CHAPTER

49

AUXILIARY POWER UNIT





737-600/700/800/900 TASK CARDS

CHAPTER 49 AUXILIARY POWER UNIT

Subject/Page	Date	COC	Subject/Page	Date	COC	Subject/Page	Date	COC
49-EFFECTIVE	E PAGES		49-072-00-01	SYS		49-220-00-01	SYS (cont)	
1	JUN 15/2016		1	Oct 15/2014		5	Oct 15/2015	
2	BLANK		2	Feb 15/2015		6	Oct 15/2015	
49-010-00-01	SYS		3	Feb 15/2015		49-240-00-01	SYS	
1	Jun 15/2015		49-082-00-01	SYS		1	Oct 15/2015	
2	Feb 15/2015		1	Oct 15/2014		2	Feb 15/2015	
3	Feb 15/2015		2	Feb 15/2015		3	Oct 15/2014	
4	Oct 15/2015		3	Feb 15/2015		4	Oct 15/2015	
5	Oct 15/2015		49-102-00-01	SYS				
49-020-00-01	SYS		1	Jun 15/2015				
1	Oct 15/2014		2	Feb 15/2015				
2	Feb 15/2015		3	Feb 15/2015				
3	Feb 15/2015		4	Feb 15/2015				
4	Feb 15/2015		5	Feb 15/2015				
5	Oct 15/2015		6	Feb 15/2015				
6	Oct 15/2015		7	Feb 15/2015				
49-030-00-01	SYS		49-140-00-01	SYS				
1	Oct 15/2014		1	Oct 15/2014				
2	Feb 15/2015		2	Feb 15/2015				
3	Oct 15/2015		3	Feb 15/2015				
4	Oct 15/2015		4	Oct 15/2015				
49-040-00-01	SYS		49-172-00-01	SYS				
1	Oct 15/2014		1	Jun 15/2015				
2	Feb 15/2015		2	Oct 15/2015				
3	Feb 15/2015		3	Oct 15/2015				
4	Feb 15/2015		4	Feb 15/2015				
5	Oct 15/2014		5	Oct 15/2015				
49-052-00-01	SYS		49-212-00-01	SYS				
1	Oct 15/2014		1	Jun 15/2015				
2	Feb 15/2015		2	Feb 15/2015				
3			3	Feb 15/2015				
	Feb 15/2015		4	Oct 15/2015				
49-062-00-01	SYS		49-220-00-01	SYS				
1	Oct 15/2014		1	Jun 15/2015				
2	Feb 15/2015		2	Feb 15/2015				
3	Feb 15/2015		3	Feb 15/2015				
			4	Feb 15/2015				

 $A = Added, \ R = Revised, \ D = Deleted, \ O = Overflow, \ C = Customer \ Originated \ Change$

49-EFFECTIVE PAGES



737-600/700/800/900 TASK CARDS

AIRLINE	AIRLINE CARD NO		TITLE	BOEING (CARD NO.		
			APU MOUNTS	49-010	49-010-00-01		
DATE	INSPECTION - GEN VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT			
	APU COMPARTMENT	1.1	5 YR	5 YR	APPLIC	ABILITY	
	COMI ARTIMENT				AIRPLANE	ENGINE	
STATION	SKILL ENGIN				ALL	ALL	
		ACCESS 315A			ZONE 315 316		

Perform a general visual inspection of the APU mounts for general condition and security of installation.

Α.	Re	fere	nces
----	----	------	------

Reference	Title
AMM 49-13-11-200-801	APU Mounts Inspection (APU Removed) (P/B 601)

EFFECTIVITY AKS ALL	SOURCE MRB	APU MOUNTS	
		D633A109-AKS 49-010-00-01	Page 1 of 5 Jun 15/2015

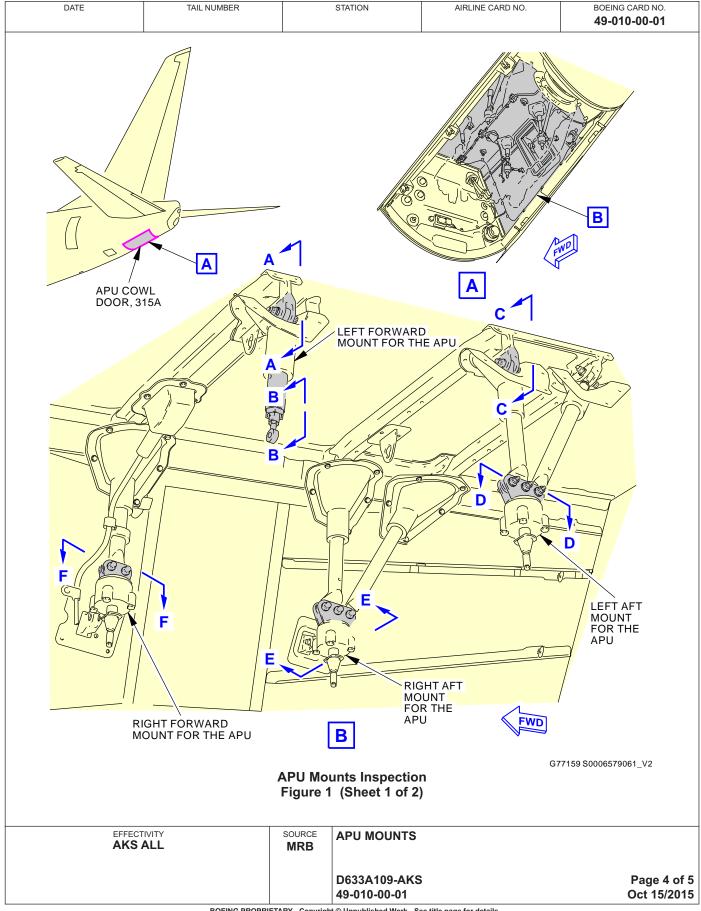


	DATE	-	TAIL NUMBER		STATION	AIRLINE CARD NO.	49-010-		
1. <u>AP</u> L		11-200-803 Inspection	n (APU Install	ed)				MECH	INSP
Α.	Prepar	e for the Ins	spection						
	-	9-13-11-860-002							
	. ,	ake sure the D-NOT-OPE		switch on t	he P5 forward	overhead panel is OFF	and install a		
	SUBTASK 4	9-13-11-860-003							
	(2) O	oen these ci	rcuit breakers	and install	safety tags:				
			l System Pan	el, P6-2					
	<u> </u>	ow Col	·	Name					
		B 19	C01344	APU FIRE	SW POWER				
	F/	O Electrica	l System Pan	el, P6-4					
	<u>F</u>	ow Col	Number	<u>Name</u>					
		A 14	C00033	AUX POW	ER UNIT CON	NT			
		9-13-11-010-005							
			ccess panel, d		eps:				
			Name/Locatio						
			APU Cowl Doc						
	(a			-	or) under the ce	enter latch.			
	(b	•	e three latches						
	,		•		ard latch, aft lat	tch, middle latch.			
	(c	•	APU Cowl D						
	(d	Remove Cowl Do		in from the	rod end of the	e forward hold-open rod	I on the APU		
	(e					the aft hold-open rod.			
	(f			•	ds from the two				
	(g) Connect comparti		nds of the	two hold-open	rods to the two bracket	ts in the APU		
	(h) Install th	e two retainer	pins in the	two rod ends.				
В.	Proced	ure							
		9-13-11-210-002							
	(1) Do				unts (Figure 1)				
	(a				APU mounts a	•			
	(b		examine these n, cracks and o	•	ne APU mounts	s that you can get acce	ess for		
			ut assemblies	-	ints)				
		,	ration isolator	•	,				
	E	FFECTIVITY		SOURCE	APU MOUNTS				



DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	49-010		
	3)) Coi	ne bolts and	nuts for the	vibration isolato	r.	·	MECH	INS
	AF	U moi	unts with the	APU remov		parts, then you must nem, do this task: AF 1-200-801.			
C. Put		•	ack to Its U	•					
	- ASK 49-13-11-4								
(1)	To close	the a	ccess panel,	do these st	eps				
	<u>Numbe</u>	<u>r 1</u>	lame/Locat	<u>ion</u>					
	315A	A	APU Cowl Do	oor					
	(a) Re	emove	the two reta	iner pins fro	m the two hold-o	pen rods in the APU	compartment.		
	(b) Di	sconne	ect the two h	old-open ro	ds from the two I	orackets.			
	(c) Pu	ut the to	wo hold-ope	n rods in the	two spring clips	on the APU Cowl D	oor, 315A.		
	(d) Ins	stall the	e retainer pir	n in the rod	end of the forwar	rd hold-open rod.			
	(e) Ins	stall the	e retainer pir	n to the sprir	ng clip on the aft	hold-open rod.			
	(f) Cl	ose the	e APU Cowl	Door, 315A.					
	(g) Cl	ose the	e three latch	es.					
	NO	OTE: I	Jse this seq	uence: midd	le latch, aft latch	, forward latch			
SUBTA	ASK 49-13-11-8		·						
(2)	Remove	e the s	afety tags ar	nd close the	se circuit breake	rs:			
	F/O Ele	ctrica	l System Pa	nel. P6-2					
	Row	Col	Number	<u>Name</u>					
	В	19	C01344	APU FIRE	SW POWER				
	E/O Ele	-4u!1	l Cuatam Da	mal DC 4					
	Row	Col	l System Pa <u>Number</u>	Name					
	A	<u>30.</u> 14	C00033		ER UNIT CONT	-			
CUPT	ASK 49-13-11-8		00000	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,					
(3)			O-NOT-OPE	ERATE tag f	rom the APU ma	ster switch on the P	5 forward		
(0)	overhea								
				— END OF	TASK ———				
	EFFECTIV			SOURCE	A DU MOUNTS				
				MRB	APU MOUNTS				
	AKS A	LL		IVIKD					
	AKS A	LL		IVIKD	D633A109-AKS			Page 3	







DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-010-00-01
	[1] ROD END			[3] ROD END	
	[2] BOLT			[4	.] BOLT
	SUPPORT BRAC	CKET		LEFT	
	(APU COMPART			FORWA BRACKE	RD ≣T
(2 LO TWO F	CATIONS ON THE ORWARD MOUNTS)			(APU)	
	A-A				
				[7] ST	RUT
	/ [5] ROD END				
	[6] BOLT		VIBRATION ISOLATOR	[8] BUSH (3 LOC	ING CATIONS)
	SUPPORT BRAC (APU COMPARTI	KET MENT)		[9] BUSI	HING
				(3 LO	CATIONS)
(4 LC TW	OCATIONS ON THE O AFT MOUNTS)		(2 LOCATIONS ON THE TWO AFT MOUNTS)	
	C-C			D-D	
				[12] \$1	 IRUT
	[10] BOLT				
	(2 LO	CATIONS)			
[11] ROD END (2 LOCATIO	VIBRATION ISO		VIBRATIO ISOLATO	,	ING CATIONS)
(2 LOCATIO			ISOLATO	\[14] BUSF (2 LO	HING (CATIONS)
	E-E			F-F	77164 S0006579062_V2
			unts Inspection (Sheet 2 of 2)		
EFFECT AKS	ALL	SOURCE MRB	APU MOUNTS		
			D633A109-AKS 49-010-00-01		Page 5 of 5 Oct 15/2015



737-600/700/800/900 TASK CARDS

AIRLINE	E CARD NO		APU MOUNTS		BOEING CARD NO. 49-020-00-01		
DATE	TASK INSPECTION - DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1	THRESHOLD 8 YR	REPEAT 8 YR	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL ENGIN				ALL	ALL	
		ACCESS 315A			ZONE 315 316		

Perform a detailed inspection of the APU mounts.

A. References

Reference	Title
AMM 49-11-00-000-801	APU Power Plant Removal (P/B 401)
AMM 49-11-00-400-801	APU Power Plant Installation (P/B 401)
AMM 49-13-11-000-802	APU Mounts Removal (P/B 401)
AMM 49-13-11-400-802	APU Mounts Installation (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	APU MOUNTS	
ANOALL	IVIKD	D633A109-AKS	Page 1 of 6
		49-020-00-01	Oct 15/2014



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				49-020-00-01

TASK 49-13-11-200-801

MECH INSP

1. APU Mounts Inspection (APU Removed)

(Figure 1)

A. Procedure

SUBTASK 49-13-11-210-001

- (1) Do these steps to inspect the APU mounts:
 - (a) Remove the APU (AMM TASK 49-11-00-000-801).
 - (b) Do the steps to remove the firewall cover and flameshield from the mount, as necessary to view the mount (AMM TASK 49-13-11-000-802).
 - (c) Visually examine these parts for corrosion, cracks and damage:
 - 1) Strut assemblies (APU mounts)
 - 2) Vibration isolators
 - 3) Cone bolts and nuts for the vibration isolator.
 - (d) If you find corrosion, cracks or damage, then do these steps:
 - Remove the parts for the APU mounts that you find with corrosion, cracks or damage (AMM TASK 49-13-11-000-802).
 - 2) Visually examine the bolts and bushings for corrosion, wear and damage.
 - a) Replace the bolts and bushings that you find with corrosion or damage.
 - b) Replace all the parts that are more than the permitted wear limits shown in (Table 1).

Table 1 APU Mount Inspection

		DIMENSION	DESIGN LIMITS		WEAR			
		INNER	DIAI	METER	PERMIT-	MAXIMUM		
ITEM NUMBER	PART	DIAMETER		MAXI- MUM	TED WEAR	CLEAR- ANCE	REPAIR	
Nomber		DIAMETER (OD)	INCH (MM)	INCH (MM)	INCH (MM)	INCH (MM)		
1	ROD END	ID	0.3120 (7.92)	0.3125 (7.94)	0.3175 (8.06)	0.0100 (0.25)	*[1]	
2	BOLT	OD	0.3115 (7.91)	0.3120 (7.92)	0.3060 (7.77)	0.0100 (0.25)	*[1]	
3	ROD END	ID	0.2495 (6.34)	0.2500 (6.35)	0.2550 (6.48)	0.0100 (0.25)	*[1]	
4	BOLT	OD	0.2490 (6.32)	0.2495 (6.34)	0.2435 (6.18)	0.0100 (0.25)	*[1]	
5	ROD END	ID	0.3120 (7.92)	0.3125 (7.94)	0.3175 (8.06)	0.0100 (0.25)	*[1]	

EFFECTIVITY AKS ALL	SOURCE MRB	APU MOUNTS	
		D633A109-AKS 49-020-00-01	Page 2 of 6 Feb 15/2015



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				49-020-00-01

Table 1 APU Mount Inspection (Continued)

MECH INSP

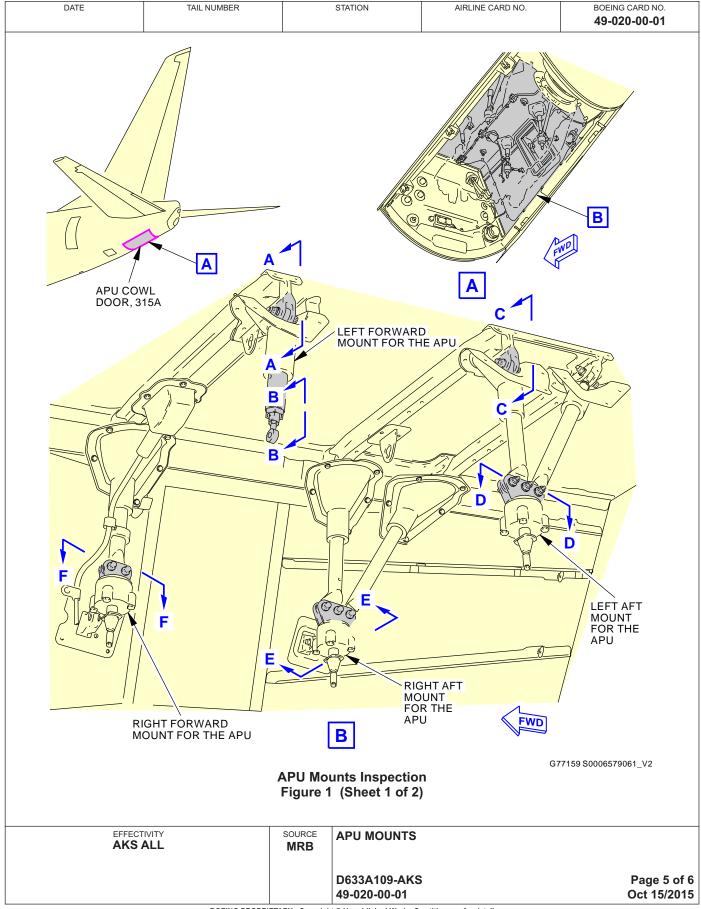
		DIMENSION	DESIG	ON LIMITS	WEAR	LIMITS	
		INNER	INNER DIAMETER		DEDMIT	MAXIMUM	
ITEM NUMBER	PART	DIAMETER (ID) / OUTER	MINI- MUM	MAXI- MUM	PERMIT- TED WEAR	CLEAR- ANCE	REPAIR
NOMBER		DIAMETER (OD)	INCH (MM)	INCH (MM)	INCH (MM)	INCH (MM)	
6	BOLT	OD	0.3115 (7.91)	0.3120 (7.92)	0.3060 (7.77)	0.0100 (0.25)	*[1]
7	STRUT	ID	0.5625 (14.29)	0.5631 (14.30)			*[2]
8	BUSHING	OD	0.4365 (11.09)	0.4370 (11.10)	0.4315 (10.96)	0.0100 (0.25)	*[1]
9	BUSHING	OD	0.5631 (14.30)	0.5638 (14.32)			*[3]
		ID	0.4400 (11.18)	0.4415 (11.21)	0.4465 (11.34)	0.0100 (0.25)	*[1]
10	BOLT	OD	0.2490 (6.32)	0.2495 (6.34)	0.2435 (6.18)	0.0100 (0.25)	*[1]
11	ROD END	ID	0.2497 (6.34)	0.2502 (6.36)	0.2552 (6.48)	0.0100 (0.25)	*[1]
12	STRUT	ID	0.5625 (14.29)	0.5631 (14.30)			*[2]
13	BUSHING	OD	0.4365 (11.09)	0.4370 (11.10)	0.4315 (10.96)	0.0100 (0.25)	*[1]
14	BUSHING	OD	0.5631 (14.30)	0.5638 (14.32)			*[3]
		ID	0.4400 (11.18)	0.4415 (11.21)	0.4465 (11.34)	0.0100 (0.25)	*[1]

- *[1] REPLACE WHEN WORN
- *[2] OVERSIZE STRUT HOLE MUST NOT BE MORE THAN 0.625 INCH (15.88 MM) IN DIAMETER
- *[3] REPLACE WITH OVERSIZE BUSHING
 - Examine the four bolts and four lockwashers that attach each housing assembly to each vibration isolator for tightness and missing part(s).
 - a) If it is necessary, tighten the bolts or replace the missing part(s).
 - 4) Visually examine the surface of each vibration isolator for scratches, nicks, burrs, corrosion, galling, fretting and wear.
 - a) If the individual damaged area is more than 0.500 in. (12.7 mm) diameter by 0.020 in. (0.51 mm) depth or 1.000 in. (25 mm) length by 0.100 in. (2.5 mm) width by 0.020 in. (0.51 mm) depth, replace the vibration isolator.



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 49-020		
			rea is more than eplace the vibra	15% of the total surfa	ace area for	MECH	INSP
		the thread	s of the cone bo	olts and nuts on the vil	oration		
	a) Replace the		_	s that you find with gal	ling, wear or		
			le part(s) for the	APU mounts (AMM			
(e)	TASK 49-13-11-4 Make sure all connection	,	ΔPII mounts ar	nd sunnort brackets a	e tiaht		
(f)	Do the steps to examir						
(.)	(AMM TASK 49-13-11-			over and namedinera,	ao moodaary		
(g)	Install the APU (AMM 7	TASK 49-1	1-00-400-801).				
		- END OF	TASK ———				
	S ALL	SOURCE MRB	APU MOUNTS				
			D633A109-AKS 49-020-00-01			Page 4 eb 15/	







DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-020-00-01
	/ [1] ROD END			/ [3] ROD END	
	[2] BOLT			[4] BOLT
	SUPPORT BRAC			LEFT	
	(APU COMPART	MENT)		FORWA BRACKE (APU)	RD ET
(2 LO TWO F	CATIONS ON THE ORWARD MOUNTS)			В-В	
	A-A				
				[7] ST	RUT
	[5] ROD END				
	[6] BOLT		VIBRATION ISOLATOR	[8] BUSH (3 LOC	ING CATIONS)
	SUPPORT BRAC (APU COMPARTI	KET MENT)		[9] BUSH	HING DCATIONS)
(410	OCATIONS ON THE		,	2 LOCATIONS ON THE	(CATIONS)
TW	C-C		,	TWO AFT MOUNTS) D-D	
				[12] \$1	rdiit.
	[10] BOLT				
	SUPPORT BRAC			[13] BUSHI	ING
[11] ROD END (2 LOCATIO	VIBRATION ISOI ONS)	_ATOR	VIBRATIO ISOLATO	DN (2 LOC	CATIONS)
	E-E			\[14] BUSH (2 LO	HING CATIONS)
			unts Inspection	G [.] 1	77164 S0006579062_V2
		Figure 1	(Sheet 2 of 2)		
EFFECT AKS	ALL	SOURCE MRB	APU MOUNTS		
			D633A109-AKS 49-020-00-01		Page 6 of 6 Oct 15/2015
			73-020-00-01		OCT 13/2013



737-600/700/800/900 TASK CARDS

AIRLIN	E CARD NO	TITLE SIGMA SEAL			BOEING CARD NO. 49-030-00-01		
DATE	INSPECTION - DETAILED				RELATE	ED CARD	
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1	THRESHOLD APU CNG	REPEAT	APPLIC	CABILITY	
STATION	SKILL ENGIN				ALL	ALL	
		ACCESS 315A			ZONE 316		

Perform a detailed inspection of the sigma seal (after APU removal).

A. References

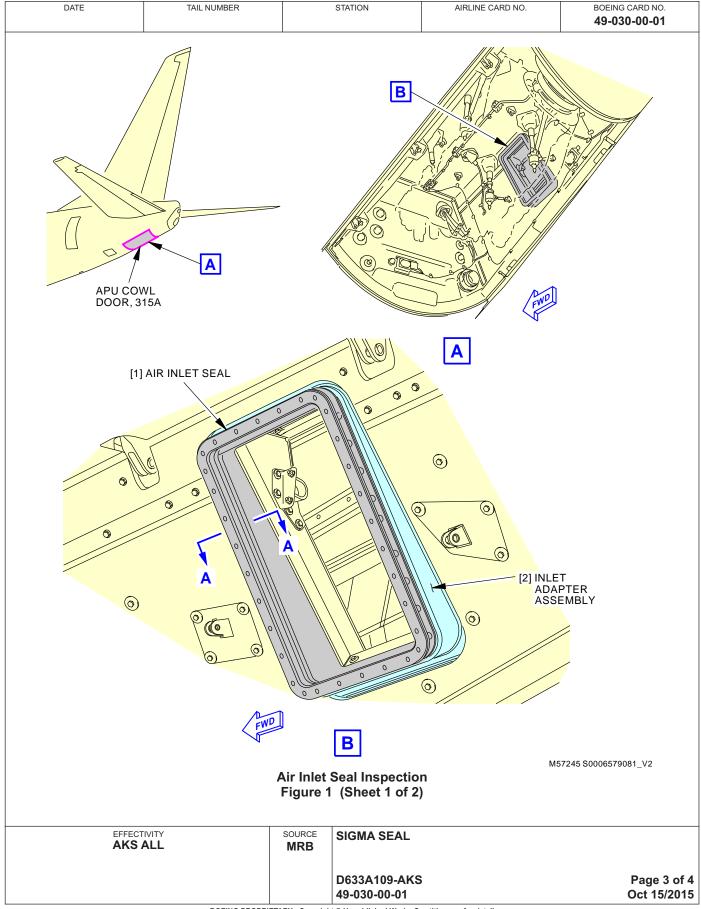
Reference	Title
AMM 49-11-00-000-801	APU Power Plant Removal (P/B 401)
AMM 49-11-00-400-801	APU Power Plant Installation (P/B 401)
AMM 49-15-11-000-801	Air Inlet Seal Removal (P/B 401)
AMM 49-15-11-400-801	Air Inlet Seal Installation (P/B 401)

EFFECTIVITY	SOURCE	SIGMA SEAL	
AKS ALL	MRB		
		D633A109-AKS 49-030-00-01	Page 1 of 4 Oct 15/2014

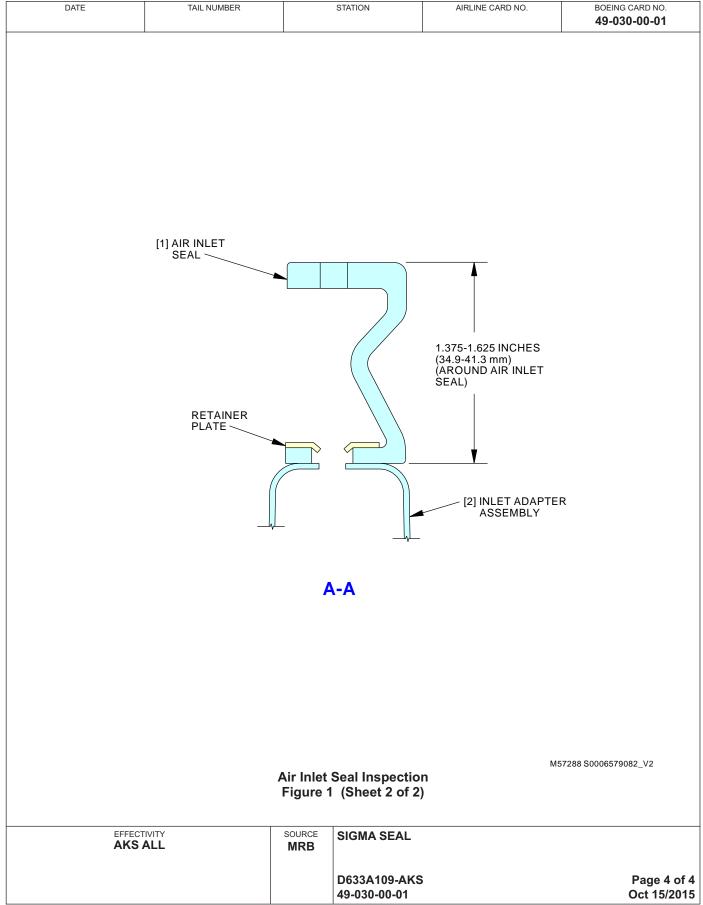


DATE				TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO.	
								49-030	MECH	INSP
١,				200-801					MECH	INSF
1.		ure 1)		spection						
	(Fig	ŕ								
	A.	-		r the Removal						
	suвтask 49-15-11-010-001 (1) Remove the APU. To remove it, do this task: APU Power Plant Removal, AMM TASK 49-11-00-000-801.									
	В.	Dro	cedure							
	В.		SK 49-15-1							
		(1)		ese steps to inspect the	e air inlet	seal [1]:				
		()		•			ling, tears or deformatio	n.		
			. ,	Measure the height of t			3			
				_	n the bott	om to the top of	the air inlet seal [1] mus	st be		
			. ,	Make sure the retainer from) the air inlet seal [•	l stiffener plate a	ire attached to (have no	t disbonded	ı	
			<u> </u>			•	the 32 screws and the a			
			. ,	Examine the mating su materials or wear dama			[1] for wrinkles, bubble r layer of fiberglass.	s, unwanted	ı	
				Examine the seven rub materials, cracks and t		s of fiberglass fo	r any separations, miss	ing		
			. ,	If you find any of the ablimits, replace the air in		-	t of the air inlet seal [1] sks:	is not in the		
			•	Air Inlet Seal Remove	al, AMM T	TASK 49-15-11-0	000-801			
			•	Air Inlet Seal Installa	tion, AMM	1 TASK 49-15-11	-400-801			
		SUBTA	NSK 49-15-1	11-410-001						
		(2)		the APU. To install it, (49-11-00-400-801.	do this tas	sk: APU Power F	Plant Installation, AMM			
					END OF	TASK ———				
			EFFEC AKS		SOURCE MRB	SIGMA SEAL				
						D633A109-AKS			Page 2 Feb 15/	













737-600/700/800/900 TASK CARDS

AIRLIN	AIRLINE CARD NO		TITLE J INSULATION PANI		BOEING CARD NO. 49-040-00-01		
DATE	INSPECTION - DETAILED				RELATEI	O CARD	
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1	THRESHOLD APU CNG	REPEAT	APPLIC/	ABILITY	
STATION	SKILL ENGIN				ALL	ALL	
		ACCESS 315A			ZONE 315 316		

Perform a detailed inspection of the APU insulation panels. (After APU removal).

A. References

Title
APU Power Plant Removal (P/B 401)
APU Power Plant Installation (P/B 401)
Insulation Panel Removal (P/B 401)
Repair of the APU Insulation Panel (P/B 801)
Insulation Panel Installation (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	APU INSULATION PANELS	
ANGALL	IVIKD	D633A109-AKS	Page 1 of 5
		49-040-00-01	Oct 15/2014



737-600/700/800/900 **TASK CARDS**

					IAS	K CARDS				
	DATE		-	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 49-040		
TAS	SK 49-	-17-11	I-200-801						MECH	INS
1. <u>Ins</u>	ulatio	n Par	nel Inspec	tion						
A.	Pre	oare 1	for the Ins	spection						
	SUBTA	ASK 49-1	7-11-860-001							
	(1)			APU master RATE tag.	switch on t	the P5 forward o	overhead panel is OFF a	and install a		
			7-11-860-002							
	(2) Open these circuit breakers and install safety tags:									
	F/O Electrical System Panel, P6-2									
		Ro		Number 004044	Name					
		В	19	C01344	APU FIRE	SW POWER				
		F/O	Electrica	l System Par	nel, P6-4					
		Ro	w Col	Number	<u>Name</u>					
		Α	14	C00033	AUX POW	/ER UNIT CON	Т			
	SUBTA	ASK 49-1	7-11-010-006							
	(3)	То о	•	ccess panel, o		eps:				
				Name/Location						
		315		APU Cowl Do						
		(a)				or) under the ce	nter latch.			
		(b)	Open the	e three latche	s.					
			NOTE: I	Use this sequ	ence: forwa	ard latch, aft late	ch, middle latch.			
		(c)	Open the	e APU Cowl D	oor, 315A.					
		(d)	Remove Cowl Do	•	oin from the	rod end of the	forward hold-open rod o	on the APU		
		(e)	Remove	the retainer p	oin from the	spring clip on t	he aft hold-open rod.			
		(f)	Disconne	ect the two ho	old-open ro	ds from the two	spring clips.			
		(g)	Connect comparts		ends of the	two hold-open r	rods to the two brackets	in the APU		
		(h)	Install the	e two retainer	pins in the	two rod ends.				
	SUBTA	ASK 49-1	7-11-010-001							
	(4)	-		a full inspectio II, AMM TASK		•	anels, do this task: APL	J Power		
		NOT	<u>ΓΕ</u> : It is ne panel.		emove the A	APU to inspect a	all the surfaces of the to	p insulation		
		NOT		ot necessary t insulation pa		he APU if you c	lo a general visual inspe	ection of the		
			ECTIVITY S ALL		SOURCE MRB	APU INSULATION	ON PANELS			
						D633A109-AKS	;		Page 2 eb 15	



737-600/700/800/900 **TASK CARDS**

	DATE			TA	AIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 49-040		
В.	Pro	cedur	re							MECH	INS
	SUBTA	SK 49-1	7-11-210	-003							
	(1)		hese age:	steps	to inspect th	e insulatio	n panels for flui	d contamination and str	ructural		
		(a)		mine [·] erial.	the insulation	panels fo	r signs of fluid c	ontamination to the cor	e insulation		
		(b)					ation panels for structural meta	missing weld stitches, r I deterioration.	missing		
		(c)		mine al she		panels fo	r holes that hav	e gone through the inne	er and outer		
		(d)	If yo		any of the a	bove dama	age, replace the	insulation panel(s). Th	ese are the		
			• In	sulatio	on Panel Rer	noval, AMI	и TASK 49-17-	11-000-801			
			• In	sulatio	on Panel Inst	allation, Al	MM TASK 49-17	7-11-400-801			
			1)	Afte	r the insulation	on panel(s)	are removed, o	do these steps:			
				a)	Examine the and damage		behind the insu	ulation panel(s) for cont	amination		
				b)			coop for blocka e a decrease ir	nge of unwanted materian air flow.	als and		
					NOTE: You par		he air inlet scoo	p behind the forward in	sulation		
				c)	If you find b	lockage of	unwanted mate	erials, remove the block	age.		
				d)	If you find c you find.	ontaminati	on or damage,	clean and repair the pro	oblems that		
	SUBTA	SK 49-1	7-11-210	-004							
	(2)	Do t	hese	steps	to inspect th	e insulatio	n panels for oth	er structural damage lir	nits:		
		(a)	Exa					panel for holes and tea	rs.		
			1)					eter are permitted.			
			2)		s less than 2 h are permitte	,	n) in length and	less than 0.05 in. (1.27	' mm) in		
		(b)	Exa	mine	the damaged	areas of t	he insulation pa	nels.			
			1)	-			ch (13 mm) of the epair the insula	ne outer metal sheet ard ation panel.	ound the		
				NOT				n the damaged area to et or attaching parts.	a grommet,		
		(c)	-			-	d the damage is ASK 49-17-11-	in the limits, do this tas 300-801.	sk: Repair of		
NOTE: CMM 49-17-00 can be us					MM 49-17-00	can be u	sed to repair da	mage less than 6 inche	s.		
		(d)			the above depanel(s). The			more than the limits, re	eplace the		
			ECTIVITY S ALL			SOURCE MRB	APU INSULATION	ON PANELS			
							D633A109-AKS	;		Page 3	



737-600/700/800/900 **TASK CARDS**

D	ATE		Т	AIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 49-040-		
		• 1	nsulat	ion Panel Re	moval AMI	M TASK 49-17-	-11-000-801	-	MECH	INSF
	Insulation Panel Installation, AMM TASK 49-17-11-400-801									
		1)					do these steps:			
		,	a)		ne structure		sulation panel(s) for con	tamination		
			b)	Examine th	ne air inlet s	scoop for block se a decrease i	age of unwanted mater	ials and		
				NOTE: Yo			op behind the forward in	nsulation		
			c)	•		unwanted mat	terials, remove the bloc	kage.		
			d)	•	•		, clean and repair the pr			
	SUBTA	SK 49-17-11-4	10-001							
	(3)					pection of the SK 49-11-00-4	insulation panel(s), do t 400-801.	his task:		
C.	C. Put the Airplane Back to Its Usual Condition									
SUBTASK 49-17-11-860-003										
	(1)	Remove	the s	afety tags an	d close the	se circuit break	(ers:			
				System Par						
		Row B	<u>Col</u> 19	Number C01344	Name	SW POWER				
				System Par		SWFOWLK				
		Row	Col	Number	Name					
			14	C00033		ER UNIT CON	NT			
	SUBTA	SK 49-17-11-8	60-004							
	(2) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.									
	SUBTASK 49-17-11-410-005									
	(3)			ccess panel,		eps				
		Number		lame/Location						
		315A		PU Cowl Do						
		` '			•		l-open rods in the APU	compartment.		
		` '			•	ds from the two		0.454		
		` '		·			ps on the APU Cowl Do	or, 315A.		
		` '		-			ard hold-open rod.			
	(e) Install the retainer pin to the spring clip on the aft hold-open rod.									
	(f) Close the APU Cowl Door, 315A.									
		(.)								
		EFFECTIVI			SOURCE MRB	APU INSULAT	ION PANELS			



DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	49-040-00-01	
(g) (Close the three latches.			MECH	INSP
	NOTE: Use this sequen	ce: middle latch, aft lat	ch, forward latch		
SUBTASK 49-17-1	1-410-007				
(4) If the A	APU was removed, do tl	his task: AMM TASK 49	9-11-00-400-801		
	——	END OF TASK ——	_		
EFFEC ⁻	TIMITY	SOURCE APU INSULAT	TION DANIEL C		
AKS	ALL	MRB APU INSULAT	IUN PANELS		
		B	· 0	_	
		D633A109-AK 49-040-00-01	.5	Page 9 Oct 15	5 of 5 /2014
	Į.				





737-600/700/800/900 TASK CARDS

AIRLIN	IE CARD NO	ENGINE	TITLE COMPRESSOR IM	BOEING CARD NO. 49-052-00-01		
DATE	TASK REPLACE				RELATEI W-49-06 W-49-07 W-49-08	2-00-01 2-00-01
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1 NOTE	THRESHOLD LIF LIM	REPEAT	APPLIC/	
STATION	SKILL ENGIN				ALL	ALL
		ACCESS 315A			ZONE 315 316	

Discard the engine compressor impeller.

INTERVAL NOTE: Refer to APU shop manual for life limits.

A. References

Reference	Title
AMM 49-11-00-000-802	APU Power Plant Removal (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-000-803	APU Power Plant Removal (Hydraulic Jack Procedure) (P/B 401)
AMM 49-11-00-400-802	APU Power Plant Installation (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-400-803	APU Power Plant Installation (Hydraulic Jack Procedure) (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	ENGINE COMPRESSOR IMPELLER	
		D633A109-AKS 49-052-00-01	Page 1 of 3 Oct 15/2014



]	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 49-052		
	TAS	K 49-1	11-00-	000-801			1		MECH	INSP
1.	APL	J Powe	er Pla	nt Removal						
	A.	APU	Powe	er Plant Removal						
				oo-o2o-oo1 ne of these tasks to ren	aovo tho A	DI I:				
		` '					le Hoist Procedure), AMM			
				TASK 49-11-00-000-80)2.					
				Do this task: APU Pow TASK 49-11-00-000-80		emoval (Hydrau	ılic Jack Procedure), AMN	1		
					END OF	TASK ———				
			EFFEC AKS	TIVITY ALL	SOURCE MRB	ENGINE COMP	RESSOR IMPELLER			
						D633A109-AKS 49-052-00-01			Page 2 eb 15/	



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C			
								49-052	1	
	TAS	SK 49-	11-00	-400-801					MECH	INSP
2.	API	J Pow	er Pla	ant Installation						
	Α.	APU	Pow	er Plant Installation						
				-00-420-001						
		(1)		ne of these tasks to ins	tall the AF	PU:				
		(-)					oole Hoist Procedure), A	MMA		
			(α)	TASK 49-11-00-400-80			oole Holst Frocedure), 7	(IVIIVI		
	(b) Do this task: APU Power Plant Installation (Hydraulic Jack Procedure), AMM TASK 49-11-00-400-803.									
					- END OF	TASK ———				
			EFFE(CTIVITY	SOURCE	ENGINE COMP	RESSOR IMPELLER		1	
			AKS	SALL	MRB					
						D633A109-AKS			Daga 1	of ?
						49-052-00-01		F	Page 3 eb 15/	2015





737-600/700/800/900 TASK CARDS

AIRLIN	E CARD NO	FIRS	TITLE T STAGE TURBINE	BOEING CARD NO. 49-062-00-01		
DATE	TASK REPLACE				RELATE W-49-05 W-49-07 W-49-08	2-00-01 2-00-01
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1 NOTE	THRESHOLD LIF LIM	REPEAT	APPLIC, AIRPLANE	ABILITY ENGINE
STATION	SKILL ENGIN				ALL	ALL
		ACCESS 315A			ZONE 315 316	

Discard the first stage turbine disk.

INTERVAL NOTE: Refer to APU shop manual for life limits.

A. References

Reference	Title
AMM 49-11-00-000-802	APU Power Plant Removal (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-000-803	APU Power Plant Removal (Hydraulic Jack Procedure) (P/B 401)
AMM 49-11-00-400-802	APU Power Plant Installation (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-400-803	APU Power Plant Installation (Hydraulic Jack Procedure) (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	FIRST STAGE TURBINE DISK	
		D633A109-AKS 49-062-00-01	Page 1 of 3 Oct 15/2014



]	DATE TAIL NUMBER				STATION	AIRLINE CARD NO.		CARD NO. 2-00-01	
	TAS	K 49-	11-00-	000-801					MECH	INSP
1.	APL	J Pow	er Pla	nt Removal						
	A.	APU	J Powe	er Plant Removal						
				00-020-001						
		(1)		ne of these tasks to ren			La lla de la Companya	48.4		
				TASK 49-11-00-000-80)2.		le Hoist Procedure), AM			
				Do this task: APU Pow TASK 49-11-00-000-80		temoval (Hydrau	ilic Jack Procedure), AN	ИM		
					END OF	TASK ———				
			CCCCC	TIMITY	SOURCE	FIDOT OT OF	TUDDINE DIOY			
			AKS	ALL	SOURCE MRB	FIRST STAGE 1	UKBINE DISK			
						D633A109-AKS			Page 2	2 of 3
						49-062-00-01			Feb 15/	2015



	[DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
							49-062	-00-01	
	TAS	K 49-11-00)-400-801					MECH	INSP
2.	APL	J Power Pl	ant Installation						
	A.		er Plant Installation						
		SUBTASK 49-1							
		` '	one of these tasks to inst						
		(a)	Do this task: APU Pow TASK 49-11-00-400-80		nstallation (Fishp	oole Hoist Procedure), A	MM		
		(b)	Do this task: APU Pow TASK 49-11-00-400-80	er Plant Ir 3.	nstallation (Hydra	aulic Jack Procedure), A	MMA		
					TASK ———				
				END OF	IASK ———				
		EFFE	ECTIVITY	SOURCE	FIRST STAGE T	URBINE DISK		1	
		AK	SALL	MRB					
					D633A109-AKS 49-062-00-01			Page 3 eb 15/	of 3





737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		SECOND STAGE TURBINE ROTOR				BOEING CARD NO. 49-072-00-01		
DATE	TASK REPLACE				RELATEI W-49-05 W-49-06 W-49-08	2-00-01 2-00-01		
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1 NOTE	THRESHOLD LIF LIM	REPEAT	APPLICA			
STATION	SKILL ENGIN				ALL	ALL		
		ACCESS 315A			ZONE 315 316			

Discard the second stage turbine rotor.

INTERVAL NOTE: Refer to APU shop manual for life limits.

A. References

Reference	Title
AMM 49-11-00-000-802	APU Power Plant Removal (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-000-803	APU Power Plant Removal (Hydraulic Jack Procedure) (P/B 401)
AMM 49-11-00-400-802	APU Power Plant Installation (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-400-803	APU Power Plant Installation (Hydraulic Jack Procedure) (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	SECOND STAGE TURBINE ROTOR	
		D633A109-AKS 49-072-00-01	Page 1 of 3 Oct 15/2014



	ı	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 49-072 -		
							49-072	MECH	INSP
		SK 49-11-00						WEOT	IIVOI
1.	APU		ant Removal						
	A.	APU Pov	ver Plant Removal						
		SUBTASK 49-1			. D. I				
		,	one of these tasks to rer			la llaiat Dua an duuna). AM	IN 4		
		(a)	TASK 49-11-00-000-80		kemovai (Fisnpo	le Hoist Procedure), AM	IVI		
		(b) Do this task: APU Power Plant Removal (Hydraulic Jack Procedure), AMM TASK 49-11-00-000-803.							
				- END OF	TASK				
		EFF AK	ECTIVITY (S ALL	SOURCE MRB	SECOND STAG	E TURBINE ROTOR		•	
								_	
					D633A109-AKS 49-072-00-01		F	Page 2 eb 15/	2 of 3 2015



	Г	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
							49-072		
	TAS	K 49-11-00	-400-801					MECH	INSP
2.	APU	J Power Pla	ant Installation						
	Α.	APII Pow	er Plant Installation						
	Λ.	SUBTASK 49-1							
			one of these tasks to inst	all the AP	·U·				
		(1) Bo c	Do this task: APU Pow			nole Hoist Procedure) A	MM		
		(a)	TASK 49-11-00-400-80		istaliation (i isni	ole Hoist Procedure), P	AIVIIVI		
		(b)	Do this task: APU Pow TASK 49-11-00-400-80	er Plant Ir 3.	nstallation (Hydra	aulic Jack Procedure), A	MMA		
				END OF	TASK ———				
		CCCC	CTIVITY	SOURCE	SECOND STAC	E TUDDINE DOTOD			
		AK	S ALL	MRB	SECOND STAG	E TURBINE ROTOR			
					D633A109-AKS 49-072-00-01		F	Page 3 eb 15/	of 3 2015



737-600/700/800/900 TASK CARDS

AIRLIN	AIRLINE CARD NO		TITLE TURBINE SHAFT		BOEING CARD NO. 49-082-00-01		
DATE	TASK REPLACE				RELATE W-49-05 W-49-06 W-49-07	2-00-01 2-00-01	
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1 NOTE	THRESHOLD LIF LIM	REPEAT	APPLIC/	ABILITY ENGINE	
STATION	SKILL ENGIN				ALL	ALL	
		ACCESS 315A			ZONE 315 316		

Discard the turbine shaft.

INTERVAL NOTE: Refer to APU shop manual for life limits.

A. References

Reference	Title
AMM 49-11-00-000-802	APU Power Plant Removal (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-000-803	APU Power Plant Removal (Hydraulic Jack Procedure) (P/B 401)
AMM 49-11-00-400-802	APU Power Plant Installation (Fishpole Hoist Procedure) (P/B 401)
AMM 49-11-00-400-803	APU Power Plant Installation (Hydraulic Jack Procedure) (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	TURBINE SHAFT	
		D633A109-AKS 49-082-00-01	Page 1 of 3 Oct 15/2014



	I	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO	
							49-082-00-0	1
	TAS	SK 49-11-0	0-000-801				MECH	H INSP
1.	API	J Power P	lant Removal					
	Α.	ADII Dov	ver Plant Removal					
	Α.							
		(1) Do	one of these tasks to r	omove the	۸DI I۰			
		` '				la Haiat Dragadura) AMM	4	
		(a)	TASK 49-11-00-000-		Removai (Fishpo	le Hoist Procedure), AMN	VI	
		(b)	Do this task: APU Po TASK 49-11-00-000-		Removal (Hydrau	ılic Jack Procedure), AMI	M	
				— END OF	TASK ———			
		EFF AK	ECTIVITY (S ALL	SOURCE MRB	TURBINE SHAF	т		
			- 					
					D633A109-AKS 49-082-00-01		Page Feb 15	2 of 3 5/2015



	I	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA		
							49-082-0	00-01	
	TAS	SK 49-11-0	0-400-801					MECH	INSP
2.	API	J Power P	ant Installation						
	Α.	A DI I Dov	ver Plant Installation						
	Α.								
		(1) Do	one of these tasks to in	etall the AE	OI I·				
		` '				oolo Hoiot Procedura) A			
		(a)	TASK 49-11-00-400-8		ristaliation (Fish	oole Hoist Procedure), A	INIINI		
		(b)	Do this task: APU Po TASK 49-11-00-400-8		nstallation (Hydr	aulic Jack Procedure), A	AMM		
				— END OF	TASK ———				
		EFF	ECTIVITY S ALL	SOURCE	TURBINE SHAF	Т			
		An	NO ALL	MRB					
					D633A109-AKS 49-082-00-01		P Fe	age 3 b 15/	of 3 2015





737-600/700/800/900 TASK CARDS

AIRLIN	E CARD NO	FUEL	TITLE INLET FILTER ELI	BOEING CARD NO. 49-102-00-01		
DATE	TASK REPLACE				RELATE	D CARD
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1	THRESHOLD 4000 AH	REPEAT 4000 AH	APPLIC.	ABILITY ENGINE
STATION	SKILL ENGIN				ALL	ALL
		ACCESS 315A			ZONE 315 316	

Discard the fuel inlet filter element on the fuel control unit (FCU).

A. References

Reference	Title
AMM 49-11-00-860-801	APU Starting and Operation - Activation (P/B 201)
AMM 49-11-00-860-802	APU Usual Shutdown (P/B 201)

B. Consumable Materials

Reference	Description	Specification
B00130	Alcohol - Isopropyl	TT-I-735
D00341	Lubricant - Polyphenyl Ether, Vacuum Pump - Santovac 5	
D00504	Grease - Petrolatum	VV-P-236
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper (Cheesecloth, Gauze)	BMS15-5 Class A

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description	
STD-1057	Air Source - Regulated, Dry Filtered, Compressed 60-105 PSIG	
	(414-723.9 KPa)(22 SCFM)	
STD-4049	Container - Fuel Resistant, 1 Gallon (4 Liters)	

EFFECTIVITY		FUEL INLET FILTER ELEMENT	
AKS ALL	MRB		
		D633A109-AKS 49-102-00-01	Page 1 of 7 Jun 15/2015



737-600/700/800/900 TASK CARDS

					TAS	K CARDS						
!	DATE		Т	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 49-102-				
TAS	K 49-	31-21	-000-801						MECH	IN		
. Inle	Inlet Fuel Filter Element Removal											
(Fig	ure 1)											
A.	A. Prepare for the Removal											
	SUBTA	NSK 49-31	-21-860-001									
	(1) Make sure the APU master switch on the P5 forward overhead panel is OFF and install a DO-NOT-OPERATE tag.											
	SUBTASK 49-31-21-860-002											
	(2) Open these circuit breakers and install safety tags:											
	F/O Electrical System Panel, P6-2											
		Rov	v <u>Col</u>	<u>Number</u>	<u>Name</u>							
		В	19	C01344	APU FIRE	SW POWER						
	F/O Electrical System Panel, P6-4											
		Rov		Number	<u>Name</u>							
		Α	14	C00033	AUX POW	/ER UNIT CON	IT					
	SUBTA	NSK 49-31	-21-010-002									
	(3)	To op	oen the ac	ccess panel,	do these st	eps:						
		Num	<u>ber</u> <u>N</u>	lame/Locati	<u>on</u>							
	315A APU Cowl Door											
(a) Support the APU panel (cowl door) under the center latch.												
(b) Open the three latches.												
NOTE: Use this sequence: forward latch, aft latch, middle latch.												
	(c) Open the APU Cowl Door, 315A.											
		(d)	Remove Cowl Do		pin from the	rod end of the	forward hold-open rod	on the APU				
(e) Remove the retainer pin from the spring clip on the aft hold-open rod.												
		(f)	Disconne	ect the two ho	old-open ro	ds from the two	spring clips.					
		(g)	Connect compartr		ends of the	two hold-open	rods to the two brackets	in the APU				
	(h) Install the two retainer pins in the two rod ends.											
В.	Inlet	t Fuel	Filter Ele	ement Remo	val							
	SUBTA	NSK 49-31	-21-020-001									
	(1) Do these steps to remove the fuel filter element [3]:											
	(a) Put the 1 gallon (4 l) fuel resistant container, STD-4049 below the fuel filter housing [1].											
			CTIVITY S ALL		SOURCE MRB	FUEL INLET FI	ILTER ELEMENT					
						D633A109-AKS	5		Page 2 eb 15/			



DATE	TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD 49-102-00-						
WAR	GLOVES WHEN FLAME, AND HE	THE FUN YOU US AT. FUE	MES FROM TH SE FUEL. KEEF L IS POISONO	EYES, OR ON YOUR S E FUEL. PUT ON GOG P FUEL AWAY FROM SI US AND FLAMMABLE. D DAMAGE TO EQUIP	GLES, AND PARKS, FUEL CAN	MECH	INS
(b)	Loosen the two nuts [4] t						
(c)	Turn the fuel filter housin two studs.						
	Remove the fuel filter ho	usina [1]	L				
` '	Remove the packing [2]			na [1]			
(0)	Discard the packing		radi inter ridaen	9[.].			
	,	emove the packing [5] from between the fuel control unit and the fuel filter					
	Discard the packing	a [5].					
(g)	Remove the fuel filter ele		l.				
(0)	Discard the fuel filter						
(h)	Make sure you install all			overs.			
()	Remove the 1 gallon (4 I						
()	•	´ END OF '		,			
			I				
	S ALL	MRB	FUEL INLET FI	LTER ELEMENT			
			D633A109-AKS 49-102-00-01	1		Page 3 eb 15/	



737-600/700/800/900

	DATE			TAIL NUMBER		STATION	AIRL	INE CARD NO.	49-102-		
TAS	K 49-	31-2 1	-400-801		·	·				MECH	INSI
Inlet	t Fuel	Filte	r Elemen	t Installatio	<u>n</u>						
(Figu	ure 1)										
A.	Ехр	endal	bles/Parts	6							
	AM	M Ite	m Desc	cription		AIPC Referen	се	AIPC Effective	rity		
		2	Pack	king		49-31-11-02-0	60	AKS ALL			
		3		filter elemer	nt	49-31-11-02-0		AKS ALL			
		5	Pack	king		49-31-11-02-0	55	AKS ALL			
B.	Proc	edur	е								
			1-21-110-001								
	(1)		•		e fuel filter h						
		(a)				ith alcohol, B00130		•			
		(b) Use the compressed 60-105 PSIG dry filtered regulated air source, STD-1057 to dry the fuel filter housing [1].					D-1057 to dry				
			NOTE: It is recommended that you use a pressure of 60-90 psig (414-620 kPa) of air or nitrogen to dry the fuel filter housing [1].								
	SUBTA	SK 49-3	1-21-420-001								
	CAU	IOITI	NECES	SARY. IF Y		COVERS FROM T REMOVE THE PICCUR.			RS,		
	(2)	Do t	hese step	s to install th	ne fuel filter e	element [3]:					
		(a)	Lubricate grease, I		icking [2] wit	h a light coat of Sa	ntova	c 5 lubricant, [D00341 or		
		(b)	Install the	e packing [2] on the fuel	filter housing [1].					
		(c)	Lubricate grease, I		icking [5] wit	h a light coat of Sa	ntova	c 5 lubricant, [D00341 or		
		(d)	Install the	e new packi	ng [5] on the	fuel filter element	[3].				
		(e)	Install the	e fuel filter e	lement [3] in	the fuel control ur	it.				
		(f)	Install the	e fuel filter h	ousing [1] or	n the fuel control u	nit.				
		(g)	Turn the	fuel filter ho	using [1] clo	ckwise until the fla	nge fu	ılly engages th	e two studs.		
		(h)	Tighten t	he two nuts	[4] to 40 pou	ınd-inches (4.5 ne	wton-r	meters).			
C.	Inlet	Fuel	Filter Ele	ement Insta	llation Test						
	SUBTA	SK 49-3	1-21-860-003								
	(1)	Rem	ove the s	afety tags aı	nd close thes	se circuit breakers:					
		F/O	Electrical	l System Pa	nel, P6-2						
		Roy	<u>v Col</u>	<u>Number</u>	<u>Name</u>						
		В	19	C01344	APU FIRE	SW POWER					
					1						
			CTIVITY		SOURCE	FUEL INLET FILTE					

Page 4 of 7 Feb 15/2015



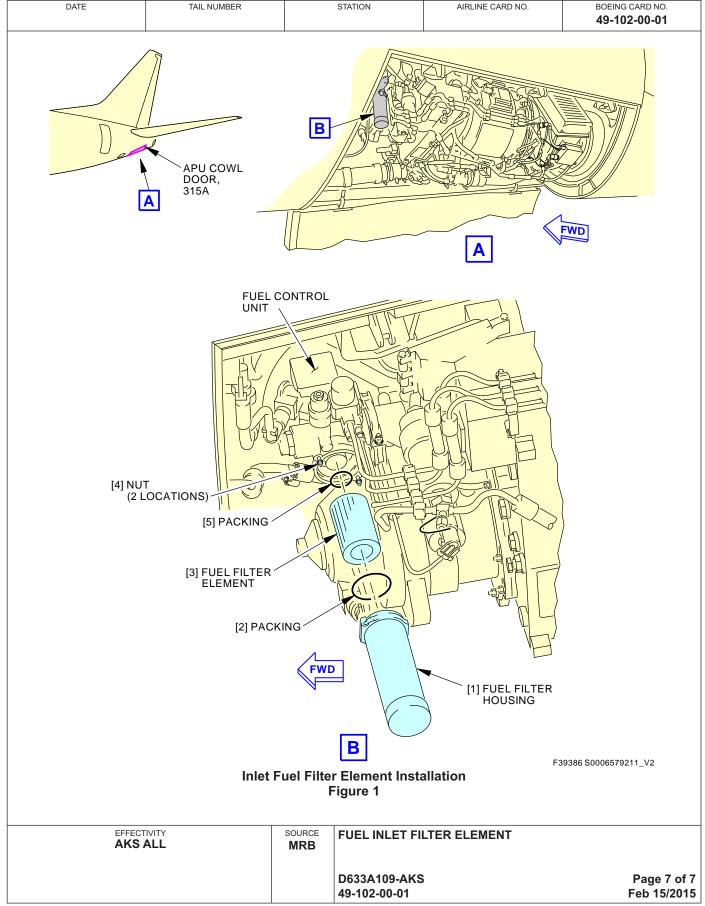
737-600/700/800/900 **TASK CARDS**

DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 49-102-0	
	Rov A	<u>w</u> <u>Co</u> 14		<u>Name</u>	/ER UNIT CON	Т		MECH IN
suвт (2)	Rem	1-21-860-004 nove the head pa		ERATE tag f	rom the APU ma	aster switch on the P5	forward	
SUBT	ASK 49-3	1-21-790-001						
(3)	Do tl	he instal	lation test for t	the fuel filter	element:			
	(a)	(a) Do this task: APU Starting and Operation - Activation, AMM TASK 49-11-00-860-801.						
		NOTE:	replace the fu	uel filter eler	ment. If you star	ore than three times afte t the APU again, make during a 15 minute inter	sure you	
		NOTE:		nance mess		el filter housing can cau o show on the control d		
	(b)	Operat	e the APU for	a minimum	of five minutes.			
	(c)	During	the APU opera	ation, exami	ine the fuel filter	housing for signs of fu	el leakage.	
	(d)	If you fi	ind fuel leakag	e, do these	steps to repair	the leakage:		
		1) D	o this task: AF	U Usual Sh	utdown, AMM T	TASK 49-11-00-860-802	2.	
		,	stall a DO-NO verhead panel		E tag to the API	J master switch on the	P5 forward	
		3) R	epair the caus	e of the fue	l leakage.			
		,	emove the DC orward overhea		RATE tag from	the APU master switch	on the P5	
		,	o this task: AF ASK 49-11-00-	_	and Operation -	Activation, AMM		
			uring the APU akage.	operation, e	examine the fue	el filter housing for signs	s of fuel	
		7) If	you find fuel le	eakage, do t	the leakage rep	air again.		
	(e)		ot necessary t 19-11-00-860-8		asks, do this tas	sk: APU Usual Shutdov	vn, AMM	
D. Put	the A	irplane	Back to Its Us	sual Condit	tion			
SUBT	ASK 49-3	1-21-410-002						
(1)	To cl	lose the	access panel,	do these st	eps			
	Num	<u>ıber</u>	Name/Locati	<u>on</u>				
	315	4	APU Cowl Do	or				
	(a)	Remov	e the two retai	iner pins fro	m the two hold-	open rods in the APU o	ompartment.	
	(b)	Discon	nect the two h	old-open ro	ds from the two	brackets.		
		S ALL		SOURCE MRB	FUEL INLET FI	LTER ELEMENT		
					D633A109-AKS 49-102-00-01	;		age 5 o



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.		G CARD NO.	
(c)	Put the two hold-open	rods in the	two spring clips	s on the APLI Cowl Do		MECH	INSF
(d)	Install the retainer pin i				01, 010/1.		
(e)	Install the retainer pin						
(f)		ose the APU Cowl Door, 315A.					
(g)	Close the three latches						
(9)	NOTE: Use this seque		lle latch, aft latch	n, forward latch			
			TASK ———				
	ECTIVITY	SOURCE	FUEL INLET FIL	TER ELEMENT			
AK	(S ALL	MRB					
			D633A109-AKS			Page 6	c of
			49-102-00-01			Feb 15/	







AIRLIN	E CARD NO	TITLE CDU'S APU MAINTENANCE PAGES				BOEING CARD NO. 49-140-00-01	
DATE	TASK OPERATIONAL				RELATE	ED CARD	
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 1600 AH	REPEAT 1600 AH	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL ENGIN				ALL	ALL	
		ACCESS			ZONE 211		

- -APU data memory module (DMM)
- -Electronics control unit (ECU)
- -Speed Sensor
- -EGT rake

EFFECTIVITY AKS ALL	SOURCE MRB	CDU'S APU MAINTENANCE PAGES	
		D633A109-AKS 49-140-00-01	Page 1 of 4 Oct 15/2014

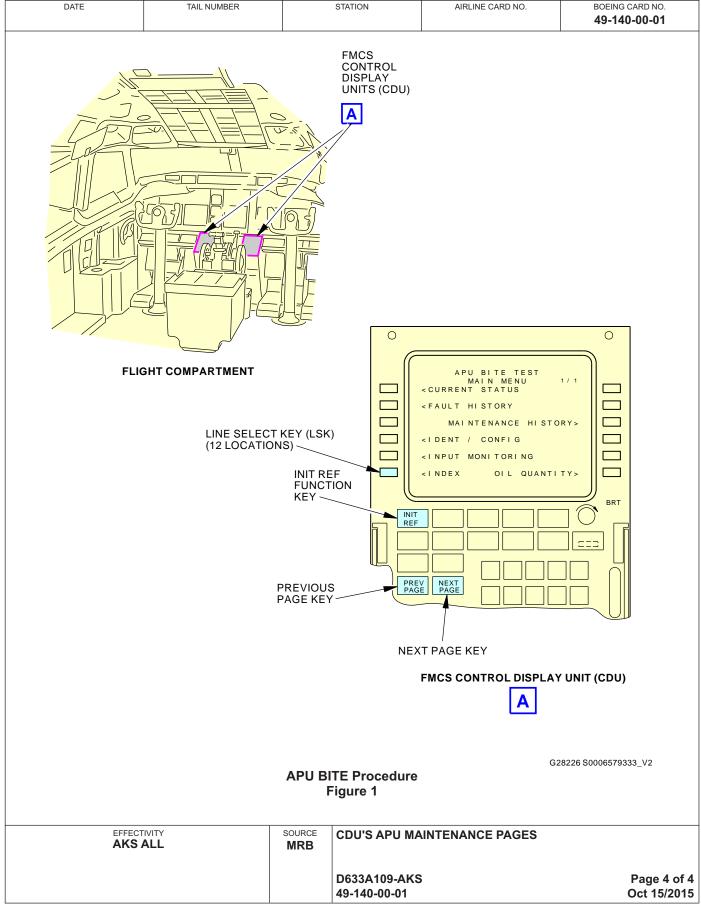


	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 0-00-01	
	TAS	K 49	-61-00)-710-801					MECH	INSP
1.	APL	J Con	trols	Operational Check						
	(Fig	ure 1))							
	A.	Pro	cedur	е						
				1-00-740-006	anal ahaak	from the MAIN	MENILI page for the AD	I I DITE		
		(1)	TES	in APU controls operation T:	orial Crieck	ITOTTI LITE IVIAIN	MENO page for the AP	OBITE		
			(a)	(a) If you get access to the CDU for the first time or were in an airplane system other than the APU, push the INIT REF function key until the PERF INT page shows on the CDU display.						
				NOTE: The PERF INT	Γpage or I	DENT page can	show on the CDU disp	lay.		
			(b)	<index.< th=""><th></th><th></th><th>ne line select key adjace</th><th>ent to</th><th></th><th></th></index.<>			ne line select key adjace	ent to		
				NOTE: The INIT/REF	•	_	, ,			
			(c)	•		0 .	ne select key adjacent	to MAINT>.		
			(d)	NOTE: The MAINT BI		. •	i the CDO display. le line select key adjace	ant to ADIIS		
			(u)	•					•	
		NOTE: If the last APU shutdown or cycle was an APU protective shutdown and/or the FAULT light is on, the FAULT HISTORY page for the APU BITE TEST shows on the CDU display. The related problem(s) that caused the APU protective shutdown with the date, Greenwich mean time (GMT) and APU cycle will show on this page.								
				TEST shows of	n the CDL		STATUS page for the Al elated problem(s) that o s page.			
				shutdowns, the	e MAIN ME You can fir	ENU page for th	nd there are no APU pro e APU BITE TEST sho nd MAINT lights on the	ws on the		
			(e)				J BITE TEST, push the MENU page for the AP			
				NOTE: The MAIN ME	NU page s	shows on the CD	OU display.			
			(f)				APU BITE TEST, push t MAIN MENU page for			
	NOTE: The MAIN MENU page shows on the CDU display.									
	(g) When the MAIN MENU page for the APU BITE TEST shows on the CDU display, push the line select key adjacent to <ident config.<="" p=""></ident>									
	NOTE: The IDENT/CONFIG page shows the first page of two pages of APU and ECU identification/configuration data.									
				S ALL	source MRB	CDU'S APU MA	INTENANCE PAGES			
						D633A109-AKS 49-140-00-01			Page 2 Feb 15/	



DATE	TAIL NU	IMBER		STATION	AIRLINE CARD NO.	BOEING C/		
	1) The first	t page shov	vs these	APU data:			МЕСН	INSP
	,	PU serial nu						
	,		`	on (APU HOURS	3)			
	•	PU cycles (A	•	•	,			
	,	d) Hours since installation on the airplane.						
	•	Make sure there is identification and configuration data for the APU.						
	,	Push the next page key (NEXT PAGE) to go to the second page which shows these ECU data:						
	a) Pa	art number f	for the E0	CU hardware (E	CU HW P/N)			
	b) EC	CU serial nu	ımber (E	CU S/N)				
	c) Pa	art number f	for the E0	CU software (E0	CU OPERATIONAL SW	′ P/N).		
	4) Make su	ure there is	configura	ation data for th	e ECU.			
		th the line select key adjacent to <index apu="" back="" bite="" for="" go="" main="" menu="" page="" td="" test.<="" the="" to=""><td></td><td></td></index>						
1	NOTE: The M	MAIN MEN	U page s	hows on the CE	OU display.			
(i) F	Push the line	select key	adjacent	to <input mo<="" td=""/> <td>NITORING.</td> <td></td> <td></td> <td></td>	NITORING.			
Ī		OTE: The INPUT MONITORING page shows the first page of four pages of APU engine data.						
	1) The first	The first page shows these APU engine data:						
	a) AF	PU speed (S	SPEED)	(%)				
	b) Ex	haust gas t	temperat	ure, T5 (EGT) (°C).			
	,	is an APU c and two EG	•		re is operational data fo	or the speed		
	Push the line the APU BITE		adjacent	to <index g<="" th="" to=""><th>o back to the MAIN ME</th><th>NU page for</th><th></th><th></th></index>	o back to the MAIN ME	NU page for		
1	NOTE: The M	MAIN MEN	U page s	hows on the CE	OU display.			
` ' .					ΓE systems, push the li ge for the APU BITE TE			
1		MAINT BITE display.	E INDEX	page for the otl	her airplane systems sh	nows on the		
		—— Е	END OF	TASK ———				
EFFECT AKS			SOURCE MRB	CDU'S APU MA	INTENANCE PAGES			
				D633A109-AKS		r	Page 3	of 4
				49-140-00-01			eb 15/	







737-600/700/800/900 TASK CARDS

AIRLINE	E CARD NO	A	TITLE APU EXHAUST SEA	BOEING CARD NO. 49-172-00-01		
DATE	INSPECTION - DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 19000 AH	REPEAT 19000 AH	APPLIC.	
STATION	SKILL ENGIN				AIRPLANE ALL	ALL ALL
		ACCESS 315A 318BR			ZONE 317 318	

Inspect (detailed) the APU exhaust seal.

A. References

Reference	Title
AMM 49-16-11-100-801	Clean the APU Drains (P/B 701)
AMM 49-81-11 P/B 401	EXHAUST DUCT MUFFLER - REMOVAL/INSTALLATION
AMM 49-81-11-000-801	Exhaust Duct Muffler Removal (P/B 401)
AMM 49-81-11-200-803	Exhaust Duct Muffler Seal Inspection (P/B 601)
AMM 49-81-11-400-801	Exhaust Duct Muffler Installation (P/B 401)

B. Consumable Materials

Reference	Description	Specification
A00160	Sealant - Firewall - Hydraulic Fluid Resistant	BMS5-63
G00440	Lockwire - MS20995C41, Corrosion Resistant Steel - 0.041 Inch (1.0414 mm) Diameter	NASM20995

EFFECTIVITY AKS ALL	SOURCE MRB	APU EXHAUST SEAL	
		D633A109-AKS 49-172-00-01	Page 1 of 5 Jun 15/2015



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				49-172-00-01

TASK 49-81-11-200-801

MECH INSP

1. Exhaust Duct Muffler Inspection

(Figure 1)

A. Expendables/Parts

AMM Item	Description	AIPC Reference	AIPC Effectivity
1	Exhaust duct muffler	49-81-00-01A-075	AKS ALL
2	Hi-lok	49-81-11-01A-040	AKS ALL
3	Collar	49-81-11-01A-045	AKS ALL
4	Upper insulation blanket	49-81-11-01A-025	AKS ALL
8	Lower insulation blanket	49-81-11-01A-030	AKS ALL

B. Procedure

SUBTASK 49-81-11-020-003

(1) Remove the exhaust duct muffler [1] (AMM TASK 49-81-11-000-801).

SUBTASK 49-81-11-020-004

(2) Do these steps to remove the upper insulation blanket [4] and lower insulation blanket [8] from the exhaust duct muffler [1]:

NOTE: The rear insulation blanket [7] on the aft end cap is installed permanently with rivets.

CAUTION: BE CAREFUL WITH THE INSULATION BLANKETS. THE BLANKETS CAN BE EASILY DAMAGED. DO NOT LET THE BLANKETS TOUCH SHARP EDGES. DAMAGE TO THE BLANKETS CAN OCCUR.

- (a) Remove the 34 lacings [13] from the upper insulation blanket [4] and lower insulation blanket [8].
- (b) Remove the upper insulation blanket [4] and lower insulation blanket [8] from the exhaust duct muffler [1].

SUBTASK 49-81-11-210-001

- (3) Do these steps to inspect the exhaust duct muffler [1]:
 - (a) Visually examine the upper insulation blanket [4], rear insulation blanket [7] and lower insulation blanket [8] for burns, holes and tears.
 - 1) If you find burns, holes or tears, replace the insulation blanket(s).
 - (b) Visually examine the external surfaces of the exhaust duct muffler [1] for cracks and missing parts.
 - 1) Cracks are not permitted on these components:
 - a) Bellows assembly [11]
 - b) Forward end cap [10]
 - c) Outer liner of the exhaust duct muffler [1]
 - d) Aft liner extension [6]
 - e) Aft end cap [5].

EFFECTIVITY AKS ALL	SOURCE MRB	APU EXHAUST SEAL		
Allena	IVIND			
		D633A109-AKS 49-172-00-01	Page 2 oct 15/2	



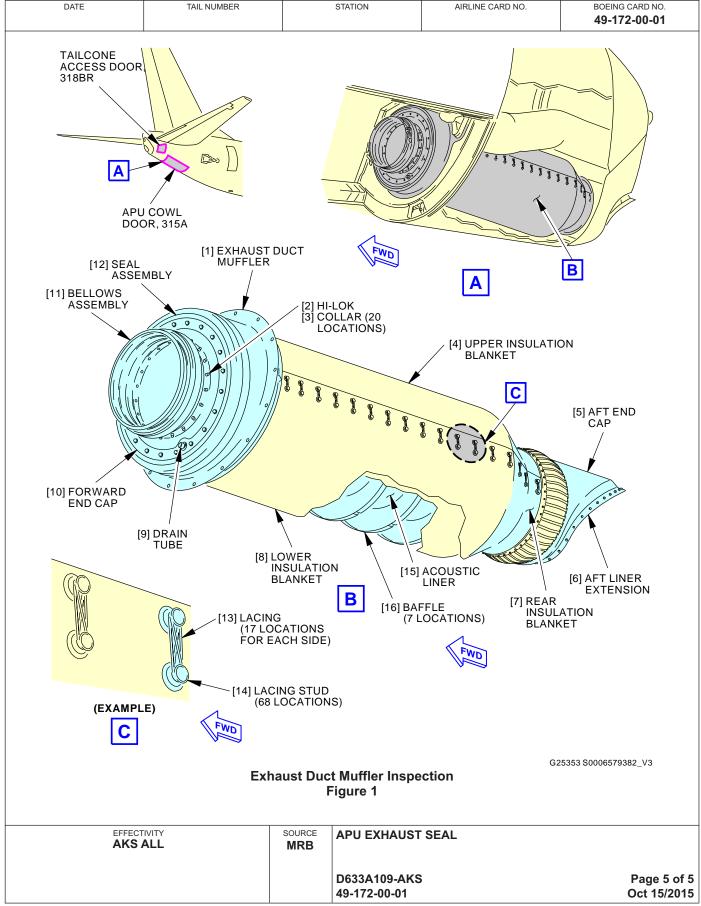
737-600/700/800/900 **TASK CARDS**

DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 2-00-01	
		,	ake sure the 20 sembly [11].	hi-loks [2]	and 20 collars	[3] are installed on the	bellows	MECH	INSP
		NO] attach the bellows ass liner [15] together.	embly [11],		
		3) Or	ne or two missir	ng rivets a	re permitted on	the aft liner extension [6].		
	(c)	•	examine the in unctures and m			aust duct muffler [1] for	cracks,		
		,	•		e the seven baff the seven baff	les [16] and acoustic lin les:	er [15], do		
		a)				ars [3] that attach the b			
		b)	Discard the	20 hi-loks	[2] and 20 colla	rs [3].			
		C)	Remove the	bellows a	ssembly [11] ar	nd acoustic liner [15].			
		d)	Cracks, tear	•	•	nitted on the seven baffl	es [16] and		
		e)	• • • • • • • • • • • • • • • • • • • •			are not permitted, repla DCK 49-81-11/401).	ce the		
		f)			-	s assembly [11] in the fo and align the 20 holes.	orward end		
		g) Apply a thin hi-loks [2].	coat of se	ealant, A00160,	to the threads of the 20	new		
		h)		_	-	urface of the exhaust du e outer surface.	ıct		
		,	the acoustic line e installed on th			make sure the 20 new h	ni-loks [2]		
	(d)	Do this	task: Exhaust D	Ouct Muffle	er Seal Inspection	on, AMM TASK 49-81-1	1-200-803.		
	(e)	Visually	examine the di	rain tube [9] for blockage	of unwanted materials.			
				-		remove or clean the dr ins, AMM TASK 49-16-			
(4)					e that is more th M TASK 49-81-1	an the permitted limits, 1-400-801).	you must		
SUBT	ASK 49-8	1-11-420-003							
(5)			os to install the st duct muffler [ulation blanket [4] and lower insulation	blanket [8]		
	TON	E: The r		olanket [7]	on the aft end o	cap [5] is installed perm	anently with		
	(a)	Put the duct mu		n blanket [[4] and lower ins	sulation blanket [8] on th	ne exhaust		
	(b)	Make sı	ure you align th	e 68 lacin	g studs [14] alor	ng the exhaust duct mu	ffler [1].		
		ECTIVITY S ALL		SOURCE MRB	APU EXHAUST	SEAL			
					D633A109-AKS 49-172-00-01			Page 3 Oct 15/	
					I .				



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING (49-172		
CAL		ETHER. I	DO NOT TIGH	THE ENDS OF THE IN TEN THE LACINGS TO ETS AND THE LACING	O MUCH.	MECH	INSP
(c)	While one person holds blanket [8] together, insta						
	NOTE: You can use 0.0				,		
	NOTE: You install each	lacing by e to six tw	a minimum of vists at the end	one full turn around ead of the lockwire and ben			
(d)	Make sure the distance linsulation blanket [8] is le	between	the upper insula	ation blanket [4] and lov	ver		
(e)	Make sure the 34 lacings	s [13] are	not broken.				
SUBTASK 49-8	1-11-420-004						
(6) Insta	all the exhaust duct muffle	r [1] (AM	1M TASK 49-81	-11-400-801).			
	—— E	END OF	TASK ———				
	ECTIVITY S ALL	SOURCE MRB	APU EXHAUST	SEAL			
			D633A109-AKS 49-172-00-01			Page 4 eb 15/	









737-600/700/800/900 TASK CARDS

AIRLIN	E CARD NO	TITLE APU EDUCTOR			BOEING CARD NO. 49-212-00-01		
DATE	TASK INSPECTION - GEN VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA APU COMPARTMENT	VERSION 1.1	THRESHOLD 10000 AH	REPEAT 10000 AH	APPLIC.	ABILITY ENGINE	
STATION	SKILL ENGIN				ALL	ALL	
		315A			ZONE 315 316		

Perform a general visual inspection of the eductor (on the APU) for general condition.

A. References

ReferenceTitleAMM 49-11-00 P/B 401APU POWER PLANT - REMOVAL/INSTALLATION

B. Consumable Materials

Reference	Description	Specification
B00130	Alcohol - Isopropyl	TT-I-735
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper	BMS15-5 Class A
	(Cheesecloth, Gauze)	

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1057	Air Source - Regulated, Dry Filtered, Compressed 60-105 PSIG
	(414-723.9 KPa)(22 SCFM)

EFFECTIVITY AKS ALL	SOURCE MRB	APU EDUCTOR	
		D633A109-AKS 49-212-00-01	Page 1 of 4 Jun 15/2015



737-600/700/800/900 TASK CARDS

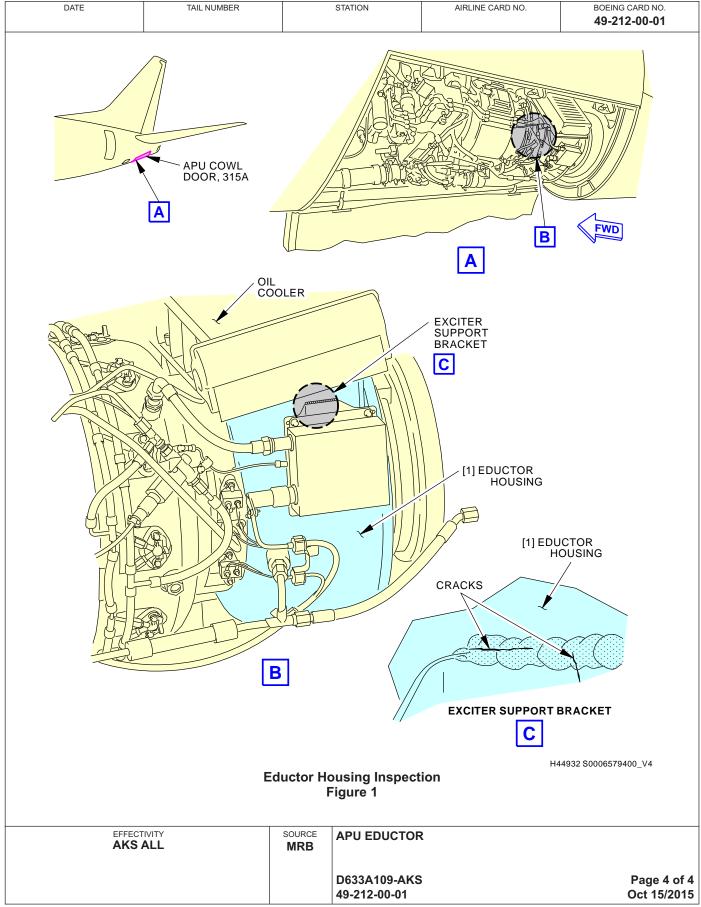
	[DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C			
									49-212		Г	
	Edu		Hous	1-200-8(ing Insp	•					MECH		
	A.	Pre	oare 1	or the I	nspection							
SUBTASK 49-81-41-010-002												
	(1) To open the access panel, do these steps:											
			<u>Nun</u>	<u>nber</u>	Name/Locatio	<u>n</u>						
			315	Д	APU Cowl Doo	r						
			(a)	Suppo	rt the APU pane	l (cowl doc	or) under the c	enter latch.				
			(b)	Open t	the three latches	5.						
				NOTE:	: Use this seque	ence: forwa	ard latch, aft la	atch, middle latch.				
			(c)	Open t	the APU Cowl De	oor, 315A.						
			(d)		ve the retainer pi Door, 315A	in from the	rod end of the	e forward hold-open rod	on the APU			
			(e)	Remov	ve the retainer pi	in from the	spring clip on	the aft hold-open rod.				
			(f)	Discon	nect the two hol	d-open roo	ds from the tw	o spring clips.				
			(g)		ct the two rod er ertment.	nds of the	two hold-open	rods to the two brackets	in the APU			
			(h)	Install	the two retainer	pins in the	two rod ends					
	B.	Pro	cedui	е								
		SUBTA	NSK 49-8	1-41-210-001	I							
		(1)	Do t	hese ste	eps to inspect the	e eductor l	housing [1] (Fi	igure 1):				
			(a)	Visuall	y examine the e	ductor hou	ısing [1] for mi	ssing and damaged bolts	s and nuts.			
				NOTE:	You examine a access from th			ctor housing [1] that you	can get			
					you find missing the grade the dama	_		nuts, install the missing p	oarts or			
			(b)	Visuall	y examine the e	ductor hou	ısing [1] for cra	acks and surface contam	ination.			
				•	lo cracks are pe ne eductor housi		eplace the edu	uctor housing [1] if you fir	nd cracks on			
				_	airplane. Honeywe	Reference II IPC 49-2	Honeywell E 26-93.	with the APU removed fr ngine Manual 49-22-00 a	and			
				2) If	•			, clean the eductor housi	_			
				ć	a) Clean the si cotton wipei			ousing with alcohol, B001	30 and a			
				S ALL		SOURCE MRB	APU EDUCTO)R				

D633A109-AKS 49-212-00-01 Page 2 of 4 Feb 15/2015



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.		G CARD NO. 12-00-01		
	•	•	60-105 PSIG dry curfaces of the e	r filtered regulated air s	source,	MECH	INSF	
	NOTE: It is (414	recomme	nded that you us	se a pressure of 60-90 en to dry the surfaces o				
SUBTASK 49-81	-41-210-002							
(2) Do th Figur	nese steps to inspect the re 1:	e APU exc	iter support brad	cket on the eductor hou	using			
(a)	Visually examine the ex	xciter supp	oort bracket (inc	luding the weld joint).				
	1) Cracks less than	67% (2/3)	of the bracket a	re permitted				
		during a sl		eductor housing is reme s cannot be repaired wi				
	2) If you see a crack (2/3) of the weld jo (AMM PAGEBLO	oint, you n	nust remove the	/3) of the bracket or mo APU	ore than 67%			
	3) If the crack (any le (AMM PAGEBLO	- /	•	enum, you must remov	e the APU			
C. Put the Ai	rplane Back to Its Usu	ıal Condit	tion					
SUBTASK 49-81								
	ose the access panel, d		eps					
<u>Num</u>		_						
315A		-						
	Remove the two retains	•		•	ompartment.			
(b)	Disconnect the two hold- Put the two hold-open in	•			or 215A			
(c)	Install the retainer pin in)I, 3 ISA.			
(a) (e)	Install the retainer pin to			·				
(e) (f)	Close the APU Cowl Do	•		пои-орентой.				
(r) (g)	Close the three latches							
(9)	NOTE: Use this seque		le latch aft latch	n forward latch				
	<u> </u>			i, ioi wara latori				
		END OF	TASK ———					
	S ALL	SOURCE MRB	APU EDUCTOR					
			D633A109-AKS 49-212-00-01			Page 3 eb 15/		







737-600/700/800/900 TASK CARDS

AIRLIN	E CARD NO	APU	TITLE EDUCTOR INLET	BOEING CARD NO. 49-220-00-01		
DATE	INSPECTION - DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 25000 FH	REPEAT 25000 FH	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL ENGIN				ALL	ALL
		ACCESS 315A 318BR			ZONE 317 318	

Inspect (detailed) the eductor inlet duct (interior and exterior).

A. References

Reference	Title
AMM 49-91-71-000-801	Eductor Inlet Duct Removal (P/B 401)
AMM 49-91-71-400-801	Eductor Inlet Duct Installation (P/B 401)

EFFECTIVITY AKS ALL	SOURCE MRB	APU EDUCTOR INLET DUCT	
		D633A109-AKS 49-220-00-01	Page 1 of 6 Jun 15/2015



737-600/700/800/900 TASK CARDS

	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO 49-220-00-01				
		71-200-801 t Duct Insp					MECH	1			
A.	-	e for the In	spection								
	(1) Ma	ake sure the	e APU master s ERATE tag.	switch on t	he P5 forward	overhead panel is OFF	and install a				
		9-91-71-860-002 Den these c	ircuit breakers	and instal	safety tags:						
	F/ <u>R</u>		ll System Pan <u>Number</u>	el, P6-2 <u>Name</u>	SW POWER						
	_	O Electrica ow Col A 14		<u>Name</u>	/ER UNIT COM	NT					
		9-91-71-010-006									
			ccess panel, d		eps:						
			Name/Locatio APU Cowl Doc								
	(a	Support	the APU pane	l (cowl dod	or) under the co	enter latch.					
	(b		e three latches	•	,						
					ard latch, aft la	tch, middle latch.					
	(c	Open th	e APU Cowl D	oor, 315A.							
	(d		e the retainer ploor, 315A	in from the	rod end of the	e forward hold-open rod	on the APU				
	(e)	Remove	the retainer p	in from the	spring clip on	the aft hold-open rod.					
	(f	Disconn	ect the two hol	ld-open ro	ds from the two	o spring clips.					
	(g	Connect compart		nds of the	two hold-open	rods to the two brackets	s in the APU				
	(h	Install th	ne two retainer	pins in the	two rod ends.	•					
		9-91-71-010-002									
	. ,	en this acc									
			<u>Name/Locatio</u> Tailcone Acces								
_			Tallcorle Acces	S D001							
В.	Procedure										
	(1) Do	9-91-71-210-001 these step mpartment	•	e eductor	inlet duct from	the APU compartment a	and tail cone				
	E	FECTIVITY		SOURCE	APU EDUCTO	R INLET DUCT					

D633A109-AKS 49-220-00-01 Page 2 of 6 Feb 15/2015



737-600/700/800/900 TASK CARDS

DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 49-220					
(. ,	Visually examine the you can get access for			aces of the eductor inle	et duct that	MECH	INS			
	 If you find blockage or contamination, remove the blockage or contamination from the eductor inlet duct. 										
(` '	Visually examine the cracks.	the eductor inlet duct	for holes and							
		If you can see h tasks:	oles or cra	cks, replace the	eductor inlet duct. The	se are the					
		 Eductor Inlet I 	Duct Remo	val, AMM TASK	49-91-71-000-801,						
		 Eductor Inlet I 	Eductor Inlet Duct Installation, AMM TASK 49-91-71-400-801.								
(. ,	Visually examine the separations of the fib			the eductor inlet duct	for					
		1) Separations of o	one or two	plies of fiberglas	s material are permitte	d.					
				nore than two plicese are the tasks:	es of fiberglass materia	al, replace					
		 Eductor Inlet I 	Duct Remo	val, AMM TASK	49-91-71-000-801,						
		 Eductor Inlet I 	Duct Install	ation, AMM TASI	K 49-91-71-400-801.						
SUBTASI	K 49-91-	71-210-002									
٠,,	Do these steps to inspect the eductor inlet duct from the upper fairing assembly on the tail cone (Figure 1):										
(. ,) Visually examine the aft end and inner surfaces of the eductor inlet duct that you can get access for blockage and contamination.									
	 If you find blockage or contamination, remove the blockage or contaminat from the eductor inlet duct. 										
((b)	Visually examine the inner surfaces of the eductor inlet duct for holes and cracks.									
		If you can see h tasks	oles or cra	cks, replace the	eductor inlet duct. The	se are the					
		 Eductor Inlet I 	Duct Remo	val, AMM TASK	49-91-71-000-801,						
		 Eductor Inlet I 	Duct Install	ation, AMM TASI	K 49-91-71-400-801.						
(` '	Visually examine the fiberglass material.	inner surfa	ces of the educto	or inlet duct for separa	tions of the					
		1) Separations of o	one or two	plies of fiberglas	s material are permitte	d.					
		, .		nore than two plicese are the tasks:	es of fiberglass materia	al, replace					
		Eductor Inlet I	Duct Remo	val, AMM TASK	49-91-71-000-801,						
		 Eductor Inlet I 	Duct Install	ation, AMM TASI	K 49-91-71-400-801.						
(,	-	le signs of	paint peeling and	or inlet duct for damag I missing paint from th tail cone.	•					
		CTIVITY S ALL	source MRB	APU EDUCTOR	INLET DUCT						
				D633A109-AKS			Page 3				

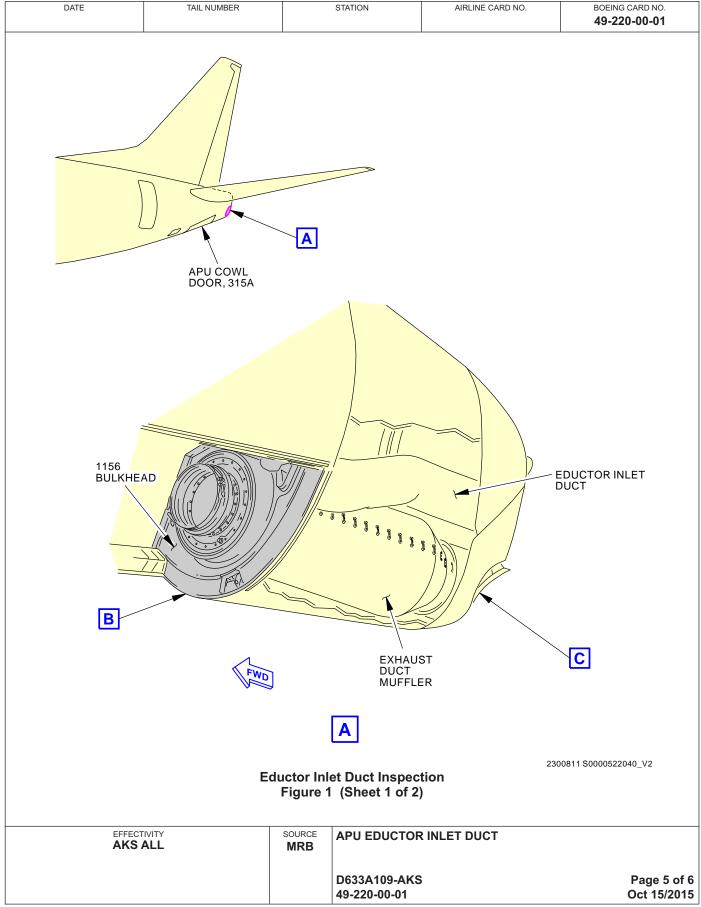
49-220-00-01

Feb 15/2015

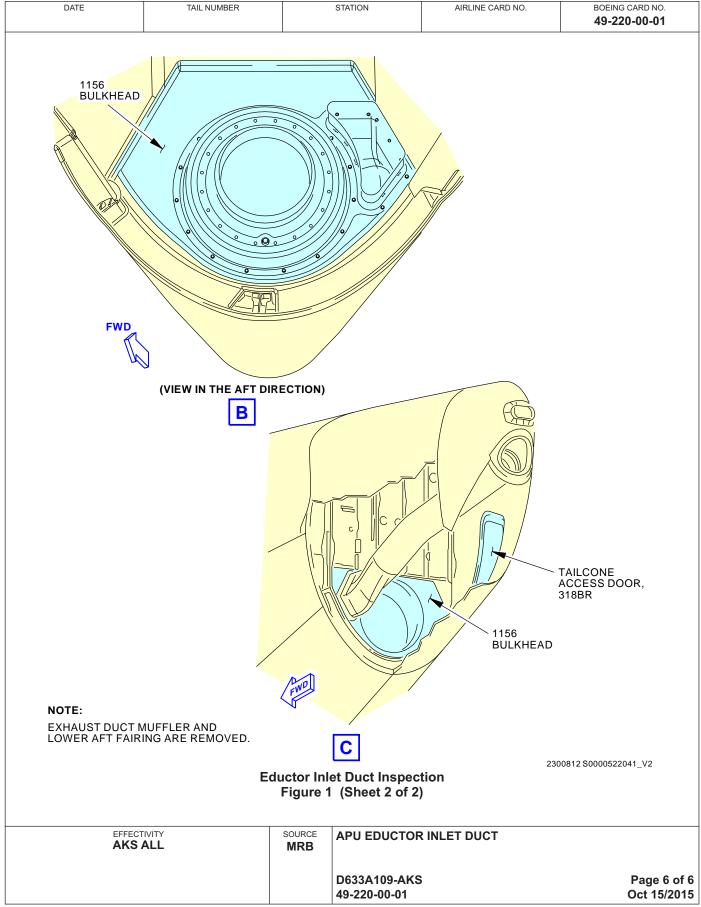


1) Signs of paint peeling and/or missing paint are permitted on the inner surfaces of the eductor inlet duct. The APU can continue in service with no further maintenance action. C. Put the Airplane Back to Its Usual Condition SUBTASK 4991-71-410-007 (1) Close this access panel: Number Name/Location 318BR Tailcone Access Door SUBTASK 4991-71-410-009 (2) To close the access panel, do these steps Number Name/Location 315A APU Cowl Door (a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-604 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel. END OF TASK ——		DATE			TA	AIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 49-220			
(1) Close this access panel: Number Name/Location 318BR Tailcone Access Door SUBTASK 49-91-71-419-097 (2) To close the access panel, do these steps Number Name/Location 315A APU Cowl Door (a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.				, (of th	e eductor inl	et duct. Th	• •	•		MECH	INS	
(1) Close this access panel: Number Name/Location 318BR Tailcone Access Door SUBTASK 49-91-71-419-097 (2) To close the access panel, do these steps Number Name/Location 315A APU Cowl Door (a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.	C.	Put	the A	irplane	Ba	ck to Its Us	ual Condit	tion					
Number Name/Location 318BR Tailcone Access Door				-									
318BR Tailcone Access Door SUBTASK 49-91-71-410-008 (2) To close the access panel, do these steps Number Name/Location 315A APU Cowl Door (a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.		(1)	Clos	e this a	acce	ess panel:							
(2) To close the access panel, do these steps Number Name/Location							<u>on</u>						
(2) To close the access panel, do these steps Number Name/Location			318	BR	Ta	ailcone Acces	ss Door						
Number Name/Location 315A APU Cowl Door (a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.		SUBTA	SK 49-9	1-71-410-00	8								
(a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch subtask 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT Subtask 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.		(2)	Тос	lose the	e ac	cess panel,	do these st	eps					
(a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			Nun	<u>nber</u>	N	ame/Locatio	<u>on</u>						
(b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			315	А	Al	PU Cowl Doo	or						
(c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch subtask 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT subtask 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			(a)	Remo	ve t	he two retair	ner pins fro	m the two hold-	open rods in the APU o	ompartment.			
(d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch subtask 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT subtask 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			(b)	Disco	nne	ct the two ho	old-open ro	ds from the two	brackets.				
(e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			(c)	Put th	e tw	o hold-open	rods in the	e two spring clip	s on the APU Cowl Dod	or, 315A.			
(e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			(d)	Install	the	retainer pin	in the rod	end of the forwa	rd hold-open rod.				
(f) Close the APU Cowl Door, 315A. (g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			` '			•			·				
(g) Close the three latches. NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			` '			•			•				
NOTE: Use this sequence: middle latch, aft latch, forward latch SUBTASK 49-91-71-860-003 (3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			٠,				•						
(3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			(3)					lle latch aft latc	h forward latch				
(3) Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-960-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.		SUDTA	SK 10.0		_	oo ano ooqa	orioo. iiilaa	no latori, art lator	n, forward laterr				
F/O Electrical System Panel, P6-2 Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.						fety tags and	d close the	se circuit breake	ers:				
Row Col Number Name B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.		(-)											
B 19 C01344 APU FIRE SW POWER F/O Electrical System Panel, P6-4 Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.						-							
Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.								SW POWER					
Row Col Number Name A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.					_								
A 14 C00033 AUX POWER UNIT CONT SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.			_			-							
SUBTASK 49-91-71-860-004 (4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.								/ED LINIT CON	т				
(4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.													
——— END OF TASK ———	(4) Remove the DO-NOT-OPERATE tag from the APU master switch on the P5 for								forward				
							- END OF	TASK ———					
I I													
							source MRB	APU EDUCTOR	INLET DUCT				
								D633A109-AKS			Page 4	1 0	
								49-220-00-01			eb 15		













737-600/700/800/900 TASK CARDS

AIRLINE	AIRLINE CARD NO		TITLE TEX GENERATOR	BOEING CARD NO. 49-240-00-01		
DATE	TASK LUBRICATE				RELATE	D CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 16000 FH	REPEAT 16000 FH	APPLIC.	ABILITY ENGINE
STATION	SKILL ENGIN	_			ALL	ALL
		ACCESS			ZONE 300	

Lubricate the vortex generator hinge pin.

A. Consumable Materials

Reference	Description	Specification
D00015	Grease - Aircraft Bearing (Use BMS 3-24 until existing stocks are depleted, BMS 3-33 supersedes BMS 3-24)	BMS3-24 (Superseded by BMS3-33)
D00633	Grease - Aircraft General Purpose	BMS3-33

EFFECTIVITY AKS ALL	SOURCE MRB	APU VORTEX GENERATOR HINGE PIN	
		D633A109-AKS 49-240-00-01	Page 1 of 4 Oct 15/2015



737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 49-240-00-01 месн I INSP TASK 49-15-22-600-801 **Vortex Generator Lubrication** (Figure 1) **Prepare for the Lubrication** SUBTASK 49-15-22-860-005 Make sure the APU master switch on the P5 forward overhead panel is OFF and install a DO-NOT-OPERATE tag. SUBTASK 49-15-22-860-006 Open these circuit breakers and install safety tags: F/O Electrical System Panel, P6-2 Col **Number** Row В C01344 APU FIRE SW POWER 19 F/O Electrical System Panel, P6-4 Row Col Number Name Α 14 C00033 AUX POWER UNIT CONT **Procedure** SUBTASK 49-15-22-640-003 Do these steps to lubricate the parts on the vortex generator: Remove the nut [9], washer [8], two bushings [5], two washers [7], washer [4] and bolt [3] that attaches the vortex generator flap [6] to the vortex generator. 1) If there is wear damage on the two bushings [5], replace the two bushings [5]. (b) Fully lubricate the surfaces of the bolt [3] with a light coat of grease, D00015 or grease, D00633. Lubricate the inner diameter of the vortex generator flap [6] with a light coat of grease, D00015 or grease, D00633. Lubricate the inner and outer diameter of the two bushings [5] with a light coat of grease, D00015 or grease, D00633. Align the vortex generator flap [6] to the vortex generator and install the bushing [5], washer [4], bolt [3], two washers [7], bushing [5], washer [8] and nut [9]. Tighten the nut [9] to 15-20 inch-pounds (1.7-2.3 newton-meters). C. Put the Airplane Back to Its Usual Condition SUBTASK 49-15-22-860-007 Remove the safety tags and close these circuit breakers: F/O Electrical System Panel, P6-2 Row Col Number Name В C01344 APU FIRE SW POWER 19 **FFFFCTIVITY** SOURCE APU VORTEX GENERATOR HINGE PIN **AKS ALL MRB** D633A109-AKS Page 2 of 4

49-240-00-01

Feb 15/2015



DATE		-	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA		
							49-240-0	I	
	E/O EI	octrica	l System Pa	nol D6-4				MECH	INSP
	Row		Number	Name					
	A	14	C00033		WER UNIT CON	т			
	A	14	C00033	AUX PUV	VER UNIT CON	ı			
	ASK 49-15-22								
(2)	Remov overhe	ve the Dead	O-NOT-OPE el.	ERATE tag	from the APU ma	aster switch on the P5 t	forward		
				— END OF	TASK ——				
	EFFECT AKS A	IVITY ALL		SOURCE MRB	APU VORTEX (GENERATOR HINGE PIN			
					D633A109-AKS			2000	OF 4
					49-240-00-01	,		age 3	



