



737 NON-DESTRUCTIVE TEST MANUAL

PART 6 - EDDY CURRENT

MAIN LANDING GEAR UPPER AND LOWER DRAG STRUT FUSE BOLT

1. Purpose

- A. To find the longitudinal cracks in the main landing gear top and bottom drag strut attachment bolts (fuse bolts) without the removal of the bolts. See Details I and II in Figure 1. The inspection is started at the bore of the bolt.
- B. Service Bulletin Reference: 737-32-1123, Rev 2

2. Equipment

- A. Instrument - Eddy current instruments that meet the necessary conditions of this procedure are permitted. The following instrument was used to make this procedure:
 - (1) MIZ-10A - Zetec, Inc.
- B. Probes - Unshielded bolthole probes that can operate at 25 kHz, go down to a depth of 4.5 inches (12 cm) and meet the necessary conditions of this procedure can be used. A reflection-type probe is recommended, See Table 1. The following probe was used to make this procedure:
 - (1) VM101BR - 1/2 X 6, 10 - 30 kHz -- V.M. Products
- C. Reference Standard - Make Reference Standard 380-X as shown in Figure 2. The reference standard must match the bore diameter of the part to be inspected. See Table 1.

NOTE: Refer to Part 1, 51-01-00 for the address information of the vendors who supply the equipment.

3. Prepare for the Inspection

- A. Remove the lubrication plug if there is one.
- B. Remove the cotter pin from the bolt.
- C. Remove all unwanted materials such as dirt and grease from the bolt and visually inspect the bore. Note all scratches, corrosion and/or the areas that look as if there is a crack in the bore.

NOTE: Scratches in the cadmium layer in the bore will cause indications that look like cracks when you do an inspection with the eddy current equipment.

4. Instrument Calibration

- A. Set the frequency of the instrument to 25 kHz.
- B. Put the probe in the reference standard in an area that does not have a notch. The probe must tightly fit but not where it causes excessive wear on the probe coil.
- C. Balance the instrument as specified in the manufacturer's instructions. Set the baseline at 20 percent of full scale.
- D. Adjust the lift-off so that the probe-to-part spacing of up to 0.006 inch (approximately two sheets of paper) gives no more than a 5 percent full scale deflection.
- E. Adjust the depth of the probe in the hole to get the maximum signal from the reference notch.
- F. Adjust the sensitivity of the instrument to give a 50 percent meter deflection when a scan is done slowly over the reference notch with the probe.

NOTE: Notice the usual sharp deflection of the meter caused by the notch.



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- G. Position the probe again in an area that is not notched. Check the balance and the lift-off. If adjustments are needed, check the sensitivity again.

5. Inspection Procedure

- A. Adjust the collar on the probe to set the depth of the penetration into the bore at 0.100 inch.
- B. Insert the probe into the pin and balance the instrument.
- C. Check the setting of the lift-off again as referred to in Paragraph 4.D.
- D. Slowly do a scan of the entire circumference of the hole and note all locations that give a sudden deflection up.
- E. Repeat Paragraph 5.B. and Paragraph 5.C. at increments of 0.100 inch depths until the entire bore has been inspected.

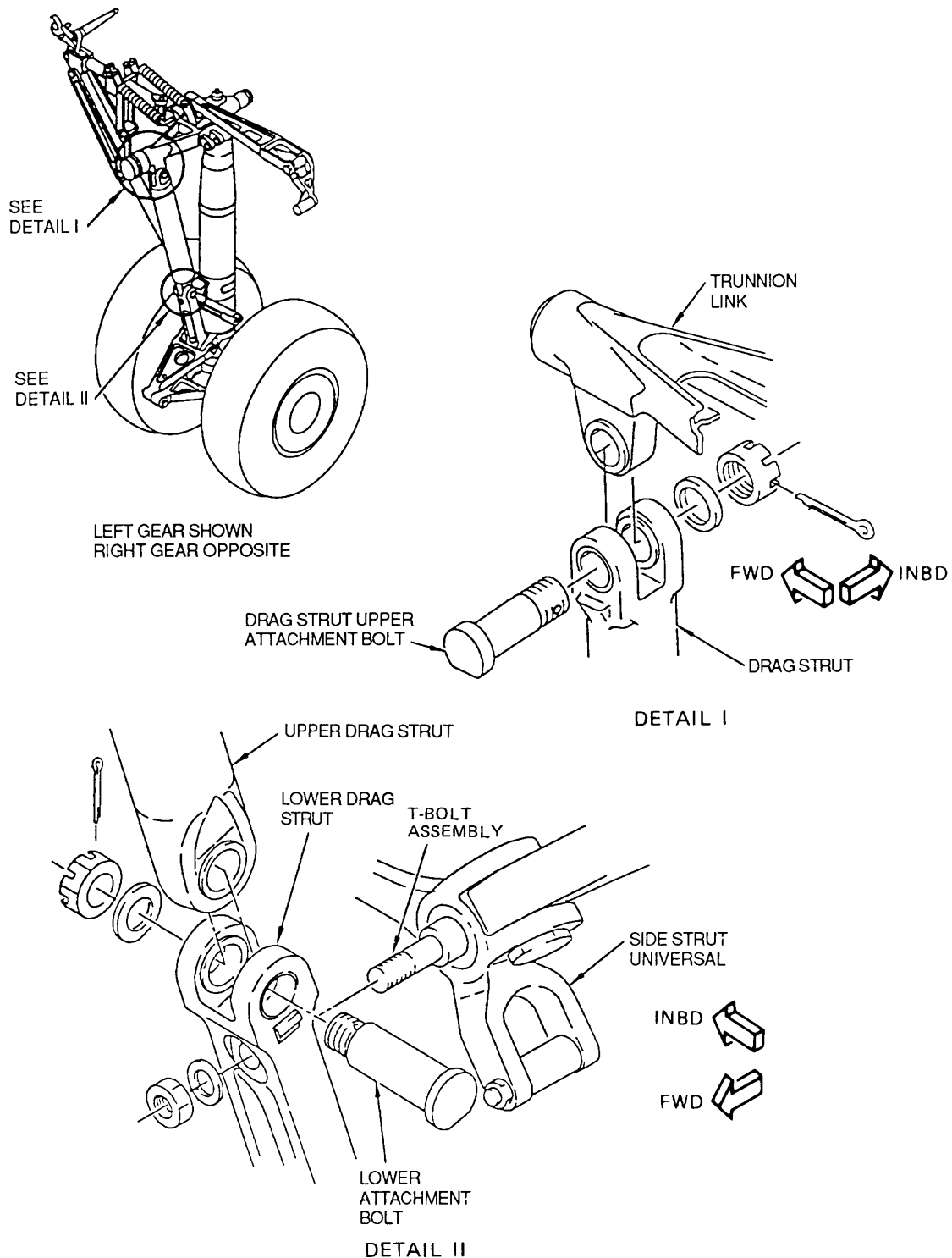
6. Inspection Results

- A. Sharp meter indications equal to or greater than the calibration response must be confirmed. To confirm the results, remove the bolt and do a magnetic particle inspection.

Table 1: Fuse Bolt Dimensions

FUSE BOLT PART NO. AND LOCATION	BORE DIAMETER
69-39473-1 (lower)	0.75 inch (PA002 and PA003 only)
69-39473-2 (lower)	0.50 inch
69-39476-5, -8 (upper)	0.87 inch
69-39476-6, -7 (upper)	0.80 inch

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Main Landing Gear Drag-Strut Attachment Bolts
Figure 1

EFFECTIVITY
ALL; 737-100, -200 AIRPLANE LINE NUMBERS 1
THRU 964

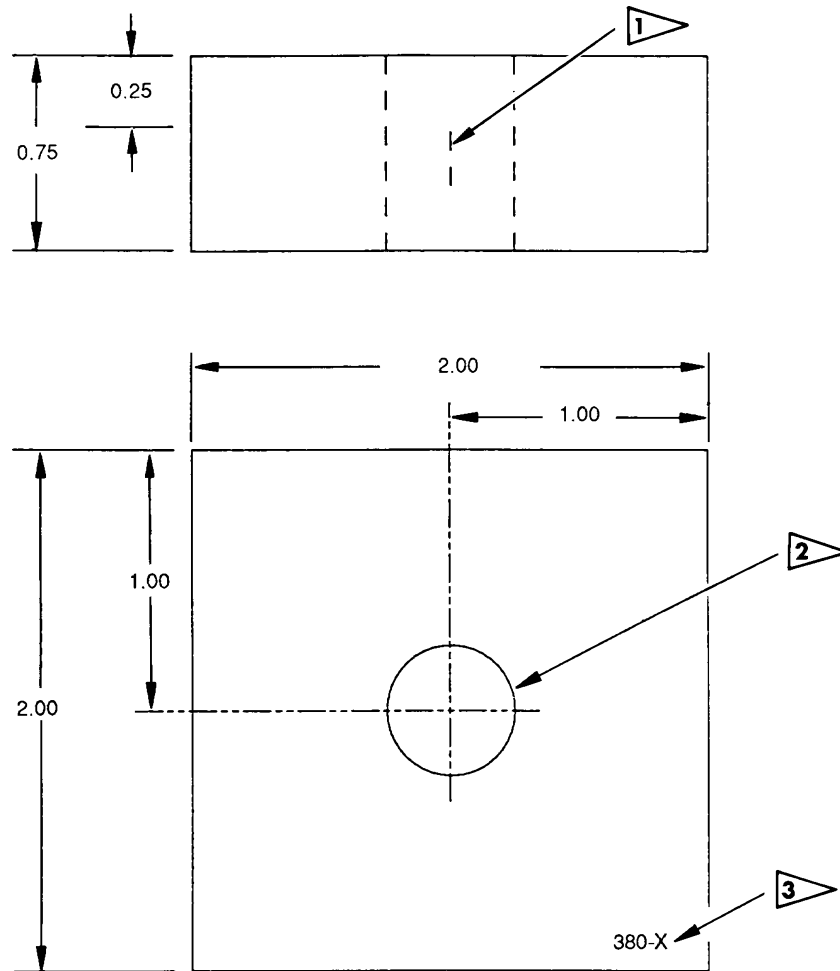
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- NOTES:
- ALL DIMENSIONS IN INCHES
 - MATERIAL – LOW ALLOY STEEL (4330M OR EQUIVALENT)
 - ALL TOLERANCES ± 0.050 EXCEPT WHERE NOTED
 - 1 EDM NOTCH, DEPTH – 0.030", WIDTH – 0.007", LENGTH – 0.250", ALL ± 0.003 "
 - 2 DIAMETER OF HOLE PER TABLE I
 - 3 ETCH OR STEEL STAMP WITH 380-X (DASH NUMBER CORRESPONDS WITH DASH NUMBER OF PART INSPECTED)

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Reference Standard
Figure 2

EFFECTIVITY
ALL; 737-100, -200 AIRPLANE LINE NUMBERS 1
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