

**F119**  
MILITARY ENGINES

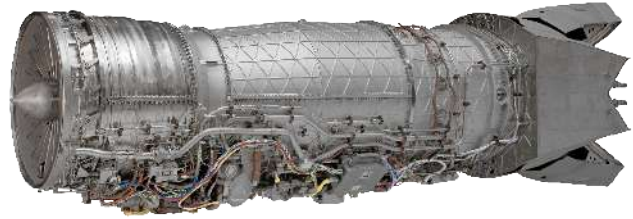
**F119-PW-100**

**PROVEN POWER FOR  
THE F-22 RAPTOR**

[PRATTWHITNEY.COM](https://prattwhitney.com)



# F119



## Proven power for the F-22 Raptor

Pratt & Whitney's F119 turbofan engine is the world's first fifth-generation fighter engine. The F119 combines stealth technologies and vectored thrust performance to provide unprecedented maneuverability and survivability with a high thrust-to-weight ratio. The ability to operate supersonically without afterburner— supercruise—gives the F-22 exceptional combat performance without compromising mission range.



### ADVANCED TECHNOLOGY

The F119 is equipped with a number of advanced technologies for unmatched operational performance and reliability. Its three-stage integrally bladed fan is powered by a single-stage low-pressure turbine. The robust, yet compact, high-pressure compressor features advanced airfoil aerodynamics and integrally bladed rotor disks for ensured durability. The engine's counter-rotating core has an aerodynamically efficient six-stage compressor driven by a single-stage high-pressure turbine featuring single-crystal superalloy blades and advanced cooling technologies. The engine delivers unparalleled aircraft maneuverability with its unique two dimensional pitch-vectoring exhaust nozzle.



### MAINTAINABILITY

Ease of assembly, maintenance and repair were designed into the F119 from its inception using a balanced team approach that included assemblers and flight line mechanics. Requirements for support equipment and labor were significantly reduced, minimizing the overall F119 logistics footprint.



### SAFETY

The F119 engine has achieved a best-in-class safety record since its introduction by outperforming legacy engine benchmarks.

### ENGINE SPECIFICATIONS

|                    |   |
|--------------------|---|
| Type               | Twin-spool, augmented turbofan  |
| Thrust             | 35,000 pounds   |
| Engine control     | FADEC (Full-Authority Digital Engine Control)   |
| Compression System | Dual-rotor, counter-rotating, axial flow, low aspect ratio<br>- Three-stage fan<br>- Six-stage high-pressure compressor |
| Combustor          | Annular, Floatwall™ configuration   |
| Turbines           | Axial flow, counter-rotating<br>- One-stage high-pressure turbine<br>- One-stage low-pressure turbine                   |
| Nozzle             | Nozzle Two-dimensional pitch-vectoring convergent/divergent   |