CHAPTER

25

EQUIPMENT/ FURNISHINGS





CHAPTER 25 EQUIPMENT/FURNISHINGS

Subject/Page	Date	coc	Subje	ct/Page	Date	coc	Subject/Page	Date	COC
25-EFFECTIVE	E PAGES		25-09	0-00-01	SYS (cont)		25-190-02-01	SYS (cont)	
1 thru 2	JUN 15/2016			4	Feb 15/2015		3	Feb 15/2015	
25-010-01-01	SYS		25-10	0-00-01	SYS		4	Jun 15/2015	
1	Jun 15/2015			1	Oct 15/2014		5	Jun 15/2015	
2	Feb 15/2016			2	Feb 15/2015		6	Jun 15/2015	
3	Jun 15/2015			3	Jun 15/2015		7	Jun 15/2015	
4	Jun 15/2015		25-13	0-00-01	SYS		25-210-00-01	SYS	
5	Jun 15/2015			1	Jun 15/2015		1	Oct 15/2014	
25-010-02-01	SYS		R	2	Jun 15/2016		2	Feb 15/2015	
1	Jun 15/2015		0	3	Jun 15/2016		3	Jun 15/2015	
2	Feb 15/2016		Α	4	Jun 15/2016		4	Jun 15/2015	
3	Jun 15/2015		Α	5	Jun 15/2016		25-220-00-01	SYS	
4	Jun 15/2015		25-16	0-00-01	SYS		1	Feb 15/2015	
5	Jun 15/2015			1	Oct 15/2014		2	Feb 15/2015	
25-020-00-01	SYS			2	Feb 15/2016		3	Jun 15/2015	
1	Oct 15/2014			3	Feb 15/2016		4	Jun 15/2015	
2	Feb 15/2015		25-16	0-00-02	SYS		25-240-00-01	SYS	
25-030-00-01	SYS			1	Oct 15/2014		1	Jun 15/2015	
1	Oct 15/2014			2	Feb 15/2016		2	Oct 15/2014	
2	Feb 15/2015			3	Feb 15/2016				
25-040-00-01	SYS		25-17	0-00-01	SYS		3	Oct 15/2014	
1	Oct 15/2014			1	Oct 15/2014		4	Feb 15/2015	
2	Oct 15/2014			2	Feb 15/2015		5	Feb 15/2015	
25-045-00-01	SYS		25-17	0-00-02	SYS		6	Feb 15/2015	
1	Jun 15/2015			1	Oct 15/2014		7	Feb 15/2015	
R 2	Jun 15/2016			2	Jun 15/2015		8	Oct 15/2014	
25-050-00-01	SYS		25-19	0-01-01	SYS		9	Oct 15/2014	
1	Jun 15/2015			1	Oct 15/2015		10	Jun 15/2015	
2	Oct 15/2014			2	Feb 15/2015		11	Jun 15/2015	
25-070-00-01	SYS			3	Feb 15/2015		12	Jun 15/2015	
1	Oct 15/2014			4	Jun 15/2015		13	Jun 15/2015	
2	Feb 15/2015			5	Jun 15/2015		14	Jun 15/2015	
3	Feb 15/2015			6	Jun 15/2015		15	Jun 15/2015	
25-090-00-01	SYS			7	Jun 15/2015		16	Feb 15/2015	
1	Oct 15/2014		25-19	0-02-01	SYS		17	Jun 15/2015	
2	Feb 15/2015			1	Oct 15/2015		18	Jun 15/2015	
3	Feb 15/2015			2	Feb 15/2015		19	Jun 15/2015	

A = Added, R = Revised, D = Deleted, O = Overflow, C = Customer Originated Change

25-EFFECTIVE PAGES





CHAPTER 25 EQUIPMENT/FURNISHINGS

Subject/Page	Date	COC	Subject/Page	Date	COC	Subject/Page	Date	COC
25-240-00-02	SYS		25-240-00-03	SYS (cont)		25-360-00-01	SYS	
1	Jun 15/2015		17	Jun 15/2015		1	Oct 15/2014	
2	Oct 15/2014		18	Jun 15/2015		2	Feb 15/2015	
3	Oct 15/2014		19	Jun 15/2015		3	Feb 15/2015	
4	Feb 15/2015		25-240-00-04	SYS		25-370-00-01	SYS	
5	Feb 15/2015		1	Jun 15/2015		1	Oct 15/2014	
6	Feb 15/2015		2	Oct 15/2014		2	Feb 15/2015	
7	Feb 15/2015		3	Oct 15/2014		25-380-00-01	SYS	
8	Oct 15/2014		4	Feb 15/2015		1	Oct 15/2014	
9	Oct 15/2014		5	Feb 15/2015		R 2	Jun 15/2016	
10	Jun 15/2015		6	Feb 15/2015		R 3	Jun 15/2016	
11	Jun 15/2015		7	Feb 15/2015		O 4	Jun 15/2016	
12	Jun 15/2015		8	Oct 15/2014		25-390-00-01	SYS	
13	Jun 15/2015		9	Oct 15/2014		1	Oct 15/2014	
14	Jun 15/2015		10	Jun 15/2015		2	Feb 15/2015	
15	Jun 15/2015		11	Jun 15/2015		25-400-00-01	SYS	
16	Feb 15/2015		12	Jun 15/2015		1	Oct 15/2014	
17	Jun 15/2015		13	Jun 15/2015		R 2	Jun 15/2016	
18	Jun 15/2015		14	Jun 15/2015		25-410-00-01	SYS	
19	Jun 15/2015		15	Jun 15/2015		1	Jun 15/2015	
25-240-00-03	SYS		16	Feb 15/2015		2	Feb 15/2015	
1	Jun 15/2015		17	Jun 15/2015		25-420-00-01	SYS	
2	Oct 15/2014		18	Jun 15/2015		1	Oct 15/2014	
3	Oct 15/2014		19	Jun 15/2015		2	Feb 15/2015	
4	Feb 15/2015		25-290-00-01	SYS		25-430-00-01	SYS	
5	Feb 15/2015		1	Oct 15/2014		1	Oct 15/2014	
6	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
7	Feb 15/2015		25-330-00-01	SYS				
8	Oct 15/2014		1	Oct 15/2014				
9	Oct 15/2014		2	Feb 15/2015				
10	Jun 15/2015		25-340-00-01	SYS				
11	Jun 15/2015		1	Oct 15/2014				
12	Jun 15/2015		2	Feb 15/2015				
13	Jun 15/2015		25-350-00-01	SYS				
14	Jun 15/2015		1	Oct 15/2014				
15	Jun 15/2015		2	Feb 15/2015				
16	Feb 15/2015		3	Feb 15/2015				

 $A = Added, \ R = Revised, \ D = Deleted, \ O = Overflow, \ C = Customer \ Originated \ Change$

25-EFFECTIVE PAGES





AIRLINI	E CARD NO	CAPTAINS	TITLE SEAT TRACKS AN		BOEING CARD NO. 25-010-01-01		
DATE	INSPECTION - DETAILED		MECHANISM	RELATED CARD			
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 7500 FH	REPEAT 7500 FH	APPLICA		
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL	
		ACCESS			ZONE 211		
		1					

Inspect (Detailed) the captains seat tracks and locking mechanism for wear, condition, and security.

A. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
SPL-1638	Gauge - Inspection, Seat Track, Flight Crew Seats (A25020-5 part of A25020-8)
	Part #: A25020-8 Supplier: 81205 Opt Part #: A25020-4 Supplier: 81205

EFFECTIVITY AKS ALL	MRB	CAPTAINS SEAT TRACKS AND LOCKING MECHAN	IISM
		D633A109-AKS 25-010-01-01	Page 1 of 5 Jun 15/2015



737-600/700/800/900 TASK CARDS

	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-010-		
TAS	SK 25	-11-01	1-200	-802					MECH	IN
				Officer Seat Track	and Lock	ing Mechanis	m Inspection			
(Fig	jure 1	and F	igure	2)						
A.	Pro	cedui	re							
	SUBT	ASK 25-1	1-01-210	-003						
	(1)	the	floor:				icer's seat tracks are fully			
		NOT		lse the horizontal ne track.	control leve	er to move the	seat forward and aft to e	xamine all of		
		(a)	Try t	to move the seat	racks to ma	ake sure they a	are securely attached.			
			1)	If you find loose	bolts on the	e seat tracks, t	ighten them.			
		(b)	Do a	a visual inspection	for cracks	and corrosion	on the seat tracks.			
			1-01-210							
	(2)			steps to inspect the						
		(a)			ons during v	which the seat	track can become worn:			
			1)	Condition 1:						
				a) The lock plant adjusted.	in rubs on t	he surface of t	he seat track while the so	eat is		
			2)	Condition 2:						
			_,	a) The lock h		-	the locking pin hits the eseat is adjusted.	dge of the		
		(b)	The	seat track is in go	-		•			
			1)	The maximum d than 0.065 in. (1			urface of the seat track is	s not more		
			2)	The lock hole is (2.54 mm) for th	•	,	s in diameter at a depth o	of 0.1 in.		
	SUBT	ASK 25-1	1-01-210	-004						
	(3)	Do t	hese	steps to do a che	ck of the su	ırface of the se	eat track:			
		(a)		the point of the in ne seat track.	spection ga	luge, SPL-163	8, in a groove that is on t	he surface		
		(b)	If the	e protrusion touch	es the surf	ace of the seat	t track, replace the seat t	rack.		
	SUBT		1-01-210							
	(4)	Do t		steps to do a che						
		(a)		·	•		1638, in the lock hole.			
		(b)		e bottom of the ins seat track.	spection ga	ge touches the	e surface of the seat trac	k, replace		
			ECTIVITY		SOURCE MRB	CAPTAINS SE	EAT TRACKS AND LOCKIN	NG MECHANIS	SM	
							_			

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DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-010		
						25-010	MECH	INSP
suвт <i>і</i> (5)	ASK 25-11-0 [.] Do. a. v	₁₋₂₁₀₋₀₀₆ visual inspection to mak	e sure th	e seat track lock	k pins are fully engaged	l at all seat		
(5)		ock positions.	o ouro ill	o ocal track look	A Pillo alo lally ellgaget	i at all 36at		
	NOTE	: Use the horizontal co			eat forward and aft to e	xamine lock		
		pin engagement in all						
	(a) N	Make sure seat track loc			in all seat track lock po	sitions.		
			END OF	TASK ——				
	EFFECT	rivity	SOURCE	CAPTAINS SEA	T TRACKS AND LOCKIN	IG MECHANIS	SM	
	AKS	ALL	MRB				-	
				D633A109-AKS		I	Page 3	3 of 5
				25-010-01-01		J	un 15/	2015



737-600/700/800/900 TASK CARDS

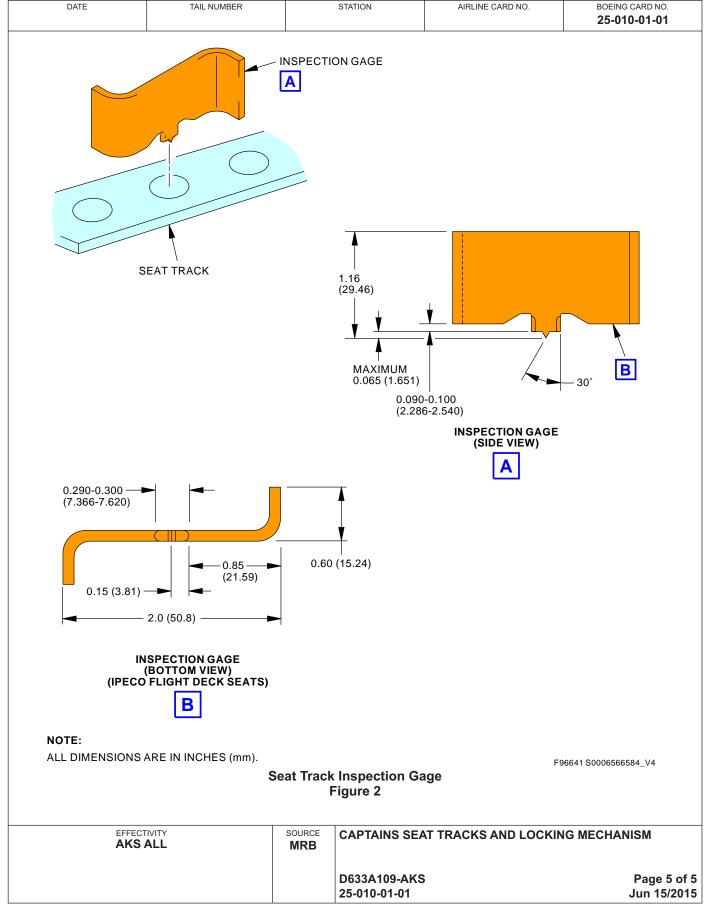
DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-010-01-01 **INSPECTION GAGE** THE POINT OF THE INSPECTION GAGE IS IN THE GROOVE ON THE SEAT TRACK THE PROTRUSION MUST NOT TOUCH SEAT TRACK THE TRACK **GROOVE WORN** IN SEAT TRACK INSPECTION ON THE SURFACE OF THE SEAT TRACK **INSPECTION GAGE SEAT TRACK** LOCK HOLE THE BOTTOM OF THE INSPECTION GAGE MUST NOT TOUCH THE TOP OF THE TRACK THE PROTRUSION OF THE INSPECTION GAGE IS IN THE LOCK HOLE INSPECTION OF THE LOCK HOLE F96655 S0006566583_V2 **Seat Track Inspection** Figure 1 **EFFECTIVITY** SOURCE CAPTAINS SEAT TRACKS AND LOCKING MECHANISM **AKS ALL MRB**

D633A109-AKS

25-010-01-01

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AIRLINI	E CARD NO	FIRST OFFICE	TITLE ER SEAT TRACKS	BOEING CARD NO. 25-010-02-01			
DATE	INSPECTION - DETAILED		MECHANISM	RELATED CARD			
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 7500 FH	REPEAT 7500 FH	APPLICABILITY AIRPLANE ENCINE		
STATION	SKILL AIRPL				AIRPLANE ALL	ALL ALL	
		ACCESS			ZONE 212		
		1					

Inspect (Detailed) the first officers seat tracks and locking mechanism for wear, condition, and security.

A. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
SPL-1638	Gauge - Inspection, Seat Track, Flight Crew Seats (A25020-5 part of A25020-8)
	Part #: A25020-8 Supplier: 81205 Opt Part #: A25020-4 Supplier: 81205

EFFECTIVITY AKS ALL	SOURCE MRB	FIRST OFFICER SEAT TRACKS AND LOCKING ME	ECHANISM
		D633A109-AKS 25-010-02-01	Page 1 of 5 Jun 15/2015



	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-010-				
TA	SK 25	-11-0 ²	1-200	-802	I				MECH	IN		
Ca	ptain a	and F	irst C	Officer Seat Trac	k and Lock	ing Mechanis	sm Inspection					
	gure 1						<u> </u>					
Α.	Pro	cedu	re									
	SUBT	ASK 25-1	SK 25-11-01-210-003									
	(1)		Do these steps to make sure the captain's or first officer's seat tracks are fully attached to he floor:									
		NO		Ise the horizontal ne track.	control leve	er to move the	seat forward and aft to e	examine all of				
		(a)	Try	to move the seat	tracks to ma	ake sure they	are securely attached.					
			1)	If you find loose	bolts on the	e seat tracks,	tighten them.					
		(b)	Do a	a visual inspectio	n for cracks	and corrosion	on the seat tracks.					
	SUBT	ASK 25-1	11-01-210	-007								
	(2)	Do t		steps to inspect								
		(a)			ions during v	which the sea	t track can become worn:					
			1)	Condition 1:								
				a) The lock padjusted.	oin rubs on t	he surface of	the seat track while the s	eat is				
			2)	Condition 2:								
				,		-	the locking pin hits the e seat is adjusted.	edge of the				
		(b)	The	seat track is in g	ood condition	on when:						
			1)	The maximum of than 0.065 in. (surface of the seat track is	s not more				
			2)	The lock hole is (2.54 mm) for the			s in diameter at a depth o	of 0.1 in.				
	SUBT		11-01-210									
	(3)	Do t		steps to do a che								
		(a)		the point of the ir ne seat track.	nspection ga	luge, SPL-163	38, in a groove that is on	the surface				
		(b)	If th	e protrusion touc	hes the surfa	ace of the sea	at track, replace the seat	track.				
			11-01-210									
	(4)			steps to do a che								
		(a)										
		(b)		e bottom of the ir seat track.	ispection ga	ge touches th	e surface of the seat trac	к, геріасе				
			ECTIVITY		SOURCE MRB	FIRST OFFIC	ER SEAT TRACKS AND L	OCKING MECI	HANIS	SIV		





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-010		
SUBTA	SK 25-11-0	1-210-006					MECH	INSP
(5)		visual inspection to ma ock positions.	ike sure th	e seat track locl	k pins are fully engaged	at all seat		
	NOTE	: Use the horizontal or pin engagement in a			eat forward and aft to e	xamine lock		
	(a) I				in all seat track lock po	sitions.		
			- END OF	TASK ——				
	AKS	rivity ALL	SOURCE MRB	FIRST OFFICER	R SEAT TRACKS AND LO	OCKING MEC	HANIS	М
				D633A109-AKS			Page 3	
				25-010-02-01		J	un 15/	2015



737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-010-02-01 **INSPECTION GAGE** THE POINT OF THE INSPECTION GAGE IS IN THE GROOVE ON THE SEAT TRACK THE PROTRUSION MUST NOT TOUCH SEAT TRACK THE TRACK **GROOVE WORN** IN SEAT TRACK INSPECTION ON THE SURFACE OF THE SEAT TRACK **INSPECTION GAGE** SEAT TRACK LOCK HOLE THE BOTTOM OF THE INSPECTION GAGE MUST NOT TOUCH THE TOP OF THE TRACK THE PROTRUSION OF THE INSPECTION GAGE IS IN THE LOCK HOLE INSPECTION OF THE LOCK HOLE F96655 S0006566583_V2 **Seat Track Inspection** Figure 1 **EFFECTIVITY** SOURCE FIRST OFFICER SEAT TRACKS AND LOCKING MECHANISM **AKS ALL MRB**

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25-010-02-01

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TAIL NUMBER

DATE

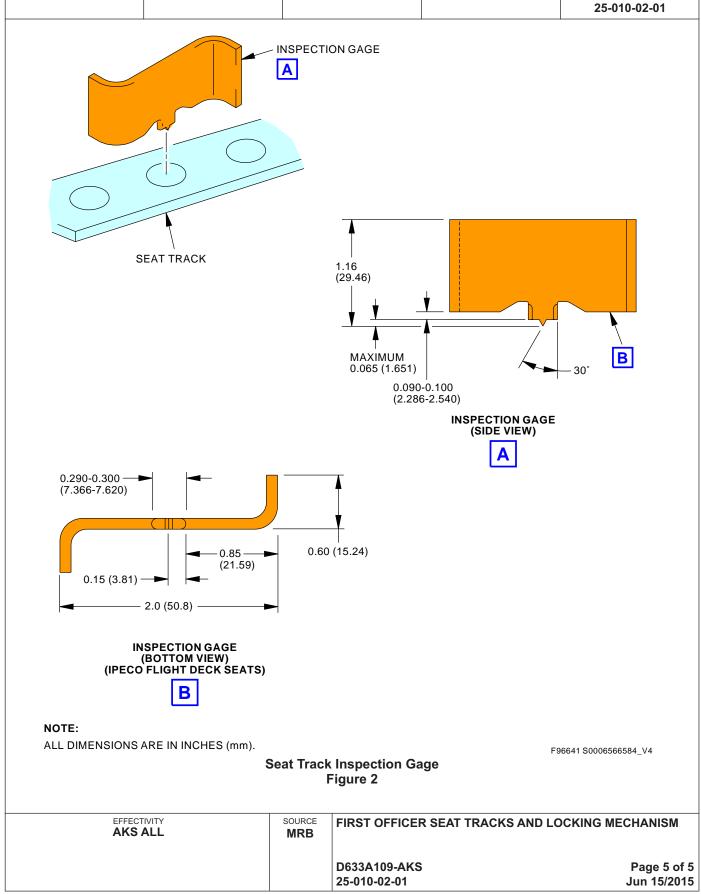


737-600/700/800/900 TASK CARDS

STATION

AIRLINE CARD NO.

BOEING CARD NO.







AIRLIN	E CARD NO	F. 101:-	001121	TITLE	EW OF AT		BOEING CAI	
				RTMENT CR STRAPS, AN			25-020-0	
DATE	INSPECTION - DETAILED	HARNE	.33L3, C	JINAF S, AN	D BEE13		RELATED (CARD
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1		RESHOLD	REPEAT 600 FH	AIRP	APPLICAB	BILITY
STATION	SKILL AIRPL					Al		ALL
		ACCESS				ZONE 210		
Straps, and s	houlder belts (as	applicable) for w	vear, co	nation, and	Security.			
	EFFECTIVITY AKS ALL		OURCE MRB	FLIGHT CON	MPARTMENT C	REW SEAT	HARNESS	SES, STRAPS,
				D633A109-A 25-020-00-0	KS			Page 1 of 2 Oct 15/2014



DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD 25-020-00-									
	TAS	K 25-11-00-2	200-802					MECH	INSP
1.			Officer's, and Observ	er's Seat	Visual Harness	s Check			
	A.	Procedure							
		SUBTASK 25-11-00	0-210-001						
			risual check of the restr						
		` '	Make sure that the strap		•				
			Look for loose stite	ching, cut	s, tears, fraying,	and discoloration.			
		SUBTASK 25-11-00	o-210-002 risual check of the restr	aint harn	occ fittings:				
			Make sure that the fitting		_	corroded			
		SUBTASK 25-11-00		go are no	n bent, 10000, or	conoded.			
			risual check of the restr	aint harn	ess securing ha	dware:			
		(a) N	Make sure that the secu	iring hard	lware is not bent	, loose, or corroded.			
				END OF	TASK ———				
		EFFECT	TIVITY	SOURCE	FLIGHT COMPA	RTMENT CREW SEAT	HARNESSES	STRA	PS
		AKS		MRB	AND BELTS	INTIMENT ONLW SEAT	HARRIEGGEG,	JINA	. 0,
					D633A109-AKS 25-020-00-01			Page 2 eb 15/	





CREW CABIN 1.1 3500 FH 3500 FH AIRPLANE ENOME ALL ALL ACCESS ZONE 210 Derationally check the captain, first officer, and the first observer seat harness inertia reels.				IASK CARDS			
APPLICABILITY WORKASEA CREW CABIN STATION SIGL AIRPL ACCESS Derationally check the captain, first officer, and the first observer seat harness inertia reels. PROPERTY OF THE STATION AIRPL ACCESS 2010 Derationally check the captain, first officer, and the first observer seat harness inertia reels. EPPECTIVITY AKS ALL MRB CHECK CAPTAIN, FIRST OFFICER, & FIRST OBSERVER S HARNESS REELS D63A109-AKS Page	AIRLINE	E CARD NO	CHECK CAF	TAIN, FIRST OFFI	CER, & FIRST		
CREW CABIN 1.1 3500 FH 3500 FH AIRPLANE ENGINE ALL ALL ACCESS ZONE 210 Derationally check the captain, first officer, and the first observer seat hamess inertia reels. SOURCE AKS ALL MRB CHECK CAPTAIN, FIRST OFFICER, & FIRST OBSERVER SHARNESS REELS D63A109-AKS Page D63A109-AKS Page	DATE		OBSERV	ER SEAT HARNES	SS REELS	RELATE	D CARD
STATION SKILL ALL ALL ACCESS 2006 210 Derationally check the captain, first officer, and the first observer seat harness inertia reels. EFFECTIVITY AKS ALL SOURCE MRS CHECK CAPTAIN, FIRST OFFICER, & FIRST OBSERVER S HARNESS REELS D633A109-AKS Page CHECK CAPTAIN, FIRST OFFICER, & FIRST OBSERVER S HARNESS REELS D633A109-AKS Page	AIL NUMBER						
perationally check the captain, first officer, and the first observer seat harness inertia reels. CHECK CAPTAIN, FIRST OFFICER, & FIRST OBSERVER S HARNESS REELS D633A109-AKS Page Pag	STATION						
EFFECTIVITY AKS ALL SOURCE MRB CHECK CAPTAIN, FIRST OFFICER, & FIRST OBSERVER S HARNESS REELS D633A109-AKS Page			ACCESS				
D633A109-AKS Page							
		EFFECTIVITY AKS ALL		URCE CHECK CA	PTAIN, FIRST OF	FFICER, & FIRST O	BSERVER S
							Page 1



		DATE		TAIL NUMBER	R		STATION	AIRLINE CARD NO.	BOEING C 25-030		
	TAS	K 25-	11-00-	200-801			-			MECH	INSI
1.	Cap	tain,	First C	Officer, and Obse	erver Se	eat Harn	ess and Inertia	Reel Operational Ch	eck		
	A.	Prod	cedure)							
				00-200-001							
		(1)	Do th	e steps that follow	w to test	the rest	raint harness ar	nd inertia reel:			
			(a)	Make sure the sh	houlder l	narness	control is in the	unlocked position.			
			(b)	Quickly pull one	shoulde	r strap ir	the forward dir	ection.			
				1) Make sure	the inert	ia reel lo	ocks.				
				2) Make sure	the strap	does n	ot extend.				
			(c)	Release the pres	ssure on	the stra	p.				
				1) Make sure	the inert	ia reel re	eleases the stra	p.			
			(d)	Quickly pull the c			•	rd direction.			
				1) Make sure							
			()	2) Make sure							
			(e)	Release the pres			•				
			(6)	,			eleases the stra	•	.9.1.		
			` '			•	•	seat back as far as pos	ssible.		
			(0)	Move the should				•	- 4:4:1		
			(h)	it is fully retracted	d.			ximately two inches at	a time, until		
								e sure the strap locks.			
			` '	Move the should				•			
						•		at back as far as possil	ole.		
			` '	Move the should				•			
			(I)	is fully retracted.		·		ximately two inches at			
				,	•		•	make sure the strap lo	ocks.		
			(m)	Put the shoulder	harness	control	in the unlocked	position.			
				_	—— E	ND OF	TASK ———				
				ALL		SOURCE MRB	CHECK CAPTA HARNESS REE	IN, FIRST OFFICER, & F LS	IRST OBSER	VER S	EΑ
							D633A109-AKS		1	Page 2	2 of
							25-030-00-01			eb 15/	





AIRLIN	E CARD NO	INSPECT	TITLE PASSENGER SE	AT BELTS	BOEING (25-040	
DATE	TASK INSPECTION - DETAILED				RELATE	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 4000 FH	REPEAT 4000 FH	APPLIC. AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS			ZONE 200 220 230 240	
Inspect (Deta	ailed) the passeng	er seat belts w/o	removal for wear	; condition, and	I security.	
	EFFECTIVITY AKS ALL	SOU MF		ASSENGER SEA	AT BELTS	
			D633A109- 25-040-00-0			Page 1 of Oct 15/201



	DATE TAIL NUMBER					STATION	AIRLINE CARD NO.	BOEING C 25-040		
								23-040	MECH	INSP
				200-801					IVILOIT	IIVOI
1.	Pas	senge	er Seat	Belt Check						
	A.	Pro	cedure							
				00-210-002						
		(1) Do a visual check to make sure the seat belts are not worn, frayed, or damaged.								
			SK 25-22-0		o curo the	ov are attached t	tightly to the cost			
		(2) Pull on the seat belts to make sure they are attached tightly to the seat.								
		(3)	Do these steps to make sure the seat belt operates correctly:							
		(a) Put the tongue end of the belt in the buckle.								
		NOTE: You can hear a click when the tongue is engaged in the buckle.								
			-	Lift the release handle						
			()			TASK ——	,			
					- END OF	IASK ———				
			EFFEC		SOURCE	INSPECT PASS	ENGER SEAT BELTS			
			AKS		MRB	2017/30				
						D633A109-AKS			Page 2	of 2
						25-040-00-01			oct 15/	





AIRLIN	IE CARD NO	PASSEN		TITLE MPARTMEN	T SEATS	1	CARD NO. 5-00-01
DATE	TASK VISUAL CHECK					RELATE	ED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1		ESHOLD 00 FH	REPEAT 12000 FH	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL					ALL	ALL
		ACCESS				ZONE 200	
Visually chec	ck the passenger c	ompartment sea	ats for pr	oper attach	nment.		
	EFFECTIVITY AKS ALL		DURCE PA	ASSENGER	COMPARTME	NT SEATS	
				633A109-A 5-045-00-01			Page 1 of 2 Jun 15/2015



	DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-045-00-01									
	TAS	SK 25-	22-00-2	200-804					MECH	INSP
1.	Trac	ck-mo	unted	Passenger Seat Che	ck and Ins	spection				
	A.	Prod	cedure							
			SK 25-22-0							
		(1)		sure the seat track fitt	ings are in	stalled correctly	<i>1</i> .			
			SK 25-22-0		o movo the	s coat forward a	nd aft. The seat assem	bly must not		
		(2)		at the seat track fitting		e seat lorward a	ind ait. The seat assem	bly must not	•	
			SK 25-22-0		:£ 41- a a a					
	(3) Remove the seat track covers if the seat moves.									
	SUBTASK 25-22-00-430-018 (4) Tighten the rear seat track fitting to the recommended torque.									
	SUBTASK 25-22-00-430-019									
		(5)		n the forward anti-ratt n front fitting to the rec			4 turn. If there is no ant	i-rattle,		
		SUBTA	SK 25-22-0	0-800-001						
		(6)		sure that the seat bott .70 in. (93.98 mm).	om cushic	n does not bloc	k the exit hatch opening	by more		
					- END OF	TASK ———				
			EFFECT AKS		SOURCE MRB	PASSENGER C	OMPARTMENT SEATS			<u> </u>
						D633A109-AKS	•		Page 2	0 6 7
						25-045-00-01	•	,	Jun 15/	





			IAON OANDO	•		
AIRLIN	E CARD NO	CHECK PAS	TITLE SSENGER SEAT B	REAK-OVER	BOEING C 25-050	
DATE	TASK OPERATIONAL		FEATURE		RELATE	D CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 12000 FH	REPEAT 12000 FH	APPLIC/	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS			ZONE 230 240	
mergency e	AIIS.					
	EFFECTIVITY AKS ALL		JRCE CHECK PA	SSENGER SEAT	BREAK-OVER FEA	TURE
			D633A109-			Page 1 - Jun 15/2



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
							25-050	T	
		SK 25-22-00-7						MECH	INSP
1.	Pas	senger Seat	Breakover Check at	Emergen	cy Exits				
	A.	Procedure							
		SUBTASK 25-22-0							
			sure the passenger se over into the pathways			aft of all the emergency	exits do not		
			•		•	ur airplanes should sho	1147		
		NOTE	requirements for the	e emergen	cy exit aisles.	ur airpianes snould sno	VV		
				- END OF	TASK ———				
		EFFECT AKS		SOURCE MRB	CHECK PASSE	NGER SEAT BREAK-OV	ER FEATURE	<u> </u>	
					D633A109-AKS 25-050-00-01			Page 2 Oct 15/	



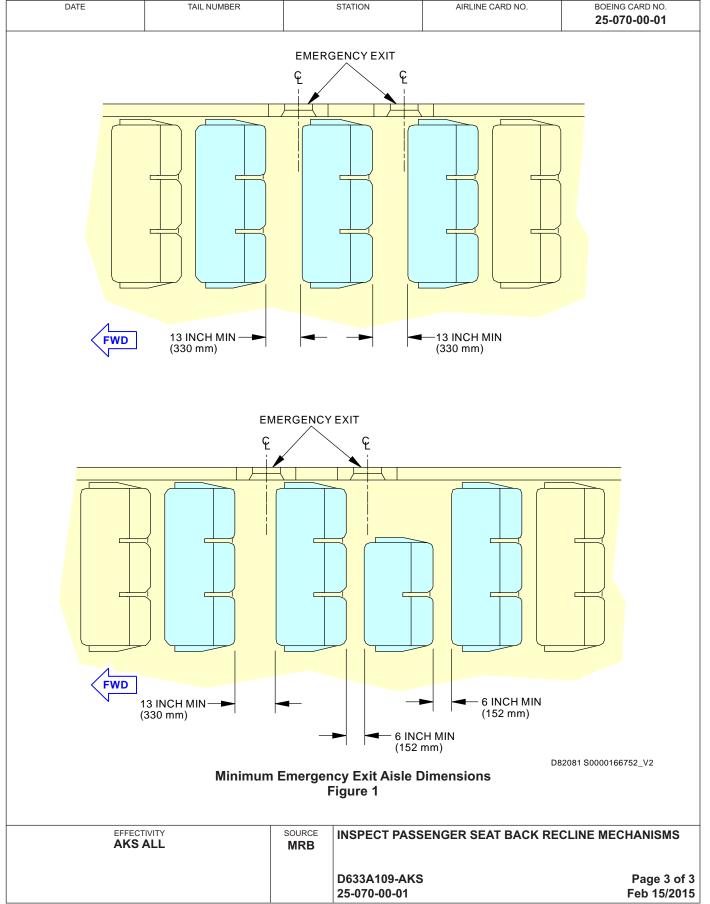


					BOEING CARD NO.		
AIRLIN	E CARD NO	INSPECT PAS	TITLE SENGER SEAT E	BACK RECLINE		CARD NO.)-00-01	
DATE	TASK FUNCTIONAL	_	MECHANISMS			ED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 6000 FH	REPEAT 6000 FH		ABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL	
		ACCESS			ZONE 220 230 240		
to emergency					n mechanisms on s		
	EFFECTIVITY AKS ALL	SOU MF	RCE INSPECT F	PASSENGER SEA	T BACK RECLINE	MECHANISMS	
		Wife	D633A109- 25-070-00-			Page 1 of 3 Oct 15/2014	



	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 25-070		
	TAS	K 25	22_00_7	│ 710-803				25-070	MECH	INSP
1.				Recline Restriction	Check at I	Emergency Exi	ts			
	Figu						<u></u>			
	A.	Pro	cedure							
			ASK 25-22-0							
		(1)		sure that the passengeline into the emergen			a of the emergency exi	it doors do		
					-		isle for a triple seat is 1	3 in.		
			(33 cm).						
				The minimum dimension 15.2 cm).	on of the e	mergency exit a	isle for a double seat is	6 in.		
					END OF	TASK ——				
			EFFECT		SOURCE	INSPECT PASS	ENGER SEAT BACK RE	CLINE MECH	│ IANISN	IS
			AKS	ALL	MRB					
						D633A109-AKS 25-070-00-01			Page 2 eb 15/	









AIRLIN	E CARD NO	INSPECT A	TITLE TTENDANT SEAT	BOEING CARD NO. 25-090-00-01				
DATE	INSPECTION - DETAILED				RELATED CARD			
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 3000 FH	REPEAT 3000 FH	APPLIC.			
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL		
		ACCESS			ZONE 221 241			
	EFFECTIVITY AKS ALL	soui MF		TTENDANT SEA	AT HARNESS			
			D633A109-A			Page 1 of Oct 15/201		



737-600/700/800/900 TASK CARDS

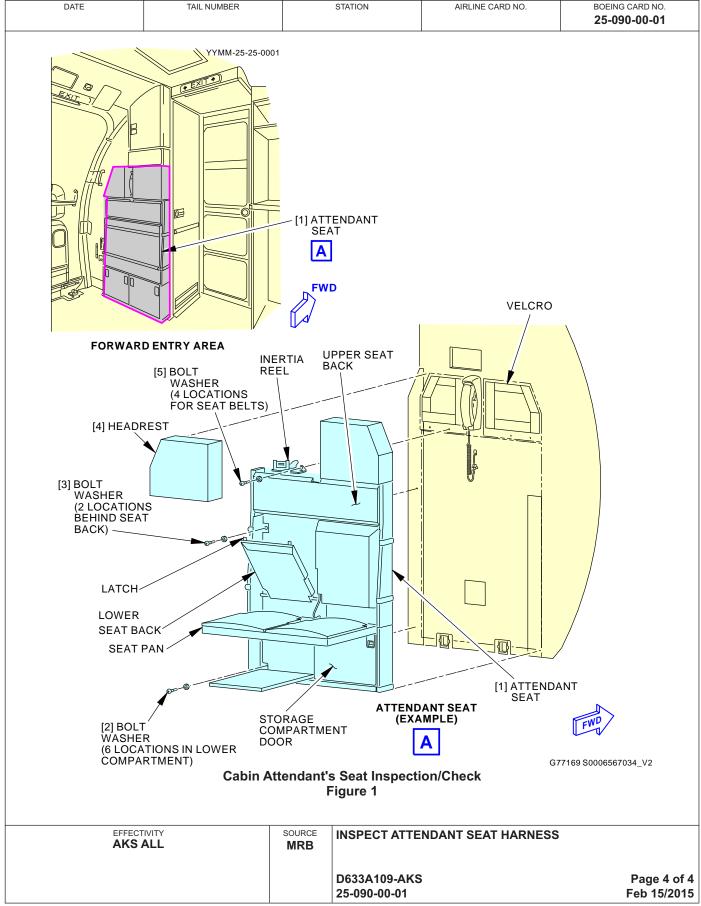
DATE TAIL NUMBER				TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-090					
TAS	SK 25-25-12-200-801												
	tendants' Seat and Harness Attachment Inspection												
	ire 1												
Α.	Prod	cedui	re										
,			5-12-210	-001									
	(1) Make sure the seat belts and the shoulder harness are not worn or damaged.												
	SUBTA	SK 25-2	5-12-210	-002			•						
	(2) Do these steps to examine the top mounting bolts [5] of the seat:												
		(a)		nove the headrest eners (Velcro).	[4]. To rem	ove it, pull on	it. It is attached with hoo	k and loop					
		(b)		e sure the mounti are not loose.	ng bolts [5]	that attach th	ne bracket for the inertia re	eels to the					
		(c)		e sure the mounti bracket are not lo	•	at attach the in	nertia reels of the shoulde	er harness to					
		(d)	Insta	all the headrest.									
	SUBTA	NSK 25-2	5-12-210	-003									
	(3) Do these steps to examine the middle mounting bolts of the seat:												
		(a)	Do t	hese steps to ope	n the lower	seat back:							
			1)	Open the seat p	an a small	distance.							
			2)	•	pull it awa	y from the upp	ower seat back and the uper seat back. It is attache						
			3)	Push the two lat back.	ches on the	e lower seat b	ack toward the center of t	he seat					
			4)	Pull the seat bac	ck out and o	down.							
		(b)	Mak	e sure the mounti	ng bolts [3]	that attach th	ne seat to the wall are not	loose.					
		(c)	Clos	se the lower seat b	oack.								
	SUBTA	SK 25-2	5-12-210	-004									
	(4)	Do t	hese	steps to examine	the lower n	nounting bolts	s of the seat:						
		(a)	Ope	n the storage con	npartment o	loor.							
		(b)	Mak	e sure the mounti	ng bolts [2]	that attach th	ne seat to the wall are not	loose.					
		(c)	Clos	se the storage con	npartment o	door.							
	SUBTA	SK 25-2	5-12-210	-005									
	(5)	Do t	hese	steps to examine	the lower a	ttachment po	ints for the seat belts:						
		(a)	Pull	the seat cushion	away from	the attachmer	nt points for the seat belts						
		(b)	Mak	e sure the bolts a	re tight.								
	SUBTA	SK 25-2	5-12-200	-001									
	(6)	Do t	hese	steps to inspect the	ne pivot poi	nts of the sea	it pan:						
		(a)	Ope	n the seat pan.									
			ECTIVITY S ALL		SOURCE MRB	INSPECT AT	TENDANT SEAT HARNESS	;					

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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 25-090-		
(b) F	Hold the seat pan near it	s back c	orner and shake	e the seat pan to make	sure it is not	MECH	INSP
	Make sure the seat pan	nas no o	bvious cracks.				
			TASK ——				
EFFECT AKS	ALL	SOURCE MRB	INSPECT ATTE	NDANT SEAT HARNESS		<u> </u>	
			D633A109-AKS		F	Page 3	of 4







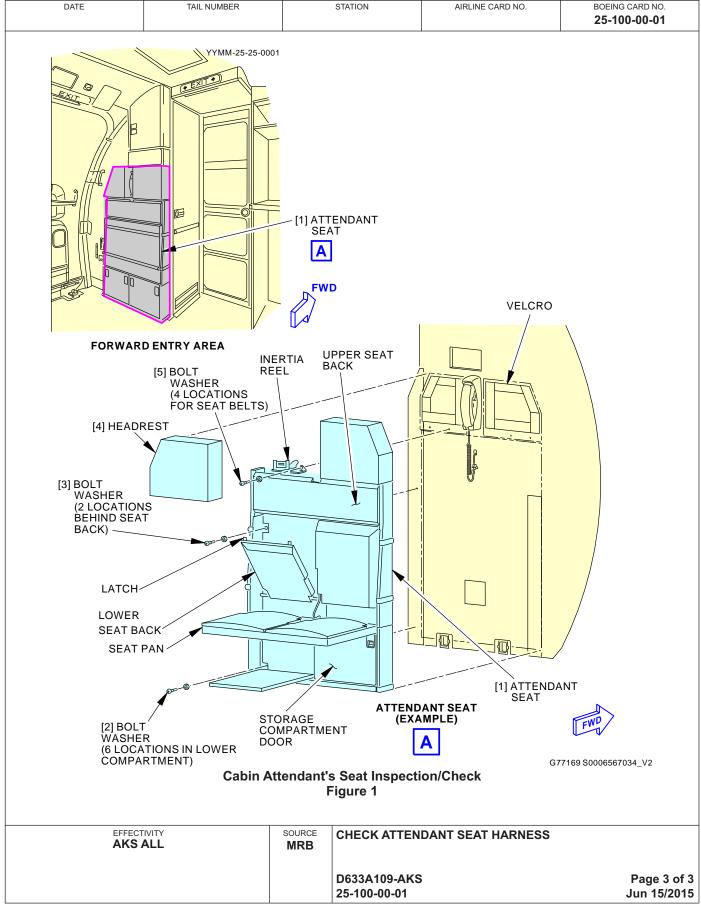


	E CARD NO	CHECK A	TITLE TTENDANT SEAT		CARD NO.)-00-01	
DATE	TASK OPERATIONAL				RELATE	ED CARD
AIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 3000 FH	REPEAT 3000 FH	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS			ZONE 221 241	
perationally	check the attenda	ant seat harness	inertia reel lock f	eature.		
	EFFECTIVITY AKS ALL	sot M	RCE CHECK ATT	ENDANT SEAT	HARNESS	



DATE TAIL NUMBER				TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-100-					
	TAS	K 25	-25-12-7	710-802					MECH	INSP			
1.	Iner	tia Re	eel, Atte	endants' Seat Harness	s, Operat	tional Check							
	(Figu	ure 1))										
	A.	Pro	cedure										
		SUBTA	ASK 25-25-1										
		(1)		ese steps to make sure									
			(a) Pull the shoulder harness out approximately half way.										
	(b) Pull suddenly on the shoulder harness.												
				 Make sure the sho the strap. 	ulder har	ness locks and	does not extend the ful	l length of					
			(c) F	Release the shoulder ha	arness.								
				1) Make sure the sho	ulder har	ness retracts.							
					END OF	TASK ——							
			EFFECT		SOURCE	CHECK ATTEN	DANT SEAT HARNESS						
			AKS	ALL	MRB								
						D633A109-AKS	;		Page 2				









AIRLIN	E CARD NO		TITLE ASTE COMPARTM	BOEING CARD NO. 25-130-00-01 RELATED CARD		
DATE	INSPECTION - DETAILED	DOOR AND	SPRING AND AC			
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 4000 FH	REPEAT 4000 FH	APPLIC/	ABILITY ENGINE
STATION	SKILL AIRPL	NOTE			ALL	ALL
		ACCESS			ZONE 221 241 242	

Inspect (Detailed) the lavatory waste compartment flapper door and the waste compartment access door latching mechanism for wear, condition, and security.

INTERVAL NOTE: FAA AD 74-08-09 interval is 1000 FH.



737-600/700/800/900 TASK CARDS

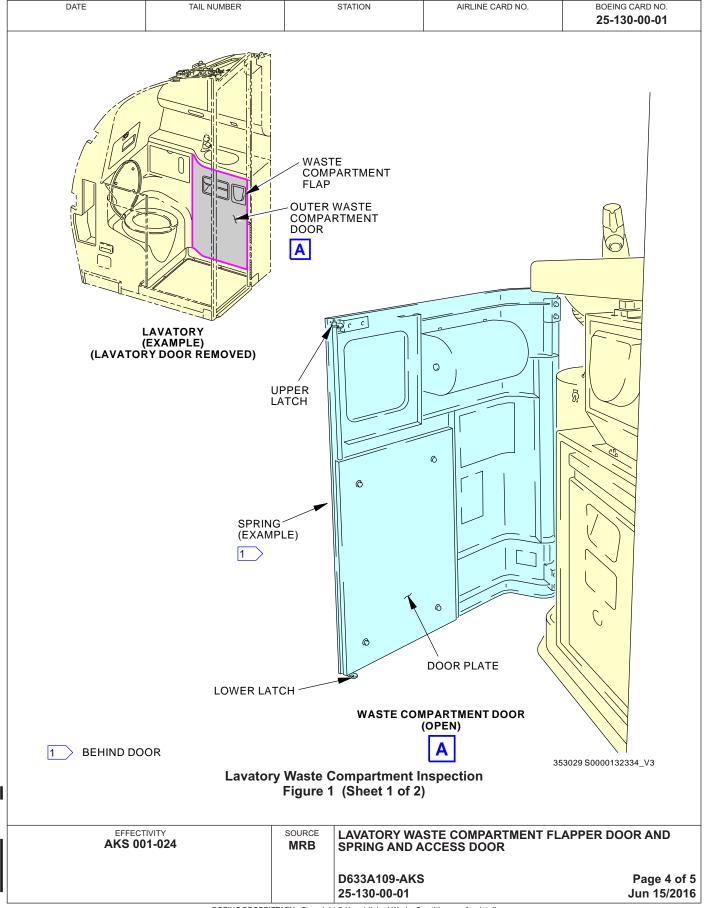
						IAS	K CARDS				
	Г	OATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 25-130- (
1.	Lava	K 25- atory ure 1)			801 npartment Inspe	ection			-	MECH	INS
	AKS	001-0)24								
		Was	te Co sk 25-40 Do th	Pull the Push 1) Do a cond Close	steps to make surthe latch at the both around the edge Make sure the downer that when the visual check of the door. The the door. The the door in the door in the door in the door in the door the lock of the door in the door in the door in the door in the door the lock of the door in the door in the door in the door in the door the d	re the door of the cor plate me springs be door is clear the door plate the upper ower latch is in easily se	closes tightly: e door to open to prepare on the interpretate on the door posed. Ite on the inner latches are full and lower latch	nner side of the door. ack to its initial position. plate. These springs cor side of the door for gene ly engaged: n area.	ral		
	AKS	025-9	99		0 0						
	B.	Was	te Co	mpar	tment Door Insp	ection Pro	ocedure				
		SUBTAS (1)		nese s	steps for lavatorie			npartment door and an ou ent doors close correctly			
			(a)	Oper	n the outer waste	compartme	ent door.				
				NOT	E: The latch is a	t the botton	n of the door.				
			(b)	Oper	n the inner waste	compartme	ent door and re	lease it.			
				1)	Make sure that t	he springs	close the inner	door without your help.			
			(c)		e sure the inner d close tightly to m			such that it does not closfire.	se fully. It		
			(d)	Close	e the outer waste	compartm	ent door.				
				1)	Make sure the la	tch holds t	he outer door o	closed.			
		AKS	ALL								
	C.	Was	te Co	mpar	tment Flap Insp	ection Pro	cedure				
		SUBTA	SK 25-40	- -08-210-(002						
		(1)	Do th	nese s	steps to make sur	e the wast	e compartment	flap operates correctly:			
			(a)	Push	the waste comp	artment fla	p open.	•			
				CTIVITY S ALL		SOURCE MRB		ASTE COMPARTMENT FLA	APPER DOOR	AND	

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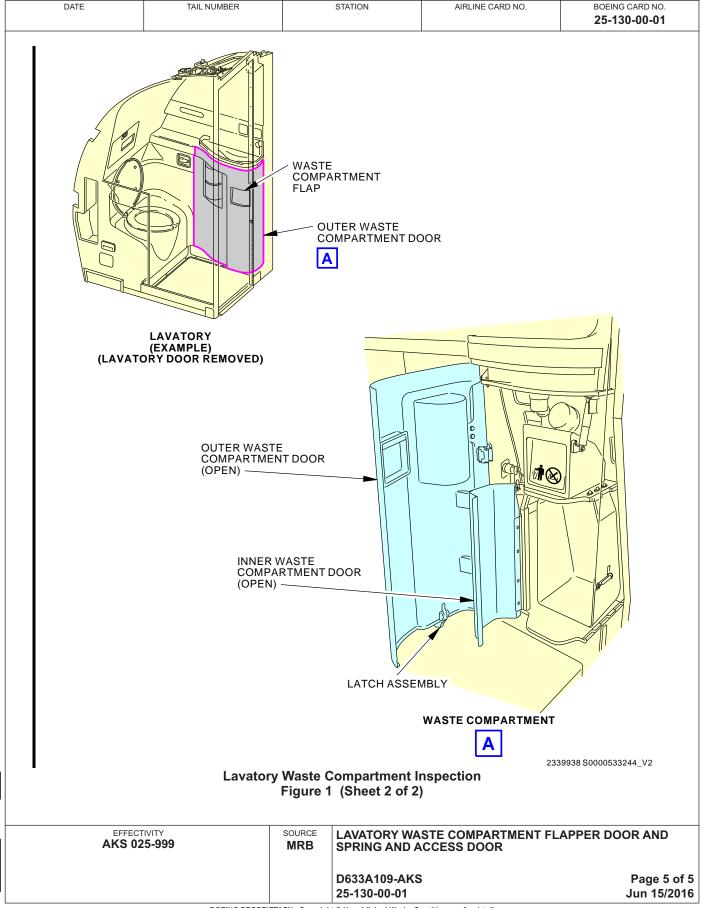


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ 25-130-		
/L\	Pologo the wests serve	nortma:	flon and make -	uro it aloogo feelle	20-100-	MECH	INSP
(c)	Release the waste com				close fully It		
(6)	must close tightly to ma	ke sure it	can contain a fi	re.	GOSE TUITY. IL		
			TASK ——				
effe AK S	S ALL	SOURCE MRB	LAVATORY WAS SPRING AND AG	STE COMPARTMENT FL CCESS DOOR	APPER DOOF	RAND	
			D633A109-AKS 25-130-00-01		F	Page 3 un 15/	of 5













AIRLINI	E CARD NO	INSPECT F	INSPECT FWD CARGO COMPARTMENT PANELS/LINERS			CARD NO. 1-00-01
DATE	TASK INSPECTION - GEN VISUAL		PANELS/LINERS		RELATE	D CARD
TAIL NUMBER	WORK AREA FWD CARGO	VERSION THRESHOLD REPEAT 1.1 450 FC 450 FC			APPLICA	
STATION	SKILL AIRPL				AIRPLANE ALL	ALL ALL
		ACCESS 821			ZONE 121 122	

Inspect (General Visual) the fwd cargo compartment floor, ceiling, sidewall, bulkhead, and blowout (pressure relief) panels/liners for holes/tears, condition, and security.

A. References

 110101011000						
Reference	Title					
AMM 21-32-05 P/B 401		COMPARTMENT BLOWOUT PANELS - AL/INSTALLATION				
AMM 25-00-00-300-801	Crushed	Core and Sandwich Panel Repair (P/B 80	11)			
AMM 25-52-01-300-802		nt Repair for Molded Liners, Shrouds, and Curves (P/B 801)	Liners with			
AMM 25-52-01-300-806	Neopren	e Coated Cargo Liner Repair (P/B 801)				
AMM 25-52-01-300-807	Silicone	mpregnated Liner Repair (P/B 801)				
AMM 25-52-01-300-808	Permane	nt Repair for the Contour of the Ceiling Sh	roud (P/B 801)			
AMM 25-52-01-340-801	Permane	nt Repair for Flat or Slightly Curved Liners	s (P/B 801)			
AMM 25-52-01-350-801	Loose or	Damaged Tape Repair (P/B 801)				
AMM 25-52-06-000-801	Cargo Co	ompartment Sidewall Lining - Removal (P/	B 401)			
AMM 25-52-06-400-801	Cargo Co	ompartment Sidewall Lining - Installation (P/B 401)			
AMM 25-52-09-000-801	Cargo Co	ompartment Ceiling Liner - Removal (P/B	401)			
AMM 25-52-09-400-801	Cargo Co	ompartment Ceiling Liner - Installation (P/E	3 401)			
AMM 25-52-16-000-801	Forward Cargo Compartment Forward Bulkhead Liner - Removal (P/B 401)					
AMM 25-52-16-400-801	Forward (P/B 401	Cargo Compartment Forward Bulkhead Li)	ner - Installation			
AMM 25-52-17-000-801	Forward	Cargo Compartment Aft Bulkhead Liner -	Removal (P/B 401)			
AMM 25-52-17-400-801	Forward (P/B 401	Cargo Compartment Aft Bulkhead Liner -	Installation			
AMM 25-52-18-000-801	Aft Cargo	Compartment Forward Bulkhead Liner -	Removal (P/B 401)			
AMM 25-52-18-400-801	Aft Cargo (P/B 401	Compartment Forward Bulkhead Liner -	Installation			
AMM 25-52-19-000-801	Aft Cargo	Compartment Aft Bulkhead Liner - Remo	val (P/B 401)			
AMM 25-52-19-400-801	Aft Cargo	Compartment Aft Bulkhead Liner - Install	ation (P/B 401)			
EFFECTIVITY	SOURCE	INSPECT FWD CARGO COMPARTMENT P	ANELS/LINERS			
AKS ALL	MRB					
		D633A109-AKS 25-160-00-01	Page 1 of 3 Oct 15/2014			



	С	ATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO 25-160-00-0				
	TAS	SK 25-52-00-200-803									
١.	Car	go Compartment Liner Inspection									
	A.	Gen	eral								
		(1)	This	task inspects all the lin	ers in the forward and	aft cargo compartments.					
		(2) It is important that all of the fasteners, seams, and punctures are sealed correctly to keep the amount of oxygen in the compartment to a minimum.3. Procedure									
	В.	Pro	cedui	re							
		SUBTA	ASK 25-5	2-00-210-001							
		(1) Make sure that there is no loose or damaged tape at the joints for the ceiling liner, sidewall liner or bulkhead liner. If there is loose or damaged tape, do this task: Loose or Damaged Tape Repair, AMM TASK 25-52-01-350-801.									
		SUBTA	ASK 25-5	2-00-211-001							
		(2) Make sure that the ceiling linings, sidewall linings, bulkhead linings and panels, blowout panels and floor panels do not have any cuts, tears, cracks, or holes.									
		(a) If there are cuts, tears, cracks or holes found on the blowout panel, replace the blowout panel (AMM PAGEBLOCK 21-32-05/401).									
		SUBTA	SUBTASK 25-52-00-200-002								
			These are the maximum damage limits for the repair of the non-honeycomb linings. These damage limits apply to all the non-honeycomb linings (the ceiling panels and shrouds, the sidewall linings, and the forward bulkhead in the aft cargo compartment) EXCEPT for cracks along the contour of the ceiling shroud in the forward cargo compartment. The cracks length limits may be longer than 35 in. (88.9 cm) and are at least 2 in. (5.1 cm) away from other damage and from the panel edge on all sides, do this task: Permanent Repair for the Contour of the Ceiling Shroud, AMM TASK 25-52-01-300-808.								
			(a)	Make sure that cuts th	at are 36 in. (91 cm) lo	ong or less.					
			(b)	Make sure that L-shap	oed tears are 9 in. (23	cm) long or less on either s	ide.				
			(c)	Make sure that holes	are 1.5 in. (3.8 cm) in o	diameter or smaller.					
		SUBTA	ASK 25-5	2-00-200-003							
		(4)	If the	e damage is less than o	r equal to the limits, do	o the applicable task to repa	air the lining:				
		 (a) Permanent Repair for Flat or Slightly Curved Liners, AMM TASK 25-52-01-340-801 (b) Permanent Repair for Molded Liners, Shrouds, and Liners with Complex Curves, AMM TASK 25-52-01-300-802 									
			(c)	Neoprene Coated Car	go Liner Repair, AMM	TASK 25-52-01-300-806					
			(d)	Silicone Impregnated	Liner Repair, AMM TAS	SK 25-52-01-300-807					
		SUBTASK 25-52-00-200-004 (5) If the damage is greater than the limits, you cannot repair the lining. You must replace the lining. Do the applicable tasks:									
				g. Do trie applicable tas	No.			J			



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-160-			
	(b)	Cargo Compartment Si Cargo Compartment Si		-			MECH	INSP	
	(c)	Forward Cargo Compa TASK 25-52-16-000-80 - Installation, AMM TAS	1 and For	ward Cargo Co					
	(d)	TASK 25-52-17-000-80	Forward Cargo Compartment Aft Bulkhead Liner - Removal, AMM TASK 25-52-17-000-801 and Forward Cargo Compartment Aft Bulkhead Liner - Installation, AMM TASK 25-52-17-400-801						
	(e)	Aft Cargo Compartmen TASK 25-52-18-000-80 Installation, AMM TASK	1 and Aft	Cargo Compart		d Liner -			
	(f)	Aft Cargo Compartmen TASK 25-52-19-000-80 Installation, AMM TASK	1 and Aft	Cargo Compart		er -			
SUBTA	ASK 25-52	-00-300-001							
(6)	forwa benc	se are the maximum dar ard cargo compartment. I of the shroud. For dam ir instructions for non-ho	These da	mage limits only flat parts of the	y apply to cracks or cut	s along the			
	(a)	Make sure that cracks the panel edge on all s		st 2 in. (5 cm) av	way from other damage	and from			
SUBTA	ASK 25-52	-00-300-002							
(7)		damage to the contour than or equal to the limi		-		artment is			
	(a)	Permanent Repair for t TASK 25-52-01-300-80		ur of the Ceiling	Shroud, AMM				
SUBTA	ASK 25-52	-00-960-003							
(8)		damage is greater thar g. Do these tasks:	the limits	s, you cannot re _l	pair the lining. You mus	st replace the			
	(a)	Cargo Compartment	_						
SUBTA	ASK 25-52	-00-200-005							
(9)	Thes	se are the maximum dar	nage limits	s for the repair o	of the nomex honeycom	nb panels.			
	(a)	For panels with core da core around the damage	•	ere must be at le	east 0.5 in. (1.3 cm) of	undamaged			
SUBTA	ASK 25-52	-00-300-003							
(10)	If the	damage is less than or	equal to t	he limits, do this	s task to repair the linin	ıg:			
	(a)	Crushed-Core and San	ndwich Pai	nel Repair, AMM	1 TASK 25-00-00-300-8	301			
SUBTA	ASK 25-52	2-00-960-004							
(11)	If the	e damage is greater thar g.	the limits	, you cannot re _l	pair the lining. You mus	st replace the			
			END OF	TASK ——					
		CTIVITY S ALL	SOURCE MRB	INSPECT FWD	CARGO COMPARTMEN	T PANELS/LIN	IERS		
				D633A109-AKS 25-160-00-01			Page 3 eb 15/2		
			TA DV 0 11						





AIRLINI	E CARD NO	INSPECT	INSPECT AFT CARGO COMPARTMENT PANELS/LINERS			CARD NO. 1-00-02
DATE	TASK INSPECTION - GEN VISUAL		PANELS/LINERS		RELATE	D CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION THRESHOLD REPEAT 1.1 450 FC 450 FC			APPLICA	
STATION	SKILL AIRPL				AIRPLANE ALL	ALL ALL
		ACCESS 822			ZONE 141 142	

Inspect (General Visual) the aft cargo compartment floor, ceiling, sidewall, bulkhead, and blowout (pressure relief) panels/liners for holes/tears, condition, and security.

A. References

Reference	Title
AMM 21-32-05 P/B 401	CARGO COMPARTMENT BLOWOUT PANELS -
	REMOVAL/INSTALLATION
AMM 25-00-00-300-801	Crushed-Core and Sandwich Panel Repair (P/B 801)
AMM 25-52-01-300-802	Permanent Repair for Molded Liners, Shrouds, and Liners with Complex Curves (P/B 801)
AMM 25-52-01-300-806	Neoprene Coated Cargo Liner Repair (P/B 801)
AMM 25-52-01-300-807	Silicone Impregnated Liner Repair (P/B 801)
AMM 25-52-01-300-808	Permanent Repair for the Contour of the Ceiling Shroud (P/B 801)
AMM 25-52-01-340-801	Permanent Repair for Flat or Slightly Curved Liners (P/B 801)
AMM 25-52-01-350-801	Loose or Damaged Tape Repair (P/B 801)
AMM 25-52-06-000-801	Cargo Compartment Sidewall Lining - Removal (P/B 401)
AMM 25-52-06-400-801	Cargo Compartment Sidewall Lining - Installation (P/B 401)
AMM 25-52-09-000-801	Cargo Compartment Ceiling Liner - Removal (P/B 401)
AMM 25-52-09-400-801	Cargo Compartment Ceiling Liner - Installation (P/B 401)
AMM 25-52-16-000-801	Forward Cargo Compartment Forward Bulkhead Liner - Removal (P/B 401)
AMM 25-52-16-400-801	Forward Cargo Compartment Forward Bulkhead Liner - Installation (P/B 401)
AMM 25-52-17-000-801	Forward Cargo Compartment Aft Bulkhead Liner - Removal (P/B 401)
AMM 25-52-17-400-801	Forward Cargo Compartment Aft Bulkhead Liner - Installation (P/B 401)
AMM 25-52-18-000-801	Aft Cargo Compartment Forward Bulkhead Liner - Removal (P/B 401)
AMM 25-52-18-400-801	Aft Cargo Compartment Forward Bulkhead Liner - Installation (P/B 401)
AMM 25-52-19-000-801	Aft Cargo Compartment Aft Bulkhead Liner - Removal (P/B 401)
AMM 25-52-19-400-801	Aft Cargo Compartment Aft Bulkhead Liner - Installation (P/B 401)
EFFECTIVITY AKS ALL	SOURCE INSPECT AFT CARGO COMPARTMENT PANELS/LINERS MRB

D633A109-AKS

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[DATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CA 25-160-				
TAS	K 25	-52-00	0-200-803			•	MECH	ı		
Car	go Co	ompa	rtment Liner Inspection	<u>n</u>						
A.	Gen	eral								
	` '		task inspects all the line	ers in the forward and a	ft cargo compartments.					
	(2)									
В.	Pro	cedui	re							
	SUBTA	ASK 25-5	2-00-210-001							
	(1)	side		er. If there is loose or d	it the joints for the ceiling amaged tape, do this task 01.					
	SUBTA	ASK 25-5	2-00-211-001							
	(2)		te sure that the ceiling ling ling ling ling ling ling ling	-	ulkhead linings and pane , cracks, or holes.	ls, blowout				
	(a) If there are cuts, tears, cracks or holes found on the blowout panel, replace the blowout panel (AMM PAGEBLOCK 21-32-05/401).									
	SUBTA	SUBTASK 25-52-00-200-002								
		shro EXC com leas task	ouds, the sidewall linings CEPT for cracks along th partment. The cracks le	s, and the forward bulkhor e contour of the ceiling ngth limits may be longe m other damage and fro	b linings (the ceiling pane ead in the aft cargo comp shroud in the forward car er than 35 in. (88.9 cm) a om the panel edge on all s ng Shroud, AMM	artment) go nd are at				
		(a)	Make sure that cuts that	at are 36 in. (91 cm) lon	g or less.					
		(b)	Make sure that L-shap	ed tears are 9 in. (23 cn	n) long or less on either s	ide.				
		(c)	Make sure that holes a	are 1.5 in. (3.8 cm) in dia	ameter or smaller.					
	SUBTA	ASK 25-5	2-00-200-003							
	(4)	If the	e damage is less than or	r equal to the limits, do t	the applicable task to repa	air the lining:				
		(a)	Permanent Repair for I	Flat or Slightly Curved L	iners, AMM TASK 25-52-	01-340-801				
	(b) Permanent Repair for Molded Liners, Shrouds, and Liners with Complex Curves, AMM TASK 25-52-01-300-802					x Curves,				
		(c)	Neoprene Coated Carg	go Liner Repair, AMM T	ASK 25-52-01-300-806					
		(d)	Silicone Impregnated L	iner Repair, AMM TASŁ	C 25-52-01-300-807					
	SUBTASK 25-52-00-200-004									
	(5)		e damage is greater than g. Do the applicable tasl	-	repair the lining. You mus	t replace the				
		(a)	Cargo Compartment C	eiling Liner - Removal,	AMM TASK 25-52-09-000)-801 and				



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-160-			
	(b)	Cargo Compartment Si Cargo Compartment Si		-			MECH I	INSP	
	(c)	Forward Cargo Compa TASK 25-52-16-000-80 - Installation, AMM TAS	1 and For	ward Cargo Co					
	(d)	TASK 25-52-17-000-80	Forward Cargo Compartment Aft Bulkhead Liner - Removal, AMM TASK 25-52-17-000-801 and Forward Cargo Compartment Aft Bulkhead Liner - Installation, AMM TASK 25-52-17-400-801						
	(e)	Aft Cargo Compartment Forward Bulkhead Liner - Removal, AMM TASK 25-52-18-000-801 and Aft Cargo Compartment Forward Bulkhead Liner - Installation, AMM TASK 25-52-18-400-801							
	(f)	Aft Cargo Compartmen TASK 25-52-19-000-80 Installation, AMM TASK	1 and Aft	Cargo Compart		er -			
SUBTA	ASK 25-52	2-00-300-001							
(6)	forwa benc	se are the maximum dar ard cargo compartment. I of the shroud. For dam ir instructions for non-ho	These da	mage limits only flat parts of the	y apply to cracks or cut	s along the			
	(a)	Make sure that cracks the panel edge on all s		st 2 in. (5 cm) av	way from other damage	and from			
SUBTA	ASK 25-52	2-00-300-002							
(7)		e damage to the contour than or equal to the limi		-		artment is			
	(a)	Permanent Repair for t TASK 25-52-01-300-80		ır of the Ceiling	Shroud, AMM				
SUBTA	ASK 25-52	2-00-960-003							
(8)		e damage is greater thar g. Do these tasks:	the limits	s, you cannot re	pair the lining. You mus	t replace the			
	(a)	Cargo Compartment	_						
SUBTA	ASK 25-52	2-00-200-005							
(9)	Thes	se are the maximum dar	nage limits	s for the repair o	of the nomex honeycom	nb panels.			
	(a)	For panels with core da core around the damage	•	ere must be at le	east 0.5 in. (1.3 cm) of	undamaged			
SUBTA	ASK 25-52	2-00-300-003							
(10)	If the	damage is less than or	equal to t	he limits, do this	s task to repair the linin	g:			
	(a)	Crushed-Core and San	ndwich Pai	nel Repair, AMM	1 TASK 25-00-00-300-8	301			
SUBTA	ASK 25-52	2-00-960-004							
(11)	If the	e damage is greater thar g.	n the limits	s, you cannot re _l	pair the lining. You mus	st replace the			
			END OF	TASK ——					
		CTIVITY S ALL	SOURCE MRB	INSPECT AFT (CARGO COMPARTMENT	PANELS/LIN	ERS		
				D633A109-AKS 25-160-00-02			Page 3 c eb 15/20		
		DOENIO DECEDIO							





	CARD NO	INSPECT FORM	TITLE WARD CARGO DO SYSTEM	OR RESTRAINT	BOEING (25-170	-00-01
DATE	TASK INSPECTION - GEN VISUAL					D CARD
AIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1	THRESHOLD 3000 FC	REPEAT 3000 FC	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 821			ZONE 121 122	
	EFFECTIVITY AKS ALL	SOUMI	RCE INSPECT FO	ORWARD CARGO	D DOOR RESTRAIN	NT SYSTEM



	[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-170 -		
	TAC	V 25	F2 00 :	⊥ 210-801				20 170	месн	INSP
1.				t (Doorway) - Visual I	nspection	1				
	A .		edure			<u> </u>				
	Α.			00-210-005						
		(1)		lly examine the cargo	restraint at	the forward and	d aft cargo door. Examii	ne the items		
				Make sure the straps a	are not wo	rn or frayed.				
				Make sure that each s			point.			
					- END OF	TASK ——				
			EFFEC AKS	TIVITY ALL	SOURCE MRB	INSPECT FORV	VARD CARGO DOOR RE	STRAINT SY	STEM	
						D633A109-AKS		1	Page 2	2 of 2
						25-170-00-01		F	eb 15/	2015





AIRLINI	E CARD NO	INSPECT AFT CA	TITLE	BOEING 0 25-170		
DATE	TASK INSPECTION - GEN VISUAL	INSPECTATION	NOO DOOK KES	TIMINI SISIEM	RELATE	
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1	THRESHOLD 3000 FC	REPEAT 3000 FC	APPLIC/	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 822			ZONE 141 142	
	EFFECTIVITY AKS ALL	SOUR MR	CE INSPECT A	FT CARGO DOOR	RESTRAINT SYS	TEM
			D633A109-			Page 1 of 2 Oct 15/2014



		[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
1. Cargo Restraint (Doorway) - Visual Inspection A. Procedure SUBTINES 25-26-26-26-26 (1) Visually examine the cargo restraint at the forward and aft cargo door. Examine the items that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point. ———————————————————————————————————									25-170		
A. Procedure SUBTASK 252-20-210-095 (1) Visually examine the cargo restraint at the forward and aft cargo door. Examine the items that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point. ———————————————————————————————————		TAS	K 25-	52-00-	210-801					MECH	INSP
SUBTIANK 25-92-00-210-005 (1) Visually examine the cargo restraint at the forward and aft cargo door. Examine the items that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point. END OF TASK END OF TASK INSPECT AFT CARGO DOOR RESTRAINT SYSTEM MRB D633A109-AKS Page 2 of 2	1.	Car	go Re	strain	t (Doorway) - Visual I	nspection	<u>1</u>				
SUBTIANK 25-92-00-210-005 (1) Visually examine the cargo restraint at the forward and aft cargo door. Examine the items that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point. END OF TASK END OF TASK INSPECT AFT CARGO DOOR RESTRAINT SYSTEM MRB D633A109-AKS Page 2 of 2		Α.	Prod	edure							
that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point. ———————————————————————————————————											
(a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point. ———————————————————————————————————			(1)	Visua	lly examine the cargo	restraint a	t the forward an	d aft cargo door. Examii	ne the items		
(b) Make sure that each strap attaches to an anchor point. ——END OF TASK—— END											
ESPECITATY AKS ALL SOURCE MRB INSPECT AFT CARGO DOOR RESTRAINT SYSTEM D633A109-AKS Page 2 of 2				(a)	Make sure the straps a	are not wo	rn or frayed.				
EFFECTIVITY AKS ALL SOURCE MRB INSPECT AFT CARGO DOOR RESTRAINT SYSTEM D633A109-AKS Page 2 of 2				(b)	Make sure that each s	trap attach	nes to an anchoi	point.			
AKS ALL MRB D633A109-AKS Page 2 of 2						- END OF	TASK ———				
AKS ALL MRB D633A109-AKS Page 2 of 2											
AKS ALL MRB D633A109-AKS Page 2 of 2											
AKS ALL MRB D633A109-AKS Page 2 of 2											
AKS ALL MRB D633A109-AKS Page 2 of 2											
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AKS ALL MRB D633A109-AKS Page 2 of 2				FFFF	TIVITY	SOURCE	INIODEOT 1 == 1	ADOO DOOD 5-0	NT OVOTES:		
							INSPECT AFT (SARGO DOOR RESTRAI	NI SYSTEM		
							D633A109-AKS 25-170-00-02			Page 2 Jun 15/	





AIRLINE CARD NO		LEFT EMERGENCY EXIT HATCH ESCAPE STRAPS			BOEING CARD NO. 25-190-01-01		
DATE	TASK INSPECTION - DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 10 YR	REPEAT 10 YR	APPLIC,	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 833			ZONE 231		

Inspect (Detailed) the left emergency exit hatch escape strap for condition and security.

A. Consumable Materials

Reference	Description	Specification
D50080	Lubricant - Solid Film	MIL-PRF-46010 Type 1
		(Supersedes MIL-L-8937)
G51031	Abrasive - 120 Grit, Aluminum Oxide, Sandpaper	
[C10-105]	Roll	

B. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-130	Brush - Soft-bristle Stainless Steel Wire

EFFECTIVITY AKS ALL			;
		D633A109-AKS 25-190-01-01	Page 1 of 7 Oct 15/2015



	DATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CAI 25-190- 0				
T/	ASK 2	5-61-1	0-710-801				MECH	INS		
			xit Door Life Line Chec	k						
	igure			=						
A		neral								
	(1) The life line (also called the escape strap) is kept in a stowage tube installed behind the sidewall and the ceiling panels of the passenger compartment. One end of the life line is attached to the doorway structure of the emergency exit door. In an emergency, the hook end of the life line is attached to the fitting on the wing. The life line is used to help the passengers when they are on the wing.									
	(2)	(2) The Left Emergency Exit Door (STA 589.5), a Right Emergency Exit Door (STA 589.5), a Left Emergency Exit Door (STA 627.5), and a Right Emergency Exit Door (STA 627.5). The stowage tube that contains the life line is located over the Left Emergency Exit Door (627.5) and the Right Emergency Exit Door (STA 627.5).								
В	. Pr	ocedu	re							
	SUE	TASK 25-	61-10-010-008							
	W	ARNIN		OOR OPENING PATH IS E. THE DOOR IS SPRIN ND INJURIES COULD C	IG LOADED TO OPEN	RELEASE				
(1) Open the emergency exit door.										
	SUE	TASK 25-	61-10-010-006							
	(2)	Оре	en the cloth cover that ho	lds the hook and life line	; [21].					
	SUE	TASK 25-	61-10-020-009							
	(3)	Pull	the hook end of the life I	ine [21] out over the win	g.					
		(a)	Make sure the life line	[21] can be attached to t	he fitting on the wing.					
	SUE		61-10-210-008							
	(4)	Do	these steps to examine the		ne [21]:					
		(a)	Make sure the life line	[21] has no worn areas.						
		(b)	Make sure the life line	[21] is not wet.						
(c) Make sure the anchor on the end of the life line [21] is not loose, corroded, or cracked.										
			61-10-210-009							
	(5)		these steps to examine the		•					
		(a)		k assembly operates co	•					
				r snap hook on the hook						
			, , , ,	close the spring keeper s	snap hook.					
		(b)	If the hook is bent, repl							
				ded that all new emerge r wear resistance.	ency hook assemblies are	e cadmium				
		(c)	Make sure that there is	no corrosion on the hoc	ok assembly.					
			1) If corrosion preve	nts the operation of the I	hook assembly, replace t	he life line.				



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 25-190		
	a) Make sure th	at there is	s no corrosion c	on the spring.		MECH	INSF
		spring is nee the life		eeper snap hook is defe	ctive,		
	b) Remove corr			er snap hook or the hoo	k with		
	2) If there is minor co	rrosion, r	emove it.				
	a) Use a soft-br G51031 [C10		less steel wire l	brush, STD-130 or sand	paper,		
(d) A	Add protection to the ho	ok assem	nbly with dry film	n lubricant.			
	Apply solid film lub the hook assembly		50080 or equiva	llent heat cured dry film	lubricant to		
	NOTE: This lubric limit metal			arts, corrosion, prevent	galling, and		
			almost the sam	ne as corrosion. If possiblor.	ole, use an		
subtask 25-61-1 (6) Fold tl	₀₋₄₂₀₋₀₁₃ he life line [21] as show	n (Fiaure	1).				
SUBTASK 25-61-1		(ga. a	.,.				
(7) Put the	e hook into the retainer	and close	e the cloth cove	r.			
SUBTASK 25-61-1							
(8) Close	the emergency exit do	or.					
		END OF	TASK ———				
EFFECT AKS		SOURCE MRB	LEFT EMERGE	NCY EXIT HATCH ESCAF	PE STRAPS		
			D633A109-AKS 25-190-01-01			Page 3 eb 15/	
			© Unnublished Work - Se		-		



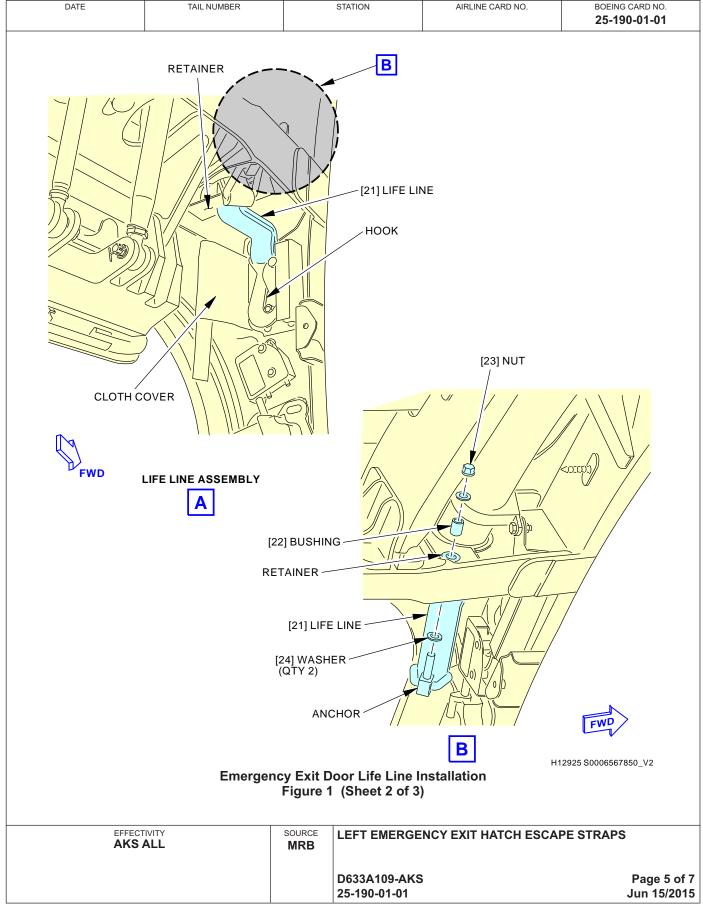
737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-190-01-01 LIFE LINE STOWAGE TUBE LIFE LINE **ASSEMBLY** Α EMERGENCY EXIT DOOR (DOOR SHOWN IN OPEN POSITION) (DOOR LINER REMOVED) (LEFT SIDE IS SHOWN, RIGHT SIDE IS OPPOSITE) F82116 S0006567849_V2 **Emergency Exit Door Life Line Installation** Figure 1 (Sheet 1 of 3) EFFECTIVITY SOURCE **LEFT EMERGENCY EXIT HATCH ESCAPE STRAPS AKS ALL MRB** D633A109-AKS Page 4 of 7

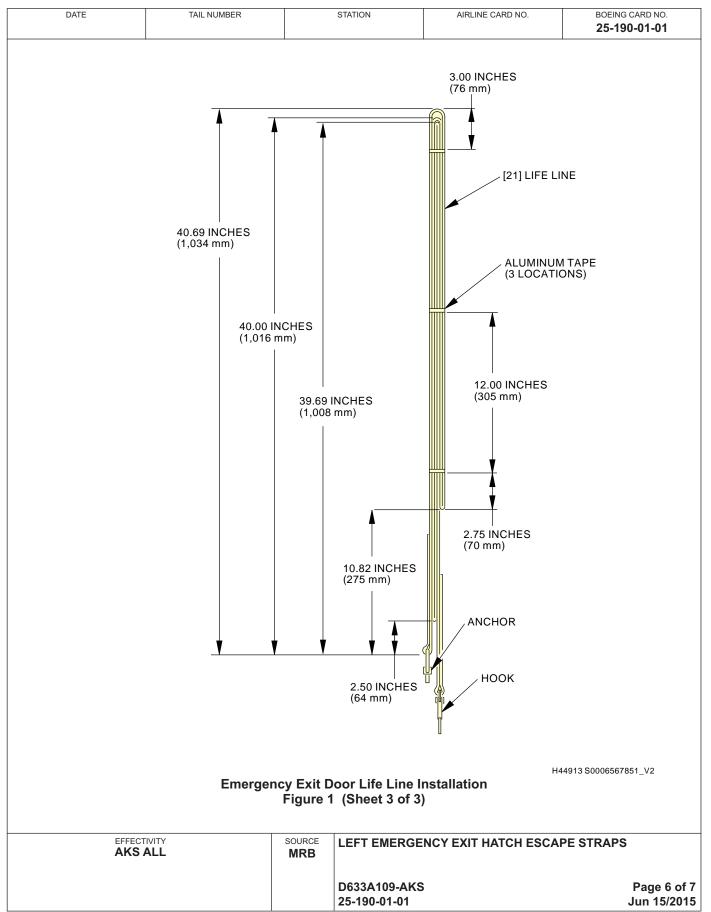
25-190-01-01

Jun 15/2015











DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-190-01-01
	RIVET		RIVET KEEPER SNAP HOOK RING KEEPER AP HOOK	HOOK	
	Emerge	ncy Exit [Door Life Line I Figure 2	nspection	2580 S0000519070_V2
EFFEC AKS	TIVITY	SOURCE MRB	LEFT EMERGE	NCY EXIT HATCH ESCAP	PE STRAPS
			D633A109-AKS 25-190-01-01		Page 7 of 7 Jun 15/2015





AIRLINI	E CARD NO	RIGHT EME	RIGHT EMERGENCY EXIT HATCH ESCAPE			BOEING CARD NO. 25-190-02-01		
DATE	INSPECTION - DETAILED	STRAPS			RELATED CARD			
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 10 YR	REPEAT 10 YR	APPLIC,	ABILITY ENGINE		
STATION	SKILL AIRPL				ALL	ALL		
		ACCESS 843			ZONE 232			

Inspect (Detailed) the right emergency exit hatch escape strap for condition and security.

A. Consumable Materials

Reference	Description	Specification
D50080	Lubricant - Solid Film	MIL-PRF-46010 Type 1
		(Supersedes MIL-L-8937)
G51031	Abrasive - 120 Grit, Aluminum Oxide, Sandpaper	
[C10-105]	Roll	

B. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-130	Brush - Soft-bristle Stainless Steel Wire

EFFECTIVITY AKS ALL	SOURCE MRB	INIGHT EMERGENCY EXIT HATCH ESCALE STRAIG	
		D633A109-AKS 25-190-02-01	Page 1 of 7 Oct 15/2015



	С	DATE			TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING C					
								25-190-	Т	1			
	TAS	K 25-	61-10	-710-	801				MECH	INS			
		Emergency Exit Door Life Line Check											
	(Figi	igure 1)											
	A. General (1) The life line (also called the escape strap) is kept in a stowage tube installed behind the												
		(1)	sidewall and the ceiling panels of the passenger compartment. One end of the life line is attached to the doorway structure of the emergency exit door. In an emergency, the hook end of the life line is attached to the fitting on the wing. The life line is used to help the passengers when they are on the wing.										
(2) The Left Emergency Exit Door (STA 589.5), a Right Emergency Exit Door (STA 589.5), a Left Emergency Exit Door (STA 627.5), and a Right Emergency Exit Door (STA 627.5). The stowage tube that contains the life line is located over the Left Emergency Exit Door (627.5) and the Right Emergency Exit Door (STA 627.5).													
	B.	Proc	edur	е									
		SUBTA	SK 25-61	-10-010-	008								
		WAR	RNING		IE DOOR HANDLE	OOR OPENING PATH IS E. THE DOOR IS SPRIN ND INJURIES COULD C	G LOADED TO OPEN	RELEASE					
(1) Open the emergency exit door.													
		SUBTA	SK 25-61	-10-010-	006								
		(2)	Ope	n the	cloth cover that ho	lds the hook and life line	[21].						
				-10-020-									
		(3)	Pull			ine [21] out over the wing							
			(a)	Make	e sure the life line	[21] can be attached to the	ne fitting on the wing.						
				-10-210-			ro 41						
		(4)			•	ne condition of the life lin	e [21]:						
			(a)			[21] has no worn areas.							
			(b)		e sure the life line	· -							
			(c)	Make cracl		on the end of the life line	[21] is not loose, corrod	ed, or					
		SUBTA	SK 25-61	-10-210-	009								
		(5)	Do tl	nese s	steps to examine tl	ne condition of the hook	assembly:						
			(a)	Make	e sure that the hoo	k assembly operates cor	rectly.						
				1)	The spring keepe	r snap hook on the hook	must open smoothly.						
				2)	The spring must of	close the spring keeper s	nap hook.						
			(b)	If the	hook is bent, repl	ace the life line.							
				NOT		ded that all new emerger er wear resistance.	ncy hook assemblies are	e cadmium					
			(c)	Make	e sure that there is	no corrosion on the hoo	k assembly.						
				1)	If corrosion preve	nts the operation of the h	nook assembly, replace t	he life line.					



DATE	DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CA 25-190-					
	a) Make sure th	at there is no corrosion of	on the spring.	ME	CH INS	
		spring is missing or the k	eeper snap hook is def	ective,		
	b) Remove corr mechanical p	osion on the spring keep procedures.	er snap hook or the hoo	ok with		
	2) If there is minor co	rrosion, remove it.				
	a) Use a soft-br G51031 [C10	istle stainless steel wire l 0-105].	brush, STD-130 or sand	dpaper,		
(d)	Add protection to the ho	ook assembly with dry filn	n lubricant.			
	 Apply solid film lub the hook assembly 	oricant, D50080 or equiva /.	alent heat cured dry film	lubricant to		
	NOTE: This lubric limit metal	ant will decrease worn paseizures.	arts, corrosion, prevent	galling, and		
		ant looks almost the samusticant but in a clear co		ble, use an		
SUBTASK 25-61-1 (6) Fold t	10-420-013 he life line [21] as show	n (Figure 1).				
SUBTASK 25-61-1	0-420-016	and close the cloth cove	r			
(7) Put th		and close the cloth cove	1.			
	the emergency exit dod	or.				
		END OF TASK ———				
EFFEC'	TIVITY	SOURCE RIGHT EMERG	ENCY EXIT HATCH ESC	ADE STRADS		
AKS		MRB RIGHT EMERG	LNOT EXIT HATCH £30/	AFE SINAPS		
		D633A109-AKS 25-190-02-01			e 3 of 1 15/201	



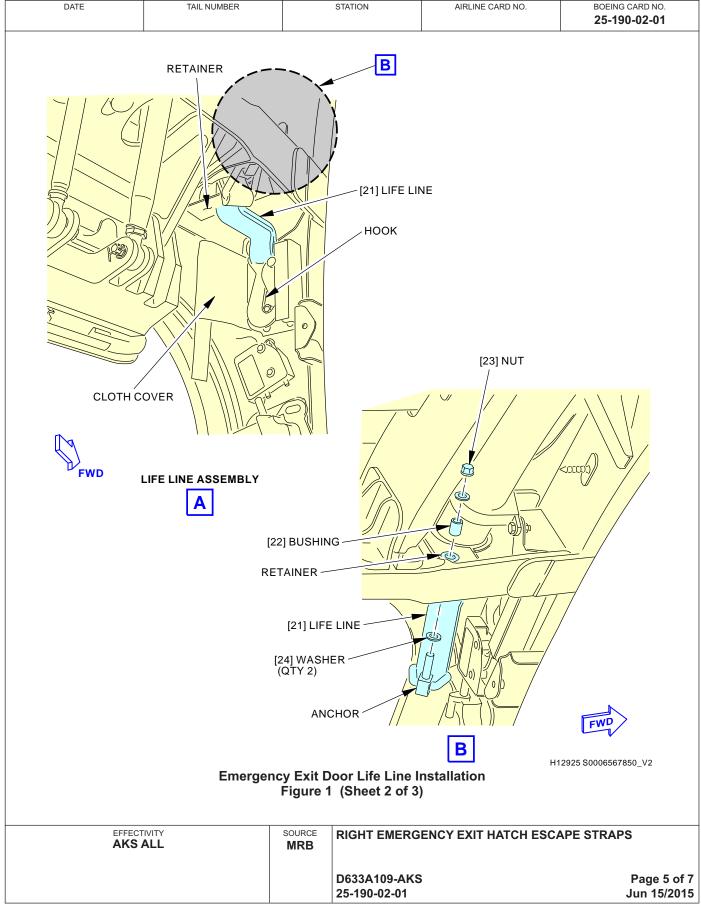
737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-190-02-01 LIFE LINE STOWAGE TUBE LIFE LINE **ASSEMBLY** Α EMERGENCY EXIT DOOR (DOOR SHOWN IN OPEN POSITION) (DOOR LINER REMOVED) (LEFT SIDE IS SHOWN, RIGHT SIDE IS OPPOSITE) F82116 S0006567849_V2 **Emergency Exit Door Life Line Installation** Figure 1 (Sheet 1 of 3) EFFECTIVITY SOURCE **RIGHT EMERGENCY EXIT HATCH ESCAPE STRAPS AKS ALL MRB** D633A109-AKS Page 4 of 7

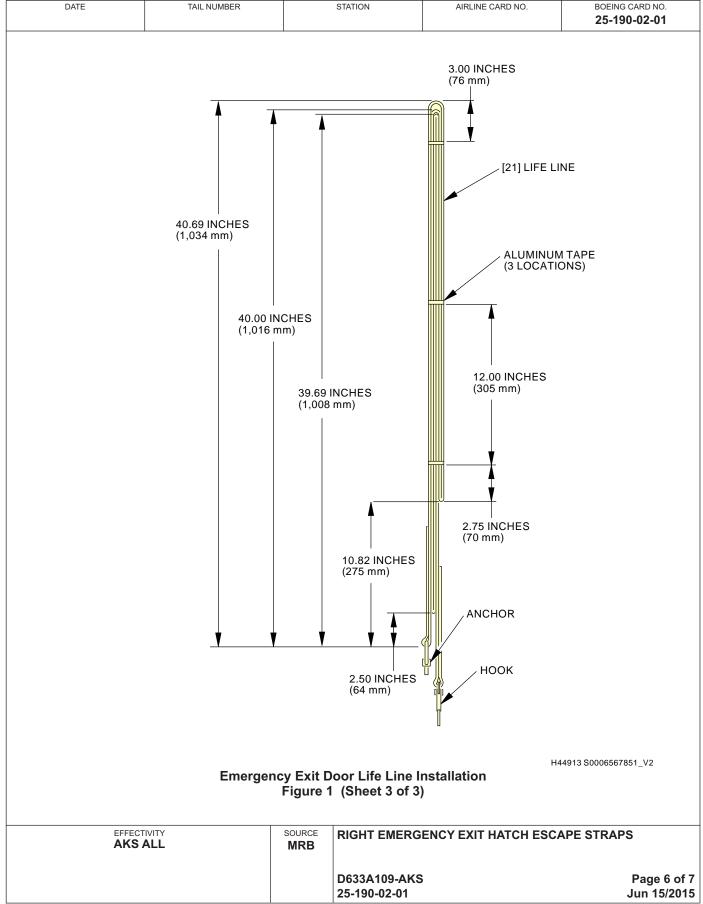
25-190-02-01

Jun 15/2015











DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-190-02-01
	RIVET		RIVET KEEPER SNAP HOOK RING KEEPER AP HOOK	HOOK	
	Emerge	ncy Exit [Door Life Line I Figure 2	nspection 2292	2580 S0000519070_V2
EFFEC AKS	TIVITY ALL	SOURCE MRB	RIGHT EMERGI	ENCY EXIT HATCH ESCA	PE STRAPS
			D633A109-AKS 25-190-02-01		Page 7 of 7 Jun 15/2015





AIRLINE CARD NO		INSPECT FLIGHT COMPARTMENT ESCAPE			BOEING CARD NO. 25-210-00-01		
DATE	INSPECTION - DETAILED		LANYARDS		RELATE	D CARD	
	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICA	ADII ITV	
	CREW CABIN	1.1	10 YR	10 YR	AIRPLANE		
STATION	SKILL AIRPL	_			ALL	ALL ALL	
		ACCESS			ZONE 211 212		

Inspect (Detailed) the flight compartment escape lanyards (2) for condition and security.

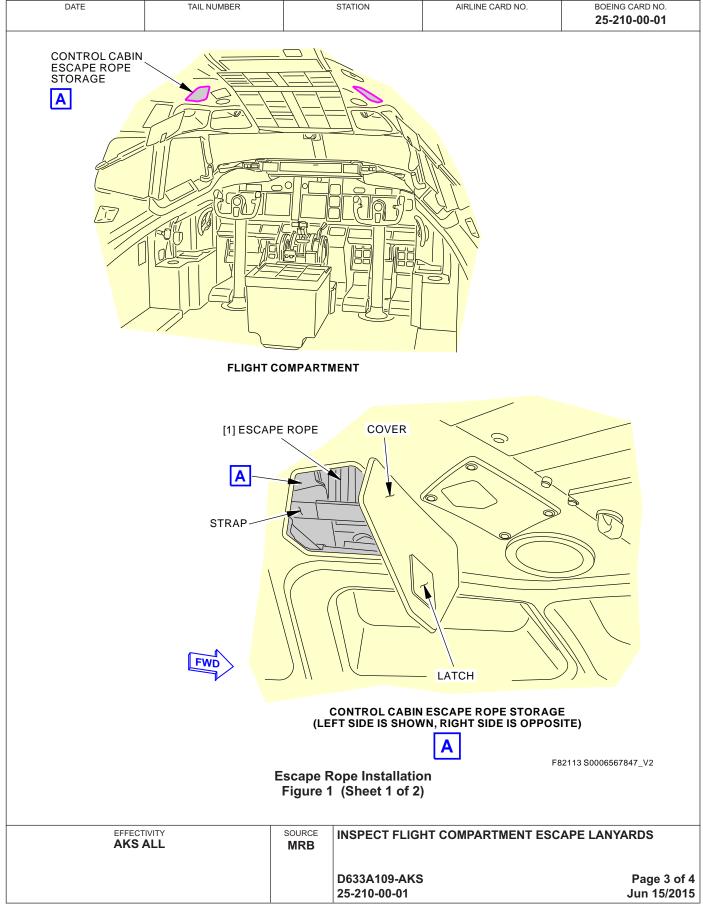
A. References

Reference	Title
AMM 25-61-10-000-801	Flight Compartment Escape Rope Removal (P/B 201)
AMM 25-61-10-400-801	Flight Compartment Escape Rope Installation (P/B 201)

EFFECTIVITY AKS ALL	SOURCE MRB	INSPECT FLIGHT COMPARTMENT ESCAPE LANYA	ARDS	
		D633A109-AKS 25-210-00-01	Page 1 of 4 Oct 15/2014	



	DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO 25-210-00-01									
	TAS	K 25-	61-10-2	210-801					MECH	INSP
1.		ht Co ure 1)		nent Escape Rope In	spection					
	Α.	Gen	eral							
		(1)	An es			• '		artment		
	В.	Prod	edure							
		SUBTA	SK 25-61-1	0-020-010						
		(1)	Do thi	s task: Flight Compartı	ment Esca	ape Rope Remo	oval, AMM TASK 25-61-	10-000-801.		
		SUBTA	SK 25-61-1	0-210-010						
		WAF	RNING:	EXAMINE THE COR	E FOR D	AMAGE. IF YOU	J DO NOT REPLACE A			
		(2)						sed. If you		
			• Fligl	ht Compartment Escap	e Rope R	Removal, AMM 1	TASK 25-61-10-000-801			
			• Fligl	ht Compartment Escap	e Rope Ir	nstallation, AMM	1 TASK 25-61-10-400-8	01		
		SUBTA	SK 25-61-1	0-210-001						
		(3)	Do thi	s check of each escap	e rope:					
			(a) I	Make sure the rope is r	not wet or	knotted.				
			(b) I	Make sure the rope is r	not fraying	g or have any w	orn areas.			
				1) If the escape rope	e is fraying	g or has worn ar	eas, replace it.			
		SUBTA	SK 25-61-1	0-210-004						
		(4)		·						
		(5)	Make	sure the anchor fitting	is not ben	nt or corroded.				
			SK 25-61-1	nent Escape Rope Inspection cape rope (also called an escape lanyard) is installed in the flight compartment the captain's seat and above the first officer's seat. PARAMETER COMPARTMENT ESCAPE ROPE REMOVAL, AMM TASK 25-61-10-000-801. DEPURATION OF TASK DEPURATIO						
		(6)		s task: Flight Comparti 25-61-10-400-801.	ment Esca	ape Rope Instal	lation, AMM			
					END OF	TASK ———				
			EFFEC*			INSPECT FLIG	HT COMPARTMENT ESC	APE LANYAR	RDS	<u> </u>
							3		_	





737-600/700/800/900 TASK CARDS

		TAS	K CARDS		
DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-210-00-01
[1] ESCAPE	EROPE			ANCHOR FITTING ANCHOR PLATE	
	FWD				
			SPRII	NG	
		ANCHO (DOOR N	DR FITTING IOT SHOWN)		
1 IF THE LAN' SURE THAT	YARD RING HAS A HOLI THE COTTER PIN IS IN	Escape R	AFT NEAR THE Rope Installati (Sheet 2 of 2	G51 ON	541 S0006567848_V3
EFFECT AKS	ALL	SOURCE MRB	INSPECT FLIC	GHT COMPARTMENT ESCA	APE LANYARDS

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AIRLINI	AIRLINE CARD NO		EMERGENCY ESCAPE SYSTEM OPERATIONAL			BOEING CARD NO. 25-220-00-01		
DATE	TASK OPERATIONAL		CHECK		RELATED) CARD		
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD NOTE	REPEAT	APPLICA AIRPLANE	BILITY ENGINE		
STATION	SKILL AIRPL	NOTE			ALL	ALL		
		ACCESS 831 834 841 844			ZONE 221 222 241 242			

Operational check of the entry and service door mounted emergency escape slide deployment system (on airplane).

INTERVAL NOTE: Each operator is to perform an operational check of its Boeing 737NG entry and service door slide system to ensure its airline specific slide maintenance program is adequate. Each check from the operator's fleet shall include a minimum of one operational check of an installed slide from the left or right side at each door position, during each 6 year period. Checks shall alternate between the left and right door position. The total set would be 2 door minimum every 6 years.

References

Reference	Title
AMM 25-66-00-840-801	Escape System Restoration (P/B 201)
AMM 25-66-01-200-801	Escape Slide Pack Inflation Cylinder Check (P/B 601)

B. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1146	Pad - Protective, Ensolite or Equivalent 1/2 Inch by 72 Inches by 96 Inches

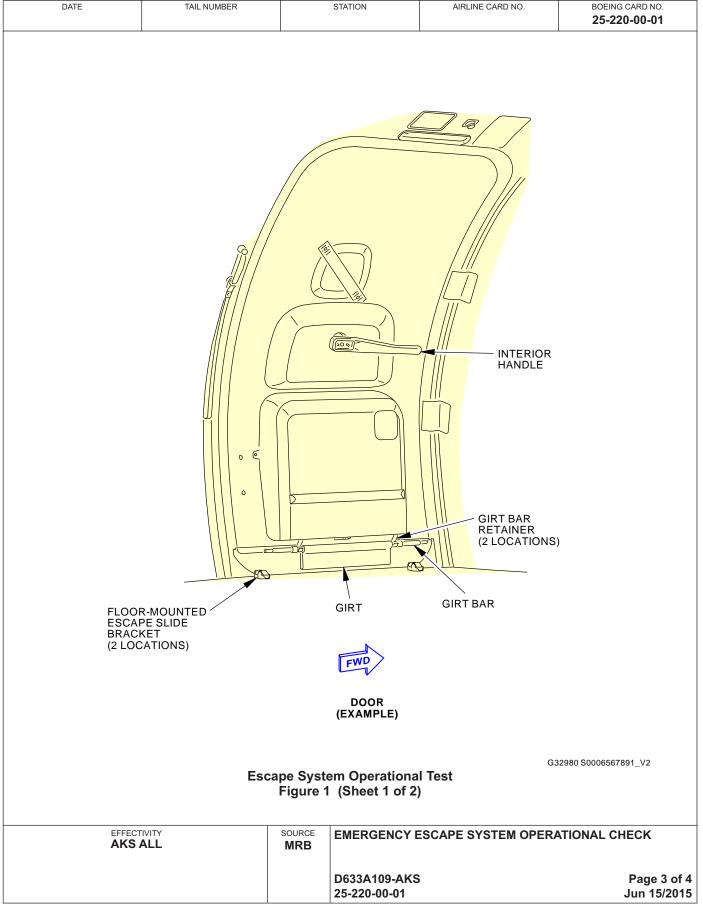
EFFECTIVITY AKS ALL	SOURCE MRB	EMERGENCY ESCAPE SYSTEM OPERATIONAL CHECK	
		D633A109-AKS 25-220-00-01	Page 1 of 4 Feb 15/2015



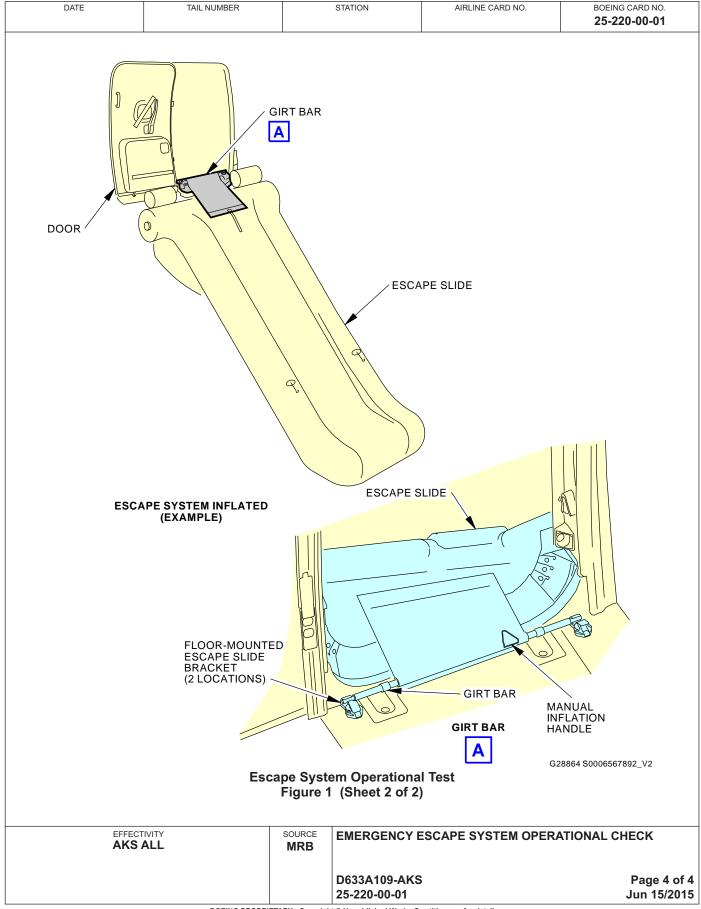
737-600/700/800/900 **TASK CARDS**

	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-220-					
TAS	SK 25-	-66-00	0-710-801	•				MECH	INSF			
Esc	ape S	yste	m Operational Test									
(Fig	ure 1)											
A.	Prep	oare f	for the Test									
			6-00-860-001									
	(1)	Prep	pare the airplane for the	test as foll	lows:							
		CAUTION: YOU MUST ATTACH THE PROTECTIVE PAD TO THE FUSELAGE BELOW THE APPLICABLE DOOR. IF YOU DO NOT INSTALL THE PROTECTIVE PAD, DAMAGE TO THE FUSELAGE SKIN AND PAINT CAN OCCUR.										
	(a) Install the protective pad, STD-1146 to the fuselage below the applicable door with the long side parallel to the ground.											
		(b)	Put a pad on the groun	nd where th	he escape slide	will inflate.						
		(c)	Make sure the door is	closed and	d latched.							
	SUBTA		6-00-210-001									
	(2)	Do t	his task: Escape Slide F	Pack Inflation	on Cylinder Che	eck, AMM TASK 25-66-	01-200-801.					
B.	Test											
			6-00-710-001		avatara.							
	(1)		the operational test of the	•	-	the cover						
	(a) Remove the girt bar from the girt bar retainers on the cover.(b) Install the girt bar in the floor-mounted escape slide brackets.											
		WAI		OOR WH	ERE THE SLIDI	S OR EQUIPMENT IN E WILL INFLATE. WHE IS OR DAMAGE TO EC	EN THE					
		(c)	Move the interior hand	lle to the o	pen position and	d open the door fully.						
		(d)	Make sure the escape releases from the door	slide fully			pe slide					
			NOTE: The escape sl	ide must ir	nflate automatica	ally to pass the operation	onal test.					
		(e)	If the escape slide doe inflate the escape slide		te automatically,	pull the manual inflation	on handle to					
		(f)	Make sure the escape	slide is ful	lly inflated.							
		(g)	Make sure the lights o	n the slide	are on.							
C.	Put	the A	airplane Back to Its Us	ual Condit	tion							
			6-00-860-002			05 00 00 040 004						
	(1)	Do t	his task: Escape Syster	n Restorat	ion, AMM TASK	25-66-00-840-801.						
				- END OF	TASK ———							
			ECTIVITY S ALL	SOURCE MRB	EMERGENCY E	SCAPE SYSTEM OPER	ATIONAL CHE	CK				
					D633A109-AKS 25-220-00-01			Page 2 eb 15/				













AIRLINE CARD NO		TITLE FWD ENTRY EMERGENCY ESCAPE SLIDE			BOEING CARD NO. 25-240-00-01		
DATE	TASK RESTORE				RELATE	D CARD	
TAIL NUMBER STATION	WORK AREA PASS CABIN SKILL AIRPL	VERSION 1.1 NOTE	THRESHOLD NOTE	REPEAT	APPLICABILITY AIRPLANE ENGINE		
					ALL	ALL	
		ACCESS 831			ZONE 221		
		-					

Restore the fwd entry emergency escape slide at the manufacturer's recommended interval.

INTERVAL NOTE: Vendor Rec

A. Consumable Materials

Reference	Description	Specification
D00113	Lubricant - Solid Film, Liquid Dispersed	BMS3-8
D00670	Oil - Penetrating (For Loosening Frozen Metallic Parts)	A-A-50493
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper (Cheesecloth, Gauze)	BMS15-5 Class A

EFFECTIVITY AKS ALL	SOURCE MRB	FWD ENTRY EMERGENCY ESCAPE SLIDE	
		D633A109-AKS 25-240-00-01	Page 1 of 19 Jun 15/2015



DATE TAIL NUMBER				STATION	AIRLINE CARD NO.	BOEING C. 25-240-					
Esc	ape S	Slide	1-000-801 Pack and Cover Rer	noval				MECH	IN		
(Fig	ure 1))									
A.	Pre	pare 1	for the Removal								
			6-01-860-001								
	(1)		Make sure the door is safe as follows:								
(a) Make sure the door is closed and latched.											
		WAI	RETAINER. RETAINERS	IF THE GIRT S, ACCIDENT IIS CAN CAU	BAR IS NOT AL INFLATION	LLED IN THE GIRT BAR INSTALLED IN THE GIR I OF THE ESCAPE SLID TO PERSONS OR DAM	RT BAR DE CAN				
		(b)	Make sure the girt b	ar [4] is insta	lled in the girt	bar retainers [5].					
В.	Ren	noval									
	SUBTA	NSK 25-6	6-01-020-001								
	(1)	Ren	nove the escape slide	pack and co	ver [2] from the	e door as follows:					
		(a)	Remove the screws [9] and washers [10] that attach the cover [3] to the door.								
			Make a note of the screw [9] lengths and locations for the subsequent installation.								
		(b)	Pull the top of the cover [3] inboard until it is clear of the door structure.								
			the work so	•	ment is open fo	cape slide pack can export a minimum time to avo					
		WAI	POUNDS (2 THE ESCAF ESCAPE SL	4.94 KILOGF PE SLIDE PAOL LIDE PACK AI	RAMS). BE CA CK AND COVI ND COVER IN	ER WEIGH APPROXIMA REFUL WHEN YOU LIF ER. IF YOU LIFT OR MC ICORRECTLY, INJURIES ENT CAN OCCUR.	T OR MOVE VE THE				
		(c)	Lift and remove the escape slide pack and cover [2] to disengage the lower fitting [15] from the bottom support bracket [14].								
		(d)	Put the escape slide pack and cover [2] on a clean horizontal surface with the cover backing pan [11] against the horizontal surface.								
	SUBTA		6-01-860-006								
	(2)					the inflation cylinder valv	re [17]:				
		(a)	Remove the latch [7] from the latch bracket [13]. Open the cover [3] to get access to the inflation cylinder valve [17].								
		(b)	Open the cover [3] t	to get access	to the inflation	i cylinder valve [17].					



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 0-00-01	
WARM	SLIDE OR SLIDE	OVER. \ E-RAFT (WITHOUT THE CAN INFLATE A	ATION CYLINDER IF SAFETY PIN, THE ES ACCIDENTALLY. IF IT I	CAPE NFLATES,	MECH	INSP
(c) F	Remove the safety pin [19] from	the pocket [18].				
	Put the safety pin [19] in			alve [17].			
	——-	END OF	TASK ———				
EFFECT AKS		SOURCE MRB	FWD ENTRY EN	MERGENCY ESCAPE SL	.IDE		
			D633A109-AKS 25-240-00-01			Page 3 o Oct 15/	



	[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
2.	Esc	ape S	lide Pa	000-802 ack Removal	1				MECH	INSI
	(Fig	ure 1)								
	A.	Prep	oare fo	r the Removal						
		(1)	If the	^{o1-010-001} escape slide pack and and Cover Removal, T			the door, do this task: E	Escape Slide		
	В.	Rem	noval							
				01-420-001						
		(1)		ove the escape slide pa						
			` ,	Remove the girt bar [4]	-		s [5].			
			` ,	Remove the latch [7] fr						
			` ,	Open the cover [3] to g	•		·	. 51		
			NOTE: When the compartment is open, the escape slide pack can expand. Plan the work so the compartment is open for a minimum time to avoid the expansion of the escape slide pack.							
		(d) Remove the pin [33], ladder tie [32], and stud [31] that attach the bonding jumpers [12] to the backing pan [11] as shown.								
			(e)	Fully open the cover [3] to get ac	cess to the esca	ape slide pack [26].			
			(f)	emove the safety pin [19] from the pocket [18] on the escape slide pack [26].						
			WAR	ESCAPE SLIDI	E CAN AC PE SLIDE	CIDENTALLY IN	FLATION CYLINDER VANFLATE. ACCIDENTAL NJURIES TO PERSON	INFLATION		
			(g)	Put the safety pin [19]	into the int	flation cylinder v	alve [17] as shown.			
			. ,	Remove the escape sli horizontal surface.	ide pack [2	26] from the bac	king pan [11] and put it	on a clean		
			ļ		ng pan. If i		ort box immediately afte , the slide will grow and			
			` '	If required, remove the girt release strap [28].	locknut [2	9] and screw [2	7] to remove the latch [7] from the		
				NOTE: The latch [7] m	nay be nee	eded for a replac	cement escape slide.			
					- END OF	TASK ———				
			AKS	ALL	SOURCE MRB	FWD ENTRY EN	MERGENCY ESCAPE SL	IDE		
						D633A109-AKS 25-240-00-01			age 4 eb 15/	



737-600/700/800/900 TASK CARDS

					TAS	K CARDS				
		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
3.				I-400-801 Pack Installation					MECH	INSF
	(Fig	ure 1)							
	A.	Pre	pare 1	or the Installation						
		SUBT	ASK 25-6	6-01-860-002						
		(1)		e sure the cover [3] and king pan [11] against the			a clean horizontal surfac	e with the		
		SUBT	ASK 25-6	6-01-860-004						
		(2)	Mak	e sure the cover [3] is o	pen.					
			NO	 ·	rtment is o		slide pack can expand. Fund the ex			
				6-01-600-001						
		(3)	Lub	ricate the latch assembl	-					
			(a)	Apply solid film lubrica		-	e slide latch.			
			(b)	Apply penetrating oil,	D00670 to	the pins [39].				
			(c)	Apply penetrating oil, penetrates the pins.	D00670 to	the pins [39] a	gain to make sure that th	ne oil fully		
			(d)	If there is too much pe the unwanted oil from	_		e a cotton wiper, G00034	to remove		
		SUBT	ASK 25-6	6-01-700-002						
		(4)	Do a	a check of the latch rele	ase, Figure	e 2.				
			(a)	Move the latch block [move.	35] to the l	eft or right side	until the latch block [35]	cannot		
				NOTE: The test starts continue to me	-	pull the latch	block [35] to the side and	l it cannot		
			(b)	Use a spring force sca	ale to pull th	ne latch chain i	in the same direction.			
				1) Pull the latch cha	ain or cable	in the same d	lirection 10 ±5°arc (0.17	±0.09 rad).		
			(c)	Make sure that the lat	ch assemb	ly releases wit	h no more than 30 lbf (13	33 N).		
		SUBT	ASK 25-6	6-01-700-003			·	,		
		(5)	Do t	he steps for the check	of the latch	release again				
			(a)	Move the latch block [35] in the c	pposite directi	on.			
				NOTE: To complete the in the opposite			is necessary to do the cl	neck again		
		SUBT	ASK 25-6	6-01-860-005						
		WA	RNIN	ESCAPE SLIDE CA	N ACCIDE	NTALLY INFL	TION CYLINDER VALVE ATE. ACCIDENTAL INFL S TO PERSONS AND D	ATION OF		
		(6)	Mak	e sure the safety pin [1	9] is in the	inflation cylind	er valve [17].			
				ECTIVITY S ALL	SOURCE MRB	FWD ENTRY E	EMERGENCY ESCAPE SL	IDE	•	

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737-600/700/800/900 TASK CARDS

[DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
В.	Insta	allatio	on of	the Escape Slide	Pack				MECH	INS
			6-01-420	-						
	(1)	Insta	all the	escape slide pac	k [26] in the	e cover [3] as fo	llows:			
		NOT	tr				immediately after reme grow and not fit into th			
		(a)	Put	the escape slide p	ack [26] in	the backing par	n [11].			
			NOT			•	n not be fitted into the must be installed.			
		(b)	Care	efully extend the g	irt [6].					
		(c)	Mak	e a "Z" fold [21] in	the girt [6]	as shown.				
		(d)		n and push togeth velcro strips [24] c			the bottom side of the g	girt [6] with		
			NO	ΓΕ: The velcro str	ips should	be clean and fu	nction as intended.			
		(e)	_	n and push togeth ro strip [24] on the		o strip [24] on tl	he girt retainer flap [20]	and the		
		(f)		quired, install the sase strap [28].	screw [27] a	and locknut [29]	to attach the latch [7]	to the girt		
			1)	Make sure that t threads.	he screw [2	27] extends thro	ugh the locknut [29] by	1 to 2		
		(g)		te sure that there in the inflation cylin			ount of slack in the firin	g cable [30]		
		(h)	If the	ere is not enough	slack in the	e firing cable [30] (Figure 1), do these s	teps:		
			1)	Do not remove the	ne safety pi	in [19].				
			2)	Make sure the sl	ide is corre	ectly installed in	the backing pan [11].			
			3)			0	correct. Make sure the lcro strip [24] on the gir	•		
			4)	Make sure that t [30] near the infl			m amount of slack in thigure 1).	e firing cable		
				a) Hold the fir slack from	-	30] at the inflatio	on cylinder valve [17] ar	nd pull the		
				dired remo mini	ction of the oved and th	inflation cylindene minimum am	e [30] away from the gir er valve [17] until all slad ount of slack is achieve , the escape slide pack	ck is ed. If the		
						in the firing cab ne inflation cyline	ole between the back of der valve [17].	fthe		
			S ALL		SOURCE MRB	FWD ENTRY EN	MERGENCY ESCAPE SL	IDE		
						D633A109-AKS		P	age 6	of '

25-240-00-01

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DATE		TAIL	IUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-240		
	WARI	THE VAL\	SAFETY P /E, THE ES	IN IS REI CAPE SL	MOVED FROM LIDE CAN ACCI	E ESCAPE SLIDE PAC THE INFLATION CYLII DENTALLY INFLATE. T AMAGE TO EQUIPMEI	NDER THIS CAN	MECH	INSP
	(i)	Remove the	safety pin	[19] from	the inflation cyli	nder valve [17].			
	(j)	Put the safe	ty pin [19] ir	nto the po	cket [18] on the	escape slide pack [26]			
	(k)	Close the co	over [3].						
	` '		•	•	ee the pressure djust as follows:	gauge [16] of the inflat	ion cylinder		
		1) Fully o	pen the cov	er [3] to g	get access to the	e escape slide pack [26].		
		,	e the safety ation cylind		•	t [18] and install safety	pin [19] in		
		,	the escape and ow [1].	slide pack	(26) to get the	pressure gauge [16] in	the center of		
		,			minimum amou cylinder valve [nt of slack in the inflation 17] (Figure 1).	on firing		
		,	lold the firin lack from th		30] at the inflation	on cylinder valve [17] ar	nd pull the		
		<	directi remov minim	on of the ed and th	inflation cylinde ne minimum am	e [30] away from the gir er valve [17] until all slad ount of slack is achieve , the escape slide pack	ck is d. If the		
		<			in the firing cat ne inflation cylin	ole between the back of der valve [17].	the		
		5) Remov	e the safety	y pin [19]	from the inflatio	n cylinder valve [17].			
		6) Put the	safety pin	[19] into t	he pocket [18] o	on the escape slide pac	k [26].		
	` '	Close the copan [11].	over [3] suffi	ciently to	connect the bo	nding jumpers [12] to th	e backing		
	. ,	Install the st the backing		der tie [32	?] and pin [33] to	attach the bonding jur	npers [12] to		
	(o)	Close the co	over [3].						
	\					e lower edge of the cov 12.0 in. (304.8 mm).	er [3] to the		
	(q)	Align the ho	les in the la	tch brack	et [13] and latch	bracket [35] to install t	he latch [7].		
	(r)	Install the la	tch [7] in the	e latch bra	acket [13] and la	atch bracket [35] as sho	own.		
	` '					and 'B' between the late an 0.01 in. (0.25 mm) (F			
	(t)	Install the gi	rt bar [4] in	the girt ba	ar retainers [5].				
				END OF	TASK ——				
	EFFEC AKS			SOURCE MRB	FWD ENTRY EN	MERGENCY ESCAPE SL	IDE		
					D633A109-AKS 25-240-00-01			age 7 e eb 15/	
			POEING PROPRIET		20-240-00-01		<u>'</u>	CD 10/	



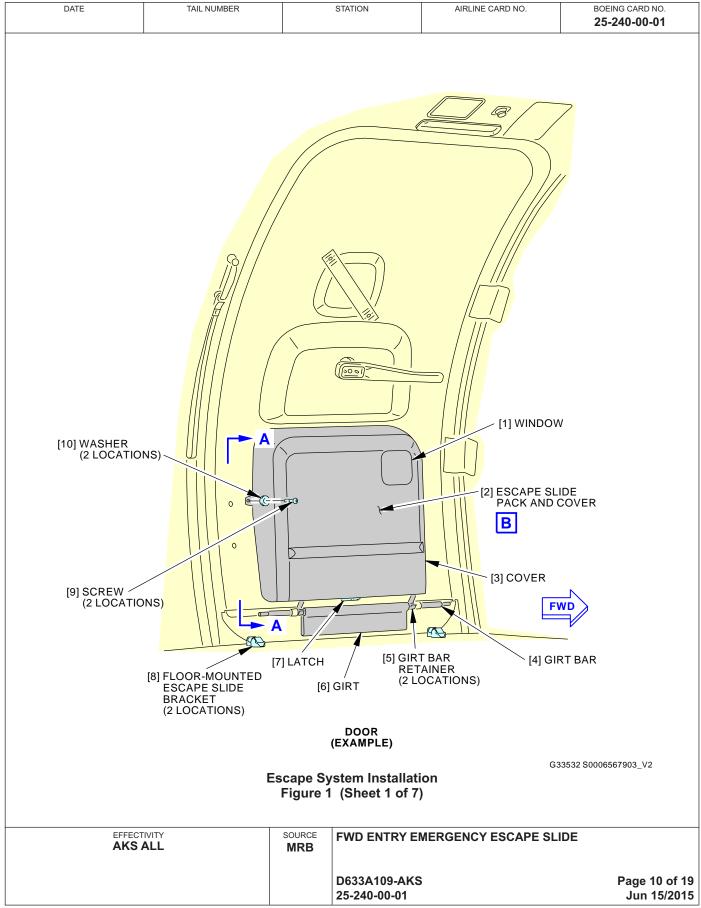
	DATE			TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CAF		
							25-240-0		INIC
		-66-01-					-	MECH	INS
	_		ack a	nd Cover Installa	tion				
(Fig	ure 1)								
A.				Installation					
		-ASK 25-66 If the			is not installed in the co	ver [3], do this task: Esca	ano Slido		
	(1)	Pack	Instal	lation, TASK 25-60		ver [3], do triis task. Esca	ape Slide		
	(-)	-NASK 25-66		15 the door is closed	and latabad				
	(2)				and latened.				
	(3)		e follo		ke sure that the safety p	oin [19] is not installed in	the inflation		
		. ,			ck and cover [2] on a cl t the clean flat horizont	ean flat horizontal surfac al surface.	e with the		
		(b)	Remo	ove the latch [7] fro	om the latch bracket [13].			
		(c)	Open	the cover [3] suffi	ciently to see the inflation	on cylinder valve [17].			
		WAR	NING	THE ESCAPE S INFLATION CYL	SLIDE WILL NOT OPER	IE INFLATION CYLINDE RATE IF THE SAFETY PI E ESCAPE SLIDE DOES ED.	N IS IN THE		
		(d)	Make	sure a safety pin	[19] is not installed in th	e inflation cylinder valve	[17].		
			CAUT	THE SAFE VALVE, TH	TY PIN IS REMOVED F E ESCAPE SLIDE CAN SE INJURIES TO PERS	THE ESCAPE SLIDE PARTION OF THE INFLATION OF THE INFLATION OF THE INFLATION OF THE INFLATIONS OR DAMAGE TO	CYLINDER		
			1)	Remove the safety	y pin [19] from the inflat	ion cylinder valve [17].			
			2)	Put the safety pin	[19] into the pocket [18]	on the escape slide pac	k [26].		
		(e)	Close	the cover [3].					
		(f)	Instal	the latch [7] into t	the latch bracket [13].				
В.	Inst	allatio	n						
	SUBTA	ASK 25-66-	01-420-0	04					
	(1)	Instal	I the e	escape slide pack	and cover [2] on the do	or as follows:			
		WAR	<u>NING</u>	THE ESCAPE S KILOGRAMS). I	SLIDE WEIGHS APPRO	SCAPE SLIDE PACK AN EXIMATELY 55 POUNDS EFUL, INJURIES TO PE OCCUR.	(24.94		
		(a)	Instal	I the escape slide	pack and cover [2] on t	he door.			
			1)	Make sure that the	hall end of the lower fi	tting [15] goes through th	a opening in		

EFFECTIVITY AKS ALL	SOURCE MRB	FWD ENTRY EMERGENCY ESCAPE SLIDE	
		D633A109-AKS 25-240-00-01	Page 8 of 19 Oct 15/2014

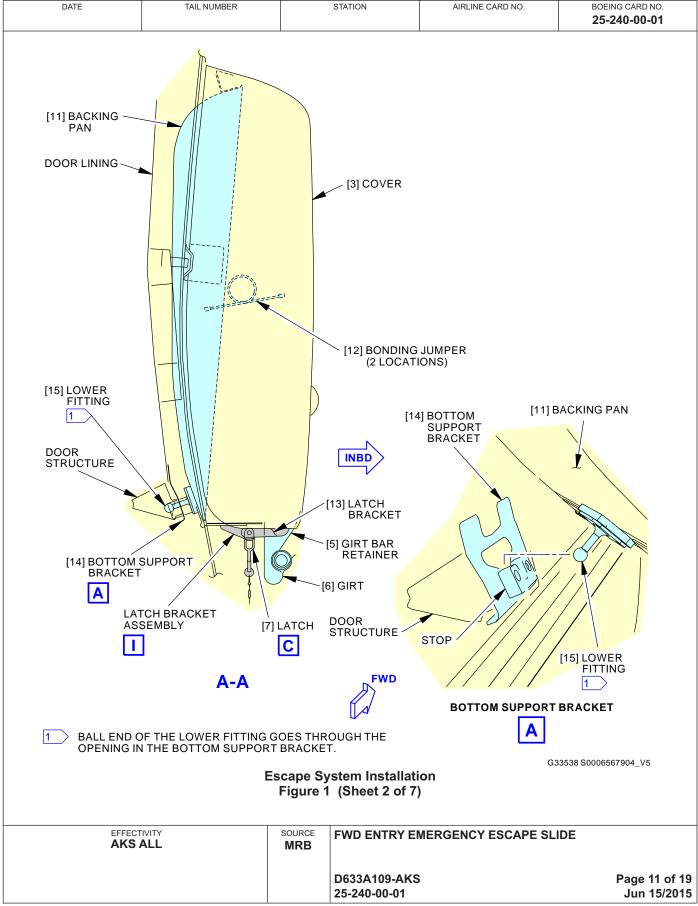


DATE	TAIL NUMBER	STATION		AIRLINE CARD NO.		CARD NO.	
(b)	Push the top of the esca	ape slide pack and	d cover	[2] outboard until it is a	gainst the	MECH	INSP
	NOTE: When the comp		open for	ape slide pack can expa a minimum time to avo			
(c)	Install the screws [9] and	d washers [10] tha	at attacl	h the cover [3] to the do	or.		
	 Make sure to use to procedure. 	he correct length	screw [9], as noted in the remo	oval		
(d)	Make sure you can arm floor-mounted brackets		by insta	alling the girt bar [4] into	the		
(e)	Install the girt bar [4] into	o the girt bar retai	iner [5].				
	NOTE: The length of the	e girt bar for the F	FWD Do	oor is 33.2 in. (84.3 cm)	long.		
	The length for the	ne girt bar for the	AFT Do	or is 29.2 in. (74.2 cm)	long.		
		END OF TASK —					
	S ALL	SOURCE FWD EN	NTRY EM	MERGENCY ESCAPE SLI	DE	1	<u> </u>
		D633A1	09-AKS			Page 9	of 19
		25-240-0				Oct 15/	

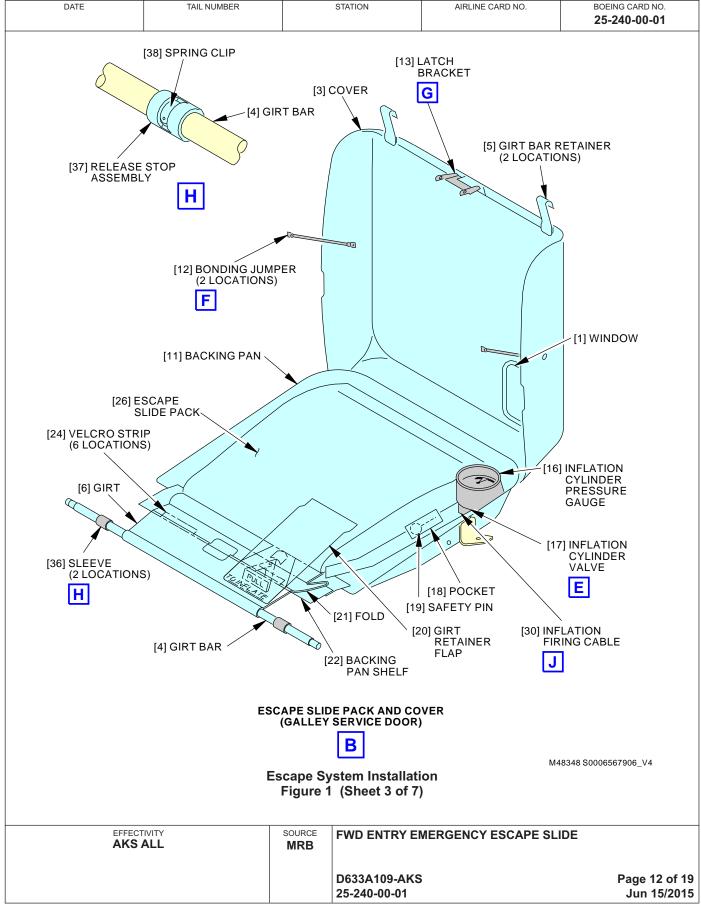




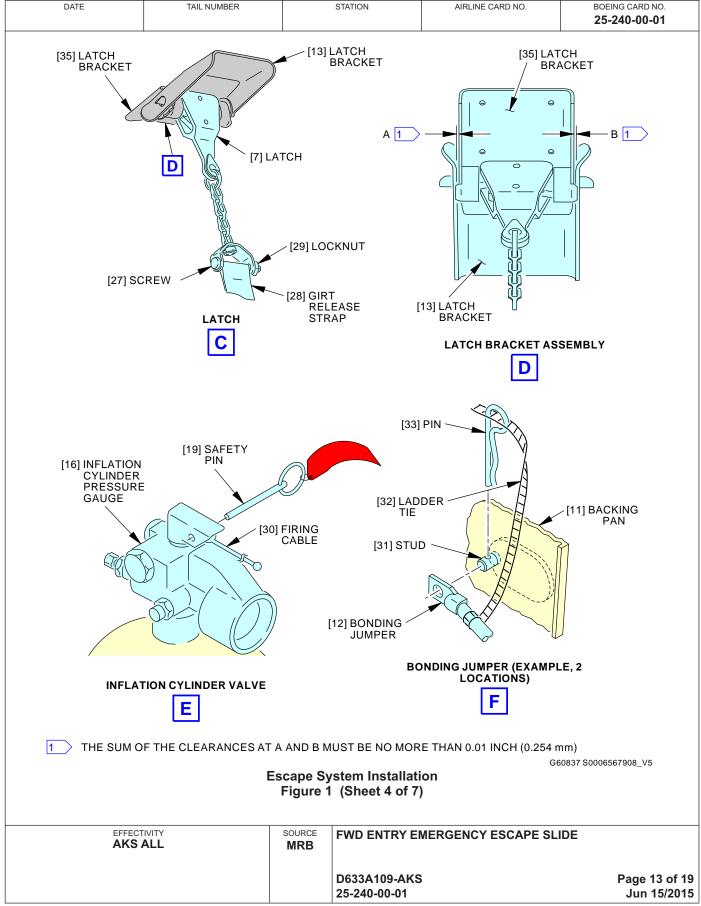














737-600/700/800/900 TASK CARDS

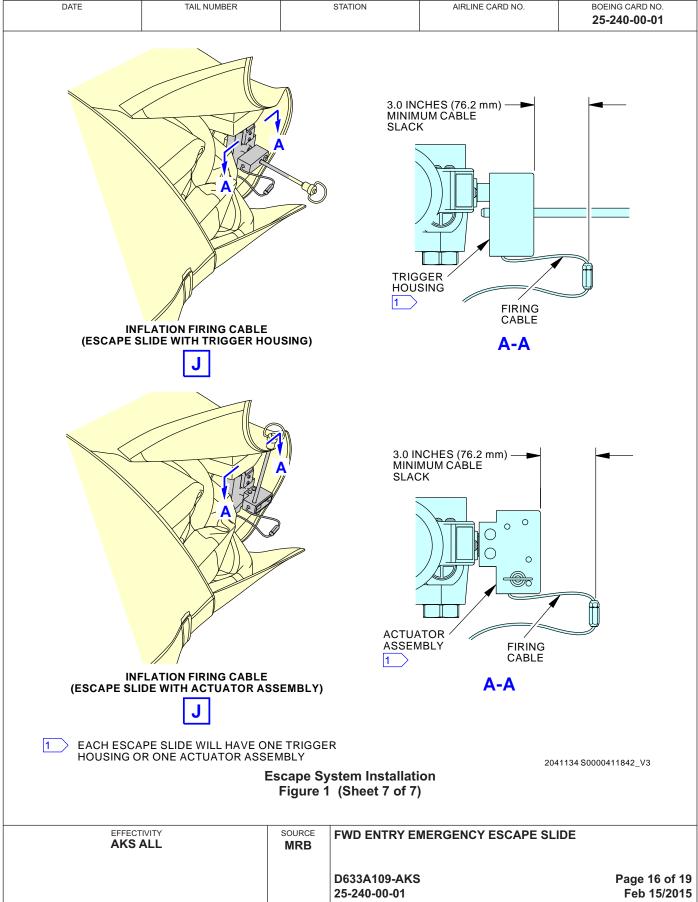
DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-01 [13] LATCH BRACKET [35] LATCH BRACKET STEP 2 (COCK LATCH) STEP 1 (ALIGN BRACKETS) G [13] LATCH [13] LATCH BRACKET **BRACKET** [35] LATCH BRACKET [35] LATCH **BRACKET** [7] LATCH STEP 4 (LATCH IN LOCKED POSITION) STEP 3 (POSITION LATCH AS SHOWN) G K62921 S0006567909_V2 **Escape System Installation** Figure 1 (Sheet 5 of 7) EFFECTIVITY SOURCE **FWD ENTRY EMERGENCY ESCAPE SLIDE AKS ALL MRB** D633A109-AKS Page 14 of 19 Jun 15/2015 25-240-00-01



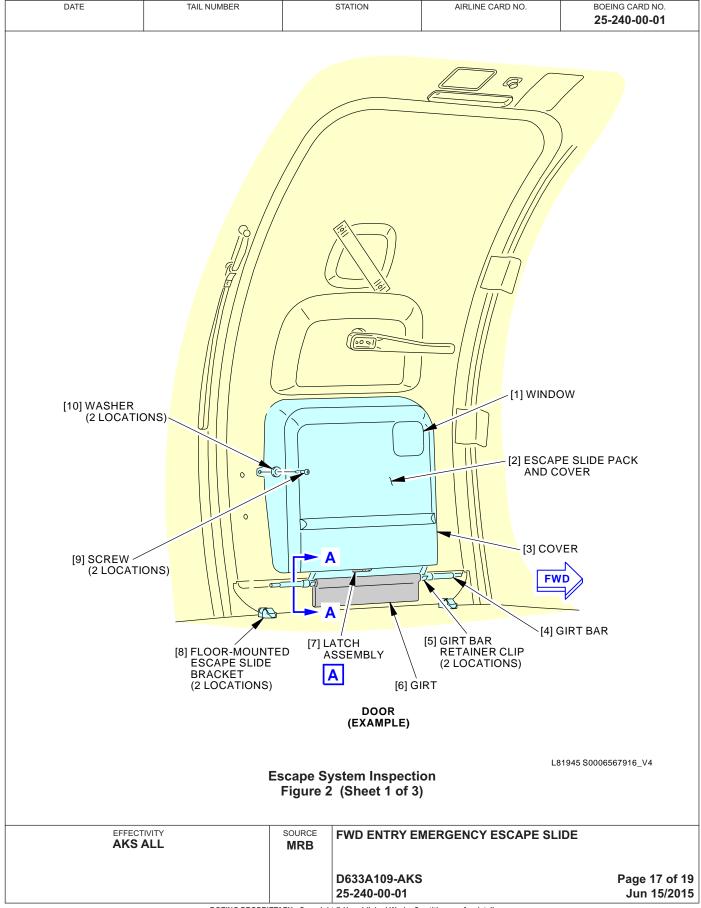
737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-01 [35] LATCH BRACKET (PAN ASSEMBLY) 0 0 LATCH BLOCK **ASSEMBLY SPLIT RING** [35] LATCH BRACKET (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH CLEVIS LATCH BLOCK 2 3 10.0° ±5.0° 10.0° ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. USE A MAXIMUM OF 30 POUNDS OF FORCE TO RELEASE THE LATCH. 1739991 S0000315072_V4 **Escape System Installation** Figure 1 (Sheet 6 of 7) EFFECTIVITY SOURCE **FWD ENTRY EMERGENCY ESCAPE SLIDE AKS ALL MRB** D633A109-AKS Page 15 of 19 Jun 15/2015 25-240-00-01











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DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-01 LATCH BRACKET 0 (PAN ASSEMBLY) 0 LATCH BLOCK **ASSEMBLY** SPLIT RING [35] LATCH BRACKET PIN (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH BLOCK LATCH CLEVIS 3 > 2 Ē 10.0° ±5.0 10.0°, ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. THE LATCH ASSEMBLY RELEASES AT 10 DEGREES, ± 5 DEGREES, WITH A MAXIMUM OF 30 LBF (133.45 NM). 3 M78780 S0006567917_V10 **Escape System Inspection** Figure 2 (Sheet 2 of 3) EFFECTIVITY SOURCE **FWD ENTRY EMERGENCY ESCAPE SLIDE AKS ALL MRB** D633A109-AKS Page 18 of 19

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737-600/700/800/900 **TASK CARDS**

		IAS	K CARDS		
DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-240-00-01
	1.980 INCH (50.292 mm		0.581 (14.7)		
	E	scape S Figure 2	ystem Inspecti ? (Sheet 3 of 3	ion)	2162267 \$0000475279_V3
EFFEC' AKS	ALL	SOURCE MRB		MERGENCY ESCAPE	
			D633A109-AKS 25-240-00-01		Page 19 of 19 Jun 15/2015





737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	FWD GALLE	TITLE Y EMERGENCY ES		BOEING CARD NO. 25-240-00-02	
DATE	TASK RESTORE				RELATE	D CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD NOTE	REPEAT	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	NOTE			ALL	ALL
		ACCESS 841			ZONE 222	
					222	

Restore the fwd galley emergency escape slide at the manufacturer's recommended interval.

INTERVAL NOTE: Vendor Rec

A. Consumable Materials

Reference	Description	Specification
D00113	Lubricant - Solid Film, Liquid Dispersed	BMS3-8
D00670	Oil - Penetrating (For Loosening Frozen Metallic Parts)	A-A-50493
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper (Cheesecloth, Gauze)	BMS15-5 Class A

EFFECTIVITY AKS ALL	SOURCE MRB	FWD GALLEY EMERGENCY ESCAPE SLIDE	
		D633A109-AKS 25-240-00-02	Page 1 of 19 Jun 15/2015



737-600/700/800/900 TASK CARDS

	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-240-		
Esc	ape S	Slide	1-000-801 Pack and Cover Re	moval			,	MECH	IN
(Fig	ure 1))							
A.	Pre	pare 1	for the Removal						
	SUBTA	ASK 25-6	66-01-860-001						
	(1)	Mak	ce sure the door is sa	fe as follows:					
		(a)	Make sure the door	is closed and	l latched.				
		WA	RETAINER:	IF THE GIRT S, ACCIDENT HIS CAN CAU	BAR IS NOT AL INFLATION	LLED IN THE GIRT BAR INSTALLED IN THE GIR I OF THE ESCAPE SLID TO PERSONS OR DAN	RT BAR DE CAN		
		(b)	Make sure the girt I	oar [4] is insta	lled in the girt	bar retainers [5].			
В.	Ren	noval							
	SUBTA	ASK 25-6	66-01-020-001						
	(1)	Ren	nove the escape slide	e pack and co	ver [2] from the	e door as follows:			
		(a)	Remove the screws	s [9] and wash	ers [10] that a	ttach the cover [3] to the	door.		
			Make a note of installation.	of the screw [9] lengths and	locations for the subsequ	ient		
		(b)	Pull the top of the o	over [3] inboa	rd until it is cle	ear of the door structure.			
			the work so	•	ment is open fo	cape slide pack can export a minimum time to avo			
		WA	POUNDS (2 THE ESCA ESCAPE S	24.94 KILOGF PE SLIDE PAO LIDE PACK AI	RAMS). BE CA CK AND COVI ND COVER IN	ER WEIGH APPROXIMA REFUL WHEN YOU LIF ER. IF YOU LIFT OR MC ICORRECTLY, INJURIES ENT CAN OCCUR.	T OR MOVE VE THE		
		(c)	Lift and remove the [15] from the bottor	•	•	er [2] to disengage the lo	wer fitting		
		(d)	Put the escape slid backing pan [11] ag	•		ean horizontal surface w	ith the cover		
	SUBTA		66-01-860-006						
	(2)					the inflation cylinder valv	re [17]:		
		(a)	Remove the latch [-	_	-			
		(b)	Open the cover [3]	to get access	to the inflation	ı cylinder valve [17].			

AKS ALL

MRB

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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
WAR	REMOVE THE SLIDE OR SLI	COVER. \	WITHOUT THE CAN INFLATE A	ATION CYLINDER IF YOU SAFETY PIN, THE ESC ACCIDENTALLY. IF IT IN GE TO EQUIPMENT CA	APE FLATES,	MECH	INSF
(c)	Remove the safety pin	[19] from	the pocket [18].				
	Put the safety pin [19]			alve [17].			
		- END OF	TASK ——				
	CTIVITY S ALL	SOURCE MRB	FWD GALLEY E	EMERGENCY ESCAPE SL	IDE		
			FWD GALLEY E			age 3 (of ⁴



		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-240					
	TAS	K 25-	66-01-	-000-802					MECH	INSI			
2.	Esc	ape S	lide P	ack Removal									
	(Figure 1)												
	A.	Prep	oare fo	or the Removal									
		SUBTA	If the	o1-o10-oo1 escape slide pack and and Cover Removal, T			the door, do this task: E	scape Slide					
	В.	Rem	noval	,									
	٥.			-01-420-001									
		(1)	Remo	ove the escape slide pa	ack [26] fro	m the cover [3]	as follows:						
			(a)	Remove the girt bar [4] from the	girt bar retainers	s [5].						
			(b)	Remove the latch [7] for	rom the lat	ch bracket [13].							
			(c)	Open the cover [3] to g	get access	to the bonding	jumpers [12].						
					e compartr	ment is open for	ape slide pack can expa a minimum time to avo						
				Remove the pin [33], la [12] to the backing par			1] that attach the bondin	g jumpers					
	(e) Fully open the cover [3] to get access to the escape slide pack [26].												
			(f)	Remove the safety pin	[19] from	the pocket [18]	on the escape slide pac	k [26].					
			WAR	ESCAPE SLID	E CAN AC PE SLIDE	CIDENTALLY II CAN CAUSE II	FLATION CYLINDER VANFLATE. ACCIDENTAL NJURIES TO PERSON:	INFLATION					
			(g)	Put the safety pin [19]	into the inf	flation cylinder v	alve [17] as shown.						
			` '	Remove the escape sl horizontal surface.	slide pack [26] from the backing pan [11] and put it on a clean								
					ing pan. If i	•	ort box immediately afte , the slide will grow and						
			. ,	If required, remove the girt release strap [28].	_	9] and screw [2	7] to remove the latch [7] from the					
				NOTE: The latch [7] n	nay be nee	eded for a replac	cement escape slide.						
					- END OF	TASK ———							
				T1) (IT) (SOURCE	EWD GALLEY	EMERGENCY ESCAPE S	LIDE	1				
				S ALL	MRB	T WD GALLET I	IMIERGENCI ESCAPE S	LIDE					



737-600/700/800/900 TASK CARDS

	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
Es		Slide	1-400-801 Pack Installation					MECH	IN
`_			iou tha luatallation						
Α			for the Installation						
	(1)	Mak	6-01-860-002 te sure the cover [3] and king pan [11] against the			a clean horizontal surface.	e with the		
			6-01-860-004						
	(2)	Mak	e sure the cover [3] is or	oen.					
		NOT		tment is op		slide pack can expand. Find the expand the expanding the e			
	SUBT		6-01-600-001						
	(3)	Lub	ricate the latch assembly						
		(a)	Apply solid film lubricar	nt, D00113	to the escape	e slide latch.			
		(b)	Apply penetrating oil, D	000670 to	the pins [39].				
		(c)	Apply penetrating oil, E penetrates the pins.	000670 to	the pins [39] a	ngain to make sure that th	e oil fully		
		(d)	If there is too much per the unwanted oil from t	-		e a cotton wiper, G00034	to remove		
	SUBT	ASK 25-6	6-01-700-002						
	(4)	Do a	a check of the latch relea	se, Figure	2.				
		(a)	Move the latch block [3 move.	5] to the le	eft or right side	e until the latch block [35]	cannot		
			NOTE: The test starts continue to mo	-	pull the latch	block [35] to the side and	l it cannot		
		(b)	Use a spring force scal	e to pull th	ne latch chain	in the same direction.			
			1) Pull the latch cha	in or cable	in the same o	direction 10 ±5°arc (0.17	±0.09 rad).		
		(c)	Make sure that the late	h assemb	ly releases wit	th no more than 30 lbf (13	33 N).		
	SUBT	` ´ ASK 25-6	6-01-700-003		•	`	,		
	(5)	Do t	he steps for the check o	f the latch	release again				
		(a)	Move the latch block [3	5] in the o	pposite direct	ion.			
			NOTE: To complete the in the opposite		ease check, it	is necessary to do the ch	neck again		
	SUBT	ASK 25-6	6-01-860-005						
	WA	RNIN	ESCAPE SLIDE CAN	N ACCIDE	NTALLY INFL	TION CYLINDER VALVE, ATE. ACCIDENTAL INFL S TO PERSONS AND D	ATION OF		
	(6)	Mak	e sure the safety pin [19] is in the	inflation cylind	er valve [17].			
			ECTIVITY	SOURCE	ı	YEMERGENCY ESCAPE S			L

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737-600/700/800/900 TASK CARDS

[DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
В.	Insta	allatio	on of	the Escape Slide	Pack				MECH	INS
			6-01-420	-						
	(1)	Insta	all the	escape slide pac	k [26] in the	e cover [3] as fo	llows:			
		<u>NOT</u>	tr				immediately after reme grow and not fit into th			
		(a)	Put	the escape slide p	ack [26] in	the backing par	n [11].			
			NOT			•	n not be fitted into the must be installed.			
		(b)	Care	efully extend the g	irt [6].					
		(c)	Mak	e a "Z" fold [21] in	the girt [6]	as shown.				
		(d)		n and push togeth velcro strips [24] c			the bottom side of the galacter.	girt [6] with		
			NO	<u>ΓΕ</u> : The velcro str	ips should	be clean and fu	nction as intended.			
		(e)	_	n and push togeth ro strip [24] on the		ro strip [24] on tl	he girt retainer flap [20]	and the		
		(f)		quired, install the asse strap [28].	screw [27]	and locknut [29]	to attach the latch [7]	to the girt		
			1)	Make sure that t threads.	he screw [2	27] extends thro	ugh the locknut [29] by	1 to 2		
		(g)		te sure that there in the inflation cylin			ount of slack in the firin	g cable [30]		
		(h)	If the	ere is not enough	slack in the	e firing cable [30] (Figure 1), do these s	teps:		
			1)	Do not remove the	ne safety p	in [19].				
			2)	Make sure the sl	ide is corre	ectly installed in	the backing pan [11].			
			3)			0	correct. Make sure the lcro strip [24] on the gir	•		
			4)	Make sure that t [30] near the infl			m amount of slack in th gure 1).	e firing cable		
				a) Hold the fir slack from		30] at the inflatio	on cylinder valve [17] ar	nd pull the		
				dired remo mini	ction of the oved and the	inflation cylindene minimum am	e [30] away from the gir or valve [17] until all slad ount of slack is achieve , the escape slide pack	ck is ed. If the		
						in the firing cab ne inflation cyline	ole between the back of der valve [17].	fthe		
			S ALL		SOURCE MRB	FWD GALLEY E	EMERGENCY ESCAPE S	SLIDE		
						D633A109-AKS		Р	age 6	of ·

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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 25-240-		
WAR	THE SAFETY I VALVE, THE E	PIN IS REI SCAPE SL	MOVED FROM LIDE CAN ACCI	E ESCAPE SLIDE PAC THE INFLATION CYLII DENTALLY INFLATE. T AMAGE TO EQUIPMEN	NDER THIS CAN	MECH	INSP
(i)	Remove the safety pin	[19] from	the inflation cyli	nder valve [17].			
(j)	Put the safety pin [19]	into the po	cket [18] on the	escape slide pack [26]			
(k)	Close the cover [3].						
1	Make sure that you ca in the window [1]. If ne	-	•	gauge [16] of the inflati	ion cylinder		
	1) Fully open the co	ver [3] to g	get access to the	e escape slide pack [26].		
	Remove the safe the inflation cyline			t [18] and install safety լ	oin [19] in		
	3) Move the escape the window [1].	slide pack	([26] to get the	pressure gauge [16] in	the center of		
	4) Make sure that the cable [30] near the			nt of slack in the inflation 17] (Figure 1).	on firing		
	a) Hold the firi slack from t		30] at the inflation	on cylinder valve [17] ar	nd pull the		
	direc remo minir	tion of the ved and th	inflation cylinde ne minimum am	e [30] away from the girter valve [17] until all slace ount of slack is achieve , the escape slide pack	ck is d. If the		
			in the firing cat ne inflation cylin	ole between the back of der valve [17].	the		
	5) Remove the safe	ty pin [19]	from the inflatio	n cylinder valve [17].			
	6) Put the safety pir	n [19] into t	he pocket [18] o	on the escape slide pac	k [26].		
	Close the cover [3] suf pan [11].	ficiently to	connect the bo	nding jumpers [12] to th	e backing		
1	Install the stud [31], lacthe backing pan [11].	dder tie [32	?] and pin [33] to	attach the bonding jun	npers [12] to		
(0)	Close the cover [3].						
	Make sure the distance center of the girt bar [4]			e lower edge of the cov 12.0 in. (304.8 mm).	er [3] to the		
(p)	Align the holes in the la	atch brack	et [13] and latch	bracket [35] to install t	he latch [7].		
(r)	Install the latch [7] in the	ne latch bra	acket [13] and la	atch bracket [35] as sho	own.		
` '				and 'B' between the lato an 0.01 in. (0.25 mm) (F			
(t)	Install the girt bar [4] ir	the girt ba	ar retainers [5].				
		- END OF	TASK ——				
	CTIVITY 5 ALL	SOURCE MRB	FWD GALLEY	EMERGENCY ESCAPE S	LIDE		
			D633A109-AKS 25-240-00-02			age 7 e eb 15/	

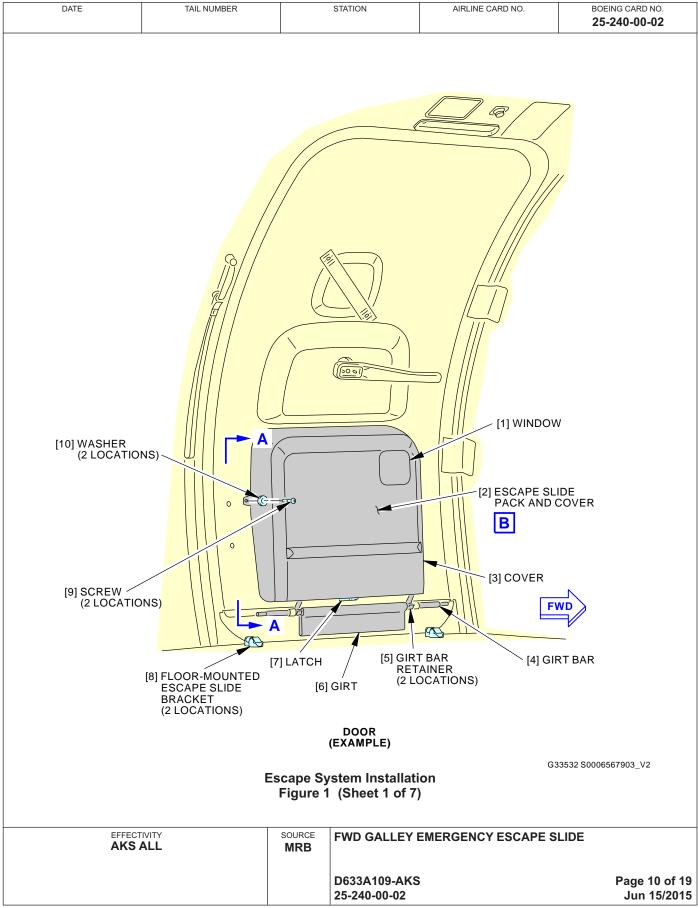


]	DATE		T/	AIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CA 25-240-		
	TAS	K 25	-66-01	-400-803					MECH	INSP
4.	Esc	ape S	Slide F	Pack and	Cover Installa	ation				
	(Fig	ure 1))							
	A.	Pre	oare f	or the Ins	tallation					
				6-01-410-004						
		(1)		•		is not installed in the cov 6-01-400-801.	ver [3], do this task: Esca	ape Slide		
		SUBTA	ASK 25-60	6-01-410-005						
		(2)	Mak	e sure the	door is closed	d and latched.				
				6-01-410-006						
		(3)		he followin der valve [ke sure that the safety pi	in [19] is not installed in	the inflation		
		(a) Put the escape slide pack and cover [2] on a clean flat horizontal surface with the backing pan [11] against the clean flat horizontal surface.					e with the			
			(b)	Remove	the latch [7] fr	om the latch bracket [13]				
			(c)	Open the	cover [3] suff	iciently to see the inflatio	n cylinder valve [17].			
			WAF	TI IN	HE ESCAPE : IFLATION CY	SAFETY PIN FROM THI SLIDE WILL NOT OPER LINDER VALVE. IF THE RSONS MAY BE INJURE	ATE IF THE SAFETY PI ESCAPE SLIDE DOES	N IS IN THE		
			(d)	Make sur	e a safety pin	[19] is not installed in the	e inflation cylinder valve	[17].		
				CAUTION	THE SAFE VALVE, Th	FUL WHEN YOU MOVE ETY PIN IS REMOVED F HE ESCAPE SLIDE CAN SE INJURIES TO PERSO NT.	ROM THE INFLATION (ACCIDENTALLY INFLA	CYLINDER		
				1) Ren	nove the safet	ty pin [19] from the inflation	on cylinder valve [17].			
				2) Put	the safety pin	[19] into the pocket [18]	on the escape slide pac	k [26].		
			(e)	Close the	cover [3].					
			(f)	Install the	atch [7] into	the latch bracket [13].				
	В.	Inst	allatio	on						
		SUBTA	ASK 25-60	6-01-420-004						
		(1)	Insta	all the esca	ipe slide pack	and cover [2] on the doo	or as follows:			
			WAF	TI K	HE ESCAPE : ILOGRAMS).	WHEN YOU LIFT THE ES SLIDE WEIGHS APPRO IF YOU ARE NOT CARE TO EQUIPMENT CAN (XIMATELY 55 POUNDS FUL, INJURIES TO PE	(24.94		
			(a)		escape slide	pack and cover [2] on the	e door.			

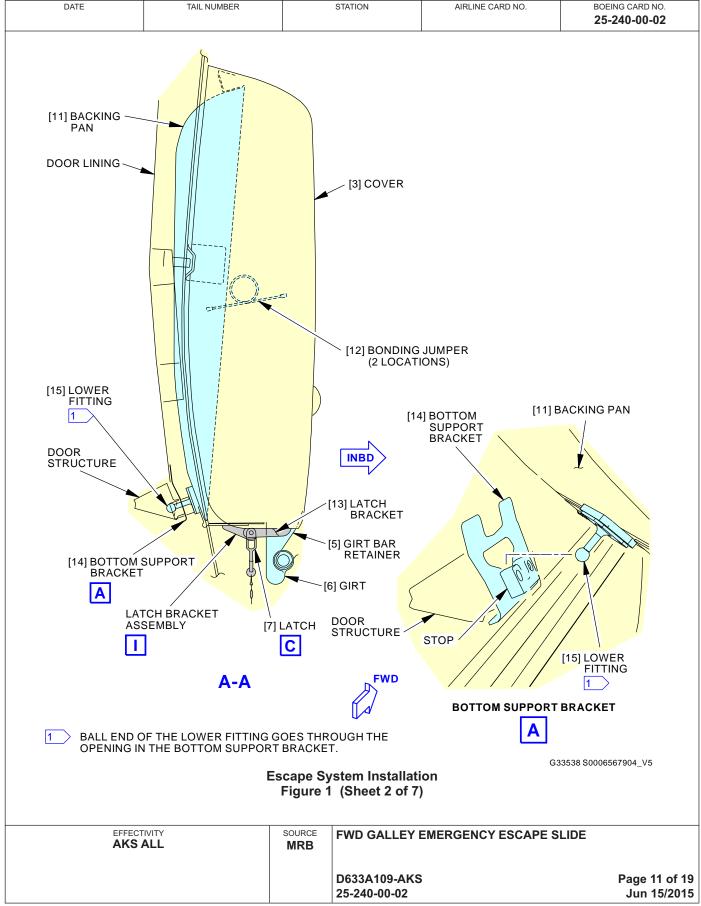


DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARE 25-240-00	
(b)	Push the top of the esca	ape slide pack and cover	[2] outboard until it is a	gainst the	ECH INSP
	NOTE: When the comp	partment is open, the esc compartment is open fo e escape slide pack.			
(c)		d washers [10] that attac	ch the cover [3] to the do	or.	
	Make sure to use procedure.	the correct length screw	[9], as noted in the remo	oval	
(d)	Make sure you can arm floor-mounted brackets	the escape slide by inst [8].	alling the girt bar [4] into	the	
(e)	Install the girt bar [4] into	o the girt bar retainer [5].			
	NOTE: The length of the	e girt bar for the FWD D	oor is 33.2 in. (84.3 cm)	long.	
	The length for the	he girt bar for the AFT De	oor is 29.2 in. (74.2 cm)	long.	
		END OF TASK ———			
	S ALL	SOURCE MRB FWD GALLEY	EMERGENCY ESCAPE S	LIDE	
		D633A109-AKS 25-240-00-02	3		e 9 of 19 : 15/2014
		20-2-0-02			. 5, 20 17

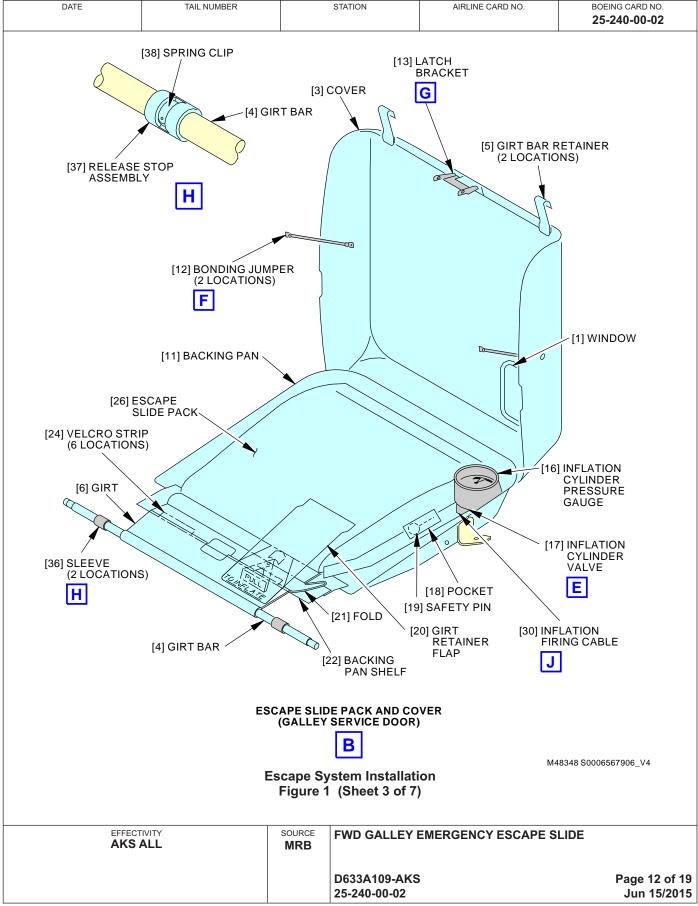




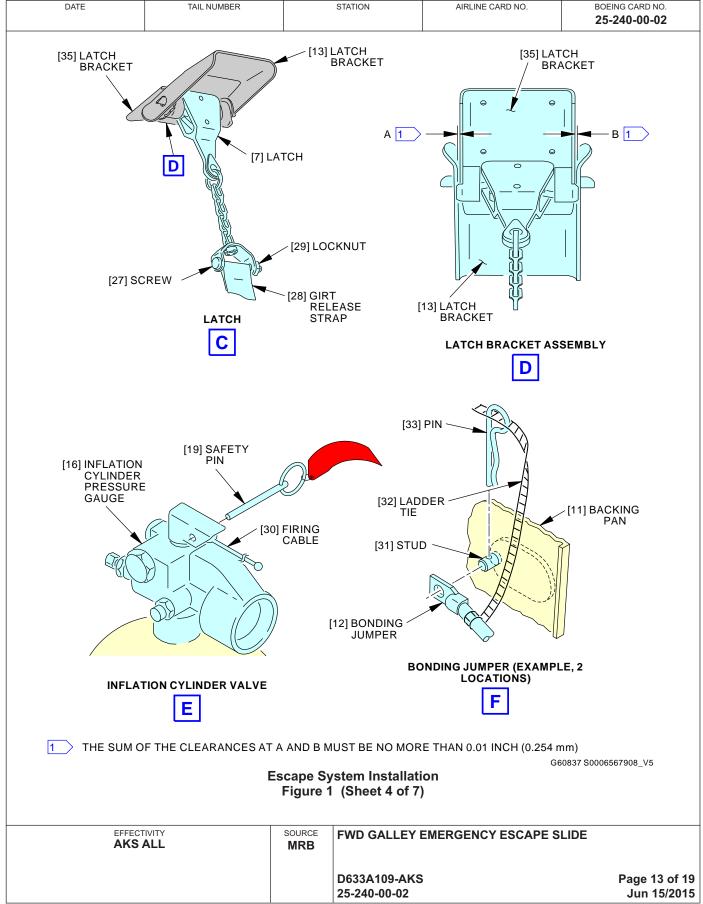














737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-02 [13] LATCH BRACKET [35] LATCH BRACKET STEP 2 (COCK LATCH) STEP 1 (ALIGN BRACKETS) G [13] LATCH [13] LATCH BRACKET **BRACKET** [35] LATCH BRACKET [35] LATCH **BRACKET** [7] LATCH STEP 4 (LATCH IN LOCKED POSITION) STEP 3 (POSITION LATCH AS SHOWN) G K62921 S0006567909_V2 **Escape System Installation** Figure 1 (Sheet 5 of 7) EFFECTIVITY SOURCE **FWD GALLEY EMERGENCY ESCAPE SLIDE AKS ALL MRB** D633A109-AKS Page 14 of 19 Jun 15/2015 25-240-00-02



737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-02 [35] LATCH BRACKET (PAN ASSEMBLY) 0 0 LATCH BLOCK **ASSEMBLY SPLIT RING** [35] LATCH BRACKET (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH CLEVIS LATCH BLOCK 2 3 10.0° ±5.0° 10.0° ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. USE A MAXIMUM OF 30 POUNDS OF FORCE TO RELEASE THE LATCH. 1739991 S0000315072_V4 **Escape System Installation** Figure 1 (Sheet 6 of 7) EFFECTIVITY SOURCE **FWD GALLEY EMERGENCY ESCAPE SLIDE AKS ALL MRB** D633A109-AKS Page 15 of 19 Jun 15/2015 25-240-00-02

TAIL NUMBER

DATE

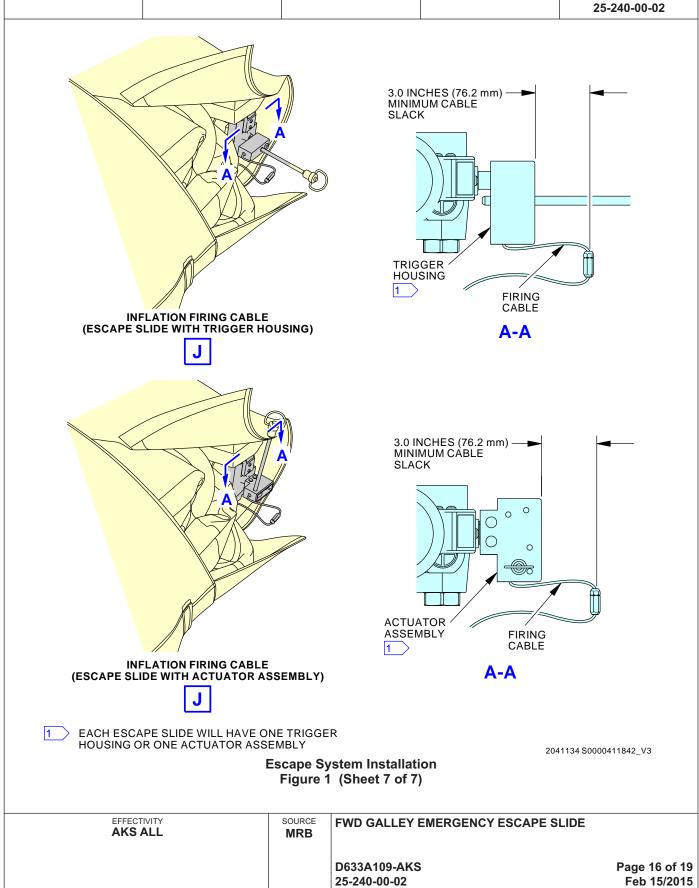


737-600/700/800/900 TASK CARDS

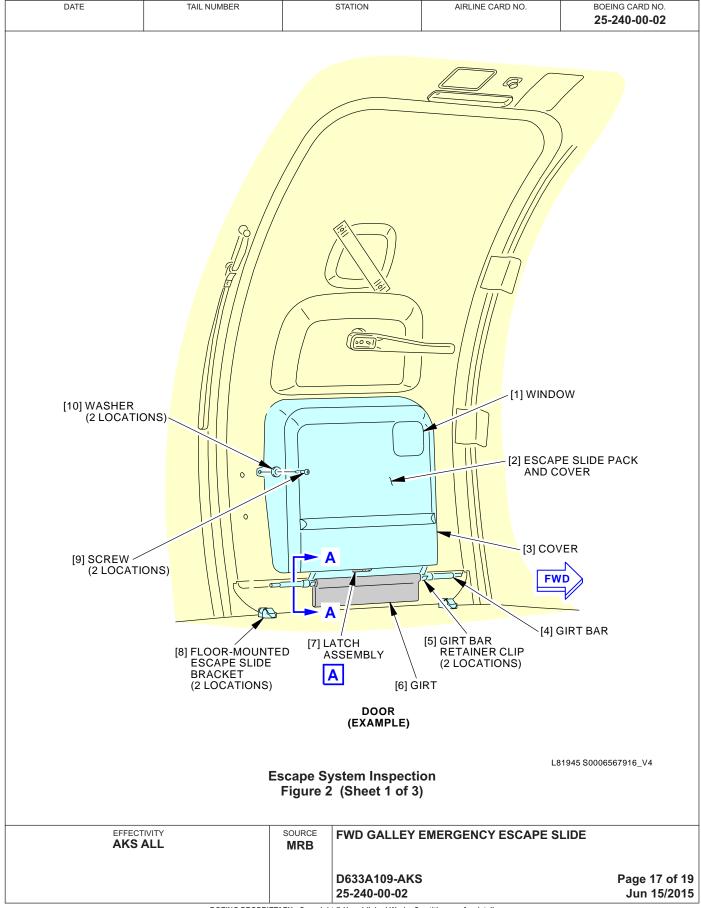
STATION

AIRLINE CARD NO.

BOEING CARD NO.









737-600/700/800/900 **TASK CARDS**

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-02 LATCH BRACKET 0 (PAN ASSEMBLY) 0 LATCH BLOCK **ASSEMBLY** SPLIT RING [35] LATCH BRACKET PIN (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH BLOCK LATCH CLEVIS 3 > 2 Ē 10.0° ±5.0 10.0°, ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. THE LATCH ASSEMBLY RELEASES AT 10 DEGREES, ± 5 DEGREES, WITH A MAXIMUM OF 30 LBF (133.45 NM). 3 M78780 S0006567917_V10 **Escape System Inspection** Figure 2 (Sheet 2 of 3) EFFECTIVITY SOURCE **FWD GALLEY EMERGENCY ESCAPE SLIDE AKS ALL MRB** D633A109-AKS Page 18 of 19 Jun 15/2015

25-240-00-02



737-600/700/800/900

		TASK CARDS		
DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-240-00-02
	1.980 INCH (50.292 mm)	0.5. (14		
	Esc F	cape System Inspectigure 2 (Sheet 3 of	ction	62267 \$0000475279_V3
EFFECTI AKS A		MRB	Y EMERGENCY ESCAPE S	
		D633A109-Al 25-240-00-02		Page 19 of 19 Jun 15/2015





737-600/700/800/900 TASK CARDS

AIRLINE	E CARD NO	AFT ENTRY	TITLE 'EMERGENCY ESC	CAPE SLIDE	BOEING 0 25-240	
DATE	TASK RESTORE				RELATE	D CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD NOTE	REPEAT	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	NOTE			ALL	ALL
		ACCESS			ZONE	
		834			241	

Restore the aft entry emergency escape slide at the manufacturer's recommended interval.

INTERVAL NOTE: Vendor Rec

A. Consumable Materials

Reference	Description	Specification
D00113	Lubricant - Solid Film, Liquid Dispersed	BMS3-8
D00670	Oil - Penetrating (For Loosening Frozen Metallic Parts)	A-A-50493
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper (Cheesecloth, Gauze)	BMS15-5 Class A

EFFECTIVITY AKS ALL	SOURCE MRB	AFT ENTRY EMERGENCY ESCAPE SLIDE	
		D633A109-AKS 25-240-00-03	Page 1 of 19 Jun 15/2015



737-600/700/800/900 TASK CARDS

	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	25-240-		
Esc		lide l	1-000-8 Pack a	801 and Cover Remo	val				MECH	INS
	,		iau tha	Demoval						
Α.				Removal						
	(1)		6-01-860-0	the door is safe a	e followe:					
	(1)	(a)		sure the door is		l latched				
		` ,								
		WAI	RNING	RETAINER. IF RETAINERS, A	THE GIRT	BAR IS NOT AL INFLATION	ILLED IN THE GIRT BAF INSTALLED IN THE GIF N OF THE ESCAPE SLIE TO PERSONS OR DAN	RT BAR DE CAN		
		(b)	Make	sure the girt bar	[4] is insta	lled in the girt	bar retainers [5].			
В.	Ren	noval								
	SUBTA	SK 25-6	6-01-020-0	001						
	(1)	Ren	nove th	ne escape slide pa	ack and co	ver [2] from th	e door as follows:			
		(a)	Remo	ove the screws [9] and wash	ers [10] that a	ttach the cover [3] to the	door.		
			•	Make a note of the installation.	ne screw [9] lengths and	locations for the subsequ	uent		
		(b)	Pull tl	he top of the cove	er [3] inboa	rd until it is cle	ear of the door structure.			
			NOTE		e compartr	nent is open f	cape slide pack can exp or a minimum time to avo			
		WAI	RNING	POUNDS (24.9 THE ESCAPE ESCAPE SLID	94 KILOGR SLIDE PA0 E PACK AI	AMS). BE CA CK AND COV ND COVER IN	ER WEIGH APPROXIMA REFUL WHEN YOU LIF ER. IF YOU LIFT OR MC ICORRECTLY, INJURIES ENT CAN OCCUR.	T OR MOVE OVE THE		
		(c)		nd remove the es rom the bottom s	•	•	er [2] to disengage the lo	wer fitting		
		(d)		ne escape slide p ing pan [11] again			ean horizontal surface w	ith the cover		
	SUBTA		6-01-860-0							
	(2)	Do t					the inflation cylinder valv	/e [17]:		
		(a)		ove the latch [7] fi		_	-			
		(b)	Open	n the cover [3] to (get access	to the inflatior	n cylinder valve [17].			

D633A109-AKS

25-240-00-03

Page 2 of 19 Oct 15/2014



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 25-240		
	WAR	REMOVE THE SLIDE OR SLI	COVER. V	WITHOUT THE CAN INFLATE A	ATION CYLINDER IF Y SAFETY PIN, THE ESC ACCIDENTALLY. IF IT IN GE TO EQUIPMENT CA	CAPE IFLATES,	MECH	INSF
	(c)	Remove the safety pir	1 [19] from	the pocket [18].				
		Put the safety pin [19]			alve [17].			
			- END OF	TASK ——				
	EFFEC AKS	TIVITY ALL	SOURCE MRB	AFT ENTRY EM	ERGENCY ESCAPE SLID)E		
				D633A109-AKS			age 3	
				25-240-00-03		(Oct 15/	201



	[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 25-240-		
2.				-000-802 ack Removal					MECH	INSF
		ure 1)								
	A.	Pre	oare fo	or the Removal						
		suвта (1)	If the	o1-010-001 escape slide pack and and Cover Removal, T			the door, do this task: E	scape Slide		
	B.	Ren	noval							
		SUBTA		01-420-001						
		(1)		ove the escape slide pa						
			` '	Remove the girt bar [4]		_				
			. ,	Remove the latch [7] fr						
			` '	Open the cover [3] to g		0.		. 5.		
					e comparti	ment is open for	ape slide pack can expa a minimum time to avo			
				Remove the pin [33], la [12] to the backing par			1] that attach the bondin	g jumpers		
			(e)	Fully open the cover [3] to get ac	cess to the esca	ape slide pack [26].			
			(f)	Remove the safety pin	[19] from	the pocket [18]	on the escape slide pac	k [26].		
			WAR	ESCAPE SLID	E CAN AC PE SLIDE	CIDENTALLY II CAN CAUSE II	FLATION CYLINDER VANFLATE. ACCIDENTAL NJURIES TO PERSON:	INFLATION		
			(g)	Put the safety pin [19]	into the inf	lation cylinder v	alve [17] as shown.			
			. ,	Remove the escape sl horizontal surface.	ide pack [2	26] from the bac	king pan [11] and put it	on a clean		
					ng pan. If i		ort box immediately afte , the slide will grow and			
			. ,	If required, remove the girt release strap [28].	locknut [2	9] and screw [2	7] to remove the latch [7] from the		
				NOTE: The latch [7] m	nay be nee	eded for a replac	cement escape slide.			
					- END OF	TASK ———				
			EFFEC AKS	ALL	SOURCE MRB	AFT ENTRY EM	ERGENCY ESCAPE SLI	DE		
						D633A109-AKS 25-240-00-03			age 4 e eb 15/	



	[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA		
								25-240-0		
				1-400-801				_	MECH	INSP
3.		_		Pack Installation						
	(Fig	ure 1)								
	Α.	_		for the Installation						
		(1)		6-01-860-002	l hacking n	an [11] are or	n a clean horizontal surfac	a with the		
		(1)		king pan [11] against the				e with the		
		SUBTA	ASK 25-6	6-01-860-004						
		(2)	Mak	e sure the cover [3] is o	pen.					
			NOT		rtment is or		e slide pack can expand. P mum time to avoid the exp			
		SUBTA		6-01-600-001						
		(3)	Lub	ricate the latch assembl	•					
			(a)	Apply solid film lubrica		-	e slide latch.			
			(b)	Apply penetrating oil, l						
			(c)	Apply penetrating oil, penetrates the pins.	D00670 to	the pins [39] a	again to make sure that th	e oil fully		
			(d)	If there is too much pe the unwanted oil from	-		se a cotton wiper, G00034	to remove		
		SUBTA	ASK 25-6	6-01-700-002						
		(4)	Do a	a check of the latch rele	ase, Figure	2.				
			(a)	Move the latch block [move.	35] to the le	eft or right sid	e until the latch block [35]	cannot		
				NOTE: The test starts continue to me	-	pull the latch	block [35] to the side and	it cannot		
			(b)	Use a spring force sca	ale to pull th	ne latch chain	in the same direction.			
				1) Pull the latch cha	ain or cable	in the same	direction 10 ±5°arc (0.17 ±	:0.09 rad).		
			(c)	Make sure that the late	ch assemb	ly releases wi	th no more than 30 lbf (13	3 N).		
		SUBTA	ASK 25-6	6-01-700-003						
		(5)	Do t	he steps for the check of	of the latch	release agair	٦.			
			(a)	Move the latch block [35] in the o	pposite direct	tion.			
				NOTE: To complete the in the opposite		ease check, i	t is necessary to do the ch	eck again		
		SUBTA	ASK 25-6	6-01-860-005						
		WAI	RNIN	ESCAPE SLIDE CA	N ACCIDE	NTALLY INFL	TION CYLINDER VALVE, LATE. ACCIDENTAL INFLA ES TO PERSONS AND DA	ATION OF		
		(6)	Mak	e sure the safety pin [1	9] is in the	inflation cylind	der valve [17].			
				ECTIVITY	SOURCE	I	EMERGENCY ESCAPE SLIC			



737-600/700/800/900 TASK CARDS

L	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
В.	Insta	allatio	on of	the Escape Slide	Pack				MECH	INS
			6-01-420	-						
	(1)	Insta	all the	escape slide pac	k [26] in the	e cover [3] as fo	llows:			
		<u>NOT</u>	tr				immediately after rem- grow and not fit into th			
		(a)	Put	the escape slide p	ack [26] in	the backing par	n [11].			
			NOT		•	•	n not be fitted into the must be installed.			
		(b)	Care	efully extend the g	irt [6].					
		(c)	Mak	ce a "Z" fold [21] ir	the girt [6]	as shown.				
		(d)		n and push togeth velcro strips [24] c			the bottom side of the 2].	girt [6] with		
			NO	<u>ΓΕ</u> : The velcro str	ips should	be clean and fu	nction as intended.			
		(e)	_	n and push togeth ro strip [24] on the		o strip [24] on tl	ne girt retainer flap [20]	and the		
		(f)		quired, install the ase strap [28].	screw [27] a	and locknut [29]	to attach the latch [7]	to the girt		
			1)	Make sure that t threads.	he screw [2	27] extends thro	ugh the locknut [29] by	1 to 2		
		(g)		te sure that there in the inflation cylin			ount of slack in the firin	ng cable [30]		
		(h)	If the	ere is not enough	slack in the	firing cable [30] (Figure 1), do these s	teps:		
			1)	Do not remove t	he safety pi	in [19].				
			2)	Make sure the s	lide is corre	ectly installed in	the backing pan [11].			
			3)			0	correct. Make sure the cro strip [24] on the gir	•		
			4)	Make sure that t [30] near the infl			ກ amount of slack in th gure 1).	e firing cable		
				a) Hold the fir slack from	-	30] at the inflatio	n cylinder valve [17] ar	nd pull the		
				dire rem mini	ction of the oved and th	inflation cylinde ne minimum am	e [30] away from the gir r valve [17] until all sla ount of slack is achieve , the escape slide pack	ck is ed. If the		
						in the firing cab ne inflation cyline	ole between the back of der valve [17].	f the		
			S ALL		SOURCE MRB	AFT ENTRY EM	ERGENCY ESCAPE SL	IDE		
						D633A109-AKS		Р	age 6	of

25-240-00-03

Feb 15/2015



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C.		
WA	T \	THE SAFETY P /ALVE, THE ES	IN IS REI CAPE SL	MOVED FROM LIDE CAN ACCI	E ESCAPE SLIDE PAC THE INFLATION CYLII DENTALLY INFLATE. 1 AMAGE TO EQUIPMEN	NDER THIS CAN	MECH	INSP
(i)	Remove	the safety pin	[19] from	the inflation cyli	nder valve [17].			
(j)	Put the s	safety pin [19] ir	nto the po	cket [18] on the	escape slide pack [26]			
(k)	Close th	e cover [3].						
(1)		•	-	ee the pressure djust as follows:	gauge [16] of the inflati	on cylinder		
	1) Ful	lly open the cov	er [3] to g	get access to the	e escape slide pack [26].		
		move the safety inflation cylind			: [18] and install safety բ	oin [19] in		
	,	ve the escape a window [1].	slide pack	([26] to get the	pressure gauge [16] in	the center of		
	,			minimum amou cylinder valve [nt of slack in the inflatio 17] (Figure 1).	n firing		
	a)	Hold the firin slack from th		30] at the inflation	on cylinder valve [17] an	d pull the		
		directi remov minim	ion of the /ed and th	inflation cylinde ne minimum am	e [30] away from the girt er valve [17] until all slac ount of slack is achieve , the escape slide pack	ck is d. If the		
				in the firing cab ne inflation cylin	ole between the back of der valve [17].	the		
	5) Re	move the safety	y pin [19]	from the inflatio	n cylinder valve [17].			
	6) Pu	t the safety pin	[19] into t	he pocket [18] o	on the escape slide pac	k [26].		
(m)	Close th pan [11].		ciently to	connect the bor	nding jumpers [12] to th	e backing		
(n)		e stud [31], ladd ing pan [11].	der tie [32	?] and pin [33] to	attach the bonding jun	npers [12] to		
(o)	Close th	e cover [3].						
(p)					e lower edge of the cov 12.0 in. (304.8 mm).	er [3] to the		
(q)	Align the	holes in the la	tch brack	et [13] and latch	bracket [35] to install t	he latch [7].		
(r)	Install th	e latch [7] in the	e latch bra	acket [13] and la	atch bracket [35] as sho	wn.		
(s)					and 'B' between the latc an 0.01 in. (0.25 mm) (F			
(t)	Install th	e girt bar [4] in	the girt ba	ar retainers [5].				
			END OF	TASK ———				
	FECTIVITY KS ALL		SOURCE MRB	AFT ENTRY EM	ERGENCY ESCAPE SLI	DE	<u>ı </u>	
				D633A109-AKS 25-240-00-03			age 7 d eb 15/2	
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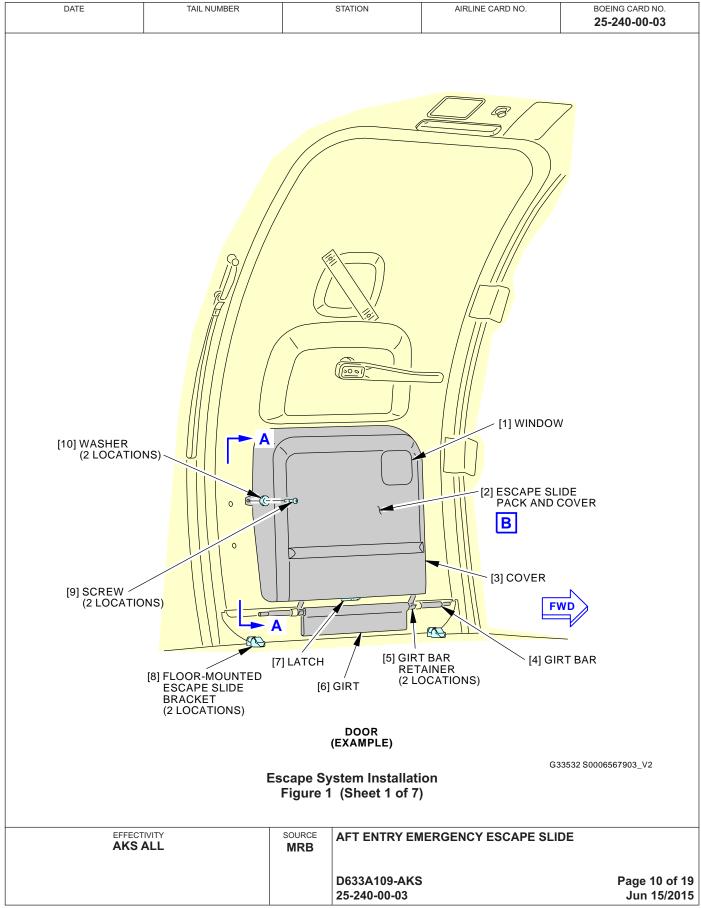


	I	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 25-240-		
	TAS	K 25	-66-0	1-400·	-803			1		MECH	INS
4.	Esc	ape S	Slide	Pack	and Cover Installa	ition					
	(Fig	ure 1))								
	A.	Pre	pare 1	for the	e Installation						
		SUBTA	ASK 25-6	6-01-410	-004						
		(1)			ape slide pack [26] i allation, TASK 25-60			ver [3], do this task: Esca	pe Slide		
		SUBTA	ASK 25-6	6-01-410	-005						
		(2)	Mak	e sure	e the door is closed	and latch	ned.				
				6-01-410							
		(3)			lowing steps to mal alve [17]:	ke sure th	at the safety p	in [19] is not installed in t	the inflation		
			(a)		the escape slide pa king pan [11] agains			ean flat horizontal surfac al surface.	e with the		
			(b)	Rem	nove the latch [7] fro	om the lat	ch bracket [13]	.			
			(c)	Ope	n the cover [3] suffi	ciently to	see the inflatio	n cylinder valve [17].			
			(d)	Mak	INFLATION CYL OPERATE, PER	LINDER V RSONS M	/ALVE. IF THE IAY BE INJURE	ATE IF THE SAFETY PII ESCAPE SLIDE DOES ED. e inflation cylinder valve	NOT		
			(-)		JTION: BE CAREF THE SAFE VALVE, TH	TUL WHEI TY PIN IS E ESCAF SE INJUR	N YOU MOVE S REMOVED F PE SLIDE CAN	THE ESCAPE SLIDE PAROM THE INFLATION OF ACCIDENTALLY INFLATIONS OR DAMAGE TO	ACK. WHEN CYLINDER		
				1)	Remove the safety	y pin [19]	from the inflati	on cylinder valve [17].			
				2)	Put the safety pin	[19] into t	he pocket [18]	on the escape slide pac	k [26].		
			(e)	Clos	se the cover [3].						
			(f)	Insta	all the latch [7] into t	the latch l	oracket [13].				
	В.	Inst	allati	on							
		SUBTA	ASK 25-6	6-01-420	-004						
		(1)	Inst	all the	escape slide pack	and cove	r [2] on the doo	or as follows:			
			WA	RNING	THE ESCAPE S	SLIDE WE	EIGHS APPRO RE NOT CARE	SCAPE SLIDE PACK AN XIMATELY 55 POUNDS FUL, INJURIES TO PEF DCCUR.	(24.94		
			(a)	Insta	all the escape slide	pack and	cover [2] on th	e door.			
				1)	Make sure that the the bottom suppor			ting [15] goes through th	e opening in		
			EEE	ECTIVITY	,	SOURCE	ACT CNTDV C	MERGENCY ESCAPE SLI	DE	-	

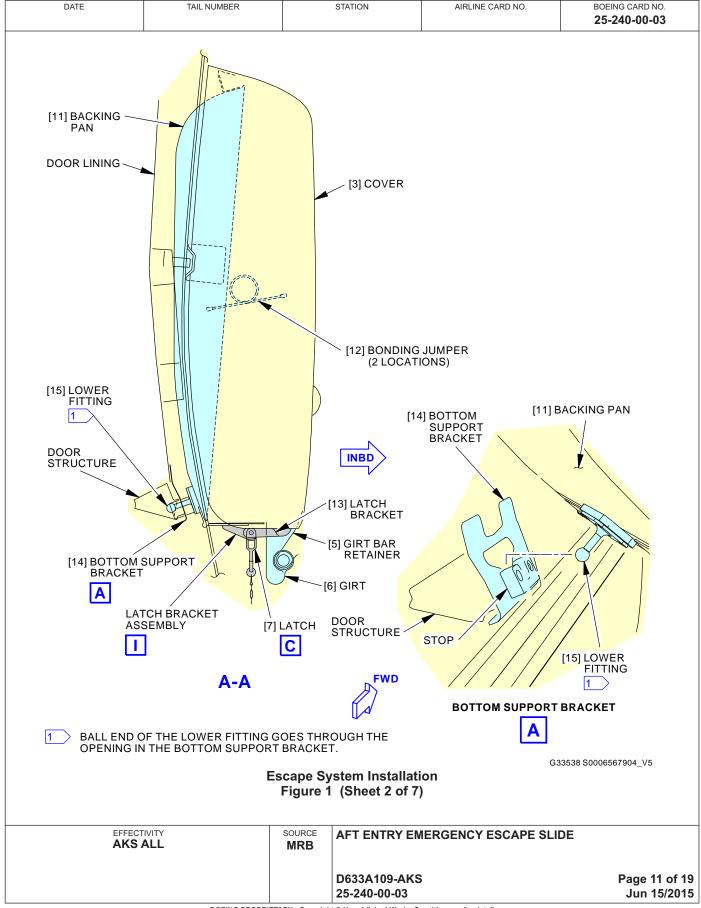


DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	ROEINO	CARD NO.	
	IAIL NOWDER	STATION	AINLINE CAND NO.)-00-03	
(b)	Push the top of the eso	cape slide pack and	cover [2] outboard until it is a	gainst the	MECH	INSI
	NOTE: When the come the work so the		e escape slide pack can exp en for a minimum time to avo			
(c)	·		attach the cover [3] to the do	oor.		
`,			crew [9], as noted in the rem			
(d)	Make sure you can arr floor-mounted brackets		installing the girt bar [4] into	the		
(e)	Install the girt bar [4] in	nto the girt bar retaine	er [5].			
	NOTE: The length of t	he girt bar for the FV	VD Door is 33.2 in. (84.3 cm)	long.		
	The length for	the girt bar for the A	FT Door is 29.2 in. (74.2 cm)	long.		
		- END OF TASK —				
	ECTIVITY S ALL	SOURCE AFT ENTE	RY EMERGENCY ESCAPE SLI	DE		
				_	_	_
		D633A109 25-240-00			Page 9 Oct 15/	
		23-240-00	-03		JUL 13/	4 0

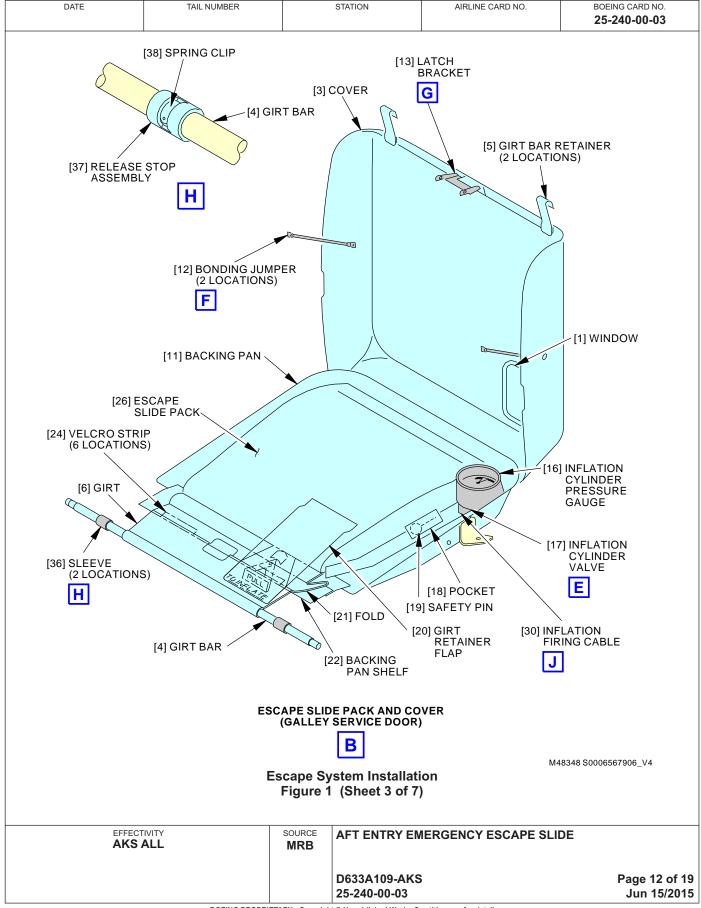




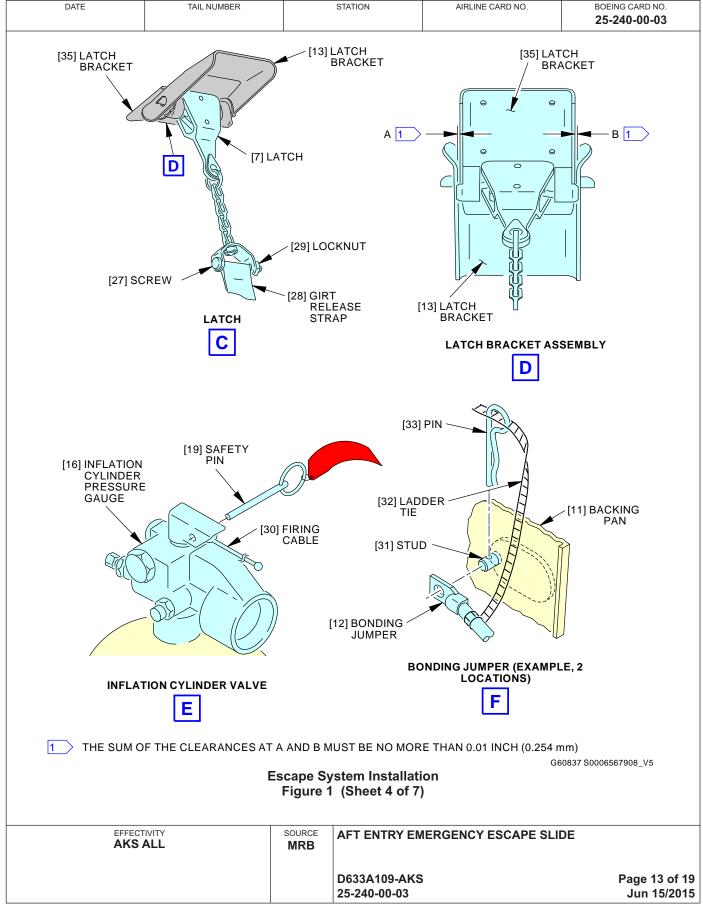














737-600/700/800/900 TASK CARDS

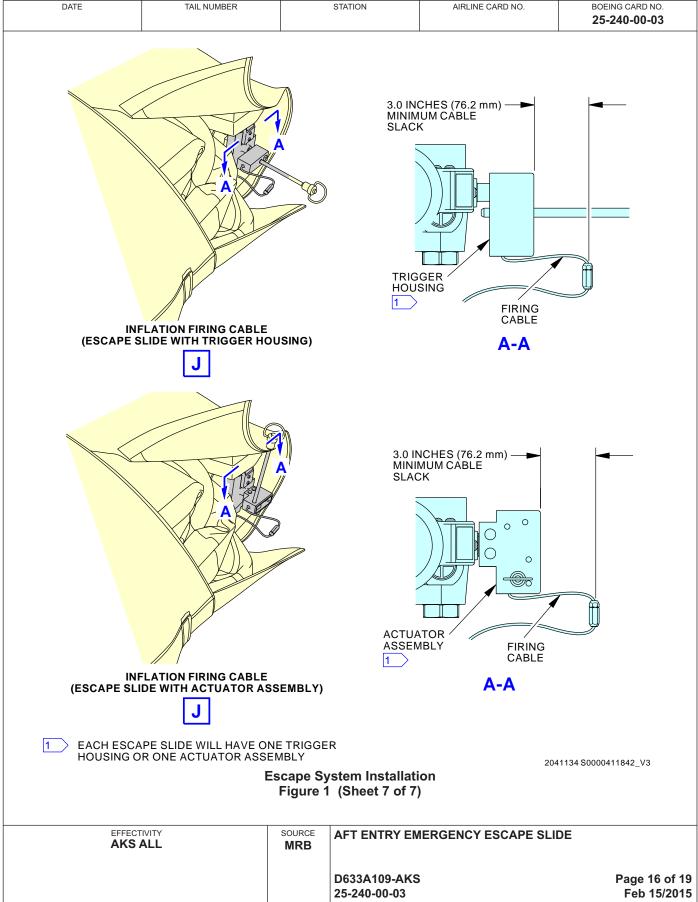
DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-03 [13] LATCH BRACKET [35] LATCH BRACKET STEP 2 (COCK LATCH) STEP 1 (ALIGN BRACKETS) G [13] LATCH [13] LATCH BRACKET **BRACKET** [35] LATCH BRACKET [35] LATCH **BRACKET** [7] LATCH STEP 4 (LATCH IN LOCKED POSITION) STEP 3 (POSITION LATCH AS SHOWN) G K62921 S0006567909_V2 **Escape System Installation** Figure 1 (Sheet 5 of 7) EFFECTIVITY SOURCE AFT ENTRY EMERGENCY ESCAPE SLIDE **AKS ALL MRB** D633A109-AKS Page 14 of 19 Jun 15/2015 25-240-00-03



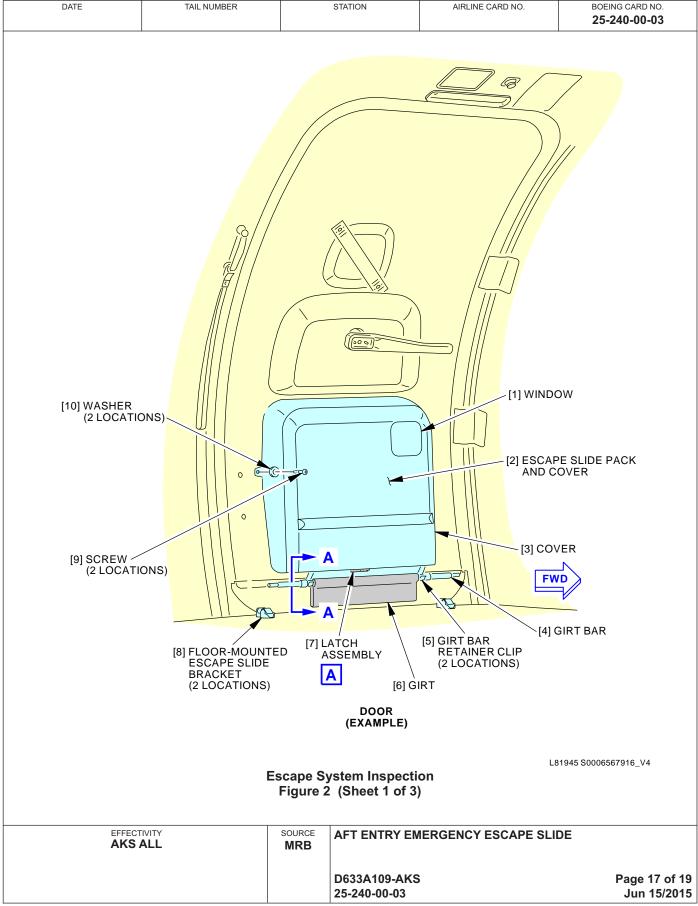
737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-03 [35] LATCH BRACKET (PAN ASSEMBLY) 0 0 LATCH BLOCK **ASSEMBLY SPLIT RING** [35] LATCH BRACKET [39] PIN (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH CLEVIS LATCH BLOCK 2 3 10.0° ±5.0° 10.0° ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. USE A MAXIMUM OF 30 POUNDS OF FORCE TO RELEASE THE LATCH. 1739991 S0000315072_V4 **Escape System Installation** Figure 1 (Sheet 6 of 7) EFFECTIVITY SOURCE AFT ENTRY EMERGENCY ESCAPE SLIDE **AKS ALL MRB** D633A109-AKS Page 15 of 19 Jun 15/2015 25-240-00-03











737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-03 LATCH BRACKET 0 (PAN ASSEMBLY) 0 LATCH BLOCK **ASSEMBLY** SPLIT RING [35] LATCH BRACKET PIN (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH BLOCK LATCH CLEVIS 3 > 2 Ē 10.0° ±5.0 10.0°, ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. THE LATCH ASSEMBLY RELEASES AT 10 DEGREES, ± 5 DEGREES, WITH A MAXIMUM OF 30 LBF (133.45 NM). 3 M78780 S0006567917_V10 **Escape System Inspection** Figure 2 (Sheet 2 of 3) EFFECTIVITY SOURCE AFT ENTRY EMERGENCY ESCAPE SLIDE **AKS ALL MRB** D633A109-AKS Page 18 of 19 Jun 15/2015 25-240-00-03



737-600/700/800/900 TASK CARDS

		TAS	K CARDS		
DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-240-00-03
	1.980 INCH (50.292 mm)		0.690 ±0 (17.526 0.581 (14.75 0.690	410 INCH	
	E:	scape Sy Figure 2	rstem Inspection (Sheet 3 of 3)	on	2162267 \$0000475279_V3
EFFECT AKS A	IVITY ALL	SOURCE MRB	AFT ENTRY EM	ERGENCY ESCAPE	SLIDE

D633A109-AKS

25-240-00-03

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737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	AFT GALLE	TITLE Y EMERGENCY ES	BOEING CARD NO. 25-240-00-04		
DATE	TASK RESTORE				RELATE	D CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD NOTE	REPEAT	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	NOTE			ALL	ALL
		ACCESS			ZONE	
		844			242	

Restore the aft galley emergency escape slide at the manufacturer's recommended interval.

INTERVAL NOTE: Vendor Rec

A. Consumable Materials

Reference	Description	Specification
D00113	Lubricant - Solid Film, Liquid Dispersed	BMS3-8
D00670	Oil - Penetrating (For Loosening Frozen Metallic Parts)	A-A-50493
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper (Cheesecloth, Gauze)	BMS15-5 Class A

EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY EMERGENCY ESCAPE SLIDE	
		D633A109-AKS 25-240-00-04	Page 1 of 19 Jun 15/2015



737-600/700/800/900 TASK CARDS

L	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	25-240-			
TAS	K 25	66-01	-000	801					MECH	IN	
	_		Pack	and Cover Remov	<u>al</u>						
(Figi	ure 1)										
A.	Pre	oare f	or the	e Removal							
			6-01-860-								
	(1)			e the door is safe as							
		(a)	Mak	e sure the door is c	losed and	l latched.					
		WAI	RNINC	RETAINER. IF 1 RETAINERS, A	THE GIRT	BAR IS NOT AL INFLATIO	ALLED IN THE GIRT BAF INSTALLED IN THE GIF N OF THE ESCAPE SLIE TO PERSONS OR DAM	RT BAR DE CAN			
		(b)	Mak	e sure the girt bar [4] is insta	lled in the girt	bar retainers [5].				
В.	Ren	noval									
	SUBTA	SK 25-6	6-01-020-	001							
	(1) Remove the escape slide pack and cover [2] from the door as follows:										
	(a) Remove the screws [9] and washers [10] that attach the cover [3] to the door.										
			1)	Make a note of the installation.	e screw [9] lengths and	locations for the subsequ	uent			
		(b)	Pull	the top of the cover	[3] inboa	rd until it is cl	ear of the door structure.				
			NOT		compartr	nent is open f	scape slide pack can exp or a minimum time to avo				
		WAI	RNING	POUNDS (24.94 THE ESCAPE S ESCAPE SLIDE	4 KILOGR SLIDE PAC E PACK AI	AMS). BE CA CK AND COV ND COVER IN	ER WEIGH APPROXIMA AREFUL WHEN YOU LIF ER. IF YOU LIFT OR MC NCORRECTLY, INJURIES ENT CAN OCCUR.	T OR MOVE OVE THE			
		(c)		and remove the esc from the bottom su			er [2] to disengage the lo	wer fitting			
		(d)		he escape slide pa ing pan [11] agains			lean horizontal surface w	ith the cover			
	SUBTA		6-01-860-								
	(2)						the inflation cylinder valv	/e [17]:			
		(a)		ove the latch [7] fro		-	-				
		(b)	Ope	n the cover [3] to go	et access	to the inflation	n cylinder valve [17].				
									1	_	

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25-240-00-04

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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
WAR	SLIDE OR SLID	COVER. \ E-RAFT (WITHOUT THE CAN INFLATE A	ATION CYLINDER IF Y SAFETY PIN, THE ESC ACCIDENTALLY. IF IT IN GE TO EQUIPMENT C	CAPE NFLATES,	MECH	INSP
(c)	Remove the safety pin	[19] from	the pocket [18].				
(d)	Put the safety pin [19] in	nto the inf	lation cylinder v	alve [17].			
		END OF	TASK ——				
EFFEC AKS	CTIVITY S ALL	SOURCE MRB	AFT GALLEY E	MERGENCY ESCAPE SL	.IDE		
ANG	, ALL	IVIKD					
			D633A109-AKS 25-240-00-04			age 3 o Oct 15/2	
<u> </u>			I.				



	[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-240 -						
2.				-000-802 ack Removal	-				MECH	INSI				
	(Fig	ure 1)												
	A.	Prep	oare fo	re for the Removal										
		SUBTA (1)	Pack and Cover Removal, TASK 25-66-01-000-801.											
	B.	Rem	noval											
				01-420-001			6.11							
		(1)		·	ove the escape slide pack [26] from the cover [3] as follows:									
			. ,	emove the girt bar [4] from the girt bar retainers [5].										
			. ,		emove the latch [7] from the latch bracket [13]. ben the cover [3] to get access to the bonding jumpers [12].									
			` ,			0.		and Dian						
					e comparti	ment is open for	ape slide pack can expa a minimum time to avo							
				Remove the pin [33], la [12] to the backing par			l] that attach the bondin	g jumpers						
			(e)	Fully open the cover [3] to get access to the escape slide pack [26].										
			(f)	(f) Remove the safety pin [19] from the pocket [18] on the escape slide pack [26].										
		WARNING: IF THE SAFETY PIN IS NOT IN THE INFLATION CYLINDER VALVE, THE ESCAPE SLIDE CAN ACCIDENTALLY INFLATE. ACCIDENTAL INFLATION OF THE ESCAPE SLIDE CAN CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.												
			(g)	Put the safety pin [19]	into the inf	lation cylinder v	alve [17] as shown.							
			. ,	Remove the escape sl horizontal surface.	ide pack [2	26] from the bac	king pan [11] and put it	on a clean						
					ng pan. If i		ort box immediately afte , the slide will grow and							
			. ,	If required, remove the girt release strap [28].	locknut [2	9] and screw [2	7] to remove the latch [7] from the						
				NOTE: The latch [7] m	nay be nee	eded for a replac	cement escape slide.							
					- END OF	TASK ———								
				CTIVITY ALL	SOURCE MRB	AFT GALLEY E	MERGENCY ESCAPE SL	IDE	1					
						D633A109-AKS 25-240-00-04			age 4 eb 15/					



	[DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA		
									MECH	INSP
3.		_		Pack Installation						
	(Fig	ure 1))							
	A.	Pre	oare 1	for the Installation						
					م بعدادات م	an [44] ana an				
		(1)						e with the		
		(2)			•					
			NO	work so the compa	rtment is or	•				
		SK 25-66-01-400-801 cape Slide Pack Installation pure 1) Prepare for the Installation subtask 25-66-1-800-002 (1) Make sure the cover [3] and backing pan [11] are on a clean horizontal surface with the backing pan [11] against the clean horizontal surface. subtask 25-66-1-800-004 (2) Make sure the cover [3] is open. NOTE: When the compartment is open, the escape slide pack can expand. Plan the work so the compartment is open for a minimum time to avoid the expansion of the escape slide pack. subtask 25-66-1-800-001 (3) Lubricate the latch assembly. (a) Apply solid film lubricant, D00113 to the escape slide latch. (b) Apply penetrating oil, D00670 to the pins [39]. (c) Apply penetrating oil, D00670 to the pins [39] again to make sure that the oil fully penetrates the pins. (d) If there is too much penetrating oil, D00670, use a cotton wiper, G00034 to remove the unwanted oil from the latch assembly. subtask 25-66-1-700-002 (4) Do a check of the latch release, Figure 2. (a) Move the latch block [35] to the left or right side until the latch block [35] cannot move. NOTE: The test starts when you pull the latch block [35] to the side and it cannot continue to move. (b) Use a spring force scale to pull the latch chain in the same direction. 1) Pull the latch chain or cable in the same direction 10 ±5°arc (0.17 ±0.09 rad). (c) Make sure that the latch assembly releases with no more than 30 lbf (133 N). subtask 25-66-1-700-003 (5) Do the steps for the check of the latch release again. (a) Move the latch block [35] in the opposite direction. NOTE: To complete the latch release acheck, it is necessary to do the check again in the opposite direction.								
		(3)	Lubi							
						-	e slide latch.			
		ASK 25-66-01-400-801 scape Slide Pack Installation igure 1) Prepare for the Installation subtrask 25-66-01-800-002 (1) Make sure the cover [3] and backing pan [11] are on a clean horizontal surface with the backing pan [11] against the clean horizontal surface. SUBTASK 25-66-01-800-004 (2) Make sure the cover [3] is open. NOTE: When the compartment is open, the escape slide pack can expand. Plan the work so the compartment is open for a minimum time to avoid the expansion of the escape slide pack. SUBTASK 25-60-01-00-001 (3) Lubricate the latch assembly. (a) Apply solid film lubricant, D00113 to the escape slide latch. (b) Apply penetrating oil, D00670 to the pins [39]. (c) Apply penetrating oil, D00670 to the pins [39] again to make sure that the oil fully penetrates the pins. (d) If there is too much penetrating oil, D00670, use a cotton wiper, G00034 to remove the unwanted oil from the latch assembly. SUBTASK 25-66-01-700-002 (4) Do a check of the latch release, Figure 2. (a) Move the latch block [35] to the left or right side until the latch block [35] cannot move. NOTE: The test starts when you pull the latch block [35] to the side and it cannot continue to move. (b) Use a spring force scale to pull the latch chain in the same direction. 1) Pull the latch chain or cable in the same direction 10 ±5° arc (0.17 ±0.09 rad). (c) Make sure that the latch assembly releases with no more than 30 lbf (133 N). SUBTASK 25-69-01-700-003 (5) Do the steps for the check of the latch release again. (a) Move the latch block [35] in the opposite direction. NOTE: To complete the latch release check, it is necessary to do the check again in the opposite direction.								
			(c)		D00670 to	the pins [39] a	again to make sure that the	e oil fully		
			(d)	-	-		se a cotton wiper, G00034	to remove		
		SUBTA	ASK 25-6	6-01-700-002						
		(4)	Do a	a check of the latch rele	ase, Figure	2.				
		subtask 25-66-01-700-002 (4) Do a check of the latch r (a) Move the latch block	-	35] to the le	eft or right sid	e until the latch block [35]	cannot			
					-	pull the latch	block [35] to the side and	it cannot		
			(b)	Use a spring force sca	ale to pull th	ne latch chain	in the same direction.			
				1) Pull the latch cha	ain or cable	in the same	direction 10 ±5°arc (0.17 ±	:0.09 rad).		
			(c)	Make sure that the lat	ch assemb	ly releases wi	ith no more than 30 lbf (13	3 N).		
		SUBTA	ASK 25-6	6-01-700-003						
		(5)	Do t	he steps for the check	of the latch	release agair	١.			
			(a)	Move the latch block [35] in the c	pposite direct	tion.			
							t is necessary to do the ch	eck again		
		WAI	RNIN	ESCAPE SLIDE CA THE ESCAPE SLID	N ACCIDE	NTALLY INFL	ATE. ACCIDENTAL INFLA	ATION OF		
		(6)	Mak	e sure the safety pin [1	9] is in the	inflation cylind	der valve [17].			
				=OTN (IT) (2011205	T				



737-600/700/800/900 TASK CARDS

[DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-240		
В.	Insta	allatio	on of	the Escape Slide	Pack				MECH	INS
			6-01-420	-						
	(1)	Insta	all the	escape slide pac	k [26] in the	e cover [3] as fo	llows:			
		<u>NOT</u>	tr				immediately after reme grow and not fit into th			
		(a)	Put	the escape slide p	ack [26] in	the backing par	n [11].			
			NOT	OTE: If the escape slide pack expands and can not be fitted into the compartment, a new escape slide pack must be installed.						
		(b)	Care	efully extend the g	irt [6].					
		(c)	Mak	ce a "Z" fold [21] ir	the girt [6]	as shown.				
		(d)		n and push togeth velcro strips [24] c			the bottom side of the galacter.	girt [6] with		
			NO	<u>ΓΕ</u> : The velcro str	ips should	be clean and fu	nction as intended.			
		(e)	_	n and push togeth ro strip [24] on the		ro strip [24] on tl	he girt retainer flap [20]	and the		
		(f)		quired, install the ase strap [28].	screw [27] a	and locknut [29]	to attach the latch [7]	to the girt		
			1)	Make sure that t threads.	he screw [2	27] extends thro	ugh the locknut [29] by	1 to 2		
		(g)		te sure that there in the inflation cylin			ount of slack in the firin	g cable [30]		
		(h)	If the	ere is not enough	slack in the	e firing cable [30] (Figure 1), do these s	teps:		
			1)	Do not remove t	he safety pi	in [19].				
			2)	Make sure the s	lide is corre	ectly installed in	the backing pan [11].			
			3)	Make sure the "Z" fold [21] in the girt [6] is correct. Make sure the velcro strip [24] on the girt retainer flap [20] and the velcro strip [24] on the girt [6] are attached.						
			4)	Make sure that t [30] near the infl			m amount of slack in th gure 1).	e firing cable		
				a) Hold the fir slack from	-	30] at the inflatio	on cylinder valve [17] ar	nd pull the		
				<1> Pull the slack in the firing cable [30] away from the girt [6], in the direction of the inflation cylinder valve [17] until all slack is removed and the minimum amount of slack is achieved. If the minimum slack is not achieved, the escape slide pack [26] must be repacked.						
						in the firing cab ne inflation cyline	ole between the back of der valve [17].	f the		
			S ALL		SOURCE MRB	AFT GALLEY E	MERGENCY ESCAPE S	LIDE		<u></u>
						D633A109-AKS		P	age 6	of

25-240-00-04

Feb 15/2015



DATE		TA	IL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-240-		
<u> </u>	VARN	TH VA	HE SAFETY F ALVE, THE ES	PIN IS REI SCAPE SL	MOVED FROM LIDE CAN ACCI	E ESCAPE SLIDE PAC THE INFLATION CYLII IDENTALLY INFLATE. T AMAGE TO EQUIPMEN	NDER THIS CAN	MECH	INSP
	(i) R	temove t	he safety pin	[19] from	the inflation cyli	nder valve [17].			
	(j) P	ut the sa	ifety pin [19] i	nto the po	cket [18] on the	e escape slide pack [26]			
(1	k) C	lose the	cover [3].						
	. ,		•	-	ee the pressure djust as follows:	gauge [16] of the inflati :	on cylinder		
	,	1) Fully	open the co	ver [3] to g	get access to the	e escape slide pack [26].		
	2		nove the safet nflation cylind			t [18] and install safety	oin [19] in		
	3	•	e the escape vindow [1].	slide pack	([26] to get the	pressure gauge [16] in	the center of		
	4	,			minimum amou cylinder valve [nt of slack in the inflation 17] (Figure 1).	n firing		
		a)	Hold the firir slack from the		30] at the inflation	on cylinder valve [17] ar	d pull the		
			direct remo minin	tion of the ved and th	inflation cylinde ne minimum am	e [30] away from the girter valve [17] until all slad ount of slack is achieve , the escape slide pack	ck is d. If the		
					in the firing cat ne inflation cylin	ole between the back of der valve [17].	the		
	5	5) Rem	ove the safet	ty pin [19]	from the inflatio	n cylinder valve [17].			
	6	6) Put t	the safety pin	[19] into t	he pocket [18] o	on the escape slide pac	k [26].		
(n	•	lose the an [11].	cover [3] suff	ficiently to	connect the bo	nding jumpers [12] to th	e backing		
(1			stud [31], lad ng pan [11].	lder tie [32	2] and pin [33] to	o attach the bonding jur	npers [12] to		
(0	o) C	lose the	cover [3].						
(1	• /					e lower edge of the cov 12.0 in. (304.8 mm).	er [3] to the		
(0	q) A	lign the I	noles in the la	atch brack	et [13] and latch	n bracket [35] to install t	he latch [7].		
	(r) Ir	nstall the	latch [7] in th	e latch bra	acket [13] and la	atch bracket [35] as sho	own.		
(:	,					and 'B' between the lato an 0.01 in. (0.25 mm) (F			
	(t) Ir	nstall the	girt bar [4] in	the girt ba	ar retainers [5].				
				END OF	TASK ———				
	EFFECTI AKS A			SOURCE MRB	AFT GALLEY E	MERGENCY ESCAPE SI	_IDE		
					D633A109-AKS 25-240-00-04	:		age 7 d eb 15/2	
L.									



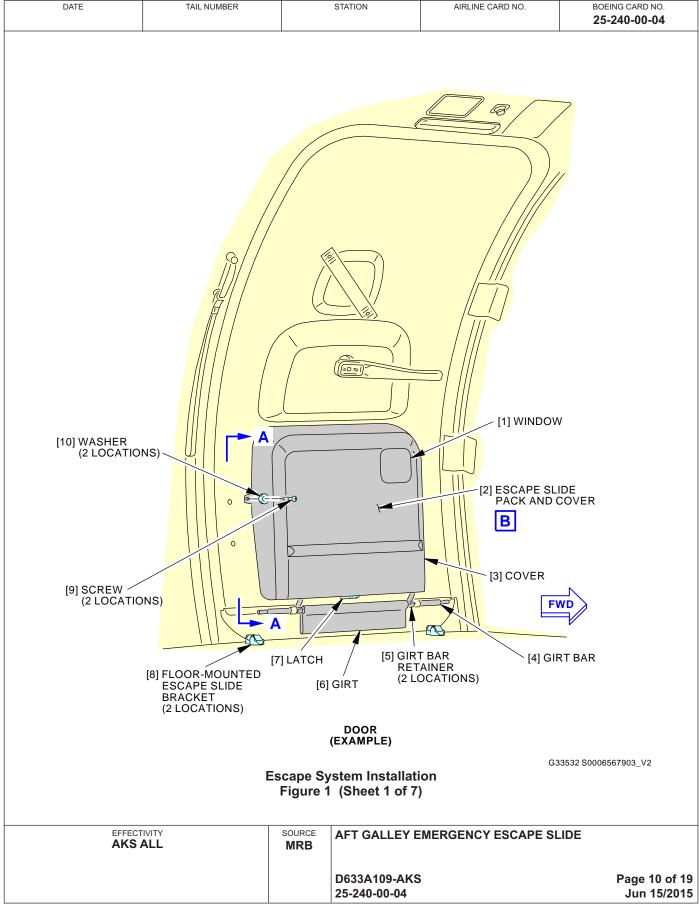
	DA	TE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-240-00-04					
Т	TASK	(25-66	-01-400	-803			MECH					
. <u>E</u>	Esca	pe Slic	le Pack	and Cover Installat	ion							
(F	Figu	re 1)										
A	A.	Prepa	e for th	e Installation								
		SUBTASK 25-66-01-410-004 (1) If the escape slide pack [26] is not installed in the cover [3], do this task: Escape Slide										
		` '		ape slide pack [26] is allation, TASK 25-66		over [3], do this task: Esca	ape Slide					
		SUBTASK	25-66-01-410	-005								
		(2) N	lake sur	e the door is closed	and latched.							
			25-66-01-410									
		. ,		llowing steps to mak alve [17]:	e sure that the safety p	oin [19] is not installed in	the inflation					
		(8			ck and cover [2] on a cl the clean flat horizont	lean flat horizontal surfac al surface.	e with the					
		(1) Ren	nove the latch [7] from	m the latch bracket [13	3].						
		(c) Ope	n the cover [3] suffic	ciently to see the inflation	on cylinder valve [17].						
	THE ESCAPE SLIDE WILL NOT OPERATE IF THE SAFETY PIN IS IN THE INFLATION CYLINDER VALVE. IF THE ESCAPE SLIDE DOES NOT OPERATE, PERSONS MAY BE INJURED.											
		(0	d) Mak	ce sure a safety pin [19] is not installed in th	ne inflation cylinder valve	[17].					
	CAUTION: BE CAREFUL WHEN YOU MOVE THE ESCAPE SLIDE PACK. WHEN THE SAFETY PIN IS REMOVED FROM THE INFLATION CYLINDER VALVE, THE ESCAPE SLIDE CAN ACCIDENTALLY INFLATE. THIS CAN CAUSE INJURIES TO PERSONS OR DAMAGE TO EQUIPMENT.											
			1)	Remove the safety	pin [19] from the inflat	ion cylinder valve [17].						
			2)	Put the safety pin [19] into the pocket [18]	on the escape slide pac	k [26].					
		(6	e) Clos	se the cover [3].								
		(f) Insta	all the latch [7] into th	ne latch bracket [13].							
E	В.	Install	ation									
		SUBTASK	25-66-01-420	-004								
		(1) Ir	stall the	escape slide pack a	and cover [2] on the do	or as follows:						
		<u>v</u>	/ARNIN	THE ESCAPE SI KILOGRAMS). IF	LIDE WEIGHS APPRO	ESCAPE SLIDE PACK AN DXIMATELY 55 POUNDS EFUL, INJURIES TO PEI OCCUR.	(24.94					
		(8	a) Insta	all the escape slide p	back and cover [2] on t	he door.						
			1)	Make sure that the	ball end of the lower fi	itting [15] goes through th	ne opening in					

EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY EMERGENCY ESCAPE SLIDE	
		D633A109-AKS 25-240-00-04	Page 8 of 19 Oct 15/2014

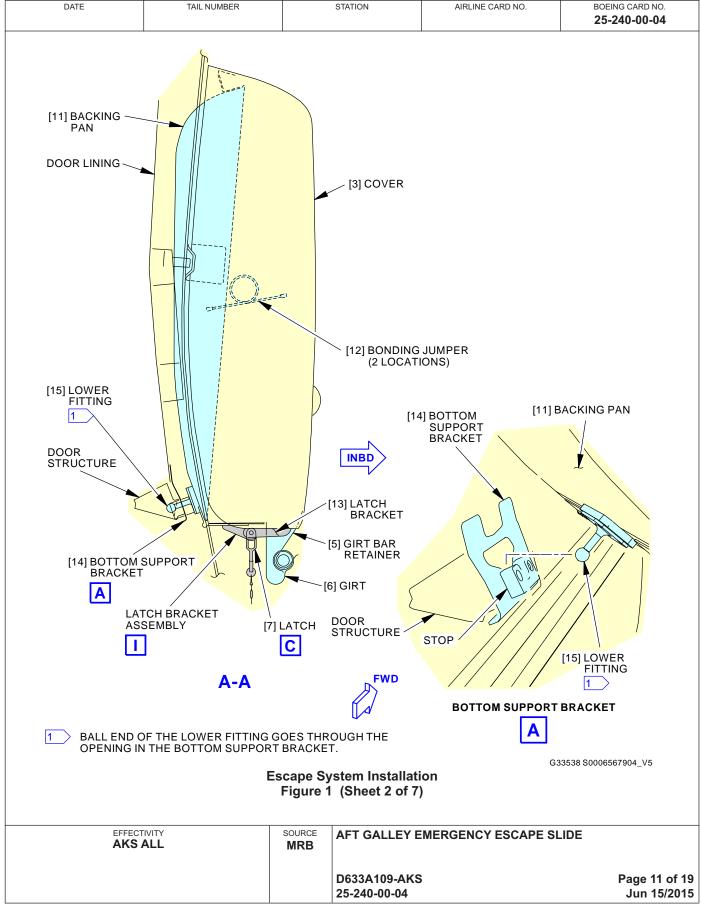


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 0-00-04				
(b)	Push the top of the esc door.	cape slide	pack and cover	[2] outboard until it is a	gainst the	MECH	INSF			
	NOTE: When the com	e compartı	ment is open for	ape slide pack can expa a minimum time to avo						
(c)	·	•	-	h the cover [3] to the do	or.					
	Make sure to use procedure.	,								
(d)	Make sure you can arn floor-mounted brackets		pe slide by insta	alling the girt bar [4] into	the					
(e)	Install the girt bar [4] in	to the girt	bar retainer [5].							
	NOTE: The length of t	he girt bar	for the FWD Do	oor is 33.2 in. (84.3 cm)	long.					
	The length for	the girt ba	r for the AFT Do	oor is 29.2 in. (74.2 cm)	long.					
		- END OF	TASK ———							
	FECTIVITY KS ALL	SOURCE MRB	AFT GALLEY E	MERGENCY ESCAPE SL	IDE					
			D633A109-AKS	;		Page 9				
			25-240-00-04			Oct 15/	201			

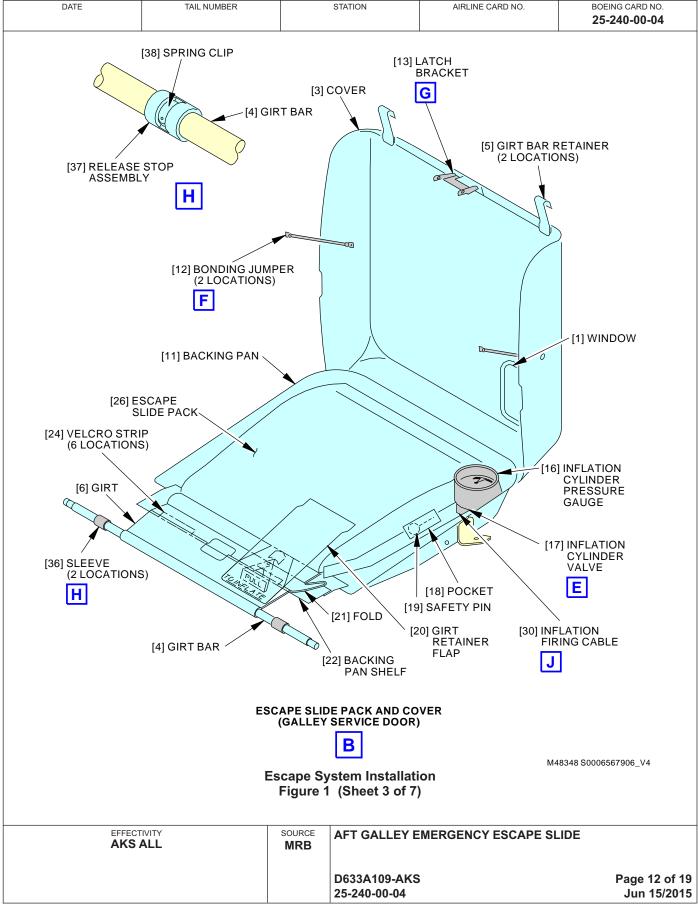












TAIL NUMBER

DATE

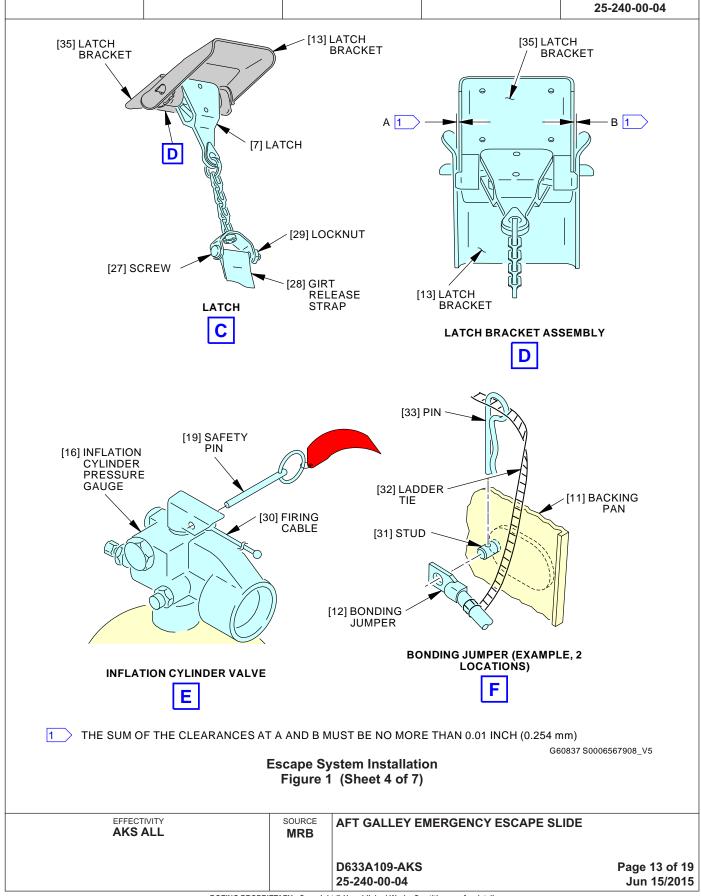


737-600/700/800/900 TASK CARDS

STATION

AIRLINE CARD NO.

BOEING CARD NO.





737-600/700/800/900 TASK CARDS

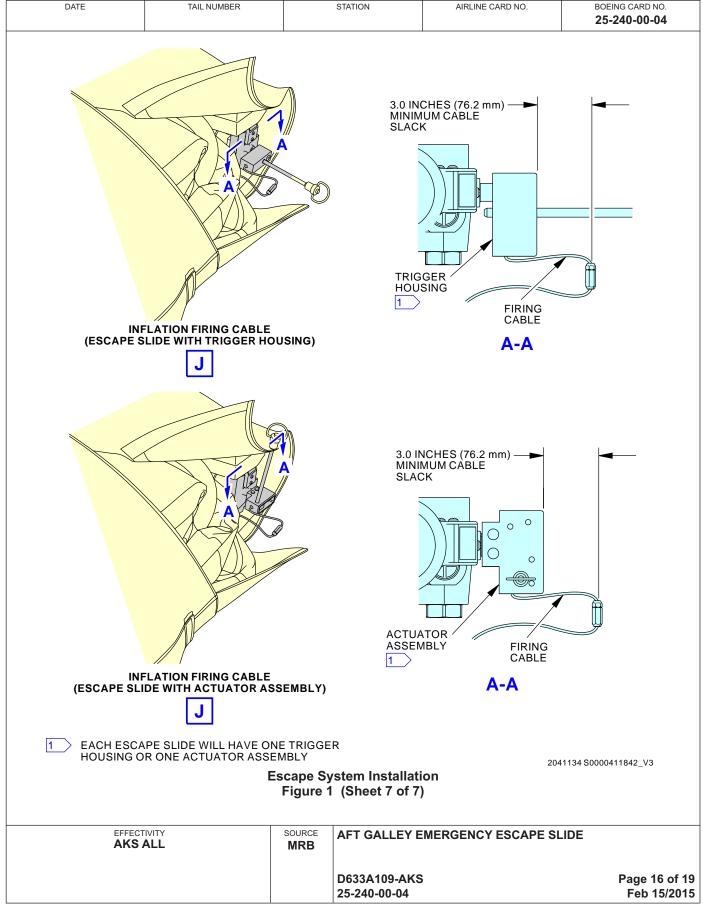
DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-04 [13] LATCH BRACKET [35] LATCH BRACKET STEP 2 (COCK LATCH) STEP 1 (ALIGN BRACKETS) G [13] LATCH [13] LATCH BRACKET **BRACKET** [35] LATCH BRACKET [35] LATCH **BRACKET** [7] LATCH STEP 4 (LATCH IN LOCKED POSITION) STEP 3 (POSITION LATCH AS SHOWN) G K62921 S0006567909_V2 **Escape System Installation** Figure 1 (Sheet 5 of 7) EFFECTIVITY SOURCE AFT GALLEY EMERGENCY ESCAPE SLIDE **AKS ALL MRB** D633A109-AKS Page 14 of 19 Jun 15/2015 25-240-00-04



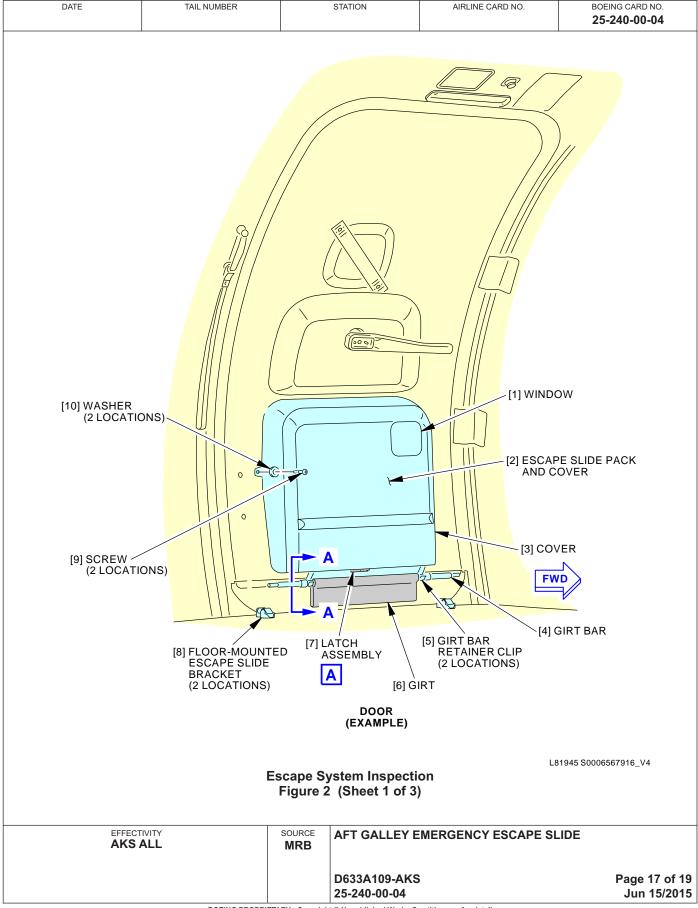
737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-04 [35] LATCH BRACKET (PAN ASSEMBLY) 0 0 LATCH BLOCK **ASSEMBLY SPLIT RING** [35] LATCH BRACKET (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH CLEVIS LATCH BLOCK 2 3 10.0° ±5.0° 10.0° ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. USE A MAXIMUM OF 30 POUNDS OF FORCE TO RELEASE THE LATCH. 1739991 S0000315072_V4 **Escape System Installation** Figure 1 (Sheet 6 of 7) EFFECTIVITY SOURCE AFT GALLEY EMERGENCY ESCAPE SLIDE **AKS ALL MRB** D633A109-AKS Page 15 of 19 Jun 15/2015 25-240-00-04











DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 25-240-00-04 LATCH BRACKET 0 (PAN ASSEMBLY) 0 LATCH BLOCK **ASSEMBLY** SPLIT RING [35] LATCH BRACKET PIN (2 LOCATIONS) (PAN ASSEMBLY) LATCH BRACKET ASSEMBLY WITH SPLIT RING LATCH BLOCK LATCH CLEVIS 3 > 2 Ē 10.0° ±5.0 10.0°, ±5.0° 30 LBS MAX 30 LBS MAX LATCH BRACKET ASSEMBLY MAKE SURE THAT THE CLEVIS CAN ROTATE FREELY IN THE LATCH BLOCK ASSEMBLY. EACH ESCAPE SLIDE WILL HAVE ONE LATCH CLEVIS OR ONE SPLIT RING. THE LATCH ASSEMBLY RELEASES AT 10 DEGREES, ± 5 DEGREES, WITH A MAXIMUM OF 30 LBF (133.45 NM). 3 M78780 S0006567917_V10 **Escape System Inspection** Figure 2 (Sheet 2 of 3) EFFECTIVITY SOURCE AFT GALLEY EMERGENCY ESCAPE SLIDE **AKS ALL MRB** D633A109-AKS Page 18 of 19 Jun 15/2015

25-240-00-04



737-600/700/800/900

		TASK CARDS		
DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 25-240-00-04
	1.980 INCH (50.292 mm)	0.56 (14		
	Esc F	cape System Inspec igure 2 (Sheet 3 of	ction	62267 S0000475279_V3
EFFECTI AKS A		OURCE MRB AFT GALLEY	'EMERGENCY ESCAPE SI	LIDE
		D633A109-AI 25-240-00-04	See title page for details	Page 19 of 19 Jun 15/2015



AIRLINE	E CARD NO	RESTOR	TITLE RE LIFE JACKETS		CARD NO.)-00-01
DATE	TASK RESTORE	_ KESTON	LI L UNOILLIO		ED CARD
TAIL NUMBER	WORK AREA UPR FUSELAGE		THRESHOLD REPEAT VEN REC	APPLIC	ABILITY
STATION	SKILL AIRPL			AIRPLANE ALL	ENGINE ALL
	7	ACCESS		ZONE 200	
	fe jackets (if insta		urer's recommended inte	erval.	
	EFFECTIVITY AKS ALL	SOURCE MRB	RESTORE LIFE JACKE	TS	
			D633A109-AKS 25-290-00-01		Page 1 of 2 Oct 15/2014



		DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ 25-290-		
	TAC	IZ 05 04	00 000 004				20 200	месн	INSP
1.			00-900-801 Restoration						
١.									
	A.	Proced							
			5-64-00-900-001 eplace the life jackets.						
) Use the standard met	hod of you	r airline to replac	ce the life jackets			
		(-,			TASK ——	o uno mo juonoso.			
				- END OF	IASK ———				
		A	FFECTIVITY IKS ALL	SOURCE MRB	RESTORE LIFE	JACKETS		ı	
					D633A109-AKS 25-290-00-01		F Ji	Page 2 un 15/2	of 2 2015



737-600/700/800/900 TASK CARDS

A. Refer	check the power r	version 1.1 Access	THRESHOLD 2 YR	REPEAT 2 YR		ABILITY ENGINE ALL
perationally of A. Referen	PASS CABIN SKILL AIRPL check the power r	1.1			AIRPLANE ALL ZONE	ENGINE
perationally o	AIRPL				ALL ZONE	
A. Refer	check the power r					
A. Refer		negaphones.				
					,	
	rences	Title				
	25-64-00-900-80		aphone Battery Ro	 eplacement (F	P/B 201)	

25-330-00-01

Oct 15/2014



737-600/700/800/900 **TASK CARDS**

	DATE			TA	IL NUMBER		STATION	AIRLINE CARD NO.	25-330-		
TAS	K 25-	64-00)-710·	-801						MECH	INS
Meg	apho	ne O	perat	ional	Test						
A.	Prod	edur	e								
	SUBTA	SK 25-6	4-00-020	-001							
	(1)	Disc	onne	ct the	clamp and r	emove the	megaphone fro	m the mounting bracke	et.		
	SUBTA (2)	Put	the m		none near yo	our mouth v	vhile you point t	he horn at the opposite	e end of the		
	suвта (3)				while you co	unt slowly	in a clear voice.				
	` ,				•	·					
	(4)			e you	r voice can b	e heard cle	early by another	person at the opposite	e end of the		
	SUBTA										
	(5)						•				
		(a)			• .	•	•	AMM TASK 25-64-00-9	900-804.		
						•					
		(c)	If the	e voic	e output rem	iains weak	, replace the me	egaphone.			
					oganhana ti	no that ha	o o long longord	and siven de the sten	a that fallow		
	(0)	to ex	xamin	e the	megaphone	siren (opti	onal):				
		NOT				ay have a l	ong lanyard with	n an alarm plug which o	operates as		
		(a)	Rem	nove t	he alarm plu	g pin to sta	art the siren auto	omatically.			
			1)	If the		•					
				a)		•	-	•			
			2)		•	w, sound c	haracter is unus	sual, or there is no sou	nd, do these		
				a)				lacement, AMM			
				b)	Do the ope	rational tes	st above again.				
				c)	If new batte	eries do no	t help, replace th	ne megaphone.			
								n back and release the	handle on		
	(7)	Atta	ch the	e meg	aphone on th	ne mountin	ig bracket with t	he clamp.			
						- END OF	TASK ———				
			ECTIVITY			SOURCE	POWER MEGA	PHONES			
		AN	S ALL	-		MRB					
	Meg	Megapho A. Proc SUBTA (1) SUBTA (2) SUBTA (3) SUBTA (4) SUBTA (5)	Megaphone O A. Procedur SUBTASK 25-6 (1) DISC SUBTASK 25-6 (2) Put airpl SUBTASK 25-6 (3) Pull SUBTASK 25-6 (4) Mak airpl SUBTASK 25-6 (5) If the (a) (b) (c) SUBTASK 25-6 (6) If you to expected by the process of the	A. Procedure SUBTASK 25-64-00-020 (1) Disconne SUBTASK 25-64-00-710 (2) Put the mairplane. SUBTASK 25-64-00-710 (3) Pull the tr SUBTASK 25-64-00-710 (4) Make surairplane. SUBTASK 25-64-00-710 (5) If the void (a) Dott (b) Dott (c) If the SUBTASK 25-64-00-710 (6) If you have to examinate	A. Procedure SUBTASK 25-64-00-020-001 (1) Disconnect the SUBTASK 25-64-00-710-001 (2) Put the microph airplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger SUBTASK 25-64-00-710-003 (4) Make sure your airplane. SUBTASK 25-64-00-710-004 (5) If the voice outplace (a) Do this tale (b) Do the oplace (c) If the voice SUBTASK 25-64-00-710-006 (6) If you have a material to examine the NOTE: The mean anatimatic (a) Remove to the step (a) 2) If the step (a) b) c)	A. Procedure SUBTASK 25-64-00-020-001 (1) Disconnect the clamp and results as 25-64-00-710-001 (2) Put the microphone near your airplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger while you consult as 25-64-00-710-003 (4) Make sure your voice can be airplane. SUBTASK 25-64-00-710-004 (5) If the voice output is weak, and airplane. SUBTASK 25-64-00-710-004 (6) If you have a megaphone to examine the megaphone man anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (b) Do the ope consult is longly in the alarm of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (b) Do the ope consult is longly in the alarm pluth of the alarm is of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (b) Do the ope consult is longly in the alarm pluth of the alarm is of an anti-theft alarm. (c) If the output is longly in the alarm is of an anti-theft alarm. (d) Remove the alarm pluth of the alarm is of an anti-theft alarm. (e) Remove the alarm pluth of the alarm is of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (b) Do the ope consult it is longly in the alarm is of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm. (b) Do the ope consult it is longly in the alarm is of an anti-theft alarm. (c) If the output is longly in the alarm is of an anti-theft alarm. (d) Remove the alarm pluth of the alarm is of an anti-theft alarm. (e) If the output is longly in the alarm is of an anti-theft alarm. (a) Remove the alarm pluth of the alarm is of an anti-theft alarm.	A. Procedure SUBTASK 25-64-00-020-001 (1) Disconnect the clamp and remove the SUBTASK 25-64-00-710-001 (2) Put the microphone near your mouth vairplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger while you count slowly SUBTASK 25-64-00-710-003 (4) Make sure your voice can be heard clearinglane. SUBTASK 25-64-00-710-004 (5) If the voice output is weak, do these steed in the voice output remains weak (a) Do this task: Megaphone Battery (b) Do the operational test above ag (c) If the voice output remains weak (b) Do the operational test above ag (c) If the voice output remains weak (a) If you have a megaphone type that had to examine the megaphone siren (option NOTE: The megaphone may have all an anti-theft alarm. (a) Remove the alarm plug pin to state (a) If the output is low, sound contents the state (b) Do the operational test (c) If new batteries do no NOTE: Make sure to the bad megas subtask 25-64-00-420-001 (7) Attach the megaphone on the mounting the subtask 25-64-00-420-001 (7) Attach the megaphone on the mounting subtask 25-64-00-400-101	Megaphone Operational Test A. Procedure SUBTASK 25-64-00-020-001 (1) Disconnect the clamp and remove the megaphone from SUBTASK 25-64-00-710-001 (2) Put the microphone near your mouth while you point the airplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger while you count slowly in a clear voice. SUBTASK 25-64-00-710-003 (4) Make sure your voice can be heard clearly by another airplane. SUBTASK 25-64-00-710-004 (5) If the voice output is weak, do these steps: (a) Do this task: Megaphone Battery Replacement, (b) Do the operational test above again. (c) If the voice output remains weak, replace the messubtask 25-64-00-710-006 (6) If you have a megaphone type that has a long lanyard to examine the megaphone may have a long lanyard with an anti-theft alarm. (a) Remove the alarm plug pin to start the siren autoful in the start of usual amplitude and sound a) Insert the alarm pin to stop the siren were steps: a) Do this task: Megaphone Battery Replacement of usual amplitude and sound a) Insert the alarm pin to stop the siren were steps: a) Do this task: Megaphone Battery Replacement of the output is low, sound character is unusual steps: a) Do the operational test above again. c) If new batteries do not help, replace the NOTE: Make sure to put the alarm pin the bad megaphone.	A. Procedure SUBTASK 25-64-00-020-001 (1) Disconnect the clamp and remove the megaphone from the mounting bracket subtask 25-64-00-710-001 (2) Put the microphone near your mouth while you point the horn at the opposite airplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger while you count slowly in a clear voice. SUBTASK 25-64-00-710-003 (4) Make sure your voice can be heard clearly by another person at the opposite airplane. SUBTASK 25-64-00-710-004 (5) If the voice output is weak, do these steps: (a) Do this task: Megaphone Battery Replacement, AMM TASK 25-64-00-6 (b) Do the operational test above again. (c) If the voice output remains weak, replace the megaphone. SUBTASK 25-64-00-710-006 (6) If you have a megaphone type that has a long lanyard and siren, do the step to examine the megaphone siren (optional): NOTE: The megaphone may have a long lanyard with an alarm plug which an anti-theft alarm. (a) Remove the alarm plug pin to start the siren automatically. 1) If the alarm is of usual amplitude and sound, the megaphone is greated and site of usual amplitude and sound, the megaphone is greated and steps: a) Insert the alarm pin to stop the siren when you release the first the site of the megaphone and the stack: Megaphone Battery Replacement, AMM TASK 25-64-00-900-804. b) Do the operational test above again. c) If new batteries do not help, replace the megaphone. NOTE: Make sure to put the alarm pin back and release the the bad megaphone.	TASK 25-64-00-710-801 Megaphone Operational Test A. Procedure SUBTASK 25-64-00-220-001 (1) Disconnect the clamp and remove the megaphone from the mounting bracket. SUBTASK 25-64-00-20-001 (2) Put the microphone near your mouth while you point the horn at the opposite end of the airplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger while you count slowly in a clear voice. SUBTASK 25-64-00-710-003 (4) Make sure your voice can be heard clearly by another person at the opposite end of the airplane. SUBTASK 25-64-00-710-004 (5) If the voice output is weak, do these steps: (a) Do this task: Megaphone Battery Replacement, AMM TASK 25-64-00-900-804. (b) Do the operational test above again. (c) If the voice output remains weak, replace the megaphone. SUBTASK 25-64-01-0006 (6) If you have a megaphone type that has a long lanyard and siren, do the steps that follow to examine the megaphone siren (optional): NOTE: The megaphone may have a long lanyard with an alarm plug which operates as an anti-theft alarm. (a) Remove the alarm plug pin to start the siren automatically. 1) If the alarm is of usual amplitude and sound, the megaphone is good. a) Insert the alarm pin to stop the siren when you release the handle. 2) If the output is low, sound character is unusual, or there is no sound, do these steps: a) Do this task: Megaphone Battery Replacement, AMM TASK 25-64-00-900-804. b) Do the operational test above again. c) If new batteries do not help, replace the megaphone. NOTE: Make sure to put the alarm pin back and release the handle on the bad megaphone. NOTE: Make sure to put the alarm pin back and release the handle on the bad megaphone.	Megaphone Operational Test A. Procedure SUBTASK 25-44-00-200-01 (1) Disconnect the clamp and remove the megaphone from the mounting bracket. SUBTASK 25-44-00-200-01 (2) Put the microphone near your mouth while you point the horn at the opposite end of the airplane. SUBTASK 25-64-00-710-002 (3) Pull the trigger while you count slowly in a clear voice. SUBTASK 25-64-00-710-003 (4) Make sure your voice can be heard clearly by another person at the opposite end of the airplane. SUBTASK 25-64-00-710-004 (5) If the voice output is weak, do these steps: (a) Do this task: Megaphone Battery Replacement, AMM TASK 25-64-00-900-804. (b) Do the operational test above again. (c) If the voice output remains weak, replace the megaphone. SUBTASK 25-64-01-0066 (6) If you have a megaphone type that has a long lanyard and siren, do the steps that follow to examine the megaphone siren (optional): NOTE: The megaphone may have a long lanyard with an alarm plug which operates as an anti-theft alarm. (a) Remove the alarm plug pin to start the siren automatically. 1) If the alarm is of usual amplitude and sound, the megaphone is good. a) Insert the alarm pin to stop the siren when you release the handle. 2) If the output is low, sound character is unusual, or there is no sound, do these steps: a) Do this task: Megaphone Battery Replacement, AMM TASK 25-64-00-900-804. b) Do the operational test above again. c) If new batteries do not help, replace the megaphone. NOTE: Make sure to put the alarm pin back and release the handle on the bad megaphone. NOTE: Make sure to put the alarm pin back and release the handle on the bad megaphone.





	CARD NO	POWER	TITLE MEGAPHONES BAT	TTERIES	BOEING 0 25-340	CARD NO.)-00-01
DATE	TASK REPLACE				RELATE	D CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT	APPLIC, AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	NOTE			ALL	ALL
		ACCESS			ZONE 221 241	
NI ENVAL N	OTE: Vendor Re					

EFFECTIVITY AKS ALL	SOURCE MRB	POWER MEGAPHONES BATTERIES	
		D633A109-AKS 25-340-00-01	Page 1 of 2 Oct 15/2014



	ı	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
							25-340	1	
	TAS	K 25-64-00-	900-804					MECH	INSP
1.	Meg	japhone Bat	ttery Replacement						
	Α.	Procedure	1						
		SUBTASK 25-64-							
			ace the batteries in the	megapho	nes.				
						dations for the megaph	one or the		
		(4)	standard method of yo	ur airline t	to replace the ba	itteries.			
					TASK ———				
				LIID OI	IAON				
			T11/1T1/	0011005	DOM/===	DUONEO DA ETTORO			
		AKS	TIVITY ALL	SOURCE MRB	POWER MEGA	PHONES BATTERIES			
					D633A109-AKS 25-340-00-01		I F	Page 2 eb 15/	of 2 2015





AIRLINE	E CARD NO		TITLE CY LOCATOR TRA		BOEING 0 25-350	
DATE	TASK FUNCTIONAL	(SURV	/IVAL / PORTABLE	RELATE	D CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	NOTE			ALL NOTE	ALL
		ACCESS			ZONE 221 241	

Functionally check (off-aircraft) the Emergency Locator Transmitter (Survival / Portable Type) per Vendor's CMM.

INTERVAL NOTE: At manufacturer's recommended interval or national regulatory requirement. Whichever

comes first.

AIRPLANE NOTE: If Installed. Applicable to dry cell type ELT's only.

SOURCE MRB EMERGENCY LOCATOR TRANSMITTER (SURVIVAL / PORTABLE TYPE)

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	Г	DATE	TAIL NUM	BER		STATION	AIRLINE CARD NO.	BOEING C 25-350		
	TAS	K 25-64	-00-000-801						MECH	INSP
1.	Eme	rgency	Locator Transmit	tter (ELT)	Remova	<u>al</u>				
	A.	Proce	dure							
			25-64-00-020-011							
		(1) L	oosen the straps o	n the eme	ergency lo	ocator transmitte	er from the support brac	cket.		
			25-64-00-020-003	ممماليممم	tor Trope	mittor (FLT) fro	m ita atawaga braskat			
		. ,	emove me ⊑merge 25-64-00-020-012	ency Loca	ilor mans	smiller (ELI) IIO	m its stowage bracket.			
		(3) N				nstalled) is set t	o the correct position pe	er the		
			OTE: The four-pos	•		d be set to the (OFF position.			
				have a th	ree-posit	ion ON/OFF/AF	RMED switch. The three	-position		
				—— г	END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE MRB	EMERGENCY L PORTABLE TY	OCATOR TRANSMITTER	R (SURVIVAL	1	
						D633A109-AKS 25-350-00-01			Page 2 eb 15/	
								<u> </u>		



	D	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	25-350		
_			64-00-4						MECH	INS
2. <u>E</u>	Eme	rgen	cy Loca	ator Transmitter (E	LT) Installa	<u>tion</u>				
	A.	Proc	edure							
		SUBTA	SK 25-64-00	-010-001						
		(1)	Remov	e the packing label	(if installed)	from the ELT.				
		SUBTA	SK 25-64-00							
		(2)	Make s retaine		a is in its sto	owed position in	the lanyard guide and	under the		
		SUBTA	SK 25-64-00	-420-015						
		(3)	Put the	e switch (if installed)	to the corre	ect position per t	he airline's regulatory r	equirements.		
			NOTE:	The four-position s	switch shoul	d be set to the A	ARM position.			
			NOTE:	Some ELTs have a switch should be s			RMED switch. The three	-position		
			NOTE:	successful self-tes flashes) if the resu failures during self	t by blinking ilt is accepta -test. The L continuously	g eight flashes (vable or by not bli ED will also indi y every two seco	E LED indicator will indicator with a 0.5 second delay nking if there is one or cate that the unit is in "onds. The LED remains	between more Transmit"		
		SUBTA	SK 25-64-00	-420-004						
		(4)	Install	the Emergency Loca	ator Transm	itter (ELT) on its	stowage bracket.			
		SUBTA	SK 25-64-00	-420-016						
		(5)	Tighter	n the straps on the s	towage bra	cket.				
					— END OF	TASK ——				
			EFFECTI AKS A		SOURCE MRB	EMERGENCY L PORTABLE TYI D633A109-AKS 25-350-00-01	•		/ Page 3	3 of /201





AIRLINE	E CARD NO		TITLE ICY LOCATOR TRA		BOEING C 25-360	
DATE	TASK DISCARD	(SURVIVAL /	PORTABLE TYPE)	- BATTERIES	RELATEI	D CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT	APPLICA	
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL NOTE	ALL
		ACCESS			ZONE 221 241	

Discard the Emergency Locator Transmitter (Survival / Portable Type) batteries.

INTERVAL NOTE: At manufacturer's recommended interval or national regulatory requirement. Whichever

comes first.

AIRPLANE NOTE: If Installed. Applicable to non-dry cell type ELT's only.

EFFECTIVITY
AKS ALL

SOURCE
MRB

EMERGENCY LOCATOR TRANSMITTER (SURVIVAL / PORTABLE TYPE) - BATTERIES

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		DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 25-360-		
	TAS	K 25-64-00	0-000-801				1	MECH	INSF
1.	Eme	ergency Lo	ocator Transmitter (EL	T) Remova	<u>al</u>				
	A.	Procedu	e						
		SUBTASK 25-6							
		, ,	•	mergency l	ocator transmitte	er from the support brac	cket.		
		subtask 25-6 (2) Rem	4-00-020-003 nove the Emergency Lo	cator Trans	smitter (FLT) fro	m its stowage bracket			
		SUBTASK 25-6		routor fruit	51111101 (221) 110	m no otomago brachou			
		(3) Mak			nstalled) is set t	o the correct position p	er the		
		NOT	ΓΕ: The four-position s	witch shoul	d be set to the C	OFF position.			
		NOT	<u>FE</u> : Some ELTs have a switch should be se			RMED switch. The three	-position		
				– END OF	TASK ———				
			ECTIVITY S ALL	SOURCE MRB	EMERGENCY L	OCATOR TRANSMITTEI	R (SURVIVAL	<u> </u> 	
					D633A109-AKS			Page 2	of
					25-360-00-01			eb 15/	



	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-360		
	TAS	K 25-6	64-00-4	l00-801					MECH	INS
2.	Eme	rgeno	cy Loca	ator Transmitter (EL	.T) Installa	tion				
	A.	Proc	edure							
		SUBTAS	SK 25-64-00	0-010-001						
		(1)	Remov	ve the packing label ((if installed)	from the ELT.				
		SUBTAS	SK 25-64-00							
		(2)	Make s		a is in its sto	wed position in	the lanyard guide and	under the		
			SK 25-64-00							
		(3)		,			he airline's regulatory r	equirements.		
				: The four-position s			•			
	NOTE: Some ELTs have a three-position ON/OFF/ARMED switch. The three-position switch should be set to the ARMED position.					e-position				
	NOTE: Some portable ELTs have LED indicators. The LED indicator will indicate a successful self-test by blinking eight flashes (with a 0.5 second delay between flashes) if the result is acceptable or by not blinking if there is one or more failures during self-test. The LED will also indicate that the unit is in "Transmit" mode, by blinking continuously every two seconds. The LED remains unlit when the unit is in "Armed" mode or "Off" mode.									
SUBTASK 25-64-00-420-004										
		(4)	Install	the Emergency Loca	tor Transm	itter (ELT) on its	stowage bracket.			
		SUBTAS	SK 25-64-00	0-420-016						
		(5) Tighten the straps on the stowage bracket.								
					– END OF	TASK ———				
			EFFECT AKS A		SOURCE MRB	EMERGENCY L PORTABLE TYPE	OCATOR TRANSMITTE PE) - BATTERIES	R (SURVIVAL	l I	





AIRLIN	NE CARD NO		TITLE		BOEING	CARD NO.
		DETACHABI	LE EMERGENCY	EQUIPMENT	25-370	0-00-01
DATE	VISUAL CHECK				RELATE	ED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 2 YR	REPEAT 2 YR	APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS			ZONE 221 241	
and medical	kits, as applicable) for condition and	d presence.		rash axe, flashlights	
	EFFECTIVITY AKS ALL	SOUF MR		BLE EMERGENO	CY EQUIPMENT	
			D633A109-/			Page 1 of 2 Oct 15/2014



	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 25-370		
	TAS	K 25-64	4-00-	210-802	I				MECH	INSP
1.				ergency Equipment	: Visual Che	eck				
	Α.	Proce	dure	1						
				00-210-002						
		(1) N	Make	sure the following pi	eces of eme	ergency equipme	ent are installed in the a	pplicable		
			ocatio							
		<u> </u>	NOTE	E: Also, make sure the airline to do this cl	ney have no neck.	visible defects.	Use the standard meth	od of your		
		(a)	Smoke goggles						
		(b)	Crash ax						
		((c)	Flash lights						
		(d)	First aid kits						
			•	Medical kits						
			(f)	Fire gloves						
					— END OF	TASK ———				
			EFFEC AKS	TIVITY ALL	SOURCE MRB	DETACHABLE	EMERGENCY EQUIPME	NT		<u> </u>
						D633A109-AKS 25-370-00-01			Page 2 eb 15/	



737-600/700/800/900 **TASK CARDS**

AIRLIN	IE CARD NO	EMER	TITLE GENCY FLASHLIG	HTS		CARD NO. 0-00-01
DATE	TASK OPERATIONAL	-				ED CARD
AIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD NOTE	REPEAT		CABILITY
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 221 241	
	y check the emerg		or hattery restors	ation		
TERVAL N	NOTE: At schedul	ed battery change	or battery restora	ation.		
	EFFECTIVITY AKS ALL	SOUR MR	CE EMERGENCY	Y FLASHLIGH	rs	



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				25-380-00-01

TASK 25-64-00-200-802

MECH INSP

1. Flashlight Check (With Flashing or Push To Test - LED Indicator)

A. Procedure

SUBTASK 25-64-00-210-005

- (1) Do a check of the LED and compare with the limits in the table below. Follow the instructions if you find that the operation is not in the limits.
- (2) Remove the flashlight from the mounting bracket.
- (3) If the light does not come on or is weak, refer to the table to identify the problem and do the corrective action.

AKS 001-010, 013, 015-018, 020-025, 027

Table 1

Symptom	Probable Cause	Corrective Action
LED is not flashing.	Battery is depleted.	Replace battery.
	Lamp is burned out.	Replace lamp.
	LED is inoperative.	Return unit for service.
	Defective Bracket. Flashlight did not turn off when installed in bracket.	Return entire system to factory for check.
LED is flashing at interval of 10 seconds or greater.	Battery is at or beyond cutoff point.	Replace battery within 10 days.
Flashlight does not turn on when removed from bracket.	Unauthorized use of flashlight. Battery depleted.	Replace battery.
	Lamp is burned out.	Replace lamp.
	Battery pack not installed or connected.	Check battery and connection.
	Broken contacts in flashlight body.	Return flashlight for service.
	Defective bracket. Flashlight did not turn off when installed in bracket.	Return entire system to factory for check.

EFFECTIVITY AKS ALL	SOURCE MRB	EMERGENCY FLASHLIGHTS		
			Page 2 Jun 15/	





DATE TAIL NU	MBER		STATION	AIRLINE CARD NO.	BOEING 0 25-380		
AKS 011, 012, 014, 019,	026, 028-	.999				MECH	INS
		Table	2				
Mounting Symptom		Probable	Cause	Corrective Act	ion		
Mounting bracket LED does not illuminate when the push-to test button is pressed	failed.			Remove the handlight from the mounting bracket. If the handlight does not turn ON immediately remove and replace the battery pack. After replacing the battery pack, verify that the handlight is ON when the handlight is not in the mounting bracket. Reinsert the handlight in the mounting bracket and verify that the LED turns on green when the push-to-test button is pressed			
	Electrical failure in handlight electronics.			If changing the battery paturn the handlight ON, the electrical failure in the hat electronics. Refer to REF repair procedures	ere is an andlight		
	Electrical failure in mounting bracket electronics.			If the handlight turns ON removed from the mount there is an electrical failu mounting bracket. Repla mounting bracket. Return unit to repair station.	ing bracket, ire in the ce the		
	Pogo contacts not making contact.			Rotate and ensure pogo making contact. Refer to for handlight installation	ASSEMBLY		
	Battery pack installed incorrectly (backwards).			Install battery correctly. Refer to ASSEMBLY for handlight installation procedures.			
Handlight does not turn ON when it is removed from the mounting bracket.	Battery p	eack is deple	eted or has failed	Replace the battery pack replacing the battery pack the handlight is ON when handlight is not in the mo- bracket. Reinsert the har mounting bracket and ve LED turns on green when push-to-test button is pre-	k, verify that the punting ndlight in the rify that the n the		
	Electrical failure in handlight electronics.		If changing the battery pack does not turn the handlight ON, there is an electrical failure in the handlight electronics. Refer to REPAIR for repair procedures				
EFFECTIVITY AKS ALL		source MRB	EMERGENCY F			Page 3	o.f
			25-380-00-01	· 		un 15/	



DATE	TAIL NU	JMBER		STATION	AIRLINE CARD NO.		CARD NO.)-00-01	
KS 011, 012, 014, 0	19, 026, 028-99	99 (Continu	ed)			•	MECH	INS
		Tab	le 2 (C	ontinued)				
Mounting Sy	mptom	Р	robable	e Cause	Corrective Action			
Handlight does not tur eturned to its mountir	Electrical fa or handligh electronics.	t head a	mounting bracket ssembly	If the handlight does not when it is returned to the bracket, there may be a failure in the mounting lanother handlight in the bracket. If that handlight turn OFF, the failure is in mounting bracket electron Replace the mounting become handlight turns is inserted in the mount the electrical failure is in handlight head assemb Replace handlight.				
Mounting bracket LED ed (flashes red once) bush-to test button is promitoring circuit cont he battery every ten semit a red flash until the placed.	The battery	pack ch	narge is low.	The battery pack should at earliest possible servoperator has approximate weeks to replace the battery of the battery of the page 1.5 miles of the battery o	rice. The ately three			
AKS ALL		1					_	
SUBTASK 25-6								
(4) Insta	all the flashligh	t in the mou	ınting b	racket.				
EFFE AK		source MRB	EMERGENCY F	CLASHLIGHTS				
				D633A109-AKS 25-380-00-01			Page 4 Jun 15/	





AIRLINE	CARD NO	TITLE EMERGENCY FLASHLIGHTS BATTERIES			BOEING CARD NO. 25-390-00-01		
DATE	TASK REPLACE				RELATE	D CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT	APPLIC/	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL NOTE	ALL	
		ACCESS			ZONE 200		

Replace the emergency flashlight batteries at the manufacturer's recommended interval.

AIRPLANE NOTE: Applicable to airplanes with non-rechargeable flashlight batteries.

A. References

Reference	Title
AMM 25-64-00-200-802	Flashlight Check (With Flashing or Push To Test - LED Indicator) (P/B 201)

EFFECTIVITY AKS ALL	SOURCE MRB	EMERGENCY FLASHLIGHTS BATTERIES	
		D633A109-AKS 25-390-00-01	Page 1 of 2 Oct 15/2014



	ı	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 25-390		
	TAS	K 25	-64-00-	900-806					MECH	INSP
1.	Flas	shligh	t Batte	ery Replacement (Wit	h Flashin	g or Push To Te	est - LED Indicator)			
	A.		cedure							
		SUBT/ (1)		oo-900-007 ace the batteries in the	flashlights					
		(1)					dations for the flashligh	t or the		
			, ,	standard method of yo	ur airline t	o replace the ba	tteries.			
		(2)		is task: Flashlight Cheo 325-64-00-200-802	ck (With FI	ashing or Push	To Test - LED Indicator), AMM		
					- END OF	TASK ———				
				ALL	SOURCE MRB	EMERGENCY F	LASHLIGHTS BATTERIE	ES		
						D633A109-AKS 25-390-00-01			Page 2 eb 15/	





AIRLIN	E CARD NO		TITLE SMOKE HOODS			CARD NO.)-00-01
DATE	TASK INSPECTION - DETAILED					ED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 24 MO	REPEAT 24 MO		CABILITY
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 220 240	
Inspect (Deta	ailed) the smoke h	oods for conditior	1.			
	EFFECTIVITY	SOUF	RCE SMOKE HOO	ากร		
	EFFECTIVITY AKS ALL	MR	B SWICKE HOU	<i>J</i> <u>J</u> J		
			D633A109-A			Page 1 of 2 Oct 15/2014



	Г	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO 25-400-00-01				
	TAS	K 25-64-0	0-200-801	<u>'</u>			1	MECH	INS		
1.	Smo	ke Hood	(Protective Breathin	g Equipmen	t) Inspection						
	A.	Procedu	ire								
			64-00-210-004								
		` '			•	n of the smoke hoods:					
		<u>NO</u>				hoods. Most smoke how when the smoke hood					
		AKS ALL	.; AIRPLANES WITH S	MOKE HOO	D CONTAINERS	WITH TAMPER TAGS					
		(a)	Do a visual check of	f the tamper	tags to make su	re they are not broken of	or ripped.				
		AKS ALL	.; AIRPLANES WITH S	MOKE HOO	D CONTAINERS	WITH AN INDICATOR \	WINDOW				
		<u>CA</u>	CONTAINER	R. IF YOU OF	EN THE CONTA	DICATOR WINDOW ON AINER, YOU CAN CAU THER COMPONENTS.	SE				
		(b)	Do a visual check of hood has not filled t		r window on the	container to make sure	the smoke				
		AKS ALL STRIPS				with Moisture Indicate container to make sure					
		(0)	moisture indicator s	trip has not c	changed to pink.						
			serviceable		of the moisture i	change, but not to pink, ndicator changes to gra					
		AKS ALL		SMOKE HOO	D CONTAINERS	WITHOUT AN INDICAT	OR				
		(d)	Feel the smoke hoo								
			NOTE: The smoke package ha		ıum packed. If th	ne smoke hood feels so	ft, then the				
				— END OF	TASK ———						
			FECTIVITY (S ALL	SOURCE MRB	SMOKE HOODS	S					
					D633A109-AKS 25-400-00-01	;		Page 2 un 15/			



DATE TASK	E CARD NO		SMOKE HOODS			BOEING CARD NO. 25-410-00-01		
DATE	TASK DISCARD				RELATE	D CARD		
AIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT	APPLIC AIRPLANE	ABILITY ENGINI		
STATION	SKILL AIRPL	NOTE			ALL	ALL		
		ACCESS			ZONE 221 241			
eard the a	moke beeds at th	o manufacturer's	rocommonded int	onvol				
card the s	moke hoods at th	e manufacturer's	recommended int	erval.				
TERVAI N	IOTE: Vendor Re	eC.						
ICKVALN	ioie: vendor Re	eC						

EFFECTIVITY AKS ALL	SOURCE MRB	SMOKE HOODS	
		D633A109-AKS 25-410-00-01	Page 1 of 2 Jun 15/2015



	С	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ 25-410 -		
	TAS	K 25-	-64-00-	-900-807					MECH	INSP
1.				Protective Breathing	Equipmen	t) Restoration				
	A.	Pro	cedure	•						
				00-900-008						
		(1)		ace the smoke hoods. Use the standard meth	had of your	r airlina ta ranlaa	a tha smaka haads			
			(a)	Ose the standard met			e the shloke hoods.			
					- END OF	TASK ———				
			EFFEC AKS	CTIVITY S ALL	SOURCE MRB	SMOKE HOODS	;			
				_						
						D633A109-AKS 25-410-00-01		F	Page 2 eb 15/2	of 2



			FIRST AID KITS	BOEING CARD NO. 25-420-00-01		
DATE	TASK RESTORE			RELATE	ED CARD	
AIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD REPEAT VEN REC	APPLIC AIRPLANE	ABILITY ENGIN	
STATION	SKILL AIRPL	NOTE		ALL	ALL	
		ACCESS		ZONE 221 241		
				221 241		
		1				
		-		221 241		
	1	1		I		
otoro tha f	rot old kits of the	manufacture:	recommended interval			
store the fi	rst aid kits at the	manufacturer's	recommended interval.			
Store the li	ist aiu kits at the	manulacturer S	recommended fillerval.			
TED\/A! \	OTE: \/a=da=D=					
TERVAL N	OTE: Vendor Re	ec				

EFFECTIVITY AKS ALL	SOURCE MRB	FIRST AID KITS	
		D633A109-AKS 25-420-00-01	Page 1 of 2 Oct 15/2014



	L	DATE	I AIL NUMBER		STATION	AIRLINE CARD NO.	25-420-		
	TAS	K 25-64-00-9	900-808					MECH	INSP
1.		t Aid Kit Res							
	Α.	Procedure							
	Α.	SUBTASK 25-64-00	0-900-009						
			ce the first aid kit.						
			Jse the standard method	d of your	airline to replac	ce the first aid kit.			
					TASK —				
			_						
		FFFOT	TIVITY	SOLIBOE	FIDOT AID 1//TO				
		EFFECT AKS	ALL	SOURCE MRB	FIRST AID KITS	•			
					D0004400 4160		_	\ ^	
					D633A109-AKS 25-420-00-01		F ₀	Page 2 eb 15/2	of 2 2015





AIRLINE	E CARD NO	MEDICAL KITS			BOEING CARD NO. 25-430-00-01		
DATE	TASK RESTORE				RELATE	D CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT	APPLICABILITY AIRPLANE ENGI		
STATION	SKILL AIRPL		NOTE			ALL	ALL
		ACCESS			ZONE 221 241		
estore the n	nedical kits at the	manufacturer's	recommended inte	rval			

DATE	TASK RESTORE				RELATI	ED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD VEN REC	REPEAT		CABILITY
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 221 241	
		-				
Restore the n	nedical kits at the	manufacturer's re	ecommended int	erval.		
INTERVAL N	OTE: Vendor Re	С				
	AKS ALL	SOUI	RB MEDICAL P	KITS		
			D633A109-			Page 1 of 2
			25-430-00-0)1		Oct 15/2014

EFFECTIVITY AKS ALL	SOURCE MRB	MEDICAL KITS	
		D633A109-AKS 25-430-00-01	Page 1 of 2 Oct 15/2014



DATE		DATE	TAIL NUMBER				CARD NO. -00-01		
		K 25-64-00-9						MECH	INSP
1.	Medical Kit Restoration								
	A.	A. Procedure							
		(1) Pople	o-900-010 ce the medical kit.						
			Jse the standard method	d of your	airline to renla	ce the medical kit			
		(α)			TASK ——	oo tiio iiiodiodi kit.			
				IND OF	IASK ———				
		EFFECT	TIVITY	SOURCE	MEDICAL KITS				
		AKS	ALL	MRB					
					D633A109-AKS	;	j.	Page 2	of 2