



				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
20-010-01-01	MRB	05-55-25-200-802 05-55-25-200-803 05-55-25-200-804 05-55-25-200-805 05-55-25-200-806	1.1	15000 FH	15000 FH	ALL	ALL
		al inspection of HIRF/L s s of damage and lack of			ssure vessel on l	eft side of the airpla	ne. Look for
20-010-02-01	MRB	05-55-26-200-802 05-55-26-200-803 05-55-26-200-804 05-55-26-200-805 05-55-26-200-806	1.1	15000 FH	15000 FH	ALL	ALL
		al inspection of HIRF/L s s of damage and lack of			ssure vessel on r	ight side of the airpl	lane. Look for
20-020-00-01	MRB	05-55-15-200-806	1.1	30000 FH	30000 FH	ALL	ALL
		al inspection of HIRF/L sity of the wire runs.	ensitive wire r	uns inside the pres	sure vessel. Look	for obvious signs o	of damage and
20-030-01-01	MRB	05-55-23-200-802 05-55-23-200-803 05-55-23-200-804 05-55-23-200-805 05-55-23-200-806	1.1	12000 FH	12000 FH	ALL	ALL
		inspection of the Lightni ondition of security and s			outside the pressu	ire vessel on the lef	t side of airplan
20-030-02-01	MRB	05-55-24-200-802 05-55-24-200-803 05-55-24-200-804 05-55-24-200-805 05-55-24-200-806	1.1	12000 FH	12000 FH	ALL	ALL
		inspection of the Lightni pect for condition of secu			outside the pressu	ire vessel on the rig	ht side of
20-040-01-01	MRB	05-55-40-200-802 05-55-40-200-803 05-55-40-200-804 05-55-40-200-805 05-55-40-200-806 05-55-44-200-801 05-55-44-200-802 05-55-44-200-803 05-55-44-200-804 05-55-44-200-804	1.1	15000 FH	15000 FH	ALL	ALL

Perform a functional check of the HIRF/L sensitive connectors outside the pressure vessel on the left side of the airplane. Check DC resistance from the backshell to ground.

AIRPLANE NOTE: Functional check using the Loop Resistance Test in AMM 05-55-44-200-XXX is the Boeing preferred method. An alternate Bond Resistance Test is provided for operators in lieu of a Loop Resistance Test for operators choosing to utilize the approved Bond Resistance Test method. Please complete this functional check using a Loop Resistance Test or a Bond Resistance Test.





				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
20-040-02-01	MRB	05-55-41-200-802 05-55-41-200-803 05-55-41-200-804 05-55-41-200-805 05-55-41-200-806 05-55-45-200-801 05-55-45-200-802 05-55-45-200-803 05-55-45-200-804 05-55-45-200-805	1.1	15000 FH	15000 FH	ALL	ALL			
		nctional check of the HIF sistance from the backs		connectors outside	the pressure vess	sel on the right side	of the airplar			
	AIRPLANE I	Resistance Test	d. An alternate for operators o	Bond Resistance Thoosing to utilize the	Test is provided fo ne approved Bond	-200-XXX is the Bo r operators in lieu o d Resistance Test m t or a Bond Resista	f a Loop ethod.			
20-040-04-01	MRB	05-55-42-200-805 05-55-46-200-803	1.1	15000 FH	15000 FH	ALL	ALL			
	Perform a functional check of the Lightning/HIRF protection components outside the pressure vessel in the vertical stabilizer. Check DC resistance from the backshell to ground.									
	AIRPLANE NOTE: Applicable to airplanes with the M2445 Rudder position sensor and M2446 Rudder actuator (servo) installed.									
		preferred metho Resistance Test	d. An alternate for operators o	Bond Resistance Thoosing to utilize the	Test is provided fo ne approved Bond	-200-803 is the Boe r operators in lieu o d Resistance Test m t or a Bond Resista	of a Loop nethod.			
20-040-05-01	MRB	05-55-44-200-806	1.1	2 YR	2 YR	ALL	ALL			
		check the Lightning/HIR				esistance Test) in the	e left wing to			
	AIRPLANE N	NOTE: Applicable to air	planes line nur	nber 1 thru 1856 th	at have not incorp	oorated SB 737-24-	1172.			
20-040-05-02	MRB	05-55-45-200-806	1.1	2 YR	2 YR	ALL	ALL			
	•	check the Lightning/HIR associated with disconne	•			esistance Test) in the	e right wing to			
	AIRPLANE I	NOTE: Applicable to air	planes line nur	nber 1 thru 1856 th	at have not incorp	oorated SB 737-24-	1172.			
20-040-06-01	MRB	05-55-44-200-806	1.1	16 YR	16 YR	ALL	ALL			

Functionally check the Lightning/HIRF protection components (by performing a Loop Resistance Test) in the left wing to body fairing associated with disconnect bracket AC0520 for bond degradation.

AIRPLANE NOTE: Applicable to airplanes line number 1 thru 1856 that have incorporated SB 737-24-1172. Also applicable to airplanes line number 1857 and on.





				APPLICABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
20-040-06-02	MRB	05-55-45-200-806	1.1	16 YR	16 YR	ALL	ALL			
		check the Lightning/HIR associated with disconne				esistance Test) in th	e right wing to			
	AIRPLANE I	NOTE: Applicable to air applicable to air	•	nber 1 thru 1856 th nber 1857 and on.	at have incorpora	ted SB 737-24-117	2. Also			
20-050-00-01	MRB 05-55-15-200-804 1.1 20000 FH 20000 FH ALL ALL									
		etail visual inspection of lonnectors. Look for cond				ssel. During the ins	pection do no			
20-060-00-01	MRB	05-55-43-200-801 05-55-43-200-802 05-55-43-200-803 05-55-43-200-804 05-55-43-200-805 05-55-43-200-807 05-55-43-200-808	1.1	30000 FH	30000 FH	ALL	ALL			
	Functional ch ground.	neck of HIRF/L sensitive	connectors ins	side the pressure ve	essel by DC resis	tance check from b	ackshell to			
20-070-00-01	MRB	05-55-15-200-805	1.1	30000 FH	30000 FH	ALL	ALL			
	Perform a de and signs of	tail visual inspection of t corrosion.	he HIRF/L sen	sitive pig tails insid	e the pressure ve	essel look for condition	ion of security			
20-100-00-01	MRB	05-55-08-200-801 05-55-08-200-802 05-55-08-200-803 05-55-08-200-804 05-55-08-200-806 05-55-08-200-806 05-55-08-200-807 05-55-08-200-808	1.1	15000 FH	15000 FH	ALL	ALL			
	aileron, leadi	neral visual inspection on ng edge flap, spoiler and pack compartment door	d each elevator	surface. Strut to w	ing bonding uses	1 straps per wing.				
20-110-01-01	MRB	05-55-10-200-801	1.1	6000 FH	6000 FH	ALL	ALL			
	General visu	al inspection of external	(cowl open) ha	arness condition an	d security of left of	engine.				
20-110-02-01	MRB	05-55-10-200-801	1.1	6000 FH	6000 FH	ALL	ALL			
	General visu	al inspection of external	(cowl open) ha	arness condition an	d security of right	engine.				
20-120-01-01	MRB	05-55-10-220-801	1.1	15000 FH	15000 FH	ALL	ALL			

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MW0302, MW0303 AND MW0304) on the left engine.





			APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
20-120-02-01	MRB	05-55-10-220-801	1.1	15000 FH	15000 FH	ALL	ALL
		ection of connectors for W0303 AND MW0304) o			ess J5, J6, J7, J8	, J9, CJ9, J10, CJ10	0, MW0301,
20-130-01-01	MRB	05-55-11-200-801	1.1	15000 FH	15000 FH	ALL	ALL
	General visu engine.	al inspection of the two e	engine bonding	g straps related to H	HRF/Lightning for	condition and secu	rity on left
20-130-02-01	MRB	05-55-11-200-801	1.1	15000 FH	15000 FH	ALL	ALL
	General visu engine.	al inspection of the two	engine bonding	g straps related to F	IIRF/Lightning for	condition and secu	rity on right
20-140-00-01	MRB	05-55-42-200-802 05-55-42-200-804 05-55-46-200-801 05-55-46-200-802	1.1	15000 FH	15000 FH	ALL	ALL
	AIRPLANE I	NOTE: Functional check	k using the Loc	n Resistance Test	in AMM 05-55-46	-200-803 is the Boe	
		Resistance Test	for operators of	Bond Resistance Theosing to utilize the	Test is provided for the approved Bond	or operators in lieu of Resistance Test met or a Bond Resista	f a Loop ethod.
20-141-00-01	MRB	Resistance Test	for operators of	Bond Resistance Theosing to utilize the	Test is provided for the approved Bond	or operators in lieu o d Resistance Test m	f a Loop ethod.
20-141-00-01	Functional ch	Resistance Test Please complete	for operators of this functional 1.1 RF protection of	Bond Resistance Tehoosing to utilize the I check using a Locarity	Test is provided for approved Bond pp Resistance Tes 48000 FH	or operators in lieu o d Resistance Test m st or a Bond Resista ALL	f a Loop nethod. nce Test.
20-141-00-01	Functional ch	Resistance Test Please complete 05-55-46-200-804 neck of the Lightning/HIF	for operators of this functional 1.1 RF protection of RF)	Bond Resistance Tehoosing to utilize the length of the len	Test is provided for approved Bond p Resistance Tes 48000 FH ail cone for electrical positions.	or operators in lieu o d Resistance Test m st or a Bond Resista ALL ical bond degradatio	f a Loop nethod. nce Test. ALL on using the
20-141-00-01	Functional ch	Resistance Test Please complete 05-55-46-200-804 neck of the Lightning/HIF ince Tester (LRT). (L/HIF	for operators of this functional 1.1 RF protection of RF)	Bond Resistance Tehoosing to utilize the length of the len	Test is provided for approved Bond p Resistance Tes 48000 FH ail cone for electrical positions.	or operators in lieu o d Resistance Test m st or a Bond Resista ALL ical bond degradatio	f a Loop nethod. nce Test. ALL on using the
	Functional chapter in the second seco	Resistance Test Please complete 05-55-46-200-804 neck of the Lightning/HIF ince Tester (LRT). (L/HIF NOTE: Applicable to air 737-33-1146.	for operators of this functional functional functional functional function cases and function cases function functional function	Bond Resistance Tehoosing to utilize the Icheck using a Local 48000 FH omponents in the temporal and on the Islands of the Islands on the Islands of the Isl	Test is provided for approved Bond p Resistance Test 48000 FH ail cone for electror those airplanes 18000 FH RRA Antenna inst	or operators in lieu of Resistance Test met or a Bond Resistance Test met or a Bond Resistance ALL fical bond degradation that have incorporated ALL	f a Loop nethod. nce Test. ALL on using the ated SB ALL
	Functional ch Loop Resista AIRPLANE I	Resistance Test Please complete 05-55-46-200-804 neck of the Lightning/HIF Ince Tester (LRT). (L/HIF NOTE: Applicable to air 737-33-1146. 05-55-47-200-801 check the Lightning/HIR enna to structure using the complete to air AG723000-40. The complete to air AG72300-40. The complete to air AG723000-40. The complete to air AG723	1.1 RF protection complete line nur 1.1 F protection complete line nur 1.1 F protection complete line nur 1.1 F protection complete line line line line line line line lin	Bond Resistance Tehoosing to utilize the Icheck using a Local 48000 FH components in the temporary and on components in the Litance Tester (LRT).	Test is provided for approved Bond p Resistance Test 48000 FH ail cone for electror those airplanes 18000 FH RRA Antenna installata antenna installata	or operators in lieu of Resistance Test met or a Bond Resistance Test met or a Bond Resistance ALL It ical bond degradation that have incorporate ALL ALL allation for degrada	f a Loop nethod. nce Test. ALL on using the ated SB ALL tion of the bo
	Functional ch Loop Resista AIRPLANE I	Resistance Test Please complete 05-55-46-200-804 neck of the Lightning/HIF Ince Tester (LRT). (L/HIF NOTE: Applicable to air 737-33-1146. 05-55-47-200-801 check the Lightning/HIR enna to structure using the complete to air AG723000-40. The complete to air AG72300-40. The complete to air AG723000-40. The complete to air AG723	1.1 RF protection complete line nur 1.1 F protection complete line nur 1.1 F protection complete line nur 1.1 F protection complete line line line line line line line lin	Bond Resistance Tehoosing to utilize the Icheck using a Local 48000 FH components in the temponents in the Litance Tester (LRT).	Test is provided for approved Bond p Resistance Test 48000 FH ail cone for electror those airplanes 18000 FH RRA Antenna installata antenna installata	ar operators in lieu of Resistance Test met or a Bond Resistance Test met or a Bond Resistance ALL ical bond degradation that have incorporated ALL allation for degradated without Gasket	f a Loop nethod. nce Test. ALL on using the ated SB ALL tion of the bo

between antenna to structure using the Loop Resistance Tester (LRT).

AIRPLANE NOTE: Applicable to airplanes with the S67-2002-18 LRRA antenna and Gasket AG723000-40 installed. The production configuration for L/N 1 through 4306 is with LRRA antenna S67-2002-18 installed without Gasket AG723000-40.





				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
20-142-03-01	MRB	05-55-47-200-801	1.1	40000 FH	40000 FH	ALL	ALL				
		check the Lightning/HIRI enna to structure using the				allation for degradat	tion of the bon				
	AIRPLANE NOTE: Applicable to airplanes with the S67-2002-28 LRRA antenna and Gasket AG723000-40 installed. The production configuration for L/N 4307 and on is with LRRA antenna S67-2002-28 and Gasket AG723000-40 installed.										
20-290-00-01	MRB	05-42-01-100-801 20-60-02-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL				
	Restore (Cle	an) area forward of Nose	Wheel Well.	(EZAP)							
	INTERVAL N	IOTE: Whichever come	s first.								
20-300-00-01	MRB	05-42-01-100-803 20-60-02-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL				
	Restore (Cle	an) area above and outb	oard of Nose	Wheel Well. (EZAF	?)						
	INTERVAL N	IOTE: Whichever come	s first.								
	ACCESS NO	TE: Remove/displace in	nsulation blanl	kets as required.							
20-305-00-01	MRB	05-42-01-211-802 20-60-03-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL				
	Side). (EZAP	ailed) the external power One of the external power OTE: Whichever comes	· ·	and connected EW	/IS in the area out	board of Nose Whe	el Well (Right				
20-310-00-01	MRB	05-42-01-100-802	1.1	18000 FC	18000 FC	ALL	ALL				
20-310-00-01	IVIND	20-60-02-100-801	1.1	6 YR	6 YR	ALL	ALL				
	Restore (Clean) areas behind the equipment racks in the Electrical and Electronics Compartment, and inside the Airstair Compartment (if installed). (EZAP)										
	INTERVAL N	IOTE: Whichever come	s first.								
	ACCESS NO	OTE: Access panel 117E and E4 electronics					E2, E3				
20-320-00-01	MRB	05-42-01-211-801 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Inspect (Deta	ailed) the IDG, APU start	er/generator, b	attery, and externa	al power feeder wi	ring and connected	EWIS. (EZAP				
	INTERVAL N	IOTE: Whichever come	s first.	-			•				
	ACCESS NO	OTE: Access through pa	nels 121JW, 1	21KW, 121LW and	122HW is from F	orward Cargo Com _l	oartment.				
20-325-00-01	MRB	05-42-01-210-803 20-60-04-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
		eral Visual) all exposed ator, battery, and externa	EWIS in the E	ectrical and Electro	onics Compartmer	-	S, APU				
	INTERVAL N	IOTE: Whichever come	s first.								

ACCESS NOTE: Access through panels 121JW, 121KW, 121LW and 122HW is from Forward Cargo Compartment.







		APPLICA	ABILITY								
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
20-330-00-01	MRB	05-42-01-100-807 20-60-02-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Restore (Clean) areas behind ceiling and sidewalls in the Forward Cargo Compartment. (EZAP)										
	INTERVAL NOTE: Whichever comes first.										
	ACCESS NO	OTE: Ceiling and sidewa	all panels remo	val required.							
20-335-00-01	MRB	05-42-01-210-802 20-60-04-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Inspect (Ger	neral Visual) all exposed	EWIS in the F	orward Cargo Com	partment. (EZAP))					
	INTERVAL I	NOTE: Whichever come	s first.								
	ACCESS NOTE: Ceiling and sidewall panels removal required.										
20-340-00-01	MRB	05-42-01-100-806 20-60-02-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Restore (Cle	Restore (Clean) area below Forward Cargo Compartment. (EZAP)									
	INTERVAL NOTE: Whichever comes first.										
	ACCESS NOTE: Center floor panels removal required. Cargo loading system removed/displaced as required.										
20-350-00-01	MRB	05-42-01-100-805 20-60-02-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL				
	Restore (Clean) area aft of Forward Cargo Compartment. (EZAP)										
	INTERVAL I	NOTE: Whichever come	s first.								
20-360-00-01	MRB	05-42-01-210-801	1.1	5500 FC	5500 FC	ALL	ALL				
		20-60-04-100-801	1.2	30 MO	30 MO						
	Inspect (Ger	neral Visual) all exposed	EWIS in the M	ain Landing Gear V	Wheel Well. (EZA	P)					
	INTERVAL I	NOTE: Whichever come	s first.								
	ACCESS NO	ACCESS NOTE: Through main landing gear opening.									
20-370-00-01	MRB	05-42-01-100-804	1.1	18000 FC	18000 FC	ALL	ALL				
	Dantona (OL	20-60-02-100-801	1.2	6 YR	6 YR	(EZAD)					
		ean) areas behind ceiling		paneis in the Aft Ca	rgo Compartment	i. (EZAP)					
		NOTE: Whichever come									
	ACCESS NO	OTE: Ceiling and sidewa	all panels remo	val required.							
20-375-00-01	MRB	05-42-01-210-804 20-60-04-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Inspect (Ger	neral Visual) all exposed	FWIS in the A	ft Cargo Compartm	ent. (FZAP)						

Inspect (General Visual) all exposed EWIS in the Aft Cargo Compartment. (EZAP)

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Ceiling and sidewall panels removal required.







				INTERVAL		APPLICABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGIN			
20-380-00-01	MRB	05-42-01-100-809 20-60-02-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Restore (Clean) area below Aft Cargo Compartment. (EZAP)									
	INTERVAL NOTE: Whichever comes first.									
	ACCESS NO	OTE: Center floor panels required.	s removal requ	ired. Cargo loading	system (if install	ed) removed/displac	ced as			
20-390-00-01	MRB	05-42-01-100-808 20-60-02-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Restore (Cle	an) area in Aft Cargo Co	mpartment. (E	ZAP)						
	INTERVAL N	IOTE: Whichever come	s first.							
	ACCESS NO	OTE: Removal of aft car	go panels at S	ta. 947 bulkhead re	equired.					
20-400-00-01	MRB	05-42-02-100-806 20-60-02-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Restore (Clean) area in Flight Compartment from Sta. 186 to 211, WL 208 to 232. (EZAP)									
	INTERVAL NOTE: Whichever comes first.									
	ACCESS NOTE: Accessible areas forward and above the rudder pedals.									
20-410-00-01	MRB	05-42-02-211-801 20-60-03-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL			
	Inspect (Deta	ailed) all exposed EWIS	in the P5 (fwd/	aft), P6 and P18 pa	anels. (EZAP)					
	INTERVAL N	IOTE: Whichever come	s first.							
	ACCESS NO	OTE: Opening P5 (fwd/a	aft), P6 and P18	3 panels required.						
20-415-00-01	MRB	05-42-02-210-801 20-60-04-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL			
		Inspect (General Visual) all exposed EWIS in the Flight Compartment excluding exposed EWIS in the P5, P6 and P18 panels. (EZAP)								
	INTERVAL N	IOTE: Whichever come	s first.							
	ACCESS NO	OTE: With access provio glareshield, instrur			d access panels,	overhead & sidewal	l panels,			
20-420-00-01	MRB	05-42-02-100-805 20-60-02-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL			
	D t (OL.	` ' '	In a fettional set also con-							

Restore (Clean) areas above ceiling, behind sidewalls and under the raceway cover plates on both sides of the passenger compartment, aft of Flight Compartment to forward entry door. (EZAP)

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: With access provided. Galleys and lavs removed. Ceiling and sidewall panels removal required.





				INTERVAL		APPLICABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE					
20-430-00-01	MRB	05-42-02-100-804 20-60-02-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL					
	`	Restore (Clean) areas above ceiling, and behind sidewalls and return air grilles in Forward Passenger Compartment, Dry Area, Sta. 360 to 663.75. (EZAP)										
	INTERVAL NOTE: Whichever comes first.											
	ACCESS NOTE: Removal of sidewall panels, ceiling panels and return air grilles required.											
20-435-00-01	MRB	05-42-02-100-803 20-60-02-100-801	1.1 1.2	36000 FC 8 YR	36000 FC 8 YR	ALL	ALL					
	,	ean) areas above ceiling, 360 to 663.75. (EZAP)	and behind sid	dewalls and return a	air grilles in Forwa	ard Passenger Com	partment, We					
	INTERVAL N	NOTE: Whichever come	s first.									
	ACCESS NO	OTE: Galleys and lavs re required in areas v		val of sidewall pand and lavs are located		and return air grille	es					
20-440-00-01	MRB	05-42-02-100-802 20-60-02-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL					
	Restore (Clean) areas above ceiling, and behind sidewalls and return air grilles in Aft Passenger Compartment, Dry Area, Sta. 663.75 to Aft Pressure Bulkhead. (EZAP)											
	INTERVAL NOTE: Whichever comes first.											
	ACCESS NO	OTE: Removal of sidewa	all panels, ceili	ng panels and retur	n air grilles requir	ed.						
20-445-00-01	MRB	05-42-02-100-801 20-60-02-100-801	1.1 1.2	36000 FC 8 YR	36000 FC 8 YR	ALL	ALL					
	Restore (Clean) areas above ceiling, and behind sidewalls and return air grilles in Aft Passenger Compartment, Wet Areas, Sta. 663.75 to Aft Pressure Bulkhead. (EZAP)											
	INTERVAL N	NOTE: Whichever come	s first.									
	ACCESS NO	OTE: Galleys and lavs re required in areas v		val of sidewall pand and lavs are located		and return air grille	es					
20-450-00-01	MRB	05-42-03-211-802 20-60-03-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL					
	Inspect (Deta (EZAP)	ailed) the APU starter/ge	nerator power	feeder wiring and c	connected EWIS i	n the area aft of pre	ssure bulkhe					
	INTERVAL N	NOTE: Whichever come	s first.									
20-460-00-01	MRB	05-42-03-211-801 20-60-03-100-801	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL					

INTERVAL NOTE: Whichever comes first.

Compartment. (EZAP)







				INTERVAL		APPLICA	ABILITY		
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
20-465-00-01	MRB	05-42-03-210-801 20-60-04-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL		
	Inspect (Gen	eral Visual) all exposed	EWIS in the Ta	ail Cone. (EZAP)					
	INTERVAL N	NOTE: Whichever come	s first.						
20-470-00-01	MRB	05-42-04-211-801 20-60-03-100-801	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL		
	Inspect (Deta	ailed) the IDG power fee	der wiring and	connected EWIS -	Engine No. 1. (Ez	ZAP)			
	INTERVAL N	NOTE: Whichever come	s first.						
20-480-00-01	MRB	05-42-04-211-802 20-60-03-100-801	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL		
	Inspect (Deta	ailed) the IDG power fee	der wiring and	connected EWIS -	Engine No. 2. (E	ZAP)			
	INTERVAL N	NOTE: Whichever come	s first.						
20-490-00-01	MRB	05-42-04-211-803 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL		
	Inspect (Detailed) all exposed EWIS in the Forward Strut Fairing - Engine No. 1. (EZAP)								
	INTERVAL N	NOTE: Whichever come	s first.						
20-500-00-01	MRB	05-42-04-211-804 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL		
	Inspect (Deta	ailed) all exposed EWIS				<u> </u>			
	. ,	NOTE: Whichever come			,				
20-510-00-01	MRB	05-42-04-211-805 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL		
	Inspect (Deta	ailed) all exposed EWIS	in Strut Torque	Box - Engine No.	1. (EZAP)				
	INTERVAL NOTE: Whichever comes first.								
20-520-00-01	MRB	05-42-04-211-806 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL		
	Inspect (Deta	ailed) all exposed EWIS	in the Forward	Strut Fairing - Eng	gine No. 2. (EZAP)			
	INTERVAL N	NOTE: Whichever come	s first.						
20-530-00-01	MRB	05-42-04-211-807 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL		
	Inanaet (Date	ailed) all expected EMIS	in Fon Coud C	unnort Doom - Eng	ing No. 2 (EZAD)				

Inspect (Detailed) all exposed EWIS in Fan Cowl Support Beam - Engine No. 2. (EZAP)

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
20-540-00-01	MRB	05-42-04-211-808 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Inspect (Detailed) all exposed EWIS in Strut Torque Box - Engine No. 2. (EZAP)									
	INTERVAL I	NOTE: Whichever come	s first.							
20-550-00-01	MRB	05-42-05-211-801 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Inspect (Det	ailed) all exposed EWIS	in the area fro	m Leading Edge to	Front Spar - Left	Wing. (EZAP)				
	INTERVAL I	NOTE: Whichever come	s first.							
	ACCESS NO	OTE: Leading edge flaps	s extended.							
20-560-00-01	MRB	05-42-05-211-802 20-60-03-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL			
	Inspect (Det	ailed) all exposed EWIS	in the area fro	m Leading Edge to	Front Spar - Left	Wing. (EZAP)				
	INTERVAL I	NOTE: Whichever come	s first.							
	ACCESS NOTE: Access is gained by extending slats, and through lower wing surface access panels.									
				_	-					
20-570-00-01	MRB	05-42-05-211-803 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Inspect (Det	ailed) all exposed EWIS	in the area fro	m Rear Spar to Ma	nin Landing Gear S	Support Beam - Left	Wing. (EZAF			
	INTERVAL I	NOTE: Whichever come	s first.							
20-580-00-01	MRB	05-42-05-211-804 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL			
	Inspect (Det	ailed) all exposed EWIS	in the area fro	m Rear Spar to Tra	ailing Edge - Left V	Ving. (EZAP)				
		NOTE: Whichever come			0 0					
	ACCESS NO	OTE: Flaps extended, sp	poilers raised.							
20-590-00-01	MRB	05-42-06-211-801	1.1	18000 FC	18000 FC	ALL	ALL			
		20-60-03-100-801	1.2	6 YR	6 YR	(====)				
	. `	ailed) all exposed EWIS		m Leading Edge to	Front Spar - Righ	it Wing. (EZAP)				
		NOTE: Whichever come								
	ACCESS NO	OTE: Leading edge flaps	s extended.							
20-600-00-01	MRB	05-42-06-211-802 20-60-03-100-801	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL			
	Inancet (Det	siled) all avenued FM/C	in the area from	m Laadina Edaa ta	Front Cnor Digh	+ \\\/:n~ (FZAD)				

Inspect (Detailed) all exposed EWIS in the area from Leading Edge to Front Spar - Right Wing. (EZAP)

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Slats extended.







			APPLICA	ABILITY							
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
20-610-00-01	MRB	05-42-06-211-803 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Inspect (Deta	ailed) all exposed EWIS	in the area fro	m Rear Spar to Mai	in Landing Gear S	Support Beam - Rig	ht Wing. (EZAI				
	INTERVAL NOTE: Whichever comes first.										
20-620-00-01	MRB	05-42-06-211-804 20-60-03-100-801	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Inspect (Deta	ailed) all exposed EWIS	in the area fro	n Rear Spar to Trai	iling Edge - Right	Wing. (EZAP)					
	INTERVAL N	NOTE: Whichever come	s first.								
	ACCESS NO	OTE: Flaps extended, sp	ooilers raised.								
21-010-00-01	MRB	21-25-01-000-801 21-25-01-400-801	1.1	7500 FH	7500 FH	ALL	ALL				
	•	recirculation fan HEPA fi 26 is applicable to all 73	()	7one 125 is annlica	ble to 737-800/90	n aircraft only					
		OTE: Access panel 1220		• • •		·	hle to				
	ACCESS NO	737-800/900 aircra		ne to all 737 NO all c	Statt. Access parte	ει τετένν ιδ αρριίσα	bie to				
21-015-00-01	MRB	21-25-03-200-801	1.1	6000 FH	6000 FH	ALL	ALL				
	Perform a detailed inspection of the Recirculation Fan Check Valve(s) for condition, security, and proper operation.										
21-020-00-01	MRB	21-27-00-890-801	1.1	8000 FH	8000 FH	ALL	ALL				
	Operationally check the alternate E/E cooling supply fan. Note: This task also checks the normal E/E cooling supply fan check valve.										
21-030-00-01	MRB	21-27-00-700-804	1.1	8000 FH	8000 FH	ALL	ALL				
	Operationally	y check the alternate E/E	cooling exhau	ust fan.							
	Note: This ta	sk also checks the norm	al E/E cooling	exhaust fan check	valve.						
21-040-00-01	MRB	21-27-01-000-801 21-27-01-400-801	1.1	7500 FH	7500 FH	ALL	ALL				
	Replace the	E/E cooling supply fan fi	lter.								
21-050-00-01	MRB	21-27-00-700-807	1.1	9000 FH	9000 FH	ALL	ALL				
		y check the equipment co	•	rd exhaust valve, s	upply fans, exhau	ust fans (if applicabl	e), and				
	AIRPLANE I	NOTE: Exhaust fan ope and on, and L/N		is applicable to 737 ave incorporated SE		irplanes Line Numb	er 1701				
21-060-00-01	MRB	21-31-00-710-801	1.1	12000 FH	12000 FH	ALL	ALL				
- 1-000-00-01	רואוו	21-01-00-710-001	1.1	12000111	12000111	ALL	ALL				

 $Operationally\ check\ the\ outflow\ valve\ manual\ mode\ (motor),\ selector\ panel,\ indicator,\ indicator\ feedback\ module.$







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
21-070-00-01	MRB	21-32-01-700-801 21-32-01-700-802	1.1	17000 FH	17000 FH	ALL	ALL
	•	check the positive pressoneach other, either test of		,	ng and the Hamilt	on Sundstrand test	equipment are
21-080-00-01	MRB	21-32-02-000-801 21-32-02-400-801	1.1	20000 FH	20000 FH	ALL	ALL
	Replace the	positive pressure relief v	/alve filters.				
21-090-00-01	MRB	21-32-03-700-801	1.1	10 YR	10 YR	ALL	ALL
	Functionally	check the negative pres	sure relief door	.			
21-100-00-01	MRB	21-51-03-000-801	1.1	2000 FC	2000 FC	ALL	ALL
	Clean the pri	imary and secondary he	at exchangers.				
21-150-00-01	MRB	21-61-06-000-801 21-61-06-100-801 21-61-06-400-801	1.1	1200 FH	1200 FH	ALL	ALL
		lace the cabin temperator has one in the control c		*		control and passen	ger cabins,
21-190-00-01	MRB	36-13-01-020-803 36-13-01-420-803	1.1	6 YR	6 YR	ALL	ALL
		bleed air duct flexible p		ed with part number	r BOE2003-0052	seal.	
23-030-00-01	MRB	23-61-00-760-801	1.1	4 YR	4 YR	ALL	ALL
	Functional cl	heck of resistance of sta	tic discharges.				
23-040-00-01	MRB	23-71-00-710-801	1.1	300 FH	300 FH	ALL	ALL
	Operational	check of the voice record	der and Record	ler Independent Po	ower Supply (RIPS	S) (if installed).	
	INTERVAL N	NOTE: Or national requi	rement.				
23-050-00-01	MRB	23-71-00-730-801	1.1	4 YR	4 YR	ALL	ALL
	-						

Functional check of the voice recorder for audio fidelity.

INTERVAL NOTE: Or national requirement.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
23-060-00-01	MRB	23-71-21-000-801 23-71-21-400-801 23-71-21-700-801 23-71-21-700-802 23-71-21-700-803 23-71-21-700-804	1.1	NOTE		ALL	ALL			
	Operational	check of the ULB at batte	ery replacemer	nt.						
	INTERVAL N	NOTE: At battery replace	ement.							
23-070-00-01	MRB	23-71-21-000-801 23-71-21-400-801 23-71-21-960-801 23-71-21-960-802	1.1	VEN REC		ALL	ALL			
	Replace ULE	B battery at vendor's reco	ommendation.							
	INTERVAL N	NOTE: At vendor's recor	nmendation or	national requireme	ent.					
23-080-00-01	MRB	23-51-00-710-801	1.1	6000 FH	6000 FH	ALL	ALL			
	Operational	check of oxygen mask m	icrophone.							
23-090-00-01	MRB	23-31-00-740-801	1.1	30000 FH	30000 FH	ALL	ALL			
	Operational	Operational check of the passenger address speakers.								
23-100-00-02	MRB	23-24-00-730-802-002	1.1	NOTE		ALL	ALL			
	Operationally	y check the Emergency I	_ocator Transm	nitter (Automatic / F	ixed Type).					
		NOTE: If Installed. NOTE: At Vendor's Reco	ommendation.							
23-110-00-02	MRB	23-24-00-000-802-002 23-24-00-400-802-002	1.1	NOTE		ALL	ALL			
	Discard the E	Emergency Locator Tran	smitter (Autom	atic / Fixed Type) E	Battery.					
	AIRPLANE I	NOTE: If Installed.								
	INTERVAL N	NOTE: At Vendor's Reco	ommendation.							
24-010-01-01	MRB	12-13-21-600-802	1.1	1800 FH	1800 FH	ALL	ALL			
	Change left I	DG oil.								
24-010-02-01	MRB	12-13-21-600-802	1.1	1800 FH	1800 FH	ALL	ALL			
	Change right	t IDG oil.								
24-020-01-01	MRB	12-13-21-200-802	1.1	800 FH	800 FH	ALL	ALL			
	Detailed Insp	pection of left IDG delta F	o indicator.							
24-020-02-01	MRB	12-13-21-200-802	1.1	800 FH	800 FH	ALL	ALL			
	Detailed Insp	pection of right IDG delta	P indicator.							





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
24-030-01-01	MRB	12-13-21-200-801	1.1	800 FH	800 FH	ALL	ALL
	Detailed Insp	pection of left IDG oil leve	l.				
24-030-02-01	MRB	12-13-21-200-801	1.1	800 FH	800 FH	ALL	ALL
	Detailed Insp	pection of right IDG oil lev	el.				
24-040-01-01	MRB	24-11-41-000-801 24-11-41-200-801 24-11-41-400-801	1.1	1800 FH	1800 FH	ALL	ALL
	Replace left	IDG charge and scaveng	e filters.				
24-040-02-01	MRB	24-11-41-000-801 24-11-41-200-801 24-11-41-400-801	1.1	1800 FH	1800 FH	ALL	ALL
	Replace righ	t IDG charge and scaven	ge filters.				
24-050-01-01	MRB	24-11-61-200-801	1.1	3600 FH	3600 FH	ALL	ALL
	Torque chec	k the left engine IDG quic	k attach/detac	ch (QAD) coupling.			
24-050-02-01	MRB	24-11-61-200-801	1.1	3600 FH	3600 FH	ALL	ALL
	Torque chec	k the right engine IDG qu	ick attach/deta	ach (QAD) coupling].		
24-100-00-01	MRB	24-34-00-710-802	1.1	15 DY	15 DY	ALL	ALL
	Operational	check of the standby pow	er control unit	(SPCU).			
	AIRPLANE	NOTE: Applies to airplar	nes with dual b	attery installation o	only.		
24-110-00-01	MRB	24-31-41-710-801	1.1	15000 FC	15000 FC	ALL	ALL
	Check remot	e control circuit breaker.					
	AIRPLANE	NOTE: Applies to airplar	nes with dual b	attery installation of	only.		
24-120-00-01	MRB	24-31-11-000-802-002 24-31-11-400-802-002	1.1	1000 FH	1000 FH	ALL	ALL
	Restore the	main and auxiliary batteri	es.				
	AIRPLANE	NOTE: Applies to airplar	nes with dual b	attery installations	only.		
	INTERVAL N	IOTE: Restore interval for	or 36 AMP/HR	(small) battery is 1	1000 FH. 48 AMP	/HR (large) is 2000	FH.
24-130-00-01	MRB	24-41-11-200-802	1.1	5000 FC	5000 FC	ALL	ALL
	Functional cl	neck of the external power	er receptacle p	ins for excessive w	vear.		
24-140-00-01	MRB	24-41-11-200-804	1.1	5000 FC	5000 FC	ALL	ALL





				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
25-010-01-01	MRB	25-11-01-200-802	1.1	7500 FH	7500 FH	ALL	ALL				
	Inspect (Deta	ailed) the captains seat to	racks and lock	ing mechanism for	wear, condition, a	and security.					
25-010-02-01	MRB	25-11-01-200-802	1.1	7500 FH	7500 FH	ALL	ALL				
	Inspect (Detailed) the first officers seat tracks and locking mechanism for wear, condition, and security.										
25-020-00-01	MRB 25-11-00-200-802 1.1 600 FH 600 FH ALL ALL										
		ailed) the captain, first off belts (as applicable) for		•	bserver (if installe	ed) seat harnesses,	crotch straps				
25-030-00-01	MRB	25-11-00-200-801	1.1	3500 FH	3500 FH	ALL	ALL				
	Operationally	check the captain, first	officer, and the	e first observer seat	harness inertia r	eels.					
25-040-00-01	MRB	25-22-00-200-801	1.1	4000 FH	4000 FH	ALL	ALL				
	Inspect (Deta	ailed) the passenger sea	t belts w/o rem	oval for wear, cond	lition, and securit	y.					
25-045-00-01	MRB	25-22-00-200-804	1.1	12000 FH	12000 FH	ALL	ALL				
	Visually chec	k the passenger compar	rtment seats fo	r proper attachmen	nt.						
25-050-00-01	MRB	25-22-00-710-802	1.1	12000 FH	12000 FH	ALL	ALL				
	Operationally exits.	/ check, where applicable	e, the passeng	er seat break-over	lock out feature of	on seats adjacent to	emergency				
25-070-00-01	MRB	25-22-00-710-803	1.1	6000 FH	6000 FH	ALL	ALL				
	Functionally emergency e	check, where applicable exits.	, the passenge	r seat back recline	restriction mecha	anisms on seats adja	acent to				
25-090-00-01	MRB	25-25-12-200-801	1.1	3000 FH	3000 FH	ALL	ALL				
	Inspect (Detailed) the attendant seat harness and attachments without removal for wear, condition, and security.										
25-100-00-01	MRB	25-25-12-710-802	1.1	3000 FH	3000 FH	ALL	ALL				
	Operationally	check the attendant sea	at harness ine	tia reel lock feature) .						
25-130-00-01	MRB	25-40-08-200-801	1.1	4000 FH	4000 FH	ALL	ALL				
	mechanism f	ailed) the lavatory waste for wear, condition, and s	security.		e waste compartr	ment access door la	tching				
	INTERVAL N	IOTE: FAA AD 74-08-09	≀interval is 100)0 FH.							
25-160-00-01	MRB	25-52-00-200-803	1.1	450 FC	450 FC	ALL	ALL				
	I 1 / O				wall bullthand a	111					

Inspect (General Visual) the fwd cargo compartment floor, ceiling, sidewall, bulkhead, and blowout (pressure relief) panels/liners for holes/tears, condition, and security.





				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
25-160-00-02	MRB	25-52-00-200-803	1.1	450 FC	450 FC	ALL	ALL			
	. ,	neral Visual) the aft cargo for holes/tears, condition	•		vall, bulkhead, an	d blowout (pressure	relief)			
25-170-00-01	MRB	25-52-00-210-801	1.1	3000 FC	3000 FC	ALL	ALL			
	Inspect (Ger	neral Visual) the fwd carg	o door restrain	t system for condit	ion and security.					
25-170-00-02	MRB	25-52-00-210-801	1.1	3000 FC	3000 FC	ALL	ALL			
	Inspect (Ger	neral Visual) the aft cargo	door restraint	system for condition	on and security.					
25-190-01-01	MRB	25-61-10-710-801	1.1	10 YR	10 YR	ALL	ALL			
	Inspect (Deta	ailed) the left emergency	exit hatch esc	ape strap for condi	tion and security.					
25-190-02-01	MRB	25-61-10-710-801	1.1	10 YR	10 YR	ALL	ALL			
	Inspect (Detailed) the right emergency exit hatch escape strap for condition and security.									
25-210-00-01	MRB	25-61-10-210-801	1.1	10 YR	10 YR	ALL	ALL			
	Inspect (Deta	ailed) the flight compartn	nent escape la	nyards (2) for cond	ition and security.					
25-220-00-01	MRB	25-66-00-710-801	1.1	NOTE		ALL	ALL			
	Operational check of the entry and service door mounted emergency escape slide deployment system (on airplane).									
	INTERVAL N	from the operator	nsure its airline r's fleet shall in ght side at eacl	e specific slide mair clude a minimum o n door position, dur	ntenance program of one operational ring each 6 year p	IG entry and service in is adequate. Each check of an installe period. Checks shall or minimum every 6	check d slide alternate			
25-240-00-01	MRB	25-66-01-000-801 25-66-01-000-802 25-66-01-400-801 25-66-01-400-803	1.1	NOTE		ALL	ALL			
	Restore the	fwd entry emergency esc	cape slide at th	e manufacturer's re	ecommended inte	rval.				
	INTERVAL N	NOTE: Vendor Rec								

Restore the fwd galley emergency escape slide at the manufacturer's recommended interval.

INTERVAL NOTE: Vendor Rec







				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
25-240-00-03	MRB	25-66-01-000-801 25-66-01-000-802 25-66-01-400-801 25-66-01-400-803	1.1	NOTE		ALL	ALL			
	Restore the aft entry emergency escape slide at the manufacturer's recommended interval.									
	INTERVAL I	NOTE: Vendor Rec								
25-240-00-04	MRB	25-66-01-000-801 25-66-01-000-802 25-66-01-400-801 25-66-01-400-803	1.1	NOTE		ALL	ALL			
	Restore the	aft galley emergency esc	cape slide at th	e manufacturer's re	commended inte	erval.				
	INTERVAL I	NOTE: Vendor Rec								
25-290-00-01	MRB	25-64-00-900-801	1.1	VEN REC		ALL	ALL			
	Restore the	life jackets (if installed) a	it the manufact	urer's recommende	d interval.					
	INTERVAL I	NOTE: Vendor Rec								
25-330-00-01	MRB	25-64-00-710-801	1.1	2 YR	2 YR	ALL	ALL			
	Operationally check the power megaphones.									
25-340-00-01	MRB	25-64-00-900-804	1.1	VEN REC		ALL	ALL			
	Replace the	power megaphone batte	eries at the mar	nufacturer's recomn	nended interval.					
	INTERVAL I	NOTE: Vendor Rec								
25-350-00-01	MRB	25-64-00-000-801 25-64-00-400-801	1.1	VEN REC		ALL	ALL			
	Functionally	check (off-aircraft) the E	mergency Loca	ator Transmitter (Su	ırvival / Portable	Type) per Vendor's	CMM.			
	AIRPLANE	NOTE: If Installed. Appli	icable to dry ce	II type ELT's only.						
	INTERVAL I	NOTE: At manufacturer's first.	s recommende	d interval or nationa	al regulatory requ	uirement. Whichever	comes			
25-360-00-01	MRB	25-64-00-000-801 25-64-00-400-801	1.1	VEN REC		ALL	ALL			
	Discard the	Emergency Locator Tran	smitter (Surviv	al / Portable Type) l	batteries.					
	AIRPLANE	NOTE: If Installed. Appli	icable to non-d	ry cell type ELT's or	nly.					
	INTERVAL I	NOTE: At manufacturer's first.	s recommende	d interval or nationa	al regulatory requ	uirement. Whichever	comes			
25-370-00-01	MRB	25-64-00-210-802	1.1	2 YR	2 YR	ALL	ALL			
		ck all detachable emerge								

Visually check all detachable emergency equipment (gloves, smoke goggles, crash axe, flashlights, first aid kits, and medical kits, as applicable) for condition and presence.





				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
25-380-00-01	MRB	25-64-00-200-802	1.1	NOTE		ALL	ALL			
	Operationally check the emergency flashlights.									
	INTERVAL N	NOTE: At scheduled bat	tery change or	battery restoration						
25-390-00-01	MRB	25-64-00-900-806	1.1	VEN REC		ALL	ALL			
	Replace the	emergency flashlight ba	tteries at the m	anufacturer's recor	mmended interval					
	AIRPLANE	NOTE: Applicable to air	planes with no	n-rechargeable flas	shlight batteries.					
25-400-00-01	MRB	25-64-00-200-801	1.1	24 MO	24 MO	ALL	ALL			
	Inspect (Deta	ailed) the smoke hoods f	or condition.							
25-410-00-01	MRB	25-64-00-900-807	1.1	VEN REC		ALL	ALL			
	Discard the	smoke hoods at the man	ufacturer's rec	ommended interval	l.					
	INTERVAL N	NOTE: Vendor Rec								
25-420-00-01	MRB	25-64-00-900-808	1.1	VEN REC		ALL	ALL			
	Restore the	first aid kits at the manuf	acturer's recor	nmended interval.						
	INTERVAL N	NOTE: Vendor Rec								
25-430-00-01	MRB	25-64-00-900-809	1.1	VEN REC		ALL	ALL			
	Restore the medical kits at the manufacturer's recommended interval.									
	INTERVAL N	NOTE: Vendor Rec								
26-010-00-01	MRB	26-14-00-730-801	1.1	7500 FH	7500 FH	ALL	ALL			
	Operational	check of the lavatory sm	oke detectors.							
26-020-00-02	MRB	26-14-01-100-802	1.1	7500 FH	7500 FH	ALL	ALL			
	Restore lavatory smoke detector grill by cleaning (Photoelectric type detectors).									
26-030-00-01	MRB	26-15-00-710-801	1.1	12000 FH	12000 FH	ALL	ALL			
	Operational panel).	check of the APU remote	e fire detection	system (M279 fire	detection control	module and P28 rei	mote APU co			
26-050-00-01	MRB	26-20-00-210-801 26-20-00-210-802	1.1	3600 FH	3600 FH	ALL	ALL			
	Visually chec	ck engine fire bottle pres	sure gauge for	correct pressure. (and APU fire bottl	e pressure gauge i	f installed).			
26-070-00-01	MRB	26-21-02-000-801 26-21-02-400-801	1.1	NOTE		ALL	ALL			

Replace the engine fire bottle squib cartridges.

INTERVAL NOTE: AT VENDORS RECOMMENDATION







				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGIN				
26-080-00-01	MRB										
	Functional ch	neck of the engine squib	firing circuit us	sing the engine fire	handle switch.						
26-090-00-01	MRB	26-21-01-000-801	1.1	NOTE		ALL	ALL				
	Operational	26-21-01-400-801 Operational check of the engine fire extinguisher system check valves for freedom of movement.									
	This operation	onal check is accomplish eral motion while listenir	ed at bottle dis	charge or at hydros			noving chec				
	INTERVAL N	IOTE: At bottle change	•								
26-100-00-01	MRB	26-21-01-000-801 26-21-01-400-801	1.1	NOTE		ALL	ALL				
	Functional ch	neck engine fire bottle p	ressure switch	(off aircraft).							
	INTERVAL N	IOTE: At vendor's reco	mmendation.								
26-110-00-01	MRB	26-21-00-720-802	1.1	15000 FH	15000 FH	ALL	ALL				
	Operational of	check of the engine fire	handle (Includi	ng fire handle lock	override circuitry)	for engine shutdow	n and isolat				
26-120-00-01	MRB	26-21-01-000-801	1.1	NOTE		ALL	ALL				
		26-21-01-400-801									
	Functional check the engine fire bottle pressure gauge (off aircraft). INTERVAL NOTE: At vendor's recommendation.										
	INTERVAL N	IOTE: At vendor's reco	mmendation.								
26-130-00-01	MRB	26-21-01-000-801	1.1	NOTE		ALL	ALL				
		26-21-01-400-801									
	Replace the engine fire bottle over pressure relief disc.										
	INTERVAL N	IOTE: At vendor's reco	mmendation.								
26-150-00-01	MRB	26-21-00-720-801	1.1	15000 FH	15000 FH	ALL	ALL				
	Functional ch	26-21-00-730-803 Functional check the engine fire extinguishing distribution system (flow and pressure check).									
		J			·	,					
26-170-00-01	MRB	26-21-01-210-801	1.1	15000 FH	15000 FH	ALL	ALL				
	General visu	al inspection of engine f	ire extinguishe	r distribution systen	n (including exting	guisher bottles).					
26-210-00-01	MRB	26-22-02-000-801	1.1	NOTE		ALL	ALL				
		26-22-02-400-801									
	·	J fire bottle squib.									
	INTERVAL N	IOTE: At vendors recor	nmendation.								
26-220-00-01	MRB	26-22-00-730-801	1.1	15000 FH	15000 FH	ALL	ALL				





				INTERVAL		APPLICA				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
26-230-00-01	MRB 26-22-00-720-801 1.1 15000 FH 15000 FH ALL ALL									
		check of the APU fire han wn and isolation.	ndle (including	fire handle lock over	erride circuitry an	d P28 remote APU	control panel)			
26-250-00-01	MRB	26-22-03-210-801	1.1	15000 FH	15000 FH	ALL	ALL			
	General visu	al inspection of the APU	remote contro	I panel for condition	and security of i	nstallation.				
26-260-00-01	MRB	26-22-01-000-801 26-22-01-400-801	1.1	NOTE		ALL	ALL			
	Functional ch	neck the APU fire bottle p	oressure switcl	h (off aircraft).						
	INTERVAL N	NOTE: At vendor's recor	mmendation.							
26-280-00-01	MRB	26-22-01-210-801	1.1	15000 FH	15000 FH	ALL	ALL			
	Detail visual	inspection of the APU fir	e extinguishin	g distribution syster	n and APU fire bo	ttle.				
26-290-00-01	MRB	26-24-01-900-801	1.1	NOTE		ALL	ALL			
	Inspect (Detailed) lavatory waste compartment fire extinguishing bottles for correct weight.									
	INTERVAL N	NOTE: At vendor recomme	mendation.							
26-300-00-01	MRB	26-24-01-200-801	1.1	7500 FH	7500 FH	ALL	ALL			
	Detail visual	inspection of the lavatory	y fire bottle fus	ible tips and discha	arge tubes.					
26-330-00-01	MRB	26-23-02-000-801 26-23-02-400-801	1.1	LIF LIM		ALL	ALL			
	Replace cargo fire bottle squib.									
	INTERVAL NOTE: At vendors recommendation.									
	ACCESS NO	OTE: Removal of air con	ditioning ducti	ng may be required	l.					
26-340-00-01	MRB	26-23-00-720-801 26-23-00-730-803	1.1	12000 FH	12000 FH	ALL	ALL			
	Functional check cargo fire extinguishing distribution system (flow and pressure check).									
	ACCESS NO	OTE: Removal of air con	ditioning ducti	ng may be required	l.					
26-360-00-01	MRB	26-23-01-000-801-001 26-23-01-400-802-001	1.1	NOTE		ALL	ALL			
	Functional ch	neck the cargo fire bottle	pressure swite	ch (off aircraft).						
	INTERVAL N	NOTE: At vendor's recor	nmendation.							
	ACCESS NO	OTE: Removal of air con	ditioning ducti	ng may be required	l.					
26-370-00-01	MRB	26-23-00-730-801	1.1	7500 FH	7500 FH	ALL	ALL			
	Operationally	y check the cargo fire ext	tinguishing arn	n/discharge switche	es to verify circuitr	V.				

ACCESS NOTE: Removal of air conditioning ducting may be required.







				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
26-390-00-01	MRB	26-23-04-000-801 26-23-04-400-801	1.1	NOTE		ALL	ALL			
	Replace the cargo compartment fire extinguishing system filter/drier.									
	AIRPLANE NOTE: Filter/drier and metering orifice are applicable if second halon bottle installed.									
	INTERVAL N	IOTE: At fire extinguish	er bottle discha	arge.						
	ACCESS NO	OTE: Removal of air con	ditioning ducti	ng may be required	i.					
26-400-00-01	MRB	26-23-00-730-801	1.1	7500 FH	7500 FH	ALL	ALL			
	Functionally	check the cargo fire exti	nguishing time	r.						
	AIRPLANE I	NOTE: If second halon I	oottle installed.							
	ACCESS NO	OTE: Removal of air con	ditioning ducti	ng may be required	i.					
26-450-00-01	MRB	26-26-01-200-801	1.1	NOTE		ALL	ALL			
	Inspect (Deta	ailed) the portable halon	fire extinguish	ers for proper press	sure (if gauge inst	alled), weight, and	condition.			
	INTERVAL N	IOTE: At vendors recon	nmendation.							
26-470-00-01	MRB	26-26-02-200-801	1.1	1 YR	1 YR	ALL	ALL			
	Inspect (deta	illed) the portable water	fire extinguishe	ers for condition.						
26-550-02-01	CMR	05-41-01-211-802	1.1	5500 FC	5500 FC	ALL	ALL			
	vertical struc	etailed inspection of the C tural members. Do the ir sk card satisfies 26-CMF	spection from			he fiberglass panel	between the			
	AIRPLANE NOTE: Applicable to airplanes line number 2093, 2216 and on.									
	INTERVAL N	IOTE: Whichever come	s first.							
27-011-00-01	MRB	27-11-00-210-801	1.1	8000 FH	8000 FH	ALL	ALL			
	General visual inspection of the forward aileron mechanical components.									
27-012-00-01	MRB	27-11-61-210-801	1.1	15000 FH	15000 FH	ALL	ALL			
	Perform a ge	eneral visual inspection o	of the aileron tr	ansfer mechanism.						
27-013-01-01	MRB	27-11-00-210-802	1.1	8000 FH	8000 FH	ALL	ALL			
	-	eneral visual inspection of spoiler mechanical conf	-	aileron mechanical	components from	n the aileron PCU's	to the aileror			
27-013-02-01	MRB	27-11-00-210-802	1.1	8000 FH	8000 FH	ALL	ALL			
		eneral visual inspection of								

Perform a general visual inspection of the right wing aileron mechanical components from the aileron PCU's to the aileron and the flight spoiler mechanical control path.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
27-016-00-01	MRB										
	Operationally check the aileron spring cartridge and transfer mechanism.										
27-018-00-01	MRB	27-11-00-700-806	1.1	25000 FH	25000 FH	ALL	ALL				
	Functionally	check the aileron spring	cartridge and	transfer mechanism	n.						
27-022-00-01	MRB	27-11-00-720-801	1.1	25000 FH	25000 FH	ALL	ALL				
	Functionally	check the force required	I to extend and	collapse the A and	B system aileron	power control unit	input pogo's.				
27-024-00-01	MRB	29-00-00-790-809	1.1	25000 FH	25000 FH	ALL	ALL				
	Functionally	check the A and B syste	m aileron powe	er control unit interr	nal leakage in a lo	paded condition.					
27-026-01-01	MRB	12-22-11-600-801 12-22-11-640-801 12-22-11-640-802 12-22-11-640-803 12-22-11-640-804	1.1	4000 FH	4000 FH	ALL	ALL				
	Lubricate the	Lubricate the left wing aileron mechanical control path and aileron power control units.									
27-026-02-01	MRB	not using BMS 3-3 12-22-11-600-801 12-22-11-640-801 12-22-11-640-802 12-22-11-640-803	1.1	4000 FH	4000 FH	ALL	ALL				
	Lubricate the right wing aileron mechanical control path.										
	SPECIAL NO	OTE: CMR Task (27-CM airplanes using BI not using BMS 3-3	MS 3-33 Greas	e and 3,000 FH / 9		(whichever comes er comes first) for a	,				
27-028-00-01	MRB	27-11-81-210-801	1.1	15000 FH	15000 FH	ALL	ALL				
	Perform a de	etail visual inspection of	the aileron feel	and centering sprii	ngs.						
27-030-00-01	MRB	27-11-00-700-807	1.1	15000 FH	15000 FH	ALL	ALL				
	Operationally	/ check, hydraulic power	r off, the aileror	control surfaces fo	or full range of tra	vel and freedom of	movement.				
27-032-00-01	MRB	27-11-00-700-803	1.1	25000 FH	25000 FH	ALL	ALL				
	Functionally	check the aileron systen	n control wheel	forces.							
27-033-00-01	MRB	27-09-91-200-806	1.1	8000 FH	8000 FH	ALL	ALL				

Functionally check the left wing aileron tab freeplay.

SPECIAL NOTE: CMR task (27-CMR-12) interval for this task is 8,000 FH / 24 months, whichever comes first. See MPD Section 9.







				INTERVAL	APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
27-033-00-02	MRB	27-09-91-200-806	1.1	8000 FH	8000 FH	ALL	ALL			
	Functionally check the right wing aileron tab freeplay.									
	SPECIAL NO	OTE: CMR task (27-CM MPD Section 9.	R-12) interval	or this task is 8,000	0 FH / 24 months	, whichever comes t	irst. See			
27-034-01-01	MRB	27-11-31-210-801	1.1 1.2	8000 FH 3 YR	8000 FH 3 YR	ALL	ALL			
	Perform a de	etail visual inspection of	he left wing ail	eron balance bay s	eals.					
	INTERVAL N	IOTE: Whichever occur	s first.							
27-034-02-01	MRB	27-11-31-210-801	1.1 1.2	8000 FH 3 YR	8000 FH 3 YR	ALL	ALL			
	Perform a de	etail visual inspection of t	he right wing a	ileron balance bay	seals.					
	INTERVAL N	IOTE: Whichever occur	s first.							
27-035-00-01	MRB	27-09-91-200-801	1.1	16000 FH	16000 FH	ALL	ALL			
27-000-01	Functionally check the left wing aileron surface freeplay.									
	ranouonany	oneon the feet wing allon	on canaco nec	piay.						
27-035-00-02	MRB	27-09-91-200-801	1.1	16000 FH	16000 FH	ALL	ALL			
	Functionally check the right wing aileron Surface freeplay.									
27-036-00-01	MRB	27-21-61-210-801	1.1	12000 FH	12000 FH	ALL	ALL			
	Detail Inspection of Aft Rudder Quadrant, Torque Tube Assembly, Feel and Centering Unit, and associated Input/Output Rods.									
27-038-00-01	MRB	27-21-51-210-801	1.1	15000 FH	15000 FH	ALL	ALL			
	General visual inspection of the rudder forward mechanical control path.									
27-040-00-01	MRB	12-22-21-600-802	1.1	6000 FH	6000 FH	ALL	ALL			
	Lubricate the	rudder feel and centerion	ng unit spring s	lider.						
27-041-00-01	MRB	27-21-00-700-824-002	1.1	10000 FH	10000 FH	ALL	ALL			
		check the main rudder				7.22				
		NOTE: Applicable to air incorporation of	planes line nur	nber 596; and 1268	3 and on; and L/N	1-595 and 597-126	7 with			
27-043-00-01	MRB	27-21-00-700-823-002	1.1	12500 FH	12500 FH	ALL	ALL			
	Operationally	check the rudder PCU	overrides.							

AIRPLANE NOTE: Applicable to airplanes line number 596; and 1268 and on; and L/N 1-595 and 597-1267 with incorporation of SB 737-27-1253.







				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
27-046-00-01	MRB	12-22-21-600-801 12-22-21-640-801	1.1	6000 FH	6000 FH	ALL	ALL			
	Lubricate the main and standby rudder power control unit rod ends and rudder hinges.									
27-047-00-01	MRB	27-21-00-700-822-002	1.1	10000 FH	10000 FH	ALL	ALL			
	Functionally	y check the rudder power of	control unit int	ernal leakage in a	loaded condition.					
	SPECIAL N	IOTE: CMR Task (27-CMF	R-10) interval	for this task is 10,0	000 FH. See MPD	Section 9.				
	AIRPLANE	NOTE: Applicable to airp incorporation of S			8 and on; and L/N	1-595 and 597-126	37 with			
	INTERVAL	NOTE: MSG-3 analysis for Section 9.	or this task is	12500 FH. CMR in	terval for this task	is 10000 FH. See N	MPD			
27-048-00-01	MRB	27-21-00-700-822-002	1.1	10000 FH	10000 FH	ALL	ALL			
		y check the rudder PCU releask is satisfied by accomp	•	,	ng rudder PCU int	ernal leakage.				
	AIRPLANE	NOTE: Applicable to airp incorporation of S		•	8 and on; and L/N	1-595 and 597-126	37 with			
27-054-00-01	MRB	29-00-00-790-808	1.1	25000 FH	25000 FH	ALL	ALL			
	Functionally check the standby rudder power control unit for internal leakage in a loaded condition.									
27-056-00-01	MRB	27-21-24-210-801	1.1	5000 FH	5000 FH	ALL	ALL			
	Perform a general visual inspection of the standby rudder power control unit with hydraulic power on.									
27-058-00-01	MRB	27-21-24-210-802	1.1	15000 FH	15000 FH	ALL	ALL			
	Perform a control to rudder su	letail visual inspection of thurface).	ne standby rud	dder power control	unit fore and aft a	ttachment points (s	tructure to PC			
27-060-00-02	MRB	27-21-00-700-820-002	1.1	15000 FH	15000 FH	ALL	ALL			
	Operationa	lly check the wheel to rudd	er interconne	ct system (WTRIS)						
27-062-00-02	MRB	27-21-00-700-819-002	1.1	15000 FH	15000 FH	ALL	ALL			
	Operationa	lly check the rudder trim th	rough the full	range of movemer	nt.					
27-064-00-02	MRB	27-21-00-700-814-002	1.1	16000 FH	16000 FH	ALL	ALL			
	Functionally	y check the rudder surface	freeplay.							
27-068-00-01	MRB	27-31-20-740-801	1.1	15000 FH	15000 FH	ALL	ALL			
	Operations	U	- h:ft f ti							

Operationally check the elevator feel shift function.

AIRPLANE NOTE: If Installed







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-070-00-01	MRB	12-22-31-600-801 12-22-31-600-802 12-22-31-640-801 12-22-31-640-802	1.1	6000 FH	6000 FH	ALL	ALL
	Lubricate the	left elevator mechanica	I control path.				
27-070-00-02	MRB	12-22-31-600-801	1.1	6000 FH	6000 FH	ALL	ALL
27-070-00-02	WIND	12-22-31-600-801 12-22-31-600-802 12-22-31-640-801 12-22-31-640-802	1.1	0000111	0000 FTT	ALL	ALL
	Lubricate the	right elevator mechanic	al control path				
27-073-00-01	MRB	27-31-61-210-801	1.1	15000 FH	15000 FH	ALL	ALL
	Perform a ge	neral visual inspection o	of the elevator r	mechanical control	path.		
27-074-00-01	MRB	27-31-00-700-807	1.1	15000 FH	15000 FH	ALL	ALL
	Operationally	check, hydraulic power	off, the elevate	or control surfaces	for full range of tr	avel and freedom of	f movement.
27-075-01-01	MRB	27-31-34-210-801	1.1	7500 FH	7500 FH	ALL	ALL
	Perform a ge	neral visual inspection o	of the left eleva	tor balance weight	installation and e	evator tab control m	nechanism.
27-075-02-01	MRB	27-31-34-210-801	1.1	7500 FH	7500 FH	ALL	ALL
	Perform a ge	neral visual inspection o	of the right elev	ator balance weigh	t installation and	elevator tab control	mechanism.
27-076-00-01	MRB	27-31-14-210-801	1.1	12000 FH	12000 FH	ALL	ALL
	Perform a ge	neral visual inspection o	of the elevator p	power control units	with hydraulic po	wer on.	
27-078-00-01	MRB	29-00-00-790-809	1.1	25000 FH	25000 FH	ALL	ALL
	Functionally	check the A and B syste	m elevator pov	ver control unit for i	nternal leakage ir	a loaded condition	
27-080-00-01	MRB	27-31-17-200-801 27-31-17-790-801	1.1	14000 FH	14000 FH	ALL	ALL
	Drain and lea	ak check the elevator pite	ot-static systen	1.			
27-084-00-01	MRB	27-31-00-720-801	1.1	25000 FH	25000 FH	ALL	ALL
	Functionally	check the force necessa	ry to breakout	the elevator contro	l column override	assembly.	
27-086-00-01	MRB	27-31-00-700-815	1.1	6000 FH	6000 FH	ALL	ALL
	Operationally	check the elevator tab	control system				
27-088-00-01	MRB	27-32-00-710-801 27-32-00-740-803	1.1	15000 FH	15000 FH	ALL	ALL





				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
27-092-00-01	MRB	27-31-00-820-809	1.1	25000 FH	25000 FH	ALL	ALL				
	Functionally	check the force necessa	ry to collapse	and extend the elev	ator input rod po	go's.					
27-093-00-01	MRB	1.2 4000 FH 4000 FH									
	Perform a detailed visual inspection of the left elevator tab and left elevator tab mechanism.										
	SPECIAL NO	DTE: CMR Task (27-CM MPD Section 9.	IR-07) interval	for this task is 2,00	0 CYC / 4,000 FH	H, whichever comes	first. See				
	AIRPLANE I	NOTE: Applicable to airg		· ·	and on; and L/N	1-595 and 597-117	4 that				
	INTERVAL N	IOTE: Whichever come	s first.								
27-093-00-02	MRB	27-31-00-220-802	1.1 1.2	2000 FC 4000 FH	2000 FC 4000 FH	ALL	ALL				
	Perform a de	tailed visual inspection o				anism.					
		OTE: CMR Task (27-CM	Ü	· ·			first. See				
		MPD Section 9.	,								
	AIRPLANE I	NOTE: Applicable to airguize have incorporate		·	and on; and L/N	1-595 and 597-117	4 that				
	INTERVAL N	IOTE: Whichever come	s first.								
27-094-00-01	MRB	22-11-26-710-801	1.1	15000 FH	15000 FH	ALL	ALL				
	Perform a BI	TE check of the elevator	autopilot serv	o pressure regulato	or.						
27-098-01-01	MRB	27-31-32-200-804	1.1	8000 FH	8000 FH	ALL	ALL				
	Functionally	check the left and right e	elevator surfac	e freeplay.							
27-099-00-01	MRB	27-31-32-200-805	1.1	2000 FC	2000 FC	ALL	ALL				
			1.2	4000 FH	4000 FH						
	•	check the elevator tab fr									
	SPECIAL NOTE: CMR Task (27-CMR-08) interval for this task is 2,000 CYC / 4,000 FH, whichever comes first. See MPD Section 9.										
	AIRPLANE I	NOTE: Applicable to all SB 737-55-1081	•	pt 737-900 line nur	mber 683 to 1174	that have not incorp	oorated				
	INTERVAL N	IOTE: Whichever come	s first.								
27-099-00-02	MRB	27-31-32-200-805	1.1 1.2	2000 FC 4000 FH	2000 FC 4000 FH	ALL	ALL				
	Functionally	check the elevator tab fr									
	•	OTE: CMR Task (27-CM		for this task is 2 00	0 CYC / 4 000 FF	L whichever comes	first. See				
	JI LUIAL III	MPD Section 9.	voj intorvar	10. 1110 1431 15 2,00	5 5 1 5 7 7,000 1 1	.,					
	AIRPLANE I	NOTE: Applicable to all	airplanes exce	pt 737-900 line nur	mber 683 to 1174	that have not incom	oorated				
		SB 737-55-1081		-		·					

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INTERVAL NOTE: Whichever comes first.





				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
27-100-00-01	MRB	27-31-00-200-801	1.1	6 YR	6 YR	ALL	ALL				
	Perform a detailed visual inspection of the elevator push rods (between output torque tube and control surface) and attachment bolts.										
27-101-00-01	MRB	27-31-00-200-802	1.1	6 YR	6 YR	ALL	ALL				
	Perform a de	etailed visual inspection of	of the single ele	ement dual load pa	th feel and center	ing unit output rod.					
27-102-00-01	MRB	12-22-41-600-801	1.1 1.2	1600 FH 1 YR	1600 FH 1 YR	ALL	ALL				
	Lubricate the	stabilizer trim actuator	and actuator gi	mbal pins and ballr	nut.						
	INTERVAL N	NOTE: Whichever come	s first.								
27-104-00-01	MRB	12-22-41-600-802	1.1	25000 FH	25000 FH	ALL	ALL				
	Lubricate the	e forward stabilizer trim n	nechanism driv	re train.							
27-106-00-01	MRB	27-41-00-700-808	1.1	15000 FH	15000 FH	ALL	ALL				
	Functionally	check the secondary sta	abilizer trim bra	ke.							
27-107-00-01	MRB	12-22-41-610-802	1.1	7500 FH	7500 FH	ALL	ALL				
	Service the H	HORIZONTAL TRIM ACT	TUATOR BRAK	Œ.							
	AIRPLANE I	NOTE: Applicable to air Applicable to air	•	nber 350 and on. nber 1 to 349 that h	nave incorporated	I SB 737-27-1210.					
27-108-00-01	MRB	27-41-81-000-801 27-41-81-400-801	1.1	25000 FH	25000 FH	ALL	ALL				
	Remove the	stabilizer trim actuator for	or restoration.								
	SPECIAL NO	OTE: CMR Task (27-CM	1R-02) interval	for this task is 4,50	0 FH. See MPD S	Section 9.					
	AIRPLANE I	NOTE: For 251A4510-4 For 251A4510-5		ore both primary an ore only the second	•		rakes.				
	INTERVAL N	NOTE: For 251A4510-6, -5 actuators is 45		actuators, interval	is 25000FH. CMF	R interval for 251A4	510-4 and				
27-110-00-01	MRB	27-41-81-210-801	1.1 1.2	6400 FH 2 YR	6400 FH 2 YR	ALL	ALL				
	Perform deta	ail visual inspection of the	e stabilizer trim	jackscrew, ballnut	, ballnut return tub	pes, and the upper a	and lower gin				
	INTERVAL N	NOTE: Whichever come	s first.								
27-112-00-01	MRB	27-41-00-700-805	1.1	15000 FH	15000 FH	ALL	ALL				
			n limit switches	(flaps up and flaps	down).						
	Operationally	y check the stabilizer trin									
27-114-00-01	Operationally MRB	27-41-00-710-801	1.1	6000 FH	6000 FH	ALL	ALL				





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-116-00-01	MRB	27-41-91-700-801	1.1	12000 FH	12000 FH	ALL	ALL
	Operationally	y check the Main Electric	: Horizontal Sta	abilizer Trim Cutout	Switch (Control S	Stand).	
27-118-00-01	MRB	27-41-00-700-807	1.1	15000 FH	15000 FH	ALL	ALL
27-110-00-01					13000 FTT	ALL	ALL
	Operationally	y check the aisle stand s	tabilizer trim ov	vernue switch.			
27-120-00-01	MRB	27-41-41-210-801	1.1	15000 FH	15000 FH	ALL	ALL
	Detail visual attachment p	inspection of the upper a points.	and lower stabi	lizer trim forward n	nechanism retenti	on turnbuckles and	turnbuckle
27-121-00-01	MRB	27-41-41-210-803 27-62-00-210-801	1.1	15000 FH	15000 FH	ALL	ALL
	U	eneral visual inspection c electric actuator, actuato				orake lever assemb	ly and auto
		07.44.07.700.004	1.1	15000 FH	15000 FH	ALL	ALL
27-122-00-01	MRB	27-41-97-700-801					
27-122-00-01	-	y check the control colun					
27-132-00-01	Operationally	y check the control colun					
	Operationally stabilizer mo	y check the control colun vement.	nn stabilizer trii	m arm and direction	nal switches for m	ovement of a single	e switch to cau
	Operationally stabilizer mo	y check the control colunvement. 12-22-51-610-801	nn stabilizer trii	m arm and direction	nal switches for m	ovement of a single	e switch to cau
27-132-00-01	Operationally stabilizer mo	y check the control colunvement. 12-22-51-610-801 ower drive unit oil level a	nn stabilizer trii 1.1 and service as	m arm and direction 5000 FC required.	nal switches for m	ovement of a single	e switch to cau
27-132-00-01	Operationally stabilizer mo	y check the control colunvement. 12-22-51-610-801 ower drive unit oil level a	nn stabilizer trii 1.1 and service as	m arm and direction 5000 FC required.	nal switches for m	ovement of a single	e switch to cau
27-132-00-01 27-134-00-01	Operationally stabilizer moderate moder	y check the control column vement. 12-22-51-610-801 ower drive unit oil level a 12-22-51-610-802 flap power drive unit oil. 12-22-51-640-807 12-22-51-640-808 12-22-51-640-809	1.1 and service as 1.1 1.1 1.1	5000 FC required. 25000 FC 4000 FH 24 MO	5000 FC 25000 FC 4000 FH	ovement of a single ALL ALL	ALL
27-132-00-01 27-134-00-01	Operationally stabilizer modern stabilizer moder	y check the control colunvement. 12-22-51-610-801 ower drive unit oil level a 12-22-51-610-802 flap power drive unit oil. 12-22-51-640-807 12-22-51-640-808 12-22-51-640-809 12-22-51-640-810	1.1 and service as 1.1 1.1 1.2 sor mechanism	5000 FC required. 25000 FC 4000 FH 24 MO	5000 FC 25000 FC 4000 FH	ovement of a single ALL ALL	ALL
27-132-00-01 27-134-00-01	MRB Check flap p MRB Replace the	12-22-51-610-802 flap power drive unit oil. 12-22-51-640-807 12-22-51-640-808 12-22-51-640-809 12-22-51-640-810 e left wing flap skew sense	1.1 and service as 1.1 1.1 1.2 sor mechanism	5000 FC required. 25000 FC 4000 FH 24 MO	5000 FC 25000 FC 4000 FH	ovement of a single ALL ALL	ALL

Lubricate the right wing flap skew sensor mechanism.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Flaps deployed.





TASK CARD NO. 27-138-00-01	SOURCE	AMM TASK REF	1/5501011						
27-138-00-01		7 (IIIII) 17 (OTT TCE	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
27-130-00-01	MRB	12-22-51-640-801	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL		
	Lubricate the	left wing flap drive torqu	ue tube suppor	ts and couplings.					
	INTERVAL NOTE: Whichever occurs first.								
	ACCESS NO	TE: Flaps deployed.							
27-138-00-02	MRB	12-22-51-640-801	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL		
	Lubricate the	right wing flap drive tore	que tube suppo	orts and couplings.					
	INTERVAL N	IOTE: Whichever occur	s first.						
	ACCESS NO	TE: Flaps deployed.							
27-140-01-01	MRB	12-22-51-610-803	1.1	5000 FC	5000 FC	ALL	ALL		
	Check the let	ft wing flap drive transmi	ssion oil level a	and service as requ	uired.				
	ACCESS NO	TE: Flaps deployed.		·					
27-140-02-01	MRB	12-22-51-610-803	1.1	5000 FC	5000 FC	ALL	ALL		
	Check the rig	ıht wing flap drive transn	nission oil level	l and service as red	quired.				
	ACCESS NO	TE: Flaps deployed.							
27-142-01-01	MRB	12-22-51-610-804	1.1	25000 FC	25000 FC	ALL	ALL		
	Replace the	left wing flap drive transı	mission oil.						
	ACCESS NO	OTE: Flaps deployed.							
27-142-02-01	MRB	12-22-51-610-804	1.1	25000 FC	25000 FC	ALL	ALL		
	Replace the	right wing flap drive tran	smission oil.						
	ACCESS NO	OTE: Flaps deployed.							
27-144-00-01	MRB	12-22-51-640-802 12-22-51-640-803 12-22-51-640-804 12-22-51-640-805	1.1	1000 FC	1000 FC	ALL	ALL		
	Lubricate the	left wing trailing edge fl	ap ballscrew as	ssemblies and flap	transmission univ	versal joints.			
		OTE: Flaps deployed.	-	·		-			
27-144-00-02	MRB	12-22-51-640-802 12-22-51-640-803 12-22-51-640-804	1.1	1000 FC	1000 FC	ALL	ALL		
	Ludada - t - 0	12-22-51-640-805	flan hallerer	annumble and 6		hans all labots			

Lubricate the right wing trailing edge flap ballscrew assemblies and flap transmission universal joints.

ACCESS NOTE: Flaps deployed.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
27-148-01-01	MRB	27-51-32-220-801 27-51-32-220-802 27-51-42-220-801 27-51-42-220-802	1.1	25000 FC	25000 FC	ALL	ALL			
	Functionally check the left wing trailing edge flaps ballscrew assembly backlash.									
	ACCESS NO	OTE: Flaps deployed.								
27-148-02-01	MRB	27-51-32-220-801 27-51-32-220-802 27-51-42-220-801 27-51-42-220-802	1.1	25000 FC	25000 FC	ALL	ALL			
	Functionally	check the right wing trail	ing edge flaps	ballscrew assembl	y backlash.					
	ACCESS NO	OTE: Flaps deployed.								
27-148-03-01	MPD	27-51-32-200-803 27-51-32-200-804 27-51-42-200-801 27-51-42-200-802	1.1	13200 FC	6600 FC	ALL	ALL			
	Detailed Insp	pection of the left wing tra	ailing edge flap	ballscrew actuator	r for grease leaka	ge, wear and condi	tion			
27-148-04-01	MPD	27-51-32-200-803 27-51-32-200-804 27-51-42-200-801 27-51-42-200-802	1.1	13200 FC	6600 FC	ALL	ALL			
	Detailed Insp	pection of the right wing	trailing edge fla	ap ballscrew actuate	or for grease leak	age, wear and cond	dition			
27-152-01-01	MRB	12-22-51-640-806	1.1 1.2	2000 FC 12 MO	2000 FC 12 MO	ALL	ALL			
	Lubricate the	#4 flap transmission an	gle/tee gearbo	x universal joints.						
	INTERVAL N	NOTE: Whichever come	s first.							
27-152-02-01	MRB	12-22-51-640-806	1.1 1.2	2000 FC 12 MO	2000 FC 12 MO	ALL	ALL			
		e #5 flap transmission an	_	x universal joints.						
27-154-00-01	MRB	27-51-00-740-803	1.1	5000 FH	5000 FH	ALL	ALL			
	Operationally	y check the flap load reli	ef system.							
27-156-00-01	MRB	27-51-00-720-804	1.1	25000 FH	25000 FH	ALL	ALL			
	Functionally	check the flap load relief	f system.							
27-158-00-01	MRB	27-51-00-720-803	1.1	15000 FH	15000 FH	ALL	ALL			
	Francisco II	alara la tha a alta ma a t								

Functionally check the alternate flap drive system.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-162-00-01	MRB	27-51-00-740-802	1.1	750 FH	750 FH	ALL	ALL
	Operationally (FSEU).	rcheck the flap skew an	d flap asymme	try systems by initia	ating a BITE ched	k of the Flap/Slat E	lectronics Un
27-164-00-01	MRB	27-51-00-740-801	1.1	15000 FH	15000 FH	ALL	ALL
	Operationally unit.	check the flap uncomm	anded motion	protection system l	by initiating a bite	check of the flap sl	at electronics
27-166-00-01	MRB	27-51-06-210-801	1.1	25000 FH	25000 FH	ALL	ALL
	Perform a de	tail visual inspection of t	he trailing edge	e flap lever sensor,	linkage, rod and	rod ends.	
27-168-01-01	MRB	27-51-00-710-802	1.1	25000 FC	25000 FC	ALL	ALL
	Operationally	check the left wing trail	ing edge flap tı	ransmission no-bac	k brakes.		
	ACCESS NO	OTE: Flaps deployed.					
27-168-02-01	MRB	27-51-00-710-802	1.1	25000 FC	25000 FC	ALL	ALL
	Operationally	check the right wing tra	iling edge flap	transmission no-ba	ick brakes.		
	ACCESS NO	OTE: Flaps deployed.					
27-170-01-01	MRB	12-22-51-640-811 12-22-51-640-812	1.1 1.2	1250 FC 8 MO	1250 FC 8 MO	ALL	ALL
	Lubricate the	e left wing trailing edge fla	ap actuation m	echanism.			
	INTERVAL N	IOTE: Whichever occur	s first.				
	ACCESS NO	OTE: Flaps deployed.					
27-170-02-01	MRB	12-22-51-640-811 12-22-51-640-812	1.1 1.2	1250 FC 8 MO	1250 FC 8 MO	ALL	ALL
	Lubricate the	right wing trailing edge	flap actuation i	mechanism.			
	INTERVAL N	IOTE: Whichever occur	s first.				
	ACCESS NO	OTE: Flaps extended.					
27-171-01-01	MRB	27-51-00-210-801	1.1	12000 FH	12000 FH	ALL	ALL
		eneral visual inspection on the couplings a				_G beam angle gea	r box, flap dri
	•	DTE: Flaps deployed.	1 1 2 2 2 2 2	,			
27-171-02-01	MRB	27-51-00-210-801	1.1	12000 FH	12000 FH	ALL	ALL
		eneral visual inspection o					

ACCESS NOTE: Flaps deployed.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
27-172-01-01	MRB	27-51-00-210-802	1.1	12000 FC	12000 FC	ALL	ALL			
	Perform a detail visual inspection of the left wing trailing edge flap actuation mechanism.									
	ACCESS NOTE: Flaps deployed.									
27-172-02-01	MRB	27-51-00-210-802	1.1	12000 FC	12000 FC	ALL	ALL			
	Perform a de	etail visual inspection of t	the right wing to	railing edge flap ac	tuation mechanisi	m.				
	ACCESS NO	OTE: Flaps deployed.								
27-174-01-01	MRB	12-22-51-640-814 12-22-51-640-815 12-22-51-640-816	1.1	4000 FC	4000 FC	ALL	ALL			
	Lubricate the	e left wing #'s 1, 2, and 3	trailing edge fl	ap track forward at	tachment point pi	ns.				
27-174-02-01	MRB	12-22-51-640-814 12-22-51-640-815 12-22-51-640-816	1.1	4000 FC	4000 FC	ALL	ALL			
	Lubricate the	e right wing #'s 6, 7, and	8 trailing edge	flap track forward a	attachment point p	oins.				
27-176-01-01	MRB	12-22-51-640-813	1.1	1000 FC	1000 FC	ALL	ALL			
	Lubricate the	e left wing #4 inboard flap	o track attachm	ent fittings.						
27-176-02-01	MRB	12-22-51-640-813	1.1	1000 FC	1000 FC	ALL	ALL			
	Lubricate the	e right wing #5 inboard fla	ap track attach	ment fittings.						
27-178-00-01	MRB	12-22-51-610-805	1.1	7500 FH	7500 FH	ALL	ALL			
	Check altern	ate flap drive gearbox oi	l level and serv	vice as required.						
27-182-00-01	MRB	12-22-61-600-801	1.1	4000 FH	4000 FH	ALL	ALL			
	Lubricate the	e spoiler mixer.								
27-182-01-01	MRB	12-22-61-600-802 12-22-61-640-801	1.1	4000 FH	4000 FH	ALL	ALL			
	Lubricate the	e left wing spoiler mecha	nical control pa	ath.						
	ACCESS NO	OTE: Flaps deployed.								
27-182-02-01	MRB	12-22-61-600-802 12-22-61-640-801	1.1	4000 FH	4000 FH	ALL	ALL			
		e right wing spoiler mech	anical control p	oath.						
27-184-00-01	MRB	27-61-00-820-809	1.1	22400 FH	22400 FH	ALL	ALL			
	Eupotionally	check the torque of the	anailar ratio ab	anger ne beek eee	a ma hala r					

Functionally check the torque of the spoiler ratio changer no-back assembly.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-186-00-01	MRB	12-22-81-600-801	1.1	6000 FH	6000 FH	ALL	ALL
	Lubricate the	speedbrake lever no-ba	ack brake.				
27-187-00-01	MSG3	27-62-00-710-802	1.1	6000 FH	6000 FH	ALL	ALL
	Operationally	Check the Speedbrake	Handle Stop				
	AIRPLANE I	NOTE: Applicable to 90	0ER and airpla	nes with Short Fiel	d Performance Pa	ackage (if installed).	
27-188-00-02	MRB	27-62-00-820-809	1.1	15000 FH	15000 FH	ALL	ALL
	Operationally	check the speedbrake	refused takeoff	(RTO) system.			
	Note: This ta	sk is applicable to airpla	nes with Short	Field Performance	Package (if insta	lled).	
27-190-00-01	MRB	27-62-00-820-810	1.1	11000 FH	11000 FH	ALL	ALL
	Operationally	check the speedbrakes	s extended ligh	t.			
27-192-01-01	MRB	27-61-00-820-810	1.1	25000 FH	25000 FH	ALL	ALL
	Perform an o	perational check of each	h left wing fligh	t spoiler actuator o	verride quandrant		
	ACCESS NO	OTE: Flaps deployed.					
27-192-02-01	MRB	27-61-00-820-810	1.1	25000 FH	25000 FH	ALL	ALL
	Perform an o	perational check of each	h right wing flig	ht spoiler actuator	override quadrant	t.	
	ACCESS NO	TE: Flaps deployed.					
27-194-00-01	MRB	27-61-00-710-802	1.1	20000 FH	20000 FH	ALL	ALL
		check the spoiler mixer				7.22	
		OTE: CMR (27-CMR-04	ŭ		H. See MPD Sect	ion 9.	
27-196-00-01	MSG3	27-62-00-760-801	1.1	7000 FH	7000 FH	ALL	ALL
	Functionally	Check the Spoiler Electr	rical Control Sy	stem Relays for Co	ontinuity		
	AIRPLANE I	NOTE: Applicable to 90	0ER and airpla	nes with Short Fiel	d Performance Pa	ackage (if installed).	
27-210-00-01	MRB	29-00-00-790-809	1.1	25000 FH	25000 FH	ALL	ALL
	Functionally	check the internal leaka	ge of leading e	dge slat actuators.			
27-212-00-01	MRB	29-00-00-790-809	1.1	25000 FH	25000 FH	ALL	ALL
	Functionally	check the internal leaka	ge of leading e	dge flap actuators.			
27-214-00-01	MRB	27-81-00-710-801	1.1	7500 FH	7500 FH	ALL	ALL
		check the leading edge			· · · · · · · · · · · · · · · · · · ·		





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-215-01-01	MRB	27-62-00-210-802	1.1	6000 FH	6000 FH	ALL	ALL
	Perform a ge	neral visual inspection o	of the left wing	spoiler actuators.			
27-215-02-01	MRB	27-62-00-210-802	1.1	6000 FH	6000 FH	ALL	ALL
	Perform a ge	neral visual inspection of	of the right wing	spoiler actuators.			
27-215-03-01	MRB	27-61-00-210-801	1.1	5000 FH	5000 FH	ALL	ALL
	Perform a ge	neral visual inspection o	of the spoiler m	echanical control p	ath.		
27-216-00-01	MRB	27-83-00-710-801	1.1	3000 FH	3000 FH	ALL	ALL
	Functionally	check the autoslat syste	m.				
27-218-00-01	MRB	27-81-00-700-804	1.1	5000 FH	5000 FH	ALL	ALL
	Operationally	check the leading edge	uncommande	d motion protection	ı system.		
27-220-01-01	MRB	12-22-71-600-801	1.1 1.2	1250 FC 8 MO	1250 FC 8 MO	ALL	ALL
	Lubricate the	left wing leading edge s	slat rollers.				
	INTERVAL N	IOTE: Whichever occur	s first.				
	ACCESS NO	OTE: Leading edges ext	ended.				
27-220-02-01	MRB	12-22-71-600-801	1.1 1.2	1250 FC 8 MO	1250 FC 8 MO	ALL	ALL
	Lubricate the	right wing leading edge	slat rollers.				
	INTERVAL N	IOTE: Whichever occur	s first.				
	ACCESS NO	OTE: Leading edges ext	ended.				
27-222-01-01	MRB	12-22-71-640-801	1.1 1.2	2500 FC 16 MO	2500 FC 16 MO	ALL	ALL
	Lubricate the	left wing leading edge s	slat tracks.				
	INTERVAL N	IOTE: Whichever occur	s first.				
	ACCESS NO	OTE: Leading edges ext	ended.				
27-222-02-01	MRB	12-22-71-640-801	1.1 1.2	2500 FC 16 MO	2500 FC 16 MO	ALL	ALL
	Lubricate the	right wing leading edge	slat tracks.				
	INTERVAL N	IOTE: Whichever occur	s first.				
	ACCESS NO	TE: Leading edges ext	ended.				
27-224-00-01	MRB	27-81-00-710-802	1.1	1250 FH	1250 FH	ALL	ALL
					•		

 $Operation ally \ check \ the \ leading \ edge \ devices \ uncommanded \ motion \ protection \ using \ the \ standby \ hydraulic \ system.$







				INTERVAL	APPLICA	APPLICABILITY	
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-225-01-01	MRB	27-81-00-210-801	1.1	6000 FH	6000 FH	ALL	ALL
	•	eneral visual inspection on mechanisms.	of the left wing	leading edge flap a	nd slat actuators	and left wing leadin	g edge flap and
	ACCESS NO	OTE: Leading edges ext	tended.				
27-225-02-01	MRB	27-81-00-210-801	1.1	6000 FH	6000 FH	ALL	ALL
27-225-02-01	Perform a ge	27-81-00-210-801 eneral visual inspection of ation mechanisms.					
27-225-02-01	Perform a ge	eneral visual inspection of	of the right wing				

Perform a detail visual inspection of the control cables within the left main landing gear wheel well for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located in the left MLG wheel well:

- A. Aileron control cables
- B. Spoiler control cables
- C. Speed brake control cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

27-226-00-02	MRB	20-20-31-200-801	1.1	4000 FC	4000 FC	ALL	ALL
		20-20-31-200-802	1.2	24 MO	24 MO		
		20-20-31-200-805					

Perform a detail visual inspection of the control cables within the right main landing gear wheel well for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the right MLG wheel well:

- A. Aileron Cables
- B. Flap control cables
- C. Spoiler control cables
- D. Speed brake control cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.





			INTERVAL			APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
27-226-00-03	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	4000 FC 24 MO	4000 FC 24 MO	ALL	ALL	

Perform a detail visual inspection of the control cables within the left wing aft spar area for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the left wing aft spar area:

A. Aileron control cables

B. Wing spoiler control cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Extend Flaps

27-226-00-04	MRB	20-20-31-200-801	1.1	4000 FC	4000 FC	ALL	ALL
		20-20-31-200-802	1.2	24 MO	24 MO		

Perform a detail visual inspection of the control cables within the right wing aft spar area for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the right wing aft spar area:

A. Aileron control cables

B. Wing spoiler control cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Extend Flaps

27-228-00-01	MRB	20-20-31-200-801	1.1	6600 FC	6600 FC	ALL	ALL
		20-20-31-200-802	1.2	3 YR	3 YR		
		20-20-31-200-805					

Perform a detail visual inspection of all internal portions of the flight control cables above the MLG wheel well from B.S. 663.75 to B.S. 727 for broken wires, associated pulleys, brackets, and mechanisms for condition and security of installation. Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Passenger cabin floor panels between B.S. 663.75 and B.S. 727

27-228-00-02	MRB	20-20-31-200-801	1.1	6600 FC	6600 FC	ALL	ALL
		20-20-31-200-802	1.2	3 YR	3 YR		
		20-20-31-200-805					

Perform a detail visual inspection of all internal portions of the flight control cables for broken wires, associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.





					APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
27-228-00-03	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	6600 FC 3 YR	6600 FC 3 YR	ALL	ALL

Perform a detail visual inspection of all internal portions of the flight control cables for broken wires within the electronics compartment for associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

27-228-00-04	MRB	20-20-31-200-801	1.1	6600 FC	6600 FC	ALL	ALL
		20-20-31-200-802 20-20-31-200-805	1.2	3 YR	3 YR		

Perform a detail visual inspection of all internal portions of the flight control cables for broken wires within the air conditioning distribution bay, associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Forward Cargo Compartment Aft Bulkhead.

27-228-00-05	MRB	20-20-31-200-801	1.1	6600 FC	6600 FC	ALL	ALL
		20-20-31-200-802	1.2	3 YR	3 YR		

Perform a detail visual inspection of all internal portions of the flight control cables for broken wires within the forward cargo compartment, associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Forward Cargo Compartment Ceiling Panels or Floor Panels between B.S.396 to B.S. 540.

27-228-00-06	MRB	20-20-31-200-801 20-20-31-200-802	1.1 1.2	6600 FC 3 YR	6600 FC 3 YR	ALL	ALL
		20-20-31-200-805					

Perform a detail visual inspection of all internal portions of the flight control cables for broken wires within the aft cargo compartment, associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Aft Cargo Compartment Ceiling Panels





				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
27-228-00-07	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	6600 FC 3 YR	6600 FC 3 YR	ALL	ALL				
	Perform a detail visual inspection of all internal portions of the flight control cables for broken wires within the aft cargo equipment bay, associated pulleys, brackets, and mechanisms for condition and security of installation. Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, an fairlead areas.										
	INTERVAL N	IOTE: Whichever occurs	s first.								
	ACCESS NO	DTE: Aft cargo compartn assembly.	nent aft bulkhe	ead panels and wat	er tank, or the pre	ssurization aft outflo	ow valve				
27-228-00-08	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	6600 FC 3 YR	6600 FC 3 YR	ALL	ALL				
	compartmen Note: The co fairlead area	etail visual inspection of a t, associated pulleys, bra entrol cable system must s. IOTE: Whichever occur	ckets, and me be displaced f	chanisms for cond	ition and security of	of installation.					
27-229-00-01	MRB	20-20-31-200-801 20-20-31-200-802	1.1 1.2	21600 FC 6 YR	21600 FC 6 YR	ALL	ALL				
	Inspect (deta	ailed) inboard trailing edg	e aft flap drive	cable.							
	INTERVAL N	IOTE: Whichever come:	s first.								
	ACCESS NO	OTE: Flaps extended.									
27-229-00-02	MRB	20-20-31-200-801 20-20-31-200-802	1.1 1.2	21600 FC 6 YR	21600 FC 6 YR	ALL	ALL				
	Inspect (deta	niled) inboard trailing edg	e aft flap drive	cable.							
	INTERVAL N	IOTE: Whichever come	s first.								
	ACCESS NO	OTE: Flaps extended.									
27-230-00-01	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL				
	Douform a d	tail viewal inanastian of s	II dialet eesteel		i.a.aitlain tlana						

Perform a detail visual inspection of all flight control cables for broken wires within the passenger compartment over the wing center section from B.S. 540 to B.S. 663.75. Check associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Passenger cabin floor panels between B.S. 540 to B.S. 663.75.







				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
27-230-00-02	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL				
	Perform a detail visual inspection of all flight control cables for broken wires within the pilot's control quadrant. Check associated pulleys, brackets, and mechanisms for condition and security of installation. Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, ar fairlead areas.										
	INTERVAL I	NOTE: Whichever occurs	s first.								
27-235-00-02	MRB	27-11-00-820-801 27-21-00-820-808-002 27-31-00-820-801	1.1 1.2	6600 FC 3 YR	6600 FC 3 YR	ALL	ALL				
	•	check flight control cable rith the rudder system enh									
	INTERVAL I	NOTE: Whichever comes	s first.								
28-010-00-01	MRB	28-11-00-210-801	1.1	8000 FH	8000 FH	ALL	ALL				
		eral visual) the main and lap joints for obvious leak		,	ower surfaces inc	luding tank vents, s	ump drain				
28-020-01-01	MRB	28-13-41-200-801	1.1	3 YR	3 YR	ALL	ALL				
	Operationall	y check the left surge tan	k pressure reli	ef valves.							
28-020-02-01	MRB	28-13-41-200-801	1.1	3 YR	3 YR	ALL	ALL				
	Operationall	y check the right surge ta	nk pressure re	elief valves.							
28-030-01-01	MRB	28-13-31-100-801	1.1	3 YR	3 YR	ALL	ALL				
	Inspect (deta	ailed) the left surge tank v	ent flame arre	stor for clogging, c	ondition, and sec	urity.					
28-030-02-01	MRB	28-13-31-100-801	1.1	3 YR	3 YR	ALL	ALL				
	Inspect (deta	ailed) the right surge tank	vent flame an	restor for clogging,	condition, and se	curity.					
28-040-00-01	MRB	28-22-00-720-802	1.1	24000 FH	24000 FH	ALL	ALL				
	Operationall	y check the center tank fu	iel scavenge s	system (jet pump, ir	nduced port check	valve, float valve,	and line).				
28-050-00-01	MRB	28-22-00-710-802	1.1	7500 FH	7500 FH	ALL	ALL				

Operationally check engine fuel suction feed system.

SPECIAL NOTE: AWL task (28-AWL-101) interval for this task is 7500 FH or 3 YRS, whichever occurs first. See MPD section 9.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
28-054-00-01	MRB	28-22-41-720-802	1.1	7500 FH	7500 FH	ALL	ALL			
	Operationally	(BITE) check the fuel p	ump GFI relay							
	SPECIAL NOTE: AWL task (28-AWL-20) interval for this task is 1 YR. See MPD section 9.									
	AIRPLANE NOTE: Applicable to airplanes line number 1981 and 2093.									
				nber 2210 and on.	20 1- 2000 4 20	004 to 0000bish h				
		incorporated SB		·	32 to 2092 and 20	194 to 2209 which h	ave			
28-056-00-01	MRB	28-22-00-720-806	1.1	12000 FH	12000 FH	ALL	ALL			
	Functionally	check Center Tank Fuel	Boost Pump P	ower Failed On Pro	otection System.					
	SPECIAL NO	OTE: AWL task (28-AWL	L-23) interval fo	or this task is 1 YR.	See MPD section	n 9.				
	AIRPLANE N	NOTE: Applicable to air								
		Applicable to air	plane line num	ber 1 to 1972 incor	porating SB 737 2	28A1248.				
28-060-01-01	MRB	28-22-13-200-802	1.1	4000 FH	4000 FH	ALL	ALL			
	Restore (clea	an) the left main fuel tank	k water scaven	ge jet pump (witho	ut defueling the ta	ank).				
28-060-02-01	MRB	28-22-13-200-802	1.1	4000 FH	4000 FH	ALL	ALL			
	Restore (clea	an) the right main fuel tar	nk water scave	nge jet pump (with	out defueling the	tank).				
28-060-03-01	MRB	28-22-13-200-801	1.1	4000 FH	4000 FH	ALL	ALL			
	Restore (clea	an) the left and right water	er scavenge je	t pumps in the cent	er wing fuel tank	(without defueling th	ne tank).			
28-070-00-01	MRB	28-25-05-790-801	1.1	12000 FH	12000 FH	ALL	ALL			
	Functionally	(pressure decay) check	the APU fuel li	ne shroud.						
28-080-00-01	MRB	28-41-00-710-801	1.1	15000 FH	15000 FH	ALL	ALL			
	Operationally	(BITE) check the fuel q	uantity indicati	ng system.						
28-090-00-01	ALI	28-22-00-200-802	1.1	10 YR	10 YR	ALL	ALL			
	Perform a de fuel tanks.	tailed inspection of the c	out of tank wire	bundles installed of	on specified brack	ets that are mounte	ed directly on			
	SPECIAL NO	OTE: AWL task (28-AWI	L-29) interval fo	or this task is 10 YF	R. See MPD section	on 9.				
28-115-00-01	ALI	28-22-00-720-805	1.1	1 YR	1 YR	ALL	ALL			
	Functionally	check the center tank bo	oost pump auto	shutoff system.						
	SPECIAL NO	OTE: AWL task (28-AWL	L-19) interval fo	or this task is 1 YR.	See MPD section	n 9.				
	AIRPLANE N	NOTE: Applicable to air	plane line num	bers 1494 and on,	and operators tha	at have incorporated	l Boeing			

INTERVAL NOTE: 1 YR interval is from 28-AWL-19. MSG-3 interval is 4 years or 15000 hours, whichever comes

first.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
28-125-00-01	MRB	28-22-00-720-801	1.1	15000 FH	15000 FH	ALL	ALL
	Operationally	y check the fuel shutoff v	alve battery.				
28-130-00-01	MRB	28-22-14-000-801 28-22-14-400-801	1.1	7 YR	7 YR	ALL	ALL
	Replace the	fuel shut-off valve batter	y.				
28-140-00-01	MRB	28-00-00-280-801	1.1	10 YR	10 YR	ALL	ALL
	Inspect (deta degradation.	ailed) in-tank tubing and ((SFAR 88)	equipment stat	ic ground straps an	nd clamps for con	dition, security and	other
	INTERVAL N	NOTE: Perform concurre damage.	ently with other	tank inspections to	o minimize tank ei	ntries and possible a	accidental
28-140-00-02	MRB	28-00-00-280-802	1.1	10 YR	10 YR	ALL	ALL
	Inspect (deta	ailed) in-tank tubing and ((SFAR 88)	equipment stat	ic ground straps an	nd clamps for con-	dition, security and	other
	INTERVAL N	NOTE: Perform concurre damage.	ently with other	tank inspections to	o minimize tank ei	ntries and possible a	accidental
28-140-00-03	MRB	28-00-00-280-803	1.1	10 YR	10 YR	ALL	ALL
28-140-00-03		ailed) in-tank tubing and					
28-140-00-03	Inspect (deta	ailed) in-tank tubing and	equipment stat	ic ground straps an	nd clamps for con	dition, security and	other
28-140-00-03 28-150-00-01	Inspect (deta	ailed) in-tank tubing and o (SFAR 88) NOTE: Perform concurre	equipment stat	ic ground straps an	nd clamps for con	dition, security and	other
	Inspect (deta degradation. INTERVAL M MRB Perform a ful	ailed) in-tank tubing and ((SFAR 88) NOTE: Perform concurre damage.	equipment state ently with other 1.1 se measuremer	tank inspections to 6 YR nt) the bonding beto	o minimize tank en 6 YR ween fuel pumps	dition, security and on the dition, security and on the dition of the di	other accidental ALL
	Inspect (deta degradation. INTERVAL M MRB Perform a ful	illed) in-tank tubing and (SFAR 88) NOTE: Perform concurred damage. 28-00-00-760-801 Inctional check (resistance)	equipment state ently with other 1.1 se measuremer	tank inspections to 6 YR nt) the bonding beto	o minimize tank en 6 YR ween fuel pumps	dition, security and on the dition, security and on the dition of the di	other accidental ALL
28-150-00-01	Inspect (deta degradation. INTERVAL N MRB Perform a fur and center p MRB Perform a fur and center p	illed) in-tank tubing and (SFAR 88) NOTE: Perform concurred damage. 28-00-00-760-801 Inctional check (resistance umps located external to	ently with other 1.1 e measurement the left wing for the left wing for the measurement of the measurement the measurement the measurement that the measurement the measurement that the measurement the measurement that the measurement that the measurement the measurement that the m	tank inspections to 6 YR nt) the bonding betweel tank. (SFAR 88) 6 YR nt) the bonding between tank.	6 YR ween fuel pumps 6 YR ween fuel pumps	ALL ALL ALL	other accidental ALL ture for the lef
28-150-00-01	Inspect (deta degradation. INTERVAL N MRB Perform a fur and center p MRB Perform a fur and center p	ailed) in-tank tubing and of (SFAR 88) NOTE: Perform concurred damage. 28-00-00-760-801 Inctional check (resistance umps located external to 28-00-00-760-802 Inctional check (resistance under the concurred to 28-00-00-760-802) Inctional check (resistance under the concurred to 28-00-00-760-802) Inctional check (resistance under the concurred to 28-00-00-760-802)	ently with other 1.1 e measurement the left wing for the left wing for the measurement of the measurement the measurement the measurement that the measurement the measurement that the measurement the measurement that the measurement that the measurement the measurement that the m	tank inspections to 6 YR nt) the bonding betweel tank. (SFAR 88) 6 YR nt) the bonding between tank.	6 YR ween fuel pumps 6 YR ween fuel pumps	ALL ALL ALL	other accidental ALL ture for the lef
28-150-00-01 28-150-00-02	Inspect (deta degradation. INTERVAL M MRB Perform a fur and center p MRB Perform a fur and center p MRB Functionally	Ailed) in-tank tubing and (SFAR 88) NOTE: Perform concurred damage. 28-00-00-760-801 Inctional check (resistance umps located external to 28-00-00-760-802 Inctional check (resistance umps located external to 28-00-00-760-802) Inctional check (resistance umps located external to 28-00-00-760-802)	equipment state ently with other 1.1 ee measurement the left wing for 1.1 ee measurement the right wing 1.1 urement) the box	tank inspections to 6 YR 10 the bonding between tank. (SFAR 88) 6 YR 11 the bonding between tank. (SFAR 88) 6 YR 12 the bonding between tank. (SFAR 88)	6 YR ween fuel pumps 6 YR ween fuel pumps 6 YR ween fuel pumps 6 YR 6 YR	ALL ALL ALL ALL ALL ALL ALL ALL	ALL ture for the lef
28-150-00-01 28-150-00-02	Inspect (deta degradation. INTERVAL N MRB Perform a fur and center p MRB Perform a fur and center p MRB Functionally (if APU DC F	illed) in-tank tubing and (SFAR 88) NOTE: Perform concurre damage. 28-00-00-760-801 Inctional check (resistancumps located external to 28-00-00-760-802 Inctional check (resistancumps located external to 28-00-00-760-803 Check (resistance measurements)	1.1 te measurement the left wing for the right wing 1.1 the measurement of the right wing 1.1	tank inspections to 6 YR 10 the bonding between tank. (SFAR 88) 6 YR 11 the bonding between tank. (SFAR 88) 6 YR 12 the bonding between tank. (SFAR 88) 6 YR 13 the bonding between tank. (SFAR 88)	6 YR ween fuel pumps 6 YR ween fuel pumps 6 YR ween fuel pumps 6 YR 6 YR	ALL ALL ALL ALL ALL ALL ALL ALL	ALL ture for the lef

Inspect (detailed) the in-tank FQIS exposed Electrical Wiring Interconnection System (EWIS) and EWIS support for damage, adequate separation with structure and proper security. (SFAR 88) (EZAP)

INTERVAL NOTE: Perform task concurrently with other fuel tank inspection tasks to minimize tank entries and possible accidental damage.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
28-170-00-02	MRB	20-60-03-100-801 28-41-44-280-802	1.1	10 YR	10 YR	ALL	ALL			
		ailed) the in-tank FQIS exequate separation with s	•	-	• ,	:WIS) and EWIS รบ _ุ	pport for			
	INTERVAL NOTE: Perform task concurrently with other fuel tank inspection tasks to minimize tank entries and possible accidental damage.									
28-170-00-03	MRB	20-60-03-100-801 28-41-44-280-803	1.1	10 YR	10 YR	ALL	ALL			
		ailed) the in-tank FQIS exequate separation with s				:WIS) and EWIS รบ _ุ	pport for			
	INTERVAL N	NOTE: Perform task cor possible acciden	•	other fuel tank insp	ection tasks to m	inimize tank entries	and			
28-171-00-01	MRB	28-41-44-280-801	1.1	10 YR	10 YR	ALL	ALL			
	Inspect (deta security. (SF	ailed) the in-tank FQIS co AR 88)	omponents for	chaffing, rubbing, a	idequate separati	on from structure ar	nd condition f			
28-171-00-02	MRB	28-41-44-280-802	1.1	10 YR	10 YR	ALL	ALL			
	Inspect (deta security. (SF	ailed) the in-tank FQIS co AR 88)	omponents for	chaffing, rubbing, a	idequate separati	on from structure ar	nd condition f			
28-171-00-03	MRB	28-41-44-280-803	1.1	10 YR	10 YR	ALL	ALL			
	Inspect (deta security. (SF	ailed) the in-tank FQIS co AR 88)	omponents for	chaffing, rubbing, a	idequate separati	on from structure ar	nd condition f			
28-173-00-01	MRB	05-55-54-200-801	1.1	12 YR	12 YR	ALL	ALL			
	Functionally	check (resistance meas	urement) out ta	ank FQIS wire bund	lle lightning shield	to ground terminat	ion. (SFAR 8			
	SPECIAL NO	OTE: AWL task (28-AW	L-03) interval f	or this task is 10 YF	RS. See MPD sec	tion 9.				
28-200-00-01	ALI	28-11-00-211-801	1.1	10 YR	10 YR	ALL	ALL			
	Perform a de deck floor bo	etailed inspection of the voords.	wire bundles ro	uted on the main d	eck over the cent	er fuel tank and und	der the main			
	SPECIAL NO	OTE: AWL task (28-AW	L-01) interval f	or this task is 10 YF	R. See MPD secti	on 9.				
28-201-00-01	MRB	29-11-04-200-801	1.1	12 YR	12 YR	ALL	ALL			
		check (resistance meas and the adjoining struct			petween the hydra	aulic line fitting at fu	el tank wall			
28-204-00-01	MRB	28-13-31-200-801	1.1	12 YR	12 YR	ALL	ALL			
		check (resistance meas								

Functionally check (resistance measurement) the bonding between the lower air vent stack and the door structure. (SFAR 88)





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
28-205-00-01	MRB	28-13-41-200-802	1.1	12 YR	12 YR	ALL	ALL
	Functionally	check (resistance measu	urement) the b	onding between the	e pressure relief v	valve and the structu	ire. (SFAR 88)
28-207-00-01	MRB	28-21-51-200-801	1.1	12 YR	12 YR	ALL	ALL
	Functionally structure. (SI	check (resistance measu FAR 88)	urement) the b	onding of the fueling	g shutoff valve a	ctuator (solenoid) to	the adjoining
28-208-00-01	MRB	28-21-11-200-801	1.1	12 YR	12 YR	ALL	ALL
	Functionally (SFAR 88)	check (resistance measu	urement) the b	onding between the	e fueling receptac	cle (manifold) and th	e structure.
28-211-00-01	MRB	28-22-11-200-801	1.1	12 YR	12 YR	ALL	ALL
	adjoining stru	check (resistance measu ucture. (SFAR 88) sk is performed from out	,	onding between the	engine spar mo	tor operated valve a	ctuator and
28-211-00-02	MRB	28-22-21-200-801	1.1	12 YR	12 YR	ALL	ALL
	adjoining stru	check (resistance measu ucture. (SFAR 88) sk is performed from out	,	onding between the	e fuel crossfeed n	notor operated valve	e actuator and
28-213-00-01	MRB	28-25-02-200-801	1.1	12 YR	12 YR	ALL	ALL
	Functionally structure. (SI	check (resistance measu FAR 88)	urement) the b	onding resistance b	etween the APU	shutoff valve actuat	or and adjoinir
29-010-00-01	MRB	29-00-00-790-802	1.1	24000 FH	24000 FH	ALL	ALL
	Gross interna	al hydraulic system leaka	age check.				
29-020-00-01	MRB	29-11-71-000-802 29-11-71-400-802	1.1	8000 FH	8000 FH	ALL	ALL
	Replace the	hydraulic system "A" pre	ssure filter ele	ments for electric m	notor driven pum	ps (EMDP).	
29-020-00-02	MRB	29-11-71-000-802 29-11-71-400-802	1.1	8000 FH	8000 FH	ALL	ALL
	Replace the	hydraulic system "B" pre	ssure filter ele	ments for electric m	notor driven pum	ps (EMDP).	
29-030-01-01	MRB	29-11-41-000-801 29-11-41-400-801	1.1	600 FH	600 FH	ALL	ALL
	Replace the	"A" hydraulic system ele	ctric motor driv	ven pump (EMDP) o	case drain filter.		
29-030-02-01	MRB	29-11-41-000-801 29-11-41-400-801	1.1	600 FH	600 FH	ALL	ALL

Replace the "B" hydraulic system electric motor driven pump (EMDP) case drain filter.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
29-040-00-01	MRB	29-11-71-000-802 29-11-71-400-802	1.1	8000 FH	8000 FH	ALL	ALL
	Replace the	hydraulic system "A" pres	sure filter ele	ments for engine di	iven pumps (EDI	P).	
29-040-00-02	MRB	29-11-71-000-802 29-11-71-400-802	1.1	8000 FH	8000 FH	ALL	ALL
	Replace the	hydraulic system "B" pres	sure filter ele	ments for engine di	iven pumps (EDI	P).	
29-050-01-01	MRB	29-11-51-000-801 29-11-51-400-801	1.1	2400 FH	2400 FH	ALL	ALL
	Replace the	"A" system EDP case dra	in filters.				
29-050-02-01	MRB	29-11-51-000-801 29-11-51-400-801	1.1	2400 FH	2400 FH	ALL	ALL
	Replace the	"B" system EDP case dra	in filters.				
		00.00.01.000.000	1.1	4000 FH	4000 FH	ALL	ALL
29-070-00-02	MRB Clean the re	29-09-01-000-803 29-09-01-400-803 servoir pressurization filte					
29-070-00-02	Clean the res		r assembly.	e number 1345 and	on, or line numb	er 1-1344 that have	incorporated
29-070-00-02 29-080-00-01	Clean the res	29-09-01-400-803 servoir pressurization filte e: This task is applicable to	r assembly.	e number 1345 and	on, or line numb	er 1-1344 that have	incorporated
	Clean the reading Airplane note Boeing Servi	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106.	r assembly. o airplane line 1.1	5000 FH			·
	Clean the reading Airplane note Boeing Servi	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802	r assembly. o airplane line 1.1	5000 FH			·
29-080-00-01	Clean the red Airplane note Boeing Servi	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa	r assembly. o airplane line 1.1 ult protection 1.1	5000 FH system. 600 FH	5000 FH 600 FH	ALL	ALL
29-080-00-01	Clean the red Airplane note Boeing Servi	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa 29-11-61-210-801	r assembly. o airplane line 1.1 ult protection 1.1	5000 FH system. 600 FH	5000 FH 600 FH	ALL	ALL
29-080-00-01 29-090-00-01	Clean the real Airplane note Boeing Servior MRB Operational of MRB Inspect (Ger	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa 29-11-61-210-801 heral Visual) the Delta "P"	r assembly. o airplane line 1.1 ult protection 1.1 indication of A	5000 FH system. 600 FH A & B system return 7500 FH	5000 FH 600 FH filter module. 7500 FH	ALL	ALL
29-080-00-01 29-090-00-01	Clean the real Airplane note Boeing Servior MRB Operational of MRB Inspect (Ger	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa 29-11-61-210-801 neral Visual) the Delta "P"	r assembly. o airplane line 1.1 ult protection 1.1 indication of A	5000 FH system. 600 FH A & B system return 7500 FH	5000 FH 600 FH filter module. 7500 FH	ALL	ALL
29-080-00-01 29-090-00-01 29-100-00-01	Clean the real Airplane note Boeing Servior MRB Operational of MRB Inspect (Gerom MRB Operational of MRB Operational of MRB	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa 29-11-61-210-801 neral Visual) the Delta "P" 29-11-81-710-801 check of the A & B system 32-33-71-000-803	r assembly. o airplane line 1.1 ult protection 1.1 indication of A 1.1 n engine pump 1.1	5000 FH system. 600 FH A & B system return 7500 FH o (EDP) shutoff valv	5000 FH 600 FH filter module. 7500 FH res.	ALL ALL	ALL
29-080-00-01 29-090-00-01 29-100-00-01	Clean the real Airplane note Boeing Servior MRB Operational of MRB Inspect (Gerom MRB Operational of MRB Operational of MRB	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa 29-11-61-210-801 heral Visual) the Delta "P" 29-11-81-710-801 check of the A & B system 32-33-71-000-803 32-33-71-400-802	r assembly. o airplane line 1.1 ult protection 1.1 indication of A 1.1 n engine pump 1.1	5000 FH system. 600 FH A & B system return 7500 FH o (EDP) shutoff valv	5000 FH 600 FH filter module. 7500 FH res.	ALL ALL	ALL
29-080-00-01 29-090-00-01 29-100-00-01	Clean the real Airplane note Boeing Service MRB Operational of MRB Inspect (Gerein MRB Operational of MRB MRB MRB Remove the	29-09-01-400-803 servoir pressurization filte e: This task is applicable to ice Bulletin 737-29-1106. 29-11-21-700-802 check of EMDP ground fa 29-11-61-210-801 neral Visual) the Delta "P" 29-11-81-710-801 check of the A & B system 32-33-71-000-803 32-33-71-400-802 nose landing gear (NLG)	r assembly. o airplane line 1.1 ult protection 1.1 indication of A 1.1 n engine pump 1.1 down line fus 1.1	5000 FH system. 600 FH A & B system return 7500 FH b (EDP) shutoff valv 25000 FH e for functional test	5000 FH 600 FH filter module. 7500 FH res. 25000 FH off aircraft.	ALL ALL ALL	ALL ALL ALL





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
29-140-00-01	MRB	27-81-61-000-801 27-81-61-400-801	1.1	25000 FH	25000 FH	ALL	ALL
	Remove the	leading edge B system f	fuses for function	onal check off aircr	aft.		
29-150-00-01	MRB	32-41-72-000-801 32-41-72-400-801	1.1	25000 FH	25000 FH	ALL	ALL
	Remove the	B system brake fuses fo	r functional che	eck off aircraft.			
29-160-00-01	MRB	32-41-72-020-801 32-41-72-420-801	1.1	25000 FH	25000 FH	ALL	ALL
	Remove the	A system brake fuses fo	r functional che	eck off aircraft.			
29-170-00-01	MRB	27-21-95-000-801 27-21-95-400-801	1.1	25000 FH	25000 FH	ALL	ALL
	Remove the	B system rudder fuse fo	r functional che	eck off aircraft.			
29-180-00-01	MRB	29-18-11-000-802 29-18-11-400-802	1.1	16000 FH	16000 FH	ALL	ALL
	Replace the	ground hydraulic reservo	oir fill filter.				
29-200-00-01	MRB	29-21-00-700-801	1.1	15000 FH	15000 FH	ALL	ALL
	Operational ogoing out.	check standby hydraulic	electric motor	driven pump to incl	ude observing lov	v pressure light illur	mination and
29-210-00-01	MRB	29-21-51-000-802 29-21-51-400-802	1.1	12000 FH	12000 FH	ALL	ALL
	Replace the	standby hydraulic syster	n pressure filte	er.			
29-220-00-01	MRB	29-21-41-000-801 29-21-41-400-801	1.1	12000 FH	12000 FH	ALL	ALL
	Replace the	standby hydraulic syster	n case drain fil	ter.			
29-230-00-01	MRB	29-21-00-700-803	1.1 1.2	1200 FH 180 DY	1200 FH 180 DY	ALL	ALL
		check the standby rudo		include observing l	ow pressure light	illumination and the	en extinguishi
29-240-00-01	MRB	27-81-00-860-802	1.1	7500 FH	7500 FH	ALL	ALL
		check the alternate lea					
29-250-00-01	MRB	29-22-00-710-801	1.1	7500 FH	7500 FH	ALL	ALL
					-		•

Operationally check the power transfer unit control system and check the control valve for proper position.







				APPLICABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
29-260-00-01	MRB	29-22-00-730-801	1.1	25000 FH	25000 FH	ALL	ALL
	Functionally	check the power transfer	r unit.				
29-270-00-01	MRB	29-22-21-020-801 29-22-21-400-802	1.1	12000 FH	12000 FH	ALL	ALL
	Replace the	power transfer unit press	sure filter.				
29-290-00-01	MRB	27-81-61-000-801 27-81-61-400-801	1.1	25000 FH	25000 FH	ALL	ALL
	Remove the	leading edge standby hy	draulic fuse fo	r functional check o	off aircraft.		
29-300-00-01	MRB	F78-34-07-000-802-F00 F78-34-07-400-802-F00		25000 FH	25000 FH	ALL	ALL
	Remove the	standby thrust reverser l	hydraulic fuse	for functional check	off aircraft.		
29-310-00-01	MRB	29-32-00-730-802 29-32-00-730-803	1.1	25000 FH	25000 FH	ALL	ALL
30-010-00-01		0-803 (Alternate Method with their maintenance pr	ractices.			·	
30-010-00-01	IVII (D	30-31-00-750-801	1 1	11000 FH	11000 FH	ΔΙΙ	ΔΙΙ
	NOTE: The A	30-31-00-750-801 y check automatic activa AMM task provides a star reakout box is used in pla	ndard and an a	alternate procedure	. The alternate me		ALL irre engines
	NOTE: The Arunning, a br	y check automatic activa AMM task provides a star eakout box is used in pla NOTE: Applicable to air	tion of the Air I ndard and an a ace of the Disp planes with au	Data Sensor heating alternate procedure lay Electronics Unit	g for system A and The alternate me t 1. nsor heating insta	d B, if installed. ethod does not requ	ire engines
31-010-00-01	NOTE: The Arunning, a br	y check automatic activa AMM task provides a star eakout box is used in pla NOTE: Applicable to air	tion of the Air I ndard and an a ace of the Disp planes with au	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Se	g for system A and The alternate me t 1. nsor heating insta	d B, if installed. ethod does not requ	ire engines
31-010-00-01	NOTE: The Arunning, a br	y check automatic activa AMM task provides a star reakout box is used in pla NOTE: Applicable to air 3424 and on, an	tion of the Air I ndard and an a ace of the Disp planes with au d airplanes inc	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Secorporating SB 737-	g for system A and The alternate mo t 1. nsor heating insta 30A1063.	d B, if installed. ethod does not requ alled. Airplanes Line	iire engines Number
31-010-00-01	NOTE: The Arunning, a br	y check automatic activa AMM task provides a star reakout box is used in pla NOTE: Applicable to air 3424 and on, an	tion of the Air I ndard and an a ace of the Disp planes with au d airplanes inc	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Secorporating SB 737-	g for system A and The alternate mo t 1. nsor heating insta 30A1063.	d B, if installed. ethod does not requ alled. Airplanes Line	iire engines Number
	MRB MRB	y check automatic activa AMM task provides a star reakout box is used in pla NOTE: Applicable to air 3424 and on, an 31-51-00-740-801 heck of the aural warning	tion of the Air I ndard and an a ace of the Disp planes with au d airplanes inc 1.1 g module (AWM	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Secorporating SB 737-15000 FH	g for system A and The alternate mo t 1. nsor heating insta 30A1063.	d B, if installed. ethod does not requal alled. Airplanes Line ALL	Number
	MRB MRB	y check automatic activa AMM task provides a star eakout box is used in pla NOTE: Applicable to air 3424 and on, an 31-51-00-740-801 heck of the aural warning	tion of the Air I ndard and an a ace of the Disp planes with au d airplanes inc 1.1 g module (AWM	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Secorporating SB 737-15000 FH	g for system A and The alternate mo t 1. nsor heating insta 30A1063.	d B, if installed. ethod does not requal alled. Airplanes Line ALL	Number
31-020-00-01	MRB Functional ch	y check automatic activa AMM task provides a star reakout box is used in pla NOTE: Applicable to air 3424 and on, an 31-51-00-740-801 heck of the aural warning 21-33-00-000-801 heck of the cabin pressur	tion of the Air I ndard and an a ace of the Disp planes with au d airplanes inc 1.1 g module (AWN 1.1 re switch.	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Secorporating SB 737-15000 FH M) using bite check.	g for system A and The alternate me t 1. nsor heating insta 30A1063. 15000 FH	d B, if installed. ethod does not requalled. Airplanes Line ALL ALL ALL	Number ALL
31-020-00-01	MRB Functional ch	y check automatic activa AMM task provides a star reakout box is used in pla NOTE: Applicable to air 3424 and on, an 31-51-00-740-801 heck of the aural warning 21-33-00-000-801 heck of the cabin pressur	tion of the Air I ndard and an a ace of the Disp planes with au d airplanes inc 1.1 g module (AWN 1.1 re switch.	Data Sensor heating alternate procedure lay Electronics Unit tomatic Air Data Secorporating SB 737-15000 FH M) using bite check.	g for system A and The alternate me t 1. nsor heating insta 30A1063. 15000 FH	d B, if installed. ethod does not requalled. Airplanes Line ALL ALL ALL	Number ALL





				INTERVAL	APPLICABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
31-050-00-01	MRB	31-51-00-730-803	1.1	6000 FH	6000 FH	ALL	ALL				
	Functional check of the take off warning system. This task tests: Trailing edge and leading edge flap, thrust lever, upper and lower stabilizer trim limit, ground spoiler pressure, speed brake, park brake and ground spoiler bypass valve switches for the aural warning module (AWM).										
31-120-00-04	MRB	31-31-00-700-801 31-31-00-970-807	1.1	7500 FH	7500 FH	ALL	ALL				
	Download da	ata from flight data record	der (FDR) to cl	neck interfacing sys	tem output to FD	R (off aircraft).					
31-120-00-05	MRB	31-31-00-700-801 31-31-00-970-808	1.1	7500 FH	7500 FH	ALL	ALL				
	Download da	ata from flight data record	der (FDR) to cl	neck interfacing sys	tem output to FD	R (off aircraft).					
31-130-00-04	MRB	31-31-00-970-807	1.1	7500 FH	7500 FH	ALL	ALL				
	Functional ch	Functional check of required parameters (FDR, DFDAU output)									
31-130-00-05	MRB	31-31-00-970-808	1.1	7500 FH	7500 FH	ALL	ALL				
	Functional ch	neck of required parame	ters (FDR, DFI	DAU output)							
31-140-00-01	MRB	31-31-09-000-801 31-31-09-400-801 31-31-09-700-801 31-31-09-700-802 31-31-09-960-803 31-31-09-960-804	1.1	NOTE		ALL	ALL				
	Operational	check of the ULB at batte	ery replaceme	nt.							
	INTERVAL N	IOTE: At battery replace	ement or nation	nal requirement.							
	ACCESS NO	TE: FDR Hinged Ceilin	g Panel.								
31-150-00-01	MRB	31-31-09-000-801 31-31-09-400-801 31-31-09-960-801 31-31-09-960-802	1.1	NOTE		ALL	ALL				

Replace ULB battery at vendor's recommendation.

INTERVAL NOTE: At vendor's recommendation or national requirement.

ACCESS NOTE: FDR Hinged Ceiling Panel.





			INTERVAL		APPLICA	ABILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
31-160-01-01	MRB	31-62-11-000-801 31-62-11-100-803 31-62-11-400-801	1.1	6000 FH	6000 FH	ALL	ALL				
	Restore (Clean) all lint, dust and debris from the ventilation holes located on the aft side of the captain's primary flight, navigation, and center lower engine display units.										
	INTERVAL N	737NG operating and implement the	nd time of the y genvironment, ne most effective eir local regula	ear. Airline operato accompanied with e and economic m	ors are encourage debris findings at aintenance interv	on operator environ ed to evaluate their l time of cleaning to al. Operators shoul est fit for their indivi	oarticular identify d				
31-160-02-01	MRB	31-62-11-000-801 31-62-11-100-803 31-62-11-400-801	1.1	3000 FH	3000 FH	ALL	ALL				
		an) all lint, dust and deb nd upper center engine		ntilation holes locat	ed on the aft side	of the first officer's	primary flight				
		and implement th	ne most effectiv	e and economic m	aintenance interv	time of cleaning to al. Operators should	d				
		and implement th	ne most effective eir local regulat	e and economic m	aintenance interv	0	d				
32-010-01-01	MRB	and implement the negotiate with the	ne most effective eir local regulat	e and economic m	aintenance interv	al. Operators should	d				
32-010-01-01		and implement the negotiate with the operational envir	ne most effectiveir local regulation onments.	ve and economic m tory agency to adju	aintenance interv st intervals to a b	al. Operators shoulest fit for their indivi	d dual				
32-010-01-01 32-010-02-01		and implement the negotiate with the operational envir	ne most effectiveir local regulation onments.	ve and economic m tory agency to adju	aintenance interv st intervals to a b	al. Operators shoulest fit for their indivi	d dual				
	Clean expose	and implement the negotiate with the operational environment of the second surfaces of the left means and implement the negotiate with the negotiate of the left means and implement the negotiate with the negotiate w	ne most effective ir local regular onments. 1.1 ain landing gea	ye and economic m tory agency to adju- 50 FC ar shock strut.	aintenance interv st intervals to a b 50 FC	al. Operators shoulest fit for their indivi	d dual ALL				
	Clean expose	and implement the negotiate with the operational environment of the surfaces of the left metals and surfaces of the left metal	ne most effective ir local regular onments. 1.1 ain landing gea	ye and economic m tory agency to adju- 50 FC ar shock strut.	aintenance interv st intervals to a b 50 FC	al. Operators shoulest fit for their indivi	d dual ALL				
32-010-02-01	MRB Clean expose MRB	and implement the negotiate with the operational environment of the surfaces of the left median surfaces of the right of the surfaces of the surfaces of the right of the surfaces	ne most effective ir local regular onments. 1.1 ain landing geat 1.1 main landing ge	50 FC ar shock strut. 50 FC ar shock strut. 4000 FC	st intervals to a b 50 FC 50 FC 4000 FC	al. Operators shoulest fit for their indivi	d dual ALL ALL				
32-010-02-01	MRB Clean expose MRB	and implement the negotiate with the operational environment of the negotiate with the operational environment of the negotiate with the operational environment of the negotiate of the left metallic order of the right of the negotiate of the right of the negotiate of the negoti	ne most effective ir local regular onments. 1.1 ain landing geat 1.1 main landing ge	50 FC ar shock strut. 50 FC ar shock strut. 4000 FC	st intervals to a b 50 FC 50 FC 4000 FC	al. Operators shoulest fit for their indivi	d dual ALL ALL				
32-010-02-01 32-020-01-01	MRB Clean expose MRB Service the le	and implement the negotiate with the operational environment of the surfaces of the left medical surfaces of the left medical surfaces of the right of the surfaces of the surf	ne most effective ir local regular onments. 1.1 ain landing geat 1.1 main landing geat 1.1 ock strut. Servi	50 FC ar shock strut. 50 FC ear shock strut. 4000 FC decing can be perform	st intervals to a b 50 FC 50 FC 4000 FC and either with at	al. Operators shoulest fit for their indiving the ALL ALL ALL ALL ALL ALL ALL	ALL ALL ALL ALL ALL				

Lubricate the left main landing gear assembly.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
32-030-02-01	MRB	12-21-11-640-801 12-21-11-640-802 12-25-07-600-801	1.1 1.2	560 FC 90 DY	560 FC 90 DY	ALL	ALL
	Lubricate the	right main landing gear	assembly.				
	INTERVAL N	IOTE: Whichever come	s first.				
32-040-01-01	MRB	32-11-00-000-801 32-11-00-400-801 32-11-61-000-803 32-11-61-400-803 32-11-71-000-801 32-11-71-420-801 32-11-83-000-801 32-11-83-000-801 32-11-89-000-801 32-11-89-420-801 32-32-31-000-801 32-32-31-000-801 32-32-31-400-801 57-16-01-000-801 57-16-01-000-801 57-16-01-400-801 57-16-02-000-801	1.1 1.2	21000 FC 10 YR	21000 FC 10 YR	ALL	ALL
	Restore the I	eft main landing gear as	sembly				
		on man landing goal ao	combiy.				
		IOTE: Whichever come	,				
32-040-02-01		0.0	,	21000 FC 10 YR	21000 FC 10 YR	ALL	ALL
32-040-02-01	MRB	32-11-00-000-801 32-11-00-400-801 32-11-00-400-801 32-11-61-000-803 32-11-61-400-801 32-11-71-000-801 32-11-83-000-801 32-11-83-000-801 32-11-89-420-801 32-11-89-420-801 32-32-31-400-801 32-32-31-400-801 32-32-31-400-801 32-32-31-400-801 32-32-31-400-801 32-32-31-400-801 57-16-01-000-801 57-16-01-400-801	1.1 1.2			ALL	ALL
32-040-02-01	MRB Restore the r	32-11-00-000-801 32-11-00-400-801 32-11-61-000-803 32-11-61-400-803 32-11-61-400-801 32-11-71-000-801 32-11-83-000-801 32-11-83-400-801 32-11-89-000-801 32-11-89-420-801 32-32-11-000-801 32-32-31-400-801 32-32-31-400-801 57-16-01-000-801 57-16-01-400-801 57-16-02-000-801 57-16-02-400-801	s first. 1.1 1.2			ALL	ALL

Discard the left main landing gear life limited parts.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
32-050-02-01	MRB	32-11-00-000-801 32-11-00-400-801	1.1	75000 FC	75000 FC	ALL	ALL			
	Discard the right main landing gear life limited parts.									
32-060-00-01	MRB	32-00-10-100-802	1.1	50 FC	50 FC	ALL	ALL			
	Clean expos	ed surface of the nose la	anding gear str	ut.						
32-070-00-01	MRB	12-15-41-610-802 12-15-41-610-805	1.1	3000 FC	3000 FC	ALL	ALL			
	Service the r	nose landing gear shock	strut. Servicino	g can be performed	either with airpla	ne on the ground or	r on jacks.			
32-080-00-01	MRB	12-21-21-640-801 12-21-21-640-802	1.1 1.2	560 FC 90 DY	560 FC 90 DY	ALL	ALL			
	Lubricate the	nose landing gear asse	embly.							
	INTERVAL N	IOTE: Whichever come	s first.							
32-085-00-01	MRB	32-21-71-200-803	1.1	30 MO	30 MO	ALL	ALL			
	Inspect (deta	iled) nose landing gear	axle.							
32-090-00-01	MRB	32-21-00-000-801 32-21-00-400-801 32-21-21-000-801 32-21-21-400-801 32-33-11-000-801 32-33-11-400-801 32-33-51-000-801 32-33-51-400-801	1.1 1.2	18000 FC 10 YR	18000 FC 10 YR	ALL	ALL			
	Restore the	nose landing gear assen	nbly.							
	INTERVAL N	IOTE: Whichever come	s first.							
32-100-00-01	MRB	32-21-00-000-801 32-21-00-400-801	1.1	75000 FC	75000 FC	ALL	ALL			
	Discard the r	nose landing gear life lim	ited parts.							
	ACCESS NO	OTE: Remove the aft ac	cess panel on	the side wall of the	nose wheel well.					
32-110-00-01	MRB	32-31-51-200-801	1.1	8000 FC	8000 FC	ALL	ALL			
	Perform a de	tail visual inspection of t	he landing gea	ır selector valve.						
32-120-00-01	MRB	32-21-00-200-801	1.1	6600 FC	6600 FC	ALL	ALL			
		tail visual inspection of t , and lock mechanism)	he nose landin	g gear extension a	nd retraction med	chanism. (Includes r	etract actuato			
32-150-00-01	MRB	32-11-00-200-801	1.1	6600 FC	6600 FC	ALL	ALL			

Perform a detail visual inspection of the left main landing gear extension and retraction mechanism. (Includes retraction actuators, up lock actuators, up lock actuators, up lock mechanism, and down lock mechanism).





		INTERV				APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
32-150-00-02	MRB	32-11-00-200-801	1.1	6600 FC	6600 FC	ALL	ALL			
		tail visual inspection of lock actuators, down lo	•	0 0		,	des retraction			
32-200-00-01	MRB	32-35-00-730-801	1.1	36 MO	36 MO	ALL	ALL			
	In-flight chec	nctional check of the not k of the manual extension n optional method to ac	on system unde	er airline controlled	•	edures (non-revenu	e flight) is			
32-220-00-01	MRB	32-34-00-730-801	1.1	36 MO	36 MO	ALL	ALL			
	In-flight chec	nctional check of the left k of the manual extension n optional method to ac	on system unde	er airline controlled						
32-220-00-02	MRB	32-34-00-730-801	1.1	36 MO	36 MO	ALL	ALL			
	In-flight chec	nctional check of the rigl k of the manual extension n optional method to ac	on system unde	er airline controlled ask.						
32-230-00-01	MRB	32-32-21-000-801 32-32-21-400-801	1.1	25000 FH	25000 FH	ALL	ALL			
	Remove the left main landing gear wheel well protection retract pressure fuse for functional test.									
32-230-00-02	MRB	32-32-21-000-801 32-32-21-400-801	1.1	25000 FH	25000 FH	ALL	ALL			
	Remove the	right main landing gear	wheel well prot	ection retract press	sure fuse for funct	tional test.				
32-240-00-01	MRB	32-31-71-400-802	1.1	8000 FH	8000 FH	ALL	ALL			
	Operationally	check the landing gear	transfer valve							
32-250-00-01	MRB	32-41-81-000-802	1.1	6600 FC	6600 FC	ALL	ALL			
	Perform a de	tail visual inspection of	the forward and	d aft brake control l	inkages and cable	e quadrants.				
32-260-00-01	MRB	32-41-31-000-802	1.1	5000 FC	5000 FC	ALL	ALL			
	Perform a de	tail visual inspection of	the brake mete	ring valves.						
32-270-01-01	MRB	32-41-41-700-801	1.1	50 FC	50 FC	ALL	ALL			
	Visually chec	k the left brake wear pir	ns for minimum	extension.						
32-270-02-01	MRB	32-41-41-700-801	1.1	50 FC	50 FC	ALL	ALL			

Visually check the right brake wear pins for minimum extension.







				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
32-290-00-01	MRB	32-41-00-710-803 32-41-93-700-801	1.1	15000 FC	15000 FC	ALL	ALL			
	Perform an operational check of the alternate brake system and alternate brake selector valve.									
32-300-00-01	MRB	12-15-11-420-801 12-15-11-610-801	1.1	1200 FH	1200 FH	ALL	ALL			
	Check brake	accumulator precharge	pressure, serv	ice as required.						
32-310-00-01	MRB	32-41-00-720-801	1.1	12500 FC	12500 FC	ALL	ALL			
	Operationally	check the brake accum	ulator isolation	valve.						
32-330-00-01	MRB	32-44-11-000-805	1.1	8000 FC	8000 FC	ALL	ALL			
	Perform a de	tail visual inspection of t	he parking bra	ke mechanical con	rol path.					
32-340-00-01	MRB	32-44-00-790-801	1.1	4000 FC	4000 FC	ALL	ALL			
	Perform a fu	nctional bleed down ched	ck of the parkir	ng brake system an	d brake accumula	ator precharge pres	sure.			
32-350-00-01										
32-350-00-01	MRB	12-15-51-610-802 12-15-51-780-801	1.1	48 HR	48 HR	ALL	ALL			
32-350-00-01					-	ALL	ALL			
32-350-00-01	Check nose	12-15-51-780-801	res for proper i		-	ALL	ALL			
32-350-00-01	Check nose	12-15-51-780-801 and main landing gear ti	res for proper i		-	ALL	ALL			
	Check nose a INTERVAL M	12-15-51-780-801 and main landing gear til IOTE: 48 elapsed clock 32-45-00-700-801	res for proper i hours.	nflation. Service as	required. 48 HR					
	Check nose of INTERVAL N MRB Visually check	12-15-51-780-801 and main landing gear tin IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803	res for proper i hours. 1.1 g gear tires and	nflation. Service as	required. 48 HR					
	Check nose of INTERVAL N MRB Visually check	12-15-51-780-801 and main landing gear til IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803 sk nose and main landing	res for proper i hours. 1.1 g gear tires and	nflation. Service as	required. 48 HR					
32-360-00-01	MRB Visually chec	12-15-51-780-801 and main landing gear til IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803 ck nose and main landing IOTE: 48 Elapsed clock 32-45-21-000-801	nes for proper i hours. 1.1 g gear tires and hours. 1.1	nflation. Service as 48 HR d wheels for conditi	required. 48 HR	ALL	ALL			
32-360-00-01	MRB Visually checkinterval MRB MRB Restore the instance of the	12-15-51-780-801 and main landing gear tin IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803 ck nose and main landing IOTE: 48 Elapsed clock 32-45-21-000-801 32-45-21-400-801	nes for proper i hours. 1.1 g gear tires and hours. 1.1	nflation. Service as 48 HR d wheels for conditi	required. 48 HR	ALL	ALL			
32-360-00-01	MRB Visually checkinterval MRB MRB Restore the instance of the	12-15-51-780-801 and main landing gear til IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803 ek nose and main landing IOTE: 48 Elapsed clock 32-45-21-000-801 32-45-21-400-801 nose landing gear wheel	nes for proper i hours. 1.1 g gear tires and hours. 1.1	nflation. Service as 48 HR d wheels for conditi	required. 48 HR	ALL	ALL			
32-360-00-01 32-380-00-01	MRB Visually checkinterval MRB MRB Restore the INTERVAL MRB	12-15-51-780-801 and main landing gear til IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803 ck nose and main landing IOTE: 48 Elapsed clock 32-45-21-000-801 32-45-21-400-801 nose landing gear wheel IOTE: Tire change.	1.1 g gear tires and hours. 1.1 assemblies.	48 HR d wheels for condition NOTE	required. 48 HR	ALL	ALL			
32-360-00-01 32-380-00-01	MRB MRB Wisually check INTERVAL M MRB Restore the INTERVAL M MRB Restore the INTERVAL M	12-15-51-780-801 and main landing gear til IOTE: 48 elapsed clock 32-45-00-700-801 32-45-00-700-803 ek nose and main landing IOTE: 48 Elapsed clock 32-45-21-000-801 32-45-21-400-801 nose landing gear wheel IOTE: Tire change. 32-45-11-000-801 32-45-11-400-801	1.1 g gear tires and hours. 1.1 assemblies.	48 HR d wheels for condition NOTE	required. 48 HR	ALL	ALL			

Restore the right main landing gear wheel assemblies.

INTERVAL NOTE: Tire change.







				INTERVAL		APPLICA	BILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
32-390-00-01	MRB	32-51-00-200-801	1.1	8000 FC	8000 FC	ALL	ALL
	control path	etail visual inspection of the including rotary actuator OTE: Access 114AW is a		-			ing interconne
32-400-00-01	MRB	32-51-51-200-801	1.1	4000 FC	4000 FC	ALL	ALL
	Perform a de	etail visual inspection of t	the nose wheel	steering actuator r	od ends and ass	sociated hardware.	
32-420-00-01	MRB	32-09-10-710-801	1.1	25000 FH	25000 FH	ALL	ALL
	Perform an o	pperational check (bite cl	heck) of the pro	oximity switch elect	ronics unit.		
32-430-00-01	MRB	32-71-00-200-801	1.1	300 FC	300 FC	800 900 900ER	ALL
	Perform a vi	sual check of the tail skid	d crushable car	tridge for evidence	of a tail strike.		
32-440-00-01	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	and brackets A. NLG man B. NLG stee Note: The co	etail visual inspection of the for condition and securional extension cables ring cables ontrol cable system must so Observe that the man	ity of installatio	n. The following ca full travel in each di	bles are located	in the nose landing go	ear wheel well s, pulleys, and
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: NLG covers P/N 2	73A4520-1 or 2	273A4520-2 need t	o be removed to	perform this task.	
32-440-00-02	MRB	20-20-31-200-801 20-20-31-200-802	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	wires. Check	etail visual inspection of t c associated pulleys, bra nanual extension cables	ckets, and med	chanisms for condit	ion and security	of installation.	ell for broken
	INTERVAL N	NOTE: Whichever come	s first.				
32-440-00-03	MRB	20-20-31-200-801	1.1	4800 FC	4800 FC	ALL	ALL

Perform a detail visual inspection of the MLG manual extension cables within the right main landing gear wheel well for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation.

Note: MLG manual extension cables do not require displacement for complete inspection.

INTERVAL NOTE: Whichever comes first.







				INTERVAL			ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
32-450-00-01	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	10400 FC 6 YR	10400 FC 6 YR	ALL	ALL

Perform a detail visual inspection of the control cables above the MLG wheel well for broken wires from B.S. 663.75 to B.S. 727. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located above the MLG wheel well:

- A. Main landing gear control cables
- B. MLG manual extension cables
- C. MLG brake control cables

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Passenger cabin floor panels between B.S. 663.75 and B.S. 727

32-450-00-02	MRB	20-20-31-200-801	1.1	10400 FC	10400 FC	ALL	ALL
		20-20-31-200-802	1.2	6 YR	6 YR		
		20-20-31-200-805					
		32-31-22-211-802					

Perform a detail visual inspection of the control cables for broken wires within the lower nose compartment. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the lower nose compartment:

- A. MLG control cables
- B. MLG manual extension cables
- C. MLG brake control cables
- D. NLG Manual extension cables
- E. NLG Steering cables

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

32-450-00-03	MRB	20-20-31-200-801	1.1	10400 FC	10400 FC	ALL	ALL
		20-20-31-200-802	1.2	6 YR	6 YR		
		20-20-31-200-805					

Perform a detail visual inspection of the control cables for broken wires within the electronics compartment. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located in the electronics compartment:

- A. MLG control cables
- B. MLG manual extension cables
- C. MLG brake control cables

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.





				INTERVAL		APPLICABILITY	
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
32-450-00-04	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	10400 FC 6 YR	10400 FC 6 YR	ALL	ALL

Perform a detail visual inspection of the control cables for broken wires within the air conditioning distribution bay. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the air conditioning distribution bay:

A. MLG control cables

B. MLG manual extension cables

C. MLG brake control cables

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Forward cargo compartment aft bulkhead panels

32-450-00-05	MRB	20-20-31-200-801	1.1	10400 FC	10400 FC	ALL	ALL
		20-20-31-200-802	1.2	6 YR	6 YR		
		20-20-31-200-805					

Perform a detail visual inspection of the control cables within the forward cargo compartment for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the forward cargo compartment:

A. MLG control cables

B. MLG manual extension cables

C. MLG brake control cables

Note: The control cable system must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

ACCESS NOTE: Forward Cargo Compartment Ceiling Panels or Floor Panels between B.S.396 to B.S. 540.

32-460-00-01	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL
		32-31-22-211-801					

Perform a detail visual inspection of the control cables within the control cabin for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located within the control cabin:

A. Landing gear control lever cables

B. Nose wheel steering cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas

INTERVAL NOTE: Whichever occurs first.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
32-460-00-02	MRB	20-20-31-200-801 20-20-31-200-802 20-20-31-200-805	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL
	center section	etail visual inspection of on from B.S. 540 to B.S. stallation. The following	663.75. Check	associated pulleys	, brackets, and m	echanisms for cond	lition and
	A. MLG cont						
		ual extension cables					
		e control cables					
	Note: The coareas.	ontrol cables must be dis	placed full trav	el in each direction	for complete insp	ection at seals, pul	eys, and fairl
	INTERVAL N	NOTE: Whichever occur	s first.				
32-720-01-01	MRB	32-05-03-210-801 51-05-01-210-802	1.1	10 YR	10 YR	ALL	ALL
	retraction linl lugs, lug bore exceeding 10	nain landing gear assem kage. Landing gear remes, bolts and pins. Norm D years are adequate to If the basic tasks and rep	oval is required al overhaul pro maintain corro	l. Disassemble as rocedures, applied w sion at safe levels o	equired to accomith the landing geon main landing g	plish CPCP basic ta ar removed, at inte	ask on all fittir vals not
32-720-02-01	MRB	32-05-03-210-802 51-05-01-210-802	1.1	10 YR	10 YR	ALL	ALL
	retraction linl lugs, lug bore exceeding 10	main landing gear asse kage. Landing gear reme es, bolts and pins. Norm O years, are adequate to f the basic tasks and rep	oval is required al overhaul pro maintain corro	Disassemble as rocedures, applied wosion at safe levels	equired to accomith the landing geon main landing g	plish CPCP basic ta ar removed, at inte	ask on all fittir vals not
32-750-00-01	MRB	32-05-03-210-803 51-05-01-210-802	1.1	10 YR	10 YR	ALL	ALL
	mechanism (on all fittings intervals not	landing gear assembly, (plates and collar). Land, lugs, lug bores, bolts a exceeding 10 years, are oplication of the basic tas	ing gear remov nd pins. Norma adequate to n	ral is required. Disa al overhaul procedu naintain corrosion a	ssemble as requi res, applied with t safe levels on n	red to accomplish C the landing gear rer ose landing gear co	PCP basic ta noved, at
32-800-00-01	MRB	05-41-07-210-801	1.1	120 DY	120 DY	ALL	ALL
		external zonal inspection und, without the use of s			d landing gear do	ors. Inspection is a	ccomplished
	INTERVAL N	NOTE: The EZAP inspe	ction requireme	ent with interval 550	00 FC/30 MO is sa	atisfied by this zona	I

Perform an external zonal inspection (GV) of the left main landing gear and landing gear doors. Inspection is accomplished from the ground, without the use of stands or ladders.

120 DY 745 FC 120 DY 745 FC ALL

INTERVAL NOTE: Whichever comes first.

05-41-07-210-802

32-804-01-01

MRB



ALL

1.1 1.2





				INTERVAL		APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
32-806-01-01	MRB	05-41-07-210-803	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL	
	Perform an e	external zonal inspection	(GV) of the lef	t main landing gea	r and landing gea	r doors. (EZAP)		
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	val 5500 FC/30 MO	is	
32-808-02-01	MRB	05-41-07-210-804	1.1 1.2	120 DY 745 FC	120 DY 745 FC	ALL	ALL	
		external zonal inspection und, without the use of s			ear and landing ge	ar doors. Inspectior	is accomplis	
	INTERVAL N	NOTE: Whichever come	s first.					
32-810-02-01	MRB	05-41-07-210-805	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL	
	Perform an e	external zonal inspection	(GV) of the rig	ht main landing ge	ear and landing ge	ar doors. (EZAP)		
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	val 5500 FC/30 MO	is	
33-010-00-01	MRB	33-51-00-710-801	1.1	600 FH	600 FH	ALL	ALL	
	Operational of	check of the emergency	lights.					
33-020-00-01	MRB	33-51-00-720-801	1.1	2 YR	2 YR	ALL	ALL	
	Operational of light switch	check of the flight deck e on" position.	mergency ligh	ting switch "on" an	d "armed" position	n and attendant pan	el emergency	
33-055-00-01	MRB	33-51-06-200-801	1.1	1 YR	1 YR	ALL	ALL	
	Functionally	check the emergency lig	hts battery pac	cks for capacity (15	5 min. minimum) a	and one complete de	eep cycle.	
33-060-00-01	MRB	33-51-06-600-802	1.1	2 YR	2 YR	ALL	ALL	
		or more complete deep econd deep cycle is acco						
33-070-00-01	MRB	33-51-15-860-802	1.1 1.2	3 YR 12000 FH	3 YR 12000 FH	ALL	ALL	
	Functional ch	neck of the photolumines	scent floor prox	rimity lighting.				
	AIRPLANE I	NOTE: If installed.						
	INTERVAL N	NOTE: Whichever occur	s first.					
33-080-00-01	MRB	33-51-15-960-801	1.1	10 YR	10 YR	ALL	ALL	
	Replace pho	toluminescent floor prox	imity lighting a	t manufacturer's life	e limit.			

AIRPLANE NOTE: If installed.







				INTERVAL		APPLICABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
33-090-00-01	MPD	57-21-22-200-801	1.1	6 MO	6 MO	ALL	ALL		
	•	neral visual inspection of	•	•	•				
	Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 33-W01-00. AIRPLANE NOTE: Single lens configuration only.								
	AIRPLANE N	NOTE: Single lens confi	iguration only.						
33-090-01-01	MPD	57-21-22-200-802	1.1	24 MO	24 MO	ALL	ALL		
	•	neral visual inspection on sk satisfies the requirem	•	•	•	•	n (Glass).		
	AIRPLANE N	NOTE: Dual Lens Confi	guration (Glass	3).					
34-010-00-01	MRB	34-11-00-790-810	1.1	24 MO	24 MO	ALL	ALL		
	Functional le	ak check of captain's pit	ot system.						
34-020-00-01	MRB	34-11-00-790-811	1.1	24 MO	24 MO	ALL	ALL		
	Functional le	ak check of first officer's	pitot system.						
34-030-00-01	MRB	34-11-00-790-812	1.1	24 MO	24 MO	ALL	ALL		
	Functional le	ak check of standby pito	ot system.						
34-040-00-01	MRB	34-11-00-790-804	1.1	24 MO	24 MO	ALL	ALL		
	Functional le	ak check of captain's sta	atic system.						
34-050-00-01	MRB	34-11-00-790-806	1.1	24 MO	24 MO	ALL	ALL		
	Functional le	ak check of first officer's	static system.						
34-060-00-01	MRB	34-11-00-790-808	1.1	24 MO	24 MO	ALL	ALL		
	Functional le	ak check of standby sta	tic system.						
34-060-10-01	MPD	34-11-00-780-802	1.1	72 MO	72 MO	ALL	ALL		
	Functional ch	neck of the air data syste	em altimetry sy	stem.					
34-070-00-01	MRB	34-11-01-200-804	1.1	7500 FH	7500 FH	ALL	ALL		
	Detail visual	inspection of the pitot pr	robes.						
34-080-00-01	MRB	34-11-02-200-803	1.1	15000 FH	15000 FH	ALL	ALL		
	Detailed insp	ection of the static ports	5.						
34-090-00-01	MRB	34-11-00-210-801	1.1	15000 FH	15000 FH	600 700 800 900 900ER	ALL		

Detail inspection for moisture in the pitot systems.





				INTERVAL		APPLICA	BILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
34-100-00-01	MRB	34-11-00-210-802	1.1	15000 FH	15000 FH	600 700 800 900 900ER	ALL
	Detail inspec	ction for moisture in the s	tatic systems.				
34-110-00-01	MRB	34-53-00-730-802	1.1	24 MO	24 MO	ALL	ALL
		nctional check of the ATC (ELS), Enhanced Survei AR 91.413)				•	•
	INTERVAL I	NOTE: or national requir	ement.				
34-110-00-02	MRB	34-53-00-730-803 34-53-00-730-805 34-53-00-730-806	1.1	24 MO	24 MO	ALL	ALL
	Surveillance	nctional check of the ATC (ELS), Enhanced Survei st Set. (See reference FA	illance (EHS),			•	
	INTERVAL I	NOTE: or national requir	ement.				
34-130-00-01	MRB	34-24-03-000-801 34-24-03-400-801	1.1	3 YR	3 YR	ALL	ALL
	Discard the	dedicated battery/charge	r internal batte	ry for the integrated	d standby flight d	splay.	
	AIRPLANE	NOTE: If Installed.					
	INTERVAL I	NOTE: At manufacture's	life limit.				
34-140-00-01	MRB	34-24-02-710-802	1.1	9000 FH	9000 FH	ALL	ALL
	Operationall	y check the Integrated St	andby Flight D	Display Dedicated B	Battery/Charger.		
	AIRPLANE	NOTE: If Installed.					
35-010-00-01	MRB	35-12-00-700-802	1.1	6000 FH	6000 FH	ALL	ALL
	Operationall	y check each flight crew	oxygen mask/ı	egulator (out of the	e box assembly).		
35-020-00-01	MRB	35-12-85-000-802 35-12-85-400-802	1.1	16000 FH	16000 FH	ALL	ALL
	Functionally	check (off the airplane) e	each flight crev	v oxygen mask/reg	ulator per the ma	nufacturer's compon	ent manual.
35-040-00-02	MRB	12-15-21-600-803-002	1.1	VEN REC		ALL	ALL
	Discard the	flight crew oxygen cylinde	er.				
	INTERVAL I	NOTE: At Vendors recon	nmendation.				
35-050-00-01	MRB	35-12-00-710-801	1.1	2400 FH	2400 FH	ALL	ALL
		ss) check the flight crew					

Visually (cross) check the flight crew oxygen cylinder pressure indicator and the control compartment flight crew oxygen indicator.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
35-060-00-01	MRB	35-22-00-210-801	1.1	6 YR	6 YR	ALL	ALL
	Visually chee oxygen gene	ck the temperature sensiterator.	tive tape on ea	ach passenger cabi	n, (including lava	tory and cabin atten	dant's), chemica
	AIRPLANE	NOTE: Not applicable to	airplanes with	n all gaseous passe	nger oxygen sys	tem.	
35-065-00-01	MRB	35-22-00-210-802	1.1	6 YR	6 YR	ALL	ALL
		of the passenger lavato verify oxygen activation	, , ,		ystem (CDS) disc	charge indicator tap	e for evidence o
	AIRPLANE	NOTE: If lavatory oxyge	n Constant Dis	spensing System (C	CDS) installed.		
35-070-00-01	MRB	35-22-11-000-804-001 35-22-11-000-811-001 35-22-11-400-804-001 35-22-11-400-811-001	1.1	NOTE		ALL	ALL
	Discard the	chemical oxygen genera	tors.				
	AIRPLANE	NOTE: Not applicable to	airplanes with	n all gaseous passe	nger oxygen sys	tem.	
	INTERVAL I	NOTE: At Vendors recor	mmendation.				
35-070-00-03	MRB	35-22-11-000-805-001	1.1	NOTE		ALL	ALL
33-070-00-03	WIND	35-22-11-000-811-001 35-22-11-400-805-001 35-22-11-400-811-001	1.1	NOTE		ALL	ALL
	Discard the	chemical oxygen genera	tors.				
	AIRPLANE	NOTE: Not applicable to	airplanes with	n all gaseous passe	nger oxygen sys	tem.	
	INTERVAL I	NOTE: At Vendors recor	nmendation.				
35-070-00-04	MRB	35-22-11-000-806-001 35-22-11-000-811-001 35-22-11-400-806-001 35-22-11-400-811-001	1.1	NOTE		ALL	ALL
	Discard the	chemical oxygen genera	tors.				
		NOTE: Not applicable to		n all gaseous passe	enger oxygen sys	tem.	
		NOTE: At Vendors recor		g p	3 , , , , .		
35-075-00-01	MRB	35-22-51-000-801 35-22-51-040-801 35-22-51-400-801 35-22-51-440-801	1.1	LIF LIM		ALL	ALL
	Discord the	passenger lavatory Cons	stant Dianansia	a System (CDS) as	avgon ovlindo:		

Discard the passenger lavatory Constant Dispensing System (CDS) oxygen cylinder.

AIRPLANE NOTE: If lavatory oxygen Constant Dispensing System (CDS) installed.

INTERVAL NOTE: Vendor recommended life limit.







				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
35-080-00-01	MRB	35-22-00-700-801 35-22-00-700-802	1.1	15000 FH	15000 FH	ALL	ALL				
	Functionally check the passenger oxygen system automatic and manual modes, (this check includes the PSU/ASU door latch actuator/solenoids and oxygen system altitude pressure switch(es), and for gaseous systems this also includes the voltage averaging unit, cylinder pressure transducer and flight deck pressure indication).										
35-090-00-01	MRB	35-22-31-210-801-001	1.1	12000 FH	12000 FH	ALL	ALL				
	Detailed vis	ual inspection 10% (rotati :	onal inspection	n) of the passenger	, lavatory, and att	endant oxygen mas	ks for condition				
35-100-00-01	MRB	35-31-00-710-801	1.1	8000 FH	8000 FH	ALL	ALL				
	Visually che	ck all the portable oxyger	n cylinders for	presence, condition	, and security.						
35-120-00-01	MRB	35-31-01-960-801	1.1	NOTE		ALL	ALL				
	Discard the	portable oxygen cylinder.									
	INTERVAL	NOTE: At Vendors recon	nmendation.								
36-020-01-01	MRB	36-12-00-710-802	1.1	16000 FH	16000 FH	ALL	ALL				
	Functionally	check the left precooler	control valve a	nd wing TAI soleno	id.						
36-020-02-01	MRB	36-12-00-710-802	1.1	16000 FH	16000 FH	ALL	ALL				
	Functionally	check the right precoole	r control valve	and wing TAI solen	oid.						
36-030-01-01	MRB	36-12-03-000-801 36-12-03-400-801	1.1	16000 FH	16000 FH	ALL	ALL				
		check (off-airplane) the l airplane portion only (rem	•		per vendor's ove	rhaul manual. Task	card procedu				
36-030-02-01	MRB	36-12-03-000-801 36-12-03-400-801	1.1	16000 FH	16000 FH	ALL	ALL				
		check (off-airplane) the rapply to on-airplane porti			or per vendor's ov	erhaul manual. Tas	k card				
38-030-00-01	MRB	38-32-05-960-801	1.1	2500 FH	2500 FH	ALL	ALL				
	Restore (cle	an) or replace the vacuu	m blower filter	(if installed).							
	AIRPLANE	NOTE: Applicable to air	planes with va	cuum blower equipp	oed with a filter.						

Restore the waste drain ball-valve by replacing the seals (2) (off aircraft).

Note: The intent of the task is to restore the ball valve assembly by replacing two internal seals (P/N RS823-1). The seals are located above and below the ball valve inside the valve assembly.







				INTERVAL	APPLICABILITY		
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
38-060-00-01	MRB	38-42-10-960-801	1.1	6500 FH	6500 FH	ALL	ALL
	Discard and	replace the bleed air in-li	ne filter.				
38-090-00-01	MRB	38-32-02-000-801 38-32-02-400-801	1.1	2500 FH	2500 FH	ALL	ALL
	Pestore (clar	an or replace) the waste	tank water ser	arator filter hackets	-		
	,	NOTE: Applicable to filte					
	AIRI LAIL	NOTE: Applicable to little	or basket part i		1 4114 0 1000 000.		
38-100-00-01	MRB	38-32-02-000-801	1.1	6000 FH	6000 FH	ALL	ALL
		38-32-02-100-801 38-32-02-400-801					
	Pestore (clar	an or replace) the waste	tank water ser	arator filter hackets	-		
	`	NOTE: Applicable to all				1956-000	
	AINFLANL	NOTE. Applicable to all	iliter basket pe	it numbers except	01940-001 and 0	1930-000.	
47-200-00-02	MSG3	47-21-00-700-802	1.1	6500 FH	6500 FH	ALL	ALL
	Visual check	fluid accumulation in the	nitrogen gene	eration system (NG	S) tubing through	the drain cap.	
		NOTE: If Nitrogen Gene			, 0 0	·	
		, and the second	·				
47-210-00-01	14000	17 00 00 710 001					
47-210-00-01	MSG3	47-00-00-710-801	1.1	13000 FH	13000 FH	ALL	ALL
47-210-00-01		y check center tank Vent			13000 FH	ALL	ALL
47-210-00-01	Operationally		Cross Flow Cl	neck Valve.			ALL
47-210-00-01	Operationally SPECIAL NO	y check center tank Vent	Cross Flow Cl	neck Valve. or this task is 13000			ALL
47-210-00-01	Operationally SPECIAL NO	y check center tank Vent OTE: AWL task (47-AWL	Cross Flow Cl	neck Valve. or this task is 13000			ALL
47-220-00-01	Operationally SPECIAL NO	y check center tank Vent OTE: AWL task (47-AWL	Cross Flow Cl	neck Valve. or this task is 13000			ALL
	Operationally SPECIAL NO AIRPLANE I MSG3 Inspect (deta	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriche	Cross Flow Cl -06) interval for ration System 1.1	neck Valve. or this task is 13000 is installed. 6500 FH) FH. See MPD S 6500 FH	ection 9.	ALL
	Operationally SPECIAL NO AIRPLANE I MSG3 Inspect (detarear spar for	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriched damage and leaks.	Cross Flow Cl -06) interval for ration System 1.1 ed air (NEA) di	neck Valve. or this task is 13000 is installed. 6500 FH stribution lines from	0 FH. See MPD S 6500 FH n the air separation	ection 9. ALL n module (ASM) to	ALL
	MSG3 Inspect (detarear spar for SPECIAL NO	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriched damage and leaks. OTE: AWL task (47-AWL	Cross Flow Cl 2-06) interval for ration System 1.1 ed air (NEA) di 2-07) interval for	or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500	0 FH. See MPD S 6500 FH n the air separation	ection 9. ALL n module (ASM) to	ALL
	MSG3 Inspect (detarear spar for SPECIAL NO	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriched damage and leaks.	Cross Flow Cl 2-06) interval for ration System 1.1 ed air (NEA) di 2-07) interval for	or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500	0 FH. See MPD S 6500 FH n the air separation	ection 9. ALL n module (ASM) to	ALL
47-220-00-01	MSG3 Inspect (detarear spar for SPECIAL NO AIRPLANE I	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriche damage and leaks. OTE: AWL task (47-AWL NOTE: If Nitrogen Gene	Cross Flow Cl -06) interval for ration System 1.1 ed air (NEA) di -07) interval for ration System	neck Valve. or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500 is installed.	0 FH. See MPD S 6500 FH In the air separation	ALL on module (ASM) to	ALL the fuel tank
	MSG3 Inspect (detarear spar for SPECIAL NO	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriched damage and leaks. OTE: AWL task (47-AWL	Cross Flow Cl 2-06) interval for ration System 1.1 ed air (NEA) di 2-07) interval for	or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500	0 FH. See MPD S 6500 FH n the air separation	ection 9. ALL n module (ASM) to	ALL
47-220-00-01	MSG3 Inspect (deta rear spar for SPECIAL NO AIRPLANE I	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 hiled) the nitrogen enriched damage and leaks. OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-32-02-000-801	Cross Flow Cl 2-06) interval for ration System 1.1 ed air (NEA) di 2-07) interval for ration System 1.1	neck Valve. or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500 is installed.	0 FH. See MPD S 6500 FH In the air separation	ALL on module (ASM) to	ALL the fuel tank
47-220-00-01	MSG3 Inspect (detarear spar for SPECIAL NO AIRPLANE I	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriched damage and leaks. OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-32-02-000-801 47-32-02-400-801	Cross Flow Cl -06) interval for ration System 1.1 ed air (NEA) di -07) interval for ration System 1.1 at a control of the ration System 1.1	neck Valve. or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500 is installed. 12000 FH	0 FH. See MPD S 6500 FH In the air separation	ALL on module (ASM) to	ALL the fuel tank
47-220-00-01	MSG3 Inspect (detarear spar for SPECIAL NO AIRPLANE I	y check center tank Vent OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-00-00-790-802 ailed) the nitrogen enriched damage and leaks. OTE: AWL task (47-AWL NOTE: If Nitrogen Gene 47-32-02-000-801 47-32-02-400-801 ozone converter (off-aircr	Cross Flow Cl -06) interval for ration System 1.1 ed air (NEA) di -07) interval for ration System 1.1 at a control of the ration System 1.1	neck Valve. or this task is 13000 is installed. 6500 FH stribution lines from or this task is 6500 is installed. 12000 FH	0 FH. See MPD S 6500 FH In the air separation	ALL on module (ASM) to	ALL the fuel tank

Clean the nitrogen generation system heat exchanger (off aircraft).

AIRPLANE NOTE: If Nitrogen Generation System is installed.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
47-400-00-01	MSG3	47-43-02-720-801	1.1	22500 FH	22500 FH	ALL	ALL			
	Functional te	est of the thermal switch	(off aircraft).							
	SPECIAL NOTE: AWL task (47-AWL-04) interval for this task is 22500 FH. See MPD Section 9.									
	AIRPLANE NOTE: If Nitrogen Generation System is installed.									
49-010-00-01	MRB	49-13-11-200-803	1.1	5 YR	5 YR	ALL	ALL			
	Perform a ge	eneral visual inspection of	of the APU mou	unts for general cor	ndition and securit	y of installation.				
49-020-00-01	MRB	49-13-11-200-801	1.1	8 YR	8 YR	ALL	ALL			
	Perform a de	etailed inspection of the	APU mounts.							
49-030-00-01	MRB	49-15-11-200-801	1.1	APU CNG		ALL	ALL			
	Perform a de	etailed inspection of the	sigma seal (aft	er APU removal).						
49-040-00-01	MRB	49-17-11-200-801	1.1	APU CNG		ALL	ALL			
	Perform a de	etailed inspection of the A	APU insulation	panels. (After APU	removal).					
49-052-00-01	MRB	49-11-00-000-801 49-11-00-400-801	1.1	LIF LIM		ALL	ALL			
	Discard the	engine compressor impe	ller							
		NOTE: Refer to APU sho		ife limits						
		TOTAL MOION TO AN O ONE	op manaar for i	no mine.						
49-062-00-01	MRB	49-11-00-000-801 49-11-00-400-801	1.1	LIF LIM		ALL	ALL			
	Discard the f	first stage turbine disk.								
		NOTE: Refer to APU sho	op manual for l	ife limits.						
49-072-00-01	MRB	49-11-00-000-801 49-11-00-400-801	1.1	LIF LIM		ALL	ALL			
	Discard the	second stage turbine rote	or .							
		NOTE: Refer to APU sho		ife limits.						
49-082-00-01	MRB	49-11-00-000-801	1.1	LIF LIM		ALL	ALL			
		49-11-00-400-801								
	Discard the t	urbine shaft.								
	INTERVAL N	NOTE: Refer to APU sho	op manual for l	ife limits.						
49-102-00-01	MRB	49-31-21-000-801 49-31-21-400-801	1.1	4000 AH	4000 AH	ALL	ALL			
	Discard the f	fuel inlet filter element or	the fuel contr	ol unit (FCU)						

Discard the fuel inlet filter element on the fuel control unit (FCU).







				INTERVAL		APPLICA	
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
49-140-00-01	MRB	49-61-00-710-801	1.1	1600 AH	1600 AH	ALL	ALL
	-APU data m	operational check of the formation of th	following by int	errogating the CDL	l's APU maintena	nce pages.	
49-172-00-01	MRB	49-81-11-200-801	1.1	19000 AH	19000 AH	ALL	ALL
	Inspect (deta	ailed) the APU exhaust so	eal.				
49-212-00-01	MRB	49-81-41-200-801	1.1	10000 AH	10000 AH	ALL	ALL
	Perform a ge	eneral visual inspection of	of the eductor (on the APU) for ger	neral condition.		
49-220-00-01	MRB	49-91-71-200-801	1.1	25000 FH	25000 FH	ALL	ALL
	Inspect (deta	niled) the eductor inlet du	ıct (interior and	exterior).			
49-240-00-01	MRB	49-15-22-600-801	1.1	16000 FH	16000 FH	ALL	ALL
	Lubricate the	e vortex generator hinge	pin.				
52-010-00-01	MRB	12-25-11-640-802	1.1	2 YR	2 YR	ALL	ALL
		e forward entry door hand ninge torque tube and the		, ,	•	and latch control roc	ls), the bearin
52-010-00-02	MRB	12-25-13-640-802	1.1	2 YR	2 YR	ALL	ALL
		e forward service door ha the door hinge torque tul		chanisms (latch tor	que tube bearing	s and latch control r	ods) and the
52-010-00-03	MRB	12-25-12-640-802	1.1	2 YR	2 YR	ALL	ALL
		e aft entry door handle, la ninge torque tube.	atch mechanisr	ns (latch torque tub	oe bearings and la	atch control rods) ar	nd the bearing
52-010-00-04	MRB	12-25-13-640-802	1.1	2 YR	2 YR	ALL	ALL
		e aft service door handle, ninge torque tube.	, latch mechan	isms (latch torque t	ube bearings and	l latch control rods)	and the beari
52-020-00-01	MRB	12-25-11-640-801	1.1	1 YR	1 YR	ALL	ALL
		e forward entry door guid arm bushing, gate hinges				gs and threads), the	upper and
52-020-00-02	MRB	12-25-13-640-801	1.1	1 YR	1 YR	ALL	ALL
	Lubricate the	e forward service door gu	ide plate track	s and arm assembl	lies (rod end bear	ings and threads), t	he upper and





				INTERVAL		APPLICA	
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
52-020-00-03	MRB	12-25-12-640-801	1.1	1 YR	1 YR	ALL	ALL
		e aft entry door guide pla ushing and gate hinges.	te tracks and a	rm assemblies (roc	d end bearings ar	nd threads), the upp	er and lower
52-020-00-04	MRB	12-25-13-640-801	1.1	1 YR	1 YR	ALL	ALL
		e aft service door guide pushings and gate hinges.		l arm assemblies (r	od end bearings	and threads), the up	oper and lowe
52-030-00-01	MRB	52-11-00-200-802	1.1	3 YR	3 YR	ALL	ALL
	Inspect (deta	iled) the forward entry d	oor centering o	juide stud and nylo	n track pads for o	condition.	
52-040-00-01	MRB	52-41-00-200-803	1.1	3 YR	3 YR	ALL	ALL
	Inspect (Deta	ailed) the fwd service do	or centering gu	ide bearings for co	ndition.		
52-040-00-02	MRB	52-13-00-200-803	1.1	3 YR	3 YR	ALL	ALL
	Inspect (Deta	ailed) the aft entry door o	entering guide	bearings for condi	tion.		
52-040-00-03	MRB	52-41-00-200-803	1.1	3 YR	3 YR	ALL	ALL
	Inspect (Deta	ailed) the aft service doo	r centering gui	de bearing for cond	lition.		
52-050-00-01	MRB	52-11-00-200-803 52-11-00-200-804	1.1	6000 FH	6000 FH	ALL	ALL
	Inspect (Gen	eral Visual) the forward	entry door pres	ssure and flapper so	eals for degradtio	on.	
52-050-00-02	MRB	52-41-00-200-802	1.1	6000 FH	6000 FH	ALL	ALL
	Inspect (Gen	eral Visual) the forward	service door pr	essure seal for deg	gradation.		
52-050-00-03	MRB	52-13-00-200-802	1.1	6000 FH	6000 FH	ALL	ALL
	Inspect (Gen	eral Visual) the aft entry	door pressure	seal for degradation	on.		
52-050-00-04	MRB	52-41-00-200-802	1.1	6000 FH	6000 FH	ALL	ALL
	Inspect (Gen	eral Visual) the aft servi	ce door pressu	re seal for degrada	tion.		
52-090-00-01	MRB	12-25-31-640-801	1.1	1 YR	1 YR	ALL	ALL
	Lubricate the	forward cargo compartr	nent door latch	torque tube bearir	ngs and the coun	ter balance idler cra	nk.
52-090-00-02	MRB	12-25-31-640-801	1.1	1 YR	1 YR	ALL	ALL
	Lubricate the	aft cargo compartment	door latch torq	ue tube bearings a	nd the counter ba	alance idler crank.	
52-100-00-01	MRB	52-31-00-200-802	1.1	5000 FH	5000 FH	ALL	ALL
		eral Visual) the forward					





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
52-100-00-02	MRB	52-31-00-200-802	1.1	5000 FH	5000 FH	ALL	ALL			
	Inspect (General Visual) the aft cargo compartment door pressure seal for degradation.									
50 400 00 04	MDD	40.05.44.040.004	4.4	0.1/D	0.70	A1.1	A.I.I.			
52-120-00-01	MRB	12-25-41-640-801	1.1	2 YR	2 YR	ALL	ALL			
	Lubricate the	E/E access door handle	e latching mecr	nanism (rack and p	inion gear and the	e lock pins).				
52-130-00-01	MRB	52-48-41-200-802	1.1	8000 FH	8000 FH	ALL	ALL			
	Inspect (gene	eral visual) the E/E acce	ss door pressu	re seal for degrada	ation.					
52-140-00-01	MRB	52-48-31-200-802	1.1	15000 FH	15000 FH	ALL	ALL			
	Inspect (gene	eral visual) the forward a	ccess door pre	essure seal for degr	radation.					
52-200-00-01	MRB	52-71-11-710-801	1.1	5000 FH	5000 FH	ALL	ALL			
32-200-00-01	IVIND	52-71-22-710-803	1.1	3000 FTT	3000 FTT	ALL	ALL			
		52-71-31-710-801 52-71-41-710-801								
		52-71-42-710-801								
	Operationally check the door sensors (proximity or mechanical switches as applicable) for the passenger cabin									
	entry/service, E/E access, automatic overwing emergency exit, forward access and cargo doors.									
	ACCESS NO	TE: Access panels 832	2 and 842 are a	applicable to 737-80	00 and 737-900 o	nly.				
52-210-00-01	MRB	52-22-00-710-802	1.1	15000 FH	15000 FH	ALL	ALL			
	Operationally check the flight lock mechanical switches for the automatic overwing emergency exit doors.									
	ACCESS NO	OTE: Zones and access	panels 832 an	d 842 are applicab	le to 737-800 and	737-900 only.				
52-220-00-01	MRB	52-22-00-710-801	1.1	9 YR	9 YR	ALL	ALL			
	Operationally	check (cycle) the auton	natic overwing	emergency exit do	ors.					
	ACCESS NOTE: Zones and access panels 832 and 842 are applicable to 737-800 and 737-900 only.									
52-230-00-01	MRB	52-22-00-210-801	1.1	6 YR	6 YR	ALL	ALL			
	Perform a de fittings/joints.	Perform a detailed visual inspection of the automatic overwing emergency exit door latch rollers, links and pivot fittings/joints.								
	ACCESS NO	OTE: Zones and access	panels 832 an	d 842 are applicab	le to 737-800 and	737-900 only.				
52-240-00-01	MRB	52-22-00-710-803	1.1	6 YR	6 YR	ALL	ALL			
		check the flight lock en								
		TE: Zones and access		0 0	le to 737-800 and	737-900 only.				
52-250-00-01	MRB	52-22-00-210-802	1.1	6 YR	6 YR	ALL	ALL			

Perform a detailed visual inspection of automatic overwing exit door flight locks for corrosion and condition.

ACCESS NOTE: Zones and access panels 832 and 842 are applicable to 737-800 and 737-900 only.







				INTERVAL		APPLICABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
52-360-00-01	MRB	52-51-00-710-802	1.1	30000 FH	30000 FH	ALL	ALL			
	Perform a functional check of the locking and unlocking latch bolt mechanism on the flight deck door decompression panel									
	SPECIAL N	OTE: CMR task (52-CMI	R-01) interval	for this task is 3000	FH. See MPD S	ection 9.				
	AIRPLANE	NOTE: Applicable to airp door installed by		1 and on and to the specific Boeing se		the new flight deck	security			
	INTERVAL I	NOTE: The equivalent C the MRB interval	,	, .	ed at 3000 hours,	which has precede	ence over			
52-370-00-01	MRB	52-51-00-710-803	1.1	30000 FH	30000 FH	ALL	ALL			
	Operationall	y check the flight deck do	or decompres	sion panel hinges.						
	AIRPLANE	NOTE: Applicable to airp door installed by		1 and on and to tho specific Boeing se	•	the new flight deck	security			
52-380-00-01	MRB	52-51-00-210-801	1.1	30000 FH	30000 FH	ALL	ALL			
	General visual inspection (GVI) of the flight deck door decompression panel hinges for condition and security.									
52-390-00-01	MRB	NOTE: Applicable to airp door installed by 52-51-01-200-801		specific Boeing ser	•	ALL	ALL			
52-390-00-01						ALL	ALL			
	General visual inspection (GVI) of the flight deck door seals for condition and security. AIRPLANE NOTE: Applicable to airplane L/N 1221 and on and to those airplanes with the new flight deck security									
	AIRPLANE			and on and to the specific Boeing ser	•	the new flight deck	security			
52-400-00-01	MRB	52-51-00-210-802	1.1	30000 FH	30000 FH	ALL	ALL			
	General visu	al inspection of the flight	deck door ded	compression panel	seals for condition	n and security.				
	AIRPLANE	NOTE: Applicable to airg door installed by		1 and on and to the specific Boeing se	•	the new flight deck	security			
52-410-00-01	MRB	52-51-00-700-802	1.1	11000 FH	11000 FH	ALL	ALL			
	•	Functionally check the "deny" time delay function of the flight deck security door access system to verify; operation of the three position rotary switch in the P8 panel, the deny function, and reversion to the default mode.								
	AIRPLANE	NOTE: Applicable to airg door installed by		1 and on and to tho specific Boeing se	•	the new flight deck	security			
52-450-00-01	MRB	51-05-01-210-803 52-05-03-211-802	1.1	9 YR	9 YR	ALL	ALL			
	Inspect Fligh	nt Deck Security Door Ass	sembly, includ	ing Main Door Pan	el Assy, Main Doo	r Panel Bond Assy,	and Armor			

Inspect Flight Deck Security Door Assembly, including Main Door Panel Assy, Main Door Panel Bond Assy, and Armor Laminate Assy.

AIRPLANE NOTE: Applicable to airplane L/N 1221 and on and to those airplanes with the New Flight Deck Security Door installed by the customer specific Boeing Service Bulletins.

ACCESS NOTE: Disassemble door only if evidence of damage, fatigue, delamination, and or bulging is found.





				INTERVAL			ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
52-460-00-01	MRB	51-05-01-210-803 52-05-03-210-803	1.1	9 YR	9 YR	ALL	ALL			
	Inspect Flight Deck Security Door Surround Assembly, including Header Assembly, Post Assembly (right hand post), Latch and Deadbolt Receiver Assembly, Support Structure, Post Cover Armor Assembly and Hinge Assembly.									
	AIRPLANE	NOTE: Applicable to airp Door installed by		and on and to thos specific Boeing Se		the New Flight Dec	k Security			
	INTERVAL I	NOTE: Whichever comes	s first.							
	ACCESS NO	OTE: As visible with carp	et, tapestries ((if equipped) and ki	ck strips displace	d.				
52-470-00-01	MRB	51-05-01-210-803 52-05-03-211-803	1.1	9 YR	9 YR	ALL	ALL			
		nt Deck Security Door Su It Receiver Assembly, Su					nd post), Latcl			
	AIRPLANE	NOTE: Applicable to airp Door installed by		and on and to thos specific Boeing Se		the New Flight Dec	k Security			
	ACCESS NO	OTE: For access displac door frames, and c		-	sets, lavs, galley	s (if equipped) adjac	cent to			
52-490-00-01	MRB	51-05-01-210-809 52-05-03-211-828	1.1	9 YR	9 YR	ALL	ALL			
	Inspect the flight deck door latch and hinge support assemblies.									
	AIRPLANE NOTE: Applicable to airplane L/N 1221 and on and to those airplanes with the New Flight Deck Security									
	Door installed by the customer specific Boeing Service Bulletins.									
	ACCESS N	OTE: Disassemble door	only if evidenc	e of damage, fatigu	ie, delamination,	and or bulging is for	und.			
50 540 00 04	MDD	54.05.04.040.000		22.110	22.110					
52-510-00-01	MRB	51-05-01-210-809	1.1	36 MO			A.I.I.			
	Inspect forward access door stop fittings and pins.									
	Inspect forw	52-05-03-211-805 ard access door stop fittii	1.2 ngs and pins.	6600 FC	36 MO 6600 FC	ALL	ALL			
	•		ngs and pins.			ALL	ALL			
	INTERVAL I	ard access door stop fittii	ngs and pins.	6600 FC		ALL	ALL			
52-530-00-01	INTERVAL I	ard access door stop fittin NOTE: Whichever comes OTE: Inspect with door of	ngs and pins. s first. pened and lini	ng not removed.	6600 FC 8 YR	ALL	ALL			
52-530-00-01	ACCESS NO	ard access door stop fittii NOTE: Whichever come: OTE: Inspect with door of 51-05-01-210-804 52-05-03-210-806	ngs and pins. s first. pened and lini 1.1 1.2	6600 FC	6600 FC					
52-530-00-01	MRB	ard access door stop fittin NOTE: Whichever comes OTE: Inspect with door of 51-05-01-210-804 52-05-03-210-806 ard access door skin and	ngs and pins. s first. pened and lini 1.1 1.2 structure.	ng not removed.	6600 FC 8 YR					
52-530-00-01	MRB	ard access door stop fittii NOTE: Whichever come: OTE: Inspect with door of 51-05-01-210-804 52-05-03-210-806	ngs and pins. s first. pened and lini 1.1 1.2 structure.	ng not removed.	6600 FC 8 YR					
52-530-00-01	MRB Inspect forw	ard access door stop fittin NOTE: Whichever comes OTE: Inspect with door of 51-05-01-210-804 52-05-03-210-806 ard access door skin and	ngs and pins. s first. pened and lini 1.1 1.2 structure. s first.	ng not removed.	6600 FC 8 YR					
52-530-00-01 52-540-00-01	MRB Inspect forw	ard access door stop fittin NOTE: Whichever comes OTE: Inspect with door of 51-05-01-210-804 52-05-03-210-806 ard access door skin and NOTE: Whichever comes	ngs and pins. s first. pened and lini 1.1 1.2 structure. s first.	ng not removed.	6600 FC 8 YR					

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Inspect with door removed as required.







				INTERVAL			ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
52-550-00-01	MRB	51-05-01-210-809 52-05-03-211-807	1.1 1.2	8 YR 18000 FC	8 YR 18000 FC	ALL	ALL		
	Inspect E/E equipment compartment access door stop fittings and pins.								
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with door a	and access par	nel removed. Remo	ove dagger pins a	s required.			
52-570-00-01	MRB	51-05-01-210-804 52-05-03-210-807	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL		
	Inspect E/E	equipment compartment	door skin and	structure.					
	INTERVAL N	NOTE: Whichever come	s first.						
	ACCESS NO	OTE: Inspect with door a	and access par	nel removed. Remo	ove dagger pins a	s required.			
52-610-00-01	MRB	51-05-01-210-809 52-05-03-211-809	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL		
	Inspect forwa	Inspect forward entry door stop fittings and pins.							
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with doors	opened and lin	ning not removed.					
52-610-00-02	MRB	51-05-01-210-809 52-05-03-211-810	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL		
	Inspect forward galley service door stop fittings and pins.								
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with doors	opened and lin	ning not removed.					
52-610-00-03	MRB	51-05-01-210-809 52-05-03-211-811	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL		
	Inspect aft e	ntry door stop fittings and	d pins.						
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with doors	opened and lin	ning not removed.					
52-610-00-04	MRB	51-05-01-210-809 52-05-03-211-812	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL		
	Inspect aft g	Inspect aft galley service door stop fittings and pins.							
	INTERVAL N	INTERVAL NOTE: Whichever comes first.							
		OTE: Inspect with doors		ning not removed.					
52-620-00-01	MRB	51-05-01-210-810 52-05-03-211-813	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL		
	Inopost form	52-05-03-211-813 1.2 18000 FC 18000 FC							

Inspect forward entry door stop fittings and pins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove insulation, interior liners and access panels as required.





TASK CARD NO. SOURCE AMM TASK REF VERSION THRESHOLD REPEAT AIRPLANE 52-620-00-02 MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL Inspect forward galley service door stop fittings and pins. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-620-00-03 MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL Inspect aft entry door stop fittings and pins. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-620-00-04 MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL 52-620-00-04 MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL 52-620-00-04 MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL 52-650-00-04 MRB 51-05-01-210-808 1.2 18000 FC 18000 FC ALL 52-650-00-01 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL <t< th=""><th></th><th></th><th></th><th></th><th>INTERVAL</th><th colspan="2">APPLICABILITY</th></t<>					INTERVAL	APPLICABILITY				
S2-620-00-02 Inspect forward galley service door stop fittings and pins.	TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
INTERVAL NOTE: Whichever comes first.	52-620-00-02	MRB					ALL	ALL		
MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL		Inspect forward galley service door stop fittings and pins.								
MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL		INTERVAL N	NOTE: Whichever come	s first.						
S2-05-03-211-815		ACCESS NO	OTE: Remove insulation	, interior liners	and access panels	s as required.				
INTERVAL NOTE: Whichever comes first.	52-620-00-03	MRB					ALL	ALL		
### ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-620-00-04 MRB		Inspect aft e	ntry door stop fittings and	d pins.						
MRB 51-05-01-210-810 1.1 9 YR 8 YR ALL		INTERVAL N	NOTE: Whichever come	s first.						
Inspect aft galley service door stop fittings and pins. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-01 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-810 1.2 18000 FC 18000 FC Inspect forward entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-811 1.2 18000 FC 18000 FC Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL S2-05-03-210-811 1.2 18000 FC 18000 FC IS000 F		ACCESS NO	OTE: Remove insulation	ı, interior liners	and access panels	s as required.				
INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-01 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-810 1.2 18000 FC 18000 FC Inspect forward entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-811 1.2 18000 FC 18000 FC Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.	52-620-00-04	MRB					ALL	ALL		
ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-01 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-810 1.2 18000 FC 18000 FC 18000 FC Inspect forward entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 18000 FC 18000 FC Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.		Inspect aft galley service door stop fittings and pins.								
MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL		INTERVAL NOTE: Whichever comes first.								
S2-05-03-210-810 1.2 18000 FC 18000 FC		ACCESS NO	OTE: Remove insulation	ı, interior liners	and access panels	s as required.				
INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-811 1.2 18000 FC 18000 FC Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.	52-650-00-01	MRB					ALL	ALL		
ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-811 1.2 18000 FC 18000 FC Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.										
52-650-00-02 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-811 1.2 18000 FC 18000 FC Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.										
S2-05-03-210-811 1.2 18000 FC 18000 FC		ACCESS NO	OTE: Remove insulation	, interior liners	and access panels	s as required.				
Inspect forward galley service door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.	52-650-00-02	MRB					ALL	ALL		
INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.		Inspect forwa								
ACCESS NOTE: Remove insulation, interior liners and access panels as required. 52-650-00-03 MRB 51-05-01-210-808 1.1 9 YR 8 YR ALL 52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.										
52-05-03-210-812 1.2 18000 FC 18000 FC Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.										
Inspect aft entry door skin and structure. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.	52-650-00-03	MRB					ALL	ALL		
INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Remove insulation, interior liners and access panels as required.		Inspect aft e								
ACCESS NOTE: Remove insulation, interior liners and access panels as required.										
52-650-00-04 MRB 51-05-01-210-808 1.1 9.ΥR 8.ΥR ΔΙΙ					and access panels	s as required.				
52-05-03-210-813 1.2 18000 FC 18000 FC	52-650-00-04	MRB	51-05-01-210-808 52-05-03-210-813	1.1	9 YR 18000 FC	8 YR	ALL	ALL		

Inspect aft galley service door skin and structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove insulation, interior liners and access panels as required.





				INTERVAL			ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
52-670-00-01	MRB	51-05-01-210-809 52-05-03-211-817	1.1 1.2	36 MO 6600 FC	36 MO 6600 FC	ALL	ALL		
	Inspect the f	orward cargo door stop f							
	•	NOTE: Whichever come							
	ACCESS NO	OTE: Inspect with doors	opened and li	ning not removed.					
52-670-00-02	MRB	51-05-01-210-809 52-05-03-211-818	1.1 1.2	36 MO 6600 FC	36 MO 6600 FC	ALL	ALL		
	Inspect the a	aft cargo door stop fitting	s and pins.						
	INTERVAL I	NOTE: Whichever come	s first.						
	ACCESS NO	OTE: Inspect with doors	opened and lin	ning not removed.					
	1400	54.05.04.040.044		0.1/5	0.1/D	A			
52-680-00-01	MRB	51-05-01-210-811 52-05-03-211-819	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL		
	Inspect forward cargo door stop fittings and pins.								
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with insula	tion blanket re	moved.					
52-680-00-02	MRB	51-05-01-210-811	1.1	9 YR	8 YR	ALL	ALL		
		52-05-03-211-820	1.2	18000 FC	18000 FC				
	Inspect aft cargo door stop fittings and pins.								
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with insula	tion blanket re	moved.					
52-710-00-01	MRB	51-05-01-210-808	1.1	9 YR	8 YR	ALL	ALL		
		52-05-03-210-814	1.2	18000 FC	18000 FC				
	Inspect forward cargo door skin and structure.								
	INTERVAL NOTE: Whichever comes first.								
	ACCESS NO	OTE: Inspect with insula	tion blanket re	moved.					
52-710-00-02	MRB	51-05-01-210-808	1.1	9 YR	8 YR	ALL	ALL		
	Inchest off a	52-05-03-210-815	1.2	18000 FC	18000 FC				
	Inspect aft cargo door skin and structure.								
	INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Inspect with insulation blanket removed.								
	ACCESS NO	OIE: Inspect with insula	uon dianket re	movea.					
52-730-00-01	MRB	51-05-01-210-809 52-05-03-211-821	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL		
	Inchest suts	matic overwing exit door							

AIRPLANE NOTE: Zone 832 and 842 are applicable to 737-800 and 737-900 only.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Inspect with doors opened and lining not removed.





				INTERVAL		APPLICABILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
52-730-00-02	MRB	51-05-01-210-809 52-05-03-211-822	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL				
	Inspect autor	matic overwing exit door	stop fittings ar	nd pins.							
	AIRPLANE I	NOTE: Zone 832 and 84	42 are applicab	le to 737-800 and	737-900 only.						
	INTERVAL N	IOTE: Whichever come	s first.								
	ACCESS NO	OTE: Inspect with doors	opened and lir	ning not removed.							
52-740-00-01	MRB	51-05-01-210-810 52-05-03-211-823	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL				
	Inspect the le	eft automatic overwing e			1000010						
	•	NOTE: Zone 832 and 84	·		737-900 only						
		IOTE: Whichever come			707 000 01.ily.						
		OTE: Inspect with hatch		e door opened or r	emoved Remove	linings and insulati	ons				
	ACCESO NO	TE. Inspect with hater	os removed, ur	e door opened or re	omoved. Remove	illings and insulati	0113.				
52-740-00-02	MRB	51-05-01-210-810 52-05-03-211-824	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL				
	Inspect right automatic overwing exit door stop fittings and pins.										
	AIRPLANE NOTE: Zone 832 and 842 are applicable to 737-800 and 737-900 only.										
	INTERVAL NOTE: Whichever comes first.										
	ACCESS NOTE: Inspect with hatches removed, the door opened or removed. Remove linings and insulations.										
52-760-00-01	MRB	51-05-01-210-808 52-05-03-210-816	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL				
	Inspect the le	eft automatic overwing e	xit door skin ar	nd structure.							
	AIRPLANE I	AIRPLANE NOTE: Zone 832 and 842 are applicable to 737-800 and 737-900 only.									
	INTERVAL N	INTERVAL NOTE: Whichever comes first.									
	ACCESS NOTE: Inspect with hatches removed, door opened or removed. Remove linings and insulations.										
52-760-00-02	MRB	51-05-01-210-808 52-05-03-210-817	1.1 1.2	9 YR 18000 FC	8 YR 18000 FC	ALL	ALL				
52-760-00-02		02 00 00 210 011		Inspect the right automatic overwing exit door skin and structure.							
52-760-00-02	Inspect the r		exit door skin a	and structure.							
52-760-00-02	•				737-900 only.						
52-760-00-02	AIRPLANE I	ight automatic overwing	42 are applicab		737-900 only.						
52-760-00-02	AIRPLANE I	ight automatic overwing NOTE: Zone 832 and 84	42 are applicab s first.	le to 737-800 and	·	linings and insulati	ons.				
52-760-00-02	AIRPLANE I	ight automatic overwing NOTE: Zone 832 and 84 NOTE: Whichever come	42 are applicab s first.	le to 737-800 and	·	linings and insulati	ons.				

See Doc. D626A001-DTR, DTR check form 52-21-04 for alternative inspections.

AIRPLANE NOTE: Panels 832AZ and 842AZ are applicable to 737-800 and 737-900 only.

ACCESS NOTE: Removal of the lining and the stop guide is required (note that the stop guide fitting is attached

with removable bolts).







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
52-794-00-02	AWL	52-05-02-211-801	1.1	50000 FC	18000 FC	ALL	ALL
		ailed) the lower frame at a 26A001-DTR, DTR chec		•			
	AIRPLANE	NOTE: Panels 832AZ ar	nd 842AZ are a	applicable to 737-8	00 and 737-900 o	nly.	
	ACCESS NO	OTE: Removal of the lining with removable boli	-	p guide is required	(note that the sto	p guide fitting is atta	ached
52-794-00-03	AWL	52-05-02-211-801	1.1	50000 FC	18000 FC	ALL	ALL
		ailed) the lower frame at a 26A001-DTR, DTR chec		•			
	AIRPLANE	NOTE: Panels 832AZ ar	nd 842AZ are a	applicable to 737-8	00 and 737-900 o	nly.	
	ACCESS NO	OTE: Removal of the lining with removable bolice.		p guide is required	(note that the sto	p guide fitting is atta	ached
52-794-00-04	AWL	52-05-02-211-801	1.1	50000 FC	18000 FC	ALL	ALL
		ailed) the lower frame at a					
	OCC DOC. DO						
		NOTE: Panels 832AZ ar	nd 842AZ are a	applicable to 737-8	00 and 737-900 o	nly.	
	AIRPLANE	NOTE: Panels 832AZ ar OTE: Removal of the lining with removable bole.	ng and the stop	• •		·	ached
52-796-00-01	AIRPLANE	OTE: Removal of the lining	ng and the stop	• •		·	ached ALL
52-796-00-01	AWL Inspect (Deta (4) door pin I	DTE: Removal of the lining with removable boles 52-05-02-211-802 ailed) the area around the ocations bounded by LBI	ng and the stop ts). 1.1 e fastener loca L 5.70, RBL 14	p guide is required 50000 FC tions common to tl	(note that the sto 4000 FC ne inner panel and 5.07 and STA 349	p guide fitting is atta ALL I the door frame ad	ALL
52-796-00-01	AWL Inspect (Deta (4) door pin I See Doc. D6	DTE: Removal of the lining with removable bole 52-05-02-211-802 ailed) the area around the	ng and the stop ts). 1.1 e fastener loca L 5.70, RBL 14 k form 52-48-0	p guide is required 50000 FC tions common to tl	(note that the sto 4000 FC ne inner panel and 5.07 and STA 349	p guide fitting is atta ALL I the door frame ad	ALL
52-796-00-01 52-800-00-01	AWL Inspect (Deta (4) door pin I See Doc. D6	DTE: Removal of the lining with removable bold to the second seco	ng and the stop ts). 1.1 e fastener loca L 5.70, RBL 14 k form 52-48-0	p guide is required 50000 FC tions common to tl	(note that the sto 4000 FC ne inner panel and 5.07 and STA 349	p guide fitting is atta ALL I the door frame ad	ALL
	AWL Inspect (Deta (4) door pin I See Doc. D6 ACCESS NO	DTE: Removal of the lining with removable bole	1.1 e fastener loca L 5.70, RBL 14 ek form 52-48-0 of the door. 1.1 (GV) of the do	50000 FC tions common to tl 1.12, frame STA 32 04-2 for alternative 120 DY ors. Inspection is a	4000 FC ne inner panel and 5.07 and STA 349 inspections.	ALL d the door frame ad 1.13.	ALL jacent to the f
	AWL Inspect (Deta (4) door pin I See Doc. D6 ACCESS NC	DTE: Removal of the lining with removable bold to the area around the ocations bounded by LBI (26A001-DTR, DTR check) access the interior 05-41-08-210-801 external zonal inspection	1.1 e fastener loca L 5.70, RBL 14 k form 52-48-0 of the door. 1.1 (GV) of the do ss panel is req	50000 FC Itions common to tl 1.12, frame STA 32 04-2 for alternative 120 DY ors. Inspection is a puired.	4000 FC the inner panel and 5.07 and STA 349 inspections.	ALL d the door frame ad 1.13.	ALL jacent to the f
	AWL Inspect (Deta (4) door pin I See Doc. Do ACCESS NO MRB Perform an estands or lace AIRPLANE I	DTE: Removal of the lining with removable bole 52-05-02-211-802 called) the area around the ocations bounded by LBI 26A001-DTR, DTR chec DTE: Access the interior 05-41-08-210-801 external zonal inspection liders. No additional acces	1.1 e fastener loca L 5.70, RBL 14 k form 52-48-0 of the door. 1.1 (GV) of the do ss panel is req	50000 FC Itions common to tl 1.12, frame STA 32 04-2 for alternative 120 DY ors. Inspection is a puired.	4000 FC the inner panel and 5.07 and STA 349 inspections.	ALL d the door frame ad 1.13.	ALL jacent to the f
	AWL Inspect (Deta (4) door pin I See Doc. Do ACCESS NO MRB Perform an estands or lace AIRPLANE I	DTE: Removal of the lining with removable bold to the second of the lining with removable bold to the second of th	1.1 e fastener loca L 5.70, RBL 14 k form 52-48-0 of the door. 1.1 (GV) of the do ss panel is req	50000 FC Itions common to tl 1.12, frame STA 32 04-2 for alternative 120 DY ors. Inspection is a puired.	4000 FC the inner panel and 5.07 and STA 349 inspections.	ALL d the door frame ad 1.13.	ALL jacent to the f
52-800-00-01	AWL Inspect (Deta (4) door pin I See Doc. Do ACCESS NO MRB Perform an estands or lace AIRPLANE I ACCESS NO MRB	DTE: Removal of the lining with removable bold to the area around the ocations bounded by LBI (26A001-DTR, DTR check) and the ocations to the ocations bounded by LBI (26A001-DTR, DTR check) and the ocations to the ocations of the ocations ocations of the ocations ocations ocations ocations ocations ocations ocations	1.1 e fastener loca L 5.70, RBL 14 ck form 52-48-0 of the door. 1.1 (GV) of the do ss panel is req nly on 900ER.	50000 FC tions common to tl 1.12, frame STA 32 04-2 for alternative 120 DY ors. Inspection is a juired. 2000 FC 240 DY	4000 FC ne inner panel and 5.07 and STA 349 inspections. 120 DY ccomplished from	ALL d the door frame add.13. ALL at the ground, without	ALL jacent to the f
52-800-00-01	AWL Inspect (Deta (4) door pin I See Doc. D6 ACCESS NC MRB Perform an e stands or lac AIRPLANE I ACCESS NC	DTE: Removal of the lining with removable bold of the lining with removal of the lining with removable bold of the lining wi	1.1 e fastener loca L 5.70, RBL 14 ek form 52-48-0 of the door. 1.1 (GV) of the do ss panel is req nly on 900ER. 1.1 1.2 (GV) of the for	50000 FC tions common to tl 1.12, frame STA 32 04-2 for alternative 120 DY ors. Inspection is a juired. 2000 FC 240 DY	4000 FC ne inner panel and 5.07 and STA 349 inspections. 120 DY ccomplished from	ALL d the door frame add.13. ALL at the ground, without	ALL jacent to the f

Perform an internal zonal inspection (GV) of the forward cargo door - section 43, sta 460.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
52-806-02-01	MRB	05-41-08-210-804	1.1 1.2	2000 FC 240 DY	2000 FC 240 DY	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the aft	cargo door - section	on 46, sta 827.		
	INTERVAL N	NOTE: Whichever come	s first.				
52-808-02-01	MRB	05-41-08-210-805	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the aft	cargo door - sectio	n 46, sta 827.		
	INTERVAL N	NOTE: Whichever come	s first.				
52-810-01-01	MRB	05-41-08-210-806	1.1 1.2	1500 FC 180 DY	1500 FC 180 DY	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the for	ward passenger do	oor - section 41, s	ta 345.	
	INTERVAL N	NOTE: Whichever come	s first.				
52-812-01-01	MRB	05-41-08-210-807	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the for	ward passenger do	or - section 41, st	a 345.	
	INTERVAL N	NOTE: Whichever come	s first.				
52-814-01-01	MRB	05-41-08-210-808	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	800 900	ALL
	Perform an e	external zonal inspection	(GV) of the au	tomatic overwing e	xit - section 44, s	ta 589.5.	
	INTERVAL N	NOTE: Whichever come	s first.				
52-816-01-01	MRB	05-41-08-210-809	1.1 1.2	18000 FC 9 YR	18000 FC 8 YR	800 900	ALL
	Perform an i	nternal zonal inspection	(GV) of the aut	omatic overwing ex	kit - section 44, st	a 589.5.	
	INTERVAL N	NOTE: Whichever come	s first.	· ·			
	ACCESS NO	OTE: Automatic overwing	g exit door line	r removal required.			
52-818-01-01	MRB	05-41-08-210-810	1.1	5500 FC	5500 FC	ALL	ALL
			1.2	30 MO	30 MO	0.7	
		external zonal inspection	` '	tomatic overwing e	exit - section 44, s	ta 627.	
	INTERVAL N	NOTE: Whichever come	s first.				
52-820-01-01	MRB	05-41-08-210-811	1.1 1.2	18000 FC 9 YR	18000 FC 8 YR	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the aut	omatic overwing ex	rit - section 44 S	ΤΔ 627	

Perform an internal zonal inspection (GV) of the automatic overwing exit - section 44, STA 627.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Automatic overwing exit door liner removal required.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
52-822-01-01	MRB	05-41-08-210-812	1.1 1.2	1500 FC 180 DY	1500 FC 180 DY	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the aft	passenger door - s	section 47, sta 98	0.	
	INTERVAL I	NOTE: Whichever come	s first.				
52-824-01-01	MRB	05-41-08-210-813	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection)	
		NOTE: Whichever come	,	passenger ass. s	.,,		
52-826-02-01	MRB	05-41-08-210-816	1.1 1.2	1500 FC 180 DY	1500 FC 180 DY	ALL	ALL
	Perform an	external zonal inspection	(GV) of the for	ward galley service	door - section 4	1, sta 340.	
	INTERVAL I	NOTE: Whichever come	es first.				
				2000 50	0000 FC	ALL	ALL
52-828-02-01	MRB	05-41-08-210-817	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
52-828-02-01			1.2	36 MO	36 MO		ALL
52-828-02-01	Perform an i	05-41-08-210-817 Internal zonal inspection NOTE: Whichever come	1.2 (GV) of the for	36 MO	36 MO		ALL
52-828-02-01	Perform an i	nternal zonal inspection	1.2 (GV) of the for	36 MO	36 MO		ALL
52-828-02-01 52-830-02-01	Perform an i	nternal zonal inspection	1.2 (GV) of the for	36 MO	36 MO		ALL
	Perform an i	nternal zonal inspection	1.2 (GV) of the formula first.	36 MO ward galley service 5500 FC 30 MO	36 MO door - section 41 5500 FC 30 MO	, STA 340. 800 900	
	Perform an i	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the au	36 MO ward galley service 5500 FC 30 MO	36 MO door - section 41 5500 FC 30 MO	, STA 340. 800 900	
52-830-02-01	Perform an i	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the aues first.	36 MO ward galley service 5500 FC 30 MO tomatic overwing e	36 MO door - section 41 5500 FC 30 MO xit - section 44, s	800 900 ta 589.5.	ALL
	Perform an i	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the au	36 MO ward galley service 5500 FC 30 MO	36 MO door - section 41 5500 FC 30 MO	, STA 340. 800 900	
52-830-02-01	Perform an i	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come	1.2 (GV) of the formula first. 1.1 1.2 (GV) of the august first. 1.1 1.2	36 MO ward galley service 5500 FC 30 MO tomatic overwing e 18000 FC 9 YR	36 MO door - section 41 5500 FC 30 MO xit - section 44, s 18000 FC 8 YR	800 900 ta 589.5.	ALL
52-830-02-01	Perform an i INTERVAL I MRB Perform an e INTERVAL I MRB Perform an i	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come 05-41-08-210-819	1.2 (GV) of the formula first. 1.1 1.2 (GV) of the autor first. 1.1 1.2 (GV) of the autor first.	36 MO ward galley service 5500 FC 30 MO tomatic overwing e 18000 FC 9 YR	36 MO door - section 41 5500 FC 30 MO xit - section 44, s 18000 FC 8 YR	800 900 ta 589.5.	ALL
52-830-02-01	Perform an i INTERVAL I MRB Perform an e INTERVAL I MRB Perform an i INTERVAL I	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come 05-41-08-210-819 internal zonal inspection	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the autor first. 1.1 1.2 (GV) of the autor first.	36 MO ward galley service 5500 FC 30 MO tomatic overwing example of the service overwing example of the service overwing example overwing exa	36 MO door - section 41 5500 FC 30 MO xit - section 44, st 18000 FC 8 YR	800 900 ta 589.5.	ALL
52-830-02-01	Perform an i INTERVAL I MRB Perform an i INTERVAL I MRB Perform an i INTERVAL I ACCESS NO	internal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come 05-41-08-210-819 internal zonal inspection NOTE: Whichever come	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the autos first. 1.1 1.2 (GV) of the autos first. (GV) of the autos first. g exit door line	36 MO ward galley service 5500 FC 30 MO tomatic overwing exervice 18000 FC 9 YR omatic overwing exervice	36 MO door - section 41 5500 FC 30 MO xit - section 44, st 18000 FC 8 YR	800 900 ta 589.5.	ALL
52-830-02-01 52-832-02-01	Perform an i INTERVAL I MRB Perform an e INTERVAL I MRB Perform an i INTERVAL I	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come 05-41-08-210-819 Internal zonal inspection NOTE: Whichever come	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the autor first. 1.1 1.2 (GV) of the autor first.	36 MO ward galley service 5500 FC 30 MO tomatic overwing example of the service overwing example of the service overwing example overwing exa	36 MO door - section 41 5500 FC 30 MO xit - section 44, st 18000 FC 8 YR	800 900 ta 589.5.	ALL
52-830-02-01 52-832-02-01	Perform an i INTERVAL I MRB Perform an i INTERVAL I MRB Perform an i INTERVAL I ACCESS NO	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come 05-41-08-210-819 Internal zonal inspection NOTE: Whichever come	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the autor first. 1.1 1.2 (GV) of the autor first. g exit door line 1.1 1.2	36 MO ward galley service 5500 FC 30 MO tomatic overwing extra removal required. 5500 FC 30 MO	36 MO door - section 41 5500 FC 30 MO xit - section 44, s 18000 FC 8 YR cit - section 44, st 5500 FC 30 MO	800 900 ta 589.5. 800 900 a 589.5.	ALL
52-830-02-01 52-832-02-01	Perform an ininterval initerval init	nternal zonal inspection NOTE: Whichever come 05-41-08-210-818 external zonal inspection NOTE: Whichever come 05-41-08-210-819 internal zonal inspection NOTE: Whichever come OTE: Automatic overwin 05-41-08-210-820	1.2 (GV) of the forces first. 1.1 1.2 (GV) of the autor first. 1.1 1.2 (GV) of the autor first. g exit door line 1.1 1.2 (GV) of the autor first.	36 MO ward galley service 5500 FC 30 MO tomatic overwing extra removal required. 5500 FC 30 MO	36 MO door - section 41 5500 FC 30 MO xit - section 44, s 18000 FC 8 YR cit - section 44, st 5500 FC 30 MO	800 900 ta 589.5. 800 900 a 589.5.	ALL

Perform an internal zonal inspection (GV) of the automatic overwing exit - section 44, sta 627.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Automatic overwing exit door liner removal required.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
52-838-02-01	MRB	05-41-08-210-822	1.1 1.2	1500 FC 180 DY	1500 FC 180 DY	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the aft	galley service door	r - section 47, sta	a 980.	
	INTERVAL N	NOTE: Whichever come	s first.				
52-840-02-01	MRB	05-41-08-210-823	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the aft	galley service door	- section 47, sta	980.	
	INTERVAL N	NOTE: Whichever come	s first.				
53-010-00-01	MRB	51-05-01-210-801 53-05-03-210-801	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL
	Inspect door	cutout at forward access	s door.				
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Open forward acce	ess door.				
53-010-00-02	MRB	51-05-01-210-801 53-05-03-210-802	1.1 1.2	36 MO 4000 FC	36 MO 4000 FC	ALL	ALL
	Inspect door	cutout at EE Compartme	ent door.				
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Open EE compartr	ment door.				
53-020-00-01	MRB	51-05-01-210-806 53-05-03-210-804	1.1	12 YR	6 YR	ALL	ALL
		landing gear wheel well nion support fitting, actua				Sta 294.5 bulkhead,	side and top
53-030-00-01	MRB	51-05-01-210-809 53-05-03-211-801	1.1 1.2	36 MO 6600 FC	36 MO 6600 FC	ALL	ALL
	Inspect the fe	orward cargo door surro	und structure, f	ittings and stops.			
	INTERVAL N	NOTE: Whichever come	s first.				
53-030-00-02	MRB	NOTE: Whichever come 51-05-01-210-809 53-05-03-211-802	1.1 1.2	36 MO 6600 FC	36 MO 6600 FC	ALL	ALL
53-030-00-02	MRB	51-05-01-210-809	1.1 1.2	6600 FC		ALL	ALL
53-030-00-02	MRB Inspect the a	51-05-01-210-809 53-05-03-211-802	1.1 1.2 tructure, fitting	6600 FC		ALL	ALL

Inspect main landing gear wheel well, including:



^{1.} Pressure deck web and stiffeners, including attachment to wing center section rear spar at Sta 663; 2. Bulkhead at STA 663; 3. Bulkhead and pressure web at STA 727; 4. Keel beam chords, webs, stiffeners and splice, keel beam/rear spar attachment angles; 5. Stringer 18A web, chord and links; 6. Side strut support frame at STA 706; 7. Main landing gear support frame at STA 695 and 716; 8. Wheel well frame at STA 685; 9. Flap track support fittings.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-060-00-01	MRB	51-05-01-210-806 53-05-03-210-806	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	Inspect the f	orward side of STA 178 I	bulkhead.				
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Open nose radom	e.				
53-070-00-01	MRB	51-05-01-210-808 53-05-03-210-807	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
		age lower lobe from STAs, frames, stringers), long					
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove weather r	adar RT moun	t. Remove/displace	insulation blanke	ets as required.	
53-080-00-01	MRB	51-05-01-210-806 53-05-03-210-808	1.1	10 YR	10 YR	ALL	ALL
	Inspect flight	compartment floor struc	cture from lowe	er lobe.			
	ACCESS NO	OTE: Access through for	rward access o	door			
53-090-00-01	MRB	51-05-01-210-808 53-05-03-210-809	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	 Skin pane 259.5; Nose whe 	age lower lobe from can ls (skins, frames, stringe el well cutout surround s	ers), longitudina	al lap splices, circur wheel well side and	mferential skin an d top panels;		ulkhead at St
		support fitting, actuator s		nd drag brace fitting	g.		
		NOTE: Whichever come					
	ACCESS NO	DTE: Access through no floor. Remove/disp		side and top acces blankets as require		ough access panel i	n crew
53-100-00-01	MRB	51-05-01-210-808 53-05-03-210-810	1.1	12 YR	12 YR	ALL	ALL

Inspect passenger compartment floor structure in dry areas (away from doors, galleys and lavs) from lower lobe.

ACCESS NOTE: Remove ceiling and sidewall panels as required. Remove/displace insulation blankets as required. Remove or displace auxiliary fuel tank as required (business jet only). Remove forward airstairs and airstairs compartment (if installed).







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-110-00-01	MRB	51-05-01-210-808 53-05-03-210-811	1.1	NOTE		ALL	ALL
	Inspect pass from lower lo	senger compartment floo bbe.	r structure in w	et areas (within ap	proximately 20 inc	ches from doors, ga	lleys and lav
	INTERVAL N		those that have SYR, whichever	e not incorporated occurs first, applic	the HI-TAK Gel T	t, applicable to Airp ape. Threshold Inte L/N# 2413 and on,	rval 9YR /
	ACCESS NO	OTE: Remove ceiling ar Remove or displac and airstair compa	e auxiliary fuel	tank as required (•	sulation blankets as Remove forward a	•
53-120-00-01	MRB	51-05-01-210-808 53-05-03-210-812	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	longitudinal l	compartment (STA 294.5 lap splices, circumferenti r cutout surround structu	ial skin and stri	nger splices, bulkh	ead at STA 294.5	; 2. EE compartmer	nt door and
	INTERVAL N	NOTE: Whichever come	s first.				
53-130-00-01	ACCESS NO	DTE: Remove LRUs and insulation blankets 51-05-01-210-808	as required.	12 YR	8 YR	ure. Remove/displa	ALL
	Inspect forward	53-05-03-210-813 ard cargo compartment	1.2 skin panels incl	36000 FC uding skins, frame	24000 FC s, and stringers (r	note: inspection incl	udes the
	circumferent	ial skin and stringer splic	ce at STA 500E	for the -900 mode	ls).		
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove sidewalls displace auxiliary f	٠.	nels. Remove/displ uired (business jet		nkets as required.	Remove/
53-140-00-01	MRB	51-05-01-210-808 53-05-03-210-814	1.1 1.2	8 YR 24000 FC	6 YR 18000 FC	ALL	ALL
	Inspect forward	ard cargo compartment t	floor structure.				
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove cargo floo	or panels.				
53-140-00-02	MRB	51-05-01-210-808 53-05-03-210-815	1.1 1.2	8 YR 24000 FC	6 YR 18000 FC	ALL	ALL
	Inspect Aft c	argo compartment floor:	structure.				

Inspect Aft cargo compartment floor structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cargo floor panels.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-150-00-01	MRB	51-05-01-210-808 53-05-03-211-803	1.1 1.2	9 YR 24000 FC	6 YR 18000 FC	ALL	ALL
	Inspect forwa	ard cargo door cutout su	rround structur	e.			
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove door reve required.	eals. Remove si	dewalls as required	d. Remove/displa	ce insulation blanke	ets as
53-160-00-01	MRB	51-05-01-210-808 53-05-03-210-816	1.1 1.2	8 YR 24000 FC	6 YR 18000 FC	ALL	ALL
		ard bilge skin panels incl bilge (note: inspection inc	•		•		
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove cargo floo	or panels and s	cuff plates. Remove	e/Displace insulat	ion blankets as req	uired.
53-170-00-01	MRB	51-05-01-210-808 53-05-03-210-817	1.1 1.2	12 YR 36000 FC	10 YR 36000 FC	ALL	ALL
		aft of forward cargo con ial skin and stringer splic	•	. ,		- , -	lap splices,
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove forward c	argo compartm ducting as requ		anels. Remove/dis	splace insulation bla	
		required. Remove	adding ad rook	uirea.			ankets as
53-180-00-01	MRB	51-05-01-210-806 53-05-03-210-818	1.1	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
53-180-00-01	Inspect area	51-05-01-210-806	1.1 1.2 dy fairing (forwa	12 YR 36000 FC ard of wing box), inc	24000 FC		ALL
53-180-00-01	Inspect area beam extens	51-05-01-210-806 53-05-03-210-818 under lower wing-to-boo	1.1 1.2 dy fairing (forwa	12 YR 36000 FC ard of wing box), inc	24000 FC		ALL
53-180-00-01 53-190-00-01	Inspect area beam extens	51-05-01-210-806 53-05-03-210-818 under lower wing-to-boo sion, wing-to-body drag a	1.1 1.2 dy fairing (forwa	12 YR 36000 FC ard of wing box), inc	24000 FC		ALL
	Inspect area beam extens INTERVAL M MRB	51-05-01-210-806 53-05-03-210-818 under lower wing-to-boo sion, wing-to-body drag a NOTE: Whichever come	1.1 1.2 dy fairing (forward angles, and States first. 1.1 1.2 ing box center states and states first.	12 YR 36000 FC and of wing box), inc 540 bulkhead. 12 YR 36000 FC section upper pane	24000 FC cluding skin panel 10 YR 36000 FC	s, longitudinal lap s ALL	ALL splices, keel
	Inspect area beam extens INTERVAL N MRB Inspect fusel stringers), St	51-05-01-210-806 53-05-03-210-818 under lower wing-to-bod sion, wing-to-body drag a NOTE: Whichever come 51-05-01-210-808 53-05-03-210-819 age lower lobe above wi	1.1 1.2 dy fairing (forwaringles, and States first. 1.1 1.2 ing box center serwing frames a	12 YR 36000 FC and of wing box), inc 540 bulkhead. 12 YR 36000 FC section upper pane	24000 FC cluding skin panel 10 YR 36000 FC	s, longitudinal lap s ALL	ALL splices, keel
	Inspect area beam extens INTERVAL M MRB Inspect fusel stringers), St	51-05-01-210-806 53-05-03-210-818 under lower wing-to-bod sion, wing-to-body drag a NOTE: Whichever come 51-05-01-210-808 53-05-03-210-819 age lower lobe above with a 540 bulkhead, and over	1.1 1.2 dy fairing (forward angles, and States and States) 1.1 1.2 ing box center serwing frames as s first.	12 YR 36000 FC and of wing box), inc 540 bulkhead. 12 YR 36000 FC section upper pane and stub beams.	24000 FC cluding skin panel 10 YR 36000 FC I, including side s	s, longitudinal lap s ALL kin panels (skins, fr	ALL ALL
	Inspect area beam extens INTERVAL M MRB Inspect fusel stringers), St	51-05-01-210-806 53-05-03-210-818 under lower wing-to-bod sion, wing-to-body drag a NOTE: Whichever come 51-05-01-210-808 53-05-03-210-819 age lower lobe above with 540 bulkhead, and over	1.1 1.2 dy fairing (forward angles, and States and States) 1.1 1.2 ing box center serwing frames as s first.	12 YR 36000 FC and of wing box), inc 540 bulkhead. 12 YR 36000 FC section upper pane and stub beams.	24000 FC cluding skin panel 10 YR 36000 FC I, including side s	s, longitudinal lap s ALL kin panels (skins, fr	ALL splices, keel

Inspect fuselage lower lobe above main landing gear wheel well, including:

1. Pressure deck web to stiffeners, stiffener attachment to floor beam at STA 727; 2. Side skin panels, circumferential skin and stringer splice; 3. Bulkheads at STA 663 and 727; 4. Side strut support frame at STA 706; 5. Main landing gear support frame at STA 695 and 716; 6. Wheel well frame at STA 685.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove floor panels. Remove/displace insulation blankets as required.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-210-00-01	MRB	51-05-01-210-804 53-05-03-210-821	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
		beam under wing-to-boo lice, keel beam/rear spa) to 663.75), inclu	ding keel beam cho	rds, webs,
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Option 1: Remove access door. Option	•	, ,		192F) and open AC n access to access	•
53-220-00-01	MRB	51-05-01-210-806 53-05-03-210-823	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	Inspect keel spar attachm	beam in wheel well (Sta ent angles.	663.75 to 727)	, including keel bea	am chords, webs,	stiffeners, splice, ke	eel beam/rea
	INTERVAL N	IOTE: Whichever come	s first.				
53-230-00-01	MRB	51-05-01-210-808 53-05-03-210-824	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	splices, (note	argo compartment, includes: located at Sta 727I for ord and links; 4. Aft side	-900 and 727L	for -900ER model	s); 2. Stringer 18		•
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove sidewall a displace insulation Remove/displace a	blankets as re	quired. Remove/dis	splace vacuum lav	vacuum lav tank. Re v components as re	
53-240-00-01	MRB	51-05-01-210-808 53-05-03-211-804	1.1 1.2	8 YR 24000 FC	6 YR 18000 FC	ALL	ALL
	Inspect aft ca	argo door cutout surroun	d structure.				
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	PTE: Remove door reverequired.	als. Remove si	dewalls as required	d. Remove/displa	ce insulation blanke	ts as
53-250-00-01	MRB	51-05-01-210-808 53-05-03-210-825	1.1 1.2	8 YR 24000 FC	6 YR 18000 FC	ALL	ALL

Inspect aft bilge skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, (note: located at Sta 727I for -900 and 727L for -900ER models); Sta 727 bulkhead and pressure web, and cargo door cutout surround structure in bilge.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cargo floor panels and scuff plates. Remove/Displace insulation blankets as required.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-260-00-01	MRB	51-05-01-210-808 53-05-03-210-826	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	circumferent	aft of cargo compartmer ial skin and stringer splic cluding chords, pressure	es; 2. Aft entry	and galley door co	utout surround stru	cture in lower lobe	3. STA 1016
	AIRPLANE	NOTE: Task not applical	ble to -900ER a	and -800 with Flat	Pressure Bulkhea	d installed.	
	INTERVAL N	NOTE: Whichever comes	s first.				
	ACCESS NO	OTE: Remove aft cargo	compartment a	ft bulkhead panel	and potable water	tank.	
		Remove/displace i	nsulation blank	ets as required.			
53-270-00-01	MRB	51-05-01-210-808 53-05-03-210-827	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	•	under lower wing-to-bod	, ,	, .	•		
		ial skin splice, stringer 18	·	of body, stringer 18	BA (web, chords ar	nd links), and keel b	eam extensio
	INTERVAL N	NOTE: Whichever comes	s first.				
53-280-00-01	MRB	51-05-01-210-806 53-05-03-210-828	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
		33-03-03-210-020	1.2	30000 i C	2400010		
	Inspect area at side of bo	under above-wing wing-				al skin splices, and	stringer 18 str
	at side of bo	under above-wing wing-	to-body fairing			al skin splices, and	stringer 18 str
53-290-00-01	at side of bo	under above-wing wing- dy.	to-body fairing			al skin splices, and	stringer 18 str
53-290-00-01	at side of bo INTERVAL N MRB	under above-wing wing- dy. NOTE: Whichever come: 51-05-01-210-809	to-body fairing s first. 1.1 1.2	, including skin par 9 YR 24000 FC	nels, circumferenti 8 YR 24000 FC	,	
53-290-00-01	MRB Inspect the a	under above-wing wing- dy. NOTE: Whichever come: 51-05-01-210-809 53-05-03-210-829	to-body fairing s first. 1.1 1.2 utout structure	, including skin par 9 YR 24000 FC	nels, circumferenti 8 YR 24000 FC	,	
53-290-00-01	MRB Inspect the a	under above-wing wing-dy. NOTE: Whichever come: 51-05-01-210-809 53-05-03-210-829 automatic overwing exit c	to-body fairing s first. 1.1 1.2 utout structure s first.	, including skin par 9 YR 24000 FC	nels, circumferenti 8 YR 24000 FC	,	
53-290-00-01 53-310-00-01	MRB Inspect the a	under above-wing wing-dy. NOTE: Whichever comes 51-05-01-210-809 53-05-03-210-829 automatic overwing exit c	to-body fairing s first. 1.1 1.2 utout structure s first.	, including skin par 9 YR 24000 FC	nels, circumferenti 8 YR 24000 FC	,	
	MRB Inspect the a INTERVAL N ACCESS NO	under above-wing wing-dy. NOTE: Whichever come: 51-05-01-210-809 53-05-03-210-829 automatic overwing exit complete. NOTE: Whichever come: OTE: Open automatic overwing exit complete.	to-body fairing s first. 1.1 1.2 utout structure s first. verwing exits.	9 YR 24000 FC , fittings and stops	8 YR 24000 FC	ALL	ALL
	MRB Inspect the a INTERVAL N ACCESS NO	under above-wing wing-dy. NOTE: Whichever comes 51-05-01-210-809 53-05-03-210-829 automatic overwing exit complete. NOTE: Whichever comes DTE: Open automatic overwing exit complete.	to-body fairing s first. 1.1 1.2 utout structure s first. rerwing exits. 1.1 1.2	9 YR 24000 FC , fittings and stops	8 YR 24000 FC ;	ALL	ALL
	MRB Inspect the a INTERVAL N ACCESS NO MRB Inspect door	under above-wing wing-dy. NOTE: Whichever comes 51-05-01-210-809 53-05-03-210-829 automatic overwing exit comes NOTE: Whichever comes DTE: Open automatic ov 51-05-01-210-809 53-05-03-211-805	to-body fairing s first. 1.1 1.2 utout structure s first. verwing exits. 1.1 1.2 nd hinges on the	9 YR 24000 FC , fittings and stops	8 YR 24000 FC ;	ALL	ALL
	MRB Inspect the a INTERVAL N ACCESS NO MRB Inspect door INTERVAL N	under above-wing wing-dy. NOTE: Whichever comes 51-05-01-210-809 53-05-03-210-829 automatic overwing exit comes NOTE: Whichever comes TE: Open automatic overwing exit comes 51-05-01-210-809 53-05-03-211-805 frames, stops, latches a	to-body fairing s first. 1.1 1.2 utout structure s first. verwing exits. 1.1 1.2 nd hinges on the s first.	9 YR 24000 FC , fittings and stops	8 YR 24000 FC ;	ALL	ALL

Inspect door frames, stops, latches and hinges on forward galley door cutout surround structure.

INTERVAL NOTE: Whichever comes first.

 $\begin{tabular}{ll} \textbf{ACCESS NOTE:} & Open forward galley service door. \\ \end{tabular}$





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-310-00-03	MRB	51-05-01-210-809 53-05-03-211-807	1.1 1.2	36 MO 6600 FC	36 MO 6600 FC	ALL	ALL
	Inspect door	frames, stops, latches a	and hinges on a	aft entry cutout surre	ound structure.		
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Open aft entry doc	or.				
53-310-00-04	MRB	51-05-01-210-809 53-05-03-211-808	1.1 1.2	36 MO 6600 FC	36 MO 6600 FC	ALL	ALL
	Inspect door	frames, stops, latches a	and hinges on a	aft galley door cutou	ut surround struct	ure.	
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Open aft galley do	or.				
53-330-00-01	MRB	51-05-01-210-808 53-05-03-210-830	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
	cabin window include: Forv Forward and window fram Structure for	compartment, including v cutout structure, and s vard side of frame at ST, aft side of Frame 259.5 e structure); Forward sidward of STA 203.8 to ST er pedal); skin, frames and structure are structure.	tructure adjace A 259.5 and str and structure de of Frame 25 A 178 and fron	ent to ground block ructure 3 inches for from STA 249 to ST 9.5 and structure 3 n floor up to window	behind rudder pe ward of STA 259. IA 263 between fl inches forward o	dal. Inspection area 5; BL 0 + 4 inches (oor and S-5L (exclu f STA 259.5 betwee	does not Left and Right) ding window ar In floor and S-3
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	DTE: Remove glare shie blankets as require		head units and pan	nels as required. F	Remove/displace in:	sulation
53-335-00-01	MRB	51-05-01-210-808 53-05-03-210-861	1.1 1.2	20 YR 50000 FC	8 YR 24000 FC	ALL	ALL
	cabin window 259.5 and strand structure side of Fram	compartment, including v cutout structure, and for ructure 3 inches forward a from STA 249 to STA 2 e 259.5 and structure 3 in from floor up to window	orward pressur of STA 259.5; 63 between flo inches forward	e bulkhead within the BL 0 + 4 inches (Le or and S-5L (excluded) of STA 259.5 between	he following areas eft and Right); Fo ding window and een floor and S-3	s: Forward side of fi rward and aft side o window frame struc R; Structure forwar	rame at STA of Frame 259.5 ture); Forward

ACCECC NOTE Described the little

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove glare shield, liners, overhead units and panels as required. Remove/displace insulation blankets as required.

53-340-00-01 MRB 51-05-01-210-808 1.1 10 YR 10 YR ALL ALL 53-05-03-210-831

Inspect flight compartment floor structure.

ACCESS NOTE: Remove sidewalls and floor panels as required. Remove/displace insulation blankets as required.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-350-00-01	MRB	51-05-01-210-808 53-05-03-210-832	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL
		enger compartment fron gers), longitudinal lap sp				s), including skin pa	nels (skins,
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove galleys/la required.	vs. Remove ca	bin interior as requ	ired. Remove/dis	place insulation bla	nkets as
53-360-00-01	MRB	51-05-01-210-808 53-05-03-211-809	1.1	NOTE		ALL	ALL
		ard entry door cutout sur oth the forward and aft o		e (the door cutout to	o, and including, t	he door side of the	first frame from
			and L/N# 5660	and on, or those the		first), applicable to ated the doorsill cor	
	ACCESS NO	OTE: Remove galleys/la required.			ired. Remove/dis	place insulation bla	
53-360-00-02	MRB) TE: Remove galleys/la			ired. Remove/dis	place insulation bla	
53-360-00-02	MRB Inspect forward	DTE: Remove galleys/la required. 51-05-01-210-808	vs. Remove ca	bin interior as requ		ALL	nkets as
53-360-00-02	MRB Inspect forwa	TE: Remove galleys/larequired. 51-05-01-210-808 53-05-03-211-810 ard galley door cutout suoth the forward and aft composition of the succession of the forward and after the succession of the	1.1 Irround structur lirections). al 8YR / 24000 to Airplanes L/N doorsill corrosi Repeat interval and L/N# 5660	NOTE Te (the door cutout to the first and Repeat into the first 1-5645 and L/Nation protection/enhatics (18000 FC (Nation) and on, or those the first 1-1645 and to first 1-1645 and to first 1-1645 and first 1-	erval 6YR / 18000 \$\delta 5653-5659 and \$\delta 6000 comes \$\delta 6000 comes	ALL the door side of the	ALL e first frame from the serval 9YR Airplanes
53-360-00-02	MRB Inspect forwathe door in but INTERVAL N	portain and any of the forward and aft of the	1.1 Irround structur lirections). al 8YR / 24000 to Airplanes L/N doorsill corrosi Repeat interval and L/N# 5660 ced moisture b	NOTE The (the door cutout to the cutout to	erval 6YR / 18000 \$5653-5659 and aced moisture ba Whichever comes nat have incorpor	ALL the door side of the DFC (Whichever co those that have not rrier. Threshold inte first), applicable to ated the doorsill cor	ALL first frame fromes rival 9YR Airplanes rrosion

Inspect aft entry door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

INTERVAL NOTE: Threshold interval 8YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes

first), applicable to Airplanes L/N# 1-5645 and L/N# 5653-5659 and those that have not incorporated the doorsill corrosion protection/enhanced moisture barrier. Threshold interval 9YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 5646-5652 and L/N# 5660 and on, or those that have incorporated the doorsill corrosion

protection/enhanced moisture barrier.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as

required.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-360-00-04	MRB	51-05-01-210-808 53-05-03-211-812	1.1	NOTE		ALL	ALL
		alley door cutout surroun the forward and aft direc	,	e door cutout to, an	d including, the	door side of the first	frame from the
	INTERVAL N	incorporated the / 24000 FC and F	to Airplanes L/N doorsill corrosi Repeat interval and L/N# 5660	N# 1-5645 and L/Na ion protection/enha 6YR / 18000 FC (V and on, or those the	# 5653-5659 and nced moisture ba Whichever comes	0 FC (Whichever co those that have not arrier. Threshold inte s first), applicable to rated the doorsill cor	rval 9YR Airplanes
	ACCESS NO	OTE: Remove galleys/lar required.	vs. Remove ca	abin interior as requ	ired. Remove/dis	splace insulation bla	nkets as
53-370-00-01	MRB	51-05-01-210-808 53-05-03-210-833	1.1	12 YR	12 YR	ALL	ALL
	Sta 540 to 72 NOTE: Not a	enger compartment floor 27. applicable to airplanes wi DTE: Remove floor pane	ith flat aft press	sure bulkhead.		,	
53-380-00-01	Sta 540 to 72 NOTE: Not a	27. applicable to airplanes wi	ith flat aft press	sure bulkhead.		,	
53-380-00-01	NOTE: Not a ACCESS NO MRB Inspect pass and the floor aft directions	pplicable to airplanes wince the property of t	ith flat aft pressels and sidewald 1.1 r structure in word to, and include the structure in word to the s	sure bulkhead. Ils as required. Ren NOTE et area (within appling, the door side of	nove/displace ins	ALL hes from doors, gall	required. ALL eys and lavs,
53-380-00-01	MRB Inspect pass and the floor aft directions NOTE: Not a	pplicable to airplanes wince the properties of t	th flat aft pressels and sidewald 1.1 r structure in war to, and including the flat aft pressels 8YR / Repeathose that have by R, whichever	NOTE et area (within appriling, the door side cure bulkhead. t interval 6YR, which is not incorporated or comes first, applic	roximately 20 incomes first frame for the first frame for the HI-TAK Gel 1	ALL hes from doors, gall	ALL eys and lavs, the forward a
53-380-00-01	MRB Inspect pass and the floor aft directions NOTE: Not a	27. Ipplicable to airplanes wi DTE: Remove floor pane 51-05-01-210-808 53-05-03-210-834 enger compartment floor structure below the door). Ipplicable to airplanes wi L/N# 1-2412 and Repeat interval 6	1.1 r structure in w r to, and includ ith flat aft press al 8YR / Repea those that hav YR, whichever orated the HI-T, and lavs. Remov	sure bulkhead. Ils as required. Ren NOTE et area (within appring, the door side of the county of	roximately 20 incomplete first frame for the first frame for the HI-TAK Gel Table to Airplanes	ALL hes from doors, gall from the door in both st, applicable to Airpl Tape. Threshold intel L/N# 2413 and on,	ALL eys and lavs, the forward a

Inspect galley and lav attach fittings and any other easily visible portions of the floor structure in wet area (within approximately 20 inches from galleys and lavs, and the floor structure below the door to, and including, the door side of the first frame from the door in both the forward and aft directions).

ACCESS NOTE: Galleys and lavs removal is not required. Remove galley kick-plates and any other easily removable panels that may help inspect areas under galleys and lavs.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-400-00-01	MRB	51-05-01-210-808 53-05-03-210-836	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL

Inspect passenger compartment from STA 360 to 663.75, including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices; 2. Window belt structure; 3. Overwing emergency exit cutout structure; 4. Forward cargo door cutout surround structure (portion in upper lobe); 5. STA 540 and 663 bulkheads and splices; 6. Overwing frames and stub beams.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cabin interior as required. Remove/displace insulation blankets as

required.

53-410-00-01	MRB	51-05-01-210-808	1.1	12 YR	8 YR	ALL	ALL
		53-05-03-210-837	12	36000 FC	24000 FC		

Inspect passenger compartment from STA 663.75 to 1016 (except areas around door cutouts), including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices (note: inspection includes the circumferential skin and stringer splice at Sta 727l for the -900 models); 2. Window belt structure; 3. STA 663 bulkhead and splices; 4. STA 727 bulkhead; 5. Side strut support frame at STA 706; 6. Main landing gear support frames at STA 695 and 716; 7. Wheel well frame at STA 685; 8. Aft cargo door cutout surround structure (portion in upper lobe); 9. Forward side of STA 1016 bulkhead (chords, pressure web, stiffeners, chord/web attachments), including vertical fin front spar fittings; 10. Stringer splice fittings and tension bolts at STA 1016.

AIRPLANE NOTE: Task not applicable to -900ER and -800 with Flat Pressure Bulkhead installed.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/replace insulation blankets as

required.

53-420-00-01	MRB	51-05-01-210-806	1.1	12 YR	8 YR	ALL	ALL
		53-05-03-210-838	1.2	36000 FC	24000 FC		

Inspect area aft of STA 1016 pressure bulkhead to STA 1088, including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices; 2. Aft side of STA 1016 bulkhead (chords, pressure web, stiffeners, chord/web attachments); 3. Stringer splice fittings and tension bolts at STA 1016; 4. STA 1088 bulkhead, including vertical fin rear spar fittings and horizontal stabilizer center section jackscrew fitting lugs and bolts.

AIRPLANE NOTE: Task not applicable to -900ER and -800 with Flat Pressure Bulkhead installed.

INTERVAL NOTE: Whichever comes first.

53-430-00-01	MRB	51-05-01-210-804	1.1	12 YR	8 YR	ALL	ALL
		53-05-03-210-839	1.2	36000 FC	24000 FC		

Inspect stabilizer torsion box compartment and APU compartment, including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices; 2. STA 1088 bulkhead, including vertical fin rear spar fittings; 3. Forward side of STA 1156 bulkhead, including horizontal stabilizer hinge fittings and bolts; 4. Upper horizontal deck (at stringer 6) and lower horizontal deck (at stringer 12).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: For area below stringer 12, remove APU and firewalls; remove APU plenum as required. For area above stringer 12, adjust stabilizer trim as required. For access to Sta 1156 horizontal stabilizer hinge fitting lugs and bolts, remove gap seal and horizontal stabilizer rear spar sliding seal as required.





				INTERVAL	APPLICABILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-440-00-01	MRB	51-05-01-210-806 53-05-03-210-840	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect aft si	de of STA 1156 bulkhead	d.							
	INTERVAL N	IOTE: Whichever come	s first.							
53-450-00-01	MRB	51-05-01-210-806 53-05-03-210-841	1.1 1.2	9 YR 24000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect fusel	age skin under dorsal fir	and aft to Sta	1016, including cir	cumferential splic	e.				
	INTERVAL N	IOTE: Whichever come	s first.							
53-460-00-01	MRB	51-05-01-210-806 53-05-03-210-842	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect vertice	cal fin front spar fitting lu			2400010					
	•	IOTE: Whichever come	`	, ta 1010).						
		OTE: Pin removal is not								
	ACCESO NO	TE. Till Tellioval is not	required.							
53-470-00-01	MRB	51-05-01-210-804 53-05-03-210-843	1.1	20 YR	8 YR	ALL	ALL			
	Inspect vertical fin front spar fitting lugs and bolts (STA 1016).									
	ACCESS NO	OTE: Pin removal is requ	uired. Remove	only one pin at a ti	me.					
53-480-00-01	MRB	51-05-01-210-806 53-05-03-210-844	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect vertical fin rear spar fitting lugs and bolts at Sta 1088. Inspect fuselage skin under vertical fin from Sta 1016 to 1088									
	INTERVAL N	IOTE: Whichever come	s first.							
53-510-00-01	MRB	51-05-01-210-808 53-05-03-210-862	1.1 1.2	9 YR 24000 FC	9 YR 24000 FC	800 900ER	ALL			
	Inspect the s	kin and lugs under the a	ntenna base p	late Sta. 727D to 7	27H+5, S-4L to S	-4R.				
	AIRPLANE NOTE: Applicable to airplanes with a KU antenna radome installed at Sta. 727D to 727H+5, S-4L to S-4R (737-800 and 737-900ER Only).									
	INTERVAL N	IOTE: Whichever come		,						
	ACCESS NO	OTE: Antenna radome a	nd adapter pla	te removal required	i.					
53-600-00-01	AWL	53-05-02-250-801	1.1	50000 FC	36000 FC	600 700	ALL			
						700IGW 800 900 900ER				
	Inspect (High	n Frequency Eddy Curre	nt) the unner s	kin along the upper	fastener row at s	tringers S-4L and S	-4R from ST			

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360.

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.







53-600-00-02 Ir 2 5 5 5 5 5 5 5 5 5	AWL ASPECT (Low 159.5 to STA	26A001-DTR, DTR checknod(s) necessary to according to according to the inspection procedure 53-05-02-250-803 Frequency Eddy Curren 360. 26A001-DTR, DTR checknod(s) necessary to according to according the inspection procedure.	k form 53-10-0 complish the integes are contained 1.1 t) the lower sk k form 53-10-0 complish the integer	23-1, for alternative tent of this inspecticed in Part 6, Subject 50000 FC tin along the lower for alternative tent of this inspecticent of this inspectice.	inspections. on is contained in of 53-30-41. 36000 FC fastener row at structure inspections. on is contained in	600 700 700IGW 800 900 900ER ringers S-4L and S-4	ALL R from STA			
Ir 2 S T (I	nspect (High 159.5 to STA See Doc. D6. The NDI met D6-37239). The NDI AWL nspect (Low 159.5 to STA See Doc. D6. The NDI met D6-37239). The NDI met	Frequency Eddy Currer 360. 26A001-DTR, DTR check hod(s) necessary to accommodate and the inspection procedure 53-05-02-250-803 Frequency Eddy Curren 360. 26A001-DTR, DTR check hod(s) necessary to accommod the inspection procedure inspection procedure.	ht) the upper slow form 53-10-0 complish the integer are contained 1.1 t) the lower skills form 53-10-0 complish the integer are contained to the contained to	kin along the upper 03-1, for alternative tent of this inspecticed in Part 6, Subject 50000 FC	inspections. on is contained in ot 53-30-41. 36000 FC fastener row at structure inspections. on is contained in other inspections.	700IGW 800 900 900ER stringers S-4L and S- the 737 Nondestruct 600 700 700IGW 800 900 900ER ringers S-4L and S-4	4R from STA rive Test Manua ALL R from STA			
2 S T (I	AWL ASPECT (Low 159.5 to STA 239). AWL ASPECT (Low 159.5 to STA 259.5 to STA 259	360. 26A001-DTR, DTR checknod(s) necessary to according to the inspection procedure 53-05-02-250-803 Frequency Eddy Curren 360. 26A001-DTR, DTR checknod(s) necessary to according to the inspection procedure.	k form 53-10-0 complish the integes are contained 1.1 t) the lower sk k form 53-10-0 complish the integer	23-1, for alternative tent of this inspecticed in Part 6, Subject 50000 FC tin along the lower for alternative tent of this inspecticent of this inspectice.	inspections. on is contained in of 53-30-41. 36000 FC fastener row at structure inspections. on is contained in	600 700 700IGW 800 900 900ER ringers S-4L and S-4	ALL R from STA			
53-600-20-01 Ir 2 S T (I	AWL AWL ASPECT (Low 159.5 to STA 160.5 Doc. D6. The NDI met D6-37239).	hod(s) necessary to according to the inspection procedure 53-05-02-250-803 Frequency Eddy Curren 360. 26A001-DTR, DTR check hod(s) necessary to according the inspection procedure	omplish the intres are contained 1.1 t) the lower skink form 53-10-0 complish the intrescent interests.	tent of this inspecticed in Part 6, Subjection 50000 FC tin along the lower for alternative tent of this inspectice.	on is contained in at 53-30-41. 36000 FC astener row at strinspections. on is contained in	600 700 700IGW 800 900 900ER ringers S-4L and S-4	ALL R from STA			
(I	AWL ASPECT (Low 159.5 to STA 160.5 Doc. D6. The NDI met D6-37239).	The inspection procedure 53-05-02-250-803 Frequency Eddy Curren 360. 26A001-DTR, DTR check hod(s) necessary to according to the control of t	1.1 It) the lower sk k form 53-10-0 complish the into	50000 FC tin along the lower for alternative tent of this inspection	astener row at strinspections.	600 700 700IGW 800 900 900ER ringers S-4L and S-4	ALL R from STA			
Ir 2 S T (L	nspect (Low 159.5 to STA See Doc. D6 The NDI met D6-37239).	Frequency Eddy Curren 360. 26A001-DTR, DTR chec hod(s) necessary to according The inspection procedure	t) the lower sk k form 53-10-0 omplish the inte	in along the lower to 33-2, for alternative tent of this inspection	fastener row at strainspections.	700IGW 800 900 900ER ringers S-4L and S-4	R from STA			
2 S T (I	59.5 to STA See Doc. D6. The NDI met D6-37239).	360. 26A001-DTR, DTR chechod(s) necessary to accurate inspection procedure	k form 53-10-0 omplish the into	03-2, for alternative ent of this inspection	inspections.					
T (I	he NDI met D6-37239).	hod(s) necessary to according to the inspection procedure	omplish the into	ent of this inspection	on is contained in	the 737 Nondestruct	ive Test Manu			
])	D6-37239). [·]	The inspection procedure	•			the 737 Nondestruct	ive Test Manu			
			es are containe	ed in Part 6, Subjec	ct 53-30-50.					
53-600-20-02	AWL	F0 0F 00 0F0 000								
		53-05-02-250-803	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL			
	nspect (Low 259.5 to STA	Frequency Eddy Curren 360.	t) the lower ski	in along the lower f	astener row at st	ringers S-4L and S-4	R from STA			
Т	he NDI met	26A001-DTR, DTR chec hod(s) necessary to acco The inspection procedure	omplish the into	ent of this inspection	on is contained in	the 737 Nondestruct	ive Test Manu			
53-600-30-01	AWL	53-05-02-250-805	1.1	50000 FC	36000 FC	ALL	ALL			
_										
S		requency Eddy Currer nd from STA 350 to STA								
	See Doc D626A001-DTR, DTR check form 53-10-03-3 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual									
		hod(s) necessary to according to according to the inspection procedure	•			the 737 Nondestruct	ive Test Manu			
53-600-30-02	AWL	53-05-02-250-805	1.1	50000 FC	36000 FC	ALL	ALL			
S	Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringer S-14L (from STA 259.5 to STA 294.5, and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277, and from STA 344 to STA 360).									
Т	he NDI met	26A001-DTR, DTR check hod(s) necessary to accor The inspection procedure	omplish the inte	ent of this inspection	on is contained in	the 737 Nondestruct	ive Test Manu			
53-600-40-01	AWL	53-05-02-250-806	1.1	50000 FC	36000 FC	ALL	ALL			
		Frequency Eddy Curren								

from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277 and from STA 344 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.





			APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-600-40-02	AWL	53-05-02-250-806	1.1	50000 FC	36000 FC	ALL	ALL
	294.5 and fro See Doc D62	Frequency Eddy Currer om STA 350 to STA 360) 26A001-DTR, DTR check hod(s) necessary to acc	and at stringe k form 53-10-0	r S-14R (from STA 3-4 for alternative i	259.5 to STA 277 inspections.	and from STA 344	to STA 360).
	(D6-37239).	The inspection procedur	es are contain	ed in Part 6, Subje	ct 53-30-50.		
53-600-50-01	AWL	53-05-02-250-807	1.1	50000 FC	36000 FC	ALL	ALL
	STA 334) and See Doc D62 The NDI met	Frequency Eddy Currer d at stringer S-24R (from 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur	STA 259.5 to k form 53-10-0 omplish the int	STA 360). 3-5 for alternative i ent of this inspection	inspections. on is contained in		
53-600-50-02	AWL	53-05-02-250-807	1.1	50000 FC	36000 FC	ALL	ALL
33-000-30-02		Frequency Eddy Currer					
	STA 334) and See Doc D62 The NDI met	d at stringer S-24R (from 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur	STA 259.5 to k form 53-10-0 omplish the int	STA 360). 3-5 for alternative i ent of this inspection	inspections. on is contained in		
53-601-01-01	AWL	53-05-02-250-809	1.1	50000 FC	12000 FC	ALL	ALL
	See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currer 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur NOTE: Applicable to air	k form 53-10-0 omplish the int es are contain	4-1a for alternative ent of this inspection ed in Part 6, Subject	inspections. on is contained in		
53-601-21-01	AWL	53-05-02-211-802	1.1	50000 FC	4000 FC	ALL	ALL
	Inspect (Deta	ailed) the CD post, from 026A001-DTR, DTR check	outside the air	-			
	AIRPLANE I	NOTE: Applicable to air	olanes L/N 138	9 and on.			
53-601-21-02	AWL	53-05-02-211-802	1.1	50000 FC	4000 FC	ALL	ALL
	See Doc D62	ailed) the CD post, from (26A001-DTR, DTR check	k form 53-10-0	4-2a for alternative		both the left and rig	ht sides.
	AIRPLANE I	NOTE: Applicable to air	olanes L/N 138	9 and on.			
53-601-30-01	AWL	53-05-02-211-803	1.1	50000 FC	18000 FC	ALL	ALL
		ailed) the EF post, from o 26A001-DTR, DTR check				ooth the left and rig	ht sides.
53-601-30-02	AWL	53-05-02-211-803	1.1	50000 FC	18000 FC	ALL	ALL
		ailed) the EF post, from o					

Inspect (Detailed) the EF post, from outside the aircraft, along the entire post length on both the left and right sides. See Doc D626A001-DTR, DTR check form 53-10-04-3 for alternative inspections.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-601-41-01	AWL	53-05-02-250-811	1.1	50000 FC	4000 FC	ALL	ALL			
	Inspect (Low Frequency Eddy Current) both rows of fasteners attaching the skin to the BD Sill, from outside the aircraft, between LBL 8 and LBL 26.5. Repeat the process between RBL 8 and RBL 26.5. See Doc D626A001-DTR, DTR check form 53-10-04-4a for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manus (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-76. AIRPLANE NOTE: Applicable to airplanes L/N 1389 and on.									
53-601-41-02	AWL	53-05-02-250-811	1.1	50000 FC	4000 FC	ALL	ALL			
	The NDI met (D6-37239).	26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur NOTE: Applicable to air	omplish the intees are contained	ent of this inspection and in Part 6, Subject	on is contained in	the 737 Nondestruc	ctive Test Manu			
53-601-50-01	AWL	53-05-02-250-812	1.1	50000 FC	36000 FC	ALL	ALL			
	common to the See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Curre he BD sill, on both the le 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Remove glareshie	ft and right side k form 53-10-0 omplish the int es are containe	es. 4-5 for alternative i ent of this inspection ed in Part 6, Subject	inspections. on is contained in ct 53-10-72.					
53-601-50-02	AWL	53-05-02-250-812	1.1	50000 FC	36000 FC	ALL	ALL			
	common to the See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Curre he BD sill, on both the le 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Remove glareshie	ift and right side k form 53-10-0 omplish the int es are containe	es. 4-5 for alternative i ent of this inspection ed in Part 6, Subject	inspections. on is contained in ct 53-10-72.					

53-601-60-01 AWL 53-05-02-250-813 1.1 50000 FC 36000 FC ALL ALL

Inspect (High Frequency Eddy Current) the Inconel angle around the seven fasteners that join the angle to the B-D Sill Web and Point "D" Fitting.

See Doc D626A001-DTR, DTR check form 53-10-04-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-71.

ACCESS NOTE: Remove glareshield if/as required to perform inspection. There are three (3) fasteners that join the angle, web and fitting. There are four (4) fasteners that join the angle and the fitting.





				INTERVAL		APPLICA	ABILITY					
TASK CARD NO.												
53-601-60-02	AWL	53-05-02-250-813	1.1	50000 FC	36000 FC	ALL	ALL					
	and Point "D See Doc D62 The NDI met	Inspect (High Frequency Eddy Current) the Inconel angle around the seven fasteners that join the angle to the B-D Sill Web and Point "D" Fitting. See Doc D626A001-DTR, DTR check form 53-10-04-6 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-71.										
	ACCESS NO	OTE: Remove glareshie angle, web and fitt				` '	at join the					
53-602-10-01	AWL	53-05-02-250-814	1.1	50000 FC	24000 FC	ALL	ALL					
	between the See Doc D62 The NDI met in the 737 No	r Frequency Eddy Currer cab window cutout and 26A001-DTR, DTR checthod(s) necessary to accondestructive Test Manu-Part 6, Subject 53-30-57	STA 259.5 pan k form 53-10-0 complish the int al (D6-37239).	el splice. 5-1 for alternative ent of this inspecti	inspections.	io for and fight side	, of BE 0.0					
53-602-10-02	AWL	53-05-02-250-814	1.1	50000 FC	24000 FC	ALL	ALL					
	between the See Doc D62 The NDI met in the 737 No	r Frequency Eddy Currer cab window cutout and 26A001-DTR, DTR chec thod(s) necessary to acc ondestructive Test Manu Part 6, Subject 53-30-57	STA 259.5 pan k form 53-10-0 complish the int al (D6-37239).	el splice. 5-1 for alternative ent of this inspecti	inspections.	ie ieπ and right side	, of BL 0.0					
53-602-20-01	AWL	53-05-02-211-862	1.1	50000 FC	4000 FC	ALL	ALL					
	except at the	ailed) the skin around all a lap splices and antenna 26A001-DTR, DTR chec	as. (53-10-08-1).	· ·	0R, from STA 259.	5 to STA 360,					
53-605-10-01	AWL	53-05-02-211-804	1.1	50000 FC	4000 FC	ALL	ALL					
	Inspection is RBL 15.47 a	ailed) the skin at the edg performed along the en nd LBL 6.74. 26A001-DTR, DTR chec	tire perimeter o	of the E/E door cuto	out and bounded b	•	•					
53-606-40-01	AWL	53-05-02-250-816	1.1	50000 FC	24000 FC	ALL	ALL					
		n Frequency Eddy Curre	nt) the two row	rs of fasteners com	mon to the forwar	d edge frame and s	skin at STA 303					

from stringers S-11L and S-12L.
See Doc. D626A001-DTR, DTR check form 53-10-14-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-89.







				INTERVAL		APPLICA	APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
53-606-50-01	AWL	53-05-02-250-817	1.1	50000 FC	36000 FC	ALL	ALL		
	STA 303.9 fr See Doc. Do The NDI me	h Frequency Eddy Currer rom stringers S-7L to S-1 626A001-DTR, DTR chec thod(s) necessary to accurate inspection procedure	1L and stringer k form 53-10- omplish the int	rs S-12L to S-13L. 14-5 for alternative ent of this inspection	inspections. on is contained in				
53-606-70-01	AWL	53-05-02-211-805	1.1	50000 FC	9000 FC	ALL	ALL		
	See Doc. Do	ailed) around the fastene 626A001-DTR, DTR chec OTE: Remove interior pa	k form 53-10-	14-7 for alternative	inspections.	ingers S-7 through S	S-16.		
53-618-00-01	AWL	53-05-02-250-818	1.1	50000 FC	30000 FC	ALL	ALL		
	The NDI me (D6-37239).	626A001-DTR, DTR check thod(s) necessary to accumulate the inspection procedure. OTE: Remove interior pages.	omplish the int es are contain	ent of this inspection ed in Part 6, Subject	on is contained in ct 53-11-01.	the 737 Nondestruc	ctive Test Manu		
53-619-00-01	AWL	53-05-02-250-819	1.1	50000 FC	18000 FC	ALL	ALL		
	S-15 and S- See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Currer 16. 526A001-DTR, DTR chec thod(s) necessary to accor The inspection procedur OTE: Remove interior pa	k form 53-10- omplish the int es are contain	14-9 for alternative ent of this inspection ed in Part 6, Subject	inspections. on is contained in ct 53-11-01.				
53-620-00-01	AWL	53-05-02-211-807	1.1	50000 FC	24000 FC	600 700 800 900 900ER	ALL		
	See Doc. Do	ailed) the S-15 and S-16 326A001-DTR, DTR chec OTE: Open FWD Entry D	k form 53-10-	14-10 for alternative	e inspections.				
53-621-00-01	AWL	53-05-02-211-808	1.1	50000 FC	18000 FC	ALL	ALL		
	Inspect (Detailed) the Aft frame stops at the inner flange holes near STA 348.2 from stringer S-7 to S-14. See Doc. D626A001-DTR, DTR check form 53-10-14-11 for alternative inspections. ACCESS NOTE: Open FWD Entry Door. Removal of interior panel is required to perform the inspection.								
53-622-00-01	AWL	53-05-02-250-820	1.1	50000 FC	36000 FC	ALL	ALL		
	Inspect (Hig	h Frequency Eddy Curre	nt) the skin arc	und the fastener h	oles and along th	e edge of the cutout	hidden by the		

Inspect (High Frequency Eddy Current) the skin around the fastener holes and along the edge of the cutout hidden by the scuff plates from STA 303 to STA 350.

See Doc. D626A001-DTR, DTR check form 53-10-14-12 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-43.

ACCESS NOTE: Removal of scuff plate is required to perform the inspection.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-623-00-01	AWL	53-05-02-250-931	1.1	50000 FC	36000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the skin around the fastener holes and along the edge of the cutout hidden by the scuff plates from STA 303 to STA 350. (53-10-15). See Doc. D626A001-DTR, DTR check form 53-10-14-12 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-43. ACCESS NOTE: Removal of scuff plate is required to perform this inspection.									
53-624-00-01	AWL	53-05-02-250-821	1.1	50000 FC	18000 FC	ALL	ALL			
	Inspect (Hig the forward See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curreledge frame at stops #1, #626A001-DTR, DTR checthod(s) necessary to accombe inspection procedur	nt) the four (4) #2, #5, #6. ck form 53-10- omplish the int es are contain	fastener at each s 15-2 for alternative ent of this inspectied in Part 6, Subje	inspections. on is contained in ct 53-10-98.	the 737 Nondestruc	I tension strap at			
		bottom up.								
53-625-00-01	AWL	53-05-02-250-822	1.1	50000 FC	9000 FC	ALL	ALL			
	See Doc. Do The NDI me	e forward and aft edge fra 626A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-10-7 omplish the int	15-4 for alternative ent of this inspection	inspections. on is contained in	the 737 Nondestruc	itive Test Manual			
53-626-00-01	AWL	53-05-02-250-823	1.1	50000 FC	36000 FC	ALL	ALL			
	the Nose Wi See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre heel Well AFT bulkhead. 526A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Access through E/	ck form 53-10- omplish the int es are contain	18-1 for alternative ent of this inspecti ed in Part 6, Subje	inspections.					
53-626-00-02	AWL	53-05-02-250-823	1.1	50000 FC	36000 FC	ALL	ALL			
	the Nose Wi See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre heel Well AFT bulkhead. 626A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur OTE: Access through E/	ck form 53-10- omplish the int es are contain	18-1 for alternative ent of this inspecti ed in Part 6, Subje	inspections.					
53-627-00-01	AWL	53-05-02-211-809	1.1	50000 FC	24000 FC	600 700 800 900 900ER	ALL			

Inspect (Detailed) the AFT access cutout forward vertical beam at STA 260, from WL 170 to WL 184.

See Doc. D626A001-DTR, DTR check form 53-10-19-4 for alternative inspections.

ACCESS NOTE: Access through Left Aft Nose Wheel Well Panel.





				INTERVAL		APPLICA	BILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-627-00-02	AWL	53-05-02-211-809	1.1	50000 FC	24000 FC	600 700 800 900 900ER	ALL			
	Inspect (Detailed) the AFT access cutout forward vertical beam at STA 260, from WL 170 to WL 184.									
	See Doc. D6	26A001-DTR, DTR chec	k form 53-10-	19-4 for alternative	inspections.					
	ACCESS NO	OTE: Access through Ri	ght Aft Nose W	/heel Well Panel.						
53-628-00-01	AWL	53-05-02-211-810	1.1	50000 FC	17000 FC	ALL	ALL			
	189.3.	ailed) the inboard and ou 26A001-DTR, DTR chec	J	9	•	the bushings at STA	262, BL 16,			
	ACCESS NO	OTE: For Direction 1, ren	moval of Drag	Brace is required.	·					
53-628-00-02	AWL	53-05-02-211-810	1.1	50000 FC	17000 FC	ALL	ALL			
	189.3.	ailed) the inboard and ou	· ·	· ·	•	the bushings at STA	262, BL 16,			
		26A001-DTR, DTR ched			inspections.					
	ACCESS NO	OTE: For Direction 1, re	moval of Drag	Brace is required.						
53-629-00-01	AWL	53-05-02-211-811	1.1	50000 FC	24000 FC	ALL	ALL			
	Inspect (Detailed) the inboard and outboard fitting segments of the Trunnion Support Fitting around the pin socket at BS 294.5, WL 156.1, and BL 16. See Doc. D626A001-DTR, DTR check form 53-10-20-3 for alternative inspections.									
53-629-00-02	AWL	53-05-02-211-811	1.1	50000 FC	24000 FC	ALL	ALL			
	294.5, WL 1	ailed) the inboard and ou 56.1, and BL 16. 26A001-DTR, DTR chec				tting around the pin s	socket at BS			
53-630-00-01	AWL	53-05-02-211-812	1.1	50000 FC	9000 FC	ALL	ALL			
		ailed) the upper and lower 26A001-DTR, DTR chec								
	AIRPLANE	NOTE: Applicable to air	planes with airs	stairs installed.	•					
	ACCESS NO	OTE: Opening of Air Sta	ir Door is requi	red to perform this	inspection.					
53-631-00-01	AWL	53-05-02-250-824	1.1	50000 FC	36000 FC	ALL	ALL			
	Inchest (Lev	Frequency Eddy Currer	at) the ferward	and aft adda frame	wohe around the	factonore common	to the corne			

Inspect (Low Frequency Eddy Current) the forward and aft edge frame webs around the fasteners common to the corner clips at STA 303.9 and STA 351.2.

See Doc. D626A001-DTR, DTR check form 53-10-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-27.

AIRPLANE NOTE: Applicable to airplanes with airstairs installed.

ACCESS NOTE: Opening of Air Stair Door is required to perform this inspection.





				APPLICA	BILITY											
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE									
53-633-00-01	AWL	53-05-02-211-813	1.1	50000 FC	4000 FC	ALL	ALL									
	except at the	ailed) the skin around all lap splices and antenna 26A001-DTR, DTR chec	as.			-10R, from STA 360	to STA 540,									
53-634-00-01	AWL	53-05-02-211-814	1.1	50000 FC	36000 FC	ALL	ALL									
	the GPS Ant	ailed) the exterior surfact enna at STA 500A. 26A001-DTR, DTR chec				the ATC Antenna at	STA 430, and									
	AIRPLANE I	NOTE: For the 737-600	, GPS Antenna	s are located at ST	A 482A.											
	ACCESS NO	OTE: Removal of extern	al antenna fairi	ngs and base plate	es are required											
53-635-00-01	AWL	53-05-02-211-815	1.1	50000 FC	36000 FC	600 700 800	ALL									
00 000 00 01	the left and r	ailed) the skin near the fa ight sides of the aircraft 26A001-DTR, DTR chec	at STA 500 bet	ween stringers S-6	and S-7.	ers, and antenna nu	tplates on bot									
	AIRPLANE I	NOTE : For the 737-600	GPS Antennas	are located at Sta	482A + 5, LBL 5	and RBL 5.										
	ACCESS NO	OTE: Removal of antenr	na is required.													
53-635-00-02	AWL	53-05-02-211-815	1.1	50000 FC	36000 FC	600 700 800	ALL									
	the left and r	ailed) the skin near the fa ight sides of the aircraft 26A001-DTR, DTR chec	at STA 500 bet	ween stringers S-6	and S-7.	ers, and antenna nu	tplates on bot									
	AIRPLANE I	NOTE: For the 737-600	GPS Antennas	are located at Sta	482A + 5, LBL 5	and RBL 5.	See Doc D626A001-DTR, DTR check form 53-30-01-5 for alternative inspections. AIRPLANE NOTE: For the 737-600 GPS Antennas are located at Sta 482A + 5, LBL 5 and RBL 5.									
	ACCESS NO															
		OTE: Removal of antenr	na is required.													
53-636-00-01	AWL	OTE: Removal of antenr 53-05-02-210-801	na is required. 1.1	50000 FC	4000 FC	ALL	ALL									
53-636-00-01	AWL Inspect (Gen		1.1 n STA 360 to S	TA 540 between str	ringers S-14 to S		ALL									
53-636-00-01	AWL Inspect (Gen See Doc D62	53-05-02-210-801 neral Visual) the skin from	1.1 n STA 360 to S k form 53-30-0	TA 540 between str 2-1 for alternative in	ringers S-14 to S	-17.	ALL									
53-636-00-01	AWL Inspect (Gen See Doc D62	53-05-02-210-801 neral Visual) the skin from 26A001-DTR, DTR chec	1.1 n STA 360 to S k form 53-30-0	TA 540 between str 2-1 for alternative in	ringers S-14 to S	-17.	ALL									
	AWL Inspect (Gen See Doc D62 ACCESS NC AWL Inspect (Gen	53-05-02-210-801 neral Visual) the skin from 26A001-DTR, DTR chec DTE: Remove or displace	1.1 n STA 360 to S k form 53-30-0 ce wing to body 1.1 n STA 360 to S	TA 540 between str 2-1 for alternative in fairings as required 50000 FC TA 540 between str	ringers S-14 to S nspections. d to perform this 4000 FC ringers S-14 to S	-17. inspection. ALL										
	AWL Inspect (Gen See Doc D62 ACCESS NC AWL Inspect (Gen See Doc D62	53-05-02-210-801 peral Visual) the skin from 26A001-DTR, DTR checontering the skin from 253-05-02-210-801 peral Visual) the skin from	1.1 m STA 360 to S k form 53-30-00 ee wing to body 1.1 m STA 360 to S k form 53-30-00	TA 540 between str 2-1 for alternative in fairings as required 50000 FC TA 540 between str 2-1 for alternative in	ringers S-14 to S nspections. d to perform this 4000 FC ringers S-14 to S nspections.	-17. inspection. ALL -17.										
	AWL Inspect (Gen See Doc D62 ACCESS NC AWL Inspect (Gen See Doc D62	53-05-02-210-801 neral Visual) the skin from 26A001-DTR, DTR chec DTE: Remove or displace 53-05-02-210-801 neral Visual) the skin from 26A001-DTR, DTR chec	1.1 m STA 360 to S k form 53-30-00 ee wing to body 1.1 m STA 360 to S k form 53-30-00	TA 540 between str 2-1 for alternative in fairings as required 50000 FC TA 540 between str 2-1 for alternative in	ringers S-14 to S nspections. d to perform this 4000 FC ringers S-14 to S nspections.	-17. inspection. ALL -17.										

See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.







				INTERVAL		APPLICA	BILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
53-636-10-02	AWL	53-05-02-211-816	1.1	50000 FC	8000 FC	ALL	ALL				
		Inspect (Detailed) the fuselage skin panels under the Wing to Body Fairing from STA 360 to STA 540. See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.									
	ACCESS NO	OTE: Remove or displac	e wing to body	fairings as require	ed to perform this	inspection.					
53-637-00-01	AWL	53-05-02-250-932	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
	360 to STA 5 See Doc. D6	h Frequency Eddy Currer 540. (PSE 53-30-04-1). 526A001-DTR, DTR chec thod(s) necessary to acc	ck form 53-10-0	03-1, for alternative	e inspections.						
		The inspection procedur									
53-637-00-02	AWL	53-05-02-250-932	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
53-638-00-01		thod(s) necessary to acc The inspection procedur 53-05-02-250-826				600 700 700IGW 800	tive Test Man				
	Inspect (Low to STA 540.	900 900ER Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 36									
		626A001-DTR, DTR chec	ck form 53-30-0	04-2. for alternative	e inspections.						
	The NDI me	thod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspecti	on is contained in	the 737 Nondestruc	tive Test Man				
53-638-00-02	AWL	53-05-02-250-826	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (Low to STA 540.	Frequency Eddy Currer	nt) the lower sk	in along the lower	fastener row at st	ringers S-4L and S-4	4R from STA 3				
	The NDI met	\$26A001-DTR, DTR check thod(s) necessary to accepte the inspection procedure.	omplish the int	ent of this inspecti	on is contained in	the 737 Nondestruc	tive Test Man				
53-639-00-01	AWL	53-05-02-250-828	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (High 360 to STA 5	h Frequency Eddy Curre	nt) the upper s	kin along the uppe	r fastener row at s	stringers S-10L and	S-10R from				

See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.







				INTERVAL		APPLICA	BILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-639-00-02	AWL	53-05-02-250-828	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL
	Inspect (High 360 to STA 5	r Frequency Eddy Currer	nt) the upper s	kin along the uppe	r fastener row at	stringers S-10L and	S-10R from STA
	See Doc. D6	26A001-DTR, DTR chec	ck form 53-30-0	04-3, for alternative	inspections.		
		hod(s) necessary to according to the inspection procedure in the inspection procedure	•			the 737 Nondestruc	tive Test Manua
53-640-00-01	AWL	53-05-02-211-817	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL
		ailed) the lower skin alon 26A001-DTR, DTR chec	•	ŭ		10R from STA 360 to	STA 540.
	ACCESS NO	OTE: Removal or displac	cement of inter	ior sidewall panels	and insulation bla	ankets are required.	
53-640-00-02	AWL	53-05-02-211-817	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL
	. ,	ailed) the lower skin alon 26A001-DTR, DTR chec	-	-		10R from STA 360 to	STA 540.
	ACCESS NO	TE: Removal or displac	cement of inter	ior sidewall panels	and insulation bla	ankets are required.	
53-641-00-01	AWL	53-05-02-211-819	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL
		ailed) the upper skin alon 26A001-DTR, DTR chec	•		-	-14R from STA 360 t	o STA 540.
53-641-00-02	AWL	53-05-02-211-819	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL
		ailed) the upper skin alon 26A001-DTR, DTR chec	•		-	-14R from STA 360 t	o STA 540.
53-642-00-01	AWL	53-05-02-250-830	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
53-642-00-02	AWL	53-05-02-250-830	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
		Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 360 to STA 540.									
	See Doc. D6	326A001-DTR, DTR ched	ck form 53-30-0	04-6, for alternative	inspections.						
		thod(s) necessary to acc The inspection procedur	•			the 737 Nondestruc	ctive Test Manu				
53-643-00-01	AWL	53-05-02-250-832	1.1	50000 FC	36000 FC	ALL	ALL				
	from STA 36 See Doc. D6 The NDI met (D6-37239).	r Frequency Eddy Currer 0 to STA 540, except at to 326A001-DTR, DTR check thod(s) necessary to according to the The inspection procedure	the cargo door ck form 53-30-0 complish the interes are contain	cutout. 04-7, for alternative tent of this inspection ed in Part 6, Subject	inspections. on is contained in ct 53-30-50.						
	ACCESS NO	OTE: Remove Wing to E	Body Fairing as	required to perforr	n this inspection.						
53-643-00-02	AWL	53-05-02-250-832	1.1	50000 FC	36000 FC	ALL	ALL				
	See Doc. D6 The NDI mei (D6-37239).	0 to STA 540, except at to 326A001-DTR, DTR check thod(s) necessary to according The inspection procedure DTE: Remove Wing to E	ck form 53-30-0 complish the interes are contain	04-7, for alternative tent of this inspection ed in Part 6, Subject	on is contained in ct 53-30-50.		ctive Test Manu				
				- 1,							
53-644-00-01	AWL	53-05-02-211-821	1.1	50000 FC	4000 FC	ALL	ALL				
		ailed) the lower (outer) s at the cargo door cutout.		ower fastener row a	t stringers S-24L	and S-24R from ST	A 360 to STA				
		326A001-DTR, DTR ched		04-8, for alternative	inspections.						
	ACCESS NO	OTE: Remove Wing to E	Body Fairing as	required to perform	n this inspection.						
53-644-00-02	AWL	53-05-02-211-821	1.1	50000 FC	4000 FC	ALL	ALL				
	540, except	ailed) the lower (outer) s at the cargo door cutout. 226A001-DTR, DTR chec			-	and S-24R from ST	A 360 to STA				
	ACCESS NO	OTE: Remove Wing to E	Body Fairing as	required to perforn	n this inspection.						
53-645-00-01	AWL	53-05-02-211-863	1.1	50000 FC	24000 FC	ALL	ALL				
		ailed) the window frames 26A001-DTR, DTR chec			,	PSE 53-30-05).					

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets is required.





				INTERVAL		APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-645-00-02	AWL	53-05-02-211-863	1.1	50000 FC	24000 FC	ALL	ALL	
		ailed) the window frames 26A001-DTR, DTR check			,	SE 53-30-05).		
	ACCESS NO	OTE: Removal and/or dis insulation blankets	•	passenger cabin s	idewalls or sidewa	ıll window assembli	es and	
53-645-01-01	AWL	53-05-02-211-864	1.1	50000 FC	4000 FC	ALL	ALL	
		ailed) the window frames 26A001-DTR, DTR check			,	SE 53-30-05).		
53-645-01-02	AWL	53-05-02-211-864	1.1	50000 FC	4000 FC	ALL	ALL	
		ailed) the window frames 26A001-DTR, DTR checl			,	SE 53-30-05).		
53-646-00-01	AWL	53-05-02-250-934	1.1	50000 FC	9000 FC	ALL	ALL	
	The NDI me	26A001-DTR, DTR check thod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Man	
53-646-10-01	AWL	53-05-02-250-833	1.1	50000 FC	18000 FC	ALL	ALL	
		n Frequency Eddy Curre pt at the door stops and		d portion of the fra	me web for damaç	ge between stringer	s S-17R and	
		26A001-DTR, DTR check			•			
		thod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	ctive Test Mar	
	ACCESS NO	OTE: Perform inspection inspection.	with door ope	n. Remove or disp	lace cargo liners a	s required to perfor	m this	
53-646-20-01	AWL	53-05-02-130-801	1.1	50000 FC	18000 FC	ALL	ALL	
	See Doc D6. The NDI me	asonic) the outboard port 26A001-DTR, DTR chect thod(s) necessary to acc The inspection procedur	k form 53-30-0 omplish the int	8-2 for alternative ent of this inspection	inspections. on is contained in	-		
	ACCESS NO	OTE: Perform inspection inspection.	with door ope	n. Remove or disp	lace cargo liners a	s required to perfor	m this	
53-646-30-01	AWL	53-05-02-250-834	1.1	50000 FC	18000 FC	ALL	ALL	
		n Frequency Eddy Curre 26A001-DTR, DTR check	,	•		ween stringers S-18	8R and S-26F	

ACCESS NOTE: Perform inspection with door open. Remove sealer at door stops as required.

(D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-90.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual





ASK CARD NO.			INTERVAL		APPLICABILITY		
ASK CAKE NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-646-40-01	AWL	53-05-02-250-835	1.1	50000 FC	18000 FC	ALL	ALL
	Inspect (High	Frequency Eddy Curre	nt) the forward	and aft edge frame	e inner chords bet	ween stringers S-1	7R and S-18R.
		26A001-DTR, DTR chec			•		
		hod(s) necessary to acc The inspection procedur	•			the 737 Nondestruc	ctive Test Manı
	ACCESS NO	OTE: Remove cargo line	ers as required	to perform the insp	ection.		
53-646-50-01	AWL	53-05-02-250-836	1.1	50000 FC	18000 FC	ALL	ALL
		Frequency Eddy Currer	,	•		tringer S-18R.	
		26A001-DTR, DTR chec			•		
		hod(s) necessary to acc The inspection procedur	•			the 737 Nondestruc	ctive Test Man
	,	OTE: Remove cargo line		•			
53-646-60-01	AWL	53-05-02-250-837	1.1	50000 FC	9000 FC	ALL	ALL
	Inspect (High	n Frequency Eddy Curre	nt) the bearstra	an for two inches or	n each side of stri	nger S-24R at STA	440 and STA
	492.4.	oquoo, _ua, ouo	, 200			go. o a. o	
	See Doc D62	26A001-DTR, DTR chec	k form 53-30-0	8-6 for alternative i	nspections.		
	The NDI met	hod(s) necessary to acc	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Man
	(D6-37239).	The inspection procedur	es are containe	ed in Part 6, Subjec	ct 53-10-91.		
E2 646 64 04							
53-646-61-01	AWL	53-05-02-130-802	1.1	50000 FC	9000 FC	ALL	ALL
53-646-61-01		53-05-02-130-802 asonic) the bearstrap for					
53-646-61-01	Inspect (Ultra		hidden damag	e under the stop ba	ackup fitting at str		
53-646-61-01	Inspect (Ultra 492.4. See Doc D62 The NDI met	asonic) the bearstrap for 26A001-DTR, DTR chec hod(s) necessary to acc	hidden damag k form 53-30-0 complish the int	e under the stop be 8-6 for alternative i ent of this inspection	ackup fitting at str nspections. on is contained in	inger S-24R at STA	440 and STA
53-646-61-01	Inspect (Ultra 492.4. See Doc D62 The NDI met	asonic) the bearstrap for 26A001-DTR, DTR chec	hidden damag k form 53-30-0 complish the int	e under the stop be 8-6 for alternative i ent of this inspection	ackup fitting at str nspections. on is contained in	inger S-24R at STA	440 and STA
	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239).	asonic) the bearstrap for 26A001-DTR, DTR chec hod(s) necessary to acc The inspection procedur	hidden damag k form 53-30-0 complish the int res are contain	e under the stop be 8-6 for alternative i ent of this inspectic ed in Part 4, Subject	ackup fitting at str nspections. on is contained in ct 53-10-07.	inger S-24R at STA	440 and STA
53-646-61-01	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239).	asonic) the bearstrap for 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur 53-05-02-211-865	k form 53-30-0 complish the intres are contained	8-6 for alternative i ent of this inspection ed in Part 4, Subject	nspections. on is contained in ct 53-10-07.	inger S-24R at STA the 737 Nondestruc ALL	440 and STA ctive Test Manu ALL
	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239).	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to acc The inspection procedur 53-05-02-211-865 ailed) the skin around the	k form 53-30-0 complish the intres are contained	8-6 for alternative i ent of this inspection ed in Part 4, Subject	nspections. on is contained in ct 53-10-07.	inger S-24R at STA the 737 Nondestruc ALL	440 and STA ctive Test Man
	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239).	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to acc The inspection procedur 53-05-02-211-865 ailed) the skin around the	k form 53-30-0 complish the intres are contained 1.1 e entire edge o	8-6 for alternative i ent of this inspectic ed in Part 4, Subject 34000 FC	nspections. on is contained in ct 53-10-07. 18000 FC	inger S-24R at STA the 737 Nondestruc ALL	440 and STA ctive Test Man
	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to according the inspection procedur 53-05-02-211-865 called) the skin around the 3-30-08-9).	k form 53-30-0 complish the interes are contained 1.1 e entire edge o	8-6 for alternative i ent of this inspectic ed in Part 4, Subject 34000 FC f the scuff plates at	nspections. on is contained in ct 53-10-07. 18000 FC all four corners (the 737 Nondestructure ALL upper/lower/fwd/aft/	A440 and STA ctive Test Man ALL) of the cargo
	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to accommodate inspection procedur 53-05-02-211-865 ailed) the skin around the 3-30-08-9).	k form 53-30-0 complish the interes are contained 1.1 e entire edge o	8-6 for alternative i ent of this inspectic ed in Part 4, Subject 34000 FC f the scuff plates at	nspections. on is contained in ct 53-10-07. 18000 FC all four corners (the 737 Nondestructure ALL upper/lower/fwd/aft/	A440 and STA ctive Test Man ALL) of the cargo
	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to accommodate inspection procedur 53-05-02-211-865 ailed) the skin around the 3-30-08-9).	k form 53-30-0 complish the interes are contained 1.1 e entire edge o	8-6 for alternative i ent of this inspectic ed in Part 4, Subject 34000 FC f the scuff plates at	nspections. on is contained in ct 53-10-07. 18000 FC all four corners (the 737 Nondestructure ALL upper/lower/fwd/aft/	ALL of the cargo
53-646-62-01	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6 ACCESS NO	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to accommodate the inspection procedur 53-05-02-211-865 called) the skin around the 3-30-08-9). 26A001-DTR, DTR check DTE: Forward cargo doc	k form 53-30-0 complish the intres are contained 1.1 e entire edge of the contained ck form 53-60-0 or must be open 1.1	8-6 for alternative i ent of this inspectic ed in Part 4, Subject 34000 FC f the scuff plates at 08-9, for alternative in to perform this ins	nspections. on is contained in ct 53-10-07. 18000 FC all four corners (inspections. spection. Scuff pla	the 737 Nondestructure ALL upper/lower/fwd/aft, ate removal required ALL	ALL of the cargo
53-646-62-01	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6 ACCESS NO AWL Inspect (Gen	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to accommodate inspection procedur 53-05-02-211-865 ailed) the skin around the 3-30-08-9). 26A001-DTR, DTR check DTE: Forward cargo documents of the 53-05-02-210-808	k form 53-30-0 complish the intres are contained 1.1 e entire edge of the contained of the contained 1.1 e entire edge of	8-6 for alternative i ent of this inspecticed in Part 4, Subject 34000 FC f the scuff plates at 08-9, for alternative in to perform this inspection of the scuff plates at 08-9, for alternative in the perform this inspection of the scuff plates at 08-9, for alternative in the perform this inspection of the scuff plates at 08-9, for alternative in the perform this inspection of the scuff plates at 08-9, for alternative in the perform this inspection of the scuff plates at 08-9, for alternative in the perform this inspection of the scuff plates at 08-9, for alternative in the scuff plates at 08-9, for	nspections. on is contained in ct 53-10-07. 18000 FC t all four corners (inspections. spection. Scuff plate the care found and the care found an	the 737 Nondestructure ALL upper/lower/fwd/aft, ate removal required ALL	ALL of the cargo d. ALL
53-646-62-01	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6 ACCESS NO AWL Inspect (Gen See Doc. D6	asonic) the bearstrap for 26A001-DTR, DTR checkhod(s) necessary to accommodate inspection procedur 53-05-02-211-865 ailed) the skin around the 3-30-08-9). 26A001-DTR, DTR check DTE: Forward cargo documents of the 3-05-02-210-808 areal Visual) the bearstra	k form 53-30-0 complish the intres are contained 1.1 e entire edge of the contained state of the contained 1.1 e entire edge of the contained 1.1 e entire edge of the contained the con	8-6 for alternative i ent of this inspecticed in Part 4, Subject 34000 FC f the scuff plates at 08-9, for alternative in to perform this insulation 34000 FC rners (upper/lower/08-9, for alternative	nspections. on is contained in ct 53-10-07. 18000 FC t all four corners (inspections. spection. Scuff plate the care found and the care found an	the 737 Nondestructure ALL upper/lower/fwd/aft, ate removal required ALL	A440 and STA ctive Test Man ALL) of the cargo d. ALL
53-646-62-01	Inspect (Ultra 492.4. See Doc D62 The NDI met (D6-37239). AWL Inspect (Deta door. (PSE 5 See Doc. D6 ACCESS NO AWL Inspect (Gen See Doc. D6	asonic) the bearstrap for 26A001-DTR, DTR check chod(s) necessary to accomplete the inspection procedur 53-05-02-211-865 ailed) the skin around the 3-30-08-9). 26A001-DTR, DTR check chock care of the inspection procedur 53-05-02-210-808 areal Visual) the bearstrae 26A001-DTR, DTR check chock c	k form 53-30-0 complish the intres are contained 1.1 e entire edge of the contained state of the contained 1.1 e entire edge of the contained 1.1 e entire edge of the contained the con	8-6 for alternative i ent of this inspecticed in Part 4, Subject 34000 FC f the scuff plates at 08-9, for alternative in to perform this insulation 34000 FC rners (upper/lower/08-9, for alternative	nspections. on is contained in ct 53-10-07. 18000 FC t all four corners (inspections. spection. Scuff plate the care found and the care found an	the 737 Nondestructure ALL upper/lower/fwd/aft, ate removal required ALL	A440 and STA ctive Test Man ALL) of the cargo d. ALL

Inspect (High Frequency Eddy Current) the outer chord around the fasteners common to the chord and bearstrap. See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-48.





			APPLICABILITY							
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-646-71-01	AWL	53-05-02-250-839	1.1	50000 FC	24000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the bearstrap along the upper edge of the forward cargo door. See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-48.									
53-646-75-01	AWL	53-05-02-211-806	1.1	50000 FC	6000 FC	600 700 800 900 900ER	ALL			
		ailed) the upper sill inner 26A001-DTR, DTR chec	•	,	e inspections.					
53-646-80-01	AWL	53-05-02-250-840	1.1	50000 FC	36000 FC	ALL	ALL			
	(D6-37239).	The inspection procedur TE: Removal of forwar	es are containe	ed in Part 6, Subjec	et 53-30-33.	the 737 Nondestruction.	ctive rest Mani			
53-647-00-01	(D6-37239).	The inspection procedur	es are containe	ed in Part 6, Subjec	et 53-30-33.		ALL			
53-647-00-01	AWL Inspect (High the forward a See Doc D6: The NDI met (D6-37239).	The inspection procedur	1.1 nt) the intercos nmon to the ba k form 53-30-0 complish the interes are contained	50000 FC tal web for cracks a ckup fitting and inte 9-3 for alternative in ent of this inspecticed in Part 6, Subject	et 53-30-33. d to perform this 9000 FC adjacent to rivets ercostal. nspections. on is contained in	ALL and fastener holes	ALL (five locations			
53-647-00-01 53-648-00-01	AWL Inspect (High the forward a See Doc D6: The NDI met (D6-37239).	The inspection procedur DTE: Removal of forwar 53-05-02-250-841 The Frequency Eddy Curre and aft edge frames) con 26A001-DTR, DTR check thod(s) necessary to accomplete inspection procedur	1.1 nt) the intercos nmon to the ba k form 53-30-0 complish the interes are contained	50000 FC tal web for cracks a ckup fitting and inte 9-3 for alternative in ent of this inspecticed in Part 6, Subject	et 53-30-33. d to perform this 9000 FC adjacent to rivets ercostal. nspections. on is contained in	ALL and fastener holes	ALL (five locations			
	AWL Inspect (High the forward a See Doc D6: The NDI met (D6-37239). AWL Inspect (High the forward a See Doc D6: The NDI met (D6-37239). AWL Inspect (High body interset See Doc D6: The NDI met (D6-37239).	The inspection procedur The inspection procedur The inspection procedur 53-05-02-250-841 The Frequency Eddy Curre and aft edge frames) con 26A001-DTR, DTR chect thod(s) necessary to acc The inspection procedur The inspection procedur The inspection procedur The inspection procedur The inspection procedur	1.1 nt) the intercos nmon to the ba k form 53-30-0 complish the interes are contained ars as required. 1.1 nt) around all fa s STA 518 to ST k form 53-30-1 complish the interes are contained	sed in Part 6, Subject cuff plate is required 50000 FC tal web for cracks a ckup fitting and inte 9-3 for alternative in ent of this inspectic ed in Part 6, Subject 50000 FC asteners in the ang A 555, above string 1-1 for alternative in ent of this inspectic ed in Part 6, Subject ded in Part 6, Subject	2t 53-30-33. d to perform this 9000 FC adjacent to rivets ercostal. Inspections. In is contained in the term of	ALL ALL ALL ALL ALL ALL ard and 12 inches at	ALL (five locations of the details) ALL (ft) of the wing of the w			

Inspect (High Frequency Eddy Current) around all fasteners in the angle (9 inches forward and 12 inches aft) of the wing to body intersection (STA 536) between STA 518 to STA 555, above stringer 24.

See Doc D626A001-DTR, DTR check form 53-30-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings is required.





				INTERVAL		APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-649-00-01	AWL	53-05-02-250-843	1.1	50000 FC	9000 FC	ALL	ALL	
	to the fusela See Doc D6. The NDI me (D6-37239).	n Frequency Eddy Curre ge (FWD of STA 536) an 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Removal of wing to body contour and	d the lower wirk form 53-30-1 complish the interest are contain to body fairings	ng skin (AFT of STA 1-02 for alternative ent of this inspectic ed in Part 6, Subject duct located aft of	A 536). inspections. on is contained in ct 53-30-47.	the 737 Nondestruc	ctive Test Manual	
53-649-00-02	AWL	53-05-02-250-843	1.1	50000 FC	9000 FC	ALL	ALL	
	to the fusela See Doc D6. The NDI me (D6-37239).	n Frequency Eddy Curre ge (FWD of STA 536) an 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Removal of wing to	nd the lower wirk form 53-30-1 complish the interest are contain	ng skin (AFT of STA 1-02 for alternative ent of this inspection ed in Part 6, Subject	A 536). inspections. on is contained in ct 53-30-47.	the 737 Nondestruc	ctive Test Manual	
53-649-01-01	AWL	53-05-02-250-844	lower wing skir	is required. 50000 FC	9000 FC	ALL	ALL	
	fuselage into See Doc D6. The NDI me (D6-37239).	n Frequency Eddy Curre erface (FWD of STA 536) 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Removal of wing to body contour and	and the angle k form 53-30-1 complish the intess are contain to body fairings	to lower wing skin 1-02 for alternative ent of this inspectic ed in Part 6, Subject duct located aft of	interface (AFT of inspections. on is contained in ot 53-30-47.	STA 536). the 737 Nondestruc	ctive Test Manual	
		·						
53-649-01-02	fuselage into See Doc D6. The NDI me (D6-37239).	53-05-02-250-844 In Frequency Eddy Curre Prace (FWD of STA 536) 26A001-DTR, DTR check thod(s) necessary to acc The inspection procedur DTE: Removal of wing to body contour and	and the angle k form 53-30-1 complish the intess are contain to body fairings	to lower wing skin 1-02 for alternative ent of this inspectic ed in Part 6, Subject duct located aft of	interface (AFT of inspections. on is contained in ot 53-30-47.	STA 536). the 737 Nondestruc	ctive Test Manual	
53-650-00-01	AWL	53-05-02-250-845						

AWL 53-05-02-250-845 1.1 50000 FC 36000 FC ALL ALL

Inspect (High Frequency Eddy Current) around the fasteners in the inboard and outboard angles eight inches forward and

See Doc. D626A001-DTR, DTR check form 53-30-11-03, for alternative inspections.

aft of STA 536.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings and duct located aft of STA 536 is required.





TASK CARD NO.				INTERVAL		APPLICA	BILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-650-00-02	AWL	53-05-02-250-845	1.1	50000 FC	36000 FC	ALL	ALL
	aft of STA 53 See Doc. D6 The NDI met	n Frequency Eddy Curre 66. 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-30-7 omplish the int	11-03, for alternativent of this inspection	re inspections. on is contained in		
	,	OTE: Removal of wing to		•		equired.	
53-652-00-01	AWL	53-05-02-211-866	1.1	50000 FC	4000 FC	ALL	ALL
	except at the	ailed) the skin around all lap splices and antenna 26A001-DTR, DTR chec	as. (53-40-01-1).		10R, from STA 540 to	STA 727,
53-653-00-01	AWL	53-05-02-250-935	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL
53-653-00-02		thod(s) necessary to acc The inspection procedur					
33-033-00-02	AWL	53-05-02-250-935	1.1	50000 FC	36000 FC	600 700 700IGW 800	ALL
33-033-00-02	Inspect (High	n Frequency Eddy Curre				700IGW 800 900 900ER	
33-033-00-02	Inspect (High 540 to STA 7	n Frequency Eddy Curre 27. (PSE 53-40-03-1).	nt) the upper s	kin along the uppe	r fastener row at s	700IGW 800 900 900ER	
33-033-00-02	Inspect (High 540 to STA 7 See Doc. Do The NDI met	n Frequency Eddy Curre	nt) the upper s ck form 53-10-0 omplish the int	kin along the upper 03-1, for alternative ent of this inspection	r fastener row at s inspections. on is contained in	700IGW 800 900 900ER stringers S-4L and S-	-4R from STA
53-654-00-01	Inspect (High 540 to STA 7 See Doc. Do The NDI met	n Frequency Eddy Curre (27. (PSE 53-40-03-1). (26A001-DTR, DTR check (thod(s) necessary to acc	nt) the upper s ck form 53-10-0 omplish the int	kin along the upper 03-1, for alternative ent of this inspection	r fastener row at s inspections. on is contained in	700IGW 800 900 900ER stringers S-4L and S-	-4R from STA
	Inspect (High 540 to STA 7 See Doc. D6 The NDI met (D6-37239). AWL Inspect (Low to STA 727. See Doc. D6 The NDI met	n Frequency Eddy Curre (27. (PSE 53-40-03-1). (26A001-DTR, DTR check (thod(s) necessary to acc The inspection procedur	nt) the upper s ck form 53-10-(omplish the int res are contain 1.1 nt) the lower sk ck form 53-30-(omplish the int	kin along the upper 03-1, for alternative ent of this inspectic ed in Part 6, Subject 50000 FC tin along the lower 04-2, for alternative ent of this inspectic	r fastener row at see inspections. on is contained in ct 53-30-41. 18000 FC fastener row at stee inspections. on is contained in	700IGW 800 900 900ER stringers S-4L and S- the 737 Nondestruc 600 700 700IGW 800 900 900ER ringers S-4L and S-4	-4R from STA tive Test Manu ALL -RR from STA 5

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (53-40-03-2).

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.







				INTERVAL		APPLICABILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
53-655-00-01	AWL	53-05-02-250-847	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
		Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 540 to STA 727.									
		26A001-DTR, DTR ched		•	•						
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	tive Test Man				
53-655-00-02	AWL	53-05-02-250-847	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
	540 to STA 7 See Doc. D6 The NDI met	n Frequency Eddy Curre 27. 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-40-0 omplish the int	03-3, for alternative ent of this inspection	inspections.	Ü					
53-656-00-01	AWL	53-05-02-211-822	1.1	50000 FC	24000 FC	ALL	ALL				
	See Doc. D6	ailed) stringers S-11 and 26A001-DTR, DTR chec DTE: Removal and/or di	ck form 53-40-0	04-1, for alternative		lation blankets is rec	uuired				
	ACCESS NO		spiacement of		dewalls and insu		julieu.				
53-656-00-02	AWL	53-05-02-211-822	1.1	50000 FC	24000 FC	ALL	ALL				
		Inspect (Detailed) stringers S-11 and S-13 from STA 540 to 727. See Doc. D626A001-DTR, DTR check form 53-40-04-1, for alternative inspections.									
	ACCESS NO	OTE: Removal and/or di	splacement of	passenger cabin si	idewalls and insu	lation blankets is req	uired.				
53-657-00-01	AWL	53-05-02-211-823	1.1	50000 FC	18000 FC	ALL	ALL				
		ailed) the skin from string 26A001-DTR, DTR chec	,			540 to STA 727.					
53-657-00-02	AWL	53-05-02-211-823	1.1	50000 FC	18000 FC	ALL	ALL				
		ailed) the skin from string 26A001-DTR, DTR chec				540 to STA 727.					
53-658-00-01	AWL	53-05-02-211-824	1.1	50000 FC	18000 FC	ALL	ALL				
		ailed) stringers S-11 to S 26A001-DTR, DTR chec				TA 663.					
	ACCESS NO	OTE: Remove and/or dis	splace passenç	ger cabin sidewall p	anels and insulat	tion blankets.					
53-658-00-02	AWL	53-05-02-211-824	1.1	50000 FC	18000 FC	ALL	ALL				

See Doc. D626A001-DTR, DTR check form 53-40-07-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall panels and insulation blankets.





			APPLICA	ABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-659-00-01	AWL	53-05-02-211-825	1.1	50000 FC	18000 FC	ALL	ALL			
		ailed) the keel beam side 26A001-DTR, DTR chec	•							
53-659-00-02	AWL	53-05-02-211-825	1.1	50000 FC	18000 FC	ALL	ALL			
	. ,	ailed) the keel beam side 26A001-DTR, DTR chec	•							
53-660-00-01	AWL	53-05-02-250-848	1.1	50000 FC	36000 FC	ALL	ALL			
	 See Doc. D626A001-DTR, DTR check form 53-40-10-1 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-29. ACCESS NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection. 									
53-660-00-02	AWL	53-05-02-250-848	1.1	50000 FC	36000 FC	ALL	ALL			
	attachment fr at LBL and R See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre itting that joins the stiffer RBL 45 and WL 202.6. i26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-40-1 complish the intres are contained	beam (at five location of this inspection of this inspection of the part 6, Section of the	ions) around the finspections. on is contained in 53-11-29.	astener/collar on the	e outboard side			
	ACCESS NO	OTE: Removal and/or di insulation blankets				sidewall panels and				
53-661-00-01	AWL	53-05-02-250-849	1.1	50000 FC	36000 FC	ALL	ALL			
	and skin fron	r Frequency Eddy Currer n STA 717 to STA 727. 26A001-DTR, DTR chec	,	·		n the fasteners com	mon to the strap			

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-49.

53-661-00-02 AWL 53-05-02-250-849 50000 FC ALL ALL 1.1 36000 FC

> Inspect (Low Frequency Eddy Current) the skin under the strap at stringer S-18 between the fasteners common to the strap and skin from STA 717 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-49.





			APPLICABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-662-00-01	AWL	53-05-02-250-850	1.1	50000 FC	36000 FC	ALL	ALL
	surface at S ⁻ See Doc. D6 The NDI met	n Frequency Eddy Currer FA 663. 26A001-DTR, DTR chec hod(s) necessary to acc The inspection procedur	ck form 53-40- omplish the int	12-1 for alternative ent of this inspection	inspections. on is contained in		
53-662-00-02	AWL	53-05-02-250-850	1.1	50000 FC	36000 FC	ALL	ALL
	surface at S ⁻ See Doc. D6 The NDI met	n Frequency Eddy CurrentA 663. 26A001-DTR, DTR chechod(s) necessary to acc The inspection procedur	ck form 53-40-7 omplish the int	12-1 for alternative ent of this inspection	inspections.	•	
53-663-00-01	AWL	53-05-02-250-851	1.1	50000 FC	18000 FC	ALL	ALL
	splice plate b	n Frequency Eddy Curre between stringers S-8 an 26A001-DTR, DTR chec	d S-9.			o the inner chord ar	nd to the inner
	ACCESS NO	OTE: Removal and/or dis	splacement of	passenger cabin si	dewalls and insul	ation blankets as re	quired.
53-663-00-02	AWL	53-05-02-250-851	1.1	50000 FC	18000 FC	ALL	ALL
	splice plate b See Doc. D6	n Frequency Eddy Current between stringers S-8 and 26A001-DTR, DTR chec DTE: Removal and/or dis	d S-9. ck form 53-40-	14-1 for alternative	inspections.		
53-664-00-01	AWL	53-05-02-211-826	1.1	50000 FC	18000 FC	ALL	ALL
	the frame to	ailed) the skin panels at t skin fastener holes. 26A001-DTR, DTR chec		-		ch side of splice 540), for cracks at
53-665-00-01	AWL	53-05-02-211-827	1.1	50000 FC	18000 FC	ALL	ALL
		ailed) the skin on each si 26A001-DTR, DTR chec				the left and right sid	les.
	AWL		1.1	50000 FC	18000 FC	ALL	A1.1
53-665-00-02	~vvL	53-05-02-211-827					ALL
53-665-00-02	Inspect (Deta	53-05-02-211-827 ailed) the skin on each si 26A001-DTR, DTR chec	ide of STA 663	,	,	the left and right sid	
53-665-00-02 53-666-00-01	Inspect (Deta	ailed) the skin on each si	ide of STA 663	,	,	the left and right sid	

hand sides.

See Doc. D626A001-DTR, DTR check form 53-40-15-3 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets as required.







			INTERVAL			APPLICA	ABILITY	
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-666-00-02	AWL	53-05-02-250-852	1.1	50000 FC	24000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the bulkhead inner chord from stringers S-10 and S-17 on both the left and right hand sides. See Doc. D626A001-DTR, DTR check form 53-40-15-3 for alternative inspections.							
	ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets as required.							
53-667-00-01	AWL	53-05-02-250-854	1.1	50000 FC	16000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the fail safe angle from inside the aft cargo bay at frame 727, from stringers S-21L to S-27L and stringers S-21R to S-27R. See Doc. D626A001-DTR, DTR check form 53-40-16-1a for alternative inspections.							
	ACCESS NO	OTE: Removal and/or di insulation blankets		aft cargo forward b perform the inspec		sidewall panels and		
53-667-00-02	AWL	53-05-02-250-854	1.1	50000 FC	16000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the fail safe angle from inside the aft cargo bay at frame 727, from stringers S-21L to S-27L and stringers S-21R to S-27R. See Doc. D626A001-DTR, DTR check form 53-40-16-1a for alternative inspections.							
	ACCESS NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.							
53-667-10-01	AWL	53-05-02-250-A98	1.1	50000 FC	18000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the chord at frame 727 from stringers S-21L to S-27L and stringers S-21R to S-27l See Doc. D626A001-DTR, DTR check form 53-40-16-1b for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manu (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-93.							
	See Doc. D6 The NDI met	626A001-DTR, DTR chec thod(s) necessary to acc	ck form 53-40- omplish the int	16-1b for alternative ent of this inspection	e inspections. on is contained in	-	S-21R to S-27	
53-667-10-02	See Doc. D6 The NDI met	626A001-DTR, DTR chec thod(s) necessary to acc	ck form 53-40- omplish the int	16-1b for alternative ent of this inspection	e inspections. on is contained in	-	S-21R to S-27	
53-667-10-02	See Doc. D6 The NDI met (D6-37239). AWL Inspect (High See Doc. D6 The NDI met	326A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-40- complish the interes are contain 1.1 nt) the chord a ck form 53-40- complish the interes.	16-1b for alternative ent of this inspectic ed in Part 6, Subjection 50000 FC trame 727 from strange for alternative ent of this inspectic	e inspections. on is contained in ct 53-10-93. 18000 FC tringers S-21L to be inspections. on is contained in	the 737 Nondestructure ALL S-27L and stringers	S-21R to S-27 ctive Test Manu ALL S-21R to S-27	
53-667-10-02 53-668-10-01	See Doc. D6 The NDI met (D6-37239). AWL Inspect (High See Doc. D6 The NDI met	526A001-DTR, DTR check thod(s) necessary to acc The inspection procedur 53-05-02-250-A98 in Frequency Eddy Curre 526A001-DTR, DTR check thod(s) necessary to acc	ck form 53-40- complish the interes are contain 1.1 nt) the chord a ck form 53-40- complish the interes.	16-1b for alternative ent of this inspectic ed in Part 6, Subjection 50000 FC trame 727 from strange for alternative ent of this inspectic	e inspections. on is contained in ct 53-10-93. 18000 FC tringers S-21L to be inspections. on is contained in	the 737 Nondestructure ALL S-27L and stringers	S-21R to S-27 ctive Test Manu ALL S-21R to S-27	
	AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239). AWL Inspect (High See Doc. D6 The NDI met (D6-37239). AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239).	526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedur 53-05-02-250-A98 in Frequency Eddy Curre 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedur	ck form 53-40- complish the intes are contain 1.1 nt) the chord a ck form 53-40- complish the intes are contain 1.1 d the six fasten ck form 53-40- complish the intex form 53-40- complish the intex fasten ck form 53-40- complish the intex fasten intex form 53-40- complish the intex fasten intex fast	16-1b for alternative ent of this inspecticed in Part 6, Subjection of the Subject of the Subject of the Subject of the Subject of this inspecticed in Part 6, Subject of the Subject of this inspecticed in Part 6 of the Subject of this inspectice of this inspectice.	e inspections. on is contained in ct 53-10-93. 18000 FC ringers S-21L to be inspections. on is contained in ct 53-10-93. er chord and webe inspections. on is contained in ct 53-10-93.	ALL S-27L and stringers the 737 Nondestruct ALL at STA 727 and WI	S-21R to S-27 ctive Test Manu ALL S-21R to S-27 ctive Test Manu ALL ALL 201.	
	AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239). AWL Inspect (High See Doc. D6 The NDI met (D6-37239). AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239).	526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 53-05-02-250-A98 in Frequency Eddy Curre 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 53-05-02-130-804 in the inspection procedure 53-05-02-130-804 in the inspection procedure 53-05-02-130-804 in the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR check thod(s) necessary to accommod the inspection procedure 526A001-DTR check thod(s) necessary the inspection procedure 526A001-DTR check thod(s) necessary	ck form 53-40- complish the intes are contain 1.1 nt) the chord a ck form 53-40- complish the intes are contain 1.1 If the six fasten ck form 53-40- complish the intes are contain	16-1b for alternative ent of this inspectice ed in Part 6, Subject 50000 FC thrame 727 from standard from the foliation of this inspectice ed in Part 6, Subject NOTE ers through the innumber of this inspectice ent of this inspectice ed in Part 4, Subject ed in Par	e inspections. on is contained in ot 53-10-93. 18000 FC tringers S-21L to be inspections. on is contained in ot 53-10-93. er chord and webe inspections. on is contained in ot 53-10-09.	ALL S-27L and stringers the 737 Nondestruct ALL at STA 727 and WI	S-21R to S-27 ctive Test Manu ALL S-21R to S-27 ctive Test Manu ALL ALL 201.	
	AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239). AWL Inspect (High See Doc. D6 The NDI met (D6-37239). AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239).	526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 53-05-02-250-A98 in Frequency Eddy Curres 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 53-05-02-130-804 in the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure thod(s) necessary to accommod the inspection procedure the inspection procedure.	2k form 53-40- complish the intres are contain 1.1 nt) the chord a ck form 53-40- complish the intres are contain 1.1 If the six fasten ck form 53-40- complish the intres are contain is effective for	16-1b for alternative ent of this inspectice ed in Part 6, Subject 50000 FC thrame 727 from standard from the foliation of this inspectice ed in Part 6, Subject NOTE ers through the innumber of this inspectice ent of this inspectice ed in Part 4, Subject ed in Par	e inspections. on is contained in ct 53-10-93. 18000 FC tringers S-21L to be inspections. on is contained in ct 53-10-93. er chord and webe inspections. on is contained in ct 53-10-09. Il 1193 and On.	ALL S-27L and stringers the 737 Nondestruct ALL at STA 727 and WI the 737 Nondestruct	S-21R to S-27 ctive Test Manu ALL S-21R to S-27 ctive Test Manu ALL ALL 201.	
	AWL Inspect (High See Doc. D6 The NDI met (D6-37239). AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239). AWL Inspect (Ultra See Doc. D6 The NDI met (D6-37239). AIRPLANE I	526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 53-05-02-250-A98 in Frequency Eddy Curre 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure 53-05-02-130-804 in the inspection procedure 526A001-DTR, DTR check thod(s) necessary to accommod the inspection procedure thod(s) necessary to accommod the inspection procedure thod(s) necessary to accommod the inspection procedure the inspection procedure. This DTR Form in See interval notes that the inspection procedure the inspection procedure. The inspection procedure the inspection procedure the inspection procedure the inspection procedure.	2k form 53-40- complish the intes are contain 1.1 Int) the chord a ck form 53-40- complish the intes are contain 1.1 If the six fasten ck form 53-40- complish the intes are contain is effective for e for airplane s -700 LN # 119	16-1b for alternative ent of this inspecticed in Part 6, Subject 50000 FC thrame 727 from standard from Ent of this inspecticed in Part 6, Subject NOTE ers through the innumber of this inspecticed in Part 4, Subject ent of this inspection of this inspect	e inspections. on is contained in ct 53-10-93. 18000 FC ringers S-21L to be inspections. on is contained in ct 53-10-93. er chord and webe inspections. on is contained in ct 53-10-09. I 1193 and On. ond repeat intervals	ALL S-27L and stringers the 737 Nondestruct ALL at STA 727 and WI the 737 Nondestruct s.	S-21R to S-27 ctive Test Manu ALL S-21R to S-27 ctive Test Manu ALL 201. ctive Test Manu ctive Test Manu	





				INTERVAL		APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-668-10-02	AWL	53-05-02-130-804	1.1	NOTE		ALL	ALL	
	Inspect (Ultrasonic) the frame around the six fasteners through the inner chord and web at STA 727 and WL 201. See Doc. D626A001-DTR, DTR check form 53-40-16-2a for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Mani (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-09.							
	AIRPLANE	NOTE: This DTR Form See interval note		all models from L/N pecific threshold a		S.		
	INTERVAL I	NOTE: For 737-600 and For 737-700C/-7				Repeat is 24,000 FC 0 FC, Repeat is 24,		
	ACCESS NO	OTE: Remove and/or dis blankets as require			l, ceiling, sidewall	panels and insulation	on	
53-669-00-01	AWL	53-05-02-250-853	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL	
	and right hand sides. See Doc. D626A001-DTR, DTR check form 53-40-16-3 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manu (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-25. ACCESS NOTE: Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation							
53-669-00-02	AWL	blankets as require 53-05-02-250-853	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL	
	Inspect (High Frequency Eddy Current) the frame inner chord at STA 727 between stringers S-17 and S-21 on both the leand right hand sides. See Doc. D626A001-DTR, DTR check form 53-40-16-3 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Mani (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-25. ACCESS NOTE: Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation							
		blankets as require	ed to perform the	he inspection.	-			
53-670-00-01	AWL	53-05-02-211-832	1.1	50000 FC	9000 FC	ALL	ALL	
	Inspect (Detailed) the frame inner chord and web between stringers S-9L and S-9R at STA 727. See Doc. D626A001-DTR, DTR check form 53-40-16-5 for alternative inspections. ACCESS NOTE: Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the							
	ACCESS NO	OTE: Remove and/or dis inspection.	spiace passenç	ger cabin ceiling pa	ineis and insulatio	on as required to per	torm the	
53-671-00-01	AWL	53-05-02-250-855	1.1	50000 FC	24000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the visible portion of the frame web above the splice angle on the forward side of the frame between stringers S-9 and S-10 on both sides of the aircraft at STA 727. See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections							

See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-92.

ACCESS NOTE: Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.







			INTERVAL			APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-671-00-02	AWL	53-05-02-250-855	1.1	50000 FC	24000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the visible portion of the frame web above the splice angle on the forward side of the frame between stringers S-9 and S-10 on both sides of the aircraft at STA 727. See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-92. ACCESS NOTE: Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.							
53-672-00-01	AWL	53-05-02-250-856	1.1	50000 FC	9000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the frame inner chord and fail-safe angle (around the fasteners common to the fail-safe angle), the forward frame web (around the fasteners common to the fail-safe angle), and the frame outer chord (around the fasteners common to the skin) between stringers S-10 and S-13. See Doc. D626A001-DTR, DTR check form 53-40-17-1 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-19.							
		TE: Remove and/or dis the inspection.				·	· 	
53-672-00-02	AWL	53-05-02-250-856	1.1	50000 FC	9000 FC	ALL	ALL	
	Inspect (High Frequency Eddy Current) the frame inner chord and fail-safe angle (around the fasteners common to the fail-safe angle), the forward frame web (around the fasteners common to the fail-safe angle), and the frame outer chord (around the fasteners common to the skin) between stringers S-10 and S-13. See Doc. D626A001-DTR, DTR check form 53-40-17-1 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-19. ACCESS NOTE: Remove and/or displace passenger cabin sidewalls and insulation blankets as required to perform the inspection.							
53-673-00-01	AWL	53-05-02-210-802	1.1	50000 FC	24000 FC	ALL	ALL	
	Inspect (General Visual) the upper fastener through the web. See Doc. D626A001-DTR, DTR check form 53-40-18-2 for alternative inspections.							
	ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.							
53-673-00-02	AWL	53-05-02-210-802	1.1	50000 FC	24000 FC	ALL	ALL	
	Inanast (Can	eral Visual) the upper fa	otopor through	thowah				

See Doc. D626A001-DTR, DTR check form 53-40-18-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.





53-676-00-01



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
53-674-00-01	AWL	53-05-02-250-857	1.1	50000 FC	18000 FC	ALL	ALL		
	the forward a See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre and aft flanges of the fra 26A001-DTR, DTR check thod(s) necessary to acc The inspection procedur DTE: Remove and/or dis blankets as require	me inner chord ck form 53-40- complish the intres are contain splace passeng	from 6 inches abo 19-1 for alternative ent of this inspection ed in Part 6, Section ger cabin sidewall p	ve and below strir inspections. on is contained in n 53-40-05.	nger S-16.	ctive Test Manu		
53-674-00-02	AWL	53-05-02-250-857	1.1	50000 FC	18000 FC	ALL	ALL		
33 014 33 32	Inspect (High the forward a See Doc. D6 The NDI met	n Frequency Eddy Curre and aft flanges of the fra 26A001-DTR, DTR cheat thod(s) necessary to acc The inspection procedure	nt) the frame w me inner chord ck form 53-40- complish the int	veb around the fast from 6 inches abo 19-1 for alternative ent of this inspection	eners common to ve and below strir inspections. on is contained in	the stringer clip at nger S-16.	stringer S-16 ar		
		DTE: Remove and/or dis	ed to perform the	ne inspection.	· 				
53-675-00-01	AWL 53-05-02-250-858 1.1 50000 FC 36000 FC ALL ALL								
	inner chord. See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre (26A001-DTR, DTR cheichod(s) necessary to acc The inspection procedur (DTE: Remove and/or dispersion as required to perf	ck form 53-40- complish the intres are contain- splace passeng	19-2 for alternative ent of this inspection ed in Part 6, Section ger cabin sidewall a	inspections. on is contained in n 53-11-14.	the 737 Nondestru	ctive Test Manu		
53-675-00-02	AWL	53-05-02-250-858	1.1	50000 FC	36000 FC	ALL	ALL		
	inner chord. See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre (26A001-DTR, DTR check (thod(s) necessary to acc The inspection procedure (DTE: Remove and/or dispersion as required to perf	ck form 53-40- complish the intres are contain- splace passeng	19-2 for alternative ent of this inspection ed in Part 6, Section ger cabin sidewall a	inspections. on is contained in n 53-11-14.	the 737 Nondestru	ctive Test Manu		

AWL 53-05-02-250-859 1.1 50000 FC 24000 FC ALL ALL

Inspect (High Frequency Eddy Current) the web of the stub beam around the fasteners common to the floor clip.

See Doc. D626A001-DTR, DTR check form 53-40-19-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.







			APPLICABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
53-676-00-02	AWL	53-05-02-250-859	1.1	50000 FC	24000 FC	ALL	ALL		
	See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre 626A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur OTE: Remove and/or dis	ck form 53-40-1 complish the interes are contained	9-3 for alternative in ent of this inspection and in Part 6, Section	inspections. on is contained in n 53-11-09.	the 737 Nondestruc	ctive Test Manu		
		as required to perf			ge.,e p.a.				
53-676-10-01	AWL	53-05-02-250-860	1.1	50000 FC	15000 FC	ALL	ALL		
	BL 64.6 on b See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre both the right and left side 526A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur OTE: Remove and/or dis	es. ck form 53-40-1 complish the inte es are containe	9-3a for alternative ent of this inspection ed in Part 6, Section	e inspections. on is contained in n 53-11-09.	the 737 Nondestruc	ctive Test Manu		
		as required to perf			3 - 13, 13 p.				
53-676-10-02	AWL	53-05-02-250-860	1.1	50000 FC	15000 FC	ALL	ALL		
	BL 64.6 on both the right and left sides. See Doc. D626A001-DTR, DTR check form 53-40-19-3a for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09. ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.								
			orm the inspect	tion.			olankets		
			orm the inspec	tion.			olankets		
53-677-00-01	AWL		orm the inspect	50000 FC	24000 FC	ALL	olankets ALL		
53-677-00-01	Inspect (Deta	as required to perf	1.1 ord from stringe	50000 FC ers S-13 to S-15.		ALL			
53-677-00-01	Inspect (Deta See Doc. D6	as required to perform 53-05-02-211-833 railed) the frame inner cho	1.1 ord from stringe ck form 53-40-2 splace passeng	50000 FC ers S-13 to S-15. 21-2 for alternative iter cabin sidewalls,	inspections.		ALL		
53-677-00-01 53-677-00-02	Inspect (Deta See Doc. D6	as required to perform a requi	1.1 ord from stringe ck form 53-40-2 splace passeng	50000 FC ers S-13 to S-15. 21-2 for alternative iter cabin sidewalls,	inspections.		ALL		
	Inspect (Deta See Doc. D6 ACCESS NO AWL Inspect (Deta	as required to perform as required to perform as required to perform as required to perform	1.1 ord from stringeck form 53-40-2 splace passengen the inspection 1.1 ord from stringe	50000 FC ers S-13 to S-15. t1-2 for alternative intercabin sidewalls, inc. 50000 FC ers S-13 to S-15.	inspections. sidewall air grille 24000 FC	s, and insulation bla	ALL ankets as		
	AWL Inspect (Deta See Doc. D6 ACCESS NC AWL Inspect (Deta See Doc. D6	as required to perform 53-05-02-211-833 cailed) the frame inner che 626A001-DTR, DTR chec OTE: Remove and/or dis required to perform 53-05-02-211-833 cailed) the frame inner che	1.1 ord from stringeck form 53-40-2 splace passengen the inspection 1.1 ord from stringeck form 53-40-2 splace passeng	50000 FC ers S-13 to S-15. et-2 for alternative in the control of	inspections. sidewall air grille 24000 FC inspections.	s, and insulation bla ALL	ALL ankets as ALL		

Inspect (High Frequency Eddy Current) the frame inner chord flange and around accessible fasteners common to the inner chord and stringer clips from stringers S-17 to S-14.

See Doc. D626A001-DTR, DTR check form 53-40-21-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-17.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.







				INTERVAL			APPLICABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-678-00-02	AWL	53-05-02-250-861	1.1	50000 FC	36000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the frame inner chord flange and around accessible fasteners common to the inner chord and stringer clips from stringers S-17 to S-14.									
	See Doc. D626A001-DTR, DTR check form 53-40-21-3 for alternative inspections.									
	The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual									
	(D6-37239). The inspection procedures are contained in Part 6, Section 53-11-17.									

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.

53-679-00-01 AWL 53-05-02-250-862 1.1 50000 FC 36000 FC ALL ALL

Inspect (High Frequency Eddy Current) the stub beam upper chord from two inches inside the skin to a distance of twelve inches inboard and around any fasteners through the upper web and chord in this area.

See Doc. D626A001-DTR, DTR check form 53-40-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-06.

ACCESS NOTE: Remove and/or displace passenger cabin floor panels as required to perform the inspection.

53-679-00-02 AWL 53-05-02-250-862 1.1 50000 FC 36000 FC ALL ALL

Inspect (High Frequency Eddy Current) the stub beam upper chord from two inches inside the skin to a distance of twelve inches inboard and around any fasteners through the upper web and chord in this area.

See Doc. D626A001-DTR, DTR check form 53-40-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-06.

ACCESS NOTE: Remove and/or displace passenger cabin floor panels as required to perform the inspection.

53-682-00-01 AWL 53-05-02-250-867 1.1 50000 FC 4000 FC 800 900 900 ER ALL

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the STA 578 cutout forward edge frame outer chord from stringers S-10 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection must be work in conjunction with either fatigue task 53-682-01 or 53-682-03 to meet DTR requirements..

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

53-682-00-02 AWL 53-05-02-250-867 1.1 50000 FC 4000 FC 800 900 900 ER ALL

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the STA 578 cutout forward edge frame outer chord from stringers S-10 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection must be work in conjunction with either fatigue task 53-682-01 or 53-682-03 to meet DTR requirements..

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.



53-685-00-01

AWL



737-600/700/800/900 **TASK CARDS**

			INTERVAL		APPLICA	BILITY
SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
AWL	53-05-02-250-868	1.1	50000 FC	4000 FC	800 900 900ER	ALL
See Doc. D6 NOTE: This The NDI met (D6-37239).	\$26A001-DTR, DTR check inspection may be option thod(s) necessary to accept the inspection procedure. NOTE: Applicable to 73	ck form 53-40-2 nal. See DTR 5 omplish the inte es are containe 7-800 (line num	22-3 for alternative 3-40-22-3 for inspe- ent of this inspection ed in Part 6, Section on the substance of t	inspections. ection requireme on is contained in n 53-11-15. 737-900 airplan	nts. n the 737 Nondestruct	ive Test Mani
AWL	53-05-02-250-868	1.1	50000 FC	4000 FC	800 900 900ER	ALL
NOTE: This The NDI met (D6-37239).	inspection may be option thod(s) necessary to acc The inspection procedur NOTE: Applicable to 73	nal. See DTR 5 omplish the interes are contained 7-800 (line num	3-40-22-3 for inspection of this inspection of the inspection of t	ection requireme on is contained in n 53-11-15. 737-900 airplan	n the 737 Nondestruct	ive Test Man
AWL	53-05-02-250-869	1.1	50000 FC	4000 FC	800 900 900ER	ALL
See Doc. D6	326A001-DTR, DTR ched	ck form 53-40-2 nal. See DTR 5	22-3 for alternative	inspections.	nts.	
	thod(s) necessary to acc The inspection procedur				n the 737 Nondestruct	ive Test Man
(D6-37239).	` '	es are containe	ed in Part 6, Sectio	n 53-11-15.		ive Test Man
(D6-37239). AIRPLANE	The inspection procedur	es are containe 7-800 (line num oor must be op	ed in Part 6, Sectionber 9 and on) and en to perform this i	n 53-11-15. 737-900 airplan	es.	
(D6-37239). AIRPLANE	The inspection procedur NOTE: Applicable to 73 OTE: Emergency Exit December 2015	es are containe 7-800 (line num oor must be op	ed in Part 6, Sectionber 9 and on) and en to perform this i	n 53-11-15. 737-900 airplan	es.	
	AWL Inspect (High See Doc. Doc. Doc. Doc. Doc. Doc. Doc. Doc.	AWL 53-05-02-250-868 Inspect (High Frequency Eddy Curre See Doc. D626A001-DTR, DTR check NOTE: This inspection may be option The NDI method(s) necessary to acc (D6-37239). The inspection procedur AIRPLANE NOTE: Applicable to 73 ACCESS NOTE: Emergency Exit Document December 2015 ACCESS NOTE: Emergency Exit Document December 2015 ACCESS NOTE: Emergency Eddy Curre See Doc. D626A001-DTR, DTR check NOTE: This inspection may be option The NDI method(s) necessary to acc (D6-37239). The inspection procedur AIRPLANE NOTE: Applicable to 73 ACCESS NOTE: Emergency Exit Document December 2015 ACCESS NOTE: Emergency Eddy Curre See Doc. D626A001-DTR, DTR checks	AWL 53-05-02-250-868 1.1 Inspect (High Frequency Eddy Current) the edges of See Doc. D626A001-DTR, DTR check form 53-40-2 NOTE: This inspection may be optional. See DTR 5 The NDI method(s) necessary to accomplish the int (D6-37239). The inspection procedures are contained AIRPLANE NOTE: Applicable to 737-800 (line number ACCESS NOTE: Emergency Exit Door must be open AWL 53-05-02-250-868 1.1 Inspect (High Frequency Eddy Current) the edges of See Doc. D626A001-DTR, DTR check form 53-40-2 NOTE: This inspection may be optional. See DTR 5 The NDI method(s) necessary to accomplish the int (D6-37239). The inspection procedures are contained AIRPLANE NOTE: Applicable to 737-800 (line number ACCESS NOTE: Emergency Exit Door must be open AWL 53-05-02-250-869 1.1 Inspect (High Frequency Eddy Current) the inner do See Doc. D626A001-DTR, DTR check form 53-40-250-869 DTR DTR DTR Check form 53-40-250-869 DTR DTR DTR Check form 53-40-250-869 DTR	AWL 53-05-02-250-868 1.1 50000 FC Inspect (High Frequency Eddy Current) the edges of the doublers from See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection may be optional. See DTR 53-40-22-3 for inspection procedures are contained in Part 6, Section AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and ACCESS NOTE: Emergency Exit Door must be open to perform this inspect (High Frequency Eddy Current) the edges of the doublers from See Doc. D626A001-DTR, DTR check form 53-40-22-3 for inspect (NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspect (D6-37239). The inspection procedures are contained in Part 6, Section AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and ACCESS NOTE: Emergency Exit Door must be open to perform this inspection for the Note of the inspection of the in	AWL 53-05-02-250-868 1.1 50000 FC 4000 FC Inspect (High Frequency Eddy Current) the edges of the doublers from stringers S-11 to See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections. NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirement The NDI method(s) necessary to accomplish the intent of this inspection is contained in (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15. AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplant ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. AWL 53-05-02-250-868 1.1 50000 FC 4000 FC Inspect (High Frequency Eddy Current) the edges of the doublers from stringers S-11 to See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections. NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirement The NDI method(s) necessary to accomplish the intent of this inspection is contained in (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15. AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplant ACCESS NOTE: Emergency Exit Door must be open to perform this inspection.	AWL 53-05-02-250-868 1.1 50000 FC 4000 FC 800 900 900 ER Inspect (High Frequency Eddy Current) the edges of the doublers from stringers S-11 to S-13. See Doc. D626A001-DTR, DTR check form 53-40-22-3 for inspection requirements. NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection is contained in the 737 Nondestructi (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15. AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes. ACCESS NOTE: Emergency Exit Door must be open to perform this inspections. NOTE: This inspection procedures are contained in Part 6, Section 53-11-15. AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes. ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. AWL 53-05-02-250-868 1.1 50000 FC 4000 FC 800 900 900ER Inspect (High Frequency Eddy Current) the edges of the doublers from stringers S-11 to S-13. See Doc. D626A001-DTR, DTR check form 53-40-22-3 for inspection requirements. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestruction (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15. AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes. ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. AWL 53-05-02-250-869 1.1 50000 FC 4000 FC 800 900 900ER Inspect (High Frequency Eddy Current) the inner doubler from stringers S-11 to S-13. See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. Seal removal or displacement is required to perform the inspection.

1.1

53-05-02-250-874 50000 FC 6000 FC 800 900 900ER Inspect (High Frequency Eddy Current) the edges of the doublers, on the upper edge, at stringer S-11(from STA 578 to STA

601 and from STA 616 to STA 639). See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection.

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ALL





				INTERVAL		APPLICA	BILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
53-685-00-02	AWL	53-05-02-250-874	1.1	50000 FC	6000 FC	800 900 900ER	ALL		
	Inspect (High Frequency Eddy Current) the edges of the doublers, on the upper edge, at stringer S-11(from STA 578 to ST 601 and from STA 616 to STA 639).								
		26A001-DTR, DTR checks with external doublers			•	2-22 for area covered	by the doubler.		

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection.

53-685-01-01 AWL 53-05-02-250-875 1.1 50000 FC 6000 FC 800 900 900 ER

Inspect (High Frequency Eddy Current) the inner doubler between the seal retainer and the frames and sills, on the upper edge, at stringer S-11 (from STA 578 to STA 601 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform this inspection. Remove or displace passenger cabin sidewall lining as required to perform this inspection.

53-685-01-02 AWL 53-05-02-250-875 1.1 50000 FC 6000 FC 800 900 900 ER ALL

Inspect (High Frequency Eddy Current) the inner doubler between the seal retainer and the frames and sills, on the upper edge, at stringer S-11 (from STA 578 to STA 601 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform this inspection. Remove or displace passenger cabin sidewall lining as required to perform this inspection.

53-686-00-01 AWL 53-05-02-130-805 1.1 50000 FC 36000 FC ALL ALL

Inspect (Ultrasonic) the edge frames outer chord under the stop backup fittings at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-22-7 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-10.

AIRPLANE NOTE: For the 737-600 and -700, STA 616 and STA 639.

For the 737-800, STA 578 and STA 639.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.

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ALL





				INTERVAL		APPLICA	BILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-686-00-02	AWL	53-05-02-130-805	1.1	50000 FC	36000 FC	ALL	ALL
	See Doc. D6 The NDI met	asonic) the edge frames 326A001-DTR, DTR chec thod(s) necessary to acco The inspection procedure	k form 53-40-2 omplish the int	22-7 for alternative ent of this inspection	inspections.		ve Test Man
	AIRPLANE I	NOTE : For the 737-600	and -700, STA	616 and STA 639.			
		For the 737-800,	STA 578 and	STA 639.			
	ACCESS NO	OTE: Emergency Exit do cabin sidewall linin	•	•	•	ve or displace passen	ger
53-688-00-01	AWL	53-05-02-250-878	1.1	50000 FC	36000 FC	800 900 900ER	ALL
	and edge fra	h Frequency Eddy Currer imes) at stringer S-14 (fro 326A001-DTR, DTR chec	om STA 578 to k form 53-40-2	STA 607 and from 22-9 for alternative	STA 616 to STA inspections.	,	
	The NDI met	is with external doublers a thod(s) necessary to acco The inspection procedure	omplish the int	ent of this inspection	on is contained ir	n the 737 Nondestruct	
	The NDI met (D6-37239).	thod(s) necessary to acco	omplish the int es are contain	ent of this inspection and in Part 6, Section	on is contained in on 53-11-15.	n the 737 Nondestruct	
53-688-00-02	The NDI met (D6-37239).	thod(s) necessary to according to the inspection procedure.	omplish the int es are contain	ent of this inspection and in Part 6, Section	on is contained in on 53-11-15.	800 900 900ER	

53-690-00-01

AWL

53-05-02-250-880

1.1

50000 FC

36000 FC

800 900 900ER

ALL

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the edge frame at STA 578 (from stringers S-13 to S-15) and at STAs 601, 616 and 639 (from stringers S-10 to S-15).

See Doc. D626A001-DTR, DTR check form 53-40-22-11 for alternative inspections.

NOTE: Doors with external doublers at the lower door corners refer to DTR 53-40-22-22 for area covered by doubler.

53-690-00-02

AWL

53-05-02-250-880

1.1

50000 FC

36000 FC

800 900 900ER

ALL

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the edge frame at STA 578 (from stringers S-13 to S-15) and at STAs 601, 616 and 639 (from stringers S-10 to S-15).

See Doc. D626A001-DTR, DTR check form 53-40-22-11 for alternative inspections.

NOTE: Doors with external doublers at the lower door corners refer to DTR 53-40-22-22 for area covered by doubler.





				APPLICA	BILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGIN		
53-692-00-01	AWL	53-05-02-210-803	1.1	50000 FC	18000 FC	800 900 900ER	ALL		
	578, 601, 610 Note: Fasten	eral Visual) the width and 6 and 639. Per location is three FWD 26A001-DTR, DTR chec	and two AFT a	at STAs 578/616 ar	nd two FWD and				
	ACCESS NO	OTE: Emergency Exit Do cabin sidewall lining	•	•	•	ove or displace passer	nger		
53-692-00-02	AWL	53-05-02-210-803	1.1	50000 FC	18000 FC	800 900 900ER	ALL		
	578, 601, 610 Note: Fasten See Doc. D6	peral Visual) the width and 6 and 639. Per location is three FWD 26A001-DTR, DTR chec DTE: Emergency Exit Do cabin sidewall lining	and two AFT a k form 53-40-2 por must be op	at STAs 578/616 ar 2-13 for alternative en to perform the in	nd two FWD and e inspections. nspection. Remo	three AFT at STAs 60	01/639.		
53-695-00-01	AWL	53-05-02-211-835	1.1	50000 FC	9000 FC	ALL	ALL		
	See Doc. D6	ailed) the door stops atta 26A001-DTR, DTR chec DTE: Emergency Exit Do	k form 53-40-2	3-1 for alternative	inspections.	gs per door.			
53-695-00-02	AWL	AWL 53-05-02-211-835 1.1 50000 FC 9000 FC ALL ALL							
00 000 00 02	See Doc. D6	ailed) the door stops atta 26A001-DTR, DTR chec DTE: Emergency Exit Do	k form 53-40-2	3-1 for alternative	inspections.	gs per door.			
				· 					
53-696-00-01	AWL	53-05-02-211-836	1.1	50000 FC	36000 FC	800 900 900ER	ALL		
53-696-00-01	AWL Inspect (Deta		1.1	50000 FC	36000 FC STA 601 to STA		ALL		
53-696-00-01	AWL Inspect (Deta	53-05-02-211-836 ailed) the door stop interc	1.1 costals (three look form 53-40-2	50000 FC ocations) between 3-2 for alternative en to perform this i	36000 FC STA 601 to STA inspections.	616.			
53-696-00-01	AWL Inspect (Deta	53-05-02-211-836 ailed) the door stop interce 26A001-DTR, DTR chece DTE: Emergency Exit Do	1.1 costals (three look form 53-40-2	50000 FC ocations) between 3-2 for alternative en to perform this i	36000 FC STA 601 to STA inspections.	616.	in		

See Doc. D626A001-DTR, DTR check form 53-40-23-2 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.







				INTERVAL			APPLICABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-697-00-01	AWL	53-05-02-250-884	1.1	50000 FC	36000 FC	600 700 800 900 900ER	ALL			
		Inspect (High Frequency Eddy Current) the window frame edge inboard of the fasteners common to the door stop backup fitting attachment at stringers S-11 and S-12.								
	See Doc. D6	26A001-DTR, DTR ched	ck form 53-40-2	23-3 for alternative	inspections.					
	The NDI met	hod(s) necessary to acc	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	tive Test Man			

AIRPLANE NOTE: For the 737-600 and -700, STA 616.

For the 737-800, STA 578.

ACCESS NOTE: Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform

the inspection.

53-697-00-02 AWL 53-05-02-250-884 1.1 50000 FC 36000 FC 600 700 800 ALL 900 900 PR

Inspect (High Frequency Eddy Current) the window frame edge inboard of the fasteners common to the door stop backup fitting attachment at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-23-3 for alternative inspections.

(D6-37239). The inspection procedures are contained in Part 6. Section 53-11-24.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-24.

AIRPLANE NOTE: For the 737-600 and -700, STA 616.

For the 737-800, STA 578.

ACCESS NOTE: Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform

the inspection.

53-697-10-01 AWL 53-05-02-210-837 1.1 50000 FC 36000 FC ALL ALL

Inspect (General Visual) the angle between the rear spar extension and the pressure deck from the AFT or FWD side, including the bend radius.

Note: Either the AFT or FWD side inspection may be performed.

See Doc. D626A001-DTR, DTR check form 53-40-24-1 for alternative inspections.

53-698-00-01 AWL 53-05-02-211-867 1.1 50000 FC 4000 FC ALL ALL

Inspect (Detailed) the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 727 to STA 887, except at the lap splices and antennas. (PSE 53-60-01-2).

See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.

ACCESS NOTE: Remove Dorsal Fin as required to perform the inspection.

53-699-00-01 AWL 53-05-02-211-837 1.1 50000 FC 36000 FC ALL ALL

Inspect (Detailed) the crown skin panel at the ADF Antenna cutout (STA 727+9, RBL 5) and the SATCOM Antenna cutout (STA 747, stringer S-1).

See Doc. D626A001-DTR. DTR check form 53-60-01-4 for alternative inspections.

ACCESS NOTE: Removal of antenna, fairing and base plate as required to expose the skin to perform the inspection.





				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
53-700-00-01	AWL	53-05-02-250-885	1.1	50000 FC	36000 FC	ALL	ALL		
	Inspect (Low Frequency Eddy Current) the skin for hidden cracks between the adaptor plate and stringers S-1 and S-2L.								
	See Doc. D626A001-DTR, DTR check form 53-60-01-5 for alternative inspections. ACCESS NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to								
	ACCESS NO	perform the inspec		abin ceiling paneis	and air conditioni	ng duct as required	το		
53-701-00-01	AWL	53-05-02-211-838	1.1	50000 FC	18000 FC	ALL	ALL		
		ailed) the skin under the 26A001-DTR, DTR chec				STA 767.			
	ACCESS NO	DTE: Remove or displace perform the inspec		abin ceiling panels	and air conditioni	ng duct as required	to		
53-702-00-01	AWL	53-05-02-250-886	1.1	50000 FC	36000 FC	ALL	ALL		
		Frequency Eddy Currer 26A001-DTR, DTR chec	,			plate and stringers	S-1 and S-2L		
	ACCESS NO	OTE: Remove or displace perform the inspec		abin ceiling panels	and air conditioni	ng duct as required	to		
53-702-01-01	AWL	53-05-02-211-839	1.1	50000 FC	12000 FC	ALL	ALL		
	Inspect (Detailed) the skin under the antenna adaptor plate from stringers S-1 to S-2L at STA 767. See Doc. D626A001-DTR, DTR check form 53-60-01-6 for alternative inspections.								
	ACCESS NO	DTE: Remove or displace perform the inspec		abin ceiling panels	and air conditioni	ng duct as required	to		
53-703-00-01	AWL	53-05-02-210-805	1.1	50000 FC	4000 FC	ALL	ALL		
	Inspect (General Visual) the skin from STA 727 to STA 887 between stringers S-14 to S-17. (PSE 53-60-02-1). See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.								
	ACCESS NO	OTE: Remove or displace	e wing to body	r fairings as require	d to perform the in	nspection.			
53-703-00-02	AWL	53-05-02-210-805	1.1	50000 FC	4000 FC	ALL	ALL		
	Inspect (General Visual) the skin from STA 727 to STA 887 between stringers S-14 to S-17. (PSE 53-60-02-1). See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.								
	ACCESS NO	OTE: Remove or displace	e wing to body	fairings as require	d to perform the in	nspection.			
53-703-10-01	AWL	53-05-02-211-868	1.1	50000 FC	8000 FC	ALL	ALL		
		ailed) the fuselage skin p 26A001-DTR, DTR chec				7 to STA 887. (53-6	0-02-4).		
	ACCESS NO	OTE: Remove or displace	e wing to body	fairings as require	d to perform this i	nspection.			
53-703-10-02	AWL	53-05-02-211-868	1.1	50000 FC	8000 FC	ALL	ALL		

Inspect (Detailed) the fuselage skin panels under the Wing to Body Fairing from \$1A 727 to \$1A 887. (53-60-02-4) See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.





				INTERVAL		APPLICA	BILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
53-704-00-01	AWL	53-05-02-250-940	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-1).										
	See Doc. D6	26A001-DTR, DTR ched	ck form 53-10-	03-1, for alternative	e inspections.						
		thod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	tive Test Manua				
53-704-00-02	AWL	53-05-02-250-940	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
		n Frequency Eddy Curre 887. (PSE 53-60-04-1).	nt) the upper s	kin along the uppe	r fastener row at	stringers S-4L and S	-4R from STA				
		626A001-DTR, DTR chec	ck form 53-10-	03-1, for alternative	e inspections.						
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	tive Test Manua				
53-705-00-01	AWL	53-05-02-250-942	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
	. ,	Frequency Eddy Currer (PSE 53-60-04-2).	nt) the lower sk	in along the lower	fastener row at st	ringers S-4L and S-4	IR from STA 727				
	See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.										
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	tive Test Manual				
53-705-00-02	AWL	53-05-02-250-942	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
		Frequency Eddy Currer (PSE 53-60-04-2).	nt) the lower sk	in along the lower	fastener row at st	ringers S-4L and S-4	IR from STA 727				
		26A001-DTR, DTR ched		•	•						
		thod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	tive Test Manual				
53-706-00-01	AWL	53-05-02-250-944	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (High	n Frequency Eddy Curre	nt) the upper s	kin along the uppe	r fastener row at s	stringers S-10L and S	S-10R from STA				

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-3)

See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.





				INTERVAL		APPLICA	BILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
53-706-00-02	AWL	53-05-02-250-944	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL				
	727 to STA 8	Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from ST/727 to STA 887. (PSE 53-60-04-3) See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.									
	The NDI met	thod(s) necessary to acc The inspection procedur	complish the int	ent of this inspection	on is contained in	the 737 Nondestruc	tive Test Manı				
53-707-00-01	AWL	53-05-02-211-869	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (Deta 53-60-04-4)	ailed) the lower skin alor	ng the lower fas	stener row at string	ers S-10L and S-	10R from STA 727 to	STA 887. (PS				
	See Doc. D6	26A001-DTR, DTR ched	ck form 53-30-0	04-4, for alternative	e inspections.						
	ACCESS NO	OTE: Removal or displa	cement of inter	ior sidewall panels	and insulation bla	ankets are required.					
53-707-00-02	AWL	53-05-02-211-869	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (Detailed) the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-4)										
	See Doc. D6	26A001-DTR, DTR ched	ck form 53-30-0	04-4, for alternative	e inspections.						
	ACCESS NO	OTE: Removal or displa	cement of inter	ior sidewall panels	and insulation bla	ankets are required.					
53-708-00-01	AWL	53-05-02-211-871	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL				
		Inspect (Detailed) the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-5)									
	See Doc. D6	26A001-DTR, DTR ched	ck form 53-30-0	04-5, for alternative	e inspections.						
53-708-00-02	AWL	53-05-02-211-871	1.1	50000 FC	24000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (Deta (PSE 53-60-	ailed) the upper skin alor 04-5)	ng the upper fa	stener row at string	gers S-14L and S	-14R from STA 727 t	o STA 887.				
	See Doc. D6	26A001-DTR, DTR ched	ck form 53-30-0	04-5, for alternative	e inspections.						
53-709-00-01	AWL	53-05-02-250-946	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL				
	Inspect (Low	Frequency Eddy Curre	nt) the lower sk	in along the lower	fastener row at st	ringers S-14L and S	-14R from \$7				

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-6)

See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-709-00-02	AWL	53-05-02-250-946	1.1	50000 FC	18000 FC	600 700 700IGW 800 900 900ER	ALL			
	Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-6) See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.									
53-710-00-01	AWL	53-05-02-250-887	1.1	50000 FC	36000 FC	ALL	ALL			
	from STA 72 See Doc. Do The NDI me	w Frequency Eddy Curre 7 to STA 887, except at 526A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	the cargo door ck form 53-60-0 complish the int	cutout. 04-7, for alternative ent of this inspection	inspections.	-				
53-710-00-02	AWL	53-05-02-250-887	1.1	50000 FC	36000 FC	ALL	ALL			
50.744.00.04	The NDI met (D6-37239).	526A001-DTR, DTR check thod(s) necessary to accomplete inspection procedure.	complish the intres are contain	ent of this inspection ed in Part 6, Section	on is contained in n 53-30-50.					
53-711-00-01	AWL	53-05-02-211-840	1.1	50000 FC	24000 FC	ALL	ALL			
	Inspect (Detailed) the window frames around each window from STA 727 to STA 888. See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.									
	ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.									
53-711-00-02	AWL	53-05-02-211-840	1.1	50000 FC	24000 FC	ALL	ALL			
	Inspect (Detailed) the window frames around each window from STA 727 to STA 888. See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.									
	ACCESS NO	OTE: Removal and/or di insulation blankets	•	passenger cabin si	dewalls or sidewa	all window assembli	es and			
53-711-01-01	AWL	53-05-02-211-841	1.1	50000 FC	4000 FC	ALL	ALL			
		ailed) the window frames 26A001-DTR, DTR chec								
53-711-01-02	AWL	53-05-02-211-841	1.1	50000 FC	4000 FC	ALL	ALL			

Inspect (Detailed) the window frames around each window from STA 727 to STA 888. See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-712-00-01	AWL	53-05-02-250-948	1.1	50000 FC	36000 FC	ALL	ALL
	807 and STA See Doc. Do The NDI me	r Frequency Eddy Currer A 827. (PSE 53-60-08). G26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-30-lomplish the in	08-12, for alternativ	ve inspections. on is contained in		
	ACCESS NO	OTE: Removal of aft care	go door scuff p	plate is required to	perform the inspec	ction.	
53-713-00-01	AWL	53-05-02-250-949	1.1	50000 FC	24000 FC	ALL	ALL
	(PSE 53-60- See Doc. D6	h Frequency Eddy Currer 08). 626A001-DTR, DTR chec DTE: Remove or displac	ck form 53-30-	08-10, for alternativ	ve inspections.		d and bearstrap.
53-713-01-01	AWL	53-05-02-250-974	1.1	50000 FC	24000 FC	ALL	ALL
	See Doc. Do	h Frequency Eddy Currer 326A001-DTR, DTR chec DTE: Remove or displac	ck form 53-30-	08-10, for alternativ	ve inspections.		3).
53-714-00-01	AWL	53-05-02-250-950	1.1	50000 FC	9000 FC	ALL	ALL
	847. (PSE 5	h Frequency Eddy Currer 3-60-08) 26A001-DTR, DTR chect				nger S-24R at STA	794.4 and STA
53-714-01-01	AWL	53-05-02-130-809	1.1	50000 FC	9000 FC	ALL	ALL
	847. (PSE 5	asonic) the bearstrap for 3-60-08) 26A001-DTR, DTR checl	·			inger S-24R at STA	794.4 and STA
53-715-00-01	AWL	53-05-02-250-888	1.1	50000 FC	18000 FC	ALL	ALL
	S-26R. See Doc D6 The NDI me (D6-37239).	h Frequency Eddy Current 26A001-DTR, DTR check thod(s) necessary to acc The inspection procedur DTE: Cargo Door must b	k form 53-60-0 omplish the in es are contain	08-1 for alternative tent of this inspecti ed in Part 6, Subje	inspections. on is contained in ect 53-10-88.	·	
53-716-00-01	AWL	53-05-02-250-889	1.1	50000 FC	18000 FC	ALL	ALL
	Inspect (Hig	h Frequency Eddy Curre	nt) the forward	and aft edge fram	e inner chord fail-	safe straps betweer	n stringers S-17R

Inspect (High Frequency Eddy Current) the forward and aft edge frame inner chord fail-safe straps between stringers S-17R and S-18R.

See Doc D626A001-DTR, DTR check form 53-60-08-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-88.

ACCESS NOTE: Cargo Door must be open to perform the inspection. Remove or displace aft cargo sidewall and ceiling lining as required to perform the inspection.





TASK CARD NO.			INTERVAL		APPLICA	ABILITY	
IASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-717-00-01	AWL	53-05-02-250-890	1.1	50000 FC	18000 FC	ALL	ALL
		Frequency Eddy Currer	,	•		s at stringer S-18R.	
		26A001-DTR, DTR check			•	707 N	
		thod(s) necessary to acc The inspection procedur	•			the 737 Nondestruc	ctive Test Manua
	ACCESS NO	OTE: Remove or displac	e aft cargo sid	ewall and ceiling lir	ning as required to	o perform the inspec	ction.
53-718-00-01	AWL	53-05-02-250-891	1.1	50000 FC	18000 FC	ALL	ALL
	S-26R at ST	h Frequency Eddy Curre	except at door	stops and sills locat	tion).	ord between stringe	rs S-16R and
		26A001-DTR, DTR chec			•	the 207 New death	To at Man
		thod(s) necessary to acc The inspection procedur		•		the 737 Nondestruc	ctive Test Manu
	ACCESS NO	OTE: Aft cargo door mus	st be open to p	erform the inspection	on.		
53-719-00-01	AWL	53-05-02-130-806	1.1	50000 FC	18000 FC	ALL	ALL
	Inspect (Ultr	asonic) the outboard por	tion of the fram	e web under the do	oor stop fittings a	nd sill clips at STA 7	94.37 and STA
	847.						
	See Doc D6	26A001-DTR, DTR chec	k form 53-60-0	8-5 for alternative i	nspections.		
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	ctive Test Manu
	ACCESS NO	TE. Aft source door mile					
		required to perform		•	on. Remove or di	splace aft cargo linir	ng as
53-720-00-01	AWL	•		•	on. Remove or di 9000 FC	splace aft cargo linir	ng as
53-720-00-01	AWL Inspect (Hig	required to perform 53-05-02-250-892 h Frequency Eddy Curre	1.1 nt) the exposed	50000 FC	9000 FC	ALL	ALL
53-720-00-01	AWL Inspect (Hig STA 794.4 a	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer	1.1 nt) the exposers S-18R to S-2	50000 FC d edge of the bears	9000 FC strap at both the f	ALL	ALL
53-720-00-01	AWL Inspect (Hig STA 794.4 a See Doc D6	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer 26A001-DTR, DTR chec	1.1 nt) the exposers S-18R to S-2k form 53-60-0	50000 FC d edge of the bears 25R. 8-8 for alternative i	9000 FC strap at both the formspections.	ALL orward and aft edge	ALL of the door at
53-720-00-01	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the int	50000 FC d edge of the bears 25R. 8-8 for alternative in	9000 FC strap at both the formspections.	ALL orward and aft edge	ALL of the door at
	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239).	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringel 26A001-DTR, DTR chec thod(s) necessary to acc	1.1 nt) the expose rs S-18R to S-2 k form 53-60-0 omplish the interest are contained.	50000 FC d edge of the bears 25R. 8-8 for alternative in	9000 FC strap at both the formula of	ALL orward and aft edge the 737 Nondestruc	ALL of the door at ctive Test Manu
53-720-00-01	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer 26A001-DTR, DTR check thod(s) necessary to acc The inspection procedur 53-05-02-211-842 ailed) the skin at all four	1.1 nt) the exposer rs S-18R to S-2 k form 53-60-0 omplish the interest are contained. 1.1 corners (upper	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspectic ed in Part 6, Sectio 34000 FC	9000 FC strap at both the formspections. on is contained in 153-11-21. 18000 FC of the cargo door	ALL orward and aft edge the 737 Nondestruct ALL	ALL of the door at
	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det See Doc. D6	required to perform 53-05-02-250-892 The Frequency Eddy Currend STA 847 from stringer 26A001-DTR, DTR check thod(s) necessary to accommodity the inspection procedur 53-05-02-211-842 ailed) the skin at all four accommodity the skin at all four accommodity the skin at all four accommodity to the skin accommodity to the skin accommodity to the skin accommodity to the skin accommodity t	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the interes are contained. 1.1 corners (upper ck form 53-60-0	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspectic ed in Part 6, Sectio 34000 FC	9000 FC strap at both the formspections. on is contained in 153-11-21. 18000 FC of the cargo door	ALL orward and aft edge the 737 Nondestruct ALL	ALL of the door at ctive Test Manu
	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det See Doc. D6	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer 26A001-DTR, DTR check thod(s) necessary to acc The inspection procedur 53-05-02-211-842 ailed) the skin at all four	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the interes are contained. 1.1 corners (upper ck form 53-60-0	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspectic ed in Part 6, Sectio 34000 FC	9000 FC strap at both the formspections. on is contained in 153-11-21. 18000 FC of the cargo door	ALL orward and aft edge the 737 Nondestruct ALL	ALL of the door at ctive Test Manu
	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det See Doc. D6	required to perform 53-05-02-250-892 The Frequency Eddy Currend STA 847 from stringer 26A001-DTR, DTR check thod(s) necessary to accommodity the inspection procedur 53-05-02-211-842 ailed) the skin at all four accommodity the skin at all four accommodity the skin at all four accommodity to the skin accommodity to the skin accommodity to the skin accommodity to the skin accommodity t	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the interes are contained. 1.1 corners (upper ck form 53-60-0	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspectic ed in Part 6, Sectio 34000 FC	9000 FC strap at both the formspections. on is contained in 153-11-21. 18000 FC of the cargo door	ALL orward and aft edge the 737 Nondestruct ALL	ALL of the door at ctive Test Manu
53-721-00-01	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det See Doc. D6 ACCESS NG	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer 26A001-DTR, DTR check thod(s) necessary to accommodity 53-05-02-211-842 ailed) the skin at all four 626A001-DTR, DTR check DTE: Scuff plate removal	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the interes are contained. 1.1 corners (upperck form 53-60-0 all required. 1.1 p at all four columns	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspecticed in Part 6, Section 34000 FC //lower/FWD/AFT) co 08-9, for alternative 34000 FC mers (upper/lower/fi	9000 FC strap at both the formspections. on is contained in n 53-11-21. 18000 FC of the cargo door inspections. 18000 FC fwd/aft) of the car	ALL orward and aft edge the 737 Nondestruct ALL cutout.	ALL of the door at etive Test Manual
53-721-00-01	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det See Doc. D6 ACCESS NG AWL Inspect (Ger See Doc. D6	required to perform 53-05-02-250-892 The Frequency Eddy Currend STA 847 from stringer 26A001-DTR, DTR chected thod(s) necessary to accommodity to accommodity the inspection procedur 53-05-02-211-842 ailed) the skin at all four accommodity to accommodity the skin at all four accommodity the skin at all four accommodity to accommodity the skin at all four accommodity to accommodity the skin at all four accommodity to accommodity the skin at all four accommodity the skin at all four accommodity to accommodity the skin at all four accommodity the skin	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the interest are contained. 1.1 corners (upper sk form 53-60-0 lrequired. 1.1 p at all four cock form 53-60-0 cock	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspectic ed in Part 6, Sectio 34000 FC //lower/FWD/AFT) c 18-9, for alternative 34000 FC mers (upper/lower/f 18-9, for alternative	9000 FC strap at both the formspections. on is contained in n 53-11-21. 18000 FC of the cargo door inspections. 18000 FC fwd/aft) of the car	ALL orward and aft edge the 737 Nondestruct ALL cutout.	ALL of the door at ctive Test Manu
53-721-00-01	AWL Inspect (Hig STA 794.4 a See Doc D6 The NDI me (D6-37239). AWL Inspect (Det See Doc. D6 ACCESS NG AWL Inspect (Ger See Doc. D6	required to perform 53-05-02-250-892 h Frequency Eddy Curre nd STA 847 from stringer 26A001-DTR, DTR check thod(s) necessary to accommodition 53-05-02-211-842 ailed) the skin at all four 626A001-DTR, DTR check DTE: Scuff plate removation 53-05-02-210-804 neral Visual) the bearstral 626A001-DTR, DTR check DTE: DTR check DTE: Scuff plate removation	1.1 nt) the exposers S-18R to S-2 k form 53-60-0 omplish the interest are contained. 1.1 corners (upper sk form 53-60-0 lrequired. 1.1 p at all four cock form 53-60-0 cock	50000 FC d edge of the bears 25R. 8-8 for alternative is ent of this inspectic ed in Part 6, Sectio 34000 FC //lower/FWD/AFT) c 18-9, for alternative 34000 FC mers (upper/lower/f 18-9, for alternative	9000 FC strap at both the formspections. on is contained in n 53-11-21. 18000 FC of the cargo door inspections. 18000 FC fwd/aft) of the car	ALL orward and aft edge the 737 Nondestruct ALL cutout.	ALL of the door at

Inspect (DET) the upper sill inner chord.

See Doc. D626A001-DTR, DTR check form 53-60-08-11, for alternative inspections.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-722-00-01	AWL	53-05-02-250-893	1.1	50000 FC	9000 FC	ALL	ALL
	stops location See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre ons on both the forward a \$26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Aft cargo door must	nd aft edge frack form 53-60-6 omplish the intest are contain	mes. 09-3, for alternative ent of this inspection ed in Part 6, Subject	inspections. on is contained in ct 53-10-82.	the 737 Nondestruc	ctive Test Manu
	AUULUU IK	lining as required t			on. Remove of di	spiace are earge sid	owan
53-723-00-01	AWL	53-05-02-250-951	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL
	887 to STA 1 See Doc. D6 The NDI me	n Frequency Eddy Curre 1016. (PSE 53-70-03-1). 326A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-10-0 omplish the int	03-1, for alternative ent of this inspection	inspections. on is contained in	Ü	
53-723-00-02	AWL	53-05-02-250-951	1.1	50000 FC	36000 FC	600 700 700IGW 800 900 900ER	ALL
	887 to STA 1 See Doc. D6 The NDI me	h Frequency Eddy Curre 1016. (PSE 53-70-03-1). 326A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 53-10-0 omplish the int	03-1, for alternative ent of this inspection	inspections.	Ü	
53-724-00-01	AWL	53-05-02-250-894	1.1	50000 FC	18000 FC	ALL	ALL
	to STA 1016 See Doc. Do The NDI me	r Frequency Eddy Currer . 626A001-DTR, DTR check thod(s) necessary to acc The inspection procedur	ck form 53-70-0 omplish the int	03-2, for alternative ent of this inspection	inspections. on is contained in		
53-724-00-02	AWL	53-05-02-250-894	1.1	50000 FC	18000 FC	ALL	ALL
	Inspect (Low to STA 1016 See Doc. D6 The NDI me	/ Frequency Eddy Currer	nt) the lower sk ck form 53-70-0 omplish the int	in along the lower 03-2, for alternative ent of this inspection	fastener row at st inspections. on is contained in	ringers S-4L and S-	4R from STA 88
53-725-00-01	AWL	53-05-02-211-845	1.1	50000 FC	9000 FC	ALL	ALL
	Inspect (Deta	ailed) the upper skin alor 1016) and at stringer S-	14R (from STA	stener row at string 888 to STA 947, a	ger S-14L (from S nd from STA 996	TA 888 to STA 947,	

See Doc D626A001-DTR, DTR check form 53-70-03-3 for alternative inspections.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-725-00-02	AWL	53-05-02-211-845	1.1	50000 FC	9000 FC	ALL	ALL
	1006 to STA	ailed) the upper skin alor a 1016) and at stringer S- 26A001-DTR, DTR chec	14R (from STA	888 to STA 947, ai	nd from STA 996		and from STA
53-726-00-01	AWL	53-05-02-250-895	1.1	50000 FC	36000 FC	ALL	ALL
	947 and from See Doc D6 The NDI me	v Frequency Eddy Currer m STA 1006 to STA 1016 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur) and at stringe k form 53-70-0 complish the int	er S-14R (from STA 3-4 for alternative in ent of this inspection	888 to STA 947 anspections. On is contained in	and from STA 996 to	STA 1016).
53-726-00-02	AWL	53-05-02-250-895	1.1	50000 FC	36000 FC	ALL	ALL
	See Doc D6 The NDI me	m STA 1006 to STA 1016 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	k form 53-70-0 complish the int	3-4 for alternative in tent of this inspection	nspections. on is contained in		
53-727-00-01	AWL	53-05-02-211-873	1.1	50000 FC	4000 FC	ALL	ALL
	except at the See Doc D6	ailed) the skin around all e lap splices and antenna 26A001-DTR, DTR chec OTE: Remove Dorsal Fi	as. (PSE 53-70 k form 53-30-0	-04-1). 1-2 for alternative i	nspections.	10R, from STA 887 t	o STA 1016,
53-728-00-01	AWL	53-05-02-250-896	1.1	50000 FC	9000 FC	ALL	ALL
	See Doc D6 The NDI me (D6-37239).	h Frequency Eddy Curre 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur OTE: Remove or displace	k form 53-70-0 complish the int es are contain	7-3 for alternative interest of this inspection and in Part 6, Subjection	nspections. on is contained in ot 53-30-35.	the 737 Nondestruc	ctive Test Manua
53-729-00-01	AWL	53-05-02-250-897	1.1	50000 FC	36000 FC	ALL	ALL
	the web and See Doc D6	h Frequency Eddy Curre I doublers at the #1, #2, # 26A001-DTR, DTR chec thod(s) necessary to acc	#6 and #7 stop k form 53-70-0	locations. 7-4 for alternative in	nspections. on is contained in		
	(D6-37239).	The inspection procedur OTE: Remove or displace				rform the inspection	
53-729-01-01	(D6-37239).	The inspection procedur				rform the inspection	

Inspect (General Visual) the door stop intercostals along the inner chord and around fasteners common to the web and doublers at the #1, #2, #6 and #7 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-07-4 for alternative inspections.

ACCESS NOTE: Remove or displace interior sidewall and door lining as required to perform the inspection.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-730-00-01	AWL	53-05-02-211-846	1.1	50000 FC	4000 FC	ALL	ALL
	Inspect (Deta	ailed) the perimeter of th	e cutout and a	round the fasteners	s common to the	edge frames and up	per sill outer
		26A001-DTR, DTR chec	k form 53-70-0	7-5 for alternative i	nspections.		
	ACCESS NO	OTE: Remove or displace	ce interior side	wall and door lining	as required to pe	rform the inspection	١.
53-731-00-01	AWL	53-05-02-211-847	1.1	50000 FC	4000 FC	ALL	ALL
		ailed) the skin around th 26A001-DTR, DTR chec	•	•	nspections.		
53-731-01-01	AWL	53-05-02-250-898	1.1	50000 FC	18000 FC	ALL	ALL
		Frequency Eddy Curre				e scuff plate.	
		26A001-DTR, DTR chec hod(s) necessary to acc			•	the 727 Nondoctrue	stive Test Mar
		The inspection procedu				the 737 Nondestruc	clive rest ivial
	ACCESS NO	OTE: Remove scuff plat	e.				
53-732-00-01	AWL	53-05-02-211-848	1.1	50000 FC	22000 FC	ALL	ALL
		ailed) the inner chord an 26A001-DTR, DTR chec				ΓA 1006.	
	ACCESS NO	DTE: Remove or displace inspection.	ce passenger c	abin sidewall and c	ceiling lining as re	quired to perform th	е
53-732-01-01	AWL	53-05-02-211-981	1.1	50000 FC	22000 FC	ALL	ALL
		ailed) the inner chord str 26A001-DTR, DTR chec				6.	
		OTE: Remove or displace inspection.			•	quired to perform th	е
53-732-10-01	AWL	53-05-02-250-997	1.1	50000 FC	36000 FC	ALL	ALL
	Inspect (High	n Frequency Eddy Curre	ent) the first five	fasteners on the u	pper flange of the	e lower main sill out	er chord, aft c
	•	26A001-DTR, DTR chec	k form 53-70-0	7-12 for alternative	inspections.		
		hod(s) necessary to acc The inspection procedu				the 737 Nondestruc	ctive Test Mar
	ACCESS NO	OTE: Remove scuff plat	es as required	for access to the o	uter chord.		
53-733-00-01	AWL	53-05-02-211-874	1.1	50000 FC	22000 FC	ALL	ALL
	Inspect (Deta	ailed) the inner chord an	d web along th	e unner main sill fr	om STA 951 to ST	TA 1006 (PSE 53-7)	0-08)

Inspect (Detailed) the inner chord and web along the upper main sill from STA 951 to STA 1006. (PSE 53-70-08). See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-733-01-01	AWL	53-05-02-211-982	1.1	50000 FC	22000 FC	ALL	ALL
	See Doc D62	ailed) the inner chord str 26A001-DTR, DTR chec DTE: Remove or displac	k form 53-70-0	7-11 for alternative	inspections.	,	e
		inspection.					
53-734-00-01	AWL	53-05-02-250-899	1.1	50000 FC	4000 FC	ALL	ALL
	See Doc D62 The NDI met	Frequency Eddy Currer 26A001-DTR, DTR chec hod(s) necessary to acc The inspection procedur	k form 53-70-0 complish the int	8-3 for alternative i ent of this inspection	nspections. on is contained in	·	ctive Test Mar
	ACCESS NO	OTE: Remove or displace inspection.	ce passenger c	abin sidewall and c	eiling lining as red	quired to perform the	е
53-734-01-01	AWL	53-05-02-250-900	1.1	50000 FC	12000 FC	ALL	ALL
	400500 110						
	ACCESS NO	OTE: Remove or displace inspection.	ce passenger c	abin sidewall and c	eiling lining as red	quired to perform the	e
53-735-00-01	AWL		ce passenger c	abin sidewall and c	eeiling lining as red	quired to perform the	e ALL
53-735-00-01	AWL Inspect (High the web and See Doc D62 The NDI met (D6-37239).	inspection.	1.1 nt) the door sto 5, and #6 stop k form 53-70-0 complish the int	50000 FC op intercostals alon locations. 8-4 for alternative i ent of this inspectic ed in Part 6, Subject	36000 FC g the inner chord nspections. on is contained in ct 53-10-84.	ALL and around fastene the 737 Nondestruc	ALL ers common to ctive Test Man
53-735-00-01 53-735-01-01	AWL Inspect (High the web and See Doc D62 The NDI met (D6-37239).	inspection. 53-05-02-250-901 Frequency Eddy Curre doubler at the #1, #2, #826A001-DTR, DTR checkhod(s) necessary to accomb inspection procedur DTE: Remove or displace	1.1 nt) the door sto 5, and #6 stop k form 53-70-0 complish the int	50000 FC op intercostals alon locations. 8-4 for alternative i ent of this inspectic ed in Part 6, Subject	36000 FC g the inner chord nspections. on is contained in ct 53-10-84.	ALL and around fastene the 737 Nondestruc	ALL ers common to ctive Test Man
	AWL Inspect (High the web and See Doc D62 The NDI met (D6-37239). ACCESS NO AWL Inspect (Gen doubler at the	inspection. 53-05-02-250-901 Frequency Eddy Curre doubler at the #1, #2, #526A001-DTR, DTR checkhod(s) necessary to accomplete inspection procedur DTE: Remove or displace inspection.	1.1 nt) the door sto 5, and #6 stop k form 53-70-0 complish the interes are contained per passenger of 1.1 p intercostals as a polocations.	50000 FC op intercostals alon locations. 8-4 for alternative i ent of this inspecticed in Part 6, Subject abin sidewall and compared to the subject along the inner choice.	36000 FC g the inner chord nspections. on is contained in ot 53-10-84. reiling lining as receiting lining lining as receiting lining as receiting lining linin	ALL and around fastene the 737 Nondestruc quired to perform the	ALL ers common to extive Test Man
	AWL Inspect (High the web and See Doc D62 The NDI met (D6-37239). ACCESS NO AWL Inspect (Gen doubler at the See Doc D62	inspection. 53-05-02-250-901 Frequency Eddy Curre doubler at the #1, #2, #26A001-DTR, DTR checkhod(s) necessary to accomplete the procedure of the inspection procedure inspection. 53-05-02-210-843 Derail Visual) the door stole #1, #2, #5, and #6 stole	1.1 nt) the door sto 5, and #6 stop k form 53-70-0 complish the interes are contained passenger contained to the passenger contained passenger contained to the passenger contained passenger contained to the pa	50000 FC op intercostals alon locations. 8-4 for alternative is ent of this inspecticed in Part 6, Subject abin sidewall and company of the second of the se	36000 FC g the inner chord nspections. on is contained in ct 53-10-84. reiling lining as receiling lining	ALL and around fastene the 737 Nondestructured to perform the ALL teners common to t	ALL ers common to ctive Test Man e ALL he web and
	AWL Inspect (High the web and See Doc D62 The NDI met (D6-37239). ACCESS NO AWL Inspect (Gen doubler at the See Doc D62	inspection. 53-05-02-250-901 Frequency Eddy Curre doubler at the #1, #2, #826A001-DTR, DTR checkhod(s) necessary to accomply the inspection procedure inspection. 53-05-02-210-843 Deeral Visual) the door stoe #1, #2, #5, and #6 stop 26A001-DTR, DTR checknot inspection.	1.1 nt) the door sto 5, and #6 stop k form 53-70-0 complish the interes are contained passenger contained to the passenger contained passenger contained to the passenger contained passenger contained to the pa	50000 FC op intercostals alon locations. 8-4 for alternative is ent of this inspecticed in Part 6, Subject abin sidewall and company of the second of the se	36000 FC g the inner chord nspections. on is contained in ct 53-10-84. reiling lining as receiling lining	ALL and around fastene the 737 Nondestructured to perform the ALL teners common to t	ALL ers common to extive Test Man e ALL he web and

Inspect (Detailed) the perimeter of the cutout and around the fasteners common to the edge frames and upper sill outer chords.

See Doc D626A001-DTR, DTR check form 53-70-08-5 for alternative inspections.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-737-00-01	AWL	53-05-02-211-875	1.1	50000 FC	4000 FC	ALL	ALL
		ailed) the skin around the	-				
	See Doc D62	26A001-DTR, DTR chec	k form 53-70-0	7-6 for alternative i	nspections.		
53-737-01-01	AWL	53-05-02-250-953	1.1	50000 FC	18000 FC	ALL	ALL
		n Frequency Eddy Curre 26A001-DTR, DTR chec	,		•	e scuff plate. (PSE s	53-70-08-6).
	ACCESS NO	OTE: Remove scuff plate	Э.				
53-737-10-01	AWL	53-05-02-250-A79	1.1	50000 FC	36000 FC	ALL	ALL
		Frequency Eddy Curre	nt) the first five	fasteners on the u	pper flange of the	lower main sill oute	er chord, aft of
	_	me. (53-70-08-12).	le forma E2 70 0	7 10 for alternative	inanaatiana		
		26A001-DTR, DTR chec hod(s) necessary to acc			•	the 737 Nondestruc	ctive Test Mani
		The inspection procedur		•			
	ACCESS NO	OTE: Remove scuff plate	es as required	for access to the or	uter chord.		
53-738-00-01	AWL	53-05-02-211-876	1.1	50000 FC	24000 FC	ALL	ALL
		ailed) the window frames				PSE 53-70-09)	
		26A001-DTR, DTR chec			·		
	ACCESS NO	OTE: Removal and/or di insulation blankets		passenger cabin si	dewalls or sidewa	all window assembli	es and
		modiation blankets	ao roquirou.				
53-738-00-02	AWL	53-05-02-211-876	1.1	50000 FC	24000 FC	ALL	ALL
		ailed) the window frames			,	°SE 53-70-09)	
	3ee D00 D02	26A001-DTR, DTR chec	K 101111 55-60-0				
	ACCECC NO	TE. Domoval and/or di			·	المصمومين الم	aa and
	ACCESS NO	OTE: Removal and/or di insulation blankets	splacement of		·	all window assembli	es and
	ACCESS NO		splacement of		·	all window assembli	es and
53-738-01-01	ACCESS NO		splacement of		·	all window assembli	es and
53-738-01-01	AWL Inspect (Deta	insulation blankets 53-05-02-211-877 ailed) the window frames	splacement of as required. 1.1 s around each v	passenger cabin si 50000 FC window from STA 8	dewalls or sideward 4000 FC 88 to STA 927. (F	ALL	
53-738-01-01	AWL Inspect (Deta	insulation blankets 53-05-02-211-877	splacement of as required. 1.1 s around each v	passenger cabin si 50000 FC window from STA 8	dewalls or sideward 4000 FC 88 to STA 927. (F	ALL	
53-738-01-01 53-738-01-02	AWL Inspect (Deta	insulation blankets 53-05-02-211-877 ailed) the window frames	splacement of as required. 1.1 s around each v	passenger cabin si 50000 FC window from STA 8	dewalls or sideward 4000 FC 88 to STA 927. (F	ALL	
	AWL Inspect (Deta See Doc D62 AWL Inspect (Deta	insulation blankets 53-05-02-211-877 ailed) the window frames 26A001-DTR, DTR chec	splacement of as required. 1.1 s around each v k form 53-60-0	50000 FC window from STA 8 5-2 for alternative i 50000 FC window from STA 8	4000 FC 888 to STA 927. (Finspections. 4000 FC	ALL PSE 53-70-09)	ALL
	AWL Inspect (Deta See Doc D62 AWL Inspect (Deta	insulation blankets 53-05-02-211-877 ailed) the window frames 26A001-DTR, DTR chec 53-05-02-211-877 ailed) the window frames	splacement of as required. 1.1 s around each v k form 53-60-0	50000 FC window from STA 8 5-2 for alternative i 50000 FC window from STA 8	4000 FC 888 to STA 927. (Finspections. 4000 FC	ALL PSE 53-70-09)	ALL

Inspect (Detailed) the pressure dome webs between the stiffeners and tear straps. See Doc D626A001-DTR, DTR check form 53-80-01-2 for alternative inspections.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-740-00-01	AWL	53-05-02-211-851	1.1	50000 FC	9000 FC	ALL	ALL
		ailed) the pressure dome 26A001-DTR, DTR check		-	-	o the radial stiffene	rs.
	AIRPLANE	NOTE: Applicable to all I	models except	-800FPB and -900	ER.		
53-741-00-01	AWL	53-05-02-250-902	1.1	50000 FC	36000 FC	ALL	ALL
	web to the p stiffener. See Doc D6 The NDI me	h Frequency Eddy Currer ressure chord. Inspect at 26A001-DTR, DTR check thod(s) necessary to according to the characteristic of the	the edge of each of the edge of each of the edge of each of the edge of the ed	ach stiffener/clip and 1-4 for alternative in ent of this inspection	nspections.	fasteners on each s	side of the
	AIRPLANE	NOTE: For L/N 721 and					
	ACCESS NO	stringers S-1 and all models excep OTE: Remove necessary	t -800FPB and				
53-742-00-01	ACCESS NO	all models excep	t -800FPB and	-900ER.			
53-742-00-01	AWL Inspect (Higher pressure choose Doc D6 The NDI me	all models excep OTE: Remove necessary	1.1 1.1 1.1 1.1 1.1 1.1 1.1 1.1	-900ER. bin and aft cargo in 50000 FC e dome webs along from stringers S-5L 1-5A for alternative ent of this inspection	12000 FC g the aft fastener r to S-7L and S-5F inspections. on is contained in	ALL row attaching the was to S-9R.	ALL eb to the
53-742-00-01	AWL Inspect (Hig pressure che See Doc D6 The NDI me (D6-37239).	all models excep OTE: Remove necessary 53-05-02-250-903 h Frequency Eddy Currer ord and between the stiffe 26A001-DTR, DTR check thod(s) necessary to according	t -800FPB and y passenger ca 1.1 ht) the pressure ener locations for form 53-80-0 complish the intelligence are contained	-900ER. bin and aft cargo in 50000 FC e dome webs along from stringers S-5L 1-5A for alternative ent of this inspectic ed in Part 6, Sectio	12000 FC g the aft fastener r to S-7L and S-5F inspections. on is contained in n 53-11-38.	ALL row attaching the was to S-9R.	ALL eb to the
53-742-00-01	AWL Inspect (Higher pressure choose Doc D6 The NDI me (D6-37239). AIRPLANE	all models excep DTE: Remove necessary 53-05-02-250-903 h Frequency Eddy Currer ord and between the stiffe 26A001-DTR, DTR check thod(s) necessary to according	1.1 nt) the pressure ener locations form 53-80-0 complish the integer are contained models except	-900ER. bin and aft cargo in 50000 FC e dome webs along from stringers S-5L 1-5A for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900	12000 FC g the aft fastener reto S-7L and S-5F inspections. on is contained in 53-11-38.	ALL ow attaching the was to S-9R. the 737 Nondestructure.	ALL eb to the
53-742-00-01	AWL Inspect (Higher pressure choose Doc D6 The NDI me (D6-37239). AIRPLANE	all models excep OTE: Remove necessary 53-05-02-250-903 The Frequency Eddy Currer ord and between the stiffer 26A001-DTR, DTR check thod(s) necessary to accompliance to the inspection procedure NOTE: Applicable to all in	1.1 nt) the pressure ener locations form 53-80-0 complish the integer are contained models except	-900ER. bin and aft cargo in 50000 FC e dome webs along from stringers S-5L 1-5A for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900	12000 FC g the aft fastener reto S-7L and S-5F inspections. on is contained in 53-11-38.	ALL ow attaching the was to S-9R. the 737 Nondestructure.	ALL eb to the
	AWL Inspect (Hig pressure che See Doc D6 The NDI me (D6-37239). AIRPLANE ACCESS NO	all models excep OTE: Remove necessary 53-05-02-250-903 h Frequency Eddy Currer ord and between the stiffed 26A001-DTR, DTR check thod(s) necessary to accompliance The inspection procedure NOTE: Applicable to all in OTE: Remove necessary	1.1 1.1 1.1 1.1 1.1 1.1 1.1 1.1	-900ER. bin and aft cargo in 50000 FC e dome webs along from stringers S-5L 1-5A for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as required 50000 FC e dome webs along putside of stringers	12000 FC g the aft fastener r to S-7L and S-5F inspections. on is contained in a n 53-11-38. DER. guired to perform t 24000 FC g the aft fastener r S-5L to S-7L and	ALL Ow attaching the was to S-9R. the 737 Nondestructure the inspection. ALL ow attaching the was the inspection.	ALL eb to the ctive Test Manu

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform this inspection.

53-744-00-01 AWL 53-05-02-250-905 50000 FC 24000 FC ALL ALL 1.1

> Inspect (Low Frequency Eddy Current) around the fasteners common to the pressure chord splices between stringers S-2 and S-3 and S-16 and S-17A.

See Doc D626A001-DTR, DTR check form 53-80-01-6 for alternative inspections.

AIRPLANE NOTE: For L/N 1057 and on, this inspection applies only to the splices between stringer S-16 and

S-17A. Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.





TASK CARD NO. 53-745-00-01				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-745-00-01	AWL	53-05-02-250-906	1.1	50000 FC	9000 FC	ALL	ALL
	. ,	Frequency Eddy Curren	nt) the forward	side of the pressur	e dome web arour	nd the fasteners co	mmon to the lap
	splice and th	ie stiffeners. 26A001-DTR, DTR check	k form 53-80-0	1-7 for alternative i	nspections		
		NOTE: Applicable to all			·		
		OTE: Remove necessary	•			ne inspection	
	ACCECC NO	JIE. Remove necessary	y passenger ee	biii iiitoriora da rec	funca to perioriii t	пе тврестоп.	
53-746-00-01	AWL	53-05-02-250-907	1.1	50000 FC	24000 FC	ALL	ALL
		h Frequency Eddy Currer	nt) the forward	side of the pressur	re dome web arou	nd the fasteners co	mmon to the
	tear strap. See Doc D62	26A001-DTR, DTR check	k form 53-80-0	1-8 for alternative i	nspections.		
	The NDI met	thod(s) necessary to according to the inspection procedure.	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Manua
	,	NOTE: Applicable to all					
		OTE: Remove necessary	•			no inspection	
	ACCESS NO	JIE. Remove necessary	y passenger ca	bill illeriors as rec	quired to perioriti t	ne inspection.	
53-747-00-01	AWL	53-05-02-250-908	1.1	50000 FC	9000 FC	ALL	ALL
	Inspect (Low	Frequency Eddy Curren	nt) the aft side	of the pressure dor	ne web at the inte	rsection of the tear	straps and lap
	•	the stiffeners.					
		26A001-DTR, DTR check			•	No. 707 No. o do otro	T4 NA
		thod(s) necessary to according The inspection procedure		'		ine 737 Nondestruc	ctive rest Manua
	,	NOTE: Applicable to all					
	,		modele except)FR		
53-748-00-01				ooon i B and ooo	JEK.		
	AWL	53-05-02-250-909	1.1	50000 FC	24000 FC	ALL	ALL
	Inspect (High	h Frequency Eddy Currer	nt) the forward	50000 FC	24000 FC		
	Inspect (High stiffeners/lap	h Frequency Eddy Currer splices and the tear stra	nt) the forward	50000 FC side of the pressur	24000 FC re dome webs at t		
	Inspect (High stiffeners/lap See Doc D62	n Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check	nt) the forward aps. k form 53-80-0	50000 FC side of the pressur 1-11 for alternative	24000 FC re dome webs at the inspections.	ne junction of the ra	adial
	Inspect (High stiffeners/lap See Doc D62 The NDI met	h Frequency Eddy Currer splices and the tear stra	nt) the forward aps. k form 53-80-0 omplish the int	50000 FC side of the pressur 1-11 for alternative ent of this inspection	24000 FC re dome webs at the inspections.	ne junction of the ra	adial
	Inspect (High stiffeners/lap See Doc D62 The NDI met (D6-37239).	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to acco	nt) the forward aps. k form 53-80-0 omplish the int es are containe	50000 FC side of the pressur 1-11 for alternative ent of this inspection and in Part 6, Section	24000 FC re dome webs at the inspections. on is contained in the on 53-11-07.	ne junction of the ra	adial
	Inspect (High stiffeners/lap See Doc D67 The NDI met (D6-37239).	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to according The inspection procedure	nt) the forward aps. k form 53-80-0 omplish the int es are containe models except	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900	24000 FC re dome webs at the inspections. on is contained in the first 53-11-07.	he junction of the ra	adial
53-749-00-01	Inspect (High stiffeners/lap See Doc D67 The NDI met (D6-37239).	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to according The inspection procedure NOTE: Applicable to all	nt) the forward aps. k form 53-80-0 omplish the int es are containe models except	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900	24000 FC re dome webs at the inspections. on is contained in the first 53-11-07.	he junction of the ra	adial
53-749-00-01	Inspect (High stiffeners/lap See Doc D6: The NDI met (D6-37239). AIRPLANE II ACCESS NO	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to according The inspection procedure NOTE: Applicable to all of DTE: Remove necessary	nt) the forward aps. k form 53-80-0 omplish the int es are containe models except y passenger ca	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as rec	24000 FC re dome webs at the inspections. on is contained in the following 53-11-07. DER. quired to perform the contained the contained to perform the contained to perform the contained the contained to perform the contained to perform the contained the contained to perform the contained th	the junction of the rather the 737 Nondestruction. ALL	adial ctive Test Manua ALL
53-749-00-01	Inspect (High stiffeners/lap See Doc D6: The NDI met (D6-37239). AIRPLANE II ACCESS NO AWL Inspect (Low doubler, Y-C.	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to accommodate The inspection procedure NOTE: Applicable to all of DTE: Remove necessary 53-05-02-250-910 or Frequency Eddy Curren thord and the tear strap b	nt) the forward aps. k form 53-80-0 omplish the intes are contained models except by passenger cannot be forward etween stringer.	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as rec 50000 FC side of the pressur rs S-1 and S-3.	24000 FC re dome webs at the inspections. on is contained in the first state of the second state of the se	the junction of the rather the 737 Nondestruction. ALL	adial ctive Test Manua ALL
53-749-00-01	Inspect (High stiffeners/lap See Doc D6: The NDI met (D6-37239). AIRPLANE II ACCESS NO AWL Inspect (Low doubler, Y-C.	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to accommodate The inspection procedure NOTE: Applicable to all of DTE: Remove necessary 53-05-02-250-910 or Frequency Eddy Curren	nt) the forward aps. k form 53-80-0 omplish the intes are contained models except by passenger cannot be forward etween stringer.	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as rec 50000 FC side of the pressur rs S-1 and S-3.	24000 FC re dome webs at the inspections. on is contained in the first state of the second state of the se	the junction of the rather the 737 Nondestruction. ALL	adial ctive Test Manua ALL
53-749-00-01	Inspect (High stiffeners/lap See Doc D6: The NDI met (D6-37239). AIRPLANE I ACCESS NO AWL Inspect (Low doubler, Y-C: See Doc D6:	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to accommodate The inspection procedure NOTE: Applicable to all of DTE: Remove necessary 53-05-02-250-910 or Frequency Eddy Curren thord and the tear strap b	nt) the forward aps. k form 53-80-0 complish the interest are contained models except by passenger cannot be forward etween stringer k form 53-80-0	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as rec 50000 FC side of the pressur rs S-1 and S-3. 1-13 for alternative	24000 FC re dome webs at the inspections. on is contained in the first state of the first	the junction of the rather the 737 Nondestruction. ALL	adial ctive Test Manua ALL
53-749-00-01	Inspect (High stiffeners/lap See Doc D6: The NDI met (D6-37239). AIRPLANE I ACCESS NO AWL Inspect (Low doubler, Y-C; See Doc D6: AIRPLANE I	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to according The inspection procedure NOTE: Applicable to all of the inspection procedure The inspection procedure NOTE: Remove necessary 53-05-02-250-910 or Frequency Eddy Curren thord and the tear strap be 26A001-DTR, DTR check	nt) the forward aps. k form 53-80-0 omplish the intes are contained models except by passenger care. 1.1 nt) the forward etween stringer k form 53-80-0 L/N 1057 an original aps.	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as rec 50000 FC side of the pressur rs S-1 and S-3. 1-13 for alternative n, except -800FPB	24000 FC re dome webs at the inspections. on is contained in the first state of the	the junction of the rather the 737 Nondestruction. ALL and the fasteners contact the properties of the fasteners contact the properties of the fasteners contact the fasteners contact the properties of the fasteners contact the fasteners contact the properties of the fasteners contact the properties of the fasteners contact the properties of the prope	adial ctive Test Manua ALL
53-749-00-01	Inspect (High stiffeners/lap See Doc D6: The NDI met (D6-37239). AIRPLANE I ACCESS NO AWL Inspect (Low doubler, Y-C; See Doc D6: AIRPLANE I	h Frequency Eddy Currer o splices and the tear stra 26A001-DTR, DTR check thod(s) necessary to accommodate The inspection procedure NOTE: Applicable to all of DTE: Remove necessary 53-05-02-250-910 or Frequency Eddy Curren thord and the tear strap be 26A001-DTR, DTR check NOTE: Applicable to all of	nt) the forward aps. k form 53-80-0 omplish the intes are contained models except by passenger care. 1.1 nt) the forward etween stringer k form 53-80-0 L/N 1057 an original aps.	50000 FC side of the pressur 1-11 for alternative ent of this inspectic ed in Part 6, Sectio -800FPB and -900 bin interiors as rec 50000 FC side of the pressur rs S-1 and S-3. 1-13 for alternative n, except -800FPB	24000 FC re dome webs at the inspections. on is contained in the first state of the	the junction of the rather the 737 Nondestruction. ALL and the fasteners contact the properties of the fasteners contact the properties of the fasteners contact the fasteners contact the properties of the fasteners contact the fasteners contact the properties of the fasteners contact the properties of the fasteners contact the properties of the prope	adial ctive Test Manua ALL

Note: Refer to Structural Repair Manual, Section 53-80-08, for definition of oil can.

No DTR form available for PSE 53-80-01-14.

AIRPLANE NOTE: All aircraft L/N 1756 and on, except -800FPB and -900ER.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-750-00-01	AWL	53-05-02-211-852	1.1	50000 FC	33000 FC	ALL	ALL
	Inspect (Deta	ailed) the fittings on both	sides of the bu	ılkhead at STA 101	6.		
	See Doc D62	26A001-DTR, DTR check	k form 53-80-0	2-1 for alternative i	nspections.		
	AIRPLANE I	NOTE: Applicable to all	models except	-800FPB and -900	ER.		
	ACCESS NO	OTE: Remove necessary	/ passenger ca	bin interiors as req	uired to perform t	the inspection.	
53-751-00-01	AWL	53-05-02-250-912	1.1	50000 FC	36000 FC	ALL	ALL
		Frequency Eddy Curre				ng lugs. Bolt remova	al is not require
		26A001-DTR, DTR check			•	the 727 Nondoetrus	ativo Toot Mon
		hod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	ctive rest Man
	,			•			
53-752-00-01	AWL	53-05-02-250-913	1.1	50000 FC	36000 FC	ALL	ALL
	Inspect (High	r Frequency Eddy Curre	nt) the stringer	splice fittings from	stringer S-9L to S	S-9R at the first two	fastener
		ward and aft of the STA 1	, -		· ·		
	See Doc D62	26A001-DTR, DTR check	k form 53-80-0	3-1 for alternative i	nspections.		
		hod(s) necessary to acc	•	•		the 737 Nondestruc	ctive Test Man
	,	The inspection procedur		•			
	ACCESS NO	OTE: Remove or displace	e passenger c	abin interior as requ	uired to perform t	he inspection.	
53-753-00-01	AWL	53-05-02-250-914	1.1	50000 FC	24000 FC	ALL	ALL
53-753-00-01	Inspect (High	n Frequency Eddy Curre					
53-753-00-01	Inspect (High	n Frequency Eddy Curre ad.	nt) both primar	y jackscrew fitting I	ugs on both sides		
53-753-00-01	Inspect (High 1088 bulkhea See Doc D62	n Frequency Eddy Curre	nt) both primar	y jackscrew fitting l 5-1 for alternative i	ugs on both sides	s around the bushin	g at the STA
53-753-00-01	Inspect (High 1088 bulkhea See Doc D62 The NDI met	n Frequency Eddy Currer ad. 26A001-DTR, DTR check	nt) both primar k form 53-80-0 omplish the int	y jackscrew fitting l 5-1 for alternative i ent of this inspection	ugs on both sides nspections.	s around the bushin	g at the STA
53-753-00-01	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currel ad. 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur DTE: Remove access pa	nt) both primar of form 53-80-0 complish the interest are contained anel as required	y jackscrew fitting I 5-1 for alternative in ent of this inspection and in Part 6, Subject	ugs on both sides nspections. on is contained in ct 53-10-95.	s around the bushin the 737 Nondestruc	g at the STA
53-753-00-01	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currel ad. 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur	nt) both primar of form 53-80-0 complish the interest are contained anel as required	y jackscrew fitting I 5-1 for alternative in ent of this inspection and in Part 6, Subject	ugs on both sides nspections. on is contained in ct 53-10-95.	s around the bushin the 737 Nondestruc	g at the STA
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currel ad. 26A001-DTR, DTR check chod(s) necessary to acc The inspection procedur DTE: Remove access pa aside for access to	nt) both primar of form 53-80-0 complish the intess are contained anel as required the lugs.	y jackscrew fitting I 5-1 for alternative i ent of this inspection ed in Part 6, Subject d. Remove/disconn	ugs on both sides nspections. on is contained in ct 53-10-95. nect the jackscrew	s around the bushin the 737 Nondestruc v from the fitting and	g at the STA ctive Test Man d move
53-753-00-01	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NO	n Frequency Eddy Currel ad. 26A001-DTR, DTR check hod(s) necessary to acc The inspection procedur DTE: Remove access paraside for access to	nt) both primar c form 53-80-0 complish the intes are contained anel as required the lugs.	y jackscrew fitting I 5-1 for alternative is ent of this inspectic ed in Part 6, Subject d. Remove/disconn	nspections. on is contained in ct 53-10-95. nect the jackscrew	s around the bushin the 737 Nondestruct or from the fitting and ALL	g at the STA ctive Test Man d move ALL
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NO	n Frequency Eddy Currel ad. 26A001-DTR, DTR check chod(s) necessary to accommodate inspection procedure. 26E Remove access paraside for access to 53-05-02-130-807. 26Sonic) the top two faster.	nt) both primar of form 53-80-0 complish the interest are contained as required the lugs.	y jackscrew fitting I 5-1 for alternative is ent of this inspectic ed in Part 6, Subject d. Remove/disconn 50000 FC	ugs on both sides nspections. on is contained in ct 53-10-95. nect the jackscrew 36000 FC gs common to the	s around the bushin the 737 Nondestruct or from the fitting and ALL	g at the STA ctive Test Man
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NO AWL Inspect (Ultra See Doc D62	n Frequency Eddy Currel ad. 26A001-DTR, DTR check chod(s) necessary to accomplete inspection procedure. 26A001-DTE: Remove access paraside for access to 53-05-02-130-807. 26A001-DTR, DTR check add.	ont) both primar of form 53-80-0 complish the interes are contained as required the lugs. 1.1 mers in the outtoor form 53-80-0	y jackscrew fitting I 5-1 for alternative in ent of this inspection of the inspection of the properties of the propertie	ugs on both sides nspections. on is contained in ot 53-10-95. hect the jackscrew 36000 FC gs common to the nspections.	the 737 Nondestruct from the fitting and ALL STA 1088 bulkhea	g at the STA ctive Test Man
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NO AWL Inspect (Ultra See Doc D62 The NDI met	n Frequency Eddy Currel ad. 26A001-DTR, DTR check chod(s) necessary to accommodate inspection procedure. 26E Remove access paraside for access to 53-05-02-130-807. 26Sonic) the top two faster.	ont) both primar of form 53-80-0 complish the interes are contained as required the lugs. 1.1 There in the outtook form 53-80-0 complish the interes in the interes in the contained as the contained are contained as the lugs.	y jackscrew fitting I 5-1 for alternative is ent of this inspectic ed in Part 6, Subject d. Remove/disconn 50000 FC coard primary fitting 6-1 for alternative is ent of this inspectic	ugs on both sides nspections. on is contained in ot 53-10-95. hect the jackscrew 36000 FC gs common to the nspections. on is contained in	the 737 Nondestruct from the fitting and ALL STA 1088 bulkhea	g at the STA ctive Test Man
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NC AWL Inspect (Ultra See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currel ad. 26A001-DTR, DTR check thod(s) necessary to accomplete the inspection procedure. 26EX. The inspection procedure aside for access to aside for access to 53-05-02-130-807. 26EX. The inspection procedure.	ont) both primar of form 53-80-0 complish the interest are contained anel as required the lugs. 1.1 ners in the outle of form 53-80-0 complish the interest are contained as are contained as required the lugs.	y jackscrew fitting I 5-1 for alternative is ent of this inspectic ed in Part 6, Subject d. Remove/disconn 50000 FC booard primary fitting 6-1 for alternative is ent of this inspectic ed in Part 4, Subject disconnection	ugs on both sides nspections. on is contained in ct 53-10-95. nect the jackscrew 36000 FC gs common to the nspections. on is contained in ct 53-80-01.	the 737 Nondestruct from the fitting and ALL STA 1088 bulkhea	g at the STA ctive Test Man d move ALL d. ctive Test Man
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NC AWL Inspect (Ultra See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currel ad. 26A001-DTR, DTR check chod(s) necessary to accomplete inspection procedure. 26EEEEEEEEEEEEEEEEEEEEEEEEEEEEEEEEEEE	ont) both primar of form 53-80-0 complish the interest are contained anel as required the lugs. 1.1 There in the outle of form 53-80-0 complish the interest are contained the tailcone	y jackscrew fitting I 5-1 for alternative is ent of this inspectic ed in Part 6, Subject d. Remove/disconn 50000 FC board primary fitting 6-1 for alternative is ent of this inspectic ed in Part 4, Subject on the forward and	ugs on both sides nspections. on is contained in ct 53-10-95. nect the jackscrew 36000 FC gs common to the nspections. on is contained in ct 53-80-01. d aft side of the S	the 737 Nondestruct from the fitting and ALL STA 1088 bulkheat TA 1088 Bulkhead.	g at the STA ctive Test Manual d move ALL d. ctive Test Manual
	Inspect (High 1088 bulkher See Doc D62 The NDI met (D6-37239). ACCESS NC AWL Inspect (Ultra See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Currel ad. 26A001-DTR, DTR check thod(s) necessary to accomplete the inspection procedure. The inspection procedure aside for access to aside for access to 53-05-02-130-807. asonic) the top two faster 26A001-DTR, DTR check thod(s) necessary to accomplete inspection procedure. TE: Access fittings inside.	ont) both primar of form 53-80-0 complish the interest are contained anel as required the lugs. 1.1 There in the outle of form 53-80-0 complish the interest are contained the tailcone	y jackscrew fitting I 5-1 for alternative is ent of this inspectic ed in Part 6, Subject d. Remove/disconn 50000 FC board primary fitting 6-1 for alternative is ent of this inspectic ed in Part 4, Subject on the forward and	ugs on both sides nspections. on is contained in ct 53-10-95. nect the jackscrew 36000 FC gs common to the nspections. on is contained in ct 53-80-01. d aft side of the S	the 737 Nondestruct from the fitting and ALL STA 1088 bulkheat TA 1088 Bulkhead.	g at the STA ctive Test Man d move ALL d. ctive Test Man

Inspect (Detailed) the four primary fitting lugs, from the inside of the lugs, at STA 1088. See Doc D626A001-DTR, DTR check form 53-80-06-2 for alternative inspections.

ACCESS NOTE: Remove vertical fin including primary and fail-safe bolts.





				INTERVAL		APPLICABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-756-00-01	AWL	53-05-02-250-915	1.1	50000 FC	36000 FC	ALL	ALL	
	See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Curre 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur NOTE: Applicable to L/N	k form 53-80-0 complish the interes are contain	7-1 for alternative itent of this inspection	inspections. on is contained in			
53-757-00-01	AWL	53-05-02-250-916	1.1	50000 FC	36000 FC	ALL	ALL	
	down 16 inch See Doc D62 The NDI met	n Frequency Eddy Curre nes vertically. 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	k form 53-80-0 complish the int	7-2 for alternative i	inspections. on is contained in			
	AIRPLANE I	NOTE: Applicable to all	aircrafts from l	_/N 1199 and on.				
	ACCESS NO	OTE: Enter aircraft throu	igh the tail con	e access panel.				
53-758-00-01	AWL	53-05-02-250-917	1.1	50000 FC	24000 FC	ALL	ALL	
	The NDI met (D6-37239).	26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Remove sliding se	complish the intress are contain	ent of this inspection ed in Part 6, Subject	on is contained in	the 737 Nondestru	ctive Test Man	
53-758-00-02	AWL	53-05-02-250-917	1.1	50000 FC	24000 FC	ALL	ALL	
	See Doc D62 The NDI met (D6-37239).	n Frequency Eddy Curre 26A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Remove sliding se	k form 53-80-0 complish the intess are contain	8-1 for alternative i ent of this inspection ed in Part 6, Subject	inspections. on is contained in			
53-759-00-01	AWL	53-05-02-230-801	1.1	50000 FC	14000 FC	ALL	ALL	
		Penetrant) both the inne 26A001-DTR, DTR chec	•	•				
	ACCESS NO	OTE: Removal and sepa	aration of pivot	pins is required to	perform the inspe	ction.		
53-759-00-02	AWL	53-05-02-230-801	1.1	50000 FC	14000 FC	ALL	ALL	
		Penetrant) both the inne 26A001-DTR, DTR chec		•				
	ACCESS NO	OTE: Removal and sepa	aration of pivot	pins is required to	perform the inspe	ction.		
53-760-00-01	AWL	53-05-02-211-854	1.1	50000 FC	18000 FC	ALL	ALL	
				100				

Inspect (Detailed) the skin panels around the STA 1138 cutout.

See Doc D626A001-DTR, DTR check form 53-80-10-1 for alternative inspections.







ASK CARD NO. 53-760-00-02				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-760-00-02	AWL	53-05-02-211-854	1.1	50000 FC	18000 FC	ALL	ALL
	Inspect (Deta	ailed) the skin panels ard	ound the STA 1	138 cutout.			
	See Doc D6	26A001-DTR, DTR chec	k form 53-80-1	0-1 for alternative in	nspections.		
53-800-00-01	MRB	05-41-01-210-801	1.1	120 DY	120 DY	ALL	ALL
	wheel well a	external zonal inspection re also included. Inspect cess panels required. (E	ion is accompli				
	INTERVAL N	NOTE: The EZAP inspection.	ction requireme	ent with interval 550	00 FC/30 MO is s	atisfied by this zona	I
53-802-00-01	MRB	05-41-01-210-802	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the rad	ome.			
	INTERVAL N	NOTE: Whichever come	s first.				
53-804-00-01	MRB	05-41-01-210-803	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
		nternal zonal inspection	(GV) of the are	a forward of the no	se wheel well - S	ection 41, Sta 178 t	o Sta 224.8.
	(EZAP) INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	val 36000 FC/12 YF	
53-806-00-01	, ,	NOTE: Whichever come satisfied by this z	zonal inspection 1.1	6600 FC	6600 FC	val 36000 FC/12 YF ALL	
53-806-00-01	MRB	satisfied by this z	20nal inspection 1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	R is
53-806-00-01	MRB Perform an i	satisfied by this z	1.1 1.2 (GV) of the are	6600 FC 36 MO a above and outbo	6600 FC 36 MO ard of the nose w	ALL rheel well. (EZAP)	R is
53-806-00-01	MRB Perform an i	satisfied by this z	1.1 1.2 (GV) of the are	6600 FC 36 MO a above and outbo	6600 FC 36 MO ard of the nose w	ALL rheel well. (EZAP)	R is
53-806-00-01 53-808-00-01	MRB Perform an i	satisfied by this z 05-41-01-210-804 nternal zonal inspection NOTE: Whichever come	1.1 1.2 (GV) of the are	6600 FC 36 MO a above and outbo	6600 FC 36 MO ard of the nose w	ALL rheel well. (EZAP)	R is
	MRB Perform an i	satisfied by this z 05-41-01-210-804 Internal zonal inspection NOTE: Whichever come satisfied by this z	1.1 1.2 (GV) of the are s first. The EZ/ conal inspection 1.1 1.2	6600 FC 36 MO a above and outbo AP inspection requin. 6600 FC 36 MO	6600 FC 36 MO ard of the nose w rement with inter 6600 FC 36 MO	ALL wheel well. (EZAP) wal 36000 FC/12 YR ALL	ALL R is
	MRB Perform an i INTERVAL N MRB Perform an i	satisfied by this z 05-41-01-210-804 Internal zonal inspection NOTE: Whichever come satisfied by this z 05-41-01-210-805	1.1 1.2 (GV) of the are s first. The EZ/conal inspection 1.1 1.2 (GV) of the ele s first. The EZ/	6600 FC 36 MO a above and outbo AP inspection requi a. 6600 FC 36 MO ctrical and electron AP inspection requi	6600 FC 36 MO ard of the nose water with intended the following of the nose water for the following of the f	ALL /heel well. (EZAP) /al 36000 FC/12 YR ALL (EZAP)	ALL ALL
	MRB Perform an i INTERVAL M MRB Perform an i INTERVAL M	satisfied by this z 05-41-01-210-804 Internal zonal inspection NOTE: Whichever come satisfied by this z 05-41-01-210-805 Internal zonal inspection NOTE: Whichever come	1.1 1.2 (GV) of the are s first. The EZ/conal inspection 1.1 1.2 (GV) of the ele s first. The EZ/conal inspection	6600 FC 36 MO a above and outbo AP inspection requin. 6600 FC 36 MO ctrical and electron AP inspection requin. coanel must be remo	6600 FC 36 MO ard of the nose w rement with interes 6600 FC 36 MO ics compartment.	ALL wheel well. (EZAP) val 36000 FC/12 YR ALL (EZAP) val 18000 FC/6 YR i	ALL ALL

Perform an internal zonal inspection (GV) of the electrical and electronics compartment access door.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-812-00-01	MRB	05-41-01-210-807	1.1	120 DY	120 DY	ALL	ALL
		nternal zonal inspection aft bulkhead.	(GV) of the for	ward cargo compar	tment - section 43	3, STA 396 to forwa	rd cargo
	AIRPLANE I	NOTE: For aircraft equip	pped with optic	nal auxiliary fuel ta	nks, removal of ta	inks not required.	
	ACCESS NO	TE: No cargo liners rer	moved.				
53-814-00-01	MRB	05-41-01-210-808	1.1 1.2	13200 FC 72 MO	13200 FC 72 MO	ALL	ALL
		nternal zonal inspection o compartment aft bulkh	` '	a above the forwar	d cargo compartn	nent - section 43, S	TA 396 to
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	TE: Ceiling panels rem	oval required.				
53-816-00-01	MRB	05-41-01-210-809	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL
		nternal zonal inspection aft bulkhead.	(GV) of the for	ward cargo compar	tment - section 43	3, STA 396 to forwa	rd cargo
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	TE: Sidewall panels re	moval required	l.			
53-818-00-01	MRB	05-41-01-210-810	1.1	2000 FC	2000 FC	ALL	ALL
			1.2	240 DY	240 DY		
	Perform an e 460.	xternal zonal inspection	(GV) of the for	ward cargo door su	urround structure	fittings and stops - s	section 43, S
		IOTE: Whichever come	s first				
			· ·				
53-820-00-01	MRB	05-41-01-210-811	1.1 1.2	13200 FC 72 MO	13200 FC 72 MO	ALL	ALL
		nternal zonal inspection rtment aft bulkhead. (EZ	` '	a below the forwar	d cargo compartn	nent - Section 43, S	ta 396 to forv
	INTERVAL N	IOTE: Whichever come satisfied by this z			rement with interv	/al 18000 FC/6 YR	is
	ACCESS NO	OTE: Center floor panels	s removal requ	ired. Cargo loading	system removed	/displaced as requir	ed.
53-822-00-01	MRB	05-41-01-210-812	1.1	6600 FC	6600 FC	ALL	ALL
			1.2	36 MO	36 MO		
	Perform an ir						

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Forward cargo compartment aft bulkhead center panel removal required.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-824-01-01	MRB	05-41-01-210-813	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL			
	663.75.	nternal zonal inspection (,		o o	,	540 to STA			
		·		complianing tasks t	55-070-00-01 and	33-000-00-01.				
		INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Floor panels removal required. Insulation as required.								
		·	·	·						
53-826-00-01	MRB	05-41-01-210-814	1.1 1.2	13200 FC 72 MO	13200 FC 72 MO	ALL	ALL			
		nternal zonal inspection (a 727. (EZAP)	(GV) of the pre	essure deck above	the main landing	gear wheel well - Se	ection 44, Sta			
	INTERVAL N	NOTE: Whichever comes satisfied by this z			irement with interv	/al 36000 FC/12 YR	l is			
	ACCESS NO	OTE: Floor panels remov	/al required.							
53-828-00-01	MRB	05-41-01-210-815	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL			
	Perform an i	nternal zonal inspection ((GV) of the kee	el beam (part) Sta	540 to 727 - Section	on 44. (EZAP)				
	INTERVAL N	NOTE: Whichever comes satisfied by this z			irement with interv	/al 36000 FC/12 YR	l is			
53-830-00-01	MRB	05-41-01-210-817	1.1	120 DY	120 DY	ALL	ALL			
53-830-00-01		05-41-01-210-817								
53-830-00-01	Perform an in		(GV) of the aft	cargo compartmer	nt - section 46 and	47 (part), sta 727 to				
53-830-00-01	Perform an ii	nternal zonal inspection ((GV) of the aft	cargo compartmer	nt - section 46 and	47 (part), sta 727 to				
	Perform an in	nternal zonal inspection (NOTE: For aircraft equip OTE: No cargo liners ren	(GV) of the aft oped with option	cargo compartmer	nt - section 46 and	47 (part), sta 727 t	o sta 947.5.			
53-830-00-01 53-832-00-01	Perform an ii	nternal zonal inspection ((GV) of the aft	cargo compartmer	nt - section 46 and	47 (part), sta 727 to				
	Perform an in AIRPLANE I ACCESS NO	nternal zonal inspection (NOTE: For aircraft equip OTE: No cargo liners ren	(GV) of the aft oped with option noved.	cargo compartmer onal fuel tanks, rem 13200 FC 72 MO	nt - section 46 and noval of tanks not n 13200 FC 72 MO	47 (part), sta 727 to required. ALL	o sta 947.5.			
	Perform an in AIRPLANE I ACCESS NO MRB Perform an in	nternal zonal inspection (NOTE: For aircraft equip OTE: No cargo liners ren 05-41-01-210-818	(GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft	cargo compartmer onal fuel tanks, rem 13200 FC 72 MO	nt - section 46 and noval of tanks not n 13200 FC 72 MO	47 (part), sta 727 to required. ALL	o sta 947.5.			
	Perform an in AIRPLANE II ACCESS NO MRB Perform an in INTERVAL IN	nternal zonal inspection (NOTE: For aircraft equip DTE: No cargo liners ren 05-41-01-210-818 Internal zonal inspection ((GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft is first.	cargo compartment on al fuel tanks, rem 13200 FC 72 MO cargo compartment	nt - section 46 and noval of tanks not not 13200 FC 72 MO nt - section 46 and	47 (part), sta 727 to required. ALL	o sta 947.5. ALL			
	Perform an in AIRPLANE II ACCESS NO MRB Perform an in INTERVAL IN	nternal zonal inspection (NOTE: For aircraft equip OTE: No cargo liners ren 05-41-01-210-818 Internal zonal inspection (NOTE: Whichever comes	(GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft s first. wall and ceiling	cargo compartment on al fuel tanks, rem 13200 FC 72 MO cargo compartment of panels removal r	13200 FC 72 MO nt - section 46 and equired.	47 (part), sta 727 to required. ALL	o sta 947.5.			
53-832-00-01	Perform an in AIRPLANE I ACCESS NO MRB Perform an in INTERVAL N ACCESS NO	nternal zonal inspection (NOTE: For aircraft equip DTE: No cargo liners ren 05-41-01-210-818 Internal zonal inspection (NOTE: Whichever comes DTE: Upper angled sides 05-41-01-210-819	(GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft is first. wall and ceiling 1.1 1.2	cargo compartment on al fuel tanks, rem 13200 FC 72 MO cargo compartment of panels removal ref 36000 FC 12 YR	13200 FC 72 MO nt - section 46 and equired.	47 (part), sta 727 to required. ALL 47 (part), sta 727 to ALL	ALL o sta 947.5.			
53-832-00-01	Perform an in AIRPLANE I ACCESS NO MRB Perform an in INTERVAL N ACCESS NO MRB Perform an in	nternal zonal inspection (NOTE: For aircraft equip DTE: No cargo liners ren 05-41-01-210-818 Internal zonal inspection (NOTE: Whichever comes DTE: Upper angled sides 05-41-01-210-819 Internal zonal inspection ((GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft is first. wall and ceiling 1.1 1.2 (GV) of the aft	cargo compartment on al fuel tanks, rem 13200 FC 72 MO cargo compartment of panels removal ref 36000 FC 12 YR	13200 FC 72 MO nt - section 46 and equired.	47 (part), sta 727 to required. ALL 47 (part), sta 727 to ALL	ALL o sta 947.5.			
53-832-00-01	Perform an in AIRPLANE I ACCESS NO MRB Perform an in INTERVAL N Perform an in INTERVAL N	nternal zonal inspection (NOTE: For aircraft equip DTE: No cargo liners ren 05-41-01-210-818 Internal zonal inspection (NOTE: Whichever comes DTE: Upper angled sides 05-41-01-210-819 Internal zonal inspection (NOTE: Whichever comes	(GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft is first. wall and ceiling 1.1 1.2 (GV) of the aft is first.	cargo compartmer onal fuel tanks, rem 13200 FC 72 MO cargo compartmer g panels removal re 36000 FC 12 YR cargo compartmer	13200 FC 72 MO nt - section 46 and equired.	47 (part), sta 727 to required. ALL 47 (part), sta 727 to ALL	ALL o sta 947.5.			
53-832-00-01	Perform an in AIRPLANE I ACCESS NO MRB Perform an in INTERVAL N Perform an in INTERVAL N	nternal zonal inspection (NOTE: For aircraft equip DTE: No cargo liners ren 05-41-01-210-818 Internal zonal inspection (NOTE: Whichever comes DTE: Upper angled sides 05-41-01-210-819 Internal zonal inspection ((GV) of the aft oped with option noved. 1.1 1.2 (GV) of the aft is first. wall and ceiling 1.1 1.2 (GV) of the aft is first.	cargo compartmer onal fuel tanks, rem 13200 FC 72 MO cargo compartmer g panels removal re 36000 FC 12 YR cargo compartmer	13200 FC 72 MO nt - section 46 and equired.	47 (part), sta 727 to required. ALL 47 (part), sta 727 to ALL	ALL o sta 947.5.			

 $Perform\ an\ external\ zonal\ inspection\ (GV)\ of\ the\ aft\ cargo\ door\ surround\ structure\ fittings\ and\ stops\ -\ section\ 46,\ sta\ 827.$

INTERVAL NOTE: Whichever comes first.







- A O I / O A D D N I O				INTERVAL		APPLICA	
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-838-00-01	MRB	05-41-01-210-821	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the aft	cargo compartmer	nt vacuum waste o	compartment.	
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Vacuum waste con	npartment pane	els removal require	ed.		
53-840-00-01	MRB	05-41-01-210-822	1.1 1.2	13200 FC 72 MO	13200 FC 72 MO	ALL	ALL
	Perform an into Sta 947.5.	nternal zonal inspection (EZAP)	(GV) of the are	a below the aft car	go compartment -	Section 46 and 47	(part), Sta 7
	INTERVAL N	IOTE: Whichever come satisfied by this z			irement with interv	/al 18000 FC/6 YR	is
	ACCESS NO	OTE: Center floor panels	removal requi	ired. Cargo loading	g system removed	/displaced as requi	red.
53-842-00-01	MRB	05-41-01-210-823	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
		nternal zonal inspection (khead. (EZAP)	(GV) of the aft	cargo compartmer	nt equipment bay	- Section 47, Sta 94	7.5. to aft
	INTERVAL N	IOTE: Whichever come			irement with interv	/al 18000 FC/6 YR	is
		satisfied by this z	onal inspection	٦.			
	ACCESS NO	OTE: Aft cargo panels at	Sta 947 bulkh		red.		
	ACCESS NO	OTE: Aft cargo panels at	Sta 947 bulkh		red.		
53-844-00-01	MRB	OTE: Aft cargo panels at 05-41-01-210-824	Sta 947 bulkh		red. 120 DY	ALL	ALL
53-844-00-01	MRB Perform an e		1.1 (GV) of the wir	ead removal requi 120 DY ng to body fairing.	120 DY		
53-844-00-01	MRB Perform an e	05-41-01-210-824 external zonal inspection	1.1 (GV) of the wir	ead removal requi 120 DY ng to body fairing.	120 DY		
	MRB Perform an ethe use of sta	05-41-01-210-824 external zonal inspection ands or ladders. No addi	1.1 (GV) of the wir tional access p 1.1 1.2	120 DY ng to body fairing. banels required. 6600 FC 36 MO	120 DY Inspection is acco	mplished from the o	ground, witho
	MRB Perform an et the use of state MRB Perform an in	05-41-01-210-824 external zonal inspection ands or ladders. No addi	1.1 (GV) of the wintional access p 1.1 1.2 (GV) of the lows first. The EZA	120 DY Ing to body fairing. Inanels required. 6600 FC 36 MO Iter wing to body fairing.	120 DY Inspection is acco	ALL ving box. (EZAP)	ground, witho
	MRB Perform an ethe use of state MRB Perform an in	05-41-01-210-824 external zonal inspection ands or ladders. No addi 05-41-01-210-825 external zonal inspection of the control	1.1 (GV) of the wintional access p 1.1 1.2 (GV) of the low s first. The EZA	120 DY Ing to body fairing. Inanels required. 6600 FC 36 MO Iter wing to body fairing.	120 DY Inspection is acco	ALL ving box. (EZAP)	ground, withou
	MRB Perform an ethe use of state MRB Perform an in	05-41-01-210-824 external zonal inspection ands or ladders. No addi 05-41-01-210-825 external zonal inspection of the companion of the compani	1.1 (GV) of the wintional access p 1.1 1.2 (GV) of the low s first. The EZA	120 DY Ing to body fairing. Inanels required. 6600 FC 36 MO Iter wing to body fairing.	120 DY Inspection is acco	ALL ving box. (EZAP)	ground, withou
53-846-00-01	MRB Perform an ethe use of state MRB Perform an in INTERVAL M ACCESS NO	05-41-01-210-824 external zonal inspection ands or ladders. No addi 05-41-01-210-825 Internal zonal inspection of the companion of the compani	1.1 (GV) of the wintional access positional access positional access positional in 1.2 (GV) of the lowest first. The EZA conal inspection ovided.	120 DY 120 DY 130 body fairing. 130 per sequired. 130 per sequired. 130 per sequired. 130 per wing to body fairing. 130 per wing to body fairing. 130 per wing to body fairing. 131 per wing to body fairing. 132 per wing to body fairing. 133 per wing to body fairing. 134 per wing to body fairing. 135 per wing to body fairing.	120 DY Inspection is accordance 6600 FC 36 MO iring - forward of virement with intended	ALL ving box. (EZAP) val 36000 FC/12 YF	ground, without ALL
53-846-00-01	MRB Perform an ethe use of statement of stat	05-41-01-210-824 external zonal inspection ands or ladders. No addi 05-41-01-210-825 Internal zonal inspection of the satisfied by this zonal inspection of the satisfied by the sat	1.1 (GV) of the wintional access p 1.1 1.2 (GV) of the low s first. The EZA conal inspection ovided. 1.1 1.2 (GV) of the low s first. The EZA conal inspection ovided.	120 DY Ing to body fairing. In	120 DY Inspection is accordance 6600 FC 36 MO iring - forward of virement with intervirement with intervirem	ALL ving box. (EZAP) val 36000 FC/12 YF	ALL ALL
53-846-00-01	MRB Perform an in INTERVAL M Perform an in INTERVAL M INTERVAL M	05-41-01-210-824 external zonal inspection ands or ladders. No additional control of the state o	1.1 (GV) of the wintional access positional access positional access positional access positional in 1.2 (GV) of the lowest first. The EZA conal inspection positional inspection access to the lowest first. The EZA conal inspectional inspection	120 DY Ing to body fairing. In	120 DY Inspection is accordance 6600 FC 36 MO iring - forward of virement with intervirement with intervirence and more services.	ALL ving box. (EZAP) val 36000 FC/12 YF	ALL ALL
53-846-00-01	MRB Perform an in INTERVAL M Perform an in INTERVAL M INTERVAL M	05-41-01-210-824 external zonal inspection ands or ladders. No addi 05-41-01-210-825 Internal zonal inspection of the state of the stat	1.1 (GV) of the wintional access positional access positional access positional access positional in 1.2 (GV) of the lowest first. The EZA conal inspection positional inspection access to the lowest first. The EZA conal inspectional inspection	120 DY Ing to body fairing. In	120 DY Inspection is accordance 6600 FC 36 MO iring - forward of virement with intervirement with intervirence and more services.	ALL ving box. (EZAP) val 36000 FC/12 YF	Al

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.







TASK CARD NO.				INTERVAL		APPLICA	ABILITY
ASK CAKD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-852-00-01	MRB	05-41-01-210-828	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection ((GV) of the low	ver wing to body fai	ring - aft of wheel	well.	
	INTERVAL I	NOTE: Whichever comes	s first.				
	ACCESS NO	OTE: Through access pr	ovided.				
53-854-02-01	MRB	05-41-01-210-829	1.1 1.2	24000 FC 9 YR	24000 FC 8 YR	ALL	ALL
	Perform an i	internal zonal inspection ((GV) of the abo	ove wing, wing to b	ody fairing - right	side.	
	INTERVAL I	NOTE: Whichever comes	s first.				
53-856-01-01	MRB	05-41-01-210-830	1.1 1.2	24000 FC 9 YR	24000 FC 8 YR	ALL	ALL
	Perform an i	nternal zonal inspection ((GV) of the abo	ove wing, wing to b	ody fairing - left si	de.	
	INTERVAL I	NOTE: Whichever comes	s first.				
53-858-00-01	MRB	05-41-02-210-801	1.1	6600 FC	6600 FC	ALL	ALL
			1 2	26 MO	36 MO		
			1.2	36 MO			
		external zonal inspection	(GV) of the flig			STA 178 to STA 2	70.
		external zonal inspection	(GV) of the flig			STA 178 to STA 2	70.
53-860-00-01		·	(GV) of the flig			STA 178 to STA 2	70.
53-860-00-01	MRB	NOTE: Whichever comes	(GV) of the flig s first.	ght control compart 5500 FC 24 MO	ment - section 41 5500 FC 24 MO	ALL	ALL
53-860-00-01	MRB Perform an i	NOTE: Whichever comes 05-41-02-210-802	(GV) of the flig s first.	ght control compart 5500 FC 24 MO	ment - section 41 5500 FC 24 MO	ALL	ALL
53-860-00-01	MRB Perform an i	NOTE: Whichever comes 05-41-02-210-802 Internal zonal inspection ((GV) of the flig s first. 1.1 1.2 (GV) of the flig s first.	5500 FC 24 MO ht control compartr	5500 FC 24 MO nent - section 41,	ALL	ALL
53-860-00-01 53-862-00-01	MRB Perform an i	NOTE: Whichever comes 05-41-02-210-802 Internal zonal inspection (NOTE: Whichever comes	(GV) of the flig s first. 1.1 1.2 (GV) of the flig s first.	5500 FC 24 MO ht control compartr	5500 FC 24 MO nent - section 41,	ALL	ALL
	MRB Perform an i INTERVAL I ACCESS NO	05-41-02-210-802 Internal zonal inspection (NOTE: Whichever comes	(GV) of the flig s first. 1.1 1.2 (GV) of the flig s first. ss panels 211. 1.1 1.2	5500 FC 24 MO ht control compartr A and 212A remova 36000 FC 10 YR	5500 FC 24 MO nent - section 41, al required. 36000 FC 10 YR	ALL sta 178 to sta 270. ALL	ALL
	MRB Perform an ii INTERVAL I ACCESS NO MRB Perform an ii	05-41-02-210-802 Internal zonal inspection (NOTE: Whichever comes OTE: Control stand acce	(GV) of the flig s first. 1.1 1.2 (GV) of the flig s first. ss panels 211. 1.1 1.2 (GV) of the flig s first. The EZ	5500 FC 24 MO ht control compartr A and 212A remova 36000 FC 10 YR ht control compartr	5500 FC 24 MO ment - section 41, al required. 36000 FC 10 YR ment - section 41,	ALL sta 178 to sta 270. ALL STA 178 to STA 27	ALL O. (EZAP)
	MRB Perform an i INTERVAL I ACCESS NO MRB Perform an i INTERVAL I	05-41-02-210-802 Internal zonal inspection (NOTE: Whichever comes OTE: Control stand acce 05-41-02-210-803 Internal zonal inspection (NOTE: Whichever comes	(GV) of the flig s first. 1.1 1.2 (GV) of the flig s first. ss panels 211. 1.1 1.2 (GV) of the flig s first. The EZ, onal inspection	5500 FC 24 MO ht control compartr A and 212A remova 36000 FC 10 YR ht control compartr AP inspection requine.	5500 FC 24 MO ment - section 41, al required. 36000 FC 10 YR ment - section 41, rement with interv	ALL Sta 178 to sta 270. ALL STA 178 to STA 27 ral 36000 FC/12 YF	ALL TO. (EZAP) R is

Perform an external zonal inspection (GV) of the passenger compartment - aft of the control compartment to forward entry door - left and right - section 41, sta 270 to sta 360.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-866-00-01	MRB	05-41-02-210-805	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the pas	ssenger compartme	ent - left and right	- Section 41, STA 2	270 to STA 36
	INTERVAL N	NOTE: Whichever come	s first.				
53-868-00-01	MRB	05-41-02-210-806	1.1 1.2	36000 FC 9 YR	36000 FC 8 YR	ALL	ALL
		nternal zonal inspection d right - Section 41, Sta	` '		ent - aft of the cor	ntrol compartment to	forward entr
	INTERVAL N	NOTE: Whichever come satisfied by this z			rement with inter	val 36000 FC/12 YR	R is
	ACCESS NO	OTE: With access provide removal required.	led. Galleys ar	nd lavs removed. Fl	oor panels, sidew	all panels, and ceili	ng panels
53-870-00-01	MRB	05-41-02-210-807	1.1 1.2	2000 FC 240 DY	2000 FC 240 DY	ALL	ALL
		external zonal inspection	(GV) of the for	ward passenger er	ntry door stops, la	tches and hinges -	section 41, st
	345.	NOTE: Whichever come	a firet				
	INTERVALI	OTE. Whichever come	5 11151.				
53-872-00-01	MRB	05-41-02-210-808	1.1 1.2	2000 FC 240 DY	2000 FC 240 DY	ALL	ALL
53-872-00-01		05-41-02-210-808 external zonal inspection	1.2	240 DY	240 DY		
53-872-00-01	Perform an e		1.2 (GV) of the for	240 DY	240 DY		
53-872-00-01 53-874-00-01	Perform an e	external zonal inspection	1.2 (GV) of the for	240 DY	240 DY		
	Perform an 6 340. INTERVAL N	external zonal inspection NOTE: Whichever come 05-41-02-210-810 external zonal inspection	1.2 (GV) of the for s first.	240 DY ward galley service 6600 FC 36 MO	240 DY e door stops, latch 6600 FC 36 MO	nes, and hinges - se	ection 41, sta
	Perform an e 340. INTERVAL N MRB Perform an e section 43 an	external zonal inspection NOTE: Whichever come 05-41-02-210-810 external zonal inspection	1.2 (GV) of the forms first. 1.1 1.2 (GV) of the forms	240 DY ward galley service 6600 FC 36 MO	240 DY e door stops, latch 6600 FC 36 MO	nes, and hinges - se	ection 41, sta
	Perform an e 340. INTERVAL N MRB Perform an e section 43 an	external zonal inspection NOTE: Whichever come 05-41-02-210-810 external zonal inspection and 44 (part).	1.2 (GV) of the forms first. 1.1 1.2 (GV) of the forms	240 DY ward galley service 6600 FC 36 MO	240 DY e door stops, latch 6600 FC 36 MO	nes, and hinges - se	ection 41, sta
53-874-00-01	Perform an e 340. INTERVAL M MRB Perform an e section 43 au INTERVAL M	external zonal inspection NOTE: Whichever come 05-41-02-210-810 external zonal inspection and 44 (part). NOTE: Whichever come 05-41-02-210-811 Internal zonal inspection	1.2 (GV) of the forms first. 1.1 1.2 (GV) of the forms first.	240 DY ward galley service 6600 FC 36 MO ward passenger co	240 DY e door stops, latch 6600 FC 36 MO empartment - sta 4800 FC 24 MO	ALL ALL ALL	ALL ALL
53-874-00-01	Perform an e 340. INTERVAL M MRB Perform an e section 43 au INTERVAL M MRB Perform an is section 43 au	external zonal inspection NOTE: Whichever come 05-41-02-210-810 external zonal inspection and 44 (part). NOTE: Whichever come 05-41-02-210-811 Internal zonal inspection	1.2 (GV) of the formula s first. 1.1 1.2 (GV) of the formula s first. 1.1 1.2 (GV) of the formula s first.	240 DY ward galley service 6600 FC 36 MO ward passenger co	240 DY e door stops, latch 6600 FC 36 MO empartment - sta 4800 FC 24 MO	ALL ALL ALL	ALL ALL

Perform an internal zonal inspection (GV) of the forward passenger compartment - Sta 360 to Sta 663.75 wet areas (within approximately 20 inches from galley or lav) - left and right - Section 43 and 44 (part). (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required in areas where galleys and lavs are located.





LV C V DD NO	INTERVAL					APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
53-880-00-01	MRB	05-41-02-210-813	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL
		nternal zonal inspection galleys, and lavs) - left a				360 to Sta 663.75 d	ry area (away
	INTERVAL N	NOTE: Whichever come satisfied by this z			rement with inter	val 36000 FC/12 YR	is
	ACCESS NO	OTE: Floor panels, sidev	wall panels, an	d ceiling panels rer	noval required.		
53-882-00-01	MRB	05-41-02-210-814	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	and right - se	external zonal inspection ection 44 (part), 46, and a applicable to airplanes wi	47, sta 663.75	to the aft pressure		to the aft pressure	bulkhead - le
	INTERVAL N	IOTE: Whichever come	s first.				
53-884-00-01	MRB	05-41-02-210-815	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	and right- se	nternal zonal inspection ction 44 (part), 46 and 4 applicable to airplanes wi	7, STA 663.75	to aft pressure bulk		75 to the aft pressur	e bulkhead -
	INTERVAL N	IOTE: Whichever come	s first.				
53-886-00-01	MRB	05-41-02-210-816	1.1 1.2	36000 FC 8 YR	36000 FC 8 YR	ALL	ALL
53-886-00-01	MRB Perform an in area (within a	05-41-02-210-816 Internal zonal inspection approximately 20 inches	1.1 1.2 (GV) of the aft from galley se	8 YR passenger compar rvice door, passen	8 YR tment - Sta 663.7	5 to the aft pressure	e bulkhead w
53-886-00-01	MRB Perform an ir area (within a (part), 46, an	05-41-02-210-816	1.1 1.2 (GV) of the aft from galley se	8 YR passenger compar rvice door, passengead. (EZAP)	8 YR tment - Sta 663.7	5 to the aft pressure	e bulkhead w
53-886-00-01	Perform an ii area (within a (part), 46, an NOTE: Not a	05-41-02-210-816 Internal zonal inspection approximately 20 inches id 47, Sta 663.75 to aft p	1.1 1.2 (GV) of the aft from galley se pressure bulkhe th flat aft press s first. The EZA	8 YR passenger compar rvice door, passenged. (EZAP) sure bulkhead. AP inspection requi	8 YR tment - Sta 663.7 ger door, galley on	5 to the aft pressure lav) - left and right	e bulkhead wo - Section 44
53-886-00-01	Perform an in area (within a (part), 46, an NOTE: Not a	05-41-02-210-816 Internal zonal inspection approximately 20 inches id 47, Sta 663.75 to aft pupplicable to airplanes willote: Whichever come	1.1 1.2 (GV) of the aft from galley se pressure bulkhe th flat aft press s first. The EZ/ conal inspection emoved. Floor	8 YR passenger compar rvice door, passenged. (EZAP) sure bulkhead. AP inspection requin. panels, sidewall pa	8 YR tment - Sta 663.7 ger door, galley or rement with inter-	5 to the aft pressure lav) - left and right val 36000 FC/12 YR	e bulkhead w - Section 44
53-886-00-01 53-888-00-01	Perform an in area (within a (part), 46, an NOTE: Not a	05-41-02-210-816 Internal zonal inspection approximately 20 inches d 47, Sta 663.75 to aft pupplicable to airplanes with satisfied by this zone. OTE: Galleys and lavs researched.	1.1 1.2 (GV) of the aft from galley se pressure bulkhe th flat aft press s first. The EZ/ conal inspection emoved. Floor	8 YR passenger compar rvice door, passenged. (EZAP) sure bulkhead. AP inspection requin. panels, sidewall pa	8 YR tment - Sta 663.7 ger door, galley or rement with inter-	5 to the aft pressure lav) - left and right val 36000 FC/12 YR	e bulkhead w - Section 44
	MRB Perform an ir area (within a (part), 46, an NOTE: Not a INTERVAL NACCESS NOTE: MRB	05-41-02-210-816 Internal zonal inspection approximately 20 inches id 47, Sta 663.75 to aft pupplicable to airplanes willotte: Whichever come satisfied by this zotte: Galleys and lavs reareas where galley	1.1 1.2 (GV) of the aft from galley se pressure bulkhe th flat aft press s first. The EZ/ conal inspection emoved. Floor ys and lavs are 1.1 1.2 (GV) of the aft	8 YR passenger compar rvice door, passenge ead. (EZAP) sure bulkhead. AP inspection requin. panels, sidewall par located. 36000 FC 12 YR passenger compar	8 YR tment - Sta 663.7 ger door, galley or rement with intervenels, and ceiling 36000 FC 12 YR tment - Sta 663.7	5 to the aft pressure lav) - left and right val 36000 FC/12 YR panels removal requested. ALL	e bulkhead we - Section 44 is is uired in ALL
	MRB Perform an in area (within a (part), 46, an NOTE: Not a INTERVAL MACCESS NOTE: MRB Perform an in area (away fin (EZAP)	05-41-02-210-816 Internal zonal inspection approximately 20 inches id 47, Sta 663.75 to aft pupplicable to airplanes with the satisfied by this zone. OTE: Whichever come satisfied by this zones where galley of the satisfied by this zones where galley of the satisfied by this zones.	1.1 1.2 (GV) of the aft from galley se pressure bulkhe th flat aft press s first. The EZ/ conal inspection emoved. Floor ys and lavs are 1.1 1.2 (GV) of the aft avs) - left and r	8 YR passenger compar rvice door, passenge ead. (EZAP) sure bulkhead. AP inspection requin. panels, sidewall par located. 36000 FC 12 YR passenger compar ight - Section 44 (pr	8 YR tment - Sta 663.7 ger door, galley or rement with intervenels, and ceiling 36000 FC 12 YR tment - Sta 663.7	5 to the aft pressure lav) - left and right val 36000 FC/12 YR panels removal requested. ALL	e bulkhead we - Section 44 is is uired in ALL
	MRB Perform an ir area (within a (part), 46, an NOTE: Not a INTERVAL MACCESS NOTE:	05-41-02-210-816 Internal zonal inspection approximately 20 inches id 47, Sta 663.75 to aft pupplicable to airplanes with satisfied by this zonal satisfied by this zonal satisfied by the careas where galley 05-41-02-210-817 Internal zonal inspection rom doors, galleys and laternal zonal inspection rom doors.	1.1 1.2 (GV) of the aft from galley se pressure bulkhe th flat aft press s first. The EZ/ conal inspection emoved. Floor /s and lavs are 1.1 1.2 (GV) of the aft avs) - left and r th flat aft press s first. The EZ/	8 YR passenger compar rvice door, passenge ead. (EZAP) sure bulkhead. AP inspection requinal n. panels, sidewall par located. 36000 FC 12 YR passenger compar right - Section 44 (pasure bulkhead. AP inspection requi	8 YR tment - Sta 663.7 ger door, galley or rement with intervenels, and ceiling 36000 FC 12 YR tment - Sta 663.7 art), 46, and 47, 5	5 to the aft pressure ray) - left and right val 36000 FC/12 YR panels removal requals to the aft pressure 5 to the aft pressure 5 to aft p	e bulkhead we - Section 44 is is uired in ALL e bulkhead dr essure bulkhe
	MRB Perform an in area (within a (part), 46, an NOTE: Not a INTERVAL MACCESS NOTE: NOTE: Not a area (away fi (EZAP) NOTE: Not a INTERVAL MACCESS NOTE: NOTE: Not a INTERVAL MACCESS NOTE: Not a INTERVAL MACCESS NOTE: NOTE	05-41-02-210-816 Internal zonal inspection approximately 20 inches d 47, Sta 663.75 to aft pupplicable to airplanes with states of the satisfied by this zero. The satisfied by the satisfied by this zero. The satisfied by the satisfied by this zero. The satisfied by the satisfied	1.1 1.2 (GV) of the aft from galley se pressure bulkher th flat aft press is first. The EZ/conal inspection emoved. Floor ys and lavs are 1.1 1.2 (GV) of the aft avs) - left and rest in that aft press is first. The EZ/conal inspection	8 YR passenger compar rvice door, passenge ead. (EZAP) sure bulkhead. AP inspection requin. panels, sidewall par located. 36000 FC 12 YR passenger compar right - Section 44 (passenger) sure bulkhead. AP inspection requine.	8 YR tment - Sta 663.7 ger door, galley or rement with intervals, and ceiling 36000 FC 12 YR tment - Sta 663.7 art), 46, and 47, 3	5 to the aft pressure ray) - left and right val 36000 FC/12 YR panels removal requals to the aft pressure 5 to the aft pressure 5 to aft p	e bulkhead we - Section 44 is is uired in ALL e bulkhead dr essure bulkhe

Perform an external zonal inspection (GV) of the aft passenger entry door stops, latches, and hinges - section 47, sta 980.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY	
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
53-892-00-01	MRB	05-41-02-210-819	1.1 1.2	2000 FC 240 DY	2000 FC 240 DY	ALL	ALL	
	Perform an e	external zonal inspection	(GV) of the aft	galley service doo	r stops, latches, a	and hinges - section	47, sta 980.	
	INTERVAL N	NOTE: Whichever come	s first.					
53-894-00-01	MRB	05-41-03-210-801	1.1	120 DY	120 DY	ALL	ALL	
	ground, with	external zonal inspection out the use of stands or lapplicable to airplanes wi	ladders. No ad	ditional access pan		ection is accomplis	hed from the	
53-896-00-01	MRB	05-41-03-210-802	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL	
	Sta 1088. (E NOTE: Not a	applicable to airplanes wi	th flat aft press	sure bulkhead.				
	INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.							
53-898-00-01	MRB	05-41-03-210-803	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL	
53-898-00-01		05-41-03-210-803	1.2	36 MO	36 MO			
53-898-00-01	Perform an i (EZAP)		1.2 (GV) of the sta s first. The EZ/	36 MO bilizer torsion box of the control of the con	36 MO compartment - Se	oction 48, Sta 1088 t	to Sta 1156.	
53-898-00-01 53-900-00-01	Perform an i (EZAP)	nternal zonal inspection	1.2 (GV) of the sta s first. The EZ/	36 MO bilizer torsion box of the control of the con	36 MO compartment - Se	oction 48, Sta 1088 t	to Sta 1156.	
	Perform an i (EZAP) INTERVAL N	nternal zonal inspection NOTE: Whichever come satisfied by this z	1.2 (GV) of the stars s first. The EZ/conal inspection 1.1 1.2	36 MO bilizer torsion box of the control of the con	36 MO compartment - Se rement with inter 5500 FC 30 MO	ction 48, Sta 1088 f val 36000 FC/12 YF ALL	to Sta 1156.	
	Perform an i (EZAP) INTERVAL N MRB Perform an i	nternal zonal inspection NOTE: Whichever come satisfied by this z	1.2 (GV) of the state of the st	36 MO bilizer torsion box of the properties of t	36 MO compartment - Se rement with inter 5500 FC 30 MO ection 48, Sta 10	ction 48, Sta 1088 to Sta 1156. (EZ	to Sta 1156. R is ALL AP)	
	Perform an i (EZAP) INTERVAL N MRB Perform an i	nternal zonal inspection NOTE: Whichever come satisfied by this z 05-41-03-210-804 Internal zonal inspection NOTE: Whichever come	1.2 (GV) of the state of the st	36 MO bilizer torsion box of the properties of t	36 MO compartment - Se rement with inter 5500 FC 30 MO ection 48, Sta 10	ction 48, Sta 1088 to Sta 1156. (EZ	to Sta 1156. R is ALL AP)	
53-900-00-01	Perform an i (EZAP) INTERVAL N MRB Perform an i INTERVAL N	nternal zonal inspection NOTE: Whichever come satisfied by this z 05-41-03-210-804 Internal zonal inspection NOTE: Whichever come satisfied by this z	1.2 (GV) of the states first. The EZ/conal inspection 1.1 1.2 (GV) of the AP is first. The EZ/conal inspection 1.1 1.1 1.1 1.2	36 MO bilizer torsion box of the properties of t	36 MO compartment - Serement with interest 5500 FC 30 MO ection 48, Sta 10 rement with interest 6600 FC 36 MO	ALL 88 to Sta 1156. (EZ	to Sta 1156. R is ALL AP) is	
53-900-00-01	Perform an i (EZAP) INTERVAL N MRB Perform an i INTERVAL N MRB	nternal zonal inspection NOTE: Whichever come satisfied by this z 05-41-03-210-804 Internal zonal inspection NOTE: Whichever come satisfied by this z 05-41-03-210-805	1.2 (GV) of the stars of the st	36 MO bilizer torsion box of the properties of t	36 MO compartment - Serement with inter 5500 FC 30 MO ection 48, Sta 10 rement with inter 6600 FC 36 MO Sta 1156 to Sta	ALL 88 to Sta 1156. (EZ. val 5500 FC/30 MO ALL	ALL AP) is ALL	

Perform an internal zonal inspection (GV) of the flap support no. 4 - left wing - Section 46, Sta 727. (EZAP)

 $\textbf{INTERVAL NOTE:} \ \ \textbf{Whichever comes first.} \ \ \textbf{The EZAP inspection requirement with interval 5500 FC/30 MO is}$

satisfied by this zonal inspection.

ACCESS NOTE: Flaps extended.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
53-906-02-01	MRB	05-41-06-210-820	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL			
	Perform an internal zonal inspection (GV) of the flap support no. 5 - right wing - Section 46, Sta 727. (EZAP)									
	INTERVAL N	NOTE: Whichever come satisfied by this a			irement with inter	val 5500 FC/30 MO	is			
	ACCESS NO	OTE: Flaps extended.								
54-010-01-01	MRB	51-05-01-210-809 54-05-03-210-801	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL			
	Inspect left for	orward engine mount as	sembly, includi	ng fan case fitting,	side links, hangei	r, and link pins.				
	INTERVAL N	NOTE: Whichever come	s first.							
	ACCESS NO	OTE: Open fan cowl.								
54-010-02-01	MRB	51-05-01-210-809 54-05-03-210-802	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL			
	Inspect right	forward engine mount a	ssembly, include	ding fan case fitting	, side links, hang	er, and link pins.				
	INTERVAL N	NOTE: Whichever come	s first.							
	ACCESS NO	OTE: Open fan cowl.								
54-015-01-01	MDD	E4 0E 04 240 900	1.1	0 VD	0 VD	A1.1	A1.1			
54-015-01-01	MRB	51-05-01-210-809 54-05-03-210-803	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL			
	Inspect left s	strut attach bolts at forwa	rd engine mou	nt.						
	INTERVAL N	NOTE: Whichever come	s first.							
	ACCESS NO	OTE: Remove fan cowl.								
54-015-02-01	MRB	51-05-01-210-809 54-05-03-210-804	1.1	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL			
	Inspect right	54-05-03-210-804 1.2 18000 FC 18000 FC Inspect right strut attach bolts at forward engine mount.								
		NOTE: Whichever come	· ·							
		OTE: Remove fan cowl.								
	A00200 III	J. I. Romovo lan cowi.								
54-020-01-01	MRB	51-05-01-210-809 54-05-03-210-805	1.1	ENG CNG		ALL	ALL			
	Inspect forward	ard and aft engine moun	t to strut shear	pins.						
	Engine remo									
		02 -POWER PLANT - R	EMOVAL/INST	ALLATION						
	Related proc	cedures: 01 - FORWARD ENGIN	F MOUNT REM	MOVAL/INISTALLA	TION					
	CIVIIVI I 1-21-	OI - I OIMMAIND FINGIN	L MOONT KE	VIO VAL/IINO IALLAI	1014					

AMM 71-21-03 - AFT ENGINE MOUNT REMOVAL/INSTALLATION

INTERVAL NOTE: At engine removal. ACCESS NOTE: Engine removal required.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
54-020-02-01	MRB	51-05-01-210-809 54-05-03-210-806	1.1	ENG CNG		ALL	ALL
	Inspect forwa	ard and aft engine mount	t to strut shear	pins.			
	Engine remo	val:					
	AMM 71-00-	02 - POWER PLANT - R	EMOVAL/INS	TALLATION			
	Related proc						
		01 - FORWARD ENGINE			ION		
	AIVIIVI /1-21-	03 - AFT ENGINE MOUN	NI REMOVAL/	INSTALLATION			
	INTERVAL N	NOTE: At engine remova	al.				
	ACCESS NO	OTE: Engine removal re	quired.				
54-030-01-01	MRB	51-05-01-210-809	1.1	6 YR	6 YR	ALL	ALL
		54-05-03-210-807	1.2	18000 FC	18000 FC		
		ngine mount assembly, in c pins; hanger and evene			k pins; mount to e	ngine left, center ar	nd right links
	INTERVAL N	NOTE: Whichever come	s first.				
54-030-02-01	MRB	51-05-01-210-809	1.1	6 YR	6 YR	ALL	ALL
		54-05-03-210-808	1.2	18000 FC	18000 FC		
		-	-		k piris, mount to e	ngine left, center ar	nd right links
	including link	opins; hanger and evene NOTE: Whichever come	er bar; attach b		k pins, mount to e	ngine iert, center ar	id right links
54-040-01-01	including link	pins; hanger and evene	er bar; attach b		48 MO 9000 FC	ALL	ad right links
54-040-01-01	including link INTERVAL N MRB	NOTE: Whichever come 51-05-01-210-809	er bar; attach b s first. 1.1 1.2	48 MO 9000 FC	48 MO 9000 FC		_
54-040-01-01	INTERVAL N MRB Inspect strut	NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809	er bar; attach b s first. 1.1 1.2 onal brace, side	48 MO 9000 FC	48 MO 9000 FC		_
54-040-01-01 54-040-02-01	INTERVAL N MRB Inspect strut	NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809 to wing upper link, diago	er bar; attach b s first. 1.1 1.2 onal brace, side s first. 1.1	48 MO 9000 FC	48 MO 9000 FC		_
	MRB Inspect strut	NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809 to wing upper link, diago	er bar; attach b s first. 1.1 1.2 onal brace, side s first.	48 MO 9000 FC e links, and strut att	48 MO 9000 FC tachment fittings.	ALL	ALL
	MRB Inspect strut INTERVAL M	NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809 to wing upper link, diago NOTE: Whichever come	er bar; attach b s first. 1.1 1.2 conal brace, side s first. 1.1 1.2	48 MO 9000 FC e links, and strut att 48 MO 9000 FC	48 MO 9000 FC tachment fittings. 48 MO 9000 FC	ALL	ALL
	MRB Inspect strut MRB Inspect strut INTERVAL M	spins; hanger and evene NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809 to wing upper link, diagon NOTE: Whichever come 51-05-01-210-809 54-05-03-210-810	er bar; attach b s first. 1.1 1.2 onal brace, side s first. 1.1 1.2 onal brace, side	48 MO 9000 FC e links, and strut att 48 MO 9000 FC	48 MO 9000 FC tachment fittings. 48 MO 9000 FC	ALL	ALL
54-040-02-01	MRB Inspect strut INTERVAL M MRB Inspect strut INTERVAL M INTERVAL M	spins; hanger and evener spins; hanger and evener spins; hanger and evener spins; hanger and evener spins; whichever come spins; hanger link, diagonal spins; whichever come spins; spin	1.1 1.2 conal brace, side s first. 1.1 1.2 conal brace, side s first. 1.1 1.2 conal brace, side s first.	48 MO 9000 FC e links, and strut att 48 MO 9000 FC e links, and strut att	48 MO 9000 FC tachment fittings. 48 MO 9000 FC tachment fittings.	ALL	ALL
	MRB Inspect strut MRB Inspect strut INTERVAL M	k pins; hanger and evene NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809 to wing upper link, diago NOTE: Whichever come 51-05-01-210-809 54-05-03-210-810 to wing upper link, diago	er bar; attach b s first. 1.1 1.2 onal brace, side s first. 1.1 1.2 onal brace, side	48 MO 9000 FC e links, and strut att 48 MO 9000 FC	48 MO 9000 FC tachment fittings. 48 MO 9000 FC	ALL	ALL
54-040-02-01	INTERVAL M MRB Inspect strut INTERVAL M MRB Inspect strut INTERVAL M MRB	spins; hanger and evener spins; hanger and evener spins; hanger and evener spins; hanger and evener spins; whichever come spins; whichever come spins; spins	er bar; attach b s first. 1.1 1.2 conal brace, side s first. 1.1 1.2 conal brace, side s first. 1.1 1.2 conal brace, side s first.	48 MO 9000 FC e links, and strut att 48 MO 9000 FC e links, and strut att	48 MO 9000 FC tachment fittings. 48 MO 9000 FC tachment fittings.	ALL	ALL
54-040-02-01	INTERVAL M MRB Inspect strut INTERVAL M MRB Inspect strut INTERVAL M MRB Inspect strut INTERVAL M	spins; hanger and evene NOTE: Whichever come 51-05-01-210-809 54-05-03-210-809 to wing upper link, diagon NOTE: Whichever come 51-05-01-210-809 54-05-03-210-810 to wing upper link, diagon NOTE: Whichever come 51-05-01-210-809 54-05-03-211-801	ar bar; attach b s first. 1.1 1.2 chal brace, side s first. 1.1 1.2 chal brace, side s first. 1.1 1.2 chal brace, side s first.	48 MO 9000 FC e links, and strut att 48 MO 9000 FC e links, and strut att	48 MO 9000 FC tachment fittings. 48 MO 9000 FC tachment fittings.	ALL	ALL
54-040-02-01	INTERVAL M MRB Inspect strut INTERVAL M MRB Inspect strut INTERVAL M MRB Inspect strut INTERVAL M	spins; hanger and evener spins; hanger and evener spins; hanger and evener spins; hanger and evener spins; whichever come spins; whichever come spins; spins	ar bar; attach b s first. 1.1 1.2 chal brace, side s first. 1.1 1.2 chal brace, side s first. 1.1 1.2 chal brace, side s first.	48 MO 9000 FC e links, and strut att 48 MO 9000 FC e links, and strut att	48 MO 9000 FC tachment fittings. 48 MO 9000 FC tachment fittings.	ALL	ALL

Inspect pins and fuse pins on upper link, midspar, diagonal brace, and side links. Pin removal is not required.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	BILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
54-060-01-01	MRB	51-05-01-210-809 54-05-03-211-803	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	Inspect the bores of pins and fuse pins on upper link, midspar, diagonal brace, and side links. Pin removal is not						s not required
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: TR's must be oper	n to remove ac	cess panels 431EL	and 431ER.		
54-060-02-01	MRB	51-05-01-210-809 54-05-03-211-804	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	Inspect the b	ores of pins and fuse pi	ns on upper lin	k, midspar, diagona	al brace, and side	links. Pin removal is	s not required
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: TR's must be oper	n to remove ac	cess panels 441EL	and 441ER.		
54-070-01-01	MRB	51-05-01-210-803 54-05-03-210-811	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL
	Inspect exterskins.	rnal areas of strut box, in	cluding upper	and lower spars, fo	rward engine mo	unt bulkhead, aft bu	lkhead, and s
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	PTE: Remove MID and a remove access pa		blanket/heat shield d 431ER. Engine re			open to
54-070-02-01	MRB	51-05-01-210-803 54-05-03-210-812	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL
	Inspect exter	rnal areas of strut box, in	cluding upper	and lower spars, fo	rward engine mo	unt bulkhead, aft bu	lkhead, and s
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	PTE: Remove MID and remove access pa		blanket/heat shield d 441ER. Engine re			open to
54-080-01-01	MRB	51-05-01-210-803 54-05-03-210-813	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL
		nal areas of strut box, included and side skins.	cluding upper a	and lower spars, for	ward and aft eng	ine mount bulkhead	s, aft and mid
	INTERVAL N	NOTE: Whichever come	s first.				
		NOTE: Whichever come OTE: Disassemble pneu and 431ER.		required. TR's mus	st be open to rem	ove access panels	431EL
54-080-02-01		OTE: Disassemble pneu		required. TR's must 9 YR 18000 FC	st be open to rem 9 YR 18000 FC	ove access panels of	431EL ALL

bulkheads, and side skins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Disassemble pneumatic ducts as required. TR's must be open to remove access panels 441EL

and 441ER.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
54-090-01-01	MRB	54-55-01-720-801 54-55-02-100-802	1.1 1.2	30 MO 9000 FC	30 MO 9000 FC	ALL	ALL
	Functionally	check the left engine for	ward strut and	aft strut fairing drai	ns.		
	INTERVAL N	NOTE: Whichever occur	s first.				
54-090-02-01	MRB	54-55-01-720-801 54-55-02-100-802	1.1 1.2	30 MO 9000 FC	30 MO 9000 FC	ALL	ALL
	Functionally	check the right engine fo	orward strut an	d aft strut fairing dra	ains.		
	INTERVAL N	NOTE: Whichever occur	s first.				
54-600-00-01	AWL	54-05-02-210-801	1.1	56000 FC	6000 FC	ALL	ALL
	Inspect (Gen	neral Visual) the lugs and	l clevises for al	I the links, fittings a	nd pins.		
		26A001-DTR, DTR chec -51-09, 54-51-15 for alte			-03, 54-51-04, 54	-51-05, 54-51-06, 5 ₉	4-51-07,
54-600-00-02	AWL	54-05-02-210-801	1.1	56000 FC	6000 FC	ALL	ALL
		26A001-DTR, DTR chec					
		-51-09, 54-51-15 for alte	·	tions.			
54-610-00-01	AWL	54-05-02-250-801	1.1	56000 FC	18000 FC	ALL	ALL
54-610-00-01	AWL Inspect (High Nacelle STA nacelle STA and right har See Doc. D6 The NDI met (D6-37239).	· 	1.1 nt) both legs of ht hand chords, thand chords, 43.5-250.6 left ck form 54-51-7 omplish the interes are contain.	56000 FC If the lower spar chos, nacelle STA 224.7- and right hand cho 10, for alternative in ent of this inspection ed in Part 6, Subjection	18000 FC ords between the -231.8 left hand coords. ispections. on is contained in the total section of the total s	ALL forward and aft eng hord, nacelle STA 2	ALL ine mounts:
54-610-00-01 54-610-00-02	AWL Inspect (High Nacelle STA nacelle STA and right har See Doc. D6 The NDI met (D6-37239).	54-05-02-250-801 The Frequency Eddy Currer 203.6 -209.9 left and right 212.3-222.0 left and right d chords, nacelle STA 2 26A001-DTR, DTR check chod(s) necessary to accommodity	1.1 nt) both legs of ht hand chords, thand chords, 43.5-250.6 left ck form 54-51-complish the interes are contain.	56000 FC If the lower spar chos, nacelle STA 224.7- and right hand cho 10, for alternative in ent of this inspection ed in Part 6, Subjection	18000 FC ords between the -231.8 left hand coords. ispections. on is contained in the total section of the total s	ALL forward and aft eng hord, nacelle STA 2	ALL ine mounts:
	AWL Inspect (High Nacelle STA nacelle STA and right har See Doc. D6 The NDI met (D6-37239). ACCESS NC AWL Inspect (High Nacelle STA and right har See Doc. D6 The NDI met (D6-37239).	54-05-02-250-801 In Frequency Eddy Currer 203.6 -209.9 left and right 212.3-222.0 left and right and chords, nacelle STA 2 (26A001-DTR, DTR check (26A005) necessary to acc (26A06) The inspection procedur (26A06) Remove/displace for the inspection procedur (26A06) Remove/displace for the inspection procedur (26A06) TE: Remove/displace for the inspection procedur (26A06) Remove/displace for the inspection procedur (26A06) Remove/displace for the inspection procedur (26A06) Remove/displace for the inspection procedure (26A06) Remove/displace for the inspecti	1.1 nt) both legs of ht hand chords, thand chords, 43.5-250.6 left ck form 54-51-omplish the interest shields and 1.1 nt) both legs of ht hand chords, 43.5-250.6 left ck form 54-51-omplish the interest are containing than the chords and the chords are containing the same containing t	56000 FC If the lower spar chose, nacelle STA 224.7- and right hand chosello, for alternative in ent of this inspection and part 6, Subject of the lower spar chosello, for alternative in ent of this inspection and right hand chosello, for alternative in ent of this inspection ent of this inspection ent of this inspection ent of this inspection and right for alternative in ent of this inspection ent of this inspection ent of this inspection and right for alternative in ent of this inspection ent of t	18000 FC ords between the -231.8 left hand coords. spections. on is contained in at 54-40-03. ired. 18000 FC ords between the -231.8 left hand coords. spections. on is contained in at 54-40-03.	ALL forward and aft eng shord, nacelle STA 2 the 737 Nondestruct ALL forward and aft eng	ALL ine mounts: 234.4-240.4 lef
	AWL Inspect (High Nacelle STA nacelle STA and right har See Doc. D6 The NDI met (D6-37239). ACCESS NC AWL Inspect (High Nacelle STA and right har See Doc. D6 The NDI met (D6-37239).	54-05-02-250-801 The Frequency Eddy Currer 203.6 -209.9 left and right and chords, nacelle STA 2 226A001-DTR, DTR check and chod(s) necessary to accommodate the commodate of t	1.1 nt) both legs of ht hand chords, thand chords, 43.5-250.6 left ck form 54-51-omplish the interest shields and 1.1 nt) both legs of ht hand chords, 43.5-250.6 left ck form 54-51-omplish the interest are containing than the chords and the chords are containing the same containing t	56000 FC If the lower spar chose, nacelle STA 224.7- and right hand chosello, for alternative in ent of this inspection and part 6, Subject of the lower spar chosello, for alternative in ent of this inspection and right hand chosello, for alternative in ent of this inspection ent of this inspection ent of this inspection ent of this inspection and right for alternative in ent of this inspection ent of this inspection ent of this inspection and right for alternative in ent of this inspection ent of t	18000 FC ords between the -231.8 left hand coords. spections. on is contained in at 54-40-03. ired. 18000 FC ords between the -231.8 left hand coords. spections. on is contained in at 54-40-03.	ALL forward and aft eng shord, nacelle STA 2 the 737 Nondestruct ALL forward and aft eng	ALL ine mounts: 234.4-240.4 let have been mounts: 234.4-240.4 let have been mounts: 234.4-240.4 let

nacelle STA 207.8 for both the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
54-611-00-02	AWL	54-05-02-250-802	1.1	56000 FC	18000 FC	ALL	ALL
	nacelle STA See Doc. D6 The NDI me	h Frequency Eddy Curre 207.8 for both the left ar 326A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	nd right hand cl ck form 54-51- complish the int	nords. 10, for alternative in ent of this inspection	nspections.		
54-611-01-01	AWL	54-05-02-130-801	1.1	56000 FC	18000 FC	ALL	ALL
	on the left ar	asonic) the lower spar cl nd right hand chords. 326A001-DTR, DTR chec		·		ΓA 203.4 and at nac	elle STA 207.8
54-611-01-02	AWL	54-05-02-130-801	1.1	56000 FC	18000 FC	ALL	ALL
	on the left ar	asonic) the lower spar cl nd right hand chords. 326A001-DTR, DTR chec				ΓA 203.4 and at nac	elle STA 207.8
54-612-00-01	AWL	54-05-02-130-802	1.1	56000 FC	18000 FC	ALL	ALL
		DTE: Removal of insular required.		•		ıst reversers, and eı	ngines as
54-612-00-02	AWL	54-05-02-130-802	1.1	56000 FC	18000 FC	ALL	ALL
	right hand si See Doc. D6	326A001-DTR, DTR checords: Removal of insula	ck form 54-51-	10, for alternative re	epeat inspection.		
		required.					
54-613-00-01	AWL	54-05-02-130-803	1.1	56000 FC	18000 FC	ALL	ALL
	nacelle STA See Doc. D6	asonic) the lower spar cl 242.7 on the left and rigi 326A001-DTR, DTR chec DTE: Remove fan cowls	ht chords. ck form 54-51-	10, for alternative ir	nspections.		ght chords and
54-613-00-02	AWL	54-05-02-130-803	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (Ultr	asonic) the lower spar ch 242.7 on the left and rigi	nord at the fran				

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

ACCESS NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shields as required.







	INTERVAL					APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
54-614-00-01	AWL	54-05-02-250-803	1.1	56000 FC	18000 FC	ALL	ALL		
	See Doc. Do	h Frequency Eddy Curre 626A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 54-51-1 complish the int	 for alternative reent of this inspection 	epeat inspection. on is contained in		ctive Test Manu		
54-614-00-02	AWL	54-05-02-250-803	1.1	56000 FC	18000 FC	ALL	ALL		
	See Doc. Do	h Frequency Eddy Curre 626A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	ck form 54-51-1 complish the int	 for alternative reent of this inspection 	epeat inspection. on is contained in		ctive Test Manu		
54-614-01-01	AWL	54-05-02-250-804	1.1	56000 FC	18000 FC	ALL	ALL		
	The NDI me (D6-37239).	626A001-DTR, DTR check thod(s) necessary to accommodate inspection procedure. The inspection requires	complish the intres are contained	ent of this inspection ed in Part 6, Subject	on is contained in	the 737 Nondestruc	ctive Test Manu		
54-614-01-02	AWL	54-05-02-250-804	1.1	56000 FC	18000 FC	ALL	ALL		
	See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre 526A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Inspection requires	ck form 54-51-1 complish the intres are contained	10, for alternative in ent of this inspection ed in Part 6, Subject	nspections. on is contained in		ctive Test Manu		
54-615-00-01	AWL	54-05-02-250-805	1.1	56000 FC	9000 FC	ALL	ALL		
	See Doc. Do The NDI me (D6-37239).	Inspect (High Frequency Eddy Current) only the lower right spar web from nacelle STA 224.7 to nacelle STA 231.8. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03. ACCESS NOTE: Open/Remove Thrust Reverser as Required. Removal of Pneumatic Ducting Required.							
54-615-00-02	AWL	54-05-02-250-805	1.1	56000 FC	9000 FC	ALL	ALL		
		h Freguency Eddy Curre			OTA /	004 7 to			

Inspect (High Frequency Eddy Current) only the lower right spar web from nacelle STA 224.7 to nacelle STA 231.8.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Open/Remove Thrust Reverser as Required. Removal of Pneumatic Ducting Required.





			APPLICA	ABILITY							
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
54-616-00-01	AWL	54-05-02-250-806	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (High	r Frequency Eddy Curre	nt) only the low	ver right spar chord	from nacelle STA	A 224.7 to nacelle S	TA 231.8.				
		26A001-DTR, DTR ched		•	•						
		hod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	ctive Test Manu				
	ACCESS NO	OTE: Remove thrust rev	erser as requir	ed. Removal of ins	ulation heat shiel	ds is required.					
54-616-00-02	AWL	54-05-02-250-806	1.1	56000 FC	18000 FC	ALL	ALL				
	See Doc. D6 The NDI met	n Frequency Eddy Curre 26A001-DTR, DTR chech hod(s) necessary to acc The inspection procedur	ck form 54-51- complish the int	10, for alternative in ent of this inspection	nspections. on is contained in						
	,	TTE: Remove thrust rev				ds is required.					
54-617-00-01	AWL	54-05-02-250-807	1.1	56000 FC	18000 FC	ALL	ALL				
	See Doc. D6 The NDI met	celle STA 244.9. 26A001-DTR, DTR chechod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Manı				
	,	OTE: Remove thrust rev				ds is required.					
54-617-00-02	AWL	54-05-02-250-807	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (High Frequency Eddy Current) the horizontal leg of the left and right hand chords, common to the compression particles at nacelle STA 244.9.										
	See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.										
	The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.										
	ACCESS NO	OTE: Remove thrust rev	erser as requir	ed. Removal of ins	ulation heat shiel	ds is required.					
54-617-01-01	AWL	54-05-02-130-804	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (Ultra	Inspect (Ultrasonic) the horizontal leg of the left and right hand chords, common to the compression pad bracket at nacelle STA 244.9.									
	See Doc. D6	26A001-DTR, DTR ched	ck form 54-51-	10, for alternative ir	nspections.						
	ACCESS NO	OTE: Remove thrust rev	erser as requir	ed. Removal of ins	ulation heat shiel	d is required.					
54-617-01-02	AWL	54-05-02-130-804	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (Ultra	asonic) the horizontal leg	g of the left and	I right hand chords,	, common to the o	compression pad bra	acket at nacelle				

ACCESS NOTE: Remove thrust reverser as required. Removal of insulation heat shield is required.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.



54-619-01-02



737-600/700/800/900 TASK CARDS

				APPLICA	ABILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
54-618-00-01	AWL	54-05-02-250-808	1.1	56000 FC	18000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the vertical leg of the compression pad bracket on the lower left and right spar chords at nacelle STA 244.9. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manu (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.									
54-618-00-02	AWL	54-05-02-250-808	1.1	56000 FC	18000 FC	ALL	ALL			
	chords at na See Doc. D6 The NDI me	h Frequency Eddy Curre celle STA 244.9. 626A001-DTR, DTR checthod(s) necessary to acc The inspection procedur	ck form 54-51-1 omplish the int	10, for alternative ir ent of this inspection	nspections.					
54-619-00-01	AWL	54-05-02-130-805	1.1	56000 FC	9000 FC	ALL	ALL			
	Inspect (Ultrasonic) the internal side of the vertical leg at nacelle STA 209.0 - 212.3, nacelle STA 231.8 - 234.4 on the left and right hand chords. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.									
54-619-00-02	AWL	54-05-02-130-805	1.1	56000 FC	9000 FC	ALL	ALL			
	and right har	asonic) the internal side nd chords. 326A001-DTR, DTR chec			·	celle STA 231.8 - 23	4.4 on the left			
54-619-01-01	AWL	54-05-02-250-809	1.1	56000 FC	9000 FC	ALL	ALL			
	231.8 - 234.4 See Doc. D6 The NDI met (D6-37239).	r Frequency Eddy Currer 4 on the left and right had 526A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur DTE: Remove thrust rev	nd chords. ck form 54-51-1 omplish the int res are contain	10, for alternative ir ent of this inspection ed in Part 6, Subject	nspections. on is contained in					

AWL 54-05-02-250-809 1.1 56000 FC 9000 FC ALL ALL

Inspect (Low Frequency Eddy Current) the external side of the vertical leg at nacelle STA 209.0 - 212.3, and nacelle STA

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Remove thrust reversers as required.

231.8 - 234.4 on the left and right hand chords.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
54-620-00-01	AWL	54-05-02-250-810	1.1	56000 FC	18000 FC	ALL	ALL			
	212.3 and from See Doc. D6 The NDI me (D6-37239).	h Frequency Eddy Curre om nacelle STA 231.8 to 326A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	nacelle STA 23 ck form 54-51-1 omplish the int es are containe	34.4. 0, for alternative in ent of this inspecticed in Part 6, Subjection	nspections. on is contained in					
	ACCESS NO	OTE: Remove thrust rev	ersers as requi	rea.						
54-620-00-02	AWL	54-05-02-250-810	1.1	56000 FC	18000 FC	ALL	ALL			
	See Doc. D6 The NDI me (D6-37239).	om nacelle STA 231.8 to 326A001-DTR, DTR checthod(s) necessary to accente inspection procedure. Remove thrust rev	ck form 54-51-1 omplish the int es are containe	0, for alternative in ent of this inspection and in Part 6, Subjection	on is contained in	the 737 Nondestruc	ctive Test Manua			
54-621-00-01	AWL	54-05-02-211-801	1.1	56000 FC	18000 FC	ALL	ALL			
	Inspect (Detailed) the lower spar chord aft of the aft engine mount bulkhead. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections. ACCESS NOTE: Remove thrust reversers as required. Removal of insulation heat shields required.									
54-621-00-02	AWL	54-05-02-211-801	1.1	56000 FC	18000 FC	ALL	ALL			
	See Doc. D6	ailed) the lower spar cho 326A001-DTR, DTR chec		•						
	7100200 111	OTE: Remove thrust rev	ersers as requi	red. Removal of in	sulation heat shie	lds required.				
54-622-00-01	AWL	54-05-02-250-811	ersers as requi	red. Removal of in	sulation heat shie	lds required. ALL	ALL			
54-622-00-01	AWL Inspect (High See Doc. Do The NDI me (D6-37239).		1.1 nt) the end pace of form 54-51-1 omplish the interest are contained smust be sequent	56000 FC I bolt holes (4 locat 1, for alternative in ent of this inspectic ed in Part 6, Subject entially (one at a ti	36000 FC ions) at the FWD spections. on is contained in ct 54-40-03. me) removed for	ALL engine mount bulkt	nead. ctive Test Manua			

Inspect (High Frequency Eddy Current) the end pad bolt holes (4 locations) at the FWD engine mount bulkhead.

See Doc. D626A001-DTR, DTR check form 54-51-11, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Bathtub fitting bolts must be sequentially (one at a time) removed for bolt hole eddy current inspection. Retorque tension bolts per dwg requirements.





				APPLICA	ABILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
54-623-00-01	AWL	54-05-02-250-812	1.1	56000 FC	18000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the FWD and Aft flanges of both tension fittings common to the R1 fitting (4) attachment bolts. See Doc. D626A001-DTR, DTR check form 54-51-14, for alternative inspections.									
		•		14, for alternative if	ispections.					
	ACCESS NO	OTE: Internal access red	quirea.							
54-623-00-02	AWL	AWL 54-05-02-250-812 1.1 56000 FC 18000 FC ALL ALL								
	attachment b		,	-	_	common to the R1	fitting (4)			
		26A001-DTR, DTR chec		14, for afternative in	ispections.					
	ACCESS NO	OTE: Internal access red	quirea.							
54-624-00-01	AWL	54-05-02-250-813	1.1	56000 FC	18000 FC	ALL	ALL			
	See Doc. D6	270 left and right hand s 26A001-DTR, DTR chec DTE: Remove thrust rev	ck form 54-51-1		nspections.					
54-624-00-02	AWL	54-05-02-250-813	1.1	56000 FC	18000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) all exposed side skin surfaces within 4.5 inches of cutout at nacelle STA 252 and nacelle STA 270 left and right hand sides. See Doc. D626A001-DTR, DTR check form 54-51-16, for alternative inspections.									
	ACCESS NO	OTE: Remove thrust rev	ersers as requ	ired.	•					
54-625-00-01	AWL	54-05-02-130-806	1.1	56000 FC	36000 FC	ALL	ALL			
		asonic) the strut side skin 26A001-DTR, DTR chec			•	elle STA 242.7, left	and right side			
54-625-00-02	AWL	54-05-02-130-806	1.1	56000 FC	36000 FC	ALL	ALL			
		asonic) the strut side skin 26A001-DTR, DTR chec		•		elle STA 242.7, left	and right side			
54-626-00-01	AWL	54-05-02-250-814	1.1	56000 FC	9000 FC	ALL	ALL			
		r Frequency Eddy Curre								

Inspect (High Frequency Eddy Current) the upper spar chord between the forward and aft engine mounts: Nacelle STA 200.9 - 211.5 on the left chords, nacelle STA 213.6 - 225.2 on the left chords, nacelle STA 226.1 - 233.6 on the left chords, nacelle STA 236.3 - 241.8 on the left and right chords.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
54-626-00-02	AWL	54-05-02-250-814	1.1	56000 FC	9000 FC	ALL	ALL
		F	() ()			<i>c</i>	

Inspect (High Frequency Eddy Current) the upper spar chord between the forward and aft engine mounts: Nacelle STA 200.9 - 211.5 on the left chords, nacelle STA 213.6 - 225.2 on the left chords, nacelle STA 226.1 - 233.6 on the left chords, nacelle STA 236.3 - 241.8 on the left and right chords.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

54-627-00-01

AWL 54-05-02-250-815

1.1

56000 FC

9000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

54-627-00-02

AWL

54-05-02-250-815

1.1

56000 FC

9000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

54-627-01-01

AWL 54-05-02-250-816

1.1

56000 FC

9000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - Nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

54-627-01-02

AWL

54-05-02-250-816

1.1

1.1

56000 FC

9000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - Nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

54-628-00-01

AWL 54-05-02-250-817

56000 FC

18000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar web on the (horizontal) chord only near the cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.





			APPLICABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
54-628-00-02	AWL	54-05-02-250-817	1.1	56000 FC	18000 FC	ALL	ALL		
	cutouts: Nac 233.6. See Doc. D6	h Frequency Eddy Curre celle STA 200.9 to nacello 626A001-DTR, DTR chec thod(s) necessary to acc	e STA 211.5, na	acelle STA 213.6 to	nacelle STA 225	2, nacelle STA 226	.1 to nacelle ST		
	(D6-37239).	The inspection procedure	res are contain	ed in Part 6, Subjec	ct 54-40-07.				
54-628-01-01	AWL	54-05-02-250-818	1.1	56000 FC	18000 FC	ALL	ALL		
	cutouts: Nac 233.6. See Doc. D6 The NDI me	h Frequency Eddy Curre celle STA 200.9 to nacello 626A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur	e STA 211.5, na ck form 54-51-7 complish the int	acelle STA 213.6 to 17, for alternative ir ent of this inspection	nacelle STA 225 aspections. on is contained in	.2, nacelle STA 226	.1 to nacelle ST		
54-628-01-02	AWL	54-05-02-250-818	1.1	56000 FC	18000 FC	ALL	ALL		
	cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.								
54-629-00-01	AWL	54-05-02-250-819	1.1	56000 FC	18000 FC	ALL	ALL		
	200.9 - to na	h Frequency Eddy Curre acelle STA 211.5, nacelle 326A001-DTR, DTR chec	STA 213.6 - to	nacelle STA 225.2	, nacelle STA 226				
54-629-00-02	AWL	54-05-02-250-819	1.1	56000 FC	18000 FC	ALL	ALL		
	200.9 - to na	h Frequency Eddy Curre acelle STA 211.5, nacelle 326A001-DTR, DTR chec	STA 213.6 - to	nacelle STA 225.2	, nacelle STA 226				
54-630-00-01	AWL	54-05-02-250-820	1.1	56000 FC	9000 FC	ALL	ALL		
	outside of th	Frequency Eddy Currel e strut, and all structure 226A001-DTR, DTR chec	buried in the sp	olices using low free	quency subsurfac		ide and the		
54-630-00-02	AWL	54-05-02-250-820	1.1	56000 FC	9000 FC	ALL	ALL		
		0.0002200020	111	30000.0		,	,,,,,		

Inspect (Low Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inside and the outside of the strut, and all structure buried in the splices using low frequency subsurface eddy current. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.







				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
54-630-01-01	AWL	54-05-02-250-821	1.1	56000 FC	9000 FC	ALL	ALL				
	outside of the	n Frequency Eddy Curren e strut, and all visible stru 26A001-DTR, DTR chec	ucture in the sp	olices using high fre	equency surface	•	ide and the				
54-630-01-02	AWL	54-05-02-250-821	1.1	56000 FC	9000 FC	ALL	ALL				
	outside of the	n Frequency Eddy Currer e strut, and all visible stru 26A001-DTR, DTR chec	ucture in the sp	olices using high fre	equency surface	•	ide and the				
54-631-00-01	AWL	54-05-02-250-822	1.1	56000 FC	18000 FC	ALL	ALL				
		Inspect (High Frequency Eddy Current) the left and right upper spar chords at nacelle STA 222.6 and nacelle STA 210.6. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.									
54-631-00-02	AWL	54-05-02-250-822	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (High Frequency Eddy Current) the left and right upper spar chords at nacelle STA 222.6 and nacelle STA 210.6. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.										
54-632-00-01	AWL	54-05-02-130-807	1.1	56000 FC	18000 FC	ALL	ALL				
		asonic) the hidden portio 26A001-DTR, DTR chec				the left and right cho	ords.				
54-632-00-02	AWL	54-05-02-130-807	1.1	56000 FC	18000 FC	ALL	ALL				
		asonic) the hidden portio 26A001-DTR, DTR chec				the left and right cho	ords.				
54-632-01-01	AWL	54-05-02-250-823	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (High Frequency Eddy Current) the visible portion of the upper spar chord at nacelle STA 242.7 on the left and right chords. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.										
54-632-01-02	AWL	54-05-02-250-823	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (High chords.	n Frequency Eddy Currer 26A001-DTR, DTR chec	nt) the visible p	oortion of the upper	spar chord at na						
54-633-00-01	AWL	54-05-02-250-824	1.1	56000 FC	9000 FC	ALL	ALL				

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.





			APPLICA	ABILITY							
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
54-633-00-02	AWL	54-05-02-250-824	1.1	56000 FC	9000 FC	ALL	ALL				
	Inspect (High Frequency Eddy Current) all exposed surfaces of the R3/R4 first fastener row on the vertical leg right and left sides. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.										
54-634-00-01	AWL	54-05-02-230-801	1.1	56000 FC	18000 FC	ALL	ALL				
	See Doc. D6	etrant) the entire forward 226A001-DTR, DTR ched	ck form 54-55-0	03, for alternative in		t hole detail at the to	op of the mount				
	ACCESS NO	OTE: Removal of engine	e and engine m	ount is required.							
54-634-00-02	AWL	54-05-02-230-801	1.1	56000 FC	18000 FC	ALL	ALL				
	See Doc. D6	etrant) the entire forward 26A001-DTR, DTR checontributes the control of engine cont	ck form 54-55-0	03, for alternative in		t hole detail at the to	op of the moun				
54-635-00-01	AWL	54-05-02-210-802	1.1	56000 FC	6000 FC	ALL	ALL				
	Inspect (General Visual) the thrust link and the thrust link clevis lug. Lead crack is the failed thrust link. Critical detail is the intact thrust link clevis lug. See Doc. D626A001-DTR, DTR check form 54-55-05, for alternative inspections.										
54-635-00-02	AWL	54-05-02-210-802	1.1	56000 FC	6000 FC	ALL	ALL				
	Inspect (General Visual) the thrust link and the thrust link clevis lug. Lead crack is the failed thrust link. Critical detail is the intact thrust link clevis lug. See Doc. D626A001-DTR, DTR check form 54-55-05, for alternative inspections.										
54-636-00-01	AWL	54-05-02-210-803	1.1	56000 FC	6000 FC	ALL	ALL				
	Lead crack is	neral Visual) the thrust lir s the failed thrust link pir 26A001-DTR, DTR chec	n. Critical detail		ŭ						
54-636-00-02	AWL	54-05-02-210-803	1.1	56000 FC	6000 FC	ALL	ALL				
	Inspect (General Visual) the thrust link pin. Lead crack is the failed thrust link pin. Critical detail is the intact thrust link clevis lug. See Doc. D626A001-DTR, DTR check form 54-55-06, for alternative inspections.										
54-637-00-01	AWL	54-05-02-230-802	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (Pen	etrant) the entire aft eng	ine mount ass	embly.							

The aft mount critical detail is the shear pin hole.

See Doc. D626A001-DTR, DTR check form 54-55-10, for alternative inspections.

ACCESS NOTE: The inspection requires the removal of the engine and disassembly of the engine mount.





			APPLICA	ABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
54-637-00-02	AWL									
		etrant) the entire aft eng		embly.						
	The aft mount critical detail is the shear pin hole. See Doc. D626A001-DTR, DTR check form 54-55-10, for alternative inspections.									
		TE: The inspection req		•	·	f the engine mount				
	ACCESS NO	TIE. THE INSPECTION TEQ	ulles the remo	ival of the engine at	nd disassembly o	i the engine mount.				
54-638-00-01	AWL 54-05-02-210-804 1.1 56000 FC 6000 FC ALL ALL									
	Inspect (Gen	eral Visual) the evener b	ar assembly o	outboard lugs.						
		s the failed evener bar (o	٠,			levis lug.				
		26A001-DTR, DTR chec		•	•					
	ACCESS NO	OTE: The inspection req	uires the remo	val of the engine ar	nd disassembly o	the engine mount.				
54-638-00-02	AWL	54-05-02-210-804	1.1	56000 FC	6000 FC	ALL	ALL			
	Inspect (Gen	eral Visual) the evener b	ar assembly o	outboard lugs.						
		s the failed evener bar (o	٠,			levis lug.				
	See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.									
	ACCESS NOTE: The inspection requires the removal of the engine and disassembly of the engine mount.									
54-639-00-01	AWL	54-05-02-230-803	1.1	56000 FC	75000 FC	ALL	ALL			
	Inspect (Penetrant) the entire aft engine mount evener bar.									
	See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.									
	ACCESS NO	OTE: Inspection requires	the removal a	and thorough cleani	ing of the evener	bar.				
54-639-00-02	AWL	54-05-02-230-803	1.1	56000 FC	75000 FC	ALL	ALL			
04 000 00 02										
	Inspect (Penetrant) the entire aft engine mount evener bar. See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.									
	ACCESS NO	OTE: Inspection requires	the removal a	and thorough cleani	ng of the evener	bar.				
54-640-00-01	AWL	54-05-02-700-801	1.1	56000 FC	6000 FC	ALL	ALL			
	Verify (Torque Check) all strut attach bolts on the forward and aft mounts.									
	See Doc. D6			13. for alternative in	ispections.					
		26A001-DTR, DTR chec	K form 54-55-	,	•					
54-640-00-02	AWL	26A001-DTR, DTR chec 54-05-02-700-801	1.1	56000 FC	6000 FC	ALL	ALL			
54-640-00-02		· 	1.1	56000 FC	6000 FC	ALL	ALL			
54-640-00-02	Verify (Torqu	54-05-02-700-801	1.1 bolts on the fo	56000 FC	6000 FC	ALL	ALL			
54-640-00-02 54-640-01-01	Verify (Torqu	54-05-02-700-801 e Check) all strut attach	1.1 bolts on the fo	56000 FC	6000 FC	ALL	ALL			

See Doc. D626A001-DTR, DTR check form 54-55-13 for alternative repeat inspection.

ACCESS NOTE: Removal of engine and engine mounts is required.







			APPLICABILITY								
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
54-640-01-02	AWL	54-05-02-230-804	1.1	56000 FC	75000 FC	ALL	ALL				
	Inspect (Pen	etrant) the strut attach b	olts on the for	ward and aft moun	ts.						
	See Doc. D626A001-DTR, DTR check form 54-55-13 for alternative repeat inspection.										
	ACCESS NO	OTE: Removal of engine	and engine m	ounts is required.							
54-800-01-01	MRB	05-41-04-210-811	1.1	9000 FC	9000 FC	ALL	ALL				
	1.2 36 MO 36 MO										
	Perform an ir	Perform an internal zonal inspection (GV) of the forward strut fairing - engine no. 1. (EZAP)									
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	/al 18000 FC/6 YR	is				
54-802-01-01	MRB	05-41-04-210-812	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an ir	nternal zonal inspection	(GV) of the fan	cowl support bear	n - engine no. 1. (EZAP)					
	INTERVAL N	IOTE : Whichever come satisfied by this z			irement with inter	/al 18000 FC/6 YR	is				
54-804-01-01	MRB	05-41-04-210-813	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Perform an internal zonal inspection (GV) of the strut torque box - engine no. 1. (EZAP)										
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	/al 18000 FC/6 YR	is				
54-806-01-01	MRB	05-41-04-210-814	1.1 1.2	9000 FC 36 MO	9000 FC 36 MO	ALL	ALL				
	Perform an ir	nternal zonal inspection	(GV) of the aft	strut fairing - engin	e no. 1. (EZAP)						
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	/al 12000 FC/4 YR	is				
54-808-02-01	MRB	05-41-04-210-815	1.1 1.2	9000 FC 36 MO	9000 FC 36 MO	ALL	ALL				
	Perform an in	nternal zonal inspection	(GV) of the for	ward strut fairing -	engine no. 2. (EZ	AP)					
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with interv	/al 18000 FC/6 YR	is				
	MRB	05-41-04-210-816	1.1	6600 FC	6600 FC	ALL	ALL				

Perform an internal zonal inspection (GV) of the fan cowl support beam - engine no. 2. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.







				INTERVAL		APPLICA	LICABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
54-812-02-01	MRB	05-41-04-210-817	1.1 1.2	18000 FC 6 YR	18000 FC 6 YR	ALL	ALL				
	Perform an i	nternal zonal inspection	(GV) of the stru	ut torque box - eng	ine no. 2. (EZAP)						
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with interv	/al 18000 FC/6 YR i	S				
54-814-02-01	MRB 05-41-04-210-818 1.1 9000 FC 9000 FC ALL ALL 1.2 36 MO 36 MO										
	Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP)										
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with interv	/al 12000 FC/4 YR i	S				
55-010-00-01	MRB	51-05-01-210-809 55-05-03-210-801	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	Inspect horiz	ontal stabilizer center se	ection jackscrev	w fitting.							
	INTERVAL NOTE: Whichever comes first.										
55-020-00-01	MRB	51-05-01-210-804 55-05-03-210-802	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	Inspect horizontal stabilizer center section front spar, rear spar, pivot fittings, and primary and secondary thrust beams.										
	INTERVAL NOTE: Whichever comes first.										
	ACCESS NO	OTE: Adjust stabilizer tri to inspect pivot fitti	•	Remove gap seal	and rear spar slid	ing seal					
55-030-00-01	MRB	51-05-01-210-806 55-05-03-210-803	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	Inspect the forward side of the vertical fin front spar, including front spar chords, webs, and terminal fittings.										
	INTERVAL N	NOTE: Whichever come	s first.								
55-050-00-01	MRB	51-05-01-210-804 55-05-03-210-804	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	•	Inspect vertical fin from front spar to rear spar, including spar chords and webs, inspar skins, lower closure rib, and rear sparterminal fittings.									
	INTERVAL N	NOTE: Whichever come	s first.								
	ACCESS NO	OTE: Remove upper ren	novable web. F	Pin removal is not re	equired for termin	al fitting					
		inspection.									
55-060-00-01	MRB	51-05-01-210-806 55-05-03-210-805	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-070-01-01	MRB	51-05-01-210-804 55-05-03-210-806	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect forwa	ard side of front spar, inc	luding front sp	ar chords and web	s, terminal fittings	, and center section	front spar lug			
	INTERVAL NOTE: Whichever comes first.									
	ACCESS NO	OTE: Bolt removal is required spar lugs.	uired to insped	ct terminal fittings a	and center section	front				
55-070-02-01	MRB	51-05-01-210-804 55-05-03-210-807	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect forwa	ard side of front spar, inc	luding front sp	ar chords and web	s, terminal fittings	, and center section	front spar lug			
	INTERVAL N	IOTE: Whichever come	s first.							
	ACCESS NO	OTE: Bolt removal is required spar lugs.	uired to insped	ct terminal fittings a	and center section	front				
55-080-01-01	MRB	51-05-01-210-801 55-05-03-210-808	1.1	8 YR	4 YR	ALL	ALL			
	Inspect left h	orizontal stabilizer front	spar terminal f	ittings and center s	section front spar I	ugs.				
	ACCESS NO	OTE: Bolt removal is not	required.							
55-080-02-01	MRB	51-05-01-210-801 55-05-03-210-809	1.1	8 YR	4 YR	ALL	ALL			
	Inspect right	horizontal stabilizer fron	t spar terminal	fittings and center	section front spar	· lugs.				
	ACCESS NO	OTE: Bolt removal is not	required.							
55-100-01-01	MRB	51-05-01-210-806 55-05-03-210-810	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect left h inspar skins.	orizontal stabilizer from	front spar to re	ar spar including s	par chords, webs,	terminal fittings, up	per and lowe			
	INTERVAL N	IOTE: Whichever come	s first.							
55-100-02-01	MRB	51-05-01-210-806 55-05-03-210-811	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Inspect right inspar skins.	horizontal stabilizer fron	n front spar to i	ear spar including	spar chords, web	s, terminal fittings, ι	ipper and low			
	INTERVAL N	IOTE: Whichever come	s first.							
55-110-01-01	MRB	51-05-01-210-804 55-05-03-210-812	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL			
	Increat off of	de of left horizontal stab				minal fittings, alare	tor tab actual			

Inspect aft side of left horizontal stabilizer rear spar, including spar chords and webs, terminal fittings, elevator tab actuato support fitting, center section rear spar lugs, elevator hinge ribs, and elevator tab leading edge.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Disconnect balance panels in balance bays. Bolt removal is required to inspect terminal fittings and center section rear spar lugs, except at rear spar upper lugs and clevis. Remove tab hinge covers to inspect elevator tab leading edge.







				INTERVAL	APPLICABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
55-110-02-01	MRB	51-05-01-210-804 55-05-03-210-813	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	Inspect aft side of right horizontal stabilizer rear spar, including spar chords and webs, terminal fittings, elevator tab actuat support fitting, center section rear spar lugs, elevator hinge ribs, and elevator tab leading edge.										
	INTERVAL N	NOTE: Whichever come	s first.								
	ACCESS NO	OTE: Disconnect balanc and center section covers to inspect e	rear spar lugs	, except at rear spa	•	to inspect terminal t clevis. Remove tab	•				
55-115-01-01	MRB	51-05-01-210-804 55-05-03-211-801	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	Inspect left e	elevator tab mechanism s	support fittings	on elevator front s	par and elevator to	ab spar at leading e	dge cutouts.				
	INTERVAL N	NOTE: Whichever come	s first.								
	ACCESS NO	OTE: Remove upper or I	ower horizonta	al stabilizer trailing	edge seal and ele	vator					
		inboard hinge cove	er panel to insp	ect support fittings	. Remove tab hing	ge covers					
		to locally inspect for	orward face of	spar.							
55-115-02-01	MRB	51-05-01-210-804 55-05-03-211-802	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	ALL	ALL				
	Inspect right	elevator tab mechanism	support fitting	s on elevator front	spar and elevator	tab spar at leading	edge cutouts				
	Inspect right elevator tab mechanism support fittings on elevator front spar and elevator tab spar at leading edge cutouts. INTERVAL NOTE: Whichever comes first.										
	ACCESS NO	OTE: Remove upper or I panel to inspect su		•	•	vator inboard hinge nspect forward face					
55-120-01-01	MRB	51-05-01-210-801 55-05-03-210-814	1.1	8 YR	4 YR	ALL	ALL				
	Inspect left h	norizontal stabilizer rear s	par terminal fi	ttings and center se	ection rear spar lu	gs.					
	ACCESS NO	OTE: Bolt removal is not	required.								
55-120-02-01	MRB	51-05-01-210-801 55-05-03-210-815	1.1	8 YR	4 YR	ALL	ALL				
	Inspect right	horizontal stabilizer rear	spar terminal	fittings and center	section rear spar l	lugs.					
	ACCESS NO	OTE: Bolt removal is not	required.								
55-130-00-01	MRB	51-05-01-210-806	1.1	12 YR	10 YR	ALL	ALL				
		55-05-03-210-816	1.2	36000 FC	30000 FC						

Inspect rudder, elevator and elevator tab skin panels, rudder and elevator spars, rudder and elevator inspar ribs, rudder and elevator leading edge skins, rudder and elevator leading edge ribs, and rudder and elevator leading edge spars.

INTERVAL NOTE: Whichever comes first.





55-601-00-01

AWL



737-600/700/800/900 **TASK CARDS**

				APPLICABILITY									
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE						
55-135-01-01	MRB	51-05-01-210-803 55-05-03-211-807	1.1 1.2	8 YR 18000 FC	8 YR 18000 FC	ALL	ALL						
	Inspect the left elevator hinge fittings, left elevator actuator fittings, left elevator balance weight support structure, left elevator tab mast arm fitting and left elevator tab hinge fittings.												
	INTERVAL I	NOTE: Whichever comes	s first.										
	ACCESS NO	OTE: After removal of ac horizontal stabilizer to gain access.			,								
55-135-02-01	MRB	51-05-01-210-803 55-05-03-211-808	1.1 1.2	8 YR 18000 FC	8 YR 18000 FC	ALL	ALL						
		right elevator hinge fitting mast arm fitting and right	. •	•	ight elevator balaı	nce weight support	structure, rigl						
	elevator tab	mast arm mung and ngm	colovator tab ii	INTERVAL NOTE: Whichever comes first.									
	INTERVAL I	NOTE: Whichever comes	s first.		have Alternative	Pomovo olovator	from						
EE COO OO O4	INTERVAL I	NOTE: Whichever comes OTE: After removal of ac horizontal stabilizer to gain access.	s first. cess panels, c r. For elevator	lisconnect balance tab mast arm fitting	gs, remove fairing	on upper surface o	f elevator						
55-600-00-01	AWL	NOTE: Whichever comes OTE: After removal of achorizontal stabilized to gain access. 55-05-02-130-801	s first. cess panels, c r. For elevator	lisconnect balance tab mast arm fitting 56000 FC	gs, remove fairing 4000 FC	on upper surface o	f elevator ALL						
55-600-00-01	AWL Inspect (Ultr See Doc. Do The NDI me (D6-37239).	NOTE: Whichever comes OTE: After removal of ac horizontal stabilizer to gain access. 55-05-02-130-801 asonic) the spar chord be 626A001 - DTR, DTR che thod(s) necessary to accome the inspection procedure.	s first. ccess panels, cr. For elevator 1.1 etween the welleck form 55-10 omplish the intes are contained	56000 FC o and shear tie at E -01-1, alternative ir ent of this inspectic ed in Part 4, Subject	4000 FC BL 1.3 on the left a spection. on is contained in tot 55-10-06.	ALL nd right hand sides	ALL						
55-600-00-01	AWL Inspect (Ultr See Doc. Do The NDI me (D6-37239).	NOTE: Whichever comes OTE: After removal of ac horizontal stabilized to gain access. 55-05-02-130-801 asonic) the spar chord be 626A001 - DTR, DTR che thod(s) necessary to access	s first. ccess panels, cr. For elevator 1.1 etween the welleck form 55-10 omplish the intes are contained	56000 FC o and shear tie at E -01-1, alternative ir ent of this inspectic ed in Part 4, Subject	4000 FC BL 1.3 on the left a spection. on is contained in tot 55-10-06.	ALL nd right hand sides	ALL						
55-600-00-01 55-600-00-02	AWL Inspect (Ultr See Doc. Do The NDI me (D6-37239).	NOTE: Whichever comes OTE: After removal of ac horizontal stabilizer to gain access. 55-05-02-130-801 asonic) the spar chord be 626A001 - DTR, DTR che thod(s) necessary to accome the inspection procedure.	s first. ccess panels, cr. For elevator 1.1 etween the welleck form 55-10 omplish the intes are contained	56000 FC o and shear tie at E -01-1, alternative ir ent of this inspectic ed in Part 4, Subject	4000 FC BL 1.3 on the left a spection. on is contained in tot 55-10-06.	ALL nd right hand sides	ALL						
	AWL Inspect (Ultr See Doc. D6 The NDI me (D6-37239). ACCESS NG AWL Inspect (Ultr See Doc. D6 The NDI me (D6-37239).	NOTE: Whichever comes OTE: After removal of ac horizontal stabilized to gain access. 55-05-02-130-801 asonic) the spar chord be 626A001 - DTR, DTR che thod(s) necessary to accomple inspection procedure OTE: Access horizontal stables.	1.1 etween the well estabilizer center 1.1 etween the well estabilizer the well established the interest the well established the well establish	56000 FC o and shear tie at E-01-1, alternative in section through of the section through o	4000 FC BL 1.3 on the left a spection. on is contained in to to 55-10-06. opening in center contained in the spection of the spection. BL 1.3 on the left and spection. on is contained in the spection.	ALL nd right hand sides the 737 Nondestruct of 1088 bulkhead. ALL nd right hand sides	ALL ALL ALL ALL						

56000 FC Inspect (High Frequency Eddy Current) the upper chord around the fasteners common to the web at BL 2.7 and BL 6.7 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-2, for alternative inspection.

1.1

55-05-02-250-801

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-16.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

INDEX

ALL

ALL

4000 FC





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-601-00-02	AWL	55-05-02-250-801	1.1	56000 FC	4000 FC	ALL	ALL
	the left and r See Doc. Do The NDI me (D6-37239).	n Frequency Eddy Currency Fright hand sides. Fright	ck form 55-10 complish the intest are contain	-01-2, for alternativent of this inspection of the inspection of t	re inspection. on is contained in ct 55-10-16.	the 737 Nondestruc	
55-602-00-01	AWL	55-05-02-250-802	1.1	56000 FC	36000 FC	ALL	ALL
		h Frequency Eddy Curre					
		326A001 - DTR, DTR che				3	
		thod(s) necessary to acc The inspection procedur		•		the 737 Nondestruc	ctive Test Manua
	ACCESS NO	OTE: Removal of fastene	ers from both L	BL and RBL 8.1 to	19.7 is required.		
		Access horizontal	stabilizer cente	er section through o	ppening in center	of 1088 bulkhead.	
55-602-00-02	AWL	55-05-02-250-802	1.1	56000 FC	36000 FC	ALL	ALL
	(D6-37239).	thod(s) necessary to acc The inspection procedur DTE: Removal of fastend Access horizontal	es are containers from both L	ed in Part 6, Subje BL and RBL 8.1 to	ct 55-10-22. 19.7 is required.		ctive Test Manu
55-603-00-01	AWL	55-05-02-130-802	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (Ultra	asonic) the rear spar forv	vard flange at	he four Texas Star	attachment point	S.	
		326A001 - DTR, DTR che		·	•		
		thod(s) necessary to acc The inspection procedur		•		the 737 Nondestruc	ctive Test Manu
	ACCESS NO	OTE: Access horizontal	stabilizer cente	r section through o	ppening in center	of 1088 bulkhead	
55-603-00-02	AWL	55-05-02-130-802	1.1	56000 FC	18000 FC	ALL	ALL
	See Doc. D6 The NDI mer (D6-37239).	asonic) the rear spar forw 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	ck form 55-10 complish the int es are contain	-01-4, for alternativent of this inspection of the inspection of t	ve inspection. on is contained in ct 55-10-07.	the 737 Nondestruc	ctive Test Manu
	ACCESS NO	OTE: Access horizontal:	stabilizer cente	r section through o	ppening in center	OI IUOO DUIKNEAD	
55-604-00-01	AWL	55-05-02-130-804	1.1	56000 FC	10000 FC	ALL	ALL
		asonic) the pivot fittings of 26A001 - DTR, DTR che			-	use pivot lug.	

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.





				INTERVAL		APPLIC	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-604-00-02	AWL	55-05-02-130-804	1.1	56000 FC	10000 FC	ALL	ALL			
	Inspect (Ultrasonic) the pivot fittings on the horizontal stabilizer center section hinge house pivot lug.									
	See Doc. D626A001 - DTR, DTR check form 55-10-02-1, for alternative inspection.									
	ACCESS NO	OTE: Access horizontal	stabilizer cente	er section through o	pening in center	of 1088 bulkhead.				
55-605-00-01	AWL	55-05-02-130-824	1.1	56000 FC	3000 FC	ALL	ALL			
		asonic) the pivot fitting p 26A001 - DTR, DTR che		0	•					
	ACCESS NO	OTE: Removal of gap co	overs and slidir	ig seals is required.						
55-605-00-02	AWL	55-05-02-130-824	1.1	56000 FC	3000 FC	ALL	ALL			
		asonic) the pivot fitting p 26A001 - DTR, DTR che		-	•					
	ACCESS NO	OTE: Removal of gap co	overs and slidir	ig seals is required.						
55-606-00-01	AWL	55-05-02-130-819	1.1	56000 FC	1600 FC	ALL	ALL			
	Inspect (Ultrasonic) the fitting lugs on the horizontal stabilizer center section thrust link fittings. There are 4 lugs per fitting. See Doc. D626A001 - DTR, DTR check form 55-10-02-3, for alternative inspection.									
	ACCESS NOTE: Remove fittings to inspect the fitting lugs, 4 lugs per fitting.									
		Access horizontal	stabilizer cente	er section through o	pening in center	of 1088 bulkhead.				
55-606-00-02	AWL	55-05-02-130-819	1.1	56000 FC	1600 FC	ALL	ALL			
		asonic) the fitting lugs or 26A001 - DTR, DTR che				ittings. There are 4	lugs per fittin			
	ACCESS NO	OTE: Remove fittings to Access horizontal	•	ing lugs, 4 lugs per er section through o	Ü	of 1088 bulkhead.				
55-607-00-01	AWL	55-05-02-130-823	1.1	56000 FC	6000 FC	ALL	ALL			
	See Doc. D6 The NDI met	asonic) the horizontal sta 26A001 - DTR, DTR cha hod(s) necessary to acc The inspection procedur	eck form 55-10 complish the int	-03-1, for alternative ent of this inspection	e inspection. on is contained in	the 737 Nondestru	ctive Test Ma			
	,	OTE: Removal of bolts a		-						
55-607-00-02	AWL	55-05-02-130-823	1.1	56000 FC	6000 FC	ALL	ALL			
	Inapport /I Iltre	aconia) the herizontal etc	abilizar aantar	antina independent fi	tting lugg					

Inspect (Ultrasonic) the horizontal stabilizer center section jackscrew fitting lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-03-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-04.

ACCESS NOTE: Removal of bolts and bushings required.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-608-00-01	AWL	55-05-02-250-806	1.1	56000 FC	36000 FC	ALL	ALL			
	Inspect (High	h Frequency Eddy Curre	ent) the visible p	portion of the lower	chord between s	tabilizer STA 129.5	and stabilizer			
	The NDI me	526A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedur	complish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Manu			
55-608-00-02	AWL	55-05-02-250-806	1.1	56000 FC	36000 FC	ALL	ALL			
	Inspect (High	h Frequency Eddy Curre	ent) the visible p	portion of the lower	chord between s	tabilizer STA 129.5	and stabilizer			
		626A001 - DTR, DTR cho			•					
		thod(s) necessary to acc The inspection procedure				the 737 Nondestruc	ctive Test Manu			
55-609-00-01	AWL	55-05-02-250-807	1.1	56000 FC	36000 FC	ALL	ALL			
		v Frequency Eddy Currer	nt) both fastene	er rows in the FWD	flange at, and be	tween, the ribs at s	tabilizer STA			
		626A001 - DTR, DTR che								
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	ctive Test Manı			
55-609-00-02	AWL	55-05-02-250-807	1.1	56000 FC	36000 FC	ALL	ALL			
	Inspect (Low Frequency Eddy Current) both fastener rows in the FWD flange at, and between, the ribs at stabilizer STA 212.3 to stabilizer STA 310.54.									
	See Doc. D6	626A001 - DTR, DTR che	eck form 55-10	-04-2, for alternativ	e inspection.					
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	ctive Test Manu			
55-610-00-01	AWL	55-05-02-250-805	1.1	56000 FC	36000 FC	ALL	ALL			
		h Frequency Eddy Curre bilizer STA 310.54.	ent) the rear spa	ar lower chord AFT	flange, at and be	tween the ribs, from	ı stabilizer STA			
	See Doc. D6	626A001 - DTR, DTR che	eck form 55-10	-04-3 for alternative	e inspection.					
	ACCESS NO	OTE: Inspection requires	s removal of the	e Trailing Edge (TE) panel and faste	ners.				
		The TE panels are	e attached by n	utplates.						
55-610-00-02	AWL	55-05-02-250-805	1.1	56000 FC	36000 FC	ALL	ALL			
	212.3 to stal	h Frequency Eddy Curre bilizer STA 310.54. 626A001 - DTR, DTR che	, .		3 /	tween the ribs, from	ı stabilizer STA			
		OTE: Inspection requires			•	nere				
	ACCESS NO	The TE panels are		σ σ ,	.) Panei anu iaste	11013.				
55-611-00-01	AWL	55-05-02-130-803	1.1	56000 FC	36000 FC	ALL	ALL			

Inspect (Ultrasonic) the rear spar lower chord web flange at the Trailing Edge (TE) ribs and stiffeners from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-4, for alternative inspection.





				INTERVAL		APPLICA	ABILITY				
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
55-611-00-02	AWL	AWL 55-05-02-130-803 1.1 56000 FC 36000 FC ALL ALL									
	212.3 to stat	asonic) the rear spar low bilizer STA 310.54. 26A001 - DTR, DTR che				and stiffeners from s	stabilizer STA				
55-611-10-01	AWL	55-05-02-250-845	1.1	56000 FC	24000 FC	ALL	ALL				
	stabilizer ST	n Frequency Eddy Curre A 212.3 to stabilizer STA 26A001 - DTR, DTR che	310.54.		-	ne Trailing Edge (T	E) ribs from				
55-611-10-02	AWL	55-05-02-250-845	1.1	56000 FC	24000 FC	ALL	ALL				
55-611-12-01		A 212.3 to stabilizer STA 26A001 - DTR, DTR che 55-05-02-250-846		-04-5, for alternativ	ve inspection.	ALL	ALL				
	ribs, from sta See Doc. D6 The NDI met	Frequency Eddy Currer abilizer STA 310.54 to ou 26A001 - DTR, DTR che chod(s) necessary to acc The inspection procedur	tboard tip. eck form 55-10 complish the int	-04-6, for alternativ	ve inspection.						
55-611-12-02	AWL	55-05-02-250-846	1.1	56000 FC	24000 FC	ALL	ALL				
	ribs, from sta See Doc. D6 The NDI met	Frequency Eddy Currer abilizer STA 310.54 to out 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	tboard tip. eck form 55-10 complish the int	-04-6, for alternativ	ve inspection. on is contained in						
55-611-14-01	AWL	55-05-02-250-815	1.1	56000 FC	15000 FC	ALL	ALL				
		Frequency Eddy Currer	nt) the rear spa	ar lower chord aft fla	ange, at and betw	een the ribs, from s					

See Doc. D626A001-DTR, DTR check form 55-10-04-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-55.

55-611-14-02 AWL 55-05-02-250-815 1.1 56000 FC 15000 FC ALL ALL

Inspect (Low Frequency Eddy Current) the rear spar lower chord aft flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001-DTR, DTR check form 55-10-04-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-55.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-611-16-01 AWL 55-05-02-250-847 1.1 56000 FC 24000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar lower chord from stabilizer STA 310.54 to outboard tip web flange, at							
	and between See Doc. D6 The NDI met		eck form 55-10 complish the int	-04-8, for alternative	ve inspection. on is contained in		
55-611-16-02	AWL	55-05-02-250-847	1.1	56000 FC	24000 FC	ALL	ALL
	and between See Doc. D6 The NDI met	Frequency Eddy Currer , the ribs. 26A001 - DTR, DTR chr hod(s) necessary to acc The inspection procedure	eck form 55-10 complish the int	-04-8, for alternative	ve inspection. on is contained in	·	
55-611-18-01	AWL	55-05-02-250-848	1.1	56000 FC	36000 FC	ALL	ALL
	See Doc. D6 The NDI met	zer STA 247, and elevat 26A001 - DTR, DTR ch hod(s) necessary to acc The inspection procedul	eck form 55-10 complish the int	-04-9, for alternative	on is contained in	the 737 Nondestruc	ctive Test Manu
55-611-18-02	AWL	55-05-02-250-848	1.1	56000 FC	36000 FC	ALL	ALL
	242 to stabili See Doc. D6 The NDI met	n Frequency Eddy Curre zer STA 247, and elevat 26A001 - DTR, DTR chi hod(s) necessary to acc The inspection procedur	or STA 197 to eck form 55-10 complish the int	elevator STA 203. -04-9, for alternativ ent of this inspection	ve inspection.		
55-611-20-01	AWL	55-05-02-250-849	1.1	56000 FC	2000 FC	ALL	ALL
	Note: Inspec See Doc. D6 The NDI met	n Frequency Eddy Curre tion is applicable to all for 26A001 - DTR, DTR cho hod(s) necessary to acc The inspection procedur	our bolt location eck form 55-10 complish the int	ns common to the or -04-10A, for alternated the of this inspection	chord and termina ative inspection. on is contained in	al fitting.	ctive Test Manu
55-611-20-02	AWL	55-05-02-250-849	1.1	56000 FC	2000 FC	ALL	ALL
	Inspect (High Note: Inspec	n Frequency Eddy Curre tion is applicable to all fo 26A001 - DTR, DTR ch	our bolt location	ns common to the	chord and termina	-	

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(D6-37239). The inspection procedures are contained in Part 6, Section 55-10-27.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual





TASK CARD NO.				INTERVAL		APPLICA	ABILITY			
IASK CAKE NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-611-22-01	AWL	55-05-02-250-804	1.1	56000 FC	3550 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork. Note: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.									
		See Doc. D626A001-DTR, DTR check form 55-10-04-10B, for alternative inspection.								
	ACCESS NO	OTE: The aft side is acce								
		through the access	panel in the id	ower inspar skin. S	ealant removal is	requirea.				
55-611-22-02	AWL	55-05-02-250-804	1.1	56000 FC	3550 FC	ALL	ALL			
	Note: Inspec	n Frequency Eddy Currer tion is applicable to the o 26A001-DTR, DTR chec	utboard three	bolt locations com	mon to the chord	-				
	ACCESS NO	OTE: The aft side is acce	essed from the	trailing edge. The	forward side is ad	ccessed				
		through the access	panel in the lo	ower inspar skin. S	ealant removal is	required.				
55-611-23-01	AWL	55-05-02-130-814	1.1	56000 FC	3550 FC	ALL	ALL			
	Note: Inspec	Inspect (Ultrasonic) the rear spar upper chord between the terminal fitting fork. Note: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting. See Doc. D626A001 - DTR, DTR check form 55-10-04-10B, for alternative inspection.								
	ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed									
	through the access panel in the lower inspar skin. Sealant removal is required.									
55-611-23-02	AWL	55-05-02-130-814	1.1	56000 FC	3550 FC	ALL	ALL			
33-011-23-02	1 (// 11)									
	Note: Inspec	asonic) the rear spar upp tion is applicable to the o 26A001 - DTR, DTR che	utboard three	bolt locations com	mon to the chord	and terminal fitting.				
	Note: Inspec See Doc. D6	tion is applicable to the observation is applicable to the observation of the observation	outboard three ock form 55-10 essed from the	bolt locations comi -04-10B, for alternations trailing edge. The	mon to the chord ative inspection. forward side is ac	ccessed				
	Note: Inspec See Doc. D6	tion is applicable to the o 26A001 - DTR, DTR che	outboard three ock form 55-10 essed from the	bolt locations comi -04-10B, for alternations trailing edge. The	mon to the chord ative inspection. forward side is ac	ccessed				
55-611-24-01	Note: Inspec See Doc. D6	tion is applicable to the observation is applicable to the observation of the observation	outboard three ock form 55-10 essed from the	bolt locations comi -04-10B, for alternations trailing edge. The	mon to the chord ative inspection. forward side is ac	ccessed	ALL			
55-611-24-01	AWL Inspect (High Note: Inspect	etion is applicable to the of 26A001 - DTR, DTR che DTE: The aft side is access through the access	outboard three ck form 55-10- cessed from the capanel in the load. 1.1 nt) the rear spanboard bolt local.	bolt locations come -04-10B, for alternative trailing edge. The ower inspar skin. S 56000 FC ar upper chord betwation common to the	mon to the chord ative inspection. forward side is ac ealant removal is 12000 FC ween the terminal he chord and term	ccessed required. ALL fitting fork.				
55-611-24-01 55-611-24-02	AWL Inspect (High Note: Inspect	tion is applicable to the of 26A001 - DTR, DTR che DTE: The aft side is accessive through the access 55-05-02-250-850 In Frequency Eddy Currer tion is applicable to the in	outboard three ck form 55-10- cessed from the capanel in the load. 1.1 nt) the rear spanboard bolt local.	bolt locations come -04-10B, for alternate trailing edge. The ower inspar skin. S 56000 FC ar upper chord betwation common to the	mon to the chord ative inspection. forward side is ac ealant removal is 12000 FC ween the terminal he chord and term	ccessed required. ALL fitting fork.				
	AWL Inspect (High Note: In	etion is applicable to the of 26A001 - DTR, DTR che DTE: The aft side is access through the access 55-05-02-250-850 in Frequency Eddy Currer etion is applicable to the in 226A001 - DTR, DTR che	extraction three color form 55-10- extraction the color form the color form the color form the color form 55-10- 1.1 1.1 1.1 1.1 1.1 1.1 1.1	bolt locations come -04-10B, for alternate trailing edge. The ower inspar skin. S 56000 FC ar upper chord betwation common to the code of the code o	mon to the chord ative inspection. forward side is acceptant removal is 12000 FC ween the terminal the chord and terminative inspection.	ALL fitting fork. ALL fitting fork.	ALL			
	AWL Inspect (High Note: In	tion is applicable to the of 26A001 - DTR, DTR che DTE: The aft side is accessive through the access 55-05-02-250-850 in Frequency Eddy Currer 25A001 - DTR, DTR che 55-05-02-250-850 in Frequency Eddy Currer 25Ton is applicable to the information in the information in the information is applicable to the information in the infor	extraction three color form 55-10- extraction the color form the color form the color form the color form 55-10- 1.1 1.1 1.1 1.1 1.1 1.1 1.1	bolt locations come -04-10B, for alternate trailing edge. The ower inspar skin. S 56000 FC ar upper chord betwation common to the code of the code o	mon to the chord ative inspection. forward side is acceptant removal is 12000 FC ween the terminal the chord and terminative inspection.	ALL fitting fork. ALL fitting fork.	ALL			

Note: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.







ASK CARD NO.				INTERVAL		APPLICA	ABILITY
	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-611-25-02	AWL	55-05-02-130-816	1.1	56000 FC	12000 FC	ALL	ALL
		asonic) the rear spar upp					
		tion is applicable to the i				inal fitting.	
	See Doc. D6	326A001 - DTR, DTR che	eck form 55-10	-04-10C, for alterna	tive inspection.		
55-611-26-01	AWL	55-05-02-250-851	1.1	56000 FC	2000 FC	ALL	ALL
		Frequency Eddy Currer	nt) both the for	ward and aft rear sp	ar upper chord fl	anges from stabilize	er STA 67.78
	stabilizer STA	A 203.10. 326A001 - DTR, DTR che	ook form EE 10	04 11A for alterna	tivo inonoction		
	See Doc. Do	ZOAOOT - DTK, DTK CHE	ECK IOIIII 33-10	-04-TTA, IOI allema	live irispection.		
55-611-26-02	AWL	55-05-02-250-851	1.1	56000 FC	2000 FC	ALL	ALL
		Frequency Eddy Currer	nt) both the for	ward and aft rear sp	ar upper chord fl	anges from stabilize	er STA 67.78
	stabilizer STA	A 203.10. 326A001 - DTR, DTR che	eck form 55-10	-04-11A for alterna	tive inspection		
	OCC DOC. DO	20/1001 - 10111, 10111 (11)	JOK TOTTI JO-10	-04-11A, for allerna	пус пізроспоп.		
55-611-27-01	AWL	55-05-02-250-852	1.1	56000 FC	3500 FC	ALL	ALL
		n Frequency Eddy Curre				78 to stabilizer STA	203.10.
	See Doc. D6	326A001 - DTR, DTR che	eck form 55-10	-04-11A, for alterna	tive inspection.		
55-611-27-02	AWL	55-05-02-250-852	1.1	56000 FC	3500 FC	ALL	ALL
		n Frequency Eddy Curre				78 to stabilizer STA	203.10.
	See Doc. D6	326A001 - DTR, DTR che	eck form 55-10	-04-11A, for alterna	tive inspection.		
55-611-28-01	AWL	55-05-02-250-853	1.1	56000 FC	15000 FC	ALL	ALL
		Frequency Eddy Currer					
		abilizer STA 258.28.	ity boar are rer	ward and are roar op	ar appor onora o	ian nangoo nom ota	DIII201 0 17 (
	See Doc. D6	26A001 - DTR, DTR che	eck form 55-10	-04-11B, for alterna	tive inspection.		
FF 044 00 00			4.4	50000 50	45000 50	A1.1	A1.1
55-611-28-02	AWL	55-05-02-250-853	1.1	56000 FC	15000 FC	ALL	ALL
55-611-28-02	Inspect (Low	Frequency Eddy Currer					
55-611-28-02	Inspect (Low 203.10 to sta	r Frequency Eddy Currer abilizer STA 258.28.	nt) both the for	ward and aft rear sp	ear upper chord s		
55-611-28-02	Inspect (Low 203.10 to sta	Frequency Eddy Currer	nt) both the for	ward and aft rear sp	ear upper chord s		
55-611-28-02 55-611-30-01	Inspect (Low 203.10 to sta See Doc. D6	r Frequency Eddy Currer abilizer STA 258.28. 326A001 - DTR, DTR che 55-05-02-130-817	nt) both the for eck form 55-10 1.1	ward and aft rear sp -04-11B, for alterna 56000 FC	nar upper chord stive inspection.	kin flanges from sta	bilizer STA
	Inspect (Low 203.10 to sta See Doc. D6	r Frequency Eddy Currer abilizer STA 258.28. 326A001 - DTR, DTR che 55-05-02-130-817 asonic) the rear spar upp	nt) both the for eck form 55-10 1.1	ward and aft rear sp -04-11B, for alterna 56000 FC	nar upper chord stive inspection.	kin flanges from sta	bilizer STA
	Inspect (Low 203.10 to sta See Doc. D6 AWL Inspect (Ultra stabilizer STA	r Frequency Eddy Currer abilizer STA 258.28. 326A001 - DTR, DTR che 55-05-02-130-817 asonic) the rear spar upp	nt) both the form eck form 55-10 1.1 per chord web	ward and aft rear sp -04-11B, for alterna 56000 FC flange at the rib and	tive inspection. 18000 FC stiffener location	kin flanges from sta	bilizer STA
	Inspect (Low 203.10 to sta See Doc. D6 AWL Inspect (Ultra stabilizer STA	r Frequency Eddy Currer abilizer STA 258.28. 326A001 - DTR, DTR che 55-05-02-130-817 asonic) the rear spar upp A 258.28.	nt) both the form eck form 55-10 1.1 per chord web	ward and aft rear sp -04-11B, for alterna 56000 FC flange at the rib and	tive inspection. 18000 FC stiffener location	kin flanges from sta	bilizer STA

stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11C, for alternative inspection.





AVIL 55-05-02-250-854 1.1 56000 FC 24000 FC ALL ALL					INTERVAL		APPLICA	ABILITY
Inspect (High Frequency Eddy Current) the rear spar upper chord web flange between the ribs and stiffeners from stabiliz STA 203.10 to stabilizer STA 258.28. Note: Remove sealant in excess of .20" on either side of the fastener head or collar. See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection. See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection.	ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
STA 203.10 to stabilizer STA 258.28. Note: Remove sealant in excess of .20" on either side of the fastener head or collar.	55-611-32-01	AWL	55-05-02-250-854	1.1	56000 FC	24000 FC	ALL	ALL
Inspect (High Frequency Eddy Current) the rear spar upper chord web flange between the ribs and stiffeners from stabilizer STA 203.10 to stabilizer STA 258.28. Note: Remove sealant in excess of .20" on either side of the fastener head or collar. See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection. 55-611-34-01 AWL 55-05-02-250-855 1.1 56000 FC 15000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. 55-611-34-02 AWL 55-05-02-250-855 1.1 56000 FC 15000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. 55-611-36-01 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-36-01 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.		STA 203.10 Note: Remov	to stabilizer STA 258.28. ve sealant in excess of .2	20" on either si	de of the fastener h	nead or collar.	he ribs and stiffene	rs from stabilizer
STA 203 :10 to stabilizer STA 258.28 Note: Remove sealant in excess of .20" on either side of the fastener head or collar.	55-611-32-02	AWL	55-05-02-250-854	1.1	56000 FC	24000 FC	ALL	ALL
Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. 55-611-34-02 AWL 55-05-02-250-855		STA 203.10 Note: Remov	to stabilizer STA 258.28. ve sealant in excess of .2	20" on either si	de of the fastener h	nead or collar.	he ribs and stiffene	rs from stabilizer
258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. 55-611-34-02 AWL 55-05-02-250-855 1.1 56000 FC 15000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. 55-611-36-01 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-36-02 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.	55-611-34-01	AWL	55-05-02-250-855	1.1	56000 FC	15000 FC	ALL	ALL
Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.		258.28 to the	e tip.	,		· ·	of the chord, from s	stabilizer STA
258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection. 55-611-36-01 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-36-02 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.	55-611-34-02	AWL	55-05-02-250-855	1.1	56000 FC	15000 FC	ALL	ALL
Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-36-02 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.		258.28 to the	e tip.	,		· ·	of the chord, from s	stabilizer STA
the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-36-02 AWL 55-05-02-250-856 1.1 56000 FC 21000 FC ALL ALL Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.	55-611-36-01	AWL	55-05-02-250-856	1.1	56000 FC	21000 FC	ALL	ALL
Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.		the tip.		,		· ·	cord, from stabilize	r STA 258.28 to
the tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection. 55-611-38-01 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.	55-611-36-02	AWL	55-05-02-250-856	1.1	56000 FC	21000 FC	ALL	ALL
Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.		the tip.				-	cord, from stabilize	r STA 258.28 to
stabilizer STA 310.54. See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Man (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.	55-611-38-01	AWL	55-05-02-130-818	1.1	56000 FC	18000 FC	ALL	ALL
55-611-38-02 AWL 55-05-02-130-818 1.1 56000 FC 18000 FC ALL ALL		stabilizer ST. See Doc. D6 The NDI me	A 310.54. 326A001 - DTR, DTR che thod(s) necessary to acc	eck form 55-10 complish the int	-04-13A, for alternated of this inspection	ative inspection. on is contained in		
	55-611-38-02	AWL	55-05-02-130-818	1.1	56000 FC	18000 FC	ALL	ALL

Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.





				APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-611-40-01	AWL	55-05-02-250-871	1.1	56000 FC	24000 FC	ALL	ALL
	stabilizer ST. See Doc. D6 The NDI me	n Frequency Eddy Curre A 258.28 to stabilizer ST 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	A 310.54. eck form 55-10 complish the int	-04-13B, for alternated the control of this inspection	ative inspection.		·
55-611-40-02	AWL	55-05-02-250-871	1.1	56000 FC	24000 FC	ALL	ALL
	stabilizer ST. See Doc. D6 The NDI me	n Frequency Eddy Curre A 258.28 to stabilizer ST 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	A 310.54. eck form 55-10 complish the int	-04-13B, for alternated the control of this inspection	ative inspection.		·
55-611-42-01	AWL	55-05-02-250-857	1.1	56000 FC	36000 FC	ALL	ALL
	stabilizer ST. See Doc. D6 The NDI mei (D6-37239).	r Frequency Eddy Currer A 310.54 to tip. 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Access is through the chord is through	eck form 55-10 complish the interes are contain the lower inspa	-04-13C, for alternated and the control of this inspection and in Part 6, Section	ative inspection. on is contained in on 55-10-60 Subsurface insp	the 737 Nondestruc	
55-611-42-02	AWL	55-05-02-250-857	1.1	56000 FC	36000 FC	ALL	ALL
	stabilizer ST. See Doc. D6 The NDI met (D6-37239).	r Frequency Eddy Currer A 310.54 to tip. 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Access is through the chord is throug	eck form 55-10 complish the interes are contain the lower inspa	-04-13C, for alternation of this inspection of the inspection of t	ative inspection. on is contained in on 55-10-60 Subsurface insp	the 737 Nondestrud	·
55-611-44-01	AWL	55-05-02-250-858	1.1	56000 FC	18000 FC	ALL	ALL

AVVL 55-05-02-250-

Inspect (High Frequency Eddy Current) the rear spar web at the lower edge of the fail-safe chord, at and between the stiffeners, from stabilizer STA 83.5 to stabilizer STA 184.7.

See Doc. D626A001 - DTR, DTR check form 55-10-04-20, for alternative inspection.

ACCESS NOTE: Access is through the removable trailing edge lower skin panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

55-611-44-02

AWL 55

55-05-02-250-858

1.1

56000 FC

18000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the rear spar web at the lower edge of the fail-safe chord, at and between the stiffeners, from stabilizer STA 83.5 to stabilizer STA 184.7.

See Doc. D626A001 - DTR, DTR check form 55-10-04-20, for alternative inspection.

ACCESS NOTE: Access is through the removable trailing edge lower skin panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
55-611-46-01	AWL	55-05-02-250-859	1.1	56000 FC	18000 FC	ALL	ALL				
	stiffeners, fro	n Frequency Eddy Currer om stabilizer STA 184.7 to 26A001 - DTR, DTR che	o stabilizer ST	A 285.9.		r chord, at and betw	veen the				
	ACCESS NO	OTE: Access is through the fillet seal. Remove			er Skin Panels. Tl	ne DTR curve assur	nes no				
55-611-46-02	AWL	55-05-02-250-859	1.1	56000 FC	18000 FC	ALL	ALL				
	stiffeners, fro See Doc. D6	n Frequency Eddy Currer om stabilizer STA 184.7 to 326A001 - DTR, DTR che DTE: Access is through to fillet seal. Remove	o stabilizer ST eck form 55-10 the removable	A 285.9. -04-21, for alternati Trailing Edge Lowe	ive inspection.	·					
55-611-48-01	AWL	55-05-02-250-860	1.1	56000 FC	24000 FC	ALL	ALL				
	Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower chord edges, at and between the stiffeners, from stabilizer STA 285.9 to stabilizer STA 310.5. See Doc. D626A001 - DTR, DTR check form 55-10-04-22, for alternative inspection. ACCESS NOTE: Access is through the removable Trailing Edge Panel Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.										
55-611-48-02	AWL	55-05-02-250-860	1.1	56000 FC	24000 FC	ALL	ALL				
	stiffeners, fro See Doc. D6	n Frequency Eddy Currer om stabilizer STA 285.9 to 226A001 - DTR, DTR che DTE: Access is through to assumes no fillet so	o stabilizer ST eck form 55-10 the removable	A 310.5. -04-22, for alternati Trailing Edge Pane	ive inspection. el Lower Skin Par						
55-611-50-01	AWL	55-05-02-250-861	1.1	56000 FC	15000 FC	ALL	ALL				
	the stiffeners	Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower web to chord fastener locations, between the stiffeners, from stabilizer STA 310.5 to outboard tip. See Doc. D626A001 - DTR, DTR check form 55-10-04-23, for alternative inspection.									
	ACCESS NO	OTE: Access is through the Remove any cap s		er skin access pane	els. The DTR curv	e assumes no cap	seal.				
			·								
55-611-50-02	AWL	55-05-02-250-861	1.1	56000 FC	15000 FC	ALL	ALL				

Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower web to chord fastener locations, between the stiffeners, from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-23, for alternative inspection.

ACCESS NOTE: Access is through the inspar lower skin access panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.





55-613-00-02

AWL



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-611-52-01	AWL	55-05-02-130-820	1.1	56000 FC	15000 FC	ALL	ALL
	310.5 to out See Doc. Do	asonic) the rear spar well poard tip. 326A001 - DTR, DTR che DTE: Access is through	eck form 55-10-	04-24, for alternat	ive inspection.	wer chords from sta	abilizer STA
55-611-52-02	AWL	55-05-02-130-820	1.1	56000 FC	15000 FC	ALL	ALL
	310.5 to outl See Doc. D6	asonic) the rear spar well coard tip. 326A001 - DTR, DTR che DTE: Access is through	eck form 55-10-	04-24, for alternat	ive inspection.	wer chords from sta	abilizer STA
55-612-00-01	AWL	55-05-02-130-821	1.1	56000 FC	18000 FC	ALL	ALL
	The NDI me (D6-37239).	i26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of gap co	omplish the inte	ent of this inspection and in Part 4, Subject	on is contained in	the 737 Nondestruc	ctive Test Manu
55-612-00-02	AWL	55-05-02-130-821	1.1	56000 FC	18000 FC	ALL	ALL
	See Doc. Do The NDI me (D6-37239).	asonic) the rear spar upp 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of gap co	eck form 55-10- omplish the into es are containe	05-1, for alternativent of this inspection of the land in Part 4, Subjection	on is contained in	the 737 Nondestruc	ctive Test Manu
55-613-00-01	AWL	55-05-02-130-822	1.1	56000 FC	3550 FC	ALL	ALL
	chord. Note: Bolts a	asonic) the forward and a and bushings should rem s26A001 - DTR, DTR che	ain installed for	the inspection. Re	emove any sealar		mon to the spar
	ACCESS NO	OTE: The aft side is accordance panel in the inspar		0 0		•	access

Inspect (Ultrasonic) the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord

3550 FC

ALL

56000 FC

Note: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

1.1

55-05-02-130-822

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

INDEX

ALL





				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
55-613-01-01	AWL	55-05-02-250-812	1.1	56000 FC	3550 FC	ALL	ALL		
	common to the Note: Bolts at See Doc. D6 The NDI met	n Frequency Eddy Curre he spar chord. and bushings should rem 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	nain installed for eck form 55-10 complish the inf	or the inspection. Re -05-2A, for alternat tent of this inspection	emove any sealarive inspection.	nt present.			
	ACCESS NO	OTE: The aft side is acc panel in the inspar		trailing edge. The ealant, if present, m			e access		
55-613-01-02	AWL	55-05-02-250-812	1.1	56000 FC	3550 FC	ALL	ALL		
	common to the Note: Bolts at See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre he spar chord. Ind bushings should rem 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	nain installed for eck form 55-10 complish the interes are contain	or the inspection. Re -05-2A, for alternat ent of this inspection ed in Part 6, Subject	emove any sealar ive inspection. on is contained in ct 55-10-18.	nt present. the 737 Nondestrue	ctive Test Manu		
	ACCESS NO	OTE: The aft side is acc panel in the inspar		e trailing edge. The ealant, if present, m			e access		
55-613-10-01	AWL	55-05-02-211-813	1.1	56000 FC	9000 FC	ALL	ALL		
	Inspect (Detailed) the forward and aft sides of the terminal fitting at the outboard attach bolt location. Note: Bolts and bushings should remain installed for the inspection. Sealant removal is required. See Doc. D626A001 - DTR, DTR check form 55-10-05-2B, for alternative inspection. ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Use of a Bore scope is required.								
55-613-10-02	AWL	55-05-02-211-813	1.1	56000 FC	9000 FC	ALL	ALL		
	Note: Bolts a	ailed) the forward and af and bushings should rem 26A001 - DTR, DTR che	ain installed fo	r the inspection. Se	ealant removal is				
	ACCESS NO	The aft side is acc through the access		trailing edge. The ower inspar skin. U					
55-614-00-01	AWL	55-05-02-230-801	1.1	56000 FC	9000 FC	ALL	ALL		
	for both the I	etrant) the upper, lower eft and right sides. 26A001 - DTR, DTR che	·			tal stabilizer rear sp	ar terminal fittii		
	ACCESS NO	OTE: Removal of pins is	required.						
55-614-00-02	AWL	55-05-02-230-801	1.1	56000 FC	9000 FC	ALL	ALL		
		etrant) the upper, lower							

Inspect (Penetrant) the upper, lower and failsafe pins at the side of body on the horizontal stabilizer rear spar terminal fitting for both the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-05-3, for alternative inspection.

ACCESS NOTE: Removal of pins is required.





TASK CARD NO.				INTERVAL			APPLICABILITY	
	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
55-614-10-01	AWL	55-05-02-250-810	1.1	56000 FC	21000 FC	ALL	ALL	
	(SOB) rib ten See Doc. D6 The NDI met	n Frequency Eddy Currentsion bolt hole. 26A001 - DTR, DTR chentshous hod(s) necessary to accument to the inspection procedure.	eck form 55-10 complish the int	0-05-4, for alternativ tent of this inspection	ve inspection. on is contained in		·	

ACCESS NOTE: Removal of the tension bolt is required.

55-614-10-02

AWL 55-05-02-250-810

1.1

56000 FC

21000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

55-614-11-01

AWL

55-05-02-250-883

1.1

56000 FC

21000 FC

ALI

ALL

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the lower side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

55-614-11-02

AWL

55-05-02-250-883

1.1

56000 FC

21000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the lower side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

55-614-12-01

AWL

55-05-02-250-863

1.1

56000 FC

36000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the rear spar lower chord at the side of body, common to the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-48.

ACCESS NOTE: The inspection requires the removal of the bolt.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-614-12-02	AWL	55-05-02-250-863	1.1	56000 FC	36000 FC	ALL	ALL
	tension bolt See Doc. Do The NDI me	n Frequency Eddy Currenthole. 626A001 - DTR, DTR chesthod(s) necessary to accurrenthe inspection procedur	eck form 55-10 omplish the int	-05-5, for alternativ ent of this inspectio	e inspection. on is contained in		,
	ACCESS NO	OTE: The inspection req	uires the remo	val of the bolt.			
55-615-00-01	AWL	55-05-02-250-808	1.1	56000 FC	2000 FC	ALL	ALL
	See Doc. Do The NDI me	r Frequency Eddy Currer 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the int	-06-1, for alternativent of this inspection	e inspection. on is contained in		
55-615-00-02	AWL	55-05-02-250-808	1.1	56000 FC	2000 FC	ALL	ALL
55-616-00-01	AWL Inspect (Low See Doc. Do	The inspection procedur 55-05-02-250-803 Frequency Eddy Currer 326A001 - DTR, DTR che thod(s) necessary to acc	1.1 nt) the front spa	56000 FC ar upper chord aft fl -06-2, for alternativ	2000 FC ange from stabiliz e inspection.		•
		The inspection procedur				the 737 Nondestruc	ctive rest Maric
55-616-00-02	AWL	55-05-02-250-803	1.1	56000 FC	2000 FC	A1.1	
				0000010	200010	ALL	ALL
	See Doc. Do The NDI me	r Frequency Eddy Currer 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the int	ar upper chord aft fl -06-2, for alternativ ent of this inspection	ange from stabilize inspection.	zer STA 66.5 to stab	oilizer tip.
55-616-05-01	See Doc. Do The NDI me	626A001 - DTR, DTR che thod(s) necessary to acc	eck form 55-10 omplish the int	ar upper chord aft fl -06-2, for alternativ ent of this inspection	ange from stabilize inspection.	zer STA 66.5 to stab	oilizer tip.
55-616-05-01	See Doc. D6 The NDI me (D6-37239). AWL Inspect (Low except at we See Doc. D6	326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the int es are contained 1.1 nt) the front spa 11.1. eck form 55-10	ar upper chord aft fl -06-2, for alternativent of this inspection ed in Part 6, Subjection 56000 FC ar upper chord web	ange from stabilize inspection. on is contained in of 55-10-32. 6000 FC flange from stabilize inspection.	zer STA 66.5 to stab the 737 Nondestruc ALL	oilizer tip. ctive Test Manu ALL
55-616-05-01 55-616-05-02	See Doc. D6 The NDI me (D6-37239). AWL Inspect (Low except at we See Doc. D6	526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur 55-05-02-250-885 Frequency Eddy Currer b splice, stabilizer STA 1 526A001 - DTR, DTR che	eck form 55-10 omplish the int es are contained 1.1 nt) the front spa 11.1. eck form 55-10	ar upper chord aft fl -06-2, for alternativent of this inspection ed in Part 6, Subjection 56000 FC ar upper chord web	ange from stabilize inspection. on is contained in of 55-10-32. 6000 FC flange from stabilize inspection.	zer STA 66.5 to stab the 737 Nondestruc ALL	oilizer tip. ctive Test Manu ALL

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 75.0 to Stab Sta 175.5, except at web splice, stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
55-616-06-01	AWL	55-05-02-250-886	1.1	56000 FC	6000 FC	ALL	ALL				
	except at we See Doc. D6	n Frequency Eddy Curre b splice, stabilizer STA 1 26A001 - DTR, DTR cho DTE: Removal of the Le	111.1. eck form 55-10	-06-4C, for alternat	tive inspection.	oilizer STA 75.0 to S	tab Sta 175.5,				
55-616-06-02	AWL	55-05-02-250-886	1.1	56000 FC	6000 FC	ALL	ALL				
	except at we See Doc. D6	n Frequency Eddy Curre b splice, stabilizer STA 1 26A001 - DTR, DTR cho DTE: Removal of the Le	111.1. eck form 55-10	-06-4C, for alternat	tive inspection.	oilizer STA 75.0 to S	tab Sta 175.5,				
55-616-07-01	AWL	55-05-02-250-887	1.1	56000 FC	15000 FC	ALL	ALL				
	See Doc. D6	Inspect (Low Frequency Eddy Current) the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96. See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection. ACCESS NOTE: Removal of the Leading Edge is required for inspection access.									
55-616-07-02	AWL	55-05-02-250-887	1.1	56000 FC	15000 FC	ALL	ALL				
	Inspect (Low Frequency Eddy Current) the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96 See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection. ACCESS NOTE: Removal of the Leading Edge is required for inspection access.										
55-616-08-01	AWL	55-05-02-250-888	1.1	56000 FC	15000 FC	ALL	ALL				
	Inspect (High See Doc. D6	n Frequency Eddy Curre 26A001 - DTR, DTR cho DTE: Removal of the Le	eck form 55-10	ar upper chord web	tive inspection.		24 to STA 111.9				
55-616-08-02	AWL	55-05-02-250-888	1.1	56000 FC	15000 FC	ALL	ALL				
	See Doc. D6	n Frequency Eddy Curre 26A001 - DTR, DTR che	eck form 55-10	-06-4D, for alternat	tive inspection.	olice from STA 110.2	24 to STA 111.9				
	ACCESS NO	OTE: Removal of the Le	adıng Edge is ı	required for inspect	tion access.						
55-616-09-01	AWL	55-05-02-250-884	1.1	56000 FC	6000 FC	ALL	ALL				
	Inspect (High	r Frequency Eddy Curre	nt) the front sp	ar upper chord web	o flange from stab	oilizer STA 258.28 to	Outboard Tip.				

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 258.28 to Outboard Tip. See Doc. D626A001 - DTR, DTR check form 55-10-06-4E, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-54.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-616-09-02	AWL	55-05-02-250-884	1.1	56000 FC	6000 FC	ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of the Le	eck form 55-10 complish the intended are contained	-06-4E, for alternati ent of this inspection ed in Part 6, Subject	ive inspection. on is contained in ct 55-10-54.					
	ACCESS NO	TIE. Removal of the Le	ading Edge is i	equired for inspect	ion access.					
55-616-10-01	AWL	55-05-02-250-864	1.1	56000 FC	6000 FC	ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	tion requires probe place 26A001 - DTR, DTR che 26A001 - DTR, DTR che 26A00(s) necessary to accepte inspection procedure. The inspection procedure. Removal of the hor	eck form 55-10 complish the intended are contained	-06-3, for alternativent of this inspection of the control of the	on is contained in ct 55-10-46.		ctive Test Manu			
55-616-10-02	AWL	55-05-02-250-864	1.1	56000 FC	6000 FC	ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	tion requires probe place 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of the ho	eck form 55-10 complish the intended are contained	-06-3, for alternativent of this inspectioned in Part 6, Subjection	on is contained in ct 55-10-46.		ctive Test Manu			
55-616-11-01	AWL	55-05-02-250-889	1.1	56000 FC	2000 FC	ALL	ALL			
	Note: Inspec See Doc. D6	r Frequency Eddy Currer tion requires probe place 26A001 - DTR, DTR che DTE: Removal of the ho	ement on prime eck form 55-10	ed metal surface. -06-3, for alternativ	e inspection.		o stabilizer tip.			
55-616-11-02	AWL	55-05-02-250-889	1.1	56000 FC	2000 FC	ALL	ALL			
	Note: Inspec See Doc. D6	Inspect (Low Frequency Eddy Current) the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip. Note: Inspection requires probe placement on primed metal surface. See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection. ACCESS NOTE: Removal of the horizontal stabilizer removable leading edge is required.								
55-616-12-01	AWL	55-05-02-250-865	1.1	56000 FC	6000 FC	ALL	ALL			
13 0.0 12 01	Inspect (Low	Frequency Eddy Currer	nt) the front spa	ar upper chord web	flange from stab					

See Doc. D626A001 - DTR, DTR check form 55-10-06-4A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual

(D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-45.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
55-616-12-02	AWL	55-05-02-250-865	1.1	56000 FC	6000 FC	ALL	ALL				
	See Doc. D6 The NDI met	Inspect (Low Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 66.5 to stabilizer STA 75.0 See Doc. D626A001 - DTR, DTR check form 55-10-06-4A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-45.									
	ACCESS NO	OTE: Removal of the Lea	ading Edge is r	equired for inspect	tion access.						
55-616-14-01	AWL	55-05-02-250-868	1.1	56000 FC	24000 FC	ALL	ALL				
	258.28.	n Frequency Eddy Currer	,		o flange from stab	ilizer STA 175.5 to s	stabilizer STA				
		26A001 - DTR, DTR che	•		ive inspection.						
	The NDI met	hod(s) necessary to according to the inspection procedure.	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Manu				
	ACCESS NO	OTE: Removal of the hor inspection access.	izontal stabiliz	er leading edge an	d the removable f	ront spar web is red	quired for				
55-616-14-02	AWL	55-05-02-250-868	1.1	56000 FC	24000 FC	ALL	ALL				
	Note: Inspection requires probe placement on primed metal surface. See Doc. D626A001 - DTR, DTR check form 55-10-06-4B, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-47. ACCESS NOTE: Removal of the horizontal stabilizer leading edge and the removable front spar web is required for inspection access.										
55-616-16-01	AWL	55-05-02-211-804	1.1	56000 FC	24000 FC	ALL	ALL				
	Inspect (Detailed) the front spar lower chord forward flange from stabilizer STA 66.50 to stabilizer STA 175.50. See Doc. D626A001 - DTR, DTR check form 55-10-06-6A, for alternative inspection.										
	ACCESS NO	OTE: Removal of the lea flange.	ding edge pan	els is required to g	ain access to the	lower surface of the	e forward				
55-616-16-02	AWL	55-05-02-211-804	1.1	56000 FC	24000 FC	ALL	ALL				
		Inspect (Detailed) the front spar lower chord forward flange from stabilizer STA 66.50 to stabilizer STA 175.50. See Doc. D626A001 - DTR, DTR check form 55-10-06-6A, for alternative inspection.									
	ACCESS NO	DTE: Removal of the lea flange.	ding edge pan	els is required to g	ain access to the	lower surface of the	e forward				
55-616-18-01	AWL	55-05-02-250-869	1.1	56000 FC	36000 FC	ALL	ALL				
	Inspect (Low	Frequency Eddy Curren	4\ 4l= = f===4 ===								

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-53.

INDEX

See Doc. D626A001 - DTR, DTR check form 55-10-06-6B, for alternative inspection.



55-616-24-01

AWL



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-616-18-02	AWL	55-05-02-250-869	1.1	56000 FC	36000 FC	ALL	ALL			
	Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 66.50 to stabilizer STA 175.50. See Doc. D626A001 - DTR, DTR check form 55-10-06-6B, for alternative inspection.									
	The NDI me	thod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspecti	ion is contained in	the 737 Nondestru	ctive Test Manual			
55-616-20-01	AWL	55-05-02-250-870	1.1	56000 FC	24000 FC	ALL	ALL			
	Inspect (High STA 258.28.	h Frequency Eddy Curre	nt) the front sp	ar lower chord for	ward flange from s	tabilizer STA 175.50	to stabilizer			
	The NDI me	s26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspecti	ion is contained in	the 737 Nondestruc	ctive Test Manual			
	ACCESS NO	OTE: Removal of the lea	ading edge pan	el is required to ga	ain access to the lo	ower surface of the	forward			
55-616-20-02	AWL	55-05-02-250-870	1.1	56000 FC	24000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 175.50 to stabilizer STA 258.28. See Doc. D626A001 - DTR, DTR check form 55-10-06-7A, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51. ACCESS NOTE: Removal of the leading edge panel is required to gain access to the lower surface of the forward flange.									
55-616-22-01	AWL	55-05-02-250-872	1.1	56000 FC	36000 FC	ALL	ALL			
	258.28. See Doc. D6 The NDI me	r Frequency Eddy Currer 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the int	-06-7B, for alterna ent of this inspecti	ntive inspection.					
55-616-22-02	AWL	55-05-02-250-872	1.1	56000 FC	36000 FC	ALL	ALL			
	Inspect (Low 258.28.	Frequency Eddy Currer	nt) the front spa	ar lower chord aft f	flange from stabiliz	er STA 175.50 to st	abilizer STA			
	The NDI me	s26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspecti	ion is contained in	the 737 Nondestrud	ctive Test Manual			

Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 258.28 to stabilizer BL 281.81.

24000 FC

ALL

56000 FC

See Doc. D626A001 - DTR, DTR check form 55-10-06-8A, for alternative inspection.

1.1

55-05-02-250-874

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

ACCESS NOTE: Leading edge skin assembly removal is required for the inspection.

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ALL





			APPLICA	ABILITY						
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-616-24-02	AWL	55-05-02-250-874	1.1	56000 FC	24000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 258.28 to stabilizer BL 281.81.									
	See Doc. D626A001 - DTR, DTR check form 55-10-06-8A, for alternative inspection.									
		ethod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	ctive Test Manu			
	ACCESS N	OTE: Leading edge skin	assembly rem	oval is required for	the inspection.					
55-616-26-01	AWL	55-05-02-250-873	1.1	56000 FC	4000 FC	ALL	ALL			
	281.81 (tip). See Doc. Do The NDI me	v Frequency Eddy Currer 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the int	-06-8B, for alternat	tive inspection. on is contained in					
55-616-26-02	AWL	55-05-02-250-873	1.1	56000 FC	4000 FC	ALL	ALL			
		ethod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestruc	ctive Test Man			
55-616-28-01	AWL	55-05-02-250-875	1.1	56000 FC	18000 FC	ALL	ALL			
	STA 66.5 to See Doc. Do The NDI me (D6-37239).	th Frequency Eddy Currer stabilizer STA 111.1. 626A001 - DTR, DTR che ethod(s) necessary to acc The inspection procedur OTE: The inspection req	eck form 55-10 omplish the int es are contain	-06-W1, for alterna ent of this inspection ed in Part 6, Section	ative inspection. on is contained in on 55-10-58.	the 737 Nondestruc				
55-616-28-02	AWL	55-05-02-250-875	1.1	56000 FC	18000 FC	ALL	ALL			
	STA 66.5 to See Doc. Do The NDI me (D6-37239).	th Frequency Eddy Currer stabilizer STA 111.1. 626A001 - DTR, DTR che ethod(s) necessary to acc The inspection procedur OTE: The inspection req	eck form 55-10 omplish the int es are contain	-06-W1, for alterna ent of this inspection ed in Part 6, Section	ative inspection. on is contained in on 55-10-58.	the 737 Nondestruc				
55-616-30-01	AWL	55-05-02-250-876	1.1	56000 FC	15000 FC	ALL	ALL			
	Inspect (Hig	th Frequency Eddy Current ction requires probe place	nt) the front sp	ar web splice, at th	e upper and lowe					

See Doc. D626A001 - DTR, DTR check form 55-10-06-WS, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.







				INTERVAL			APPLICABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
55-616-30-02	AWL	55-05-02-250-876	1.1	56000 FC	15000 FC	ALL	ALL				
	Inspect (High Frequency Eddy Current) the front spar web splice, at the upper and lower web edges at stabilizer STA 111.1										
	Note: Inspection requires probe placement on primed metal surface.										
	See Doc. D6	26A001 - DTR, DTR che	eck form 55-10	-06-WS, for alterna	ative inspection.						
	The NDI met	hod(s) necessary to acc	omplish the int	ent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Man				
	(D6-37239).	The inspection procedur	es are contain	ed in Part 6, Sectio	n 55-10-58.						

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

55-616-32-01 AWL 55-05-02-211-814 1.1 56000 FC 12000 FC ALL ALL

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 111.1 to stabilizer STA 175.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any sealant which exceeds .30" on either side of a fastener head or collar.

55-616-32-02 AWL 55-05-02-211-814 1.1 56000 FC 12000 FC ALL ALL

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 111.1 to stabilizer STA 175.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any sealant which exceeds .30" on either side of a fastener head or collar.

55-616-34-01 AWL 55-05-02-250-878 1.1 56000 FC 18000 FC ALL ALL

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 175.5 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

Remove any cap seal, that is present, on nut plated BACB30NM fasteners.

55-616-34-02 AWL 55-05-02-250-878 1.1 56000 FC 18000 FC ALL ALL

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 175.5 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

Remove any cap seal, that is present, on nut plated BACB30NM fasteners.





				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
55-616-36-01	AWL	55-05-02-250-879	1.1	56000 FC	18000 FC	ALL	ALL		
	STA 258.28 t See Doc. D6 The NDI met	n Frequency Eddy Curre to stabilizer tip. 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the int	-06-W4, for alterna tent of this inspection	tive inspection. on is contained in				
	,	The inspection req	uires removal	of the horizontal sta		lge.			
55-616-36-02	AWL	55-05-02-250-879	1.1	56000 FC	18000 FC	ALL	ALL		
	STA 258.28 t See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre to stabilizer tip. 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	eck form 55-10 omplish the interest are contain	-06-W4, for alterna tent of this inspection ed in Part 6, Section	ntive inspection. on is contained in on 55-10-58.	the 737 Nondestruc			
		OTE: The inspection req	sealant that is p	present.					
55-617-00-01	AWL	55-05-02-211-803	1.1	56000 FC	18000 FC	ALL	ALL		
	Inspect (Detailed) the exposed surface of the front spar upper chord between the upper skin and Leading Edge (LE) skin. See Doc. D626A001 - DTR, DTR check form 55-10-07-2, for alternative inspection.								
55-617-00-02	AWL	55-05-02-211-803	1.1	56000 FC	18000 FC	ALL	ALL		
		ailed) the exposed surface 26A001 - DTR, DTR che				skin and Leading E	dge (LE) skin.		
55-618-00-01	AWL	55-05-02-230-802	1.1	56000 FC	12000 FC	ALL	ALL		
	Inspect (Penetrant) the upper and lower Side of Body (SOB) spar bolts on the horizontal stabilizer front spar terminal fitting. See Doc. D626A001 - DTR, DTR check form 55-10-07-3, for alternative inspection. ACCESS NOTE: Removal of upper and lower spar bolts is required.								
55-618-00-02	AWL	55-05-02-230-802	1.1	56000 FC	12000 FC	ALL	ALL		
22 0.0 00 02	Inspect (Pen See Doc. D6	etrant) the upper and lov 26A001 - DTR, DTR che	wer Side of Bo	dy (SOB) spar bolts -07-3, for alternativ	s on the horizonta				
	ACCESS NO	TE: Removal of upper	and lower spa	r doits is required.					
55-618-10-01	AWL	55-05-02-250-880	1.1	56000 FC	36000 FC	ALL	ALL		
	Inanaat /I liak	frequency Eddy Currer	t) the lower ch	ord at the and rib to	oncion fitting incto	lation halt halo			

See Doc. D626A001 - DTR, DTR check form 55-10-07-4, for alternative inspection.

ACCESS NOTE: The inspection requires the removal of the bolt.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-618-10-02	AWL	55-05-02-250-880	1.1	56000 FC	36000 FC	ALL	ALL
	See Doc. D6	n frequency Eddy Curren 26A001 - DTR, DTR che DTE: The inspection req	ck form 55-10	-07-4, for alternativ	· ·	llation bolt hole.	
55-619-00-01	AWL	55-05-02-250-816	1.1	56000 FC	36000 FC	ALL	ALL
	176. See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Currer 26A001 - DTR, DTR che thod(s) necessary to according The inspection procedur DTE: Removal of hinge p not be removed.	eck form 55-10 omplish the int es are contain	-08-1, for alternativent of this inspection of the control of the	ve inspection. on is contained in to	the 737 Nondestruc	ctive Test Manual
55-619-00-02	AWL	55-05-02-250-816	1.1	56000 FC	36000 FC	ALL	ALL
	The NDI met (D6-37239).	226A001 - DTR, DTR che thod(s) necessary to according The inspection procedure DTE: Removal of hinge p not be removed.	omplish the int es are contain	ent of this inspecti ed in Part 6, Subje	on is contained in to		
55-620-00-01	AWL	55-05-02-250-817	1.1	56000 FC	36000 FC	ALL	ALL
	250. See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Current 26A001 - DTR, DTR che thod(s) necessary to accommod The inspection procedure DTE: Removal of hinge procedure not be removed.	eck form 55-10 omplish the int es are contain	-08-2, for alternativent of this inspection of the control of the	ve inspection. on is contained in to	the 737 Nondestrud	ctive Test Manual
55-620-00-02	AWL	55-05-02-250-817	1.1	56000 FC	36000 FC	ALL	ALL
	250. See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Current 226A001 - DTR, DTR che shod(s) necessary to according The inspection procedure DTE: Removal of hinge protection of be removed.	eck form 55-10 omplish the int es are contain	-08-2, for alternativent of this inspection of this inspection of the contract	ve inspection. on is contained in tot 55-10-35.	the 737 Nondestrud	ctive Test Manual
55-621-00-01	AWL	55-05-02-250-818	1.1	56000 FC	36000 FC	ALL	ALL
		Fraguanay Eddy Curra					

Inspect (High Frequency Eddy Current) all of the holes on the hinge plates around the bore at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

ACCESS NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-621-00-02	AWL	55-05-02-250-818	1.1	56000 FC	36000 FC	ALL	ALL
	See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of hinge	eck form 55-10 omplish the int es are contain	-08-3, for alternativent of this inspection of the control of the	ve inspection. on is contained in ct 55-10-36.	the 737 Nondestruc	
55-622-00-01	AWL	55-05-02-250-819	1.1	56000 FC	36000 FC	ALL	ALL
	See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of hinge	eck form 55-10 omplish the int es are contain	-08-4, for alternativent of this inspection of the inspection of t	ve inspection. on is contained in ct 55-10-36.		ctive Test Manu
55-622-00-02	AWL	55-05-02-250-819	1.1	56000 FC	36000 FC	ALL	ALL
	The NDI met (D6-37239).	626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of hinge	omplish the int es are contain	ent of this inspection and in Part 6, Subject	on is contained in ct 55-10-36.		ctive Test Manu
55-623-01-01	AWL	55-05-02-250-821	1.1	56000 FC	36000 FC	ALL	ALL
	edge clevis I See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre ugs at elevator STA 121 326A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur NOTE: This inspection a	and elevator S ck form 55-10-0 omplish the int es are contain	TA 176. 08-5B, for alternativent of this inspection ed in Part 6, Subject	ve inspection. on is contained in ct 55-10-23.	the 737 Nondestruc	ctive Test Manu
55-623-01-02	AWL	55-05-02-250-821	1.1	56000 FC	36000 FC	ALL	ALL
	edge clevis I See Doc. D6 The NDI met (D6-37239).	h Frequency Eddy Curre ugs at elevator STA 121 226A001-DTR, DTR chec thod(s) necessary to acc The inspection procedur NOTE: This inspection a	and elevator S ck form 55-10-0 omplish the int es are contain	TA 176. 08-5B, for alternativent of this inspection ed in Part 6, Subjection	ve inspection. on is contained in ct 55-10-23.	the 737 Nondestruc	ctive Test Manu
55-624-00-01	AWL	55-05-02-250-822	1.1	56000 FC	36000 FC	ALL	ALL

See Doc. D626A001-DTR, DTR check form 55-10-08-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-624-00-02	AWL	55-05-02-250-822	1.1	56000 FC	36000 FC	ALL	ALL
	Inspect (Hig 66.	h Frequency Eddy Curre	nt) around the	bushing flanges or	both the upper ar	nd lower clevis lugs	at elevator ST/
	See Doc. D6	626A001-DTR, DTR chec	ck form 55-10-0	8-6, for alternative	inspection.		
		thod(s) necessary to acc				he 737 Nondestruc	ctive Test Manu
	(D6-37239).	The inspection procedur	es are contain	ed in Part 6, Subje	ct 55-10-23.		
55-625-00-01	AWL	55-05-02-250-823	1.1	56000 FC	18000 FC	ALL	ALL
		h Frequency Eddy Curre	nt) around the	bushing flanges at	both the upper an	d lower clevis lugs	at elevator STA
		vator STA 250. 626A001 - DTR, DTR che	ok form EE 10	09.7 alternative in	anaction		
		thod(s) necessary to acc		·	•	he 737 Nondestru	ctive Test Manu
		The inspection procedur	•			ine 131 Nondestruc	Stive lest Maric
	(= = = = = =).						
55-625-00-02	AWL	55-05-02-250-823	1.1	56000 FC	18000 FC	ALL	ALL
55-625-00-02	Inspect (Hig	55-05-02-250-823 h Frequency Eddy Currer vator STA 250.					
55-625-00-02	Inspect (Hig	h Frequency Eddy Curre	nt) around the	bushing flanges at	both the upper an		
55-625-00-02	Inspect (High 213 and elev See Doc. De	h Frequency Eddy Curre vator STA 250.	nt) around the	bushing flanges at	both the upper an	d lower clevis lugs	at elevator STA
55-625-00-02	Inspect (Hig 213 and elev See Doc. Do The NDI me	h Frequency Eddy Currer vator STA 250. 626A001 - DTR, DTR che	nt) around the eck form 55-10 omplish the int	bushing flanges at -08-7, alternative ir ent of this inspection	both the upper an aspection.	d lower clevis lugs	at elevator STA
55-625-00-02 55-626-00-01	Inspect (Hig 213 and elev See Doc. Do The NDI me	h Frequency Eddy Currel vator STA 250. 626A001 - DTR, DTR che thod(s) necessary to acc	nt) around the eck form 55-10 omplish the int	bushing flanges at -08-7, alternative ir ent of this inspection	both the upper an aspection.	d lower clevis lugs	at elevator STA
	Inspect (Hig 213 and election See Doc. Doc. The NDI me (D6-37239).	h Frequency Eddy Currel vator STA 250. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur 55-05-02-250-824	nt) around the eck form 55-10 omplish the int es are containe	bushing flanges at -08-7, alternative ir ent of this inspection ed in Part 6, Subjection	both the upper an aspection. on is contained in tot 55-10-23.	d lower clevis lugs the 737 Nondestruc	at elevator STA ctive Test Manu ALL
	Inspect (Hig 213 and election See Doc. Doc. The NDI me (D6-37239).	h Frequency Eddy Currel vator STA 250. 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	nt) around the eck form 55-10 omplish the int es are containe	bushing flanges at -08-7, alternative ir ent of this inspection ed in Part 6, Subjection	both the upper an aspection. on is contained in tot 55-10-23.	d lower clevis lugs the 737 Nondestruc	at elevator STA ctive Test Manu ALL
	Inspect (Hig 213 and elev See Doc. Do The NDI me (D6-37239). AWL Inspect (Hig STA 265.	h Frequency Eddy Currel vator STA 250. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur 55-05-02-250-824	nt) around the eck form 55-10 omplish the int es are containe	bushing flanges at -08-7, alternative ir ent of this inspection ed in Part 6, Subjection	both the upper an aspection. on is contained in tot 55-10-23.	d lower clevis lugs the 737 Nondestruc	at elevator STA ctive Test Manu ALL
	Inspect (Hig 213 and elex See Doc. Do The NDI me (D6-37239). AWL Inspect (Hig STA 265. Note: Bushir	h Frequency Eddy Currel vator STA 250. 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur 55-05-02-250-824 h Frequency Eddy Currel	nt) around the eck form 55-10 omplish the int es are contained 1.1 nt) around the	bushing flanges at -08-7, alternative ir ent of this inspectic ed in Part 6, Subject 56000 FC edge of the bushin	both the upper an inspection. on is contained in the ct 55-10-23. 18000 FC g at both the upper	d lower clevis lugs the 737 Nondestruc	at elevator ST/ ctive Test Manu ALL
	Inspect (Hig 213 and elex See Doc. Do The NDI me (D6-37239). AWL Inspect (Hig STA 265. Note: Bushir See Doc. Do The NDI me	h Frequency Eddy Currel vator STA 250. 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur 55-05-02-250-824 h Frequency Eddy Currel	nt) around the eck form 55-10 omplish the interest are contained. 1.1 nt) around the eck form 55-10 omplish the interest are contained.	bushing flanges at -08-7, alternative ir ent of this inspectic ed in Part 6, Subject 56000 FC edge of the bushin -08-8, alternative ir ent of this inspectic	both the upper an inspection. on is contained in the ct 55-10-23. 18000 FC g at both the upper an inspection. on is contained in the containe	d lower clevis lugs the 737 Nondestruc ALL r and lower clevis l	at elevator STA ctive Test Manu ALL ugs at elevator

55-626-00-02

AWL 55-05-02-250-824

1.1

56000 FC

18000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) around the edge of the bushing at both the upper and lower clevis lugs at elevator STA 265.

Note: Bushing removal not required.

See Doc. D626A001 - DTR, DTR check form 55-10-08-8, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

55-627-00-01

AWL

55-05-02-250-825

1.1

56000 FC

12000 FC

ALL

ALL

Inspect (High Frequency Eddy Current) the elevator hinge rib chord around the fasteners common to the upper and lower spar chords at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-08-9, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of lower composite skin panels is required.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-627-00-02	AWL	55-05-02-250-825	1.1	56000 FC	12000 FC	ALL	ALL
	spar chords a See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre at elevator STA 66 and e 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of lower of	elevator STA 12 eck form 55-10 complish the interes are contain	2108-9, alternative interpretent of this inspection of the subjection of the subjection of the subject in Part 6, Subjection of the subjection of the subject in Part 6, Subjection of the subject in Part 6, Subjection of the subjection of the subject in Part 6,	nspection. on is contained in ect 55-10-37.	·	
55-628-00-01	AWL	55-05-02-250-826	1.1	56000 FC	24000 FC	ALL	ALL
	The NDI met (D6-37239).	226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of lower to	complish the intress are contain	ent of this inspecti ed in Part 6, Subje	on is contained in	the 737 Nondestruc	ctive Test Manua
55-628-00-02	AWL	55-05-02-250-826	1.1	56000 FC	24000 FC	ALL	ALL
	spar chord a See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre t elevator STA 176. :26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of lower	eck form 55-10 complish the interes are contain	-8-10, alternative in ent of this inspection ed in Part 6, Subje	nspection. on is contained in		
	ACCESS NO	JIE. Removal of lower		· 			
55-629-00-01	AWL	55-05-02-250-827	1.1	56000 FC	30000 FC	ALL	ALL
	elevator STA See Doc. D6	n Frequency Eddy Curre 213, elevator STA 250 a 26A001 - DTR, DTR che	and elevator S eck form 55-10	TA 265. -8-11, alternative ir	nspection.		chords, at

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of elevator and cover panel is required. No fastener removal is required.

55-629-00-02 AWL 55-05-02-250-827 1.1 56000 FC 30000 FC ALL ALL

Inspect (High Frequency Eddy Current) the hinge rib to rear spar attachment, on both the upper and lower chords, at elevator STA 213, elevator STA 250 and elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-8-11, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of elevator and cover panel is required. No fastener removal is required.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-630-00-01	AWL	55-05-02-250-828	1.1	56000 FC	24000 FC	ALL	ALL
	rib and the u and elevator See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre upper chord and lower chord and lower chord and lower chord and 121. 526A001 - DTR, DTR chord thod(s) necessary to accurate inspection procedur OTE: Removal of lower of the inspection of lower of the inspection procedur	ord fastener lo eck form 55-10 omplish the int es are contain	cations between th -8-13, for alternative tent of this inspection ed in Part 6, Subject	re rear spar and tr re inspection. on is contained in ct 55-10-33.	railing edge beam a	t elevator STA 6
	A00200 III	ore. Removar or lower (oompoone onn	r pariolo lo required	. No lasterior rem	ovar roquirou.	
55-630-00-02	AWL	55-05-02-250-828	1.1	56000 FC	24000 FC	ALL	ALL
	rib and the u and elevator See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre upper chord and lower chord and lower chord 121. 526A001 - DTR, DTR che thod(s) necessary to accomb inspection procedur The inspection procedur The Removal of lower of the comb inspection procedur	ord fastener lo eck form 55-10 omplish the int es are contain	cations between the -8-13, for alternative tent of this inspection ed in Part 6, Subject	re rear spar and tr re inspection. on is contained in ct 55-10-33.	railing edge beam a	t elevator STA 6
55-631-00-01	AWL	55-05-02-250-829	1.1	56000 FC	18000 FC	ALL	ALL
	edge panels See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Currer between the stabilizer respectively. DTR, DTR chesthod(s) necessary to accomple inspection procedure. Removal of the low	ear spar and the eck form 55-10 omplish the intees are contain	ne trailing edge bea 1-8-14, for alternativ tent of this inspection ed in Part 6, Subject	m at elevator STA re inspection. on is contained in ct 55-10-33.	A 176.	, and the second
55-631-00-02	AWL	55-05-02-250-829	1.1	56000 FC	18000 FC	ALL	ALL
	edge panels See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre between the stabilizer re 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Removal of the low	ear spar and the eck form 55-10 omplish the intes es are contain	te trailing edge bea 1-8-14, for alternative tent of this inspection ed in Part 6, Subject	m at elevator STA re inspection. on is contained in ct 55-10-33.	A 176.	J
55-631-10-01	AWL	55-05-02-250-809	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (High	h Frequency Eddy Curre zer STA. 157.1. 626A001 - DTR, DTR che	,	astener locations in	n the upper inspar		
55-631-10-02	AWL	55-05-02-250-809	1.1	56000 FC	18000 FC	ALL	ALL
00 00: 10-0 <u>2</u>	/ \	00 00 02-200-009	1.1	0000010	1000010	/ \	/\LL

Inspect (High Frequency Eddy Current) around all fastener locations in the upper inspar skin between the SOB rib and the rib at stabilizer STA. 157.1.

See Doc. D626A001 - DTR, DTR check form 55-10-09-1, alternative inspection.





55-633-00-01 Ins Se 55-633-00-02 Ins Se AC 55-634-00-02 Ins Se AC	AWL spect (Detage Doc. D6 AWL spect (Detage Doc. D6 AWL spect (Detage Doc. D6 CCESS NC AWL spect (Detage Doc. D6	AMM TASK REF 55-05-02-211-805 ailed) the horizontal stabilities of the portion	1.1 dilizer lower instack form 55-10 1.1 ar upper side of eck form 55-10 g edge and gal 1.1 ar upper side of eck form 55-10 g edge and gal	56000 FC par skin from side- 10-1, for alternative 56000 FC of body clevis lugs. 11-1, for alternative p seals is required. 56000 FC of body clevis lugs.	9000 FC of-body to the tip. re inspection. 18000 FC re inspection.	ALL ALL ALL	ALL ALL
55-633-00-02 Ins Se 55-634-00-01 Ins Se AC	AWL spect (Deta ee Doc. D6	ailed) the horizontal stabilizedA001 - DTR, DTR ches 55-05-02-211-805 ailed) the horizontal stabilizedA001 - DTR, DTR ches 55-05-02-211-806 ailed) all four (4) front spansor of leading 55-05-02-211-806 ailed) all four (4) front spansor of leading 55-05-02-211-806 ailed) all four (4) front spansor of leading 55-05-02-211-806 ailed) all four (4) front spansor of leading 526A001 - DTR, DTR ches 526A001 - DTR, DTR ches	1.1 Alizer lower insect form 55-10 1.1 Alizer lower insect form 55-10 1.1 ar upper side of the control of	par skin from side-i-10-1, for alternative 56000 FC par skin from side-i-10-1, for alternative 56000 FC of body clevis lugs. i-11-1, for alternative p seals is required. 56000 FC of body clevis lugs.	of-body to the tip. we inspection. 9000 FC of-body to the tip. we inspection. 18000 FC the inspection.	ALL	ALL
55-633-00-02 Ins Se AC 55-634-00-02 Ins Se AC	AWL spect (Detage Doc. D6 AWL spect (Detage Doc. D6 AWL spect (Detage Doc. D6 CCESS NC AWL spect (Detage Doc. D6	55-05-02-211-805 ailed) the horizontal stabilized A001 - DTR, DTR che 55-05-02-211-806 ailed) all four (4) front spansized A001 - DTR, DTR che DTE: Removal of leading 55-05-02-211-806 ailed) all four (4) front spansized A001 - DTR, DTR che 26A001 - DTR, DTR che 26A001 - DTR, DTR che	1.1 dilizer lower instack form 55-10 1.1 ar upper side of eck form 55-10 g edge and gal 1.1 ar upper side of eck form 55-10 g edge and gal	56000 FC par skin from side- 10-1, for alternative 56000 FC of body clevis lugs. 11-1, for alternative p seals is required. 56000 FC of body clevis lugs.	9000 FC of-body to the tip. re inspection. 18000 FC re inspection.	ALL	ALL
55-634-00-01 Ins Se AC 55-634-00-02 Ins Se AC	AWL Spect (Deta ee Doc. D6 AWL CCESS NC AWL Spect (Deta ee Doc. D6	ailed) the horizontal stabilizedA001 - DTR, DTR ches 55-05-02-211-806 ailed) all four (4) front spansizedA001 - DTR, DTR ches 55-05-02-211-806 ailed) all four (4) front spansizedA001 - DTR, DTR ches 226A001 - DTR, DTR ches 226A001 - DTR, DTR ches	1.1 ar upper side of gedge and gaper side of the side	par skin from side-i-10-1, for alternative 56000 FC of body clevis lugs. i-11-1, for alternative p seals is required. 56000 FC of body clevis lugs.	of-body to the tip. ve inspection. 18000 FC ve inspection.	ALL	ALL
55-634-00-01 Ins Se AC 55-634-00-02 Ins Se AC	AWL spect (Detable Doc. D6 AWL spect (Detable Doc. D6	55-05-02-211-806 ailed) all four (4) front spa 526A001 - DTR, DTR che DTE: Removal of leading 55-05-02-211-806 ailed) all four (4) front spa 526A001 - DTR, DTR che	1.1 ar upper side of the control of	56000 FC of body clevis lugs. 1-11-1, for alternative p seals is required. 56000 FC of body clevis lugs.	18000 FC re inspection. 18000 FC		
55-634-00-02 Ins	AWL spect (Deta	ailed) all four (4) front spa 226A001 - DTR, DTR che DTE: Removal of leading 55-05-02-211-806 ailed) all four (4) front spa 226A001 - DTR, DTR che	ar upper side of the control of the	of body clevis lugs11-1, for alternativ p seals is required. 56000 FC of body clevis lugs.	e inspection.		
55-634-00-02 Ins Se AC	AWL spect (Deta	226A001 - DTR, DTR che DTE: Removal of leading 55-05-02-211-806 ailed) all four (4) front spa 226A001 - DTR, DTR che	eck form 55-10 g edge and gap 1.1 ar upper side c	p seals is required. 56000 FC of body clevis lugs.	18000 FC	ALL	ALL
55-634-00-02 Ins Se AC	AWL spect (Deta	55-05-02-211-806 ailed) all four (4) front spa 26A001 - DTR, DTR che	1.1 ar upper side o	56000 FC of body clevis lugs.	18000 FC	ALL	ALL
Ins Se AC	spect (Deta	ailed) all four (4) front spa 26A001 - DTR, DTR che	ar upper side o	of body clevis lugs.		ALL	ALL
Se A C	ee Doc. D6	626A001 - DTR, DTR che		,	re inspection.		
			edge and gai	n seals is required.	-		
	AWL	55-05-02-130-805	1.1	56000 FC	36000 FC	ALL	ALL
Se Th (De	ee Doc. D6 ne NDI met 6-37239).	asonic) all four lower pivo 226A001 - DTR, DTR che thod(s) necessary to acco The inspection procedure DTE: Removal of the gap	eck form 55-10 complish the intest es are contain	1-12-1, for alternative tent of this inspection ed in Part 4, Subjection	re inspection. on is contained in to		ctive Test Ma
55-635-00-02	AWL	55-05-02-130-805	1.1	56000 FC	36000 FC	ALL	ALL
Se Th	ee Doc. D6 ne NDI met	asonic) all four lower pivo 26A001 - DTR, DTR che thod(s) necessary to acco The inspection procedure	eck form 55-10 omplish the int	-12-1, for alternative	re inspection. on is contained in t		ctive Test Ma
AC	CCESS NO	OTE: Removal of the gap	o covers and s	liding seals is requ	ired.		
55-636-00-01	AWL	55-05-02-130-806	1.1	56000 FC	4000 FC	ALL	ALL

See Doc. D626A001 - DTR, DTR check form 55-10-12-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-08.

ACCESS NOTE: Removal of gap covers is required.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-636-00-02	AWL	55-05-02-130-806	1.1	56000 FC	4000 FC	ALL	ALL
	See Doc. D6 The NDI mer (D6-37239).	asonic) the side of body s26A001 - DTR, DTR char thod(s) necessary to acc The inspection procedur DTE: Removal of gap co	eck form 55-10 complish the intres are contain	-12-2, for alternativent of this inspectioned in Part 4, Subjection	e inspection. on is contained in		
55-637-00-01	AWL	55-05-02-211-807	1.1	56000 FC	36000 FC	ALL	ALL
		ailed) the thrust beam to 26A001 - DTR, DTR ch				n front spar upper cl	nord.
55-637-00-02	AWL	55-05-02-211-807	1.1	56000 FC	36000 FC	ALL	ALL
		ailed) the thrust beam to 26A001 - DTR, DTR ch				n front spar upper cl	nord.
55-638-00-01	AWL	55-05-02-130-807	1.1	56000 FC	9000 FC	ALL	ALL
	See Doc. D6	asonic) the upper and lo 326A001 - DTR, DTR cho DTE: Access horizontal	eck form 55-10	-14-1, for alternativ	e inspection.		
55-638-00-02	AWL	55-05-02-130-807	1.1	56000 FC	9000 FC	ALL	ALL
	See Doc. D6	asonic) the upper and lo 326A001 - DTR, DTR cho DTE: Access horizontal	eck form 55-10	-14-1, for alternativ	e inspection.	, ,	
55-639-00-01	AWL	55-05-02-250-830	1.1	56000 FC	9000 FC	ALL	ALL
	and rear spa See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre ir joint plates and angles (26A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedur DTE: Access horizontal	eck form 55-10 complish the int res are contain	-14-2, for alternativ ent of this inspectic ed in Part 6, Subjec	e inspection. on is contained in ct 55-10-38.	the 737 Nondestru	
55-639-00-02	AWL	55-05-02-250-830	1.1	56000 FC	9000 FC	ALL	ALL

and rear spar joint plates and angles.

See Doc. D626A001 - DTR, DTR check form 55-10-14-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-38.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-640-00-01	AWL	55-05-02-130-808	1.1	56000 FC	9000 FC	ALL	ALL
	the two chorn See Doc. Do The NDI me	asonic) the upper primar ds and splice angle. 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10- omplish the int	-14-3, for alternativent of this inspection	re inspection. on is contained in		
	ACCESS NO	OTE: Access horizontal	stabilizer cente	r section through o	ppening in center	of 1088 bulkhead.	
55-640-00-02	AWL	55-05-02-130-808	1.1	56000 FC	9000 FC	ALL	ALL
	the two chorn See Doc. D6 The NDI met (D6-37239).	asonic) the upper primar ds and splice angle. 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Access horizontal	eck form 55-10- omplish the int es are containe	-14-3, for alternativent of this inspectioned in Part 4, Subjection	re inspection. on is contained in ct 55-10-14.	the 737 Nondestruc	
55-641-00-01	AWL	55-05-02-130-809	1.1	56000 FC	21000 FC	ALL	ALL
	members at See Doc. D6	asonic) the thrust beam of both the front and rear s i26A001 - DTR, DTR che DTE: Access horizontal	par joints. eck form 55-10	-14-4, for alternativ	re inspection.		e splice
55-641-00-02	AWL	55-05-02-130-809	1.1	56000 FC	21000 FC	ALL	ALL
	members at See Doc. D6	asonic) the thrust beam of both the front and rear s i26A001 - DTR, DTR che DTE: Access horizontal	par joints. eck form 55-10	-14-4, for alternativ	re inspection.		e splice
55-642-00-01	AWL	55-05-02-250-831	1.1	56000 FC	34000 FC	ALL	ALL
	to the splice	n Frequency Eddy Curre members at both the fro 26A001 - DTR, DTR che	nt and rear spa	ırs. -14-5, for alternativ		the first row of fast	eners common

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-39.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

55-642-00-02 AWL 55-05-02-250-831 1.1 56000 FC 34000 FC ALL ALL

> Inspect (High Frequency Eddy Current) the upper and lower thrust beam chords around the first row of fasteners common to the splice members at both the front and rear spars.

See Doc. D626A001 - DTR, DTR check form 55-10-14-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-39.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-643-00-01	AWL	55-05-02-130-810	1.1	56000 FC	14000 FC	ALL	ALL
	Inspect (Ultr	asonic) upper and lower	thrust beam ch	nords forward and a	aft of the intersect	tion joint (4 places).	
	See Doc. D6	326A001 - DTR, DTR che	eck form 55-10	-14-6, for alternativ	e inspection.		
	The NDI me (D6-37239).	the 737 Nondestruc	ctive Test Manu				
	ACCESS NO	OTE: Access horizontal	stabilizer cente	er section through o	ppening in center	of 1088 bulkhead.	
55-643-00-02	AWL	55-05-02-130-810	1.1	56000 FC	14000 FC	ALL	ALL
		asonic) upper and lower				tion joint (4 places).	
		326A001 - DTR, DTR che		•	•	the 707 New Jeetse	
		thod(s) necessary to acc The inspection procedur		•		the 737 Nondestruc	ctive lest Manu
	ACCESS NO	OTE: Access horizontal	stabilizer cente	er section through o	pening in center	of 1088 bulkhead.	
55-644-00-01	AWL	55-05-02-250-832	1.1	56000 FC	27000 FC	ALL	ALL
	See Doc. Do	h Frequency Eddy Curre 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 complish the int	-14-7, for alternativent of this inspection	re inspection. on is contained in	the 737 Nondestruc	ctive Test Manu
	ACCESS NO	OTE: Access horizontal	stabilizer cente	er section through o	pening in center	of 1088 bulkhead.	
55-644-00-02	AWL	55-05-02-250-832	1.1	56000 FC	27000 FC	ALL	ALL
	See Doc. Do The NDI me	h Frequency Eddy Curre 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 complish the int	-14-7, for alternativent of this inspection	re inspection. on is contained in	the 737 Nondestruc	ctive Test Manu
	ACCESS NO	OTE: Access horizontal	stabilizer cente	er section through o	ppening in center	of 1088 bulkhead.	
55-645-00-01	AWL	55-05-02-130-811	1.1	56000 FC	13000 FC	ALL	ALL
	See Doc. Do The NDI me	asonic) the upper thrust 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-10 complish the int	-14-8, for alternativent of this inspection	re inspection. on is contained in	·	·
	,	OTE: Access horizontal				of 1088 bulkhead.	
55-645-00-02	AWL	55-05-02-130-811	1.1	56000 FC	13000 FC	ALL	ALL
	. ,	asonic) the upper thrust 326A001 - DTR, DTR che		-		common to the prim	ary beam chor

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-10.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.





				INTERVAL		APPLIC	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-646-00-01	AWL	55-05-02-250-833	1.1	56000 FC	8000 FC	ALL	ALL
	front and rea See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre ar spars from STA 1216 to 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	o STA 1242. eck form 55-10 omplish the int es are contain	-14-9, for alternative ent of this inspection ed in Part 6, Subject	e inspection. on is contained in ot 55-10-25.	the 737 Nondestru	
	ACCESS NO	OTE: Access horizontal	stabilizer cente	er section through o	pening in center of	ot 1088 buiknead.	
55-646-00-02	AWL	55-05-02-250-833	1.1	56000 FC	8000 FC	ALL	ALL
	front and rea See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre ar spars from STA 1216 to 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur OTE: Access horizontal	o STA 1242. eck form 55-10 omplish the int es are contain	-14-9, for alternative ent of this inspection ed in Part 6, Subject	e inspection. on is contained in ot 55-10-25.	the 737 Nondestru	
55-647-00-01	AWL	55-05-02-250-834	1.1	56000 FC	18000 FC	ALL	ALL
	See Doc. Do The NDI me	h Frequency Eddy Curre 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 55-20 omplish the int	-05, for alternative i ent of this inspectio	inspection. on is contained in		
55-647-00-02	AWL	55-05-02-250-834	1.1	56000 FC	18000 FC	ALL	ALL
55-647-00-02	Inspect (Hig See Doc. Do The NDI me	55-05-02-250-834 h Frequency Eddy Curre 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	nt) all hinge fitt eck form 55-20 omplish the int	ing clevis lugs on the -05, for alternative i ent of this inspectio	ne lug face around inspection. on is contained in	d the circumference	ALL e of the bushing
55-647-00-02 55-648-00-01	Inspect (Hig See Doc. Do The NDI me	h Frequency Eddy Curre 326A001 - DTR, DTR che thod(s) necessary to acc	nt) all hinge fitt eck form 55-20 omplish the int	ing clevis lugs on the -05, for alternative i ent of this inspectio	ne lug face around inspection. on is contained in	d the circumference	ALL e of the bushing
	Inspect (Hig See Doc. Do The NDI me (D6-37239). AWL Inspect (Hig See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	nt) all hinge fitteck form 55-20 omplish the interest are contained. 1.1 nt) each lug fareck form 55-20 omplish the interest are contained.	ing clevis lugs on the control of this inspection of this inspection of the control of this inspection of this inspection of the control of the	ne lug face around inspection. on is contained in n 55-50-07. 24000 FC ctuator fittings. inspection. on is contained in is contained in	d the circumference the 737 Nondestru ALL	ALL e of the bushing ctive Test Manu ALL
	Inspect (Hig See Doc. Do The NDI me (D6-37239). AWL Inspect (Hig See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre 626A001 - DTR, DTR che othod(s) necessary to acc The inspection procedur 55-05-02-250-835 h Frequency Eddy Curre 626A001 - DTR, DTR che othod(s) necessary to acc The inspection procedur	nt) all hinge fitteck form 55-20 omplish the interest are contained. 1.1 nt) each lug fareck form 55-20 omplish the interest are contained.	ing clevis lugs on the control of this inspection of this inspection of the control of this inspection of this inspection of the control of the	ne lug face around inspection. on is contained in n 55-50-07. 24000 FC ctuator fittings. inspection. on is contained in is contained in	d the circumference the 737 Nondestru ALL	ALL e of the bushing ctive Test Manu

Inspect (High Frequency Eddy Current) each lug face of the elevator actuator fittings.

See Doc. D626A001 - DTR, DTR check form 55-20-06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-09.

ACCESS NOTE: Remove the actuator rod assembly as required.



55-655-00-02



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
55-649-00-01	AWL	55-05-02-210-801	1.1	56000 FC	18000 FC	ALL	ALL			
		neral Visual) the elevator 26A001 - DTR, DTR che		0		STA 260.				
	ACCESS NOTE: Removal of the elevator horn balance weight fairing (183A7400) is required. Outboard rib is inspected through hole in inboard rib.									
55-649-00-02	AWL	55-05-02-210-801	1.1	56000 FC	18000 FC	ALL	ALL			
		neral Visual) the elevator 26A001 - DTR, DTR che		0		STA 260.				
	ACCESS NO	OTE: Removal of the ele Outboard rib is insp		0 0	,	quired.				
55-652-00-01	AWL	55-05-02-250-838	1.1	56000 FC	18000 FC	ALL	ALL			
	upper flange See Doc. D6	n Frequency Eddy Currer :26A001 - DTR, DTR che :hod(s) necessary to acco	eck form 55-20	-08-2, for alternativ	e inspection.					
	upper flange See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che	eck form 55-20- omplish the int es are containe I 596 and ALL	-08-2, for alternativent of this inspectioned in Part 6, Subjection	e inspection. on is contained in ct 55-10-15.	the 737 Nondestru	ctive Test Mar			
55-652-00-02	upper flange See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che thod(s) necessary to acco The inspection procedure NOTE: Applicable to L/N	eck form 55-20- omplish the int es are containe I 596 and ALL	-08-2, for alternativent of this inspectioned in Part 6, Subjection	e inspection. on is contained in ct 55-10-15.	the 737 Nondestru	ctive Test Mar			
55-652-00-02	upper flange See Doc. D6 The NDI met (D6-37239). AIRPLANE I AWL Inspect (High upper flange See Doc. D6 The NDI met (D6-37239).	226A001 - DTR, DTR che chod(s) necessary to according The inspection procedure NOTE: Applicable to L/N 55A1080 or SB 7	eck form 55-20- complish the intes are contained. I 596 and ALL. 737-55-1081. 1.1 nt) around the eck form 55-20- complish the intes are contained. I 596 and ALL.	-08-2, for alternativent of this inspecticed in Part 6, Subject 1175 and on and L 56000 FC circumference of the 108-2, for alternativent of this inspecticed in Part 6, Subjected 15 to	e inspection. on is contained in ot 55-10-15. /N 1 to 1174 that 18000 FC ne washer at bolt e inspection. on is contained in ot 55-10-15.	the 737 Nondestructure thave incorporated \$\frac{ALL}{hole 2 and 3 on the the 737 Nondestructure \$\frac{A}{A} \text{The substitute of the 737 Nondestructure}	ctive Test Mar SB 737- ALL mast arm fitti ctive Test Mar			
55-652-00-02 55-655-00-01	upper flange See Doc. D6 The NDI met (D6-37239). AIRPLANE I AWL Inspect (High upper flange See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che chod(s) necessary to according to the inspection procedure. NOTE: Applicable to L/N 55A1080 or SB 7 55-05-02-250-838 The Frequency Eddy Current. 26A001 - DTR, DTR checkhod(s) necessary to according the inspection procedure. NOTE: Applicable to L/N	eck form 55-20- complish the intes are contained. I 596 and ALL. 737-55-1081. 1.1 nt) around the eck form 55-20- complish the intes are contained. I 596 and ALL.	-08-2, for alternativent of this inspecticed in Part 6, Subject 1175 and on and L 56000 FC circumference of the 108-2, for alternativent of this inspecticed in Part 6, Subjected 15 to	e inspection. on is contained in ot 55-10-15. /N 1 to 1174 that 18000 FC ne washer at bolt e inspection. on is contained in ot 55-10-15.	the 737 Nondestructure thave incorporated \$\frac{ALL}{hole 2 and 3 on the the 737 Nondestructure \$\frac{A}{A} \text{The substitute of the 737 Nondestructure}	SB 737- ALL mast arm fitti			

Inspect (X-ray) the upper and lower flanges of hinges 3, 4, 5, and 6.

55-05-02-260-802

See Doc. D626A001 - DTR, DTR check form 55-20-09-2, for alternative inspection.

1.1

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 2, Subject 55-10-01.

56000 FC

36000 FC

ALL

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and On, and L/N 1 - 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.



ALL





A CIZ C A DD NO				INTERVAL		APPLICA	
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-657-00-01	AWL	55-05-02-250-841	1.1	56000 FC	36000 FC	ALL	ALL
	Inspect (High bore.	n Frequency Eddy Curre	nt) the surface	of the fitting and do	oublers of each b	racket assembly aro	und the lug
		26A001 - DTR, DTR che	eck form 55-20	-12-1, for alternativ	e inspection.		
	AIRPLANE I	NOTE: Applicable to L/N 55A1080 or SB 7		1175 and on, and L	_/N 1 - 1174 that I	have incorporated S	B 737-
	ACCESS NO	DTE: Remove upper or I panel. Bushing ren		-	edge seal and ele	evator inboard hinge	cover
55-657-00-02	AWL	55-05-02-250-841	1.1	56000 FC	36000 FC	ALL	ALL
	bore.	n Frequency Eddy Curre				racket assembly aro	und the lug
		NOTE: Applicable to L/N 55A1080 or SB 7	N 596 and ALL		•	have incorporated S	В 737-
	ACCESS NO	DTE: Remove upper or I panel. Bushing ren		•	edge seal and ele	evator inboard hinge	cover
55-658-00-01	AWL	55-05-02-250-842	1.1	56000 FC	24000 FC	600 700 800 900 900ER	ALL
		26A001 - DTR, DTR che		,			
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	tive Test Mar
55-658-00-02		` '				600 700 800 900 900ER	tive Test Man
55-658-00-02	AWL Inspect (High 73.4 to vertic See Doc. D6 The NDI met	The inspection procedur 55-05-02-250-842 The Frequency Eddy Curre	1.1 nt) the expose eck form 55-30 omplish the interest of the second of	56000 FC d chord between th -03-2, for alternativent of this inspection	24000 FC e inspar and trailing inspection. on is contained in	600 700 800 900 900ER ing edge skin from v	ALL ertical fin STA
55-658-00-02 55-660-00-01	AWL Inspect (High 73.4 to vertic See Doc. D6 The NDI met	The inspection procedur 55-05-02-250-842 The Frequency Eddy Curre cal fin tip. 26A001 - DTR, DTR checked by the control of	1.1 nt) the expose eck form 55-30 omplish the interest of the second of	56000 FC d chord between th -03-2, for alternativent of this inspection	24000 FC e inspar and trailing inspection. on is contained in	600 700 800 900 900ER ing edge skin from v	ALL ertical fin STA
	AWL Inspect (High 73.4 to vertice See Doc. D6 The NDI met (D6-37239). AWL Inspect (Deta fitting.	The inspection procedur 55-05-02-250-842 Frequency Eddy Curre cal fin tip. 26A001 - DTR, DTR che chod(s) necessary to acc The inspection procedur	1.1 nt) the expose eck form 55-30 omplish the integrate contain 1.1 ange and land	56000 FC d chord between th -03-2, for alternative ent of this inspection of the company of the	24000 FC 24000 FC The inspar and trailing in the inspection. The inspection is contained in the contained	600 700 800 900 900ER ing edge skin from v the 737 Nondestruc	ALL ertical fin STA tive Test Mar
	AWL Inspect (High 73.4 to vertice See Doc. D6 The NDI met (D6-37239). AWL Inspect (Deta fitting.	The inspection procedur 55-05-02-250-842 The Frequency Eddy Curre cal fin tip. 26A001 - DTR, DTR che chod(s) necessary to accurate inspection procedur 55-05-02-211-808 ailed) the chords lower fine	1.1 nt) the expose eck form 55-30 omplish the integrate contain 1.1 ange and land	56000 FC d chord between th -03-2, for alternative ent of this inspection of the company of the	24000 FC 24000 FC The inspar and trailing in the inspection. The inspection is contained in the contained	600 700 800 900 900ER ing edge skin from v the 737 Nondestruc	ALL ertical fin STA tive Test Mar
55-660-00-01	AWL Inspect (High 73.4 to vertice See Doc. D6 The NDI met (D6-37239). AWL Inspect (Deta fitting. See Doc. D6 AWL Inspect (Deta fitting. See Doc. D6	The inspection procedur 55-05-02-250-842 The Frequency Eddy Curre and fin tip. 26A001 - DTR, DTR check thod(s) necessary to accomplete inspection procedur 55-05-02-211-808 ailed) the chords lower file 26A001 - DTR, DTR checked	1.1 nt) the expose eck form 55-30 omplish the interest are contain 1.1 ange and land eck form 55-30 1.1 ange and land	56000 FC d chord between th -03-2, for alternativ ent of this inspectic ed in Part 6, Subject 56000 FC -up at the forward e -05-1, for alternativ 56000 FC -up at the forward e	24000 FC 24000 FC The inspar and trailing in the inspection. The inspection is contained in the contained	600 700 800 900 900ER ing edge skin from v the 737 Nondestruc ALL ion common to rear	ALL ertical fin STA tive Test Mar ALL spar terminal
55-660-00-01	AWL Inspect (High 73.4 to vertice See Doc. D6 The NDI met (D6-37239). AWL Inspect (Deta fitting. See Doc. D6 AWL Inspect (Deta fitting. See Doc. D6	The inspection procedur 55-05-02-250-842 The Frequency Eddy Curre and fin tip. 26A001 - DTR, DTR check thod(s) necessary to accomplete the inspection procedur 55-05-02-211-808 ailed) the chords lower flated the chords	1.1 nt) the expose eck form 55-30 omplish the interest are contain 1.1 ange and land eck form 55-30 1.1 ange and land	56000 FC d chord between th -03-2, for alternativ ent of this inspectic ed in Part 6, Subject 56000 FC -up at the forward e -05-1, for alternativ 56000 FC -up at the forward e	24000 FC 24000 FC The inspar and trailing in the inspection. The inspection is contained in the contained	600 700 800 900 900ER ing edge skin from v the 737 Nondestruc ALL ion common to rear	ALL ertical fin STA tive Test Mar ALL spar terminal

Inspect (Detailed) the hinge fitting lugs at the rib attachment for hinges #1, #2, #3, #4, #5, #6, #7, #7A and #8. See Doc. D626A001 - DTR, DTR check form 55-30-06-1, for alternative inspection.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-663-00-01	AWL	55-05-02-211-810	1.1	56000 FC	36000 FC	ALL	ALL
		ailed) rudder hinge ribs 1	-				
		326A001 - DTR, DTR che		•	•		
	ACCESS NO	OTE: Removal of the ski	in panels aft of	the fin spar is requ	iired.		
55-664-00-01	AWL	55-05-02-211-811	1.1	56000 FC	9000 FC	ALL	ALL
		ailed) all the rudder hing 626A001 - DTR, DTR che	•	0	•	s to the skin and the	spar.
55-665-00-01	AWL	55-05-02-250-844	1.1	56000 FC	18000 FC	ALL	ALL
		h Frequency Eddy Curre				STA 60.85 and rudde	er STA 70.65, a
		ally inspect the hinge fitting	-				
		326A001 - DTR, DTR che thod(s) necessary to acc		•	•	the 737 Nondestruc	ctive Test Manu
		The inspection procedur	•				
	ACCESS NO	OTE: Removal of the ac	tuator rod for a	ccess to the inner I	ug faces is requir	ed.	
55-666-00-01	AWL	55-05-02-211-812	1.1	56000 FC	18000 FC	ALL	ALL
	front spar.	ailed) the casting in the a				assembly attachme	ent to the ruda
55-800-00-01	MRB	05-41-03-210-806	1.1	120 DY	120 DY	ALL	ALL
		external zonal inspection out the use of stands or				pection is accompli	
55-802-00-01	MRB						shed from the
		05-41-03-210-807	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	shed from the
	Perform an e	05-41-03-210-807	1.2	36 MO	36 MO	ALL	
			1.2 (GV) of the ve	36 MO	36 MO	ALL	
55-804-00-01		external zonal inspection	1.2 (GV) of the ve	36 MO	36 MO	ALL	
55-804-00-01	INTERVAL I	external zonal inspection	1.2 (GV) of the ve s first.	36 MO rtical fin - dorsal fin	36 MO i.		ALL
55-804-00-01	MRB	external zonal inspection	1.2 (GV) of the ve s first.	36 MO rtical fin - dorsal fin 6600 FC 36 MO	36 MO	ALL	ALL
55-804-00-01	MRB Perform an i	external zonal inspection NOTE: Whichever come 05-41-03-210-808	1.2 (GV) of the verse first. 1.1 1.2 (GV) of the verse first.	36 MO rtical fin - dorsal fin 6600 FC 36 MO	36 MO	ALL	ALL
55-804-00-01	MRB Perform an i	external zonal inspection NOTE: Whichever come 05-41-03-210-808 internal zonal inspection	1.2 (GV) of the verse first. 1.1 1.2 (GV) of the verse minstalled.	36 MO rtical fin - dorsal fin 6600 FC 36 MO	36 MO	ALL	ALL
55-804-00-01	MRB Perform an i	external zonal inspection NOTE: Whichever come 05-41-03-210-808 internal zonal inspection NOTE: Only if HF system	1.2 (GV) of the verse first. 1.1 1.2 (GV) of the verse minstalled.	36 MO rtical fin - dorsal fin 6600 FC 36 MO	36 MO	ALL	shed

Perform an external zonal inspection (GV) of the vertical fin - leading edge.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-808-00-01	MRB	05-41-03-210-810	1.1 1.2	36000 FC 12 YR	36000 FC 12 YR	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the ver	tical fin - leading ed	lge.		
	INTERVAL N	NOTE: Whichever come	s first.				
55-810-00-01	MRB	05-41-03-210-811	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the ver	tical fin - front spar	to rear spar. (EZ	AP)	
	INTERVAL N	NOTE: Whichever come satisfied by this z			rement with inter	val 36000 FC/12 YR	is
55-812-00-01	MRB	05-41-03-210-812	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the ve	rtical fin - front spa	to rear spar.		
	INTERVAL N	NOTE: Whichever come	s first.				
55-814-00-01	MRB	05-41-03-210-813	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
55-814-00-01		05-41-03-210-813	1.2	24 MO	24 MO		ALL
55-814-00-01	Perform an i		1.2 (GV) of the ver s first. The EZ/	24 MO tical fin - rear spar AP inspection requi	24 MO to trailing edge. (I	EZAP)	
55-814-00-01 55-816-00-01	Perform an i	nternal zonal inspection	1.2 (GV) of the ver s first. The EZ/	24 MO tical fin - rear spar AP inspection requi	24 MO to trailing edge. (I	EZAP)	
	Perform an in	nternal zonal inspection NOTE: Whichever come satisfied by this z	1.2 (GV) of the verse first. The EZA conal inspection 1.1 1.2	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO	24 MO to trailing edge. (I rement with inten	EZAP) val 36000 FC/12 YR	is
	Perform an in INTERVAL N MRB Perform an e	nternal zonal inspection (NOTE: Whichever come satisfied by this z	1.2 (GV) of the verse first. The EZ/conal inspection 1.1 1.2 (GV) of the verse first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO	24 MO to trailing edge. (I rement with inten	EZAP) val 36000 FC/12 YR	is
	Perform an in INTERVAL N MRB Perform an e	NOTE: Whichever come satisfied by this z	1.2 (GV) of the verse first. The EZ/conal inspection 1.1 1.2 (GV) of the verse first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO	24 MO to trailing edge. (I rement with inten	EZAP) val 36000 FC/12 YR	is
55-816-00-01	Perform an in INTERVAL N MRB Perform an e INTERVAL N	nternal zonal inspection (NOTE: Whichever come satisfied by this zonal-210-814) external zonal inspection (NOTE: Whichever come	1.2 (GV) of the verse first. The EZ/conal inspection 1.1 1.2 (GV) of the verse first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO rtical fin - rear spar	24 MO to trailing edge. (I rement with intended of the control of	EZAP) val 36000 FC/12 YR ALL	ALL
55-816-00-01	Perform an in INTERVAL MARB Perform an el INTERVAL MARB Perform an el Perform an el INTERVAL MARB	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal-10-41-03-210-814 external zonal inspection NOTE: Whichever come 05-41-03-210-815	1.2 (GV) of the verse first. The EZA conal inspection 1.1 1.2 (GV) of the verse first. 1.1 1.2 (GV) of the verse first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO rtical fin - rear spar	24 MO to trailing edge. (I rement with intended of the control of	EZAP) val 36000 FC/12 YR ALL	ALL
55-816-00-01	Perform an in INTERVAL MARB Perform an el INTERVAL MARB Perform an el Perform an el INTERVAL MARB	nternal zonal inspection NOTE: Whichever come satisfied by this z 05-41-03-210-814 external zonal inspection NOTE: Whichever come of the satisfied by this z	1.2 (GV) of the verse first. The EZA conal inspection 1.1 1.2 (GV) of the verse first. 1.1 1.2 (GV) of the verse first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO rtical fin - rear spar	24 MO to trailing edge. (I rement with intended of the control of	EZAP) val 36000 FC/12 YR ALL	ALL
55-816-00-01 55-818-00-01	Perform an in INTERVAL MARB Perform an element of INTERVAL MARB Perform an element of INTERVAL MARB	nternal zonal inspection of NOTE: Whichever come satisfied by this z 05-41-03-210-814 external zonal inspection NOTE: Whichever come external zonal inspection systemal zonal inspection external zonal inspection NOTE: Whichever come	1.2 (GV) of the verse first. The EZA conal inspection 1.1 1.2 (GV) of the verse first. 1.1 1.2 (GV) of the runs first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO rtical fin - rear spar 5500 FC 30 MO dder. 6600 FC 36 MO	24 MO to trailing edge. (I rement with interv 6600 FC 36 MO to trailing edge. 5500 FC 30 MO	EZAP) val 36000 FC/12 YR ALL	ALL ALL
55-816-00-01 55-818-00-01	Perform an in INTERVAL MARB Perform an existence in INTERVAL MARB Perform an existence in INTERVAL MARB Perform an existence in INTERVAL MARB	nternal zonal inspection of NOTE: Whichever come satisfied by this z 05-41-03-210-814 external zonal inspection NOTE: Whichever come o5-41-03-210-815 external zonal inspection NOTE: Whichever come o5-41-03-210-816	1.2 (GV) of the verse first. The EZ/conal inspection 1.1 1.2 (GV) of the verse first. 1.1 1.2 (GV) of the runs first. 1.1 1.2 (GV) of the verse first.	24 MO rtical fin - rear spar AP inspection requin. 6600 FC 36 MO rtical fin - rear spar 5500 FC 30 MO dder. 6600 FC 36 MO	24 MO to trailing edge. (I rement with interv 6600 FC 36 MO to trailing edge. 5500 FC 30 MO	EZAP) val 36000 FC/12 YR ALL	ALL ALL

 $Perform\ an\ external\ zonal\ inspection\ (GV)\ of\ the\ horizontal\ stabilizer\ -\ leading\ edge\ -\ left.$







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-824-01-01	MRB	05-41-03-210-818	1.1 1.2	4000 FC 18 MO	4000 FC 18 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the ho	rizontal stabilizer -	ront spar to rear	spar - left.	
	INTERVAL N	NOTE: Whichever come	s first.				
55-826-01-01	MRB	05-41-03-210-819	1.1 1.2	4000 FC 18 MO	4000 FC 18 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the inb	oard horizontal stat	oilizer - front spar	to rear spar - left.	
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: As an alternate to lower side of stabil		vided, access pane	l 332AB can be r	removed for inspecti	on from
55-828-01-01	MRB	05-41-03-210-820	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the hor	rizontal stabilizer - r	ear spar to trailin	g edge - left. (EZAP	')
	INTERVAL	NOTE: Whichever come			ement with inter-	vai 30000 FC/12 TN	. 15
55-830-01-01	MRB	satisfied by this z	1.1	4800 FC	4800 FC	ALL	ALL
55-830-01-01		05-41-03-210-821	1.1 1.2	4800 FC 24 MO	24 MO		ALL
55-830-01-01	Perform an e		1.1 1.2 (GV) of the ho	4800 FC 24 MO	24 MO		ALL
55-830-01-01 55-832-01-01	Perform an e	05-41-03-210-821	1.1 1.2 (GV) of the ho	4800 FC 24 MO	24 MO		ALL
	Perform an e	05-41-03-210-821 external zonal inspection NOTE: Whichever come	1.1 1.2 (GV) of the ho s first.	4800 FC 24 MO rizontal stabilizer - 1 6600 FC 36 MO	24 MO rear spar to traillin 6600 FC 36 MO	ng edge - left.	
	Perform an e	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822	1.1 1.2 (GV) of the hose s first.	4800 FC 24 MO rizontal stabilizer - 1 6600 FC 36 MO	24 MO rear spar to traillin 6600 FC 36 MO	ng edge - left.	
	Perform an e	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822 external zonal inspection	1.1 1.2 (GV) of the hose s first.	4800 FC 24 MO rizontal stabilizer - 1 6600 FC 36 MO	24 MO rear spar to traillin 6600 FC 36 MO	ng edge - left.	
55-832-01-01	Perform an e INTERVAL N MRB Perform an e INTERVAL N	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822 external zonal inspection NOTE: Whichever come	1.1 1.2 (GV) of the hors first. 1.1 1.2 (GV) of the hors first.	4800 FC 24 MO rizontal stabilizer - 1 6600 FC 36 MO rizontal stabilizer - 1 6600 FC 36 MO	24 MO rear spar to trailin 6600 FC 36 MO elevator - left. 6600 FC 36 MO	ng edge - left. ALL ALL	ALL
55-832-01-01	Perform an e INTERVAL N MRB Perform an e INTERVAL N MRB	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822 external zonal inspection NOTE: Whichever come 05-41-03-210-823	1.1 1.2 (GV) of the horse first. 1.1 1.2 (GV) of the horse first. 1.1 1.2 (GV) of the horse first.	4800 FC 24 MO rizontal stabilizer - 1 6600 FC 36 MO rizontal stabilizer - 1 6600 FC 36 MO	24 MO rear spar to trailin 6600 FC 36 MO elevator - left. 6600 FC 36 MO	ng edge - left. ALL ALL	ALL
55-832-01-01	Perform an e INTERVAL N MRB Perform an e INTERVAL N MRB	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822 external zonal inspection NOTE: Whichever come 05-41-03-210-823 external zonal inspection	1.1 1.2 (GV) of the horse first. 1.1 1.2 (GV) of the horse first. 1.1 1.2 (GV) of the horse first.	4800 FC 24 MO rizontal stabilizer - 1 6600 FC 36 MO rizontal stabilizer - 1 6600 FC 36 MO	24 MO rear spar to trailin 6600 FC 36 MO elevator - left. 6600 FC 36 MO	ng edge - left. ALL ALL	ALL
55-832-01-01 55-834-01-01	Perform an e INTERVAL N MRB Perform an e INTERVAL N MRB Perform an e INTERVAL N	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822 external zonal inspection NOTE: Whichever come 05-41-03-210-823 external zonal inspection NOTE: Whichever come	1.1 1.2 (GV) of the horse first. 1.1 1.2 (GV) of the horse first. 1.1 1.2 (GV) of the horse first. 1.1 1.2 1.1 1.2 1.1 1.2 1.1 1.2 1.1 1.2 1.1 1.2 1.1 1.2 1.1 1.2	4800 FC 24 MO rizontal stabilizer - 6 6600 FC 36 MO rizontal stabilizer - 6 6600 FC 36 MO rizontal stabilizer - 6 6600 FC 36 MO	24 MO rear spar to trailin 6600 FC 36 MO elevator - left. 6600 FC 36 MO stabilizer tip -left.	ALL ALL	ALL
55-832-01-01 55-834-01-01	Perform an e INTERVAL N MRB Perform an e INTERVAL N MRB Perform an e INTERVAL N	05-41-03-210-821 external zonal inspection NOTE: Whichever come 05-41-03-210-822 external zonal inspection NOTE: Whichever come 05-41-03-210-823 external zonal inspection NOTE: Whichever come	1.1 1.2 (GV) of the horse first.	4800 FC 24 MO rizontal stabilizer - 6 6600 FC 36 MO rizontal stabilizer - 6 6600 FC 36 MO rizontal stabilizer - 6 6600 FC 36 MO	24 MO rear spar to trailin 6600 FC 36 MO elevator - left. 6600 FC 36 MO stabilizer tip -left.	ALL ALL	ALL

Perform an external zonal inspection (GV) of the horizontal stabilizer - front spar to rear spar - right.







				INTERVAL		APPLICA	BILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
55-840-02-01	MRB	05-41-03-210-826	1.1 1.2	4000 FC 18 MO	4000 FC 18 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the inb	oard horizontal sta	bilizer - front spaı	to rear spar - right.	
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: With access provide side of the stabilize		nel 342AB can be	used as an altern	ate to inspect from t	he lower
55-842-02-01	MRB	05-41-03-210-827	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the hor	rizontal stabilizer - ı	rear spar to trailin	g edge - right. (EZA	P)
	INTERVAL N	NOTE: Whichever come satisfied by this z			rement with inter	val 36000 FC/12 YR	is
55-844-02-01	MRB	05-41-03-210-828	1.1 1.2	4800 FC 24 MO	4800 FC 24 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the ho	rizontal stabilizer -	rear spar to trailing	ng edge - right.	
	INTERVAL N	NOTE: Whichever come	s first.				
55-846-02-01	MRB	05-41-03-210-829	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
		external zonal inspection NOTE: Whichever come	` '	rizontal stabilizer -	elevator - right.		
55-848-02-01	MRB	05-41-03-210-830	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the ho	rizontal stabilizer -	stabilizer tip - rig	ht.	
	INTERVAL N	NOTE: Whichever come	s first.				
56-010-00-01	MRB	12-25-81-600-801	1.1	2 YR	2 YR	600 700 800 900	ALL
		inside release mechani for the co-pilots #2 sliding		ots and co-pilots #2	sliding windows	. Lubricate the outsion	le release
56-030-00-01	MRB	56-12-11-710-803	1.1	4 YR	4 YR	600 700 800 900	ALL
		y check the inside releas sliding window from the o		for the pilots and o	co-pilots #2 slidin	g windows. Operatio	nally check
56-050-00-01	MRB	56-12-11-200-802	1.1	4 YR	4 YR	600 700 800 900	ALL

Perform a general visual inspection of the pilots and co-pilots #2 sliding window sill drain for obvious damage, clogging, condition, and security.

AIRPLANE NOTE: Applicable to airplanes line number 145 and on.

Applicable to airplanes line number 1 to 144 incorporating SB 737-56-1011.







ASK CARD NO.			INTERVAL		APPLICA	ABILITY	
AON OAND NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-010-00-01	MRB	51-05-01-210-804 57-05-03-210-801	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	Inspect forwa	ard side of wing center se	ection front spa	ar, including the sid	le of body/termina	l fitting.	
	INTERVAL N	IOTE: Whichever comes	s first.	-		-	
	ACCESS NO	OTE: Remove aft panels	in forward car	go compartment.			
57-020-00-01	MRB	51-05-01-210-809 57-05-03-210-802	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	Inspect insid	e wing center section:					
	(including sk attachment to stiffeners, up chord, lower	face (including skins, typ ins, typical stringers, spli o fuselage drag angle, at oper and lower spar chord tee chord, front and rear IOTE: Whichever comes	ice stringers, a t attachment to ds, attachment r spar terminal	t attachment to kee lower beam at BL s to skin); 4. Side o	el beam, at drain in 41); 3. Front and of body rib (includ	nstallation, at acces rear spars (includin ng webs and stiffen	s hole, at g webs and
57-030-00-01	MRB	51-05-01-210-806 57-05-03-210-803	1.1	9 YR	3 YR	ALL	ALL
	Inspect aft si	de of rear spar, including	keel beam sti	ffeners at BL 6.2, a	and side of body re	ear spar terminal fitt	ing.
57-040-00-01	MRB	51-05-01-210-804	1.1	6 YR	6 YR	ALL	ALL
		EZ OE OO O4O OO4	4 0				
		57-05-03-210-804	1.2	18000 FC	18000 FC		
	1. Skins, typi plates; 3. Lo	r side of lower surface of cal stringers, splice strinwer beams at BL 41; 4. And access holes.	f wing center so	ection: rear spar lower ch	nords; 2. Side of b		
	1. Skins, typi plates; 3. Lov installation a	r side of lower surface of cal stringers, splice strin wer beams at BL 41; 4. A	f wing center s gers, front and At attachments	ection: rear spar lower ch	nords; 2. Side of b		
	1. Skins, typi plates; 3. Lov installation a	r side of lower surface of cal stringers, splice strinwer beams at BL 41; 4. And access holes.	f wing center s gers, front and At attachments s first.	ection: rear spar lower ch to keel beam, to lo	nords; 2. Side of b		
57-050-00-01	1. Skins, typi plates; 3. Lov installation a	r side of lower surface of cal stringers, splice strin wer beams at BL 41; 4. And access holes.	f wing center s gers, front and At attachments s first.	ection: rear spar lower ch to keel beam, to lo	nords; 2. Side of b		
57-050-00-01	1. Skins, typi plates; 3. Lov installation a INTERVAL N ACCESS NO	r side of lower surface of cal stringers, splice strin wer beams at BL 41; 4. And access holes. HOTE: Whichever come: DTE: Remove ECS heat	f wing center s gers, front and at attachments s first. exchanger ac	ection: rear spar lower ch to keel beam, to lo cess panel. 6 YR 18000 FC	ords; 2. Side of b wer beams at BL 6 YR 18000 FC	41, to fuselage draφ	g angles; at d
57-050-00-01	1. Skins, typi plates; 3. Lovinstallation a INTERVAL N ACCESS NC	r side of lower surface of cal stringers, splice strinwer beams at BL 41; 4. And access holes. NOTE: Whichever come: 51-05-01-210-806 57-05-03-210-805	f wing center s gers, front and at attachments s first. exchanger ac 1.1 1.2	ection: rear spar lower ch to keel beam, to lo cess panel. 6 YR 18000 FC	ords; 2. Side of b wer beams at BL 6 YR 18000 FC	41, to fuselage draφ	g angles; at d
57-050-00-01 57-060-01-01	1. Skins, typi plates; 3. Lovinstallation a INTERVAL N ACCESS NC	r side of lower surface of cal stringers, splice strin wer beams at BL 41; 4. And access holes. NOTE: Whichever comes TE: Remove ECS heat 51-05-01-210-806 57-05-03-210-805 ard side of front spar, inc	f wing center s gers, front and at attachments s first. exchanger ac 1.1 1.2	ection: rear spar lower ch to keel beam, to lo cess panel. 6 YR 18000 FC	ords; 2. Side of b wer beams at BL 6 YR 18000 FC	41, to fuselage draφ	g angles; at c
	1. Skins, typi plates; 3. Lovinstallation a INTERVAL MACCESS NO INTERVAL MINTERVAL MIN	r side of lower surface of cal stringers, splice strin wer beams at BL 41; 4. And access holes. NOTE: Whichever come: 51-05-01-210-806 57-05-03-210-805 and side of front spar, inc. NOTE: Whichever come: 51-05-01-210-806	f wing center s gers, front and at attachments s first. exchanger ac 1.1 1.2 luding side of I s first.	ection: rear spar lower ch to keel beam, to lo cess panel. 6 YR 18000 FC cody/terminal fitting	6 YR 18000 FC 6 YR 18000 FC 18000 FC	ALL ALL	g angles; at d
	1. Skins, typi plates; 3. Lovinstallation a INTERVAL MACCESS NO INTERVAL MINTERVAL MIN	r side of lower surface of cal stringers, splice strin wer beams at BL 41; 4. And access holes. NOTE: Whichever comes 51-05-01-210-806 57-05-03-210-805 ard side of front spar, inc. NOTE: Whichever comes 51-05-01-210-806 57-05-03-210-806	f wing center s gers, front and at attachments s first. exchanger ac 1.1 1.2 luding side of I s first. 1.1 1.2 ace (under lowe	ection: rear spar lower ch to keel beam, to lo cess panel. 6 YR 18000 FC cody/terminal fitting	6 YR 18000 FC 6 YR 18000 FC 18000 FC	ALL ALL	g angles; at d

Inspect right outboard wing lower surface (under lower side of body fairing), including attachment locations.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-070-01-01	MRB	51-05-01-210-806 57-05-03-210-808	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL
	Inspect uppe	er side of left outboard wi	ing upper surfa	ice (under side-of-b	ody fairing), inclu	iding:	
	1. Wing upp	er surface at side-of-bod	y splice, includ	ing upper rib chord	; 2. Wing upper s	urface at attachmer	nt locations.
	INTERVAL I	NOTE: Whichever come	s first.				
57-070-02-01	MRB	51-05-01-210-806 57-05-03-210-809	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL
	Inspect uppe	er side of right outboard v	wing upper sur	face (under side-of-	-body fairing), inc	luding:	
	1. Wing upp	er surface at side-of-bod	y splice, includ	ing upper rib chord	; 2. Wing upper s	urface at attachmer	nt locations.
	INTERVAL I	NOTE: Whichever come	s first.				
57-090-00-01	MRB	51-05-01-210-808 57-05-03-210-810	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	side of body	er side of upper wing sur upper rib chord; 2. Floor secondary vapor barrier.	beams from S				
	INTERVAL I	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove floor pane	els and insulati	on as required in pa	assenger compar	tment as for access	i.
57-100-01-01	MRB	51-05-01-210-801 57-05-03-210-811	1.1 1.2	48 MO 9000 FC	48 MO 9000 FC	ALL	ALL
	(R3); 4. Inbo	following fittings: 1. Front pard side load fitting (R4) attachment to nacelle fitt	; 5. Side brace	• , ,	0 0 1	, .	•
	INTERVAL I	NOTE: Whichever come	s first.				
57-100-02-01	MRB	51-05-01-210-801 57-05-03-210-812	1.1 1.2	48 MO 9000 FC	48 MO 9000 FC	ALL	ALL
	(R3); 4. Inbo	following fittings: 1. Front pard side load fitting (R4) attachment to nacelle fitt	; 5. Side brace				
	INTERVAL I	NOTE: Whichever come	s first.				
57-110-01-01	MRB	51-05-01-210-809	1.1	48 MO	48 MO	ALL	ALL
		57-05-03-210-813	1.2	9000 FC	9000 FC		
	Inspect left v	ving lower surface under	strut fairing, in	cluding all attachm	ent locations.		
	INTERVAL I	NOTE: Whichever come	s first.				
57-110-02-01	MRB	51-05-01-210-809	1.1	48 MO	48 MO	ALL	ALL
		57-05-03-210-814	1.2	9000 FC	9000 FC		

Inspect right wing lower surface under strut fairing, including all attachment locations.







TASK CARD NO				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-120-01-01	MRB	51-05-01-210-806 57-05-03-210-815	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	Inspect left f	ront spar chords, webs a	nd stiffeners, in	ncluding at side of b	oody joint and at i	nacelle fitting attach	ment.
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Deploy Krueger Fla	aps.				
57-120-02-01	MRB	51-05-01-210-806 57-05-03-210-816	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	Inspect right	front spar chords, webs	and stiffeners,	including at side of	body joint and at	nacelle fitting attac	hment.
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Deploy Krueger Fla	aps.				
57-130-01-01	MRB	51-05-01-210-806	1.1	10 YR	10 YR	ALL	ALL
37-130-01-01	IVIND	57-05-03-210-817	1.1	10 11	10 110	ALL	ALL
	Inspect left v	ving leading edge cavity,	including flaps	and slats.			
	ACCESS NO	OTE: Extend Krueger fla	ps and slats.				
57-130-02-01	MRB	51-05-01-210-806 57-05-03-210-818	1.1	10 YR	10 YR	ALL	ALL
	Inspect right	wing leading edge cavity	y, including flap	s and slats.			
	ACCESS NO	OTE: Extend Krueger fla	ps and slats.				
	ACCESS NO	OTE: Extend Krueger fla	ps and slats.				
57-140-01-01	MRB	51-05-01-210-809 57-05-03-211-801	ps and slats.	6 YR	6 YR	ALL	ALL
57-140-01-01	MRB	51-05-01-210-809		6 YR	6 YR	ALL	ALL
57-140-01-01	MRB Inspect left v	51-05-01-210-809 57-05-03-211-801		6 YR	6 YR	ALL	ALL
57-140-01-01	MRB Inspect left v	51-05-01-210-809 57-05-03-211-801 ving slat tracks.		6 YR	6 YR	ALL	ALL
57-140-01-01 57-140-02-01	MRB Inspect left v	51-05-01-210-809 57-05-03-211-801 ving slat tracks.		6 YR 6 YR	6 YR	ALL	ALL
	MRB Inspect left v	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats.	1.1				
	MRB Inspect left v ACCESS NO MRB Inspect right	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats. 51-05-01-210-809 57-05-03-211-802	1.1				
	MRB Inspect left v ACCESS NO MRB Inspect right	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats. 51-05-01-210-809 57-05-03-211-802 wing slat tracks.	1.1	6 YR	6 YR		
57-140-02-01	MRB Inspect left v ACCESS NO MRB Inspect right ACCESS NO	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats. 51-05-01-210-809 57-05-03-211-802 wing slat tracks. DTE: Extend stats.	1.1	6 YR	6 YR	ALL	ALL
57-140-02-01	MRB Inspect left v ACCESS NO MRB Inspect right ACCESS NO	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats. 51-05-01-210-809 57-05-03-211-802 wing slat tracks. DTE: Extend stats.	1.1	6 YR 6 YR 18000 FC	6 YR 6 YR 18000 FC	ALL	ALL
57-140-02-01	MRB Inspect left v ACCESS NO MRB Inspect right ACCESS NO MRB	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats. 51-05-01-210-809 57-05-03-211-802 wing slat tracks. DTE: Extend stats. 51-05-01-210-809 57-05-03-210-819	1.1 1.1 1.1 1.2 pard wing lowe	6 YR 6 YR 18000 FC	6 YR 6 YR 18000 FC	ALL	ALL
57-140-02-01	MRB Inspect left v ACCESS NO MRB Inspect right ACCESS NO MRB	51-05-01-210-809 57-05-03-211-801 ving slat tracks. DTE: Extend slats. 51-05-01-210-809 57-05-03-211-802 wing slat tracks. DTE: Extend stats. 51-05-01-210-809 57-05-03-210-819 access holes in left outboom	1.1 1.1 1.1 1.2 pard wing lowe	6 YR 6 YR 18000 FC	6 YR 6 YR 18000 FC	ALL	ALL

Inspect fuel access holes in right outboard wing lower surface. (Tank entry is not required.)







			APPLICA				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-170-01-01	MRB	51-05-01-210-809 57-05-03-210-821	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	Inspect inside	e left outboard wing from	side of body I	Rib to Rib 5:			
	and rear spa attachments webs, stiffend	dy rib (including webs an rs); 2. Upper and lower s to front and rear spars; a ers and rib posts; 4. Sheatort: Whichever comes	surfaces (include at attachments ar tied and no	ding skins; typical s to shear tied ribs a	plice, vent and rai	l stringers; at drain	installations;
57-170-02-01	MRB	51-05-01-210-809 57-05-03-210-822	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
		e right outboard wing from			2		
	attachments webs, stiffen	rs); 2. Upper and lower s to front and rear spars; a ers and rib posts; 4. She	at attachments ar tied and nor	to shear tied ribs a			
57-180-01-01	attachments webs, stiffen	to front and rear spars; a ers and rib posts; 4. She IOTE: Whichever comes 51-05-01-210-809	at attachments ar tied and nor s first.	to shear tied ribs an-shear tied ribs.	and support fittings		
57-180-01-01	attachments webs, stiffend INTERVAL N	to front and rear spars; a ers and rib posts; 4. She IOTE: Whichever comes 51-05-01-210-809 57-05-03-210-823	at attachments ar tied and nor s first.	to shear tied ribs an-shear tied ribs. 10 YR 36000 FC	nnd support fittings	s); 3. Front and rear	spar chords,
57-180-01-01	MRB Inspect inside 1. Upper and attachments chords, webs posts, flap tracking line including Rill	to front and rear spars; a ers and rib posts; 4. She IOTE: Whichever comes 51-05-01-210-809	at attachments ar tied and nor s first. 1.1 1.2 1 Rib 5 to Rib 2 g skins; typica pport fittings; a, including at r ding rear spar s 6 and 7 at na	10 YR 36000 FC 22: Il splice, vent and rat attachment to na nacelle fitting attach at major fitting attaches support fitting attaches support fitting	10 YR 36000 FC ail stringers; at atta celle fittings; at dra ments; 3. Rear sp chments; 4. Shear s, Ribs 10 and 14	ALL achments to front allain installations); 2. par chords, webs, sir tied and non-shear	ALL nd rear spars: Front spar tiffeners and r
57-180-01-01	MRB Inspect inside attachments chords, webs posts, flap tracking line luding Rill Nacelle supp	to front and rear spars; as ers and rib posts; 4. She ers and rib posts; 4. She ers and rib posts; 4. She ers and rib posts ack support fittings, include 6 structural doors, Ribs	at attachments ar tied and nor s first. 1.1 1.2 I Rib 5 to Rib 2 g skins; typica pport fittings; a, including at r ding rear spar s 6 and 7 at na nk, R4 back up	10 YR 36000 FC 22: Il splice, vent and rat attachment to na nacelle fitting attach at major fitting attaches support fitting attaches support fitting	10 YR 36000 FC ail stringers; at atta celle fittings; at dra ments; 3. Rear sp chments; 4. Shear s, Ribs 10 and 14	ALL achments to front allain installations); 2. par chords, webs, sir tied and non-shear	ALL nd rear spars Front spar tiffeners and i
57-180-01-01	MRB Inspect inside 1. Upper and attachments chords, webs posts, flap trace (including Ril Nacelle supplinterval N	to front and rear spars; a ers and rib posts; 4. Sheaters and rib posts; 4. Sheaters and rib posts; 4. Sheaters and rib posts are left outboard wing from a lower surfaces (including to shear tied ribs and support fittings, include 6 structural doors, Ribsport fittings (R2 backup limited and ribs and ribs and ribsport fittings).	at attachments ar tied and nor s first. 1.1 1.2 I Rib 5 to Rib 2 g skins; typica pport fittings; a, including at r ding rear spar 6 6 and 7 at nank, R4 back ups first.	to shear tied ribs and shear tied ribs. 10 YR 36000 FC 22: all splice, vent and reat attachment to nath an acelle fitting attact at major fitting attact celle support fitting belink, R7/8 backup	10 YR 36000 FC ail stringers; at atta celle fittings; at dra nments; 3. Rear sp chments; 4. Shea s, Ribs 10 and 14 fitting).	ALL achments to front allain installations); 2. par chords, webs, sir tied and non-shear	ALL nd rear spars Front spar tiffeners and r
57-180-01-01 57-180-02-01	MRB Inspect inside 1. Upper and attachments chords, webs posts, flap trace (including Ril Nacelle supplinterval N	to front and rear spars; a ers and rib posts; 4. She ers and rib posts; 4. She ers and rib posts; 4. She ers and rib posts; 51-05-01-210-809 57-05-03-210-823 ers left outboard wing from a lower surfaces (including to shear tied ribs and support fittings, include 6 structural doors, Ribs ort fittings (R2 backup ling to the support fittings).	at attachments ar tied and nor s first. 1.1 1.2 I Rib 5 to Rib 2 g skins; typica pport fittings; a, including at r ding rear spar 6 6 and 7 at nank, R4 back ups first.	to shear tied ribs and shear tied ribs. 10 YR 36000 FC 22: all splice, vent and reat attachment to nath an acelle fitting attact at major fitting attact celle support fitting belink, R7/8 backup	10 YR 36000 FC ail stringers; at atta celle fittings; at dra nments; 3. Rear sp chments; 4. Shea s, Ribs 10 and 14 fitting).	ALL achments to front allain installations); 2. par chords, webs, sir tied and non-shear	ALL nd rear spars Front spar tiffeners and i

1. Upper and lower surfaces (including skins; typical splice, vent and rail stringers; at attachments to front and rear spars; at attachments to shear tied ribs and support fittings; at attachment to nacelle fittings; at drain installations); 2. Front spar chords, webs, stiffeners and rib posts, including at nacelle fitting attachments; 3. Rear spar chords, webs, stiffeners and rib posts, flap track support fittings, including rear spar at major fitting attachments; 4. Shear tied and non-shear tied ribs (including Rib 6 structural doors, Ribs 6 and 7 at nacelle support fittings, Ribs 10 and 14 at flap track support fittings); 5. Nacelle support fittings (R2 backup link, R4 back up link, R7/8 backup fitting).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Do not remove the 632AZ and 632BZ access doors at the same time.

57-190-01-01	MRB	51-05-01-210-806	1.1	10 YR	10 YR	ALL	ALL
		57-05-03-210-825	1.2	36000 FC	36000 FC		

Inspect inside left outboard wing surge tank and dry bay (from Rib 22 to Rib 27), including upper and lower skins, stringers, front and rear spars, rib posts, WBL 656.17 closure rib (if provisioned for winglets, L/N 778 and on), shear tied and non-shear tied ribs, and access holes.

INTERVAL NOTE: Whichever comes first.





				INTERVAL			ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-190-02-01	MRB	51-05-01-210-806 57-05-03-210-826	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL
	stringers, from	e right outboard wing sunt and rear spars, rib postd ribs, and access holes	sts, WBL 656.1	• • •	, .	0	

INTERVAL NOTE: Whichever comes first.

57-200-01-01 MRB 51-05-01-210-802 1.1 6 YR 6 YR ALL ALL 57-05-03-210-827 1.2 18000 FC 18000 FC

Inspect left inboard flap inboard track assembly, carriage assembly, forward fitting assembly, aft link and aft link pins. Normal overhaul procedures, applied with the flap track assemblies, carriage assemblies and forward fitting assemblies removed, at intervals not exceeding 10 years, are adequate to maintain corrosion at safe levels on these components. Therefore application of the basic tasks and reporting are not required on these components.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove flap drive lube access door and access door from MLG wheel well.

57-200-02-01 MRB 51-05-01-210-802 1.1 6 YR 6 YR ALL ALL 57-05-03-210-828 1.2 18000 FC 18000 FC

Inspect right inboard flap inboard track assembly, carriage assembly, forward fitting assembly, aft link and aft link pins. Normal overhaul procedures, applied with the flap track assemblies, carriage assemblies and forward fitting assemblies removed, at intervals not exceeding 10 years, are adequate to maintain corrosion at safe levels on these components. Therefore application of the basic tasks and reporting are not required on these components.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove flap drive lube access door and access door from MLG wheel well.

57-210-01-01 MRB 51-05-01-210-802 1.1 6 YR 6 YR ALL ALL 57-05-03-210-829 1.2 18000 FC 18000 FC

Inspect left inboard flap outboard track assembly, carriage assembly, forward fitting assembly, and aft attach fitting. Normal overhaul procedures, applied with the flap track assemblies, carriage assemblies and forward fitting assemblies removed, at intervals not exceeding 10 years, are adequate to maintain corrosion at safe levels on these components. Therefore application of the basic tasks and reporting are not required on these components.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove flap support forward fairing and deploy flaps.

57-210-02-01 MRB 51-05-01-210-802 1.1 6 YR 6 YR ALL ALL 57-05-03-210-830 1.2 18000 FC 18000 FC

Inspect right inboard flap outboard track assembly, carriage assembly, forward fitting assembly, and aft attach fitting. Normal overhaul procedures, applied with the flap track assemblies, carriage assemblies and forward fitting assemblies removed, at intervals not exceeding 10 years, are adequate to maintain corrosion at safe levels on these components. Therefore application of the basic tasks and reporting are not required on these components.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove flap support forward fairing and deploy flaps.

57-220-01-01 MRB 51-05-01-210-801 1.1 6 YR 6 YR ALL ALL 57-05-03-210-831 1.2 18000 FC 18000 FC

Inspect lower side of lower surface (under flap support No. 3 fairing), including all attachment locations and access holes.

INTERVAL NOTE: Whichever comes first.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-220-02-01	MRB	51-05-01-210-801 57-05-03-210-832	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	Inspect lowe	r side of lower surface (ι	ınder flap supp	ort No. 6 fairing), ir	ncluding all attach	ment locations and	access hole
	INTERVAL N	IOTE: Whichever come	s first.				
57-230-01-01	MRB	51-05-01-210-803 57-05-03-210-833	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	links. Normal assemblies r	utboard flap inboard and l overhaul procedures, a emoved, at intervals not Therefore application of	pplied with the exceeding 10	flap track assembly years, are adequat	ies, carriage asse e to maintain corr	mblies and forward osion at safe levels	fitting
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Remove flap forwa	ard fairings and	deploy flaps.			
57-230-02-01	MRB	51-05-01-210-803 57-05-03-210-834	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	links. Normal assemblies r	outboard flap inboard ar I overhaul procedures, a emoved, at intervals not . Therefore application o	pplied with the exceeding 10	flap track assembly years, are adequat	ies, carriage asse e to maintain corr	mblies and forward osion at safe levels	fitting
	links. Normal assemblies r components.	l overhaul procedures, a emoved, at intervals not	pplied with the exceeding 10 f the basic task	flap track assembly years, are adequat	ies, carriage asse e to maintain corr	mblies and forward osion at safe levels	fitting
	links. Normal assemblies r components.	l overhaul procedures, a emoved, at intervals not Therefore application of	pplied with the exceeding 10 f the basic task s first.	flap track assembli years, are adequat as and reporting are	ies, carriage asse e to maintain corr	mblies and forward osion at safe levels	fitting
57-240-01-01	links. Normal assemblies r components.	I overhaul procedures, a emoved, at intervals not Therefore application of IOTE: Whichever come	pplied with the exceeding 10 f the basic task s first.	flap track assembli years, are adequat as and reporting are	ies, carriage asse e to maintain corr	mblies and forward osion at safe levels	fitting
57-240-01-01	links. Normal assemblies r components. INTERVAL N ACCESS NO	l overhaul procedures, a emoved, at intervals not Therefore application of NOTE: Whichever come OTE: Remove flap forwards 51-05-01-210-801	pplied with the exceeding 10 f the basic task s first. Indicate the fairings and the fairings are the fairi	flap track assembly years, are adequat as and reporting are deploy flaps. 6 YR 18000 FC	ies, carriage asse te to maintain corr e not required on t 6 YR 18000 FC	emblies and forward cosion at safe levels these components.	fitting on these
57-240-01-01	links. Normal assemblies r components. INTERVAL N ACCESS NO MRB Inspect lower holes.	l overhaul procedures, a emoved, at intervals not. Therefore application of IOTE: Whichever come DTE: Remove flap forwards 51-05-01-210-801 57-05-03-210-835	pplied with the exceeding 10 f the basic task is first. and fairings and 1.1 1.2 inder flap supp	flap track assembly years, are adequat as and reporting are deploy flaps. 6 YR 18000 FC	ies, carriage asse te to maintain corr e not required on t 6 YR 18000 FC	emblies and forward cosion at safe levels these components.	fitting on these
57-240-01-01 57-240-02-01	links. Normal assemblies r components. INTERVAL N ACCESS NO MRB Inspect lower holes.	l overhaul procedures, a emoved, at intervals not. Therefore application of IOTE: Whichever come ITE: Remove flap forwards 151-05-01-210-801 57-05-03-210-835 r side of lower surface (united to the control of the cont	pplied with the exceeding 10 f the basic task is first. and fairings and 1.1 1.2 inder flap supp	flap track assembly years, are adequat as and reporting are deploy flaps. 6 YR 18000 FC	ies, carriage asse te to maintain corr e not required on t 6 YR 18000 FC	emblies and forward cosion at safe levels these components.	fitting on these
	Inks. Normal assemblies recomponents. INTERVAL N ACCESS NO MRB Inspect lower holes. INTERVAL N MRB	l overhaul procedures, a emoved, at intervals not. Therefore application of IOTE: Whichever come DTE: Remove flap forwards 151-05-01-210-801 57-05-03-210-835 r side of lower surface (UIOTE: Whichever come 51-05-01-210-801	pplied with the exceeding 10 f the basic task s first. In a fairings and fairings	flap track assembly years, are adequates and reporting are deploy flaps. 6 YR 18000 FC ort No. 1 & 2 fairing 6 YR 18000 FC	es, carriage asset to maintain correct not required on the not required on the not required on the not required and the not required an	amblies and forward osion at safe levels these components. ALL attachment locations	ALL ALL
	Inks. Normal assemblies recomponents. INTERVAL N ACCESS NO MRB Inspect lower holes. INTERVAL N INTERVAL N INTERVAL N	l overhaul procedures, a emoved, at intervals not. Therefore application of IOTE: Whichever come DTE: Remove flap forward 51-05-01-210-835 or side of lower surface (UIOTE: Whichever come DTE: Whichever come	pplied with the exceeding 10 f the basic task is first. In and fairings and 1.1 f.2 inder flap suppose its first. In and fairings and 1.2 inder flap suppose its first.	flap track assembly years, are adequates and reporting are deploy flaps. 6 YR 18000 FC ort No. 1 & 2 fairing 6 YR 18000 FC	es, carriage asset to maintain correct not required on the not required on the not required on the not required and the not required an	amblies and forward osion at safe levels these components. ALL attachment locations	ALL ALL

Inspect left main landing gear support structure: 1. Main landing gear beam assembly; 2. Outboard support (dog house) assembly; 3. Inboard support (hanger link) assembly; 4. Trunnion support assembly; 5. Stabilizer links, including attach fittings and fuse pins.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-250-02-01	MRB	51-05-01-210-806 57-05-03-210-838	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
		main landing gear supp Inboard support (hange use pins.		0 0	•	• • • • • • • • • • • • • • • • • • • •	` •
	INTERVAL N	NOTE: Whichever come	s first.				
57-260-01-01	MRB	51-05-01-210-806 57-05-03-210-839	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	Inspect aft si at trunnion a	ide of rear spar (chords, ttachment.	webs and stiffe	eners), including at	main landing gea	r outboard support	attachment, a
	INTERVAL N	NOTE: Whichever come	s first.				
57-260-02-01	MRB	51-05-01-210-806 57-05-03-210-840	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL
	Inspect aft si at trunnion a	ide of rear spar (chords, ttachment.	webs and stiffe	eners), including at	main landing gea	r outboard support	attachment, a
	INTERVAL N	NOTE: Whichever come	s first.				
57-270-01-01	MRB	51-05-01-210-806 57-05-03-210-841	1.1	10 YR	5 YR	ALL	ALL
	Inspect the in	nterior of left wing trailing	g edge cavity, i	ncluding skins, ribs	, ailerons and spo	oilers.	
	ACCESS NO	OTE: Flap extension req	luired for inspe	ction.			
57-270-02-01	MRB	51-05-01-210-806 57-05-03-210-842	1.1	10 YR	5 YR	ALL	ALL
	Inspect the in	nterior of right wing trailing	ng edge cavity,	including skins, rib	s, ailerons and sp	poilers.	
	ACCESS NO	OTE: Flap extension req	uired for inspe	ction.			
57-280-01-01	MRB	51-05-01-210-801 57-05-03-210-843	1.1 1.2	9 YR 18000 FC	9 YR 18000 FC	ALL	ALL
	Inspect left in	nboard ground spoiler ac					
		NOTE: Whichever come	•				
		OTE: Extend flaps and g					
57-280-02-01	MRB	51-05-01-210-801	1.1	9 YR	9 YR	ALL	ALL
37-200-02-01		57-05-03-210-844	1.2	18000 FC	18000 FC	,	/ \

Inspect right inboard ground spoiler actuator fittings.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Extend flaps and ground spoilers.







			INTERVAL		APPLICA	ABILITY			
SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
MRB	51-05-01-210-806 57-05-03-210-845	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL			
Inspect left inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspar ribs, torque tube, torque tube ribs 2. Aft flap track support assembly attachment on main flap rear spar.									
INTERVAL N	NOTE: Whichever come	s first.							
MRB	51-05-01-210-806 57-05-03-210-846	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL			
					oar ribs, torque tube	, torque tube			
INTERVAL N	NOTE: Whichever come	s first.							
MRB	51-05-01-210-806 57-05-03-210-847	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL			
Inspect aft si	ide of rear spar (chords,	webs and stiffe	eners), including fla	p track 1 & 2 sup	port fittings.				
INTERVAL N	NOTE: Whichever come	s first.							
MRB	51-05-01-210-806 57-05-03-210-848	1.1 1.2	6 YR 18000 FC	6 YR 18000 FC	ALL	ALL			
Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 7 & 8 support fittings.									
INTERVAL N	NOTE: Whichever come	s first.							
MRB	51-05-01-210-806 57-05-03-210-849	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL			
Inspect left outboard flap internally: 1. Front spar (aft side), including support fittings at WBL 254 and 358; 2. Rear spar (forward side); 3. Inspar ribs and aft flap track support ribs.									
INTERVAL NOTE: Whichever comes first.									
ACCESS NO	OTE: Deploy aft flap so t	that aft flap trad	cks do not block vie	ew of rear spar lov	wer chord.				
MRB	51-05-01-210-806 57-05-03-210-850	1.1 1.2	10 YR 36000 FC	10 YR 36000 FC	ALL	ALL			
	'	. ,	,,	support fittings at	WBL 254 and 358;	2. Rear spar			
INTERVAL N	NOTE: Whichever come	s first.							
ACCESS NOTE: Deploy aft flap so that aft flap tracks do not block view of rear spar lower chord.									
ACCESS NO									
	MRB Inspect left in 2. Aft flap training interval interva	MRB 51-05-01-210-806 57-05-03-210-845 Inspect left inboard flap internally: 1. 2. Aft flap track support assembly att INTERVAL NOTE: Whichever come MRB 51-05-01-210-806 57-05-03-210-846 Inspect right inboard flap internally: 1 ribs; 2. Aft flap track support assembly INTERVAL NOTE: Whichever come MRB 51-05-01-210-806 57-05-03-210-847 Inspect aft side of rear spar (chords, INTERVAL NOTE: Whichever come MRB 51-05-01-210-806 57-05-03-210-848 Inspect aft side of rear spar (chords, INTERVAL NOTE: Whichever come MRB 51-05-01-210-806 57-05-03-210-849 Inspect left outboard flap internally: 1 (forward side); 3. Inspar ribs and aft interval Note: Deploy aft flap so interval outboard flap internally: 1 (forward side); 3. Inspar ribs and aft interval outboard flap internally: 1 (forward side); 3. Inspar ribs and aft interval outboard flap internally: 1 (forward side); 3. Inspar ribs and aft interval outboard flap internally: 1 (forward side); 3. Inspar ribs and aft interval outboard flap internally: 1 (forward side); 3. Inspar ribs and aft interval side side	MRB 51-05-01-210-806 1.1 Inspect left inboard flap internally: 1. Front spar (aft 2. Aft flap track support assembly attachment on material interval interv	SOURCE AMM TASK REF VERSION THRESHOLD MRB 51-05-01-210-806 1.1 10 YR 57-05-03-210-845 1.2 36000 FC Inspect left inboard flap internally: 1. Front spar (aft side), rear spar (for 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 57-05-03-210-846 1.2 36000 FC Inspect right inboard flap internally: 1. Front spar (aft side), rear spar (fribs; 2. Aft flap track support assembly attachment on main flap rear spar (fribs; 2. Aft flap track support assembly attachment on main flap rear spar (INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR Inspect aft side of rear spar (chords, webs and stiffeners), including flat INTERVAL NOTE: Whichever comes first. 1.1 6 YR MRB 51-05-01-210-806 1.1 10 YR 57-05-03-210-849 1.2 36000 FC Inspect left outboard flap internally: 1. Front spar (aft side), including side); 3. Inspar ribs and aft flap track support ribs. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Deploy aft flap so that aft flap tracks do not block views for the proper space of the proper ribs. </td <td>SOURCE AMM TASK REF VERSION THRESHOLD REPEAT MRB 51-05-01-210-806 1.1 10 YR 36000 FC 36000 FC Inspect left inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspect 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 10 YR 57-05-03-210-846 1.2 36000 FC 36000 FC Inspect right inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspirits; 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 1 & 2 sup INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 7 & 8 sup INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR MRB 51-05-01-210-806 1.1 10 YR 36000 FC Inspect left outboard flap internally: 1. F</td> <td>SOURCE AMM TASK REF VERSION THRESHOLD REPEAT AIRPLANE MRB 51-05-01-210-806 1.1 10 YR 10 YR ALL Inspect left inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspar ribs, torque tube, 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 10 YR ALL 57-05-03-210-846 1.2 36000 FC 36000 FC Inspect right inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspar ribs, torque tube ribs; 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR ALL 57-05-03-210-847 1.2 18000 FC 18000 FC Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 1 & 2 support fittings. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR ALL 57-05-03-210-848 1.2 18000 FC 18000 FC Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 1 & 2 support fittings. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR ALL 57-05-03-210-848 1.2 18000 FC 18000 FC Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 7 & 8 support fittings. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 10 YR ALL 57-05-03-210-849 1.2 36000 FC 36000 FC Inspect left outboard flap internally: 1. Front spar (aft side), including support fittings at WBL 254 and 358; 2 (forward side); 3. Inspar ribs and aft flap track support ribs. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Deploy aft flap so that aft flap tracks do not block view of rear spar lower chord.</td>	SOURCE AMM TASK REF VERSION THRESHOLD REPEAT MRB 51-05-01-210-806 1.1 10 YR 36000 FC 36000 FC Inspect left inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspect 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 10 YR 57-05-03-210-846 1.2 36000 FC 36000 FC Inspect right inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspirits; 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 1 & 2 sup INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 7 & 8 sup INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR MRB 51-05-01-210-806 1.1 10 YR 36000 FC Inspect left outboard flap internally: 1. F	SOURCE AMM TASK REF VERSION THRESHOLD REPEAT AIRPLANE MRB 51-05-01-210-806 1.1 10 YR 10 YR ALL Inspect left inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspar ribs, torque tube, 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 10 YR ALL 57-05-03-210-846 1.2 36000 FC 36000 FC Inspect right inboard flap internally: 1. Front spar (aft side), rear spar (forward side), inspar ribs, torque tube ribs; 2. Aft flap track support assembly attachment on main flap rear spar. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR ALL 57-05-03-210-847 1.2 18000 FC 18000 FC Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 1 & 2 support fittings. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR ALL 57-05-03-210-848 1.2 18000 FC 18000 FC Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 1 & 2 support fittings. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 6 YR 6 YR ALL 57-05-03-210-848 1.2 18000 FC 18000 FC Inspect aft side of rear spar (chords, webs and stiffeners), including flap track 7 & 8 support fittings. INTERVAL NOTE: Whichever comes first. MRB 51-05-01-210-806 1.1 10 YR 10 YR ALL 57-05-03-210-849 1.2 36000 FC 36000 FC Inspect left outboard flap internally: 1. Front spar (aft side), including support fittings at WBL 254 and 358; 2 (forward side); 3. Inspar ribs and aft flap track support ribs. INTERVAL NOTE: Whichever comes first. ACCESS NOTE: Deploy aft flap so that aft flap tracks do not block view of rear spar lower chord.			

panels and spars.

Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 57-340-01.

AIRPLANE NOTE: All airplanes equipped with winglets.

ACCESS NOTE: Access through cover 527AB.







TACK CADD NO				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-340-02-01	MRB	51-05-01-210-806 57-05-03-211-804	1.1	6 YR	6 YR	ALL	ALL			
	Inspect aluminum rib structure at winglet stations 0, 1, and 4. Utilize borescope to inspect the flanges adjacent to skin panels and spars.									
	Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 57-340-02. AIRPLANE NOTE: All airplanes equipped with winglets.									
			• •	giets.						
	ACCESS NC	OTE: Access through co	over 627Ab.							
57-351-01-01	MRB	51-05-01-210-804 57-05-03-211-805	1.1	6 YR	6 YR	ALL	ALL			
		r and lower flanges and sk satisfies the requiren	•	•	•	•	. 658.17.			
	AIRPLANE N	NOTE: All airplanes equ	uipped with win	glets.						
	ACCESS NO	OTE: Remove winglet a	ssembly. Remo	ove winglet access	panels as noted.					
		Remove barrel nu	ts to facilitate in	nspection of recess	es.					
57-351-02-01	MRB	51-05-01-210-804 57-05-03-211-806	1.1	6 YR	6 YR	ALL	ALL			
	Inspect upper and lower flanges and webs, including barrel nut holes, Winglet Rib 0 and Wing Rib 27, WBL 658.17. Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 57-350-02.									
	AIRPLANE NOTE: All airplanes equipped with winglets.									
	ACCESS NOTE: Remove winglet assembly. Remove winglet access panels as noted.									
		Remove barrel nu	ts to facilitate in	spection of recess	es.					
57-600-00-01	AWL	57-05-02-130-801	1.1	56000 FC	36000 FC	ALL	ALL			
		asonic) the upper side o 26A001 - DTR, DTR ch		•		terface at STA 639.				
	ACCESS NO	OTE: Inspection require	s removal of wi	ng-to-body fairing.						
57-600-00-02	AWL	57-05-02-130-801	1.1	56000 FC	36000 FC	ALL	ALL			
		asonic) the upper side o 26A001 - DTR, DTR ch				terface at STA 639.				
	ACCESS NO	OTE: Inspection require	s removal of wi	ng-to-body fairing.	-					
57-601-00-01	AWL	57-05-02-250-801	1.1	56000 FC	18000 FC	ALL	ALL			
01-001-00-01	AVVL.	01-00-02-200-001	1.1	3000010	1000010	ALL	ALL			

Inspect (High Frequency Eddy Current) the double plus chord at stringer 18-A interface located at aft end of upper vertical flange radius at vertical flange/horizontal flange.

See Doc. D626A001 - DTR, DTR check form 57-10-05-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-84.

ACCESS NOTE: Inspection requires removal of wing-to-body fairing.







				INTERVAL		APPLICA	BILITY					
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE					
57-601-00-02	AWL	57-05-02-250-801	1.1	56000 FC	18000 FC	ALL	ALL					
	flange radius See Doc. Do The NDI me	Inspect (High Frequency Eddy Current) the double plus chord at stringer 18-A interface located at aft end of upper vertical flange radius at vertical flange/horizontal flange. See Doc. D626A001 - DTR, DTR check form 57-10-05-2, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-84.										
	ACCESS NO	OTE: Inspection requires	s removal of wi	ng-to-body fairing.								
57-601-10-01	AWL	57-05-02-130-802	1.1	56000 FC	18000 FC	600 700 700C 700IGW 800 900	ALL					
	663, on top (asonic) the double plus of of wing, aft of overwing e 326A001 - DTR, DTR cho	exits.			, located between ST	A 639 and S					
	See Doc. D626A001 - DTR, DTR check form 57-10-05-3a, for alternative inspection. AIRPLANE NOTE: All except 900ER											
	ACCESS NOTE: Inspection requires removal of wing-to-body fairing.											
57-601-10-02	AWL	57-05-02-130-802	1.1	56000 FC	18000 FC	600 700 700C 700IGW 800 900	ALL					
	Inspect (Ultrasonic) the double plus chord vertical flange at body stringer 18A interface, located between STA 639 and STA 663, on top of wing, aft of overwing exits. See Doc. D626A001 - DTR, DTR check form 57-10-05-3a, for alternative inspection.											
	AIRPLANE NOTE: All except 900ER											
	ACCESS NOTE: Inspection requires removal of wing-to-body fairing.											
57-601-22-01	AWL	57-05-02-210-823	1.1	NOTE		ALL	ALL					
	on both the i	Inspect (General Visual) the upper horizontal flange at the double plus chord from the front spar to rear spar. Inspection is on both the inboard and outboard locations at BBL 70.85. See Doc. D626A001 - DTR, DTR check form 57-10-05-4, for alternative inspection.										
	INTERVAL I	NOTE: Flight length sen recommended re		sk. See Section 9, F n interval is 18000F		mine threshold. Boeir	ng					
	ACCESS NO	OTE: Inspection requires	s removal of th	e wing-to-body fairi	ing and floor pan	els.						
57-601-22-02	AWL	57-05-02-210-823	1.1	NOTE		ALL	ALL					
	on both the i	neral Visual) the upper he nboard and outboard loc 326A001 - DTR, DTR che	cations at BBL	70.85.		front spar to rear spar	r. Inspection i					

See Doc. D626A001 - DTR, DTR check form 57-10-05-4, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing

recommended repeat inspection interval is 18000FC.

ACCESS NOTE: Inspection requires removal of the wing-to-body fairing and floor panels.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
57-601-25-01	AWL	57-05-02-250-846	1.1	NOTE		ALL	ALL				
	616 and 639		,	-		tub beams at STAs	559, 578, 597,				
	See Doc. D626A001 - DTR, DTR check form 57-10-05-5, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual										
	The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-43.										
	INTERVAL	NOTE: Flight length sens recommended re	, ,	sk. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng				
	ACCESS N	OTE: Floor panel remova	al is required.								
57-601-25-02	AWL	57-05-02-250-846	1.1	NOTE		ALL	ALL				
	Inspect (Hig 616 and 639	h Frequency Eddy Curre 9.	nt) the upper h	orizontal flange of t	the chord at the s	tub beams at STAs	559, 578, 597,				
		626A001 - DTR, DTR che		•	•						
	The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-43.										
	INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.										
	ACCESS N	OTE: Floor panel remova	al is required.								
57-601-27-01	AWL	57-05-02-210-824	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (General Visual) the upper vertical flange at the double plus chord forward of STA 639. See Doc D626A001-DTR, DTR Check Form 57-10-05-6 For alternative inspections.										
	ACCESS N	OTE: Inspection requires	s removal of w	ing-to-body fairing.							
57-601-27-02	AWL	57-05-02-210-824	1.1	56000 FC	18000 FC	ALL	ALL				
		neral Visual) the upper ve 26A001-DTR, DTR Chec	ŭ	•		ГА 639.					
	ACCESS N	OTE: Inspection requires	s removal of w	ing-to-body fairing.	·						
57-601-28-01	AWL	57-05-02-250-921	1.1	56000 FC	18000 FC	ALL	ALL				
	. , .	h Frequency Eddy Currer	,	-	•	rd forward of STA 63	39.				
	The NDI me	thod(s) necessary to acc The inspection procedur	omplish the in	tent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Manua				
	ACCESS N	OTE: Inspection requires	s removal of w	ing-to-body fairing.							
57-601-28-02	AWL	57-05-02-250-921	1.1	56000 FC	18000 FC	ALL	ALL				
	Inspect (Hig	h Frequency Eddy Curre	nt) the unner w	ertical flance at the	double plus cho	rd forward of STA 63	39				

 $Inspect \ (High\ Frequency\ Eddy\ Current)\ the\ upper\ vertical\ flange\ at\ the\ double\ plus\ chord\ forward\ of\ STA\ 639.$

See Doc. D626A001 - DTR, DTR check form 57-10-05-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-77.

ACCESS NOTE: Inspection requires removal of wing-to-body fairing.







			APPLICA	ABILITY							
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
57-601-30-01	AWL	57-05-02-130-804	1.1	NOTE		ALL	ALL				
	Inspect (Ultrasonic) the wing center section upper skin at floor beams and shear ties located at BL 0, BL 25, BL 45, from front spar to rear spar. See Doc. D626A001 - DTR, DTR check form 57-10-06, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 4, Subject 57-10-12. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing										
	recommended repeat inspection interval is 36000FC. ACCESS NOTE: Floor panel removal is required.										
57-601-30-02	AWL	57-05-02-130-804	1.1	NOTE		ALL	ALL				
	front spar to See Doc. D6 The NDI met	asonic) the wing center s rear spar. 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	eck form 57-10 omplish the int	-06, for alternative i	inspection. on is contained in						
		recommended re TE: Floor panel remove	peat inspectio	sk. See Section 9, F n interval is 36000F		nine threshold. Boeii	ng				
57-602-00-01	AWL	57-05-02-211-802	1.1	NOTE		ALL	ALL				
	No. 8 from LI See Doc. D6	ailed) the wing center se BL 67.0 to RBL 67.0. 26A001 - DTR, DTR che IOTE: Flight length sense recommended re	eck form 57-10 sitive (FLS) tas	-07-1, for alternative	e inspection. igure 1 to detern		Ü				
57-602-00-02	AWL	57-05-02-211-802	1.1	NOTE		ALL	ALL				
	No. 8 from LI	ailed) the wing center se BL 67.0 to RBL 67.0. 26A001 - DTR, DTR che		-		No. 4 and stringer	No. 6 through				
	INTERVAL N	IOTE: Flight length sense recommended re	' '	sk. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng				
57-602-10-01	AWL	57-05-02-211-803	1.1	NOTE		ALL	ALL				
		ailed) wing center section 26A001 - DTR, DTR che				and side of body to	side of body.				
	INTERVAL N	IOTE: Flight length sense recommended re	, ,	sk. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng				
57-602-10-02	AWL	57-05-02-211-803	1.1	NOTE		ALL	ALL				
	Inchest (Deta	ailed) wing center section	n lower panel	ekin from the front s	enar to rear enar	and side of body to	side of body				

Inspect (Detailed) wing center section, lower panel skin from the front spar to rear spar and side of body to side of body. See Doc. D626A001 - DTR, DTR check form 57-10-07/08, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.





				INTERVAL		APPLIC	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-603-00-01	AWL	57-05-02-211-804	1.1	NOTE		ALL	ALL			
	Inspect (Deta	ailed) wing center sectio	n lower splice	stringer No. 5 and N	No. 9 from LBL 6	7.0 to RBL 67.0.				
	See Doc. D6	326A001 - DTR, DTR ch	eck form 57-10	-08, for alternative	inspection.					
	INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 15000FC.									
57-603-00-02	AWL 57-05-02-211-804 1.1 NOTE ALL ALL									
		ailed) wing center sectio 326A001 - DTR, DTR ch		-		7.0 to RBL 67.0.				
	INTERVAL N	NOTE: Flight length sen recommended re		sk. See Section 9, F n interval is 15000F	-	nine threshold. Boei	ng			
57-604-00-01	AWL	57-05-02-210-801	1.1	NOTE		ALL	ALL			
		neral Visual) the wing ce 326A001 - DTR, DTR ch				n LBL 67.0 to RBL 6	7.0.			
	INTERVAL N	NOTE: Flight length sen recommended re		sk. See Section 9, F n interval is 18000F		nine threshold. Boei	ng			
57-604-00-02	AWL	57-05-02-210-801	1.1	NOTE		ALL	ALL			
	. ,	neral Visual) the wing ce		•	•	n LBL 67.0 to RBL 6	7.0.			
	See Doc. D626A001 - DTR, DTR check form 57-10-09, for alternative inspection.									
	INTERVAL N	NOTE: Flight length sen recommended re		sk. See Section 9, F n interval is 18000F		nine threshold. Boei	ng			
57-605-00-01	AWL	57-05-02-250-802	1.1	NOTE		ALL	ALL			
	rear spar at I See Doc. D6 The NDI met	r Frequency Eddy Curre BBL 70.85. Inspection is 326A001 - DTR, DTR ch thod(s) necessary to acc The inspection procedu	on both the in eck form 57-10 complish the in	board and outboard -11-1, for alternativ tent of this inspection	d locations at BBI re inspection. on is contained in	_ 70.85.	·			
		NOTE: Flight length sen	sitive (FLS) tas		Figure 2 to deterr	nine threshold. Boei	ng			
	ACCESS NO	OTE: Inspection require								
57-605-00-02	AWL	57-05-02-250-802	1.1	NOTE		ALL	ALL			
	rear spar at I See Doc. D6 The NDI met	r Frequency Eddy Curre BBL 70.85. Inspection is 326A001 - DTR, DTR ch thod(s) necessary to acc The inspection procedu	on both the in eck form 57-10 complish the in	board and outboard 1-11-1, for alternativ tent of this inspection	d locations at BBI re inspection. on is contained in	_ 70.85.	·			
	,	The inspection procedu				nine threshold Roei	na			

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing

recommended repeat inspection interval is 18000FC.

ACCESS NOTE: Inspection requires removal of Wing-to-Body fairing.





				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
57-605-01-01	AWL	57-05-02-250-803	1.1	NOTE		ALL	ALL				
	the rear spa See Doc. Do The NDI me (D6-37239).	Inspect (High Frequency Eddy Current) the side of body splice lower surface at the lower tee chord from the front spar to the rear spar at BBL 70.85. See Doc. D626A001 - DTR, DTR check form 57-10-11-1, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-35. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.									
57.005.04.00	A1A#	F7 0F 00 0F0 000	4.4	NOTE		A11	A. I.				
57-605-01-02	AWL	57-05-02-250-803 h Frequency Eddy Curre	1.1	NOTE		ALL	ALL				
	See Doc. Do The NDI me (D6-37239).	r at BBL 70.85. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen recommended re	omplish the in es are contain sitive (FLS) tas	tent of this inspection ed in Part 6, Subject	on is contained in et 57-10-35. Figure 2 to deterr						
57-605-10-01	AWL	57-05-02-250-850	1.1	NOTE		ALL	ALL				
	See Doc. Do The NDI me (D6-37239).	ear spar at BBL 70.85. 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen- recommended re OTE: Inspection requires	omplish the interest are contain sitive (FLS) tast peat inspection	tent of this inspectic ed in Part 6, Subject sk. See Section 9, F n interval is 18000F	on is contained in ot 57-10-39. Figure 2 to deterr FC.						
57-605-10-02	AWL	57-05-02-250-850	1.1	NOTE		ALL	ALL				
	spar to the r See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre ear spar at BBL 70.85. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen- recommended re	eck form 57-10 omplish the in es are contain sitive (FLS) tas epeat inspectio	1-11-2, for alternative tent of this inspection ed in Part 6, Subjection sk. See Section 9, F n interval is 18000F	e inspection. on is contained in ot 57-10-39. Figure 2 to deterr FC.	the 737 Nondestruc	ctive Test Manual				
57-605-20-01	AWL	57-05-02-250-852	1.1	NOTE		ALL	ALL				
	the rear spa See Doc. Do The NDI me	h Frequency Eddy Curre r at BBL 70.85. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-10 omplish the in	-11-3, for alternative	e inspection. on is contained in		·				

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing

recommended repeat inspection interval is 18000FC.

ACCESS NOTE: Inspection requires removal of Wing-to-Body fairing.





				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-605-20-02	AWL	57-05-02-250-852	1.1	NOTE		ALL	ALL			
	Inspect (High Frequency Eddy Current) the side of body splice, lower surface skin in the hidden areas from the front spar to the rear spar at BBL 70.85. See Doc. D626A001 - DTR, DTR check form 57-10-11-3, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-70. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.									
	ACCESS NO	OTE: Inspection requires	s removal of W	ing-to-Body fairing.						
57-606-00-01	AWL	57-05-02-130-805	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met	asonic) the lower panel a 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-10 complish the int	-12, for alternative i	nspection. In is contained in	·	ctive Test Man			
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	k. See Section 9, F n interval is 9000F0	•	nine threshold. Boei	ng			
57-606-00-02	AWL	57-05-02-130-805	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	asonic) the lower panel as 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen recommended re	eck form 57-10 complish the int res are containd sitive (FLS) tas	-12, for alternative i ent of this inspection ed in Part 4, Subject	nspection. on is contained in ot 57-10-13. igure 2 to detern	the 737 Nondestruc				
					··					
57-606-01-01	AWL	57-05-02-250-806	1.1	NOTE	,. 	ALL	ALL			
57-606-01-01	Inspect (High LBBL 3.5. See Doc. D6 The NDI met (D6-37239).	57-05-02-250-806 The Frequency Eddy Curre 266A001 - DTR, DTR che 26hod(s) necessary to acc 27the inspection procedur 28THE: Flight length sen	nt) the lower parents form 57-10 complish the intres are contained sitive (FLS) tas	anel skin at the drai -13, for alternative i ent of this inspectic ed in Part 6, Subjec	n installation bet nspection. In is contained in to 57-10-75. Gure 1 to detern	ween stringer S-7 and the 737 Nondestructure	nd stringer S-8			

Inspect (High Frequency Eddy Current) the lower panel skin at the drain installation between stringer S-7 and stringer S-8 at LBBL 3.5.

See Doc. D626A001 - DTR, DTR check form 57-10-13, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-75.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.





				INTERVAL		APPLIC	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-607-00-01	AWL	57-05-02-130-806	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	Inspect (Ultrasonic) the lower skin at the lower beam attachment from the front spar to the rear spar. See Doc. D626A001 - DTR, DTR check form 57-10-15, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 4, Subject 57-10-13.								
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 9000F0	•	nine threshold. Boei	ng			
57-607-00-02	AWL	57-05-02-130-806	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met	asonic) the lower skin at 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-10 complish the inf	1-15, for alternative tent of this inspection	inspection. on is contained in		ctive Test Manı			
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	sk. See Section 9, F n interval is 9000F0	-	nine threshold. Boei	ng			
57-607-10-01	AWL	57-05-02-211-805	1.1	NOTE		ALL	ALL			
	Inspect (Detailed) the web common to the fuel tank from LBBL 70.85 to RBBL 70.85. See Doc. D626A001 - DTR, DTR check form 57-10-17-1, for alternative inspection.									
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 4000F0	•	nine threshold. Boei	ng			
57-607-10-02	AWL	57-05-02-211-805	1.1	NOTE		ALL	ALL			
	Inspect (Detailed) the web common to the fuel tank from LBBL 70.85 to RBBL 70.85. See Doc. D626A001 - DTR, DTR check form 57-10-17-1, for alternative inspection.									
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 4000F0	•	nine threshold. Boei	ng			
57-607-11-01	AWL	57-05-02-250-807	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met	r Frequency Eddy Currer 26A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedur	eck form 57-10 complish the inf	-17-1, for alternativ	e inspection. on is contained in		ctive Test Man			
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 36000F	•	nine threshold. Boei	ng			
57-607-11-02	AWL	57-05-02-250-807	1.1	NOTE		ALL	ALL			
	Inspect (Low	Frequency Eddy Curre	nt) the web cor	mmon to the fuel tai	nk from LBBL 70.	85 to RBBL 70.85.				

Inspect (Low Frequency Eddy Current) the web common to the fuel tank from LBBL 70.85 to RBBL 70.85.

See Doc. D626A001 - DTR, DTR check form 57-10-17-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-81.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.





				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-607-20-01	AWL	57-05-02-211-806	1.1	NOTE		ALL	ALL			
		ailed) the web common t 326A001 - DTR, DTR che								
	INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 8000FC.									
57-607-20-02	AWL	57-05-02-211-806	1.1	NOTE		ALL	ALL			
		ailed) the web common t 326A001 - DTR, DTR che								
	INTERVAL I	NOTE: Flight length sense recommended re		k. See Section 9, F n interval is 8000F0		nine threshold. Boei	ng			
57-607-21-01	AWL	57-05-02-250-922	1.1	NOTE		ALL	ALL			
	The NDI me (D6-37239).	i26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	omplish the into es are containe sitive (FLS) tas	ent of this inspection and in Part 6, Subject	on is contained in et 57-10-81. Figure 1 to detern					
57-607-21-02	AWL	57-05-02-250-922	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI me (D6-37239).	rFrequency Eddy Currer (26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens	eck form 57-10- omplish the intended are contained	-17-2, for alternative ent of this inspection ed in Part 6, Subject	e inspection. on is contained in ct 57-10-81.	the 737 Nondestruc	ctive Test Man			
		recommended re	peat inspection	n interval is 36000F	C.		ng			
F7 C07 20 04	A)A/I			n interval is 36000F						
57-607-30-01	AWL	57-05-02-250-808	1.1	56000 FC	22600 FC	ALL	ng ALL			
57-607-30-01	Inspect (High See Doc. Do The NDI me		1.1 nt) rear spar keeck form 57-10- omplish the interpretation	56000 FC eel beam stiffeners -18, for alternative i	22600 FC at LBL 6.2 and R inspection. on is contained in	ALL BL 6.2.	ALL			
57-607-30-01 57-607-30-02	Inspect (High See Doc. Do The NDI me	57-05-02-250-808 in Frequency Eddy Curre is26A001 - DTR, DTR che thod(s) necessary to acc	1.1 nt) rear spar keeck form 57-10- omplish the interpretation	56000 FC eel beam stiffeners -18, for alternative i	22600 FC at LBL 6.2 and R inspection. on is contained in	ALL BL 6.2.	ALL			

Inspect (High Frequency Eddy Current) rear spar keel beam stiffeners at LBL 6.2 and RBL 6.2.

See Doc. D626A001 - DTR, DTR check form 57-10-18, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-74.







				INTERVAL		APPLICA	ABILITY			
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-608-00-01	AWL	57-05-02-250-809	1.1	NOTE		ALL	ALL			
	center section both sides of See Doc. D6	asonic) around the four A in lower skin at BBL 70.8 the joint. 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	5 as well as the eck form 57-10 omplish the int	e upper rear spar th -20, for alternative i ent of this inspectio	nrough the doubl nspection. n is contained in	e plus chord horizor	ntal flange on			
	INTERVAL N	IOTE: Flight length sens recommended re	` '	k. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng			
	ACCESS NO	OTE: Inspection requires	s removal of ex	ternal panels.						
57-608-00-02	AWL	57-05-02-250-809	1.1	NOTE		ALL	ALL			
	center section both sides of See Doc. D6	26A001 - DTR, DTR che hod(s) necessary to acc	5 as well as the eck form 57-10 omplish the int	e upper rear spar the -20, for alternative i ent of this inspection	nrough the doubl nspection. n is contained in	e plus chord horizor	ntal flange on			
	(D6-37239). The inspection procedures are contained in Part 4, Subject 57-10-17. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.									
	ACCESS NO	OTE: Inspection requires	s removal of ex	ternal panels.						
57-609-00-01	AWL	57-05-02-250-811	1.1	NOTE		ALL	ALL			
57-609-00-01	AWL Inspect (Ultra and center so both sides of See Doc. D6 The NDI met contained in procedures a INTERVAL N	57-05-02-250-811 asonic) around the four Fection lower skin at BBL the joint. 26A001 - DTR, DTR checked the 737 Nondestructive are contained in Part 4, S	1.1 FWD fasteners 70.85 as well a eck form 57-10 omplish the int Test Manual (D Subject 57-10-1 sitive (FLS) tas speat inspection	NOTE along the front spa as the upper front s -21, for alternative i ent of this inspectio 16-37239). The insp 7. k. See Section 9, F or interval is 18000F	par through the on the conspection. In is section sigure 2 to deterning the constant in the c	splice plate to the w double plus chord ho	ing lower skin orizonal flange o			
	AWL Inspect (Ultra and center suboth sides of See Doc. D6 The NDI met contained in procedures a INTERVAL MACCESS NO	57-05-02-250-811 asonic) around the four Fection lower skin at BBL the joint. 26A001 - DTR, DTR checked the 737 Nondestructive are contained in Part 4, SHOTE: Flight length sense recommended reDTE: Inspection requires	1.1 FWD fasteners 70.85 as well a eck form 57-10 omplish the int Test Manual (E Subject 57-10-1 sitive (FLS) tas peat inspection s removal of the	NOTE along the front spa as the upper front s -21, for alternative i ent of this inspection 6-37239). The insp 7. k. See Section 9, Fin interval is 18000F e external fairings.	par through the on the conspection. In is section sigure 2 to deterning the constant in the c	splice plate to the w double plus chord ho	ing lower skin orizonal flange o			
57-609-00-02	AWL Inspect (Ultra and center so both sides of See Doc. D6 The NDI met contained in procedures a INTERVAL N ACCESS NO	57-05-02-250-811 asonic) around the four Fection lower skin at BBL the joint. 26A001 - DTR, DTR checknod(s) necessary to accurate 737 Nondestructive are contained in Part 4, S	1.1 FWD fasteners 70.85 as well a eck form 57-10 omplish the int Test Manual (D Subject 57-10-1 sitive (FLS) tas peat inspection s removal of the	NOTE along the front spa as the upper front s -21, for alternative i ent of this inspectio 6-37239). The insp 7. k. See Section 9, F n interval is 18000F e external fairings.	par through the onspection. In is section Sigure 2 to determine.	splice plate to the w double plus chord ho nine threshold. Boei	ing lower skin orizonal flange o ng ALL			

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INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing

recommended repeat inspection interval is 18000FC.

ACCESS NOTE: Inspection requires removal of the external fairings.



57-610-01-02

STA 727B.



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-609-10-01	AWL	57-05-02-250-813	1.1	56000 FC	11000 FC	ALL	ALL			
	Inspect (High Frequency Eddy Current) the BL 0 and BL 25 floor beams from BS 655 to BS 675, and BL 0 floor beam from BS 716 to BS 727B, and BL 45 floor beam from BS 685 to BS 716. See Doc. D626A001 - DTR, DTR check form 57-10-23-1, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-83. ACCESS NOTE: Remove passenger cabin floor panels as required.									
57-609-10-02	AWL	57-05-02-250-813	1.1	56000 FC	11000 FC	ALL	ALL			
37-003-10-02		n Frequency Eddy Curre								
	See Doc. D6 The NDI met (D6-37239).	S 727B, and BL 45 floor 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Remove passenge	eck form 57-10 complish the interes are contain	-23-1, for alternative the control of this inspection of the control of the contr	on is contained in	the 737 Nondestrue	ctive Test Manual			
57-610-00-01	AWL	57-05-02-211-807	1.1	56000 FC	24000 FC	ALL	ALL			
	STA 727A. See Doc. D6	ailed) the BL0, 25, and 4 326A001 - DTR, DTR che DTE: Remove passenge	eck form 57-10	-23-2, for alternativ		. 25 floor beams fro	m STA 685 to			
57-610-00-02	AWL	57-05-02-211-807	1.1	56000 FC	24000 FC	ALL	ALL			
	STA 727A. See Doc. D6	ailed) the BL0, 25, and 4 326A001 - DTR, DTR che DTE: Remove passenge	eck form 57-10	-23-2, for alternativ		. 25 floor beams fro	m STA 685 to			
57-610-01-01	AWL	57-05-02-250-814	1.1	56000 FC	18000 FC	ALL	ALL			
	STA 727B. See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: Remove passenge	eck form 57-10 complish the interes are contain	-23-3, for alternativent of this inspection of the control of the	ve inspection. on is contained in					

AWL 57-05-02-250-814 1.1 56000 FC 18000 FC ALL ALL

Inspect (High Frequency Eddy Current) the lower chord along the radius at BL0, 25, and 45 floor beams from STA 664 to

See Doc. D626A001 - DTR, DTR check form 57-10-23-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-83.

ACCESS NOTE: Remove passenger cabin floor panels as required.





TASK CARD NO.			INTERVAL			APPLICABILITY	
	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-611-00-01	AWL	57-05-02-250-815	1.1	56000 FC	36000 FC	ALL	ALL
	Inspect (High Frequency Eddy Current) the lower chord along the radius at the BL 0, 25, 45 floor beams between STA 574 through STA 664. See Doc. D626A001 - DTR, DTR check form 57-10-23-4, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manu (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-83. ACCESS NOTE: Remove passenger cabin floor panels as required. Sealant present in chord radius must be						
		removed for full ins	spection credit				
57-611-00-02	AWL	57-05-02-250-815	1.1	56000 FC	36000 FC	ALL	ALL
	through STA 664. See Doc. D626A001 - DTR, DTR check form 57-10-23-4, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manus (D6-37239). The inspection procedures are contained in Part 6, Section 57-10-83. ACCESS NOTE: Remove passenger cabin floor panels as required. Sealant present in chord radius must be removed for full inspection credit.						
57-611-10-01	AWL	57-05-02-211-808	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (Detailed) the lower chord along the BL 0 25, and 45 floor beams between STA 540 through STA 574. See Doc. D626A001 - DTR, DTR check form 57-10-23-5, for alternative inspection. ACCESS NOTE: Remove passenger cabin floor panels as required. Remove any sealant beyond specifications for full inspection credit.						
57-611-10-02	AWL	57-05-02-211-808	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (Detailed) the lower chord along the BL 0 25, and 45 floor beams between STA 540 through STA 574. See Doc. D626A001 - DTR, DTR check form 57-10-23-5, for alternative inspection. ACCESS NOTE: Remove passenger cabin floor panels as required. Remove any sealant beyond specifications for						
		full inspection crec	lit.				
57-611-20-01	AWL	57-05-02-211-809	1.1	NOTE		ALL	ALL
	Inspect (Detailed) the typical web to stiffener attach points on the side of body rib. See Doc. D626A001 - DTR, DTR check form 57-10-25, for alternative inspection.						
	INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.						
57-611-20-02	AWL	57-05-02-211-809	1.1	NOTE		ALL	ALL
	Inspect (Deta	ailed) the typical web to	stiffener attach	points on the side	of body rib.		

See Doc. D626A001 - DTR, DTR check form 57-10-25, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-611-21-01	AWL	57-05-02-211-810	1.1	NOTE		ALL	ALL
		ailed) the typical web to s 326A001 - DTR, DTR che		•	,		
	INTERVAL I	NOTE: Flight length sense recommended re	` '	k. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng
	ACCESS NO	OTE: Remove passenge	er cabin floor pa	anels as required.			
57-611-21-02	AWL	57-05-02-211-810	1.1	NOTE		ALL	ALL
		ailed) the typical web to s 326A001 - DTR, DTR che					
	INTERVAL I	NOTE: Flight length sens		k. See Section 9, F n interval is 18000F		nine threshold. Boei	ng
	ACCESS NO	OTE: Remove passenge	er cabin floor pa	anels as required.			
57-612-00-01	AWL	57-05-02-210-802	1.1	NOTE		ALL	ALL
	areas.	neral Visual) stringers S-2		· ·		rib 10 at the non-hi	dden, fairing
	INTERVAL I	NOTE: Flight length sense recommended re	` ,	k. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-612-00-02	AWL	57-05-02-210-802	1.1	NOTE		ALL	ALL
	areas.	neral Visual) stringers S-2 326A001 - DTR, DTR che		· ·		rib 10 at the non-hi	dden, fairing
	INTERVAL I	NOTE: Flight length sense recommended re	` ,	k. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-612-01-01	AWL	57-05-02-210-804	1.1	NOTE		ALL	ALL
	non hidden a	neral Visual) stringers S- areas. 326A001 - DTR, DTR che	· ·	ŭ		rib 15 adjacent to sp	oar chords at
	INTERVAL I	NOTE: Flight length sense recommended re	` ,	k. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-612-01-02	AWL	57-05-02-210-804	1.1	NOTE		ALL	ALL
	non hidden a	neral Visual) stringers Sareas.	· ·	· ·		rib 15 adjacent to sp	oar chords at

See Doc. D626A001 - DTR, DTR check form 57-20-01-2, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing

recommended repeat inspection interval is 24000FC.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-613-00-01	AWL	57-05-02-250-817	1.1	NOTE		ALL	ALL
	hidden by se See Doc. D6 The NDI met (D6-37239).	Frequency Eddy Currer al pans & sealant. 26A001 - DTR, DTR che hod(s) necessary to accommended recommended re	eck form 57-20 omplish the intes are contain sitive (FLS) tas	-01-3, for alternative ent of this inspection ed in Part 6, Subject	e inspection. In is contained in the total the	the 737 Nondestruc	ctive Test Manua
57-613-00-02	AWL	57-05-02-250-817	1.1	NOTE		ALL	ALL
	hidden by se See Doc. D6 The NDI met (D6-37239).	Frequency Eddy Currer al pans & sealant. 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sens recommended re	eck form 57-20 omplish the intes are contain sitive (FLS) tas	-01-3, for alternative tent of this inspection ed in Part 6, Subject	e inspection. In is contained in the total total total total to the total tota	the 737 Nondestruc	ctive Test Manua
57-614-00-01	AWL	57-05-02-130-807	1.1	NOTE		ALL	ALL
	The NDI met (D6-37239).	26A001 - DTR, DTR che hod(s) necessary to according The inspection procedur IOTE: Flight length sens recommended re	omplish the intest are contain sitive (FLS) tas	ent of this inspectioned in Part 4, Subject	n is contained in at 57-10-15. Figure 2 to determ	the 737 Nondestruc	
57-614-00-02	AWL	57-05-02-130-807	1.1	NOTE		ALL	ALL
	See Doc. D6 The NDI met (D6-37239).	asonic) the skin panel at 26A001 - DTR, DTR che hod(s) necessary to according The inspection procedur IOTE: Flight length sens recommended re	eck form 57-20 omplish the int es are contain sitive (FLS) tas	-01, -02, -03, -04, -0 cent of this inspection ed in Part 4, Subject	05, -08-1, for alte in is contained in it 57-10-15. igure 2 to determ	rnative inspection. the 737 Nondestruc	
57-614-10-01	AWL	57-05-02-210-806	1.1	NOTE		ALL	ALL
	See Doc. D6	eral Visual) rib 1 to rib 1- 26A001 - DTR, DTR che IOTE: Flight length sens recommended re	eck form 57-20 sitive (FLS) tas	-01, -02, -03, -04, -0	05, -08-2, for alte igure 2 to determ	rnative inspection.	ng
57-614-10-02	AWL	57-05-02-210-806	1.1	NOTE		ALL	ALL
	Inspect (Gen	eral Visual) rib 1 to rib 1 26A001 - DTR, DTR che		ally visible areas fro		to the rear spar.	

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 4000FC.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGIN
57-614-20-01	AWL	57-05-02-211-813	1.1	NOTE		ALL	ALL
		ailed) rib 1 to rib 27 at the 26A001 - DTR, DTR che			•	rnative inspection	
		NOTE: Flight length sens	sitive (FLS) tas		Figure 2 to determ	·	ng
	ACCESS NO	OTE: Fairing removal re	quired.				
57-614-20-02	AWL	57-05-02-211-813	1.1	NOTE		ALL	ALL
		ailed) rib 1 to rib 27 at the 26A001 - DTR, DTR che		, ,	O .	ernative inspection.	
	INTERVAL N	NOTE: Flight length sens	` '	k. See Section 9, F n interval is 3000F0	•	nine threshold. Boei	ng
	ACCESS NO	OTE: Fairing removal re	quired.				
57-614-30-01	AWL	57-05-02-210-808	1.1	NOTE		ALL	ALL
		neral Visual) rib 14 to rib 226A001 - DTR, DTR che		,			
	INTERVAL N	NOTE: Flight length sense recommended re	` '	k. See Section 9, F n interval is 5000F0	•	nine threshold. Boei	ng
57-614-30-02	AWL	57-05-02-210-808	1.1	NOTE		ALL	ALL
		neral Visual) rib 14 to rib 26A001 - DTR, DTR che		•	•	•	
	INTERVAL N	NOTE: Flight length sense recommended re	, ,	k. See Section 9, For interval is 5000F0	-	nine threshold. Boei	ng
57-614-40-01	AWL	57-05-02-211-815	1.1	NOTE		ALL	ALL
		ailed) stringer S-6 and S- 26A001 - DTR, DTR che				non-faired areas.	
	INTERVAL N	NOTE: Flight length sense recommended re	` ,	k. See Section 9, F n interval is 18000F	O	nine threshold. Boei	ng
57-614-40-02	AWL	57-05-02-211-815	1.1	NOTE		ALL	ALL
	Inspect (Deta	ailed) stringer S-6 and S-	-8 from rib 1 to	rib 19 at the non h	idden, faired and	non-faired areas.	

Inspect (Detailed) stringer S-6 and S-8 from rib 1 to rib 19 at the non hidden, faired and non-faired areas. See Doc. D626A001 - DTR, DTR check form 57-20-02-1, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-615-00-01	AWL	57-05-02-130-809	1.1	NOTE		ALL	ALL
	stringers are See Doc. D6 The NDI me (D6-37239).	asonic) the vertical flange hidden under seal pan a 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	and sealant. eck form 57-20 complish the interes are contained	-02-2, for alternative ent of this inspection ed in Part 4, Subject	e inspection. on is contained in ot 57-10-07. Figure 1 to detern	the 737 Nondestruc	ctive Test Manu
57-615-00-02	AWL	57-05-02-130-809	1.1	NOTE		ALL	ALL
	stringers are See Doc. Do The NDI me (D6-37239).	asonic) the vertical flange hidden under seal pan a 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen recommended re	and sealant. eck form 57-20 complish the intended are contained sitive (FLS) tas	-02-2, for alternative ent of this inspection ed in Part 4, Subject	e inspection. on is contained in ot 57-10-07. Figure 1 to detern	the 737 Nondestruc	ctive Test Manu
57-615-01-01	AWL	57-05-02-250-855	1.1	NOTE		ALL	ALL
	Inspect (High	h Frequency Eddy Curre	nt) the lower w		er, S-8, from rib 6		
	Inspect (High nacelle supp See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre fort strut attachment fittin 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens	nt) the lower wig. eck form 57-20 complish the intres are contained	ing panel rail string -02-3, for alternative ent of this inspection ed in Part 6, Subject	e inspection. on is contained in ot 57-10-80. Figure 1 to detern	6 to rib 7 at the area	s hidden by the
57-615-01-02	Inspect (High nacelle supp See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre fort strut attachment fittin 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens	nt) the lower wig. eck form 57-20 complish the intres are contained	ing panel rail string -02-3, for alternativent of this inspection of the inspection of the control of the cont	e inspection. on is contained in ot 57-10-80. Figure 1 to detern	6 to rib 7 at the area	s hidden by the
	Inspect (High nacelle supp See Doc. Doc The NDI mer (D6-37239). INTERVAL Note: AWL Inspect (High nacelle supp See Doc. Doc The NDI mer (D6-37239).	th Frequency Eddy Curre fort strut attachment fittin 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	nt) the lower wing. eck form 57-20 complish the intres are contained in the strike of	ing panel rail string -02-3, for alternative ent of this inspection ed in Part 6, Subject sk. See Section 9, For interval is 36000F NOTE ing panel rail string -02-3, for alternative ent of this inspection ed in Part 6, Subject sk. See Section 9, For string 1, 1000 sk. See Section 9, For alternative ed in Part 6, Subject sk. See Section 9, For alternative ed in Part 6, Subject sk. See Section 9, For alternative	e inspection. on is contained in ot 57-10-80. Figure 1 to detern FC. er, S-8, from rib 6 e inspection. on is contained in ot 57-10-80. Figure 1 to detern	the 737 Nondestructure threshold. Boei ALL 5 to rib 7 at the area the 737 Nondestructure threshold.	s hidden by the ctive Test Manual ALL s hidden by the ctive Test Manual Course Test Manua
	Inspect (High nacelle supp See Doc. Doc The NDI mer (D6-37239). INTERVAL Note: AWL Inspect (High nacelle supp See Doc. Doc The NDI mer (D6-37239).	h Frequency Eddy Curre fort strut attachment fittin 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sense recommended research for the first attachment fittin 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sense of the first length sense of	nt) the lower wing. eck form 57-20 complish the intres are contained in the strike of	ing panel rail string -02-3, for alternative ent of this inspection ed in Part 6, Subject sk. See Section 9, For interval is 36000F NOTE ing panel rail string -02-3, for alternative ent of this inspection ed in Part 6, Subject sk. See Section 9, For string 1, 1000 sk. See Section 9, For alternative ed in Part 6, Subject sk. See Section 9, For alternative ed in Part 6, Subject sk. See Section 9, For alternative	e inspection. on is contained in ot 57-10-80. Figure 1 to detern FC. er, S-8, from rib 6 e inspection. on is contained in ot 57-10-80. Figure 1 to detern	the 737 Nondestruction of the 737 Nondestruc	s hidden by the ctive Test Manual ALL s hidden by the ctive Test Manual Course Test Manua
57-615-01-02	Inspect (High nacelle supp See Doc. De The NDI me (D6-37239). INTERVAL N AWL Inspect (High nacelle supp See Doc. De The NDI me (D6-37239). INTERVAL N AWL Inspect (Ger See Doc. De S	h Frequency Eddy Curre fort strut attachment fittin 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sense recommended research thod(s) necessary to acc 57-05-02-250-855 h Frequency Eddy Curre fort strut attachment fittin 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur recommended research thod(s) necessary to acc The inspection procedur recommended research thod(s) necessary to acc The inspection procedur recommended research thousand the sense recommended research Visual) the lower with 526A001 - DTR, DTR che NOTE: Flight length sense NOTE:	nt) the lower way. g. eck form 57-20 complish the interes are containstive (FLS) tasepeat inspection 1.1 nt) the lower way. eck form 57-20 complish the interes are containstive (FLS) tasepeat inspection 1.1 ng panel rail steck form 57-20 sitive (FLS) tasepeat inspection	ing panel rail string -02-3, for alternativent of this inspecticed in Part 6, Subjects See Section 9, For interval is 36000F NOTE ing panel rail string -02-3, for alternativent of this inspecticed in Part 6, Subjects See Section 9, For interval is 36000F NOTE ringers, S-6 and S02-4, for alternativent	e inspection. on is contained in to 57-10-80. Figure 1 to detern FC. er, S-8, from rib 6 e inspection. on is contained in to 57-10-80. Figure 1 to detern FC. 8, from rib 19 to e inspection.	the 737 Nondestruction the Table 1 the Table 1 the Table 2 the Tab	s hidden by the ctive Test Manuary ALL shidden by the ctive Test Manuary ALL Iden areas.

Inspect (General Visual) the lower wing panel rail stringers, S-6 and S-8, from rib 19 to rib 25 at the non-hidden areas. See Doc. D626A001 - DTR, DTR check form 57-20-02-4, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-616-10-01	AWL	57-05-02-130-811	1.1	NOTE		ALL	ALL
	sealant.	asonic) lower wing pane 326A001 - DTR, DTR che			·	the flange of rib 22,	shims and
		thod(s) necessary to acc The inspection procedur				the 737 Nondestru	ctive Test Manu
	INTERVAL N	NOTE: Flight length sen recommended re		sk. See Section 9, F n interval is 24000F		nine threshold. Boei	ng
57-616-10-02	AWL	57-05-02-130-811	1.1	NOTE		ALL	ALL
	Inspect (Ultra	asonic) lower wing pane	I rail stringers,	S-6 and S-8, at the	areas hidden by	the flange of rib 22,	shims and
		26A001 - DTR, DTR che		*	•		
		thod(s) necessary to acc The inspection procedu				the 737 Nondestru	ctive Test Man
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-617-00-01	AWL	57-05-02-210-812	1.1	NOTE		ALL	ALL
	without the fa	0		-		o rib 10 at the non-h	idden areas
	See Doc. D6	26A001 - DTR, DTR che	eck form 57-20	-03-1, for alternativ	e inspection.		
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	sk. See Section 9, F n interval is 24000F	-	nine threshold. Boei	ng
57-617-00-02	AWL	57-05-02-210-812	1.1	NOTE		ALL	ALL
	without the fa	neral Visual) the lower wi airing. 326A001 - DTR, DTR che				o rib 10 at the non-h	idden areas
		NOTE: Flight length sen	sitive (FLS) tas		Figure 2 to detern	nine threshold. Boei	ng
57-617-01-01	AWL	57-05-02-250-819	1.1	NOTE		ALL	ALL
		r Frequency Eddy Current reas without the fairing.	nt) the lower wi	ng panel splice stri	ngers, S-5 and S	-9, from rib 1 to rib	10 at the

See Doc. D626A001 - DTR, DTR check form 57-20-03-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-55.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 5500FC.





				INTERVAL		APPLIC	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-617-01-02	AWL	57-05-02-250-819	1.1	NOTE		ALL	ALL
	non-hidden See Doc. Do The NDI me (D6-37239).	w Frequency Eddy Currer areas without the fairing. 626A001 - DTR, DTR cho withod(s) necessary to acc. The inspection procedur NOTE: Flight length sen recommended re	eck form 57-20 complish the int res are contain sitive (FLS) tas	-03-1, for alternative ent of this inspection ed in Part 6, Subject	e inspection. on is contained in ot 57-10-55. igure 2 to detern	the 737 Nondestru	ctive Test Manual
57-618-00-01	AWL	57-05-02-211-875	1.1	NOTE		ALL	ALL
37-010-00-01		th Frequency Eddy Curre		-	ngers S-5 and 9		
	non-hidden See Doc. Do	areas under the fairing. 626A001 - DTR, DTR che	eck form 57-20	-03-2, for alternativ	e inspection.		
		ethod(s) necessary to acc The inspection procedur	•	•		the 737 Nondestru	ctive rest Manual
	INTERVAL	NOTE: Flight length sen recommended re	` ,	k. See Section 9, F n interval is 15000F	•	nine threshold. Boei	ng
57-618-00-02	AWL	57-05-02-211-875	1.1	NOTE		ALL	ALL
	non-hidden See Doc. Do The NDI me	th Frequency Eddy Curre areas under the fairing. 626A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the int	-03-2, for alternative	e inspection. on is contained in		
	,	NOTE: Flight length sen	sitive (FLS) tas		igure 2 to detern	nine threshold. Boei	ng
57-618-01-01	AWL	57-05-02-250-917	1.1	NOTE		ALL	ALL
		h Frequency Eddy Curre	ent) the lower w	ing panel splice stri	ingers, S-5 and S	6-9, from rib 1 to rib	10 at the
	See Doc. Doc. The NDI me	areas under the fairing. 626A001-DTR, DTR cheethod(s) necessary to acc. The inspection procedure	complish the int	ent of this inspection	n is contained in	the 737 Nondestru	ctive Test Manual
	See Doc. Do The NDI me (D6-37239).	626A001-DTR, DTR check thod(s) necessary to according to the inspection procedure. NOTE: Flight length sen	complish the intres are containdstitute (FLS) tas	ent of this inspection ed in Part 6, Subject	on is contained in at 57-10-33. Figure 2 to detern		

Inspect (High Frequency Eddy Current) the lower wing panel splice stringers, S-5 and S-9, from rib 1 to rib 10 at the non-hidden areas under the fairing.

See Doc. D626A001-DTR, DTR check form 57-20-03-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-33.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 15000FC.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-619-00-01	AWL	57-05-02-130-813	1.1	NOTE		ALL	ALL
	and sealant. See Doc. D6 The NDI me	asonic) the lower wing page 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the int	-03-3, for alternative	e inspection. In is contained in		
	,	NOTE: Flight length sen	sitive (FLS) tas		igure 2 to detern	nine threshold. Boei	ng
57-619-00-02	AWL	57-05-02-130-813	1.1	NOTE		ALL	ALL
	and sealant. See Doc. Do The NDI me (D6-37239).	asonic) the lower wing page 226A001 - DTR, DTR che thod(s) necessary to accepte inspection procedure. Flight length sense recommended re	eck form 57-20 complish the interes are containantesistive (FLS) tas	-03-3, for alternative ent of this inspection ed in Part 4, Subjection	e inspection. in is contained in it 57-10-07. igure 2 to detern	the 737 Nondestruc	ctive Test Manua
57-619-01-01	AWL	57-05-02-250-821	1.1	NOTE		ALL	ALL
	hidden by se See Doc. Do The NDI me (D6-37239).	r Frequency Eddy Currer eal pans and sealant. 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen- recommended re	eck form 57-20 complish the interes are containa sitive (FLS) tas	-03-3, for alternative ent of this inspection ed in Part 6, Subject	e inspection. In is contained in the thick that the thick the thick the thick the thick the thick the thick the	the 737 Nondestruc	ctive Test Manua
57-619-01-02	AWL	57-05-02-250-821	1.1	NOTE		ALL	ALL
	hidden by se	r Frequency Eddy Currer	,			-9, from rib 5 to rib 8	3 at the areas
	The NDI me (D6-37239).	i26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen- recommended re	complish the int res are contain sitive (FLS) tas	ent of this inspection ed in Part 6, Subject	n is contained in at 57-10-55. Figure 2 to detern		
57-620-00-01	The NDI me (D6-37239).	thod(s) necessary to acc The inspection procedur NOTE: Flight length sen	complish the int res are contain sitive (FLS) tas	ent of this inspection ed in Part 6, Subject kk. See Section 9, F	n is contained in at 57-10-55. Figure 2 to detern		

Inspect (Detailed) the web and free flange of the lower wing panel splice stringers, S-5 and S-9, from rib 10 to rib 19 except at areas externally covered by rub strips.

See Doc. D626A001 - DTR, DTR check form 57-20-03-4, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 15000FC.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-620-00-02	AWL	57-05-02-211-817	1.1	NOTE		ALL	ALL
	at areas exte	ailed) the web and free f ernally covered by rub st 326A001 - DTR, DTR che	rips.			and S-9, from rib 10	to rib 19 excep
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 15000F	•	nine threshold. Boei	ng
57-620-01-01	AWL	57-05-02-250-825	1.1	NOTE		ALL	ALL
	areas extern See Doc. Do The NDI me	r Frequency Eddy Currer ally covered by rub strips 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	s. eck form 57-20 complish the int	-03-4, for alternativent of this inspection	e repeat inspection is contained in	on.	
		NOTE: Flight length sen recommended re OTE: Fairing removal re	epeat inspectio	sk. See Section 9, F n interval is 5500F0	-	nine threshold. Boei	ng
57-620-01-02	AWL	57-05-02-250-825	1.1	NOTE		ALL	ALL
	areas extern See Doc. Do The NDI me	r Frequency Eddy Currer ally covered by rub strips 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	s. eck form 57-20 complish the int	-03-4, for alternativent of this inspection	re repeat inspection is contained in	on.	
		NOTE: Flight length sen recommended re	epeat inspectio	sk. See Section 9, F n interval is 5500F0	•	nine threshold. Boei	ng
			· 				
57-621-00-01	areas extern	57-05-02-211-878 ailed) the web and free fi ally covered by rub strips 326A001 - DTR, DTR che	S.			ALL and S-9, from rib 10	ALL to rib 19 at the
		NOTE: Flight length sen	sitive (FLS) tas	•	igure 2 to detern	nine threshold. Boei	ng
57-621-00-02	AWL	57-05-02-211-878	1.1	NOTE		ALL	ALL
	. ,	ailed) the web and free f	•	ng lower panel splid	ce stringers, S-5	and S-9, from rib 10	to rib 19 at the

See Doc. D626A001 - DTR, DTR check form 57-20-03-5, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 15000 FC.



57-621-02-02

AWL



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-621-01-01	AWL	57-05-02-250-827	1.1	NOTE		ALL	ALL
	externally co See Doc. D6 The NDI met	n Frequency Eddy Curre vered by rub strips. 26A001 - DTR, DTR che rhod(s) necessary to acc The inspection procedur	eck form 57-20 complish the int	-03-5, for alternative ent of this inspection	e inspection. on is contained in	·	
	INTERVAL N	IOTE: Flight length sen recommended re	, ,	k. See Section 9, F n interval is 15000 I	•	nine threshold. Boeii	ng
	ACCESS NO	OTE: Remove minimal a	amount of seala	ant to facilitate direc	ction 3 HFEC in t	he lower stringer rac	lius.
57-621-01-02	AWL	57-05-02-250-827	1.1	NOTE		ALL	ALL
	and a second of the control	والمرابع والمرابع المرابع والمرابع					
	See Doc. D6 The NDI met (D6-37239).	vered by rub strips. 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re	complish the intres are contained sitive (FLS) tas	ent of this inspection ed in Part 6, Subject k. See Section 9, F	on is contained in at 57-10-33. Figure 2 to detern		
	See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen	complish the intres are containe sitive (FLS) tase epeat inspection	ent of this inspecticed in Part 6, Subject k. See Section 9, F n interval is 15000 I	on is contained in at 57-10-33. Figure 2 to detern FC.	nine threshold. Boei	ng
57-621-02-01	See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re	complish the intres are containe sitive (FLS) tase epeat inspection	ent of this inspecticed in Part 6, Subject k. See Section 9, F n interval is 15000 I	on is contained in at 57-10-33. Figure 2 to detern FC.	nine threshold. Boei	ng
57-621-02-01	See Doc. D6 The NDI met (D6-37239).* INTERVAL N ACCESS NC AWL Inspect (High externally co See Doc. D6 The NDI met	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re DTE: Remove minimal a	complish the interest are contained as are contained as a containe	ent of this inspecticed in Part 6, Subjection Part 6, Subjection 9, Fin interval is 15000 I and to facilitate direction NOTE wer panel splice striction of this inspection in Part 6, Subjection 1, S	on is contained in the tot 57-10-33. Figure 2 to determine the tot of the to	nine threshold. Boein he lower stringer rac ALL 6-9, from rib 10 to rib	ng lius. ALL o 19, at the area
57-621-02-01	ACCESS NC AWL Inspect (High externally cordinated See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che chod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re OTE: Remove minimal a 57-05-02-250-828 Trequency Eddy Curre vered by rub strips. 26A001 - DTR, DTR che chod(s) necessary to acc The inspection procedur IOTE: Flight length sen	complish the interes are contained sitive (FLS) tas expeat inspection amount of seals amount of seals are contained to the same contained are contained to the same contained to	ent of this inspecticed in Part 6, Subjection Part 6, Subjection 9, Final interval is 15000 In ant to facilitate direction NOTE Wer panel splice striction of this inspectice in Part 6, Subjection i	on is contained in the tot 57-10-33. Figure 2 to determine the tot of the to	he lower stringer rac ALL S-9, from rib 10 to rib	ALL o 19, at the area

Inspect (High Frequency Eddy Current) the wing lower panel splice stringers, S-5 and S-9, from rib 10 to rib 19, at the areas externally covered by rub strips.

NOTE

See Doc. D626A001 - DTR, DTR check form 57-20-03-5, for alternative inspection.

1.1

57-05-02-250-828

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-33.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 15000 FC.

ACCESS NOTE: Remove minimal amount of sealant to facilitate direction 4 HFEC at all fasteners.

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ALL

ALL



57-621-11-02

AWL



737-600/700/800/900 TASK CARDS

				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-621-10-01	AWL	57-05-02-250-831	1.1	NOTE		ALL	ALL
	externally co See Doc. D6 The NDI met	n Frequency Eddy Curre vered by fairing and rub 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	strips. eck form 57-20 complish the int	-03-6, for alternative	e inspection. on is contained in	,	
	INTERVAL N	IOTE: Flight length sens	` ,	sk. See Section 9, F n interval is 15000F	•	nine threshold. Boei	ng
	ACCESS NO	OTE: Remove minimal a	mount of seala	ant to facilitate direc	tion 3 HFEC in t	he lower stringer rac	lius.
57-621-10-02	AWL	57-05-02-250-831	1.1	NOTE		ALL	ALL
	externally co	vered by fairing and rub	STrips.				
	See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen	eck form 57-20 complish the int res are containd sitive (FLS) tas	ent of this inspection ed in Part 6, Subject	n is contained in at 57-10-33.		
	See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen	eck form 57-20 complish the intres are containe sitive (FLS) tas epeat inspection	ent of this inspection ed in Part 6, Subject kk. See Section 9, F in interval is 15000F	n is contained in at 57-10-33. Gure 2 to detern	nine threshold. Boeiı	ng
57-621-11-01	See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sens recommended re	eck form 57-20 complish the intres are containe sitive (FLS) tas epeat inspection	ent of this inspection ed in Part 6, Subject kk. See Section 9, F in interval is 15000F	n is contained in at 57-10-33. Gure 2 to detern	nine threshold. Boeiı	ng
57-621-11-01	ACCESS NO AWL Inspect (High externally co See Doc. D6 The NDI met	26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen- recommended re DTE: Remove minimal a	eck form 57-20 complish the integration of the inte	ent of this inspection and in Part 6, Subject in Part 6, Subject is k. See Section 9, For interval is 15000F and to facilitate direct ing panel splice strictly and the subject in Part 1 in Part 2 in Part 2 in Part 2 in Part 3	on is contained in at 57-10-33. Figure 2 to determine a HFEC in the second sec	nine threshold. Boein he lower stringer rac ALL 6-9, from rib 1 to rib	dius. ALL 10 at the areas
57-621-11-01	ACCESS NC AWL Inspect (High externally co See Doc. D6 The NDI met (D6-37239).	26A001 - DTR, DTR che chod(s) necessary to acc The inspection procedur IOTE: Flight length sen- recommended re OTE: Remove minimal a 57-05-02-250-924 Trequency Eddy Curre vered by fairing and rub 26A001 - DTR, DTR che chod(s) necessary to acc The inspection procedur IOTE: Flight length sen-	eck form 57-20 complish the intres are contained sitive (FLS) tas expeat inspection amount of seals 1.1 nt) the lower w strips. eck form 57-20 complish the intres are contained sitive (FLS) tas	ent of this inspection and in Part 6, Subject of the Part 6, Subject	in is contained in it 57-10-33. Figure 2 to determ C. Ition 3 HFEC in the imagers, S-5 and See inspection. In is contained in it 57-10-33. Figure 2 to determ is contained in it 57-10-33.	ALL S-9, from rib 1 to rib	ALL 10 at the areas

Inspect (High Frequency Eddy Current) the lower wing panel splice stringers, S-5 and S-9, from rib 1 to rib 10 at the areas externally covered by fairing and rub strips.

NOTE

See Doc. D626A001 - DTR, DTR check form 57-20-03-6, for alternative inspection.

1.1

57-05-02-250-924

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-33.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 15000FC.

ACCESS NOTE: Remove minimal amount of sealant to facilitate direction 4 HFEC at all fasteners.

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ALL

ALL





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-621-20-01	AWL	57-05-02-250-833	1.1	NOTE		ALL	ALL
	17 to rib 18; See Doc. D6 The NDI mei (D6-37239).	Frequency Eddy Currer S-5 from rib 18 to rib 19 S26A001 - DTR, DTR che thod(s) necessary to accor The inspection procedur NOTE: Flight length sens recommended re	and upper string and upper string the complish the interest are contained string (FLS) tas	nger, S-14 from rib -03/15, for alternati ent of this inspection ed in Part 6, Subject	19 to rib 20. ve inspection. on is contained in ot 57-10-60. Figure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-621-20-02	AWL	57-05-02-250-833	1.1	NOTE		ALL	ALL
	17 to rib 18; See Doc. D6 The NDI mei (D6-37239).	w Frequency Eddy Currer S-5 from rib 18 to rib 19 626A001 - DTR, DTR che thod(s) necessary to acco The inspection procedur NOTE: Flight length sens recommended re	and upper string the ck form 57-20 complish the interest are contained strikes (FLS) tas	nger, S-14 from rib -03/15, for alternati ent of this inspection ed in Part 6, Subject	19 to rib 20. ve inspection. on is contained in ot 57-10-60. Figure 2 to deterr	the 737 Nondestruc	ctive Test Manu
57-622-00-01	AWL	57-05-02-250-835	1.1	NOTE		ALL	ALL
		r Frequency Eddy Curren the hidden areas from rib				eas from rib 1 to rib 5	5, and rib 19 to
	Rib 22, and the See Doc. Doc. The NDI met (D6-37239).	the hidden areas from rib s26A001 - DTR, DTR che thod(s) necessary to according The inspection procedur NOTE: Flight length sens	o 1 to rib 6, and eck form 57-20 omplish the int es are contain sitive (FLS) tas	I from rib 19 to rib 2 -04-1, for alternativ ent of this inspection ed in Part 6, Subject	22. e inspection. on is contained in ot 57-10-54. Figure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-622-00-02	Rib 22, and the See Doc. Doc. The NDI met (D6-37239).	the hidden areas from rib s26A001 - DTR, DTR che thod(s) necessary to according The inspection procedur NOTE: Flight length sens	o 1 to rib 6, and eck form 57-20 omplish the int es are contain sitive (FLS) tas	I from rib 19 to rib 2 -04-1, for alternativ ent of this inspectic ed in Part 6, Subjectives. sk. See Section 9, F	22. e inspection. on is contained in ot 57-10-54. Figure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-622-00-02	AWL Inspect (Low Rib 22, and see Doc. D6 The NDI mei (D6-37239). AWL Inspect (Low Rib 22, and see Doc. D6 The NDI mei (D6-37239).	the hidden areas from rib 526A001 - DTR, DTR che thod(s) necessary to acc. The inspection procedur. NOTE: Flight length sens recommended re 57-05-02-250-835 Frequency Eddy Current the hidden areas from rib 526A001 - DTR, DTR che thod(s) necessary to acc. The inspection procedur. NOTE: Flight length sens	o 1 to rib 6, and ock form 57-20 complish the interest are contained sitive (FLS) tas peat inspection 1.1 In the front span of the front 57-20 complish the interest are contained sitive (FLS) tas	I from rib 19 to rib 2 -04-1, for alternativent of this inspection of the inspection of the results of the resu	e inspection. on is contained in ot 57-10-54. Figure 2 to deterred. e non-hidden are 22. e inspection. on is contained in ot 57-10-54. Figure 2 to deterred.	ALL eas from rib 1 to rib 5	ALL 5, and rib 19 to
57-622-00-02 57-623-00-01	AWL Inspect (Low Rib 22, and see Doc. D6 The NDI mei (D6-37239). AWL Inspect (Low Rib 22, and see Doc. D6 The NDI mei (D6-37239).	the hidden areas from rib 526A001 - DTR, DTR che thod(s) necessary to acc. The inspection procedur. NOTE: Flight length sens recommended re 57-05-02-250-835 Frequency Eddy Current the hidden areas from rib 526A001 - DTR, DTR che thod(s) necessary to acc. The inspection procedur. NOTE: Flight length sens	o 1 to rib 6, and ock form 57-20 complish the interest are contained sitive (FLS) tas peat inspection 1.1 In the front span of the front 57-20 complish the interest are contained sitive (FLS) tas	I from rib 19 to rib 2 -04-1, for alternativent of this inspecticed in Part 6, Subjection 9, For interval is 12000F NOTE Ar lower chord at the 1 from rib 19 to rib 2 -04-1, for alternativent of this inspecticed in Part 6, Subjecticed in Part 6, Subjectices.	e inspection. on is contained in ot 57-10-54. Figure 2 to deterred. e non-hidden are 22. e inspection. on is contained in ot 57-10-54. Figure 2 to deterred.	ALL eas from rib 1 to rib 5	ALL 5, and rib 19 to
	AWL Inspect (Detasted Detasted	the hidden areas from rib 526A001 - DTR, DTR che thod(s) necessary to accommended re NOTE: Flight length sens recommended re 57-05-02-250-835 Frequency Eddy Current the hidden areas from rib 526A001 - DTR, DTR che thod(s) necessary to accommended re NOTE: Flight length sens recommended re 57-05-02-211-819 ailed) the front spar lowe 526A001 - DTR, DTR che 526A001 - DTR, DTR che S26A001 - DTR, DTR che	of 1 to rib 6, and ock form 57-20 complish the intest are contained sitive (FLS) tast peat inspection 1.1 Into the front space of 1 to rib 6, and ock form 57-20 complish the intest are contained sitive (FLS) tast peat inspection 1.1 In the front space of 1 to rib 6, and ock form 57-20 complish the intest are contained sitive (FLS) tast peat inspection 1.1 In chord at the reck form 57-20 sitive (FLS) tast specifically tast sitive (FLS) tast sitive (FLS) tast specifically the form 57-20 sitive (FLS) tast specifically the siting tast specifically	If from rib 19 to rib 2 -04-1, for alternativent of this inspecticed in Part 6, Subjective Risk. See Section 9, For interval is 12000F NOTE ar lower chord at the form rib 19 to rib 2 -04-1, for alternativent of this inspecticed in Part 6, Subjective Risk. See Section 9, For interval is 12000F NOTE NOTE non-hidden areas frequency for alternativent of this inspection for alternativent of this inspection for the part of this inspection for the part of this inspection for the part of the part o	e inspection. on is contained in ot 57-10-54. Figure 2 to deterred. e non-hidden are 22. e inspection. on is contained in ot 57-10-54. Figure 2 to deterred.	ALL tas from rib 1 to rib 5 the 737 Nondestructure ALL ALL ALL ALL ALL ALL ALL ALL	ALL 5, and rib 19 to ctive Test Manu

See Doc. D626A001 - DTR, DTR check form 57-20-04-2, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-624-00-01	AWL	57-05-02-130-815	1.1	NOTE		ALL	ALL
	22 to 25. See Doc. D6 The NDI met	asonic) the front spar low 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 omplish the int	-04-3, for alternative	e inspection. on is contained in	· ·	
	INTERVAL N	NOTE: Flight length sens recommended re		k. See Section 9, F n interval is 24000F		nine threshold. Boei	ng
57-624-00-02	AWL	57-05-02-130-815	1.1	NOTE		ALL	ALL
	22 to 25. See Doc. D6 The NDI met (D6-37239).	asonic) the front spar low 326A001 - DTR, DTR che thod(s) necessary to accor- The inspection procedur NOTE: Flight length sens recommended re	eck form 57-20 omplish the int es are contain sitive (FLS) tas	-04-3, for alternative ent of this inspection ed in Part 4, Subject	e inspection. on is contained in ot 57-10-09. Gigure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-625-00-01	AWL	57-05-02-250-837	1.1	NOTE		ALL	ALL
						as from rib 7 to rib 1	19, and from ric
	22 to rib 25, See Doc. D6 The NDI met (D6-37239).	and the hidden areas fro 226A001 - DTR, DTR che thod(s) necessary to according The inspection procedur NOTE: Flight length sens	om rib 6 to rib 1 eck form 57-20 omplish the intes are contained sitive (FLS) tas	9, and from rib 22 t -04-4, for alternative ent of this inspection ed in Part 6, Subjection	o rib 25. e inspection. on is contained in ot 57-10-54. figure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-625-00-02	22 to rib 25, See Doc. D6 The NDI met (D6-37239).	and the hidden areas fro 226A001 - DTR, DTR che thod(s) necessary to according The inspection procedur NOTE: Flight length sens	om rib 6 to rib 1 eck form 57-20 omplish the intes are contained sitive (FLS) tas	9, and from rib 22 t -04-4, for alternative ent of this inspection ed in Part 6, Subjection k. See Section 9, F	o rib 25. e inspection. on is contained in ot 57-10-54. figure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-625-00-02	AWL Inspect (Low 22 to rib 25, See Doc. D6 The NDI met (D6-37239). AWL Inspect (Low 22 to rib 25, See Doc. D6 The NDI met (D6-37239).	and the hidden areas fro 26A001 - DTR, DTR che thod(s) necessary to accommended results. Flight length sense recommended results. Frequency Eddy Current and the hidden areas fro 26A001 - DTR, DTR che thod(s) necessary to accommended results.	m rib 6 to rib 1 eck form 57-20 complish the int es are contain- sitive (FLS) tas peat inspection 1.1 ht) the front spa eck form 57-20 complish the int es are contain- sitive (FLS) tas	9, and from rib 22 to -04-4, for alternative ent of this inspection ed in Part 6, Subject like. See Section 9, For interval is 24000F NOTE ar lower chord at the 9, and from rib 22 to -04-4, for alternative ent of this inspection ed in Part 6, Subject	o rib 25. e inspection. on is contained in at 57-10-54. Figure 2 to determ C. e non-hidden are o rib 25. e inspection. on is contained in at 57-10-54. Figure 2 to determ	the 737 Nondestruction threshold. Boein ALL as from rib 7 to rib 1 the 737 Nondestruction t	ALL 19, and from rib
57-625-00-02 57-626-00-01	AWL Inspect (Low 22 to rib 25, See Doc. D6 The NDI met (D6-37239). AWL Inspect (Low 22 to rib 25, See Doc. D6 The NDI met (D6-37239).	and the hidden areas fro 26A001 - DTR, DTR che thod(s) necessary to accommended results. Flight length sense recommended results. Frequency Eddy Current and the hidden areas fro 26A001 - DTR, DTR che thod(s) necessary to accommended results.	m rib 6 to rib 1 eck form 57-20 complish the int es are contain- sitive (FLS) tas peat inspection 1.1 ht) the front spa eck form 57-20 complish the int es are contain- sitive (FLS) tas	9, and from rib 22 to -04-4, for alternative ent of this inspectioned in Part 6, Subject Rk. See Section 9, For interval is 24000F NOTE ar lower chord at the 9, and from rib 22 to -04-4, for alternative ent of this inspectioned in Part 6, Subject Rk. See Section 9, F	o rib 25. e inspection. on is contained in at 57-10-54. Figure 2 to determ C. e non-hidden are o rib 25. e inspection. on is contained in at 57-10-54. Figure 2 to determ	the 737 Nondestruction threshold. Boein ALL as from rib 7 to rib 1 the 737 Nondestruction t	ALL 19, and from rib
	AWL Inspect (Deta AWL Inspect (D6-37239). INTERVAL N AWL Inspect (Low 22 to rib 25, See Doc. D6 The NDI met (D6-37239). INTERVAL N AWL Inspect (Deta See Doc. D6	and the hidden areas fro (26A001 - DTR, DTR che (26A001 - DTR, DTR che (26A001 - DTR), DTR che (26A001 - DTR), DTR che (26A001 - DTR), DTR che (26A001 - DTR, DTR), DTR), and (26A001 - DTR), DTR), DTR che (26A001 - DTR),	m rib 6 to rib 1 eck form 57-20 omplish the int es are contain sitive (FLS) tas peat inspection 1.1 nt) the front spa m rib 6 to rib 1 eck form 57-20 omplish the int es are contain sitive (FLS) tas peat inspection 1.1 r chord at the reck form 57-20 sitive (FLS) tas peative (FLS) tas	9, and from rib 22 to 104-4, for alternative ent of this inspectioned in Part 6, Subject of the Section 9, For interval is 24000F NOTE ar lower chord at the 9, and from rib 22 to 104-4, for alternative ent of this inspectioned in Part 6, Subject in Part 6, Subject in Part 104-104-105, in interval is 24000F NOTE non-hidden areas from rib 22 to 104-5, for alternative ent of this inspectioned in Part 6, Subject in Part 6, Subject in Part 104-104-104-104-104-104-104-104-104-104-	or rib 25. e inspection. on is contained in it 57-10-54. Figure 2 to determ or rib 25. e inspection. on is contained in it 57-10-54. Figure 2 to determ or rib 25. e inspection. on is contained in it 57-10-54. Figure 2 to determ or rib 1 to rib 19. e inspection. Figure 2 to determ or rib 1 to rib 19. e inspection.	the 737 Nondestruction the threshold. Boein ALL as from rib 7 to rib 1 the 737 Nondestruction threshold. Boein ALL	ALL 19, and from rib

See Doc. D626A001 - DTR, DTR check form 57-20-04-5, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.





ASK CARD NO. 57-627-00-01	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIDDL AND	ENGINE			
-				THIRESHOLD	REFEAT	AIRPLANE	ENGINE			
	AWL	57-05-02-250-868	1.1	NOTE		ALL	ALL			
	Inspect (High Frequency Eddy Current) the front spar lower chord at the non-hidden areas from rib 19 to rib 22. See Doc. D626A001 - DTR, DTR check form 57-20-04-6, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-44. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.									
57-627-00-02	AWL	57-05-02-250-868	1.1	NOTE		ALL	ALL			
-		n Frequency Eddy Curre 26A001 - DTR, DTR che	, .			eas from rib 19 to rib	22.			
		thod(s) necessary to acc The inspection procedur				the 737 Nondestruc	ctive Test Manu			
	INTERVAL N	NOTE: Flight length sense recommended re	` '	k. See Section 9, F n interval is 24000F	•	nine threshold. Boein	ng			
57-627-01-01	AWL	57-05-02-250-839	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	r Frequency Eddy Currer (26A001 - DTR, DTR che (thod(s) necessary to acc The inspection procedur (IOTE: Flight length sen- recommended re	eck form 57-20 omplish the int es are containe sitive (FLS) tas	-04-6, for alternative ent of this inspectio ed in Part 6, Subjec	e inspection. on is contained in ot 57-10-54. Figure 2 to determ	the 737 Nondestruc	ctive Test Manu			
57-627-01-02	AWL	57-05-02-250-839	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	rFrequency Eddy Currer (26A001 - DTR, DTR che (thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	eck form 57-20 omplish the int es are containe sitive (FLS) tas	-04-6, for alternative ent of this inspectio ed in Part 6, Subjec	e inspection. on is contained in ot 57-10-54. Figure 2 to determ	the 737 Nondestruc	ctive Test Manu			
57-628-00-01	AWL	57-05-02-130-817	1.1	NOTE		ALL	ALL			

See Doc. D626A001 - DTR, DTR check form 57-20-04-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 57-10-09.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-628-00-02	AWL	57-05-02-130-817	1.1	NOTE		ALL	ALL
	See Doc. D6 The NDI met (D6-37239).	asonic) the front spar low 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	eck form 57-20 complish the intes are contain sitive (FLS) tas	-04-7, for alternative ent of this inspectio ed in Part 4, Subjec	e inspection. in is contained in it 57-10-09. igure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-628-01-01	AWL	57-05-02-250-925	1.1	NOTE		ALL	ALL
	19 to rib 22. See Doc. D6 The NDI met (D6-37239).	r Frequency Eddy Currer (26A001 - DTR, DTR che (thod(s) necessary to accommended re (The inspection procedure (NOTE: Flight length sensing recommended re	eck form 57-20 complish the intest are contain sitive (FLS) tas	-04-7, for alternative ent of this inspectio ed in Part 6, Subjec	e inspection. In is contained in the total the	the 737 Nondestruc	ctive Test Manu
57-628-01-02	AWL	57-05-02-250-925	1.1	NOTE		ALL	ALL
	The NDI met	:26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	omplish the int	ent of this inspectio	n is contained in	the 737 Nondestruc	ctive Test Manu
	INTERVAL N	NOTE: Flight length sens recommended re	, ,		igure 2 to detern	nine threshold. Boei	ng
57-628-10-01	AWL	• •	, ,	k. See Section 9, F	igure 2 to detern	nine threshold. Boein	ng ALL
57-628-10-01	AWL Inspect (Gen See Doc. D6	recommended re 57-05-02-210-814 peral Visual) the front spa (26A001 - DTR, DTR che	1.1 Ir lower chord eck form 57-20 sitive (FLS) tas	NOTE at the non-hidden a	reas from rib 22 e inspection.	ALL to rib 25.	ALL
57-628-10-01 57-628-10-02	AWL Inspect (Gen See Doc. D6	recommended re 57-05-02-210-814 peral Visual) the front spa (26A001 - DTR, DTR che	1.1 Ir lower chord eck form 57-20 sitive (FLS) tas	NOTE at the non-hidden a -04-8, for alternative k. See Section 9, F	reas from rib 22 e inspection.	ALL to rib 25.	ALL
	AWL Inspect (Gen See Doc. D6 INTERVAL N AWL Inspect (Gen See Doc. D6	recommended re 57-05-02-210-814 peral Visual) the front spate (26A001 - DTR, DTR check (26A001 - DTR) the commended re 57-05-02-210-814 peral Visual) the front spate (26A001 - DTR, DTR check (26A001 - DTR, DTR) check (26A001 - DTR)	1.1 Ir lower chord eck form 57-20 sitive (FLS) tas peat inspectio 1.1 Ir lower chord to ck form 57-20 sitive (FLS) tas peat inspection for the ck form 57-20 sitive (FLS) tas	NOTE at the non-hidden a -04-8, for alternative	reas from rib 22 e inspection. reas from rib 22 e inspection. reas from rib 22 e inspection.	ALL to rib 25. nine threshold. Boein ALL to rib 25.	ALL ALL
	AWL Inspect (Gen See Doc. D6 INTERVAL N AWL Inspect (Gen See Doc. D6	recommended re 57-05-02-210-814 peral Visual) the front spate (26A001 - DTR, DTR check (26A001 - DTR) the commended re 57-05-02-210-814 peral Visual) the front spate (26A001 - DTR, DTR check (26A001 - DTR, DTR) check (26A001 - DTR)	1.1 Ir lower chord eck form 57-20 sitive (FLS) tas peat inspectio 1.1 Ir lower chord to ck form 57-20 sitive (FLS) tas peat inspection for the ck form 57-20 sitive (FLS) tas	NOTE at the non-hidden a -04-8, for alternative at the non-hidden a -04-8, for alternative at the non-hidden a -04-8, for alternative at the see Section 9, For alternative at the section 2 and 2 a	reas from rib 22 e inspection. reas from rib 22 e inspection. reas from rib 22 e inspection.	ALL to rib 25. nine threshold. Boein ALL to rib 25.	ALL ALL

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ACCESS NOTE: Nacelle fairing should be removed for inspection.





				INTERVAL		APPLIC	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-629-00-02	AWL	57-05-02-130-819	1.1	NOTE		ALL	ALL
	See Doc. Do The NDI me	asonic) all fasteners, out 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the inf	-04-9, for alternativ	e inspection. on is contained ir		ctive Test Manu
	INTERVAL N	NOTE: Flight length sense recommended re	. ,	sk. See Section 9, F n interval is 7000F0	•	nine threshold. Boei	ng
	ACCESS NO	OTE: Nacelle fairing sho	uld be remove	d for inspection.			
57-630-00-01	AWL	57-05-02-250-841	1.1	NOTE		ALL	ALL
	See Doc. Do The NDI me	n Frequency Eddy Curre 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the inf	-04-9, for alternativ	e inspection. on is contained ir		
	,	NOTE: Flight length sen	sitive (FLS) tas	-	Figure 2 to deterr	nine threshold. Boei	ing
	ACCESS NO	OTE: Nacelle fairing sho	ould be remove	d for inspection.			
57-630-00-02	AWL	57-05-02-250-841	1.1	NOTE		ALL	ALL
	See Doc. Do The NDI me	n Frequency Eddy Curre 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the inf	-04-9, for alternativ	e inspection. on is contained ir		
	INTERVAL N	NOTE: Flight length sense	, ,	sk. See Section 9, F n interval is 7000F0	-	mine threshold. Boei	ng
	ACCESS NO	OTE: Nacelle fairing sho	ould be remove	d for inspection.			
57-631-00-01	AWL	57-05-02-130-820	1.1	NOTE		ALL	ALL
	See Doc. Do The NDI me	asonic) all fasteners, inno 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the inf	-04-9, for alternativ	e inspection. on is contained ir		ctive Test Manu
	INTERVAL N	NOTE: Flight length sens		sk. See Section 9, F n interval is 18000F		nine threshold. Boei	ng
	ACCESS NO	OTE: Nacelle fairing sho	ould be remove	d for inspection.			
57-631-00-02	AWL	57-05-02-130-820	1.1	NOTE		ALL	ALL
	See Doc. Do	asonic) all fasteners, inne 326A001 - DTR, DTR che thod(s) necessary to acc	eck form 57-20 complish the inf	-04-9, for alternativ	e inspection. on is contained ir		ctive Test Manu
	,	The inspection procedur		•			

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing

recommended repeat inspection interval is 18000FC.

ACCESS NOTE: Nacelle fairing should be removed for inspection.





				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
57-632-00-01	AWL	57-05-02-250-842	1.1	NOTE		ALL	ALL		
	Inspect (High Frequency Eddy Current) all fasteners, inner location, common to the R7/R8 nacelle fitting attachment. See Doc. D626A001 - DTR, DTR check form 57-20-04-9, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manu (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-51. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 18000FC. ACCESS NOTE: Nacelle fairing should be removed for inspection.								
57-632-00-02	AWL	57-05-02-250-842	1.1	NOTE		ALL	ALL		
	See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen	eck form 57-20 complish the intres are contain sitive (FLS) tas	-04-9, for alternativent of this inspection of the inspection of t	re inspection. on is contained in ct 57-10-51. Figure 2 to determ	the 737 Nondestruc	ctive Test Man		
57-632-10-01	ACCESS NO	OTE: Nacelle fairing sho				ALL	ALL		
0.002.000	Inspect (Low See Doc. D6 The NDI met	r Frequency Eddy Currer 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	nt) the front spa eck form 57-20 complish the int	ar lower chord from -04-10, for alternati ent of this inspection	ive inspection. on is contained in				
	INTERVAL N	NOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 24000F	•	ine threshold. Boei	ng		
57-632-10-02	AWL	57-05-02-250-872	1.1	NOTE		ALL	ALL		
	See Doc. D6 The NDI met (D6-37239).	Frequency Eddy Currer (26A001 - DTR, DTR che (thod(s) necessary to acc The inspection procedur (NOTE: Flight length sen recommended re	eck form 57-20 complish the intres are contain sitive (FLS) tas	-04-10, for alternati ent of this inspection ed in Part 6, Subject	ive inspection. on is contained in ct 57-10-54. Figure 2 to determ				
57-633-00-01	AWL	57-05-02-250-919	1.1	NOTE		ALL	ALL		
	to rib 19, and	Frequency Eddy Currer	ord from rib 19	to rib 20.		·	chord from rib		

See Doc. D626A001 - DTR, DTR check form 57-20-04 / 05 / 16, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-60.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 2750FC.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-633-00-02	AWL	57-05-02-250-919	1.1	NOTE		ALL	ALL
	to rib 19, and See Doc. D6 The NDI mei (D6-37239).	r Frequency Eddy Currer d the front spar upper che 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	ord from rib 19 eck form 57-20 omplish the intes are contain sitive (FLS) tas	to rib 2004 / 05 / 16, for alto tent of this inspection ed in Part 6, Subjection	ernative inspection is contained in the street of the stre	on. the 737 Nondestru	ctive Test Manua
57-633-01-01	AWL	57-05-02-130-821	1.1	NOTE		ALL	ALL
		asonic) the front spar low			rear spar lower		
	See Doc. D6 The NDI met (D6-37239).	per chord from rib 19 to 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur VOTE: Flight length sens	eck form 57-20 omplish the in es are contain	ent of this inspection ed in Part 4, Subject	n is contained in t 57-10-14.	the 737 Nondestruc	
		0 0	` ,	n interval is 24000F	0		3
57-633-01-02	AWL	57-05-02-130-821	1.1	NOTE		ALL	ALL
	front spar up See Doc. D6 The NDI mei (D6-37239).	asonic) the front spar low oper chord from rib 19 to 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	rib 20. eck form 57-20 omplish the intes are contain sitive (FLS) tas	-04 / 05 / 16, for alto ent of this inspection ed in Part 4, Subject	ernative inspection is contained in the total transfer to the transfer transfer to the transfer transfer to the transfer transfe	on. the 737 Nondestru	ctive Test Manua
57-634-00-01	AWL	57-05-02-210-816	1.1	NOTE		ALL	ALL
	Inspect (Ger	neral Visual) the rear spa	r lower chord a	at the non-hidden ar	eas from rib 1 to	rib 7.	
	See Doc. D6	26A001 - DTR, DTR che	eck form 57-20	-05-1, for alternative	e inspection.		
	INTERVAL N	NOTE: Flight length sens recommended re	` '	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-634-00-02	AWL	57-05-02-210-816	1.1	NOTE		ALL	ALL
		neral Visual) the rear spa 626A001 - DTR, DTR che				rib 7.	
	INTERVAL N	NOTE: Flight length sense recommended re		sk. See Section 9, F n interval is 24000F		nine threshold. Boei	ng
57-635-00-01	AWL	57-05-02-210-818	1.1	NOTE		ALL	ALL
	. ,	neral Visual) rear spar lov 626A001 - DTR, DTR che				14.	

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-635-00-02	AWL	57-05-02-210-818	1.1	NOTE		ALL	ALL
		neral Visual) rear spar lov				14.	
		26A001 - DTR, DTR che			•		
	INTERVAL N	NOTE: Flight length sens	` ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
			pour moposito		·		
57-636-00-01	AWL	57-05-02-130-823	1.1	NOTE		ALL	ALL
		asonic) rear spar lower c 26A001 - DTR, DTR che		•		fitting from rib 1 to r	ib 25.
		thod(s) necessary to acc The inspection procedur		•		the 737 Nondestruc	ctive Test Man
	INTERVAL N	NOTE: Flight length sens recommended re	, ,	sk. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng
57-636-00-02	AWL	57-05-02-130-823	1.1	NOTE		ALL	ALL
		asonic) rear spar lower c 26A001 - DTR, DTR che		•		fitting from rib 1 to r	ib 25.
	The NDI met	thod(s) necessary to acc The inspection procedur	omplish the in	tent of this inspection	on is contained in	the 737 Nondestruc	ctive Test Man
	INTERVAL N	NOTE: Flight length sens recommended re	, ,	sk. See Section 9, F n interval is 18000F	•	nine threshold. Boei	ng
57-637-00-01	AWL	57-05-02-211-823	1.1	NOTE		ALL	ALL
		ailed) the rear spar lower 26A001 - DTR, DTR che					
	INTERVAL N	NOTE: Flight length sens recommended re	` ,	sk. See Section 9, F n interval is 36000F	•	nine threshold. Boei	ng
57-637-00-02	AWL	57-05-02-211-823	1.1	NOTE		ALL	ALL
		ailed) the rear spar lower 26A001 - DTR, DTR che					
	INTERVAL N	NOTE: Flight length sens recommended re	, ,	sk. See Section 9, F n interval is 36000F	•	nine threshold. Boei	ng
57-638-00-01	AWL	57-05-02-250-845	1.1	NOTE		ALL	ALL
		Fraguency Eddy Currer	1) 11		1:11 1 00		

Inspect (Low Frequency Eddy Current) the rear spar lower chord areas hidden by a stiffener from rib 1 to rib 22 and the areas hidden by rib post or fitting from rib 1 to rib 25.

See Doc. D626A001 - DTR, DTR check form 57-20-05-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-58.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 18000FC.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-638-00-02	AWL	57-05-02-250-845	1.1	NOTE		ALL	ALL			
	Inspect (Low Frequency Eddy Current) the rear spar lower chord areas hidden by a stiffener from rib 1 to rib 22 and the areas hidden by rib post or fitting from rib 1 to rib 25. See Doc. D626A001 - DTR, DTR check form 57-20-05-5, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-58. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing									
		recommended re	peat inspection	n interval is 18000F	-C.					
57-639-00-01	AWL	57-05-02-211-825	1.1	NOTE		ALL	ALL			
		ailed) the rear spar lower 26A001 - DTR, DTR che				2.				
	INTERVAL N	IOTE: Flight length sens recommended re	, ,	sk. See Section 9, F n interval is 24000F	-	nine threshold. Boei	ng			
57-639-00-02	AWL	57-05-02-211-825	1.1	NOTE		ALL	ALL			
	See Doc. D6	ailed) the rear spar lower 26A001 - DTR, DTR che IOTE: Flight length sens recommended re	eck form 57-20 sitive (FLS) tas	-05-6, for alternativ	e inspection. Figure 2 to determ		ng			
57-640-00-01	AWL	57-05-02-211-827	1.1	NOTE		ALL	ALL			
		ailed) the rear spar lower 26A001 - DTR, DTR che				2.				
	INTERVAL N	IOTE: Flight length sens recommended re	` ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng			
57-640-00-02	AWL	57-05-02-211-827	1.1	NOTE		ALL	ALL			
	. ,	ailed) the rear spar lower 26A001 - DTR, DTR che				2.				
	INTERVAL N	IOTE: Flight length sens recommended re	, ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng			
57-641-00-01	AWL	57-05-02-250-874	1.1	NOTE		ALL	ALL			
	Inspect (Low	Frequency Eddy Currer	nt) the rear spa	ar lower chord from	rib 25 to rib 27.					
		26A001 - DTR, DTR che			•					
		hod(s) necessary to acc The inspection procedur				the 737 Nondestruc	ctive Test Mani			

(D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-58.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.





				INTERVAL		APPLICA	BILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-641-00-02	AWL	57-05-02-250-874	1.1	NOTE		ALL	ALL			
	Inspect (Low Frequency Eddy Current) the rear spar lower chord from rib 25 to rib 27. See Doc. D626A001 - DTR, DTR check form 57-20-05-10, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-58. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.									
57-642-00-01	AWL	57-05-02-250-847	1.1	56000 FC	28000 FC	600 700 700C 700IGW 800 900	ALL			
	the gear bea See Doc. D6 The NDI met	n Frequency Eddy Curre m outboard support fittin 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	ngs and at WS ⁻ eck form 57-20 complish the int	TA 180 and WSTA 1 -06 / 07-1, for alternent of this inspection	190 on the forwar native repeat insp on is contained in	d trunnion support finection.	tting.			
	AIRPLANE I	NOTE: All except 900EF	₹							
	ACCESS NO	OTE: Fairing removal re	quired at WST	A 228.25.						
57-642-00-02	AWL	57-05-02-250-847	1.1	56000 FC	28000 FC	600 700 700C 700IGW 800 900	ALL			
	the gear bea See Doc. D6 The NDI met (D6-37239). AIRPLANE I	n Frequency Eddy Curre m outboard support fittin 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur NOTE: All except 900ER DTE: Fairing removal re	ngs and at WS- eck form 57-20 complish the interes are contain	FA 180 and WSTA ² -06 / 07-1, for alter ent of this inspectic ed in Part 6, Sectio	190 on the forwar native repeat insp on is contained in	d trunnion support fi	tting.			
57-643-00-01	AWL	57-05-02-130-835	1.1	NOTE		ALL	ALL			
	See Doc. D6 The NDI met (D6-37239).	asonic) the lower wing sl 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re	eck form 57-20 complish the intes res are contain sitive (FLS) tas	-09, for alternative ent of this inspection ed in Part 4, Subject	inspection. on is contained in ct 57-10-06. Figure 2 to detern	the 737 Nondestruc				
57-643-00-02	AWL	57-05-02-130-835	1.1	NOTE		ALL	ALL			
		econic) the lower wing sl								

Inspect (Ultrasonic) the lower wing skin at the R2, R3 and R4 nacelle fitting attachments.

See Doc. D626A001 - DTR, DTR check form 57-20-09, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 57-10-06.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 2750FC.







				INTERVAL		APPLICA	ABILITY		
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
57-643-01-01	AWL	57-05-02-250-849	1.1	NOTE		ALL	ALL		
	Inspect (High Frequency Eddy Current) the lower wing skin at the R2 and R4 nacelle fitting attachments. See Doc. D626A001 - DTR, DTR check form 57-20-09, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manu (D6-37239). The inspection procedures are contained in Part 6, Subject 57-30-08. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing								
	INTERVAL	• •	, ,	n interval is 2750FC	•	illine tillesheld. Beer	9		
57-643-01-02	AWL	57-05-02-250-849	1.1	NOTE		ALL	ALL		
	See Doc. Do The NDI me (D6-37239).	h Frequency Eddy Curre 626A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedul NOTE: Flight length sen	eck form 57-20 complish the int res are contain sitive (FLS) tas	-09, for alternative i ent of this inspectio ed in Part 6, Subjec k. See Section 9, F	nspection. In is contained in It 57-30-08. Iigure 2 to detern	the 737 Nondestruc			
		recommended re	epeat inspection	n interval is 2750FC	> .				
57-643-10-01	AWL	57-05-02-250-851	1.1	NOTE		ALL	ALL		
57-643-10-01	AWL Inspect (Hig See Doc. Do The NDI me (D6-37239).	57-05-02-250-851 h Frequency Eddy Curre 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedul NOTE: Flight length sen	1.1 ent) the lower week form 57-20 complish the intres are contained sitive (FLS) task	NOTE ing panel skin at the -10 for alternative ir ent of this inspection ed in Part 6, Subject	e shear tied rib anspection. In is contained in to 57-10-64. It gure 1 to detern	ttachments at rib 14	ctive Test Man		
57-643-10-01 57-643-10-02	AWL Inspect (Hig See Doc. Do The NDI me (D6-37239).	57-05-02-250-851 h Frequency Eddy Curre 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedul NOTE: Flight length sen	1.1 ent) the lower week form 57-20 complish the intres are contained sitive (FLS) task	NOTE ing panel skin at the -10 for alternative ir ent of this inspection ed in Part 6, Subjection ek. See Section 9, F	e shear tied rib anspection. In is contained in to 57-10-64. It gure 1 to detern	ttachments at rib 14	ctive Test Man		
	AWL Inspect (Hig See Doc. D6 The NDI me (D6-37239). INTERVAL I AWL Inspect (Hig See Doc. D6 The NDI me (D6-37239).	57-05-02-250-851 The Frequency Eddy Curre S26A001 - DTR, DTR chothod(s) necessary to accommended research to the inspection procedure of the	1.1 ent) the lower week form 57-20 complish the interes are contained sitive (FLS) task epeat inspection 1.1 ent) the lower week form 57-20 complish the interes are contained sitive (FLS) task epitement of the interes are contained sitive (FLS) task epitement of the interes are contained sitive (FLS) task epitement of the interest are contained sitive (FLS) task epitement of the interest are contained as it is in the interest are contained as it is interest.	NOTE ing panel skin at the -10 for alternative ir ent of this inspectio ed in Part 6, Subject sk. See Section 9, For interval is 18000F NOTE ing panel skin at the -10 for alternative ir ent of this inspectio ed in Part 6, Subjection	e shear tied rib anspection. In is contained in the thick that the	ttachments at rib 14 the 737 Nondestruction threshold. Boei ALL ttachments at rib 14 the 737 Nondestruction threshold.	active Test Manual		

fittings at flap tracks 1, 2, 7 and 8.

See Doc. D626A001 - DTR, DTR check form 57-20-12, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-30-09.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 12000FC.

ACCESS NOTE: Removal of flap track fairing required.





				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
57-644-00-02	AWL	57-05-02-250-853	1.1	NOTE		ALL	ALL		
	Inspect (High Frequency Eddy Current) the wing lower skin, area under the flap fairing, between the forward and aft attach fittings at flap tracks 1, 2, 7 and 8. See Doc. D626A001 - DTR, DTR check form 57-20-12, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 57-30-09.								
	INTERVAL	NOTE: Flight length sen recommended re	` '	n interval is 12000F	•	nine threshold. Boel	ng		
	ACCESS NO	OTE: Removal of flap tra	ack fairing requ	ired.					
57-644-01-01	AWL	57-05-02-130-825	1.1	NOTE		ALL	ALL		
	(D6-37239). INTERVAL I	thod(s) necessary to acc The inspection procedur NOTE: Flight length sen recommended re OTE: Removal of flap tra	res are containe sitive (FLS) tas epeat inspection	ed in Part 4, Subjectsk. See Section 9, Fin interval is 12000F	ct 57-10-08. Figure 2 to detern				
57-644-01-02	AWL	57-05-02-130-825	1.1	NOTE		ALL	ALL		
	See Doc. Do The NDI me (D6-37239).	asonic) the perimeter of 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen recommended re	eck form 57-20 complish the intres are contain sitive (FLS) tas epeat inspection	-12, for alternative ent of this inspectic ed in Part 4, Subjec sk. See Section 9, F n interval is 12000F	inspection. on is contained in ot 57-10-08. Figure 2 to detern	the 737 Nondestruc	ctive Test Manu		
57-644-02-01	AWL	57-05-02-250-854	1.1	NOTE		ALL	ALL		
31- 044- 02-01	Inspect (High rub strips at See Doc. Do The NDI me	n Frequency Eddy Curre flap tracks 1, 2, 7 and 8. 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	nt) the wing loveck form 57-20 complish the int	wer skin forward of -12, for alternative ent of this inspection	inspection. on is contained in	ach fittings and betw	reen the fairing		

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing

recommended repeat inspection interval is 12000FC.

ACCESS NOTE: Removal of flap track fairing required.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-644-02-02	AWL	57-05-02-250-854	1.1	NOTE		ALL	ALL
	rub strips at See Doc. D6 The NDI me	h Frequency Eddy Curre flap tracks 1, 2, 7 and 8. 526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the int	-12, for alternative i	inspection. on is contained in	Ü	Ü
		NOTE: Flight length sen recommended re DTE: Removal of flap tra	epeat inspection	n interval is 12000F	•	nine threshold. Boeir	ng
57-645-00-01	AWL	57-05-02-211-829	1.1	NOTE		ALL	ALL
	See Doc. D6	ailed) the typical stringer 326A001 - DTR, DTR che NOTE: Flight length sen	eck form 57-20	-13-1, for alternative	e inspection.		
	INTERVALI			n interval is 6875 F		nine threshold. Boeir	ig
57-645-00-02	AWL					ALL	ALL
57-645-00-02	AWL Inspect (Det	recommended re	1.1 rs at rib 5 that a	n interval is 6875 Fo	C. al pans and seal	ALL	
57-645-00-02	AWL Inspect (Det See Doc. De	recommended re 57-05-02-211-829 ailed) the typical stringer 526A001 - DTR, DTR che NOTE: Flight length sen	1.1 s at rib 5 that a eck form 57-20 sitive (FLS) tas	NOTE re hidden under se -13-1, for alternative	al pans and seal e inspection.	ALL ant.	ALL
57-645-00-02 57-645-05-01	AWL Inspect (Det See Doc. De	recommended re 57-05-02-211-829 ailed) the typical stringer 526A001 - DTR, DTR che NOTE: Flight length sen	1.1 s at rib 5 that a eck form 57-20 sitive (FLS) tas	NOTE re hidden under se -13-1, for alternative k. See Section 9, F	al pans and seal e inspection.	ALL ant.	ALL
	AWL Inspect (Det See Doc. Det INTERVAL INTERVAL INTERVAL INTERVAL INSPECT (High inboard and See Doc. Det The NDI me (D6-37239).	recommended re 57-05-02-211-829 ailed) the typical stringer 526A001 - DTR, DTR che NOTE: Flight length sen recommended re	1.1 rs at rib 5 that a eck form 57-20 sitive (FLS) tas epeat inspection 1.1 rnt) the upper s BL 70.85. eck form 57-20 complish the interes are contain.	NOTE The hidden under set of	al pans and seale inspection. igure 1 to deterr C. tachment at BBL 1, for alternative on is contained in at 57-10-34.	ALL ant. nine threshold. Boein ALL .70.85. Inspection is inspection. the 737 Nondestruct	ALL ang ALL and on both the
	AWL Inspect (Det See Doc. Det INTERVAL INTERVAL INTERVAL INTERVAL INSPECT (High inboard and See Doc. Det The NDI me (D6-37239).	recommended re 57-05-02-211-829 ailed) the typical stringer 326A001 - DTR, DTR che NOTE: Flight length sen recommended re 57-05-02-250-923 h Frequency Eddy Curre outboard locations at BE 326A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	1.1 rs at rib 5 that a eck form 57-20 sitive (FLS) tas epeat inspection 1.1 rnt) the upper s BL 70.85. eck form 57-20 complish the interes are contains sitive (FLS) tas	NOTE The hidden under set of	al pans and seale inspection. Gigure 1 to deterr C. tachment at BBL 1, for alternative on is contained in its 57-10-34.	ALL ant. nine threshold. Boein ALL .70.85. Inspection is inspection. the 737 Nondestruct	ALL ALL on both the

Inspect (High Frequency Eddy Current) the upper skin-to-plus chord attachment at BBL 70.85. Inspection is on both the inboard and outboard locations at BBL 70.85.

See Doc. D626A001 - DTR, DTR check form 57-20-13/-14/-15/-16/-17-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-34.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-645-10-01	AWL	57-05-02-250-876	1.1	NOTE		ALL	ALL
	Inspection is See Doc. D6 The NDI met (D6-37239).	on Frequency Eddy Curre on both the inboard and 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re	I outboard loca eck form 57-20 omplish the int es are contain sitive (FLS) tas	tions at BBL 70.85. -13/-14/-15/-16/-17 ent of this inspectic ed in Part 6, Subjec	-2, for alternative in is contained in it 57-10-34. igure 2 to detern	inspection. the 737 Nondestruc	ctive Test Manua
57-645-10-02	AWL	57-05-02-250-876	1.1	NOTE		ALL	ALL
	Inspection is See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre on both the inboard and 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen	outboard loca eck form 57-20 omplish the int es are contain	tions at BBL 70.85. -13/-14/-15/-16/-17 ent of this inspectic ed in Part 6, Subjec	-2, for alternative on is contained in of 57-10-34.	inspection. the 737 Nondestruc	ctive Test Manua
57-645-11-01	AWL	recommended re 57-05-02-250-878	peat inspection	n interval is 36000F NOTE	C.	ALL	ALL
	See Doc. D6 The NDI met	n Frequency Eddy Curre 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	eck form 57-20 omplish the int	-13/-14/-15/-16/-17 ent of this inspectio	-2, for alternative on is contained in	inspection.	
	INTERVAL N	IOTE: Flight length sen recommended re	` ,	k. See Section 9, F n interval is 36000F	•	nine threshold. Boei	ng
57-645-11-02	AWL	57-05-02-250-878	1.1	NOTE		ALL	ALL
57-645-11-02	Inspect (High See Doc. D6 The NDI met	57-05-02-250-878 n Frequency Eddy Curre 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur	nt) the aft skin- eck form 57-20 omplish the int	to-plus chord attac -13/-14/-15/-16/-17- ent of this inspectio	-2, for alternative on is contained in	stringer S-14, BBL 7	70.85.
57-645-11-02	Inspect (High See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen	nt) the aft skin- eck form 57-20 omplish the int es are contain sitive (FLS) tas	to-plus chord attac -13/-14/-15/-16/-17- ent of this inspection ed in Part 6, Subject	2, for alternative in is contained in the total strain the strain the total strain the strain th	stringer S-14, BBL 7 inspection. the 737 Nondestruc	70.85. ctive Test Manua
57-645-11-02 57-645-12-01	Inspect (High See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedur IOTE: Flight length sen	nt) the aft skin- eck form 57-20 omplish the int es are contain sitive (FLS) tas	to-plus chord attact-13/-14/-15/-16/-17-ent of this inspectioned in Part 6, Subject. k. See Section 9, F	2, for alternative in is contained in the total strain the strain the total strain the strain th	stringer S-14, BBL 7 inspection. the 737 Nondestruc	70.85. ctive Test Manu

See Doc. D626A001 - DTR, DTR check form 57-20-13/-14/-15/-16/-17-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-53.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 36000FC.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-645-12-02	AWL	57-05-02-250-880	1.1	NOTE		ALL	ALL
	See Doc. D6 The NDI met (D6-37239).	Frequency Eddy Currer 26A001 - DTR, DTR chi- hod(s) necessary to acc The inspection procedur	eck form 57-20 complish the infres are contain	-13/-14/-15/-16/-17 tent of this inspection ed in Part 6, Subject	-2, for alternative on is contained in ct 57-10-53.	inspection. the 737 Nondestruc	ctive Test Manu
	INTERVAL	IOTE: Flight length sen recommended re	` ,	n interval is 36000F	•	nine threshold. Boei	ng
57-645-13-01	AWL	57-05-02-250-882	1.1	NOTE		ALL	ALL
	See Doc. D6 The NDI met (D6-37239).	Frequency Eddy Currer 26A001 - DTR, DTR chi- thod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re	eck form 57-20 complish the intres are contain sitive (FLS) tas	-13/-14/-15/-16/-17 tent of this inspection ed in Part 6, Subject	-2, for alternative on is contained in ct 57-10-53. Figure 2 to detern	inspection. the 737 Nondestruc	ctive Test Manu
57-645-13-02	AWL	57-05-02-250-882	1.1	NOTE		ALL	ALL
	The NDI met (D6-37239).	26A001 - DTR, DTR chechod(s) necessary to acc The inspection procedur IOTE: Flight length sen recommended re	complish the interes are contain sitive (FLS) tas	tent of this inspection ed in Part 6, Subject	on is contained in ct 57-10-53. Figure 2 to detern	the 737 Nondestruc	
57-645-15-01	AWL	57-05-02-210-821	1.1	NOTE		ALL	ALL
		eral Visual) upper wing 26A001 - DTR, DTR ch		•		ocations adjacent to	the spar chord
	INTERVAL N	IOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-645-15-02	AWL	57-05-02-210-821	1.1	NOTE		ALL	ALL
		eral Visual) upper wing 26A001 - DTR, DTR ch		-		ocations adjacent to	the spar chord
	INTERVAL N	IOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-646-00-01	AWL	57-05-02-211-831	1.1	NOTE		ALL	ALL
		ailed) rear spar upper ch 26A001 - DTR, DTR ch					

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-646-00-02	AWL	57-05-02-211-831	1.1	NOTE		ALL	ALL
	Inspect (Deta	ailed) rear spar upper ch	ord at the non-	hidden areas from	rib 1 to rib 13.		
	See Doc. D6	626A001 - DTR, DTR cho	eck form 57-20	-17-1, for alternativ	e inspection.		
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	k. See Section 9, F n interval is 24000F	•	nine threshold. Boei	ng
57-647-00-01	AWL	57-05-02-250-857	1.1	NOTE		ALL	ALL
	rib 1 to rib 13 See Doc. D6 The NDI met (D6-37239).	526A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedul NOTE: Flight length sen	eck form 57-20 complish the intres are containastive (FLS) tas	-17-2, for alternative ent of this inspection ed in Part 6, Subject	e inspection. on is contained in ot 57-10-59. Figure 2 to detern	the 737 Nondestruc	ctive Test Manu
57-647-00-02	AWL	57-05-02-250-857	1.1	NOTE		ALL	ALL
		Frequency Eddy Curre	nt) the rear spa		e areas hidden b		
	rib 1 to rib 13		,			, , . , . , , , , , , , , , , , , , ,	,
	See Doc. D6	326A001 - DTR, DTR che	eck form 57-20	-17-2, for alternative	e inspection.		
	The NDI met	thod(s) necessary to acc	omnlish the int	ent of this inspection	n is contained in	the 727 Needestrie	
						i ine 737 Nondesiru	ctive Test Manu
	(D6-37239).	The inspection procedur				the 737 Nondestruc	ctive Test Manu
	,	The inspection procedure NOTE: Flight length sen	res are contain sitive (FLS) tas	ed in Part 6, Subjec	et 57-10-59. Figure 2 to detern		
57-647-10-01	,	The inspection procedure NOTE: Flight length sen	res are contain sitive (FLS) tas	ed in Part 6, Subjec k. See Section 9, F	et 57-10-59. Figure 2 to detern		
57-647-10-01	AWL Inspect (High	The inspection procedur NOTE: Flight length sen recommended re	res are contain sitive (FLS) tas epeat inspection 1.1 nt) the rear spa	ed in Part 6, Subjective. See Section 9, Fin interval is 18000F NOTE ar upper chord at the	et 57-10-59. Figure 2 to detern FC. e non-hidden are	nine threshold. Boei	ng ALL
57-647-10-01	AWL Inspect (High See Doc. Do The NDI mei	NOTE: Flight length sen recommended re 57-05-02-250-859 h Frequency Eddy Curre	res are contain sitive (FLS) tas epeat inspection 1.1 nt) the rear spaceck form 57-20 complish the int	ed in Part 6, Subject in Part 6, Subject in Part 6, Subject in Part 1,	et 57-10-59. Figure 2 to determed. FC. The non-hidden are a inspection. The inspection is contained in	ALL eas from rib 1 to rib	ALL 13.
57-647-10-01	AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239).	The inspection procedur NOTE: Flight length sen recommended re 57-05-02-250-859 h Frequency Eddy Curre 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen	res are contain- sitive (FLS) tas speat inspection 1.1 int) the rear spa eck form 57-20 complish the int res are contain- sitive (FLS) tas	ed in Part 6, Subject ik. See Section 9, For interval is 18000F NOTE ar upper chord at the 17-3, for alternative ent of this inspectice in Part 6, Subjected in Part 6, Subject	et 57-10-59. Figure 2 to deternice. The non-hidden are enon-hidden are enspection. In is contained in the third straight of the t	ALL eas from rib 1 to rib	ALL 13. ctive Test Manu
57-647-10-01 57-647-10-02	AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239).	The inspection procedur NOTE: Flight length sen recommended re 57-05-02-250-859 h Frequency Eddy Curre 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen	res are contain- sitive (FLS) tas speat inspection 1.1 int) the rear spa eck form 57-20 complish the int res are contain- sitive (FLS) tas	NOTE ar upper chord at the 17-3, for alternative et in Part 6, Subjective the 18-17-8. The subjective et in Part 6, Sub	et 57-10-59. Figure 2 to deternice. The non-hidden are enon-hidden are enspection. In is contained in the third straight of the t	ALL eas from rib 1 to rib	ALL 13. ctive Test Manu
	AWL Inspect (High See Doc. Doc. The NDI met (D6-37239). INTERVAL M	The inspection procedur NOTE: Flight length sen recommended re 57-05-02-250-859 h Frequency Eddy Curre (26A001 - DTR, DTR che (16A001) checks ary to accommended resource (16A001) recommended reco	res are contain sitive (FLS) tas epeat inspection 1.1 nt) the rear spaceck form 57-20 complish the interes are contain sitive (FLS) tas epeat inspection 1.1 nt) the rear space point in the sitive (FLS) tas epeat inspection 1.1 nt) the rear space are contained to the sitive (FLS) tas epeat inspection 1.1	ed in Part 6, Subject in Part 6, Subject in Part 6, Subject in Interval is 18000F NOTE ar upper chord at the 17-3, for alternative in Part 6, Subject in Part 6, Subject in Part 1, Subject in Part 1, Subject in Part 1, Subject in Part 1, Subject in Part 2, Subject in Part 1, Su	et 57-10-59. Figure 2 to deterned. e non-hidden are e inspection. on is contained in the 57-10-42. Figure 2 to deterned. e non-hidden are	ALL eas from rib 1 to rib the 737 Nondestructure threshold. Boein	ALL 13. ctive Test Manu
	AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239). INTERVAL N AWL Inspect (High See Doc. Doc. The NDI mer (The NDI mer (The NDI mer))	The inspection procedur NOTE: Flight length sen recommended re 57-05-02-250-859 h Frequency Eddy Curre 626A001 - DTR, DTR che 626A001 - DTR che 626A001	res are contain sitive (FLS) tas epeat inspection 1.1 nt) the rear spaceck form 57-20 complish the interes are contain sitive (FLS) tas epeat inspection 1.1 nt) the rear spaceck form 57-20 complish the interes are contain sitive (FLS) tas epeat inspection 2.1	ed in Part 6, Subject in Part 6, Subject in Part 6, Subject in Interval is 18000F NOTE ar upper chord at the 17-3, for alternative in Part 6, Subject in Part 6, Subject in Part 1, Subject in Part 1, Subject in Part 1, Subject in Part 2, Subject in Part 3, For alternative in Interval is 24000F	et 57-10-59. Figure 2 to deterned. e non-hidden are e inspection. on is contained in et 57-10-42. Figure 2 to deterned. e non-hidden are e inspection. on is contained in et inspection. on is contained in et inspection.	ALL eas from rib 1 to rib the 737 Nondestruct nine threshold. Boein ALL eas from rib 1 to rib	ALL 13. ctive Test Manuary ALL 13.
	AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239). INTERVAL N AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239).	The inspection procedur NOTE: Flight length sen recommended re 57-05-02-250-859 In Frequency Eddy Curre (26A001 - DTR, DTR che (16A001) NOTE: Flight length sen recommended re 57-05-02-250-859 In Frequency Eddy Curre (26A001) The inspection procedure (26A001)	1.1 Int) the rear spaceck form 57-20 complish the interest inspection 1.1 Int) the rear spaceck form 57-20 complish the interest inspection 1.1 Int) the rear spaceck form 57-20 complish the interest inspection Int) the rear spaceck form 57-20 complish the interest are contained in the interest i	NOTE ar upper chord at the 17-3, for alternative interval is 24000F NOTE ar upper chord at the 17-3, for alternative in Part 6, Subject in Part 8, Subject in Part 9, For interval is 24000F NOTE ar upper chord at the 17-3, for alternative in part 6, Subject i	et 57-10-59. Figure 2 to deternice. The non-hidden are enspection. The instruction is contained in the struction. The non-hidden are enspection. The non-hidden are enspection. The instruction is contained in the struction. The struction is contained in the struction. The struction is contained in the struction is contained in the struction.	ALL eas from rib 1 to rib the 737 Nondestruct nine threshold. Boein ALL eas from rib 1 to rib the 737 Nondestruct	ALL 13. ctive Test Manual 13. ALL 13. ctive Test Manual
	AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239). INTERVAL N AWL Inspect (High See Doc. Doc. The NDI mer (D6-37239).	The inspection procedur NOTE: Flight length sen recommended re 57-05-02-250-859 In Frequency Eddy Curre (26A001 - DTR, DTR che (16A001) NOTE: Flight length sen recommended re 57-05-02-250-859 In Frequency Eddy Curre (26A001) The inspection procedure (26A001)	1.1 Int) the rear spaceck form 57-20 complish the interest inspection 1.1 Int) the rear spaceck form 57-20 complish the interest inspection 1.1 Int) the rear spaceck form 57-20 complish the interest inspection Int) the rear spaceck form 57-20 complish the interest are contained in the interest i	NOTE ar upper chord at the 17-3, for alternative on interval is 24000F NOTE ar upper chord at the 17-3, for alternative of this inspectice of in Part 6, Subject on the 17-3, for alternative of the 17-3, for alternative on the 17-3, for alternati	et 57-10-59. Figure 2 to deternice. The non-hidden are enspection. The instruction is contained in the struction. The non-hidden are enspection. The non-hidden are enspection. The instruction is contained in the struction. The struction is contained in the struction. The struction is contained in the struction is contained in the struction.	ALL eas from rib 1 to rib the 737 Nondestruct nine threshold. Boein ALL eas from rib 1 to rib the 737 Nondestruct	ALL 13. ctive Test Manual ng ALL 13. ctive Test Manual

See Doc. D626A001 - DTR, DTR check form 57-20-17-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 57-10-10.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing recommended repeat inspection interval is 24000FC.





LVEK CVBD NO				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-647-20-02	AWL	57-05-02-130-827	1.1	NOTE		ALL	ALL
	See Doc. Do The NDI me (D6-37239).	asonic) the rear spar upp 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sens recommended re	eck form 57-20- omplish the int es are containe sitive (FLS) tas	17-4, for alternative ent of this inspection ed in Part 4, Subjec	e inspection. In is contained in It 57-10-10. Igure 2 to determ	the 737 Nondestruc	ctive Test Manu
57-648-00-01	AWL	57-05-02-250-861	1.1	56000 FC	18000 FC	ALL	ALL
	fairing brack See Doc. Do The NDI me	h Frequency Eddy Currer let from the adjacent edgo 326A001 - DTR, DTR che thod(s) necessary to acco The inspection procedur	e. eck form 57-20- omplish the int	-19, for alternative i	nspection. on is contained in		
57-648-00-02	AWL	57-05-02-250-861	1.1	56000 FC	18000 FC	ALL	ALL
57.040.00.04		The inspection procedure	es are containe	ed in Part 6, Section	n 57-30-10.		
7-649-00-01	AWL	57-05-02-211-833	1.1	NOTE		ALL	ALL
57-649-00-01	Inspect (Det	57-05-02-211-833 ailed) the web at the stiffe 526A001 - DTR, DTR che	eners or fittings	from rib 1 to rib 6			ALL
57-649-00-01	Inspect (Det See Doc. De	ailed) the web at the stiffe 326A001 - DTR, DTR che NOTE: Flight length sens	eners or fittings eck form 57-20- sitive (FLS) tas	from rib 1 to rib 6 22, for alternative i	nspection. igure 1 to determ	rib 27.	
57-649-00-01 57-649-00-02	Inspect (Det See Doc. De	ailed) the web at the stiffe 326A001 - DTR, DTR che NOTE: Flight length sens	eners or fittings eck form 57-20- sitive (FLS) tas	from rib 1 to rib 6 22, for alternative i k. See Section 9, F	nspection. igure 1 to determ	rib 27.	
	Inspect (Det See Doc. Do INTERVAL I	ailed) the web at the stiffe 326A001 - DTR, DTR che NOTE: Flight length sens recommended re	eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1	s from rib 1 to rib 6 22, for alternative i k. See Section 9, F interval is 18000F NOTE s from rib 1 to rib 6	nspection. rigure 1 to determ C. and from rib 7 to	rib 27. ine threshold. Boeii ALL	ng
	Inspect (Det See Doc. Do INTERVAL I AWL Inspect (Det See Doc. Do	ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-833 ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens	eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 eners or fittings eck form 57-20- sitive (FLS) tas	s from rib 1 to rib 6 22, for alternative i k. See Section 9, F interval is 18000F NOTE from rib 1 to rib 6 22, for alternative i	nspection. igure 1 to determ C. and from rib 7 to nspection. igure 1 to determ	rib 27. ine threshold. Boeir ALL rib 27.	ng ALL
	Inspect (Det See Doc. Do INTERVAL I AWL Inspect (Det See Doc. Do	ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-833 ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens	eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 eners or fittings eck form 57-20- sitive (FLS) tas	s from rib 1 to rib 6 22, for alternative i k. See Section 9, F n interval is 18000F NOTE s from rib 1 to rib 6 22, for alternative i k. See Section 9, F	nspection. igure 1 to determ C. and from rib 7 to nspection. igure 1 to determ	rib 27. ine threshold. Boeir ALL rib 27.	ng ALL
57-649-00-02	AWL Inspect (Det See Doc. Do INTERVAL I Inspect (Det See Doc. Do INTERVAL I INTERVAL I INTERVAL I	ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-833 ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re	eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 rear spar web	s from rib 1 to rib 6 s.22, for alternative i k. See Section 9, F. interval is 18000F NOTE s from rib 1 to rib 6 s.22, for alternative i k. See Section 9, F. interval is 18000F NOTE NOTE at ribs 1 to 27.	nspection. igure 1 to determ C. and from rib 7 to nspection. igure 1 to determ C.	rib 27. ine threshold. Boeir ALL rib 27. ine threshold. Boeir	ALL
57-649-00-02	AWL Inspect (Det See Doc. Det INTERVAL II AWL AWL Inspect (Det See Doc. Det INTERVAL II AWL Inspect (Det See Doc. Det S	ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-833 ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-835 ailed) the outboard wing S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight lengt	eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 rear spar web eck form 57-20- sitive (FLS) tas	s from rib 1 to rib 6 a.22, for alternative i k. See Section 9, F. interval is 18000F NOTE s from rib 1 to rib 6 a.22, for alternative i k. See Section 9, F. interval is 18000F NOTE at ribs 1 to 2724-1, for alternative	nspection. rigure 1 to determ C. and from rib 7 to nspection. rigure 1 to determ C.	rib 27. ine threshold. Boein ALL rib 27. ine threshold. Boein ALL	ALL ALL
57-649-00-02	AWL Inspect (Det See Doc. Det INTERVAL II AWL AWL Inspect (Det See Doc. Det INTERVAL II AWL Inspect (Det See Doc. Det S	ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-833 ailed) the web at the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens recommended re 57-05-02-211-835 ailed) the outboard wing S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR, DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight length sens for the stiffe S26A001 - DTR che NOTE: Flight lengt	eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 eners or fittings eck form 57-20- sitive (FLS) tas peat inspection 1.1 rear spar web eck form 57-20- sitive (FLS) tas	s from rib 1 to rib 6 and rib 22, for alternative in k. See Section 9, For interval is 18000F NOTE s from rib 1 to rib 6 and rib 22, for alternative in k. See Section 9, For interval is 18000F NOTE at ribs 1 to 27. 24-1, for alternative k. See Section 9, For alternative k.	nspection. rigure 1 to determ C. and from rib 7 to nspection. rigure 1 to determ C.	rib 27. ine threshold. Boein ALL rib 27. ine threshold. Boein ALL	ALL ALL

Inspect (Detailed) the outboard wing rear spar web at ribs 1 to 27.

See Doc. D626A001 - DTR, DTR check form 57-20-24-1, for alternative inspection.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 8000FC.







				INTERVAL		APPLICA	BILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
57-650-10-01	AWL	57-05-02-250-884	1.1	NOTE		ALL	ALL		
	Inspect (High Frequency Eddy Current) the outboard wing rear spar at the hidden areas from rib 1 to rib 27, except at locations covered by PSE 57-20-24/25/26. See Doc. D626A001 - DTR, DTR check form 57-20-24-2, for alternative inspection. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manua (D6-37239). The inspection procedures are contained in Part 6, Subject 57-10-52. INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing recommended repeat inspection interval is 12000FC.								
57-650-10-02	AWL	57-05-02-250-884	1.1	NOTE		ALL	ALL		
	locations cov See Doc. D6 The NDI met (D6-37239).	n Frequency Eddy Curre vered by PSE 57-20-24/2 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Flight length sen recommended re	25/26. eck form 57-20 complish the interest are contain sitive (FLS) tas	-24-2, for alternativ tent of this inspection ed in Part 6, Subject	e inspection. on is contained in ot 57-10-52. Figure 1 to detern	the 737 Nondestruc	tive Test Manu		
57-651-00-01	AWL	57-05-02-130-829	1.1	NOTE		ALL	ALL		
	Inspect (Ultra	asonic) the rear spar we	b at the trunnic	on attachment and r	main landing gea	r fitting locations from	rib 1 to rib 27		
	INTERVAL N	The inspection procedur NOTE: Flight length sen recommended re OTE: Deploy flaps and s	sitive (FLS) tas epeat inspectio	sk. See Section 9, F n interval is 18000F	igure 1 to deterr	nine threshold. Boein	g		
57-651-00-02	AWL	57-05-02-130-829	1.1	NOTE		ALL	ALL		
07 001 00 02		asonic) the rear spar we			main landing gea				
	See Doc. D6 The NDI met	526A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the interior	-24/25/26, for alterr tent of this inspection	native inspection on is contained in				
	INTERVAL N	NOTE: Flight length sen recommended re		sk. See Section 9, F n interval is 18000F	-	nine threshold. Boein	g		
	ACCESS NO	OTE: Deploy flaps and s	poilers to gain	access.					
57-651-10-01	AWL	57-05-02-250-863	1.1	NOTE		700 800 900ER	ALL		
	spar at WBL See Doc. D6 The NDI met	n Frequency Eddy Curre 616.75. 26A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedur	eck form 57-20 complish the int	-29-1, for alternativ	re inspection. on is contained in				
	AIRPLANE	NOTE: Applies to 737-7 Table 9-1.	00, 737-800 aı	nd 737-900 ER with	n production insta	alled winglets per Sec	tion 9,		

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recommended repeat inspection interval is 24000FC.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 2 to determine threshold. Boeing





				INTERVAL		APPLICA	BILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-651-10-02	AWL	57-05-02-250-863	1.1	NOTE		700 800 900ER	ALL
	spar at WBL See Doc. D6 The NDI me	h Frequency Eddy Curre .616.75. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-20 omplish the int	-29-1, for alternative ent of this inspection	e inspection. on is contained i		
	AIRPLANE	NOTE: Applies to 737-7 Table 9-1.	00, 737-800 ar	nd 737-900 ER with	production inst	alled winglets per Sec	tion 9,
	INTERVAL I	NOTE: Flight length sense recommended re	` '	k. See Section 9, F n interval is 24000F	•	mine threshold. Boein	9
57-652-00-01	AWL	57-05-02-250-865	1.1	NOTE		700 800 900ER	ALL
	See Doc. Do The NDI me (D6-37239).	WBL 658.17. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Applies to 737-7 Table 9-1. NOTE: Flight length sense	omplish the int es are contain 00, 737-800 ar sitive (FLS) tas	ent of this inspectioned in Part 6, Subjectioned 737-900 ER with	on is contained in the strategy of the strateg	alled winglets per Sec	tion 9,
57-652-00-02	AWL	57-05-02-250-865	1.1	NOTE		700 800 900ER	ALL
	rear spar at See Doc. Do The NDI me (D6-37239).	v Frequency Eddy Currer WBL 658.17. 626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur NOTE: Applies to 737-7 Table 9-1.	eck form 57-20 omplish the int es are containe 00, 737-800 ar	-29-2, for alternative ent of this inspection ed in Part 6, Subject and 737-900 ER with	e inspection. on is contained i ot 57-10-48. production inst	n the 737 Nondestruct	ive Test Manu
	INTERVAL I	NOTE: Flight length sense recommended re	` '	n interval is 36000F	•	mine threshold. Boeing	9
57-653-00-01	AWL	• •	` '		•	700 800 900ER	ALL

AIRPLANE NOTE: Applies to 737-700, 737-800 and 737-900 ER with production installed winglets per Section 9, Table 9-1.

INTERVAL NOTE: Flight length sensitive (FLS) task. See Section 9, Figure 1 to determine threshold. Boeing

recommended repeat inspection interval is 24000FC.

ACCESS NOTE: Fastener removal required.







				INTERVAL		APPLICA	BILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-653-00-02	AWL	57-05-02-250-867	1.1	NOTE		700 800 900ER	ALL			
	locations from See Doc. D6 The NDI met	n Frequency Eddy Curre m the outboard side pass 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	sing through ril eck form 57-20 complish the int	o 27. -29-3, for alternativ tent of this inspection	re inspection. on is contained in					
	AIRPLANE NOTE: Applies to 737-700, 737-800 and 737-900 ER with production installed winglets per Section 9, Table 9-1.									
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	sk. See Section 9, F n interval is 24000F	•	nine threshold. Boein	g			
	ACCESS NO	OTE: Fastener removal	required.							
57-654-00-01	AWL	57-05-02-211-839	1.1	56000 FC	5000 FC	ALL	ALL			
		ailed) the links, fittings in 26A001 - DTR, DTR che				57-20-39, for alternat	ive inspection.			
57-654-00-02	AWL	57-05-02-211-839	1.1	56000 FC	5000 FC	ALL	ALL			
57-655-00-01		ailed) the links, fittings in 26A001 - DTR, DTR che 57-05-02-250-869		•		57-20-39, for alternat	ive inspection.			
37-033-00-01	Inspect (Ope See Doc. Do The NDI met	en Hole Eddy Current) al 226A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	I eighteen uppo eck form 57-31 complish the int	er and lower tension -02-1, for alternative tent of this inspection	re inspection. on is contained in	mon to the winglet ST	A 0 Rib.			
		NOTE: Applicable to 73				er Section 9, Table 9	-1.			
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	sk. See Section 9, F n interval is 21000F	•	nine threshold. Boein	g			
	ACCESS NO	OTE: Removal of wingle	t and tension b	oolts is required.						
57-655-00-02	AWL	57-05-02-250-869	1.1	NOTE		800 900ER	ALL			
	See Doc. D6 The NDI met	en Hole Eddy Current) al 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-31 complish the int	-02-1, for alternative tent of this inspection	re inspection. on is contained in					
		NOTE: Applicable to 73				er Section 9, Table 9	-1.			
	INTERVAL N	NOTE: Flight length sen recommended re	, ,	sk. See Section 9, F n interval is 21000F	-	nine threshold. Boein	g			
	ACCESS NO	OTE: Removal of wingle	t and tension b	oolts is required.						







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-656-00-01	AWL	57-05-02-250-871	1.1	NOTE		800 900ER	ALL
	Inspect (Ope	n Hole Eddy Current) th	e in-spar lower	flange of the wingl	et STA 0 (root) rib).	
		26A001 - DTR, DTR cho		*	•		
		hod(s) necessary to acc The inspection procedu	•	•		the 737 Nondestruc	ctive Test Manu
	AIRPLANE I	NOTE: Applicable to 73	7-800, -900ER	with production ins	stalled winglets pe	er Section 9, Table 9	9-1.
	INTERVAL N	IOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 21000F	•	nine threshold. Boei	ng
	ACCESS NO	OTE: Removal of wingle front spar commor	•	•	quired. Remove 2	fasteners forward of	of the
57-656-00-02	AWL	57-05-02-250-871	1.1	NOTE		800 900ER	ALL
	See Doc. D6 The NDI met	en Hole Eddy Current) th 26A001 - DTR, DTR che hod(s) necessary to acc The inspection procedu	eck form 57-31 complish the int	-02-2, for alternativent of this inspection	e inspection. on is contained in		ctive Test Manu
	AIRPLANE I	NOTE: Applicable to 73	7-800, -900ER	with production ins	stalled winglets pe	er Section 9, Table 9	9-1.
	INTERVAL N	IOTE: Flight length sen recommended re	` ,	sk. See Section 9, F n interval is 21000F	O	nine threshold. Boei	ng
	ACCESS NO	DTE: Removal of wingle front spar commor	•	•	quired. Remove 2	fasteners forward of	of the
57-658-00-01	AWL	57-05-02-211-841	1.1	56000 FC	18000 FC	ALL	ALL
		ailed) all the lugs of both 26A001 - DTR, DTR che				ings on the main lar	nding gear bean
	ACCESS NO	OTE: Access requires de	eployment of ir	board flaps.			
57-658-00-02	AWL	57-05-02-211-841	1.1	56000 FC	18000 FC	ALL	ALL
		ailed) all the lugs of both 26A001 - DTR, DTR che			• • •	ings on the main lar	nding gear bean
	ACCESS NO	DTE: Access requires de	eployment of ir	board flaps.			
57-659-00-01	AWL	57-05-02-250-873	1.1	56000 FC	18000 FC	ALL	ALL
		n Frequency Eddy Curre 26A001 - DTR, DTR cho		-		of the main landing	gear beam.
57-659-00-02	AWL	57-05-02-250-873	1.1	56000 FC	18000 FC	ALL	ALL
		Frequency Eddy Curre					

Inspect (High Frequency Eddy Current) the four main lugs on the inboard support fitting of the main landing gear beam. See Doc. D626A001 - DTR, DTR check form 57-51-16, for alternative inspection.







				INTERVAL		APPLICA	BILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-660-00-01	AWL	57-05-02-211-843	1.1	56000 FC	4000 FC	600 700 700C 700IGW 800 900	ALL
		ailed) the outboard main 626A001 - DTR, DTR che					
	AIRPLANE	NOTE: All except 900EF	₹				
57-660-00-02	AWL	57-05-02-211-843	1.1	56000 FC	4000 FC	600 700 700C 700IGW 800 900	ALL
		ailed) the outboard main 326A001 - DTR, DTR che	• •				
	AIRPLANE	NOTE: All except 900EF	3				
57-661-00-01	AWL	57-05-02-211-845	1.1	56000 FC	4000 FC	ALL	ALL
	254.0 and 16	ailed) the catcher attache 64.0 on the inside bottom 226A001 - DTR, DTR che	n surface of the	catcher.	•	right and left side at V	VBL 357.7,
	254.0 and 16 See Doc. D6	64.0 on the inside bottom	n surface of the eck form 57-53	catcher02, for alternative	inspection.		·
57-661-00-02	254.0 and 16 See Doc. D6	64.0 on the inside bottom 626A001 - DTR, DTR che	n surface of the eck form 57-53	catcher02, for alternative	inspection.		·
57-661-00-02	254.0 and 10 See Doc. Do ACCESS NO AWL Inspect (Det 254.0 and 10	64.0 on the inside bottom 626A001 - DTR, DTR che DTE: Remove catcher. A	n surface of the eck form 57-53 Access to catch 1.1 ed to the forwant surface of the	catcher02, for alternative her requires remova 56000 FC rd fitting assemblie catcher.	inspection. If of the forward 4000 FC is for flap tracks in the forward in the f	fixed flap support fairi ALL	ng.
57-661-00-02	AWL Inspect (Det 254.0 and 16 See Doc. D6	54.0 on the inside bottom 526A001 - DTR, DTR che 526E Remove catcher. A 57-05-02-211-845 ailed) the catcher attache 64.0 on the inside bottom	n surface of the eck form 57-53 Access to catch 1.1 ed to the forwan surface of the eck form 57-53	catcher02, for alternative er requires remova 56000 FC rd fitting assemblie catcher02, for alternative	inspection. al of the forward 4000 FC s for flap tracks inspection.	fixed flap support fairi ALL right and left side at V	ALL VBL 357.7,
57-661-00-02 57-662-00-01	AWL Inspect (Det 254.0 and 16 See Doc. D6	54.0 on the inside bottom 526A001 - DTR, DTR che DTE: Remove catcher. A 57-05-02-211-845 ailed) the catcher attache 64.0 on the inside bottom 526A001 - DTR, DTR che	n surface of the eck form 57-53 Access to catch 1.1 ed to the forwan surface of the eck form 57-53	catcher02, for alternative er requires remova 56000 FC rd fitting assemblie catcher02, for alternative	inspection. al of the forward 4000 FC s for flap tracks inspection.	fixed flap support fairi ALL right and left side at V	ALL VBL 357.7,
	AWL AWL Inspect (Det 254.0 and 16 See Doc. Det ACCESS NO ACCESS NO AWL	54.0 on the inside bottom 526A001 - DTR, DTR che DTE: Remove catcher. A 57-05-02-211-845 ailed) the catcher attache 54.0 on the inside bottom 526A001 - DTR, DTR che DTE: Remove catcher. A	1.1 ed to the forwant surface of the eck form 57-53 Access to catch 1.1 ed to the forwant surface of the eck form 57-53 Access to catch 1.1 flap, inboard tr	catcher02, for alternative per requires removal 56000 FC rd fitting assemblie catcher02, for alternative per requires removal 56000 FC	4000 FC s for flap tracks inspection. al of the forward 4000 FC 4000 FC	fixed flap support fairi ALL right and left side at V fixed flap support fairi 600 700 700C 700IGW 800	ALL VBL 357.7,
	AWL Inspect (Det ACCESS NO	54.0 on the inside bottom 526A001 - DTR, DTR che 57-05-02-211-845 ailed) the catcher attache 54.0 on the inside bottom 526A001 - DTR, DTR che 57-05-02-211-847 ailed) the outboard main	n surface of the eck form 57-53 Access to catch 1.1 ed to the forwa in surface of the eck form 57-53 Access to catch 1.1 flap, inboard to eck form 57-53 of the track the	catcher02, for alternative per requires remova 56000 FC rd fitting assemblie catcher02, for alternative per requires remova 56000 FC rack at WBL 254.003-1a, for alternatice flap must be deplo	inspection. 4000 FC s for flap tracks inspection. al of the forward 4000 FC 4000 FC ve inspection. byed so the fairir	fixed flap support fairing ALL right and left side at V fixed flap support fairing 600 700 700C 700IGW 800 900 and clears the track. For	ALL VBL 357.7, ng. ALL

Inspect (Detailed) the outboard main flap, inboard track at WBL 254.0.

See Doc. D626A001 - DTR, DTR check form 57-53-03-1a, for alternative inspection.

ACCESS NOTE: For the aft portion of the track the flap must be deployed so the fairing clears the track. For the forward portion of the track the flap support fairing must be removed.







				INTERVAL		APPLICA	BILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-664-00-01	AWL	57-05-02-211-849	1.1	56000 FC	4000 FC	600 700 700C 700IGW 800 900	ALL
	Inspect (Det	ailed) the inboard main fl	lap, outboard tr	ack at WBL 164.0.			
	See Doc. D6	626A001 - DTR, DTR che	eck form 57-53	-06-1a, for alternati	ve inspection.		
	ACCESS NO	OTE: For the aft portion forward portion of		flap must be deplo ap support fairing m	•	•	or the
57-664-00-02	AWL	57-05-02-211-849	1.1	56000 FC	4000 FC	600 700 700C 700IGW 800 900	ALL
		ailed) the inboard main fl 326A001 - DTR, DTR che	• •		ve inspection.		
	ACCESS NO	OTE: For the aft portion forward portion of the		flap must be deplo ap support fairing m	•	•	or the
57-666-00-01	AWL	57-05-02-130-831	1.1	56000 FC	36000 FC	ALL	ALL
57-666-00-01	Inspect (Ultr	57-05-02-130-831 asonic) the inboard flap, 326A001 - DTR, DTR che	inboard track a	oft link pins at WBL	64.0.	ALL	ALL
57-666-00-01	Inspect (Ultr. See Doc. D6	asonic) the inboard flap,	inboard track a eck form 57-53	oft link pins at WBL	64.0.	ALL	ALL
57-666-00-01 57-666-00-02	Inspect (Ultr. See Doc. D6	asonic) the inboard flap, 326A001 - DTR, DTR che	inboard track a eck form 57-53	oft link pins at WBL	64.0.	ALL	ALL
	Inspect (Ultr See Doc. Do ACCESS NO AWL Inspect (Ultr	asonic) the inboard flap, 626A001 - DTR, DTR che DTE: Inner pin removal r	inboard track a eck form 57-53 required. 1.1 inboard track a	off link pins at WBL -09, for alternative in 56000 FC off link pins at WBL	64.0. nspection. 36000 FC 64.0.	112	
	AWL Inspect (Ultr. See Doc. D6 ACCESS NC	asonic) the inboard flap, 526A001 - DTR, DTR che DTE: Inner pin removal in 57-05-02-130-831 asonic) the inboard flap,	inboard track a eck form 57-53- required. 1.1 inboard track a eck form 57-53-	off link pins at WBL -09, for alternative in 56000 FC off link pins at WBL	64.0. nspection. 36000 FC 64.0.	112	
	AWL Inspect (Ultr. See Doc. D6 ACCESS NC	asonic) the inboard flap, 526A001 - DTR, DTR che DTE: Inner pin removal r 57-05-02-130-831 asonic) the inboard flap, 526A001 - DTR, DTR che	inboard track a eck form 57-53- required. 1.1 inboard track a eck form 57-53-	off link pins at WBL -09, for alternative in 56000 FC off link pins at WBL	64.0. nspection. 36000 FC 64.0.	112	
57-666-00-02	AWL Inspect (Ultr. See Doc. Doc. ACCESS NO. AWL AWL Inspect (Ultr. See Doc. Doc. ACCESS NO. AWL	asonic) the inboard flap, 326A001 - DTR, DTR che DTE: Inner pin removal r 57-05-02-130-831 asonic) the inboard flap, 326A001 - DTR, DTR che DTE: Inner pin removal r	inboard track a eck form 57-53-required. 1.1 inboard track a eck form 57-53-required. 1.1 1.1	aft link pins at WBL -09, for alternative in 56000 FC aft link pins at WBL -09, for alternative in 56000 FC	64.0. nspection. 36000 FC 64.0. nspection.	ALL 600 700 700C 700IGW 800	ALL
57-666-00-02	AWL Inspect (Ultr. See Doc. Doc. ACCESS NO. ACCESS NO. ACCESS NO. ACCESS NO. AWL Inspect (Det. See Doc. Doc. Doc. Doc. Doc. Doc. Doc. Doc.	asonic) the inboard flap, 626A001 - DTR, DTR che DTE: Inner pin removal responsible for the pin removal responsible for the inboard flap, 626A001 - DTR, DTR che DTE: Inner pin removal responsible for the inboard main flated) the inboard main flated.	inboard track a eck form 57-53-required. 1.1 inboard track a eck form 57-53-required. 1.1 lap, inboard track a eck form 57-53-g must be remo	aft link pins at WBL -09, for alternative in 56000 FC aft link pins at WBL -09, for alternative in 56000 FC	64.0. nspection. 36000 FC 64.0. nspection. 4000 FC ve inspection.	ALL 600 700 700C 700IGW 800 900	ALL

Inspect (Detailed) the inboard main flap, inboard track at WBL 64.0.

See Doc. D626A001 - DTR, DTR check form 57-53-10-1a, for alternative inspection.

ACCESS NOTE: Side of body fairing must be removed to gain access. For HFEC inspections the carriage must be cycled to gain access.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-668-00-01	AWL	57-05-02-250-883	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (High	h Frequency Eddy Currer	nt) the primary	lug of the inboard	main flap forward	fitting on the inboar	d track.
		626A001 - DTR, DTR che			•		
		thod(s) necessary to according to the inspection procedure.	•	•		the 737 Nondestruc	tive Test Man
	ACCESS NO	OTE: Flap track removal	required to pe	rform this inspection	on.		
57-668-00-02	AWL	57-05-02-250-883	1.1	56000 FC	18000 FC	ALL	ALL
	See Doc. Do	h Frequency Eddy Currer 626A001 - DTR, DTR che thod(s) necessary to acco The inspection procedure	eck form 57-53 omplish the int	-11, for alternative ent of this inspection	inspection. on is contained in	J	
	,	OTE: Flap track removal		•			
57-669-00-01	AWL	57-05-02-211-853	1.1	56000 FC	18000 FC	ALL	ALL
	main flap ST	ailed) the front spar uppe 「A 73 to 167, away from tl 626A001 - DTR, DTR che	he rib.		·	Inboard Trailing Ed	ge Flap (11E
	ACCESS NO	OTE: Access requires re	moval of inspa	ır skin panel.			
57-669-00-02	AWL	57-05-02-211-853	1.1	56000 FC	18000 FC	ALL	ALL
		ailed) the front spar uppe FA 73 to 167, away from t		edge on the inboa	ard main flap from	Inboard Trailing Ed	ge Flap (ITE
	See Doc. D6	626A001 - DTR, DTR che	ck form 57-53	-12-1, for alternativ	e inspection.		
	ACCESS NO	OTE: Access requires re	moval of inspa	r skin panel.			
57-669-10-01	AWL	57-05-02-250-885	1.1	56000 FC	36000 FC	ALL	ALL
	from Inboard	v Frequency Eddy Curren d Trailing Edge Flap (ITER	=) main flap S⁻		rib location.	ng edge and the inb	ooard main fla
		520A001 - DTIX, DTIX CHE		,			
57-669-10-02	AWL	57-05-02-250-885	1.1	56000 FC	36000 FC	ALL	ALL
57-669-10-02	Inspect (Low from Inboard		1.1 it) the front spa i) main flap S	56000 FC ar upper and lower FA 73 to 167, at the	36000 FC chord on the traili		
57-669-10-02 57-669-20-01	Inspect (Low from Inboard	57-05-02-250-885 v Frequency Eddy Curren d Trailing Edge Flap (ITEF	1.1 it) the front spa i) main flap S	56000 FC ar upper and lower FA 73 to 167, at the	36000 FC chord on the traili		

Edge Flap (ITEF) main flap STA 155.00.

See Doc. D626A001 - DTR, DTR check form 57-53-12-3, for alternative inspection.

ACCESS NOTE: Nose skin over the cutout must be removed, and flaps deployed.





				APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-669-20-02	AWL	57-05-02-210-820	1.1	56000 FC	18000 FC	ALL	ALL
	Edge Flap (I	eral Visual) the front spa TEF) main flap STA 155. 26A001 - DTR, DTR cho	.00.			ard main flap from I	nboard Trailing
	ACCESS NO	OTE: Nose skin over the	e cutout must b	e removed, and fla	ps deployed.		
57-670-00-01	AWL	57-05-02-211-855	1.1	56000 FC	9000 FC	ALL	ALL
	flap at Inboa	ailed) the upper and low rd Trailing Edge Flap (IT 26A001 - DTR, DTR cho	EF) and main t	flap STA 85.225 and	d 143.3.	out on the trailing ed	ge inboard ma
	ACCESS NO	OTE: Lower inspar skin	panel removal	is required.			
57-670-00-02	AWL	57-05-02-211-855	1.1	56000 FC	9000 FC	ALL	ALL
	flap at Inboa	ailed) the upper and low rd Trailing Edge Flap (IT 26A001 - DTR, DTR cho	EF) and main t	flap STA 85.225 and	d 143.3.	out on the trailing ed	ge inboard ma
	ACCESS NO	OTE: Lower inspar skin	panel removal	is required.			
57-670-10-01	AWL	57-05-02-211-857	1.1	56000 FC	9000 FC	ALL	ALL
	pushrod cuto	ailed) the horizontal and out on the trailing edge, i 26A001 - DTR, DTR cho	nboard main fla	ap at Inboard Trailir	ng Edge Flap (ITE		
	ACCESS NO	OTE: Skin panel remova	al is required.				
57-670-10-02	AWL	57-05-02-211-857	1.1	56000 FC	9000 FC	ALL	ALL
	pushrod cuto	ailed) the horizontal and out on the trailing edge, i 26A001 - DTR, DTR cho	nboard main fla	ap at Inboard Trailir	ng Edge Flap (ITE		
	ACCESS NO	OTE: Skin panel remova	al is required.				
57-670-11-01	AWL	57-05-02-211-858	1.1	56000 FC	9000 FC	ALL	ALL
	pushrod cuto	ailed) the horizontal and out on the trailing edge, i 26A001 - DTR, DTR cho	nboard main fla	ap at Inboard Trailir	ng Edge Flap (ITE		
	ACCESS NO	OTE: Flap deployment is	s required.				
57-670-11-02	AWL	57-05-02-211-858	1.1	56000 FC	9000 FC	ALL	ALL
		silad) the besisental and				anda at tha inhaard	

Inspect (Detailed) the horizontal and vertical flanges of the rear spar upper and lower chords at the inboard and outboard pushrod cutout on the trailing edge, inboard main flap at Inboard Trailing Edge Flap (ITEF) main flap STA 91.5 and 149.0. See Doc. D626A001 - DTR, DTR check form 57-53-13-2, for alternative inspection.

ACCESS NOTE: Flap deployment is required.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-670-20-01	AWL	57-05-02-211-861	1.1	56000 FC	9000 FC	ALL	ALL
	and main flag	ailed) the rear spar uppe p STA 73 to 167, away fr 26A001 - DTR, DTR che	om the rib.		·	Inboard Trailing Ed	ge Flap (ITEF
	ACCESS NO	OTE: Removal of skin pa	anels is require	ed.			
57-670-20-02	AWL	57-05-02-211-861	1.1	56000 FC	9000 FC	ALL	ALL
	and main flag	ailed) the rear spar uppe p STA 73 to 167, away fr 26A001 - DTR, DTR che	om the rib.		·	Inboard Trailing Ed	ge Flap (ITEF
	ACCESS NO	OTE: Removal of skin pa	anels is require	ed.			
57-670-30-01	AWL	57-05-02-250-887	1.1	56000 FC	36000 FC	ALL	ALL
	Trailing Edge	r Frequency Eddy Currer e Flap (ITEF) and main fl 226A001 - DTR, DTR che	lap STA 73 to 1	67 at the rib.		nboard main flap fro	n Inboard
57-670-30-02	AWL	57-05-02-250-887	1.1	56000 FC	36000 FC	ALL	ALL
	Trailing Edge	r Frequency Eddy Currer e Flap (ITEF) and main fl 26A001 - DTR, DTR che	lap STA 73 to 1	67 at the rib.		nboard main flap fro	m Inboard
57-671-00-01	AWL	57-05-02-211-863	1.1	56000 FC	9000 FC	ALL	ALL
	STA 73 to 16	ailed) the skin at the rear 37. 326A001 - DTR, DTR che				g Edge Flap (ITEF)	and main flap
57-671-00-02	AWL	57-05-02-211-863	1.1	56000 FC	9000 FC	ALL	
	Inspect (Deta						ALL
	STA 73 to 16	ailed) the skin at the rear 37.	r spar chord, in	board main flap from	m Inboard Trailin	g Edge Flap (ITEF)	
		,	•	•		g Edge Flap (ITEF)	
57-672-00-01		67. ´	•	•		g Edge Flap (ITEF) ALL	
57-672-00-01	AWL Inspect (High See Doc. D6 The NDI met	57. S26A001 - DTR, DTR che 57-05-02-250-889 In Frequency Eddy Curre 526A001 - DTR, DTR che thod(s) necessary to acc	1.1 nt) the inboard eck form 57-53	56000 FC main flap, inboard -17, for alternative i	nspection. 18000 FC torque tube rib of torque tube rib of the name of the nam	ALL n the lower chord.	and main flap
57-672-00-01	AWL Inspect (High See Doc. D6 The NDI met (D6-37239).	57. S26A001 - DTR, DTR che 57-05-02-250-889 In Frequency Eddy Curre 526A001 - DTR, DTR che	1.1 nt) the inboard eck form 57-53 complish the interest are contain	-14, for alternative i 56000 FC main flap, inboard -17, for alternative i ent of this inspection ed in Part 6, Section	nspection. 18000 FC torque tube rib ornspection. on is contained in n 57-50-27.	ALL n the lower chord.	and main flap

Inspect (High Frequency Eddy Current) the inboard main flap, inboard torque tube rib on the lower chord.

See Doc. D626A001 - DTR, DTR check form 57-53-17, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-50-27.

ACCESS NOTE: Removal of flap lower skin is required for inspection.





				INTERVAL		APPLIC	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
57-673-00-01	AWL	57-05-02-210-822	1.1	56000 FC	18000 FC	ALL	ALL		
	Inspect (Ge	neral Visual) the inboard	main flap torqu	ie tube on all of the	exposed torque	tube surfaces.			
	See Doc. D626A001 - DTR, DTR check form 57-53-20, for alternative inspection.								
57-673-00-02	AWL	57-05-02-210-822	1.1	56000 FC	18000 FC	ALL	ALL		
		neral Visual) the inboard 626A001 - DTR, DTR cho				tube surfaces.			
57-674-00-01	AWL	57-05-02-250-891	1.1	56000 FC	36000 FC	ALL	ALL		
	support ass	h Frequency Eddy Curre emblies at the Inboard Tr 626A001 - DTR, DTR cho	ailing Edge Fla	p (ITEF) and main	flap STA 85 and		aft flap track		
	The NDI me	ethod(s) necessary to acc The inspection procedu	complish the int	ent of this inspection	on is contained in	the 737 Nondestru	ctive Test Manu		
	ACCESS N	OTE: Lower skin panel r flap tracks to exter			rack pushrods dis	sconnected to allow	the aft		
57-674-00-02	AWL	57-05-02-250-891	1.1	56000 FC	36000 FC	ALL	ALL		
	support asso See Doc. Do The NDI me	th Frequency Eddy Curre emblies at the Inboard Tr 626A001 - DTR, DTR che ethod(s) necessary to acc The inspection procedur	railing Edge Fla eck form 57-53 complish the int	p (ITEF) and main -21, for alternative i ent of this inspectio	flap STA 85 and inspection. on is contained in	143.			
	ACCESS N	OTE: Lower skin panel r flap tracks to exter			rack pushrods dis	sconnected to allow	the aft		
57-675-00-01	AWL	57-05-02-250-893	1.1	56000 FC	36000 FC	ALL	ALL		
		h Frequency Eddy Curre 626A001 - DTR, DTR cho				y.			
	ACCESS N	OTE: The flaps must be removed to gain a		ain inspection acces and Outbd side			ust be		
57-675-00-02	AWL	57-05-02-250-893	1.1	56000 FC	36000 FC	ALL	ALL		
		th Frequency Eddy Curre 626A001 - DTR, DTR che		-	-	y.			
	ACCESS N	OTE: The flaps must be		ain inspection acces		•	ust be		
		removed to gain a							
57-676-00-01	AWL	57-05-02-250-895	1.1	56000 FC	18000 FC	ALL	ALL		

See Doc. D626A001 - DTR, DTR check form 57-53-23-1, alternative inspection.

ACCESS NOTE: The side of body fairing must be removed to gain access. In addition, the aft roller cartridge assembly must be removed to gain access to the inboard and outboard side surfaces of the aft roller lug.







				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
57-676-00-02	AWL	57-05-02-250-895	1.1	56000 FC	18000 FC	ALL	ALL		
	Inspect (High Frequency Eddy Current) the forward and aft lugs on the carriage plates. See Doc. D626A001 - DTR, DTR check form 57-53-23-1, alternative inspection.								
	ACCESS NO	The side of body for assembly must be roller lug.				the aft roller cartrid rd side surfaces of			
57-676-10-01	AWL	57-05-02-211-865	1.1	NOTE		ALL	ALL		
		ailed) the torque tube att			0 .				
	INTERVAL N	NOTE: 56000 FC thresh 1-4020) that hav		to all airplanes (L/N SB 737-57A-1314.	,	. ,	L/N		
		32000 FC thresh 57A-1314. Repe	• • •	. ,	-4020) that have r	not incorporated SB	737-		
57-676-10-02	AWL	57-05-02-211-865	1.1	NOTE		ALL	ALL		
		ailed) the torque tube att			0 .				
	INTERVAL N	NOTE: 56000 FC thresh 1-4020) that hav		to all airplanes (L/N SB 737-57A-1314.	,	. ,	L/N		
		32000 FC thresh 57A-1314. Repe		. ,	-4020) that have r	not incorporated SB	737-		
57-676-20-01	AWL	57-05-02-211-867	1.1	56000 FC	9000 FC	ALL	ALL		
		ailed) the aft upper chord			•				
	ACCESS NO	OTE: Removal of rub pa	d attachment f	astener is required.					
57-676-20-02	AWL	57-05-02-211-867	1.1	56000 FC	9000 FC	ALL	ALL		
		ailed) the aft upper chord 26A001 - DTR, DTR che			•				
	ACCESS NO	OTE: Removal of rub pa	d attachment f	astener is required.					
57-677-00-01	AWL	57-05-02-250-897	1.1	56000 FC	18000 FC	ALL	ALL		
	To a second di Pari	- Francisco as Eddy Curra	() (lb (b)	haranda a meta manan da Hi		th	- L M/DL 057		

Inspect (High Frequency Eddy Current) the four outboard carriage roller bosses on the outboard main flap at WBL 357.7. See Doc. D626A001 - DTR, DTR check form 57-53-24-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-50-23.

ACCESS NOTE: Flaps must be deployed and roller pin assemblies removed.





				INTERVAL		APPLIC	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-677-00-02	AWL	57-05-02-250-897	1.1	56000 FC	18000 FC	ALL	ALL
		h Frequency Eddy Curre 326A001 - DTR, DTR che	,	•		outboard main flap	at WBL 357.7.
		thod(s) necessary to acc		*	•	the 737 Nondestru	ctive Test Manu
		The inspection procedur		•		tile 737 Noritiestitu	ctive rest Maria
	ACCESS NO	OTE: Flaps must be dep	loyed and rolle	er pin assemblies re	emoved.		
57-677-10-01	AWL	57-05-02-250-899	1.1	56000 FC	18000 FC	ALL	ALL
	aft bridge su	h Frequency Eddy Curre pport on the outboard m	ain flap at WBI	_ 357.7.		dge fitting at the ou	tboard carriage
		626A001 - DTR, DTR che		•		th = 707 N = = d = = t	-ti Tt N4
		thod(s) necessary to acc The inspection procedur				the 737 Nondestru	ctive lest Manu
	ACCESS NO	OTE: The flaps must be	deployed and	the aft bridge fitting	g and bolts must b	e removed for insp	ection.
57-677-10-02	AWL	57-05-02-250-899	1.1	56000 FC	18000 FC	ALL	ALL
	The NDI me (D6-37239).	626A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur DTE: The flaps must be	complish the intress are contain	ent of this inspection ed in Part 6, Section	on is contained in on 57-50-23.		
57-678-00-01	AWL	57-05-02-250-901	1.1	56000 FC	18000 FC	ALL	ALL
37-078-00-01							
		h Frequency Eddy Curre 326A001 - DTR, DTR che	,	•		utboard main fiap a	t WBL 254.0.
	The NDI me	thod(s) necessary to acc	complish the int	ent of this inspection	on is contained in	the 737 Nondestru	ctive Test Manu
	,	The inspection procedur					
	ACCESS NO	OTE: The flaps must be	deployed and	roller piri assemblik	es removed.		
57-678-00-02	AWL	57-05-02-250-901	1.1	56000 FC	18000 FC	ALL	ALL
	Inspect (High	h Frequency Eddy Curre	nt) the four inb	oard carriage roller	r bosses on the ou	utboard main flap a	t WBL 254.0.
		326A001 - DTR, DTR che thod(s) necessary to acc		*	•	the 727 Nondoctru	otivo Toot Manu
		The inspection procedur		•		tile 737 Noridestid	clive lest Mariu
	ACCESS NO	OTE: The flaps must be	deployed and	roller pin assemblie	es removed.		
57-678-10-01	AWL	57-05-02-250-903	1.1	56000 FC	18000 FC	ALL	ALL
					<u> </u>		

Inspect (High Frequency Eddy Current) the holes in the carriage which attach the aft bridge fitting on the outboard main flap at WBL 254.0.

See Doc. D626A001 - DTR, DTR check form 57-53-25-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 57-50-24.

ACCESS NOTE: The flaps must be deployed and the aft bridge fitting and bolts must be removed for the inspection.







				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-678-10-02	AWL	57-05-02-250-903	1.1	56000 FC	18000 FC	ALL	ALL			
		Inspect (High Frequency Eddy Current) the holes in the carriage which attach the aft bridge fitting on the outboard main fla at WBL 254.0.								
	The NDI me	326A001 - DTR, DTR cho thod(s) necessary to acc The inspection procedu	complish the int	ent of this inspection	on is contained in	the 737 Nondestrue	ctive Test Manu			
	ACCESS NO	OTE: The flaps must be	deployed and	the aft bridge fitting	g and bolts must b	e removed for the i	nspection.			
57-679-00-01	AWL	57-05-02-250-905	1.1	56000 FC	9000 FC	ALL	ALL			
	at the ribs fro	v Frequency Eddy Currel om WBL 254 to WBL 358 326A001 - DTR, DTR che	3.			o WBL 327 and the	spar lower cho			
		OTE: Deployment of flag		·	·					
57-679-00-02	AWL	57-05-02-250-905	1.1	56000 FC	9000 FC	ALL	ALL			
	at the ribs fro	v Frequency Eddy Curre om WBL 254 to WBL 358 326A001 - DTR, DTR cho	3.			o WBL 327 and the	spar lower ch			
	ACCESS NO	OTE: Deployment of flap	os provides acc	ess at the carriage	e support ribs.					
57-680-00-01	AWL	57-05-02-250-907	1.1	56000 FC	36000 FC	ALL	ALL			
	lower chord	h Frequency Eddy Curre between the ribs from W 526A001 - DTR, DTR cho	BL 254 to WBL	_ 358.		L 280 to WBL 327 a	and the spar			
	ACCESS NO	OTE: Removal of the sk	in is required.							
57-680-00-02	AWL	57-05-02-250-907	1.1	56000 FC	36000 FC	ALL	ALL			
	lower chord	h Frequency Eddy Curre between the ribs from W 326A001 - DTR, DTR che	BL 254 to WBI	_ 358.		L 280 to WBL 327 a	and the spar			
	ACCESS NO	OTE: Removal of the sk	in is required.		·					
57-681-00-01	AWL	57-05-02-250-909	1.1	56000 FC	36000 FC	ALL	ALL			
		-: 00 02 200 000	••••							

Inspect (High Frequency Eddy Current) the aft flap track cutouts 2 & 3 at the rear spar outboard main flap.

See Doc. D626A001 - DTR, DTR check form 57-53-27-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 57-50-21.

ACCESS NOTE: Internal access is required. Removal of the flap upper skin is required to gain access. The aft flap track pushrods need to be disconnected to allow the aft flap tracks to extend to their stops.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-681-00-02	AWL	57-05-02-250-909	1.1	56000 FC	36000 FC	ALL	ALL
	See Doc. D6 The NDI met	n Frequency Eddy Currer 26A001 - DTR, DTR che thod(s) necessary to acc The inspection procedur	eck form 57-53 omplish the int	-27-1, for alternative	ve inspection. on is contained in	•	ctive Test Manu
	ACCESS NO	OTE: Internal access is r track pushrods nee	•			I to gain access. Th extend to their stop	
57-682-00-01	AWL	57-05-02-250-911	1.1	56000 FC	18000 FC	ALL	ALL
	lower chord	r Frequency Eddy Currer at the ribs from WBL 254 26A001 - DTR, DTR che	to WBL 358.			280 to WBL 327 and	d the rear spar
	ACCESS NO	OTE: Deployment of flap	s provides acc	cess at the carriage	e support ribs.		
57-682-00-02	AWL	57-05-02-250-911	1.1	56000 FC	18000 FC	ALL	ALL
	lower chord	r Frequency Eddy Currer at the ribs from WBL 254 26A001 - DTR, DTR che	to WBL 358.			280 to WBL 327 and	d the rear spar
	ACCESS NO	OTE: Deployment of flap	s provides acc	cess at the carriage	e support ribs.		
57-683-00-01	AWL	57-05-02-250-913	1.1	56000 FC	36000 FC	ALL	ALL
	spar lower cl	n Frequency Eddy Currer nord between the ribs fro 26A001 - DTR, DTR che	m WBL 254 to	WBL 358.		n WBL 280 to WBL 3	327 and the rea
	ACCESS NO	OTE: Removal of flap up	per skin panel	is required for acc	ess.		
57-683-00-02	AWL	57-05-02-250-913	1.1	56000 FC	36000 FC	ALL	ALL
	spar lower cl	n Frequency Eddy Currer nord between the ribs fro 26A001 - DTR, DTR che	m WBL 254 to	WBL 358.		n WBL 280 to WBL 3	327 and the rea
	ACCESS NO	OTE: Removal of flap up	per skin panel	is required for acc	ess.		
57-684-00-01	AWL	57-05-02-250-915	1.1	56000 FC	36000 FC	ALL	ALL
	fastener loca	n Frequency Eddy Curre ations at both the front an 26A001 - DTR, DTR che	ıd rear spar.		•	ower skin at the skin	n-to-chord
	ACCESS NO	OTE: Deployment of flap	s provides acc	cess at the carriage	e support ribs.		
57-684-00-02	AWL	57-05-02-250-915	1.1	56000 FC	36000 FC	ALL	ALL
	fastener loca	n Frequency Eddy Currentions at both the front and 126A001 - DTR, DTR che	ıd rear spar.		•	ower skin at the skin	n-to-chord

See Doc. D626A001 - DTR, DTR check form 57-53-29, for alternative inspection.

ACCESS NOTE: Deployment of flaps provides access at the carriage support ribs.





				INTERVAL		APPLICA	ABILITY			
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-685-00-01	AWL	57-05-02-240-801	1.1	56000 FC	18000 FC	ALL	ALL			
		gnetic Particle) the front a 226A001 - DTR, DTR ch		•	•	it WBL 254.0 and W	BL 358.0.			
	ACCESS NOTE: Access requires removal of the flap from the carriages.									
57-685-00-02	AWL	57-05-02-240-801	1.1	56000 FC	18000 FC	ALL	ALL			
		gnetic Particle) the front 326A001 - DTR, DTR ch		•	•	it WBL 254.0 and W	BL 358.0.			
	ACCESS NO	OTE: Access requires re	emoval of the fl	ap from the carriage	es.					
57-686-00-01	AWL	57-05-02-211-877	1.1	56000 FC	9000 FC	ALL	ALL			
	Inspect (Deta 358.0.	ailed) the carriage aft lin	k assembly (aft	spigot fitting, aft lir	nk, aft bridge fittin	g clevis, aft pin) at \	WBL 254.0 a			
	See Doc. D6	26A001-DTR, DTR che	ck form 57-53-3	32, for alternative in	spections.					
	ACCESS NO	DTE: Flap deployment is	s required to ga	ain access.						
57-686-00-02	AWL	57-05-02-211-877	1.1	56000 FC	9000 FC	ALL	ALL			
	Inspect (Deta 358.0.	ailed) the carriage aft lin	k assembly (aft	spigot fitting, aft lir	nk, aft bridge fittin	g clevis, aft pin) at \	WBL 254.0 a			
	See Doc. D6	26A001-DTR, DTR che	ck form 57-53-3	32, for alternative in	spections.					
	ACCESS NO	OTE: Flap deployment is	s required to ga	ain access.						
57-687-00-01	AWL	57-05-02-211-871	1.1	56000 FC	9000 FC	ALL	ALL			
		ailed) the lower flanges				from WBL 254.0 an	d 358.0.			
		326A001 - DTR, DTR ch			•					
	ACCESS NO	OTE: Flaps deployed ar	id the lower cov	ve parieis removed	ioi access.					
57-687-00-02	AWL	57-05-02-211-871	1.1	56000 FC	9000 FC	ALL	ALL			
		ailed) the lower flanges				from WBL 254.0 an	d 358.0.			
		326A001 - DTR, DTR ch		,	•					
	ACCESS NO	OTE: Flaps deployed ar	id the lower co	ve panels removed	for access.					
57-688-00-01	AWL	57-05-02-211-873	1.1	56000 FC	18000 FC	ALL	ALL			
		ailed) the lugs on both h	•	•						
		326A001 - DTR, DTR ch			inspection.					
	ACCESS NO	OTE: Access requires d	eployment of in	board flap.						
57-688-00-02	AWL	57-05-02-211-873	1.1	56000 FC	18000 FC	ALL	ALL			

See Doc. D626A001 - DTR, DTR check form 57-70-01, for alternative inspection.

ACCESS NOTE: Access requires deployment of inboard flap.







				APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-800-00-01	MRB	05-41-01-210-816	1.1 1.2	36000 FC 10 YR	36000 FC 10 YR	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the cer	nter section wing bo	ox - section 44, st	a 540 to sta 663.75	•
	INTERVAL N	NOTE: Whichever come	s first.				
57-802-01-01	MRB	05-41-05-210-801	1.1	120 DY	120 DY	ALL	ALL
		external zonal inspection Iders. No additional acce	` '	•	s accomplished fr	om the ground, with	nout the use o
	ACCESS NO	OTE: Control surfaces ex	xtended.				
57-804-01-01	MRB	05-41-05-210-804	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the lea	ading edge to front	spar - inboard of	nacelle strut - left w	ing.
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Excluding surface	under wing to	body fairing.			
57-806-01-01	MRB	05-41-05-210-805	1.1	6600 FC	6600 FC	ALL	ALL
57-806-01-01			1.2	36 MO	36 MO		
57-806-01-01		05-41-05-210-805	1.2	36 MO	36 MO		
57-806-01-01	Perform an i		1.2 (GV) of the lease s first. The EZA	36 MO ding edge to front s AP inspection requi	36 MO spar - inboard of r	nacelle strut - left wi	ng. (EZAP)
57-806-01-01	Perform an i	nternal zonal inspection	1.2 (GV) of the lease s first. The EZA	36 MO ding edge to front s AP inspection requi	36 MO spar - inboard of r	nacelle strut - left wi	ng. (EZAP)
57-806-01-01 57-808-01-01	Perform an i	nternal zonal inspection of NOTE: Whichever come satisfied by this z	1.2 (GV) of the lease s first. The EZA	36 MO ding edge to front s AP inspection requi	36 MO spar - inboard of r	nacelle strut - left wi	ng. (EZAP)
	Perform an i	nternal zonal inspection (NOTE: Whichever come satisfied by this zonate: Flaps deployed.	1.2 (GV) of the lease first. The EZD conal inspection	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO	36 MO spar - inboard of r rement with inten	nacelle strut - left wi val 18000 FC/6 YR	ng. (EZAP) is
	Perform an i INTERVAL N ACCESS NO MRB Perform an e	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal. OTE: Flaps deployed. 05-41-05-210-806	1.2 (GV) of the least street. The EZ conal inspection 1.1 1.2 (GV) of the Kr	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO	36 MO spar - inboard of r rement with inten	nacelle strut - left wi val 18000 FC/6 YR	ng. (EZAP) is
	Perform an i INTERVAL N ACCESS NO MRB Perform an e	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal inspection of the notation of the not	1.2 (GV) of the least street. The EZI tonal inspection 1.1 1.2 (GV) of the Kr street.	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO ueger flaps no. 1 a	36 MO spar - inboard of rement with intervented from FC 36 MO and 2 - left wing.	nacelle strut - left wi val 18000 FC/6 YR	ng. (EZAP) is
57-808-01-01	Perform an i INTERVAL N ACCESS NO MRB Perform an e INTERVAL N	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal inspection. 05-41-05-210-806 external zonal inspection. NOTE: Whichever come.	1.2 (GV) of the lease s first. The EZ conal inspection 1.1 1.2 (GV) of the Kr s first.	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO ueger flaps no. 1 a	36 MO spar - inboard of r rement with intended from the following of the following of the following follow	nacelle strut - left wi val 18000 FC/6 YR ALL	ng. (EZAP) is ALL
57-808-01-01	Perform an i INTERVAL N ACCESS NO MRB Perform an e INTERVAL N MRB Perform an i	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal inspection. 05-41-05-210-806 external zonal inspection. NOTE: Whichever come. 05-41-05-210-807 Internal zonal inspection.	1.2 (GV) of the least street. The EZ tonal inspection 1.1 1.2 (GV) of the Kr street. 1.1 1.2 (GV) of the Kr street. 1.1 1.2	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO ueger flaps no. 1 a	36 MO spar - inboard of r rement with intended from the following of the following of the following follow	nacelle strut - left wi val 18000 FC/6 YR ALL	ng. (EZAP) is ALL
57-808-01-01	Perform an i INTERVAL N ACCESS NO MRB Perform an e INTERVAL N MRB Perform an i	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal inspection. 05-41-05-210-806 external zonal inspection. NOTE: Whichever come. 05-41-05-210-807 Internal zonal inspection of the satisfied by this zonal.	1.2 (GV) of the least street. The EZ tonal inspection 1.1 1.2 (GV) of the Kr street. 1.1 1.2 (GV) of the Kr street. 1.1 1.2	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO ueger flaps no. 1 a	36 MO spar - inboard of r rement with intended from the following of the following of the following follow	nacelle strut - left wi val 18000 FC/6 YR ALL	ng. (EZAP) is ALL
57-808-01-01	Perform an i INTERVAL N ACCESS NO MRB Perform an e INTERVAL N MRB Perform an i	nternal zonal inspection of NOTE: Whichever come satisfied by this zonal inspection. 05-41-05-210-806 external zonal inspection. NOTE: Whichever come. 05-41-05-210-807 Internal zonal inspection.	1.2 (GV) of the least street. The EZ tonal inspection 1.1 1.2 (GV) of the Kr street. 1.1 1.2 (GV) of the Kr street. 1.1 1.2	36 MO ding edge to front s AP inspection requin. 6600 FC 36 MO ueger flaps no. 1 a	36 MO spar - inboard of r rement with intended from the following of the following of the following follow	nacelle strut - left wi val 18000 FC/6 YR ALL	ng. (EZAP) is ALL

Perform an external zonal inspection (GV) of the leading edge to front spar - outboard of nacelle strut - left wing.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Slats deployed.







TASK CARD NO.				INTERVAL		APPLICA	ABILITY	
	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
57-814-01-01	MRB	05-41-05-210-809	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL	
	Perform an i	nternal zonal inspection	(GV) of the lea	ding edge to front s	spar - outboard of	nacelle strut - left v	ving. (EZAP)	
	INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.							
	ACCESS NO	OTE: Slats deployed.						
57-816-01-01	MRB	05-41-05-210-810	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL	
	Perform an e	external zonal inspection	(GV) of the sla	ats no. 1, 2, 3, 4 - le	eft wing.			
	INTERVAL N	NOTE: Whichever come	s first.					
	ACCESS NO	OTE: Slats in full extend	position.					
57-822-01-01	MRB	05-41-05-210-813	1.1 1.2	36000 FC 10 YR	36000 FC 10 YR	ALL	ALL	
	Perform an i	nternal zonal inspection	(GV) of the cer	nter fuel tank - left v	ving.			
	INTERVAL N	NOTE: Whichever come	s first.					
57-824-01-01	MRB	05-41-05-210-814	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL	
	Perform an e	external zonal inspection	(GV) of the ce	nter fuel tank - left	wing.			
		external zonal inspection NOTE: Whichever come	,	nter fuel tank - left	wing.			
	INTERVAL N	•	s first.		wing.			
57-826-01-01	INTERVAL N	NOTE: Whichever come	s first.		36000 FC 10 YR	ALL	ALL	
57-826-01-01	MRB	NOTE: Whichever come OTE: Excluding surface	s first. under wing to l 1.1 1.2	obody fairing. 36000 FC 10 YR	36000 FC 10 YR	ALL	ALL	
57-826-01-01	MRB Perform an i	NOTE: Whichever come OTE: Excluding surface 05-41-05-210-815	s first. under wing to l 1.1 1.2 (GV) of the ma	obody fairing. 36000 FC 10 YR	36000 FC 10 YR	ALL	ALL	
57-826-01-01	MRB Perform an i	NOTE: Whichever come OTE: Excluding surface 05-41-05-210-815 Internal zonal inspection	s first. under wing to l 1.1 1.2 (GV) of the mass first.	36000 FC 10 YR in fuel tank - left wi	36000 FC 10 YR ng.	ALL	ALL	
57-826-01-01 57-828-01-01	MRB Perform an i	NOTE: Whichever come OTE: Excluding surface 05-41-05-210-815 Internal zonal inspection of the come NOTE: Whichever come	1.1 1.2 (GV) of the mass first.	36000 FC 10 YR in fuel tank - left wi and 532BZ at the sa	36000 FC 10 YR ng. ame time.	ALL	ALL	
	MRB Perform an i INTERVAL N ACCESS NO	NOTE: Whichever come OTE: Excluding surface 05-41-05-210-815 Internal zonal inspection of the come OTE: Whichever come OTE: Caution - Do not recome of the come	s first. under wing to l 1.1 1.2 (GV) of the mass first. emove 532AZ a 1.1 1.2	36000 FC 10 YR in fuel tank - left wi and 532BZ at the sa 5500 FC 30 MO	36000 FC 10 YR ng. ame time. 5500 FC 30 MO			
	MRB Perform an i INTERVAL N ACCESS NO MRB Perform an e	NOTE: Whichever come OTE: Excluding surface 05-41-05-210-815 Internal zonal inspection of the come OTE: Whichever come OTE: Caution - Do not recome	1.1 1.2 (GV) of the mass first. emove 532AZ at 1.1 1.2 (GV) of the mass first.	36000 FC 10 YR in fuel tank - left wi and 532BZ at the sa 5500 FC 30 MO	36000 FC 10 YR ng. ame time. 5500 FC 30 MO			

Perform an internal zonal inspection (GV) of the surge tank - left wing.

INTERVAL NOTE: Whichever comes first.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-832-01-01	MRB	05-41-05-210-818	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the su	rge tank - left wing.			
	INTERVAL N	NOTE: Whichever come	s first.				
57-834-01-01	MRB	05-41-05-210-819	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the dry	/ bay - left wing.			
	INTERVAL N	NOTE: Whichever come	s first.				
57-836-01-01	MRB	05-41-05-210-821	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the fai	ring flap support no	. 3 - left wing.		
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Flaps extended.					
57-838-01-01	MRB	05-41-05-210-822	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the fair	ing flap support no.	. 3 - left wing. (EZ	ZAP)	
	INTERVAL N	NOTE: Whichever come satisfied by this z			rement with inter	val 18000 FC/6 YR i	s
	ACCESS NO	OTE: Flaps extended, in	side of fairing.				
57-840-01-01	MRB	05-41-05-210-823	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the fai	ring flap support no	. 2 - left wing.		
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Flaps extended.					
57-842-01-01	MRB	05-41-05-210-824	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an in	nternal zonal inspection	(GV) of the fair	ing flap support no.	. 2 - left wing. (EZ	ZAP)	
		NOTE: Whichever come	s first. The EZ/	AP inspection requi			is
		satistied by this 7					
	ACCESS NO	satisfied by this z	•				

Perform an external zonal inspection (GV) of the fairing flap support no. 1 - left wing.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Flaps extended.





				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-846-01-01	MRB	05-41-05-210-826	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection ((GV) of the fair	ing flap support no	o. 1 - left wing. (EZ	AP)	
	INTERVAL N	NOTE: Whichever comes satisfied by this z			irement with inter	/al 18000 FC/6 YR i	is
	ACCESS NO	OTE: Flaps extended, in:	side of fairing.				
57-848-01-01	MRB	05-41-05-210-827	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the rea	ar spar to landing	gear support bean	n - left wing.	
	INTERVAL N	NOTE: Whichever come	s first.				
	ACCESS NO	OTE: Excluding surface	under wing to b	oody fairing.			
57-850-01-01	MRB	05-41-05-210-828	1.1 1.2	21600 FC 6 YR	21600 FC 6 YR	ALL	ALL
	Perform an i	nternal zonal inspection ((GV) of the rea	r spar to landing g	ear support beam	- left wing.	
	INTERVAL N	NOTE: Whichever come	s first.			-	
57-852-01-01	MRB	05-41-05-210-829	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
57-852-01-01		05-41-05-210-829	1.2	36 MO	36 MO	ALL	ALL
57-852-01-01	Perform an e		1.2 (GV) of the inb	36 MO	36 MO	ALL	ALL
57-852-01-01	Perform an e	external zonal inspection	1.2 (GV) of the inb	36 MO	36 MO	ALL	ALL
57-852-01-01	Perform an e	external zonal inspection	1.2 (GV) of the inb	36 MO	36 MO	ALL	ALL
57-852-01-01 57-854-01-01	Perform an e	external zonal inspection	1.2 (GV) of the inb	36 MO	36 MO	ALL	ALL
	Perform an e INTERVAL N ACCESS NO	external zonal inspection NOTE: Whichever come: OTE: Spoiler raised.	1.2 (GV) of the inb s first.	36 MO poard spoiler no. 6 5500 FC 24 MO	36 MO - left wing. 5500 FC 24 MO		
	Perform an e INTERVAL N ACCESS NO MRB Perform an e	external zonal inspection NOTE: Whichever come: DTE: Spoiler raised. 05-41-05-210-830	1.2 (GV) of the into s first. 1.1 1.2 (GV) of the into	36 MO poard spoiler no. 6 5500 FC 24 MO	36 MO - left wing. 5500 FC 24 MO		
	Perform an e INTERVAL N ACCESS NO MRB Perform an e INTERVAL N	external zonal inspection NOTE: Whichever come: OTE: Spoiler raised. 05-41-05-210-830 external zonal inspection	1.2 (GV) of the into s first. 1.1 1.2 (GV) of the into	36 MO poard spoiler no. 6 5500 FC 24 MO	36 MO - left wing. 5500 FC 24 MO		
	Perform an e INTERVAL N ACCESS NO MRB Perform an e INTERVAL N	external zonal inspection NOTE: Whichever comes OTE: Spoiler raised. 05-41-05-210-830 external zonal inspection NOTE: Whichever comes	1.2 (GV) of the into s first. 1.1 1.2 (GV) of the into	36 MO poard spoiler no. 6 5500 FC 24 MO	36 MO - left wing. 5500 FC 24 MO		
57-854-01-01	Perform an e INTERVAL N ACCESS NO MRB Perform an e INTERVAL N ACCESS NO	external zonal inspection NOTE: Whichever come: DTE: Spoiler raised. 05-41-05-210-830 external zonal inspection NOTE: Whichever come: DTE: Flaps extended. 05-41-05-210-831	1.2 (GV) of the into s first. 1.1 1.2 (GV) of the into s first.	36 MO poard spoiler no. 6 5500 FC 24 MO poard flaps - left with	36 MO - left wing. 5500 FC 24 MO ng.	ALL	ALL
57-854-01-01	Perform an e INTERVAL M ACCESS NO MRB Perform an e INTERVAL M ACCESS NO MRB	external zonal inspection NOTE: Whichever comes OTE: Spoiler raised. 05-41-05-210-830 external zonal inspection NOTE: Whichever comes OTE: Flaps extended.	1.2 (GV) of the into s first. 1.1 1.2 (GV) of the into s first.	36 MO poard spoiler no. 6 5500 FC 24 MO poard flaps - left with	36 MO - left wing. 5500 FC 24 MO ng.	ALL	ALL
57-854-01-01	Perform an e INTERVAL N ACCESS NO MRB Perform an e INTERVAL N ACCESS NO MRB Perform an ii	external zonal inspection NOTE: Whichever comes OTE: Spoiler raised. 05-41-05-210-830 external zonal inspection NOTE: Whichever comes OTE: Flaps extended. 05-41-05-210-831 Internal zonal inspection in	1.2 (GV) of the into s first. 1.1 1.2 (GV) of the into s first.	36 MO poard spoiler no. 6 5500 FC 24 MO poard flaps - left with	36 MO - left wing. 5500 FC 24 MO ng.	ALL	ALL

Perform an external zonal inspection (GV) of the rear spar to trailing edge - outboard of inboard flap - inboard of fixed trailing edge - left wing. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is

satisfied by this zonal inspection.

ACCESS NOTE: Flaps down, spoilers raised.







				INTERVAL		APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-860-01-01	MRB	05-41-05-210-833	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the sp	oilers 1, 2, 3, 4, 5 -	left wing.		
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Spoilers raised.					
57-862-01-01	MRB	05-41-05-210-834	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the ou	tboard flaps - left w	ring.		
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Flaps extended.					
57-864-01-01	MRB	05-41-05-210-835	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the fix	ed trailing edge - le	ft wing.		
	INTERVAL N	IOTE: Whichever come	s first.				
57-866-01-01	MRB	05-41-05-210-836	1.1	10 YR	10 YR	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the fixe	ed trailing edge - lef	t wing.		
57-868-01-01	MRB	05-41-05-210-837	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the fixe	ed trailing edge - lef	t wing. (EZAP)		
	INTERVAL N	NOTE: Whichever come satisfied by this z			rement with inter	val 36000 FC/12 YF	Ris
	ACCESS NO	OTE: Aileron control are	a only.				
57-870-01-01	MRB	05-41-05-210-838	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the ail	eron - left wing.			
	INTERVAL N	IOTE: Whichever come	s first.				
57-872-02-01	MRB	05-41-06-210-801	1.1	120 DY	120 DY	ALL	ALL
		external zonal inspection Iders. No additional acce			is accomplished	from the ground, wi	thout the use
	ACCESS NO	OTE: Control surfaces e	xtended.				
57-874-02-01	MRB	05-41-06-210-804	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Darfarra an a		(C)() of the lea			the control of the control	ada Carabana

Perform an external zonal inspection (GV) of the leading edge to front spar - inboard of the nacelle strut - right wing.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Excluding surface under wing to body fairing.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
57-876-02-01	MRB	05-41-06-210-805	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an in	nternal zonal inspection	(GV) of the lea	ding edge to front	spar - inboard of t	he nacelle strut - riç	ght wing. (EZA
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	val 18000 FC/6 YR	is
	ACCESS NO	OTE: Flaps deployed.					
57-878-02-01	MRB	05-41-06-210-806	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the Kr	ueger flaps no. 3 a	nd 4 - right wing.		
	INTERVAL N	IOTE: Whichever come	s first.				
57-880-02-01	MRB	05-41-06-210-807	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the Kru	leger flaps no. 3 ar	nd 4 - right wing.		
	INTERVAL N	IOTE: Whichever come	s first.		0 0		
	ACCESS NO	OTE: Flaps deployed.					
57-882-02-01	MRB	05-41-06-210-808	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the lea	ading edge to front	spar - outboard o	f the nacelle strut -	right wing.
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Slats deployed.					
57-884-02-01	MRB	05-41-06-210-809	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an ii (EZAP)	nternal zonal inspection	(GV) of the lea	ding edge to front	spar - outboard of	the nacelle strut - r	ight wing.
	INTERVAL N	Note: Whichever come satisfied by this z			irement with inter	val 36000 FC/12 YF	Ris
	ACCESS NO	OTE: Slats deployed.					
57-886-02-01	MRB	05-41-06-210-810	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL
	Perform an e	external zonal inspection	(GV) of the sla	its no. 5, 6, 7, 8 - ri	ight wing.		
	INTERVAL N	IOTE: Whichever come	s first.				
	ACCESS NO	OTE: Slats in full extend	position.				
57-892-02-01	MRB	05-41-06-210-813	1.1 1.2	36000 FC 10 YR	36000 FC 10 YR	ALL	ALL
	Perform an i	nternal zonal inspection	(GV) of the cer	nter fuel tank - right	t wina		

Perform an internal zonal inspection (GV) of the center fuel tank - right wing.

INTERVAL NOTE: Whichever comes first.







TASK CARD NO.	INTERVAL AI									
	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-894-02-01	MRB	05-41-06-210-814	1.1 1.2	5500 FC 30 MO	5500 FC 30 MO	ALL	ALL			
	Perform an external zonal inspection (GV) of the center fuel tank - right wing.									
	INTERVAL NOTE: Whichever comes first.									
	ACCESS NO	OTE: Excluding surface	under wing to I	oody fairing.						
57-896-02-01	MRB	05-41-06-210-815	1.1 1.2	36000 FC 10 YR	36000 FC 10 YR	ALL	ALL			
	Perform an i	nternal zonal inspection	(GV) of the ma	in fuel tank - right	wing.					
	INTERVAL I	NOTE: Whichever come	s first.							
	ACCESS NO	OTE: Caution - Do not re	emove 632AZ a	and 632BZ at the s	same time.					
57-898-02-01	MRB	05-41-06-210-816	1.1	5500 FC	5500 FC	ALL	ALL			
37-030-02-01	WIND	03-41-00-210-010	1.2	30 MO	30 MO	ALL	ALL			
	Perform an external zonal inspection (GV) of the main fuel tank - right wing.									
	INTERVAL NOTE: Whichever comes first.									
== 000 00 04	1400	05.44.00.040.047		20000 50	00000 50	A. I.				
57-900-02-01	MRB	05-41-06-210-817	1.1 1.2	36000 FC 10 YR	36000 FC 10 YR	ALL	ALL			
	Perform an internal zonal inspection (GV) of the surge tank - right wing.									
	INTERVAL I	NOTE: Whichever come	s first.							
57-902-02-01	MRB	05-41-06-210-818	1.1	5500 FC	5500 FC	ALL	ALL			
			1.2	30 MO	30 MO					
	Perform an external zonal inspection (GV) of the surge tank - right wing.									
	INTERVAL NOTE: Whichever comes first.									
				5500 50	5500 50					
57-904-02-01	MRB	05-41-06-210-819	1.1	5500 FC	5500 FC	ALL	ALL			
57-904-02-01			1.2	30 MO	30 MO	ALL	ALL			
57-904-02-01	Perform an e	external zonal inspection	1.2 (GV) of the dry	30 MO		ALL	ALL			
57-904-02-01	Perform an e		1.2 (GV) of the dry	30 MO		ALL	ALL			

Perform an external zonal inspection (GV) of the fairing flap support no. 6 - right wing.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Flaps extended.







		APPLICA	ABILITY								
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
57-908-02-01	MRB	05-41-06-210-822	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an i	Perform an internal zonal inspection (GV) of the fairing flap support no. 6 - right wing. (EZAP)									
	INTERVAL N	NOTE: Whichever come satisfied by this z			irement with inter	val 18000 FC/6 YR	is				
	ACCESS NO	OTE: Flaps extended, in	side of fairing.								
57-910-02-01	MRB	05-41-06-210-823	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an e	external zonal inspection									
		NOTE: Whichever come	,	gap capport	g.						
		OTE: Flaps extended.									
57-912-02-01	MRB	05-41-06-210-824	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an i	nternal zonal inspection	(GV) of the fair	ing flap support no	. 7 - right wing. (E	ZAP)					
	INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.										
	ACCESS NOTE: Flaps extended, inside of fairing.										
57-914-02-01	MRB	05-41-06-210-825	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an e	external zonal inspection	(GV) of the fai	ring flap support no	o. 8 - right wing.						
	INTERVAL NOTE: Whichever comes first.										
	ACCESS NO	OTE: Flaps extended.									
57-916-02-01	MRB	05-41-06-210-826	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an i	Perform an internal zonal inspection (GV) of the fairing flap support no. 8 - right wing. (EZAP)									
	INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.										
	ACCESS NOTE: Flaps extended, inside of fairing.										
57-918-02-01	MRB	05-41-06-210-827	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an e	external zonal inspection				n - right wing					
		NOTE: Whichever come	` ,	a. Spar to larianty y	Jour ouppoit boain						
		OTE: Excluding surface		body fairing.							
57-920-02-01	MRB	05-41-06-210-828	1.1 1.2	21600 FC 6 YR	21600 FC 6 YR	ALL	ALL				
	D. f.	nternal zonal inspection									

 $\label{performan} \mbox{Perform an internal zonal inspection (GV) of the rear spar to landing gear support beam - right wing.}$

INTERVAL NOTE: Whichever comes first.







	INTERVAL APPLIC									
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
57-922-02-01	MRB	05-41-06-210-829	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL			
	Perform an external zonal inspection (GV) of the inboard spoiler no. 7 - right wing.									
	INTERVAL NOTE: Whichever comes first.									
	ACCESS NO	OTE: Spoiler raised.								
57-924-02-01	MRB	05-41-06-210-830	1.1 1.2	5500 FC 24 MO	5500 FC 24 MO	ALL	ALL			
	Perform an e	external zonal inspection	(GV) of the int	ooard flaps - right v	/ing.					
	INTERVAL N	IOTE: Whichever come	s first.							
	ACCESS NO	OTE: Flaps extended.								
57-926-02-01	MRB	05-41-06-210-831	1.1 1.2	21600 FC 6 YR	21600 FC 6 YR	ALL	ALL			
	Perform an i	nternal zonal inspection	(GV) of the inb	oard main flap - ric	ht wina.					
		IOTE: Whichever come	` ,		3					
	ACCESS NOTE: Flaps extended.									
	7100200 110	riano exteriada.								
57-928-02-01	MRB	05-41-06-210-832	1.1	6600 FC	6600 FC	All	ALI			
57-928-02-01	MRB	05-41-06-210-832	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL			
57-928-02-01	Perform an e	05-41-06-210-832 external zonal inspection - right wing. (EZAP)	1.2	36 MO	36 MO					
57-928-02-01	Perform an e	external zonal inspection	1.2 (GV) of the reases first. The EZ/	36 MO ar spar to trailing e AP inspection requ	36 MO dge - outboard of	inboard flap - inboa	rd of fixed			
57-928-02-01	Perform an e trailing edge	external zonal inspection - right wing. (EZAP)	1.2 (GV) of the real series first. The EZ/zonal inspection	36 MO ar spar to trailing e AP inspection requ	36 MO dge - outboard of	inboard flap - inboa	rd of fixed			
57-928-02-01 57-930-02-01	Perform an e trailing edge	external zonal inspection - right wing. (EZAP) IOTE: Whichever come satisfied by this z	1.2 (GV) of the real series first. The EZ/zonal inspection	36 MO ar spar to trailing e AP inspection requ	36 MO dge - outboard of	inboard flap - inboa	rd of fixed			
	Perform an etrailing edge INTERVAL N ACCESS NO	external zonal inspection - right wing. (EZAP) IOTE: Whichever come satisfied by this z OTE: Flaps down, spoile	1.2 (GV) of the real series first. The EZ/conal inspection ers raised.	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO	inboard flap - inboa val 18000 FC/6 YR	rd of fixed			
	Perform an etrailing edge INTERVAL M ACCESS NO MRB Perform an e	external zonal inspection - right wing. (EZAP) NOTE: Whichever come satisfied by this zone: OTE: Flaps down, spoile 05-41-06-210-833	1.2 (GV) of the real street of the street o	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO	inboard flap - inboa val 18000 FC/6 YR	rd of fixed			
	Perform an etrailing edge INTERVAL M ACCESS NO MRB Perform an etalling edge	external zonal inspection - right wing. (EZAP) IOTE: Whichever come satisfied by this z OTE: Flaps down, spoile 05-41-06-210-833	1.2 (GV) of the real street of the street o	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO	inboard flap - inboa val 18000 FC/6 YR	rd of fixed			
	Perform an etrailing edge INTERVAL M ACCESS NO MRB Perform an etalling edge	external zonal inspection - right wing. (EZAP) IOTE: Whichever come satisfied by this z OTE: Flaps down, spoile 05-41-06-210-833 external zonal inspection	1.2 (GV) of the real street of the street o	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO	inboard flap - inboa val 18000 FC/6 YR	rd of fixed			
57-930-02-01	Perform an etrailing edge INTERVAL N ACCESS NO MRB Perform an etalon INTERVAL N ACCESS NO MRB	external zonal inspection - right wing. (EZAP) IOTE: Whichever come satisfied by this z OTE: Flaps down, spoile 05-41-06-210-833 external zonal inspection IOTE: Whichever come OTE: Spoilers Raised	1.2 (GV) of the real set first. The EZ/conal inspection ers raised. 1.1 1.2 (GV) of spoiler set first.	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO rs no. 8, 9, 10, 11, 5500 FC 30 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO 12 - right wing.	inboard flap - inboa val 18000 FC/6 YR ALL	rd of fixed			
57-930-02-01	Perform an etrailing edge INTERVAL M ACCESS NO MRB Perform an etalon MRB MRB Perform an etalon MRB	external zonal inspection - right wing. (EZAP) IOTE: Whichever come satisfied by this z OTE: Flaps down, spoile 05-41-06-210-833 external zonal inspection IOTE: Whichever come OTE: Spoilers Raised 05-41-06-210-834	1.2 (GV) of the real street of the street o	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO rs no. 8, 9, 10, 11, 5500 FC 30 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO 12 - right wing.	inboard flap - inboa val 18000 FC/6 YR ALL	rd of fixed			
57-930-02-01	Perform an etrailing edge INTERVAL N ACCESS NO MRB Perform an etrailing edge INTERVAL N ACCESS NO MRB Perform an etrailing edge INTERVAL N INTERVAL N	external zonal inspection - right wing. (EZAP) NOTE: Whichever come satisfied by this z OTE: Flaps down, spoile 05-41-06-210-833 external zonal inspection NOTE: Whichever come OTE: Spoilers Raised 05-41-06-210-834	1.2 (GV) of the real street of the street o	36 MO ar spar to trailing e AP inspection requ n. 6600 FC 36 MO rs no. 8, 9, 10, 11, 5500 FC 30 MO	36 MO dge - outboard of frement with inter 6600 FC 36 MO 12 - right wing.	inboard flap - inboa val 18000 FC/6 YR ALL	rd of fixed			

Perform an external zonal inspection (GV) of the fixed trailing edge - right wing.

INTERVAL NOTE: Whichever comes first.







	INTERVAL APP										
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
57-936-02-01	MRB 05-41-06-210-836 1.1 10 YR 10 YR ALL ALL										
	Perform an internal zonal inspection (GV) of the fixed trailing edge - right wing.										
57-938-02-01	MRB	05-41-06-210-837	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Dorform on i	Perform an internal zonal inspection (GV) of the fixed trailing edge - right wing. (EZAP)									
		NOTE: Whichever come satisfied by this z	s first. The EZ/	AP inspection requi	, , ,	val 36000 FC/12 YF	? is				
	ACCESS NO	OTE: Aileron control are	a only.								
57-940-02-01	MRB	05-41-06-210-838	1.1 1.2	6600 FC 36 MO	6600 FC 36 MO	ALL	ALL				
	Perform an e	external zonal inspection	(GV) of the ail	eron - right wing.							
	INTERVAL N	NOTE: Whichever come	s first.								
57-950-01-01	MRB	05-41-05-210-802	1.1	36 MO	36 MO	ALL	ALL				
	Perform an e	external zonal inspection	(GV) of the wi	nglet - left wing.							
	Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 57-942-01-0.										
	AIRPLANE	NOTE: Applicable only	on airplanes wi	th winglet installed.							
57-952-01-01	MRB	05-41-05-210-803	1.1	3 YR	3 YR	ALL	ALL				
	Perform an i	nternal zonal inspection	(GV) of the wir	glet - left wing.							
	Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 57-944-01-0.										
	AIRPLANE	NOTE: Applicable only to	to airplanes wit	h winglet installed.							
57-960-02-01	MRB	05-41-06-210-802	1.1	36 MO	36 MO	ALL	ALL				
	Perform an external zonal inspection (GV) of the winglet - right wing. Note: This task satisfies the requirement of the Airplane Partners Boeing (APB) task 57-946-01-0.										
		•			ng (APB) task 57-	-946-01-0.					
	AIRPLANE	NOTE: Applicable only t	to airplanes wit	h winglet installed.							
57-962-02-01	MRB	05-41-06-210-803	1.1	3 YR	3 YR	ALL	ALL				
		nternal zonal inspection ask satisfies the requirem			ng (APB) task 57-	-948-01-0.					
	AIRPLANE	NOTE: Applicable only	to airplanes wit	h winglet installed.							
70-800-01-01	MRB	05-41-04-210-801	1.1	120 DY	120 DY	ALL	ALL				
		external zonal inspection without the use of stands				s. Inspection is acco	omplished fro				
70-802-01-01	MRB	05-41-04-210-802	1.1	2000 FC	2000 FC	ALL	ALL				

Perform an external zonal inspection (GV) of the engine no. 1. (EZAP)

INTERVAL NOTE: The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.







				INTERVAL		APPLICA	ABILITY				
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE				
70-804-01-01	MRB	MRB 05-41-04-210-803 1.1 6600 FC 6600 FC ALL ALL									
	Perform an external zonal inspection (GV) of the fan cowl - engine no. 1.										
	ACCESS NO	OTE: Inner and outer ex	terior surface.								
70-806-01-01	MRB	05-41-04-210-804	1.1	4000 FC	4000 FC	ALL	ALL				
	Perform an e	external zonal inspection	(GV) of the thi	rust reverser - engir	ne no. 1.						
	ACCESS NO	OTE: Thrust reverser in	stowed and de	ployed position.							
70 000 04 04	MDD	05 44 04 040 005	4.4	2000 50	0000 50	A11	A.I.I.				
70-808-01-01	MRB	05-41-04-210-805	1.1	2000 FC	2000 FC	ALL	ALL				
		external zonal inspection									
	INTERVAL N	IOTE : The EZAP insperinspection.	ction requireme	ent with interval 550	00 FC/30 MO is s	atisfied by this zona	ıl				
	ACCESS NO	TE: Inner and outer ex	terior surfaces	-							
70-810-02-01	MRB	05-41-04-210-806	1.1	120 DY	120 DY	ALL	ALL				
	Perform an external zonal inspection (GV) of engine no. 2 and engine no. 2 strut fairings. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required.										
70-812-02-01	MRB	05-41-04-210-807	1.1	2000 FC	2000 FC	ALL	ALL				
	Perform an external zonal inspection (GV) of the engine no. 2. (EZAP)										
	INTERVAL N	IOTE: The EZAP inspection.	ction requireme	ent with interval 550	00 FC/30 MO is s	atisfied by this zona	ıl				
70-814-02-01	MRB	05-41-04-210-808	1.1	6600 FC	6600 FC	ALL	ALL				
	Perform an e	external zonal inspection	(GV) of the far	n cowl - engine no.	2.						
	ACCESS NO	OTE: Inner and outer ex	terior surface.								
70.040.00.04	MDD	05 44 04 040 000	4.4	4000 50	1000 50	ALL	A1.1				
70-816-02-01	MRB	05-41-04-210-809	1.1	4000 FC	4000 FC	ALL	ALL				
	Perform an external zonal inspection (GV) of the thrust reverser - engine no. 2 ACCESS NOTE: Thrust reverser in stowed and deployed position.										
	ACCESS NO	TIE: Thrust reverser in	stowed and de	pioyea position.							
70-818-02-01	MRB	05-41-04-210-810	1.1	2000 FC	2000 FC	ALL	ALL				
	Perform an e	external zonal inspection	(GV) of the thi	rust reverser - engir	ne no. 2. (EZAP)						
	INTERVAL N	IOTE: The EZAP insperinspection.	ction requireme	ent with interval 550	00 FC/30 MO is s	atisfied by this zona	ıl				
	ACCESS NO	OTE: Inner and outer ex	terior surfaces								
71-010-01-01	MRB	F71-11-01-200-801-F00) 1.1	2500 FH	2500 FH	ALL	ALL				
. 1 0 10 -0 1 -0 1		pection of the left inlet co			2000 111	/ \	ALL				

Detailed inspection of the left inlet cowl's inner surface.







				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
71-010-02-01	MRB F71-11-01-200-801-F00 1.1 2500 FH 2500 FH ALL ALL								
	Detailed ins	spection of the right inlet co	wl's inner sur	face.					
71-040-01-01	MRB	54-55-01-200-801 F71-71-00-700-801-F00	1.1	6 YR	6 YR	ALL	ALL		
	Operationa	lly check left engine all drai	n lines.						
71-040-02-01	MRB	54-55-01-200-801 F71-71-00-700-801-F00	1.1	6 YR	6 YR	ALL	ALL		
	Operationa	lly check right engine all dra	ain lines.						
72-020-01-01	MRB	F72-21-00-220-801-F00	1.1	2500 FH	2500 FH	ALL	ALL		
	Detailed ins	spection of left engine inlet	and fan blade	es.					
72-020-02-01	MRB	F72-21-00-220-801-F00	1.1	2500 FH	2500 FH	ALL	ALL		
	Detailed inspection of right engine inlet and fan blades.								
72-025-01-01	MRB	F72-21-00-640-801-F00	1.1 1.2	5000 FH 3000 FC	5000 FH 3000 FC	ALL	ALL		
	Lubricate Left engine fan blades dovetail.								
	INTERVAL NOTE: Whichever comes first.								
72-025-02-01	MRB	F72-21-00-640-801-F00	1.1	5000 FH	5000 FH	ALL	ALL		
. 2 020 02 0.			1.2	3000 FC	3000 FC	,	, , , , ,		
	Lubricate R	Right engine fan blades dov	etail.						
	INTERVAL	NOTE: Whichever comes	first.						
72-030-01-01	MRB	F72-99-99-000-801-F00	1.1	LIF LIM		ALL	ALL		
	Discard left	engine fan disk at life limit.							
72-030-02-01	MRB	F72-99-99-000-801-F00	1.1	LIF LIM		ALL	ALL		
	Discard righ	nt engine fan disk at life lim	it.						
72-040-01-01	MRB	F72-99-99-000-802-F00	1.1	LIF LIM		ALL	ALL		
		engine booster spool at life to the engine shop manua		or life limits.					
72-040-02-01	MRB	F72-99-99-000-802-F00	1.1	LIF LIM		ALL	ALL		
	Discard righ	nt engine booster spool at l	ife limit.						

Discard right engine booster spool at life limit.

Note: Refer to the engine shop manual, Chapter 5 for life limits.







				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
72-050-01-01	MRB	F72-99-99-000-803-F00	1.1	LIF LIM		ALL	ALL
		engine fan shaft at life limit r to the engine shop manua		or life limits.			
72-050-02-01	MRB	F72-99-99-000-803-F00	1.1	LIF LIM		ALL	ALL
	•	nt engine fan shaft at life lim r to the engine shop manua		or life limits.			
72-070-01-01	MRB	F72-20-00-210-801-F00 F72-60-00-200-801-F00 F72-63-00-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL
	Visual chec mounts.	k of the left engine accesso	ory gearbox/ti	ransfer gearbox mo	ount flanges and f	an case and fan fra	me attachme
72-070-02-01	MRB	F72-20-00-210-801-F00 F72-60-00-200-801-F00 F72-63-00-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL
	Visual chec mounts.	k of the right engine access	sory gearbox	transfer gearbox m	nount flanges and	fan case and fan fr	ame attachm
72-100-01-01	MRB	F72-23-04-200-802-F00	1.1	10000 FC	10000 FC	ALL	ALL
	Visual chec	k of the left engine attachm	nent bolts for	the thrust mount fit	tings.		
72-100-02-01	MRB	F72-23-04-200-802-F00	1.1	10000 FC	10000 FC	ALL	ALL
	Visual chec	k of the right engine attach	ment bolts for	r the thrust mount f	ittings.		
72-110-01-01	MRB	F72-23-04-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL
	Visual chec	k of the left engine thrust m	nount fittings.				
72-110-02-01	MRB	F72-23-04-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL
	Visual chec	k of the right engine thrust	mount fittings	S.			
72-130-01-01	MRB	F72-99-99-000-804-F00	1.1	LIF LIM		ALL	ALL
	Discard left	engine stages 1 and 2 spo	ool at manufac	cturer's life limit.			
72-130-02-01	MRB	F72-99-99-000-804-F00	1.1	LIF LIM		ALL	ALL
	Discard righ	nt engine stages 1 and 2 sp	oool at manuf	acturer's life limit.			
72-140-01-01	MRB	F72-99-99-000-805-F00	1.1	LIF LIM		ALL	ALL
	Discard left	engine stage 3 disk at mar	nufacturer's li	fe limit.			
72-140-02-01	MRB	F72-99-99-000-805-F00	1.1	LIF LIM		ALL	ALL

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Discard right engine stage 3 disk at manufacturer's life limit.





				INTERVAL		APPLICA	ABILITY	
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE	
72-150-01-01	MRB	F72-99-99-000-806-F00	1.1	LIF LIM		ALL	ALL	
		engine stages 4 - 9 spool at to the engine shop manua						
72-150-02-01	MRB	F72-99-99-000-806-F00	1.1	LIF LIM		ALL	ALL	
	_	nt engine stages 4 - 9 spool to the engine shop manua						
72-160-01-01	MRB	F72-99-99-000-807-F00	1.1	LIF LIM		ALL	ALL	
		engine front shaft at manuate to the engine shop manua						
72-160-02-01	MRB	F72-99-99-000-807-F00	1.1	LIF LIM		ALL	ALL	
		nt engine front shaft at man to the engine shop manua						
72-170-01-01	MRB	F72-99-99-000-808-F00	1.1	LIF LIM		ALL	ALL	
		engine rear rotating (CDP) to the engine shop manua						
72-170-02-01	MRB	F72-99-99-000-808-F00	1.1	LIF LIM		ALL	ALL	
	_	nt engine rear rotating (CDF to the engine shop manua			t.			
72-180-01-01	MRB	F72-00-00-200-802-F00 F72-00-00-200-805-F00 F72-00-00-200-814-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL	
	Borescope	inspection of the left engine	combustion	chamber.				
	ENGINE N	OTE: Task is for Single Ann	nular Combus	stor engines.				
72-180-02-01	MRB	F72-00-00-200-802-F00 F72-00-00-200-805-F00 F72-00-00-200-814-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL	
	Borescope	inspection of the right engir	ne combustio	n chamber.				
	ENGINE N	OTE: Task is for Single Ann	nular Combus	stor engines.				
72-200-01-01	MRB	F72-00-00-200-802-F00 F72-00-00-200-806-F00 F72-00-00-200-814-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL	
	F72-00-00-200-814-F00 Borescope inspection of the left engine HPT nozzle. (SAC /1 and DAC engines)							
	Borescope		HPT nozzle	. (SAC /1 and DAC	engines)			

Borescope inspection of the left engine HPT nozzle. (Tech Insertion /3 and 7BE engines)







				INTERVAL		APPLICA	ABILITY		
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
72-200-02-01	MRB	F72-00-00-200-802-F00 F72-00-00-200-806-F00 F72-00-00-200-814-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL		
	Borescope	inspection of the right engi	ne HPT nozzl	e. (SAC /1 and DAG	C engines)				
72-200-02-02	MRB	F72-00-00-200-802-F00 F72-00-00-200-814-F00 F72-00-00-200-818-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL		
	Borescope	inspection of the right engi	ne HPT nozzl	e. (Tech Insertion /3	3 and 7BE engine	es)			
72-210-01-01	MRB	F72-00-00-200-802-F00 F72-00-00-200-807-F00 F72-00-00-200-814-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL		
	Borescope	inspection of the left engine	e HPT blades						
72-210-02-01	MRB	F72-00-00-200-802-F00 F72-00-00-200-807-F00 F72-00-00-200-814-F00	1.1 1.2	6600 FC 15000 FH	1600 FC 4000 FH	ALL	ALL		
	Borescope inspection of the right engine HPT blades.								
72-220-01-01	MRB	F72-99-99-000-809-F00	1.1	LIF LIM		ALL	ALL		
		engine HPT front shaft at r to the engine shop manua							
72-220-02-01	MRB	F72-99-99-000-809-F00	1.1	LIF LIM		ALL	ALL		
		nt engine HPT front shaft at r to the engine shop manua							
72-230-01-01	MRB	F72-99-99-000-810-F00	1.1	LIF LIM		ALL	ALL		
		HPT front rotating air seal to the engine shop manua							
72-230-02-01	MRB	F72-99-99-000-810-F00	1.1	LIF LIM		ALL	ALL		
		nt HPT front rotating air sea r to the engine shop manua							
72-240-01-01	MRB	F72-99-99-000-811-F00	1.1	LIF LIM		ALL	ALL		
		engine HPT disk at manufarto the engine shop manua							
72-240-02-01	MRB	F72-99-99-000-811-F00	1.1	LIF LIM		ALL	ALL		
	Discard righ	nt engine HPT disk at manu	ufacturer's life	limit.					

Discard right engine HPT disk at manufacturer's life limit. Note: Refer to the engine shop manual, chapter 5, for life limits.





				INTERVAL		APPLICA	ABILITY
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
72-250-01-01	MRB	F72-99-99-000-812-F00	1.1	LIF LIM		ALL	ALL
		engine HPT rear shaft at n to the engine shop manua					
72-250-02-01	MRB	F72-99-99-000-812-F00	1.1	LIF LIM		ALL	ALL
	•	nt engine HPT rear shaft at to the engine shop manua					
72-270-01-01	MRB	F72-99-99-000-813-F00	1.1	LIF LIM		ALL	ALL
		engine LPT rotor support are to the engine shop manua					
72-270-02-01	MRB	F72-99-99-000-813-F00	1.1	LIF LIM		ALL	ALL
	_	nt engine LPT rotor support to the engine shop manua					
72-280-01-01	MRB	F72-99-99-000-814-F00	1.1	LIF LIM		ALL	ALL
		engine LPT shaft at manufer to the engine shop manua					
72-280-02-01	MRB	F72-99-99-000-814-F00	1.1	LIF LIM		ALL	ALL
	_	nt engine LPT shaft at man to the engine shop manua					
72-290-01-01	MRB	F72-99-99-000-815-F00	1.1	LIF LIM		ALL	ALL
		engine stage 1 LPT disk, s to the engine shop manua	-	-	sk, stage 4 LPT c	lisk at manufacutrer	's life limit.
72-290-02-01	MRB	F72-99-99-000-815-F00	1.1	LIF LIM		ALL	ALL
		nt engine stage 1 LPT disk, to the engine shop manua			disk, stage 4 LPT	disk at manufacutre	er's life limit.
72-300-01-01	MRB	F72-56-00-200-802-F00	1.1	7500 FH	7500 FH	ALL	ALL
	Visual chec	k of the left engine AFT mo	ounts clevis fo	or structural integrity	/ failure.		
72-300-02-01	MRB	F72-56-00-200-802-F00	1.1	7500 FH	7500 FH	ALL	ALL
	Visual chec	k of the right engine AFT m	nount clevis fo	or structural integrity	y failure.		
72-320-01-01	MRB	F79-00-00-200-804-F00	1.1	500 FH	500 FH	ALL	ALL
	- I			100/700			

Detailed inspection of the left engine fwd sump, aft sump, AGB/TGB magnetic chip detectors or debris monitoring system detectors and scavenge screens for particles.







			INTERVAL			APPLICA	ABILITY
ASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE
72-320-02-01	MRB	F79-00-00-200-804-F00	1.1	500 FH	500 FH	ALL	ALL
		spection of the right engine nd scavenge screens for pa	• •	t sump, AGB/TGB	magnetic chip de	tectors or debris mo	nitoring syste
73-010-01-01	MRB	F73-11-02-000-801-F00 F73-11-02-400-801-F00	1.1	6000 FH	6000 FH	ALL	ALL
	Replace the	e left engine fuel filter.					
73-010-02-01	MRB	F73-11-02-000-801-F00 F73-11-02-400-801-F00	1.1	6000 FH	6000 FH	ALL	ALL
	Replace the	e right engine fuel filter.					
73-020-01-01	MRB	F73-21-00-740-803-F00	1.1	150 FH	150 FH	ALL	ALL
	Interrogate	the EMC CDLI for left engir	na faulte				

Interrogate the FMC CDU for left engine faults.

INTERVAL NOTE: A. If any short time faults are found, corrective action for their repair is required immediately. The frequency of this check may be modified provided the new interval plus the time the fault corrective action may be deferred does not exceed 150 hrs total, as required per ATA 05-17-01 of the engine shop manual CFMI-TP.SM.10.

> For example, check recent faults every 70 hrs and fix the reported short time faults within the next 80 hrs.

B. If any long time faults are found, corrective action for their repair is required with 425 hrs. The frequency of this check may be modified provided one half of the new interval plus the time the fault corrective action may be deferred does not exceed 500 hrs total, as required per ATA 05-17-01 of the engine shop manual CFMI-TP.SM.10.

For example, check recent faults every 70 hrs and fix the reported long time faults within the next 465 flight hours.

C. If any economic faults are found, repair is recommended on an opportunity basis.

73-020-02-01

MRB F73-21-00-740-803-F00 150 FH

150 FH

ALL

ALL

Interrogate the FMC CDU for right engine faults.

INTERVAL NOTE: A. If any short time faults are found, corrective action for their repair is required immediately. The frequency of this check may be modified provided the new interval plus the time the fault corrective action may be deferred does not exceed 150 hrs total, as required per ATA 05-17-01 of the engine shop manual CFMI-TP.SM.10.

> For example, check recent faults every 70 hrs and fix the reported short time faults within the next 80 hrs.

B. If any long time faults are found, corrective action for their repair is required with 425 hrs. The frequency of this check may be modified provided one half of the new interval plus the time the fault corrective action may be deferred does not exceed 500 hrs total, as required per ATA 05-17-01 of the engine shop manual CFMI-TP.SM.10.

For example, check recent faults every 70 hrs and fix the reported long time faults within the next 465 flight hours.

C. If any economic faults are found, repair is recommended on an opportunity basis.





	INTERVAL APPLICABIL									
TASK CARD NO.	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE			
73-030-01-01	CMR	F73-21-10-000-801-F00 F73-21-10-200-801-F00 F73-21-10-400-801-F00	1.1	300 FH	300 FH	800	ALL			
	Remove the	e left engine hydro mechan	ical unit for in	spection per Service	ce Bulletin CFM 5	66-7B 73-016.				
	SPECIAL N	IOTE: CMR task (73-CMR	R-01) interval f	or this task is 300 F	FH. See MPD Se	ction 9.				
	ENGINE N	OTE: Applicable to engine	hydro mecha	nical unit P/N 1853	BM56P04 or P/N	1853M56P05.				
73-030-02-01	CMR	F73-21-10-000-801-F00 F73-21-10-200-801-F00 F73-21-10-400-801-F00	1.1	300 FH	300 FH	800	ALL			
	Remove the	e right engine hydro mecha	nical unit for	inspection per Serv	rice Bulletin CFM	56-7B 73-016.				
	SPECIAL N	IOTE: CMR task (73-CMR	R-01) interval f	or this task is 300 f	FH. See MPD Se	ction 9.				
	ENGINE N	OTE: Applicable to engine	hydro mecha	nical unit P/N 1853	BM56P04 or P/N	1853M56P05.				
74-020-01-01	MRB	F74-21-01-200-801-F00	1.1	4000 FC	4000 FC	ALL	ALL			
	Detailed ins	spection of both left engine	ignition leads							
74-020-02-01	MRB	F74-21-01-200-801-F00	1.1	4000 FC	4000 FC	ALL	ALL			
	Detailed inspection of both right engine ignition leads.									
78-011-01-01	MRB	F78-11-00-210-803-F00	1.1	ENG CNG		ALL	ALL			
	Note: This t	spection of the left engine e ask is intended for on-aircr r off-aircraft use or if the sh	aft use or if th	ne short exhaust plu	ug is still installed	on the engine. Refe	er to CMM			
	AIRPLANE NOTE: Applicable to airplanes with exhaust plugs equipped with drain pan and tube system installed.									
	ACCESS N	OTE: Engine exhaust plug	g removal req	uired.						
78-011-02-01	MRB	F78-11-00-210-803-F00	1.1	ENG CNG		ALL	ALL			
	Note: This t	Detailed inspection of the right engine exhaust plug drain pan and tube for condition and security. Note: This task is intended for on-aircraft use or if the short exhaust plug is still installed on the engine. Refer to CMM 78-11-40 for off-aircraft use or if the short exhaust plug has been removed from the engine.								
	AIRPLANE	NOTE: Applicable to airpl	lanes with exl	naust plugs equippe	ed with drain pan	and tube system in	stalled.			
	ACCESS N	OTE: Engine exhaust pluç	g removal req	uired.						
78-050-01-01	MRB	F78-31-01-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL			
	Visually che	eck the left engine T/R's far	n duct walls.							
78-050-02-01	MRB	F78-31-01-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL			
	Visually che	eck the right engine T/R's fa	an duct walls.							
78-060-01-01	MRB	F78-31-07-900-801-F00	1.1	12000 FH	12000 FH	ALL	ALL			

Detailed inspection of the left engine thrust reverser drag link spherical bearings.





TASK CARD NO.				INTERVAL		APPLICA			
	SOURCE	AMM TASK REF	VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
78-060-02-01	MRB	F78-31-07-900-801-F00	1.1	12000 FH	12000 FH	ALL	ALL		
	Detailed ins	spection of the right engine	thrust reverse	er drag link spheric	al bearings.				
78-070-01-01	MRB	F78-31-06-200-801-F00	1.1	12000 FH	12000 FH	ALL	ALL		
	Visually check the left engine blocker doors.								
78-070-02-01	MRB	F78-31-06-200-801-F00	1.1	12000 FH	12000 FH	ALL	ALL		
	Visually check the right engine blocker doors.								
78-080-01-01	MRB	F78-31-23-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL		
	Visually check the left engine bullnose seal and retainer.								
78-080-02-01	MRB	F78-31-23-200-801-F00	1.1	15000 FH	15000 FH	ALL	ALL		
	Visually check the right engine bullnose seal and retainer.								
78-100-01-01	MRB	F78-31-12-200-802-F00	1.1	7500 FH	7500 FH	ALL	ALL		
	Detailed inspection of the left engine T/R fire seal.								
78-100-02-01	MRB	F78-31-12-200-802-F00	1.1	7500 FH	7500 FH	ALL	ALL		
	Detailed inspection of the right engine T/R fire seal.								
78-110-01-01	MRB	F78-31-00-700-803-F00	1.1	5000 FH	5000 FH	ALL	ALL		
	Perform an operational check of the left engine T/R sync lock.								
	SPECIAL N	IOTE: CMR task (78-CMR	-01) interval t	for this task is 5000	FH. See MPD S	ection 9.			
78-110-02-01	MRB	F78-31-00-700-803-F00	1.1	5000 FH	5000 FH	ALL	ALL		
	Perform an operational check of the right engine T/R sync lock.								
	SPECIAL NOTE: CMR task (78-CMR-01) interval for this task is 5000 FH. See MPD Section 9.								
78-120-01-01	MRB	F78-31-00-700-804-F00	1.1	3600 FH	3600 FH	ALL	ALL		
	Perform operational check (bite) on the left engine EAU.								
78-120-02-01	MRB	F78-31-00-700-804-F00	1.1	3600 FH	3600 FH	ALL	ALL		
	Perform operational check (bite) on the right engine EAU.								
78-130-01-01	MRB	F78-31-00-700-801-F00	1.1	15000 FH	15000 FH	ALL	ALL		
	Perform an operational check of the left engine "reverser" light indication system.								
78-130-02-01	MRB	F78-31-00-700-801-F00	1.1	15000 FH	15000 FH	ALL	ALL		
	Perform an operational check of the right engine "reverser" light indication system.								





TASK CARD NO.	SOURCE	AMM TASK REF	INTERVAL			APPLICABILITY			
			VERSION	THRESHOLD	REPEAT	AIRPLANE	ENGINE		
79-010-01-01	MRB	F79-21-03-000-802-F00 F79-21-03-400-801-F00	1.1	7500 FH	7500 FH	ALL	ALL		
	Remove and replace the left engine oil supply filter element.								
79-010-02-01	MRB	F79-21-03-000-802-F00 F79-21-03-400-801-F00	1.1	7500 FH	7500 FH	ALL	ALL		
	Remove and replace the right engine oil supply filter element.								
79-040-01-01	MRB	F79-21-06-000-801-F00 F79-21-06-400-801-F00	1.1	7500 FH	7500 FH	ALL	ALL		
	Remove and replace the left engine oil scavenge filter filter element.								
79-040-02-01	MRB	F79-21-06-000-801-F00 F79-21-06-400-801-F00	1.1	7500 FH	7500 FH	ALL	ALL		
	Remove and replace the right engine oil scavenge filter filter element.								
80-010-01-01	MRB	F80-11-01-200-801-F00	1.1	1600 FC	1600 FC	ALL	ALL		
	Detail inspection of the left engine starter magnetic chip detector for metal chips.								
80-010-02-01	MRB	F80-11-01-200-801-F00	1.1	1600 FC	1600 FC	ALL	ALL		

Detail inspection of the right engine starter magnetic chip detector for metal chips.