# **CHAPTER**

# 54

# NACELLES/ PYLONS





# CHAPTER 54 NACELLES/PYLONS

Subject/Page	Date	COC	Subject/Page	Date	COC	Subject/Page	Date	coc
54-EFFECTIVE	E PAGES		54-030-01-01	STR		54-060-02-01	STR	
1 thru 5	JUN 15/2016		1	Oct 15/2014		1	Oct 15/2014	
6	BLANK		2	Feb 15/2015		2	Feb 15/2015	
54-010-01-01	STR		3	Oct 15/2015		3	Oct 15/2015	
1	Oct 15/2014		4	Oct 15/2015		4	Oct 15/2015	
2	Feb 15/2015		54-030-02-01	STR		54-070-01-01	STR	
3	Oct 15/2015		1	Oct 15/2014		1	Jun 15/2015	
4	Oct 15/2015		2	Feb 15/2015		2	Feb 15/2015	
54-010-02-01	STR		3	Oct 15/2015		3	Feb 15/2015	
1	Oct 15/2014		4	Oct 15/2015		4	Oct 15/2015	
2	Feb 15/2015		54-040-01-01	STR		5	Oct 15/2014	
3	Oct 15/2015		1	Jun 15/2015		6	Oct 15/2015	
4	Oct 15/2015		2	Feb 15/2015		7	Oct 15/2015	
54-015-01-01	STR		3	Oct 15/2015		54-070-02-01	STR	
1	Jun 15/2015		4	Oct 15/2015		1	Jun 15/2015	
2	Feb 15/2015		54-040-02-01	STR		2	Feb 15/2015	
3	Oct 15/2015		1	Jun 15/2015		3	Feb 15/2015	
4	Oct 15/2015		2	Feb 15/2015		4	Oct 15/2015	
54-015-02-01	STR		3	Oct 15/2015		5	Oct 15/2014	
1	Jun 15/2015		4	Oct 15/2015		6	Oct 15/2015	
2	Feb 15/2015		54-050-01-01	STR		7	Oct 15/2015	
3	Oct 15/2015		1	Jun 15/2015		54-080-01-01	STR	
4	Oct 15/2015		2	Feb 15/2015		1	Jun 15/2015	
54-020-01-01	STR		3			2	Feb 15/2015	
1	Oct 15/2014			Oct 15/2015		3	Feb 15/2015	
2	Feb 15/2015		4	Oct 15/2015		4	Oct 15/2015	
3	Feb 15/2015		54-050-02-01	STR		5	Oct 15/2014	
4	Oct 15/2015		1	Jun 15/2015		6	Oct 15/2015	
5	Oct 15/2015		2	Feb 15/2015		7	Oct 15/2015	
6	Oct 15/2015		3	Oct 15/2015		54-080-02-01	STR	
54-020-02-01	STR		4	Oct 15/2015		1	Jun 15/2015	
1	Oct 15/2014		54-060-01-01	STR		2	Feb 15/2015	
2	Feb 15/2015		1	Oct 15/2014		3	Feb 15/2015	
3	Feb 15/2015		2	Feb 15/2015		4	Oct 15/2015	
4	Oct 15/2015		3	Oct 15/2015		5	Oct 15/2014	
5	Oct 15/2015		4	Oct 15/2015		6	Oct 15/2015	
6	Oct 15/2015					7	Oct 15/2015	

 $A = Added, \ R = Revised, \ D = Deleted, \ O = Overflow, \ C = Customer \ Originated \ Change$ 





# CHAPTER 54 NACELLES/PYLONS

Subject/Page	Date	COC	Subject/Page	Date	COC	Subject/Page	Date	COC
54-090-01-01	SYS		54-611-01-01	FAT (cont)		54-616-00-01	FAT	
1	Jun 15/2015		3	Feb 15/2015		1	Oct 15/2014	
2	Jun 15/2015		54-611-01-02	FAT		2	Feb 15/2015	
3	Oct 15/2014		1	Oct 15/2014		54-616-00-02	FAT	
4	Feb 15/2015		2	Feb 15/2015		1	Oct 15/2014	
5	Oct 15/2014		3	Feb 15/2015		2	Feb 15/2015	
54-090-02-01	SYS		54-612-00-01	FAT		54-617-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
3	Oct 15/2014		54-612-00-02	FAT		54-617-00-02	FAT	
4	Feb 15/2015		1	Oct 15/2014		1	Oct 15/2014	
5	Oct 15/2014		2	Feb 15/2015		2	Feb 15/2015	
54-600-00-01	FAT		54-613-00-01	FAT		54-617-01-01	FAT	
1	Oct 15/2014		1	Oct 15/2014				
2	Feb 15/2015		2	Feb 15/2015		1	Oct 15/2014	
54-600-00-02	FAT		54-613-00-02	FAT		2	Feb 15/2015	
1	Feb 15/2015					54-617-01-02	FAT	
2	Feb 15/2015		1	Oct 15/2014		1	Oct 15/2014	
54-610-00-01	FAT		2	Feb 15/2015		2	Feb 15/2015	
1	Oct 15/2014		54-614-00-01	FAT		54-618-00-01	FAT	
2	Feb 15/2015		1	Oct 15/2014		1	Oct 15/2014	
3	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
54-610-00-02	FAT		54-614-00-02	FAT		54-618-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
3	Feb 15/2015		54-614-01-01	FAT		54-619-00-01	FAT	
54-611-00-01	FAT		1	Oct 15/2014		1	Oct 15/2014	
1	Oct 15/2014		2	Feb 15/2015		2	Feb 15/2015	
2	Feb 15/2015		54-614-01-02	FAT		3	Feb 15/2015	
3	Feb 15/2015		1	Oct 15/2014		54-619-00-02	FAT	
54-611-00-02	FAT		2	Feb 15/2015		1	Oct 15/2014	
1	Oct 15/2014		54-615-00-01	FAT		2	Feb 15/2015	
2	Feb 15/2015		1	Oct 15/2014		3	Feb 15/2015	
3	Feb 15/2015		2	Feb 15/2015		54-619-01-01	FAT	
54-611-01-01	FAT		54-615-00-02	FAT		1	Oct 15/2014	
1	Oct 15/2014		1	Oct 15/2014		2	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		3	Oct 15/2014	

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54-619-01-02	FAT		54-624-00-02	FAT		54-628-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
3	Oct 15/2014		3	Feb 15/2015		54-628-00-02	FAT	
54-620-00-01	FAT		54-625-00-01	FAT		1	Jun 15/2015	
1	Oct 15/2014		1	Oct 15/2014		2	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		54-628-01-01	FAT	
54-620-00-02	FAT		3	Feb 15/2015		1	Jun 15/2015	
1	Oct 15/2014		54-625-00-02	FAT		2	Jun 15/2015	
2	Feb 15/2015		1	Oct 15/2014		54-628-01-02	FAT	
54-621-00-01	FAT		2	Feb 15/2015		1	Jun 15/2015	
1	Oct 15/2014		3	Feb 15/2015		2	Jun 15/2015	
2	Feb 15/2015		54-626-00-01	FAT		54-629-00-01	FAT	
54-621-00-02	FAT		1	Jun 15/2015		1	Oct 15/2014	
1	Oct 15/2014		2	Jun 15/2015		2	Feb 15/2015	
2	Feb 15/2015		3	Jun 15/2015		3	Feb 15/2015	
54-622-00-01	FAT		54-626-00-02	FAT		54-629-00-02	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
54-622-00-02	FAT		3	Jun 15/2015		3	Feb 15/2015	
1	Oct 15/2014		54-627-00-01	FAT		54-630-00-01	FAT	
2	Feb 15/2015		1	Jun 15/2015		1	Oct 15/2014	
54-623-00-01	FAT		2	Jun 15/2015		2	Feb 15/2015	
1	Oct 15/2014		3	Jun 15/2015		3	Feb 15/2015	
2	Feb 15/2015		54-627-00-02	FAT		54-630-00-02	FAT	
3	Feb 15/2015		1	Jun 15/2015		1	Oct 15/2014	
54-623-00-02	FAT		2	Jun 15/2015		2	Feb 15/2015	
1	Oct 15/2014		3	Jun 15/2015		3	Feb 15/2015	
2	Feb 15/2015		54-627-01-01	FAT		54-630-01-01	FAT	
3	Feb 15/2015		1	Jun 15/2015		1	Oct 15/2014	
54-624-00-01	FAT		2	Jun 15/2015		2	Feb 15/2015	
1	Oct 15/2014		3	Jun 15/2015		3	Feb 15/2015	
2	Feb 15/2015		54-627-01-02	FAT		54-630-01-02	FAT	
3	Feb 15/2015		1	Jun 15/2015		1	Oct 15/2014	
			2	Jun 15/2015		2	Feb 15/2015	
			3	Jun 15/2015		3	Feb 15/2015	

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54-631-00-01	FAT		54-635-00-01	FAT		54-640-01-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
3	Oct 15/2014		54-635-00-02	FAT		54-640-01-02	FAT	
54-631-00-02	FAT		1	Oct 15/2014		1	Oct 15/2014	
1	Oct 15/2014		2	Feb 15/2015		2	Feb 15/2015	
2	Feb 15/2015		54-636-00-01	FAT		54-800-01-01	ZON	
3	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
54-632-00-01	FAT		2	Feb 15/2015		2	Feb 15/2015	
1	Oct 15/2014		54-636-00-02	FAT		3	Feb 15/2015	
2	Feb 15/2015		1	Oct 15/2014		4	Feb 15/2015	
3	Oct 15/2014		2	Feb 15/2015		54-802-01-01	ZON	
54-632-00-02	FAT		54-637-00-01	FAT		1	Feb 15/2015	
1	Oct 15/2014		1	Oct 15/2014		2	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		3	Feb 15/2015	
3	Oct 15/2014		54-637-00-02	FAT		54-804-01-01	ZON	
54-632-01-01	FAT		1	Oct 15/2014		R 1	Jun 15/2016	
1	Oct 15/2014		2	Feb 15/2015		R 2	Jun 15/2016	
2	Feb 15/2015		54-638-00-01	FAT		3	Feb 15/2015	
3	Oct 15/2014		1	Oct 15/2014		4	Feb 15/2015	
54-632-01-02	FAT		2	Feb 15/2015		54-806-01-01	ZON	
1	Oct 15/2014		54-638-00-02			1	Jun 15/2015	
2	Feb 15/2015			FAT		2	Feb 15/2015	
3	Oct 15/2014		1	Oct 15/2014		3	Feb 15/2015	
54-633-00-01	FAT		2	Feb 15/2015		4	Feb 15/2015	
	Oct 15/2014		54-639-00-01	FAT		5	Feb 15/2015	
1 2			1	Oct 15/2014		54-808-02-01	ZON	
	Feb 15/2015		2	Feb 15/2015		1	Jun 15/2015	
54-633-00-02	FAT		54-639-00-02	FAT		2	Feb 15/2015	
1	Oct 15/2014		1	Oct 15/2014		3	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		4	Feb 15/2015	
54-634-00-01	FAT		54-640-00-01	FAT		54-810-02-01	ZON	
1	Oct 15/2014		1	Oct 15/2014		1	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
54-634-00-02	FAT		54-640-00-02	FAT		3	Feb 15/2015	
1	Oct 15/2014		1	Oct 15/2014		54-812-02-01	ZON	
2	Feb 15/2015		2	Feb 15/2015		R 1	Jun 15/2016	

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# CHAPTER 54 NACELLES/PYLONS

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54-812-02-01	ZON (cont)							
R 2	Jun 15/2016							
3	Feb 15/2015							
4	Feb 15/2015							
54-814-02-01	ZON							
1	Jun 15/2015							
2	Feb 15/2015							
3	Feb 15/2015							
4	Feb 15/2015							
5	Feb 15/2015							

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### 737-600/700/800/900 TASK CARDS

CARD NO	LEFT FORWA	TITLE ARD ENGINE MOU	BOEING ( <b>54-010</b>			
TASK GENERAL VISUAL				RELATE	D CARD	
WORK AREA LEFT ENGINE	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR		APPLICABILITY  LANE ENGINE	
SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL	
	ACCESS 413 414 NOTE			ZONE 413 414		
	GENERAL VISUAL WORK AREA LEFT ENGINE	TASK GENERAL VISUAL  WORK AREA LEFT ENGINE SKILL AIRPL  ACCESS 413 414	TASK GENERAL VISUAL  WORK AREA LEFT ENGINE SKILL AIRPL  LEFT FORWARD ENGINE MOU  THRESHOLD THRESHOLD THRESHOLD THRESHOLD THRESHOLD THRESHOLD THRESHOLD THRESHOLD ACCESS	TASK   GENERAL   VISUAL   WORK AREA   LEFT ENGINE   1.1   6 YR   6 YR   1.2   18000 FC   18000 FC   AIRPL   ACCESS   413 414	TASK   GENERAL VISUAL   WORK AREA   LEFT ENGINE   1.1   6 YR   6 YR   AIRPLANE   ALL   NOTE   ACCESS   413 414   413 414   413 414   54-010   54-010   54-010   54-010   6   6   6   6   6   6   6   6   6	

Inspect left forward engine mount assembly, including fan case fitting, side links, hanger, and link pins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open fan cowl.

A. References

 Reference
 Title

 AMM 51-00-58
 STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT FORWARD ENGINE MOUNT ASSEMBLY	
		D633A109-AKS 54-010-01-01	Page 1 of 4 Oct 15/2014



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-010		
	TAS	K 54-05-0	03-210-8	01					MECH	INSP
1.	EXT	ERNAL -	GENER	AL VISUAL: LEFT	FORW	ARD ENGINE N	OUNT ASSEMBLY			
	(Fig	ure 1)								
	A.	Inspecti	ion							
			I-05-03-010-00	1						
		(1) Op	en these	access panels:						
		<u>Nu</u>	<u>ımber</u>	Name/Location						
		413		Left Fan Cowl, E	-					
		414		Right Fan Cowl,	Engine 1	1				
		<u>NC</u>	<u>)TE</u> : Ope	en fan cowl.						
			-05-03-210-00			l = <b>f</b> + f =	in a manage to a second by the	lll f		
				side links, hanger			ine mount assembly, ir	icluding fan		
			I-05-03-910-00							
				Basik Task Descrip	tion, AMI	M Task 51–05–0	01–210–809.			
			1-05-03-410-00	access panels:						
			ımber	Name/Location						
		413		Left Fan Cowl, E	naine 1					
		414		Right Fan Cowl,	_	1				
				——— Е	END OF	TASK ———				
			FECTIVITY		SOURCE	LEFT FORWAR	D ENGINE MOUNT ASS	SEMBLY		
		A	NO ALL		MRB					
						D633A109-AKS	i .		Page 2	



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-010-				
	TAS	K 51	-05-01-	210-809				1	MECH	INSF		
2.	737	-6789	Basic	Task Description								
	A.	CPO	CP Bas	ic Task								
		SUBT	ASK 51-05-0									
		(1)	CPCF	Basic Task Item 1 is not applicable.								
		(-)	ASK 51-05-0	1 - 1 - 1 - 1								
		(2)		CPCP Basic Task Item 2 as follows:								
			1	Prior to inspection clean the area as required to accomplish CPCP Basic Task Item B. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the netal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.								
		SUBT	ASK 51-05-0	01-210-080								
		(3)	Do the	e CPCP Basic Task Ite	em 3 as fol	lows:						
			(	as specified in each ta visual inspections follo corrosion, such as bul etc. In the task area, c determine if removal is	sk descrip wing partiaging skins heck the ir required,	tion. Use Additional disassembly in or corrosion run ontegrity of any second and any corrosion	scription. The inspection on all non-destructive insomal non-destructive insomal fitnere are indications of the are indications of the area of the are	pections or of hidden der fittings, and to l, particularly				
		SUBT	ASK 51-05-0	01-210-081								
		(4)	Do the	e CPCP Basic Task ite	m 4 as foll	lows:						
			: -     	structure as required, i TREATMENT METHO Manual (SRM) D634A D634A211(-900), D634	including a DS, AMM 200, (-600 4A333 (BB errous meta	application of pro SUBJECT 51-00 ), D634A201 (-7 BJ), or related se	air or replace all discreptective finishes per ST/0-58, or 737 Structural (700), D634A210 (-800), ervice bulletin, as approvibe handled by normal	ANDARD Repair priate.				
		SUBT	ASK 51-05-0	01-210-082								
		(5)	CPCF	Basic Task Item 5 is	not applica	ıble.						
		SUBT	ASK 51-05-0									
		(6) Do the CPCP Basic Task Item 6 (Not Applicable)										
			CPCF	o1-210-084 P Basic Task Item 7 is I	not applica	ıble.						
					- END OF	TASK ———						
			EFFEC AKS		SOURCE MRB	LEFT FORWAR	D ENGINE MOUNT ASSI	EMBLY				

TAIL NUMBER

DATE

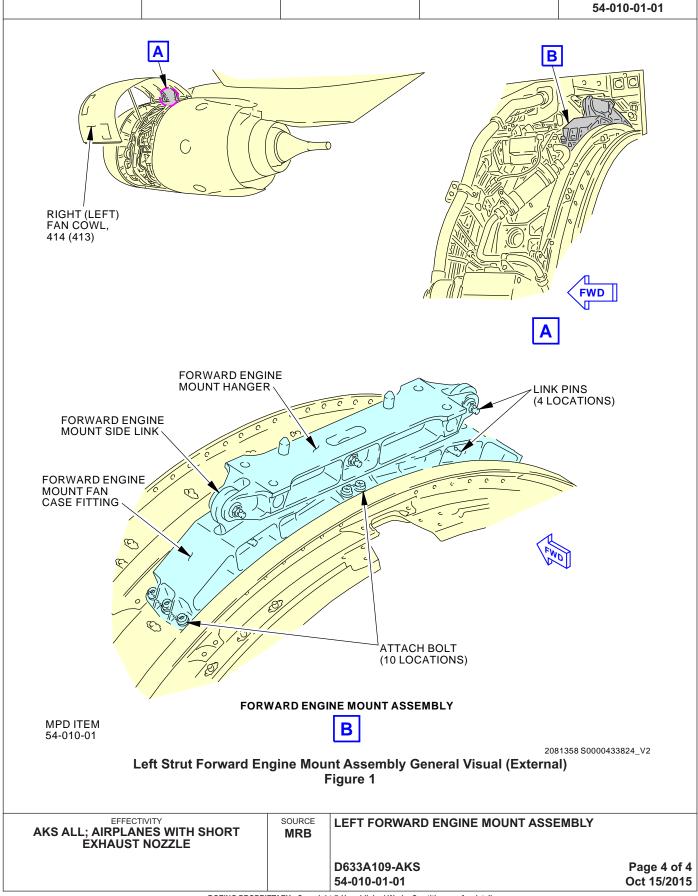


### 737-600/700/800/900 TASK CARDS

STATION

AIRLINE CARD NO.

BOEING CARD NO.





### 737-600/700/800/900 TASK CARDS

E CARD NO	RIGHT FORWA	TITLE ARD ENGINE MOU	54-010			
TASK GENERAL VISUAL				RELATE	D CARD	
WORK AREA RIGHT ENGINE	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR		APPLICABILITY  LANE ENGINE	
SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL	
	ACCESS 423 424 NOTE			ZONE 423 424		
	TASK GENERAL VISUAL WORK AREA RIGHT ENGINE SKILL	TASK GENERAL VISUAL  WORK AREA RIGHT ENGINE  SKILL AIRPL  ACCESS 423 424	TASK GENERAL VISUAL  WORK AREA RIGHT ENGINE SKILL AIRPL  ACCESS 423 424	TASK   GENERAL   VISUAL   WORK AREA   RIGHT ENGINE   MOUNT ASSEMBLY   MORK AREA   RIGHT ENGINE   1.1   6 YR   6 YR	TASK GENERAL VISUAL  WORK AREA RIGHT ENGINE  SKILL AIRPL  RIGHT FORWARD ENGINE MOUNT ASSEMBLY  TASK RELATE  STATE  STATE  STATE  TASK RELATE  THRESHOLD REPEAT APPLIC AIRPL  ACCESS  REPEAT APPLIC AIRPL ACCESS  ZONE	

Inspect right forward engine mount assembly, including fan case fitting, side links, hanger, and link pins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open fan cowl.

A. References

ReferenceTitleAMM 51-00-58STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT FORWARD ENGINE MOUNT ASSEMBLY	
		D633A109-AKS 54-010-02-01	Page 1 of 4 Oct 15/2014



г	ATE	I	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C	ARD N∩	
			MEROMBER		5.7(10)(	AUTEURE OFFICE NO.	54-010		
TAS	K 54-	05-03-210-8	302					MECH	INSP
EXT	ERN	AL - GENER	RAL VISUAL: RIC	SHT FORV	VARD ENGINE	MOUNT ASSEMBLY			
(Fig	ure 1)								
A.	Insp	ection							
		ASK 54-05-03-010-0							
	(1)		e access panels:						
		Number 423	Name/Locatio						
		423 424	Left Fan Cowl, Right Fan Cow		2				
			pen fan cowl.	, 3					
	SUBTA	NSK 54-05-03-210-0							
	(2)					gine mount assembly, i	ncluding fan		
		case fitting	, side links, hang	er, and link	c pins.				
		ASK 54-05-03-910-0	<sub>02</sub> Basic Task Descr	rintion AM	M Took 51 05 0	11 210 800			
	(3)	7 3 7 0 7 0 9 ASK 54-05-03-410-0		iption, Aivii	W 185K 51-05-0	71–210–609.			
	(4)		e access panels:						
	( )	<u>Number</u>	Name/Locatio						
		423	Left Fan Cowl,	-					
		424	Right Fan Cow	/I, Engine 2	2				
				- END OF	TASK ———				
					I				
		AKS ALL		SOURCE MRB	RIGHT FORWAR	RD ENGINE MOUNT AS	SEMBLY		
					D0004400 4175			_	
					D633A109-AKS 54-010-02-01			Page 2 eb 15/	
							<u> </u>	01	



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO.			
								54-010	0-02-01	INICD		
				-210-809					MECH	INSP		
2.	737	-6789	Basic	Task Description								
	A.	CPO	CP Bas	sic Task								
		SUBT	ASK 51-05	-01-210-078								
		(1)	CPC	P Basic Task Item 1 is	not applica	able.						
		SUBT		-01-210-079								
		(2)	Do th	ne CPCP Basic Task Ite	em 2 as fo	llows:						
			(a)	3. It is not necessary tunless it has deteriora metal. A light uniform accumulated dirt or details.	to remove ated to the film of Corebris, will nay require	normal amounts point where moi rosion Inhibiting formally allow ac	accomplish CPCP Basic of sealant/leveling com- sture can penetrate dov Compound (CIC) that had lequate inspection of the re are multiple layers an	pound vn to the as not e structure				
		SUBT	ASK 51-05	-01-210-080								
		(3)	Do th	ne CPCP Basic Task Ite	em 3 as fo	llows:						
			(a)	as specified in each to visual inspections follo corrosion, such as bul etc. In the task area, of determine if removal i	ask descrip owing parti ging skins check the i s required,	otion. Use Additional disassembly in or corrosion runderity of any solution and any corros	scription. The inspection onal non-destructive ins fithere are indications or uning into splices, or uncealant/leveling compour ion inhibiting compound lication is required per Compour ion inhibiting compound lication is required per Compound in the comp	pections or f hidden der fittings, nd to , particularly				
		SUBTASK 51-05-01-210-081										
		(4)	Do th	ne CPCP Basic Task ite	em 4 as fol	llows:						
			(a)	structure as required, TREATMENT METHO Manual (SRM) D634A D634A211(-900), D63	including a DDS, AMM 200, (-600 4A333 (BE errous met	application of pro SUBJECT 51-0 )), D634A201 (-7 BJ), or related se	air or replace all discrept otective finishes per STA 0-58, or 737 Structural I (00), D634A210 (-800), ervice bulletin, as approp to be handled by normal	ANDARD Repair oriate.				
		SUBT	ASK 51-05	-01-210-082								
		(5)	CPC	P Basic Task Item 5 is	not applica	able.						
				-01-210-100								
		(6)	Do th	ne CPCP Basic Task Ito	em 6 (Not a	Applicable)						
				-01-210-084		-1-1-						
		(7)	CPC	P Basic Task Item 7 is								
					– END OF	TASK ———						
				CTIVITY S ALL	SOURCE MRB	RIGHT FORWA	RD ENGINE MOUNT ASS	SEMBLY	I	I		
						D633A109-AKS 54-010-02-01			Page 3 Oct 15/			

TAIL NUMBER

DATE

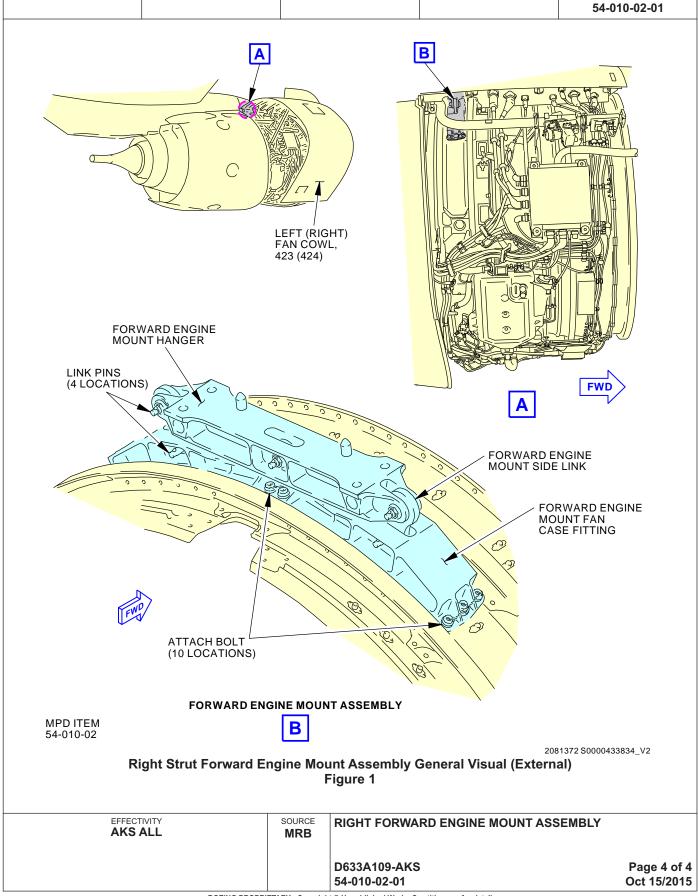


### 737-600/700/800/900 TASK CARDS

STATION

AIRLINE CARD NO.

BOEING CARD NO.





### 737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE  LEFT STRUT ATTACH BOLTS AT FORWARD			BOEING CARD NO. <b>54-015-01-01</b>		
DATE	TASK GENERAL VISUAL	-	ENGINE MOUNT		RELATE	D CARD	
TAIL NUMBER	WORK AREA LEFT ENGINE	VERSION 1.1	THRESHOLD  9 YR	REPEAT  9 YR	APPLIC		
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	AIRPLANE ALL	ALL ALL	
		ACCESS 413 414 431AT NOTE			ZONE 413 414		

Inspect left strut attach bolts at forward engine mount.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove fan cowl.

A. References

 Reference
 Title

 AMM 51-00-58
 STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE	LEFT STRUT ATTACH BOLTS AT FORWARD ENG	SINE MOUNT
		D633A109-AKS 54-015-01-01	Page 1 of 4 Jun 15/2015



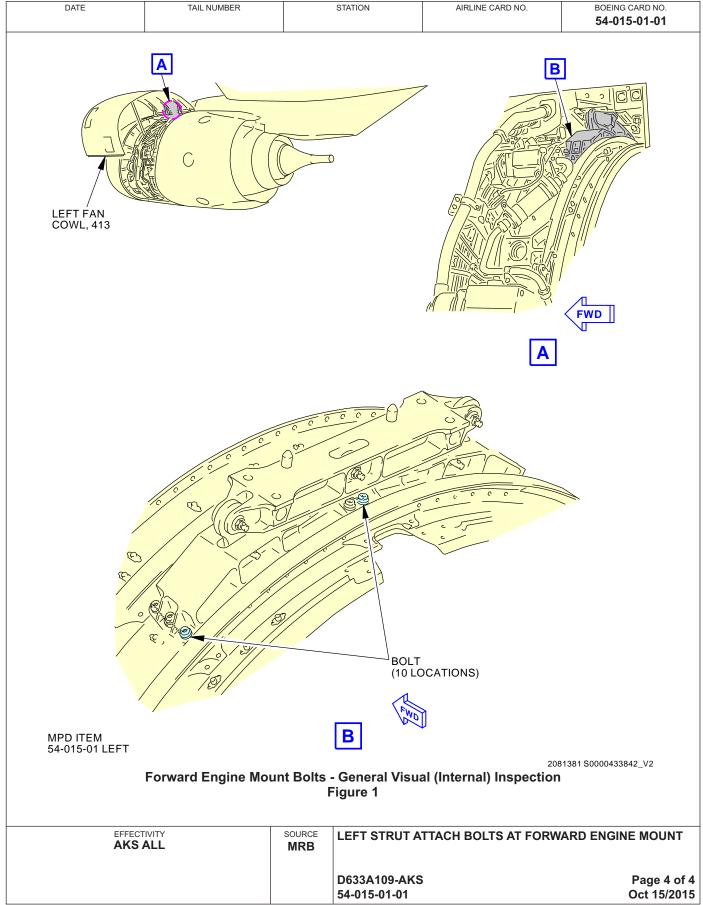
### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-015-01-01 MECH INSP TASK 54-05-03-210-803 INTERNAL - GENERAL VISUAL: LEFT STRUT ATTACH BOLTS AT FORWARD ENGINE MOUNT (Figure 1) Α. Inspection SUBTASK 54-05-03-010-003 (1) Open these access panels: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 431AT Forward Strut Fairing, Thumbnail Fairing, Strut 1 NOTE: Remove fan cowl. SUBTASK 54-05-03-210-003 (2) Do a Genral Visual inspection of the left strut attach bolts at forward engine mount. SUBTASK 54-05-03-910-003 (3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–809. SUBTASK 54-05-03-410-003 (4) Close these access panels: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 431AT Forward Strut Fairing, Thumbnail Fairing, Strut 1 — END OF TASK —— **FFFFCTIVITY** SOURCE LEFT STRUT ATTACH BOLTS AT FORWARD ENGINE MOUNT **AKS ALL MRB** D633A109-AKS Page 2 of 4 54-015-01-01 Feb 15/2015



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 5-01-01	
	TAS	K 51	-05-01-	210-809					MECH	INSP
2.	737	-6789	Basic	Task Description						
	A.	СРО	CP Bas	ic Task						
		SUBTA	ASK 51-05-	01-210-078						
		(1)	CPCF	P Basic Task Item 1 is r	not applica	able.				
				<sub>01-210-079</sub> e CPCP Basic Task Ite	m O oo fol	lourer				
		(2)					accomplish CPCP Basi	c Task Itam		
				<ol> <li>It is not necessary to unless it has deterioral metal. A light uniform fi accumulated dirt or de</li> </ol>	o remove rated to the pilm of Correling will no bris, will no nay require	normal amounts point where moi rosion Inhibiting ormally allow ac	of sealant/leveling comsture can penetrate down Compound (CIC) that he lequate inspection of the are multiple layers an	npound vn to the las not e structure		
		SUBTA		01-210-080						
		(3)		e CPCP Basic Task Ite						
				as specified in each ta visual inspections follo corrosion, such as buld etc. In the task area, c determine if removal is	sk descrip wing partia ging skins heck the ir required,	tion. Use Additional disassembly in or corrosion runategrity of any seand any corros	scription. The inspection on all non-destructive ins fithere are indications or underling into splices, or undealant/leveling compound ion inhibiting compound lication is required per Compound	pections or f hidden der fittings, nd to , particularly	1	
		SUBTA	ASK 51-05-							
		(4)		e CPCP Basic Task ite						
				structure as required, i TREATMENT METHO Manual (SRM) D634A D634A211(-900), D634	ncluding a DS, AMM 200, (-600 4A333 (BB rrous meta	application of pro SUBJECT 51-0 ), D634A201 (-7 BJ), or related se	air or replace all discrept otective finishes per STA 0-58, or 737 Structural I (00), D634A210 (-800), ervice bulletin, as appropt by be handled by normal	ANDARD Repair priate.		
		,		01-210-082						
		(5)		P Basic Task Item 5 is r	not applica	able.				
		(6)	ASK 51-05- Do th	<sub>01-210-100</sub> e CPCP Basic Task Ite	m 6 (Not A	Applicable)				
		` '		01-210-084	0 (11017	τρριισασίο)				
		(7)		P Basic Task Item 7 is r	not applica	able.				
					- END OF	TASK ——				
			FFFF	TIVITY	COLIDOR	L EET OTDUT A	TT4 OU DOLTO 4T FORM	4 DD ENOIN	- NO.	\
			AKS	ALL	MRB	LEFISIRUIA	TTACH BOLTS AT FORW	AKD ENGIN	E WIOU	IN I
						D633A109-AKS 54-015-01-01	i		Page 3 Oct 15/	







### 737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT STRUT ATTACH BOLTS AT FORWARD			BOEING CARD NO. <b>54-015-02-01</b>		
DATE	TASK GENERAL VISUAL		ENGINE MOUNTS	<b>3</b>	RELATE	D CARD	
TAIL NUMBER	WORK AREA RIGHT ENGINE	VERSION 1.1	THRESHOLD  9 YR	REPEAT  9 YR	APPLIC		
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL ALL	
		ACCESS 423 424 441AT NOTE			ZONE 423 424		

Inspect right strut attach bolts at forward engine mount.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove fan cowl.

A. References

 Reference
 Title

 AMM 51-00-58
 STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT ATTACH BOLTS AT FORWARD EMOUNTS	NGINE
		D633A109-AKS 54-015-02-01	Page 1 of 4 Jun 15/2015



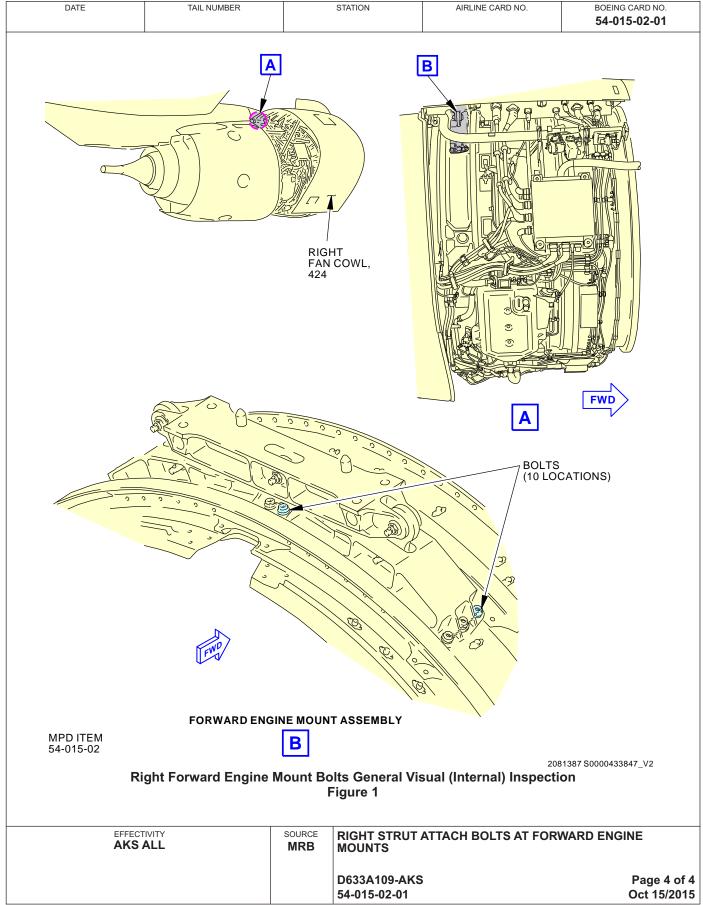
### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-015-02-01 MECH INSP TASK 54-05-03-210-804 INTERNAL - GENERAL VISUAL: RIGHT STRUT ATTACH BOLTS AT FORWARD ENGINE **MOUNTS** (Figure 1) A. Inspection SUBTASK 54-05-03-010-004 (1) Open these access panels: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 Forward Strut Fairing, Thumbnail Fairing, Strut 2 441AT NOTE: Remove fan cowl. SUBTASK 54-05-03-210-004 (2) Do a General Visual inspection of the right strut attach bolts at forward engine mount. SUBTASK 54-05-03-910-004 (3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–809. SUBTASK 54-05-03-410-004 (4) Close these access panels: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 441AT Forward Strut Fairing, Thumbnail Fairing, Strut 2 —— END OF TASK —— **FFFFCTIVITY** SOURCE RIGHT STRUT ATTACH BOLTS AT FORWARD ENGINE **AKS ALL MRB MOUNTS** D633A109-AKS Page 2 of 4 54-015-02-01 Feb 15/2015



D NO. <b>2-01</b>
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### 737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		LEFT STRUT FORWARD AND AFT ENGINE MOUNT			BOEING CARD NO. <b>54-020-01-01</b>		
DATE	TASK GENERAL VISUAL	TC	STRUT SHEAR PI	NS	RELATEI	O CARD	
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD ENG CNG	REPEAT	APPLICA		
STATION	SKILL AIRPL	NOTE			AIRPLANE <b>ALL</b>	ALL ALL	
		ACCESS NOTE			ZONE 413 414 433		

Inspect forward and aft engine mount to strut shear pins.

Engine removal:

AMM 71-00-02 -POWER PLANT - REMOVAL/INSTALLATION

Related procedures:

AMM 71-21-01 - FORWARD ENGINE MOUNT REMOVAL/INSTALLATION

AMM 71-21-03 - AFT ENGINE MOUNT REMOVAL/INSTALLATION

**INTERVAL NOTE:** At engine removal.

ACCESS NOTE: Engine removal required.

### A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT FORWARD AND AFT EN SHEAR PINS	GINE MOUNT TO STRUT
		D633A109-AKS 54-020-01-01	Page 1 of 6 Oct 15/2014



	D/	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-020</b>				
Т	ΓΔSΙ	K 54-	05-03-210-8	05	'		1		MECH	INS		
_	TASK 54-05-03-210-805 INTERNAL - GENERAL VISUAL: LEFT STRUT FORWARD AND AFT ENGINE MOUNT TO											
_			HEAR PINS					<u></u>				
_		re 1)										
•		,										
,	Α.	-	ection									
			BTASK 54-05-03-010-005 ) Open these access panels:									
		(1)			_							
			Number	Name/Location								
			413 414	Left Fan Cowl, Right Fan Cow	-	1						
			414	Left Thrust Re	_							
			416	Right Thrust R								
			416CR	•		•	Hinge Fairing, Engine 1					
			431AL	•	•		ser Disconnect, Strut 1					
			431AT			umbnail Fairing						
			431BL	Forward Strut	Fairing, Le	ft Mid Strut Fair	ing, Strut 1					
			431BR		•	ght Mid Strut Fa	•					
			431CL		•	ft Overwing Fair						
			431DL		-	ft Underwing Fa	_					
			431DR		_	ght Underwing F	_					
			431EL 431ER		•	ft T.R. Strut Fair ght T.R. Strut Fa	•					
					r airirig, ixi	giit i.ix. Stiut i a	annig, Strut i					
			Special Acc									
	Number Name/Location											
			S4001	Left Strut Forw	ard and At	t Engine Mount	Assemblies					
			NOTE: Engine removal required.									
		SUBTA	ASK 54-05-03-210-00	05								
		(2)	Do a Gener	ral Visual inspect	tion of the	forward and aft	engine mount to strut she	ar pins.				
		SUBTA	ASK 54-05-03-910-01	19								
		(3)	737–6789 E	Basic Task Desci	ription, AM	M Task 51–05–0	01–210–809.					
		SUBTA	ASK 54-05-03-410-00	05								
		(4)	Close these	e access panels:								
			Number	Name/Location	<u>n</u>							
			413	Left Fan Cowl,	Engine 1							
			414	Right Fan Cow	∕l, Engine ′	1						
			415	Left Thrust Re	verser, Enç	gine 1						
			416	Right Thrust R		-						
			416CR		-		Hinge Fairing, Engine 1					
			431AL		-		ser Disconnect, Strut 1					
			431AT		-	umbnail Fairing						
			431BL	Forward Strut	Fairing, Le	ft Mid Strut Fair	ing, Strut 1					
			EFFECTIVITY AKS ALL		SOURCE		ORWARD AND AFT ENGIN	E MOUNT 1	O STI	RU		
			ANO ALL		MRB	SHEAR PINS						
						D633A109-AKS			Page 2			
						54-020-01-01		F	eb 15/	201		



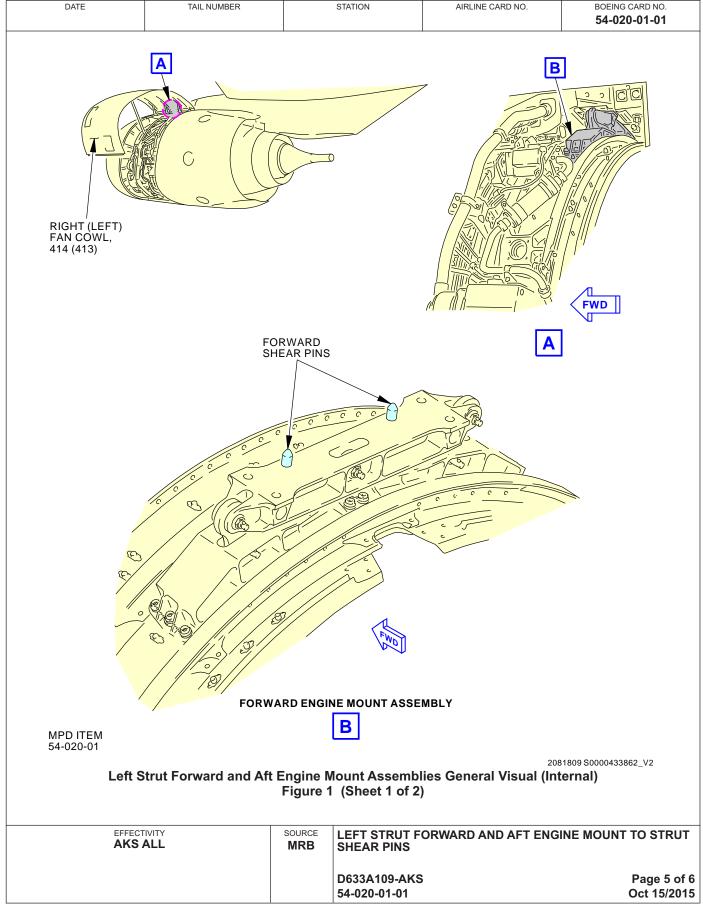


DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-020-01-01</b>	
(Cont	inued)			MECH	INSP
Number 431BR 431CL 431DL 431DR 431EL 431ER	Forward Strut Farage Forward S	airing, Right Mid Strut Fa airing, Left Overwing Fair airing, Left Underwing Fa airing, Right Underwing Fa airing, Left T.R. Strut Fair airing, Right T.R. Strut Fa	ring, Strut 1 iiring, Strut 1 Fairing, Strut 1 ing, Strut 1		
EFFECTI AKS A		SOURCE HEFT STRUT FOR SHEAR PINS  D633A109-AKS 54-020-01-01	ORWARD AND AFT ENG	GINE MOUNT TO STR Page 3 Feb 15/2	of 6

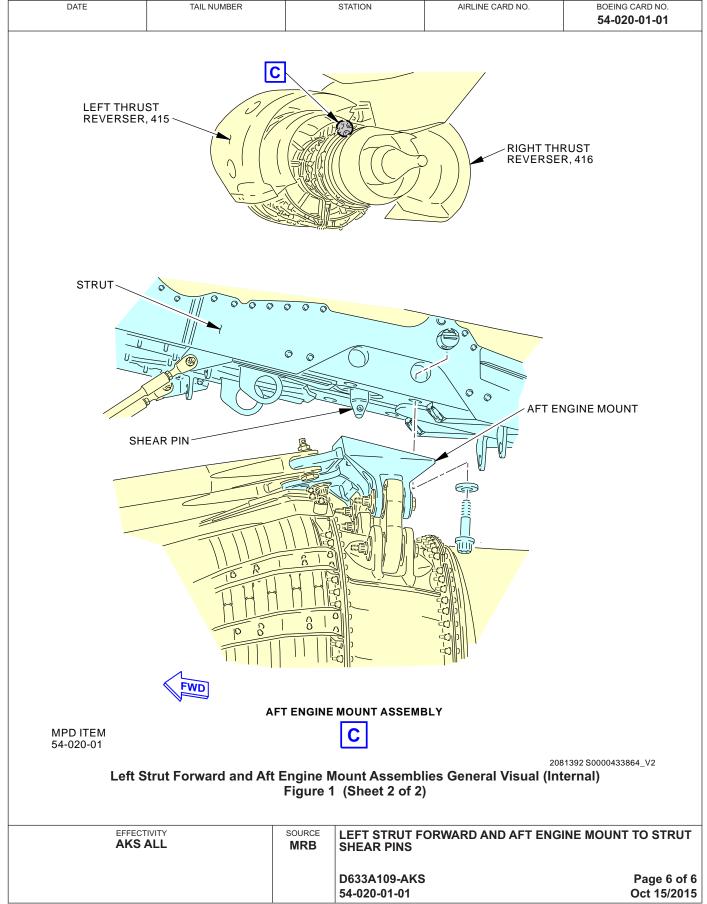


	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	54-020				
2.				I-210-809 c Task Description	'				MECH	INSF		
	Α.											
		SUBT	ASK 51-0	5-01-210-078								
		(1)	CPC	CP Basic Task Item 1 is	s not applica	able.						
		SUBT	ASK 51-0	5-01-210-079								
		(2)	Do t	he CPCP Basic Task I	tem 2 as fol	lows:						
			(a)	3. It is not necessary unless it has deterior metal. A light uniform accumulated dirt or o	to remove rated to the particular film of Correlebris, will not may require	normal amounts point where mois osion Inhibiting ormally allow ad	accomplish CPCP Bas of sealant/leveling cor sture can penetrate do Compound (CIC) that equate inspection of the e are multiple layers a	mpound wn to the has not ne structure				
		SUBTASK 51-05-01-210-080										
		(3)	Do t	he CPCP Basic Task I	tem 3 as fol	lows:						
		(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.										
		SUBT	ASK 51-0	5-01-210-081								
		(4)	Do t	he CPCP Basic Task i	tem 4 as fol	lows:						
			(a)	structure as required TREATMENT METH Manual (SRM) D634 D634A211(-900), D6	, including a ODS, AMM A200, (-600 34A333 (BB ferrous meta	application of pro SUBJECT 51-00 ), D634A201 (-7 BJ), or related se	air or replace all discreptective finishes per ST 0-58, or 737 Structural 00), D634A210 (-800), rvice bulletin, as apprope be handled by norma	ANDARD Repair ppriate.				
		SUBT	ASK 51-0	5-01-210-082								
		(5)	CPC	CP Basic Task Item 5 is	s not applica	able.						
		SUBT		5-01-210-100								
		(6)	Do t	he CPCP Basic Task I	tem 6 (Not A	Applicable)						
				5-01-210-084	4 - · · P	.h.l.						
		(7)	CPC	CP Basic Task Item 7 is								
					— END OF	TASK ——						
				ECTIVITY S ALL	SOURCE MRB	LEFT STRUT FO	ORWARD AND AFT ENG	GINE MOUNT	то ѕті	RUT		











### 737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	TITLE RIGHT STRUT FORWARD AND AFT ENGINE			BOEING CARD NO. <b>54-020-02-01</b>		
DATE	TASK GENERAL VISUAL	MOUN	T TO STRUT SHEA	RELATED CARD			
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD ENG CNG	REPEAT	APPLICABILITY  AIRPLANE ENGINE		
STATION	SKILL AIRPL	NOTE	NOTE			ALL	
		ACCESS NOTE			ZONE 423 424 443		

Inspect forward and aft engine mount to strut shear pins.

Engine removal:

AMM 71-00-02 - POWER PLANT - REMOVAL/INSTALLATION

Related procedures:

AMM 71-21-01 - FORWARD ENGINE MOUNT REMOVAL/INSTALLATION

AMM 71-21-03 - AFT ENGINE MOUNT REMOVAL/INSTALLATION

**INTERVAL NOTE:** At engine removal.

ACCESS NOTE: Engine removal required.

### A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT FORWARD AND AFT ENGINE MO STRUT SHEAR PINS	OUNT TO
		D633A109-AKS 54-020-02-01	Page 1 of 6 Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING ( <b>54-02</b> 0	CARD NO. 0-02-01			
TA	SK 54	-05-03-210-8	06					MECH	INSF		
. IN	ΓERNA	AL - GENERA	AL VISUAL: RIG	HT STRU	T FORWARD A	ND AFT ENGINE MOU	NT TO				
ST	RUT S	HEAR PINS									
(Fi	gure 1)	)									
A.	Insp	pection									
	-	ASK 54-05-03-010-00	6								
	(1)	Open these	access panels:								
		<u>Number</u>	Name/Location	<u>on</u>							
		423	Left Fan Cowl,	Engine 2							
		424		ight Fan Cowl, Engine 2							
		425	Left Thrust Re								
		425CL 426	Right Thrust R	-		linge Fairing, Engine 2					
		420 441AL	•	•	•	ser Disconnect, Strut 2					
		441AT			umbnail Fairing						
		441BL			ft Mid Strut Fair						
		441BR		_	ght Mid Strut Fa	_					
		441CL		•	ft Overwing Fair	•					
		441DL 441DR		-	ft Underwing Fa ght Underwing F	_					
		441EL		_	-	_					
441EL Forward Strut Fairing, Left T.R. Strut Fairing, Strut 2 441ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2 Special Access:											
		Number	Name/Locatio	<u>n</u>							
		S4002	Right Strut For	ward and <i>i</i>	Aft Engine Mour	nt Assemblies					
		NOTE: En	gine removal req	uired.							
	SUBTA	ASK 54-05-03-210-00	6								
	(2)	Do a Gener	al Visual inspect	tion of the	forward and aft	engine mount to strut sh	near pins.				
	SUBTA	ASK 54-05-03-910-00									
	(3)	737–6789 E	Basic Task Descr	ription, AM	M 51–05–01–21	10–809.					
		ASK 54-05-03-410-00									
	(4)		access panels:								
		<u>Number</u>	Name/Locatio								
		423	Left Fan Cowl,	•	,						
		424 425	Right Fan Cow Left Thrust Re	-							
		425CL		7		linge Fairing, Engine 2					
		426	Right Thrust R	•							
		441AL		-		ser Disconnect, Strut 2					
		441AT		-	umbnail Fairing						
		441BL	Forward Strut	Fairing, Le	ft Mid Strut Fair	ing, Strut 2					
		EFFECTIVITY		SOURCE		FORWARD AND AFT ENG	GINE MOUN	ГТО			
		AKS ALL		MRB	STRUT SHEAR	PINS					
					D633A109-AKS 54-020-02-01			Page 2 eb 15			
					I.						



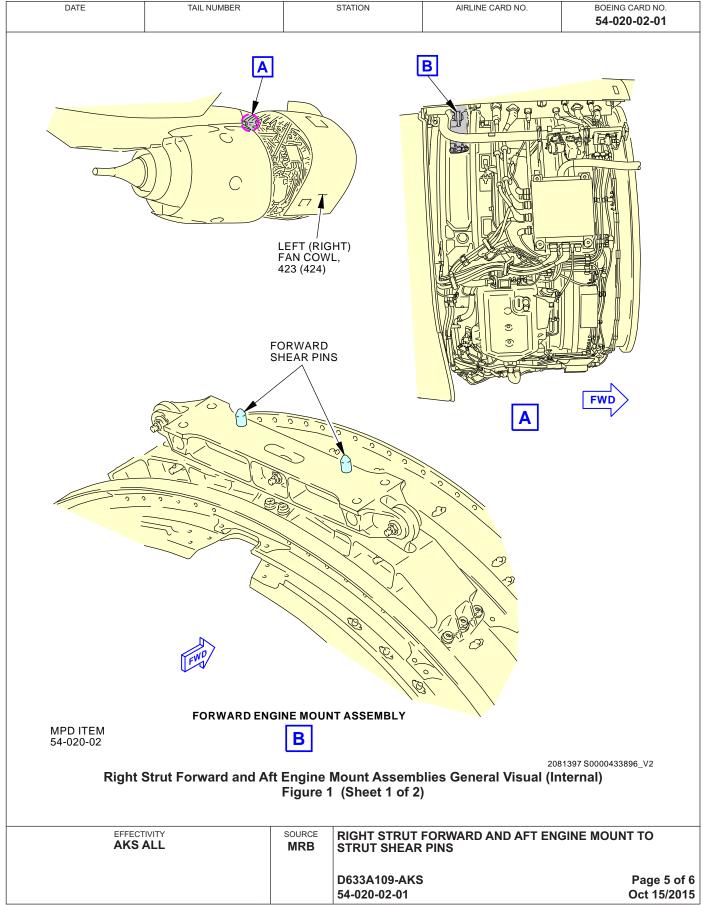


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-020-</b>		
(Con	tinued)	1	I		1	MECH	INSP
Numb 441BF 441CL 441DL 441DF 441EL 441EF	R Forward Strut	airing, Rig airing, Left airing, Left airing, Rig airing, Left	t Overwing Fair t Underwing Fa ht Underwing F t T.R. Strut Fairi	ing, Strut 2 iring, Strut 2 airing, Strut 2 ing, Strut 2			
		END OF 1	TASK ———				
EFFECT AKS	IVITY <b>ALL</b>		RIGHT STRUT F STRUT SHEAR	ORWARD AND AFT END PINS	GINE MOUNT	то	
			D633A109-AKS 54-020-02-01			age 3 b 15/2	

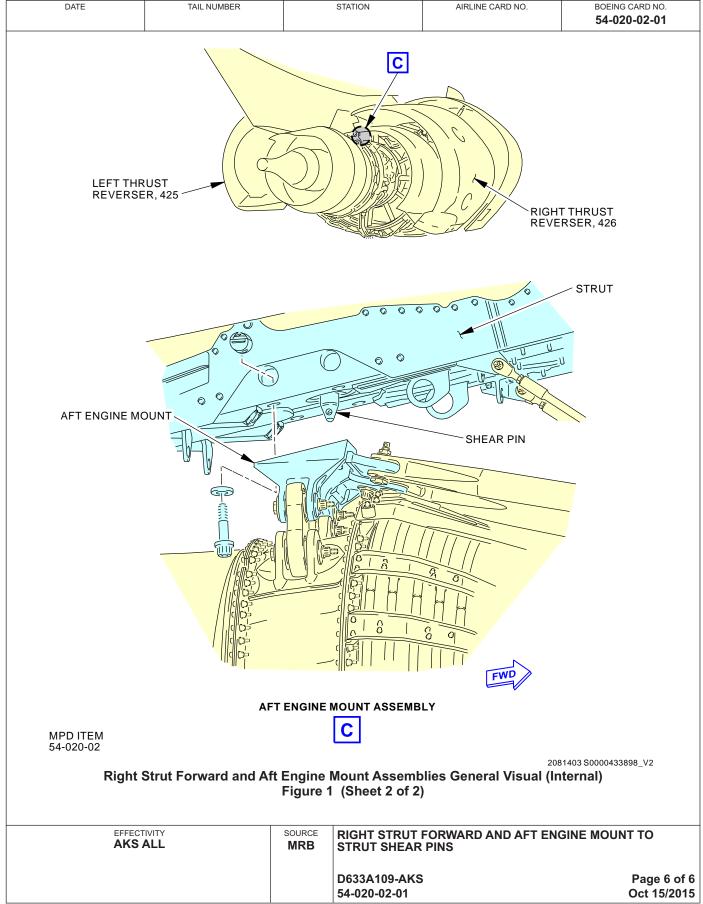


	Γ	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-020		
	TAS	K 51	-05-01	-210-809					MECH	INSP
2.	737-	-6789	Basic	Task Description						
	A.	СРС	P Bas	sic Task						
		SUBTA	ASK 51-05	-01-210-078						
		(1)	CPC	P Basic Task Item 1 is	not applica	ıble.				
				-01-210-079		1				
		(2)		ne CPCP Basic Task Ite			i-b ODOD Di	. T l. l		
		(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.								
		SUBTA	ASK 51-05	-01-210-080						
		(3)	Do th	ne CPCP Basic Task Ite	em 3 as foll	lows:				
		(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.								
		SUBTA		-01-210-081						
		(4)	Do th	ne CPCP Basic Task ite	em 4 as foll	lows:				
		(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.						ANDARD Repair priate.		
		SUBTA	ASK 51-05	-01-210-082						
		(5)	CPC	P Basic Task Item 5 is	not applica	ıble.				
				-01-210-100						
		(6)	Do th	ne CPCP Basic Task Ite	em 6 (Not A	Applicable)				
	suвтаsк 51-05-01-210-084 (7) CPCP Basic Task Item 7 is not applicable.									
					- END OF	TASK ———				
				CTIVITY S ALL	SOURCE MRB	RIGHT STRUT STRUT SHEAR	FORWARD AND AFT ENG PINS	GINE MOUNT	то	
						D633A109-AKS 54-020-02-01	}		Page 4 Oct 15/	













AIRLINE CARD NO		TITLE  LEFT STRUT AFT ENGINE MOUNT ASSEMBLY			BOEING CARD NO. <b>54-030-01-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR	APPLIC.		
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	AIRPLANE ALL	ALL ALL	
		ACCESS 415 416			ZONE 415 416		

Inspect aft engine mount assembly, including thrust links and thrust link pins; mount to engine left, center and right links, including link pins; hanger and evener bar; attach bolts.

INTERVAL NOTE: Whichever comes first.

### A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT AFT ENGINE MOUNT ASSEMBLY	
		D633A109-AKS 54-030-01-01	Page 1 of 4 Oct 15/2014

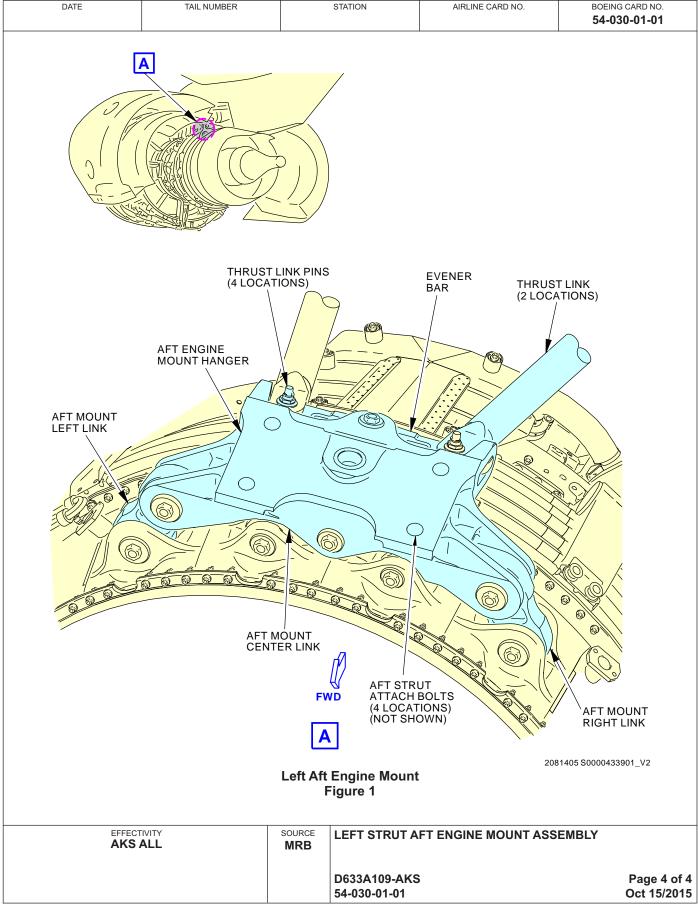


	D	ATE		TAIL NUME	BER		STATION	AIRLINE CARD NO.	BOEING C 54-030		
									54-030		20141
			05-03-21							MECH	INSP
1.				NERAL VISU	AL: LEFT	STRUT	AFT ENGINE I	MOUNT ASSEMBLY			
	(Figu	ure 1)									
	A.	Insp	ection								
		SUBTA	SK 54-05-03-								
		(1)	Open th	nese access							
			Numbe		<u>Location</u>						
			415		rust Reve						
			416	Right I	hrust Rev	erser, E	ngine 1				
			SK 54-05-03-								
		(2)	and thru		mount to			it assembly, including ght links, including lin			
		SUBTA	SK 54-05-03-	910-007							
		(3)	737–67	'89 Basic Tas	k Descrip	tion, AM	M Task 51–05–0	)1–210–809.			
		SUBTA	SK 54-05-03-	410-007							
		(4)	Close tl	hese access	panels:						
			Numbe	<u>Name/</u>	<u>Location</u>						
			415	Left Th	rust Reve	rser, Enç	gine 1				
			416	Right T	hrust Rev	erser, E	ngine 1				
					—— Е	ND OF	TASK ———				
			EFFECTIV AKS A			SOURCE MRB	LEFT STRUT AI	FT ENGINE MOUNT AS	SSEMBLY	1	<u> </u>
							D633A109-AKS			Page 2	of A
							54-030-01-01			eb 15/	



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>0-01-01</b>	
	TAC	V 54	05.04	240.900				04-00	MECH	INSP
2.				-210-809 : Task Description						
	Α.			sic Task -01-210-078						
		(1)		P Basic Task Item 1 is	not applic	able.				
		` '		-01-210-079						
		(2)		e CPCP Basic Task Ite	em 2 as fo	llows:				
			. ,	3. It is not necessary t unless it has deteriora metal. A light uniform t accumulated dirt or de	o remove ted to the ilm of Cor bris, will r nay requi	normal amounts point where moi rosion Inhibiting normally allow ac	accomplish CPCP Basi of sealant/leveling comsture can penetrate down Compound (CIC) that he lequate inspection of the are multiple layers are	npound wn to the nas not e structure		
		SUBT	ASK 51-05	-01-210-080						
		(3)	Do th	e CPCP Basic Task Ite	em 3 as fo	llows:				
			` ,	as specified in each ta visual inspections follo corrosion, such as bul etc. In the task area, o determine if removal is	sk descrip wing parti ging skins heck the i s required	otion. Use Addition ial disassembly in or corrosion rure or tegrity of any series, and any corros	scription. The inspection onal non-destructive ins fithere are indications or uning into splices, or uncealant/leveling compound in inhibiting compound lication is required per Compound in the compound in t	pections or of hidden der fittings, and to l, particularly	y	
		SUBT	ASK 51-05-	-01-210-081						
		(4)	Do th	e CPCP Basic Task ite	m 4 as fo	llows:				
			. ,	structure as required, TREATMENT METHO Manual (SRM) D634A D634A211(-900), D63	including a DS, AMM 200, (-600 4A333 (BE errous met	application of pro SUBJECT 51-0 0), D634A201 (-7 BJ), or related se	air or replace all discreptective finishes per STA 0-58, or 737 Structural I (00), D634A210 (-800), ervice bulletin, as appropted by hormal	ANDARD Repair priate.		
		SUBT		-01-210-082						
		(5)	CPCI	P Basic Task Item 5 is	not applic	able.				
				-01-210-100	O (NI)	A B b b				
		(6)		ie CPCP Basic Task Ite	em 6 (Not	Applicable)				
		SUBT/ (7)		-01-210-084 P Basic Task Item 7 is	not applic	able				
		(1)	OF CI	P Dasic Task Itelii / Is						
					- END OF	TASK ———				
				CTIVITY S ALL	SOURCE MRB	LEFT STRUT A	FT ENGINE MOUNT ASS	EMBLY		1
						D633A109-AKS 54-030-01-01			Page 3 Oct 15/	









#### 737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	RIGHT STRUT	AFT ENGINE MOL	BOEING CARD NO. <b>54-030-02-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR	APPLIC.	
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL ALL
		ACCESS 425 426			ZONE 425 426	
		-				

Inspect aft engine mount assembly, including thrust links and thrust link pins; mount to engine left, center and right links, including link pins; hanger and evener bar; attach bolts.

INTERVAL NOTE: Whichever comes first.

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

L				
	EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT AFT ENGINE MOUNT ASSEMBLY	
			D633A109-AKS 54-030-02-01	Page 1 of 4 Oct 15/2014

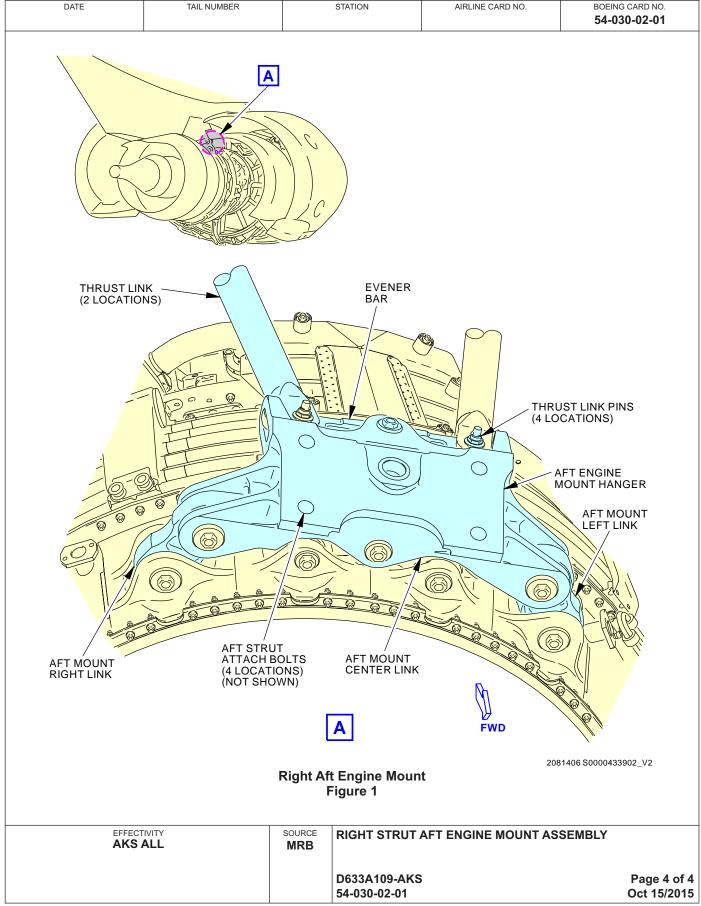


	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-030</b> -		
	TAS	K 54-	·05-03-2	10-808					MECH	INSP
1.	EXT	ERNA	AL - GEI	NERAL VISUAL: RI	GHT STRU	JT AFT ENGINE	MOUNT ASSEMBLY			
	(Fig	ure 1)	)							
	A.	Insp	ection							
			NSK 54-05-03	-010-008						
		(1)	Open tl	hese access panels:						
			Numbe	er <u>Name/Locatio</u>	<u>on</u>					
			425	Left Thrust Re		•				
			426	Right Thrust F	Reverser, E	ngine 2				
		SUBTA	NSK 54-05-03							
		(2)	thrust li				assembly, including thru inks, including link pins			
		SUBTA	NSK 54-05-03	-910-008						
		(3)	737–67	789 Basic Task Desc	ription , AM	/IM Task 51–05–	-01–210–809.			
		SUBTA	NSK 54-05-03	-410-008						
		(4)	Close t	these access panels:						
			Numbe							
			425	Left Thrust Re		-				
			426	Right Thrust R	Reverser, E	ngine 2				
					- END OF	TASK ———				
			AKS A		SOURCE MRB	RIGHT STRUT	AFT ENGINE MOUNT AS	SEMBLY	<u> </u>	<u> </u>
						D633A109-AKS	<b>3</b>		Page 2	



	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>0-02-01</b>	
	TAC	NZ 54	05.04	240,000				34-03	MECH	INSP
2.				-210-809 : Task Description						
۷.				<del>-</del>						
	A.			sic Task						
		(1)		-01-210-078 P Basic Task Item 1 is	not applic	able.				
		( )		-01-210-079	посаррно	a.5.0.				
		(2)		e CPCP Basic Task Ite	m 2 as fo	llows:				
			` ,	3. It is not necessary tunless it has deteriora metal. A light uniform faccumulated dirt or de	o remove ted to the ilm of Cor bris, will r nay requi	normal amounts point where moi rosion Inhibiting normally allow ac	accomplish CPCP Basi of sealant/leveling com sture can penetrate down Compound (CIC) that he lequate inspection of the are multiple layers are	npound vn to the las not e structure		
		SUBT		-01-210-080						
		(3)		e CPCP Basic Task Ite						
			` ,	as specified in each ta visual inspections follo corrosion, such as bull etc. In the task area, of determine if removal is	sk descrip wing parti ging skins heck the i s required	otion. Use Addition ial disassembly in or corrosion rure or tegrity of any series, and any corros	scription. The inspection onal non-destructive ins fithere are indications or uning into splices, or uncealant/leveling compound ion inhibiting compound lication is required per Compour in the compound in t	pections or if hidden der fittings, nd to , particularl	y	
		SUBT	ASK 51-05-	-01-210-081						
		(4)	Do th	e CPCP Basic Task ite	m 4 as fo	llows:				
			` ,	structure as required, TREATMENT METHO Manual (SRM) D634A D634A211(-900), D63-	ncluding a DS, AMM 200, (-600 4A333 (BE rrous met	application of pro SUBJECT 51-0 0), D634A201 (-7 BJ), or related se	air or replace all discreption of the control of th	ANDARD Repair priate.		
		SUBT		-01-210-082						
		(5)		P Basic Task Item 5 is	not applica	able.				
				-01-210-100 	m 6 /Not	Applicable)				
		(6)		e CPCP Basic Task Ite	ili o (Not	Applicable)				
		(7)		-01-210-084 P Basic Task Item 7 is	not applic	able.				
		(.)	0. 0.			TASK ——				
					- END OF	IAGN ———				
				CTIVITY 5 ALL	SOURCE MRB	RIGHT STRUT	AFT ENGINE MOUNT AS	SEMBLY	•	
						D633A109-AKS 54-030-02-01			Page 3 Oct 15/	







#### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	LEFT STR	UT TO WING ATTA	BOEING CARD NO. <b>54-040-01-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD 48 MO	REPEAT <b>48 MO</b>	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	9000 FC	9000 FC	ALL	ALL
		ACCESS 431CL 431CR 43 434AR 434BL	31DL 431DR 431EL	431ER 434AL	ZONE <b>431 434</b>	

Inspect strut to wing upper link, diagonal brace, side links, and strut attachment fittings.

INTERVAL NOTE: Whichever comes first.

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT TO WING ATTACHMENTS	
		D633A109-AKS 54-040-01-01	Page 1 of 4 Jun 15/2015

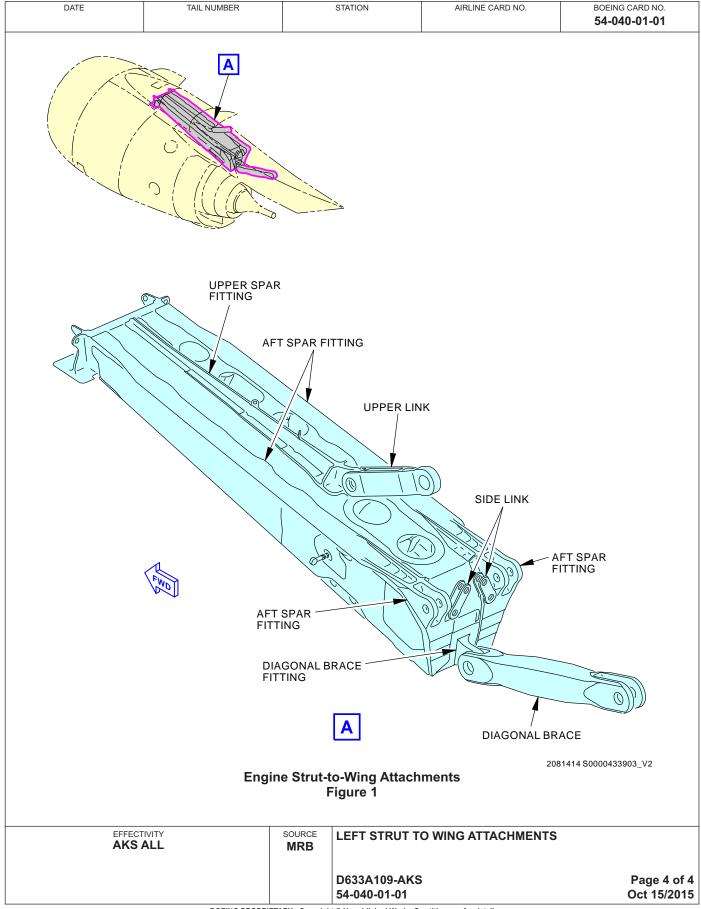


	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-040-</b>		
TASI	K 54	-05-03-210-8	809	'				MECH	INS
			AL VISUAL: LEF	T STRUT	TO WING ATTA	ACHMENTS			
(Figu						<u> </u>			
Α.	-	pection							
		ASK 54-05-03-010-0							
	(1)		e access panels:						
		Number	Name/Locatio		# O				
		431CL 431CR		_	ft Overwing Failight Overwing Fa	_			
		431DL		_	ft Underwing Fa	_			
		431DR		•	ght Underwing F				
		431EL		_	ft T.R. Strut Fai	_			
		431ER		•	ght T.R. Strut Fa	•			
		434AL	Aft Strut Fairing	g, Left For	ward Panel, Str	ut 1			
		434AR		• •	orward Panel, S	trut 1			
		434BL	Aft Strut Fairing	g, Left Aft l	Panel, Strut 1				
	SUBTA	ASK 54-05-03-210-00	09						
	(2)		ral Visual inspect ttachment fittings		strut to wing upp	per link, diagonal brace, si	de links,		
	SUBTA	ASK 54-05-03-910-00	09						
	(3)	737–6789 I	Basic Task Descr	ription, AM	M Task 51–05–0	01–210–809.			
	SUBTA	ASK 54-05-03-410-00	09						
	(4)	Close these	e access panels:						
		<u>Number</u>	Name/Locatio	<u>n</u>					
		431CL	Forward Strut	Fairing, Le	ft Overwing Fai	ring, Strut 1			
		431CR	Forward Strut	Fairing, Ri	ght Overwing Fa	airing, Strut 1			
		431DL		•	ft Underwing Fa				
		431DR		•	ght Underwing F	•			
		431EL			ft T.R. Strut Fair				
		431ER 434AL			ght T.R. Strut Fa ward Panel, Str				
		434AR	· ·	0,	orward Panel, S				
		434BL	Aft Strut Fairing						
					TACK				
				- END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE MRB	LEFT STRUT TO	O WING ATTACHMENTS			



	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>0-01-01</b>	
	TAS	K 51	_05_01	-210-809				34-04	MECH	INSP
2.				Task Description						
	Α.			sic Task						
				i-01-210-078						
		(1)	CPC	P Basic Task Item 1 is	not applica	able.				
		SUBT	ASK 51-05	-01-210-079						
		(2)	Do th	ne CPCP Basic Task Ite	em 2 as fo	llows:				
			(a)	3. It is not necessary t unless it has deteriora metal. A light uniform t accumulated dirt or de	o remove ted to the ilm of Cor bris, will n	normal amounts point where moi rosion Inhibiting normally allow ac	accomplish CPCP Basi of sealant/leveling comsture can penetrate down Compound (CIC) that he lequate inspection of the eare multiple layers are	npound vn to the las not e structure		
		SUBT	ASK 51-05	-01-210-080						
		(3)	Do th	ne CPCP Basic Task Ite	em 3 as fo	llows:				
			(a)	as specified in each ta visual inspections follo corrosion, such as bul etc. In the task area, of determine if removal is	sk descrip wing parti ging skins heck the i s required,	otion. Use Addition ial disassembly in or corrosion run on tegrity of any son, and any corrosion run and any corrosion.	scription. The inspection on all non-destructive ins fithere are indications of aning into splices, or undealant/leveling compound in inhibiting compound lication is required per Compound	pections or if hidden der fittings, nd to , particularly	y	
		SUBT		-01-210-081						
		(4)	Do th	ne CPCP Basic Task ite	m 4 as fo	llows:				
			(a)	structure as required, TREATMENT METHO Manual (SRM) D634A D634A211(-900), D63	including a DS, AMM 200, (-600 4A333 (BE errous met	application of pro SUBJECT 51-0 )), D634A201 (-7 BJ), or related se	air or replace all discrept otective finishes per STA 0-58, or 737 Structural I (00), D634A210 (-800), prvice bulletin, as appro or be handled by normal	ANDARD Repair priate.		
		SUBT		-01-210-082						
		(5)	CPC	P Basic Task Item 5 is	not applica	able.				
				i-01-210-100	C /N ! : 1	A = = !- ! - !				
		(6)		ne CPCP Basic Task Ite	em 6 (Not	Applicable)				
		(7)		-01-210-084 P Basic Task Item 7 is	not applic	able				
		(1)	OI O	I Dasic lask itelli i is						
					- END OF	TASK ———				
				CTIVITY S ALL	SOURCE MRB	LEFT STRUT TO	O WING ATTACHMENTS		1	
						D633A109-AKS 54-040-01-01			Page 3 Oct 15/	









#### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	RIGHT STE	TITLE RUT TO WING ATT	BOEING CARD NO. <b>54-040-02-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD 48 MO	REPEAT <b>48 MO</b>	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	9000 FC	9000 FC	ALL	ALL
		ACCESS 441CL 441CR 44 444AR 444BR	1DL 441DR 441EL	. 441ER 444AL	ZONE 441 444	

Inspect strut to wing upper link, diagonal brace, side links, and strut attachment fittings.

INTERVAL NOTE: Whichever comes first.

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

AKS ALL SOURCE MRB		RIGHT STRUT TO WING ATTACHMENTS	
		D633A109-AKS 54-040-02-01	Page 1 of 4 Jun 15/2015

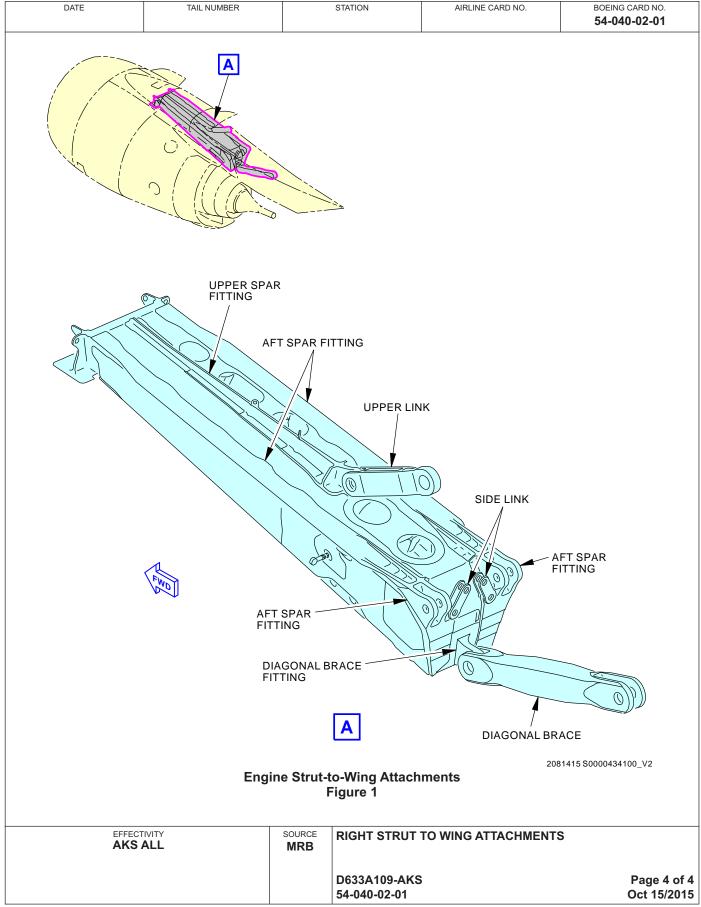


DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-040</b>		
K 54	-05-03-210-8	310					MECH	INS
ERN#	L - GENER	AL VISUAL: RIGI	HT STRU	TO WING AT	TACHMENTS			
(Figure 1)								
·								
SUBTASK 54-05-03-010-010								
441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2								
				-	_			
			•	•	•			
			•	•	•			
			•					
					_			
	444AR	•	• •					
	444BR	-						
SUBT/	ASK 54-05-03-210-0°	10						
(2)	Do a Gene	ral Visual inspecti		strut to wing up	per link, diagonal brace, s	side links,		
		•						
			ntion AMI	M Tack 51 05 0	11 210 900			
` /			ption, Aivii	VI 145K 3 1-03-C	71-210-609.			
(4)		·						
			_					
			-	_	_			
			•	•	•			
			0,	•	0,			
	441EL			-	_			
	441ER		-		•			
	444AL		•	-	•			
	444AR	Aft Strut Fairing	g, Right Fo	rward Panel, S	trut 2			
	444BR	Aft Strut Fairing	g, Right Af	t Panel, Strut 2				
			END OF	TV & K				
			END OF	TASK ———				
			END OF	TASK ———				
			END OF	TASK ———				
			END OF	TASK ———				
			END OF	TASK ———				
			END OF	TASK ———				
			END OF	TASK ———				
	EFFECTIVITY AKS ALL		SOURCE MRB		TO WING ATTACHMENTS			
ŀ	ERNA ure 1) Insp subta (1) subta (2) subta (3)	ERNAL - GENERA  ure 1)  Inspection  SUBTASK 54-05-03-010-07  (1) Open these  Number  441CL  441CR  441DL  441DR  441EL  441ER  444AL  444AR  444BR  SUBTASK 54-05-03-210-07  (2) Do a Gene  and strut at  SUBTASK 54-05-03-910-07  (3) 737-6789 I  SUBTASK 54-05-03-410-07  (4) Close these  Number  441CL  441CR  441DL  441DR  441DR  441EL  441ER	Inspection  SUBTASK 54-05-03-010-010  (1) Open these access panels:  Number Name/Location  441CL Forward Strut F  441DL Forward Strut F  441DR Forward Strut F  441EL Forward Strut F  444AL Aft Strut Fairing  444AR Aft Strut Fairing  444BR Aft Strut Fairing  SUBTASK 54-05-03-210-010  (2) Do a General Visual inspectic and strut attachment fittings.  SUBTASK 54-05-03-910-010  (3) 737-6789 Basic Task Descrit SUBTASK 54-05-03-410-010  (4) Close these access panels:  Number Name/Location  441CL Forward Strut F  441DL Forward Strut F  441DR Forward Strut F  441DR Forward Strut F  441DR Forward Strut F  441EL Forward Strut F  500 F  600	Inspection  SUBTASK 54-05-03-010-010  (1) Open these access panels:  Number Name/Location  441CL Forward Strut Fairing, Lee 441DL Forward Strut Fairing, Lee 441ER Forward Strut Fairing, Lee 441ER Forward Strut Fairing, Right Aft Strut Fairing, Ambustask 54-05-03-210-010  (2) Do a General Visual inspection of the sand strut attachment fittings.  SUBTASK 54-05-03-910-010  (3) 737–6789 Basic Task Description, AMI SUBTASK 54-05-03-410-010  (4) Close these access panels:  Number Name/Location  441CL Forward Strut Fairing, Lee 441DR Forward Strut Fairing, Right Forward Strut Fairing, Right Aft Fairing, Righ	Inspection  SUBTASK 54-05-03-010-010  (1) Open these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fair 441DL Forward Strut Fairing, Left Underwing Fair 441DL Forward Strut Fairing, Right Underwing Fair 441DL Forward Strut Fairing, Right Underwing Fair 441EL Forward Strut Fairing, Left T.R. Strut Fair 441ER Forward Strut Fairing, Right T.R. Strut Fair 444AL Aft Strut Fairing, Left Forward Panel, Strut Fairing, Right Forward Panel, Strut Fairing, Right Forward Panel, Strut Fairing, Right Aft Panel, Strut Fairing, Right Strut Fairing, Right Forward Panel, Strut Fairing Strut Staussk 54-05-03-210-010  (2) Do a General Visual inspection of the strut to wing uppand strut attachment fittings.  SUBTASK 54-05-03-210-010  (3) 737-6789 Basic Task Description, AMM Task 51-05-05-05-05-05-05-05-05-05-05-05-05-05-	Inspection  SUBTASK 54-05-03-010-010  (1) Open these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Left T.R. Strut Fairing, Strut 2  441ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  SUBTASK 54-05-03-210-010  (2) Do a General Visual inspection of the strut to wing upper link, diagonal brace, s and strut attachment fittings.  SUBTASK 54-05-03-910-010  (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.  SUBTASK 54-05-03-410-010  (4) Close these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2  441DR Forward Strut Fairing, Right Underwing Fairing, Strut 2  441DR Forward Strut Fairing, Right Underwing Fairing, Strut 2  441DR Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2  441EL Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2	Inspection Subtask 54-05-03-210-810  Inspection Subtask 54-05-03-010-010  (1) Open these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2  441EL Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2  441EL Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2  444AL Aft Strut Fairing, Right Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  SUBTASK 54-05-03-210-010  (2) Do a General Visual inspection of the strut to wing upper link, diagonal brace, side links, and strut attachment fittings.  SUBTASK 54-05-03-210-010  (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.  SUBTASK 54-05-03-410-010  (4) Close these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Left Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Left T.R. Strut Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2	Inspection  SUBTASK 54-09-03-010-010  (1) Open these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441DL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441DL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2  444ER Forward Strut Fairing, Right Forward Panel, Strut 2  444AAL Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Forward Panel, Strut 2  SUBTASK 54-05-03-210-010  (2) Do a General Visual inspection of the strut to wing upper link, diagonal brace, side links, and strut attachment fittings.  SUBTASK 54-05-03-410-010  (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.  SUBTASK 54-05-03-410-010  (4) Close these access panels:  Number Name/Location  441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2  441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441DL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Left Underwing Fairing, Strut 2  441EL Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2  441ER Forward Strut Fairing, Right Underwing Fairing, Strut 2



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>0-02-01</b>	
	TAS	K 51	-05-01-	210-809					MECH	INSF
2.	737	6789	Basic	Task Description						
	A.	CPC	P Bas	ic Task						
		SUBT	ASK 51-05-0	01-210-078						
		(1)	CPCF	P Basic Task Item 1 is r	not applica	ble.				
		SUBT	ASK 51-05-0							
		(2) Do the CPCP Basic Task Item 2 as follows:								
	(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.									
		SUBT	ASK 51-05-0	01-210-080						
		(3)	Do the	e CPCP Basic Task Ite	m 3 as foll	lows:				
			(	Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.						
		SUBT	ASK 51-05-0	01-210-081						
		(4)	Do the	e CPCP Basic Task ite	m 4 as foll	ows:				
			; -     	Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.						
		SUBT	ASK 51-05-0							
		(5)	CPCF	P Basic Task Item 5 is r	not applica	ble.				
		SUBT	ASK 51-05-0							
		(6)	Do the	e CPCP Basic Task Ite	m 6 (Not A	Applicable)				
			ивтаsк 51-05-01-210-084 7) СРСР Basic Task Item 7 is not applicable.							
		——— END OF TASK ———								
			EFFEC <b>AKS</b>		SOURCE MRB	RIGHT STRUT	TO WING ATTACHMENTS	5		
						D633A109-AKS 54-040-02-01			Page 3 Oct 15/	









#### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	LEFT STRUT TO WING ATTACHMENTS - PINS AND			BOEING CARD NO. <b>54-050-01-01</b>		
DATE	TASK DETAILED		FUSE PINS	RELATED CARD			
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD 48 MO	REPEAT 48 MO	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	9000 FC	9000 FC	ALL	ALL	
		ACCESS 431CL 431CR 43 434AR 434BL	1DL 431DR 431EL	ZONE 431 434			

 $Inspect\ pins\ and\ fuse\ pins\ on\ upper\ link,\ midspar,\ diagonal\ brace,\ and\ side\ links.\ Pin\ removal\ is\ not\ required.$ 

INTERVAL NOTE: Whichever comes first.

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT TO WING ATTACHMENTS - PINS AN	ND FUSE PINS
		D633A109-AKS 54-050-01-01	Page 1 of 4 Jun 15/2015

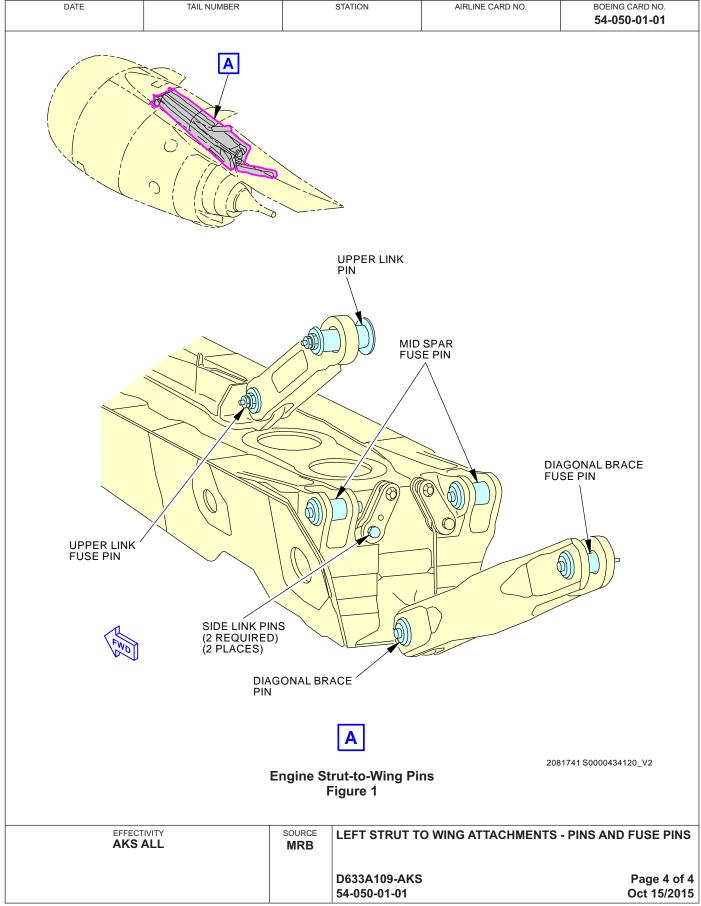


	01-01							
	MECH II							
FUSE PINS								
INTERNAL - DETAILED: LEFT STRUT TO WING ATTACHMENTS - PINS AND FUSE PINS (Figure 1)								
A. Inspection								
SUBTASK 54-05-03-010-015								
(1) Open these access panels:  Number Name/Location								
431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1								
nar diagonal brace								
jai, diagonal brace,								
	1 1							
MENTS - PINS AND F	USE PIN							
	par, diagonal brace,							



	1	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO.	
	TAC	N 54	05.04	240.900				04 000	месн	INSP
2.				-210-809 : Task Description						
	A.			sic Task						
	A.			-01-210-078						
		(1)		P Basic Task Item 1 is	not applic	able.				
		SUBT		-01-210-079						
		(2)	Do th	ne CPCP Basic Task Ite	m 2 as fo	llows:				
	(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.									
		SUBT		-01-210-080						
		(3)		ne CPCP Basic Task Ite						
	(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.						1			
		SUBT	ASK 51-05	-01-210-081						
		(4)	Do th	ne CPCP Basic Task ite	m 4 as fo	llows:				
	(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.									
		SUBT		-01-210-082						
		(5)	CPC	P Basic Task Item 5 is	not applic	able.				
				-01-210-100	m 6 /Nat	Applicable)				
		(6)		ne CPCP Basic Task Ite	em 6 (NOL	Applicable)				
		(7)		-01-210-084 P Basic Task Item 7 is	not applic	able.				
		(.)	0. 0.			TASK ——				
					- END OF	Mark ———				
				CTIVITY S ALL	SOURCE MRB	LEFT STRUT T	O WING ATTACHMENTS	- PINS AND	FUSE F	PINS
						D633A109-AKS 54-050-01-01	;		Page 3 Oct 15/	









#### 737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	RIGHT STRUT TO WING ATTACHMENTS - PINS			BOEING CARD NO. <b>54-050-02-01</b>		
DATE	TASK DETAILED		AND FUSE PINS	RELATED CARD			
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD 48 MO	REPEAT <b>48 MO</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	9000 FC	9000 FC	ALL	ALL	
		ACCESS 441CL 441CR 44 444AR 444BR	1DL 441DR 441EL	ZONE 441 444			

 $Inspect\ pins\ and\ fuse\ pins\ on\ upper\ link,\ midspar,\ diagonal\ brace,\ and\ side\ links.\ Pin\ removal\ is\ not\ required.$ 

INTERVAL NOTE: Whichever comes first.

Reference	Title			
AMM 51-00-58	STANDARD TREATMENT METHODS			

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT TO WING ATTACHMENTS - PINS A PINS	AND FUSE
		D633A109-AKS 54-050-02-01	Page 1 of 4 Jun 15/2015



### 737-600/700/800/900 **TASK CARDS**

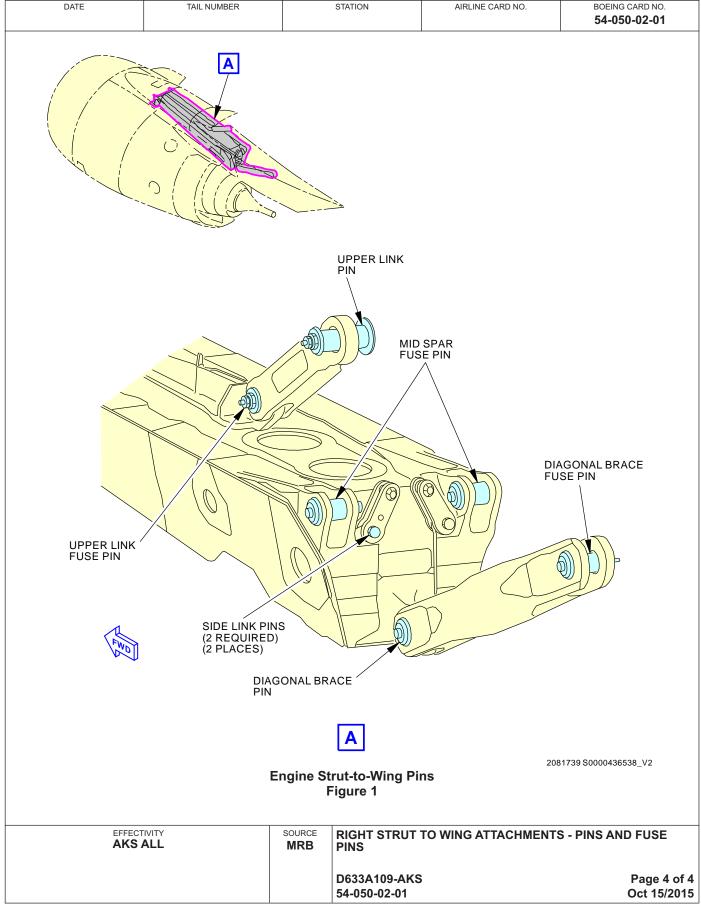
					IAS	ok CARDS				
	DA	TE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-050		
	TASK	54-	05-03-211-8	302					MECH	INS
1.					JT TO WII	NG ATTACHME	NTS - PINS AND FUS	E PINS		
	(Figu	e 1)								
	Α.	Insp	ection							
		-	SK 54-05-03-010-0	16						
		(1)	Open these	e access panels:						
			<u>Number</u>	Name/Locatio	<u>n</u>					
		441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2								
			441CR		_	-	_			
			441DL 441DR		•	eft Underwing Fa	•			
441EL Forward Strut Fairing, Left T.R. Strut Fairing, Strut 2 441ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2						-	_			
			444AL			ward Panel, Str				
444AR Aft Strut Fairing, Right Forward Panel, Strut 2 444BR Aft Strut Fairing, Right Aft Panel, Strut 2										
			444BR	Aft Strut Fairing	g, Right At	t Panel, Strut 2				
	SUBTASK 54-05-03-211-002									
	(2) Do a Detailed inspection of the pins and fuse pins on upper link, midspar, diagonal brace, and side links. Pin removal is not required.									
	SUBTASK 54-05-03-910-012									
(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–809.  SUBTASK 54-05-03-410-016										
		(4)	Number	e access panels: Name/Locatio	n					
			441CL		_	eft Overwing Fai	ring Strut 2			
			441CR		_	ght Overwing F	•			
			441DL		_	eft Underwing Fa	_			
			441DR			ght Underwing				
			441EL		_	eft T.R. Strut Fai	•			
			441ER 444AL		_	ght T.R. Strut Fa ward Panel, Str	_			
			444AR	· · · · · · · · · · · · · · · · · · ·	•	orward Panel, S				
			444BR	· · · · · · · · · · · · · · · · · · ·	• •	ft Panel, Strut 2				
					- FND OF	TASK ———				
					LIND OF	IAON				
			EFFECTIVITY AKS ALL		SOURCE MRB	RIGHT STRUT PINS	TO WING ATTACHMENT	S - PINS AND	FUSE	
						D633A109-AKS	3		Page 2 eb 15/	
						1				



### 737-600/700/800/900 **TASK CARDS**

					iAO	K CANDS				
	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0	CARD NO. 0-02-01	
_				1-210-809					MECH	INS
2.	131	-6/89	Basi	c Task Description						
	A.	CP	CP Ba	isic Task						
				5-01-210-078		la La				
		(1)		CP Basic Task Item 1 is	not applica	ible.				
		(2)		<sub>'<sup>5-01-210-079</sup></sub> :he CPCP Basic Task Ite	em 2 as foll	lows.				
		(-)	(a)				accomplish CPCP Basic	Task Item		
		3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.								
		SUBT	ASK 51-0	5-01-210-080						
	(3) Do the CPCP Basic Task Item 3 as follows:									
		(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.								
	SUBTASK 51-05-01-210-081									
		(4)	Do t	he CPCP Basic Task ite	em 4 as foll	ows:				
	(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existir maintenance practices.									
		SUBT	ASK 51-0	5-01-210-082						
		(5)	CPC	CP Basic Task Item 5 is	not applica	ble.				
				5-01-210-100	0.01					
		(6)		the CPCP Basic Task Ite	em 6 (Not A	Applicable)				
	suвтаsк 51-05-01-210-084 (7) CPCP Basic Task Item 7 is not applicable.									
		(7)	CFC	or basic task item / is						
					– END OF	TASK ———				
				ECTIVITY S ALL	SOURCE MRB	RIGHT STRUT	TO WING ATTACHMENTS	6 - PINS AND	FUSE	
						D633A109-AKS 54-050-02-01	1		Page 3 Oct 15/	
_										









#### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	LEFT STR	TITLE LUT TO WING ATTA	BOEING CARD NO. <b>54-060-01-01</b>		
DATE	TASK DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD  10 YR	REPEAT 10 YR	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	36000 FC	36000 FC	ALL	ALL
		431CL 431CR 43 434AR 434BL NOTE	31DL 431DR 431EL	ZONE 431 434		

Inspect the bores of pins and fuse pins on upper link, midspar, diagonal brace, and side links. Pin removal is not required.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: TR's must be open to remove access panels 431EL and 431ER.

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT TO WING ATTACHMENTS	
		D633A109-AKS 54-060-01-01	Page 1 of 4 Oct 15/2014

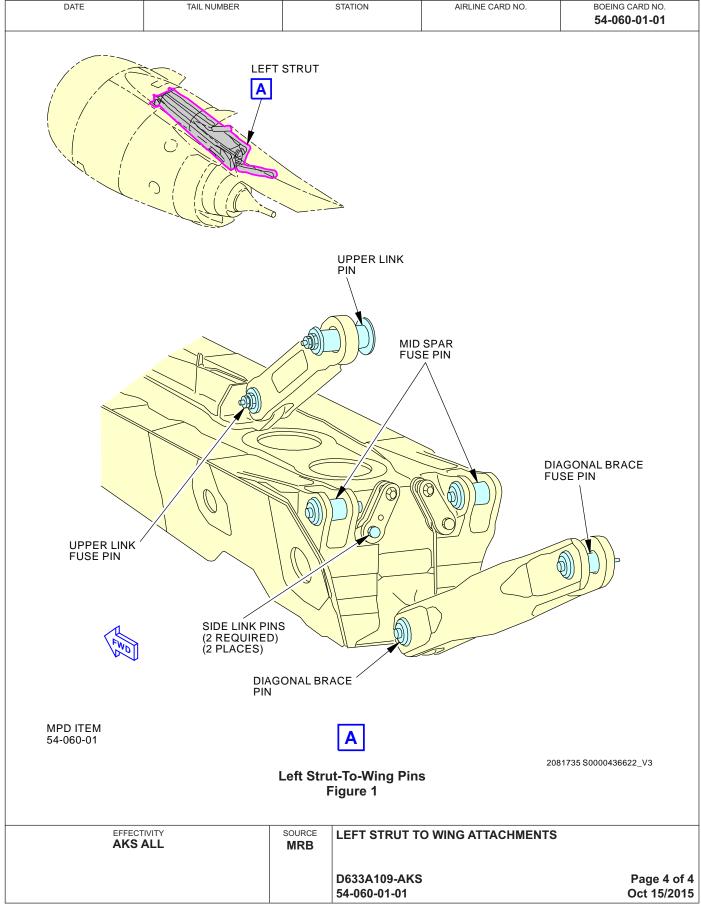


	-	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-060</b> -		
	TAS	SK 54	-05-03-211-8	03		,	<u>'</u>		MECH	INS
1.	INT	ERNA	L - DETAIL	ED: LEFT STRUT	TO WING	G ATTACHMEN	TS			
	(Fig	ure 1)	)							
	A.	Insp	ection							
			ASK 54-05-03-010-0	17						
		(1)	Open these	e access panels:						
			<u>Number</u>	Name/Location	<u>1</u>					
			431CL	Forward Strut F	_	-	_			
			431CR	Forward Strut F	_	-	_			
			431DL	Forward Strut F	_	-	_			
			431DR 431EL	Forward Strut F Forward Strut F	_	-	_			
			431ER	Forward Strut F	_		_			
			434AL		_	_	_			
	434AL Aft Strut Fairing, Left Forward Panel, Strut 1 434AR Aft Strut Fairing, Right Forward Panel, Strut 1									
			434BL	Aft Strut Fairing	, Left Aft	Panel, Strut 1				
			NOTE: TR	's must be open to	remove	access panels 4	431EL and 431ER.			
		SUBTA	ASK 54-05-03-211-0	13						
	<ul> <li>(2) Do a Detailed inspection of the bores of pins and fuse pins on upper link, midspar, diagonal brace, and side links. Pin removal is not required.</li> </ul>									
		SUBTA	ASK 54-05-03-910-0	13						
		(3)	737–6789	Basic Task Descrip	otion, AM	M Task 51–05–0	)1–210–809.			
		SUBTA	ASK 54-05-03-410-0	17						
		(4)	Close these	e access panels:						
			<u>Number</u>	Name/Location	<u>1</u>					
			431CL	Forward Strut F	-	-	_			
			431CR	Forward Strut F	_	-	_			
			431DL 431DR	Forward Strut F Forward Strut F	-	-	_			
			431EL	Forward Strut F	•	•	•			
			431ER	Forward Strut F	-		_			
			434AL	Aft Strut Fairing						
			434AR	Aft Strut Fairing	, Right Fo	orward Panel, St	trut 1			
			434BL	Aft Strut Fairing	, Left Aft I	Panel, Strut 1				
					END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE MRB	LEFT STRUT TO	O WING ATTACHMENTS			
						D633A109-AKS 54-060-01-01			Page 2 eb 15/	
						i .				



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO.			
	TAS	K 51	-05-01-	210-809					MECH	INSP		
2.	737	-6789	Basic	Task Description								
	A.	CPC	CP Bas	ic Task								
		SUBT	ASK 51-05-									
		(1)	CPCF	P Basic Task Item 1 is r	not applica	ıble.						
			ASK 51-05-	<sub>01-210-079</sub> e CPCP Basic Task Ite	m 2 ac fol	lowe:						
		(2)	(a)	3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.								
			ASK 51-05-			_						
		<ul><li>(3) Do the CPCP Basic Task Item 3 as follows:</li><li>(a) Visually inspect all structure listed in the task description. The inspection method is</li></ul>										
				Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.								
			ASK 51-05-									
		(4)		e CPCP Basic Task ite								
		(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.										
			ASK 51-05-									
		(5)		P Basic Task Item 5 is r	not applica	IDIE.						
		(6)	48K 51-05-1 Do th	<sub>01-210-100</sub> e CPCP Basic Task Ite	m 6 (Not A	Applicable)						
		` '	ASK 51-05-		(	<b>TP</b> [ ]						
		(7)	CPCF	P Basic Task Item 7 is r	not applica	ıble.						
					- END OF	TASK ———						
		EFFECTIVITY AKS ALL SOURCE MRB LEFT STRUT TO WING ATTACHMENTS						,	•			
						D633A109-AKS 54-060-01-01			Page 3 Oct 15/			









#### 737-600/700/800/900 TASK CARDS

AIRLINI	E CARD NO	RIGHT ST	TITLE RUT TO WING ATT	BOEING CARD NO. <b>54-060-02-01</b>		
DATE	TASK DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD  10 YR	REPEAT 10 YR	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	36000 FC	36000 FC	ALL	ALL
		ACCESS 441CL 441CR 44 444AR 444BR NOTE	11DL 441DR 441EL	ZONE 441 444		

Inspect the bores of pins and fuse pins on upper link, midspar, diagonal brace, and side links. Pin removal is not required.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: TR's must be open to remove access panels 441EL and 441ER.

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT TO WING ATTACHMENTS	
		D633A109-AKS 54-060-02-01	Page 1 of 4 Oct 15/2014

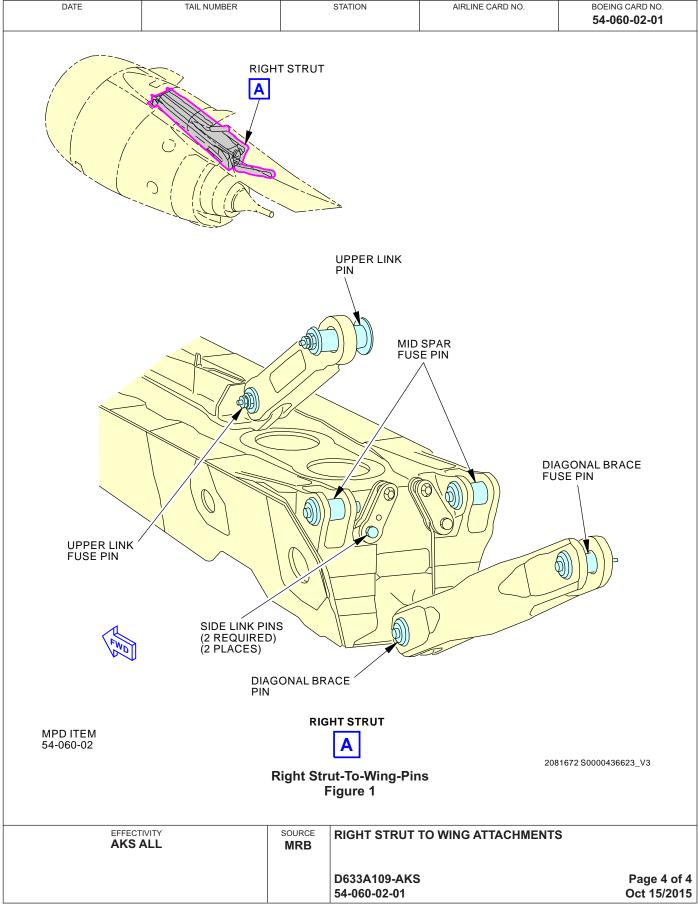


	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ <b>54-060-</b>					
TAS	K 54	-05-03-211-8	04					MECH	INS			
INT	ERNA	L - DETAILI	ED: RIGHT STRU	JT TO WII	NG ATTACHME	NTS						
	ure 1											
Α.	lnor	nation										
A.		pection										
	(1)	ASK 54-05-03-010-0 <sup>-</sup>	access panels:									
	(1)	Number	Name/Locatio	n								
		441CL			ft Overwing Fair	ring Strut 2						
		441CR		•	ght Overwing Fa							
		441DL		_	ft Underwing Fa	_						
		441DR	Forward Strut I	Fairing, Ri	ght Underwing F	Fairing, Strut 2						
		441EL		•	ft T.R. Strut Fair							
		441ER		orward Strut Fairing, Right T.R. Strut Fairing, Strut 2  If Strut Fairing, Left Forward Panel, Strut 2								
		444AL 444AR	· ·	•	ward Panel, Stri orward Panel, St							
		444AR 444BR			t Panel, Strut 2	iiui Z						
				-		141FL and 441FR						
	NOTE: TR's must be open to remove access panels 441EL and 441ER.											
	subtask 54-05-03-211-004  (2) Do a Detailed Inspection of the bores of pins and fuse pins on upper link, midspar,											
	diagonal brace, and side links. Pin removal is not required.											
	SUBTASK 54-05-03-910-014 (3) 737_6789 Basic Task Description AMM Task 51_05_01_210_809											
(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–809.												
		ASK 54-05-03-410-01										
	(4)		e access panels:									
		<u>Number</u>	Name/Locatio									
		441CL 441CR			ft Overwing Fair ght Overwing Fa							
		441DL		_	eft Underwing Fa	_						
		441DR		•	ght Underwing F							
		441EL		_	ft T.R. Strut Fair	_						
		441ER	Forward Strut I	Fairing, Ri	ght T.R. Strut Fa	airing, Strut 2						
		444AL	,	0.	ward Panel, Stru							
		444AR	· · · · · · · · · · · · · · · · · · ·		orward Panel, St	trut 2						
		444BR	Aft Strut Fairing	g, Right Af	t Panel, Strut 2							
				- END OF	TASK ———							
		EFFECTIVITY <b>AKS ALL</b>		SOURCE MRB	RIGHT STRUT	TO WING ATTACHMENTS						
					D633A109-AKS			Page 2				
					54-060-02-01		F	eb 15/	<b>Z</b> U			



DATE				TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-060-02-01</b>		
	TAS	K 51	-05-01-2	210-809					MECH	INSP
2.	737	-6789	Basic	Task Description						
	A.	СРС	P Basi	ic Task						
		SUBTA	ASK 51-05-0							
		(1)	CPCF	P Basic Task Item 1 is no	ot applica	ble.				
			NSK 51-05-0	o1-210-079 e CPCP Basic Task Item	2 oo foll	0)4/0;				
		(2)					accomplish CDCD Basi	n Tack Itam		
			) 1 3	Prior to inspection clean 3. It is not necessary to unless it has deteriorate metal. A light uniform filr accumulated dirt or debout without removal. CIC made accumulations of dirt or	remove r d to the p n of Corr ris, will no ay require	normal amounts point where moi osion Inhibiting ormally allow ac	of sealant/leveling com sture can penetrate dov Compound (CIC) that h lequate inspection of the	pound vn to the as not e structure		
		SUBTA	ASK 51-05-0							
		(3)		e CPCP Basic Task Item						
	(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.									
		SUBTA	ASK 51-05-0							
		(4)		e CPCP Basic Task item						
			? ] ] }	Remove all corrosion, estructure as required, incorrection as required, incorrection and the second and the sec	cluding a S, AMM 3 00, (-600) A333 (BB	pplication of pro SUBJECT 51-0 ), D634A201 (-7 J), or related se	otective finishes per STA 0-58, or 737 Structural I (00), D634A210 (-800), prvice bulletin, as appro	ANDARD Repair oriate.		
			ASK 51-05-0							
		(5)		P Basic Task Item 5 is no	or applica	DIE.				
		(6)	0-05 Do th	<sub>01-210-100</sub> e CPCP Basic Task Item	n 6 (Not A	applicable)				
		. ,	ASK 51-05-0		`	,				
		(7)	CPCF	Basic Task Item 7 is no	ot applica	ble.				
					END OF	TASK ———				
			EFFEC <sup>*</sup>		SOURCE MRB	RIGHT STRUT	TO WING ATTACHMENTS	3		l
						D633A109-AKS			Page 3	







#### 737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		LEFT STRUT BOX			BOEING CARD NO. <b>54-070-01-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD  9 YR	REPEAT 9 YR	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL	
			413 414 415 416 431AT 431BL 431BR 431CL 431CR 431EL 431ER S4331				

Inspect external areas of strut box, including upper and lower spars, forward engine mount bulkhead, aft bulkhead, and side skins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove MID and AFT insulation blanket/heat shields. Remove fan cowls. TR's must be open to

remove access panels 431EL and 431ER. Engine removal not required.

Reference	Title	
AMM 51-00-58	STANDARD TREATMENT METHODS	

EFFECTIVITY <b>AKS ALL</b>	SOURCE MRB	LEFT STRUT BOX	
		D633A109-AKS	Page 1 of 7
		54-070-01-01	Jun 15/2015



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 <b>54-070</b>		
	TAS	K 54-	-05-03-210-	811					MECH	INS
1.	INTI	ERNA	L - GENER	AL VISUAL: EXT	ERNAL -	LEFT STRUT B	OX			
			Figure 2)							
	A.	Insp	ection							
		-	NSK 54-05-03-010-0	011						
		(1)	Open thes	se access panels:						
			<u>Number</u>	Name/Locatio	<u>n</u>					
			413	Left Fan Cowl,	Engine 1					
			414	Right Fan Cow	l, Engine 1	1				
			415	Left Thrust Rev	verser, Eng	gine 1				
			416	Right Thrust R		-				
			431AT		•	umbnail Fairing				
			431BL		-	ft Mid Strut Fair	_			
			431BR			ght Mid Strut Fa	_			
			431CL		•	ft Overwing Fair	•			
			431CR		•	ght Overwing Fa	•			
			431EL 431ER		-	ft T.R. Strut Fair	_			
					raining, Rig	ght T.R. Strut Fa	airing, Strut 1			
			Special Ac							
			<u>Number</u>	Name/Locatio						
			S4431	Right Strut Box	External	Inspection				
			m				nields. Remove fan cow L and 431ER. Engine re			
		SUBTA	NSK 54-05-03-210-0	011						
		(2)		•			of strut box, including up nead, and side skins.	per and		
		SUBTA	NSK 54-05-03-910-0	015						
		(3)	737–6789	Basic Task Descr	iption, AM	M Task 51-050	)1-210–803.			
		SUBTA	ASK 54-05-03-410-0	011						
		(4)	Close thes	se access panels:						
			<u>Number</u>	Name/Locatio	<u>n</u>					
			413	Left Fan Cowl,	Engine 1					
			414	Right Fan Cow	-	1				
			415	Left Thrust Rev	verser, Eng	gine 1				
			416	Right Thrust R	everser, E	ngine 1				
			431AT		-	umbnail Fairing				
			431BL		•	ft Mid Strut Fair	•			
			431BR			ght Mid Strut Fa	_			
			431CL		-	ft Overwing Fair	_			
			431CR 431EL			ght Overwing Fa ft T.R. Strut Fair				
			EFFECTIVITY	1 Orward Orlat I	SOURCE	LEFT STRUT B				
			AKS ALL		MRB	LEI I OIROI B	<b>-</b> /-			
						D633A109-AKS 54-070-01-01			Page 2 eb 15/	
						3-7 0.0-01-01			5.5 15/	



DATE	TAIL NUMBER	S	TATION	AIRLINE CARD NO.	BOEING C. <b>54-070</b> -		
(Co	ontinued)				-	MECH	INS
Num		on					
431E			t T.R. Strut Fair	ing, Strut 1			
				3,			
		- END OF IA	\SK				
EFFE <b>AK</b> S	CTIVITY S ALL	SOURCE L	EFT STRUT BOX	x			
EFFE <b>AK</b> \$	CTIVITY S ALL	MRB	EFT STRUT BOX	X		Page 3	of



#### 737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				54-070-01-01

TASK 51-05-01-210-803

MECH INSP

#### 2. 737-6789 Basic Task Description

#### A. CPCP Basic Task

SUBTASK 51-05-01-210-036

- (1) Do the CPCP Basic Task Item 1 as follows:
  - (a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

#### SUBTASK 51-05-01-210-037

- (2) Do the CPCP Basic Task Item 2 as follows:
  - (a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

#### SUBTASK 51-05-01-210-038

- (3) Do the CPCP Basic Task Item 3 as follows:
  - (a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

#### SUBTASK 51-05-01-210-039

- (4) Do the CPCP Basic Task item 4 as follows:
  - (a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

#### SUBTASK 51-05-01-210-040

(5) CPCP Basic Task Item 5 is not applicable.

#### SUBTASK 51-05-01-210-110

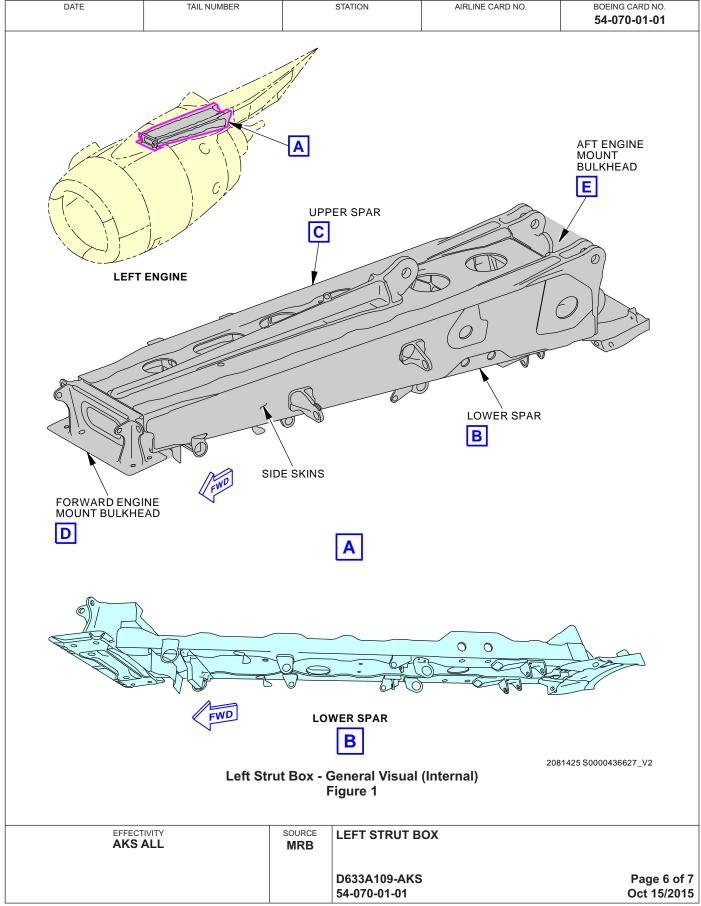
(6) Do the CPCP Basic Task Item 6 (Not Applicable)

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT BOX	
		D633A109-AKS 54-070-01-01	Page 4 of 7 Oct 15/2015

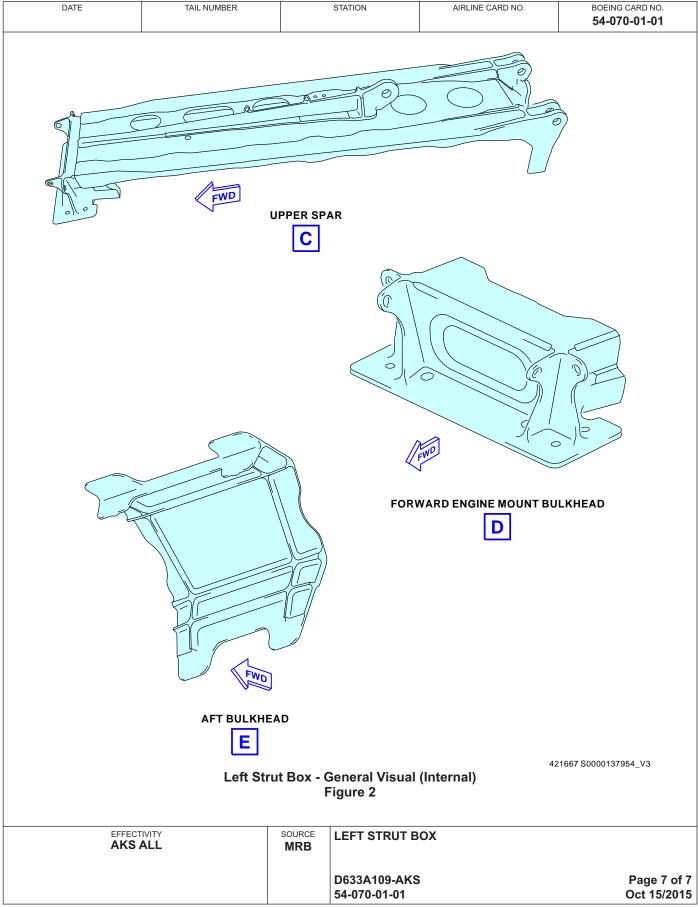


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ <b>54-070-</b>		
SUBTASK 51-05-0	1-210-042					MECH	INSP
(7) CPCP	Basic Task Item 7 is no	t applica	ble.				
	——-	END OF	TASK ———				
	,						
AKS A	ALL	SOURCE MRB	LEFT STRUT BO	ΟX			
			D633A109-AKS		F	Page 5	of 7











### 737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		RIGHT STRUT BOX			BOEING CARD NO. <b>54-070-02-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	1.1 9 YR	REPEAT 9 YR	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL <b>AIRPL</b>	1.2 NOTE	18000 FC	18000 FC	ALL	ALL	
		ACCESS 423 424 425 426 441EL 441ER S4 NOTE	441AT 441BL 441I 1431	3R 441CL 441CR	ZONE <b>443</b>		

Inspect external areas of strut box, including upper and lower spars, forward engine mount bulkhead, aft bulkhead, and side skins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove MID and AFT insulation blanket/heat shields. Remove fan cowls. TR's must be open to

remove access panels 441EL and 441ER. Engine removal not required.

#### A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT BOX	
		D633A109-AKS 54-070-02-01	Page 1 of 7 Jun 15/2015



	L	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-070	CARD NO. 0-02-01		
	TAS	K 54	-05-03-210-	812					MECH	INS	
1.				RAL VISUAL: EXT	ERNAL -	RIGHT STRUT	вох				
•			Figure 2)	<u> </u>							
	Α.	Iner	ection								
	71.	_	ASK 54-05-03-010-	012							
		(1)		se access panels:							
		( )	Number	Name/Location	n						
			423	Left Fan Cowl,							
			424	Right Fan Cow	•	2					
			425	Left Thrust Rev	-						
			426	Right Thrust Re	everser, E	ngine 2					
			441AT	Forward Strut F	airing, Th	umbnail Fairing	, Strut 2				
			441BL		_	ft Mid Strut Fair	_				
			441BR		_	ght Mid Strut Fa	_				
			441CL		_	ft Overwing Fair	_				
			441CR		•	ght Overwing Fa	•				
			441EL 441ER		•	ft T.R. Strut Fair ght T.R. Strut Fa					
					airing, Ki	giit i.R. Stiut Fa	airing, Strut Z				
			Special Ad		_						
			Number	Name/Location	_						
			S4431	Right Strut Box		•					
			m				hields. Remove fan cow L and 441ER. Engine re				
		SUBTA	NSK 54-05-03-210-	012							
		(2)		-			of strut box, including up nead, and side skins.	oper and			
		SUBTA	ASK 54-05-03-910-	016							
		(3)	737–6789	Basic Task Descri	ption, AM	M Task 51–05–0	01–210–803.				
		SUBTA	ASK 54-05-03-410-	012							
		(4)	Close thes	se access panels:							
			<u>Number</u>	Name/Location	<u>n</u>						
			423	Left Fan Cowl,	Engine 2						
			424	Right Fan Cow	l, Engine 2	2					
			425	Left Thrust Rev	erser, Enç	gine 2					
			426	Right Thrust Re		•					
			441AT		•	umbnail Fairing					
			441BL		0,	ft Mid Strut Fair	0,				
			441BR 441CL		•	ght Mid Strut Fa	•				
			441CL 441CR	G							
			441EL		_	ft T.R. Strut Fair	_				
			EFFECTIVITY AKS ALL		SOURCE MRB	RIGHT STRUT	вох				
						D633A109-AKS 54-070-02-01	:		Page 2 eb 15/		



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-070</b>		
	(Continue	d)	1			'	MECH	INSF
	Number	Name/Location	n					
	441ER			ght T.R. Strut Fa	iring, Strut 2			
					3,			
			END OF	TASK ———				
	EFFECTIVITY AKS ALL		SOURCE MRB	RIGHT STRUT E	вох			
	EFFECTIVITY AKS ALL		SOURCE MRB	RIGHT STRUT E			Page 3	of



### 737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				54-070-02-01

TASK 51-05-01-210-803

MECH INSP

#### 2. 737-6789 Basic Task Description

#### A. CPCP Basic Task

SUBTASK 51-05-01-210-036

- (1) Do the CPCP Basic Task Item 1 as follows:
  - (a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

#### SUBTASK 51-05-01-210-037

- (2) Do the CPCP Basic Task Item 2 as follows:
  - (a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

#### SUBTASK 51-05-01-210-038

- (3) Do the CPCP Basic Task Item 3 as follows:
  - (a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

#### SUBTASK 51-05-01-210-039

- (4) Do the CPCP Basic Task item 4 as follows:
  - (a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

#### SUBTASK 51-05-01-210-040

(5) CPCP Basic Task Item 5 is not applicable.

#### SUBTASK 51-05-01-210-110

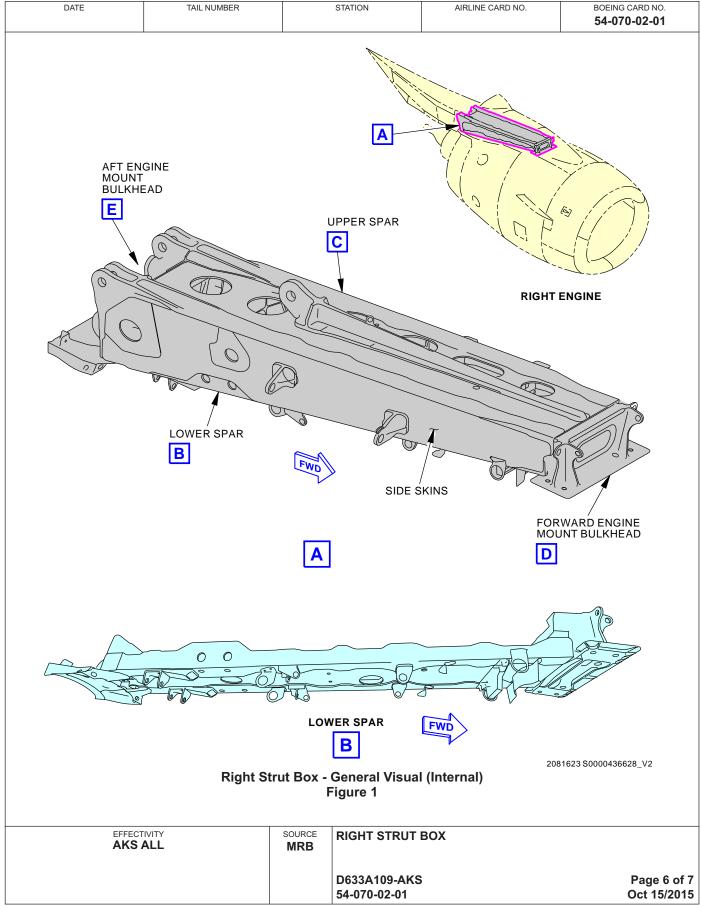
(6) Do the CPCP Basic Task Item 6 (Not Applicable)

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT BOX		
		D633A109-AKS 54-070-02-01	Page 4 of 7 Oct 15/2015	

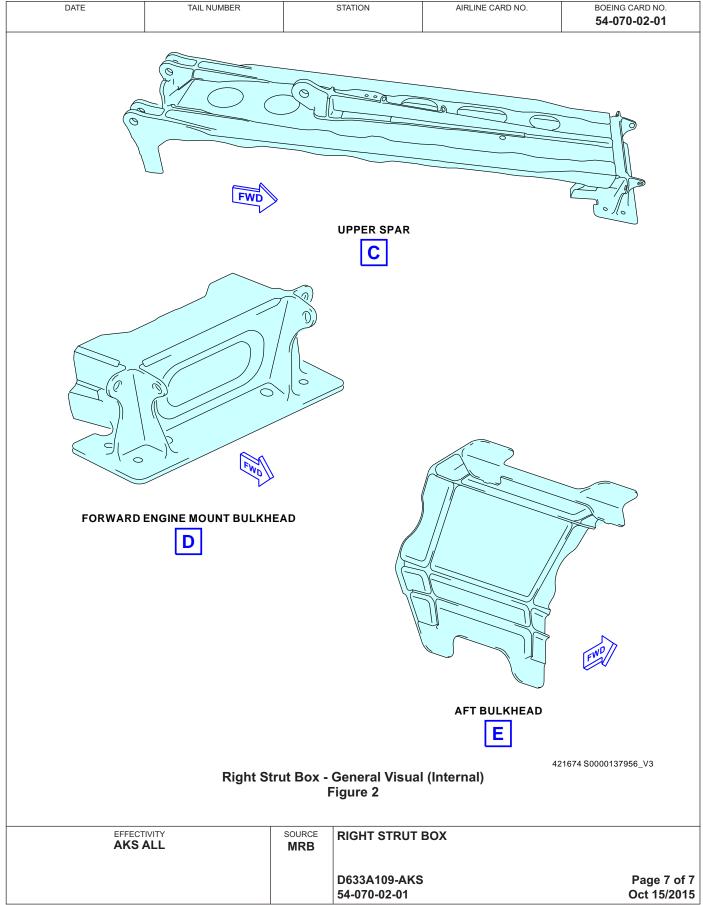


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-070</b>		
SUBTASK 51-05-01	1-210-042	'	'			MECH	INSP
(7) CPCP	Basic Task Item 7 is n	ot applica	ble.				
		END OF	TASK ———				
FFFFCT	IVITY	SOURCE	RIGHT STRUT B	ΟΥ			
EFFECT <b>AKS</b> A	ALL	MRB	MOIII SIRUI D				
			D633A109-AKS		ı	Page 5	of 7
			54-070-02-01		Ċ	Page 5 Oct 15/2	2014











### 737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		LEFT STRUT BOX			BOEING CARD NO. <b>54-080-01-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA  LEFT STRUT	VERSION 1.1	THRESHOLD 9 YR	REPEAT  9 YR	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL	
			31CL 431CR 431DL 33AR 433AT 433BT		ZONE <b>433</b>		

Inspect internal areas of strut box, including upper and lower spars, forward and aft engine mount bulkheads, aft and mid bulkheads, and side skins.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Disassemble pneumatic ducts as required. TR's must be open to remove access panels 431EL

and 431ER.

#### A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT BOX	
		D633A109-AKS 54-080-01-01	Page 1 of 7 Jun 15/2015



							54-080-01-0°	1
	SK 54	-05-03-210-8	113				MECH	H INSI
			AL VISUAL: INTE	RNAL - L	EFT STRUT BO	ox		
(Fig	gure 1)		12 1100/12: 11112			<u> </u>		
A.	-	ection						
		ASK 54-05-03-010-01						
	(1)		e access panels:					
		Number 404DL	Name/Location	_	6 M. J. O E	01 . 1.4		
		431BL 431BR		-	ft Mid Strut Fairi	_		
		431CL			ght Mid Strut Fa ft Overwing Fair			
		431CR		_	ght Overwing Fa	_		
		431DL		-	ft Underwing Fa	_		
		431DR		•	ght Underwing F			
		431EL	Forward Strut F	airing, Le	ft T.R. Strut Fair	ing, Strut 1		
		431ER		-	ght T.R. Strut Fa	iring, Strut 1		
		433AL	Strut, Left Aft D					
		433AR 433AT	Strut, Right Aft Strut, Forward					
		433AT 433BT	Strut, Forward	•				
		433CT	Strut, Upper Sp					
		433DT	Strut, Upper Sp					
		Special Acc	cess:					
		Number	Name/Location	า				
		S4332	Left Strut Box I	_	spection			
			sassemble pneum nels 431EL and 43		as required. TR	d's must be open to remo	ve access	
	SUBTA	ASK 54-05-03-210-01	13					
	(2)					strut box, including upper aft and mid bulkheads, a		
	SUBTA	ASK 54-05-03-910-01	17					
	(3)	737–6789 E	Basic Task Descri	ption, AM	M Task 51–05–0	)1–210–803.		
	SUBTA	ASK 54-05-03-410-01	13					
	(4)	Close these	e access panels:					
		<u>Number</u>	Name/Location	<u>1</u>				
		431BL	Forward Strut F	- airing, Le	ft Mid Strut Fairi	ing, Strut 1		
		431BR	Forward Strut F	airing, Ri	ght Mid Strut Fa	iring, Strut 1		
		431CL		•	ft Overwing Fair			
		431CR		_	ght Overwing Fa	_		
		431DL		-	ft Underwing Fa	_		
		431DR 431EL			ght Underwing F ft T.R. Strut Fair			
		EFFECTIVITY  AKS ALL		SOURCE MRB	LEFT STRUT B			
		- <del></del>			D633A109-AKS 54-080-01-01		Page Feb 15	



(Continued)  Number Name/Location  431ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 1  433AL Strut, Left Att Dry Bay, Strut 1  433AS Strut, Forward Spar Web, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1  ——END OF TASK——	DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO <b>54-080-01-0</b>	
Number 431ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 1 433AL Strut, Left Aft Dry Bay, Strut 1 433AR Strut, Right Aft Dry Bay, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1	(Contir	ued)			MECH	INSP
431ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 1 433AL Strut, Left Aft Dry Bay, Strut 1 433AR Strut, Right Aft Dry Bay, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1			1			
433AL Strut, Left Aft Dry Bay, Strut 1 433AR Strut, Right Aft Dry Bay, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1				airing, Strut 1		
433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1	I .	Strut, Left Aft Dr	y Bay, Strut 1	_		
433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1	I .					
433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1	I .	Strut, Forward S	Spar Web, Strut 1			
433DT Strut, Upper Spar Web, Strut 1						
	I .					
END OF TASK	433D1					
			END OF TASK ———			
EFFECTIVITY SOURCE MRB LEFT STRUT BOX	EFFECTIVI AKS AI	TY		вох		
	Alto AL	-				
D633A109-AKS Page 3 of			D633A109-AKS	5	Page	3 of 7
54-080-01-01 Feb 15/20					Feb 15	5/2015



### 737-600/700/800/900 TASK CARDS

	DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CA			
	TASK 51-05-01-210-803							
2.	737-6789 Basic	Task Description						

#### A. CPCP Basic Task

SUBTASK 51-05-01-210-036

- (1) Do the CPCP Basic Task Item 1 as follows:
  - (a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

#### SUBTASK 51-05-01-210-037

- (2) Do the CPCP Basic Task Item 2 as follows:
  - (a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

#### SUBTASK 51-05-01-210-038

- (3) Do the CPCP Basic Task Item 3 as follows:
  - (a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

#### SUBTASK 51-05-01-210-039

- (4) Do the CPCP Basic Task item 4 as follows:
  - (a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

#### SUBTASK 51-05-01-210-040

(5) CPCP Basic Task Item 5 is not applicable.

#### SUBTASK 51-05-01-210-110

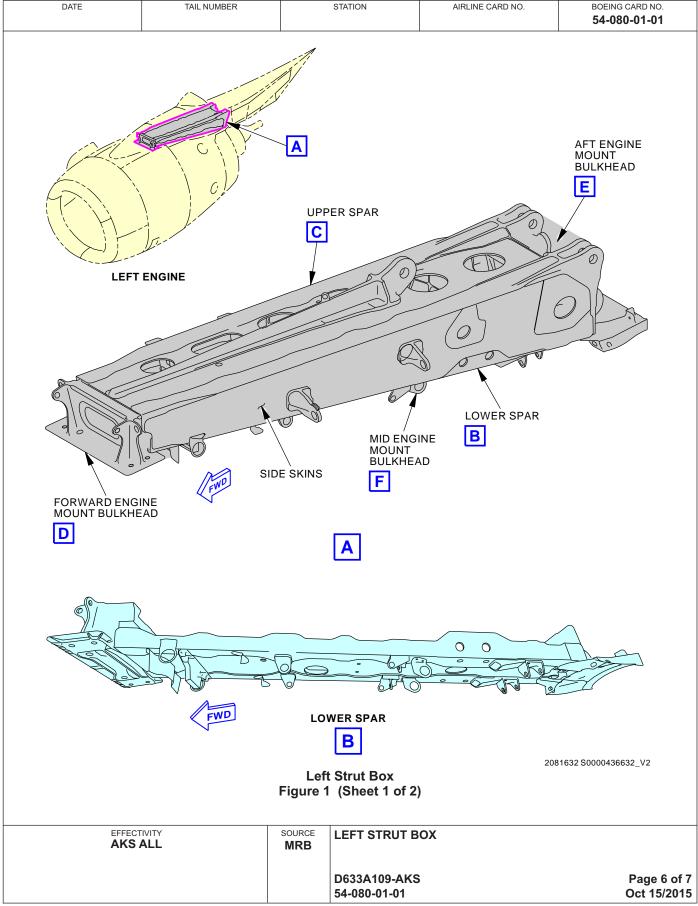
(6) Do the CPCP Basic Task Item 6 (Not Applicable)

EFFECTIVITY AKS ALL	SOURCE MRB	LEFT STRUT BOX		
		D633A109-AKS 54-080-01-01	Page 4 of 7 Oct 15/2015	

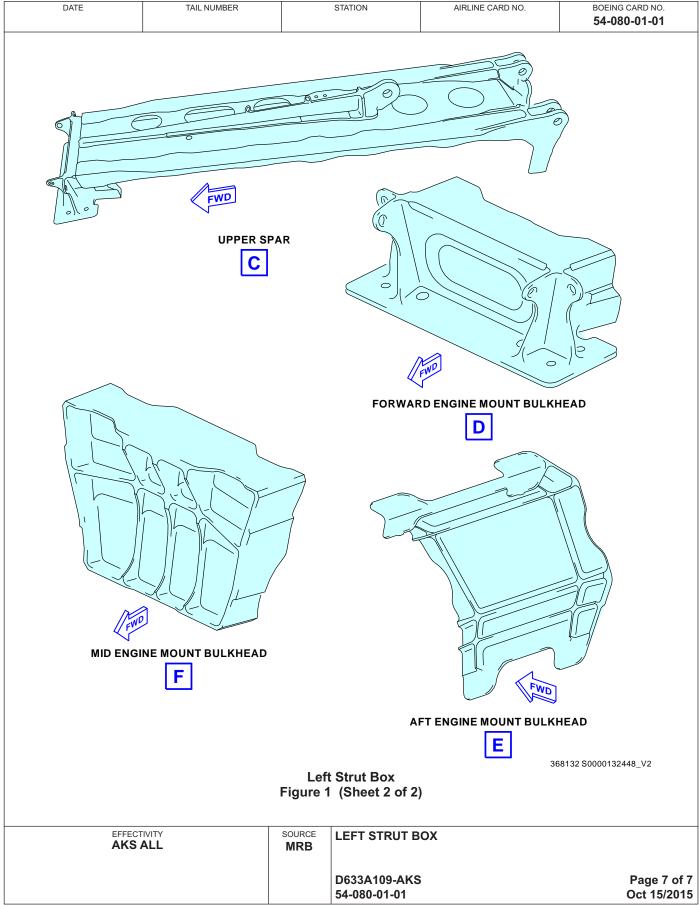


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-080</b> -		
SUBTASK 51-05-01	1-210-042	'	'			MECH	INSP
	Basic Task Item 7 is n	ot applica	ble.				
		END OF	TASK ———				
EFFECT <b>AKS</b>	IVITY	SOURCE	LEFT STRUT BO	X			
ANS A	766	MRB					
			D633A109-AKS 54-080-01-01		C	Page 5 oct 15/2	of 7 2014











### 737-600/700/800/900 TASK CARDS

AIRLINE	E CARD NO	TITLE RIGHT STRUT BOX			BOEING CARD NO. <b>54-080-02-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD  9 YR	REPEAT  9 YR	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	ALL	ALL	
			1CL 441CR 441DL 3AR 443AT 443BT		ZONE <b>443</b>		

Inspect internal areas of strut box, including upper and lower spars, forward and aft engine mount bulkheads, aft and mid bulkheads, and side skins.

**INTERVAL NOTE:** Whichever comes first.

ACCESS NOTE: Disassemble pneumatic ducts as required. TR's must be open to remove access panels 441EL

and 441ER.

#### A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT BOX	
		D633A109-AKS 54-080-02-01	Page 1 of 7 Jun 15/2015



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING ( <b>54-080</b>		
	TAS	K 54	-05-03-210-8	314		l.		1	MECH	INSP
1.	INTI	ERNA	L - GENER	AL VISUAL: INT	ERNAL - F	RIGHT STRUT E	BOX			
	(Fig	ure 1,	Figure 2)							
	Α.	Insp	ection							
		SUBTA	ASK 54-05-03-010-01	14						
		(1)	Open these	e access panels:						
			<u>Number</u>	Name/Location	<u>on</u>					
			441BL		_	ft Mid Strut Fairi	•			
			441BR		-	ght Mid Strut Fai	•			
	441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2									
			441CR			-	_			
			441DL 441DR		_	ft Underwing Fa	_			
	441DR Forward Strut Fairing, Right Underwing Fairing, Strut 2 441EL Forward Strut Fairing, Left T.R. Strut Fairing, Strut 2									
	441ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 2									
	443AL Strut, Left Aft Dry Bay, Strut 2									
	443AR Strut, Right Aft Dry Bay, Strut 2									
			443AT	Strut, Forward	•					
		443BT Strut, Forward Spar Web, Strut 2								
			443CT	, 11 ,						
			443DT Strut, Upper Spar Web, Strut 2							
			Special Acc		_					
	Number Name/Location									
			S4432	Right Strut Box		•				
				sassemble pneun nels 441EL and 4		as required. TR	's must be open to rem	iove access		
			ASK 54-05-03-210-01							
		(2)					strut box, including up aft and mid bulkheads	•		
		SUBTA	ASK 54-05-03-910-01	18						
		(3)	737–6789 E	Basic Task Descr	ription, AM	M Task 51–05–0	1–210–803.			
		SUBTA	ASK 54-05-03-410-01	14						
		(4)	Close these	e access panels:						
			Number	Name/Location	<u>n</u>					
			441BL	Forward Strut	Fairing, Le	ft Mid Strut Fairi	ng, Strut 2			
			441BR		•	ght Mid Strut Fai	•			
	441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2									
	441DL Forward Strut Fairing, Left Underwing Fairing, Strut 2									
	441DR Forward Strut Fairing, Right Underwing Fairing, Strut 2 441EL Forward Strut Fairing, Left T.R. Strut Fairing, Strut 2									
			EFFECTIVITY AKS ALL		SOURCE MRB	RIGHT STRUT E	вох			
						D633A109-AKS			Page 2	2 of 7
						54-080-02-01			eb 15/	



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-080		
	(Continue	d)		,		<u> </u>	MECH	INSF
1	Number	Name/Location	1					
	141ER	Forward Strut F		nht TR Strut Fa	iring Strut 2			
	143AL	Strut, Left Aft Dr			iiiiig, Strut Z			
	143AR	Strut, Right Aft [						
	143AT	Strut, Forward S						
	143BT	Strut, Forward S						
	143CT	Strut, Upper Spa						
	143DT	Strut, Upper Spa						
-	14301							
			END OF	TASK ——				
	EFFECTIVITY <b>AKS ALL</b>		SOURCE MRB	RIGHT STRUT	вох			
				D633A109-AKS		_	Page 3	of
				54-080-02-01		F	eb 15/	201



### 737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				54-080-02-01

TASK 51-05-01-210-803

MECH INSP

#### 2. 737-6789 Basic Task Description

#### A. CPCP Basic Task

SUBTASK 51-05-01-210-036

- (1) Do the CPCP Basic Task Item 1 as follows:
  - (a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

#### SUBTASK 51-05-01-210-037

- (2) Do the CPCP Basic Task Item 2 as follows:
  - (a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

#### SUBTASK 51-05-01-210-038

- (3) Do the CPCP Basic Task Item 3 as follows:
  - (a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

#### SUBTASK 51-05-01-210-039

- (4) Do the CPCP Basic Task item 4 as follows:
  - (a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

#### SUBTASK 51-05-01-210-040

(5) CPCP Basic Task Item 5 is not applicable.

#### SUBTASK 51-05-01-210-110

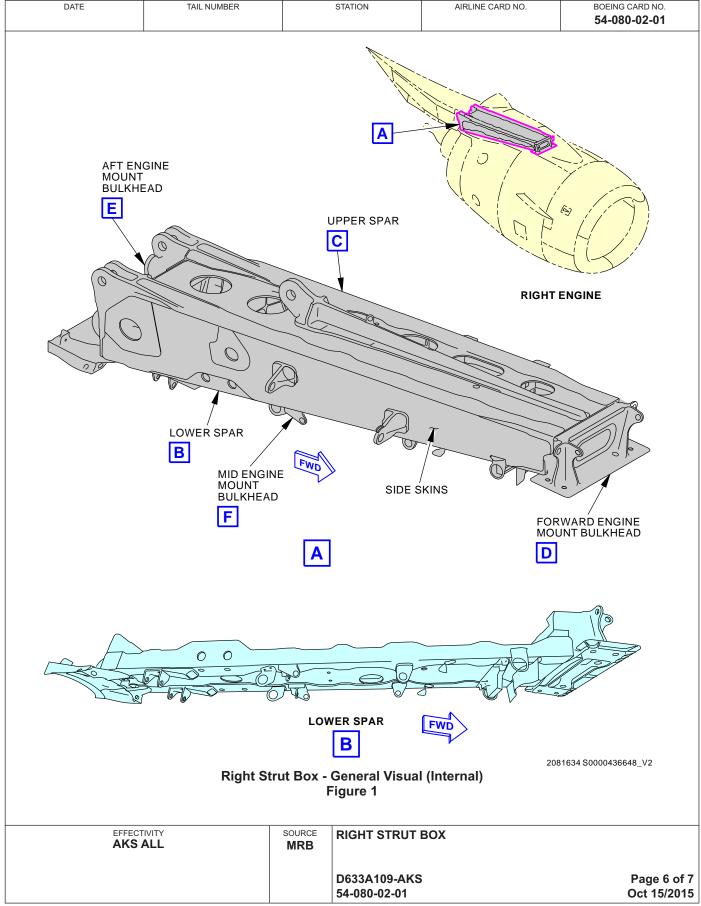
(6) Do the CPCP Basic Task Item 6 (Not Applicable)

EFFECTIVITY AKS ALL	SOURCE MRB	RIGHT STRUT BOX	
		D633A109-AKS 54-080-02-01	Page 4 of 7 Oct 15/2015

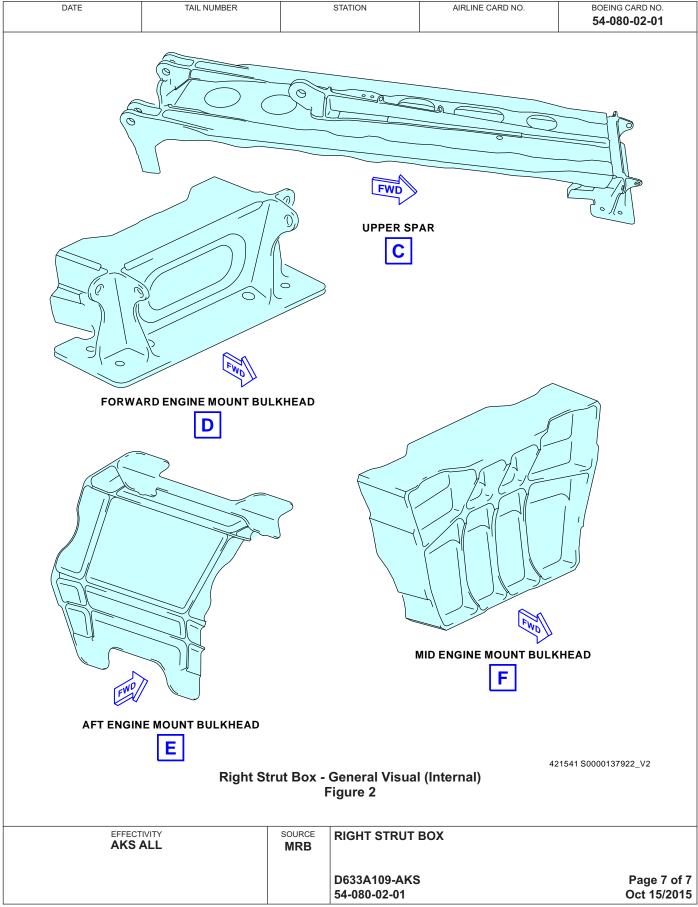


DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-080		
SUBTASK 51-05-01	I-210-042	•	'			MECH	INSP
	Basic Task Item 7 is n	ot applica	ble.				
		END OF	TASK ———				
EFFECT <b>AKS</b> A	IVITY	SOURCE MRB	RIGHT STRUT B	ОХ			
			<b>Decet</b>			_	
			D633A109-AKS 54-080-02-01		C	Page 5 Oct 15/2	of 7 2014













### 737-600/700/800/900 TASK CARDS

AIRLINE	E CARD NO	TITLE FUNCTIONAL CHECK LEFT ENGINE FORWARD STRUT AND AFT STRUT FAIRING DRAINS			BOEING CARD NO. <b>54-090-01-01</b>	
DATE	TASK FUNCTIONAL	STRUT AND	AFT STRUT FAIR	ING DRAINS	RELATE	D CARD
TAIL NUMBER	WORK AREA LEFT STRUT	VERSION 1.1	THRESHOLD  30 MO	REPEAT 30 MO	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	9000 FC	9000 FC	ALL	ALL
		ACCESS 431BL 431BR 43	1CL 431CR 434AR	ZONE 410 430		

Functionally check the left engine forward strut and aft strut fairing drains.

INTERVAL NOTE: Whichever occurs first.

#### A. References

Reference	Title
AMM 08-21-02-580-801	Make the Airplane Level (P/B 201)
AMM 24-22-00-860-812	Remove Electrical Power (P/B 201)
AMM 54-51-01-040-801	Prepare the Strut for Maintenance Operations (P/B 201)
AMM 54-51-01-440-801	Put the Strut Back to its Usual Condition (P/B 201)
AMM 54-52-06-010-801	Aft Fairing Access Panel Removal (P/B 401)
AMM 54-52-06-410-801	Aft Fairing Access Panel Installation (P/B 401)

#### B. Consumable Materials

Reference	Description	Specification
G50316	Cloth - Clean, Dry, Lint-free, White, Cotton	

#### C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1155	Funnel - Long Neck
STD-1280	Source - Air, Regulated, Dry Filtered, 0-30 PSIG
STD-13465	Bucket - 2.5 Gallon Capacity
STD-13870	.75 inch O.D., clear vinyl hose
STD-3910	Container - Plastic
STD-77	Air Source - Regulated, Dry Filtered, 0-50 psig

EFFECTIVITY AKS ALL	SOURCE MRB	FUNCTIONAL CHECK LEFT ENGINE FORW AFT STRUT FAIRING DRAINS	ARD STRUT AND
		D633A109-AKS 54-090-01-01	Page 1 of 5 Jun 15/2015



## 737-600/700/800/900 TASK CARDS

					iAO	K CAKDS			
	D	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO <b>54-090-01-0</b> 1	
T.	ASI	K 54-	55-01-720-	801	'			MECH	INS
l. <u>S</u>	Strut	t Dra	in - Functio	onal Test					
P	Δ.	Gen	eral						
		(1)	This task (	gives steps to do a	a functiona	I test of the str	ut drain.		
E	В.	Prep	oare for the	Functional Test					
		SUBTA	ASK 54-55-01-010-						
		(1)		sk: Prepare the St 51-01-040-801.	rut for Mai	ntenance Oper	rations, AMM		
			SK 54-55-01-040-			. D	Charles and Anna		
		(2)		sk to remove elect 22-00-860-812.	rical powe	r:Remove Elec	ctrical Power, AMM		
			SK 54-55-01-010-						
		(3)	•	se access panels:					
			Number	Name/Locatio		tt M:4 Ctm.t Ca	inima. Otuvit 1		
			431BL 431BR	Forward Strut   Forward Strut	-		_		
			431CL	Forward Strut	_	-	_		
			431CR	Forward Strut	-	_	_		
			434AR Aft Strut Fairing, Right Forward Panel, Strut 1						
			434BL	Aft Strut Fairing	•				
			441BL	Forward Strut	•		•		
		441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2							
			441CR	Forward Strut	_	-	_		
			444AL	Aft Strut Fairing	_	-	_		
			444BR	Aft Strut Fairing	g, Right Af	t Panel, Strut 2	2		
		SUBTA	SK 54-55-01-210-	007					
		(4)		that the internal solutions that the training the training that the training training that the training training the training tra		the strut and d	rain inlet is free of materia	ıl that can	
			SK 54-55-01-840-						
		(5)	Put a buck	ket, STD-13465 or	equivalen	t container bel	ow the aft drain.		
			SK 54-55-01-480-						
		(6)	Put a .75 i STD-1346		TD-13870	on the drain tu	be to point the water into	the bucket,	
C	C.	Fun	ctional Tes	t					
		SUBTA	SK 54-55-01-200-	001					
		(1)		g neck funnel, STI ig the aft end of th			128 fluid ounces)(3.8 liter at spar.	s) of clean	
			NOTE: Po	our water as close	to the dra	in as possible.			
			(a) Make	e sure that the wa	ter flows fr	eely from the a	aft drain.		
			EFFECTIVITY AKS ALL		source <b>MRB</b>		CHECK LEFT ENGINE FOR AIRING DRAINS	RWARD STRUT A	ND
						D633A109-AK	S	Page	

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С	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ <b>54-090-</b>		
	SUBT	ASK 54-55-01-200-0	02		·			MECH	INS
D.	(2) subtr (3) subtr (4) subtr (5) subtr (6)	from the dr ASK 54-55-01-720-0 Use a 0-30 all remainin ASK 54-55-01-210-0 Make sure ASK 54-55-01-910-0 Remove re ASK 54-55-01-910-0 Use a 0-50 internal sui	utes, make sure the rain.  psig dry filtered recommended by that there are no lecommended by the structure of the structure.  psig dry filtered recommended by the structure.  psig dry filtered recommended by the structure.	gulated a strut dra eaks at the n the stru- gulated a airing, Le airing, Right Fo Left Aft airing, Le airing, Righiring, Le airing, Righiring, Le airing, Righiring,	air source, STD- in, into the buck the strut drain hou at with a cotton of air source, STD- tion  off Mid Strut Fairi ght Mid Strut Fairi ght Overwing Fair ght Overwing Fair privard Panel, St Panel, Strut 1 off Mid Strut Fairi ght Mid Strut Fairi ght Mid Strut Fairi ght Mid Strut Fairi ght Overwing Fair ght Overwing Fair ght Overwing Fair	se to strut drain line fittelloth, G50316.  77 or (0-345 kPa), to one of the fittelloth, G50316.  The firming, Strut 1 the firming, Strut 1 the firming, Strut 1 the firming, Strut 2 the firming t	inlet to blow ting.	MECH	IN
		444BR	Aft Strut Fairing,	Right Af	t Panel, Strut 2				
				END OF	TASK ——				
		EFFECTIVITY AKS ALL		SOURCE MRB	FUNCTIONAL C AFT STRUT FAI D633A109-AKS	HECK LEFT ENGINE F		UT AN	_



### 737-600/700/800/900 TASK CARDS

	DATE TAIL NUMBER			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARI <b>54-090-0</b> 1		
2.			55-02-100-8 g Sump Dra	302 ain Inspection				M	1ECH	INS
	A.	Gen	eral							
		(1)	Do this tas	k for the aft fairin	g sump dra	ain when there	is a possible blockage.			
	B.	Prep	are to Insp	ect the Aft Fairi	ng Sump l	Drain				
			SK 54-55-02-040-0							
		(1)		•	ane Level,	AMM TASK 08	8-21-02-580-801.			
		SUBTASK 54-55-02-040-002								
		(2) Do this task: Prepare the Strut for Maintenance Operations, AMM TASK 54-51-01-040-801.								
			SK 54-55-02-010-0							
		(3)	•	pplicable aft fairi	•	panels:				
			Number	K 54-52-06-010-8 Name/Location						
			434AR	Aft Strut Fairin		orward Panel	Strut 1			
			434BL	Aft Strut Fairin			ottut 1			
			444AL	Aft Strut Fairin	•					
			444BR	Aft Strut Fairin	g, Right Af	t Panel, Strut 2	2			
	C.		_	n Inspection						
			SK 54-55-02-200-0	ո ontainer, STD-39	110 bolow t	ho oft fairing d	Irain			
		(1)	•			_	fluid ounces)(7.5 liters).			
						,	iring does not have unwant	ed		
			mate		ornar oarra	oc or the art la	ining does not have anwant			
			SK 54-55-02-160-0							
		(2)	along the fo	orward end to the	aft end of	the internal flo	d ounce (0.03 liter) of clean oor of the aft fairing.	water		
				ur water on all in			n 244 fluid ounges (7.2 lites	o) io		
			` '	ted into the conta			n 244 fluid ounces (7.2 liteı ı drain.	5) 15		
			` '	lly make sure that aft fairing.	at there are	no leaks thro	ugh the bulb seal at the for	ward end		
	<ul> <li>(c) Use a syringe to make sure that no single puddle of water is larger than 1 fluid ounce (0.03 liter).</li> </ul>				fluid					
		SUBTA	SK 54-55-02-160-0							
		(3)	Remove re	maining water fro	om the aft f	fairing.				
		(4)			-	air source, STI	D-77 or (0-345 kPa), to dry	the		
			AKS ALL		SOURCE MRB		CHECK LEFT ENGINE FOR AIRING DRAINS	WARD STRU	TAN	ID
					1	1	-	_	_	

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	AIE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-090	)-01-01	
D.	Put	the Airpl	lane Back to Its Usu	ıal Condi	tion			MECH	INS
	SUBTA	SK 54-55-02-1	160-004						
<u>CAUTION</u> :		 Τ F	JNWANTED MATER THE REMOVAL OF F	IALS IN T	HE STRUT COI HROUGH THE S	REA CLEAN. LOOSE T MPARTMENTS CAN PI STRUT DRAINS. IF YO N CAUSE DAMAGE TO	REVENT U DO NOT		
	(1)	Make su	ure that the work area	a is clean	and remove all	tools and other items.			
	SUBTA	SK 54-55-02-4	410-002						
	(2)	Close th	ne aft fairing access o	doors:					
		(AMM T	ASK 54-52-06-410-8	01)					
		Numbe	r Name/Location	<u>n</u>					
		434AR	Aft Strut Fairing	g, Right Fo	orward Panel, S	trut 1			
		434BL	Aft Strut Fairing	-					
		444AL	-	-	ward Panel, Str	ut 2			
		444BR	Aft Strut Fairing	g, Right Ai	t Panel, Strut 2				
		SK 54-55-02-4							
	(3)	Do this	task: Put the Strut Ba	ack to its I	Jsual Condition,	AMM TASK 54-51-01-4	140-801.		
				END OF	TASK ———				
		EFFECTIV AKS AI		SOURCE MRB	FUNCTIONAL O	CHECK LEFT ENGINE FO	RWARD STF	RUT AN	ND
					D633A109-AKS 54-090-01-01			Page 5 Oct 15/	





### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	FUNCTIONAL CHECK RIGHT ENGINE FORWARD STRUT AND AFT STRUT FAIRING DRAINS			BOEING CARD NO. <b>54-090-02-01</b>	
DATE	TASK FUNCTIONAL	STRUT AND	AFT STRUT FAIR	ING DRAINS	RELATE	D CARD
TAIL NUMBER	WORK AREA RIGHT STRUT	VERSION 1.1	THRESHOLD  30 MO	REPEAT 30 MO	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	9000 FC	9000 FC	ALL	ALL
		ACCESS 441BL 441BR 44	1CL 441CR 444AL	ZONE 420 440		

Functionally check the right engine forward strut and aft strut fairing drains.

INTERVAL NOTE: Whichever occurs first.

#### A. References

Reference	Title
AMM 08-21-02-580-801	Make the Airplane Level (P/B 201)
AMM 24-22-00-860-812	Remove Electrical Power (P/B 201)
AMM 54-51-01-040-801	Prepare the Strut for Maintenance Operations (P/B 201)
AMM 54-51-01-440-801	Put the Strut Back to its Usual Condition (P/B 201)
AMM 54-52-06-010-801	Aft Fairing Access Panel Removal (P/B 401)
AMM 54-52-06-410-801	Aft Fairing Access Panel Installation (P/B 401)

#### B. Consumable Materials

Reference	Description	Specification
G50316	Cloth - Clean, Dry, Lint-free, White, Cotton	

#### C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1155	Funnel - Long Neck
STD-1280	Source - Air, Regulated, Dry Filtered, 0-30 PSIG
STD-13465	Bucket - 2.5 Gallon Capacity
STD-13870	.75 inch O.D., clear vinyl hose
STD-3910	Container - Plastic
STD-77	Air Source - Regulated, Dry Filtered, 0-50 psig

EFFECTIVITY AKS ALL	SOURCE MRB	FUNCTIONAL CHECK RIGHT ENGINE FORMAFT STRUT FAIRING DRAINS	NARD STRUT AND
		D633A109-AKS 54-090-02-01	Page 1 of 5 Jun 15/2015



## 737-600/700/800/900 TASK CARDS

	DAT	Ē	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARI <b>54-090-02</b>					
TA	ASK	54-55-01-720	0-801				M	1ECH	INSF			
	trut Drain - Functional Test											
Α		General										
^			gives steps to do a	a functiona	al test of the stru	t drain.						
В	`	•	ne Functional Test									
		SUBTASK 54-55-01-01		•								
		1) Do this to	ask: Prepare the St -51-01-040-801.	rut for Mai	ntenance Opera	ations, AMM						
	8	SUBTASK 54-55-01-04	10-002									
	(	,	ask to remove elect -22-00-860-812.	rical powe	r:Remove Elect	rical Power, AMM						
	8	SUBTASK 54-55-01-01	0-003									
	(		ese access panels:									
		<u>Number</u>										
		431BL		_	ft Mid Strut Fair	_						
		431BR 431CL		•	ght Mid Strut Fa ft Overwing Fai	•						
		431CE 431CR		•	•	•						
431CR Forward Strut Fairing, Right Overwing Fairing, Strut 7 434AR Aft Strut Fairing, Right Forward Panel, Strut 1						•						
	434BL Aft Strut Fairing, Left Aft Panel, Strut 1											
		441BL		•	ft Mid Strut Fair	ing, Strut 2						
		441BR	Forward Strut	Fairing, Ri	ght Mid Strut Fa	iring, Strut 2						
		441CL		-	ft Overwing Fai	•						
		441CR		_	ght Overwing Fa	_						
		444AL		O .	ward Panel, Str	ut 2						
		444BR	Att Strut Fairin	g, Right At	t Panel, Strut 2							
		SUBTASK 54-55-01-21			(l (	ata talanta ka ara-ka ara-ka a	.1.0					
	(	,	re that the internal s blockage in the dra		the strut and dra	ain inlet is free of materia	al that can					
		SUBTASK 54-55-01-84										
	(	(5) Put a bu	cket, STD-13465 oı	r equivalen	it container beic	w the aft drain.						
	\$	SUBTASK 54-55-01-48										
	(	6) Put a .75 STD-134		TD-13870	on the drain tub	e to point the water into	the bucket,					
С	. I	Functional Te	est									
	8	SUBTASK 54-55-01-20										
(1) Use a long neck funnel, STD-1155, to pour 1 gallon (128 fluid ounces)(3.8 liters) of clear water along the aft end of the internal floor of the strut spar.							rs) of clean					
		NOTE: I	Pour water as close	to the dra	in as possible.							
		(a) Ma	ke sure that the wa	ter flows fr	eely from the af	t drain.						
		EFFECTIVIT AKS AL		SOURCE MRB	FUNCTIONAL (	CHECK RIGHT ENGINE FO	ORWARD STRU	JT A	ND			
					D633A109-AKS	<b>.</b>	Pa	ge 2				

54-090-02-01

Jun 15/2015





[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-090							
	SUBTA	SUBTASK 54-55-01-200-002												
	(2)													
	SUBTA	SUBTASK 54-55-01-720-001  (3) Lieu a 0.30 psig dry filtered regulated air source. STD 1280 at the strut drain inlet to blow												
	(3)	all remaining water out of the strut drain, into the bucket, STD-13465.												
	SUBTA	SUBTASK 54-55-01-210-009  (A) Make sure that there are no leaks at the strut drain best to strut drain line fitting												
	(4)													
	SUBTA	SUBTASK 54-55-01-910-001												
	(5)	(5) Remove remaining water from the strut with a cotton cloth, G50316.												
	SUBTA	SUBTASK 54-55-01-910-002												
	(6)													
D.	Put	Put the Airplane Back to Its Usual Condition												
	SUBTA	ASK 54-55-01-410-0	04											
	(1)	Close these	e access panels:											
		<u>Number</u>	Name/Location	<u>1</u>										
		431BL	Forward Strut F	airing, Le	ft Mid Strut Fairi	ng, Strut 1								
		431BR	Forward Strut F	airing, Ri	ght Mid Strut Fa	iring, Strut 1								
		431CL			ft Overwing Fair									
		431CR		_	ght Overwing Fa	_								
		434AR	_	_	orward Panel, St	rut 1								
		434BL 441BL	Aft Strut Fairing			na Ctrut 2								
		441BR		_	eft Mid Strut Fairi ght Mid Strut Fai	•								
		441CL			eft Overwing Fair	•								
		441CR		_	ght Overwing Fa	•								
		444AL			ward Panel, Stru	_								
		444BR	Aft Strut Fairing	, Right Af	ft Panel, Strut 2									
				END OF	TA SK									
				LIND OI	IAOR									
		EFFECTIVITY  AKS ALL		SOURCE MRB		HECK RIGHT ENGINE	FORWARD ST	RUT A	N					
		ANOALL		IVIKD	AFT STRUT FAI	RING DRAINS								
		ANOALL		WIKD	D633A109-AKS			Page 3	2 ~					



### 737-600/700/800/900 TASK CARDS

[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARE <b>54-090-02</b>		
		5-02-100-8 Sump Dra	02 in Inspection				M	ECH	INS
A.	Gene	ral							
	(1)	Oo this task	for the aft fairing	g sump dra	ain when there	is a possible blockage.			
В.	Prepa	re to Insp	ect the Aft Fairi	ng Sump l	Drain				
		C 54-55-02-040-00							
	` ,		·	ane Level,	AMM TASK 08	3-21-02-580-801.			
		( 54-55-02-040-00		rut for Mai	ntananaa Ona	rations ANNA			
			c: Prepare the St 1-01-040-801.	rut ior iviali	ntenance Opei	rations, Aiviivi			
		C 54-55-02-010-00			1.				
	` '	•	pplicable aft fairi	•	panels:				
		AMM TASI <b>Number</b>	√ 54-52-06-010-8  Name/Location  Name/Location						
	_	134AR	Aft Strut Fairin		orward Panel :	Strut 1			
		134BL	Aft Strut Fairin			Strut 1			
		144AL	Aft Strut Fairin	g, Left For	ward Panel, St				
		144BR	Aft Strut Fairin	g, Right Af	t Panel, Strut 2	2			
C.	Aft Fa	iring Draii	n Inspection						
		( 54-55-02-200-00		10 bolow t	ha aft fairing d	rain			
	. ,	•	ontainer, STD-39		_	fluid ounces)(7.5 liters).			
	_				,	iring does not have unwan	tod		
	(	mater		orial Surial	ce of the alt la	ining does not have unwan	ted		
		C 54-55-02-160-00		\/= = 1		(0.00 !!! ) 6 !			
	` ′	along the fo	orward end to the	aft end of	the internal flo	d ounce (0.03 liter) of clear por of the aft fairing.	n water		
	_		ur water on all in						
	(		three minutes, m ted into the conta			n 244 fluid ounces (7.2 lite ≀drain.	ers) is		
	(	,	ly make sure tha aft fairing.	it there are	no leaks thro	ugh the bulb seal at the for	ward end		
(c) Use a syringe to make sure that no single puddle of water is larger than 1 fluid ounce (0.03 liter).									
SUBTASK 54-55-02-160-006									
	(3) Remove remaining water from the aft fairing.								
	(4) l			-	air source, STI	O-77 or (0-345 kPa), to dry	the		
 		EFFECTIVITY AKS ALL		SOURCE MRB		CHECK RIGHT ENGINE FO AIRING DRAINS	RWARD STRU	JT A	NI
						_	_		

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				TAIL NUMBER			STATION	AIRLINE CAR	D NO.	54-090-		
D.			-	Back to It	s Usua	al Condi	tion		'		MECH	INS
		NSK 54-55-0 JTION:	MAKE UNWA THE F	ANTED MAREMOVAL	ATERIA OF FL	ALS IN T LUIDS TH	P THE STRUT A THE STRUT CO HROUGH THE RIALS, YOU CA	MPARTMENTS STRUT DRAIN	S CAN PREV S. IF YOU D	/ENT O NOT		
	(1)	Make	sure th	at the wor	k area	is clean	and remove all	tools and othe	r items.			
	SUBTA	NSK 54-55-0										
	(2)			fairing ac								
		,		54-52-06-		•						
		Numb 434AF 434BL 444AL 444BF	— - -	Aft Strut F Aft Strut F	airing, airing, airing,	Right Fo	orward Panel, S Panel, Strut 1 ward Panel, Str t Panel, Strut 2	rut 2				
	SUBTA	NSK 54-55-0			3,	3	,					
	(3)	Do thi	s task:	Put the St	trut Ba	ck to its l	Jsual Condition	, AMM TASK 5	4-51-01-440-	-801.		
						END OF	TASK ———					
		EFFECT AKS				SOURCE MRB		CHECK RIGHT I	ENGINE FOR		RU1	





### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	TITLE  LEFT STRUT TO WING ATTACHMENTS			BOEING CARD NO. <b>54-600-00-01</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>6000 FC</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				AIRPLANE ALL	ALL	
		431CL 431CR 43	4AL 434AR 434BL	. 511BT	ZONE <b>431 434</b>		

Inspect (General Visual) the lugs and clevises for all the links, fittings and pins.

See Doc D626A001-DTR, DTR check form 54-51-01, 54-51-02, 54-51-03, 54-51-04, 54-51-05, 54-51-06, 54-51-07, 54-51-08, 54-51-09, 54-51-15 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRUT TO WING ATTACHMENTS	
		D633A109-AKS 54-600-00-01	Page 1 of 2 Oct 15/2014



MECH IN
I pins.
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tions.
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)





		TITLE RIGHT STRUT TO WING ATTACHMENTS			BOEING CARD NO. <b>54-600-00-02</b>		
DATE	TASK GENERAL VISUAL				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLIC/	ABILITY ENGINE	
STATION	SKILL <b>AIRPL</b>	_			ALL	ALL	
		ACCESS 441CL 441CR 44	4AL 444AR 511BT		ZONE 441 444		

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRUT TO WING ATTACHMENTS	
		D633A109-AKS 54-600-00-02	Page 1 of 2 Feb 15/2015



	DATE			TAIL NUMBER	STATION		AIRLINE CARD NO.		BOEING CARD NO. <b>54-600-00-02</b>	
Т	TAS	K 54-	05-02-210-8	301	'				MECH	INS
1. <u>II</u>	INTERNAL - GENERAL VISUAL: STRUT TO WING ATTACHMENTS									
	Α.	Inspection								
,	Λ.	SUBTASK 54-05-02-010-010								
		(1)	Open these access panels for Enginer No. 1:							
		( ' )	Number	·						
			431CL			eft Overwing Fair	ring Strut 1			
			431CR		_	ght Overwing Fa	•			
			434AL	Aft Strut Fairing	, Left For	ward Panel, Stru	ut 1			
			434AR	•		orward Panel, St	trut 1			
			434BL	Aft Strut Fairing						
			511BT		-	Ipper Removabl	e Access Panel			
	Open these access panels for Engine No. 2:									
			Number	Name/Location						
			441CL		•	eft Overwing Fair	•			
			441CR 444AL		_	ght Overwing Fa ward Panel, Stru	•			
			444AR	Aft Strut Fairing						
			511BT		_	Ipper Removabl				
	SUBTASK 54-05-02-210-001									
	(2) Do a General Visual inspection of the lugs and clevises for all the links, fittings ar							s and pins.		
	See Doc D626A001-DTR, DTR check form 54-51-01, 54-51-02, 54-51-03, 54-51-							-		
	54-51-05, 54-51-06, 54-51-07, 54-51-08, 54-51-09, 54-51-15 for alternative inspecti									
	SUBTASK 54-05-02-410-043									
	(3) Close these access panels for Engine No. 1:									
			<u>Number</u>	Name/Location	1					
			431CL			ft Overwing Fair				
			431CR	Forward Strut F						
			434AL	J	,	ward Panel, Str				
			434AR 434BL	Aft Strut Fairing Aft Strut Fairing		orward Panel, St	trut 1			
			511BT	•		Ipper Removabl	e Access Panel			
	Close these access panels for Engine No. 2:									
			Number	Name/Location	•	110. 2.				
			441CL		_	eft Overwing Fair	ring Strut 2			
			441CR		_	ght Overwing Fai	•			
			444AL		_	ward Panel, Stru	•			
			444AR	•		orward Panel, St				
			511BT	Inboard Leading	g Edge, U	Ipper Removabl	e Access Panel			
					END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT STRUT	TO WING ATTACHMENTS	S		
						D633A109-AKS 54-600-00-02			age 2 b 15/	





AIRLINE CARD NO		LEFT LOWER SPAR CHORD - TYPICAL EXPOSED			BOEING CARD NO. <b>54-610-00-01</b>		
DATE	SPECIAL DETAILED		DETAIL		RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 431BL 431BR 43 NOTE	3AT 433BT 433CT	433DT	ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) both legs of the lower spar chords between the forward and aft engine mounts: Nacelle STA 203.6 -209.9 left and right hand chords,

nacelle STA 212.3-222.0 left and right hand chords, nacelle STA 224.7-231.8 left hand chord, nacelle STA 234.4-240.4 left and right hand chords, nacelle STA 243.5-250.6 left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Remove/displace heat shields and brackets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD - TYPICAL EXPOSED DETAIL
		D633A109-AKS Page 1 of 3 54-610-00-01 Oct 15/2014
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					TAS	K CARDS				
	DAT	ΓE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 54-610		
		54-05-02 RNAL - S			WER SPA	AR CHORD - T	YPICAL EXPOSED DE	TAIL	MECH	INS
A	<b>A</b> . I	nspectio	n							
	S	SUBTASK 54-0								
	(			access panels for	•	No. 1:				
			<u>ıber</u>	Name/Location						
		431I 433I 433I 433I 433I	BR AT BT CT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S Strut, Upper Sp Strut, Upper Sp	Fairing, Rig Spar Web Spar Web par Web, S	ght Mid Strut Fa , Strut 1 , Strut 1 Strut 1	•			
		Ope	n these	access panels fo	or Engine	No. 2:				
			<u>ıber</u>	Name/Location	-					
		441I 441I 443I 443I 443I	BL BR AT BT CT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S Strut, Upper Sp	Fairing, Rig Spar Web Spar Web oar Web, S	ght Mid Strut Fa , Strut 2 , Strut 2 Strut 2	•			
		4431		Strut, Upper Sp						
		NOT	E: Rer	move/displace he	at shields	and brackets a	is required.			
		betw chor hand	a High F veen the ds, nac d chord	Frequency Eddy ( e forward and aft elle STA 212.3-2	engine me 22.0 left a	ounts: Nacelle s nd right hand cl	legs of the lower spar STA 203.6 -209.9 left a hords, nacelle STA 224 nd chords, nacelle STA	nd right hand 7-231.8 left		
		See	Doc. D	626A001-DTR, D	TR check	form 54-51-10	, for alternative inspect	ions.		
		the	737 Noi	, ,	Manual (E	•	nt of this inspection is c inspection procedures			
	s	SUBTASK 54-0	5-02-410-01	8						
	(	(3) Clos	e these	access panels for	or Engine	No. 1:				
		<u>Nun</u>	<u>ıber</u>	Name/Location	<u>n</u>					
		4311 4311 433/ 4331	BR AT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S	airing, Rig Spar Web	ght Mid Strut Fa , Strut 1	_			
		4330		Strut, Upper Sp	ar Web, S	Strut 1				
		4331	DΤ	Strut, Upper Sp	oar Web, S	Strut 1				
			S ALL		SOURCE AWL	LEFT LOWER S	SPAR CHORD - TYPICAI	L EXPOSED D	ETAIL	
						D633A109-AKS 54-610-00-01			Page 2 eb 15/	





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ <b>54-610-</b>		
	Close these	e access panels fo	or Engine	No. 2:			MECH	INSP
	Number	Name/Location						
	441BL	Forward Strut F		ft Mid Strut Fai	ring, Strut 2			
	441BR	Forward Strut F						
	443AT	Strut, Forward S	Spar Web	, Strut 2				
	443BT	Strut, Forward S						
	443CT	Strut, Upper Sp						
	443DT	Strut, Upper Sp	ar Web, S	Strut 2				
			END OF	TASK ———				
	EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER	SPAR CHORD - TYPICAL	EXPOSED DI	ETAIL	
				D633A109-AK	S	F	Page 3	of 3
				54-610-00-01			eb 15/	





AIRLINE CARD NO		RIGHT LOWER SPAR CHORD - TYPICAL EXPOSED			BOEING CARD NO. <b>54-610-00-02</b>		
DATE	SPECIAL DETAILED		DETAIL		RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 441BL 441BR 44 NOTE	3AT 443BT 443CT	443DT	ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) both legs of the lower spar chords between the forward and aft engine mounts: Nacelle STA 203.6 -209.9 left and right hand chords,

nacelle STA 212.3-222.0 left and right hand chords, nacelle STA 224.7-231.8 left hand chord, nacelle STA 234.4-240.4 left and right hand chords, nacelle STA 243.5-250.6 left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Remove/displace heat shields and brackets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD - TYPICAL EXPOSED	DETAIL
		D633A109-AKS 54-610-00-02	Page 1 of 3 Oct 15/2014





				TAS	K CARDS				
	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-610		
		-05-02-250-6 AL - SPECIA		WER SPA	AR CHORD - T	YPICAL EXPOSED DE	ΓAIL	MECH	INS
A.	Insp	ection							
		ASK 54-05-02-010-0		or Engine	No. 1.				
	(1)	Number	e access panels for Name/Location	-	NO. I.				
		431BL 431BR 433AT 433BT 433CT 433DT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S Strut, Upper Sp Strut, Upper Sp	- Fairing, Le Fairing, Ri Spar Web Spar Web, S	ght Mid Strut Fa , Strut 1 , Strut 1 Strut 1	•			
		Open thes	e access panels fo	or Engine	No. 2:				
		Number 441BL 441BR 443AT 443BT 443CT	Name/Location Forward Strut F Forward Strut F Strut, Forward S Strut, Upper Sp	- Fairing, Le Fairing, Rig Spar Web Spar Web, S	ght Mid Strut Fa , Strut 2 , Strut 2 Strut 2				
		443DT	Strut, Upper Sp			a a ma an dua d			
			emove/displace he	at snieids	and brackets a	as required.			
	(2)	between the chords, nathand chords	Frequency Eddy ( ne forward and aft celle STA 212.3-2	engine me 22.0 left a	ounts: Nacelle nd right hand c	n legs of the lower spar of STA 203.6 -209.9 left an hords, nacelle STA 224. and chords, nacelle STA	d right hand 7-231.8 left		
		See Doc. [	D626A001-DTR, D	TR check	form 54-51-10	), for alternative inspection	ons.		
		the 737 No	` '	Manual (E	•	nt of this inspection is co inspection procedures			
	SUBTA	ASK 54-05-02-410-0	118						
	(3)	Close thes	e access panels f	or Engine	No. 1:				
		<u>Number</u>	Name/Location						
		431BL 431BR 433AT 433BT 433CT 433DT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S Strut, Upper Sp Strut, Upper Sp	Fairing, Rig Spar Web Spar Web oar Web, S	ght Mid Strut Fa , Strut 1 , Strut 1 Strut 1	•			
		EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LOWER	R SPAR CHORD - TYPICA	L EXPOSED	DETAII	L
					D633A109-AKS 54-610-00-02			Page 2 eb 15/	





DATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-610-00-02</b>	
	Close thes	e access panels fo	r Engine No. 2:		MECH	INSF
	Number	Name/Location				
	441BL		airing, Left Mid Strut l	Fairing, Strut 2		
	441BR		airing, Right Mid Stru			
	443AT	Strut, Forward S	Spar Web, Strut 2			
	443BT		Spar Web, Strut 2			
	443CT	Strut, Upper Spa				
	443DT	Strut, Upper Spa	ar Web, Strut 2			
			END OF TASK ——			
	EFFECTIVITY AKS ALL		SOURCE AWL RIGHT LOW	VER SPAR CHORD - TYPICA	L EXPOSED DETAII	L
			D633A109-A	AKS	Page 3	3 of





TASK	LEFT LOWER SPAR CHORD AT FIRE SEAL			BOEING CARD NO. <b>54-611-00-01</b>		
SPECIAL DETAILED	) DE	PRESSOR BRACI	KET	RELATE	D CARD	
WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC		ABILITY ENGINE	
SKILL AIRPL				ALL	ALL	
	ACCESS 431BL 431BR 43	3AT 433BT 433CT	433DT	ZONE <b>433</b>		
I	WORK AREA ENG/STRUT SKILL	DETAILED  WORK AREA ENG/STRUT SKILL AIRPL  ACCESS	DETAILED  WORK AREA ENG/STRUT SKILL AIRPL  ACCESS	DETAILED  WORK AREA VERSION THRESHOLD REPEAT  SKILL AIRPL  VERSION THRESHOLD REPEAT  1.1 56000 FC 18000 FC	DETAILED  WORK AREA ING/STRUT SKILL AIRPL  ACCESS  VERSION THRESHOLD REPEAT 1.1 56000 FC 18000 FC AIRPLANE ALL  ACCESS ZONE	

Inspect (High Frequency Eddy Current) the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and nacelle STA 207.8 for both the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

SOURCE AWL SOURCE BRACKET

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## 737-600/700/800/900 TASK CARDS

				IAS	K CARDS			
	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-611-00-	
l. <u>IN</u>				WER SPA	AR CHORD AT	FIRE SEAL DEPRESS	OR	CH INS
A	. Insp	ection						
	SUBTA	ASK 54-05-02-010-0	19					
	(1)	Open these	e access panels fo	or Engine	No. 1:			
		<u>Number</u>	Name/Location	<u>n</u>				
		431BL 431BR 433AT 433BT 433CT 433DT	Forward Strut F Forward Strut F Strut, Forward Strut, Forward Strut, Upper Sp Strut, Upper Sp	Fairing, Rig Spar Web Spar Web oar Web, S	ght Mid Strut Fa , Strut 1 , Strut 1 Strut 1	•		
		Open these	e access panels fo	or Engine	No. 2:			
		<u>Number</u>	Name/Location	<u>n</u>				
		441BL 441BR 443AT 443BT 443CT 443DT	Forward Strut F Forward Strut F Strut, Forward Strut, Forward Strut, Upper Sp Strut, Upper Sp	Fairing, Rig Spar Web Spar Web oar Web, S	ght Mid Strut Fa , Strut 2 , Strut 2 Strut 2	•		
	SUBTA	SK 54-05-02-250-0	02					
	(2)		bracket at nacelle			lower spar chord at the STA 207.8 for both the le		
		See Doc. D	0626A001-DTR, E	TR check	form 54-51-10	, for alternative inspecti	ons.	
		the 737 No		Manual (I		nt of this inspection is co inspection procedures		
	SUBTA	ASK 54-05-02-410-0	19					
	(3)	Close these	e access panels f	or Engine	No. 1:			
		<u>Number</u>	Name/Location	<u>n</u>				
		431BL 431BR 433AT 433BT 433CT 433DT	Forward Strut F Forward Strut F Strut, Forward Strut, Forward Strut, Upper Sp Strut, Upper Sp	airing, Rig Spar Web Spar Web oar Web, S	ght Mid Strut Fa , Strut 1 , Strut 1 Strut 1	_		
		Close these	e access panels f					
		Number	Name/Location	•				
		441BL	Forward Strut F		ft Mid Strut Fai	ring, Strut 2		
		EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER BRACKET	SPAR CHORD AT FIRE S	EAL DEPRESSOR	R
					D633A109-AK\$	3		e 2 of

54-611-00-01

Feb 15/2015





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-611-00-01</b>
(Con	tinued)			MECH INSI
(Con <u>Numb</u> 441BF 443AT 443BT 443CT 443DT	er Name/Location  Forward Strut Fa Strut, Forward S Strut, Forward S Strut, Upper Spa Strut, Upper Spa	par Web, Strut 2 ır Web, Strut 2	iring, Strut 2	MECH INSI
EFFECT AKS	IVITY <b>ALL</b>	SOURCE AWL LEFT LOWER S BRACKET  D633A109-AKS 54-611-00-01	SPAR CHORD AT FIRE SE	EAL DEPRESSOR  Page 3 of 5 Feb 15/201





AIRLINE	AIRLINE CARD NO		RIGHT LOWER SPAR CHORD AT FIRE SEAL			BOEING CARD NO. <b>54-611-00-02</b>	
DATE	SPECIAL DETAILED	DE	EPRESSOR BRACI	KET	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 441BL 441BR 44	3AT 443BT 443CT	443DT	ZONE 443		

Inspect (High Frequency Eddy Current) the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and nacelle STA 207.8 for both the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

AKS ALL

SOURCE AWL

RIGHT LOWER SPAR CHORD AT FIRE SEAL DEPRESSOR BRACKET

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54-611-00-02
Oct 15/2014



## 737-600/700/800/900 **TASK CARDS**

					1710	IN CANDS				
	С	OATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-611		
1.	INTE				OWER SPA	AR CHORD AT	FIRE SEAL DEPRESS	OR	MECH	INS
	^	lnou	_							
	Α.	-	ection	-						
		(1)	SK 54-05-02-010-01 Open these	e access panels f	or Engine	No 1:				
		(1)	Number	Name/Locatio	_	140. 1.				
			431BL	•		ft Mid Strut Fair	ing Strut 1			
			431BR		-	ght Mid Strut Fa	_			
			433AT	Strut, Forward	Spar Web	, Strut 1				
			433BT	Strut, Forward						
			433CT	Strut, Upper S						
			433DT	Strut, Upper S						
			-	access panels f	•	No. 2:				
			Number	Name/Locatio		(LM: 1 Ot ) [ F : 1	01. 1.0			
			441BL 441BR		•	ft Mid Strut Fair ght Mid Strut Fa	•			
			443AT	Strut, Forward	_	-	annig, Strut Z			
			443BT	Strut, Forward	•					
			443CT	Strut, Upper S	par Web, S	Strut 2				
			443DT	Strut, Upper S	par Web, S	Strut 2				
		SUBTA	SK 54-05-02-250-00	)2						
		(2)		oracket at nacelle			ower spar chord at the ITA 207.8 for both the Is			
			See Doc. D	626A001-DTR, [	OTR check	form 54-51-10	, for alternative inspecti	ons.		
			the 737 No	` '	: Manual (I	•	nt of this inspection is co inspection procedures			
		SUBTA	SK 54-05-02-410-01	9						
		(3)	Close these	e access panels	for Engine	No. 1:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			431BL 431BR 433AT 433BT 433CT 433DT		Fairing, Ri Spar Web Spar Web par Web, S	, Strut 1 Strut 1	•			
			Close these	e access panels	for Engine	No. 2:				
			Number	Name/Locatio	•					
			441BL	Forward Strut	– Fairing, Le	ft Mid Strut Fair	ring, Strut 2			
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LOWER BRACKET	SPAR CHORD AT FIRE	SEAL DEPRE	SSOR	
						D633A109-AKS	3		Page 2 eb 15/	
						1				





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-611-00-02</b>
(Con	tinued)			MECH INS
Numb 441BF 443AT 443BT 443CT	er Name/Location  Forward Strut Forward Strut, Forward Strut, Forward Strut, Upper Spa	airing, Right Mid Strut Fa Spar Web, Strut 2 Spar Web, Strut 2 ar Web, Strut 2	iring, Strut 2	
443DT	Strut, Upper Spa	ar Web, Strut 2		
		END OF TASK ———		
EFFECT <b>AKS</b> A	IVITY <b>ALL</b>	AWL BRACKET	SPAR CHORD AT FIRE S	
		D633A109-AKS 54-611-00-02		Page 3 of 5 Feb 15/201





AIRLINE CARD NO		LEFT LOWER SPAR CHORD AT FIRE SEAL			BOEING CARD NO. <b>54-611-01-01</b>		
DATE	SPECIAL DETAILED	- DE	EPRESSOR BRAC	KET	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION         THRESHOLD         REPEAT           1.1         56000 FC         18000 FC			ABILITY ENGINE		
STATION	SKILL AIRPL				AIRPLANE ALL	ALL	
		ACCESS 431BL 431BR 43 433DT	31CL 431CR 433A1	Г 433BT 433CT	ZONE <b>433</b>		

Inspect (Ultrasonic) the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

SOURCE AWL LEFT LOWER SPAR CHORD AT FIRE SEAL DEPRESSOR BRACKET

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54-611-01-01 Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-611-</b>		
I. <u>INT</u>				WER SPA	AR CHORD AT	FIRE SEAL DEPRESS	SOR	MECH	INS
A.	Insp	ection							
	SUBTA	ASK 54-05-02-010-00	)7						
	(1)	Open these	e access panels fo	or Engine	No. 1:				
		<u>Number</u>	Name/Location						
		431BL	Forward Strut F	-		•			
		431BR 431CL	Forward Strut F Forward Strut F	-	_	•			
		431CR	Forward Strut F	_	-	_			
		433AT	Strut, Forward S	-	-				
		433BT	Strut, Forward S	•					
		433CT 433DT	Strut, Upper Sp Strut, Upper Sp						
			access panels fo	ŕ					
		Number	Name/Location	•					
		441BL	Forward Strut F		ft Mid Strut Fair	ing, Strut 2			
		441BR	Forward Strut F	-	_	•			
		441CL	Forward Strut F	_	-	_			
		441CR 443AT	Forward Strut F Strut, Forward S	-	-	airing, Strut 2			
		443BT	Strut, Forward S						
		443CT	Strut, Upper Sp						
		443DT	Strut, Upper Sp	ar Web, S	Strut 2				
	SUBTA	ASK 54-05-02-130-00	)1						
	(2)		•			d at the fire seal depre left and right hand cho			
		See Doc. D	626A001-DTR, D	TR check	form 54-51-10,	, for alternative inspect	ions.		
		ASK 54-05-02-410-00							
	(3)		e access panels fo	•	No. 1:				
		Number	Name/Location						
		431BL 431BR	Forward Strut F Forward Strut F	•		•			
		431CL	Forward Strut F		-	_			
		431CR	Forward Strut F	•	•	•			
		433AT	Strut, Forward S						
		433BT	Strut, Forward S	•					
		433CT 433DT	Strut, Upper Sp Strut, Upper Sp						
		43301	Strut, Opper Sp	ai vveb, c	oudt 1				
									L
		AKS ALL		AWL	LEFT LOWER S BRACKET	SPAR CHORD AT FIRE S	SEAL DEPRES	SOR	





DATE	TAIL NUM	IBER	STATION	AIRLINE CARD NO.	BOEING CARD <b>54-611-01</b>	
Clo	se these access	panels for End	gine No. 2:		ME	ECH INSP
		/Location	<b>,</b>			
			, Left Mid Strut Fair	ing Strut 2		
			, Right Mid Strut Fa			
			,, Left Overwing Fair			
			, Right Overwing Fa			
		Forward Spar \		annig, oli di 2		
		Forward Spar \				
		Upper Spar We				
		Upper Spar We				
	,		OF TASK —			
		LIND	or rack			
EF A	FECTIVITY KS ALL	SOUR <b>AW</b>		SPAR CHORD AT FIRE S	EAL DEPRESSO	R
			D6224400 AKC		Dar	10 2 cf 2
			D633A109-AKS 54-611-01-01			ge 3 of 3 15/2015
			3-7-011-01-01		i eb	13/2013





AIRLINE CARD NO		RIGHT LOWER SPAR CHORD AT FIRE SEAL			BOEING CARD NO. <b>54-611-01-02</b>		
DATE	SPECIAL DETAILED	_ DE	EPRESSOR BRAC	KET	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION         THRESHOLD         REPEAT           1.1         56000 FC         18000 FC			ABILITY		
STATION	SKILL AIRPL				AIRPLANE ALL	ALL	
		ACCESS 441BL 441BR 44 443DT	11CL 441CR 443A1	Г 443BT 443CT	ZONE 443		

Inspect (Ultrasonic) the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

SOURCE AWL RIGHT LOWER SPAR CHORD AT FIRE SEAL DEPRESSOR BRACKET

D633A109-AKS Page 1 of 3 54-611-01-02 Oct 15/2014



TASK 54-05-02-130-801  INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD AT FIRE SEAL DEPRESSOR BRACKET  A. Inspection  SUBTIASK 54-05-02-19-0007  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  433CT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433BT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441BL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443BT Strut, Upper Spar Web, Strut 2  443BT Strut, Upper Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  30BTASK 54-05-02-19-0001  (2) Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  30BTASK 54-05-02-19-0007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Mid S		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-611		
(1) Open these access panels for Engine No. 1:    Number   Name/Location	. INT	ERNA	AL - SPECIA		WER SPA	AR CHORD AT	FIRE SEAL DEPRESS	OR	MECH	INS
(1) Open these access panels for Engine No. 1:    Number   Name/Location	Α.	Inst	ection							
Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Left Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Upper Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  443AT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443BT Strut, Upper Spar Web, Strut 2  443BT Strut, Upper Spar Web, Strut 2  443DT Strut, Upper Spar Web, Strut 2  308TASK \$445-02-130-401  (2) Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  Subtrask \$445-02-130-001  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1		-		07						
431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433BT Strut, Upper Spar Web, Strut 1 433BT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2 441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443BT Strut, Upper Spar Web, Strut 2 443BT Strut, Upper Spar Web, Strut 2 308TASK \$445902430401  (2) Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  Subtrask \$4459024104007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1		(1)	Open these	e access panels fo	or Engine	No. 1:				
431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 500 Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  SUBTASK \$445-92-410-007  (3) Close these access panels for Engine No. 1:  Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431CR Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431CR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1			<u>Number</u>	Name/Location	<u>n</u>					
Open these access panels for Engine No. 2:    Number   Name/Location			431BR 431CL 431CR 433AT 433BT 433CT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S Strut, Upper Sp	Fairing, Rig Fairing, Le Fairing, Rig Spar Web Spar Web, S	ght Mid Strut Fa Ift Overwing Fair ght Overwing Fa If Strut 1 If Strut 1 If Strut 1	iring, Strut 1 ring, Strut 1			
Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2  443AT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  443DT Strut, Upper Spar Web, Strut 2  3UBTASK \$4-05-02-130-001  (2) Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  SUBTASK \$4-05-02-410-007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1					ŕ					
441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443AT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443BT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2  SUBTASK \$4-05-02-130-001  (2) Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  SUBTASK \$4-05-02-410-007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1			-	·	•					
(2) Do an Ultrasonic inspection of the the lower spar chord at the fire seal depressor bracket at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  SUBTASK 54-05-02-410-007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1			441BR 441CL 441CR 443AT 443BT 443CT	Forward Strut F Forward Strut F Strut, Forward S Strut, Forward S Strut, Upper Sp	Fairing, Rig Fairing, Le Fairing, Rig Spar Web Spar Web Spar Web, S	ght Mid Strut Fa Ift Overwing Fair ght Overwing Fa If, Strut 2 If, Strut 2 If Strut 2	iring, Strut 2 ring, Strut 2			
at nacelle STA 203.4 and at nacelle STA 207.8 on the left and right hand chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  SUBTASK 54-05-02-410-007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1		SUBT	ASK 54-05-02-130-0							
(3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1		(2)		•		•	•			
(3) Close these access panels for Engine No. 1:    Number   Name/Location			See Doc. D	0626A001-DTR, D	TR check	c form 54-51-10,	for alternative inspection	ons.		
Number 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1										
431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1		(3)		-	•	No. 1:				
AND ALL			431BL 431BR 431CL 431CR 433AT 433BT 433CT	Forward Strut F Forward Strut F Forward Strut F Forward Strut F Strut, Forward Strut, Forward Strut, Forward Strut, Upper Sp	Fairing, Le Fairing, Rig Fairing, Le Fairing, Rig Spar Web Spar Web	ght Mid Strut Fa Ift Overwing Fair ght Overwing Fa I, Strut 1 I, Strut 1 Strut 1	iring, Strut 1 ring, Strut 1			
						RIGHT LOWER BRACKET	SPAR CHORD AT FIRE	SEAL DEPRE	SSOR	
									Page 2 eb 15/	





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO 54-611-01-02	
Close t	these access panels fo	or Engine No. 2:		MECH	INSP
		_			
Number 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT	Forward Strut Forward Strut Forward Strut Forward Strut Forward Strut Forward Strut, Forward Strut, Forward Strut, Upper Spaceton	airing, Left Mid Strut Fair airing, Right Mid Strut Fa airing, Left Overwing Fair airing, Right Overwing Fa Spar Web, Strut 2 Spar Web, Strut 2 ar Web, Strut 2	iring, Strut 2 ring, Strut 2		
EFFECTI AKS A		SOURCE AWL RIGHT LOWER BRACKET  D633A109-AKS 54-611-01-02	SPAR CHORD AT FIRE	SEAL DEPRESSOF Page Feb 15	3 of 3





AIRLINE CARD NO		LEFT LOWER SPAR CHORD - BRACKET			BOEING CARD NO. <b>54-612-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (Ultrasonic) the lower spar chord bracket at nacelle STA 216.0 on the left hand side and nacelle STA 218.0 on the right hand side.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative repeat inspection.

**ACCESS NOTE:** Removal of insulation heat shield is required. Remove fan cowls, thrust reversers, and engines as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD - BRACKET	
		D633A109-AKS 54-612-00-01	Page 1 of 2 Oct 15/2014



Right Thrust R Open these access panels f Number Name/Location	for Engine No. 1:  on Engine 1 VI, Engine 1 Verser, Engine 1 Leverser, Engine 1	- BRACKET	MECH IN
ction  x 54-05-02-010-002  Open these access panels to the sea access p	for Engine No. 1:  on Engine 1 VI, Engine 1 Verser, Engine 1 Leverser, Engine 1	- BRACKET	
Open these access panels for the seaccess panels for t	on Engine 1 vl, Engine 1 verser, Engine 1 Leverser, Engine 1		
Open these access panels to Number	on Engine 1 vl, Engine 1 verser, Engine 1 Leverser, Engine 1		
Number 413 Left Fan Cowl, 414 Right Fan Cow 415 Left Thrust Re 416 Right Thrust R Open these access panels f Number Name/Location	on Engine 1 vl, Engine 1 verser, Engine 1 Leverser, Engine 1		
Left Fan Cowl, 414 Right Fan Cowl 415 Left Thrust Re 416 Right Thrust R Open these access panels f Number Name/Location	— , Engine 1 vl, Engine 1 verser, Engine 1 leverser, Engine 1		
Right Fan Cow 415 Left Thrust Re 416 Right Thrust R Open these access panels t Number Name/Location	vI, Engine 1 verser, Engine 1 leverser, Engine 1		
Left Thrust Rewards 416 Right Thrust Rewards Proper these access panels the Number Name/Location Properties of the Number Name/Location Properties of the Name Properties	verser, Engine 1 Leverser, Engine 1		
Right Thrust R Open these access panels f Number Name/Location	everser, Engine 1		
Open these access panels to Number Name/Location	•		
Number Name/Location	. ogo		
	•		
+23 Leit Fan Cowi,	— , Engine 2		
Right Fan Cow	•		
	_		
· ·			
	-	ed. Remove fan cowls, thrust	
K 54-05-02-130-002			
-	-		.0 on the
See Doc. D626A001-DTR, I	DTR check form 54-51	-10, for alternative repeat insp	pection.
K 54-05-02-410-002			
·	•		
,	_		
_	_		
	-		
Close these access panels	for Engine No. 2:		
Number Name/Location	<u>on</u>		
	_		
	_		
+20 Right Hirust R	-		
	- END OF TASK	_	
EFFECTIVITY AKS ALL	SOURCE LEFT LOWI	ER SPAR CHORD - BRACKET	
	D633A109-	AKS	Page 2 o
4 N K C N 4 4 4 4 4 C N 4 4 4 4	Right Thrust Resources Removal of insulation reversers, and enging 154-05-02-130-002  Do an Ultrasonis inspection aft hand side and nacelle Scee Doc. D626A001-DTR, 154-05-02-410-002  Close these access panels Name/Location Right Fan Cowlett Right Thrust Resources Right Thrust Resources Right Thrust Resources Right Fan Cowlett Right Fa	Right Thrust Reverser, Engine 2  NOTE: Removal of insulation heat shield is requir reversers, and engines as required.  154-05-02-130-002  Do an Ultrasonis inspection of the lower spar chore of the hand side and nacelle STA 218.0 on the right of the hand side and nacelle STA 21	Right Thrust Reverser, Engine 2  NOTE: Removal of insulation heat shield is required. Remove fan cowls, thrust reversers, and engines as required.  S405-02-130-002  Do an Ultrasonis inspection of the lower spar chord bracket at nacelle STA 216. Left hand side and nacelle STA 218.0 on the right hand side.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative repeat insulations the second





AIRLINE CARD NO		RIGHT LOWER SPAR CHORD - BRACKET			BOEING CARD NO. <b>54-612-00-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (Ultrasonic) the lower spar chord bracket at nacelle STA 216.0 on the left hand side and nacelle STA 218.0 on the right hand side.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative repeat inspection.

**ACCESS NOTE:** Removal of insulation heat shield is required. Remove fan cowls, thrust reversers, and engines as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD - BRACKET	
		D633A109-AKS 54-612-00-02	Page 1 of 2 Oct 15/2014



	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 2-00-02	
	TAS	K 54-	05-02-130	0-802					MECH	INSP
1.	INT	ERNA	L - SPEC	IAL DETAILED: LO	WER SPA	AR CHORD - BI	RACKET			
	A.	Insp	ection							
		SUBTA	NSK 54-05-02-01	10-002						
		(1)	Open the	ese access panels fo	or Engine	No. 1:				
			Number		_					
			413	Left Fan Cowl,	-	1				
			414 415	Right Fan Cowl Left Thrust Rev	-					
			416	Right Thrust Re		•				
				ese access panels fo		· ·				
			Number	•	-	. 10. 2.				
			423	Left Fan Cowl,						
			424	Right Fan Cowl	-	2				
			425	Left Thrust Rev		•				
			426	Right Thrust Re	everser, E	ngine 2				
	NOTE: Removal of insulation heat shield is required. Remove fan cowls, thrust reversers, and engines as required.									
		SUBTA	NSK 54-05-02-13	30-002						
		(2)		Itrasonis inspection of side and nacelle ST		•	acket at nacelle STA 21 I side.	6.0 on the		
			See Doc	. D626A001-DTR, D	TR check	c form 54-51-10,	for alternative repeat in	spection.		
		SUBTA	ASK 54-05-02-41	10-002						
		(3)	Close the	ese access panels fo	or Engine	No. 1:				
			<u>Number</u>		_					
			413	Left Fan Cowl,	-	4				
			414 415	Right Fan Cowl Left Thrust Rev	_					
			416	Right Thrust Re		-				
			Close the	ese access panels fo		_				
			Number	·	•					
			423	Left Fan Cowl,	_					
	424 Right Fan Cowl, Engine 2									
			425	Left Thrust Rev		•				
			426	Right Thrust Re	everser, E	ngine 2				
					END OF	TASK ———				
			EFFECTIVIT AKS AL		SOURCE <b>AWL</b>	RIGHT LOWER	SPAR CHORD - BRACK	ΕT		[
						D633A109-AKS			Page 2	of 2
						54-612-00-02			eb 15/	





AIRLINE	CARD NO	LEFT LOWER S	LEFT LOWER SPAR CHORD AT FRAME WITH			BOEING CARD NO. <b>54-613-00-01</b>		
DATE	SPECIAL DETAILED		BRACKETS		RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE		
STATION	SKILL AIRPL				ALL	ALL		
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>			

Inspect (Ultrasonic) the lower spar chord at the frames with brackets at nacelle STA 222.6 on the left and right chords and nacelle STA 242.7 on the left and right chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

ACCESS NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shields as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD AT FRAME WITH BRACKETS		
		D633A109-AKS 54-613-00-01	Page 1 of 2 Oct 15/2014	



DA	TE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-613		
TASK	<b>54-</b> 0	)5-02-130-8	03					MECH	INSF
. INTER	RNAL	- SPECIA	L DETAILED: LO	OWER SPA	AR CHORD AT F	RAMES WITH BRAC	CKETS		
		ection							
	-	K 54-05-02-010-00	13						
			access panels	for Engine	No. 1:				
	` '	Number	Name/Location	•					
		413	Left Fan Cowl,	Engine 1					
		414	Right Fan Cow	-					
	415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1								
			•						
		•	access panels	•	No. 2:				
	Number Name/Location  423 Left Fan Cowl, Engine 2								
	423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2								
		426	Right Thrust R	-					
		NOTE: Re	move fan cowls.	thrust reve	ersers. engine an	d insulation heat shie	lds as		
			uired.		3				
	SUBTAS	K 54-05-02-130-00	13						
(	. ,		•		er spar chord at the left and right of	the frames with brack chords.	ets at nacelle		
		See Doc. D	626A001-DTR,	DTR check	form 54-51-10,	for alternative inspect	ions.		
:	SUBTAS	K 54-05-02-410-00	3						
(	(3)	Close these	e access panels	for Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>on</u>					
		413	Left Fan Cowl,	-					
		414 415	Right Fan Cow Left Thrust Re	-					
		416	Right Thrust R	-					
			access panels		_				
		Number	Name/Location	J	140. 2.				
		423	Left Fan Cowl,						
		424	Right Fan Cow	-	2				
		425	Left Thrust Re						
		426	Right Thrust R	everser, Ei	ngine 2				
				- END OF	TASK ———				
		FFFFOTN "TV		COURCE			E 14/1717 DD 4 0		
		AKS ALL		SOURCE AWL	LEFT LOWER SI	PAR CHORD AT FRAM	E WITH BRAC	KETS	
		ANS ALL		AVVL					
		ANS ALL		AVVL	D633A109-AKS			Page 2	





AIRLINE	CARD NO	RIGHT LOWER	RIGHT LOWER SPAR CHORD AT FRAME WITH			BOEING CARD NO. <b>54-613-00-02</b>		
DATE	SPECIAL DETAILED		BRACKET		RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC AIRPLANE	ABILITY ENGINE		
STATION	SKILL AIRPL				ALL	ALL		
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>			

Inspect (Ultrasonic) the lower spar chord at the frames with brackets at nacelle STA 222.6 on the left and right chords and nacelle STA 242.7 on the left and right chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

ACCESS NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shields as required.

SOURCE AWL RIGHT LOWER SPAR CHORD AT FRAME WITH BRACKET

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54-613-00-02 Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-613-00-02</b>		
TAS	K 54-	05-02-130-8	803	'	,			MECH	INSF
INTE	ERNA	L - SPECIA	L DETAILED: LO	OWER SPA	AR CHORD AT F	RAMES WITH BRAC	KETS		
Α.	Insp	ection							
	-		03						
	(1)	Open these	e access panels	for Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>on</u>					
		413		-					
			-	-					
				-					
			-						
		•	•	•	110. 2.				
			•						
				•	2				
		425	-	-					
		426	Right Thrust R	teverser, Ei	ngine 2				
NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shields as required.									
	SUBTA	SK 54-05-02-130-00	03						
	(2)		•		•		ets at nacelle		
		See Doc. D	0626A001-DTR,	DTR check	form 54-51-10,	for alternative inspect	ions.		
	SUBTA	SK 54-05-02-410-00	03						
	(3)	Close these	e access panels	for Engine	No. 1:				
		<u>Number</u>		<del></del>					
				-					
			-	-					
				-					
			_		_				
			•	J	. 10. 2.				
		423							
		424		-	2				
		425							
	426 Right Thrust Reverser, Engine 2								
				- END OF	TASK ———				
		EFFECTIVITY <b>AKS ALL</b>		SOURCE AWL	RIGHT LOWER S	SPAR CHORD AT FRAI	ME WITH BRA	CKET	
				7					
		INTERNA  A. Insp SUBTA (1)  SUBTA (2)	A. Inspection  SUBTASK 54-05-02-010-01  (1) Open these  Number  413  414  415  416  Open these  Number  423  424  425  426  NOTE: Re  SUBTASK 54-05-02-130-01  (2) Do an Ultra  STA 222.6  See Doc. E  SUBTASK 54-05-02-410-01  (3) Close these  Number  413  414  415  416  Close these  Number  423  424  425  426	A. Inspection  SUBTASK 54-05-02-010-003  (1) Open these access panels  Number Name/Location  413 Left Fan Cowley  414 Right Fan Cowley  415 Left Thrust Reverted Right Thrust Reverted Right Thrust Reverted Right Fan Cowley  423 Left Fan Cowley  424 Right Fan Cowley  425 Left Thrust Reverted Right Fan Cowley  (3) Close these access panels  Number Name/Location  413 Left Fan Cowley  414 Right Fan Cowley  415 Left Thrust Reverted Right Thrust Reverted Right Fan Cowley  416 Right Thrust Reverted Right Fan Cowley  427 Left Thrust Reverted Right Fan Cowley  428 Left Fan Cowley  429 Left Thrust Reverted Right Fan Cowley  420 Right Thrust Reverted Right Thrust	A. Inspection  SUBTASK 54-05-02-010-003  (1) Open these access panels for Engine Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine NOTE: Remove fan cowls, thrust reversequired.  SUBTASK 54-05-02-130-003  (2) Do an Ultrasonic inspection of the lowe STA 222.6 and nacelle STA 242.7 on the See Doc. D626A001-DTR, DTR check SUBTASK 54-05-02-410-003  (3) Close these access panels for Engine Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine Name/Location  416 Right Thrust Reverser, Engine 1  417 Right Fan Cowl, Engine 1  418 Left Thrust Reverser, Engine Number Name/Location  429 Left Fan Cowl, Engine 1  410 Right Fan Cowl, Engine 1  411 Right Fan Cowl, Engine 1  412 Left Thrust Reverser, Engine Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine Number Name/Location  426 Right Thrust Reverser, Engine Number Name/Location  427 Right Fan Cowl, Engine 2  428 Right Fan Cowl, Engine 2  429 Right Thrust Reverser, Engine Number Name/Location  420 Right Thrust Reverser, Engine Number Name/Location  421 Right Fan Cowl, Engine 2  422 Right Thrust Reverser, Engine Right Thrust Reverse	A. Inspection  SUBTASK 54-05-02-010-003  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  NOTE: Remove fan cowls, thrust reversers, engine an required.  SUBTASK 54-05-02-130-003  (2) Do an Ultrasonic inspection of the lower spar chord at 1 STA 222.6 and nacelle STA 242.7 on the left and right See Doc. D626A001-DTR, DTR check form 54-51-10, SUBTASK 54-05-02-410-003  (3) Close these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  423 Left Thrust Reverser, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  423 Left Thrust Reverser, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  427 Right Thrust Reverser, Engine 2  428 Right Thrust Reverser, Engine 2  429 Right Thrust Reverser, Engine 2  420 Right Thrust Reverser, Engine 2	INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD AT FRAMES WITH BRACE  A. Inspection  SUBTASK \$449542-01-003  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shie required.  SUBTASK \$4495-42-10-003  (2) Do an Ultrasonic inspection of the lower spar chord at the frames with brack STA 222.6 and nacelle STA 242.7 on the left and right chords.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspect SUBTASK \$4495-42-10-003  (3) Close these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 1  416 Right Thrust Reverser, Engine 2  427 Right Fan Cowl, Engine 2  428 Right Thrust Reverser, Engine 2  429 Right Thrust Reverser, Engine 2  420 Right Thrust Reverser, Engine 2  421 Right Thrust Reverser, Engine 2  422 Right Thrust Reverser, Engine 2  423 Left Thrust Reverser, Engine 2  424 Right Thrust Reverser, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  427 Left Thrust Reverser, Engine 2  428 Right Thrust Reverser, Engine 2  429 Right Thrust Reverser, Engine 2  420 Right Thrust Reverser, Engine 2  421 Left Thrust Reverser, Engine 2  422 Left Thrust Reverser, Engine 2  423 Left Thrust Reverser, Engine 2  424 Right Thrust Reverser, Engine 2  425 Left Thrust Reverser, Engine 2	TASK 54-05-02-130-803 INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD AT FRAMES WITH BRACKETS  A. Inspection  SUBTASK 54-05-02-010-003  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shields as required.  SUBTASK 54-09-02-19-003  (2) Do an Ultrasonic inspection of the lower spar chord at the frames with brackets at nacelle STA 222.6 and nacelle STA 242.7 on the left and right chords. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.  SUBTASK 54-09-02-19-003  (3) Close these access panels for Engine No. 1:  Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 417 Left Thrust Reverser, Engine 1 418 Right Fan Cowl, Engine 2 429 Left Thrust Reverser, Engine 1 420 Left Fan Cowl, Engine 2 421 Right Fan Cowl, Engine 2 422 Left Thrust Reverser, Engine 2 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 427 Right Fan Cowl, Engine 2 428 Right Fan Cowl, Engine 2 429 Left Thrust Reverser, Engine 2 420 Right Thrust Reverser, Engine 2 421 Right Fan Cowl, Engine 2 422 Left Thrust Reverser, Engine 2 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 427 Left Thrust Reverser, Engine 2 428 Left Thrust Reverser, Engine 2 429 Left Thrust Reverser, Engine 2 420 Left Thrust Reverser, Engine 2	TASK 54-05-02-130-803 INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD AT FRAMES WITH BRACKETS  A. Inspection SUBTRANK 54-05-02-01-003  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 417 Open these access panels for Engine No. 2:  Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 427 NOTE: Remove fan cowls, thrust reversers, engine and insulation heat shields as required.  SUBTRANK 54-05-02-100-003  (2) Do an Ultrasonic inspection of the lower spar chord at the frames with brackets at nacelle STA 222.6 and nacelle STA 242.7 on the left and right chords.  SUBTRANK 54-05-02-100-003  (3) Close these access panels for Engine No. 1:  Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 417 Close these access panels for Engine No. 2:  Number Name/Location 423 Left Fan Cowl, Engine 1 434 Right Fan Cowl, Engine 1 435 Left Thrust Reverser, Engine 2 424 Right Thrust Reverser, Engine 1 435 Left Fan Cowl, Engine 1 436 Right Thrust Reverser, Engine 2 427 Right Thrust Reverser, Engine 2 428 Right Thrust Reverser, Engine 2 429 Right Thrust Reverser, Engine 2 420 Right Thrust Reverser, Engine 2 421 Right Thrust Reverser, Engine 2 422 Right Thrust Reverser, Engine 2 423 Right Thrust Reverser, Engine 2 424 Right Thrust Reverser, Engine 2 425 Left Thrust Reverser, Engine 2 426 RID OF TASK



EFFECTIVITY AKS ALL



#### 737-600/700/800/900 TASK CARDS

AIRLINE	CARD NO	LEFT LOWER	SPAR CHORD (D	IRECTION 1)		CARD NO. <b>4-00-01</b>
DATE	TASK SPECIAL DETAILED		(	,		ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 413 414 415 416			ZONE 433	
ee Doc. D62 ne NDI meth	6A001-DTR, DT	Current) the lower R check form 54-5 to accomplish the inspection procedu	1-10, for alternat	ive repeat insp	ection. ained in the 737 No	ondestructiv
st Maridar (	D0-37233). The	mspection procedu	res are containe	u III i ait 0, out	Ject 34-40-03.	

LEFT LOWER SPAR CHORD (DIRECTION 1)

SOURCE

**AWL** 



	DAT	E	TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>1-00-01</b>	
T	ASK	54-05-02-	 250-803				1	MECH	INSF
1. <u>IN</u>	NTER	NAL - SP	ECIAL DETAILED: LO	OWER SPA	AR CHORD (DI	RECTION 1)			
A	A. lı	nspection							
		UBTASK 54-05-0							
	(	1) Open	these access panels	for Engine	No. 1:				
		Numb	oer Name/Location	<u>on</u>					
		413	Left Fan Cowl	, Engine 1					
		414	Right Fan Cov	_					
		415	Left Thrust Re		•				
		416	Right Thrust R		•				
		•	these access panels	_	No. 2:				
		<u>Numb</u>							
		423	Left Fan Cowl		_				
		424	Right Fan Cov	_					
		425 426	Left Thrust Re Right Thrust R	7	-				
			· ·	(CVC1301, L	rigirio 2				
		ивтаѕк 54-05-0 2) Do a I		Current ins	spection of the I	ower spar chord at the	aft engine		
	(4		t bulkhead.	Currenting	spection of the i	ower spar chord at the	an engine		
		See D	oc. D626A001-DTR.	DTR check	form 54-51-10	, for alternative repeat in	nspection.		
						nt of this inspection is co	•		
			` '	-	•	inspection procedures			
		contai	ined in Part 6, Subject	t 54-40-03.					
	s	UBTASK 54-05-0	02-410-020						
	(;	3) Close	these access panels	for Engine	No. 1:				
		Numb	<u> Name/Locatio</u>	<u>on</u>					
		413	Left Fan Cowl	-					
		414	Right Fan Cov	_					
		415	Left Thrust Re		•				
		416	Right Thrust R		•				
			these access panels	_	No. 2:				
		Numb		_					
		423	Left Fan Cowl	-	2				
		424 425	Right Fan Cov Left Thrust Re						
		425	Right Thrust R	7	-				
		120	ragile rindocr						
				– END OF	TASK ———				
		EFFEC <b>AKS</b>		SOURCE AWL	LEFT LOWER S	SPAR CHORD (DIRECTIO	ON 1)		
					D633A109-AKS 54-614-00-01	3		Page 2 Feb 15/	





AIRLINE CARD NO		RIGHT LOWER SPAR CHORD (DIRECTION 1)			54-614	
DATE	TASK SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY	
		- ···	3000010	1000010	AIRPLANE	ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS			ZONE	
		423 424 425 426			443	

DATE	SPECIAL				RELATE	ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL <b>AIRPL</b>				ALL	ALL
		ACCESS 423 424 425 426			ZONE <b>443</b>	
		Current) the lower				
		R check form 54-5				
		to accomplish the nspection procedu			ained in the 737 No biect 54-40-03.	ondestructive
CSt Maridar (	D0-07200). THE I	napection proced	nes are containe	50 III 1 art 0, 00	bjeet 94-40-00.	
	EFFECTIVITY AKS ALL	SOUR <b>AW</b>		VER SPAR CHOI	RD (DIRECTION 1)	
			D633A109-/			Page 1 of Oct 15/201

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD (DIRECTION 1)	
		D633A109-AKS 54-614-00-02	Page 1 of 2 Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-614		
TA	ASK 54	l-05-02-250	)-803					MECH	INS
1. <u>IN</u>	TERN	AL - SPEC	IAL DETAILED: LO	WER SPA	AR CHORD (DI	RECTION 1)			
Α	. Ins	pection							
		ГАЅК 54-05-02-01	0-020						
	(1)	Open the	ese access panels f	or Engine	No. 1:				
	` ,	Number	Name/Locatio	<u>n</u>					
		413	Left Fan Cowl,	Engine 1					
		414	Right Fan Cow	l, Engine	1				
		415	Left Thrust Rev		-				
		416	Right Thrust Ro	everser, E	ngine 1				
		Open the	ese access panels f	or Engine	No. 2:				
	Number Name/Location								
		423	Left Fan Cowl,	-					
		424	Right Fan Cow	_					
	425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2								
	SUBTASK 54-05-02-250-003								
	(2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine						aft engine		
		mount bulkhead.							
		See Doc	e Doc. D626A001-DTR, DTR check form 54-51-10, for alternative repeat inspection. e NDI method(s) necessary to accomplish the intent of this inspection is contained in e 737 Nondestructive Test Manual (D6-37239). The inspection procedures are						
	contained in Part 6, Subject 54-40-03.  SUBTASK 54-05-02-410-020								
	(3)		ese access panels f	•	No. 1:				
		<u>Number</u>	•						
		413	Left Fan Cowl,						
		414	Right Fan Cow	_					
		415 416	Left Thrust Rev Right Thrust Re		•				
			•		· ·				
			ese access panels f	_	No. 2:				
		<u>Number</u>	Name/Locatio						
		423	Left Fan Cowl,	-	•				
		424	Right Fan Cow						
		425 426	Left Thrust Rev Right Thrust Re		-				
		420	Right Thrust Ri	everser, E	rigine z				
				- END OF	TASK ———				
		EFFECTIVIT <b>AKS AL</b> I		SOURCE AWL	RIGHT LOWER	SPAR CHORD (DIRECT	ION 1)		
					D633A109-AKS 54-614-00-02	3		Page 2 eb 15/	
					i .				





AIRLINE CARD NO		TITLE  LEFT LOWER SPAR CHORD (DIRECTION 2)			BOEING CARD NO. <b>54-614-01-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) the lower spar chord at the aft engine mount bulkhead.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Inspection requires the removal of engine mount.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD (DIRECTION 2)	
		D633A109-AKS 54-614-01-01	Page 1 of 2 Oct 15/2014



TASK 54-05-02-250-804  1. INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD (DIRECTION 2)  A. Inspection  SUBTASK 54-05-02-010-022  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engmount bulkhead.  See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.	4-614-01-01								
A. Inspection  SUBTASK 54-05-02-010-022  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.	MECH II								
(1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engmount bulkhead.									
(1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engmount bulkhead.									
Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engmount bulkhead.									
413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2:  Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004 (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine unut bulkhead.									
414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engmount bulkhead.									
415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
A16 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2  NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
<ul> <li>425 Left Thrust Reverser, Engine 2</li> <li>426 Right Thrust Reverser, Engine 2</li> <li>NOTE: Inspection requires the removal of engine mount.</li> <li>SUBTASK 54-05-02-250-004</li> <li>(2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.</li> </ul>									
<ul> <li>426 Right Thrust Reverser, Engine 2         NOTE: Inspection requires the removal of engine mount.     </li> <li>SUBTASK 54-05-02-250-004</li> <li>(2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.</li> </ul>									
NOTE: Inspection requires the removal of engine mount.  SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.									
SUBTASK 54-05-02-250-004  (2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engineering mount bulkhead.									
(2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engineering mount bulkhead.									
mount bulkhead.	no								
See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.	ile								
See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.									
The NDI method(s) necessary to accomplish the intent of this inspection is contained	d in								
the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.									
SUBTASK 54-05-02-410-022									
(3) Close these access panels for Engine No. 1:									
Number Name/Location									
413 Left Fan Cowl, Engine 1									
414 Right Fan Cowl, Engine 1									
415 Left Thrust Reverser, Engine 1									
416 Right Thrust Reverser, Engine 1									
Close these access panels for Engine No. 2:									
Number Name/Location									
423 Left Fan Cowl, Engine 2									
424 Right Fan Cowl, Engine 2									
425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2									
——— END OF TASK ———									
EFFECTIVITY AKS ALL SOURCE AWL LEFT LOWER SPAR CHORD (DIRECTION 2)									
D633A109-AKS 54-614-01-01									





AIRLINE CARD NO		TITLE RIGHT LOWER SPAR CHORD (DIRECTION 2)			BOEING CARD NO. <b>54-614-01-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) the lower spar chord at the aft engine mount bulkhead.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Inspection requires the removal of engine mount.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD (DIRECTION 2)	
		D633A109-AKS 54-614-01-02	Page 1 of 2 Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-614		
TAS	SK 54	-05-02-250-8	804					MECH	INS
1. <u>INT</u>	ERNA	L - SPECIA	L DETAILED: LO	WER SPA	AR CHORD (DI	RECTION 2)			
Α.	Insp	ection							
	-	ASK 54-05-02-010-0	22						
	(1)	Open these	e access panels fo	or Engine	No. 1:				
		Number	Name/Location	<u>n</u>					
		413	Left Fan Cowl,	Engine 1					
		414	Right Fan Cowl	-					
		415	Left Thrust Rev		•				
		416	Right Thrust Re	everser, E	ngine 1				
		Open these	e access panels fo	_	No. 2:				
		<u>Number</u>	Name/Location	_					
		423	Left Fan Cowl,	•	_				
	424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2								
	NOTE: Inspection requires the removal of engine mount.								
	<ul> <li>SUBTASK 54-05-02-250-004</li> <li>(2) Do a High Frequency Eddy Current inspection of the lower spar chord at the aft engine mount bulkhead.</li> <li>See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.</li> <li>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.</li> </ul>						oft ongino		
							an engine		
							ions.		
	(3)								
	( )	Number	Name/Location	•					
		413	Left Fan Cowl,	_					
		414	Right Fan Cowl	_	I				
		415	Left Thrust Rev	erser, Eng	gine 1				
		416	Right Thrust Re	everser, E	ngine 1				
		Close these	e access panels f	or Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>n</u>					
		423	Left Fan Cowl,	•					
		424	Right Fan Cowl	-					
		425	Left Thrust Rev		•				
	426 Right Thrust Reverser, Engine 2								
				END OF	TASK ———				
		EFFECTIVITY <b>AKS ALL</b>		SOURCE AWL	RIGHT LOWER	SPAR CHORD (DIRECT	TION 2)		





AIRLINE CARD NO		LEFT LOWER SPAR CHORD, WEB ONLY			BOEING CARD NO. <b>54-615-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) only the lower right spar web from nacelle STA 224.7 to nacelle STA 231.8.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Open/Remove Thrust Reverser as Required. Removal of Pneumatic Ducting Required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD, WEB ONLY	
		D633A109-AKS 54-615-00-01	Page 1 of 2 Oct 15/2014



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-615-00-01 MECH INSP TASK 54-05-02-250-805 INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD - WEB ONLY Α. Inspection SUBTASK 54-05-02-010-023 Open these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Open/Remove Thrust Reverser as Required. Removal of Pneumatic Ducting Required. SUBTASK 54-05-02-250-005 Do a High Frequency Eddy Current inspection of only the lower right spar web from nacelle STA 224.7 to nacelle STA 231.8. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03. SUBTASK 54-05-02-410-023 (3) Close these access panels for Engine No. 1: Number Name/Location Left Fan Cowl, Engine 1 413 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: **Number** Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 — END OF TASK —— **FFFFCTIVITY** SOURCE LEFT LOWER SPAR CHORD, WEB ONLY **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-615-00-01 Feb 15/2015





AIRLINE CARD NO		RIGHT LOWER SPAR CHORD, WEB ONLY			BOEING CARD NO. <b>54-615-00-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLICABILITY AIRPLANE EN		
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) only the lower right spar web from nacelle STA 224.7 to nacelle STA 231.8.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Open/Remove Thrust Reverser as Required. Removal of Pneumatic Ducting Required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD, WEB ONLY	
		D633A109-AKS 54-615-00-02	Page 1 of 2 Oct 15/2014



### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-615-00-02 MECH INSP TASK 54-05-02-250-805 INTERNAL - SPECIAL DETAILED: LOWER SPAR CHORD - WEB ONLY Α. Inspection SUBTASK 54-05-02-010-023 Open these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Open/Remove Thrust Reverser as Required. Removal of Pneumatic Ducting Required. SUBTASK 54-05-02-250-005 Do a High Frequency Eddy Current inspection of only the lower right spar web from nacelle STA 224.7 to nacelle STA 231.8. See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03. SUBTASK 54-05-02-410-023 (3) Close these access panels for Engine No. 1: Number Name/Location Left Fan Cowl, Engine 1 413 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: **Number** Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 — END OF TASK —— **FFFFCTIVITY** SOURCE RIGHT LOWER SPAR CHORD, WEB ONLY **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-615-00-02 Feb 15/2015





AIRLINE CARD NO		LEFT LOWER SPAR CHORD, CHORD ONLY			BOEING CARD NO. <b>54-616-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC. AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) only the lower right spar chord from nacelle STA 224.7 to nacelle STA 231.8.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Remove thrust reverser as required. Removal of insulation heat shields is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD, CHORD ONLY		
		D633A109-AKS 54-616-00-01	Page 1 of 2 Oct 15/2014	



			TAIL NUMBER		STATION	AIRLINE CARD NO.	1	CARD NO. <b>6-00-01</b>	
TA	SK 54	-05-02-250-8	306					MECH	INSF
1. <u>IN</u> 7	ERNA	L - SPECIA	L DETAILED: LO	WER SPA	AR CHORD - CH	HORD ONLY			
Α.	Inst	ection							
7 11	-	ASK 54-05-02-010-0	24						
	(1)		e access panels fo	or Engine	No. 1:				
	( )	Number	Name/Location	•					
		413	Left Fan Cowl,						
		414	Right Fan Cow	•	1				
		415	Left Thrust Rev		•				
		416	Right Thrust Re	everser, Ei	ngine 1				
		Open these	e access panels f	or Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>n</u>					
		423	Left Fan Cowl,	Engine 2					
		424	Right Fan Cow	-					
		425	Left Thrust Rev		•				
		426	Right Thrust Re		_				
			emove Thrust Rev quired.	erser as re	equired. Remov	al of insulation heat shi	elds is		
	SUBT	ASK 54-05-02-250-0	06						
	(2)		Frequency Eddy ( A 224.7 to nacelle			the lower right spar ch	ord from		
		See Doc. [	D626A001-DTR, E	TR check	form 54-51-10,	for alternative inspecti	ons.		
		the 737 No		Manual (E		t of this inspection is co inspection procedures			
	SUBT	ASK 54-05-02-410-0	24						
	(3)	Close thes	e access panels f	or Engine	No. 1:				
	, ,	Number	Name/Location	<u>n</u>					
		413	Left Fan Cowl,	- Engine 1					
		414	Right Fan Cow	-	1				
		415	Left Thrust Rev		•				
		416	Right Thrust Re	everser, Ei	ngine 1				
		Close thes	e access panels f	or Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>n</u>					
		423	Left Fan Cowl,	Engine 2					
		424	Right Fan Cow	-					
		425	Left Thrust Rev		•				
		426	Right Thrust Re	everser, Ei	ngine 2				
				END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S	SPAR CHORD, CHORD C	DNLY		
					i .				





AIRLINE CARD NO		TITLE RIGHT LOWER SPAR CHORD, CHORD ONLY			BOEING CARD NO. <b>54-616-00-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) only the lower right spar chord from nacelle STA 224.7 to nacelle STA 231.8.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Remove thrust reverser as required. Removal of insulation heat shields is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD, CHORD ONLY	
		D633A109-AKS	Page 1 of 2
POEING PROPRIE	TARY Conveig	54-616-00-02  It is Unpublished Work - See title page for details	Oct 15/2014



		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-616		
T	ΓAS	K 54-	05-02-250-	806					MECH	INS
1. II	NTE	ERNA	L - SPECIA	AL DETAILED: LO	WER SPA	AR CHORD - CI	HORD ONLY			
_	Α.	Inen	ection							
,	Α.	_	SK 54-05-02-010-0	124						
		(1)		e access panels f	or Engine	No. 1:				
		( )	Number	Name/Locatio	•					
			413	Left Fan Cowl,						
			414	Right Fan Cow	•	1				
			415	Left Thrust Rev	7	-				
			416	Right Thrust Ro	everser, E	ngine 1				
			Open thes	e access panels f	or Engine	No. 2:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			423	Left Fan Cowl,	Engine 2					
			424	Right Fan Cow	-					
			425	Left Thrust Rev	,	-				
			426	Right Thrust Ro	everser, E	ngine 2				
				emove Thrust Rev quired.	erser as r	equired. Remov	al of insulation heat shi	elds is		
		SUBTA	SK 54-05-02-250-0	006						
		(2)	_	Frequency Eddy A 224.7 to nacelle			the lower right spar che	ord from		
			See Doc. I	D626A001-DTR, [	OTR check	form 54-51-10,	, for alternative inspecti	ons.		
			the 737 No		: Manual ([		t of this inspection is co inspection procedures			
		SUBTA	SK 54-05-02-410-0	024						
		(3)	Close thes	se access panels f	for Engine	No. 1:				
			Number	Name/Locatio	n					
			413	Left Fan Cowl,	Engine 1					
			414	Right Fan Cow	-	1				
			415	Left Thrust Rev		•				
			416	Right Thrust Ro	everser, E	ngine 1				
			Close thes	se access panels f	for Engine	No. 2:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			423	Left Fan Cowl,	Engine 2					
			424	Right Fan Cow	l, Engine 2	2				
			425	Left Thrust Rev		•				
			426	Right Thrust Ro	everser, E	ngine 2				
					- END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LOWER	SPAR CHORD, CHORD	ONLY		
						D633A109-AKS 54-616-00-02	3		Page 2 eb 15/	
						54-616-00-02		F	-eb	15/





AIRLINE CARD NO		LEFT LOWER SPAR CHORD - COMPRESSION PAD			BOEING CARD NO. <b>54-617-00-01</b>		
DATE	SPECIAL DETAILED	BRAC	KET, HORIZONTA	AL LEG	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLICABILITY  AIRPLANE ENGINE		
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) the horizontal leg of the left and right hand chords, common to the compression pad bracket at nacelle STA 244.9.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Remove thrust reverser as required. Removal of insulation heat shields is required.

EFFECTIVITY AKS ALL	source AWL	LEFT LOWER SPAR CHORD - CO	OMPRESSION PAD BRACKET,
		D633A109-AKS 54-617-00-01	Page 1 of 2 Oct 15/2014
			AKS ALL AWL HORIZONTAL LEG  D633A109-AKS



## 737-600/700/800/900 **TASK CARDS**

					iAo	K CARDS				
	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>7-00-01</b>	
1.	INT	ERNA	05-02-250-8 L - SPECIA IAL LEG		OWER SPA	AR CHORD - C	OMPRESSION PAD BE	RACKET -	MECH	INSI
	1101									
	A.	Insp	ection							
			SK 54-05-02-010-02		· -	NI. 4				
		(1)	•	access panels f	_	NO. 1:				
			Number 413	Name/Location						
			413	Left Fan Cowl, Right Fan Cow	-	1				
			415	Left Thrust Re	-					
			416	Right Thrust R						
			Open these	access panels t	for Engine	No. 2:				
			Number	Name/Locatio	_					
			423	Left Fan Cowl,						
			424	Right Fan Cow	-	2				
			425	Left Thrust Re						
			426	Right Thrust R	everser, E	ngine 2				
				move thrust reve uired.	erser as red	quired. Remova	l of insulation heat shiel	ds is		
		SUBTA	SK 54-05-02-250-00	7						
		(2)	•			•	norizontal leg of the left at nacelle STA 244.9.	and right		
			See Doc. D	626A001-DTR, I	DTR check	form 54-51-10	, for alternative inspection	ons.		
			the 737 No		t Manual ([		nt of this inspection is co inspection procedures a			
		SUBTA	SK 54-05-02-410-02	25						
		(3)	Close these	e access panels	for Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>on</u>					
			413	Left Fan Cowl,	•					
			414	Right Fan Cow	-					
			415	Left Thrust Re						
			416	Right Thrust R						
				e access panels	_	No. 2:				
			Number	Name/Location						
			423	Left Fan Cowl,	•	2				
			424 425	Right Fan Cow Left Thrust Re						
			426	Right Thrust R						
						TASK ——				
			EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S	SPAR CHORD - COMPRE LEG	SSION PAD	BRACE	KET
						D633A109-AKS	;		Page 2 Feb 15/	
				POEING PROPRI		OT-017-00-01			13/	_U I





AIRLINE	CARD NO	RIGHT LOWER SPAR CHORD - COMPRESSION			BOEING CARD NO. <b>54-617-00-02</b>		
DATE	SPECIAL DETAILED	PAD BRA	ACKET, HORIZON	ITAL LEG	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLICABILITY  AIRPLANE ENGIN		
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) the horizontal leg of the left and right hand chords, common to the compression pad bracket at nacelle STA 244.9.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Remove thrust reverser as required. Removal of insulation heat shields is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD - COMPRESSIO BRACKET, HORIZONTAL LEG	N PAD
		D633A109-AKS 54-617-00-02	Page 1 of 2 Oct 15/2014



	TE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-617</b> -				
1. INTER				OWER SPA	AR CHORD - CO	OMPRESSION PAD BE	RACKET -	MECH	INS		
	Inspect		_								
		1-05-02-010-02: nen these	access panels f	or Engine	No. 1:						
`	. , .	ımber	Name/Locatio	•	110. 1.						
	41		Left Fan Cowl,								
	41	4	Right Fan Cow	•							
	41		Left Thrust Rev		•						
	41	6	Right Thrust R	everser, E	ngine 1						
	Ор	en these	access panels f	or Engine	No. 2:						
	<u>Nu</u>	<u>ımber</u>	Name/Locatio	<u>n</u>							
	42		Left Fan Cowl,	-							
	42		Right Fan Cow								
	42: 42:		Left Thrust Rev Right Thrust R		•						
			•			. 62 1. 62 1 6 . 162 . 1					
	NOTE: Remove thrust reverser as required. Removal of insulation heat shields is required.										
	SUBTASK 54-05-02-250-007										
(	(2) Do a High Frequency Eddy Current inspection of the horizontal leg of the left and right hand chords, common to the compression pad bracket at nacelle STA 244.9.										
	See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.										
	the	737 Nor		Manual (E		t of this inspection is coinspection procedures					
5	SUBTASK 54	1-05-02-410-02	5								
(	(3) Clo	se these	access panels f	for Engine	No. 1:						
	<u>Nu</u>	<u>ımber</u>	Name/Locatio	<u>n</u>							
	41		Left Fan Cowl,	•							
	41		Right Fan Cow	-							
	4.4	<b>a</b>			1						
	41:		Left Thrust Rev		•						
	41	6	Right Thrust R	everser, E	ngine 1						
	410 Clo	6 ose these	Right Thrust Raccess panels f	everser, Ei for Engine	ngine 1						
	410 Clo <u>Nu</u>	6 ose these <u>imber</u>	Right Thrust Reaccess panels for Name/Location	everser, Ei for Engine <u>n</u>	ngine 1						
	410 Clo <u>Nu</u> 423	6 ose these <u>imber</u> 3	Right Thrust R access panels to Name/Location Left Fan Cowl,	everser, Ei for Engine <u>n</u> Engine 2	ngine 1 No. 2:						
	410 Clo <b>Nu</b> 420 420	6 ose these <u>imber</u> 3 4	Right Thrust Researcess panels to Name/Location Left Fan Cowl, Right Fan Cow	everser, Ei for Engine n Engine 2 l, Engine 2	ngine 1 No. 2:						
	410 Clo <u>Nu</u> 423	6 ose these <u>imber</u> 3 4 5	Right Thrust R access panels to Name/Location Left Fan Cowl,	everser, El for Engine <u>n</u> Engine 2 Il, Engine 2 verser, Eng	ngine 1 No. 2: 2 gine 2						
	410 Clo <u>Nu</u> 420 420 420	6 ose these <u>imber</u> 3 4 5	Right Thrust Researces panels to Name/Location Left Fan Cowl, Right Fan Cow Left Thrust Res	everser, El for Engine <u>n</u> Engine 2 Il, Engine 2 verser, Eng	ngine 1 No. 2:  gine 2 ngine 2						
	410 Clo <u>Nu</u> 420 420 420	6 ose these <u>imber</u> 3 4 5	Right Thrust Researces panels to Name/Location Left Fan Cowl, Right Fan Cow Left Thrust Res	everser, El for Engine n Engine 2 I, Engine 2 verser, Engeverser,	ngine 1 No. 2: gine 2 ngine 2 TASK	SPAR CHORD - COMPRIZONTAL LEG	ESSION PAD				





AIRLINE	CARD NO			MPRESSION PAD	BOEING CARD NO. <b>54-617-01-01</b>		
DATE	SPECIAL DETAILED	BRAC	CKET, HORIZONTA	AL LEG	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL <b>AIRPL</b>				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (Ultrasonic) the horizontal leg of the left and right hand chords, common to the compression pad bracket at nacelle STA 244.9.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

ACCESS NOTE: Remove thrust reverser as required. Removal of insulation heat shield is required.

SOURCE AWL LEFT LOWER SPAR CHORD - COMPRESSION PAD BRACKET, HORIZONTAL LEG

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	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-617			
1. <u>IN</u>	ITERNA	-05-02-130-{ AL - SPECIA NAL LEG		WER SP	AR CHORD - C	OMPRESSION PAD BE	RACKET -	MECH I	INS	
Α	. Insp	ection								
	_	ASK 54-05-02-010-0	04							
	(1)	Open thes	e access panels f	or Engine	No. 1:					
		<u>Number</u>	Name/Locatio	<u>n</u>						
		413	Left Fan Cowl,	-						
		414	Right Fan Cow	_						
		415 416	Left Thrust Rev		•					
			Right Thrust Re		•					
		-	e access panels f	•	No. 2:					
		Number	Name/Locatio	_						
		423 424	Left Fan Cowl, Right Fan Cow	•	2					
		425	Left Thrust Rev	_						
		426	Right Thrust Re		•					
			emove thrust reve quired.	rser as re	quired. Remova	l of insulation heat shiel	d is			
	SUBTASK 54-05-02-130-004									
	(2)		asonic inspection pression pad brad		-	eft and right hand chord	s, common			
		See Doc. [	D626A001-DTR, [	OTR check	k form 54-51-10	, for alternative inspecti	ons.			
	SUBT	ASK 54-05-02-410-0	04							
	(3)	Close thes	e access panels f	or Enigne	No. 1:					
		<u>Number</u>	Name/Locatio							
		413	Left Fan Cowl,	-	4					
		414 415	Right Fan Cow Left Thrust Rev	_						
		415	Right Thrust Re		•					
			e access panels f							
		Number	Name/Locatio	•	110. 2.					
		423	Left Fan Cowl,							
		424	Right Fan Cow	•	2					
		425	Left Thrust Rev							
		426	Right Thrust Re	everser, E	ingine 2					
	——— END OF TASK ———									
		EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S HORIZONTAL L	SPAR CHORD - COMPRE LEG	SSION PAD I	BRACKE	=	
					D633A109-AKS 54-617-01-01	}		Page 2 o		
					1					





AIRLINE	CARD NO		SPAR CHORD -		BOEING 0 <b>54-617</b>	CARD NO. '-01-02
DATE	SPECIAL DETAILED	PAD BR	ACKET, HORIZON	ITAL LEG	RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>	

Inspect (Ultrasonic) the horizontal leg of the left and right hand chords, common to the compression pad bracket at nacelle STA 244.9.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

ACCESS NOTE: Remove thrust reverser as required. Removal of insulation heat shield is required.

AKS ALL

SOURCE AWL

RIGHT LOWER SPAR CHORD - COMPRESSION PAD BRACKET, HORIZONTAL LEG

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	ļ	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-617			
	TAS	K 54	-05-02-130-	-804				1	MECH	INS	
1.	INT	ERNA	L - SPECIA	AL DETAILED: LO	WER SP	AR CHORD - CO	OMPRESSION PAD BE	RACKET -			
	НОІ	RIZON	NAL LEG								
	A.	Insp	ection								
		_	ASK 54-05-02-010-	-004							
		(1)	Open thes	se access panels f	or Engine	No. 1:					
			<u>Number</u>	Name/Locatio	<u>n</u>						
			413	Left Fan Cowl,	_						
			414	Right Fan Cow	_						
			415 416	Left Thrust Rev		-					
				Right Thrust R		_					
			•	se access panels f	•	No. 2:					
			Number 423	Name/Location Left Fan Cowl,							
			423 424	Right Fan Cow,	•	2					
			425	Left Thrust Rev	-						
			426	Right Thrust R	everser, E	ngine 2					
	NOTE: Remove thrust reverser as required. Removal of insulation heat shield is required.										
	SUBTASK 54-05-02-130-004										
		` '		rasonic inspection	the horizo	ntal leg of the le	ft and right hand chord	s, common			
			to the con	npression pad brad	cket at nac	celle STA 244.9					
			See Doc.	D626A001-DTR, I	OTR check	k form 54-51-10,	for alternative inspection	ons.			
			ASK 54-05-02-410-								
		(3)		se access panels	•	No. 1:					
			<u>Number</u>	Name/Locatio	_						
			413	Left Fan Cowl,	_	4					
			414 415	Right Fan Cow Left Thrust Rev							
			416	Right Thrust R		_					
				se access panels		· ·					
			Number	Name/Locatio	_						
			423	Left Fan Cowl,	_						
			424	Right Fan Cow	-	2					
			425	Left Thrust Rev		-					
			426	Right Thrust R	everser, E	ngine 2					
					- END OF	TASK ———					
						Terre					
			AKS ALL		SOURCE AWL	RIGHT LOWER BRACKET, HOP	SPAR CHORD - COMPR RIZONTAL LEG	ESSION PAD	)		
						D633A109-AKS 54-617-01-02			Page 2 eb 15/		
						1					





AIRLINE	CARD NO	LEFT LOWER SPA				8-00-01
DATE	TASK SPECIAL DETAILED	BRAC	KET, VERTICAL	LEG	RELAT	ED CARD
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	ADDLI	CABILITY
	ENG/STRUT	1.1	56000 FC	18000 FC	AIRPLANE	ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 433AT 433BT 433C	CT 433DT		ZONE <b>433</b>	

Inspect (High Frequency Eddy Current) the vertical leg of the compression pad bracket on the lower left and right spar chords at nacelle STA 244.9.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

EFFECTIVITY AKS ALL

SOURCE AWL

LEFT LOWER SPAR CHORD - COMPRESSION PAD BRACKET, VERTICAL LEG

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## 737-600/700/800/900 **TASK CARDS**

	D	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 54-618		
	TAS	K 54.	-05-02-250-8						MECH	INSF
1.					OWER SPA	AR CHORD - C	OMPRESSION PAD BE	RACKET -		
	VER	TICA	L LEG							
	A.	Insp	ection							
		-	ASK 54-05-02-010-0	26						
		(1)	Open these	e access panels f	or Engine	No. 1:				
			<u>Number</u>	Name/Location						
			433AT	Strut, Forward	•					
			433BT 433CT	Strut, Forward Strut, Upper S	•					
			433DT	Strut, Upper S						
			Open these	e access panels f	or Engine	No. 2:				
			<u>Number</u>	Name/Location	<u>on</u>					
			443AT	Strut, Forward	•					
			443BT	Strut, Forward	•					
			443CT 443DT	Strut, Upper S Strut, Upper S						
		SURTA	ASK 54-05-02-250-0		pa. 1100, c					
		(2)			Current ins	spection of the	vertical leg of the compr	ession pad		
		` '	_	the lower left and		•				
			See Doc. D	0626A001-DTR, I	OTR check	form 54-51-10	, for alternative inspection	ons.		
				` '	-	•	nt of this inspection is co			
				ndestructive Test n Part 6, Subject	,	,	inspection procedures	are		
		SUBTA	ASK 54-05-02-410-0	26						
		(3)		e access panels	-	No. 1:				
			Number	Name/Location		0				
			433AT 433BT	Strut, Forward Strut, Forward	-					
			433CT	Strut, Upper S	•					
			433DT	Strut, Upper S						
			Close these	e access panels	for Engine	No. 2:				
			<u>Number</u>	Name/Location	<u>n</u>					
			443AT	Strut, Forward		•				
			443BT	Strut, Forward	•					
			443CT 443DT	Strut, Upper S Strut, Upper S						
			11021	oudi, oppor o	,					
					- END OF	IASK ——				
			EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S VERTICAL LEG	SPAR CHORD - COMPRE	SSION PAD	BRACE	KET
						D633A109-AKS 54-618-00-01	3		Page 2 eb 15/	
						1				





AIRLINE	CARD NO		R SPAR CHORD - 0		BOEING ( <b>54-618</b>	
DATE	TASK SPECIAL DETAILED	PAD B	RACKET, VERTICA	AL LEG	RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	
STATION	SKILL AIRPL				RELAT	ALL ALL
		ACCESS 443AT 443BT 44	3CT 443DT			

Inspect (High Frequency Eddy Current) the vertical leg of the compression pad bracket on the lower left and right spar chords at nacelle STA 244.9.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

AKS ALL

SOURCE AWL

RIGHT LOWER SPAR CHORD - COMPRESSION PAD BRACKET, VERTICAL LEG

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	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. 8-00-02			
١.	INT	ERNA	05-02-250-8 L - SPECIA L LEG		OWER SPA	AR CHORD - CO	OMPRESSION PAD BE	RACKET -	MECH II	NS		
	A.	-	ection									
		SUBTA (1)	SK 54-05-02-010-02	26 e access panels f	or Engine	No. 1:						
		(1)	Number	Name/Locatio	•	NO. 1.						
			433AT	Strut, Forward		Strut 1						
			433BT	Strut, Forward	•							
			433CT	Strut, Upper S								
			433DT	Strut, Upper S	par Web, S	Strut 1						
			Open these	e access panels f	or Engine	No. 2:						
			Number	Name/Locatio	<u>n</u>							
			443AT	Strut, Forward	•							
			443BT 443CT	Strut, Forward Strut, Upper S	•							
			443DT	Strut, Upper S								
	ession pad											
(2) Do a High Frequency Eddy Current inspection of the vertical leg of the compression p bracket on the lower left and right spar chords at nacelle STA 244.9.												
See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.												
	The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.											
	contained in Part 6, Subject 54-40-03.  subtask 54-05-02-410-026  (3) Close these access panels for Engine No. 1:											
			<u>Number</u>	Name/Locatio	<u>n</u>							
			433AT	Strut, Forward	•							
			433BT	Strut, Forward Strut, Upper S	•							
			433CT 433DT	Strut, Upper S								
				e access panels								
			Number	Name/Locatio	-	NO. 2.						
			443AT	Strut, Forward		Strut 2						
			443BT	Strut, Forward	•							
			443CT	Strut, Upper S	•							
			443DT	Strut, Upper S	par Web, S	Strut 2						
					- END OF	TASK ———						
			EFFECTIVITY		SOURCE	DIGUT I CHIES	CDAD CHORD COVER	TEGGOV 54				
			AKS ALL		AWL	BRACKET, VER	SPAR CHORD - COMPR RTICAL LEG	ESSION PAL	,			
						D633A109-AKS 54-618-00-02			Page 2 o eb 15/20			





			IAON OANDO	•		
AIRLINE	E CARD NO		TITLE SPAR CHORD - T			CARD NO. 9-00-01
DATE	TASK SPECIAL DETAILED	DE	ETAIL, VERTICAL I	_EG	RELATI	ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		431BL 431BR 43 433DT	31CL 431CR 433AT	433BT 433CT	ZONE <b>433</b>	
			ical leg at nacelle	STA 209.0 - 21	2.3, nacelle STA 2	31.8 - 234.4 on
	ght hand chords.		-51-10, for alterna	ative inspections		
766 DOC. DO2	20/1001-0111, 01	TY CHECK TOTTI 34	-51-10, for alterna	ilive ilispections	•	
	EFFECTIVITY AKS ALL		WL LEFT LOWING	ER SPAR CHORI LEG	O - TYPICAL FRAM	E DETAIL,
			D633A109-			Page 1 of Oct 15/201



	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-619</b> -		
	TAS	K 54-	05-02-13	30-805					MECH	INSP
1.	INT	ERNA	L - SPE	CIAL DETAILED: LO	WER SP	AR CHORD— T	YPICAL FRAME DETA	AIL,		
	VEF	RTICA	L LEG (I	DIRECTION 1)						
	A.	Insp	ection							
		SUBTA	NSK 54-05-02-	010-005						
		(1)	•	nese access panels f	•	No. 1:				
			Numbe							
			431BL 431BR 431CL 431CR 433AT 433BT 433CT 433DT	Forward Strut I Forward Strut I Forward Strut I Strut, Forward Strut, Forward	Fairing, Ri Fairing, Le Fairing, Ri Spar Web Spar Web oar Web, S	o, Strut 1 Strut 1	iring, Strut 1 ring, Strut 1			
			Open th	nese access panels f						
			Numbe	·	-					
			441BL 441CL 441CR 443AT 443BT 443CT 443DT	Forward Strut I Forward Strut I	Fairing, Ri Fairing, Le Fairing, Ri Spar Web Spar Web oar Web, S	Strut 2	iring, Strut 2 ring, Strut 2			
		SUBTA	NSK 54-05-02-							
		(2)	Do an U				vertical leg at nacelle S hand chords.	STA. 209.0 -		
			See Do	c. D626A001-DTR, [	OTR check	k form 54-51-10,	for alternative inspecti	ions.		
		SUBTA	NSK 54-05-02-							
		(3)		hese access panels f	•	No. 1:				
			Numbe 431BL 431BR 431CL 431CR 433AT 433BT 433CT 433DT	Forward Strut I Forward Strut I Forward Strut I Forward Strut I Strut, Forward Strut, Forward	Fairing, Le Fairing, Ri Fairing, Le Fairing, Ri Spar Web Spar Web	o, Strut 1 Strut 1	iring, Strut 1 ring, Strut 1			
			EFFECTIV AKS A		source <b>AWL</b>	VERTICAL LEG				
						D633A109-AKS 54-619-00-01	i 		Page 2 eb 15/	





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ <b>54-619-</b>		
C	lose these	access panels for	or Engine	No. 2:	-		MECH	INSP
	<u>lumber</u>	Name/Location	_					
	41BL		_	ft Mid Strut Fairi	na Strut 2			
	41BR			ght Mid Strut Fai				
	41CL			ft Overwing Fair				
	41CR			ght Overwing Fa				
	43AT	Strut, Forward	_	-	9, 0 0.1 _			
	43BT	Strut, Forward						
	43CT	Strut, Upper Sp						
	43DT	Strut, Upper Sp						
			END OF					
,	EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S VERTICAL LEG	PAR CHORD - TYPICAL F	FRAME DETA	AIL,	
				D633A109-AKS			Page 3	
				54-619-00-01		F	eb 15/	2015





AIRLINE CARD NO			TITLE R SPAR CHORD - T			CARD NO. <b>9-00-02</b>
DATE	TASK SPECIAL DETAILED	DE	ETAIL, VERTICAL L	.EG	RELATE	ED CARD
AIL NUMBER	WORK AREA ENG/STRUT	5TRUT 1.1 56000 FC			APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 441BL 441BR 44 443DT	11CL 441CR 443AT	443BT 443CT	ZONE 443	
e left and ri	ght hand chords.		ical leg at nacelle -51-10, for alterna			31.0 - 234.
, o 200. 201			or re, rer anerria			

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	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CAI <b>54-619-0</b>		
	TAS	K 54-	05-02-1	30-805	1				MECH	INSP
1.	INT	ERNA	L - SPE	CIAL DETAILED: LO	WER SP	AR CHORD— T	YPICAL FRAME DETA	AIL,		
	VEF	RTICA	L LEG (	DIRECTION 1)						
	A.	Insp	ection							
		SUBTA	NSK 54-05-02-	-010-005						
		(1)	•	hese access panels f	•	No. 1:				
			Numbe			6 M. I O. I E .	. 0, 1,1			
			431BL 431BR 431CL 431CR 433AT 433BT 433CT 433DT	Forward Strut   Forward Strut   Forward Strut   Strut, Forward Strut, Forward	Fairing, Ri Fairing, Le Fairing, Ri Spar Web Spar Web oar Web, S	o, Strut 1 Strut 1	iring, Strut 1 ring, Strut 1			
			Open th	hese access panels f	or Engine	No. 2:				
	Numk			er <u>Name/Locatio</u>	<u>n</u>					
			441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT	Forward Strut   Forward Strut   Forward Strut   Strut, Forward Strut, Forward Strut, Upper S	Fairing, Ri Fairing, Le Fairing, Ri Spar Web Spar Web oar Web, S	Strut 2 Strut 2	iring, Strut 2 ring, Strut 2			
		SURTA	ASK 54-05-02-		pa. 1100, t	5.1. G. E				
		(2)	Do an l				vertical leg at nacelle S hand chords.	STA. 209.0 -		
			See Do	oc. D626A001-DTR, [	OTR check	k form 54-51-10,	for alternative inspecti	ons.		
		SUBTA	NSK 54-05-02-							
		(3)		hese access panels	•	No. 1:				
	431BR Forward Strut F 431CL Forward Strut F				Fairing, Le Fairing, Ri Fairing, Le Fairing, Ri Spar Web Spar Web	Strut 1 Strut 1	iring, Strut 1 ring, Strut 1			
			EFFECTIV AKS A		source <b>AWL</b>	VERTICAL LEG				
						D633A109-AKS 54-619-00-02			age 2 b 15/2	





DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO 54-619-00-02	
Clos	se these access panels	s for Enaine	No. 2:	<u> </u>	MECH	INSP
	nber <u>Name/Locati</u>	_				
441			ft Mid Strut Fairi	na Strut 2		
441			ght Mid Strut Fai			
441			ft Overwing Fair			
			ght Overwing Fa			
443		_	-	ming, ou at 2		
443						
	CT Strut, Upper S					
443						
		— END OF				
		LND OI	IAOR			
EFF <b>A</b> K	ECTIVITY (S ALL	SOURCE AWL	RIGHT LOWER VERTICAL LEG	SPAR CHORD - TYPICAL	FRAME DETAIL,	1
			D633A109-AKS		Page	3 of 3
			54-619-00-02		Feb 15	
		-1	1			





AIRLINE	CARD NO	LEFT LOWER SPAR CHORD - TYPICAL FRAME			BOEING CARD NO. <b>54-619-01-01</b>			
DATE	SPECIAL DETAILED	_ DE	TAIL, VERTICAL L	.EG	RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>		APPLICABILITY		
STATION	STATION SKILL AIRPL				AIRPLANE ALL	ALL		
		ACCESS 413 414 415 416 NOTE	431DL 431DR		ZONE <b>433</b>			

Inspect (Low Frequency Eddy Current) the external side of the vertical leg at nacelle STA 209.0 - 212.3, and nacelle STA 231.8 - 234.4 on the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Remove thrust reversers as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD - TYPICAL FRAME DETAIL, VERTICAL LEG		
		D633A109-AKS Page 1 of 3		
		54-619-01-01 Oct 15/2014		



### 737-600/700/800/900 TASK CARDS

DATE		TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-619-01-01</b>		
TASK 5	4-05-02-250-8	309	1		MECH		
			WER SPAR CHORD -	TYPICAL FRAME DETA	JL,		
	AL LEG (DIR						
A. In	spection						
	BTASK 54-05-02-010-0	27					
(1	Open these	e access panels fo	r Engine No. 1:				
	<u>Number</u>	Name/Location					
	413	Left Fan Cowl, E	•				
	414	Right Fan Cowl,					
	415 416	Left Thrust Reve Right Thrust Re	. •				
	431DL	-	airing, Left Underwing	Fairing, Strut 1			
	431DR		airing, Right Underwin	_			
	Open these	e access panels fo	r Engine No. 2:				
	<u>Number</u>	Name/Location					
	423	Left Fan Cowl, E	_				
	424	Right Fan Cowl,	•				
	425 426	Left Thrust Reve Right Thrust Re	. •				
	441DL	•	airing, Left Underwing	Fairing, Strut 2			
	441DR		airing, Right Underwin	•			
	NOTE: Re	emove thrust revers	ers as required.				
SU	BTASK 54-05-02-250-0	09					
(2				e external side of the vertion 234.4 on the left and right			
	See Doc. [	D626A001-DTR, D	TR check form 54-51-	10, for alternative inspection	ons.		
	the 737 No	ondestructive Test I	Manual (D6-37239). Th	ent of this inspection is co ne inspection procedures			
	contained i	in Part 6, Subject 5	4-40-03.				
	BTASK 54-05-02-410-0		. E C No. 4				
(3		e access panels fo	•				
	<u>Number</u> 413	Name/Location Left Fan Cowl, E					
	413	Right Fan Cowl,	•				
	415	Left Thrust Reve	•				
	416	Right Thrust Reverser, Engine 1					
	431DL	Forward Strut Fairing, Left Underwing Fairing, Strut 1 Forward Strut Fairing, Right Underwing Fairing, Strut 1					
	431DR	Forward Strut Fa	airing, Right Onderwin	y Fairing, Strut 1			
	EFFECTIVITY AKS ALL		SOURCE LEFT LOWEI VERTICAL LI	R SPAR CHORD - TYPICAL EG	FRAME DETAIL,		

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD <b>54-619-01-</b>	
Close the	ese access panels for	r Engine No. 2:	1	ME	CH IN
Number 423 424 425 426 441DL 441DR	Name/Location Left Fan Cowl, E Right Fan Cowl, Left Thrust Reve Right Thrust Rev Forward Strut Fa	ngine 2 Engine 2 rser, Engine 2			
		END OF TASK ——			
EFFECTIVIT AKS ALI		SOURCE LEFT LOWE VERTICAL I	ER SPAR CHORD - TYPICAL LEG	FRAME DETAIL,	





AIRLINE	CARD NO	RIGHT LOWER SPAR CHORD - TYPICAL FRAME			BOEING CARD NO. <b>54-619-01-02</b>			
DATE	TASK SPECIAL DETAILED	- DE	ETAIL, VERTICAL L	_EG	RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLICA AIRPLANE	APPLICABILITY		
STATION	SKILL AIRPL				ALL	ALL ALL		
		ACCESS 423 424 425 426 NOTE	441DL 441DR		ZONE <b>443</b>			

Inspect (Low Frequency Eddy Current) the external side of the vertical leg at nacelle STA 209.0 - 212.3, and nacelle STA 231.8 - 234.4 on the left and right hand chords.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

ACCESS NOTE: Remove thrust reversers as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD - TYPICAL FRAME DETAIL, VERTICAL LEG		
		D633A109-AKS 54-619-01-02	Page 1 of 3 Oct 15/2014	
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#### 737-600/700/800/900 TASK CARDS

DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-619-</b>		
EXTERNA			WER SP	AR CHORD - T	YPICAL FRAME DETA	IL,	MECH	IN
	ection							
-	SK 54-05-02-010-0	27						
		n these access panels for Engine No. 1:						
` '	Number	Name/Location	•	1.				
	413	Left Fan Cowl, E						
	414	Right Fan Cowl,	•					
	415	Left Thrust Reve	-					
	416	Right Thrust Rev		•				
	431DL	Forward Strut Fa						
	431DR	Forward Strut Fa			airing, Strut 1			
	•	e access panels fo	-	No. 2:				
	Number	Name/Location	'					
	423	Left Fan Cowl, E	-					
	424 425	Right Fan Cowl, Left Thrust Reve	-					
	426	Right Thrust Rev	_					
	441DL	Forward Strut Fa		-	niring, Strut 2			
	441DR	Forward Strut Fa	-	_	_			
	NOTE: Re	move thrust revers	sers as re	quired.				
SUBTAS	SK 54-05-02-250-0	09						
, ,					xternal side of the vertion 34.4 on the left and right			
	See Doc. D	626A001-DTR, DTR check form 54-51-10, for alternative inspections.						
	the 737 No		Manual (E		t of this inspection is co inspection procedures a			
SUBTAS	SK 54-05-02-410-0	27						
(3)	Close these	e access panels fo	r Engine	No. 1:				
	<u>Number</u>	Name/Location	ı					
	413	Left Fan Cowl, E	ingine 1					
	414	Right Fan Cowl,	-					
	415	Left Thrust Reverser, Engine 1						
	416 431DL	Right Thrust Reverser, Engine 1 Forward Strut Fairing, Left Underwing Fairing, Strut 1						
	431DR	Forward Strut Fa						
				,	g, c			
	EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LOWER VERTICAL LEG	SPAR CHORD - TYPICA	L FRAME DE	ΓAIL,	

54-619-01-02

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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-619</b> -		
	e these access panels for these access panels for the ber Name/Location  Left Fan Cowl, Right Fan Cowl Left Thrust Rev Right Thrust Rev Right Thrust Rev Forward Strut For	n Engine 2 I, Engine 2 verser, Engeverser, Er Fairing, Lei Fairing, Rig	No. 2: 2 gine 2	iring, Strut 2			INSP
	STIVITY S <b>ALL</b>	SOURCE AWL	RIGHT LOWER VERTICAL LEG D633A109-AKS 54-619-01-02		ı	TAIL, Page 3 oct 15/	





AIRLINE CARD NO		TITLE  LEFT LOWER SPAR CHORD - TYPICAL FRAME			BOEING CARD NO. <b>54-620-00-01</b>		
DATE	SPECIAL DETAILED	DET	AIL, HORIZONTAL	LEG	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) the horizontal leg on the left and right chords from nacelle STA 209.0 to nacelle STA 212.3 and from nacelle STA 231.8 to nacelle STA 234.4.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Remove thrust reversers as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LOWER SPAR CHORD - TYPICAL FRAME DETAIL, HORIZONTAL LEG	
			1 of 2
		54-620-00-01 Oct 15	5/2014



	ı	DATE		TAIL NUMBER STATION AIRLINE CARD NO.		AIRLINE CARD NO.	BOEING CARD N <b>54-620-00-0</b>					
	TAS	K 54-	-05-02-250-8	310					MECH	INSI		
1.	INT	ERNA	L - SPECIA	L DETAILED: LC	WER SPA	AR CHORD - TY	PICAL FRAME DETAI	<u>L,</u>				
	HOI	RIZON	NTAL LEG									
	A.	Insp	ection									
		SUBTA	ASK 54-05-02-010-0	28								
		(1)										
			<u>Number</u>	Name/Locatio								
			413	Left Fan Cowl,	-	4						
			414 415	Right Fan Cow Left Thrust Rev	-							
			416	Right Thrust R	7	-						
			Open these	e access panels f								
			Number	Name/Locatio	-							
			423	Left Fan Cowl,	Engine 2							
			424	Right Fan Cow	-							
			425	Left Thrust Rev		•						
			426	Right Thrust R								
			NOTE: Re	emove thrust reve	rsers as re	equired.						
		suвтаsк 54-05-02-250-010 (2) Do a high Frequency Eddy Current inspection of the horizontal leg on the left and right										
		(2)	•	n nacelle STA 20		•	and from nacelle STA 23	•				
			See Doc. D	0626A001-DTR, [	OTR check	c form 54-51-10,	for alternative inspection	ons.				
			the 737 No		Manual ([	D6-37239). The	t of this inspection is co inspection procedures a					
		SUBTA	ASK 54-05-02-410-0	28								
		(3)	Close thes	e access panels f	for Engine	No. 1:						
			<u>Number</u>	Name/Locatio	<u>n</u>							
			413	Left Fan Cowl,	•							
			414 415	Right Fan Cow Left Thrust Rev	-							
			416	Right Thrust R		•						
				e access panels f		· ·						
			Number	Name/Locatio	•	. 10. 2.						
			423	Left Fan Cowl,								
			424	Right Fan Cow	•	2						
			425	Left Thrust Rev		•						
			426	Right Thrust R	everser, E	ngine 2						
					- END OF	TASK ——						
			EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S HORIZONTAL L	SPAR CHORD - TYPICAL EG	FRAME DETA	AIL,			
						D633A109-AKS 54-620-00-01			Page 2 eb 15/			





AIRLINE CARD NO		RIGHT LOWER SPAR CHORD - TYPICAL FRAME			BOEING CARD NO. <b>54-620-00-02</b>		
DATE	SPECIAL DETAILED	DETA	AIL, HORIZONTAL	_ LEG	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) the horizontal leg on the left and right chords from nacelle STA 209.0 to nacelle STA 212.3 and from nacelle STA 231.8 to nacelle STA 234.4.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Remove thrust reversers as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LOWER SPAR CHORD - TYPICAL FRA HORIZONTAL LEG	AME DETAIL,
		D633A109-AKS 54-620-00-02	Page 1 of 2 Oct 15/2014



		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-620			
	TAS	K 54-	-05-02-250-8	310					MECH	INS	
1.	INT	ERNA	L - SPECIA	L DETAILED: LO	WER SPA	AR CHORD - TY	PICAL FRAME DETAI	L,			
	HOI	RIZON	NTAL LEG								
	A.	Insp	ection								
		SUBTA	ASK 54-05-02-010-0	28							
		(1) Open these access panels for Engine No. 1:									
			<u>Number</u>	Name/Locatio							
			413	Left Fan Cowl,	-	4					
			414 415	Right Fan Cow Left Thrust Rev	-						
			416	Right Thrust R	7	-					
			Open these	e access panels f							
			Number	Name/Locatio	-						
			423	Left Fan Cowl,	Engine 2						
			424	Right Fan Cow	-						
			425	Left Thrust Rev		•					
			426	Right Thrust R							
				emove thrust reve	rsers as re	equirea.					
		SUBTASK 54-05-02-250-010									
		(2)	Do a high Frequency Eddy Current inspection of the horizontal leg on the left and right chords from nacelle STA 209.0 to nacelle STA 212.3 and from nacelle STA 231.8 to nacelle STA 234.4.								
			See Doc. D	0626A001-DTR, [	OTR check	c form 54-51-10,	for alternative inspection	ons.			
			the 737 No		Manual ([	D6-37239). The	t of this inspection is co inspection procedures a				
		SUBTA	ASK 54-05-02-410-0	28							
		(3)	Close thes	e access panels t	for Engine	No. 1:					
			<u>Number</u>	Name/Locatio	<u>n</u>						
			413	Left Fan Cowl,	•						
			414 415	Right Fan Cow Left Thrust Rev	-						
			416	Right Thrust R		•					
				e access panels		· ·					
			Number	Name/Locatio	•	. 10. 2.					
			423	Left Fan Cowl,							
			424	Right Fan Cow	•	2					
			425	Left Thrust Rev		•					
			426	Right Thrust R	everser, E	ngine 2					
					- END OF	TASK ——					
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LOWER HORIZONTAL L	SPAR CHORD - TYPICA EG	L FRAME DE	TAIL,		
						D633A109-AKS 54-620-00-02			Page 2 eb 15/		





AIRLINE CARD NO		TITLE  LEFT LOWER SPAR CHORD			BOEING CARD NO. <b>54-621-00-01</b>		
DATE	TASK <b>DETAILED</b>				RELAT	ED CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLI: AIRPLANE	CABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>		

Inspect (	Detailed	) the lower s	spar chord	aft of the aft	engine n	nount bulkhead.

See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

ACCESS NOTE: Remove thrust reversers as required. Removal of insulation heat shields required.

SOURCE AWL LEFT LOWER SPAR CHORD

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	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-621		
	TAS	K 54-	05-02-211-	801					MECH	INS
<b>1.</b> ]	INTI	ERNA	L - DETAIL	ED: LOWER SPA	R CHORI	D				
	Α.									
		-	ection ASK 54-05-02-010-	014						
		(1)	Open thes	se access panels fo	or Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>n</u>					
			413	Left Fan Cowl,	-					
			414	Right Fan Cowl	-					
			415 416	Left Thrust Rev Right Thrust Re		-				
				se access panels fo		_				
			Number	Name/Location	_	NO. 2.				
			423	Left Fan Cowl,	_					
			424	Right Fan Cowl	_	2				
			425	Left Thrust Rev	-					
			426	Right Thrust Re	everser, E	ngine 2				
				emove thrust rever equired.	sers as re	equired. Remova	al of insulation heat shi	elds		
	SUBTASK 54-05-02-211-001									
	(2) Do a Detailed inspection of the lower spar chord aft of the aft engine mount bulkhea									
	See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.									
	SUBTASK 54-05-02-410-014									
	(3) Close these access panels for Engine No. 1:									
			<u>Number</u>	Name/Location						
			413	Left Fan Cowl,	-	4				
			414 415	Right Fan Cowl Left Thrust Rev	-					
			416	Right Thrust Re		•				
			Close the	se access panels fo		_				
			Number	Name/Locatio	_					
			423	Left Fan Cowl,	_					
			424	Right Fan Cowl						
			425	Left Thrust Rev		-				
			426	Right Thrust Re	everser, E	ngine 2				
					END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LOWER S	PAR CHORD			
						D633A109-AKS 54-621-00-01			Page 2 eb 15/	
						34-021-00-01		F	eb '	10/





AIRLINE	CARD NO	RIGHT	TITLE  LOWER SPAR C	BOEING CARD NO. <b>54-621-00-02</b>			
DATE	TASK DETAILED				RELATE	ED CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC		APPLICABILITY  AIRPLANE ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>		

Inspect (Detailed)	) the lower spar chord	d aft of the aft engine mount	bulkhead.
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See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.

**ACCESS NOTE:** Remove thrust reversers as required. Removal of insulation heat shields required.

RIGHT LOWER SPAR CHORD

AKS ALL

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	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-621			
	TAS	SK 54-	-05-02-211·	-801					MECH	INS	
1.	INT	ERNA	L - DETAI	LED: LOWER SPA	R CHOR	<u>D</u>					
	A.	Insp	ection								
		-	ASK 54-05-02-010	-014							
		(1)	Open the	se access panels fo	or Engine	No. 1:					
			<u>Number</u>	Name/Location	<u>n</u>						
			413	Left Fan Cowl,	-						
			414	Right Fan Cow	-						
			415 416	Left Thrust Rev Right Thrust Re		-					
				se access panels fo							
			Number	Name/Location	_	NO. 2.					
			423	Left Fan Cowl,	_						
			424	Right Fan Cow	_	2					
			425	Left Thrust Rev	-						
			426	Right Thrust Re	everser, E	ngine 2					
				demove thrust rever equired.	rsers as re	equired. Remova	al of insulation heat shi	elds			
	SUBTASK 54-05-02-211-001										
	(2) Do a Detailed inspection of the lower spar chord aft of the aft engine mount bulkhead.										
	See Doc. D626A001-DTR, DTR check form 54-51-10, for alternative inspections.										
	SUBTASK 54-05-02-410-014										
	(3) Close these access panels for Engine No. 1:										
			<u>Number</u>	Name/Location							
			413 414	Left Fan Cowl, Right Fan Cow	-	1					
			415	Left Thrust Rev	-						
			416	Right Thrust Re		•					
			Close the	se access panels f	or Engine	No. 2:					
			Number	Name/Location	_						
			423	Left Fan Cowl,	Engine 2						
			424	Right Fan Cow							
			425	Left Thrust Rev		-					
			426	Right Thrust Re	everser, E	ngine 2					
					END OF	TASK ———					
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LOWER	SPAR CHORD		<u> </u>		
						D633A109-AKS 54-621-00-02			Page 2 eb 15/		
						1					





AIRLINE	CARD NO	LEFT FORWAR	TITLE RD ENGINE MOU	BOEING CARD NO. <b>54-622-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>36000 FC</b>		APPLICABILITY AIRPLANE ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 413 414 415 416 NOTE			ZONE <b>433</b>	

Inspect (High Frequency Eddy Current) the end pad bolt holes (4 locations) at the FWD engine mount bulkhead. See Doc. D626A001-DTR, DTR check form 54-51-11, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Bathtub fitting bolts must be sequentially (one at a time) removed for bolt hole eddy current inspection. Retorque tension bolts per dwg requirements.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FORWARD ENGINE MOUNT BULKHEAD	
		D633A109-AKS 54-622-00-01	Page 1 of 2 Oct 15/2014



	D/	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>2-00-01</b>					
Т	ASI	K 54-	05-02-2	50-811					MECH	INSF				
				CIAL DETAILED: FO	DRWARD I	ENGINE MOUN	IT BULKHEAD							
_	Δ	A. Inspection												
•		-	SK 54-05-02-	010-029										
		(1)		nese access panels f	or Engine	No. 1:								
	. Nur			•	•									
	_		413	Left Fan Cowl,										
			414	Right Fan Cow	l, Engine 1	1								
			415	Left Thrust Re										
			416	Right Thrust R	everser, E	ngine 1								
	Oper			nese access panels f	for Engine	No. 2:								
			Numbe		_									
			423	Left Fan Cowl,	•	_								
			424	Right Fan Cow	-									
			425 426	Left Thrust Re Right Thrust R										
				•		•		L . 10 L . L .						
	NOTE			_			e at a time) removed for ts per dwg requirement							
		SUBTA	SK 54-05-02-	250-011										
		(2)		igh Frequency Eddy D engine mount bulk		spection of the e	end pad bolt holes (4 Lo	ocations) at						
			See Do	c. D626A001-DTR, I	DTR check	form 54-51-11,	for alternative inspection	ons.						
			the 737	` ,	t Manual ([	•	nt of this inspection is co inspection procedures							
		SUBTA	SK 54-05-02-	410-029										
		(3)	Close tl	hese access panels	for Engine	No. 1:								
			Numbe	er Name/Location	<u>on</u>									
			413	Left Fan Cowl,	Engine 1									
			414	Right Fan Cow	-									
			415	Left Thrust Re										
			416	Right Thrust R		•								
				hese access panels	-	No. 2:								
			Numbe											
			423	Left Fan Cowl,	•	_								
			424	Right Fan Cow	-									
			425 426	Left Thrust Re										
			420	26 Right Thrust Reverser, Engine 2  ——— END OF TASK ———										
			EFFECTIV		SOURCE AWL	LEFT FORWAR	D ENGINE MOUNT BULI	KHEAD						
						D633A109-AKS	<b>;</b>	J	Page 2 Feb 15/					





AIRLINE	CARD NO	RIGHT FORWAR	RD ENGINE MOU	BOEING CARD NO. <b>54-622-00-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC		APPLICABILITY  AIRPLANE ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426 NOTE			ZONE <b>443</b>	

Inspect (High Frequency Eddy Current) the end pad bolt holes (4 locations) at the FWD engine mount bulkhead. See Doc. D626A001-DTR, DTR check form 54-51-11, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-03.

**ACCESS NOTE:** Bathtub fitting bolts must be sequentially (one at a time) removed for bolt hole eddy current inspection. Retorque tension bolts per dwg requirements.

	T		
EFFECTIVITY  AKS ALL	AWL	RIGHT FORWARD ENGINE MOUNT BULKHEAD	
		D633A109-AKS 54-622-00-02	Page 1 of 2 Oct 15/2014



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING ( 54-622		
SK 54	-05-02-250-	-811					MECH	INSF
			RWARD I	ENGINE MOUN	T BULKHEAD			
		020						
			or Engine	No 1·				
(')	•	·	•	110. 1.				
	414		•	I				
	415	_	_					
	416	Right Thrust Re	everser, Eı	ngine 1				
	Open thes	se access panels f	or Engine	No. 2:				
	<u>Number</u>	Name/Locatio	<u>n</u>					
	423	Left Fan Cowl,	Engine 2					
	424	-	_					
	425			•				
	426	Right Thrust Re	everser, Ei	ngine 2				
		_			•			
SUBT	ASK 54-05-02-250	-011						
(2)				spection of the e	end pad bolt holes (4 Lo	ocations) at		
	See Doc.	D626A001-DTR, E	OTR check	form 54-51-11,	for alternative inspection	ons.		
	the 737 N	ondestructive Test	Manual (E	•	•			
SUBT	ASK 54-05-02-410	-029						
(3)	Close the	se access panels f	or Engine	No. 1:				
	<u>Number</u>	Name/Locatio	<u>n</u>					
	413	Left Fan Cowl,	Engine 1					
	414	_	_					
	415			•				
	416	Right Thrust Re	everser, Eı	ngine 1				
	Close the	se access panels f	for Engine	No. 2:				
	<u>Number</u>	Name/Locatio	<u>n</u>					
	423		•					
		-	_					
				•				
	420	Right Thrust Reverser, Engine 2						
			- END OF	TASK ———				
			SOURCE AWL	RIGHT FORWA	RD ENGINE MOUNT BU	LKHEAD		
				1				
	Insp SUBT. (1)	Inspection SUBTASK 54-05-02-010 (1) Open these Number 413 414 415 416 Open these Number 423 424 425 426 NOTE: B e SUBTASK 54-05-02-250 (2) Do a High the FWD See Doc. The NDI in the 737 N contained SUBTASK 54-05-02-410 (3) Close the Number 413 414 415 416 Close the Number 423 424 425 426	Inspection  SUBTASK 54-05-02-010-029  (1) Open these access panels for Number Name/Location  413 Left Fan Cowl, 414 Right Fan Cowl, 415 Left Thrust Rev. 416 Right Thrust Rev. 416 Right Fan Cowl, 423 Left Fan Cowl, 424 Right Fan Cowl, 425 Left Thrust Rev. 426 Right Thrust Rev. 426 Right Thrust Rev. 427 Left Thrust Rev. 428 Right Thrust Rev. 429 Right Thrust Rev. 420 Right Thrust Rev. 421 Right Fan Cowl, 422 Left Thrust Rev. 423 Left Fan Cowl, 424 Right Fan Cowl, 425 Left Thrust Rev. 426 Right Thrust Rev. 427 Left Thrust Rev. 428 Right Fan Cowl, 429 Left Fan Cowl, 429 Close these access panels for Number Name/Location 413 Left Fan Cowl, 414 Right Fan Cowl, 415 Left Thrust Rev. 416 Right Thrust Rev. 416 Right Thrust Rev. 417 Name/Location 418 Left Fan Cowl, 419 Left Fan Cowl, 419 Left Fan Cowl, 410 Right Fan Cowl, 4110 Right Fan Cowl, 4120 Left Fan Cowl, 413 Left Fan Cowl, 414 Right Fan Cowl, 415 Left Thrust Rev. 416 Right Thrust Rev. 417 Left Fan Cowl, 418 Left Fan Cowl, 419 Left Fan Cowl, 419 Left Fan Cowl, 410 Right Fan Cowl, 410 Right Fan Cowl, 4110 Right Fan Cowl, 4121 Right Fan Cowl, 4122 Left Thrust Rev. 413 Left Fan Cowl, 414 Right Fan Cowl, 415 Left Thrust Rev. 416 Right Fan Cowl, 417 Right Fan Cowl, 418 Right Fan Cowl, 419 Right Fan Cowl, 419 Right Fan Cowl, 420 Right Fan Cowl, 421 Right Fan Cowl, 4221 Left Thrust Rev. 423 Left Fan Cowl, 424 Right Fan Cowl, 425 Left Thrust Rev. 426 Right Thrust Rev. 427 Left Thrust Rev. 428 Right Thrust Rev. 429 Right Thrust Rev.	Inspection  SUBTASK 54-05-02-010-029  (1) Open these access panels for Engine Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  421 Left Fan Cowl, Engine 2  422 Right Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  427 Left Thrust Reverser, Engine 2  428 Right Thrust Reverser, Engine 2  429 Right Thrust Reverser, Engine 2  420 Right Thrust Reverser, Engine 2  421 Right Fan Cowl, Engine 2  422 Left Thrust Reverser, Engine 2  423 Left Thrust Reverser, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 3  SUBTASK 54-05-02-250-011  (2) Do a High Frequency Eddy Current institute FWD engine mount bulkhead.  See Doc. D626A001-DTR, DTR check The NDI method(s) necessary to accordine 737 Nondestructive Test Manual (Econtained in Part 6, Subject 54-40-03.)  SUBTASK 54-05-02-410-029  (3) Close these access panels for Engine Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 2  427 Right Fan Cowl, Engine 2  428 Left Thrust Reverser, Engine 2  429 Right Fan Cowl, Engine 2  420 Right Fan Cowl, Engine 2  421 Right Fan Cowl, Engine 2  422 Right Fan Cowl, Engine 2  423 Left Thrust Reverser, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  427 Right Fan Cowl, Engine 2  428 Right Thrust Reverser, Engine 2  429 Right Thrust Reverser, Engine 2  420 Right Thrust Reverser, Engine 2	Inspection  Subtrask 54-05-02-010-029  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  NOTE: Bathtub fitting bolts must be sequentially (one eddy current inspection. Retorque tension bol subtrask 54-05-02-250-011  (2) Do a High Frequency Eddy Current inspection of the edthe FWD engine mount bulkhead.  See Doc. D626A001-DTR, DTR check form 54-51-11, The NDI method(s) necessary to accomplish the intenthe 737 Nondestructive Test Manual (D6-37239). The contained in Part 6, Subject 54-40-03.  Subtrask 54-05-02-410-029  (3) Close these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  Actif Fan Cowl, Engine 2  Actif Fan	Inspection  SUBTASK \$449-92-010-029  (1) Open these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  416 Right Thrust Reverser, Engine 1  Open these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Right Thrust Reverser, Engine 2  NOTE: Bathtub fitting bolts must be sequentially (one at a time) removed for eddy current inspection. Retorque tension bolts per dwg requirement subtask \$449-92-290-911  (2) Do a High Frequency Eddy Current inspection of the end pad bolt holes (4 Lot the FWD engine mount bulkhead.  See Doc. D626A001-DTR, DTR check form 54-51-11, for alternative inspection for the 737 Nondestructive Test Manual (D6-37239). The inspection procedures contained in Part 6, Subject 54-40-03.  SUBTASK \$449-92-410-929  (3) Close these access panels for Engine No. 1:  Number Name/Location  413 Left Fan Cowl, Engine 1  414 Right Fan Cowl, Engine 1  415 Left Thrust Reverser, Engine 1  Close these access panels for Engine No. 2:  Number Name/Location  423 Left Fan Cowl, Engine 2  424 Right Fan Cowl, Engine 2  425 Left Thrust Reverser, Engine 2  426 Right Thrust Reverser, Engine 2  427 Right Fan Cowl, Engine 2  428 Right Fan Cowl, Engine 2  429 Left Thrust Reverser, Engine 2  420 Right Thrust Reverser, Engine 2  END OF TASK	SK 54-05-02-250-811  FERNAL - SPECIAL DETAILED: FORWARD ENGINE MOUNT BULKHEAD  Inspection  SUBTASK 54-05-02-019-029  (1) Open these access panels for Engine No. 1:  Number Name/Location  413	Inspection  SUBTACK 54-65-62-410-629  (1) Open these access panels for Engine No. 1:  Number   Name/Location   413





AIRLINE CARD NO		LEFT	MID STRUT BULK	BOEING CARD NO. <b>54-623-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 413 414 433AT 4 NOTE	33BT 433CT 433D	Т	ZONE 433	

Inspect (High Frequency Eddy Current) the FWD and Aft flanges of both tension fittings common to the R1 fitting (4) attachment bolts.

See Doc. D626A001-DTR, DTR check form 54-51-14, for alternative inspections.

ACCESS NOTE: Internal access required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MID STRUT BULKHEAD	
		D633A109-AKS 54-623-00-01	Page 1 of 3 Oct 15/2014



[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C <b>54-623</b>					
TAS	K 54-	05-02-2	50-812					MECH	INSF			
			CIAL DETAILED: M	ID STRUT	BULKHEAD							
A. Inspection												
Α.	-	ASK 54-05-02	040.020									
	(1)		hese access panels f	for Eniane	No 1							
	(')	Numbe	•	J	140. 1.							
		413	Left Fan Cowl,									
		414	Right Fan Cow	•	1							
		433AT	Strut, Forward	Spar Web	, Strut 1							
		433BT	,									
		433CT 433DT	, 11									
			, , , ,									
		•	hese access panels f	•	INO. ∠:							
		Number 423										
		423 424	Left Fan Cowl, Right Fan Cow	-	)							
		443AT	•									
		443BT	Strut, Forward	Spar Web	, Strut 2							
		443CT		•								
		443DT	Strut, Upper S	par Web, S	Strut 2							
		NOTE:	Internal access requ	uired.								
	SUBTA	NSK 54-05-02										
	(2)		ligh Frequency Eddy common to the R1 fit		•	FWD and Aft flanges of bo	oth tension					
		See Do	oc. D626A001-DTR, I	DTR check	form 54-51-14,	for alternative inspection	ıs.					
	SUBTA	NSK 54-05-02										
	(3)		these access panels	_	No. 1:							
		Numbe										
		413	Left Fan Cowl,	U								
		414 433AT	Right Fan Cow Strut, Forward	-								
		433BT										
		433CT										
		433DT	• • • • • • • • • • • • • • • • • • • •	•								
		Close t	these access panels	for Engine	No. 2:							
		Numbe	er <u>Name/Locatio</u>	<u>on</u>								
		423	Left Fan Cowl,	Engine 2								
		424	Right Fan Cow	_								
		443AT	,									
		443BT 443CT	•									
		EFFECTI	VITY	SOURCE	LEFT MID STRU	IT BIII KHEAD						
		AKS A		AWL	LEFT WID STRU	JI BULKHEAD						
					D633A109-AKS 54-623-00-01			Page 2 eb 15/				



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-623		
(0	Continued)	'	,			MECH	INSP
<u>Nu</u>	mber Name/Location						
443	3DT Strut, Upper S	par Web, S	Strut 2				
		- END OF	TASK ——				
EF	FECTIVITY	SOURCE	LEFT MID STRU	T BULKHEAD			
A	KS ALL	AWL					
			D633A109-AKS			Page 3 eb 15/	of 3
1			54-623-00-01		-	GD 12/	<b>∠</b> ∪15





AIRLINE	CARD NO	RIGHT	TITLE F MID STRUT BUL		BOEING CARD NO. <b>54-623-00-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 423 424 443AT 4 NOTE	43BT 443CT 443D	Т	ZONE 443		

Inspect (High Frequency Eddy Current) the FWD and Aft flanges of both tension fittings common to the R1 fitting (4) attachment bolts.

See Doc. D626A001-DTR, DTR check form 54-51-14, for alternative inspections.

ACCESS NOTE: Internal access required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MID STRUT BULKHEAD	
		D633A109-AKS 54-623-00-02	Page 1 of 3 Oct 15/2014



[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-623		
TAS	K 54	-05-02-2	50-812					MECH	INSF
			CIAL DETAILED: M	D STRUT	BULKHEAD				
Α.		ection							
A.	_	ASK 54-05-02	040.020						
	(1)		:hese access panels f	or Enjane	No 1·				
	(')	Numbe	•	•	110. 1.				
		413	Left Fan Cowl,						
		414	Right Fan Cow	-	I				
		433AT	,	•					
		433BT	,	•					
		433CT 433DT	, 11						
			, , , ,	•					
		Number 1	these access panels f	_	NO. 2.				
		423	er Name/Location  Left Fan Cowl,						
		424	Right Fan Cow	-					
		443AT	•						
		443BT	Strut, Forward	Spar Web	, Strut 2				
		443CT		•					
		443DT	, 11		Strut 2				
		NOTE:	Internal access requ	uired.					
		ASK 54-05-02		0		-1A/D   A G G   G			
	(2)	fittings	common to the R1 fit	tting (4) att	achment bolts.	FWD and Aft flanges of bo			
		See Do	oc. D626A001-DTR, I	DTR check	form 54-51-14,	for alternative inspection	S.		
		ASK 54-05-02		. – .					
	(3)		these access panels	_	No. 1:				
		Numbe	_						
		413 414	Left Fan Cowl,	O	1				
		414 433AT	Right Fan Cow Strut, Forward	-					
		433BT		•					
		433CT	Strut, Upper S	par Web, S	Strut 1				
		433DT	Strut, Upper S	par Web, S	Strut 1				
		Close t	these access panels	for Engine	No. 2:				
		Numbe	er <u>Name/Locatio</u>	<u>on</u>					
		423	Left Fan Cowl,	•					
		424	Right Fan Cow	_					
		443AT 443BT	•	•					
		443CT		•					
		EFFECTI		SOURCE AWL	RIGHT MID STR	RUT BULKHEAD			
			- <del></del>	AVIL	D633A109-AKS		ı	Page 2	2 of∶
					54-623-00-02			eb 15/	



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-623		
(Con	tinued)	'	'			MECH	INSP
Numb	er Name/Location						
443DT	Strut, Upper S	par Web, S	Strut 2				
		- END OF	TASK ———				
EFFECT <b>AKS</b>	IVITY Ali	SOURCE	RIGHT MID STR	UT BULKHEAD			
ANS	7 <b>LL</b>	AWL					
			D633A109-AKS 54-623-00-02			Page 3 eb 15/2	of 3
		1				:0//	





	7	ACCESS 413 414 415 416 NOTE	431DL 431DR		ZONE <b>433</b>	
STATION	SKILL AIRPL				ALL	ALL
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD <b>56000 FC</b>	REPEAT <b>18000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE
DATE	SPECIAL DETAILED				RELATE	D CARD
AIRLINE	CARD NO	LEF	T SIDE SKIN CUT	BOEING CARD NO. <b>54-624-00-01</b>		

Inspect (High Frequency Eddy Current) all exposed side skin surfaces within 4.5 inches of cutout at nacelle STA 252 and nacelle STA 270 left and right hand sides.

See Doc. D626A001-DTR, DTR check form 54-51-16, for alternative inspections.

ACCESS NOTE: Remove thrust reversers as required.



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-624-		
TAS	K 54	05-02-250-	 813					MECH	INS
			AL DETAILED: SII	DE SKIN (	ситоитѕ				
Α.		ection							
Α.	_	SK 54-05-02-010-0	134						
	(1)		e access panels fo	or Engine	No. 1:				
	(.,	Number	Name/Location	_					
		413	Left Fan Cowl,						
		414	Right Fan Cow	•	I				
		415	Left Thrust Rev						
		416	Right Thrust Re		•				
		431DL		_	ft Underwing Fa	•			
		431DR	Forward Strut F	-airing, Ri	ght Underwing F	airing, Strut 1			
		•	e access panels for	_	No. 2:				
		<u>Number</u>	Name/Location	_					
		423	Left Fan Cowl,	•					
		424	Right Fan Cow	-					
		425 426	Left Thrust Rev Right Thrust Re		•				
		441DL	•		ft Underwing Fa	iring, Strut 2			
		441DR		_	ght Underwing F	_			
		NOTE: Re	emove thrust reve	rsers as re	equired.				
	SUBTA	SK 54-05-02-250-0	013						
	(2)	_			•	rposed side skin surfact 70 left and right hand s			
		See Doc. I	D626A001-DTR, D	TR check	form 54-51-16,	for alternative inspecti	ions.		
	SUBTA	SK 54-05-02-410-0	031						
	(3)	Close thes	se access panels f	or Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>n</u>					
		413	Left Fan Cowl,	Engine 1					
		414	Right Fan Cow	-					
		415	Left Thrust Rev		•				
		416 431DL	Right Thrust Re		ngine 1 ft Underwing Fa	iring Strut 1			
		431DR		•	ght Underwing Fa	•			
			se access panels f		-	aming, on at 1			
		Number	Name/Location	-					
		423	Left Fan Cowl,	_					
		424	Right Fan Cow	•	2				
		425	Left Thrust Rev	erser, Eng	gine 2				
		426	Right Thrust Re		•				
		441DL	Forward Strut F	airing, Le	ft Underwing Fa	iring, Strut 2			
		EFFECTIVITY AKS ALL		SOURCE AWL	LEFT SIDE SKII	N CUTOUTS			
									2 of



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-624-</b>		
(Cont	inued)					MECH	INSP
<u>Numbe</u>		<u>1</u>					
441DR	Forward Strut F	airing, Ri	ght Underwing F	airing, Strut 2			
		END OF	TASK ———				
EFFECTI <b>AKS A</b>	VITY <b>LL</b>	SOURCE AWL	LEFT SIDE SKIN	CUTOUTS			
			B000 1105 5155		_		
			D633A109-AKS 54-624-00-01		F Fe	age 3 b 15/2	of 3 2015





AIRLINE	CARD NO	RIGH	TITLE IT SIDE SKIN CUT	54-624		
DATE	SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426 NOTE	441DL 441DR		ZONE <b>443</b>	

Inspect (High Frequency Eddy Current) all exposed side skin surfaces within 4.5 inches of cutout at nacelle STA 252 and nacelle STA 270 left and right hand sides.

See Doc. D626A001-DTR, DTR check form 54-51-16, for alternative inspections.

ACCESS NOTE: Remove thrust reversers as required.



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ 54-624-		
TAS	SK 54	-05-02-25						MECH	INS
			CIAL DETAILED: SI	DE SKIN (	ситоитѕ				
Α.		ection							
Λ.	-	ASK 54-05-02-0	110.031						
	(1)		iese access panels f	or Engine	No 1·				
	(.)	Numbe	•	_					
		413	Left Fan Cowl,						
		414	Right Fan Cow	•	1				
		415	Left Thrust Re	verser, Eng	gine 1				
		416	Right Thrust R		•				
		431DL 431DR		_	ft Underwing Fa	•			
					ght Underwing F	-airing, Strut 1			
		•	ese access panels f	_	No. 2:				
		Numbe							
		423 424	Left Fan Cowl, Right Fan Cow	-	)				
		424	Left Thrust Re	-					
		426	Right Thrust R						
		441DL	•		ft Underwing Fa	niring, Strut 2			
		441DR	Forward Strut	Fairing, Ri	ght Underwing F	Fairing, Strut 2			
		NOTE:	Remove thrust reve	rsers as re	equired.				
	SUBT	ASK 54-05-02-2	250-013						
	(2)				•	xposed side skin surfac 270 left and right hand s			
		See Do	c. D626A001-DTR, I	OTR check	form 54-51-16,	for alternative inspection	ons.		
	SUBT	ASK 54-05-02-4	110-031						
	(3)	Close th	nese access panels	for Engine	No. 1:				
		Numbe	<u>r</u> <u>Name/Locatio</u>	<u>n</u>					
		413	Left Fan Cowl,	0					
		414	Right Fan Cow	-					
		415 416	Left Thrust Rev Right Thrust R						
		431DL	•		ft Underwing Fa	niring, Strut 1			
		431DR		•	ght Underwing F	•			
		Close th	nese access panels	for Engine	No. 2:	_			
		Numbe	•	•					
		423	Left Fan Cowl,						
		424	Right Fan Cow	l, Engine 2	2				
		425	Left Thrust Re						
		426	Right Thrust R		-	:::::			
		441DL	Forward Strut	⊢aırıng, Le	ft Underwing Fa	niring, Strut 2			
		EFFECTIV AKS AI		SOURCE AWL	RIGHT SIDE SK	IN CUTOUTS			
				_	D633A109-AKS 54-624-00-02			Page 2 eb 15/	
				1	I .				



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-624		
(Con	tinued)	'				MECH	INSP
<u>Numb</u>		<u>1</u>					
441DF		airing, Ri	ght Underwing F	airing, Strut 2			
		END OF	TASK ———				
EFFECT	rivity	SOURCE	RIGHT SIDE SK	IN CUTOUTS			
AKS	ALL	AWL					
			D633A109-AKS		1	Page 3	of 3
			54-624-00-02		F	eb 15/	2015





AIRLINE	E CARD NO	I FF	TITLE TT STRUT SIDE S	SKIN		CARD NO. 5-00-01
DATE	TASK SPECIAL DETAILED		TOTAL OIDE	Sixiiv		D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>36000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 413 414 415 416 4	31DL 431DR		ZONE 433	
ight sides.					and nacelle STA 24	2.7, left and
See Doc. D6	26A001-DTR, D1	TR check form 54-5	51-16, for alterna	ative inspection	S.	
	EFFECTIVITY AKS ALL	SOUF AW		UT SIDE SKIN		
			D633A109-			Page 1 o



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-625-		
	TAS	K 54-	05-02-130	)-806					MECH	INS
1.	INT	ERNA	L - SPEC	IAL DETAILED: ST	RUT SIDE	SKIN				
	Α.	Iner	ection							
	Λ.	-	ASK 54-05-02-01	0-006						
		(1)		ese access panels fo	or Engine	No. 1:				
		( - )	Number	•	•					
			413	Left Fan Cowl,	_					
			414	Right Fan Cow	_	1				
			415	Left Thrust Rev						
			416	Right Thrust Re		•	0			
			431DL 431DR		_	ft Underwing Fa	•			
						ght Underwing F	alling, Strut 1			
			•	ese access panels fo	•	No. 2:				
			Number	•						
			423 424	Left Fan Cowl, Right Fan Cow	•	)				
			425	Left Thrust Rev						
			426	Right Thrust Re		•				
			441DL		-	ft Underwing Fa	_			
			441DR	Forward Strut F	Fairing, Ri	ght Underwing F	Fairing, Strut 2			
		SUBTA	NSK 54-05-02-13							
		(2)		trasonic inspection elle STA 242.7, left a			celle STA 222.6, left ar	nd right sides		
			See Doc	. D626A001-DTR, D	OTR check	form 54-51-16,	for alternative inspect	ions.		
		SUBTA	NSK 54-05-02-41	0-006						
		(3)	Close the	ese access panels f	or Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>n</u>					
			413	Left Fan Cowl,	-					
			414	Right Fan Cow	_					
			415 416	Left Thrust Rev Right Thrust Re		•				
			431DL	•		ft Underwing Fa	iring Strut 1			
			431DR			ght Underwing F				
			Close the	ese access panels f	or Engine	No. 2:				
			Number	•	_					
			423	Left Fan Cowl,	_					
			424	Right Fan Cow	I, Engine 2					
			425	Left Thrust Rev						
			426	Right Thrust Re		-	inima Otra t O			
			441DL 441DR			ft Underwing Fa ght Underwing F				
			44 IDK	i orwaru Strut F	anny, M	gin Onderwing F	anny, Strut Z			
			EFFECTIVIT		SOURCE	LEFT STRUT SI	DE SKIN			
			AKS AL	L	AWL					
						D633A109-AKS 54-625-00-01			Page 2 eb 15/	
						J . J_J JJ-01		' '	J. 101	



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	CARD NO. 5-00-01	
		- END OF	TASK ——		MECH	INSP
			i i i i i i i i i i i i i i i i i i i			
EFFECT <b>AKS</b>	TIVITY <b>ALL</b>	SOURCE AWL	LEFT STRUT SI	DE SKIN		
		<b>-</b>				
			D633A109-AKS 54-625-00-01		Page 3 Feb 15/2	of 3 2015





AIRLINE	E CARD NO	DI	TITLE GHT STRUT SIDE S	SKIN		CARD NO. 5-00-02
DATE	TASK SPECIAL DETAILED	_ Ni	GHI STRUT SIDE	<b>STAIN</b>		ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>36000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426	6 441DL 441DR		ZONE 443	
spect (Ultra	sonic) the strut s	ide skin at nacel	le STA 222.6, left	and right sides a	and nacelle STA 24	12.7, left and
ght sides.	064004 DTD DT	TD about form Ed	E1 16 for alterna	ativo inonoctions		
e Doc. D62	26A001-DTR, DT	R check form 54	l-51-16, for alterna	itive inspections	5.	
	EFFECTIVITY AKS ALL		PURCE RIGHT STR	EUT SIDE SKIN		
	EFFECTIVITY AKS ALL		RIGHT STR			Page 1

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Oct 15/2014



	I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-625-		
	TAS	K 54-	-05-02-130	)-806					MECH	INS
1.	INT	ERNA	L - SPEC	IAL DETAILED: ST	RUT SIDE	SKIN				
	Α.	Iner	ection							
	Λ.	-	ASK 54-05-02-01	0-006						
		(1)		ese access panels fo	or Engine	No. 1:				
		( - )	Number	•	•					
			413	Left Fan Cowl,	_					
			414	Right Fan Cow	_	1				
			415	Left Thrust Rev						
			416	Right Thrust Re		•	0			
			431DL 431DR		_	ft Underwing Fa ght Underwing F	•			
							alling, Strut 1			
			•	ese access panels fo	•	No. 2:				
			Number	•						
			423 424	Left Fan Cowl, Right Fan Cow	•	)				
			425	Left Thrust Rev						
			426	Right Thrust Re		•				
			441DL		-	ft Underwing Fa	_			
			441DR	Forward Strut F	Fairing, Ri	ght Underwing F	Fairing, Strut 2			
		SUBTA	ASK 54-05-02-13	0-006						
		(2)		trasonic inspection elle STA 242.7, left a			celle STA 222.6, left ar	nd right sides		
			See Doc	. D626A001-DTR, D	OTR check	form 54-51-16,	for alternative inspect	ions.		
		SUBTA	ASK 54-05-02-41	0-006						
		(3)	Close the	ese access panels f	or Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>n</u>					
			413	Left Fan Cowl,	-					
			414	Right Fan Cow	_					
			415 416	Left Thrust Rev Right Thrust Re		•				
			431DL	•		ft Underwing Fa	iring Strut 1			
			431DR			ght Underwing F				
			Close the	ese access panels f	or Engine	No. 2:				
			Number	•	_					
			423	Left Fan Cowl,	_					
			424	Right Fan Cow	_	2				
			425	Left Thrust Rev						
			426	Right Thrust Re		-	ining Others C			
			441DL 441DR			ft Underwing Fa ght Underwing F				
			44 IDK	i orwaru Strut F	anny, M	gin Onderwing F	anny, Strut Z			
			EFFECTIVIT		SOURCE	RIGHT STRUT	SIDE SKIN			
			AKS AL	L	AWL					
						D633A109-AKS 54-625-00-02			Page 2 eb 15/	
						U-T U-U-U-UL			J. 101	



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.		CARD NO. <b>5-00-02</b>	
		- END OF	TASK ——		·	MECH	INSP
EFFEC <b>AKS</b>	TIVITY <b>ALL</b>	SOURCE AWL	RIGHT STRUT S	IDE SKIN			
			D0004405 1155				
			D633A109-AKS 54-625-00-02			Page 3 Feb 15/2	of 3 2015





AIRLINE	CARD NO	LEF"	TITLE T UPPER SPAR CH	HORD	BOEING CARD NO. <b>54-626-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL <b>AIRPL</b>				ALL	ALL	
		ACCESS 431BL 431BR 43 433DT	31CL 431CR 433AT	433BT 433CT	ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) the upper spar chord between the forward and aft engine mounts: Nacelle STA 200.9 - 211.5 on the left chords, nacelle STA 213.6 - 225.2 on the left chords, nacelle STA 236.1 - 233.6 on the left chords, nacelle STA 236.3 - 241.8 on the left and right chords.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

EFFECTIVITY
AKS ALL

SOURCE
AWL

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54-626-00-01

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	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C <b>54-626</b>		
TAS	SK 54-	-05-02-250-8	14					MECH	INSF
I. <u>INT</u>	ERNA	L - SPECIAI	L DETAILED: UF	PPER SPA	R CHORD				
Α.	Inst	ection							
		ASK 54-05-02-010-00	ıq.						
	(1)		access panels f	or Engine	No. 1:				
	( · )	Number	Name/Locatio	_					
		431BL			ft Mid Strut Fair	ing Strut 1			
		431BR		_	ght Mid Strut Fa	•			
		431CL		•	ft Overwing Fai	•			
		431CR		_	ght Overwing Fa	airing, Strut 1			
		433AT	Strut, Forward						
		433BT	Strut, Forward						
		433CT 433DT	Strut, Upper S						
			Strut, Upper S						
		•	access panels f	-	No. 2:				
		Number	Name/Locatio		ft Mid Otmut Fair	ing Church O			
		441BL 441BR		-	ft Mid Strut Fair ght Mid Strut Fa	_			
		441CL		_	ft Overwing Fai	_			
		441CR		-	ght Overwing Fa	_			
		443AT	Strut, Forward	_	-	<b>3</b> , 2 2 2 2			
		443BT	Strut, Forward	Spar Web	, Strut 2				
		443CT	Strut, Upper S						
		443DT	Strut, Upper S	par Web, S	Strut 2				
		ASK 54-05-02-250-01							
	(2)	-			•	upper spar chord betwee			
						11.5 on the left chords, 1233.6 on the left chords,			
			- 241.8 on the lef			.55.0 On the left chords,	Hacene		
				_		, for alternative inspection	ons.		
		The NDI me	ethod(s) necessa	ary to acco	mplish the inten	t of this inspection is co	ntained in		
				,	,	inspection procedures a	are		
		contained ir	n Part 6, Subject	54–40-07					
		ASK 54-05-02-410-00			N. 0				
	(3)		e access panels	-	No. 2:				
		<u>Number</u>	Name/Locatio						
		431BL			ft Mid Strut Fair				
		431BR 431CL		_	ght Mid Strut Fa ft Overwing Fai	_			
		431CR		•	ght Overwing Fa	•			
		433AT	Strut, Forward	_	-	annig, oa ac i			
		433BT	Strut, Forward						
		EFFECTIVITY		SOURCE	LEFT UPPER S	PAR CHORD			
		AKS ALL		AWL					
					D633A109-AKS 54-626-00-01	;		Page 2 un 15/	
					1 - 1 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 -				



DATE		TAIL NUMBER	S	TATION	AIRLINE CARD NO.	BOEING C 54-626		
	(Continued	d)		I			MECH	INSP
	<u>lumber</u>	Name/Location						
_	33CT	Strut, Upper Spa		rut 1				
4	33DT	Strut, Upper Spa						
C	Close these	access panels for	r Engine N	lo. 2:				
<u>N</u>	<u>lumber</u>	Name/Location						
4	41BL	Forward Strut Fa	airing, Left	Mid Strut Fairi	ng, Strut 2			
4	41BR	Forward Strut Fa						
	41CL	Forward Strut Fa	-	-	_			
	41CR	Forward Strut Fa			iring, Strut 2			
	43AT	Strut, Forward S						
	43BT 43CT	Strut, Forward S Strut, Upper Spa						
	43DT	Strut, Upper Spa						
		—— t	END OF TA	ASK ———				
			J					
	AKS ALL		SOURCE L	EFT UPPER SI	PAR CHORD			
				D633A109-AKS			Page 3	
			5	54-626-00-01		J	un 15/	2015





CARD NO	RIGH	TITLE IT UPPER SPAR C	BOEING CARD NO. <b>54-626-00-02</b>		
SPECIAL DETAILED				RELATE	D CARD
WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLICA	ABILITY ENGINE
SKILL AIRPL				ALL	ALL
	ACCESS 441BL 441BR 44 443DT	1CL 441CR 443AT	443BT 443CT	ZONE <b>443</b>	
	SPECIAL DETAILED  WORK AREA ENG/STRUT  SKILL	TASK SPECIAL DETAILED  WORK AREA ENG/STRUT SKILL AIRPL  ACCESS 441BL 441BR 44	TASK SPECIAL DETAILED  WORK AREA ENG/STRUT SKILL AIRPL  ACCESS 441BL 441BR 441CL 441CR 443AT	RIGHT UPPER SPAR CHORD	TASK   SPECIAL   DETAILED   WORK AREA   ENG/STRUT   SKILL   AIRPL   ACCESS   441BL 441BR 441CL 441CR 443AT 443BT 443CT   AFFICIAL   SALE   ACCESS   ACCESS   A441BL 441BR 441CL 441CR 443AT 443BT 443CT   ACCESS   A443BT A43CT   ACCESS   A443BT A443CT   ACCESS   A443BT A

Inspect (High Frequency Eddy Current) the upper spar chord between the forward and aft engine mounts: Nacelle STA 200.9 - 211.5 on the left chords, nacelle STA 213.6 - 225.2 on the left chords, nacelle STA 226.1 - 233.6 on the left chords, nacelle STA 236.3 - 241.8 on the left and right chords.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

EFFECTIVITY
AKS ALL

SOURCE
AWL

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54-626-00-02

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	l	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C <b>54-626</b>		
			05-02-250						MECH	INSP
1.	INT	ERNA	L - SPECIA	AL DETAILED: UP	PER SPA	R CHORD				
	A.	Insp	ection							
			SK 54-05-02-010							
		(1)	•	se access panels fo	_	No. 1:				
			Number	Name/Location	_					
			431BL	Forward Strut F	-		•			
			431BR 431CL	Forward Strut F Forward Strut F	_	-	_			
			431CR	Forward Strut F	•	•	•			
			433AT	Strut, Forward	_	-	0,			
			433BT	Strut, Forward	•					
			433CT	Strut, Upper Sp						
			433DT	Strut, Upper Sp	ŕ					
			•	se access panels fo	•	No. 2:				
			<u>Number</u>	Name/Location	_					
			441BL	Forward Strut F	•		•			
			441BR 441CL	Forward Strut F Forward Strut F	•	•	•			
			441CR	Forward Strut F	-	_	_			
			443AT	Strut, Forward	_	-	g, • =			
			443BT	Strut, Forward	Spar Web	, Strut 2				
			443CT	Strut, Upper Sp						
			443DT	Strut, Upper Sp	ar Web, S	Strut 2				
			SK 54-05-02-250							
		(2)	forward at 213.6 - 22	nd aft engine moun	ts: Nacell ds, nacel	e STA 200.9 - 2 le STA 226.1 - 2	upper spar chord between 11.5 on the left chords, 133.6 on the left chords,	nacelle STA		
					•		, for alternative inspection	ons.		
							it of this inspection is co			
			the 737 N	. ,	Manual ([	06-37239). The	inspection procedures a			
		SUBTA	SK 54-05-02-410	-009						
		(3)		se access panels for	•	No. 2:				
			<u>Number</u>	Name/Location						
			431BL	Forward Strut F	-		•			
			431BR 431CL	Forward Strut F	_	_	•			
			431CR	Forward Strut F Forward Strut F	-	•	•			
			433AT	Strut, Forward	_	-	annig, ou at 1			
			433BT	Strut, Forward	•					
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER	SPAR CHORD			
			<del></del> -			D633A109-AKS 54-626-00-02	i.		Page 2 lun 15/	



DATE		TAIL NUMBER	ST	TATION	AIRLINE CARD NO.	BOEING 0 <b>54-626</b>		
	(Continue	ed)				,	MECH	INSP
	Number	Name/Location						
	433CT	Strut, Upper Spa	r Web Stri	ut 1				
	433DT	Strut, Upper Spa						
		se access panels for						
	Number	Name/Location	Ligillo i	J. Z.				
	441BL	Forward Strut Fa	airina Left I	Mid Strut Fairi	na Strut 2			
	441BR	Forward Strut Fa						
	441CL	Forward Strut Fa						
	441CR	Forward Strut Fa						
	443AT	Strut, Forward S						
	443BT	Strut, Forward S						
	443CT	Strut, Upper Spa						
	443DT	Strut, Upper Spa						
			END OF TA	ASK ———				
	EFFECTIVITY			IGHT UPPER	SPAR CHORD			
	AKS ALL		AWL		<del></del>			
							_	
				633A109-AKS			Page 3	
				4-626-00-02	o titlo nago for dotaile	•	lun 15/	∠∪15





AIRLINE	CARD NO	LEFT UPPER	LEFT UPPER SPAR CHORD NEAR CUTOUTS			ARD NO. -00-01	
DATE	SPECIAL DETAILED		(DIRECTION 1)	RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLIC/	CABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 431BL 431BR 43 433DT	31CL 431CR 433AT	433BT 433CT	ZONE <b>433</b>		

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AWL LEFT UPPER SPAR CHORD NEAR CUTOUTS (DIRECTION 1)

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	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-627				
TAS	K 54-05-0	<b>2-250-</b> 81	15					MECH	INSF		
INTE	ERNAL - S	PECIAL	DETAILED: UF	PER SPA	R CHORD NEA	AR CUTOUTS (DIREC	CTION 1)				
				-		,	<u> </u>				
Α.	Inspection										
	SUBTASK 54-0				NI. 4						
	. , .		access panels f	_	No. 1:						
		<u>mber</u>	Name/Location								
	431		Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 Forward Strut Fairing, Left Overheing Fairing, Strut 1								
	431										
	431 431			Forward Strut Fairing, Left Overwing Fairing, Strut 1 Forward Strut Fairing, Right Overwing Fairing, Strut 1							
	433				-	alling, Ottut i					
	433			Strut, Forward Spar Web, Strut 1 Strut, Forward Spar Web, Strut 1							
	433		Strut, Upper Sp								
	433	DT	Strut, Upper Sp								
	Оре	en these	access panels f	or Engine	No. 2:						
	<u>Nur</u>	<u>mber</u>	Name/Locatio	<u>n</u>							
	441			-	ft Mid Strut Fair	•					
	441				ght Mid Strut Fa	•					
	441			-	ft Overwing Fair	-					
	441 443				ght Overwing Fa	airing, Strut 2					
	443		Strut, Forward Strut, Forward	•							
	443		Strut, Upper Sp	•							
	443		Strut, Upper Sp								
	SUBTASK 54-05-02-250-015										
	Nac	celle STA	200.9 - nacelle	STA 211.5	on the right ch	upper spar chord near ord, nacelle STA 213. elle STA 233.6 on the	6 - nacelle				
			_			for alternative inspec	_				
						t of this inspection is					
	the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.										
	SUBTASK 54-0	05-02-410-032	!								
	` '		access panels f	•	No. 1:						
	<u>Nur</u>	Number Name/Location									
	431			-	ft Mid Strut Fair	-					
	431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1										
	431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1										
	431			_	-	airing, Strut 1					
	433 433		Strut, Forward								
	433										
		ECTIVITY		SOURCE AWL	LEFT UPPER S	PAR CHORD NEAR CU	JTOUTS (DIREC	CTION 1	l)		
					D633A109-AKS 54-627-00-01			Page 2 un 15/2			





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-627-</b>		
(0	Continued)						MECH	INSP
-		Name/Location						
		Strut, Upper Spa		Strut 1				
		access panels fo						
		Name/Location	_	140. 2.				
		Forward Strut Fa		ft Mid Strut Fairi	na Strut 2			
		Forward Strut Fa						
		Forward Strut Fa						
		Forward Strut Fa						
443		Strut, Forward S						
443	3BT	Strut, Forward S	par Web,	Strut 2				
		Strut, Upper Spa						
443	3DT	Strut, Upper Spa	ar Web, S	Strut 2				
		<del></del> I	END OF	TASK ———				
EF	FECTIVITY		SOURCE	I FFT UPPER SI	PAR CHORD NEAR CUTO	JTS (DIREC	TION	1)
Al	KS ALL		AWL		CHOILD HEAR COTO	(511/20		''
				B0001105 :::=		_		
				D633A109-AKS 54-627-00-01			age 3 un 15/2	
				J7-021-00-01		JU	411 13/	2013





AIRLINE CARD NO		RIGHT UPPER	TITLE R SPAR CHORD NE	BOEING CARD NO. <b>54-627-00-02</b>			
DATE	SPECIAL DETAILED		(DIRECTION 1)		RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLICABILITY AIRPLANE ENGINE		
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 441BL 441BR 44 443DT	11CL 441CR 443AT	443BT 443CT	ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AWL RIGHT UPPER SPAR CHORD NEAR CUTOUTS (DIRECTION 1)

D633A109-AKS Page 1 of 3
54-627-00-02 Jun 15/2015



DATE		TAIL NUMBER STATION AIRLINE		AIRLINE CARD NO.				
SK 54	-05-02-250-81	15					MECH	INS
ΓERNA	L - SPECIAL	DETAILED: UP	PER SPA	R CHORD NEA	AR CUTOUTS (DIREC	TION 1)		
					,	<del></del>		
-		_						
				NI- 4.				
(1)	•	•	_	NO. 1:				
		-	-					
			_		•			
			•		•			
			-	_	_			
					annig, Strut i			
			•					
	433CT		•					
	433DT							
	Open these	access panels fo	r Engine	No. 2:				
	<u>Number</u>	Name/Location	<u>1</u>					
	441BL	Forward Strut F	airing, Le	ft Mid Strut Fairi	ing, Strut 2			
	441BR							
	441CL		_	_	=			
			•		airing, Strut 2			
			•					
SUBTA								
			Current ins	spection of the u	upper spar chord near	the cutouts:		
( )								
	STA 225.2 o	on the right chord	, nacelle S	STA 226.1 - nac	elle STA 233.6 on the	right chord.		
	See Doc. De	626A001-DTR, D	TR check	form 54-51-17,	for alternative inspect	ions.		
		` '	•	•	•			
			•	06-37239). The	inspection procedures	are		
		-	04-40-07.					
			or Engine	No. 1:				
(3)		•	-	INU. I.				
				6 M. J. O				
			_		_			
			_	•	•			
			•		annig, Calac i			
	433BT		•					
	433CT		•					
	EFFECTIVITY		SOURCE	RIGHT UPPER	SPAR CHORD NEAR C	JTOUTS (DIRE	CTION	 N 1
	VICTI							
	AKS ALL		AWL					•
	Insp subtr (1)	SK 54-05-02-250-87  FERNAL - SPECIAL  Inspection  SUBTASK 54-05-02-010-032  (1) Open these  Number  431BL  431BR  431CL  431CR  433AT  433BT  433DT  Open these  Number  441BL  441BR  441CL  441CR  443AT  443BT  443DT  SUBTASK 54-05-02-250-019  (2) Do a High F  Nacelle STA  STA 225.2 co  See Doc. Do  The NDI me the 737 Nor contained in  SUBTASK 54-05-02-410-032  (3) Close these  Number  431BL  431BR  431CL  431CR  433AT  433BT	Inspection  SUBTASK 54-05-02-010-032  (1) Open these access panels for Number Name/Location  431BL Forward Strut F 431CL Forward Strut F 433CR Forward Strut F 433BT Strut, Forward S 433BT Strut, Upper Sp 433DT Strut, Upper Sp Open these access panels for Number Name/Location  441BL Forward Strut F 441CL Forward Strut F 441CL Forward Strut F 441CL Forward Strut F 443AT Strut, Forward S 443BT Strut, Upper Sp Subtask 54-05-02-250-015  (2) Do a High Frequency Eddy O Nacelle STA 200.9 - nacelle S STA 225.2 on the right chord See Doc. D626A001-DTR, D The NDI method(s) necessar the 737 Nondestructive Test contained in Part 6, Subject S SUBTASK 54-05-02-410-032  (3) Close these access panels for Number Name/Location 431BL Forward Strut F 431CL Forward Strut F 431CL Forward Strut F 431CR Forward Strut F 433AT Strut, Forward S Strut, Forward Strut F 431CR Forward Strut F 433AT Strut, Forward S Strut, Forward Strut F 500 SUBTASK STRUT F 500 SU	Inspection  SUBTASK 54-05-02-010-032  (1) Open these access panels for Engine  Number  A31BL  A31BL  A31CL  A31CL  A31CR  A33AT  A33CT  A33DT  A33DT  A41BL  A41BR  A33DT  A41BL  A41BR  A33DT  A43BC  A44BC  A44BC  A44BC  A44BC  A44BC  A44CC  A44CC	Inspection  Subtask 54-05-02-250-815  TERNAL - SPECIAL DETAILED: UPPER SPAR CHORD NEA  Inspection  Subtask 54-05-02-010-032  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fair 431CL Forward Strut Fairing, Left Overwing Fair 431CR Forward Strut Fairing, Right Overwing Fair 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433BT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fair 441CL Forward Strut Fairing, Left Overwing Fair 441CL Forward Strut Fairing, Right Mid Strut Fair 441CR Forward Strut Fairing, Right Overwing Fair 443AT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 30 Subtask 54-05-02-250-015  (2) Do a High Frequency Eddy Current inspection of the Unacelle STA 200.9 - nacelle STA 211.5 on the right che STA 225.2 on the right chord, nacelle STA 226.1 - nac See Doc. D626A001-DTR, DTR check form 54-51-17, The NDI method(s) necessary to accomplish the intent the 737 Nondestructive Test Manual (D6-37239). The contained in Part 6, Subject 54-40-07.  SUBTASK 54-05-02-410-032  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fair 431BR Forward Strut Fairing, Left Mid Strut Fair 431BR Forward Strut Fairing, Right Overwing Fair 431CR Forward Strut Fairing, Right Overwing Fair 431CR Forward Strut Fairing, Right Overwing Fair 433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1	SK 54-05-02-250-815  IERNAL - SPECIAL DETAILED: UPPER SPAR CHORD NEAR CUTOUTS (DIRECT Inspection  SUBTASK 54-05-02-010-032  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433BT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441BR Forward Strut Fairing, Right Wist Strut Fairing, Strut 2  441BR Forward Strut Fairing, Right Wist Strut Fairing, Strut 2  443AT Strut, Forward Spar Web, Strut 2  443AT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  43BT Strut, Upper Spar Web, Strut 2  35BTA S405-02-250-015  (2) Do a High Frequency Eddy Current inspection of the upper spar chord near Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 233.6 on the See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspect The NDI method(s) necessary to accomplish the intent of this inspection is c the 737 Nondestructive Test Manual (D6-37239). The inspection procedures contained in Part 6, Subject 54-40-07.  SUBTASK 54-05-02-2410-032  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  431BR Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1  431BR Forward Strut Fairing, Left Overwing Fairing, Strut 1  431BR Forward Strut Fairing, Left Overwing Fairing, Strut 1  431	SK 54-05-02-250-815  **TERNAL - SPECIAL DETAILED: UPPER SPAR CHORD NEAR CUTOUTS (DIRECTION 1)  **Inspection**    Inspection**   Summark 54-05-02-01-02-02-02-02-02-02-02-02-02-02-02-02-02-	SK 54-05-02-250-815  **ERNAL - SPECIAL DETAILED: UPPER SPAR CHORD NEAR CUTOUTS (DIRECTION 1)  Inspection  **SUBTASK 54-05-2010-823**  (1) Open these access panels for Engine No. 1:  **Number**  **A31BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  **431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  **431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  **433CT Forward Strut Fairing, Right Overwing Fairing, Strut 1  **433BT Strut, Forward Spar Web, Strut 1  **433BT Strut, Upper Spar Web, Strut 1  **533DT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  **Number**  **Name/Location**  **441BL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  **441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  **441CL Forward Strut Fairing, Right Wid Strut Fairing, Strut 2  **441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  **443CT Strut, Forward Spar Web, Strut 2  **443CT Strut, Upper Spar Web, Strut 2  **50CTAST Strut, Upper Spar Web, Strut 1  **50CTAST Strut, Upper Spar Strut, Forward Spar Web, S





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CAI <b>54-627-</b> 0		
(Cor	ntinued)				MECH	INSP
Numl		n				
433D		– oar Web, Strut 1				
	these access panels f					
Numl		_				
441B		 Fairing, Left Mid Strut F	airing, Strut 2			
441B		Fairing, Right Mid Strut				
441C		Fairing, Left Overwing	•			
441C		airing, Right Overwing				
443A		Spar Web, Strut 2				
443B	T Strut, Forward	Spar Web, Strut 2				
443C		oar Web, Strut 2				
443D	T Strut, Upper Sp	oar Web, Strut 2				
		END OF TASK ——	_			
EFFEC AKS		SOURCE RIGHT UPPI	ER SPAR CHORD NEAR C	UTOUTS (DIREC	CTION	l 1)
AKS	ALL	AWL				
		D633A109-A	KS	P	age 3	of 3
		54-627-00-02			n 15/2	





AIRLINE	CARD NO	LEFT UPPER SPAR CHORD NEAR CUTOUTS			BOEING CARD NO. <b>54-627-01-01</b>	
DATE	SPECIAL DETAILED		(DIRECTION 2)			D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL <b>AIRPL</b>				ALL	ALL
		ACCESS 431BL 431BR 43 433DT	31CL 431CR 433AT	433BT 433CT	ZONE <b>433</b>	

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - Nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AWL LEFT UPPER SPAR CHORD NEAR CUTOUTS (DIRECTION 2)

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		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-627		
	TAS	SK 54-	-05-02-250-8 <sup>-</sup>	16					MECH	INS
1.	INT	ERNA	L - SPECIAL	DETAILED: UP	PER SPA	R CHORD NEA	AR CUTOUTS (DIREC	TION 2)		
	Α.	Iner	ection				·			
	Α.	_								
			ASK 54-05-02-010-03		or Facino	No. 1.				
		(1)	•	access panels for	_	INO. I.				
			<u>Number</u>	Name/Locatio		6 LU LO				
			431BL		-	ft Mid Strut Fair	•			
			431BR 431CL		•	ght Mid Strut Fa ft Overwing Faiı	•			
			431CR		-	ght Overwing Fail	_			
			433AT	Strut, Forward	_	-	alling, Ottat i			
			433BT	Strut, Forward	•					
			433CT	Strut, Upper Sp	•					
			433DT	Strut, Upper Sp	oar Web, S	Strut 1				
			Open these	access panels for	or Engine	No. 2:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			441BL	Forward Strut F	airing, Le	ft Mid Strut Fair	ing, Strut 2			
			441BR			ght Mid Strut Fa				
			441CL		_	ft Overwing Fair	=			
			441CR		_	ght Overwing Fa	airing, Strut 2			
			443AT	Strut, Forward	•					
			443BT 443CT	Strut, Forward						
			443C1 443DT	Strut, Upper Sp Strut, Upper Sp						
					our vvob, c	oli di Z				
		(2)	NSK 54-05-02-250-010		Current in	enaction of the L	upper spar chord near t	the cutoute:		
		(2)	•			•	ord, nacelle STA 213.6			
							celle STA 233.6 on the			
			See Doc. D	626A001-DTR, [	OTR check	form 54-51-17,	for alternative inspect	ions.		
			The NDI me	ethod(s) necessa	ry to acco	mplish the inten	t of this inspection is c	ontained in		
					•	06-37239). The	inspection procedures	are		
			contained in	Part 6, Subject	54-40-07.					
			ASK 54-05-02-410-03			4				
		(3)		access panels f	•	no. 1:				
			<u>Number</u>	Name/Locatio						
			431BL		•	ft Mid Strut Fair	•			
			431BR		_	ght Mid Strut Fa	_			
			431CL		-	ft Overwing Fair	•			
			431CR		_	ght Overwing Fa	airing, Strut 1			
			433AT 433BT	Strut, Forward Strut, Forward	•					
			433CT	Strut, Upper Sp	•					
				7 11 -1						L
			AKS ALL		SOURCE AWL	LEFT UPPER S	PAR CHORD NEAR CUT	TOUTS (DIREC	CTION	2)
						D0004400 4100			n 1	
						D633A109-AKS 54-627-01-01			Page 2 un 15/	





DATE		TAIL NUMBER	STA	TION	AIRLINE CARD NO.	BOEING C. <b>54-627</b>		
	(Continue	d)	•	1		•	MECH	INSP
	Number	Name/Location	1					
	433DT	Strut, Upper Spa		t 1				
	Close these	e access panels fo						
	Number	Name/Location	_					
	441BL	Forward Strut Fa		lid Strut Fairi	na. Strut 2			
	441BR	Forward Strut Fa						
	441CL	Forward Strut Fa						
	441CR	Forward Strut Fa	airing, Right	Overwing Fa	iring, Strut 2			
	443AT	Strut, Forward S	•					
	443BT	Strut, Forward S						
	443CT	Strut, Upper Spa						
	443DT	Strut, Upper Spa	ar Web, Stru	t 2				
			END OF TAS	SK				
	EFFECTIVITY		SOURCE LE	FT UPPER SI	PAR CHORD NEAR CUT	OUTS (DIREC	TION	2)
	AKS ALL		AWL					
			De	33A109-AKS		ı	Page 3	of 3
				-627-01-01			un 15/	
L								





AIRLINE CARD NO		RIGHT UPPER SPAR CHORD NEAR CUTOUTS			BOEING CARD NO. <b>54-627-01-02</b>	
DATE	SPECIAL DETAILED		(DIRECTION 2)		RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 441BL 441BR 44 443DT	1CL 441CR 443AT	443BT 443CT	ZONE <b>443</b>	

Inspect (High Frequency Eddy Current) the upper spar chord near the cutouts: Nacelle STA 200.9 - nacelle STA 211.5 on the right chord, nacelle STA 213.6 - nacelle STA 225.2 on the right chord, nacelle STA 226.1 - Nacelle STA 233.6 on the right chord.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AWL RIGHT UPPER SPAR CHORD NEAR CUTOUTS (DIRECTION 2)

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54-627-01-02 Jun 15/2015



## 737-600/700/800/900 **TASK CARDS**

					IAS	K CAKDS				
	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/		
	TAS	K 54-	05-02-250-8	316					MECH	INS
1.					PER SPA	R CHORD NEA	AR CUTOUTS (DIRECT	ΓΙΟΝ 2)		
								<u> </u>		
	Α.	-	ection							
			SK 54-05-02-010-0		r Engine	No. 1.				
		(1)	•	e access panels fo	•	INO. I.				
			Number	Name/Location	_	ft Mid Ctrut Fair	rina Ctrut 1			
			431BL 431BR	Forward Strut F Forward Strut F	-		•			
			431CL	Forward Strut F	_	-	_			
			431CR	Forward Strut F	•	•	•			
			433AT	Strut, Forward S	Spar Web	, Strut 1				
			433BT	Strut, Forward S						
			433CT	Strut, Upper Sp						
			433DT	Strut, Upper Sp	ar Web, S	Strut 1				
			Open these	e access panels fo	•	No. 2:				
			<u>Number</u>	Name/Location	_					
			441BL	Forward Strut F	-		•			
			441BR	Forward Strut F	_	-	_			
			441CL 441CR	Forward Strut F Forward Strut F	-	-	_			
			443AT	Strut, Forward S	_	-	airing, Strut Z			
			443BT	Strut, Forward S	•					
			443CT	Strut, Upper Sp						
			443DT	Strut, Upper Sp	ar Web, S	Strut 2				
		SUBTA	SK 54-05-02-250-0	16						
		(2)					upper spar chord near t			
							ord, nacelle STA 213.6 celle STA 233.6 on the i			
			See Doc. D	0626A001-DTR, D	TR check	form 54-51-17	, for alternative inspecti	ons.		
							nt of this inspection is co			
				ndestructive Test n Part 6, Subject (	•	,	inspection procedures	are		
		SUBTA	SK 54-05-02-410-0	33						
		(3)	Close these	e access panels fo	or Engine	no. 1:				
			<u>Number</u>	Name/Location	<u>1</u>					
			431BL	Forward Strut F	-		•			
			431BR	Forward Strut F	_	_	•			
			431CL	Forward Strut F	-	-	_			
			431CR 433AT	Forward Strut F Strut, Forward S	_	-	airing, Strut 1			
			433BT	Strut, Forward S						
			433CT	Strut, Upper Sp						
			EFFECTIVITY AKS ALL		SOURCE	RIGHT UPPER	SPAR CHORD NEAR CU	TOUTS (DIRE	CTION	1 2
			AND ALL		AWL					
						D633A109-AKS 54-627-01-02	3		Page 2 un 15/2	





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CAR <b>54-627-0</b> °		
(Con	tinued)			N	ИЕСН	INSP
Numb		1				
433D7						
	these access panels for					
Numb		_				
441BL		<u>-</u> airing, Left Mid Strut Fair	ing. Strut 2			
441BF		airing, Right Mid Strut Fa				
441CL		airing, Left Overwing Fai				
441CF		airing, Right Overwing Fa				
443AT		Spar Web, Strut 2				
443B1	Strut, Forward	Spar Web, Strut 2				
443C7						
443D7	Strut, Upper Sp	ar Web, Strut 2				
		END OF TASK ———				
EFFECT		SOURCE RIGHT UPPER	SPAR CHORD NEAR CL	JTOUTS (DIREC	TION	1 2)
AKS	ALL	AWL		-		-
		D633A109-AKS	•	Da	ige 3	of 3
		54-627-01-02	•		ige	





AIRLINE	E CARD NO	LEFT UPPER SPAR, R1 - FITTING AND UPPER			BOEING CARD NO. <b>54-628-00-01</b>	
DATE	SPECIAL DETAILED	SPAR WEB JOI	NTS, CHORD ONL	Y (DIRECTION 1)	RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 431BL 431BR 43	1CL 431CR		ZONE 433	

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar web on the (horizontal) chord only near the cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AWL SOURCE AWL LEFT UPPER SPAR, R1 - FITTING AND UPPER SPAR WEB
JOINTS, CHORD ONLY (DIRECTION 1)

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54-628-00-01 Jun 15/2015



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C <b>54-628</b>		
			-05-02-250-8						MECH	
1.						R, R1 - FITTIN	G AND UPPER SPAR V	VEB		
	JOII	N 1 5,	CHORD ON	LY (DIRECTION	<u>1)</u>					
	A.	Insp	pection							
			ASK 54-05-02-010-0			NI. 4				
		(1)	•	e access panels f	J	No. 1:				
			Number 431BL	Name/Locatio		ft Mid Strut Fair	ring Strut 1			
			431BR		-	ght Mid Strut Fa	_			
			431CL		_	ft Overwing Fai	_			
			431CR	Forward Strut I	Fairing, Ri	ght Overwing F	airing, Strut 1			
			Open these	e access panels f	or Engine	No. 2:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			441BL		-	ft Mid Strut Fair	_			
			441BR 441CL		•	ght Mid Strut Fa ft Overwing Fai	•			
			441CR		•	ght Overwing F	•			
		SURT	ASK 54-05-02-250-0		,	g	g,			
		(2)			Current ins	spection of the	upper spar, R1 fitting an	d upper		
		( )	spar web o STA 211.5,	n the (horizontal)	chord onl	y near the cuto	uts: Nacelle STA 200.9 to nacelle STA 226.1 to nac	o nacelle		
			233.6.	06264004 DTD 1		form E4 E1 17	for alternative inequalic			
							, for alternative inspection			
			the 737 No		Manual ([		nt of this inspection is co inspection procedures a			
		SUBTA	ASK 54-05-02-410-0	34						
		(3)	Close these	e access panels f	for Engine	No. 1:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			431BL		0.	ft Mid Strut Fair	0,			
			431BR 431CL		0,	ght Mid Strut Fa ft Overwing Fai	0,			
			431CR		•	ght Overwing F	•			
			Close these	e access panels f	_	-	<b>0</b> ′			
			Number	Name/Locatio	•					
			441BL			ft Mid Strut Fair	ring, Strut 2			
			441BR		•	ght Mid Strut Fa	•			
			441CL		•	ft Overwing Fai	•			
			441CR	Forward Strut I	-airing, Ri	ght Overwing F	airing, Strut 2			
					- END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE AWL		SPAR, R1 - FITTING AND I	UPPER SPAR	WEB	
						D633A109-AKS	3		Page 2 un 15/	
						1				





TASK PECIAL ETAILED VORK AREA	SPAR WEB JOINTS  VERSION	THRESHOLD		RELATE	ED CARD
	VERSION	THRESHOLD	DEDEAT		
G/STRUT	1.1	56000 FC	18000 FC		ABILITY ENGINE
SKILL <b>AIRPL</b>				ALL	ALL
	ACCESS 441BL 441BR 441CI	L 441CR		ZONE <b>443</b>	
	SKILL	SKILL AIRPL ACCESS	SKILL AIRPL	SKILL AIRPL ACCESS	SKILL AIRPL  ACCESS  AIRPLANE ALL  ZONE

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar web on the (horizontal) chord only near the cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT UPPER SPAR, R1 - FITTING AND UPPER JOINTS, CHORD ONLY (DIRECTION 1)	SPAR WEB
		D633A109-AKS 54-628-00-02	Page 1 of 2 Jun 15/2015



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-628-</b>		
ΤΔ?	SK 54.	-05-02-250-8	 R17					MECH	INS
				PER SPA	R. R1 - FITTING	G AND UPPER SPAR V	NEB		
			LY (DIRECTION 1		,				
Α.	Insp	ection							
	_	ASK 54-05-02-010-0	34						
	(1)	Open these	e access panels fo	r Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>1</u>					
		431BL	Forward Strut F	-		•			
		431BR	Forward Strut F		-	_			
		431CL 431CR	Forward Strut F Forward Strut F	•	•	•			
			e access panels fo		9	annig, calat i			
		Number	Name/Location	•	140. 2.				
		441BL	Forward Strut F	_	ft Mid Strut Fair	ing, Strut 2			
		441BR	Forward Strut F	_		_			
		441CL	Forward Strut F	•	•	•			
		441CR	Forward Strut F	airing, Ri	ght Overwing Fa	airing, Strut 2			
		ASK 54-05-02-250-0				<b>-</b>			
	(2)	spar web o	on the (horizontal)	chord only	y near the cutou	upper spar, R1 fitting ar uts: Nacelle STA 200.9 t	to nacelle		
		233.6.				acelle STA 226.1 to nad			
						for alternative inspecti			
		the 737 No		Manual ([		t of this inspection is co inspection procedures			
	SUBTA	ASK 54-05-02-410-0	34						
	(3)	Close these	e access panels fo	or Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>1</u>					
		431BL	Forward Strut F	0,		0,			
		431BR	Forward Strut F	•	•	•			
		431CL 431CR	Forward Strut F Forward Strut F	•	•	•			
			e access panels fo		9	annig, Guar i			
		Number	Name/Location	•	140. 2.				
		441BL	Forward Strut F	_	ft Mid Strut Fair	ina. Strut 2			
		441BR	Forward Strut F	_		•			
		441CL	Forward Strut F	•	•	•			
		441CR	Forward Strut F	airing, Ri	ght Overwing Fa	airing, Strut 2			
				END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE AWL		SPAR, R1 - FITTING AND D ONLY (DIRECTION 1)	UPPER SPAI	R WE	3





AIRLINE	E CARD NO	LEFT UPPER SPAR, R1 - FITTING AND UPPER			BOEING CARD NO. <b>54-628-01-01</b>	
DATE	TASK SPECIAL DETAILED	SPAR WEB JOI	NTS, CHORD ONL	Y (DIRECTION 2)	RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 431BL 431BR 43	31CL 431CR		ZONE 433	

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar web on the (vertical) chord only near the cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AWL SOURCE AWL LEFT UPPER SPAR, R1 - FITTING AND UPPER SPAR WEB

JOINTS, CHORD ONLY (DIRECTION 2)

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54-628-01-01 Jun 15/2015



		DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 54-628-		
			-05-02-250-8						MECH	INS
1.				L DETAILED: UPI ILY (DIRECTION 2		AR, R1 - FITTING	G AND UPPER SPAR \	<u>WEB</u>		
	A.			TET (BIREOTION)	<u>- /</u>					
	A.	_	oection ASK 54-05-02-010-0	35						
		(1)		e access panels fo	r Engine	No. 1:				
		` ,	<u>Number</u>	Name/Location	-					
			431BL	Forward Strut F	airing, Le	ft Mid Strut Fair	ing, Strut 1			
			431BR	Forward Strut F	_	-	_			
			431CL	Forward Strut F	-	_	•			
			431CR	Forward Strut F		-	airing, Strut 1			
			•	e access panels fo	-	No. 2:				
			Number	Name/Location	•	6 M. LO. 4 E				
			441BL 441BR	Forward Strut Fa	-		•			
			441CL	Forward Strut F	_	-	_			
			441CR	Forward Strut F	0.	•	0,			
		SUBTA	ASK 54-05-02-250-0	18						
		(2)	Do a High	Frequency Eddy C	current ins	spection of the u	ıpper spar, R1 fitting an	nd upper		
		, ,	•	, ,	-		Nacelle STA 200.9 to re STA 226.1 to nacelle			
	See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.						ons.			
			the 737 No		Manual ([		t of this inspection is coinspection procedures			
		SUBTA	ASK 54-05-02-410-0	•						
		(3)	Close thes	e access panels fo	r Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>.</u>					
			431BL	Forward Strut F	airing, Le	ft Mid Strut Fair	ing, Strut 1			
			431BR	Forward Strut F	_	_	•			
			431CL	Forward Strut F	•	•	•			
			431CR	Forward Strut F	_		airing, Strut 1			
				e access panels fo	•	No. 2:				
			Number 444DL	Name/Location	•	£ Mid Ot	: · · · · · · · · · · · · · · · · · ·			
			441BL 441BR	Forward Strut Fa	•		•			
			441CL	Forward Strut F	•	•	•			
			441CR	Forward Strut F	-	-	•			
					END OF	TASK ———				
			EFFOT" "T"	T	COLIDOT		DAD DA =======	UDDED CT:		
			AKS ALL		AWL		PAR, R1 - FITTING AND D ONLY (DIRECTION 2)	UPPER SPAR	WEB	
						D633A109-AKS 54-628-01-01			Page 2 un 15/	





AIRLINE	CARD NO		SPAR, R1 - FITTII		BOEING CARD NO. <b>54-628-01-02</b>		
DATE	SPECIAL DETAILED	SPAR WEB JOI	NTS, CHORD ONL	Y (DIRECTION 2)	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLICA AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 441BL 441BR 44	1CL 441CR		ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar web on the (vertical) chord only near the cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 54-40-07.

SOURCE AKS ALL

SOURCE AWL

SOURCE AWL

SOURCE AWL

RIGHT UPPER SPAR, R1 - FITTING AND UPPER SPAR WEB
JOINTS, CHORD ONLY (DIRECTION 2)

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Page 1 of 2
54-628-01-02

Jun 15/2015



	l	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-628		
	TAS	K 54-	-05-02-250-	818		-		ı	MECH	INSI
1.	INT	ERNA	L - SPECIA	AL DETAILED: UP	PER SPA	R, R1 - FITTING	G AND UPPER SPAR \	WEB		
	JOI	NTS -	CHORD O	NLY (DIRECTION	2)					
	A.	Insp	ection							
		SUBTA	ASK 54-05-02-010-	035						
		(1)	Open thes	se access panels f	or Engine	No. 1:				
			<u>Number</u>	Name/Locatio						
			431BL		-	ft Mid Strut Fair	•			
			431BR		_	ght Mid Strut Fa	_			
			431CL 431CR		•	ft Overwing Fair ght Overwing Fa	•			
				se access panels for		-	annig, Guar i			
			Number	Name/Locatio	_					
			441BL	•		ft Mid Strut Fair	ing, Strut 2			
			441BR		-	ght Mid Strut Fa	_			
			441CL		•	ft Overwing Fair	•			
			441CR	Forward Strut F	Fairing, Ri	ght Overwing Fa	airing, Strut 2			
			ASK 54-05-02-250-							
		(2)					ipper spar, R1 fitting an			
	spar web on the (vertical) chord only near the cutouts: Nacelle STA 200.9 to nacelle STA 211.5, nacelle STA 213.6 to nacelle STA 225.2, nacelle STA 226.1 to nacelle STA 233.6.									
	See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.									
			The NDI n	nethod(s) necessa	ry to acco	mplish the inten	t of this inspection is co	ntained in		
						D6-37239). The	inspection procedures	are		
			contained	in Part 6, Subject	54-40-07.					
			ASK 54-05-02-410-		or Engine	No. 1:				
		(3)		se access panels f Name/Locatio	_	NO. I.				
			Number 431BL			ft Mid Strut Fair	ing Strut 1			
			431BR		-	ght Mid Strut Fa	•			
			431CL		_	ft Overwing Fair	•			
			431CR	Forward Strut F	airing, Ri	ght Overwing Fa	airing, Strut 1			
			Close thes	se access panels f	or Engine	No. 2:				
			<u>Number</u>	Name/Locatio	<u>n</u>					
			441BL		_	ft Mid Strut Fair	_			
			441BR		•	ght Mid Strut Fa	•			
			441CL 441CR		•	ft Overwing Fair ght Overwing Fa	•			
			44 ICK				annig, Strut Z			
					END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE AWL		SPAR, R1 - FITTING AND D ONLY (DIRECTION 2)	UPPER SPA	R WE	В
						D633A109-AKS 54-628-01-02			Page 2 un 15/	





AIRLINE	CARD NO	LEFT UPPER SPAR, R1 - FITTING AND UPPER			BOEING CARD NO. <b>54-629-00-01</b>	
DATE	SPECIAL DETAILED	SPA	R JOINTS, WEB (	DNLY	RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL			ALL	ALL	
		ACCESS 431BL 431BR 43 433DT	1CL 431CR 433AT	433BT 433CT	ZONE <b>433</b>	

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar joints, web only near cutouts: Nacelle STA 200.9 - to nacelle STA 211.5, nacelle STA 213.6 - to nacelle STA 225.2, nacelle STA 226.1 - to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

SOURCE AWL LEFT UPPER SPAR, R1 - FITTING AND UPPER SPAR JOINTS, WEB ONLY

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## 737-600/700/800/900 **TASK CARDS**

					1710	IN CANDS				
	1	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-629-</b>		
	TAS	K 54	-05-02-250-8	B19	'				MECH	INS
1.	INT	ERNA	L - SPECIA	L DETAILED: UF	PPER SPA	R, R1 - FITTING	G AND UPPER SPAR	JOINTS -		
		B ON								
	A.	Insr	ection							
	7 1.		ASK 54-05-02-010-0	36						
		(1)		e access panels f	for Engine	No. 1:				
		` ,	Number	Name/Locatio	•					
			431BL	Forward Strut	— Fairing, Le	ft Mid Strut Fair	ing, Strut 1			
			431BR		•	ght Mid Strut Fa	•			
			431CL		•	ft Overwing Fair	•			
			431CR		_	ght Overwing Fa	airing, Strut 1			
			433AT 433BT	Strut, Forward Strut, Forward	•	•				
			433CT	Strut, Upper S	•					
			433DT	Strut, Upper S						
			Open these	e access panels f	for Engine	No. 2:				
			Number	Name/Locatio	<u>on</u>					
			441BL	Forward Strut	Fairing, Le	ft Mid Strut Fair	ing, Strut 2			
			441BR		•	ght Mid Strut Fa	•			
			441CL		_	ft Overwing Fair	_			
			441CR 443AT	Strut, Forward	_	ght Overwing Fa	airing, Strut 2			
			443BT	Strut, Forward	•					
			443CT	Strut, Upper S	•					
	443DT Strut, Upper Spar Web, Strut 2									
		SUBTA	ASK 54-05-02-250-0	19						
		(2)	spar joints,	web only near co	utouts: Na	celle STA 200.9	upper spar, R1 fitting a - to nacelle STA 211.5 acelle STA 233.6.			
			See Doc. D	0626A001-DTR, I	DTR check	c form 54-51-17,	for alternative inspect	ions.		
		SUBTA	ASK 54-05-02-410-0	36						
		(3)	Close thes	e access panels	for Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>on</u>					
			431BL		_	ft Mid Strut Fair	_			
			431BR		•	ght Mid Strut Fa	•			
			431CL 431CR		•	ft Overwing Fair ght Overwing Fa	•			
			433AT	Strut, Forward	_	-	aning, Strut 1			
			433BT	Strut, Forward						
			433CT	Strut, Upper S						
			433DT	Strut, Upper S	par Web, S	Strut 1				
			EFFECTIVITY <b>AKS ALL</b>		SOURCE AWL	LEFT UPPER S WEB ONLY	PAR, R1 - FITTING AND	UPPER SPAR	JOIN.	TS,
						D633A109-AKS			Page 2	
						54-629-00-01		F	eb 15/	201





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 54-629-		
	Close these	access panels fo	or Engine	No. 2:			MECH	INSP
	<u>Number</u>	Name/Location	_					
_	41BL		_	ft Mid Strut Fairi	na Strut 2			
	41BR			ght Mid Strut Fai				
	41CL			ft Overwing Fair				
	41CR			ght Overwing Fa				
	43AT	Strut, Forward S	-	-				
	43BT	Strut, Forward S						
	43CT	Strut, Upper Sp						
	43DT	Strut, Upper Sp						
			END OF					
	EFFECTIVITY AKS ALL		SOURCE AWL	LEFT UPPER SI WEB ONLY	PAR, R1 - FITTING AND U	IPPER SPAR	JOIN <sup>-</sup>	ΓS,
				D633A109-AKS			Page 3	of 2
				54-629-00-01			age 3 eb 15/2	
				1		• '		





AIRLINE	CARD NO	RIGHT UPPER SPAR, R1 - FITTING AND UPPER SPAR JOINTS, WEB ONLY			BOEING CARD NO. <b>54-629-00-02</b>		
DATE	SPECIAL DETAILED	SPA	AR JOINTS, WEB (	ONLY	RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>18000 FC</b>	APPLIC.	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 441BL 441BR 44 443DT	11CL 441CR 443AT	443BT 443CT	ZONE <b>443</b>		

Inspect (High Frequency Eddy Current) the upper spar, R1 fitting and upper spar joints, web only near cutouts: Nacelle STA 200.9 - to nacelle STA 211.5, nacelle STA 213.6 - to nacelle STA 225.2, nacelle STA 226.1 - to nacelle STA 233.6.

See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.

EFFECTIVITY SOURCE RIGHT UPPER SPAR, R1 - FITTING AND UPPER SPAR JOINTS, **AKS ALL AWL WEB ONLY** D633A109-AKS Page 1 of 3 Oct 15/2014 54-629-00-02



## 737-600/700/800/900 **TASK CARDS**

Di	ATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 54-629		
TASI	< 54-05-02-	-250-819					MECH	INSF
			UPPER SPA	R, R1 - FITTING	S AND UPPER SPAR	JOINTS -		
WEB	ONLY							
Α.	Inspection	1						
71.	SUBTASK 54-05-							
		these access pane	ls for Engine	No. 1:				
	Num	•	_					
	431B	L Forward Str	ut Fairing, Le	ft Mid Strut Fair	ing, Strut 1			
	431B			ght Mid Strut Fa	_			
	431C		_	ft Overwing Fair	_			
	431C 433A		rut Fairing, Riç ard Spar Web	ght Overwing Fa	airing, Strut 1			
	433A 433B		ard Spar Web ard Spar Web					
	433C		r Spar Web, S					
	433D	T Strut, Upper	r Spar Web, S	Strut 1				
	Open	these access pane	ls for Engine	No. 2:				
	Num	ber <u>Name/Loca</u>	<u>ıtion</u>					
	441B		_	ft Mid Strut Fair	_			
	441B			ght Mid Strut Fa	_			
	441C 441C		-	ft Overwing Fair ght Overwing Fa	_			
	443A		ard Spar Web	-	annig, Otrut 2			
	443B		ard Spar Web					
	443C		r Spar Web, S					
	443D	T Strut, Uppe	r Spar Web, S	Strut 2				
	SUBTASK 54-05-							
	spar j		r cutouts: Nad	celle STA 200.9	ipper spar, R1 fitting ai - to nacelle STA 211.5 acelle STA 233.6.			
	See [	Doc. D626A001-DT	R, DTR check	form 54-51-17,	for alternative inspect	ons.		
	SUBTASK 54-05-	02-410-036						
	(3) Close	these access pane	els for Engine	No. 1:				
	<u>Num</u>	<u>ber</u> <u>Name/Loca</u>	<u>ıtion</u>					
	431B		•	ft Mid Strut Fair	•			
	431B			ght Mid Strut Fa	_			
	431C 431C		•	ft Overwing Fair ght Overwing Fa	•			
	431C		ard Spar Web	-	annig, Strut i			
	433B	•	ard Spar Web					
	433C		r Spar Web, S					
	433D	T Strut, Uppe	r Spar Web, S	Strut 1				
		CTIVITY ALL	SOURCE AWL	RIGHT UPPER	SPAR, R1 - FITTING AN	D UPPER SPA	R JOIN	NTS





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CAF <b>54-629-0</b>		
Close	these access panels for	or Engine No. 2:			MECH	INSP
Numb	•	•				
441BL 441BF 441CL 441CF	R Forward Strut F Forward Strut F R Forward Strut F	airing, Left Mid Strut Fair airing, Right Mid Strut Fa airing, Left Overwing Fai airing, Right Overwing Fa	airing, Strut 2 ring, Strut 2			
443AT 443BT 443CT	Strut, Forward S	Spar Web, Strut 2 Spar Web, Strut 2 ar Web, Strut 2				
443D1						
		END OF TASK ———				
EFFECT	TIVITY	SOURCE RIGHT LIPPER	CDAD D4 FITTING AND	NIDDED CDAD	IOIN	ITC
AKS		AWL WEB ONLY	SPAR, R1 - FITTING AND	JUFFER SPAR	JUIN	ıı <b>ɔ</b> ,
		D633A109-AKS 54-629-00-02	3		age 3 b 15/2	





			IAON OANDO	•		
AIRLINE	E CARD NO	LEFT UPPE	TITLE R AND LOWER SP	AR CHORDS	1	CARD NO. <b>0-00-01</b>
DATE	SPECIAL DETAILED		(DIRECTION 1)		RELAT	ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLI0 AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		431BL 431BR 43 433DT	31CL 431CR 433AT	433BT 433CT	ZONE <b>433</b>	
					n all splices. Inspe	
					cy subsurface eddy	/ current.
See Doc. D62	26A001-DTR, DT	R check form 54-	-51-17, for alterna	itive inspection	S.	
	EFFECTIVITY AKS ALL		JRCE LEFT UPPE	R AND LOWER	SPAR CHORDS (D	IRECTION 1)
			D633A109-A			Page 1 of Oct 15/20



### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO 54-630-00-01 MECH INSP TASK 54-05-02-250-820 INTERNAL - SPECIAL DETAILED: UPPER AND LOWER SPAR CHORDS Α. Inspection SUBTASK 54-05-02-010-037 Open these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431BR 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2: Number Name/Location 441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443AT Strut, Forward Spar Web, Strut 2 Strut, Forward Spar Web, Strut 2 443BT 443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2 SUBTASK 54-05-02-250-020 Do a Low Frequency Eddy Current inspection of the chords, skins, webs, and bulkhead in all splices. Inspect the inside and the outside of the strut, and all structure buried in the splices using low frequency subsurface eddy current. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections. SUBTASK 54-05-02-410-037 Close these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 Strut, Forward Spar Web, Strut 1 433BT 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 **FFFFCTIVITY** SOURCE LEFT UPPER AND LOWER SPAR CHORDS (DIRECTION 1) **AKS ALL AWL** D633A109-AKS Page 2 of 3 54-630-00-01 Feb 15/2015





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-630</b> -		
	Close thes	e access panels fo	or Engine	No. 2:	1		MECH	INSP
	Number	Name/Location	_					
	441BL	Forward Strut F	_	ft Mid Strut Fairi	na Strut 2			
	441BR	Forward Strut F						
	441CL	Forward Strut F						
	441CR	Forward Strut F						
	443AT	Strut, Forward S			ming, ou at 2			
	443BT	Strut, Forward S	•					
	443CT	Strut, Upper Sp.						
	443DT	Strut, Upper Sp.						
			END OF					
			LIND OI	IAON				
	EFFECTIVITY AKS ALL		SOURCE AWL	LEFT UPPER AI	ND LOWER SPAR CHORI	DS (DIRECTI	ON 1)	<u> </u>
				D633A109-AKS			Page 3	
				54-630-00-01		F	eb 15/	2015





				-		
AIRLINE	E CARD NO	RIGHT UPPER A	TITLE	PAR CHORDS		CARD NO. <b>)-00-02</b>
DATE	TASK SPECIAL DETAILED		DIRECTION 1)	TAK GHOKDO		ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 441BL 441BR 441CL 443DT	_ 441CR 443AT	443BT 443CT	ZONE 443	
		Current) the chords,				
		structure buried in the R check form 54-51-			-	current.
366 Doc. Doz	20/1001-0111, 01	IV CHECK IOIIII 34-31-	TI, IOI alterna	uive ilispections	•	
	EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT UPF	PER AND LOWER	R SPAR CHORDS (E	DIRECTION 1)
			D633A109-			Page 1 of 3 Oct 15/2014



[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-630</b> -		
		05-02-250-82 L - SPECIAL	20 . DETAILED: UP	PER AND	) LOWER SPAR	CHORDS		MECH	INS
A.	Insp	ection							
	SUBTA	SK 54-05-02-010-037	7						
	(1)	Open these	access panels for	or Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>n</u>					
		431BL	Forward Strut F	airing, Le	ft Mid Strut Fair	ing, Strut 1			
		431BR		_	ght Mid Strut Fa	•			
		431CL		-	ft Overwing Fair	_			
		431CR 433AT		•	ght Overwing Fa	airing, Strut 1			
		433BT	Strut, Forward Strut, Forward	•					
		433CT	Strut, Upper Sp	•					
		433DT	Strut, Upper Sp	•					
		Open these	access panels for	or Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>n</u>					
		441BL		•	ft Mid Strut Fair	•			
		441BR		_	ght Mid Strut Fa	•			
		441CL		•	ft Overwing Fair	•			
		441CR 443AT	Strut, Forward	_	ght Overwing Fa	airing, Strut 2			
		443AT	Strut, Forward	•					
		443CT	Strut, Upper Sp	•					
		443DT	Strut, Upper Sp						
	SUBTA	SK 54-05-02-250-020	)						
	(2)	all splices. In		and the o	outside of the str	hords, skins, webs, and rut, and all structure bu			
					-	for alternative inspection	ons.		
	SUBTA	SK 54-05-02-410-037	,			·			
	(3)	Close these	access panels f	or Engine	No. 1:				
		Number	Name/Location	<u>n</u>					
		431BL	Forward Strut F	airing, Le	ft Mid Strut Fair	ing, Strut 1			
		431BR		•	ght Mid Strut Fa	•			
		431CL		-	ft Overwing Fair	•			
		431CR		_	ght Overwing Fa	airing, Strut 1			
		433AT 433BT	Strut, Forward Strut, Forward	•					
		433CT	Strut, Upper Sp	•					
		433DT	Strut, Upper Sp	•					
		EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER	AND LOWER SPAR CHO	RDS (DIRECT	l ΓΙΟΝ 1	)
					D633A109-AKS 54-630-00-02			Page 2 eb 15/	
					1				





DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-630-</b>		
Clo	se these access panels	s for Engine	No. 2:		'	MECH	INSP
	mber Name/Locati	_					
			ft Mid Strut Fairi	ng, Strut 2			
			ght Mid Strut Fai				
			ft Overwing Fairi				
44	1CR Forward Strut	t Fairing, Ri	ght Overwing Fa	iring, Strut 2			
I	SAT Strut, Forward						
I	BBT Strut, Forward						
	SCT Strut, Upper S						
443	BDT Strut, Upper S	Spar Web, S	Strut 2				
		— END OF	TASK ——				
	FECTIVITY <b>KS ALL</b>	SOURCE AWL	RIGHT UPPER A	AND LOWER SPAR CHO	RDS (DIRECT	ION 1	)
			D0004400 1150		_		
			D633A109-AKS 54-630-00-02			age 3 b 15/2	
		_1	JT-030-00-02		Г	JU 13/4	-013





DATE  NIL NUMBER  STATION	TASK SPECIAL DETAILED WORK AREA ENG/STRUT SKILL AIRPL	VERSION 1.1	(DIRECTION 2)  THRESHOLD 56000 FC	REPEAT <b>9000 FC</b>	RELATE	O-01-01 D CARD ABILITY
	SKILL					ABILITY
STATION					AIRPLANE	ENGINE
					ALL	ALL
		ACCESS 431BL 431BR 43 433DT	31CL 431CR 433AT	433BT 433CT	ZONE 433	

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	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-630-		
		05-02-250-82 L - SPECIAL	21 . DETAILED: UP	PER AND	LOWER SPAR	CHORDS		MECH	INS
A.	Insp	ection							
	SUBTA	SK 54-05-02-010-038	3						
	(1)	Open these	access panels fo	or Engine	No. 1:				
		Number	Name/Location	<u>1</u>					
		431BL 431BR 431CL 431CR 433AT 433BT 433CT	Forward Strut F Forward Strut F	airing, Rig airing, Le airing, Rig Spar Web Spar Web	, Strut 1	iring, Strut 1 ing, Strut 1			
		433DT	Strut, Upper Sp						
		Open these	access panels fo	or Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>1</u>					
		441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT	Forward Strut F Forward Strut F	airing, Rig airing, Le airing, Rig Spar Web Spar Web ar Web, S	, Strut 2 Strut 2	iring, Strut 2 ing, Strut 2			
	SUBTA	SK 54-05-02-250-021	I						
	(2)	in all splices		de and the	e outside of the	hords, skins, webs, and bull strut, and all visible structure			
		See Doc. Do	626A001-DTR, D	TR check	form 54-51-17,	for alternative inspections.			
	SUBTA	SK 54-05-02-410-038	3						
	(3)	Close these	access panels for	or Engine	No. 1:				
		Number	Name/Location	<u>1</u>					
		431BL 431BR 431CL 431CR 433AT 433BT 433CT 433DT	Forward Strut F Forward Strut F	airing, Rigairing, Le airing, Rig Spar Web Spar Web ar Web, S	, Strut 1 Strut 1	iring, Strut 1 ing, Strut 1			
		EFFECTIVITY AKS ALL		SOURCE AWL	LEFT UPPER A	ND LOWER SPAR CHORDS (	DIRECTIO	ON 2)	





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO 54-630-01-0	
Close t	these access panels fo	or Engine No. 2:		MECH	H INSP
		-			
Number 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT	Forward Strut Factorian Strut, Forward Strut, Upper Spanish	airing, Left Mid Strut Fair airing, Right Mid Strut Fa airing, Left Overwing Fai airing, Right Overwing Fa Spar Web, Strut 2 Spar Web, Strut 2 ar Web, Strut 2	iring, Strut 2 ring, Strut 2		
EFFECTI AKS A		SOURCE AWL LEFT UPPER A D633A109-AKS 54-630-01-01	ND LOWER SPAR CHOI	Page	3 of 3 5/2015





ENG/STRUT 1.1 56000 FC 9000 FC APPLICABILITY AIRPLANE EN	SPECIAL DETAILED  MBER WORK AREA ENG/STRUT  ION SKILL AIRPL  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT  ACT (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inside the outside of the strut, and all visible structure in the splices using high frequency surface eddy current.	SPECIAL DETAILED  NUMBER WORK AREA ENG/STRUT ATION SKILL AIRPL  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT  Act (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inspect of the strut, and all visible structure in the splices using high frequency surface eddy current	REPEAT APPLICABILITY AIRPLANE ENG ALL ENG ENG ALL ENG	THRESHOLD 56000 FC  1CL 441CR 443AT  rds, skins, webs, ture in the splices	ACCESS 441BL 441BR 4 443DT  Current) the ch	SPECIAL DETAILED  WORK AREA ENG/STRUT  SKILL AIRPL  Frequency Eddy de of the strut, an	STATION  spect (High and the outside
ENG/STRUT STATION  SKILL AIRPL  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT  Pect (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the install the outside of the strut, and all visible structure in the splices using high frequency surface eddy current	ENG/STRUT  SKILL AIRPL  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT  ACT (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inside e outside of the strut, and all visible structure in the splices using high frequency surface eddy current.	ENG/STRUT ATION SKILL AIRPL  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT  ACCESS 441BL FREQUENCY Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inspect of the strut, and all visible structure in the splices using high frequency surface eddy current	DOO FC  9000 FC  APPLICABILITY AIRPLANE ALL  ZONE 443  CR 443AT 443BT 443CT  All  All  All  All  All  All  All  A	56000 FC  1CL 441CR 443AT  rds, skins, webs, ture in the splices	ACCESS 441BL 441BR 4 443DT  Current) the ch	SKILL AIRPL  Frequency Eddy de of the strut, an	spect (High
ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443  Dect (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the install the outside of the strut, and all visible structure in the splices using high frequency surface eddy current	ALL ALL  ACCESS  441BL 441BR 441CL 441CR 443AT 443BT 443CT  At the character of the strut, and all visible structure in the splices using high frequency surface eddy current.	ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT  ACCESS 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443  Exect (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inspect of the strut, and all visible structure in the splices using high frequency surface eddy current	ALL AND ALL ALL ALL ALL ALL ALL ALL ALL ALL AL	rds, skins, webs, cure in the splices	441BL 441BR 4 443DT  Current) the child all visible stru	AIRPL  Frequency Eddy de of the strut, an	spect (High
pect (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the ind the outside of the strut, and all visible structure in the splices using high frequency surface eddy current	441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT  tt (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inside outside of the strut, and all visible structure in the splices using high frequency surface eddy current.	441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT  ect (High Frequency Eddy Current) the chords, skins, webs, and bulkhead in all splices. Inspect the inspect of the strut, and all visible structure in the splices using high frequency surface eddy current	ns, webs, and bulkhead in all splices. Inspect the inshe splices using high frequency surface eddy current	rds, skins, webs, cure in the splices	441BL 441BR 4 443DT  Current) the child all visible stru	de of the strut, an	nd the outsic
d the outside of the strut, and all visible structure in the splices using high frequency surface eddy currer	e outside of the strut, and all visible structure in the splices using high frequency surface eddy current.	he outside of the strut, and all visible structure in the splices using high frequency surface eddy currer	he splices using high frequency surface eddy currer	ure in the splices	d all visible stru	de of the strut, an	nd the outsic
				•			
e Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.	oc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.	Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.	for alternative inspections.	51-17, for alterna	R check form 54	26A001-DTR, DT	ee Doc. D62

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT UPPER AND LOWER SPAR CHO	PRDS (DIRECTION 2)
		D633A109-AKS 54-630-01-02	Page 1 of 3 Oct 15/2014



### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-630-01-02 MECH INSP TASK 54-05-02-250-821 INTERNAL - SPECIAL DETAILED: UPPER AND LOWER SPAR CHORDS Α. Inspection SUBTASK 54-05-02-010-038 Open these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431BR 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2: Number Name/Location 441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443AT Strut, Forward Spar Web, Strut 2 Strut, Forward Spar Web, Strut 2 443BT 443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2 SUBTASK 54-05-02-250-021 Do a High Frequency Eddy Current inspection of the chords, skins, webs, and bulkhead in all splices. Inspect the inside and the outside of the strut, and all visible structure in the splices using high frequency surface eddy current. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections. SUBTASK 54-05-02-410-038 Close these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 Strut, Forward Spar Web, Strut 1 433BT 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 **FFFFCTIVITY** SOURCE RIGHT UPPER AND LOWER SPAR CHORDS (DIRECTION 2) **AKS ALL AWL** D633A109-AKS Page 2 of 3

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DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-630</b> -		
	Close these	access panels fo	or Engine	No. 2:			MECH	INSP
	Number	Name/Location	_					
	441BL	Forward Strut F	_	ft Mid Strut Fairi	na Strut 2			
	441BR	Forward Strut F						
	441CL	Forward Strut F						
	441CR	Forward Strut F						
	443AT	Strut, Forward S			9,			
	443BT	Strut, Forward S						
	443CT	Strut, Upper Sp						
	443DT	Strut, Upper Sp						
			END OF					
	EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER A	AND LOWER SPAR CHO	RDS (DIRECT	ΓΙΟΝ 2	)
				D633A109-AKS 54-630-01-02			Page 3 eb 15/	
				3-7-030-0 I-02		г	CD 13/	2013





AIRLINE	E CARD NO	LEFT	TITLE UPPER SPAR C	HORD		CARD NO. I-00-01
DATE	TASK SPECIAL DETAILED					ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	18000 FC	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 431BL 431BR 431 433DT	ICL 431CR 433A	T 433BT 433CT	ZONE <b>433</b>	
Inspect (High 210.6.	Frequency Eddy	Current) the left a	and right upper	spar chords at ı	nacelle STA 222.6 a	nd nacelle STA
	26A001-DTR, DT	R check form 54-	51-17, for altern	ative inspectior	is.	
	EFFECTIVITY AKS ALL	SOUF		ER SPAR CHOR	RD.	
			D633A109 54-631-00			Page 1 of 3 Oct 15/2014



### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-631-00-01 MECH INSP TASK 54-05-02-250-822 INTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD Α. Inspection SUBTASK 54-05-02-010-039 Open these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2: Number Name/Location 441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443AT Strut, Forward Spar Web, Strut 2 Strut, Forward Spar Web, Strut 2 443BT 443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2 SUBTASK 54-05-02-250-022 Do a High Frequency Eddy Current inspection of the left and right upper spar chords at nacelle STA 222.6 and nacelle STA 210.6. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections. SUBTASK 54-05-02-410-039 Close these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 Strut, Forward Spar Web, Strut 1 433AT 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 **FFFFCTIVITY** SOURCE LEFT UPPER SPAR CHORD **AKS ALL AWL** D633A109-AKS Page 2 of 3 54-631-00-01 Feb 15/2015



DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-631-00-01</b>	
Clos	e these access panels for	or Engine No. 2:		MECH	INSP
Num		=			
441E		<u>-</u> airing, Left Mid Strut Fa	airing Strut 2		
4416		airing, Right Mid Strut I	•		
4410		airing, Left Overwing F			
4410		airing, Right Overwing			
443		Spar Web, Strut 2	3,		
4438		Spar Web, Strut 2			
4430		ar Web, Strut 2			
443[	OT Strut, Upper Sp	ar Web, Strut 2			
		END OF TASK ——	_		
	ECTIVITY	SOURCE LEFT UPPER	SPAR CHORD		
AKS	SALL	AWL			
		D633A109-A	(S	Page 3	3 of 3
I		54-631-00-01		Oct 15	





AIRLINE	CARD NO		TITLE		BOEING	CARD NO.
		RIGH	IT UPPER SPAR C	CHORD		31-00-02
DATE	SPECIAL DETAILED				RELAT	TED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLI AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 441BL 441BR 44 443DT	41CL 441CR 443A	T 443BT 443CT	ZONE <b>443</b>	
10.6.			-51-17, for alterna		acelle STA 222.6 a	and nacene o
	EFFECTIVITY AKS ALL		URCE RIGHT UPI	PER SPAR CHOP	RD	



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-631-00-02 MECH INSP TASK 54-05-02-250-822 INTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD Α. Inspection SUBTASK 54-05-02-010-039 Open these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2: Number Name/Location 441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443AT Strut, Forward Spar Web, Strut 2 Strut, Forward Spar Web, Strut 2 443BT 443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2 SUBTASK 54-05-02-250-022 Do a High Frequency Eddy Current inspection of the left and right upper spar chords at nacelle STA 222.6 and nacelle STA 210.6. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections. SUBTASK 54-05-02-410-039 Close these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 Strut, Forward Spar Web, Strut 1 433AT 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 **FFFFCTIVITY** SOURCE RIGHT UPPER SPAR CHORD **AKS ALL AWL** D633A109-AKS Page 2 of 3 54-631-00-02 Feb 15/2015



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	CARD NO.	
C	Close these	access panels fo	r Engine	No. 2:		MECH	INSP
	Number	Name/Location	_				
4 4 4	141BL 141BR 141CL 141CR 143AT 143BT 143CT 143DT	Forward Strut Fa Forward Strut Fa Forward Strut Fa Forward Strut, Forward S Strut, Forward S Strut, Upper Spa Strut, Upper Spa	airing, Lei airing, Rig airing, Lei airing, Rig par Web ar Web, S	ght Mid Strut Fa ft Overwing Fair ght Overwing Fa , Strut 2 , Strut 2 Strut 2 Strut 2	iring, Strut 2 ing, Strut 2		
	EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER S	SPAR CHORD		
				D633A109-AKS 54-631-00-02		Page 3 Oct 15/	





AIRI INF	E CARD NO		т	ITLE		BOEIN	NG CARD NO.
Alleline		LE	FT UPPER		ORD		32-00-01
DATE	TASK SPECIAL DETAILED					REL	ATED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1		SHOLD 00 FC	REPEAT 18000 FC	APP AIRPLANE	LICABILITY ENGINE
STATION	SKILL AIRPL	100500				ALL	ALL
		431BL 431BR 433DT	431CL 431	CR 433AT	433BT 433CT	ZONE 433	
						242.7 on the left a	nd right chords.
See Doc. D62	26A001-DTR, DT	R check form 5	54-51-17, f	or alterna	tive inspectior	is.	
	EFFECTIVITY AKS ALL		SOURCE LE	EFT UPPE	R SPAR CHOR	RD.	
			D	633A109- <i>F</i>	AKS		Page 1 of 3

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#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-632-00-01 MECH INSP TASK 54-05-02-130-807 INTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD Α. Inspection SUBTASK 54-05-02-010-008 Open these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431BR 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2: Number Name/Location 441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2 441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2 441CL Forward Strut Fairing, Left Overwing Fairing, Strut 2 441CR Forward Strut Fairing, Right Overwing Fairing, Strut 2 443AT Strut, Forward Spar Web, Strut 2 Strut, Forward Spar Web, Strut 2 443BT 443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2 SUBTASK 54-05-02-130-007 Do an Ultrasonic inspection of the hidden portion of the upper spar chord at nacelle STA 242.7 on the left and right chords. See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections. SUBTASK 54-05-02-410-008 Close these access panels for Engine No. 1: Number Name/Location 431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1 431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1 431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1 431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1 Strut, Forward Spar Web, Strut 1 433AT 433BT Strut, Forward Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 **FFFFCTIVITY** SOURCE LEFT UPPER SPAR CHORD **AKS ALL AWL** D633A109-AKS Page 2 of 3 54-632-00-01 Feb 15/2015



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C <b>54-632</b>		
	Close these	e access panels fo	r Engine	No. 2:		1	MECH	INSP
	Number	Name/Location	_					
	441BL		_	ft Mid Strut Fairi	na. Strut 2			
	441BR			ght Mid Strut Fai				
	441CL			ft Overwing Fairi				
	441CR	Forward Strut F	airing, Ri	ght Overwing Fa	iring, Strut 2			
	443AT	Strut, Forward S						
	443BT	Strut, Forward S	•					
	443CT	Strut, Upper Spa						
	443DT	Strut, Upper Spa	ar Web, S	Strut 2				
			END OF	TASK ——				
	EFFECTIVITY AKS ALL		SOURCE	LEFT UPPER SF	PAR CHORD		1	
	ANS ALL		AWL					
				D633A109-AKS		1	Page 3	of ?
				54-632-00-01			oct 15/	
				1				





AIDI INI	E CAPD NO		TITLE		DOEINO.	CAPD NO
AIRLINE	E CARD NO	RIGH	TITLE HT UPPER SPAR	CHORD		CARD NO. 2-00-02
DATE	TASK SPECIAL DETAILED					D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 441BL 441BR 44 443DT	41CL 441CR 443A	T 443BT 443CT	ZONE <b>443</b>	



C	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.			
TAS	K 54-	-05-02-130-8	807					MECH	INSF
INTE	ERNA	L - SPECIA	L DETAILED: UP	PER SPA	R CHORD				
Α.	Insp	ection							
	-		08						
	(1)	Open these	e access panels f	or Engine	No. 1:				
		<u>Number</u>	Name/Locatio	<u>n</u>					
		431BL		-		_			
					-	_			
				-	-	_			
		433AT		•	•	annig, ou at 1			
		433BT							
				ŕ					
		•	•	_	No. 2:				
					£ M: -1 Ot + E - :	i			
				•		•			
		441CL			-	_			
		441CR	Forward Strut F	airing, Ri	ght Overwing Fa	airing, Strut 2			
		443AT							
		443DT							
	SUBTA	ASK 54-05-02-130-0	07						
	(2)		•		den portion of th	e upper spar chord at n	acelle STA		
		See Doc. D	0626A001-DTR, [	OTR check	form 54-51-17,	for alternative inspection	ons.		
	SUBTA	ASK 54-05-02-410-0	08						
	(3)	Close these	•	J	No. 1:				
		<u>Number</u>							
		431BL		-		_			
				•	•	•			
				0.	•	0,			
		433AT		•	•	<b>3</b> ,			
		433BT		•					
		433D1	Strut, Upper Sp	oar vveb, S	strut 1				
		EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER	SPAR CHORD			
I	TAS NTE	NTERNA A. Insp subt/ (1)  SUBT/ (2)	TASK 54-05-02-130-8 NTERNAL - SPECIA  A. Inspection SUBTASK 54-05-02-010-0 (1) Open these Number 431BL 431BR 431CL 431CR 433AT 433BT 433CT 433DT Open these Number 441BL 441BR 441CL 441CR 443AT 443BT 443CT 443DT  SUBTASK 54-05-02-130-0 (2) Do an Ultra 242.7 on th See Doc. E SUBTASK 54-05-02-410-0 (3) Close these Number 431BL 431BR 431CL 431BR 431CL 431CR 433AT 433BT 433CT 433BT 433CT 433BT 433CT 433DT	TASK 54-05-02-130-807  NTERNAL - SPECIAL DETAILED: UP  A. Inspection  SUBTASK 54-05-02-010-008  (1) Open these access panels for Number Name/Location  431BL Forward Strut Forward Strut Forward Strut Forward Strut Forward Strut Forward Strut Forward Strut, Forward Forward Strut, Forward Forward Strut, Upper Special Strut, U	TASK 54-05-02-130-807 NTERNAL - SPECIAL DETAILED: UPPER SPA  A. Inspection  SUBTASK 54-05-02-010-008  (1) Open these access panels for Engine  Number Name/Location  431BL Forward Strut Fairing, Le 431CL Forward Strut Fairing, Le 431CR Forward Strut Fairing, Rig 433AT Strut, Forward Spar Web 433BT Strut, Forward Spar Web 433CT Strut, Upper Spar Web, S 433DT Strut, Upper Spar Web, S Open these access panels for Engine  Number Name/Location  441BL Forward Strut Fairing, Le 441CR Forward Strut Fairing, Le 441CR Forward Strut Fairing, Rig 443AT Strut, Forward Spar Web 443BT Strut, Forward Spar Web 443BT Strut, Forward Spar Web 443BT Strut, Upper Spar Web, S SUBTASK 54-05-02-130-007  (2) Do an Ultrasonic inspection of the hidd 242.7 on the left and right chords. See Doc. D626A001-DTR, DTR check SUBTASK 54-05-02-410-008  (3) Close these access panels for Engine  Number Name/Location  431BL Forward Strut Fairing, Le 431CR Forward Strut Fairing, Le 431BR Forward Strut Fairing, Le 431CR Forward Strut Fairing, Rig 433CT Strut, Upper Spar Web, S	TASK 54-05-02-130-807 NTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD  A. Inspection  SUBTASK 54-05-02-010-008  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fair 431BR Forward Strut Fairing, Right Mid Strut Fair 431CL Forward Strut Fairing, Right Overwing Fair 431CR Forward Strut Fairing, Right Overwing Fair 433AT Strut, Forward Spar Web, Strut 1 433BT Strut, Forward Spar Web, Strut 1 433BT Strut, Upper Spar Web, Strut 1 433CT Strut, Upper Spar Web, Strut 1 433DT Strut, Upper Spar Web, Strut 1 Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fair 441CL Forward Strut Fairing, Right Overwing Fair 441CR Forward Strut Fairing, Right Overwing Fair 443AT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443BT Strut, Forward Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 308TASK 54-05-02-130-007  (2) Do an Ultrasonic inspection of the hidden portion of the 242.7 on the left and right chords. See Doc. D626A001-DTR, DTR check form 54-51-17, 308TASK 54-05-02-130-007  (2) Do an Ultrasonic inspection of Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fair 431BR Forward Strut Fairing, Right Mid Strut Fair 431BR Forward Strut Fairing, Right Mid Strut Fair 431BR Forward Strut Fairing, Right Mid Strut Fair 431CR Forward Strut Fairing, Right Mid Strut Fair 50 Forward Strut Fairing, Right Overwing Fair 50 Forward Strut Fairing, Right Mid Strut Fair 50 Forward Strut Fairing, Right Overwing Fair 50 Forward	TASK 54-05-02-130-807  NTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD  A. Inspection  SUBTRACK \$4-06-02-010-0000  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433BT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441BR Forward Strut Fairing, Right Overwing Fairing, Strut 2  441BR Forward Strut Fairing, Right Overwing Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  443AT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  3UBTASK \$40-9-02-130-007  (2) Do an Ultrasonic inspection of the hidden portion of the upper spar chord at ne 242.7 on the left and right chords.  See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspection of the left and right chords.  See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspection of Subtrask \$40-9-02-1-00000  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  433BT Strut, Forward Spar Web, Strut 1	TASK 54-05-02-130-807 NTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD  A. Inspection  Subtrack 54-05-02-10-009  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  433CF Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433BT Strut, Lopper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 2  441BL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Overwing Fairing, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443BT Strut, Forward Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  SUBTACK 54-05-02-130-007  (2) Do an Ultrasonic inspection of the hidden portion of the upper spar chord at nacelle STA  242.7 on the left and right chords.  See Doc. D626A001-DTR, DTR check form 54-51-17, for alternative inspections.  SUBTACK 54-05-22-130-007  (3) Close these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  433CT Strut, Upper Spar Web, Strut 1  433DT Strut, Upper Spar Web, Strut 1	TASK 54-05-02-130-807  NTERNAL - SPECIAL DETAILED: UPPER SPAR CHORD  A. Inspection  Subtrack 54-05-2010-0000  (1) Open these access panels for Engine No. 1:  Number Name/Location  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CL Forward Strut Fairing, Right Overwing Fairing, Strut 1  431AT Strut, Forward Spar Web, Strut 1  433AT Strut, Forward Spar Web, Strut 1  433BT Strut, Forward Spar Web, Strut 1  433CT Strut, Upper Spar Web, Strut 1  Open these access panels for Engine No. 2:  Number Name/Location  441BL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Mid Strut Fairing, Strut 2  441CL Forward Strut Fairing, Right Wewning Fairing, Strut 2  443CT Strut, Forward Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  443CT Strut, Upper Spar Web, Strut 2  510CT Strut, Upper Spar Web, Strut 2  810CT Strut, Upper Spar Web, Strut 1  431CL Forward Strut Fairing, Left Overwing Fairing, Strut 1  431BL Forward Strut Fairing, Left Mid Strut Fairing, Strut 1  431BL Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Mid Strut Fairing, Strut 1  431CR Forward Strut Fairing, Right Overwing Fairing, Strut 1  433CT Strut, Forward Spar Web, Strut 1



DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-632-00-02</b>	
Close	these access panels for	or Engine No. 2:		MECH IN	NSP
		-			
Numl 441B 441C 441C 443A 443B 443C 443D	Forward Strut F Strut, Forward S T Strut, Upper Sp T Strut, Upper Sp	-	airing, Strut 2 ring, Strut 2		
EFFEC AKS		SOURCE AWL RIGHT UPPER  D633A109-AKS 54-632-00-02		Page 3 of Oct 15/20	





AIRLINE	E CARD NO	LEI	T UP	TITLE PER SPAR C	HORD			CARD NO. <b>2-01-01</b>
DATE	TASK SPECIAL DETAILED							ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1		THRESHOLD 56000 FC	REPE 18000		APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL						ALL	ALL
		431BL 431BR 4 433DT	31CL	431CR 433A	Г 433ВТ 43	зст	433	
Inspect (High	Frequency Eddy	Current) the vis	sible p	ortion of the	upper spa	ar chord	l at nacelle STA 2	242.7 on the left
_	. 40. 26A001-DTR, DT	R check form 54	1-51-1	7, for alterna	ative inspe	ections.		
	EFFECTIVITY AKS ALL		OURCE <b>\WL</b>	LEFT UPP	ER SPAR C	HORD		
				D633A109-				Page 1 of 3 Oct 15/2014



[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-632		
		-05-02-250-8 AL - SPECIAI	23 L DETAILED: UF	PPER SPA	R CHORD			MECH	INS
A.	Insp	ection							
	SUBTA	ASK 54-05-02-010-04	10						
	(1)	Open these	access panels f	or Engine	No. 1:				
		<u>Number</u>	Name/Locatio	<u>n</u>					
		431BL	Forward Strut I	Fairing, Le	ft Mid Strut Fair	ing, Strut 1			
		431BR	Forward Strut I	Fairing, Ri	ght Mid Strut Fa	iring, Strut 1			
		431CL		•	ft Overwing Fai	•			
		431CR		•	ght Overwing Fa	airing, Strut 1			
		433AT	Strut, Forward						
		433BT	Strut, Forward	•					
		433CT 433DT	Strut, Upper Sp Strut, Upper Sp						
			access panels f						
		Number	Name/Locatio	•	NO. 2.				
		441BL			ft Mid Strut Fair	ing Strut 2			
		441BR		_	ght Mid Strut Fa	•			
		441CL		•	ft Overwing Fair	•			
		441CR		_	ght Overwing Fa	•			
		443AT	Strut, Forward	Spar Web	, Strut 2				
		443BT	Strut, Forward	•					
		443CT	Strut, Upper Sp						
		443DT	Strut, Upper Sp	oar Web, S	Strut 2				
		ASK 54-05-02-250-02		0					
	(2)	-	requency Eddy celle STA 242.7		•	visible portion of the upp	er spar		
						, for alternative inspection	ons.		
	SUBTA	ASK 54-05-02-410-04	10						
	(3)	Close these	e access panels f	for Engine	No. 1:				
		<u>Number</u>	Name/Locatio	<u>n</u>					
		431BL	Forward Strut I	Fairing, Le	ft Mid Strut Fair	ing, Strut 1			
		431BR		_	ght Mid Strut Fa	_			
		431CL		•	ft Overwing Fair	•			
		431CR		_	ght Overwing Fa	airing, Strut 1			
		433AT 433BT	Strut, Forward	•					
		433CT	Strut, Forward Strut, Upper Sp						
		433DT	Strut, Upper Sp						
	EFFECTIVITY AKS ALL			SOURCE AWL	LEFT UPPER S	PAR CHORD			
					D633A109-AKS 54-632-01-01	;		Page 2 eb 15/	



DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CAI <b>54-632-0</b>		
	Close thes	e access panels fo	or Engine	No. 2:			МЕСН	INSP
	Number	Name/Location	_					
	441BL	Forward Strut F	_	ft Mid Strut Fairi	ng, Strut 2			
	441BR	Forward Strut F	-		_			
	441CL	Forward Strut F						
	441CR	Forward Strut F	airing, Ri	ght Overwing Fa	iring, Strut 2			
	443AT	Strut, Forward S						
	443BT	Strut, Forward S						
	443CT	Strut, Upper Spa						
	443DT	Strut, Upper Spa	ar Web, S	Strut 2				
			END OF	TASK ——				
	EFFECTIVITY AKS ALL		SOURCE AWL	LEFT UPPER SI	PAR CHORD			
				D0004400 445		_		
				D633A109-AKS 54-632-01-01			age 3 ct 15/2	
				37-002-01-01		00	JE 13/4	-014





AIRLINE	E CARD NO	RIGH		TITLE ER SPAR C	CHORD			NG CARD NO. 6 <b>32-01-02</b>
DATE	TASK SPECIAL DETAILED	_						ATED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1		000 FC		PEAT 00 FC	APF AIRPLANE	PLICABILITY ENGINE
STATION	SKILL AIRPL						ALL	ALL
		441BL 441BR 44 443DT	11CL 44	1CR 443A	Г 443ВТ 4	43CT	ZONE 443	
Inspect (High		Current) the visil	ble port	tion of the	upper sp	ar chore	d at nacelle ST/	A 242.7 on the left
_		R check form 54-	-51-17,	for alterna	ative insp	ections.		
	EFFECTIVITY AKS ALL		JRCE F	RIGHT UPI	PER SPAR	R CHORE	)	
				D633A109- 54-632-01-				Page 1 of 3 Oct 15/2014



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-632		
	TAS	K 54	-05-02-250-8	823						INSF
1.	INT	ERNA	L - SPECIA	L DETAILED: UF	PPER SPA	R CHORD				
	Α.	Inst	ection							
		-	ASK 54-05-02-010-0	040						
		(1)	Open thes	e access panels f	or Engine	No. 1:				
			<u>Number</u>	Name/Locatio	<u>on</u>					
			431BL		-	ft Mid Strut Fair	_			
			431BR		_	ght Mid Strut Fa	_			
			431CL 431CR		-	ft Overwing Fai ght Overwing Fa	•			
			433AT	Strut, Forward	•	•	annig, Guat i			
			433BT	Strut, Forward	•					
			433CT	Strut, Upper S						
			433DT	Strut, Upper S	,					
			•	e access panels f	-	No. 2:				
			Number 444D	Name/Locatio		(LAC) - ( E - )	01. 10			
			441BL 441BR		•	ft Mid Strut Fair ght Mid Strut Fa	•			
			441CL		_	ft Overwing Fai	_			
			441CR		•	ght Overwing Fa	•			
			443AT	Strut, Forward						
			443BT	Strut, Forward						
			443CT 443DT	Strut, Upper Sp Strut, Upper Sp						
		CUDT	ASK 54-05-02-250-0		pa. 1100, c	5 at 2				
		(2)	Do a High			•	visible portion of the upps.	er spar		
			See Doc. I	D626A001-DTR, [	OTR check	form 54-51-17	, for alternative inspection	ons.		
		SUBTA	ASK 54-05-02-410-0	040						
		(3)	Close thes	e access panels	for Engine	No. 1:				
			<u>Number</u>	Name/Locatio	<u>on</u>					
			431BL		-	ft Mid Strut Fair	_			
			431BR		•	ght Mid Strut Fa	•			
			431CL 431CR		0.	ft Overwing Fai ght Overwing Fa	0,			
			433AT	Strut, Forward	•	•	annig, ou at 1			
			433BT	Strut, Forward	•					
			433CT	Strut, Upper S						
			433DT	Strut, Upper S	par Web, S	Strut 1				
			EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER	SPAR CHORD			<u>I</u>
						D633A109-AKS 54-632-01-02	3		Page 2 eb 15/	
						1				



DA	TE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ 54-632-		
	Close thes	se access panels fo	or Engine	No. 2:		-	MECH	INSP
	<u>Number</u>	Name/Location	_					
	441BL	Forward Strut F	airing, Le	ft Mid Strut Fairi	ng, Strut 2			
	441BR			ght Mid Strut Fai				
	441CL			ft Overwing Fair				
	441CR 443AT	Strut, Forward Strut F		ght Overwing Fa	iring, Strut 2			
	443AT	Strut, Forward S						
	443CT	Strut, Upper Sp						
	443DT	Strut, Upper Sp						
			END OF	TASK ———				
	EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT UPPER	SPAR CHORD			
				D633A109-AKS		F	Page 3	of 3
				54-632-01-02			oct 15/	





			TASK CARDS	5		
AIRLINE	E CARD NO	LEFT R3/R4 F	TITLE IRST FASTENER RO	OW - VERTICAL	BOEING C 54-633	
DATE	TASK SPECIAL DETAILED		LEG		RELATED	) CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICA AIRPLANE	BILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		431EL 431ER 43	34AL 434AR		ZONE <b>433</b>	
Inspect (High	Frequency Eddy	/ Current) all exp	osed surfaces of t	he R3/R4 first fas	stener row on the	vertical leg
right and left		y Current) all exp	osed surfaces of t	ne nome motias	sterier row on the	vertical leg
See Doc. D62	26A001-DTR, DT	R check form 54	-51-17, for alterna	tive inspections.		
	EFFECTIVITY AKS ALL		URCE LEFT R3/R4	FIRST FASTENE	R ROW - VERTICA	L LEG
			LEFT R3/R4 WL D633A109-A 54-633-00-0	AKS	R ROW - VERTICA	L LEG Page 1 o Oct 15/20



С	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C/ <b>54-633-</b>		
		.05-02-250-6 L - SPECIA		R4 FIRS	T FASTENER R	OW - VERTICAL LEG		MECH	INSP
Α.	Insp	ection							
	SUBTA	ASK 54-05-02-010-0	41						
	(1)	Open these	e access panels fo	r Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>!</u>					
		431EL	Forward Strut Fa	airing, Le	ft T.R. Strut Fair	ring, Strut 1			
		431ER	Forward Strut Fa	_	-	_			
		434AL	Aft Strut Fairing,						
		434AR	Aft Strut Fairing,	, Right Fo	orward Panel, S	trut 1			
		Open these	e access panels fo	r Engine	No. 2:				
		<u>Number</u>	Name/Location	1					
		441EL	Forward Strut Fa	airing, Le	ft T.R. Strut Fair	ing, Strut 2			
		441ER	Forward Strut Fa	•	•	•			
		444AL	Aft Strut Fairing,						
		444AR	Aft Strut Fairing,	, Right Fo	orward Panel, S	trut 2			
	SUBTA	ASK 54-05-02-250-0	24						
	(2)	_	Frequency Eddy C w on the vertical le		•	xposed surfaces of the	R3/R4 first		
				-		for alternative inspection	nns		
	CUDTA	ASK 54-05-02-410-0		117 011001	(101111 0 + 01 17,	nor alternative mopeout	5110.		
	(3)		e access panels fo	r Engine	No. 1:				
	( )	Number	Name/Location	-					
		431EL	Forward Strut Fa	•	ft T.R. Strut Fair	ina. Strut 1			
		431ER	Forward Strut Fa	•		•			
		434AL	Aft Strut Fairing,	, Left For	ward Panel, Str	ut 1			
		434AR	Aft Strut Fairing,	, Right Fo	orward Panel, S	trut 1			
		Close thes	e access panels fo	r Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>l</u>					
		441EL	Forward Strut Fa	airing, Le	ft T.R. Strut Fair	ring, Strut 2			
		441ER	Forward Strut Fa	•	•	•			
		444AL	Aft Strut Fairing,						
		444AR	Aft Strut Fairing,	, Right Fo	orward Panel, S	trut 2			
				END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE <b>AWL</b>	LEFT R3/R4 FIF	RST FASTENER ROW - V	ERTICAL LEG	3	
					D633A109-AKS 54-633-00-01			Page 2 eb 15/	





		• *	AON OANDO	,		
AIRLINE	E CARD NO	RIGHT R3/R4 FIRS		OW - VERTICAL		CARD NO. <b>3-00-02</b>
DATE	TASK SPECIAL DETAILED		LEG		RELATE	ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLIC AIRPLANE	CABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		441EL 441ER 444A	AL 444AR		ZONE <b>443</b>	
spect (High	Frequency Eddy sides.	Current) all expose	ed surfaces of the	ne R3/R4 first fa	stener row on the	vertical leg
_		R check form 54-5	1-17, for alterna	tive inspections.		
	EFFECTIVITY	SOURC	E RIGHT R3/R	4 FIRST FASTEN	IER ROW - VERTION	CAL LEG
	AKS ALL	AWL	-			<del>-</del>
			D633A109-A	AKS		Page 1
			54-633-00-0			Oct 15/2



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 54-633		
TA	SK 54	-05-02-250-8	24	1				MECH	INSF
1. <u>IN</u> 1	TERN/	L - SPECIA	L DETAILED: R3	R4 FIRS	T FASTENER R	OW - VERTICAL LEG			
Α.	Insp	ection							
	SUBT	ASK 54-05-02-010-04	11						
	(1)	Open these	access panels f	or Engine	No. 1:				
		Number	Name/Locatio	<u>n</u>					
		431EL		•	ft T.R. Strut Fair	•			
		431ER		•	ght T.R. Strut Fa	•			
		434AL 434AR		-	ward Panel, Str orward Panel, S				
			access panels f			uuti			
		Number	Name/Locatio	_	NO. 2.				
		441EL			ft T.R. Strut Fair	ring Strut 2			
		441ER		•	ght T.R. Strut Fa	•			
		444AL		•	ward Panel, Str	•			
		444AR	Aft Strut Fairing	g, Right Fo	orward Panel, S	trut 2			
	SUBT	ASK 54-05-02-250-02	24						
	(2)	-	Frequency Eddy w on the vertical		•	xposed surfaces of the	R3/R4 first		
						, for alternative inspecti	ons.		
	SUBT	ASK 54-05-02-410-04				'			
	(3)	Close these	e access panels	for Engine	No. 1:				
	, ,	Number	Name/Locatio	<u>n</u>					
		431EL	Forward Strut	Fairing, Le	ft T.R. Strut Fair	ring, Strut 1			
		431ER		•	ght T.R. Strut Fa	•			
		434AL		-	ward Panel, Str				
		434AR		-	orward Panel, S	trut 1			
			e access panels	_	No. 2:				
		<u>Number</u> 441EL	Name/Locatio		.ft T.D. Otw.it Fair	ring Ctrut 2			
		441EL 441ER		•	ft T.R. Strut Fair ght T.R. Strut Fa	•			
		444AL		-	ward Panel, Str	_			
		444AR	Aft Strut Fairing	g, Right Fo	orward Panel, S	trut 2			
				- END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE <b>AWL</b>	RIGHT R3/R4 F	IRST FASTENER ROW -	VERTICAL LE	EG	
					D633A109-AKS 54-633-00-02	3		Page 2 eb 15/	
					<u> </u>				





AIRLINE	CARD NO	LEFT FORWA	TITLE ARD ENGINE MOU	BOEING CARD NO. <b>54-634-00-01</b>			
DATE	TASK SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC		APPLICABILITY AIRPLANE ENGINI	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE 413 414		

Inspect (Penetrant) the entire forward engine mount hanger. The critical detail is the bolt hole detail at the top of the mount.

See Doc. D626A001-DTR, DTR check form 54-55-03, for alternative inspections.

ACCESS NOTE: Removal of engine and engine mount is required.



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA <b>54-634-</b>		
TAS	SK 54	-05-02-230-	801					MECH	INSI
			AL DETAILED: FO	RWARD	ENGINE MOUN	T HANGER			
Α.	Insr	ection							
Λ.	-	ASK 54-05-02-010-	015						
	(1)		se access panels fo	or Engine	No. 1:				
	( )	Number	Name/Location	_					
		413	Left Fan Cowl,	_					
		414	Right Fan Cowl	•	1				
		415	Left Thrust Rev						
		416	Right Thrust Re	everser, E	ngine 1				
		Open thes	se access panels fo	or Engine	No. 2:				
		<u>Number</u>	Name/Location	<u>n</u>					
		423	Left Fan Cowl,	-					
		424	Right Fan Cowl						
		425	Left Thrust Rev	7					
		426	Right Thrust Re		-				
		NOTE: Re	emoval of engine a	and engine	e mount is requi	red.			
		ASK 54-05-02-230-							
	(2)		etrant inspection of ole detail at the top		_	mount hanger. The cri	tical detail is		
		See Doc. I	D626A001-DTR, D	TR check	form 54-55-03,	for alternative inspecti	ons.		
	SUBT	ASK 54-05-02-410-	015						
	(3)	Close thes	se access panels for	or Engine	No. 1:				
		<u>Number</u>	Name/Location	<u>n</u>					
		413	Left Fan Cowl,	-					
		414	Right Fan Cowl	_					
		415 416	Left Thrust Rev	7					
			Right Thrust Re						
			se access panels fo	_	No. 2:				
		Number	Name/Location	_					
		423 424	Left Fan Cowl, Right Fan Cowl	-	2				
		424	Left Thrust Rev	_					
		426	Right Thrust Re						
				END OF	IASK ——				
		EFFECTIVITY AKS ALL		SOURCE AWL	LEFT FORWAR	D ENGINE MOUNT HAN	GER		
					D633A109-AKS 54-634-00-01			Page 2 eb 15/	
					1				





AIRLINE	CARD NO	RIGHT FORW	TITLE ARD ENGINE MO	BOEING CARD NO. <b>54-634-00-02</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC/	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426 NOTE			ZONE <b>423 424</b>	

Inspect (Penetrant) the entire forward engine mount hanger. The critical detail is the bolt hole detail at the top of the mount.

See Doc. D626A001-DTR, DTR check form 54-55-03, for alternative inspections.

ACCESS NOTE: Removal of engine and engine mount is required.

SOURCE AWL RIGHT FORWARD ENGINE MOUNT HANGER

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54-634-00-02 Oct 15/2014



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-634-00-02 МЕСН INSP TASK 54-05-02-230-801 INTERNAL - SPECIAL DETAILED: FORWARD ENGINE MOUNT HANGER Inspection Α. SUBTASK 54-05-02-010-015 Open these access panels for Engine No. 1: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Name/Location Number 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Removal of engine and engine mount is required. SUBTASK 54-05-02-230-001 Do a Penetrant inspection of the entire forward engine mount hanger. The critical detail is the bolt hole detail at the top of the mount. See Doc. D626A001-DTR, DTR check form 54-55-03, for alternative inspections. SUBTASK 54-05-02-410-015 Close these access panels for Engine No. 1: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 **END OF TASK — FFFFCTIVITY** SOURCE RIGHT FORWARD ENGINE MOUNT HANGER **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-634-00-02 Feb 15/2015





AIRLINE	CARD NO	LEFT 1	TITLE THRUST LINK ASS	54-635		
DATE	TASK GENERAL VISUAL				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>6000 FC</b>	APPLICA	
STATION	SKILL AIRPL				AIRPLANE ALL	ALL ALL
		ACCESS 413 414 415 416			ZONE 415 416	

STATION	SKILL <b>AIRPL</b>			ALL	ALL
		ACCESS 413 414 415 416		ZONE 415 416	
Inspect (Gene	ral Visual) the th	rust link and the thrus	t link clevis lua		
			e intact thrust link clevis lug.		
			5, for alternative inspections.		
	EFFECTIVITY AKS ALL	source AWL	LEFT THRUST LINK ASSEME	BLY	
			D633A109-AKS		Page 1 of 2

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT THRUST LINK ASSEMBLY	
		D633A109-AKS 54-635-00-01	Page 1 of 2 Oct 15/2014



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-635-00-01 месн I INSP TASK 54-05-02-210-802 **INTERNAL - GENERAL VISUAL: THRUST LINK ASSEMBLY** Inspection Α. SUBTASK 54-05-02-010-011 Open these access panels for Engine No. 1: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Name/Location Number 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 SUBTASK 54-05-02-210-002 Do a General Visual inspection of the thrust link and the thrust link clevis lug. Lead crack is the failed thrust link. Critical detail is the intact thrust link clevis lug. See Doc. D626A001-DTR, DTR check form 54-55-05, for alternative inspections. SUBTASK 54-05-02-410-011 (3) Close these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 END OF TASK — **FFFFCTIVITY** SOURCE LEFT THRUST LINK ASSEMBLY **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-635-00-01 Feb 15/2015





AIRLINE	CARD NO	RIGHT	TITLE THRUST LINK ASS		BOEING CARD NO. <b>54-635-00-02</b>		
DATE	TASK GENERAL VISUAL				RELATEI	O CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>6000 FC</b>	APPLICA		
STATION	SKILL AIRPL				AIRPLANE ALL	ALL ALL	
		ACCESS 423 424 425 426			ZONE <b>425 426</b>		

See Doc. D626A001-DTR, DTR check form 54-55-05, for alternative inspections.

EFFECTIVITY SOURCE **RIGHT THRUST LINK ASSEMBLY AKS ALL AWL** D633A109-AKS Page 1 of 2 Oct 15/2014 54-635-00-02



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-635		
TAS	SK 54	-05-02-210	-802					MECH	INSF
			RAL VISUAL: THR	UST LINE	<b>CASSEMBLY</b>				
Α.	Iner	ection							
Α.	-	ASK 54-05-02-010	)-011						
	(1)		se access panels fo	or Engine	No. 1:				
	( )	Number	Name/Location	_					
		413	Left Fan Cowl,	_					
		414	Right Fan Cow	l, Engine 1	1				
		415	Left Thrust Rev		-				
		416	Right Thrust Re	everser, E	ngine 1				
		Open the	se access panels for	or Engine	No. 2:				
		<u>Number</u>	Name/Location	_					
		423	Left Fan Cowl,	•	_				
		424	Right Fan Cow						
		425 426	Left Thrust Rev Right Thrust Re		-				
			•	everser, E	rigirie z				
		ASK 54-05-02-210		ion of the	thought limbs and th	ho throat link alovio lug	l and arnale		
	(2)	is the faile	ed thrust link. Critic	al detail is	the intact thrus	•			
		See Doc.	D626A001-DTR, L	) IR check	t form 54-55-05	, for alternative inspection	ons.		
		ASK 54-05-02-410		–	NI. 4				
	(3)		ese access panels f	_	No. 1:				
		Number	Name/Location						
		413 414	Left Fan Cowl, Right Fan Cow	•	1				
		415	Left Thrust Rev	-					
		416	Right Thrust Re		-				
		Close the	ese access panels f	or Engine	No. 2:				
		Number	•	_					
		423	Left Fan Cowl,						
		424	Right Fan Cow	-	2				
		425	Left Thrust Rev		-				
		426	Right Thrust Re	everser, E	ngine 2				
				END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT THRUST	T LINK ASSEMBLY			
					D633A109-AKS 54-635-00-02	<b>3</b>		Page 2 eb 15/	
					<u> </u>				





AIRLINE	E CARD NO	LE	TITLE FT THRUST LINK I	PIN	BOEING C <b>54-636</b>	
DATE	TASK GENERAL VISUAL				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLIC/	ABILITY ENGINE
STATION	SKILL <b>AIRPL</b>				ALL	ALL
		ACCESS 413 414 415 416			ZONE 415 416	
ad crack is		ink pin. Critical d	etail is the intact t		· ·	



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-636			
TA	<b>ASK 5</b> 4	-05-02-21	10-803					MECH	INSP	
1. <u>IN</u>	ITERN	AL - GEN	ERAL VISUAL: THE	RUST LINE	<u>CPIN</u>					
Α	. Ins	pection								
	SUBT	ASK 54-05-02-	010-012							
	(1)	Open th	nese access panels f	or Engine	No. 1:					
	Number Name/Location									
		413	Left Fan Cowl,	-						
		414 415	Right Fan Cow	-						
		415	Left Thrust Rev Right Thrust R	7	-					
			•							
	Open these access panels for Engine No. 2:  Number Name/Location									
		423	Left Fan Cowl,							
		424	Right Fan Cow	-	2					
		425	Left Thrust Rev	-						
		426	Right Thrust R	7	-					
	SUBT	ASK 54-05-02-	210-003							
	(2)		neral Visual inspectio detail is the intact thi		•	nd crack is the failed thr	rust link pin.			
		See Do	c. D626A001-DTR, [	OTR check	form 54-55-06,	for alternative inspection	ons.			
	SUBT	SUBTASK 54-05-02-410-012								
	(3)									
		Numbe	er Name/Locatio	<u>n</u>						
		413	Left Fan Cowl,	Engine 1						
		414	Right Fan Cow	-						
		415	Left Thrust Rev		-					
		416	Right Thrust R							
			hese access panels f	•	No. 2:					
			Name/Locatio							
		423 424	Left Fan Cowl,	-	2					
		424	Right Fan Cow Left Thrust Rev	-						
		426	Right Thrust R		•					
			_	- END OF						
		EFFECTIV AKS A		SOURCE AWL	LEFT THRUST I	LINK PIN				
					D633A109-AKS 54-636-00-01			Page 2 eb 15/		
				1	I .					





AIRLINE	CARD NO	RIG	TITLE  SHT THRUST LINK	PIN	54-636	
DATE	TASK GENERAL VISUAL				RELATE	D CARD
AIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>6000 FC</b>	APPLICABILITY AIRPLANE ENGINE	
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426			ZONE 425 426	
spect (Gene	eral Visual) the th	rust link pin.				

See Doc. D626A001-DTR, DTR check form 54-55-06, for alternative inspections.

EFFECTIVITY SOURCE **RIGHT THRUST LINK PIN AKS ALL AWL** D633A109-AKS Page 1 of 2 Oct 15/2014 54-636-00-02



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-636-00-02 МЕСН INSP TASK 54-05-02-210-803 **INTERNAL - GENERAL VISUAL: THRUST LINK PIN** Inspection Α. SUBTASK 54-05-02-010-012 Open these access panels for Engine No. 1: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Name/Location Number 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 SUBTASK 54-05-02-210-003 Do General Visual inspection of the thrust link pin. Lead crack is the failed thrust link pin. Critical detail is the intact thrust link clevis lug. See Doc. D626A001-DTR, DTR check form 54-55-06, for alternative inspections. SUBTASK 54-05-02-410-012 (3) Close these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 END OF TASK — **FFFFCTIVITY** SOURCE **RIGHT THRUST LINK PIN AKS ALL AWL** D633A109-AKS Page 2 of 2 54-636-00-02 Feb 15/2015





AIRLINE	CARD NO	LEFT AFT	ENGINE MOUNT		BOEING CARD NO. <b>54-637-00-01</b>		
DATE	SPECIAL DETAILED				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLIC/ AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE 415 416		

Inspect (I	Penetrant)	the entire	aft engine	mount assembly.

The aft mount critical detail is the shear pin hole.

See Doc. D626A001-DTR, DTR check form 54-55-10, for alternative inspections.

**ACCESS NOTE:** The inspection requires the removal of the engine and disassembly of the engine mount.

SOURCE AWL LEFT AFT ENGINE MOUNT ASSEMBLY

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37-00-01	
MECH	INS
е	
Page 2	
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AIRLINE	CARD NO	RIGHT AFT	ENGINE MOUNT	BOEING CARD NO. <b>54-637-00-02</b>		
DATE	TASK SPECIAL DETAILED				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426 NOTE			ZONE <b>425 426</b>	

Inspect (Penetrant) the entire aft engine mount assembly.
---

The aft mount critical detail is the shear pin hole.

See Doc. D626A001-DTR, DTR check form 54-55-10, for alternative inspections.

**ACCESS NOTE:** The inspection requires the removal of the engine and disassembly of the engine mount.

SOURCE AWL RIGHT AFT ENGINE MOUNT ASSEMBLY

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	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-637		
	TAS	K 54-	05-02-230-8	302					MECH	INS
1.	INT	ERNA	L - SPECIA	L DETAILED: A	FT ENGINE	E MOUNT ASSI	EMBLY			
	A.	Insp	ection							
		-	SK 54-05-02-010-0	16						
		(1) Open these access panels for Engine No. 1:								
			<u>Number</u>	Name/Location	<u>on</u>					
			413	Left Fan Cowl	-					
			414	Right Fan Cov	-					
			415 416	Left Thrust Re Right Thrust R						
				e access panels						
			Number	Name/Location	•	140. 2.				
			423	Left Fan Cowl						
			424	Right Fan Cov	-	2				
			425	Left Thrust Re	-					
			426	Right Thrust R	leverser, Ei	ngine 2				
				e inspection requount.	uires the re	moval of the en	gine and disassembly o	of the engine		
		SUBTA	ASK 54-05-02-230-0							
		(2)		trant inspection on the site of the site of the shear pi		ntire aft engine	mount assembly. The a	ft mount		
			See Doc. D	0626A001-DTR,	DTR check	form 54-55-10	, for alternative inspecti	ons.		
			ASK 54-05-02-410-0							
		(3)		e access panels	_	No. 1:				
			Number	Name/Location	<del></del>					
			413 414	Left Fan Cowl Right Fan Cov	-	1				
			415	Left Thrust Re	-					
			416	Right Thrust R	-					
			Close these	e access panels	for Engine	No. 2:				
			<u>Number</u>	Name/Location	<u>on</u>					
			423	Left Fan Cowl	, Engine 2					
			424	Right Fan Cov	-					
			425	Left Thrust Re		•				
			426	Right Thrust R	æverser, ⊑i	ngine 2				
					- END OF	TASK ———				
			AKS ALL		SOURCE AWL	RIGHT AFT EN	GINE MOUNT ASSEMBL	Υ		
						D633A109-AKS			Daga 1	) ^ <del>-</del>
						54-637-00-02	,		Page 2 eb 15/	





AIRLINE	CARD NO	LEFT EVENER BAR ASSEMBLY - OUTBOARD			BOEING CARD NO. <b>54-638-00-01</b>		
DATE	TASK GENERAL VISUAL		LUGS	RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLIC/		
STATION	SKILL AIRPL				AIRPLANE ALL	ALL	
		ACCESS 413 414 415 416 NOTE			ZONE 415 416		

Inspect (General Visual) the evener bar assembly outboard lugs.

Lead crack is the failed evener bar (outboard lug). Critical detail is the intact thrust link clevis lug.

See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.

**ACCESS NOTE:** The inspection requires the removal of the engine and disassembly of the engine mount.

SOURCE AWL LEFT EVENER BAR ASSEMBLY - OUTBOARD LUGS

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54-638-00-01 Oct 15/2014



	1	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING (	CARD NO. 8-00-01	
	TAS	K 54-	-05-02-210-8	804					MECH	INSP
1.					ENER BAR	R ASSEMBLY -	OUTBOARD LUGS			
	Α.		ection							
	Λ.	-	ASK 54-05-02-010-0	12						
		(1) Open these access panels for Engine No. 1:  Number Name/Location								
	413 Left Fan Cowl, Engine 1									
			414	Right Fan Cov	•	1				
			415	Left Thrust Re		-				
			416	Right Thrust R	leverser, E	ngine 1				
		Open these access panels for Engine No. 2:								
			<u>Number</u>	Name/Location						
			423	Left Fan Cowl,		2				
			424 425	Right Fan Cov Left Thrust Re	-					
			426	Right Thrust R		-				
				· ·		•	gine and disassembly o	of the engine		
				ount.	00 1.10 10		girlo arra albaccombly c	or the origine		
		SUBTA	ASK 54-05-02-210-0	04						
		(2)		•			embly outboard lugs. Le ne intact thrust link clev			
	See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.									
		SUBTA	ASK 54-05-02-410-0	13						
		(3) Close these access panels for Engine No. 1:								
			<u>Number</u>	Name/Location	<u>on</u>					
			413	Left Fan Cowl,	-					
	414 Right Fai				-					
			415 416	Left Thrust Re Right Thrust R		-				
				e access panels						
			Number	Name/Locatio	•	NO. 2.				
			423	Left Fan Cowl,						
			424	Right Fan Cov	-	2				
			425	Left Thrust Re						
			426	Right Thrust R	leverser, E	ngine 2				
					- END OF	TASK ———				
					T	I				
			AKS ALL		SOURCE AWL	LEFT EVENER	BAR ASSEMBLY - OUTE	BOARD LUGS	5	
						D633A109-AKS			Dog C	2 04 2
						54-638-00-01		ı	Page 2 eb 15/	





AIRLINE CARD NO		RIGHT EVENER	RIGHT EVENER BAR ASSEMBLY - OUTBOARD			ARD NO. -00-02
DATE	TASK GENERAL VISUAL		LUGS		RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426 NOTE			ZONE 425 426	

Inspect (General	Visual) the	evener bar	assembly	outboard lugs
------------------	-------------	------------	----------	---------------

Lead crack is the failed evener bar (outboard lug). Critical detail is the intact thrust link clevis lug.

See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.

**ACCESS NOTE:** The inspection requires the removal of the engine and disassembly of the engine mount.

RIGHT EVENER BAR ASSEMBLY - OUTBOARD LUGS
AWL

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Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	54-638-		
TA	SK 54	-05-02-210	-804					MECH	INS
1. <u>IN</u>	ΓERN <i>A</i>	AL - GENE	RAL VISUAL: EVE	NER BAR	ASSEMBLY -	OUTBOARD LUGS			
A.	Ins	pection							
	SUBT	ASK 54-05-02-010	0-013						
	(1)	Open the	se access panels f	or Engine	No. 1:				
		<u>Number</u>	Name/Locatio	<u>n</u>					
		413	Left Fan Cowl,						
		414	Right Fan Cow	_					
		415 416	Left Thrust Rev Right Thrust Re	-					
			se access panels for		•				
		Number	Name/Locatio	_	140. 2.				
		423	Left Fan Cowl,						
		424	Right Fan Cow	-	2				
		425	Left Thrust Rev	erser, Eng	gine 2				
		426	Right Thrust Re	everser, Er	ngine 2				
			he inspection requinount.	ires the re	moval of the en	gine and disassembly c	of the engine		
	SUBT	ASK 54-05-02-210	0-004						
	(2)					embly outboard lugs. Le ne intact thrust link clevi			
		See Doc.	D626A001-DTR, E	OTR check	form 54-55-11,	, for alternative inspection	ons.		
	SUBT	ASK 54-05-02-410							
	(3)		ese access panels f	-	No. 1:				
		Number	Name/Locatio						
		413	Left Fan Cowl,	-	1				
		414 415	Right Fan Cow Left Thrust Rev	-					
		416	Right Thrust Re	_					
			ese access panels f		-				
		Number	Name/Locatio	_					
		423	Left Fan Cowl,	Engine 2					
		424	Right Fan Cow						
		425	Left Thrust Rev						
		426	Right Thrust Re	everser, Er	ngine 2				
				END OF	TASK ———				
		EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT EVENER	R BAR ASSEMBLY - OUT	BOARD LUGS	<b>5</b>	1
					D633A109-AKS 54-638-00-02	•		age 2 b 15/2	





AIRLINE CARD NO		LEFT AFT ENG	TITLE SINE MOUNT EVE	BOEING CARD NO. <b>54-639-00-01</b>		
DATE	SPECIAL DETAILED		SPAN		RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>75000 FC</b>	APPLICABILITY	
STATION	SKILL <b>AIRPL</b>				AIRPLANE ALL	ALL ALL
		ACCESS 413 414 415 416 NOTE			ZONE 415 416	

Inspect (Penetrant) the entire aft engine mount evener to	bar.
---	------

See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.

**ACCESS NOTE:** Inspection requires the removal and thorough cleaning of the evener bar.

SOURCE AWL

SOURCE AWL

LEFT AFT ENGINE MOUNT EVENER BAR - MID SPAN

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54-639-00-01 Oct 15/2014



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-639-00-01 MECH INSP TASK 54-05-02-230-803 INTERNAL - SPECIAL DETAILED: AFT ENGINE MOUNT EVENER BAR - MID SPAN Inspection Α. SUBTASK 54-05-02-010-021 Open these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Name/Location Number 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Inspection requires the removal and thorough cleaning of the evener bar. SUBTASK 54-05-02-230-003 Do a Penetrant inspection of the entire aft engine mount evener bar. See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections. SUBTASK 54-05-02-410-021 (3) Close these access panels for Engine No. 1: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 END OF TASK — **FFFFCTIVITY** SOURCE LEFT AFT ENGINE MOUNT EVENER BAR - MID SPAN **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-639-00-01 Feb 15/2015





AIRLINE CARD NO		RIGHT AFT EN	GINE MOUNT EVE	BOEING CARD NO. <b>54-639-00-02</b>		
DATE	SPECIAL DETAILED		SPAN		RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>75000 FC</b>	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ALL ALL
		ACCESS 423 424 425 426 NOTE			ZONE <b>425 426</b>	

Inspect (Penetran	t) the e	entire aft	engine	mount	evener bar.
-------------------	----------	------------	--------	-------	-------------

See Doc. D626A001-DTR, DTR check form 54-55-11, for alternative inspections.

ACCESS NOTE: Inspection requires the removal and thorough cleaning of the evener bar.

SOURCE AWL RIGHT AFT ENGINE MOUNT EVENER BAR - MID SPAN

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54-639-00-02 Oct 15/2014



	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C <b>54-639</b>		
TA	SK 54	-05-02-230-8	803					MECH	INSF
1. <u>INT</u>	ERNA	L - SPECIA	L DETAILED: AI	FT ENGINE	E MOUNT EVEN	NER BAR - MID SPAN			
Α.	Insr	ection							
,	-	ASK 54-05-02-010-0	21						
	(1)		e access panels	for Engine	No. 1:				
	( )	<u>Number</u>	Name/Location	_					
		413	Left Fan Cowl	_ , Engine 1					
		414	Right Fan Cov		I				
		415	Left Thrust Re	-					
		416	Right Thrust R	Reverser, Ei	ngine 1				
		Open these	e access panels	for Engine	No. 2:				
		<u>Number</u>	Name/Location						
		423	Left Fan Cowl	-					
		424	Right Fan Cov	-					
		425 426	Left Thrust Re	-					
			Right Thrust R				•		
		NOTE: Ins	spection requires	the remova	al and thorough	cleaning of the evener I	oar.		
		ASK 54-05-02-230-0							
	(2)		trant inspection of		_				
		See Doc. D	0626A001-DTR,	DTR check	form 54-55-11,	for alternative inspectio	ns.		
		ASK 54-05-02-410-0							
	(3)		e access panels	_	No. 1:				
		<u>Number</u>	Name/Location						
		413	Left Fan Cowl	-	•				
		414 415	Right Fan Cov Left Thrust Re						
		416	Right Thrust R	-					
			e access panels						
			Name/Location	•	140. 2.				
		423	Left Fan Cowl						
		424	Right Fan Cov		2				
		425	Left Thrust Re	-					
		426	Right Thrust R	Reverser, Ei	ngine 2				
				- END OF	TASK ———				
				- LIND OI	IASK ———				
				1	Т				
		AKS ALL		SOURCE AWL	RIGHT AFT EN	GINE MOUNT EVENER BA	AR - MID SPA	AN	
					D633A109-AKS			Page 2	
					54-639-00-02		F	eb 15/	<b>/20</b> 1





AIRLINE	E CARD NO	LEFT STRUT AT	TACHI	TITLE BOLTS (FOR	RWARD AND AFT		CARD NO. <b>)-00-01</b>
DATE	TASK SPECIAL DETAILED		M	OUNTS)			ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1		RESHOLD 6000 FC	REPEAT 6000 FC	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL					ALL	ALL
		ACCESS 413 414 415 416				ZONE 413 414 415 416	
See Doc. D62	26A001-DTR, DT	R check form 54-	-55-13,	, for alterna	tive inspections.		
	EFFECTIVITY AKS ALL		JRCE	LEFT STRU	T ATTACH BOLTS	S (FORWARD AND	AFT MOUNTS)
	AND ALL	A		D633A109-A			Page 1 of 2 Oct 15/2014



	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.		G CARD NO. 40-00-01	
	TAS	K 54	-05-02-700-8	801					MECH	INSF
1.	INT	ERNA	L - SPECIA	L DETAILED: ST	RUT ATT	ACH BOLTS (F	ORWARD AND AFT N	IOUNTS)		
	A.	Insp	ection							
		-	ASK 54-05-02-010-0	42						
		(1)	Open these	e access panels fo	r Engine	No. 1:				
			<u>Number</u>	Name/Location	<u>1</u>					
			413	Left Fan Cowl, E	Engine 1					
			414	Right Fan Cowl,	•					
			415	Left Thrust Reve		-				
			416	Right Thrust Re						
			•	e access panels fo	-	No. 2:				
			<u>Number</u>	Name/Location	_					
			423	Left Fan Cowl, E	-	0				
			424 425	Right Fan Cowl, Left Thrust Reve	_					
			426	Right Thrust Re		-				
		CUDT	ASK 54-05-02-700-0							
		(2)			ıt attach l	oolts on the forw	ard and aft mounts.			
		(2)	-				for alternative inspect	ione		
					TTY CHECK	X 101111 34-33-13,	ioi aiterriative irispect	10115.		
			ASK 54-05-02-410-04	<sup>42</sup> e access panels fo	or Engine	No. 1:				
		(3)	Number	Name/Location	_	INO. I.				
			413	Left Fan Cowl, E	_					
			414	Right Fan Cowl,	-	1				
			415	Left Thrust Reve	_					
			416	Right Thrust Re		-				
			Open these	e access panels fo	r Engine	No. 2:				
			Number	Name/Location	-					
			423	Left Fan Cowl, E	- Engine 2					
			424	Right Fan Cowl,	-	2				
			425	Left Thrust Reve		-				
			426	Right Thrust Re	verser, E	ngine 2				
					END OF	TASK				
			EFFECTIVITY		SOURCE	LEFT STRUT A	TTACH BOLTS (FORWA	RD AND AF	 T MOUN	TS)
			AKS ALL		AWL					
						D633A109-AKS			Page 2	2 of 2
						54-640-00-01			Feb 15/	





AIRLINE	CARD NO	RIGHT STRUT	TITLE ATTACH BOLTS (F	ORWARD AND	BOEING C 54-640	
DATE	TASK SPECIAL DETAILED	_	AFT MOUNTS)		RELATE	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLIC/ AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 423 424 425 426			ZONE 423 424 425 426	
		attach bolts on the R check form 54-5				
	EFFECTIVITY AKS ALL	SOUR		JT ATTACH BOL	TS (FORWARD AN	D AFT MOUI



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-640-00-02 МЕСН INSP TASK 54-05-02-700-801 INTERNAL - SPECIAL DETAILED: STRUT ATTACH BOLTS (FORWARD AND AFT MOUNTS) Inspection Α. SUBTASK 54-05-02-010-042 Open these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Name/Location Number 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 SUBTASK 54-05-02-700-001 Do a Torque Check of all strut attach bolts on the forward and aft mounts. See Doc. D626A001-DTR, DTR check form 54-55-13, for alternative inspections. SUBTASK 54-05-02-410-042 (3) Close these access panels for Engine No. 1: Number Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 - END OF TASK -**FFFFCTIVITY** SOURCE RIGHT STRUT ATTACH BOLTS (FORWARD AND AFT MOUNTS) **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-640-00-02 Feb 15/2015





AIRLINE	CARD NO	LEFT STRUT ATT	LEFT STRUT ATTACH BOLTS (FORWARD AND AFT			ARD NO. • <b>01-01</b>
DATE	TASK SPECIAL DETAILED		MOUNTS)	RELATED	CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 56000 FC	REPEAT <b>75000 FC</b>	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ALL
		ACCESS 413 414 415 416 NOTE			ZONE 413 414 415 416	

Inspect (Penetrant ) the strut attach bolts on the forward and aft mounts.

See Doc. D626A001-DTR, DTR check form 54-55-13 for alternative repeat inspection.

ACCESS NOTE: Removal of engine and engine mounts is required.

SOURCE AWL

SOURCE AWL

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CEFFECTIVITY
AKS ALL

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Fage 1 of 2
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INTER	54-05-02-23					54-640-0	RD NO. <b>)1-01</b>	
INTER		30-804					MECH	INSP
			TRUT ATTA	ACH BOLTS (F	ORWARD AND AFT M	OUNTS)		
				,				
	spection							
	JBTASK 54-05-02-	nese access panels	for Engine	No. 1:				
(1	Numbe		_	110. 1.				
	413	Left Fan Cowl						
	414	Right Fan Cov	-					
	415	Left Thrust Re	-					
	416	Right Thrust F						
	Open these access panels for Engine No. 2:  Number Name/Location							
	423	Left Fan Cowl	, Engine 2					
	424	Right Fan Cov	-					
	425	Left Thrust Re	-					
	426	Right Thrust F	Reverser, Er	ngine 2				
	NOTE:	Removal of engine	and engine	e mounts is requ	uired.			
SU	SUBTASK 54-05-02-230-004							
(2	(2) Do a Penetrant inspection of the strut attach bolts on the forward and aft mounts.							
	See Do	oc. D626A001-DTR,	DTR check	form 54-55-13	for alternative repeat in	spection.		
su	JBTASK 54-05-02-	410-017						
(3	3) Close to	hese access panels	for Engine	No. 1:				
	Numbe	er <u>Name/Location</u>	<u>on</u>					
	413	Left Fan Cowl	, Engine 1					
	414	Right Fan Cov	-					
			-					
	416	Right Thrust F	Reverser, Er	ngine 1				
	Close tl	hese access panels	_	No. 2:				
	Numbe							
	423		-					
		•						
			-					
	426	Right Thrust F	Reverser, Er	ngine 2				
			- END OF	TASK ———				
	EFFECTIV AKS A		SOURCE AWL	LEFT STRUT A	TTACH BOLTS (FORWAR	RD AND AFT M	OUN	TS)
	413 414 415 416 Close th	Left Fan Cowl Right Fan Cov Left Thrust Re Right Thrust F hese access panels	, Engine 1 WI, Engine 1 everser, Eng Reverser, Er for Engine on , Engine 2 WI, Engine 2 everser, Eng Reverser, Eng	gine 1 ngine 1 No. 2: 2 gine 2 ngine 2				





AIRLINE CARD NO		RIGHT STRUT ATTACH BOLTS (FORWARD AND AFT MOUNTS)			BOEING CARD NO. <b>54-640-01-02</b>		
DATE	SPECIAL DETAILED		Al I MODITO,		RELATED	CARD	
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICA	DILITY	
	ENG/STRUT	1.1 56000 FC 75000 FC		AIRPLANE	ENGINE		
STATION	SKILL <b>AIRPL</b>				ALL	ALL	
		ACCESS 423 424 425 426 NOTE			ZONE 423 424 425 426		

Inspect (Penetrant ) the strut attach bolts on the forward and aft mounts.

See Doc. D626A001-DTR, DTR check form 54-55-13 for alternative repeat inspection.

ACCESS NOTE: Removal of engine and engine mounts is required.

SOURCE AWL RIGHT STRUT ATTACH BOLTS (FORWARD AND AFT MOUNTS)

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54-640-01-02 Oct 15/2014



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-640-01-02 МЕСН INSP TASK 54-05-02-230-804 INTERNAL - SPECIAL DETAILED: STRUT ATTACH BOLTS (FORWARD AND AFT MOUNTS) Inspection Α. SUBTASK 54-05-02-010-017 Open these access panels for Engine No. 1: Name/Location Number 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Open these access panels for Engine No. 2: Name/Location Number 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 NOTE: Removal of engine and engine mounts is required. SUBTASK 54-05-02-230-004 Do a Penetrant inspection of the strut attach bolts on the forward and aft mounts. See Doc. D626A001-DTR, DTR check form 54-55-13 for alternative repeat inspection. SUBTASK 54-05-02-410-017 (3) Close these access panels for Engine No. 1: **Number** Name/Location 413 Left Fan Cowl, Engine 1 414 Right Fan Cowl, Engine 1 415 Left Thrust Reverser, Engine 1 416 Right Thrust Reverser, Engine 1 Close these access panels for Engine No. 2: Number Name/Location 423 Left Fan Cowl, Engine 2 424 Right Fan Cowl, Engine 2 425 Left Thrust Reverser, Engine 2 426 Right Thrust Reverser, Engine 2 – END OF TASK – **FFFFCTIVITY** SOURCE RIGHT STRUT ATTACH BOLTS (FORWARD AND AFT MOUNTS) **AKS ALL AWL** D633A109-AKS Page 2 of 2 54-640-01-02 Feb 15/2015





AIRLINE CARD NO		FORWARD S	TITLE FORWARD STRUT FAIRING - ENGINE NO. 1			BOEING CARD NO. <b>54-800-01-01</b>		
DATE	TASK ZONAL (GV)				RELATE	ED CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 9000 FC	REPEAT <b>9000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE		
STATION	SKILL <b>AIRPL</b>	1.2 36 MO 36 MO NOTE		ALL	ALL			
		ACCESS 431AL 431AR 43 431DL 431DR 43	1AT 431BL 431BR 1EL 431ER	431CL 431CR	ZONE <b>431</b>			

Perform an internal zonal inspection (GV) of the forward strut fairing - engine no. 1. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

FORWARD STRUT FAIRING - ENGINE NO. 1

MRB

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54-800-01-01



### 737-600/700/800/900 TASK CARDS

Figure 1)  A. Genera (1) Tree  B. Zonal I  SUBTASK 0 (1) Or  No. 43 43 43 43 43 43 43 43 43 43 43 43 43	c ZONAL (  Al  his Zonal in equirement  conspection  conspection  pen these  comber  31AL  31AR  31AR	inspection procedure t for this zone.	<b>It Fairing - Engine N</b> re satisfies the require	<b>lo. 1</b> ed EZAP-derived Zonal ins	54-800-01-
ASK 05-41- NTERNAL - Figure 1)  A. General (1) The re  B. Zonal II SUBTASK 0 (1) Op  No 43 43 43 43 43 43 43 43 43 43 43 43 43	c ZONAL (  Al  his Zonal in equirement  conspection  conspection  pen these  comber  31AL  31AR  31AR	inspection procedure t for this zone.			spection
NTERNAL - Figure 1)  A. General (1) The rest of the subtask of the	c ZONAL (  Al  his Zonal in equirement  conspection  conspection  pen these  comber  31AL  31AR  31AR	inspection procedure t for this zone.			spection
Figure 1)  A. Genera (1) Tree  B. Zonal I  SUBTASK 0 (1) Or  No. 43 43 43 43 43 43 43 43 43 43 43 43 43	his Zonal i equirement inspection ps-41-04-010-00 pen these umber 31AL 31AR	inspection procedu t for this zone. n access panels:			spection
A. Genera (1) The re  3. Zonal II SUBTASK 0 (1) O  NI 43 43 43 43 43 43 43 43 43 43 43 43 43	his Zonal i equirement Inspection ps-41-04-010-00 pen these umber 31AL 31AR 31AT	t for this zone.  n  access panels:	re satisfies the require	ed EZAP-derived Zonal ins	spection
(1) The rest of the subtask of the s	his Zonal i equirement Inspection ps-41-04-010-00 pen these umber 31AL 31AR 31AT	t for this zone.  n  access panels:	re satisfies the require	ed EZAP-derived Zonal ins	spection
Te B. Zonal II SUBTASK 0 (1) O  Nu 43 43 43 43 43 43 43 43 43 43 43 43 43	equirement Inspection 105-41-04-010-00 pen these umber 31AL 31AR 31AR	t for this zone.  n  access panels:	e satisties the require	ed EZAP-derived Zonai ins	spection
SUBTASK 0 (1) O   Nu 43 43 43 43 43 43 43 43 43 43 43 43 43	Inspection 05-41-04-010-00 pen these umber 31AL 31AR 31AT	n access panels:			
SUBTASK 0 (1) O   Nu 43 43 43 43 43 43 43 43 43 43 43 43 43	pen these wmber 31AL 31AR 31AT	access panels:			
(1) O   Nu 43 43 43 43 43 43 43 43 43 43 43 43 43	pen these umber 31AL 31AR 31AT	access panels:			
Ni 43 43 43 43 43 43 43 43 43 43 43 43 43	umber 31AL 31AR 31AT				
43 43 43 43 43 43 43 43 43 43 43 43 43 4	31AL 31AR 31AT	<u>Name/Location</u>			
43 43 43 43 43 43 43 43 5UBTASK 0 (2) Do (3) Pe (2) Do (4) Re CO SUBTASK 0 (5) CI NI 43 43	31AR 31AT	- 101 (- 1		D:	
43 43 43 43 43 43 43 43 43 43 43 5UBTASK 0 (2) DG (3) PG (E SUBTASK 0 (5) CI M1 43 43 43 43	31AT		_	erser Disconnect, Strut 1	
43 43 43 43 43 43 43 43 43 43 43 43 8UBTASK 0 (2) DC (3) PE SUBTASK 0 (4) RE CC SUBTASK 0 (5) CI M1 43 43 43 43			ring, Right Thrust Re ring, Thumbnail Fairi	verser Disconnect, Strut 1	
43 43 43 43 43 43 43 43 43 43 SUBTASK 0 (2) Do (3) Pe (E SUBTASK 0 (4) Re CO SUBTASK 0 (5) CI Mi 43 43 43	31BL		ring, Left Mid Strut Fa	=	
43 43 43 43 43 SUBTASK 0 (2) DO (3) Pe (E SUBTASK 0 (4) Re CO SUBTASK 0 (5) CI NI 43 43	31BR		ring, Right Mid Strut	•	
43 43 43 43 43 SUBTASK 0 (2) D0 (3) P6 (E SUBTASK 0 (4) R6 C0 SUBTASK 0 (5) CI N1 43 43 43	31CL		ring, Left Overwing F	_	
43 43 43 SUBTASK 0 (2) D0 (3) P6 (E SUBTASK 0 (4) R6 CC SUBTASK 0 (5) CI N1 43 43 43	31CR	Forward Strut Fai	ring, Right Overwing	Fairing, Strut 1	
43 43 SUBTASK 0 (2) D0 (3) P6 (E SUBTASK 0 (4) R6 C0 SUBTASK 0 (5) CI N1 43 43 43	31DL	Forward Strut Fai	ring, Left Underwing	Fairing, Strut 1	
43 SUBTASK 0 (2) D0 (3) P6 (E SUBTASK 0 (4) R6 C0 SUBTASK 0 (5) CI N1 43 43 43	31DR		ring, Right Underwing		
SUBTASK 0 (2) DO (3) Pe (E SUBTASK 0 (4) Re CC SUBTASK 0 (5) CI NI 43 43	31EL		ring, Left T.R. Strut F	=	
(2) Do (3) Pe (E) SUBTASK 0 (4) Re CC SUBTASK 0 (5) CI NI 43 43 43	31ER	Forward Strut Fai	ring, Right T.R. Strut	Fairing, Strut 1	
(3) Pe (E SUBTASK 0 (4) Re CC SUBTASK 0 (5) Cl Nu 43 43 43	05-41-04-210-01				
(E SUBTASK 0 (4) Re CC SUBTASK 0 (5) CI NI 43 43 43	o the zona	al inspection followi	ng the procedures in	AMM 05-00-00-910-804.	
(4) R6 CC SUBTASK 0 (5) CI N1 43 43 43	erform an EZAP)	internal zonal inspe	ection (GV) of the for	ward strut fairing - engine ı	10. 1.
SUBTASK 0 (5) CI Mu 43 43	05-41-04-910-00	5			
(5) CI <u>Ni</u> 43 43			the protection and ca damage to EWIS duri	aution information that will ng maintenance.	minimize
Ni 43 43 43	05-41-04-410-00	7			
43 43 43	lose these	e access panels:			
43 43	<u>umber</u>	Name/Location			
43	31AL	Forward Strut Fai	ring, Left Thrust Reve	erser Disconnect, Strut 1	
	31AR			verser Disconnect, Strut 1	
43	31AT		ring, Thumbnail Fairi	_	
	31BL		ring, Left Mid Strut Fa	•	
	31BR		ring, Right Mid Strut	_	
	31CL 31CR		ring, Left Overwing F ring, Right Overwing	•	
			ring, Right Overwing ring, Left Underwing	_	
			ring, Right Underwing	_	
	31DL 31DR		<i>J, J</i> : - : : : : : : : : : : : : : : : : :	J,	

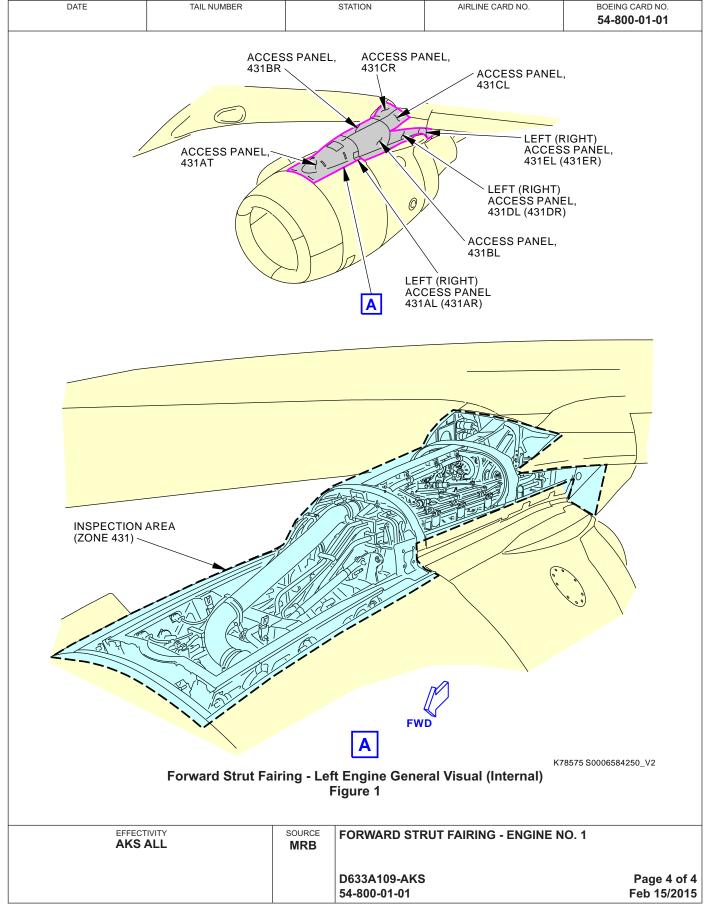
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D633A109-AKS 54-800-01-01



(Continued)  Number Name/Location  431EL Forward Strut Fairing, Left T.R. Strut Fairing, Strut 1  431ER Forward Strut Fairing, Right T.R. Strut Fairing, Strut 1  END OF TASK  ———————————————————————————————————	DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-800-01-01</b>	
	Numbe 431EL	Name/Location Forward Strut Fa	airing, Left T.R. Strut		MECH I	INSP
	431ER	Forward Strut Fa	airing, Right T.R. Stru	ut Fairing, Strut 1		
EFFECTIVITY SOURCE FORWARD STRUT FAIRING - ENGINE NO. 1	EEEECTII	VITY	SOURCE EODIMARD	STRUIT FAIRING FAIGHE NO		
AKS ALL MRB D633A109-AKS Page 3	AKS A	ILL	MRB D633A109-	AKS	. 1 Page 3 c Feb 15/20	of 4









AIRLINE CARD NO		FAN COWL S	TITLE FAN COWL SUPPORT BEAM - ENGINE NO. 1			BOEING CARD NO. <b>54-802-01-01</b>		
DATE	TASK ZONAL (GV)				RELATE	D CARD		
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	1 6600 FC 6600 FC		APPLIC.	ABILITY ENGINE		
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	ALL	ALL		
		ACCESS 431AT			ZONE <b>432</b>			

Perform an internal zonal inspection (GV) of the fan cowl support beam - engine no. 1. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

FAN COWL SUPPORT BEAM - ENGINE NO. 1

MRB

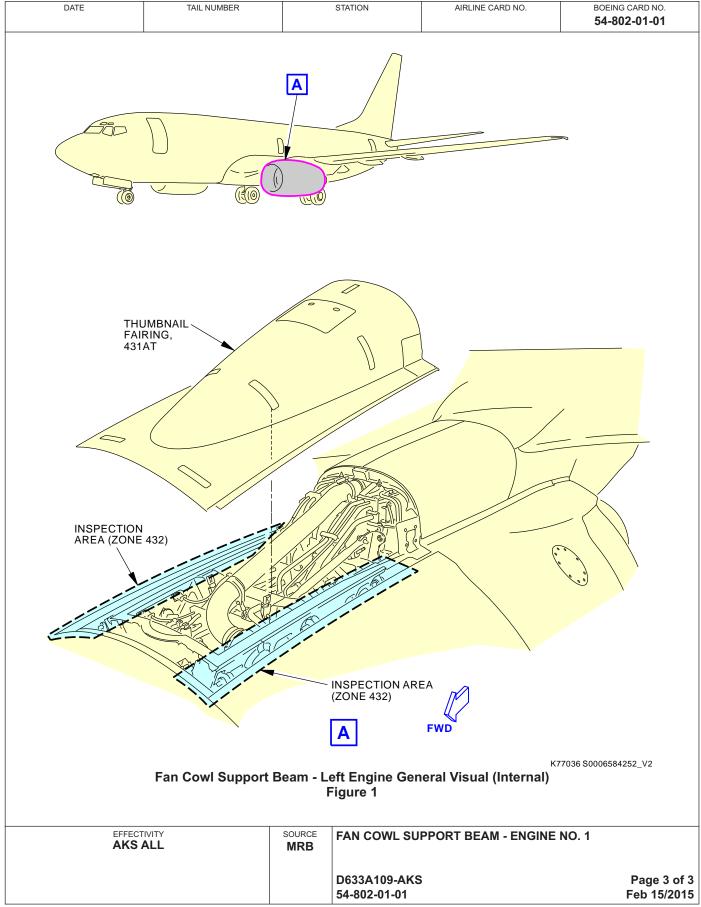
D633A109-AKS
Feb 15/2015



#### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-802-01-01 MECH INSP **EWIS** TASK 05-41-04-210-812 INTERNAL - ZONAL (GV): Fan Cowl Support Beam - Engine No. 1 (Figure 1) A. General This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone. **Zonal Inspection** SUBTASK 05-41-04-010-008 (1) Open this access panel: Number Name/Location 431AT Forward Strut Fairing, Thumbnail Fairing, Strut 1 SUBTASK 05-41-04-210-012 Do the zonal inspection following the procedures in AMM 05-00-00-910-804. Perform an internal zonal inspection (GV) of the fan cowl support beam - engine no. 1. (EZAP) SUBTASK 05-41-04-910-007 Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance. SUBTASK 05-41-04-410-008 (5) Close this access panel: Number Name/Location 431AT Forward Strut Fairing, Thumbnail Fairing, Strut 1 – END OF TASK – **FFFFCTIVITY** SOURCE FAN COWL SUPPORT BEAM - ENGINE NO. 1 **AKS ALL MRB** D633A109-AKS Page 2 of 3 54-802-01-01 Feb 15/2015







### 737-600/700/800/900 TASK CARDS

AIRLINE	AIRLINE CARD NO  TITLE  STRUT TORQUE BOX - ENGINE NO. 1			BOEING CARD NO. <b>54-804-01-01</b>		
DATE	TASK ZONAL (GV)				RELATE	ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 18000 FC	REPEAT 18000 FC	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	6 YR	6 YR	ALL	ALL
		ACCESS 433AL 433AR 43	33AT 433BT 433CT	433DT	ZONE <b>433</b>	

Perform an internal zonal inspection (GV) of the strut torque box - engine no. 1. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

#### A. References

Reference	Title
AMM 05-51-22-210-801	Inspection of Titanium Parts When Contaminated With Fire-Resistant Hydraulic Fluid (P/B 201)

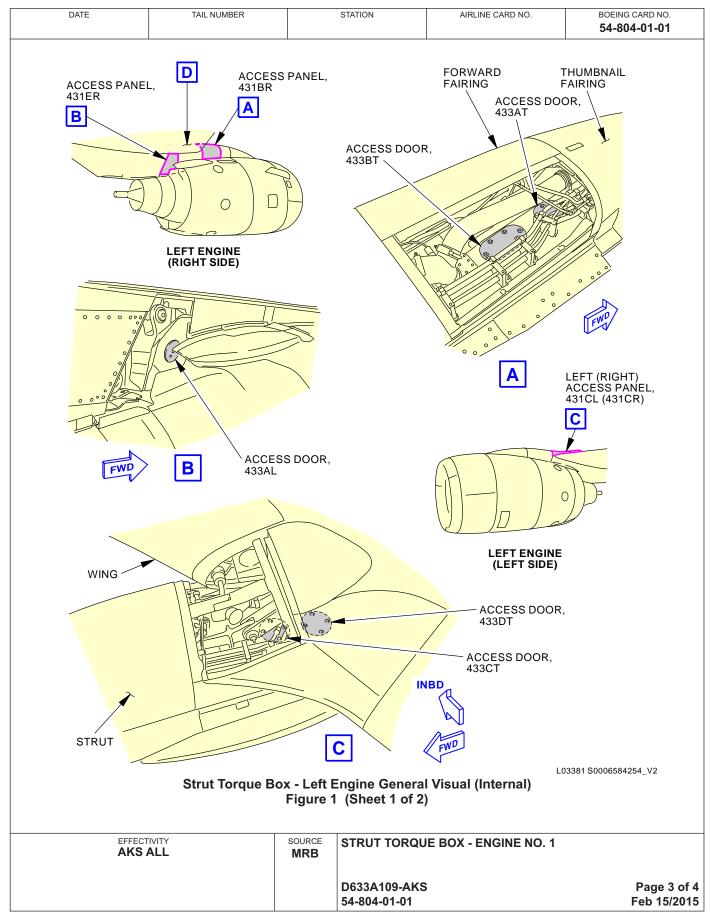
EFFECTIVITY AKS ALL	SOURCE MRB	STRUT TORQUE BOX - ENGINE NO. 1	
		D633A109-AKS 54-804-01-01	Page 1 of 4 Jun 15/2016



### 737-600/700/800/900 **TASK CARDS**

					170	N CARDS				
	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. <b>54-804</b>		
	EWI	_	41-04-210	-813					MECH	INS
1.				_ (GV): Strut Torq	ue Box - E	Engine No. 1				
	(Figi	ure 1)								
	A.	Gen	eral							
		(1)		al inspection proce ent for this zone.	dure satisf	ies the required	I EZAP-derived Zonal in	spection		
	B.	Zon	al Inspecti	on						
		SUBTA	NSK 05-41-04-010	-009						
		(1)	Open thes	se access panels:						
			<u>Number</u>	Name/Location	<u>on</u>					
			433AL	Strut, Left Aft [	Ory Bay, St	rut 1				
			433AR	Strut, Right Aft	Dry Bay, S	Strut 1				
			433AT	Strut, Forward	•					
			433BT	Strut, Forward	•					
			433CT	Strut, Upper S	•					
			433DT	Strut, Upper S	par Web, S	Strut 1				
		SUBTA	ASK 05-41-04-210							
		(2)	Do the zo	nal inspection follo	owing the p	procedures in A	MM 05-00-00-910-804.			
		(3)	Perform a	ın internal zonal in	spection (0	GV) of the strut	torque box - engine no.	1. (EZAP)		
		SUBTA	NSK 05-41-04-211	-001						
		(4)	-				spection of Titanium Pa IM TASK 05-51-22-210-			
		SUBTA	NSK 05-41-04-910	-006						
		(5)		20-60-07-913-801 ation and accident	•		tion information that will g maintenance.	minimize		
		SUBTA	NSK 05-41-04-410	-009						
		(6)	Close the	se access panels:						
			<u>Number</u>	Name/Location	<u>on</u>					
			433AL	Strut, Left Aft [	Dry Bay, St	rut 1				
			433AR	Strut, Right Aff						
			433AT	Strut, Forward	Spar Web	, Strut 1				
			433BT	Strut, Forward	•					
			433CT	Strut, Upper S	•					
			433DT	Strut, Upper S	par Web, S	Strut 1				
					- END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE MRB	STRUT TORQU	JE BOX - ENGINE NO. 1		1	ı
						D633A109-AKS	3		Page 2 un 15/	
					1	1				







### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-804-01-01 ACCESS, 433DT ACCESS, 433CT ACCESS, 433BT ACCESS, 433AT **INSPECTION AREA** (ZONE 433) WING ACCESS, 433AR 1 (RIGHT SIDE IS SHOWN, LEFT SIDE IS EQUIVALENT) D 1 TORQUE BOX SKIN INSTALLATION NOT SHOWN. L07339 S0006584255\_V2 Strut Torque Box - Left Engine General Visual (Internal) Figure 1 (Sheet 2 of 2) EFFECTIVITY SOURCE STRUT TORQUE BOX - ENGINE NO. 1 **AKS ALL MRB** D633A109-AKS Page 4 of 4 Feb 15/2015 54-804-01-01





AIRLINE CARD NO		TITLE AFT STRUT FAIRING - ENGINE NO. 1			BOEING CARD NO. <b>54-806-01-01</b>		
TASK ZONAL (GV)				RELATE	D CARD		
WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 9000 FC	9000 FC 9000 FC		ABILITY ENGINE		
SKILL <b>AIRPL</b>	1.2 NOTE	36 MO	36 MO	ALL	ALL		
	ACCESS 434AL 434AR 43	4BL 434CL		ZONE <b>434</b>			
	TASK ZONAL (GV)  WORK AREA ENG/STRUT  SKILL	TASK ZONAL (GV)  WORK AREA ENG/STRUT SKILL AIRPL ACCESS  AFT STR VERSION 1.1 1.2 ACCESS	AFT STRUT FAIRING - ENG  TASK  ZONAL (GV)  WORK AREA ENG/STRUT SKILL AIRPL  AFT STRUT FAIRING - ENG  VERSION THRESHOLD 1.1 9000 FC 1.2 36 MO  NOTE	AFT STRUT FAIRING - ENGINE NO. 1  TASK ZONAL (GV)  WORK AREA ENG/STRUT 1.1 9000 FC 9000 FC SKILL AIRPL ACCESS	AFT STRUT FAIRING - ENGINE NO. 1   54-806		

Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 1. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 12000 FC/4 YR is satisfied by this zonal inspection.

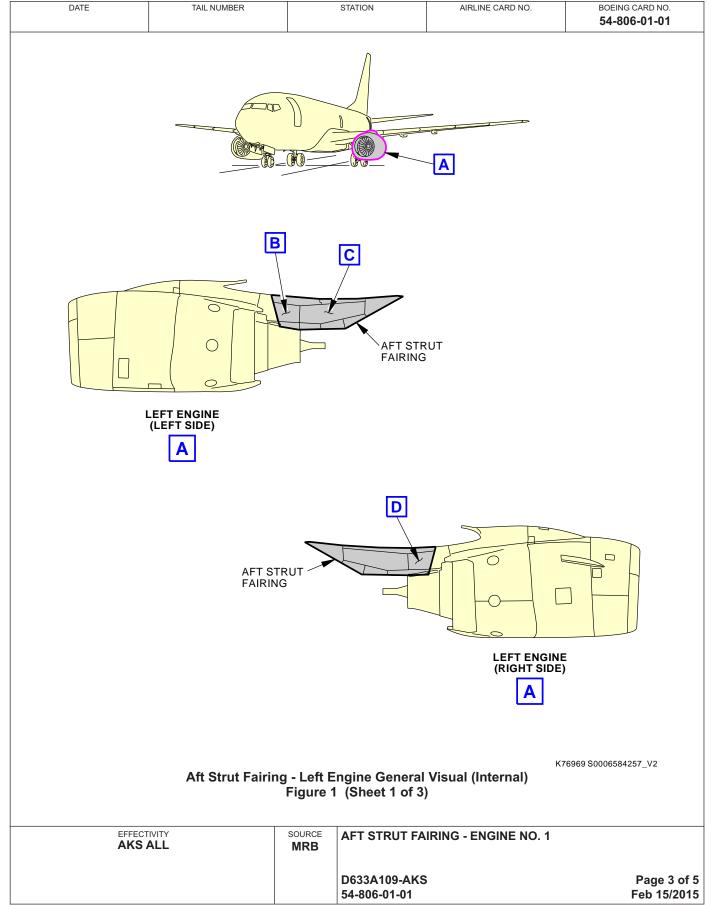
EFFECTIVITY AKS ALL	SOURCE MRB	AFT STRUT FAIRING - ENGINE NO. 1	
		D633A109-AKS 54-806-01-01	Page 1 of 5 Jun 15/2015



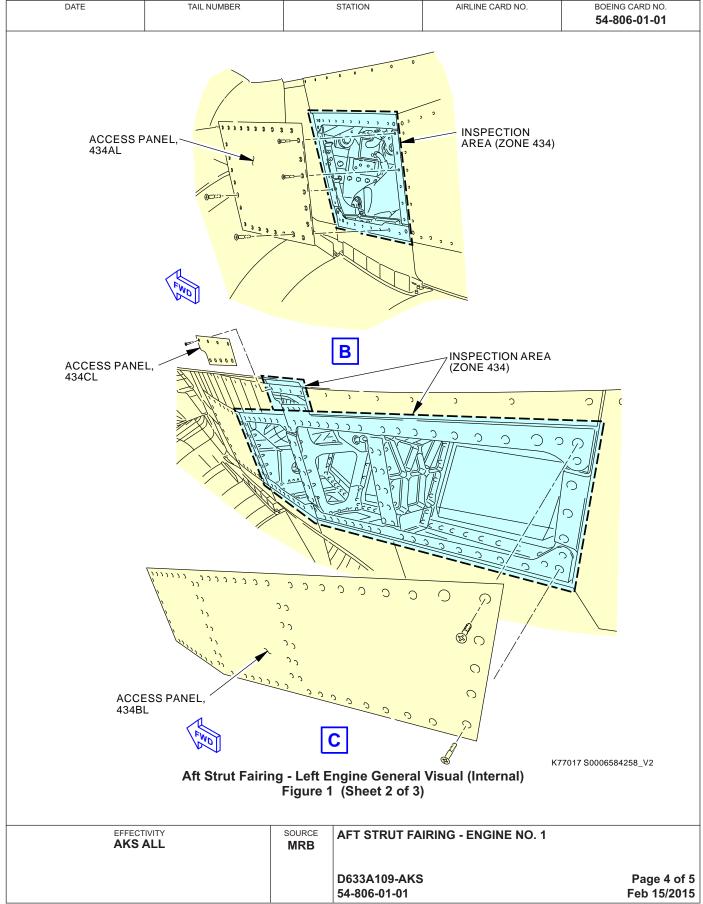
# 737-600/700/800/900

I	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD N <b>54-806-01-0</b>				
EWI	s						MEC				
		-41-04-210-8	314								
			(GV): Aft Strut F	airing - Eı	ngine No. 1						
(Fig	ure 1)	)									
A.	Gen	eral									
	(1)		inspection proced nt for this zone.	dure satisf	ies the require	d EZAP-derived Zonal in	spection				
B.	Zon	al Inspectio	n								
		ASK 05-41-04-010-0									
	(1)		e access panels:								
		Number	Name/Locatio		15 10						
		434AL 434AR	Aft Strut Fairing Aft Strut Fairing								
		434BL	Aft Strut Fairing			otrat 1					
		434CL	Aft Strut Fairing, Left Access To Fuel Door, Strut 1								
	SUBTA	ASK 05-41-04-210-0									
<ul><li>(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.</li><li>(3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 1. (EZAP)</li></ul>											
	(3)	Perform an	internal zonal ins	spection (0	GV) of the aft s	strut fairing - engine no. 1	. (EZAP)				
SUBTASK 05-41-04-910-008  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.							minimizo				
							minimize				
	SUBTASK 05-41-04-410-010										
	(5)	Close these	ose these access panels:								
	Number Name/Location										
		434AL	Aft Strut Fairing								
		434AR 434BL	Aft Strut Fairing Aft Strut Fairing			Strut 1					
		434CL	Aft Strut Fairing			oor, Strut 1					
				FND OF	TASK ———	_					
				LIND OI	IAON						
		EFFECTIVITY AKS ALL		SOURCE MRB	AFT STRUT F	AIRING - ENGINE NO. 1					
					D633A109-AK	-	Page				

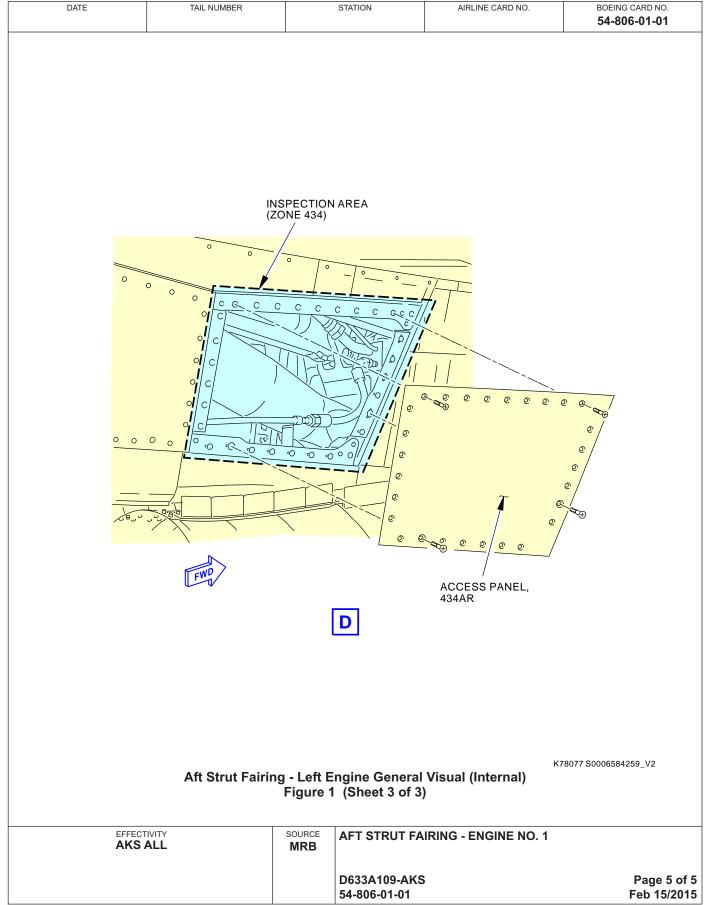
















AIRLINE CARD NO		FORWARD S	TITLE STRUT FAIRING - E	BOEING CARD NO. <b>54-808-02-01</b>		
DATE	TASK ZONAL (GV)				RELATE	ED CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 9000 FC	REPEAT <b>9000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	ALL	ALL
		ACCESS 441AL 441AR 44 441DL 441DR 44	1AT 441BL 441BR 1EL 441ER	441CL 441CR	ZONE <b>441</b>	

Perform an internal zonal inspection (GV) of the forward strut fairing - engine no. 2. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

SOURCE MRB FORWARD STRUT FAIRING - ENGINE NO. 2

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### 737-600/700/800/900 TASK CARDS

	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 54-808		
EW	ıs	I						MECH	T
		-41-04-210-8	R15						Ī
			(GV): Forward S	Strut Eairin	a Engino N	0.2			
	ure 1)		(GV). I OIWaid C	oliul i ali ii	ig - Liigille iv	<u>0. Z</u>			
(i ig	uie i,								
A.	Gen	eral							
	(1)		inspection proce nt for this zone.	dure satisf	ies the require	ed EZAP-derived Zonal in	spection		
В.	Zon	al Inspection	on						
	SUBTA	ASK 05-41-04-010-0	111						
	(1)	Open thes	e access panels:						
		Number	Name/Location	n					
		441AL	Forward Strut	— Fairing. Le	ft Thrust Reve	erser Disconnect, Strut 2			
		441AR	Forward Strut	Fairing, Rig	ght Thrust Rev	verser Disconnect, Strut 2	2		
		441AT	Forward Strut	Fairing, Th	umbnail Fairin	ng, Strut 2			
		441BL	Forward Strut	Fairing, Le	ft Mid Strut Fa	iring, Strut 2			
		441BR	Forward Strut			_			
		441CL	Forward Strut	•	•				
		441CR			-	Fairing, Strut 2			
		441DL		•	•	Fairing, Strut 2			
		441DR				Fairing, Strut 2			
		441EL 441ER	Forward Strut	•		Fairing, Strut 2			
				r airing, rai	giit i.ix. Oliut i	annig, otrat z			
		NSK 05-41-04-210-0		owing the n	rooduros in A	AMM 05-00-00-910-804.			
	(2)		•				0		
	(3)	Perform ar (EZAP)	n internal zonal in	spection (C	JV) of the forw	vard strut fairing - engine	no. 2.		
	SUBTA	ASK 05-41-04-910-0							
	(4)					ution information that will	minimize		
		contamina	tion and accident	al damage	to EVVIS durir	ng maintenance.			
		ASK 05-41-04-410-0							
(5) Close these access panels:									
		<u>Number</u>	Name/Location	<u>on</u>					
		441AL		•		erser Disconnect, Strut 2			
		441AR			-	verser Disconnect, Strut 2	2		
		441AT	Forward Strut	-					
		441BL	Forward Strut	-		_			
		441BR	Forward Strut	•		•			
		441CL 441CR	Forward Strut	-	_	Fairing, Strut 2			
		441DL			-	Fairing, Strut 2 Fairing, Strut 2			
		441DR		-	-	Fairing, Strut 2			
			. J. Maid Olidi		5.16 G/16G/ WIII/9	,			
				2011-2-	<b> </b>				
		EFFECTIVITY		SOURCE		FRUT FAIRING - ENGINE N	11 1 2		

AKS ALL

MRB

D633A109-AKS

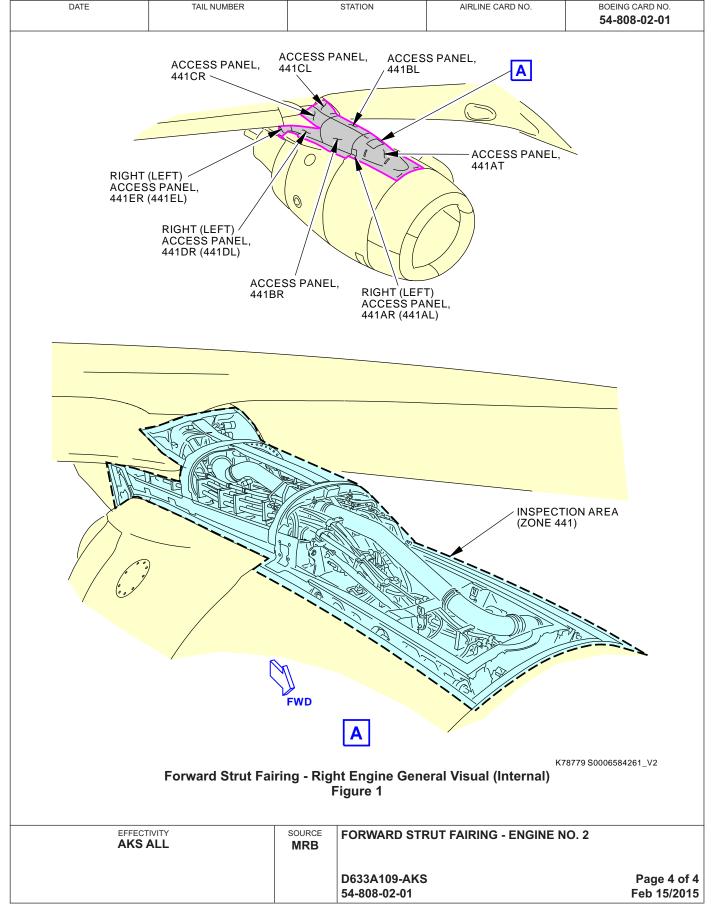
Page 2 of 4
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Feb 15/2015



			D633A109-AKS 54-808-02-01		Pa Fe	age 3 b 15/2	of 4 2015
EFFEC* AKS	ALL	MRB					
EFFEC: AKS	TIVITY <b>ALL</b>	SOURCE MRB	FORWARD STR	UT FAIRING - ENGINE NO.	2		
7712			TASK ———	ming, otracz			
(Con <u>Numb</u> 441EL 441EF	Forward Strut Fa	airing, Let				MECH	INSP
DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CAF <b>54-808-0</b>		









AIRLINE	AIRLINE CARD NO		TITLE SUPPORT BEAM -	BOEING CARD NO. <b>54-810-02-01</b>		
DATE	TASK ZONAL (GV)				RELATE	D CARD
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	ALL	ALL
		ACCESS 441AT			ZONE <b>442</b>	

Perform an internal zonal inspection (GV) of the fan cowl support beam - engine no. 2. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

FAN COWL SUPPORT BEAM - ENGINE NO. 2

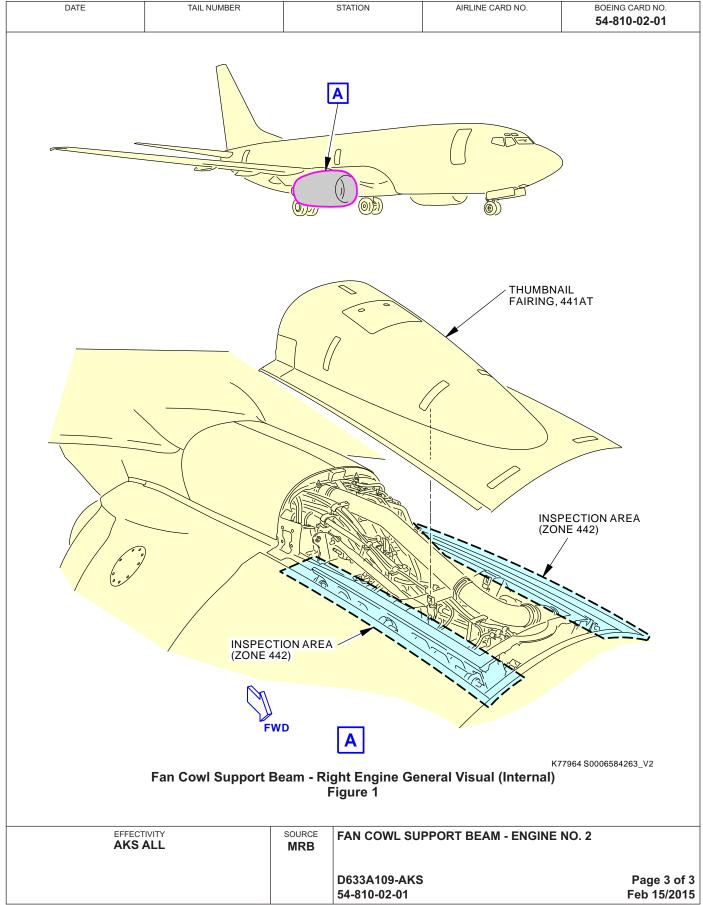
MRB

D633A109-AKS
Feb 15/2015



	DATE			TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. <b>54-810-02-01</b>		
	EWI TAS		41-04-21	10-816					MECH	INSF
1.		ERNA ure 1)		AL (GV): Fan Cowl S	Support E	Beam - Engine	No. 2			
	A.	<b>Gen</b> (1)	This Zo	nal inspection proced ment for this zone.	ure satisf	ies the required	EZAP-derived Zonal in	spection		
	B.		al Inspect SK 05-41-04-0 Open the Numbe 441AT	o10-012 nis access panel:	_	umbnail Fairing	, Strut 2			
	<ul> <li>SUBTASK 05-41-04-210-016</li> <li>(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.</li> <li>(3) Perform an internal zonal inspection (GV) of the fan cowl support beam - engine no</li> </ul>									
	(EZAP)  SUBTASK 05-41-04-910-012  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 05-41-04-410-012  (5) Close this access panel:  Number Name/Location  441AT Forward Strut Fairing, Thumbnail Fairing, Strut 2									
					END OF	TASK ———				
			EFFECTIV AKS AI		SOURCE MRB	FAN COWL SU	PPORT BEAM - ENGINE	NO. 2		
						D633A109-AKS 54-810-02-01	i		Page 2 eb 15/	









AIRLINE	CARD NO	STRUT TORQUE BOX - ENGINE NO. 2			BOEING CARD NO. <b>54-812-02-01</b>		
DATE	TASK ZONAL (GV)				RELATE	D CARD	
TAIL NUMBER	WORK AREA ENG/STRUT	VERSION 1.1	THRESHOLD 18000 FC	REPEAT <b>18000 FC</b>	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	6 YR	6 YR	ALL	ALL	
		ACCESS 443AL 443AR 44	3AT 443BT 443CT	443DT	ZONE <b>443</b>		

Perform an internal zonal inspection (GV) of the strut torque box - engine no. 2. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

#### A. References

Reference	Title
AMM 05-51-22-210-801	Inspection of Titanium Parts When Contaminated With Fire-Resistant Hydraulic Fluid (P/B 201)

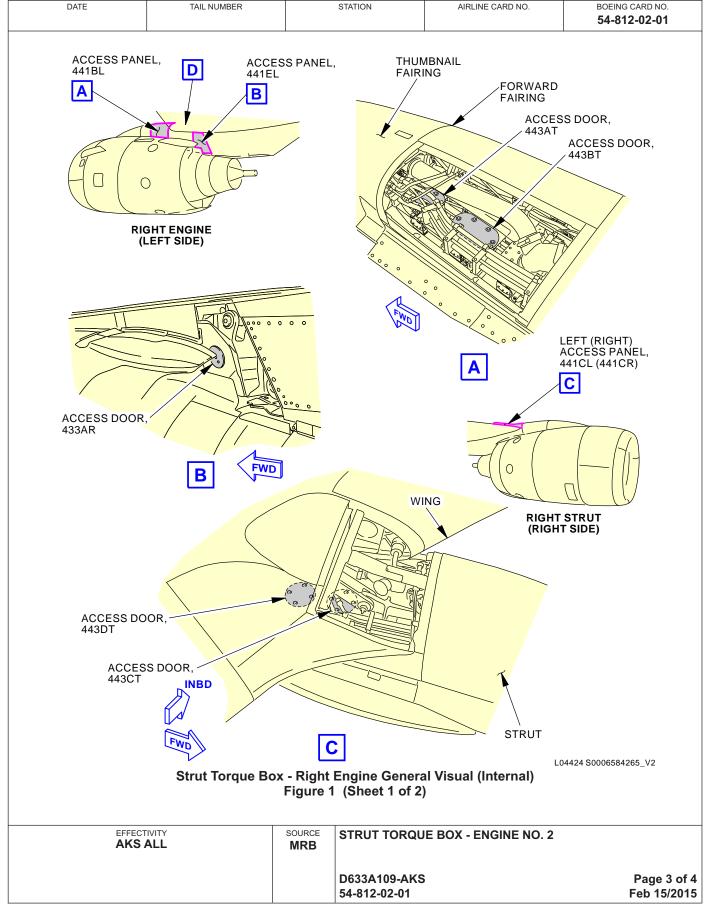
EFFECTIVITY AKS ALL	source MRB	STRUT TORQUE BOX - ENGINE NO. 2	
		D633A109-AKS 54-812-02-01	Page 1 of 4 Jun 15/2016



### 737-600/700/800/900 **TASK CARDS**

					170	N CARDS				
	[	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 54-812		
	EWI	_	-41-04-210	-817					MECH	INS
1.				L (GV): Strut Torq	ue Box - I	Engine No. 2				
	(Fig	ure 1)								
	A.	Gen								
		(1)		al inspection proce ent for this zone.	dure satisf	ies the required	I EZAP-derived Zonal in	spection		
	B.	Zon	al Inspecti	ion						
		SUBTA	ASK 05-41-04-010	-013						
		(1)	Open the	se access panels:						
			<u>Number</u>	Name/Location	<u>on</u>					
			443AL	Strut, Left Aft [						
			443AR	Strut, Right Aff						
			443AT	Strut, Forward						
			443BT 443CT	Strut, Forward						
		443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2								
					pai vveb, c	Strut Z				
			NSK 05-41-04-210		owing the r	rooduroo in Al	MM 05 00 00 010 904			
		(2)		•			MM 05-00-00-910-804.	0 (5745)		
		(3)	Perform a	an internal zonal in	spection (	V) of the strut	torque box - engine no.	2. (EZAP)		
			ASK 05-41-04-211							
		(4)	-				spection of Titanium Pa IM TASK 05-51-22-210-			
		SUBTA	ASK 05-41-04-910							
		(5)		20-60-07-913-801 ation and accident			tion information that will g maintenance.	minimize		
		SUBTA	ASK 05-41-04-410	-013						
		(6)	Close the	se access panels:						
			<u>Number</u>	Name/Location	<u>on</u>					
			443AL	Strut, Left Aft [	Ory Bay, St	rut 2				
			443AR	Strut, Right Aff						
	443AT Strut, Forward Spar Web, Strut 2									
	443BT Strut, Forward Spar Web, Strut 2 443CT Strut, Upper Spar Web, Strut 2 443DT Strut, Upper Spar Web, Strut 2									
			443DT	Strut, Opper S	par web, s	Strut Z				
					- END OF	TASK ———				
			EFFECTIVITY AKS ALL		SOURCE MRB	STRUT TORQU	JE BOX - ENGINE NO. 2			
						D633A109-AKS	3		Page 2 un 15/	
						3. 3.2 32 3.			· · · · ·	







### 737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 54-812-02-01 ACCESS, ACCESS, 443DT 443AT ACCESS, ACCESS. 443CT 443BT **INSPECTION AREA** (ZONE 443) ACCESS, 443AL WING (LEFT SIDE IS SHOWN, RIGHT SIDE IS EQUIVALENT) 1 TORQUE BOX SKIN INSTALLATION NOT SHOWN. L08081 S0006584266\_V2 Strut Torque Box - Right Engine General Visual (Internal) Figure 1 (Sheet 2 of 2) EFFECTIVITY SOURCE STRUT TORQUE BOX - ENGINE NO. 2 **AKS ALL MRB** D633A109-AKS Page 4 of 4 Feb 15/2015 54-812-02-01





TASK DNAL (GV)				RELATE	D CARD
VORK AREA					
IG/STRUT	VERSION 1.1	THRESHOLD 9000 FC	REPEAT <b>9000 FC</b>	APPLIC.	ABILITY ENGINE
SKILL AIRPL	1.2 NOTE	36 MO	36 MO	ALL	ALL
		4BR 444CR		ZONE <b>444</b>	
	SKILL AIRPL	SKILL 1.2 AIRPL NOTE  ACCESS	SKILL 1.2 36 MO AIRPL NOTE	SKILL 1.2 36 MO 36 MO AIRPL NOTE  ACCESS	SKILL 1.2 36 MO 36 MO ALL AIRPLANE ALL ACCESS ZONE

Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP)

**INTERVAL NOTE:** Whichever comes first. The EZAP inspection requirement with interval 12000 FC/4 YR is satisfied by this zonal inspection.

EFFECTIVITY	SOURCE	AFT STRUT FAIRING - ENGINE NO. 2	
AKS ALL	MRB		
		D633A109-AKS 54-814-02-01	Page 1 of 5 Jun 15/2015



#### 737-600/700/800/900 TASK CARDS

TASK 05-41-04-210-818 INTERNAL - ZONAL (GV): Aft Strut Fairing - Engine No. 2 (Figure 1)  A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.  B. Zonal Inspection SUBTRANC 64-10-419044 (1) Open these access panels: Number Name/Location 444AL Aft Strut Fairing, Left Forward Panel, Strut 2 444AR Aft Strut Fairing, Right Aft Panel, Strut 2 444AR Aft Strut Fairing, Right Aft Panel, Strut 2 444CR Aft Strut Fairing, Right Aft Panel, Strut 2 444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2 SUBTANC 65-41-04-219-018 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP) SUBTANC 65-41-04-219-011 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTANC 65-41-04-219-014 (5) Close these access panels: Number Name/Location 444AL Aft Strut Fairing, Right Forward Panel, Strut 2 444AR Aft Strut Fairing, Right Aft Panel, Strut 2 444AR Aft Strut Fairing, Right Aft Panel, Strut 2 444AR Aft Strut Fairing, Right Access To Fuel Door, Strut 2 END OF TASK  EFFECTIVITY AKS ALL  BY  BY  BY  BY  BY  BY  BY  BY  BY		BOEING C/ <b>54-814-</b>	AIRLINE CARD NO.	STATION		TAIL NUMBER		ATE	D	
(Figure 1)  A. General  (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.  B. Zonal Inspection  SUBTASK 08-41-04-01-0414  (1) Open these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444BR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  SUBTASK 08-41-04-21-018  (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.  (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP)  SUBTASK 08-41-04-01-041  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 08-41-04-14-01-14  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444BR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  ———————————————————————————————————	MECH I			ngine No. 2	iring - E			K 05-4	TAS	
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.  B. Zonal Inspection  SUBTAS 05-14-04-01-04  (1) Open these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  SUBTASK 05-14-04-210-018  (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.  (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP) SUBTASK 05-14-04-010-011  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 05-14-04-10-014  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  ——END OF TASK  EFFECTIVITY  SOURCE AFT STRUT FAIRING - ENGINE NO. 2						,	,			
B. Zonal Inspection SUBTASK 05-41-04-010-014  (1) Open these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  SUBTASK 05-41-04-210-018  (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.  (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP)  SUBTASK 05-41-04-010-014  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 05-41-04-410-014  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  END OF TASK  END OF TASK		spection	ired EZAP-derived Zonal ins	fies the required	ure satisf	•	This Zonal i	(1)	A.	
(1) Open these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Aft Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  SUBTASK 05-41-04-219-019  (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.  (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP) SUBTASK 05-41-04-919-011  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 05-41-04-419-014  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  ——END OF TASK  AFT STRUT FAIRING - ENGINE NO. 2							•		B.	
Number  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Aft Panel, Strut 2  308TASK 05-41-04-210-018  (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP)  308TASK 05-41-04-910-011  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  308TASK 05-41-04-410-014  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  ——END OF TASK  EFFECTIVITY  SOURCE AFT STRUT FAIRING - ENGINE NO. 2							-			
Aft Strut Fairing, Left Forward Panel, Strut 2 444AR Aft Strut Fairing, Right Forward Panel, Strut 2 444AR Aft Strut Fairing, Right Aft Panel, Strut 2 444CR Aft Strut Fairing, Right Aft Panel, Strut 2 444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  subtask 05-41-04-210-018  (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP)  subtask 05-41-04-910-011  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  subtask 05-41-04-910-914  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2 444AR Aft Strut Fairing, Right Forward Panel, Strut 2 444BR Aft Strut Fairing, Right Aft Panel, Strut 2 444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  END OF TASK  EFIECTIVITY  SOURCE AFT STRUT FAIRING - ENGINE NO. 2						access panels:	Open these	(1)		
(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.  (3) Perform an internal zonal inspection (GV) of the aft strut fairing - engine no. 2. (EZAP) SUBTASK 05-41-04-910-011  (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 05-41-04-410-014  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  END OF TASK  END OF TASK  SOURCE AFT STRUT FAIRING - ENGINE NO. 2			el, Strut 2 ut 2	orward Panel, S ft Panel, Strut 2	- , Left For , Right Fo , Right Af	Aft Strut Fairing, Aft Strut Fairing, Aft Strut Fairing,	444AL 444AR 444BR			
(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.  SUBTASK 05-41-04-410-014  (5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  END OF TASK  END OF TASK  AFT STRUT FAIRING - ENGINE NO. 2		(EZAP)				al inspection follow	Do the zona	(2)		
(5) Close these access panels:  Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  ———————————————————————————————————		ninimize				-60-07-913-801 fo	Refer to 20-	(4)		
Number Name/Location  444AL Aft Strut Fairing, Left Forward Panel, Strut 2  444AR Aft Strut Fairing, Right Forward Panel, Strut 2  444BR Aft Strut Fairing, Right Aft Panel, Strut 2  444CR Aft Strut Fairing, Right Access To Fuel Door, Strut 2  ———————————————————————————————————										
EFFECTIVITY SOURCE AFT STRUT FAIRING - ENGINE NO. 2			el, Strut 2 ut 2	orward Panel, S ft Panel, Strut 2	- , Left For , Right Fo , Right Af	Name/Location Aft Strut Fairing, Aft Strut Fairing, Aft Strut Fairing,	Number 444AL 444AR 444BR	, ,		
ALL OTHOLI AIRING - ENGINE NO. 2			<u> </u>		•	_				
ALL OTHOLI AINTO - ENGINE NO. 2										
ALL OTHOLI AINTO - ENGINE NO. 2										
ALL OTHOLI AINTO - ENGINE NO. 2										
			FAIRING - ENGINE NO. 2	AFT STRUT FA						

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Feb 15/2015



