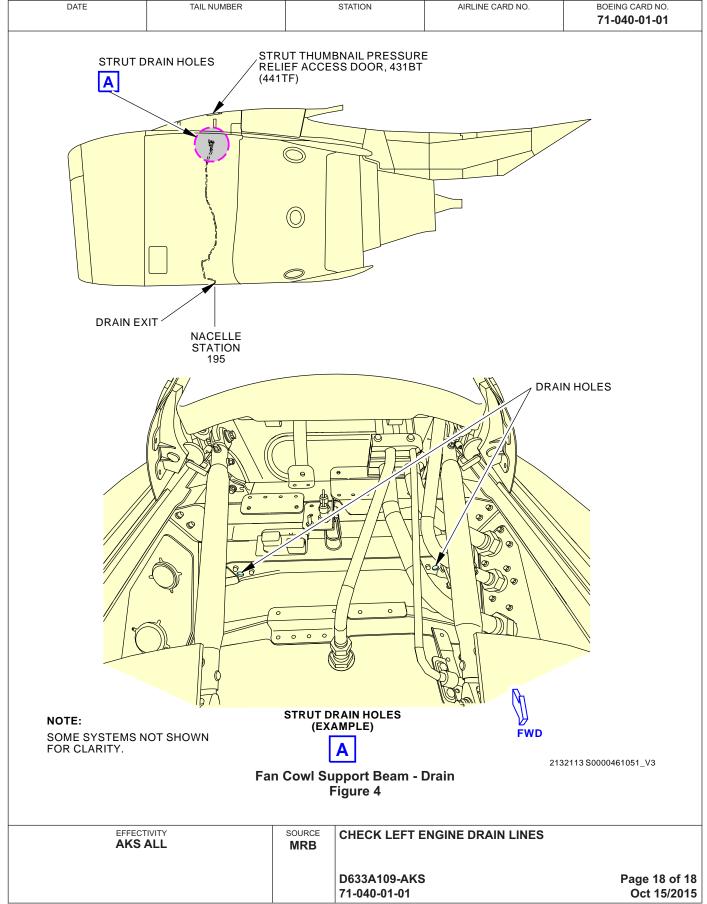
















737-600/700/800/900 TASK CARDS

AIRLIN	E CARD NO	CHECK R	TITLE RIGHT ENGINE DRA	BOEING CARD NO. 71-040-02-01		
DATE	TASK OPERATIONAL				RELATE	D CARD
TAIL NUMBER	WORK AREA RIGHT ENGINE	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 441AT			ZONE 421	

Operationally check right engine all drain lines.

A. References

Reference	Title
AMM 71-00-00-800-811-F00	Power Plant Test Reference Table (P/B 501)
AMM 71-11-02-010-801-F00	Open the Fan Cowl Panels (P/B 201)
AMM 71-11-02-410-801-F00	Close the Fan Cowl Panels (P/B 201)
AMM 78-31-00-010-801-F00	Open the Thrust Reverser (Selection) (P/B 201)
AMM 78-31-00-010-804-F00	Close the Thrust Reverser (Selection) (P/B 201)

B. Consumable Materials

Reference	Description	Specification	
D00601	High-temperature graphite compound	SAE AMS 2518	
[CP2101]			

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1280	Source - Air, Regulated, Dry Filtered, 0-30 PSIG
STD-5497	Plug/Cap - To block each port

EFFECTIVITY AKS ALL	SOURCE MRB	CHECK RIGHT ENGINE DRAIN LINES	
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	DAT	E		TAIL NUM	BER		STATION	AIRLINE CARD NO.	71-040		
T/	ASK	71-7 ⁻	1-00-7	700-801-F00						MECH	INS
. Di	rain	Lines	s Insp	ection (Oper	ational	Check)					
(F	igur	e 1, F	igure	2, and Figure	3)						
Α	۱. (Genei	ral								
	(Γhis is drain I		mainten	ance task	which does a	n operational check of the er	ngine		
В	3. I	Orain	Lines	s Inspection (Operati	ional Chec	ck)				
	8	UBTASK	C 71-71-0	0-010-013-F00							
	(1) [Oo this	s task: Open t	he Fan (Cowl Pane	ls, AMM TAS	K 71-11-02-010-801-F00.			
	8	SUBTASE	C 71-71-0	0-010-010-F00							
	_	WARNING: DO THESE SPECIFIED TASKS IN THE CORRECT SEQUENCE BEFORE YOU OPEN THE THRUST REVERSERS: RETRACT THE LEADING EDGE, DO THE DEACTIVATION PROCEDURES FOR THE LEADING EDGE AND THE THRUST REVERSERS (FOR GROUND MAINTENANCE), AND OPEN THE FAN COWL PANELS. IF YOU DO NOT OBEY THE ABOVE SEQUENCE, INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT CAN OCCUR.									
	(2) [Oo this	s task: Open t	he Thrus	st Reverse	r (Selection),	AMM TASK 78-31-00-010-80	01-F00.		
	8	SUBTASK	C 71-71-0	0-210-004-F00							
	(3) [Do the	ese steps to pr	epare fo	or the proc	edure:				
	(a)		a) F	Find the drain	lines for	the check	(Figure 1):				
		(Find the applic Figure 2 and I			nections at th	ne engine component to disco	onnect		
			<u> </u>	NOTE: Some compo		tions are fo	ound at a diffe	erent location than the engine	Э		
				1) The forward	ard sum	p drain is c	on the fan cas	se (rear) at the 5:00 o'clock p	osition.		
				2) The TBV	drain is	on the TB	V fuel manifol	ld.			
				3) The VSV	drain is	on the rod	l-end and hea	ad-end of the actuators.			
				0-700-001-F00							
	(4) [•							
		(,		•		•	•			
		(ne from the e	engine component and push t	the line		
1) It is not necessary to disconnect the oil tank scupper drain.				ank scupper drain.							
				2) It can be	necessa	ary to disco	onnect the cla	mps on some of the drain lin	es.		
		It can be necessary to disconnect the clamps on some of the drain lines. AKS ALL PRE SB CFM56-7B 73-44									
				,	nts and			sconnect the drain lines from on the drain lines to do the c			
			EFFECT	ΓΙVΙΤΥ		SOURCE	CHECK BIGE	IT ENGINE DRAIN LINES			



737-600/700/800/900 TASK CARDS

DATE	TAIL NUM	BER		STATION	AIRLINE CARD NO.	BOEING C 71-040			
AKS ALL	AKS ALL POST SB CFM56-7B 73-44								
	a) The	BSV is	not installe	ed.					
AKS ALL									
	4) For the H	PTACC	valve drai	n, do the applica	able step to get access	to the drain.			
AKS ALL	AKS ALL PRE SB 737-CFM56-7B-73-045								
	a) Disc	connect t	the drain I	ine at the fuel m	nanifold.				
AKS ALL	POST SB 737-C	FM56-7E	3-73-045						
	b) Ren	nove the	four bolts	at the fuel man	ifold.				
AKS ALL									
(c)	Connect an 0-3 the applicable			regulated air so	ource, STD-1280 to the	bottom of			
(d)	Make sure that	t the air f	lows freel	y through the lir	ne.				
(e)	If you find bloc	kage, re	move the	blockage or rep	lace the drain line.				
(f)	Re-connect the	e applica	ble drain l	ine at the engin	e component.				
	1) For the H	PTACC	valve drai	n, do the applic	able step to connect the	e drain:			
AKS ALL	PRE SB 737-CF	M56-7B-	73-045						
	a) Con	nect the	drain line	at the fuel man	ifold.				
AKS ALL	AKS ALL POST SB 737-CFM56-7B-73-045								
b) Connect the fuel manifold.									
	<1>		cate the th 01 [CP21		ur bolts with graphite co	ompound,			
	<2>	Instal valve		et between the	fuel manifold and the H	IPTACC			
		NOTE		t the gasket pric amaged or defo	or to installation. Replac ormed.	ce the gasket			
	<3>	Instal	I the four	bolts.					
	<4>	Tighte	en the fou	r bolts to 62-68	pound-inches (7-8 Nev	vton-meters).			
AKS ALL									
(g)		-	-	sened to move t					
(h)	(h) Remove the 0-30 psig dry filtered regulated air source, STD-1280 from the bottom of the applicable drain line.								
SUBTASK 71-7	or the applicable drain line.								
WARNING	WARNING: OBEY THE INSTRUCTIONS IN THE PROCEDURE TO CLOSE THE THRUST								
	REVERSER. IF YOU DO NOT OBEY THE INSTRUCTIONS, INJURIES TO								
(5) Do t	PERSONS AND DAMAGE TO EQUIPMENT CAN OCCUR. (5) Do this task: Close the Thrust Reverser (Selection), AMM TASK 78-31-00-010-804-F00.								
` '	1-00-410-008-F00	o mud		(Colodion), A	17.01.70 01-00-01	J JJ T I JU.			
		he Fan (Cowl Pane	els, AMM TASK	71-11-02-410-801-F00				
	ECTIVITY S ALL		SOURCE MRB	CHECK RIGHT	ENGINE DRAIN LINES			1	
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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 71-040		
AKS ALL P	OST SB 737-CFM56-7B	-73-045				MECH	INSP
SUBTASK 71-71-0							
	eak check of the HPTA	CC valve	fuel manifold (A	AMM TASK 71-00-00-80	00-811-F00).		
AKS ALL							
		END OF	TASK ———				
EFFECT AKS	TIVITY ALL	SOURCE MRB	CHECK RIGHT	ENGINE DRAIN LINES			
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			71-040-02-01	1	P	age 4	2016



		DATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CA	RD NO.		
							71-040-0	02-01		
	TAS	K 54-	55-01-200-8	01				MECH	INS	
2.	Stru	ıt Fan	Cowl Supp	ort Beam Drain - 0	Operational Test					
	Figu	ire 4								
	Α.	Gen	eral							
		(1)		ves steps to do an	operational test of the	Fan Cowl Support Beam	Drain.			
	В.									
			SK 54-55-01-010-00	-						
		(1)		access panels:						
		` '	<u>Number</u>	Name/Location						
			431AT	Forward Strut Fai	iring, Thumbnail Fairin	g, Strut 1				
			441AT		iring, Thumbnail Fairin	_				
	C.	Stru	t Fan Cowl	Support Beam Dra	ain - Operational Test	t				
			SK 54-55-01-210-00		•					
		(1)		that the Fan Cowl S	Support Beam and dra	in inlets are free of unwa	nted			
			material.							
		(=)	SK 54-55-01-710-00							
		(2)	Use a Plug	(Cap, STD-5497 to	plug one drain inlet.					
	SUBTASK 54-55-01-710-001									
		WAF	PRO YOU	OTECTION. DO NO	T POINT THE NOZZI	JT ON GOGGLES FOR E LE AT OTHER PERSONN S, INJURIES TO PERSO	NEL. IF			
		(3)	Use a 0-30 applicable of		gulated air source, STI	D-1280 to blow into the to	pp of the			
		SUBTA	SK 54-55-01-710-00	02						
		(4)	Make sure	that the air flows fre	eely through the drain	line.				
		SUBTA	SK 54-55-01-710-00							
		(5)	Remove the	e Plug/Cap, STD-54	197.					
		SUBTA	SK 54-55-01-710-00							
		(6)	Use a Plug	/Cap, STD-5497 to	plug the other drain in	let.				
		SUBTA	SK 54-55-01-710-00	06						
		WAF	PRO YOU	OTECTION. DO NO	T POINT THE NOZZI	JT ON GOGGLES FOR E LE AT OTHER PERSONN S, INJURIES TO PERSO	NEL. IF			
		(7)	Use a 0-30 applicable of		gulated air source, STI	D-1280 to blow into the to	pp of the			
		SUBTA	SK 54-55-01-710-00	7						
		(8)	Make sure	that the air flows fre	eely through the drain	line.				
			EFFECTIVITY		SOURCE CHECK DIGHT	T ENGINE DRAIN LINES				



	DATE	TAIL I	NUMBER		STATION	AIRLINE CARD NO.	BOEING C 71-040				
	subtask 54-	55-01-710-008 nove the Plug	/Can_STD-!	5497				MECH	INSP		
D.		Airplane Bacl			ion						
		subtask 54-55-01-410-001 (1) Close the panels removed for access.									
		(a) Close these access panels:									
	(a)	Number	Name/Loc								
		431AT			ıg, Thumbnail Fa	airing. Strut 1					
		441AT			ng, Thumbnail Fa						
				END OF	TASK ——						
	EFF A	ECTIVITY (S ALL		SOURCE MRB	CHECK RIGHT	ENGINE DRAIN LINES	6				
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